

- ~~① Just buy a pulley w/ 2.3 mm bore — UNLIKELY~~
- ② 2.3 mm to 5 mm shaft ADAPTER and buy bigger pulley
- ③ FIT A SLEEVE ON shaft, effectively INCREASING diameter
- ④ TAKE A STOCK PIECE OF METAL,
drill hole for shaft, and fit pulley on OD of
MAKE-SHIFT SLEEVE
- ⑤ TAKE STOCK ALUMINUM rod : LATHE IT DOWN TO 5mm OD,
drill 2.3 mm hole on ONE end.

TODO

- ① DRILL 2.3 mm hole INTO shaft
- ② FIT PULLEY ONTO shaft USING SET SCREWS : Loctite
- ③ DRILL HOLES FOR SET SCREWS INTO shaft
- ④ MOUNT MOTOR, ENCODER, and BELT/PULLEY SYSTEM
- ⑤ MOUNT ROD ONTO ENCODER (wood block)

→ DONE

PARTS

- ① LINEAR ENCODING STRIP - MEASURE POSITION of CART
- ② ROTATIONAL ENCODER - MEASURE Angle of PENDULUM
- ③ RAILS & CART
- ④ PENDULUM (can be wood)
- ⑤ MOTOR DRIVER - drives the MOTOR
- ⑥ DC MOTOR
- ⑦ MICROCONTROLLER

$$\frac{2\pi 8\text{mm}}{0.1\text{mm}} = 503 \text{ PPR}$$

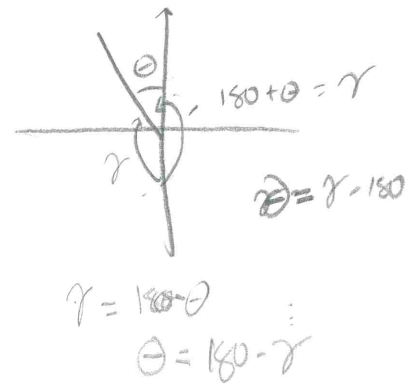
0.5

2π



$$s = r d\theta$$

	Epson	HP	
dpi	2400	600 - <u>PLENTY</u>	$\frac{t_1}{\theta_1}$ $\frac{t_2}{\theta_2}$
V	120 V	110 V AC	
MAX PRINT length	8.5 in	8.5 in	
f	50 HZ	50 Hz hz	
PPM	7.3/13	22/28	



PARTS

✓ ① BELT - GTZ, 2 mm PITCH, 6 mm wide, 550 mm long

✓ ② Timing Pulley - GTZ, 2mm pitch, 6mm wide,

DON'T NEED ③ IDLE PULLEY - 6 mm wide, teeth doesn't MATTER

✓ ④ LINEAR POSITION FINDER - ① OPTICAL range finder
② ROTARY ENCODER

✓ ⑤ belt CLAMP TO CART -

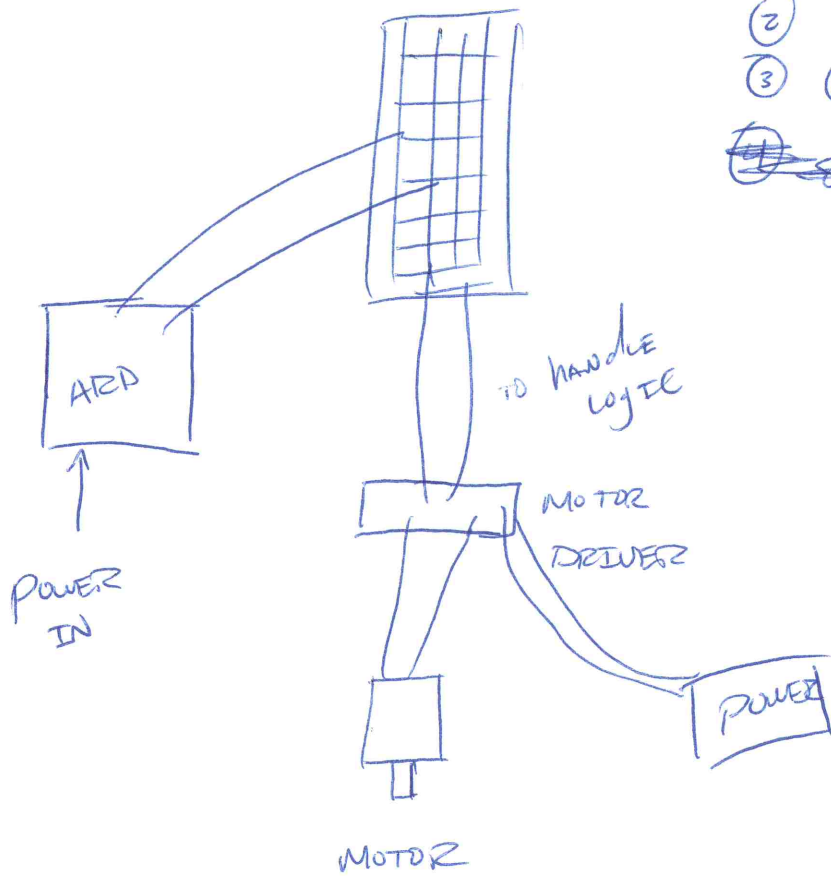
✓ ⑥ BELT TENSOWER -

$$x = r_{\text{gear}} d\theta_{\text{STEP}}$$

DON'T NEED ⑦ MOTOR - 12-18V MOTOR, Ideally we want < 0.5 mm POSITION

PARTS

- ① MULTIMETER
- ② BATTERY PACK
- ③ CLIP AND WIRES - ALLigator
- ~~④ 8 gauge~~



TOP 2

- ① MEASURE VOLTAGE ON MOTOR

V

I

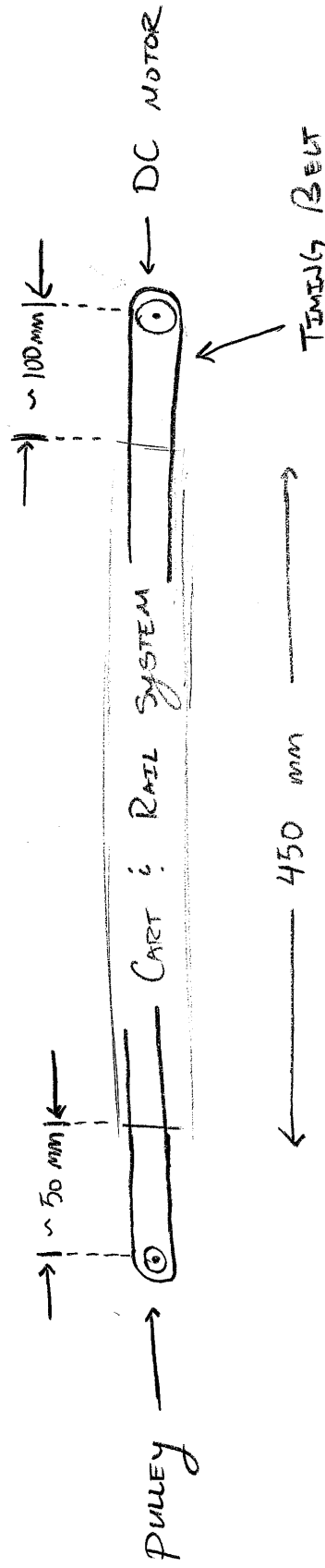
3.2

0.3 A

0.31

* ORIGINAL BELT IS TOO SHORT, SO I NEED
A NEW BELT THAT MEETS THE SPECS of the

DC MOTOR HEAD



DC MOTOR HEAD:

WAS MADE TO FIT THE ORIGINAL BELT W/ SPECS:

~ 1 mm PITCH
~ 3 mm wide

7 # of TEETH

Py CONTROL

① AS A USER

I WANT TO CREATE A PID CONTROLLER
SO I CAN CONTROL MY HARDWARE

② AS A USER

I WANT TO SPECIFY MY GAINS ON THE PID CONTROLLER
SO I CAN TUNE IT MYSELF

③ AS A USER

I WANT TO SPECIFY SATURATION LIMITS
SO MY PID CONTROLLER MEETS MY HARDWARE CONSTRAINTS

④ AS A USER

I WANT TO SPECIFY AN ANTI-WINDUP METHOD
SO MY INTEGRAL TERM DOESN'T ACCUMULATE
UNREASONABLE COMMANDS

⑤ AS A USER,

I WANT TO BE ABLE TO AUTOTUNE MY CONTROLLER
SO IT PERFORMS OPTIMALLY TO THE TF I PROVIDE

⑥

AS A USER,

I WANT TO TUNE MY PID BASED ON TIME-DOMAIN CONSTRAINTS
SO I DON'T HAVE TO DO THE MATH MYSELF

⑦ AS A USER,

I WANT TO TUNE MY PID BASED ON S-DOMAIN CONSTRAINTS
SO I DON'T HAVE TO DO THE MATH MYSELF

⑧ AS A USER,

I WANT TO TUNE MY PID BASED ON FREQUENCY-DOMAIN CONSTRAINTS
SO I DON'T HAVE TO DO THE MATH MYSELF

⑨ AS A USER,

I WANT TO HAVE THE OPTION OF CHOOSING A DERIVATIVE FILTER
SO I CAN BLOCK HIGH-FREQUENCY NOISE

pygci

① AS A USER

I WANT TO SPECIFY THE # OF POPS, # OF CHILDREN,
SO I CAN HAVE CONTROL OF WHAT THE ALGO IS PERFORMING

② AS A USER,

I WANT TO SPECIFY THE FITNESS FUNCTION TO EVALUATE RESULTS ON
SO THAT I CAN CUSTOMIZE THE RESULTS

③ AS A USER,

I WANT TO BE RETURNED ALL OF THE DATA ABOUT THE
ALGORITHM SO I CAN DO WHAT I WANT W/ THE DATA.

MEASUREMENTS

$$d_p = 0.233''$$

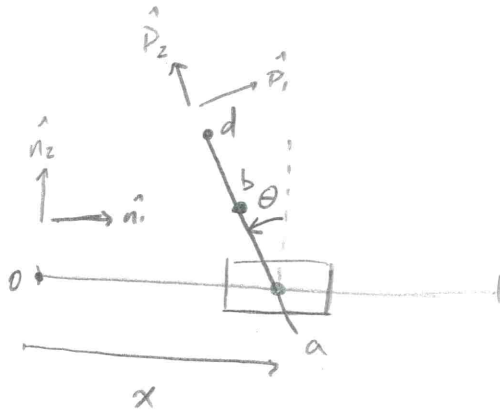
$$d_e = 0.235''$$

DYNAMIC

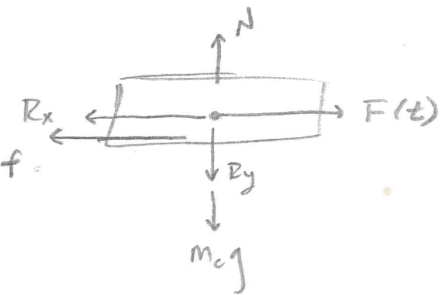
MODEL

①

① IDENTIFY SYSTEM



$$\begin{pmatrix} \hat{P}_1 \\ \hat{P}_2 \\ \hat{P}_3 \end{pmatrix} = \begin{pmatrix} \cos\theta & \sin\theta & 0 \\ -\sin\theta & \cos\theta & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{pmatrix}$$



