

# Smart Control Algorithm for 2-DOF Helicopter

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# Outline

- 1 Introduction
- 2 Background Study
  - Control Techniques
  - Modeling a 2-DOF Helicopter
  - Prior Work
- 3 Subsystem Level Functional Requirements
  - Block Diagram
- 4 Preliminary Work
  - LQR Simulation
  - LQR via USB
  - LQR via Wireless
  - Demonstration
- 5 Parts List
- 6 Future Directions

- Helicopter are important for short-distance travel
  - air-sea rescue
  - fire fighting
  - traffic control
  - tourism
- Purpose of control system
  - resistance to turbulence
  - enable use of mobile device
- Which is better?
  - Fundamental (LQR)
  - Noise Filtering (LQG)
  - Machine Learning (ADP)

# Introduction

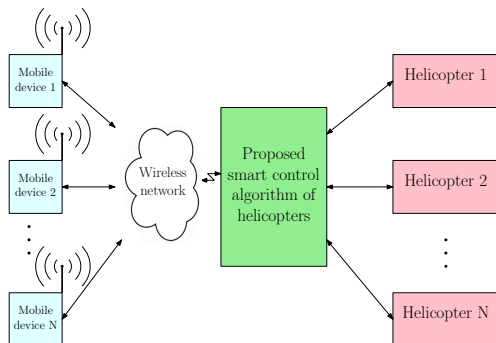


Figure 1: General High-Level System Architecture

# Introduction

- This project will:
  - use a pair of 2-DOF (2-degrees-of-freedom) testing platforms
  - implement control algorithms on embedded system
  - use mobile device for user control
  - encourage research
  - serve as an educational tool

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# Background Study

## Control Techniques

Various control techniques have been proposed for 2-DOF helicopters such as:

- Sliding mode control [?]
- Fuzzy Logic control [?] [?] [?]
- Data-driven Adaptive Optimal Output-feedback control [?]
- Decentralized discrete-time neural control [?]

These control techniques employ advanced mathematics that are difficult to implement on embedded systems.

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# Background Study

## Modeling a 2-DOF Helicopter

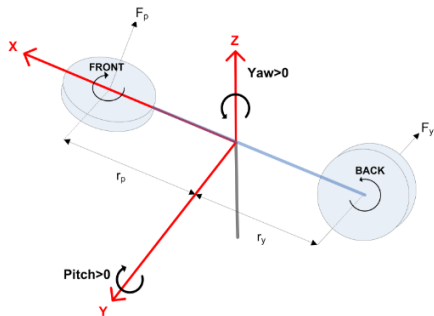


Figure 2: Model of a 2-DOF Helicopter

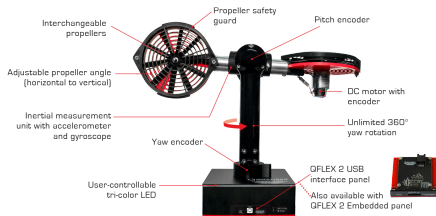


Figure 3: Quanser Aero

# Background Study

## Modeling a 2-DOF Helicopter

- Characterized by fixed base
  - Can change 2 of 3 possible orientations...
    - Pitch ( $\theta$ )
    - Yaw ( $\psi$ )
    - *Not Roll*
  - and cannot change position
    - x direction
    - y direction
    - z direction

# Background Study

## Modeling a 2-DOF Helicopter

- Motors are attached to the propellers to create thrust due to air resistance
  - Main - changes pitch angle
  - Tail - changes yaw angle
- Torque due to rotation also creates a force on opposite axes

# Background Study

## Modeling a 2-DOF Helicopter

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t), \text{ where} \quad (1)$$

$$\mathbf{A} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -\frac{K_{sp}}{J_p} & 0 & -\frac{D_p}{J_p} & 0 \\ 0 & 0 & 0 & -\frac{D_y}{J_y} \end{bmatrix} \text{ and } \mathbf{B} = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ \frac{K_{pp}}{J_p} & \frac{K_{py}}{J_p} \\ \frac{K_{yp}}{J_y} & \frac{K_{yy}}{J_y} \end{bmatrix},$$

# Background Study

## Modeling a 2-DOF Helicopter

- $K_{sp}$  - being the stiffness of the axes
- $K_{pp}$  - pitch motor thrust constant
- $K_{py}$  - thrust constant acting on the pitch angle from the yaw motor
- $K_{yp}$  - thrust constant acting on the yaw angle from the pitch motor
- $K_{yy}$  - yaw motor thrust constant
- $J_p$  - moment of inertia about pitch axis
- $J_y$  - moment of inertia about yaw axis
- $D_p$  - viscous damping of the pitch axis
- $D_y$  - viscous damping of the yaw axis

