

In [1]:

```
import numpy as np
from scipy.integrate import odeint
from scipy import signal
import matplotlib.pyplot as pl
%matplotlib inline
```

Machine parameters for PMSM

We try to create parameter set for 3 types of PMSM

1. IPMSM $l_d < l_q$
2. IPMSM $l_d > l_q$
3. EPMSM $l_d = l_q$

In [2]:

```
#Defining Machine parameters

mach_ramu = {"rs": 0.1729, "lq": 0.6986, "ld": 0.4347, "tmech":50.5}
mach_servo = {"rs": 0.0426, "ld": 2.252+0.078, "lq": 2.252+0.078, "tmech":100.0}
mach_servodq = {"rs": 0.0426, "ld": 2.252+0.078, "lq": 2.252+0.0758, "tmech":100.0}
```

Getting machine parameters

In [3]:

```

def mach_parapm(dict):
    """Takes in the dictionary containing machine parameters (normalized)
       and returns various parameters that are used in space vector equations

    Takes arguement: machine dictionary
    """
    rs = dict["rs"]
    lq = dict['lq']
    ld = dict['ld']
    tsq=lq/rs
    tsd=ld/rs
    tmech = dict['tmech']
    psirm = 0.9
    #    print("lh = {0:1.3f}".format(lh))
    return rs, lq, ld, tmech, psirm

def mach_parapmSym(dict):
    """Takes in the dictionary containing machine parameters (normalized)
       and returns various parameters that are used in space vector equations

    Takes arguement: machine dictionary
    """
    rs = dict["rs"]
    lq = dict['lq']
    ld = dict['ld']
    ls = dict['ls']
    tsq=lq/rs
    tsd=ls/rs
    tmech = dict['tmech']
    psirm = 0.95
    #    print("lh = {0:1.3f}".format(lh))
    return rs, ls, tmech, psirm

#Test
mach_parapmSym(mach_servo)

```

Out[3]:

(0.0426, 2.3299999999999996, 100.0, 0.95)

Define PMSM model in stator coordinates with $L_d = L_q$

This models is for machine in stator coordinates. For simplicity we will assume $L_d = L_q$, which is the case with most surface-mounted permanent magnets on rotor.

$$\vec{v}_s = r_s \vec{i}_s + \frac{d\vec{\psi}_s}{d\tau}$$

$$\vec{\psi}_s = l_s \vec{i}_s + \vec{\psi}_{r,m}$$

Where $\psi_{r,m}$ is the rotor flux coupling with the stator winding. Hence, we get

$$\vec{v}_s = r_s \vec{i}_s + l_s \frac{d\vec{i}_s}{d\tau} + \frac{d\vec{\psi}_{r,m}}{d\tau}$$

$$\vec{\psi}_{r,m} = |\psi_{r,m}| e^{j\delta} = \psi_{r,m\alpha} + j\psi_{r,m\beta}$$

$$\vec{v}_s = r_s \vec{i}_s + l_s \frac{d\vec{i}_s}{d\tau} + j\omega_s \vec{\psi}_{r,m}$$

$$v_{s\alpha} = r_s i_{s\alpha} + l_s \frac{di_{s\alpha}}{d\tau} - \omega_s \psi_{r,m\beta}$$

$$v_{s\beta} = r_s i_{s\beta} + l_s \frac{di_{s\beta}}{d\tau} + \omega_s \psi_{r,m\alpha}$$

In [4]:

Dynamic model in stator coordinates for symmetrical PMSM rotor

#PMSM machine 4x4 dynamic model

```
def PMSM_dynstepstatorSym(X, t, params):
    """Defines a function to calculate the derivatives for a 4x4 PMS motor dynamics
    1. Uses state variables stator current $|\vec{i}|_s$ and rotor flux space vector $|\vec{\psi}|_r$ 
    2. Is non-linear and uses the state variable $\omega$ rotor angular velocity to describe
       rotor dynamics
    3. calls mach_im function to calculate the system equation coefficients
    4. mach_im will in turn call mach_para function.
    5. uses normalized time $2*\pi*50*t$ 
    6. This is to be used in a for loop for the time span
    """
    x0 = X[0] #isalpha
    x1 = X[1] #isbeta
    x2 = X[2] #ws
    x3 = X[3] #delta

    # wx = X[5]
    u1, u2, mL, rs, ls, tmech, psirm = params

    # rs, ls, tmech, psirm = mach_parapmSym(mach_servo)
    # rs, ls, tmech, psirm = mach_parapmSym(mach_ramu)
    psrma = psirm*np.cos(x3)
    psrmb = psirm*np.sin(x3)
    mdiff = (psrma*x1 - psrmb*x0) - mL
    # print(A[1][0], A[1][1], A[1][2], A[1][3], B[4][0])
    dx0dt = -(rs/ls)*x0 + psrmb*x2 + u1/ls
    dx1dt = -(rs/ls)*x1 - psrma*x2 + u2/ls
    dx2dt = mdiff/tmech
    dx3dt = x2
    # dwxdt = rate
    return [dx0dt, dx1dt, dx2dt, dx3dt]
```

In [5]:

```
def rotorangle(X, t, params):
    del0 = X
    w = params
    ddel0dt = w
    return ddel0dt
```

In [6]:

```
def PICon(xe, y, dt, Kparams):
    """Pass on xe = [error[k], error[k-1]]
       Pass on y = [y[k-1]]
       Pass on Parameters = [kp, Ti]
    """
    kp, Ti = Kparams
    xeo = xe[1]
    xen = xe[0]
    # print(xeo, xen, y, kp, Ti)
    y1 = y
    y2 = y1 + kp*(xen - xeo) + (kp/Ti)*(xen)
    return y2

def PIConwithLim(xe, y, dt, Kparams):
    """Pass on xe = [error[k], error[k-1]]
       Pass on y = [y[k-1]]
       Pass on Parameters = [kp, Ti]
    """
    kp, Ti, uplim, dwnlim = Kparams
    xeo = xe[1]
    xen = xe[0]
    # print(xeo, xen, y, kp, Ti)
    # y2 = y1 + kp*(xen - xeo) + (kp/Ti)*(xen)
    # print(xeo, xen, y, kp, Ti)
    y1 = y
    yx = y1 + kp*(xen - xeo) + (kp/Ti)*(xen)
    if (yx > uplim):
        yx = uplim
    elif (yx < dwnlim):
        yx = dwnlim
    y2 = yx
    return y2
```