POTENTIAL: $m_c g$

CONSTRAINT: N, R_y

IMPRESSED: $R_x, F(t), f$

$$\vec{f}_c = (F(t) - R_x - f) \hat{n}_1 + (N - R_y - m_c g) \hat{n}_2$$

WHERE,

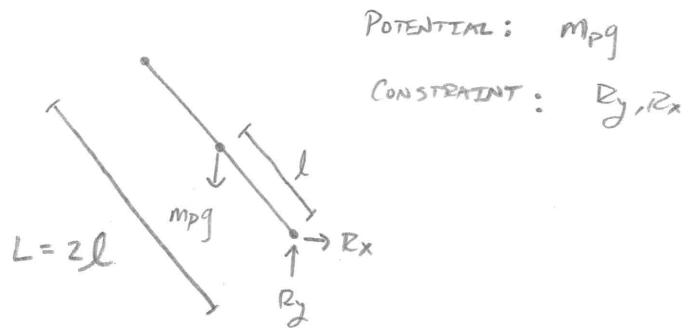
$$f = \epsilon \dot{x}$$

SO,

$$\vec{f}_c = (F(t) - R_x - \epsilon \dot{x}) \hat{n}_1 + (N - R_y - m_c g) \hat{n}_2$$

$$\vec{f}_c = 0$$

② FBD



POTENTIAL: $m_p g$

CONSTRAINT: R_y, R_x

$$\vec{f}_p = (R_x) \hat{n}_1 + (R_y - m_p g) \hat{n}_2$$

$$\vec{l}_c^p = \vec{r} \times \vec{F}$$

$$= \begin{pmatrix} 0 \\ -l \\ 0 \end{pmatrix}_P \times \begin{pmatrix} R_x \\ R_y \\ 0 \end{pmatrix}_N$$

$$= l \begin{pmatrix} \sin\theta \\ -\cos\theta \\ 0 \end{pmatrix}_N \times \begin{pmatrix} R_x \\ R_y \\ 0 \end{pmatrix}_N$$

$$= l (R_y \sin\theta + R_x \cos\theta) \hat{n}_3$$

③ KINEMATICS : CARDOINAL VECTORS

PENDULUM

$$\begin{aligned} \vec{r}_{b/o} &= \vec{r}_{a/o} + \vec{r}_{b/a} \\ &= x \hat{n}_1 + l \hat{P}_2 \end{aligned}$$

CART

$$\vec{r}_{a/o} = x \hat{n}_1$$

$$\dot{\vec{r}}_{a/o} = \dot{x} \hat{n}_1$$

$$\dot{\vec{r}}_{b/o} = \frac{d}{dt} \vec{r}_{a/o} + \frac{d}{dt} \vec{r}_{b/a} + \vec{\omega}_{P/N} \times \vec{r}_{b/a}$$

$$= \dot{x} \hat{n}_1 + \frac{d}{dt} (l \hat{P}_2) + \begin{pmatrix} 0 \\ 0 \\ \dot{\theta} \end{pmatrix}_P \times \begin{pmatrix} 0 \\ l \\ 0 \end{pmatrix}_P$$

$$= \dot{x} \hat{n}_1 - l \dot{\theta} \hat{P}_1$$

$$= \dot{x} \hat{n}_1 - l \dot{\theta} (\cos\theta \hat{n}_1 + \sin\theta \hat{n}_2)$$

$$= (\dot{x} - l \dot{\theta} \cos\theta) \hat{n}_1 + (-l \dot{\theta} \sin\theta) \hat{n}_2$$

$$\vec{\omega}_{P/N} = \dot{\theta} \hat{P}_2 = \dot{\theta} \hat{n}_3$$

CARDINAL VECTORS

$$\mathcal{L}_x^c = \frac{\partial \vec{v}_{A/O}}{\partial \dot{x}} = \hat{n}_1$$

$$\mathcal{L}_\theta^c = 0$$

$$\rho_x^c = \rho_\theta^c = 0$$

$$\mathcal{L}_x^p = \frac{\partial \vec{v}_{B/O}}{\partial \dot{x}} = \hat{n}_1$$

$$\mathcal{L}_\theta^p = \frac{\partial \vec{v}_{B/O}}{\partial \dot{\theta}} = -l \hat{r}_1$$

$$\rho_\theta^p = \frac{\partial \vec{\omega}_{FW}}{\partial \dot{\theta}} = \hat{P}_3 = \hat{n}_3$$

④ GENERALIZED FORCES

WORK RATE PRINCIPLE:

$$\dot{W} = \sum_{\text{body } i} [\vec{f}_i \cdot \vec{v}_i + \vec{l}_i^i \cdot \vec{\omega}_i]$$

$$\dot{W}_{\text{pend}} = \vec{f}_P \cdot \vec{r}_{B/O} + \vec{l}_c^P \cdot \vec{\omega}_{FW}$$

$$= \begin{pmatrix} R_x \\ R_y - m_P g \\ 0 \end{pmatrix}_N \cdot \begin{pmatrix} \dot{x} - l \dot{\theta} \cos \theta \\ -l \dot{\theta} \sin \theta \\ 0 \end{pmatrix}_N + \begin{pmatrix} 0 \\ 0 \\ l(R_y \sin \theta + R_x \cos \theta) \end{pmatrix}_N \cdot \begin{pmatrix} 0 \\ 0 \\ \dot{\theta} \end{pmatrix}_N$$

$$= R_x \dot{x} - (R_x l \cos \theta) \dot{\theta} - (R_y l \sin \theta) \dot{\theta} + (m_P g l \sin \theta) \dot{\theta} + l(R_y \sin \theta + R_x \cos \theta) \dot{\theta}$$

$$= (R_x) \dot{x} + l(m_P g \sin \theta - R_x \cos \theta - R_y \sin \theta + R_y \sin \theta + R_x \cos \theta) \dot{\theta}$$

$$= (R_x) \dot{x} + (m_P g l \sin \theta) \dot{\theta}$$

$$\dot{W}_{\text{CART}} = \vec{f}_c \cdot \vec{r}_{A/O} + \vec{f}_c \cdot \vec{v}_N$$

$$= \begin{pmatrix} F(t) - R_x - f \\ N - R_y - m_c g \\ 0 \end{pmatrix}_N \cdot \begin{pmatrix} \dot{x} \\ 0 \\ 0 \end{pmatrix}_N$$

$$= (F(t) - R_x - f) \dot{x}$$

$$\therefore \dot{W} = \dot{W}_{\text{pend}} + \dot{W}_{\text{CART}}$$

$$= (F(t) - f) \dot{x} + (m_P g l \sin \theta) \dot{\theta}$$

$$= \underbrace{(F(t) - f)}_{Q_{\dot{x}}} \dot{x} + \underbrace{(m_P g l \sin \theta)}_{Q_{\dot{\theta}}} \dot{\theta}$$

$Q_{\dot{x}}$

$Q_{\dot{\theta}}$

⑤ LAGRANGE'S EQU

②

$$L = \sum_{\text{body } i} (T_i - V_i)$$

$$= T_{\text{trans}}^P + T_{\text{rot}}^P + T_{\text{trans}}^C$$

$$= \frac{1}{2} M_P \vec{v}_P \cdot \vec{v}_P + \frac{1}{2} \vec{\omega}_{P/N} \cdot \mathbf{I}_C^P \vec{\omega}_{P/N} + \frac{1}{2} M_C \vec{v}_C \cdot \vec{v}_C$$

Assume a principal axis frame on the pendulum, so \mathbf{I}_C^P is constant and we have $\mathbf{I}_C^P = \mathbf{I}_3^P = \frac{1}{12} M_P L^2$, where L is the total length of the rod. so,

$$L = \frac{1}{2} M_P \begin{pmatrix} \dot{x} - l\dot{\theta} \cos \theta \\ -l\dot{\theta} \sin \theta \\ 0 \end{pmatrix}_N \cdot \begin{pmatrix} \dot{x} - l\dot{\theta} \cos \theta \\ -l\dot{\theta} \sin \theta \\ 0 \end{pmatrix}_N + \frac{1}{2} \begin{pmatrix} 0 \\ 0 \\ \dot{\theta} \end{pmatrix}_P \cdot \mathbf{I}_C^P \begin{pmatrix} 0 \\ 0 \\ \dot{\theta} \end{pmatrix}_P + \frac{1}{2} M_C \begin{pmatrix} \dot{x} \\ 0 \\ 0 \end{pmatrix}_N \cdot \begin{pmatrix} \dot{x} \\ 0 \\ 0 \end{pmatrix}_N$$

$$= \frac{1}{2} M_P \left[(\dot{x} - l\dot{\theta} \cos \theta)^2 + l^2 \dot{\theta}^2 \sin^2 \theta \right] + \frac{1}{2} \mathbf{I}_C^P \dot{\theta}^2 + \frac{1}{2} M_C \dot{x}^2$$

$$= \frac{1}{2} M_P \left[\dot{x}^2 - 2\dot{x}l\dot{\theta} \cos \theta + l^2 \dot{\theta}^2 \cos^2 \theta + l^2 \dot{\theta}^2 \sin^2 \theta \right] + \frac{1}{2} \mathbf{I}_C^P \dot{\theta}^2 + \frac{1}{2} M_C \dot{x}^2$$

$$= \frac{1}{2} M_P \left[\dot{x}^2 - (2l \cos \theta) \dot{x} \dot{\theta} + l^2 \dot{\theta}^2 \right] + \frac{1}{2} \mathbf{I}_C^P \dot{\theta}^2 + \frac{1}{2} M_C \dot{x}^2$$

$$= \frac{1}{2} (M_P + M_C) \dot{x}^2 + \frac{1}{2} (M_P l^2 + \mathbf{I}_C^P) \dot{\theta}^2 - (M_P l \cos \theta) \dot{x} \dot{\theta}$$

④ EQUATIONS of MOTION

$$Q_{\dot{\theta}} = \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\theta}} \right) - \frac{\partial L}{\partial \theta}$$

Derivatives

$$\frac{\partial L}{\partial \dot{\theta}} = (m_P l^2 + I_C^P) \dot{\theta} - (m_P l \cos \theta) \dot{x}$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\theta}} \right) = (m_P l^2 + I_C^P) \ddot{\theta} - m_P l (\ddot{x} \cos \theta - \dot{x} \dot{\theta} \sin \theta)$$

$$= (m_P l^2 + I_C^P) \ddot{\theta} + m_P l (\dot{x} \dot{\theta} \sin \theta - \ddot{x} \cos \theta)$$

$$\frac{\partial L}{\partial \theta} = -m_P l \dot{x} \dot{\theta} (-\sin \theta) = m_P l \dot{x} \dot{\theta} \sin \theta$$

$$\therefore Q_{\dot{\theta}} = m_P g l \sin \theta = (m_P l^2 + I_C^P) \ddot{\theta} + m_P l (\cancel{\dot{x} \dot{\theta} \sin \theta} - \ddot{x} \cos \theta) - m_P l \cancel{\dot{x} \dot{\theta} \sin \theta}$$

$$\boxed{m_P g l \sin \theta = (m_P l^2 + I_C^P) \ddot{\theta} - m_P l \ddot{x} \cos \theta} \quad \text{--- ①}$$

I think this should
be POSITIVE...

$$Q_{\dot{x}} = \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{x}} \right) - \frac{\partial L}{\partial x}$$

Derivatives

$$\frac{\partial L}{\partial \dot{x}} = (m_p + m_c) \dot{x} - m_p l \dot{\theta} \cos \theta$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{x}} \right) = (m_p + m_c) \ddot{x} - m_p l (\ddot{\theta} \cos \theta + \dot{\theta} (-\sin \theta) \dot{\theta})$$

$$= (m_p + m_c) \ddot{x} - m_p l (\ddot{\theta} \cos \theta - \dot{\theta}^2 \sin \theta)$$

$$= (m_p + m_c) \ddot{x} + m_p l (\dot{\theta}^2 \sin \theta - \ddot{\theta} \cos \theta)$$

$$\frac{\partial L}{\partial x} = 0$$

$$\therefore Q_{\dot{x}} = F(t) - \epsilon \dot{x} = (m_p + m_c) \ddot{x} + m_p l (\dot{\theta}^2 \sin \theta - \ddot{\theta} \cos \theta)$$

$$= (m_p + m_c) \ddot{x} + m_p l \dot{\theta}^2 \sin \theta - m_p l \ddot{\theta} \cos \theta$$

$$\rightarrow F(t) - \epsilon \dot{x} = (m_p + m_c) \ddot{x} + m_p l \dot{\theta}^2 \sin \theta - m_p l \ddot{\theta} \cos \theta$$