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# Background Study

## Prior Work

- extensive modeling & simulations
- implementation of two motion control algorithms (LQR & ADP)
- one helicopter

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# Subsystem Level Functional Requirements

## Block Diagram

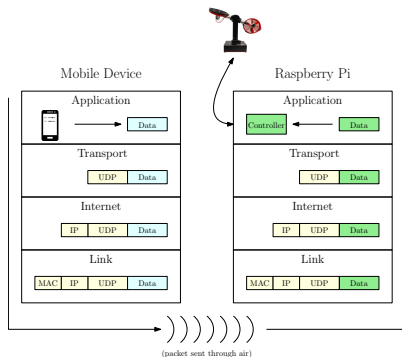


Figure 4: Communication Model

# Subsystem Level Functional Requirements

## Block Diagram

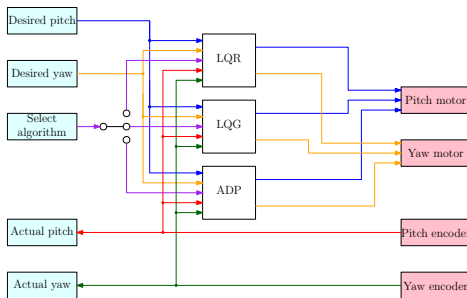


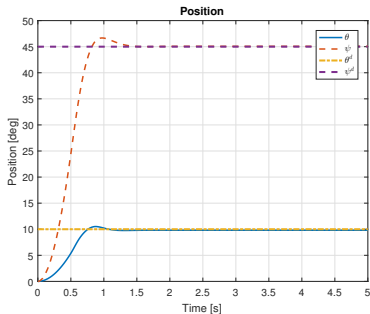
Figure 5: Low Level Smart Control Diagram

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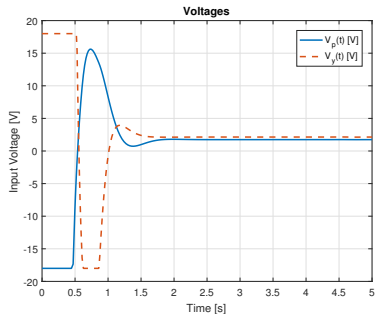
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# Preliminary Work

## LQR Simulation



(a)



(b)

Figure 6: LQR Simulation (a) Position and (b) Voltage w/ Constant Signal

# Preliminary Work

## LQR Simulation

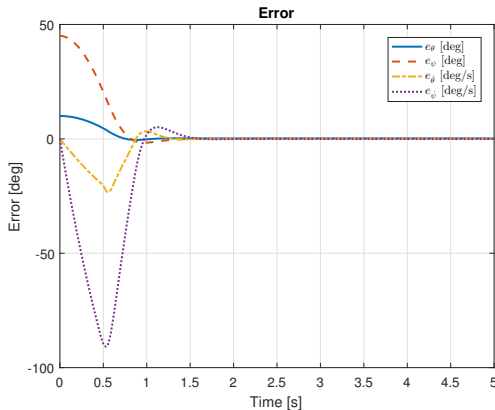


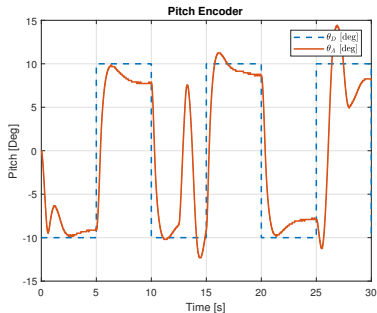
Figure 7: LQR Simulation Error w/ Constant Signal

# Outline

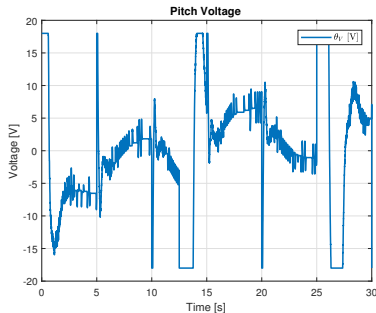
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# Preliminary Work

LQR via USB



(a)

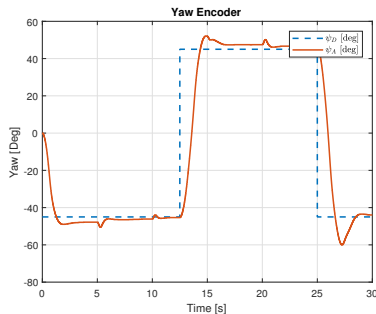


(b)

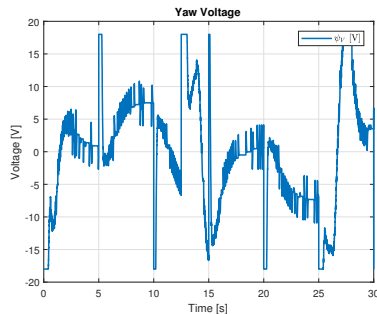
Figure 8: LQR Pitch Position (a) and Voltage (b) on PC with Square Wave Input

