

Assignment - 2

1)

$$\Delta v = I_{sp} g_0 \ln \left[\frac{M_0}{M_{pay} + M_{rocket}} \right]$$

$$\Delta v = 7.6 \text{ km/s} = 7600 \text{ m/s} \quad I_{sp} = 400 \text{ s} \quad g_0 = 9.8 \text{ m/s}^2$$

$$\Rightarrow \ln \left[1 + \frac{M_{prop}}{M_{pay} + M_{rocket}} \right] = \frac{7600}{400 \times 9.8} = 1.9388$$

$$\Rightarrow 1 + \frac{M_{prop}}{M_{pay} + M_{rocket}} = e^{1.9388} = 6.95$$

$$\Rightarrow M_{prop} = (5.95)(M_{pay} + M_{rocket})$$

$$\begin{aligned} \text{Now } \frac{M_{prop}}{M_i} &= \frac{5.95(M_{pay} + M_{rocket})}{M_{pay} + M_{rocket} + M_{prop}} \\ &= \frac{5.95(M_{pay} + M_{rocket})}{6.95(M_{pay} + M_{rocket})} = 0.856 \end{aligned}$$

2) $\ddot{y} f_{thrust} = -c \frac{dm}{dt}$

$$f_{net} = -c \frac{dm}{dt} - mg = m \frac{dv}{dt}$$

$$\begin{aligned} \Rightarrow -c dm - mg dt &= m dv \\ \Rightarrow -\int_{m_0}^{m_b} \frac{c dm}{m} - \int g dt &= \int_{v_0}^{v(t)} dv \quad \Rightarrow -c \ln \left| \frac{m_t}{m_0} \right| - g t = -v(t) - v_0 \end{aligned}$$

$$\Rightarrow v(t) = v_0 + c \ln \left| \frac{m_0}{m_b} \right| - g t$$

$$\Rightarrow \boxed{v(t) = v_0 + c \ln \left| \frac{m_0}{m} \right| - g t}$$

iv) $n = \frac{F_{thrust}}{m_0 g}$ \times acceleration of rocket.

So, for $n > 1$ the flight will be more easier because lift generated will be very high.

ii) If the burning rate = constant.

$$\Rightarrow -\frac{dm}{dt} = k \quad \Rightarrow \int_0^{t_b} k dt = \int_{m_0}^{m'} dm \quad \Rightarrow k = \frac{m_0 - m'}{t_b}$$

t_b = time to reach orbit

For $t < t_b$,

$$\int_0^t k dt = - \int_{m_0}^{m(t)} dm \quad \Rightarrow kt = -m(t) + m_0$$

$$\Rightarrow (m_0 - m') \frac{t}{t_b} - m_0 = -m(t)$$

$$\Rightarrow m(t) = m_0 - (m_0 - m') \frac{t}{t_b}$$

Now,

$$h = \int_0^t u dt \quad \Rightarrow h = \int_0^t \left(-c \ln \frac{m(t)}{m_0} - g t \right) dt$$

$$\Rightarrow h = -c \int_0^t \ln \left(\frac{m_0 - (m_0 - m') \frac{t}{t_b}}{m_0} \right) dt - g \int_0^t t dt$$

$$\Rightarrow h = \left[-ct \ln \left(m_0 - (m_0 - m') \frac{t}{t_b} \right) + ct \right. \\ \left. + \frac{cm_0 t_b}{m_0 - m'} \ln \left[m_0 - (m_0 - m') \frac{t}{t_b} \right] - \frac{gt^2}{2} \right]_0^t \\ + c t \ln m_0$$

$$\Rightarrow h = -ct \ln \left(m_0 - (m_0 - m') \frac{t}{t_b} \right) + ct - \frac{gt^2}{2} + c t \ln m_0 \\ + \frac{cm_0 t_b}{m_0 - m'} \ln \left[m_0 - (m_0 - m') \frac{t}{t_b} \right] \\ - \frac{cm_0 t_b}{m_0 - m'} \ln (m_0) - c t \ln m_0$$

$$\Rightarrow h = -c\ell \ln |m(t)| + c\ell - \frac{gt^2}{2} + \frac{cm_0 b}{m_0 - m'} \ln |m(t)| - \frac{cm_0 b}{m_0 m'} \ln(m_0)$$

