

---

# ASEN 5044 HW 2 Script

## Table of Contents

Housekeeping .....	1
Problem 1c .....	1
Problem 4a .....	2
Problem 4b .....	6

By: Ian Faber

## Housekeeping

```
clc; clear; close all;
```

## Problem 1c

```
r0 = 6678; % km
k = 398600; % km^3/s^2
w0 = sqrt(k/(r0^3)); % rad/s

dt = 10; % sec

Abar = [
    0          1          0      0
    (3*k)/(r0^3) 0          0  2*sqrt(k/r0)
    0          0          0      1
    0          -2*sqrt(k/(r0^5)) 0  0
];

Bbar = [
    0 0
    1 0
    0 0
    0 1/r0
];

Cbar = [
    1 0 0 0
    0 0 1 0
];

Dbar = zeros(2,2);

Ahat = [
    Abar Bbar
    zeros(2,6)
];

matExp = expm(Ahat*dt);
```

```
F = matExp(1:4, 1:4)
G = matExp(1:4, 5:6)
H = Cbar
M = Dbar
```

$F =$

```
    1.0002    9.9998         0    772.5749
    0.0000    0.9999         0    154.5133
   -0.0000   -0.0000    1.0000     9.9991
   -0.0000   -0.0000         0     0.9997
```

$G =$

```
    49.9994    0.3856
     9.9998    0.1157
   -0.0001    0.0075
   -0.0000    0.0015
```

$H =$

```
    1    0    0    0
    0    0    1    0
```

$M =$

```
    0    0
    0    0
```

## Problem 4a

Time vector

```
T = (2*pi)/w0; % Circular orbit, r0 = a
dt = 10; % sec
t = 0:dt:T; % sec, assuming 90 minute orbit
```

% System ICs

```
r0 = r0; % km, see above
rDot0 = 0; % rad/s
theta0 = 0; % rad
thetaDot0 = w0; % rad/s, see above
```

```
x0 = [r0; rDot0; theta0; thetaDot0];
```

% Perturbation ICs

```
dr0 = 10; % km
drDot0 = -0.5; % km/s
```

```

dtheta0 = 0; % rad
dthetaDot0 = 2.5e-5; % rad/s

dx0 = [dr0; drDot0; dtheta0; dthetaDot0];

% Simulate systems
xNom = [];
xPerturb = [];
dxLast = dx0;
for kk = t
    xNom = [xNom, [x0(1); x0(2); x0(4)*kk + x0(3); x0(4)]]; % Nominal state
    dx = F*dxLast;
    xPerturb = [xPerturb, dx]; % Perturbation state
    dxLast = dx;
end

x = xNom + xPerturb; % Total state

% Plot!
figure(1) % Perturbation states
sgtitle("Problem 4a. Perturbation states vs. time")
subplot(4,1,1)
hold on; grid on
title("\Deltar vs. time")
plot(t, xPerturb(1,:))
xlabel("Time [sec]")
ylabel("\Delta r [km]")
subplot(4,1,2)
hold on; grid on
title("\DeltarDot vs. time")
plot(t, xPerturb(2,:))
xlabel("Time [sec]")
ylabel("\DeltarDot [km/s]")
subplot(4,1,3)
hold on; grid on
title("\Delta\theta vs. time")
plot(t, xPerturb(3,:))
xlabel("Time [sec]")
ylabel("\Delta\theta [rad]")
subplot(4,1,4)
hold on; grid on
title("\Delta\thetaDot vs. time")
plot(t, xPerturb(4,:))
xlabel("Time [sec]")
ylabel("\Delta\thetaDot [rad/s]")

figure(2) % Total state
sgtitle("Problem 4a. Total state vs. time")
subplot(4,1,1)
hold on; grid on
title("r vs. time")
plot(t, x(1,:))
xlabel("Time [sec]")

