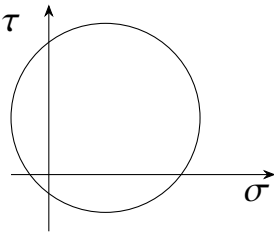


GATE Questions 19

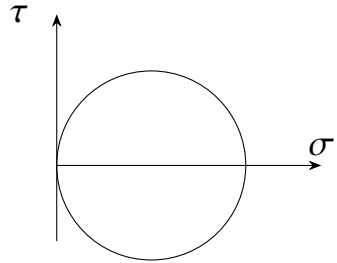
1

EE24BTECH11012 - Bhavanisankar G S

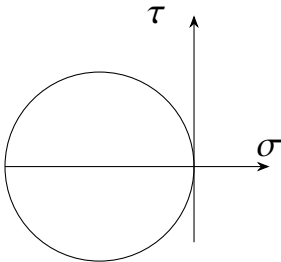
- 1) For International Standard Atmosphere (ISA) upto 11 km, which of the following statement(s) is(are) true ?
- a) The hydrostatic / aerostatic equation is used.
 - b) The temperature lapse rate is taken as $-10^{-2}K/m$
 - c) The sea level conditions are taken as $P_s = 1.01325 \times 10^5 Pa$, $T_s = 300K$, $\rho_s = 1.225kg/m^3$
 - d) Air is treated as a perfect gas
- 2) Let σ and ρ represent the normal stress and shear stress on a plane, respectively. The Mohr circle(s) that may possibly represent the state of stress at points in a beam of rectangular cross-section under **pure bending** is/are :



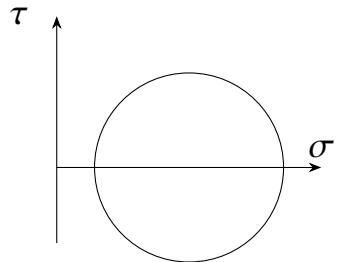
a)



c)



b)



d)

- 3) An isotropic linear elastic material point under plane strain condition in the x-y plane always obeys :
- a) out-of-plane normal strain $\epsilon_{zz} = 0$
 - b) out-of-plane normal stress $\sigma_{zz} = 0$

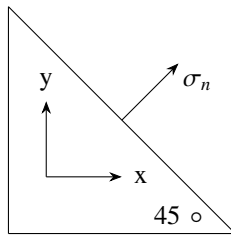
- c) out-of-plane shear stress $\tau_{xz} = 0$
 d) out-of-plane shear strain $\gamma_{xz} = 0$
- 4) A high-pressure-ratio multistage axial compressor encounters an extreme loading mismatch during starting. Which of the following technique(s) can be used to alleviate this problem ?

- a) Blade cooling
 b) Variable angle stator vanes
 c) Blow-off valves
 d) Multi-spool shaft

- 5) The arc length of the parametric curve: $x = \cos \theta, y = \sin \theta, z = \theta$ from $\theta = 0$ to $\theta = 2\pi$ is equal to
- 6) An unpowered glider is flying at a glide angle of 10° . Its lift-to-drag ratio is _
- 7) The two-dimensional plane-stress state at a point is

$$\sigma_{xx} = 110 \text{ MPa}, \sigma_{yy} = 30 \text{ MPa}, \tau_{xy} = 40 \text{ MPa}$$

The normal stress, σ_n on a plane inclined at 45° as shown in the figure is _ MPa .



- 8) In a **static** test, a turbofan engine with **by-pass ratio** of 9 has core hot exhaust speed 1.5 times that of fan exhaust speed. The engine is operated at a fuel to air ratio of $f = 0.03$. Both the fan and the core streams have no pressure thrust. The ratio of fan thrust to thrust from the core engine is _
- 9) In a single stage turbine, the hot gases come out of stator/nozzle at a speed of 500 m/s and at angle of 70° with the turbine axis as shown. The design speed of the rotor blade is 250 m/s at the mean blade radius. The rotor blade angle, β , at the leading edge is _ degrees.