(2)

↑
SHOULD BE
NEGATIVE

↑
SHOULD BE
POSITIVE

Solving ① for $\ddot{\theta}$ and ② for \ddot{x} yields

$$\ddot{\theta} = m_P l (m_P l^2 + I_c^P)^{-1} (\ddot{x} \cos \theta + g \sin \theta) \quad \text{--- ①}$$

$$\ddot{x} = (m_P + m_c)^{-1} (F(t) - f + m_P l \ddot{\theta} \cos \theta - m_P l \dot{\theta}^2 \sin \theta) \quad \text{--- ②}$$

Substituting ① into ② yields

$$\begin{aligned} (m_P + m_c) \ddot{x} &= F(t) - f - m_P l \dot{\theta}^2 \sin \theta + m_P l \cos \theta (\beta) (\ddot{x} \cos \theta + g \sin \theta) \\ &= F(t) - f - m_P l \dot{\theta}^2 \sin \theta + m_P l \beta \ddot{x} \cos^2 \theta + m_P l g \beta \sin \theta \cos \theta \end{aligned}$$

$$(m_P + m_c - m_P l \beta \cos^2 \theta) \ddot{x} = F(t) - f - m_P l \dot{\theta}^2 \sin \theta + m_P l g \beta \sin \theta \cos \theta$$

$$\therefore \ddot{x} = [m_P + m_c - m_P l \beta \cos^2 \theta]^{-1} [F(t) - f + m_P l \sin \theta (g \beta \cos \theta - \dot{\theta}^2)]$$

--- ③.

Plugging ③ into ① yields

$$\ddot{\Theta} = \beta g \sin \theta + \beta \cos \theta (\ddot{x})$$

$$\ddot{\Theta} = \beta g \sin \theta + \beta \cos \theta [m_P + m_C - m_P \beta \cos^2 \theta]^{-1} [F(t) - f + m_P l \sin \theta (g \beta \cos \theta - \dot{\theta}^2)]$$

$$= \frac{\beta}{m_C + (1 - \beta \cos^2 \theta) m_P} \left[g \sin \theta (m_C + (1 - \beta \cos^2 \theta) m_P) + \cos \theta (F(t) - f + m_P l \sin \theta (g \beta \cos \theta - \dot{\theta}^2)) \right]$$

$$= \gamma [m_C g \sin \theta + m_P g \sin \theta - m_P l g \beta \sin \theta \cos^2 \theta + F(t) \cos \theta - f \cos \theta + m_P l g \beta \sin \theta \cos^2 \theta - m_P l \dot{\theta}^2 \sin \theta \cos \theta]$$

$$= \gamma [(m_C + m_P) g \sin \theta + (F(t) - f) \cos \theta - m_P l \dot{\theta}^2 \sin \theta \cos \theta]$$

$$\ddot{x} = (m_C + m_P (1 - \beta \cos^2 \theta))^{-1} (F(t) - f + m_P l g \beta \sin \theta \cos \theta - m_P l \dot{\theta}^2 \sin \theta)$$

$$\ddot{\Theta} = \beta (m_C + m_P (1 - \beta \cos^2 \theta))^{-1} ((m_C + m_P) g \sin \theta + (F(t) - f) \cos \theta - m_P l \dot{\theta}^2 \sin \theta \cos \theta)$$

$$\beta = \frac{m_P l}{m_P l^2 + I_C^P} = \frac{1}{l + I_C^P / m_P l}$$

SWAP SIGNS

We NOW CONVERT THIS TO A SYSTEM OF FIRST ORDER
DEs by letting

$$\begin{aligned} x_1 &= \theta & x_3 &= \dot{\theta} \\ x_2 &= x & x_4 &= \dot{x} \end{aligned}$$

SO we obtain the system of FIRST-ORDER NON LINEAR DEs

$$\textcircled{b} \left\{ \begin{aligned} \dot{x}_1 &= x_3 \\ \dot{x}_2 &= x_4 \\ \dot{x}_3 &= \frac{\beta}{m_c + m_p(1 - l\beta \cos^2 x_1)} \left[(m_c + m_p)g \sin x_1 + (u(t) - c x_4) \cos x_1 - m_p l x_3^2 \sin x_1 \cos x_1 \right] \\ \dot{x}_4 &= \frac{u(t) - c x_4 + m_p l g \beta \sin x_1 \cos x_1 - m_p l x_3^2 \sin x_1}{m_c + m_p(1 - l\beta \cos^2 x_1)} \end{aligned} \right.$$

WHICH has the form $\dot{\vec{x}} = f(\vec{x})$.

(5)

Next, we find the EQ points by solving

$f(\vec{x}) = 0$. That is, $\dot{x}_1 = \dot{x}_2 = \dot{x}_3 = \dot{x}_4 = 0$. Consider the case w/ NO CONTROL INPUT, then

$$\dot{x}_1 = 0 = x_3 \Rightarrow \bar{x}_3 = 0 \Rightarrow \bar{x}_1 = \text{CONST.}$$

$$\dot{x}_2 = 0 = x_4 \Rightarrow \bar{x}_4 = 0 \Rightarrow \bar{x}_2 = \text{CONST.}$$

$$\dot{x}_3 = 0 = (m_c + m_P) g \sin(\bar{x}_1) + (\cancel{\ddot{u}(t)} - \cancel{\epsilon \bar{x}_4}) \cos \bar{x}_1 = 0$$

$$0 = \sin(\bar{x}_1)$$

$$\therefore \bar{x}_1 = \arcsin(0) = n\pi, \quad n \in \mathbb{Z}_{\geq 0}$$

$$\dot{x}_4 = 0 = \cancel{\ddot{u}(t)} - \cancel{\epsilon \bar{x}_4} + m_P g \beta \sin \bar{x}_1 \cos \bar{x}_1 = 0$$

$$0 = \sin \bar{x}_1 \cos \bar{x}_1$$

$$= (0) \cos(\bar{x}_1)$$

$$0 = 0.$$

Therefore,

$$\vec{x}_{eq} = \begin{pmatrix} \bar{x}_1 \\ \bar{x}_2 \\ \bar{x}_3 \\ \bar{x}_4 \end{pmatrix} = \begin{pmatrix} n\pi \\ c \\ 0 \\ 0 \end{pmatrix} \quad \text{for } n \in \mathbb{Z}_{\geq 0} \text{ and } c \in \mathbb{R}$$

We now linearize (5) about \vec{x}_{eq} using Taylor series expansion method. Note that,

$$\dot{x}_1 = f_1(x_3)$$

$$\dot{x}_2 = f_2(x_4)$$

$$\dot{x}_3 = f_3(x_1, x_3, x_4, u(t))$$

$$\dot{x}_4 = f_4(x_1, x_3, x_4, u(t)).$$

So,

$$\delta \vec{x} = Df(\vec{x}_{eq}) \delta \vec{x}$$

where $\delta \vec{x} = \vec{x} - \vec{x}_{eq}$, and $Df(\vec{x}_{eq})$ is the JACOBIAN of $f(\vec{x}_{eq})$.

Computing the JACOBIAN

$$Df(\vec{x}_{eq}) = \begin{pmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \frac{\partial f_1}{\partial x_3} & \frac{\partial f_1}{\partial x_4} \\ \frac{\partial f_2}{\partial x_1} & \frac{\partial f_2}{\partial x_2} & \frac{\partial f_2}{\partial x_3} & \frac{\partial f_2}{\partial x_4} \\ \frac{\partial f_3}{\partial x_1} & \frac{\partial f_3}{\partial x_2} & \frac{\partial f_3}{\partial x_3} & \frac{\partial f_3}{\partial x_4} \\ \frac{\partial f_4}{\partial x_1} & \frac{\partial f_4}{\partial x_2} & \frac{\partial f_4}{\partial x_3} & \frac{\partial f_4}{\partial x_4} \end{pmatrix}$$

yields

(6)

COMPUTATIONS

$$\delta \dot{x}_3 = \cancel{\frac{\partial f_3}{\partial x_4}} + \frac{\partial f_3}{\partial x_1} \bigg|_{eq} (x_1 - \bar{x}_1) + \frac{\partial f_3}{\partial x_3} \bigg|_{eq} (x_3 - \bar{x}_3) + \frac{\partial f_3}{\partial x_4} \bigg|_{eq} (x_4 - \bar{x}_4) + \frac{\partial f_3}{\partial u(t)} \bigg|_{eq} (u(t) - \bar{u}(t))$$

WOLFRAM ALPHA

$$\frac{\partial f_3}{\partial x_1} \bigg|_{eq} = \beta \frac{\partial}{\partial x_1} \left[\frac{(m_c + m_p)g \sin x_1 + (u(t) - \epsilon x_4) \cos x_1 - m_p l x_3^2 \sin x_1 \cos x_1}{m_c + m_p(1 - l\beta \cos^2 x_1)} \right]$$

$$= \frac{\beta}{(m_c + m_p(1 - l\beta \cos^2 \bar{x}_1))} \left((m_c + m_p)g \cos \bar{x}_1 - (u(t) - \epsilon x_4) \cancel{\sin \bar{x}_1} - m_p l x_3^2 \cancel{\sin \bar{x}_1 \cos \bar{x}_1} \right)$$

$$= \frac{(m_c + m_p)g\beta \cos \bar{x}_1}{m_c + m_p(1 - l\beta \cos^2 \bar{x}_1)}, \quad \text{but } \cos^2(\bar{x}_1 = n\pi) \text{ is ALWAYS } = 1$$

$$= \frac{(m_c + m_p)g\beta \cos(\bar{x}_1)}{m_c + m_p(1 - l\beta)}$$

$$\frac{\partial f_3}{\partial x_3} \bigg|_{eq} = \frac{\beta}{m_c + m_p(1 - l\beta \cos^2 \bar{x}_1)} \left(-2m_p l \cancel{x_3} \sin \bar{x}_1 \cos \bar{x}_1 \right) = 0$$

$$\frac{\partial f_3}{\partial x_4} \bigg|_{eq} = \frac{\beta}{m_c + m_p(1 - l\beta \cos^2(\bar{x}_1))} \left(-\epsilon \cos \bar{x}_1 \right) = \frac{-\epsilon \beta \cos \bar{x}_1}{m_c + m_p(1 - l\beta)}$$

$$\frac{\partial f_3}{\partial u(t)} \bigg|_{eq} = \frac{\beta}{m_c + m_p(1 - l\beta \cos^2(\bar{x}_1))} \left(\cos(\bar{x}_1) \right) = \frac{\beta \cos(\bar{x}_1)}{m_c + m_p(1 - l\beta)}$$

$$\dot{x}_4 \approx \dot{x}_4(x_{eq}) + \left. \frac{\partial \dot{x}_4}{\partial x_1} \right|_{eq} (x_1 - \bar{x}_1) + \left. \frac{\partial \dot{x}_4}{\partial x_3} \right|_{eq} (x_3 - \bar{x}_3) + \left. \frac{\partial \dot{x}_4}{\partial x_4} \right|_{eq} (x_4 - \bar{x}_4) + \left. \frac{\partial \dot{x}_4}{\partial u(t)} \right|_{eq} (u(t) - \bar{u}(t))$$