In [7]:

```
#Preparing simulation with current control
# We will use for loop

# ODE solver parameters
abserr = 1.0e-8
relerr = 1.0e-3

tend = 20*2*np.pi
tstart = 0.0
delta_t = 0.01
n = 100000
n2 = int(100*2*np.pi/0.01)
ws = -0.5
mL = 0.0
Tta = np.arange(tstart, tend, delta_t)
# Tta = np.linspace(tstart, tend, n2)
#delta_t = Tta[1] - Tta[0]
usa = np.zeros(len(Tta))
usb = np.zeros(len(Tta))
sol1 = np.zeros((len(Tta), 4))
sol2 = np.zeros(len(Tta))
FCangle = np.zeros(len(Tta))
gamma = np.zeros(len(Tta))
# Field coordinate currents
cosdelta = np.zeros(len(Tta))
sindelta = np.zeros(len(Tta))
isa = np.zeros(len(Tta))
isb = np.zeros(len(Tta))
w = np.zeros(len(Tta))
gamma = np.zeros(len(Tta))
isd = np.zeros(len(Tta))
isq = np.zeros(len(Tta))
usd = np.zeros(len(Tta))
usq = np.zeros(len(Tta))
eid = np.zeros(len(Tta))
eq = np.zeros(len(Tta))

#Creating reference values for isd and isq
isdrefval = 0.01
isdref = np.zeros(len(Tta))
isqref = np.zeros(len(Tta))

isqref1 = 0.0
isqref2 = 0.9
isqref3 = -0.35
for ii in range(len(Tta)):
    isdref[ii] = isdrefval
    if (Tta[ii]>=18*2*np.pi):
        isqref[ii] = isqref3
    elif(Tta[ii]>=1*2*np.pi):
        isqref[ii]=isqref2
    else:
        isqref[ii]=isqref1

#  

rs, ls, tmech, psirm = mach_parapmSym(mach_servo)
rs, ls, tmech, psirm = mach_parapmSym(mach_servo)
```

```

psird = np.zeros(len(Tta))
psirq = np.zeros(len(Tta))

me = np.zeros(len(Tta))
x0 = [0.0, 0.0, ws, 0]

kp = 5.5
Ti = 5.0e1

Kparams = [kp, Ti]

#isdref = 0.26
#isqref = 0.75
#The intial value error will never get corrected in
#rotor angle
y0 = 0
for ii in range(len(Tta)):
    #Start controller after first step
    if ii>=0:
        eid[ii] = isdref[ii-1] - isd[ii-1]
        PIed = [eid[ii], eid[ii-1]]
        usd[ii] = PICon(PIed, usd[ii-1], delta_t, Kparams)
        eqref[ii] = isqref[ii-1] - isq[ii-1]
        PEq = [eqref[ii], eqref[ii-1]]
        usq[ii] = PICon(PEq, usq[ii-1], delta_t, Kparams)
    #    usd[ii] = 0
    #    usq[ii] = .0001
        if ws>=1.0:
            a = 1.0
        else:
            a = ws
    #    usa[ii] = a*np.cos(ws*Tta[ii])
    #    usb[ii] = a*np.sin(ws*Tta[ii])
        usa[ii] = usd[ii-1]*cosdelta[ii-1] - usq[ii-1]*sindelta[ii-1]
        usb[ii] = usq[ii-1]*cosdelta[ii-1] + usd[ii-1]*sindelta[ii-1]
    #    me[ii] = kr*(sol1[ii-1][2]*sol1[ii-1][1] - sol1[ii-1][3]*sol1[ii-1][0]) - mL
        params = [usa[ii], usb[ii], mL, rs, ls, tmech, psirm]
        sol1a = odeint(PMSM_dynstepstatorSym, x0, [0, delta_t], args = (params,), atol = abserr, rtol= relerr)
        sol1[ii]= sol1a[-1]
    #    display(sol1a[-1])
        x0 = sol1a[-1]
        params2 = sol1[ii][2] # w
        sol2a = odeint(rotorangle, y0, [0, delta_t], args = (params2,), atol = abserr, rtol= relerr)
        sol2[ii] = sol2a[-1]
    #Using the internal state of the machine
    #for internal model use sol1[ii][3]
    #Change this to resolver model
    #Resolver model use sol2[ii]
    FCangle[ii] = sol2[ii] #sol1[ii][3]
    cosdelta[ii] = np.cos(FCangle[ii])
    sindelta[ii] = np.sin(FCangle[ii])
    #Coordinate transformation
    isa[ii] = sol1[ii][0]
    isb[ii] = sol1[ii][1]
    w[ii] = sol1[ii][2]
    gamma[ii] = sol1[ii][3]
    #Convert stator coordinate current to field coordinates
    isd[ii] = sol1[ii][0]*cosdelta[ii] + sol1[ii][1]*sindelta[ii]
    isq[ii] = sol1[ii][1]*cosdelta[ii] - sol1[ii][0]*sindelta[ii]
    #Rotor flux in field coordinates Using estimator output as in practice
    #actual flux will not be available for measurement

```

```
psird[ii] = 1.0
y0 = sol2a[-1]
psird[ii] = psirm

#Do not use these assignments
# isa = sol1[:, 0]
# isb = sol1[:, 1]
# w = sol1[:, 2]
# gamma = sol1[:, 3]

me = psirm*isq
```

In []:

In [8]:

```

dirfig = "C:/Users/ashwi/Documents/tex/lec/drives/figures/"
pl.figure(501, figsize = (6, 6))
pl.rc('font', size = 16)
pl.subplot(2, 1, 1)
pl.plot(Tta, isa, 'purple', Tta, isb, 'tomato', lw=2)
# pl.plot(Tta, isd, 'tomato', Tta, isq, 'crimson', lw =2)
pl.xlim(0, tend)
pl.ylabel(r'$i_{\alpha}$, $i_{\beta}$')
pl.xticks(np.linspace(0, tend, 5))
ax1= pl.subplot(2, 1, 2)
ax1.plot(Tta, me, 'crimson', lw =3)
# pl.plot(Tta,w, 'peru', lw =2)
# pl.plot(Tta,gamma, 'navy')
ax1.set_ylabel(r'$m_e$')
ax1.set_xlabel(r'$\omega_t$')
ax2 = ax1.twinx()
ax2.plot(Tta,w, 'peru', lw =3)
ax2.set_yticks(np.linspace(-0.5, 0.5, 5))
ax2.set_ylim(-0.5, 0.5)
ax1.set_yticks(np.linspace(-1.2, 1.2, 7))
ax1.set_ylim(-1.2, 1.2)
ax2.set_ylabel(r'$\omega_s$')
ax2.axhline(0, c= 'k')
ax1.set_xlim(0, tend)
ax1.set_xticks(np.linspace(0, tend, 5))
# pl.savefig(dirfig + "PMSMSpeedConcurrentsvst.pdf", bbox_inches = 'tight', transparent = True)

pl.figure(521, figsize = (6, 6))
pl.rc('font', size = 16)
pl.subplot(2, 1, 1)
pl.plot(Tta, isd, 'purple', lw =3, label = '$i_{sd}$')
pl.plot(Tta, isq , 'crimson' ,lw =3, label = '$i_{sq}$')
pl.xlim(0, tend)
pl.ylabel(r'$i_{sd}$, $i_{sq}$')
pl.xticks(np.linspace(0, tend, 5))
pl.yticks(np.linspace(-1.0, 1.0, 5))
pl.rcParams['legend.fontsize']=12
pl.legend(loc='lower left')
# pl.subplot(2, 1, 2)
# pl.plot(Tta, psra, 'tomato', lw =2)
# pl.plot(Tta,psrb, 'peru', lw =2)
# pl.ylabel(r'$\psi_r, \alpha$, $\psi_r, \beta$')
pl.xlabel(r'$\omega_t$')
# pl.xlim(0, tend)
# pl.xticks(np.linspace(0, tend, 5))
# pl.yticks(np.linspace(-1.0, 1.0, 5))
# pl.savefig(dirfig + "PMSMCurrentconsvst.pdf", bbox_inches = 'tight', transparent = True)

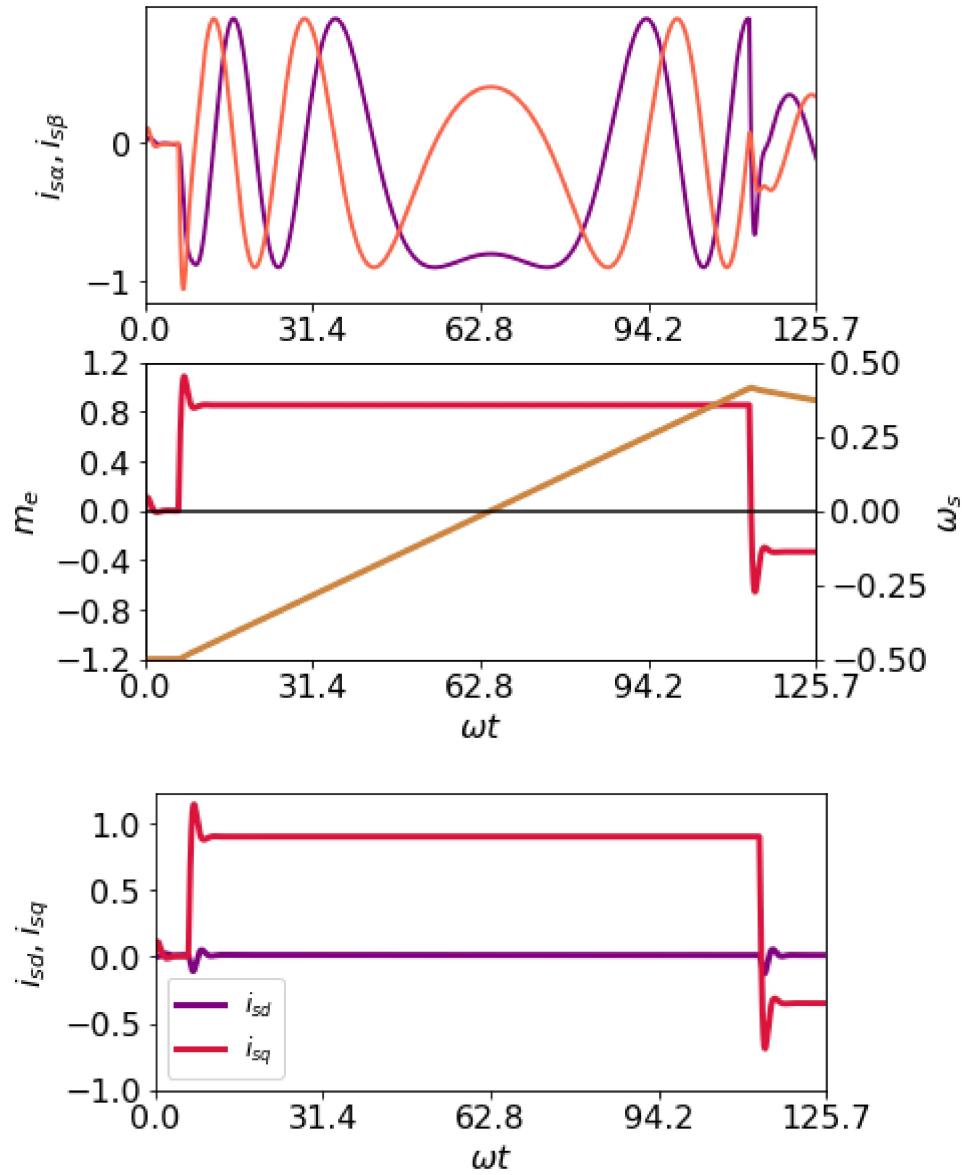
ui = np.where(Tta == 60)
index1 = ui[0][0]
pl.figure(523, figsize = (4, 4))
pl.rc('font', size = 16)
pl.plot(isd,isq,'crimson', lw =2, label = 'estimate')
# pl.plot(psra, psrb, 'blue', lw =2 , label = "actual")
pl.arrow(0,0,psird[-1],psirq[-1], fc = 'magenta', ec = 'magenta', head_width = 0.05 \
         , length_includes_head = True, lw =3)
pl.arrow(0,0,isd[-1],isq[-1], fc = 'crimson', ec = 'crimson', head_width = 0.05 \
         , length_includes_head = True, lw =3)
pl.arrow(0,0,usd[-1],usq[-1], fc = 'navy', ec = 'navy', head_width = 0.05 \
         , length_includes_head = True, lw =3)

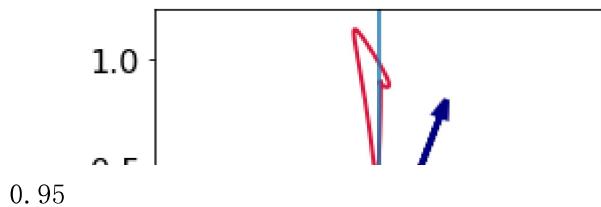
```

```

    , length_includes_head = True, lw = 3)
pl.arrow(0, 0, usd[index1], usq[index1], fc = 'blue', ec = 'blue', head_width = 0.05\
    , length_includes_head = True, lw = 3)
pl.axhline(0)
pl.axvline(0)
pl.xlim(-1.0, 1.0)
pl.xlabel('d-axis')
pl.ylabel('q-axis')
# pl.axis('equal')
# pl.savefig(dirfig + "PMSMXYwCC.pdf", bbox_inches = 'tight', transparent = True)
pl.show()
print(psirm)