```

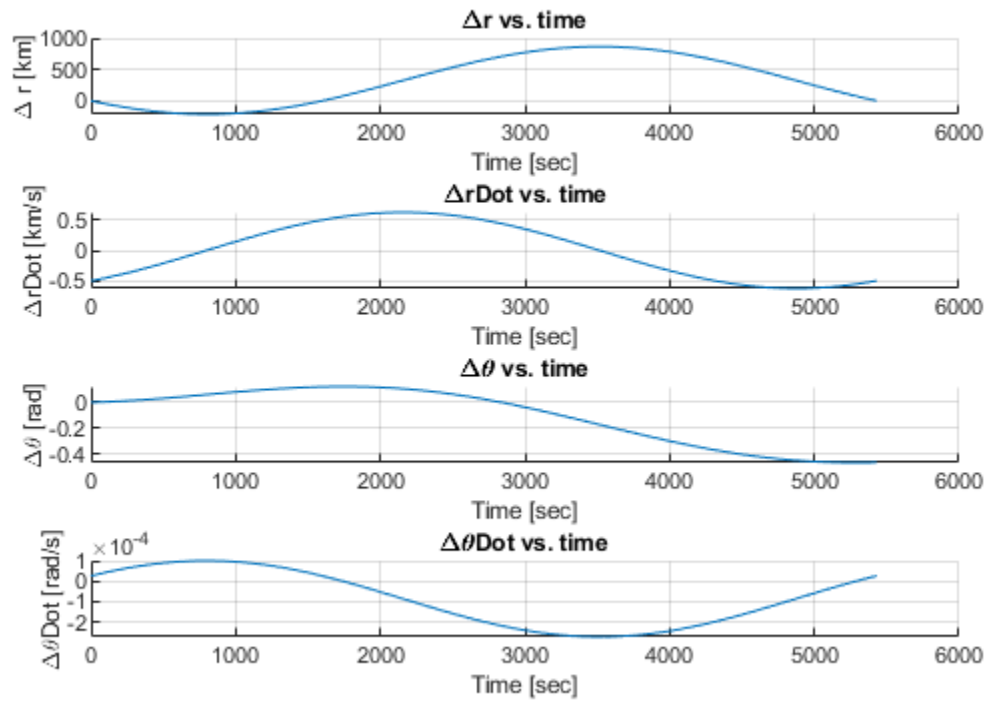
```

ylabel("r [km]")
subplot(4,1,2)
hold on; grid on
title("rDot vs. time")
plot(t, x(2,:))
xlabel("Time [sec]")
ylabel("rDot [km/s]")
subplot(4,1,3)
hold on; grid on
title("\theta vs. time")
plot(t, x(3,:))
xlabel("Time [sec]")
ylabel("\theta [rad]")
subplot(4,1,4)
hold on; grid on
title("\thetaDot vs. time")
plot(t, x(4,:))
xlabel("Time [sec]")
ylabel("\thetaDot [rad/s]")

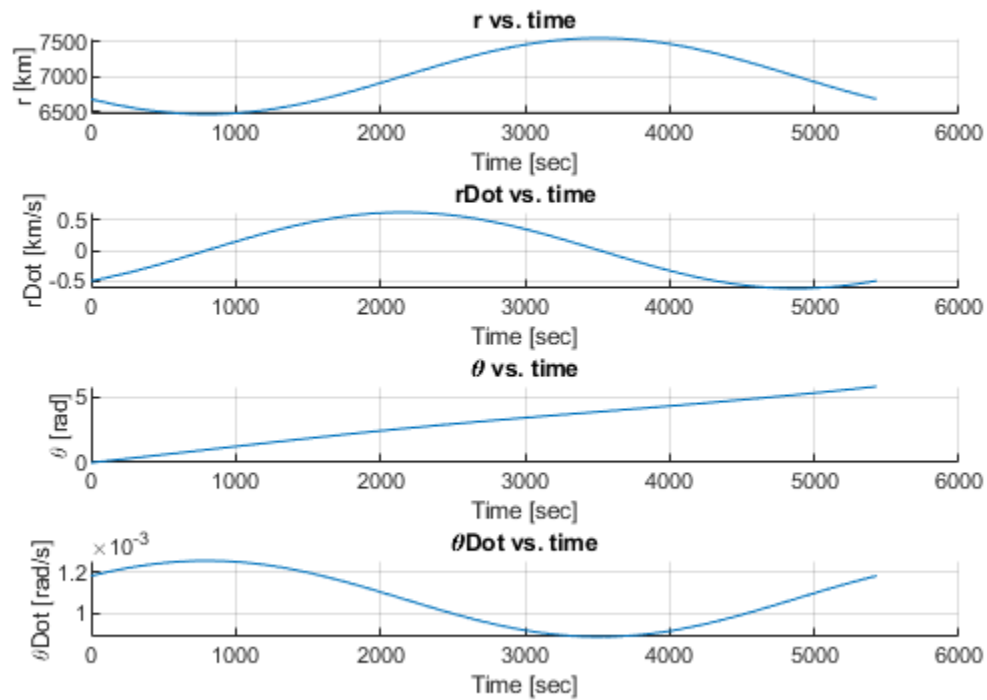
% figure(3) % Nominal state
% sgttitle("Nominal state vs. time")
% subplot(4,1,1)
% hold on; grid on
% title("r_{nom} vs. time")
% plot(t, xNom(1,:))
% xlabel("Time [sec]")
% ylabel("r_{nom} [km]")
% subplot(4,1,2)
% hold on; grid on
% title("rDot_{nom} vs. time")
% plot(t, xNom(2,:))
% xlabel("Time [sec]")
% ylabel("rDot_{nom} [km/s]")
% subplot(4,1,3)
% hold on; grid on
% title("\theta_{nom} vs. time")
% plot(t, xNom(3,:))
% xlabel("Time [sec]")
% ylabel("\theta_{nom} [rad]")
% subplot(4,1,4)
% hold on; grid on
% title("\thetaDot_{nom} vs. time")
% plot(t, xNom(4,:))
% ylim([0.9*w0, 1.1*w0])
% xlabel("Time [sec]")
% ylabel("\thetaDot_{nom} [rad/s]")