WOLFRAM ALPHA

$$\left. \frac{\partial \dot{x}_4}{\partial x_1} \right|_{eq} = \frac{(m_p l \beta \cos^2 \bar{x}_1 - m_p l \bar{x}_3 \sin(\bar{x}_1))}{m_c + m_p (1 - l \beta \cos^2 \bar{x}_1)} = \frac{m_p g l \beta}{m_c + m_p (1 - l \beta)}$$

$$\left. \frac{\partial \dot{x}_4}{\partial x_3} \right|_{eq} = \frac{1}{1} (-2 m_p l \bar{x}_3 \sin(\bar{x}_1)) = 0$$

$$\left. \frac{\partial \dot{x}_4}{\partial x_4} \right|_{eq} = \frac{1}{m_c + m_p (1 - l \beta \cos^2 \bar{x}_1)} (-\epsilon) = \frac{-\epsilon}{m_c + m_p (1 - l \beta)}$$

$$\left. \frac{\partial \dot{x}_4}{\partial u(t)} \right|_{eq} = \frac{1}{m_c + m_p (1 - l \beta \cos^2 \bar{x}_1)} = \frac{1}{m_c + m_p (1 - l \beta)}$$

$$A \equiv Df(\vec{x}_{eq}) = \begin{pmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ \frac{g\beta(m_c+m_p)\cos(\bar{x}_1)}{m_c+m_p(1-l\beta)} & 0 & 0 & \frac{-\epsilon\beta\cos(\bar{x}_1)}{m_c+m_p(1-l\beta)} \\ \frac{m_p g l \beta}{m_c+m_p(1-l\beta)} & 0 & 0 & \frac{-\epsilon}{m_c+m_p(1-l\beta)} \end{pmatrix}$$

and,

$$B \equiv \begin{pmatrix} 0 \\ 0 \\ \frac{\beta\cos(\bar{x}_1)}{m_c+m_p(1-l\beta)} \\ \frac{1}{m_c+m_p(1-l\beta)} \end{pmatrix}, \quad C \equiv \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix}$$

$$D \equiv 0.$$

THEREFORE OUR STATE-SPACE MODEL IS

$$\begin{aligned} \dot{\vec{x}} &= A\vec{x} + B\vec{u} \\ \vec{y} &= C\vec{x} \end{aligned}$$

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I N P U T f O R C E
a n d F R I C T I O N

AT 12 V, I'M MEASURING MAX CURRENT DRAW AROUND $\sim 7A$ WITH
SPIKES NORTH OF 10A AROUND STALL.

<u>V (V)</u>	<u>I (A)</u>	
12	1.27	
2	1.2	
12	$\sim 7A - \sim 15A$	<u>NEAR</u> STALLING CONDITIONS

INPUT FORCE

let $u(t) = \alpha V(t) - \gamma \dot{x}$. FROM (McGILURAY 2002),

this can be MODELED AS

$$\begin{aligned} u(t) &= \alpha V(t) - \gamma \dot{x} \\ &= \frac{K_m K_g}{r R_A} V(t) - \frac{(K_m K_g)^2}{r^2 R_A} \dot{x} \end{aligned}$$

WHERE:

K_m : MOTOR TORQUE CONST.

K_g : GEAR RATIO of gear head

r : radius of output gear

R_A : RESISTANCE of MOTOR

① MEASURING R_A

① APPLY VOLTAGE TO MOTOR

② MEASURE CURRENT DRAW

③ $R_A = \frac{V}{I}$

④ REPEAT : AVERAGE

② MEASURING K_m

THE eq for K_m IS DEFINED AS

$$K_m = \frac{\tau}{i} = \frac{60}{2\pi \cdot K_v} \quad (\text{WIKIPEDIA: MOTOR CONSTANTS})$$

WHERE K_v = MOTOR VELOCITY CONSTANT. BUT τ IS A LITTLE DIFFICULT TO MEASURE, SO LET'S FIND K_v . K_v IS DEFINED AS

$$K_v = \frac{\omega_{\text{NO-LOAD}}}{V}$$

TECHNICALLY V IS SUPPOSED TO BE THE V_{PEAK} NOT V_{RMS} . I THINK THIS ONLY APPLIES TO AC VOLTAGES, SO I'M JUST GOING TO USE THE VOLTAGE SUPPLIED TO THE MOTOR.

TWO WAYS TO DO THIS:

① speed - torque curve:

- ① MEASURE NO LOAD RPM, ($\tau=0$)
- ② MEASURE STALL TORQUE, ($\omega=0$)
- ③ MEASURE CURRENT FOR ① : ②
- ④ PLOT SPEED VS. TORQUE USING ① and ②.
- ⑤ PLOT CURRENT VS. TORQUE
- ⑥ THE INVERSE OF THE SLOPE OF ⑤ IS K_m

② FROM K_v :

① MEASURE $\omega_{\text{NO-LOAD}}$

$$\textcircled{2} K_v = \frac{\omega_{\text{NL}}}{V_{\text{IN}}} \quad , \quad K_m = \frac{60}{2\pi \cdot K_v}$$

③ MEASURING K_g

- We don't have MULTIPLE GEARS so our $K_g = 1$

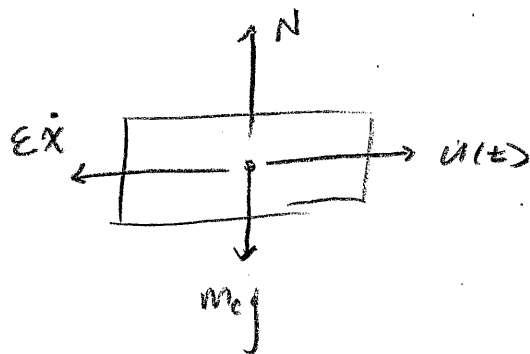
Therefore,

$$U(t) = \alpha V(t) - \gamma \dot{x}$$

$$\alpha = \frac{K_m K_g}{r R_A}$$

$$\gamma = \frac{(K_m K_g)^2}{r^2 R_A}$$

DETERMINING FRICTION



* AT FIRST I WAS GOING TO JUST MEASURE ε BY MOVING CART W/O PENDULUM ACROSS TRACK, BUT PENDULUM WILL INTRODUCE MORE COULOMB FRICTION
 ∴ I HOPE TO CAPTURE THAT IN ε BY INCLUDING (M+MP) IN NEWTON'S 2ND LAW

$$\varepsilon \dot{x} = [N]$$

$$\varepsilon [m/s] = [kg \cdot m/s^2]$$

$$\varepsilon = [kg/s]$$

$$u(t) - \varepsilon \dot{x} = (m_c + m_p) \ddot{x}$$

$$\alpha V(t) - \gamma \dot{x} - \varepsilon \dot{x} = (m_c + m_p) \ddot{x}$$

$$\ddot{x} + \frac{1}{m_c + m_p} (\varepsilon + \gamma) \dot{x} = \frac{\alpha}{m_c + m_p} V(t)$$

INCLUDING MP WILL ALLOW US TO FIND ε THAT ENCOMPASSES COULOMB FRICTION

ASSUME $V(t) = V = \text{CONST}$, and let $x(0) = \dot{x}(0) = 0$. TAKING THE LAPLACE TRANSFORM YIELDS

$$s^2 X(s) - \cancel{s x(0)} - \cancel{\dot{x}(0)} + a(s X(s) - \cancel{x(0)}) = \frac{b}{s}$$

$$(s^2 + as) X(s) = \frac{b}{s}$$

$$X(s) = \frac{b}{s(s^2 + as)}$$

$$= \frac{b}{s^2(s+a)}$$

* DOING IT THIS WAY IS EASIER THAN USING THE COMPLICATED \ddot{x} EOM, and this should be good enough.

$$\left[\frac{b}{s^2(s+a)} = \frac{y_1}{s} + \frac{y_2}{s^2} + \frac{y_3}{s+a} \right] \cdot s^2(s+a)$$

$$\begin{aligned} b &= y_1 s(s+a) + y_2(s+a) + y_3 s^2 \\ &= y_1 s^2 + y_1 s a + y_2 s + y_2 a + y_3 s^2 \\ &= (y_1 + y_3) s^2 + (a y_1 + y_2) s + a y_2 \end{aligned}$$

equating coeffs yields

$$b = a y_2 \Rightarrow y_2 = \frac{b}{a}$$

$$0 = a y_1 + y_2 \Rightarrow y_1 = \frac{-y_2}{a} = -\frac{b}{a^2}$$

$$0 = y_1 + y_3 \Rightarrow y_3 = -y_1 = \frac{b}{a^2}$$

so,

$$X(s) = \frac{b}{s^2(s+a)} = \frac{b/a}{s^2} + \frac{b/a^2}{s+a} - \frac{b/a^2}{s}$$

WITH

$$b = \frac{1}{m_{\text{comp}}} \alpha V \quad \text{and} \quad a = \frac{1}{m_{\text{comp}}} (\zeta + \gamma).$$

$$\therefore X(t) = \frac{b}{a} t + \frac{b}{a^2} e^{-at} - \frac{b}{a^2}$$

$$\dot{X}(t) = \frac{b}{a} (1 - e^{-at}).$$

How we can estimate ϵ :

It would prob be easier to just use the \dot{x} equation,
but I don't have a reliable way to measure velocity, so
I'm going to use the position eq.

We know the length of the track ($x(t_f)$), and we can
measure how much time (t) it takes for the cart to get to
the end track at the applied constant voltage V .

Let $x(t_f) =$ cm

Then,

<u>TRIAL</u>	<u>VOLTAGE (V)</u>	<u>Time (s)</u>
1		
2		
3		
4		
5		
6		
7		
8		
9		
10		

We will then NUMERICALLY APPROXIMATE ϵ in the eq.

$$X(t) = \frac{b}{a^2} \left(e^{-at} + at - 1 \right)$$

$$= \frac{\frac{1}{m+mp} (\alpha V)}{\left(\frac{1}{m+mp}\right)^2 (\epsilon + r)^2} \left(e^{-\frac{\epsilon+r}{m+mp} t} + \left(\frac{\epsilon+r}{m+mp}\right) t - 1 \right)$$

$$X(t) = \frac{\alpha V (m+mp)}{(\epsilon + r)^2} \left(e^{-\left(\frac{\epsilon+r}{m+mp}\right) t} + \left(\frac{\epsilon+r}{m+mp}\right) t - 1 \right)$$

* NO LOAD

USING AUDACITY TO GET Δt of 1 cycle, IT'S PRETTY CONSISTENT THAT

$$RPM = \frac{60s}{\Delta t_{cycle}} = \frac{60s}{0.005s} = 12,000$$

MEASUREMENTS

V (V)	I (A)	R_a
20.4	1.51	
	1.54	
	1.87	
	1.53	
	1.561	
	1.54	
	1.475	

IN ORDER TO FIND R_a , I SUPPLIED $\sim 10\%$ of RATED VOLTAGE, LOCKED THE MOTOR, and FOUND CURRENT. I SUPPLIED 2V but AFTER LOCKING the MOTOR IT ONLY WAS DRAWING ~ 0.340 V, the CURRENT WAS ~ 1.70 A, so

$$R_a = \frac{V}{I} \approx 0.25 \Omega$$

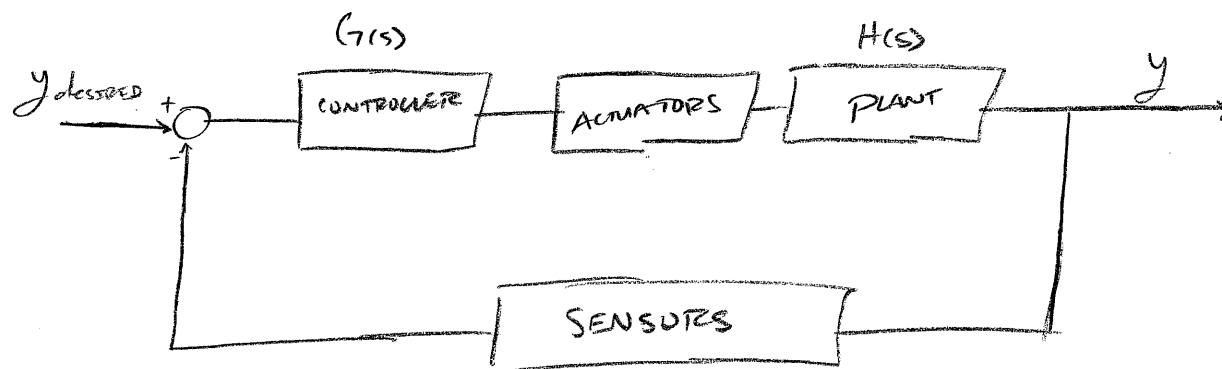
WRONG!

$$R_a \approx 13 \Omega$$

THE MULTIMETER PROBE RESISTANCE IS $\sim 0.1 \Omega$, but I'M JUST GOING TO USE 0.25Ω .

$$1.45 \text{ to } 1.65 \text{ A} \Rightarrow 14.06 \text{ to } 12.36 \Omega$$

NOTES ON TRANSFER FUNCTIONS



$$T_{OL} = G(s) H(s)$$

$$T_{CL} = \frac{G(s) H(s)}{1 + G(s) H(s)}$$

WHEN TO USE WHAT :

- $T_{CL} \rightarrow$ USE TO ANALYZE STABILITY (eg. ROOT CRITERIA)
- $T_{OL} \rightarrow$
 - ① ANALYZE STEADY STATE ERROR
 - ② USE TO START THE ROOT LOCUS
 - ③ BODE PLOT
 - ④ NYQUIST DIAGRAM

YOU CAN STILL

USE T_{CL} TO DRAW
the root locus!