```





Model in stator coordinates, asymmetrical

In []:

PMSM dynamic equations

we will define the parameters in d-q coordinates

$$\begin{aligned} \frac{di_{sd}}{d\tau} &= -\frac{r_s}{l_d}i_{sd} + \omega_s \frac{l_q}{l_d}i_{sq} + \frac{v_{sd}}{l_d} \\ \frac{di_{sq}}{d\tau} &= -\frac{r_s}{l_q}i_{sq} - \omega_s \frac{l_d}{l_q}i_{sd} - \omega_s \frac{\psi_{r,m}}{l_q} + \frac{v_{sq}}{l_q} \\ m_e &= \psi_s \times \vec{i}_s &= \psi_{r,m}i_{sq} + (l_d - l_q)i_{sd}i_{sq} \\ \frac{d\omega_s}{d\tau} &= m_e - m_L \end{aligned}$$

In [9]:

```
def PMSM_dynstepdq(X, t, params):
    """Defines a function to calculate the derivatives for a 2x2 PMS motor dynamics
    1. Uses state variables stator current $|\vec{i}_s|$ id d-q coordinates.
    2. Is non-linear and uses the state variable $\omega$ rotor angular velocity to describe
       rotor dynamics. rotor angular velocity will be go from outside as parameter
    3. calls mach_im function to calculate the system equation coefficients
    4. mach_im will in turn call mach_para function.
    5. uses normalized time $2*\pi*50*t$
    6. This is to be used in a for loop for the time span
    """
    x0 = X[0] #isd
    x1 = X[1] #isq
    # x2 = X[2] #ws
    # x3 = X[3] #rotor angle
    # wx = X[5]
    u1, u2, w, rs, ld, lq, tmech, psirm = params #needs state variable ws from outside
    #Convert voltage from stator coordinates into field coordinates
    # ud = u1*np.cos(x3) + u2*np.sin(x3)
    # uq = -u1*np.sin(x3) + u2*np.cos(x3)
    #
    ud = u1
    uq = u2
    # rs, ld, lq, tmech, psirm = mach_parapm(mach_ramu)
    # rs, ld, lq, tmech, psirm = mach_parapm(mach_servodq)
    # mdiff = psirm*x1 + (ld - lq)*x0*x1 - mL
    # print(A[1][0], A[1][1], A[1][2], A[1][3], B[4][0])
    dx0dt = -(rs/ld)*x0 + w*(lq*x1)/ld + ud/ld
    dx1dt = -w*(ld*x0)/lq - (rs/lq)*x1 - w*psirm/lq + uq/lq
    # dx2dt = mdiff/tmech
    # dx3dt = x2
    # dwdxdt = rate
    return [dx0dt, dx1dt]
```

In [10]:

```
#Rotor dynamics as mechanics
def rotor_dyndq(X, t, params):
    w = X[0]
    gamma = X[1] #rotor angle

    me, mL, tmech = params
    dwdt = (me - mL) / tmech
    dgammadt = w
    return [dwdt, dgammadt]
```

In [11]:

```
#Function for rotor angle integration
# d\delta/dt = w

def rotorangle(X, t, params):
    del0 = X
    w = params
    ddel0dt = w
    return ddel0dt
```

PMSM for loop dynamic simulation

In [12]:

```
#Preparing simulation for current control
# We will use for loop

# ODE solver parameters
abserr = 1.0e-8
relerr = 1.0e-3

tend = 20*2*np.pi
tstart = 0.0
delta_t = 0.01
n = 100000
n2 = int(100*2*np.pi/0.01)
ws = -0.5
mL = 0.0
Tta = np.arange(tstart, tend, delta_t)
# Tta = np.linspace(tstart, tend, n2)
#delta_t = Tta[1] - Tta[0]
usa = np.zeros(len(Tta))
usb = np.zeros(len(Tta))
sol1 = np.zeros((len(Tta), 2))
sol2 = np.zeros((len(Tta), 2))
FCangle = np.zeros(len(Tta))
gamma = np.zeros(len(Tta))

# Field coordinate currents
cosdelta = np.zeros(len(Tta))
sindelta = np.zeros(len(Tta))
isd = np.zeros(len(Tta))
isq = np.zeros(len(Tta))
isa = np.zeros(len(Tta))
isb = np.zeros(len(Tta))
usd = np.zeros(len(Tta))
usq = np.zeros(len(Tta))
eid = np.zeros(len(Tta))
eqi = np.zeros(len(Tta))
w = np.zeros(len(Tta))

rs, ld, lq, tmech, psirm = mach_parapm(mach_ramu)
# rs, ld, lq, tmech, psirm = mach_parapm(mach_servodq)

#Creating reference values for isd and isq
isdrefval = -0.1
isdref = np.zeros(len(Tta))
isqref = np.zeros(len(Tta))

isqref1 = 0.0
isqref2 = 0.8
isqref3 = -0.35
for ii in range(len(Tta)):
    isdref[ii] = 0.0
    if (Tta[ii]>=10*2*np.pi):
        isqref[ii] = isqref3
        isdref[ii] = -isdrefval
    elif (Tta[ii]>=1*2*np.pi):
        isqref[ii]=isqref2
        isdref[ii] = isdrefval
    else:
        isqref[ii]=isqref1
```

```

psird = np.zeros(len(Tta))
psirq = np.zeros(len(Tta))

me = np.zeros(len(Tta))
x0 = [0.0, 0.0]

kpd = ld*1
Tid = 25.0e1

kpq = lq*2
Tiq = 25.0e1

Kparamsd = [kpd, Tid]
Kparamsq = [kpq, Tiq]

#isdref = 0.26
#isqref = 0.75
#The intial value error will never get corrected in
#rotor angle
y0 = [ws, 0]

for ii in range(len(Tta)):
    #Start controller after first step
    if ii>=0:
        eid[ii] = isdref[ii-1] - isd[ii-1]
        PIed = [eid[ii], eid[ii-1]]
        usd[ii] = PIcon(PIed, usd[ii-1], delta_t, Kparamsd)
        eiq[ii] = isqref[ii-1] - isq[ii-1]
        PIeq = [eiq[ii], eiq[ii-1]]
        usq[ii] = PIcon(PIeq, usq[ii-1], delta_t, Kparamsq)
    #        usd[ii] = 0
    #        usq[ii] = .0001
        if ws>=1.0:
            a = 1.0
        else:
            a = ws
    #        usa[ii] = a*np.cos(ws*Tta[ii])
    #        usb[ii] = a*np.sin(ws*Tta[ii])
    #        usa[ii] = usd[ii-1]*cosdelta[ii-1] - usq[ii-1]*sindelta[ii-1]
    #        usb[ii] = usq[ii-1]*cosdelta[ii-1] + usd[ii-1]*sindelta[ii-1]

        params = [usd[ii], usq[ii], w[ii], rs, ld, lq, tmech, psirm]
        sol1a = odeint(PMSM_dynstepdq, x0, [0, delta_t], args = (params,), atol = abserr, rtol= relerr)
        sol1[ii] = sol1a[-1]
    #        display(sol1a[-1])
        x0 = sol1a[-1]
        me[ii] = 1.0*sol1[ii][1] + (ld -lq)*sol1[ii][0]*sol1[ii][1]
        paramsw = [me[ii], 0.0, tmech]
        sol2a = odeint(rotor_dyndq, y0, [0, delta_t], args = (paramsw,), atol = abserr, rtol= relerr)
        y0 = sol2a[-1]
        sol2[ii] = sol2a[-1]
        #Rotor angle using internal angle
        FCangle[ii] = sol2[ii][1]
        gamma[ii] = sol2[ii][1]
        w[ii] = sol2[ii][0]
        #using resolver output
    #        FCangle[ii] = sol2[ii]
        cosdelta[ii] = np.cos(FCangle[ii])
        sindelta[ii] = np.sin(FCangle[ii])
        #Coordinate transformation