```

### Problem 4a. Perturbation states vs. time



### Problem 4a. Total state vs. time



## Problem 4b

```
x0 = x0 + dx0; % Implement initial perturbations into initial condition

opt = odeset('RelTol', 1e-12, 'AbsTol', 1e-12);
[t, X] = ode45(@ (t,X) orbitEOM(t,X,k), t, x0, opt);

XPerturb = (X' - xNom)';

figure(4) % Perturbation states
sgtitle("Problem 4b. Perturbation states vs. time")
subplot(4,1,1)
hold on; grid on
title("\Deltar vs. time")
linear = plot(t, XPerturb(1,:), 'b-');
nonlinear = plot(t, XPerturb(:,1), 'r--');
xlabel("Time [sec]")
ylabel("\Deltar [km]")
legend([linear, nonlinear], ["Linear", "Non-Linear"], 'location', 'best')
subplot(4,1,2)
hold on; grid on
title("\DeltarDot vs. time")
plot(t, XPerturb(2,:), 'b-')
plot(t, XPerturb(:,2), 'r--')
xlabel("Time [sec]")
ylabel("\DeltarDot [km/s]")
subplot(4,1,3)
hold on; grid on
title("\Delta\theta vs. time")
plot(t, XPerturb(3,:), 'b-')
plot(t, XPerturb(:,3), 'r--')
xlabel("Time [sec]")
ylabel("\Delta\theta [rad]")
subplot(4,1,4)
hold on; grid on
title("\Delta\thetaDot vs. time")
plot(t, XPerturb(4,:), 'b-')
plot(t, XPerturb(:,4), 'r--')
xlabel("Time [sec]")
ylabel("\Delta\thetaDot [rad/s]")