$$\Rightarrow h = -c\ell \ln |m(t)| + c\ell - \frac{gt^2}{2} + \frac{cm_0 b}{m_0 - m'} \ln \left| \frac{m(t)}{m_0} \right|$$

$$\Rightarrow h = -c\ell \ln |u m_0| + c\ell - \frac{gt^2}{2} + \frac{cm_0 b}{m_0 - m'} \ln |u| + c\ell \ln m_0$$

$$\Rightarrow h = -c\ell \ln |u| + c\ell - \frac{gt^2}{2} + \frac{cm_0 b}{m_0 - m'} \ln |u|$$

$$\Rightarrow h = -c\ell \ln |u| \left[t - \frac{m_0 b}{m_0 - m'} \right] + c\ell - \frac{gt^2}{2}$$

$$\Rightarrow h = \frac{-c\ell b \ln |u|}{m_0 - m'} \left[(m_0 - m') \frac{t}{b} - m_0 \right] + c\ell - \frac{gt^2}{2}$$

$$\Rightarrow h = \frac{-c\ell b}{m_0 - m'} \ln |u| [m(t)] + c\ell - \frac{gt^2}{2}$$

$$\Rightarrow h = \frac{-c\ell b \ln |u|}{(m_0 - m')} \times m_0 + c\ell - \frac{gt^2}{2}$$

$$\Rightarrow \boxed{h = \frac{-c\ell b \ln |u|}{\left(1 - \frac{m'}{m_0}\right)} \times m_0 + c\ell - \frac{gt^2}{2}}$$

for $t = t_0$ then $m(t) = m'$ i.e. $u = \frac{m'}{m_0}$

$$h_b = \frac{-c\ell b \ln \left| \frac{m'}{m_0} \right|}{1 - \frac{m'}{m_0}} \times \frac{m'}{m_0} + c\ell b - \frac{gt^2}{2}$$

$$\Rightarrow h_b = \frac{-c\ell b \ln \left| \frac{m'}{m_0} \right|}{\frac{m_0 - m'}{m'}} + c\ell b - \frac{gt^2}{2}$$

$$\dot{h}_t = \frac{-c_b \ln \left| \frac{m'}{m_0} \right|}{1 - \frac{m'}{m_0}} + c_b - \frac{g t_p^2}{2} \quad \text{--- burnout altitude}$$

$$4) \Delta v = I_{sp} g_0 \ln \left(\frac{M_i}{M_f} \right)$$

$$M_{pay} = 2000 \text{ kg}$$

$$\Rightarrow e^{\frac{\Delta v}{I_{sp} g_0}} = \frac{M_i}{M_f}$$

$$\Rightarrow e^{\frac{\Delta v}{I_{sp} g_0}} = \frac{M_{prop} + M_{pay} + M_{rocket}}{M_{pay} + M_{rocket}}$$

$$\Rightarrow e^{\frac{\Delta v}{I_{sp} g_0}} = \frac{M_i}{2000 + \frac{2000\lambda}{1-\lambda}}$$

$$\Rightarrow e^{\frac{\Delta v}{I_{sp} g_0}} = \frac{M_i(1-\lambda)}{2000\lambda}$$

$$\Delta v = 8 \text{ km/s} = 8000 \text{ m/s} \quad I_{sp} = [350, 400, 450]$$

$$i) I_{sp} = 350 \text{ s}$$

$$2000 \times e^{\frac{8000}{350 \times 9.8}} = M_i(1-\lambda)$$

$$\Rightarrow \text{~~2000~~} \times M_i(1-\lambda) = 2.06 \times 10^4$$

$$ii) I_{sp} = 400 \text{ s}$$

$$M_i(1-\lambda) = e^{\frac{8000}{400 \times 9.8}} \times 2000 = \text{~~4.00~~} \times 10^3 = 1.54 \times 10^4$$

$$iii) I_{sp} = 450$$

$$M_i(1-\lambda) = e^{\frac{8000}{450 \times 9.8}} \times 2000 = 1.227 \times 10^4$$

The graph is rectangular hyperbola.