```

```
isd[ii] = sol1[ii][0]
isq[ii] = sol1[ii][1]
#Convert stator coordinate current to field coordinates
isa[ii] = sol1[ii][0]*cosdelta[ii] - sol1[ii][1]*sindelta[ii]
isb[ii] = sol1[ii][1]*cosdelta[ii] + sol1[ii][0]*sindelta[ii]
#Rotor flux in field coordinates Using estimator output as in practice
#actual flux will not be available for measurement
psird[ii] = 1.0
psirq[ii]= 0.0
```

```
# rs, ld, lq, tmech, psirm = mach_parapm(mach_servodq)
```

```
# me = 1.0*isq + (ld - lq)*isd*isq
```

In [13]:

```

dirfig = "C:/Users/ashwi/Documents/tex/lec/drives/figures/"
pl.figure(501, figsize = (6, 6))
pl.rc('font', size = 16)
pl.subplot(2, 1, 1)
pl.plot(Tta, isd, 'purple', Tta, isdref, 'skyblue', lw=3)
pl.plot(Tta, isqref, 'tomato', Tta, isq, 'crimson', lw =3)
pl.xlim(0, tend)
pl.ylabel(r'$i_{sd}$, $i_{sq}$' )
pl.xticks(np.linspace(0, tend, 5))
ax1 = pl.subplot(2, 1, 2)
ax1.axhline(0)
ax1.plot(Tta, me, 'crimson', lw =3)
ax1.set_yticks(np.linspace(-1.0, 1.0, 5))
ax1.set_ylabel('me [p.u]')
# ax1.plot(Tta, gamma, 'navy', Tta, FCangle)
ax2 = ax1.twinx()
ax2.set_ylabel('$\omega$')
ax2.plot(Tta, w, 'peru', lw =3)

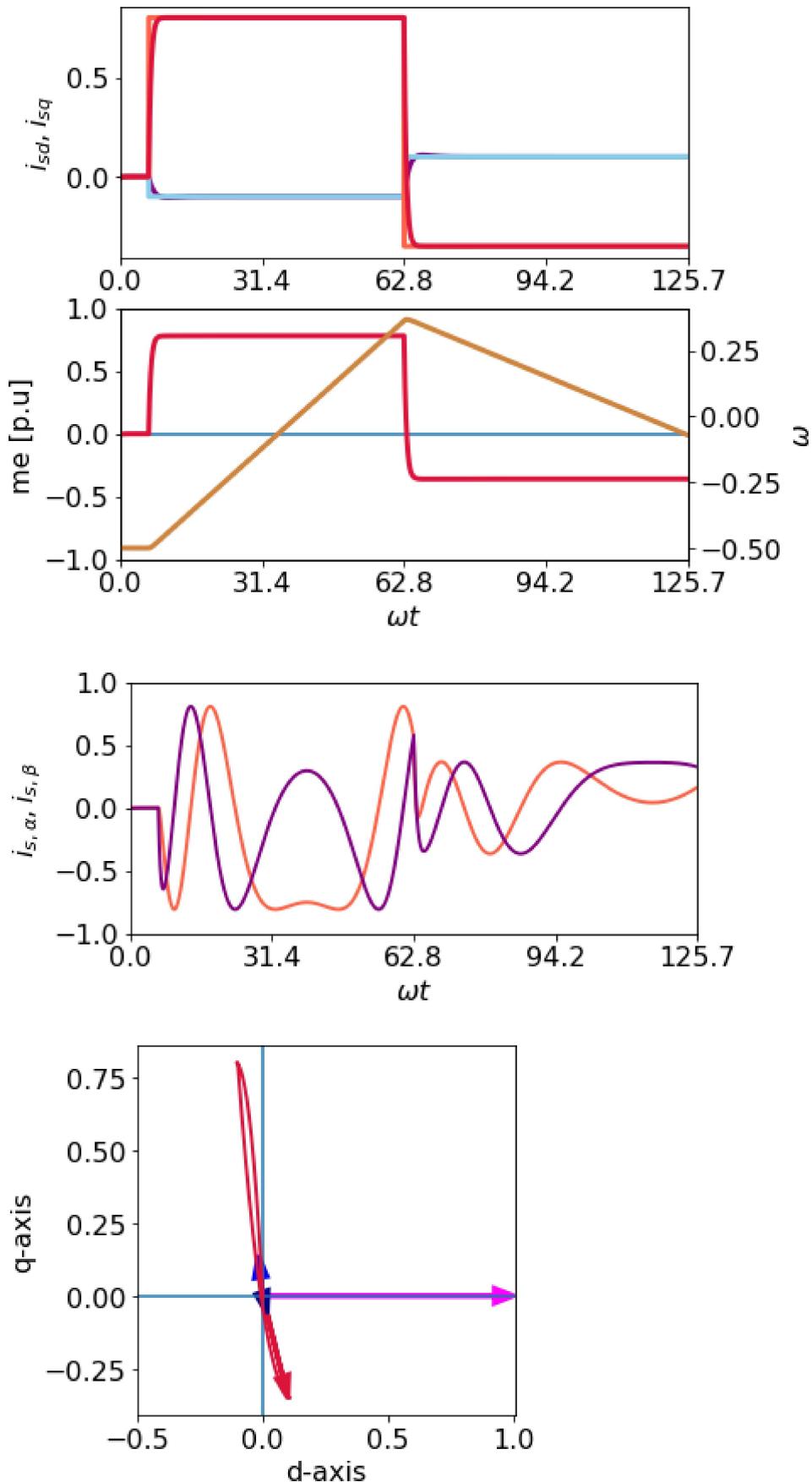
# pl.ylabel(r'$\psi_r, \alpha$, $\psi_r, \beta$' )
ax1.set_xlabel(r'$\omega$')
ax1.set_xlim(0, tend)
ax1.set_xticks(np.linspace(0, tend, 5))
# pl.savefig(dirfig + "PMSMdqSpeedConcurrentsvst.pdf", bbox_inches = 'tight', transparent = True)

pl.figure(521, figsize = (6, 6))
pl.rc('font', size = 16)
# pl.subplot(2, 1, 1)
# pl.plot(Tta, psiradash, 'r', Tta, psirbdash , 'purple' ,lw =2)
# pl.xlim(0, tend)
# # pl.ylabel(r'$\psi_s\alpha^{\prime}$, $\psi_s\beta^{\prime}$' )
# pl.xticks(np.linspace(0, tend, 5))
# pl.yticks(np.linspace(-1.0, 1.0, 5))
pl.subplot(2, 1, 2)
pl.plot(Tta, isa, 'tomato', lw =2)
pl.plot(Tta, isb, 'purple', lw =2)
pl.ylabel(r'$i_s\alpha$, $i_s\beta$' )
pl.xlabel(r'$\omega$ t$')
pl.xlim(0, tend)
pl.xticks(np.linspace(0, tend, 5))
pl.yticks(np.linspace(-1.0, 1.0, 5))
# pl.savefig(dirfig + "PMSMdqCurrentconsvst.pdf", bbox_inches = 'tight', transparent = True)

pl.figure(523, figsize = (4, 4))
pl.rc('font', size = 16)
pl.axhline(0)
pl.axvline(0)
pl.plot(isd, isq, 'crimson', lw =2, label = 'estimate')
# pl.plot(psird, psirq, 'blue', lw =2 , label = "actual")
pl.arrow(0, 0, psird[-1], psirq[-1], fc = 'magenta', ec = 'magenta', head_width = 0.05\
, length_includes_head = True, lw =3)
pl.arrow(0, 0, isd[-1], isq[-1], fc = 'crimson', ec = 'crimson', head_width = 0.05\
, length_includes_head = True, lw =3)
pl.arrow(0, 0, usd[-1], usq[-1], fc = 'navy', ec = 'navy', head_width = 0.05\
, length_includes_head = True, lw =3)
pl.arrow(0, 0, usd[index1], usq[index1], fc = 'blue', ec = 'blue', head_width = 0.05\
, length_includes_head = True, lw =3)