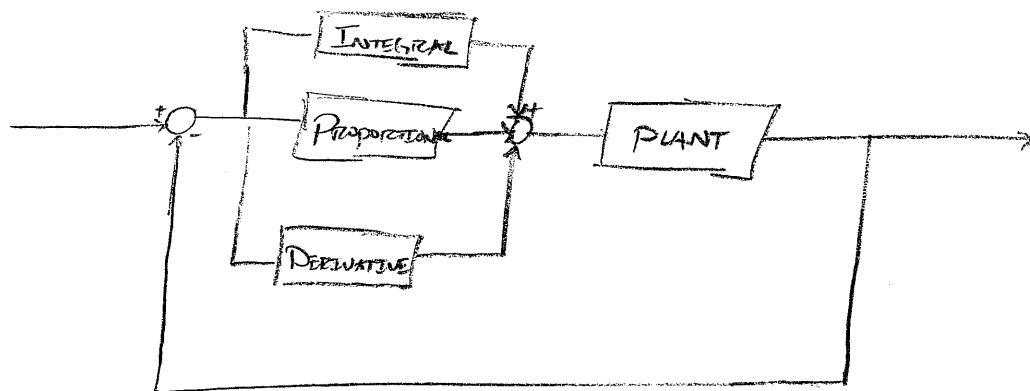
* USE T_{OL} TO ANALYZE & DESIGN CONTROLLERS FOR THE
CLOSED-LOOP SYSTEM

* CHARACTERISTIC EQ ALWAYS REFERS TO THE DENOMINATOR
of T_{CL}

PID TUNING

PID CONTROL

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→ THE IDEA:

- ① PROPORTIONAL K_P , reduces STEADY STATE error
- ② INTEGRAL K_I , "remembers the past", reduces SS error
- ③ DERIVATIVE K_D , "predicts the future", IMPROVES TRANSIENT response.

① → MULTIPLIES error by K_P . the output (eg. voltage) to the plant SCALES PROPORTIONALLY by K_P . MORE error → MORE VOLTAGE

② → INTEGRATES error. By INTEGRATING error, we can SUM the error OVER time Δt and if error CONTINUES to PERSIST we SEND higher and higher VOLTAGES.

③ → DERIVATIVE of error. THE DERIVATIVE of the error lets us KNOW how fast we're CLOSING IN ON OUR position goal, ALLOWING us TO SLOW DOWN & PREVENT OVERSHOOT (ie. TRANSIENT response).

* PID ISN'T PERFECT. We CAN HAVE INTEGRATOR WIND-UP, and high frequency NOISE CAN IMPACT the DERIVATIVE PATH A LOT.

SUMMARY of PID COEFF. effects

		<u>RISE TIME</u>	<u>OVERSHOOT</u>	<u>SETTLING TIME</u>	<u>SS error</u>	<u>STABILITY</u>
INCREASING	K_P	↓	↑	~ ↑	↓	degrades
INCREASING	K_I	~ ↓	↑	↑	↓!	degrades
INCREASING	K_D	~ ↓	↓	↓	~ 0	IMPROVES

~ : SMALL change

! : big change

PID CONTROLLER DESIGN

① OBTAIN A MATHEMATICAL MODEL

→ THIS CAN BE DONE BY DOING PHYSICS OR BY FITTING A MODEL TO DATA COLLECTED FROM YOUR SYSTEM

② TUNE YOUR PID CONTROLLER TO THE SYSTEM

① MANUAL TUNING → POLE PLACEMENT (root locus)
LOOP SHAPING (BODE PLOT : PHASE MARGIN)

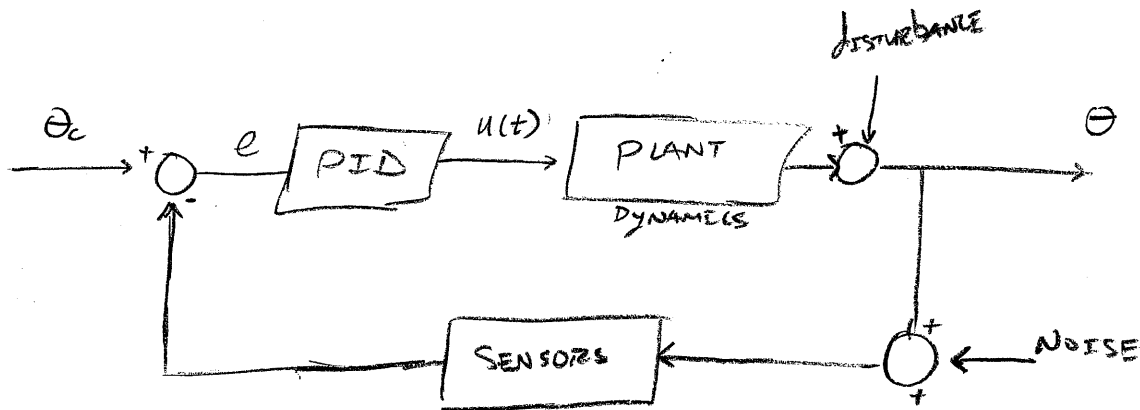
* THIS IS DONE USING THE OPEN-LOOP SYSTEM

② AUTOMATIC TUNING

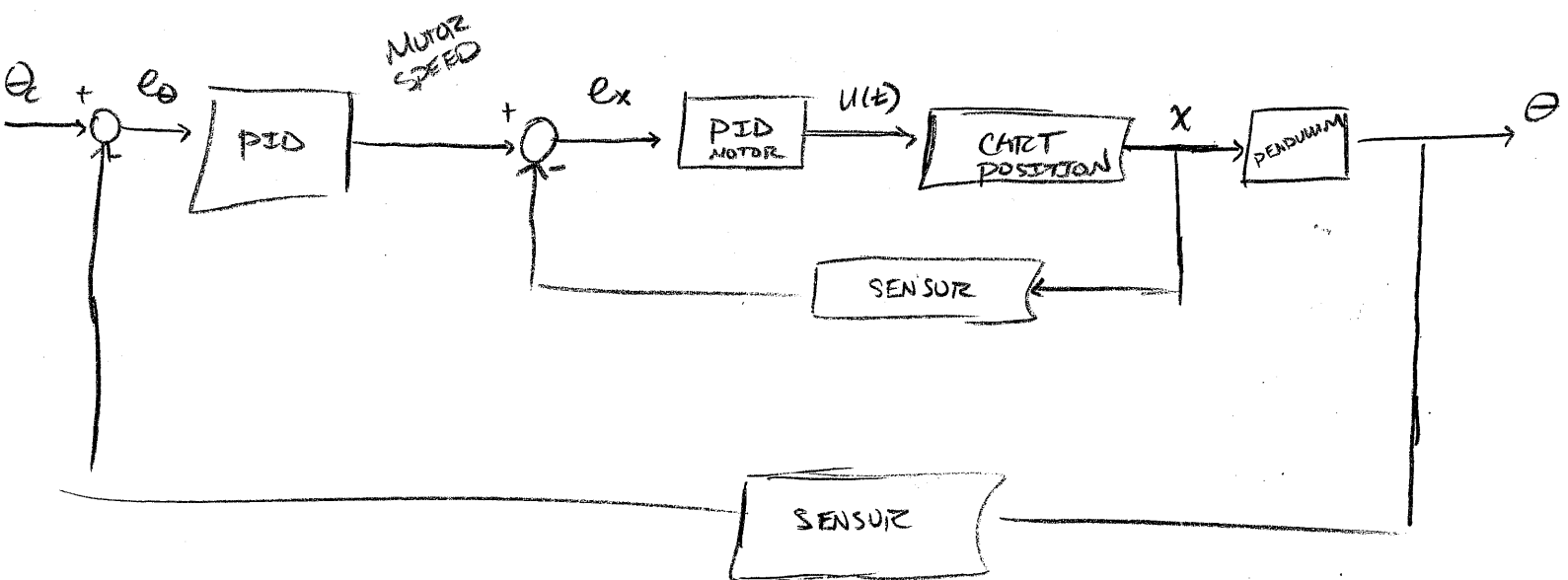
* FOR NON-LINEAR SYSTEMS, LOOK INTO GAIN SCHEDULING, LQR, H_∞ , etc., LOOK AT WIKI FOR NON-LINEAR CONTROL.

PID block diagram

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if we ADD A PID CONTROLLER to both the CART POSITION & the PENDULUM ANGLE, we get SOMETHING like



We CAN ANALYZE this system USING MIMO pole placement to find PID VALUES for each OPEN-LOOP system that will help STABILIZE the CLOSED LOOP system.

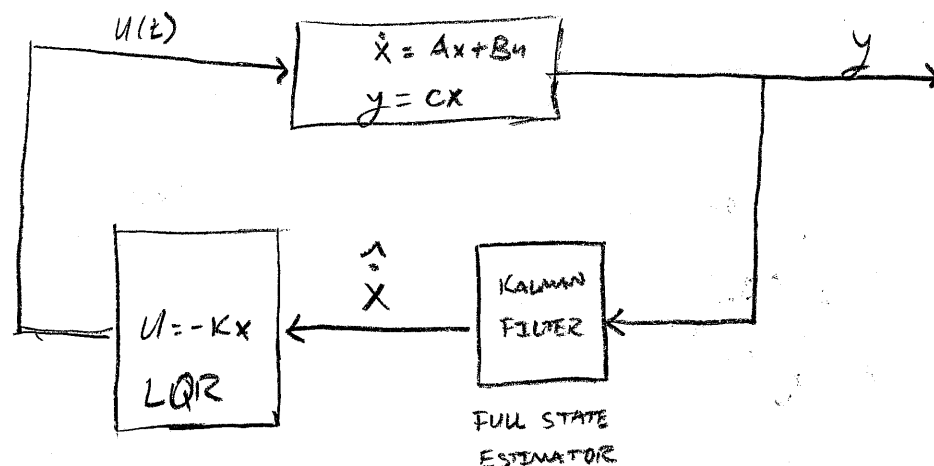
My system

- ① THERE'S TWO WAYS TO DO THIS:
- ① 2 PID CONTROLLERS
 - ② FULL STATE feedback
 - ↳ LQR
 - ↳ POLE PLACEMENT

① 2 PID CONTROLLERS : See PREVIOUS PAGE

② FULL STATE feedback: (LQR : POLE PLACEMENT)

→ SINCE we DON'T HAVE SENSORS TO MEASURE \dot{x} & $\dot{\theta}$, we CAN USE A KALMAN FILTER TO ESTIMATE these STATES. THIS WILL be OUR OBSERVER.

