# **ASEN 5050 HW 9 Main Script**

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## Housekeeping

```
clc; clear; close all;
```

### **Setup**

```
muEarth = 3.986004415e5; % km^3/s^2
muMars = 4.305e4; % km^3/s^2

rEarth = 6378.1363; % km
rMars = 3397.2; % km

Prot_Mars = 1.02595675; % days

r0 = 10000; % km
```

#### **Relative Motion matrices**

```
Phi rr = 0(t,n) [
                     4 - 3*\cos(n*t),
                                             Ο,
                                                      0;
                     6*(sin(n*t) - n*t),
                                             1,
                                                      0;
                                                      cos(n*t)
                ];
Phi rv = 0(t,n) [
                     (1/n) * sin(n*t),
                                            (2/n)*(1-cos(n*t)),
                                                                       0;
                     (2/n)*(\cos(n*t) - 1), (4/n)*\sin(n*t) - 3*t,
                                                                       0;
                                                                       (1/
n) *sin(n*t)
                ];
Phi vr = 0(t,n) [
                     3*n*sin(n*t),
                                             Ο,
                                                      0;
                     6*(n*cos(n*t)-n),
                                            Ο,
                     0,
                                              Ο,
                                                      -n*sin(n*t)
                ];
```

### **Problem 1**

```
fprintf("--- Problem 1 ---\n")
x0 = 5; % m
xDot0 = 0; % m/s
z0 = 0 % m \rightarrow Same orbit plane
zDot0 = 0 % m/s \rightarrow Same orbit plane
yMax = 15; % m
n = sqrt(muEarth/(r0^3));
yDot0 = -2*n*x0 % -> Drift terms suppressed
y0 = yMax - 2*x0
--- Problem 1 ---
z0 =
     0
zDot0 =
     0
yDot0 =
   -0.0063
y0 =
     5
```

### **Problem 2**

```
fprintf("--- Problem 2 ---\n")
format shortE
r0_vec = [2;2;0]; % m
v0minus = [-0.03; 0.01; 0.05]; % m/s
r1_vec = [-2;2;0]; % m

T = 2*pi/n; % sec
t1 = T/2;
```

```
v0plus = (Phi rv(t1,n)^-1)*(r1 vec - Phi rr(t1,n)*r0 vec)
dv1 = norm(v0plus - v0minus)
v1minus = Phi vr(t1,n)*r0 vec + Phi vv(t1,n)*v0plus
dv2 = norm(zeros(size(v1minus)) - v1minus)
    % Plot Trajectory
t = 0:t1;
markerSize = 25;
for k = 1: length(t)
    r(:,k) = Phi rr(t(k),n)*r0 vec + Phi rv(t(k),n)*v0plus;
    v(:,k) = Phi vr(t(k),n)*r0 vec + Phi vv(t(k),n)*v0plus;
end
figure
hold on; grid on; grid minor;
title("Problem 2 CubeSat Trajectory")
traj = plot3(r(1,:), r(2,:), r(3,:));
plot3(r0 vec(1), r0 vec(2), r0 vec(3), 'g.', 'MarkerSize', markerSize)
plot3(0, 0, 0, 'k.', 'MarkerSize', markerSize)
plot3(r(1,end), r(2,end), r(3,end), 'r.', 'MarkerSize', markerSize)
xlabel("Radial [m]"); ylabel("Along-track [m]"); zlabel("Cross-track [m]")
xlim([-5 \ 2.5]); ylim([-2.5 \ 5]); zlim([-2 \ 2]); view([-90 \ 90])
legend("Trajectory", "Start", "Primary s/c", "End")
labels = ["Xdot", "Ydot", "Zdot"];
values = [round(v(1,:),7); round(v(2,:),7); round(v(3,:),7)];
for k = 1:length(labels)
    extraRows(k) = dataTipTextRow(labels(k), values(k,:));
    traj.DataTipTemplate.DataTipRows(end+length(extraRows)) = extraRows(k);
end
points = [1, size(r,2)];
for k = 1:length(points)
    datatip(traj, r(1,points(k)), r(2,points(k)), r(3,points(k)),
'location', 'southwest');
end
format default
--- Problem 2 ---
v0plus =
  8.6736e-19
  -2.5254e-03
            Ω
dv1 =
```

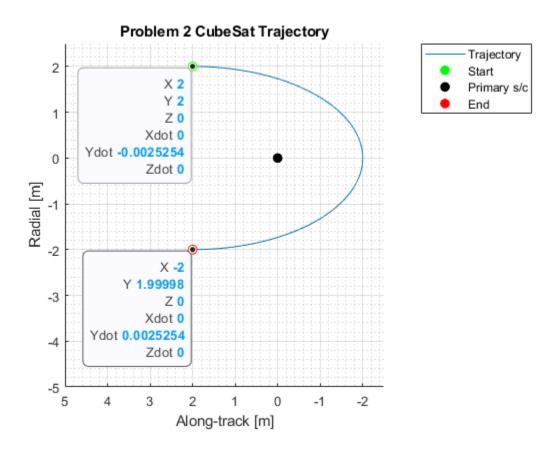
5.9640e-02

v1minus =

-1.0220e-18 2.5254e-03 0

dv2 =

2.5254e-03



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