```

```
# pl.axis('equal')
pl.xlabel('d-axis')
pl.ylabel('q-axis')
pl.xlim(-0.5, 1.01)
# pl.savefig(dirfig + "PMSMdqXYwCC.pdf", bbox_inches = 'tight', transparent = True)
pl.show()
print(psirm)
```



0.9

In [14]:

```

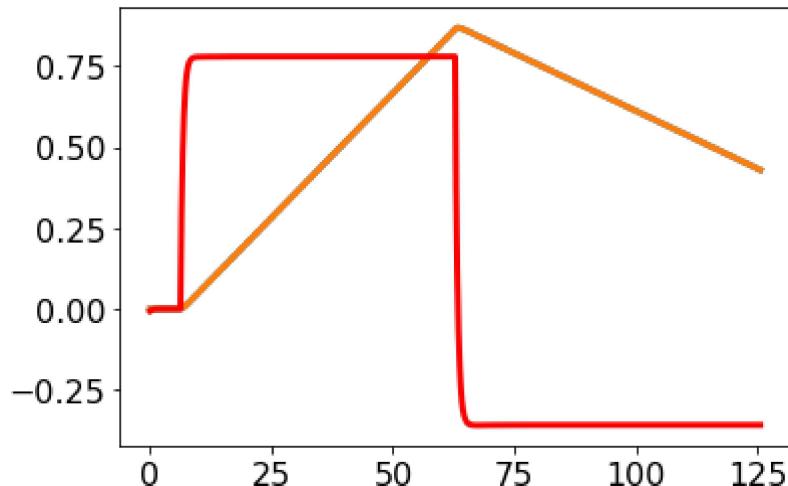
y0 = [0, 0]
wxx = np.zeros(len(Tta))
for ii in range(len(Tta)):
    params = [me[ii], 0.0, tmech]
    sol2a = odeint(rotor_dyndq, y0, [0, delta_t], args = (params,), atol = abserr, rtol= relerr)
    sol2[ii] = sol2a[-1]
    y0 = sol2a[-1]
    wxx[ii] = sol2[ii][0]

# wxx = sol2[:, 0]
pl.plot(Tta, sol2[:, 0], Tta, wxx, lw =3)
pl.plot(Tta, me, 'r', lw =3)
# pl.plot(Tta, sol2[:, 1], lw =3)

print(ws*lq/ld, ws*ld/lq, 1.0/lq, 1.0/ld, ld, )

```

-0.31112224448897796 -0.8035426731078905 2.3004370830457788 1.43143429716576 0.6986



4x4 PMSM model salient rotor

In [15]:

```
def PMSM_dynstepdq44(X, t, params):
    """Defines a function to calculate the derivatives for a 2x2 PMS motor dynamics
    1. Uses state variables stator current $|\vec{i}_s|$ id d-q coordinates.
    2. Is non-linear and uses the state variable $\omega$ rotor angular velocity to describe
       rotor dynamics. rotor angular velocity will be go from outside as parameter
    3. calls mach_im function to calculate the system equation coefficients
    4. mach_im will in turn call mach_para function.
    5. uses normalized time  $2*\pi*50*t$ 
    6. This is to be used in a for loop for the time span
    """
    x0 = X[0] #isd
    x1 = X[1] #isq
    x2 = X[2] #ws
    x3 = X[3] #rotor angle
    # wx = X[5]
    u1, u2, w, rs, ld, lq, tmech, psirm = params #needs state variable ws from outside
    #Convert voltage from stator coordinates into field coordinates
    # ud = u1*np.cos(x3) + u2*np.sin(x3)
    # uq = -u1*np.sin(x3) + u2*np.cos(x3)
    #
    ud = u1
    uq = u2
    # rs, ld, lq, tmech, psirm = mach_parapm(mach_ramu)
    # rs, ld, lq, tmech, psirm = mach_parapm(mach_servodq)
    mdiff = psirm*x1 + (ld - lq)*x0*x1 - mL
    # print(A[1][0], A[1][1], A[1][2], A[1][3], B[4][0])
    dx0dt = -(rs/ld)*x0 + w*(lq*x1)/ld + ud/ld
    dx1dt = - w*(ld*x0)/lq - (rs/lq)*x1 - w*psirm/lq + uq/lq
    dx2dt = mdiff/tmech
    dx3dt = x2
    return [dx0dt, dx1dt, dx2dt, dx3dt]
```