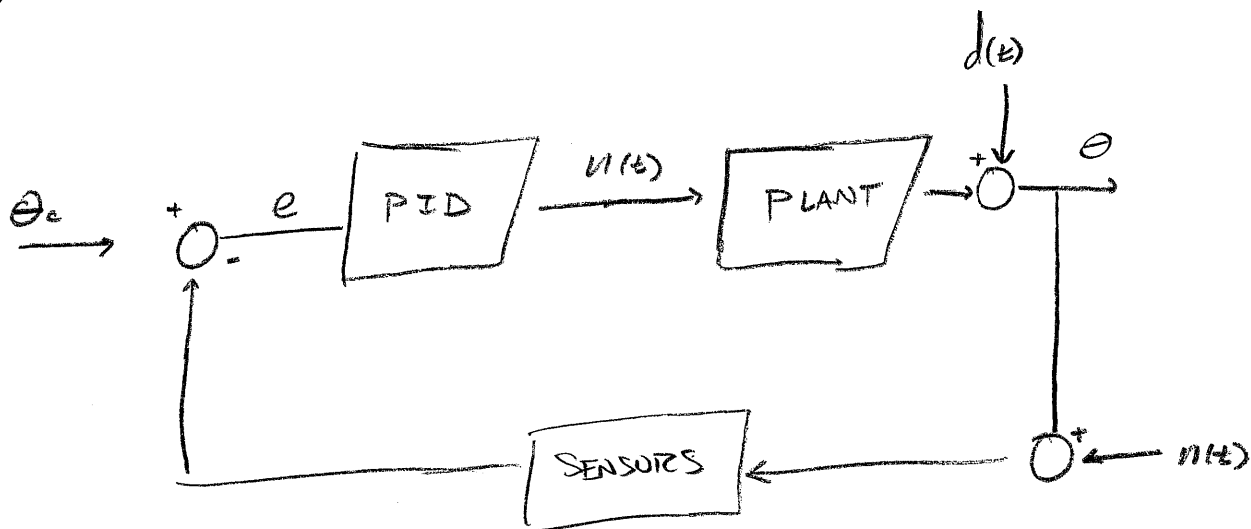


* FULL-STATE feedback (eg. LQR : MIMO POLE PLACEMENT) REQUIRES ALL STATES. SINCE I DON'T HAVE SENSORS TO MEASURE VELOCITY, this REQUIRES A STATE ESTIMATOR (ie. full-observer) like A KALMAN FILTER

APPLYING PID CONTROL

→ SINCE full-STATE feedback is MORE INVOLVED (read: KALMAN filter),
LET'S DO MULTI-LOOP PID CONTROL SO WE CAN GET A WORKING
PROJECT. THEN WE WILL TRY TO IMPLEMENT LQR.

ULTIMATELY, HERE'S WHAT WE WANT



WE PROBABLY COULD IMPLEMENT THIS FOR JUST θ AND ONLY HAVE ONE
PID CONTROLLER, BUT WE ALSO WANT TO CONTROL THE CART'S POSITION,
SO LET'S USE CASCADE CONTROL. WE NEED TO CHOOSE THE
INNER & OUTER LOOP.

* RULE of thumb: INNER LOOP SHOULD BE 3 to 5x
FASTER than OUTER LOOP.

My first intuition was to choose θ as outer loop & X as inner-loop, but this doesn't seem to be consistent w/ examples I've seen online. After some thought, I think it makes sense.

CASE 1: θ outer loop, X inner loop

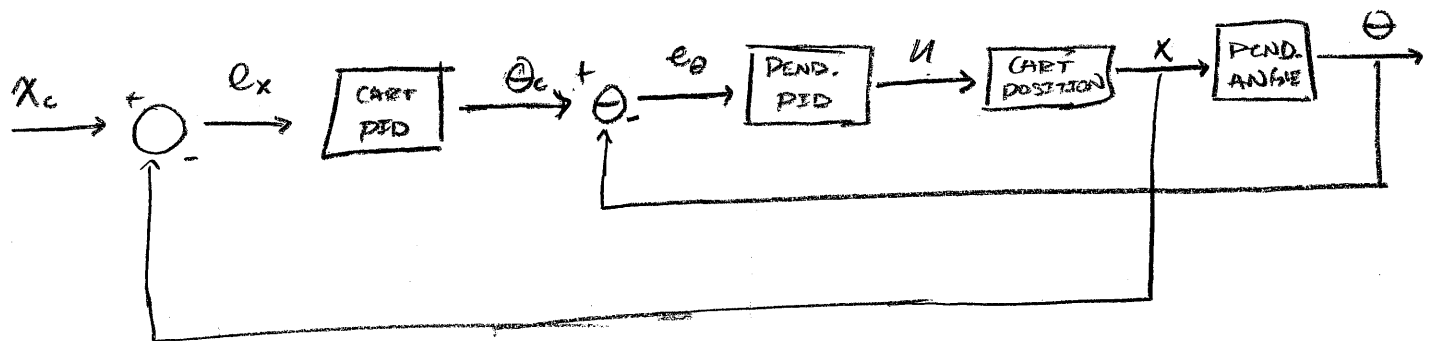
→ THIS MEANS we're driving X closer to the set point faster than we're driving θ to its set point. THIS SHOULDN'T HAPPEN, WE SHOULD

FIRST drive θ to its setpoint then drive X to the set point. Plus, I believe it is faster to drive θ to its set point since driving X to its set point requires θ to be at a set point then we can drive X to a set point.

CASE 2: X outer loop, θ inner loop

→ THIS MEANS we're constantly moving X to drive θ to its setpoint, then this means X is at some distance away from the setpoint so the outer loop drives X to the set-point. However, while X is being driven to its setpoint, the inner loop is constantly correcting θ .

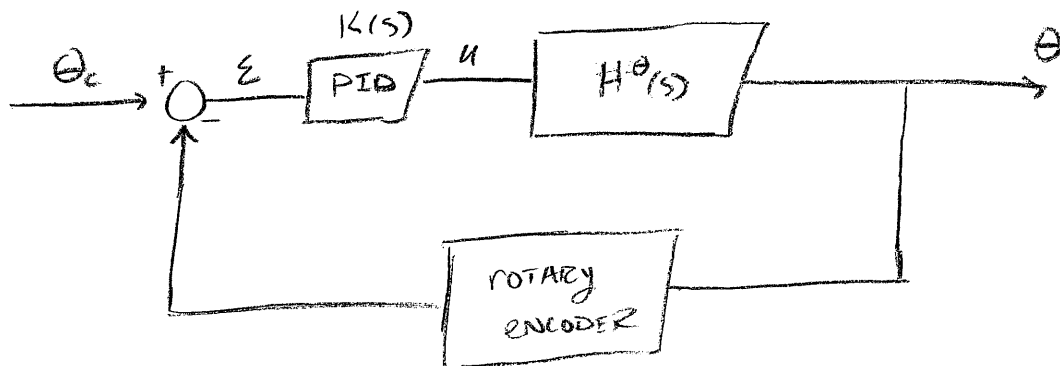
THIS SHOULD GIVE US A CONTROL-BLOCK DIAGRAM THAT LOOKS LIKE:



Is this right?

SCRATCH THIS / Let's just work on tuning the PID

CONTROLLER FOR θ .



CONTROL LAW: $U(s) = K_P E(s) + K_I \int_0^s E(s) ds + K_D \frac{dE(s)}{ds}$

$$U(s) = K_P E(s) + \frac{1}{s} K_I E(s) + K_D s E(s)$$

$$(K_P + \frac{1}{s} K_I + s K_D) E(s)$$

Plugging in some sample values like

$$M_c = 0.320 \text{ kg}$$

$$I = \frac{1}{3} M_P L_{TOT}^2$$

$$M_P = 0.25 \text{ kg}$$

$$\zeta = 0.5 \text{ kg/s}$$

$$L_{TOT} = 0.3 \text{ m}$$

$$L_{CM} = \frac{1}{2} L_{TOT}$$

We get

$$H_{OL}^{\theta}(s) = \frac{3.086s}{s^3 + 1.08s^2 - 17.25s - 15.13}$$

and,

$$H_{CL}^{\theta}(s) = \frac{K(s) H_{OL}^{\theta}(s)}{1 + K(s) H_{OL}^{\theta}(s)} = \frac{K(s) \frac{N(s)}{D(s)}}{1 + K(s) \frac{N(s)}{D(s)}} = \frac{K(s) N(s)}{D(s) + K(s) N(s)}$$

$$= \frac{K(s) (3.086)s}{(s^3 + 1.08s^2 - 17.25s - 15.13) + K(s) (3.086)s}$$

Using $K(s) = \left(K_P + \frac{1}{s} K_I + K_D \left(\frac{sN}{s+N} \right) \right)$ WHERE $\frac{N}{s+N}$ IS A low pass filter.

THE CHARACTERISTIC EQ FOR THE SYSTEM IS

$$\begin{aligned}\Delta^0 &= (s^3 + as^2 + bs + c) + \left(K_P + \frac{1}{s}K_I + K_d \frac{sN}{s+N}\right)ds \\ &= \underbrace{s^3 + as^2 + bs + c}_{\substack{\cdot \\ \cdot \\ \cdot}} + \underbrace{dK_P s}_{\cdot} + \underbrace{dK_I}_{\cdot} + \underbrace{dK_d \frac{s^2 N}{s+N}}_{\cdot} \\ &= s^3 + \left(a + dK_d \frac{N}{s+N}\right)s^2 + (b + dK_P)s + (c + dK_I)\end{aligned}$$

WHERE,

$$a = 1.08 \quad c = -15.13$$

$$b = -17.25 \quad d = 3.086$$

THE OPEN LOOP POLES ARE LOCATED AT

$$0 = s^3 + as^2 + bs + c$$

THESE ARE ALSO THE
EIGENVALUES OF A.

WOLFRAM
ALPHA

\Rightarrow

$$s_1 \approx -4.2828$$

$$s_2 \approx -0.8674$$

$$s_3 \approx 4.0707 \quad \leftarrow \text{SYSTEM IS UNSTABLE.}$$

Let's say we wish to have % OS $\leq 5\%$ and $T_s \leq \frac{1}{2} s$.

This implies

$$T_s \approx \frac{4}{\zeta \omega_n} \leq \beta$$

$$\therefore \zeta \omega_n \geq \frac{4}{\beta}$$

$$PO = 100 e^{-\zeta \pi (1 - \zeta^2)^{-1/2}} \leq \gamma$$

$$-\zeta \pi (1 - \zeta^2)^{-1/2} \leq \ln\left(\frac{\gamma}{100}\right)$$

$$\zeta \pi (1 - \zeta^2)^{-1/2} \geq -\ln\left(\frac{\gamma}{100}\right) = \ln\left(\frac{100}{\gamma}\right)$$

$$\zeta \pi \geq \ln\left(\frac{100}{\gamma}\right) (1 - \zeta^2)^{1/2}$$

$$\zeta^2 \pi^2 \geq \ln\left(\frac{100}{\gamma}\right)^2 (1 - \zeta^2)$$

$$\zeta^2 \pi^2 \geq \ln\left(\frac{100}{\gamma}\right)^2 - \ln\left(\frac{100}{\gamma}\right)^2 \zeta^2$$

$$(\pi^2 + \ln\left(\frac{100}{\gamma}\right)^2) \zeta^2 \geq \ln\left(\frac{100}{\gamma}\right)^2$$

$$\therefore \zeta \geq \ln\left(\frac{100}{\gamma}\right) [\pi^2 + \ln\left(\frac{100}{\gamma}\right)^2]^{-1/2}$$

For $\gamma = 5$ and $\beta = \frac{1}{2}$, we get

$$\zeta \geq \sim 0.69$$

$$\omega_n \geq \sim 11.59.$$

* We now have an issue because those values only have meaning for 2nd order systems and we're dealing w/ a 3rd order system. So instead of thinking about specifying ζ and ω_n , we need to think in terms of where to place the poles and zeros to achieve the desired response.