figure(5) % Total state
sgtitle("Problem 4b. Total state vs. time")
subplot(4,1,1)
hold on; grid on
title("r vs. time")
linear = plot(t, x(1,:), 'b-');
nonlinear = plot(t, X(:,1), 'r--');
xlabel("Time [sec]")
ylabel("r [km]")
legend([linear, nonlinear], ["Linear", "Non-Linear"], 'location', 'best')
subplot(4,1,2)
hold on; grid on
```

```

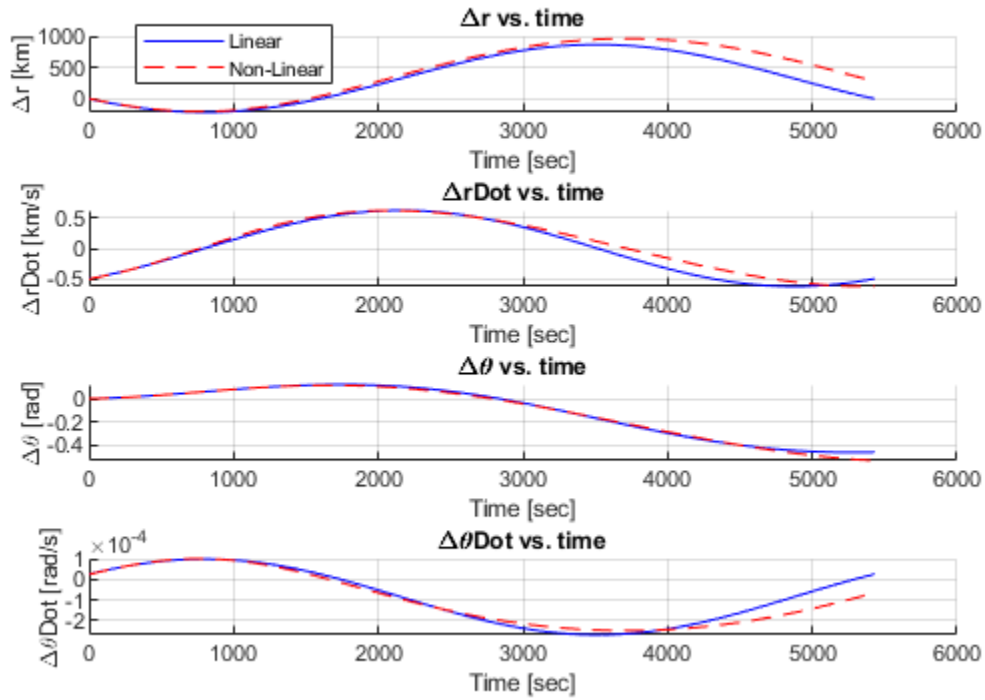
title("rDot vs. time")
plot(t, x(2,:), 'b-')
plot(t, X(:,2), 'r--')
xlabel("Time [sec]")
ylabel("rDot [km/s]")
subplot(4,1,3)
hold on; grid on
title("\theta vs. time")
plot(t, x(3,:), 'b-')
plot(t, X(:,3), 'r--')
xlabel("Time [sec]")
ylabel("\theta [rad]")
subplot(4,1,4)
hold on; grid on
title("\thetaDot vs. time")
plot(t, x(4,:), 'b-')
plot(t, X(:,4), 'r--')
xlabel("Time [sec]")
ylabel("\thetaDot [rad/s]")

function dX = orbitEOM(t, X, k)
    r = X(1);
    rDot = X(2);
    theta = X(3);
    thetaDot = X(4);

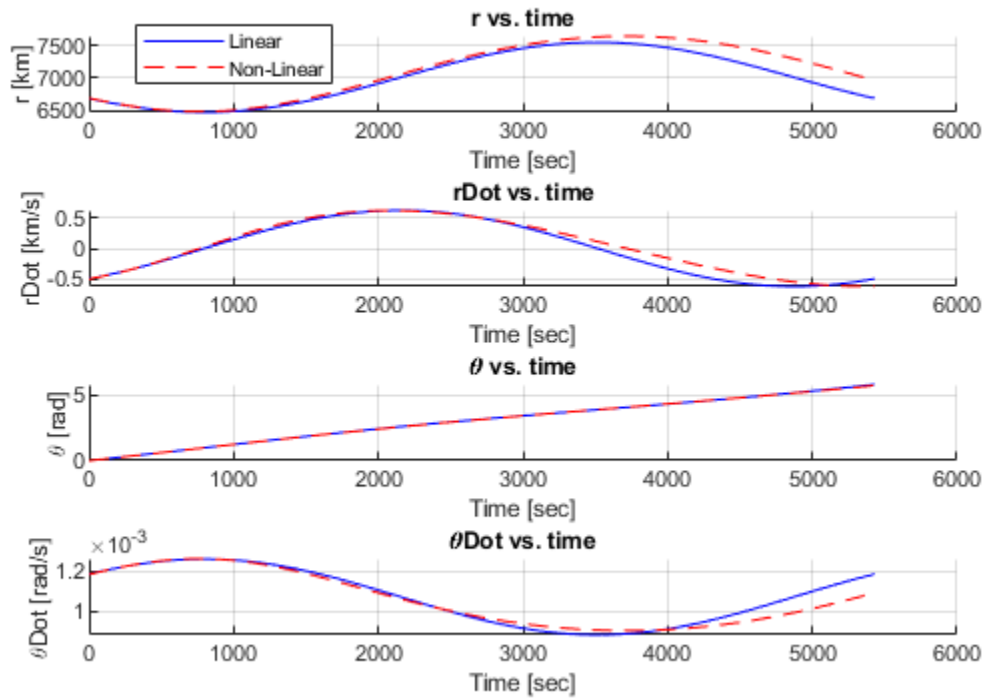
    dX = [
        rDot
        r*(thetaDot)^2 - k/(r^2)
        thetaDot
        (-2*thetaDot*rDot)/r
    ];
end

```

### Problem 4b. Perturbation states vs. time



### Problem 4b. Total state vs. time





*Published with MATLAB® R2023b*