In [16]:

```
#Preparing simulation for current control
# We will use for loop

# ODE solver parameters
abserr = 1.0e-8
relerr = 1.0e-3

tend = 20*2*np.pi
tstart = 0.0
delta_t = 0.005
n = 100000
n2 = int(100*2*np.pi/0.01)
ws = -0.5
mL = 0.0
Tta = np.arange(tstart, tend, delta_t)
# Tta = np.linspace(tstart, tend, n2)
#delta_t = Tta[1] - Tta[0]
usa = np.zeros(len(Tta))
usb = np.zeros(len(Tta))
sol1 = np.zeros((len(Tta), 4))
sol2 = np.zeros((len(Tta), 2))
FCangle = np.zeros(len(Tta))
gamma = np.zeros(len(Tta))

# Field coordinate currents
cosdelta = np.zeros(len(Tta))
sindelta = np.zeros(len(Tta))
isd = np.zeros(len(Tta))
isq = np.zeros(len(Tta))
isa = np.zeros(len(Tta))
isb = np.zeros(len(Tta))
usd = np.zeros(len(Tta))
usq = np.zeros(len(Tta))
eid = np.zeros(len(Tta))
eqi = np.zeros(len(Tta))
w = np.zeros(len(Tta))
w2 = np.zeros(len(Tta))
gamma2 = np.zeros(len(Tta))

rs, ld, lq, tmech, psirm = mach_parapm(mach_ramu)
# rs, ld, lq, tmech, psirm = mach_parapm(mach_servodq)

#Creating reference values for isd and isq
isdrefval = -0.1
isdref = np.zeros(len(Tta))
isqref = np.zeros(len(Tta))

isqref1 = 0.0
isqref2 = 0.8
isqref3 = -0.35
for ii in range(len(Tta)):
    isdref[ii] = 0.0
    if (Tta[ii]>=10*2*np.pi):
        isqref[ii] = isqref3
        isdref[ii] = -isdrefval
    elif(Tta[ii]>=1*2*np.pi):
        isqref[ii]=isqref2
        isdref[ii] = isdrefval
```

```

else:
    isqref[ii]=isqref1

psird = np.zeros(len(Tta))
psirq = np.zeros(len(Tta))

me = np.zeros(len(Tta))
x0 = [0, 0, 0, 0, ws, 0]

kpd = 1d*1
Tid = 25.0e1

kpq = 1q*2
Tiq = 25.0e1

Kparamsd = [kpd, Tid]
Kparamsq = [kpq, Tiq]

#isdref = 0.26
#isqref = 0.75
#The intial value error will never get corrected in
#rotor angle
y0 = [ws, 0]

for ii in range(len(Tta)):
    #Start controller after first step
    if ii>=0:
        eid[ii] = isdref[ii-1] - isd[ii-1]
        PIed = [eid[ii], eid[ii-1]]
        usd[ii] = PIcon(PIed, usd[ii-1], delta_t, Kparamsd)
        eiq[ii] = isqref[ii-1] - isq[ii-1]
        PIEq = [eiq[ii], eiq[ii-1]]
        usq[ii] = PIcon(PIEq, usq[ii-1], delta_t, Kparamsq)
    #    usd[ii] = 0
    #    usq[ii] = .0001
        if ws>=1.0:
            a = 1.0
        else:
            a = ws
    #    usa[ii] = a*np. cos(ws*Tta[ii])
    #    usb[ii] = a*np. sin(ws*Tta[ii])
    #    usa[ii] = usd[ii-1]*cosdelta[ii-1] - usq[ii-1]*sindelta[ii-1]
    #    usb[ii] = usq[ii-1]*cosdelta[ii-1] + usd[ii-1]*sindelta[ii-1]

        params = [usd[ii], usq[ii], w[ii], rs, ld, lq, tmech, psirm]
        sol1a = odeint(PMSM_dynstepdq44, x0, [0, delta_t], args = (params,), atol = abserr, rtol= relerr)
        sol1[ii]=sol1a[-1]
    #    display(sol1a[-1])
        x0 = sol1a[-1]
        me[ii] = 1.0*sol1[ii][1] + (ld -lq)*sol1[ii][0]*sol1[ii][1]
        paramsw = [me[ii], 0, 0, tmech]
        sol2a = odeint(rotor_dyndq, y0, [0, delta_t], args = (paramsw,), atol = abserr, rtol= relerr)
        y0 = sol2a[-1]
        sol2[ii] = sol2a[-1]
        #Rotor angle using internal angle
        FCangle[ii] = sol1[ii][3]
        gamma[ii] = sol1[ii][3]
        w[ii] = sol1[ii][2]
        #using resolver output
    #    FCangle[ii] = sol2[ii]

```

```
cosdelta[ii] = np.cos(FCangle[ii])
sindelta[ii] = np.sin(FCangle[ii])
isd[ii] = sol1[ii][0]
isq[ii] = sol1[ii][1]
w2[ii] = sol2[ii][0]
gamma2[ii] = sol2[ii][1]
#Coordinate transformation
#Convert stator coordinate current to field coordinates
isa[ii] = sol1[ii][0]*cosdelta[ii] - sol1[ii][1]*sindelta[ii]
isb[ii] = sol1[ii][1]*cosdelta[ii] + sol1[ii][0]*sindelta[ii]
#Rotor flux in field coordinates Using estimator output as in practice
#actual flux will not be available for measurement
psird[ii] = 1.0
psirq[ii] = 0.0

# isd = sol1[:, 0]
# isq = sol1[:, 1]
# w2 = sol2[:, 0]
# gamma2 = sol2[:, 1]

# rs, ld, lq, tmech, psirm = mach_param(mach_servodq)

# me = 1.0*isq + (ld - lq)*isd*isq
```