NOTES ON ROOT LOCUS : PID DESIGN

THE WORKFLOW FOR ROOT LOCUS DESIGN IN MATLAB GOES LIKE THIS:

- ① DEFINE OL TRANSFER FUNCTION
- ② PLOT the POLES & ZEROS IN the S-DOMAIN
- ③ LOOK AT HOW CHANGING GAIN effects the POLE PLACEMENT
- ④ LOOK AT HOW new POLE PLACEMENT effects CLOSED-LOOP STABILITY

THERE ARE SEVERAL different ways TO DO EACH STEP:

①

(a) $s = tf('s')$. THEN $G = \frac{ss}{4+s}$, etc.

(b) Define NUMERATOR & DENOMINATOR COEFFS:

$$n = [1] \\ d = [1 \ 5 \ 6] \rightarrow G = tf(n,d) \rightarrow G = \frac{1}{s^2 + 5s + 6}$$

(c) DEFINE USING POLES, ZEROS, and GAIN.

$$z = []$$

$$p = [0, -4, -2+4i, -2-4i] \rightarrow G = ZPK(z, p, k)$$

$$k = 1$$

②

⑨ $\text{pzmap}(G)$ - SHOWS POLES & ZEROS of transfer function
→ DOESN'T SHOW EFFECTS of gain!

