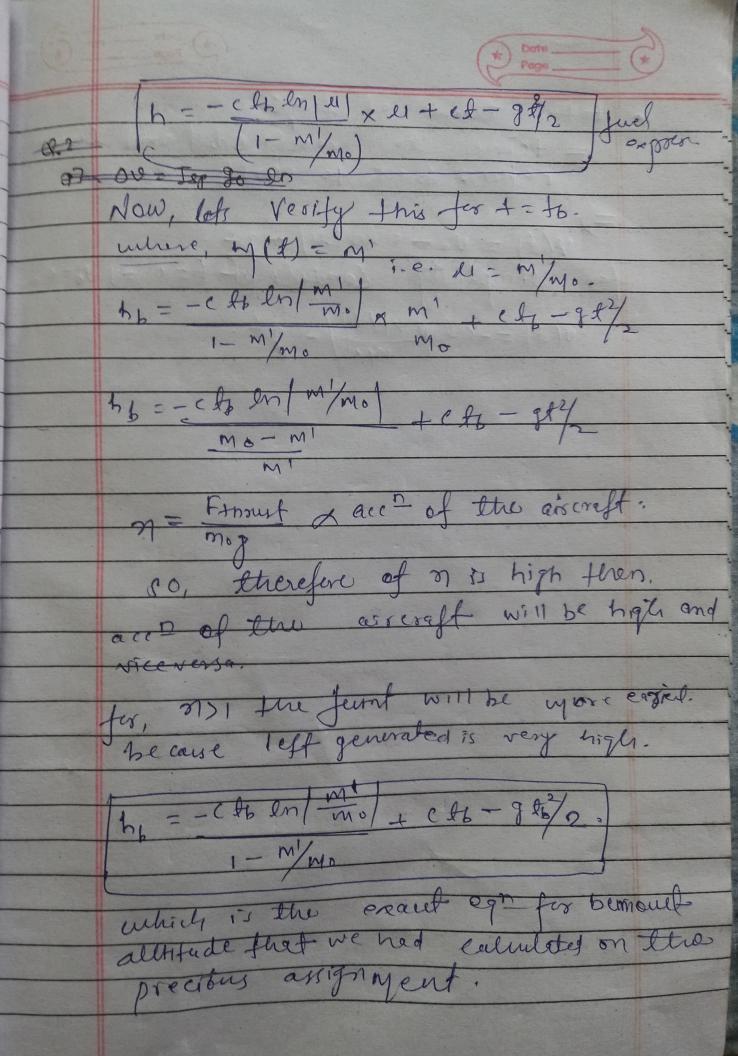


h = - (fer) m(f) + cf - gf2 + cmdb en m(f)moto en/mol, h = - (fln/m(t)) + (f-gty + (moth ln ) m(t) c moto en/mol mo-m1. h = -cf ln/mt/+ cf gfy + (moto ln/m(d)) h = - (fln | u Mo) + cf - gt + (Mo + b ln | u) + (flower) h = - ctln/e/+ ct-gt2 + (moth ln/4) h = - cto enjer[mit] + ct - 9th h = - ( to lenje) [m(t)] + ct - gt/2 n = -ch ln/4/ x umo + ct - gf2
(mo-M!)



- total time of flight.

= teme to really orbit and limet referra back to eastly. =) v(H = vo - cln |u| - 9f.

for, simplicity Conclude af t= 0 = 1 vo=0. v(+) = - (2) M(+) - gt.