# Preliminary Work

LQR via USB



(a)



(b)

Figure 9: LQR Yaw Position (a) and Voltage (b) on PC with Square Wave Input

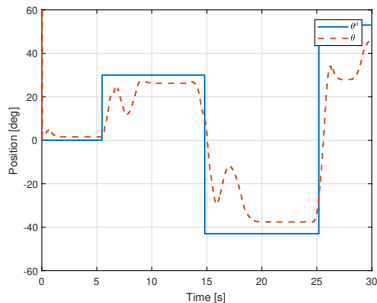


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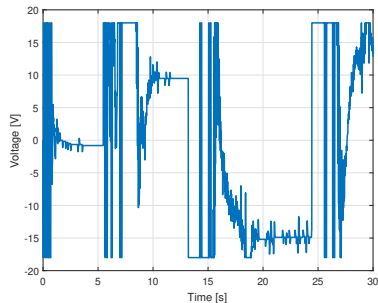
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# Preliminary Work

## LQR via Wireless



(a)

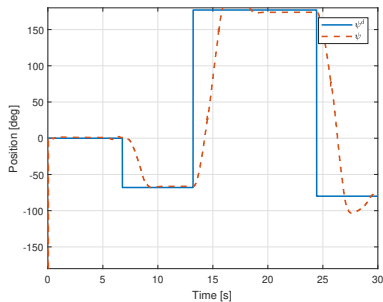


(b)

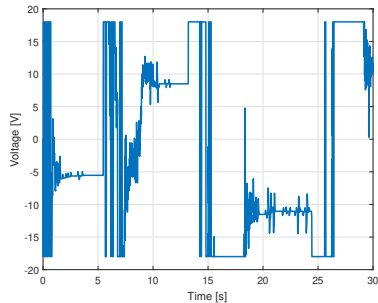
**Figure 10:** Performance in following user's command (a) tracking pitch angle, and (b) pitch motor input voltage

# Preliminary Work

## LQR via Wireless



(a)



(b)

**Figure 11:** Performance in following user's command (a) tracking yaw angle, and (b) yaw motor input voltage.

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# Demonstration

# Parts List

- Hardware

- Two Quanser Aeros
  - Q-flex2 Embedded Panel
- Two Single Board Computers (Raspberry Pi 3 Model B)
- Android Smart-phone or Tablet  
(Note that Apple devices could also be used, however modifications are needed)

- Software

- MATLAB & Simulink
  - Raspberry Pi Support Package
  - Android Support Package
- Quanser Real-Time Control (QUARC)

# Deliverables

## Schedule for Completion

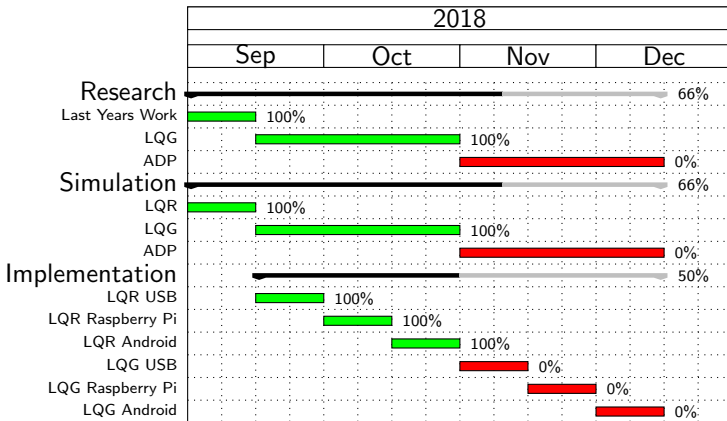


Figure 12: Gantt chart for Fall 2018

# Deliverables

## Schedule for Completion

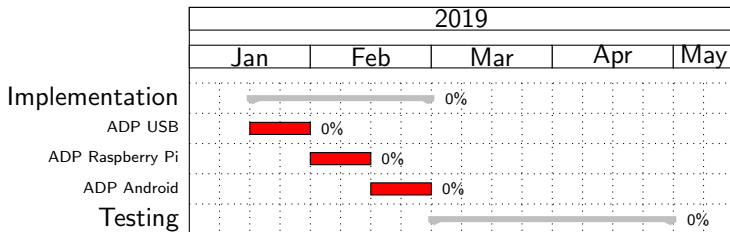


Figure 13: Gantt Chart for Spring 2019



# Future Directions

- Two more motion control algorithms (LQG & ADP)
- Test plan
- Implementation on 6-DOF Helicopter

# Summary

- Embedded implementation of control algorithms
- Mobile interface

# For Further Reading I