③

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```
for K = 1:10
    pzmap(feedback(G*K, 1))
end
```

Plots closed-loop poles
- POLES for varying
gain
→ This is difficult!

6

⑥ $r_{\text{locus}}(G)$ - easy! shows effects of gain on closed-loop poles.

4

⑨ USE NEW CLOSED LOOP POLES FROM ③ TO ANALYZE STEP RESPONSE.

$$\#CL = \#(\sim)$$

$$\text{STEP}(\#CL).$$

* All these steps are tedious !!

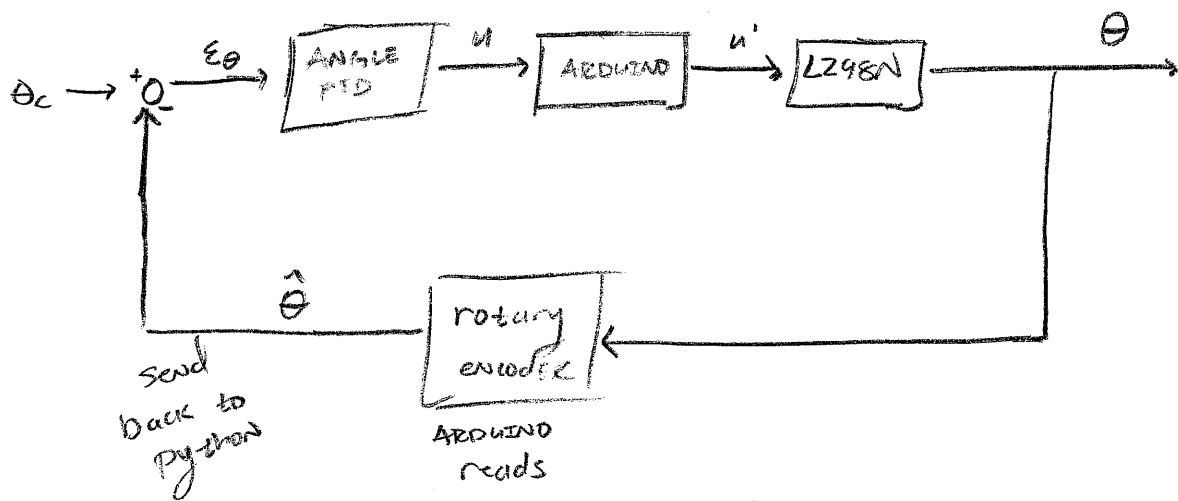
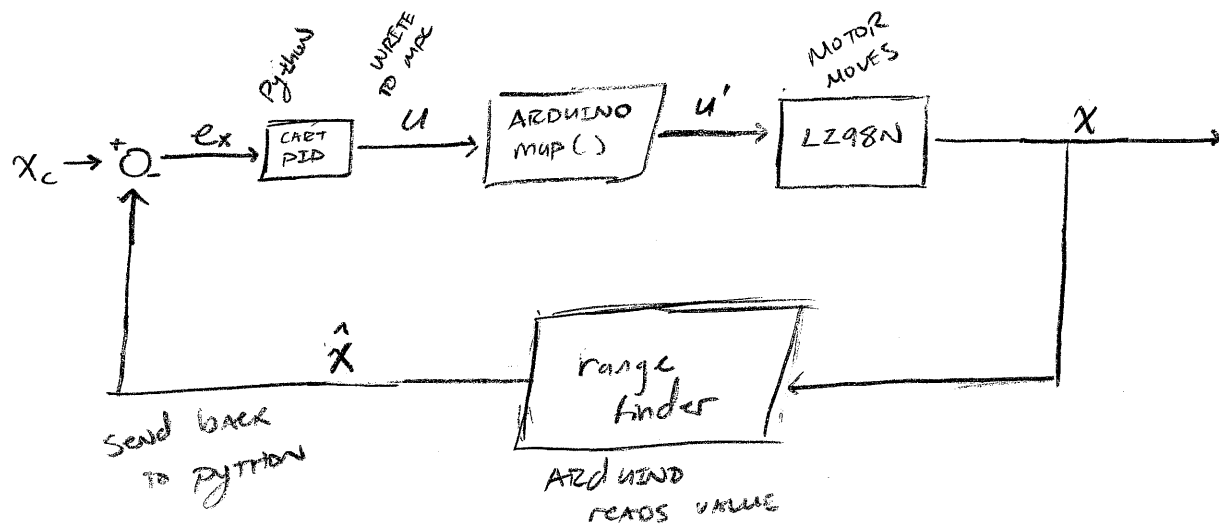
PRACTICAL APPROACH

- ① USE MATLAB CONTROL SYSTEM DESIGNER APP
- ② OPEN ROOT LOCUS and STEP PLOTS

* THIS IS FOR MANUTZ TUNING, A PID.

HERE IS MY CURRENT PLAN OF ATTACK NOW THAT I HAVE A LITTLE MORE INSIGHT ON HOW TO IMPLEMENT THIS:

- ① PLAY AROUND WITH SIMULINK AND LEARN ABOUT CONTROLLING & TUNING PIDS
- ② ONCE I FEEL COMFORTABLE DOING THIS, MOVE TO PYTHON AND WRITE A PID CONTROLLER THAT TAKES INPUTS: REFERENCE, K_P , K_I , K_D AND WILL CALCULATE THE OUTPUT. WE CAN TUNE THIS USING A GA.
- ③ CONTROL ARDUINO VIA PYTHON BY HAVING PYTHON WRITE TO THE OUTPUT TO THE ARDUINO. LOOK INTO HOW TO MAKE THIS FAST & EFFICIENT.
- ④ IF POSSIBLE, WRITE ALL THIS CODE ON THE ARDUINO MPU ITSELF FOR OPTIMAL SPEED.
- ⑤ MOVE FROM PID CONTROL TO LQR & LQG. THIS SHOULD HONESTLY BE EASIER. LQG WILL REQUIRE A KALMAN FILTER THOUGH WHICH WILL BE DIFFICULT.



★ Technically, the output of a PID can be any physical value. The output is just an electrical signal $0(\pm)$, and by tuning the gains of the PID, you're making sure $0(\pm)$ is within the range of your desired command value. For example, the command value for my motor is voltage, so how does the PID know how to convert angle error to voltage? It doesn't. By tuning the gains you're making sure the PID output is in the correct voltage range that gives the desired response.

After learning that previous fact, we can definitely choose either cascaded PID or break it into two PIDs.

→ If choosing cascaded, since the θ loop affects the x_{cart} loop, we need to first tune the inner loop then tune the outer loop.

Python Simulation

Pseudocode:

$ICs = \begin{pmatrix} \vdots \end{pmatrix}$

$tspan = ()$

$\text{Integrate}(\text{non-linear-model}, tspan, ICs) = t, \text{states}$

for i in $\text{len}(tspan)$:

$\text{state} = \text{states}(i, :)$

$x = \text{state}(1)$

$\theta = \text{state}(0)$

 square w/ center at x

 rod w/ one end at x and other end at $(L \cos \theta, L \sin \theta)$

end

REDUCING MODEL COMPUTATION USING EOMS FROM PAPER

$$M_{TOTAL} \ddot{x} + c \dot{x} + m_{pl} \ddot{\theta} \cos \theta - m_{pl} \dot{\theta}^2 \sin \theta = U(t) \quad \text{--- (1)}$$

$$m_{pl} \ddot{x} \cos \theta + (m_{pl} l^2 + I_c^p) \ddot{\theta} - m_{pl} g \sin \theta = 0 \quad \text{--- (2)}$$

SOLVING FOR $\ddot{\theta}$ IN (2) GIVES

$$\ddot{\theta} = m_{pl} (m_{pl} l^2 + I_c^p)^{-1} (g \sin \theta - \ddot{x} \cos \theta) \quad \text{--- (3)}$$

PLUGGING INTO (1) YIELDS and LETTING $\beta = m_{pl} (m_{pl} l^2 + I_c^p)^{-1}$ GIVES

$$m_T \ddot{x} + c \dot{x} - m_{pl} \dot{\theta}^2 \sin \theta + m_{pl} \cos \theta \cdot \beta (g \sin \theta - \ddot{x} \cos \theta) = U(t)$$

$$m_T \ddot{x} + c \dot{x} - m_{pl} \dot{\theta}^2 \sin \theta + \beta g m_{pl} \sin \theta \cos \theta - \beta m_{pl} \ddot{x} \cos^2 \theta = U(t)$$

$$m_T \ddot{x} - \beta m_{pl} \ddot{x} \cos^2 \theta = U(t) - c \dot{x} + m_{pl} \dot{\theta}^2 \sin \theta - \beta g m_{pl} \sin \theta \cos \theta$$

$$(m_T - \beta m_{pl} \cos^2 \theta) \ddot{x} = U(t) - c \dot{x} + m_{pl} \sin \theta (\dot{\theta}^2 - g \beta \cos \theta)$$

$$\therefore \ddot{x} = (m_T - \beta m_{pl} \cos^2 \theta)^{-1} [U(t) - c \dot{x} + m_{pl} \sin \theta (\dot{\theta}^2 - g \beta \cos \theta)] \quad \text{--- (4)}$$

Plugging (4) into (3) gives

$$\begin{aligned}\ddot{\theta} &= \beta \left[g \sin \theta - \cos \theta \cdot (m_T - \beta m_P l \cos^2 \theta)^{-1} (u(t) - \epsilon \dot{x} + m_P l \sin \theta (\dot{\theta}^2 - g \beta \cos \theta)) \right] \\ &= \frac{\beta}{m_T - \beta m_P l \cos^2 \theta} \left[g \sin \theta (m_T - \beta m_P l \cos^2 \theta) - \cos \theta \cdot (u(t) - \epsilon \dot{x} + m_P l \dot{\theta}^2 \sin \theta - m_P l g \beta \sin \theta \cos \theta) \right] \\ &= \gamma \left[m_T g \sin \theta - \cancel{\beta m_P g l \sin \theta \cos^2 \theta} \right. \\ &\quad \left. - u(t) \cos \theta + \epsilon \dot{x} \cos \theta - m_P l \dot{\theta}^2 \sin \theta \cos \theta + \cancel{\beta m_P g l \sin \theta \cos^2 \theta} \right]\end{aligned}$$

$$\therefore \ddot{\theta} = \gamma \left[m_T g \sin \theta + (\epsilon \dot{x} - u(t)) \cos \theta - m_P l \dot{\theta}^2 \sin \theta \cos \theta \right] \quad - (5)$$

Thus, we have

$$\ddot{x} = \frac{1}{d} (u(t) - \epsilon \dot{x} + m_P l \sin \theta (\dot{\theta}^2 - g \beta \cos \theta))$$

$$\ddot{\theta} = \frac{\beta}{d} (m_T g \sin \theta + (\epsilon \dot{x} - u(t)) \cos \theta - m_P l \dot{\theta}^2 \sin \theta \cos \theta)$$

With,

$$\beta = \frac{m_P l}{m_P l^2 + I_C^P}$$

$$d = m_T - \beta m_P l \cos^2 \theta$$

$$= (m_C + m_P) - \beta m_P l \cos^2 \theta$$

$$= m_C + m_P (1 - \beta l \cos^2 \theta)$$

We convert this to a system of first order DEs by letting

$$\begin{aligned} X_1 &= \theta & X_3 &= \dot{\theta} \\ X_2 &= x & X_4 &= \dot{x} \end{aligned}$$

then,

$$\begin{cases} \dot{X}_1 = X_3 \\ \dot{X}_2 = X_4 \\ \dot{X}_3 = \frac{\beta}{m_T - \beta m_P \cos^2 X_1} \left[m_T g \sin X_1 + (\epsilon X_4 - u(t)) \cos X_1 - m_P \beta X_3^2 \sin X_1 \cos X_1 \right] \\ \dot{X}_4 = \frac{1}{m_T - \beta m_P \cos^2 X_1} \left[u(t) - \epsilon X_4 + m_P \beta \sin X_1 (X_3^2 - g \beta \cos X_1) \right] \end{cases}$$

ie. $\dot{\vec{x}} = f(\vec{x})$. Now we solve for the eq. points by setting

$0 = f(\vec{x})$, so

$$\dot{X}_1 = 0 = \bar{X}_3 \Rightarrow \bar{X}_3 = 0 \Rightarrow \bar{X}_1 = \theta = \text{const.}$$

$$\dot{X}_2 = 0 = \bar{X}_4 \Rightarrow \bar{X}_4 = 0 \Rightarrow \bar{X}_2 = x = \text{const.}$$

$$\dot{X}_3 = 0 = m_T g \sin(\bar{X}_1) - \cancel{u(t)} \cos(\bar{X}_1)$$

$$\Rightarrow \bar{X}_1 = \arcsin(0) = n\pi, \quad n \in \mathbb{Z}_{\geq 0}$$

$$\dot{X}_4 = 0 = 0$$

$$\therefore \vec{X}_{eq} = \begin{pmatrix} \bar{X}_1 \\ \bar{X}_2 \\ \bar{X}_3 \\ \bar{X}_4 \end{pmatrix} = \begin{pmatrix} n\pi \\ c \\ 0 \\ 0 \end{pmatrix} \quad \text{for } \begin{matrix} n \in \mathbb{Z}_{\geq 0} \\ c \in \mathbb{R} \end{matrix}$$

Refer to pg. ⑤. Computing the Jacobian yields

$$Df(\vec{x}_{eq}) = \begin{pmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ a_1 & a_2 & a_3 & a_4 \\ b_1 & b_2 & b_3 & b_4 \end{pmatrix}.$$

Note that the only thing that changed between systems is

	OLD		NEW
$\ddot{\theta} :$	$(u(z) - \epsilon \dot{x})$	\longrightarrow	$(\epsilon \dot{x} - u(z))$
$\ddot{x} :$	$(g\beta \cos\theta - \dot{\theta}^2)$	\longrightarrow	$(\dot{\theta}^2 - g\beta \cos\theta)$

THE ONLY ELEMENTS of $Df(\vec{x}_{eq})$ that CHANGES ARE ... (I'm ABOUT TO CHECK).

Using WOLFRAM,

$$\left. \frac{\partial f_3}{\partial x_1} \right|_{\vec{x}_{eq}} = \beta \left(\frac{m_T g \cos(\bar{x}_1)}{m_T - \beta m_P l \cos(\bar{x}_1)} \right) = \frac{\beta m_T g \cos(\bar{x}_1)}{m_T - \beta m_P l} \quad \checkmark$$

$$\frac{\partial f_3}{\partial x_2}, \frac{\partial f_3}{\partial x_3} \quad \checkmark$$

$$\left. \frac{\partial f_3}{\partial x_4} \right|_{\vec{x}_{eq}} = \frac{\beta}{d(\bar{x}_1)} \left(\epsilon \cos(\bar{x}_1) \right) = \frac{\beta \epsilon \cos(\bar{x}_1)}{m_T - \beta m_P l}$$

$$\left. \frac{\partial f_3}{\partial u} \right|_{eq} = \frac{-\beta \cos(\bar{x}_1)}{m_T - \beta m_P l}$$

and,

$$a = b = d = 0$$

$$\left. \frac{\partial u}{\partial x_1} \right|_{eq} = \frac{-m_p l g \beta \cos^2(\bar{x}_1)}{m_T - \beta m_p l} = -\frac{m_p l g \beta}{m_T - \beta m_p l} = \frac{-g}{\frac{m_T + m_p}{\beta m_p l} - 1}$$

$$\left. \frac{\partial u}{\partial x_2} \right|_{eq} \checkmark, \quad \left. \frac{\partial u}{\partial x_3} \right|_{eq} \checkmark, \quad \left. \frac{\partial u}{\partial u} \right|_{eq} \checkmark$$

$$\left. \frac{\partial u}{\partial x_4} \right|_{eq} = \frac{-\varepsilon}{d(\bar{x}_1)} = \frac{-\varepsilon}{m_T - \beta m_p l} \checkmark$$

* Coefficients that changed in STATE SPACE MODEL:

$$A_{41}, A_{34}, B_{31}$$

$$A = \begin{pmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ \frac{\beta m_T g \cos(\bar{x}_1)}{m_T - \beta m_p l} & 0 & 0 & \frac{\beta \varepsilon \cos(\bar{x}_1)}{m_T - \beta m_p l} \\ -\frac{m_p l g \beta}{m_T - \beta m_p l} & 0 & 0 & \frac{-\varepsilon}{m_T - \beta m_p l} \end{pmatrix}$$

$$B = \begin{pmatrix} 0 \\ 0 \\ \frac{-\beta \cos(\bar{x}_1)}{m_T - \beta m_p l} \\ \frac{1}{m_T - \beta m_p l} \end{pmatrix}$$

$$C = \begin{pmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{pmatrix}$$

$$D = 0$$

PYTHON PID CONTROL

$$\epsilon_{i-1} =$$

$$\theta_{i-1} =$$

$$ICs = [0, 0, 0, 0]$$

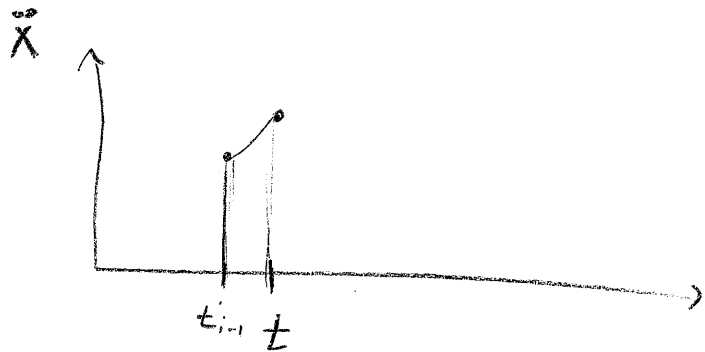
for t in T :

CALCULATE $\theta, x, \dot{\theta}, \dot{x}$

$$\text{error} = \theta_i - \theta_{i-1}$$

$$\text{Voltage} = \text{PID}(K_p, K_i, K_d, N, \epsilon_i, \epsilon_{i-1})$$

$$\text{SYSTEM_CONTROL_LAW} = \text{VOLTAGE}$$



LQR CONTROL

→ OK SCRATCH PID. Reason's why we're going w/ LQR:

- ① LQR WILL WORK BETTER (MORE ROBUST & BETTER TRACKING)
- ② I CAN ONLY CHOOSE ONE (ASSUMINGLY)
- ③ LQR WILL EASILY EXTEND TO LQG, WHICH WILL BE REALLY GOOD EXPERIENCE IF I IMPLEMENT A KALMAN FILTER.
- ④ SINCE LQR MINIMIZES CONTROL EFFORT, THIS WILL BE GOOD CONSIDERING I BOUGHT A PRETTY beefy MOTOR

MATH

- ① CHOOSE Q, R
- ② SOLVE RICCATI EQ FOR P
$$PA + A^T P + Q - PBR^{-1}B^T P = 0$$
- ③ COMPUTE FEEDBACK GAIN MATRIX
$$K = R^{-1}B^T P$$
- ④ SIMULATE the CLOSED LOOP RESPONSE
$$\dot{x} = (A - BK)x = (A - BR^{-1}B^T P)x$$

→ TO DETERMINE HOW good the RESPONSE IS, LOOK AT SINGULAR VALUES

① Choosing Q, R

Let $q = (q_1 \ q_2 \ q_3 \ q_4)$, then $Q = qI$.

$$q_1 = \frac{1}{T_{s_1} x_{1, \max}^2} = \left(0.3s \cdot \left(\frac{\pi}{4}\right)^2\right)^{-1}$$

$$q_2 = \frac{1}{T_{s_2} x_{2, \max}^2} = \left(5s \cdot (0.19m)^2\right)^{-1}$$

$$q_3 = q_4 = 1.$$

Similarly for $R = \rho \vec{r} I$,

$$r_1 = \frac{1}{u_{1, \max}^2} = (18v)^{-2}$$

$$\rho = 1$$

So,

$$Q = \begin{pmatrix} q_1 & 0 & 0 & 0 \\ 0 & q_2 & 0 & 0 \\ 0 & 0 & q_3 & 0 \\ 0 & 0 & 0 & q_4 \end{pmatrix}, \quad R = (r_1)$$

RANDOM

NOTES

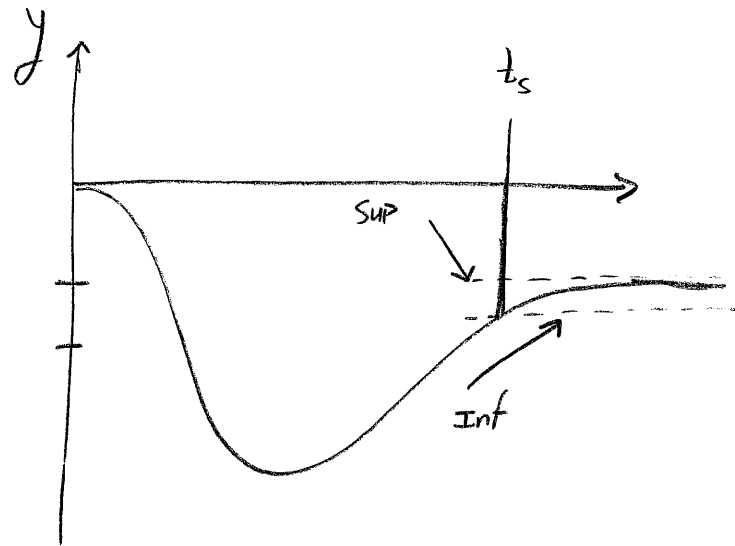
STEP RESPONSE PARAMETERS

① SETTLING TIME

for y_i from y_f to y_i

if $(y_i > \text{sup} \text{ or } y_i < \text{inf})$:

settling time = time_{i+1}



② RISE TIME