In [17]:

```

dirfig = "C:/Users/ashwi/Documents/tex/lec/drives/figures/"
pl.figure(501, figsize = (4, 4))
pl.rc('font', size = 12)
pl.subplot(2, 1, 1)
pl.plot(Tta, isd, 'purple', Tta, isdref, 'skyblue', lw=2)
pl.plot(Tta, isqref, 'tomato', Tta, isq, 'crimson', lw =2)
pl.xlim(0, tend)
pl.ylabel(r'$i_{sd}$, $i_{sq}$' )
pl.xticks(np.linspace(0, tend, 5))
pl.minorticks_on()
pl.grid(which = 'major', lw =0.5, c = 'k')
pl.grid(which = 'minor', lw =0.4, color = 'skyblue')

ax1 = pl.subplot(2, 1, 2)
ax1.axhline(0)
ax1.plot(Tta, me, 'crimson', lw =2)
ax1.set_yticks(np.linspace(-1.0, 1.0, 5))
ax1.set_ylabel('me [p.u]')
# ax1.plot(Tta, gamma, 'navy', Tta, FCangle)
ax2 = ax1.twinx()
ax2.set_ylabel('$\omega$')
ax2.plot(Tta, w, 'peru', lw =2)
# ax2.plot(Tta, w2, 'brown', lw =2)

# pl.ylabel(r'$\psi_{r,\alpha}$, $\psi_{r,\beta}$' )
ax1.set_xlabel(r'$\omega_t$')
ax1.set_xlim(0, tend)
ax1.set_xticks(np.linspace(0, tend, 5))
ax1.minorticks_on()
ax1.grid(which = 'major', lw =0.5, c = 'k')
ax1.grid(which = 'minor', lw =0.4, color = 'skyblue')
# pl.savefig(dirfig + "PMSMdqSpeedConcurrentsvst.pdf", bbox_inches = 'tight', transparent = True)

pl.figure(521, figsize = (4, 3.6))
pl.rc('font', size = 14)
# pl.subplot(2, 1, 1)
# pl.plot(Tta, psiradash, 'r', Tta, psirbdash, 'purple', lw =2)
# pl.xlim(0, tend)
# # pl.ylabel(r'$\psi_{s\alpha}'^{'}, $\psi_{s\beta}'^{'})'
# pl.xticks(np.linspace(0, tend, 5))
# pl.yticks(np.linspace(-1.0, 1.0, 5))
pl.subplot(1, 1, 1)
pl.plot(Tta, isa, 'tomato', lw =2)
pl.plot(Tta, isb, 'purple', lw =2)
pl.ylabel(r'$i_{s,\alpha}$, $i_{s,\beta}$' )
pl.xlabel(r'$\omega_t$')
pl.xlim(0, tend)
pl.xticks(np.linspace(0, tend, 5))
pl.yticks(np.linspace(-1.0, 1.0, 5))
pl.minorticks_on()
pl.grid(which = 'major', lw =0.5, c = 'k')
pl.grid(which = 'minor', lw =0.4, color = 'skyblue')
# pl.savefig(dirfig + "PMSMdqCurrentconsvst.pdf", bbox_inches = 'tight', transparent = True)

pl.figure(523, figsize = (4, 4))
pl.rc('font', size = 14)
pl.axhline(0)
pl.axvline(0)

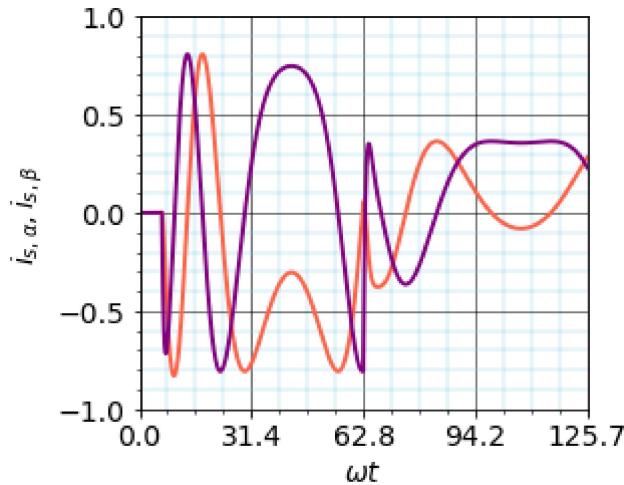
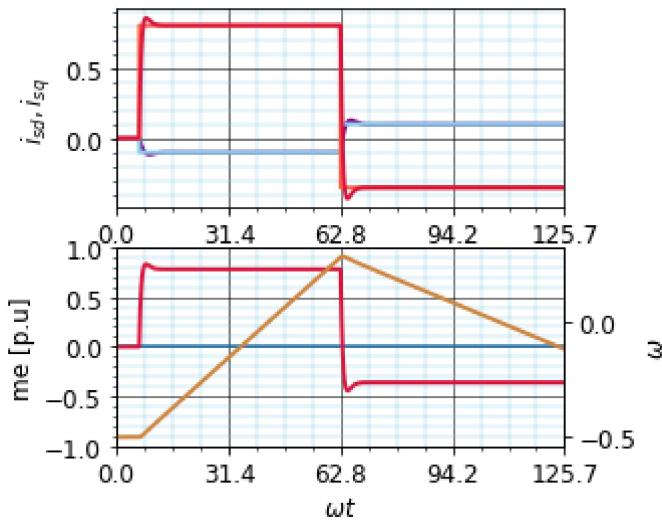
```

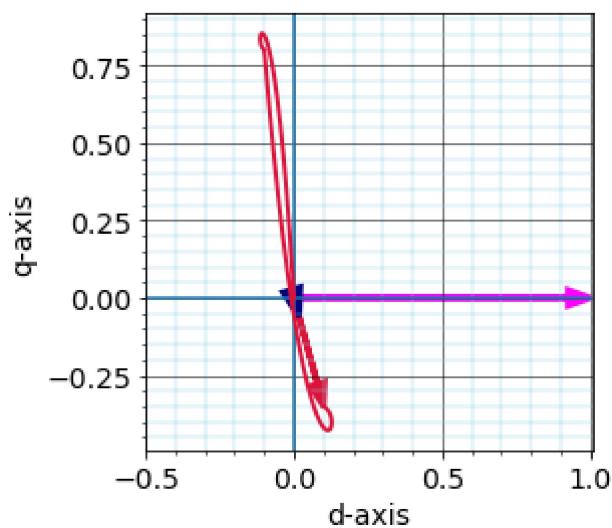
```

pl.plot(isd,isq, 'crimson', lw =2, label = 'estimate')
# pl.plot(psird, psirq, 'blue', lw =2 , label = "actual")
pl.arrow(0,0,psird[-1],psirq[-1], fc = 'magenta', ec = 'magenta', head_width = 0.05\
, length_includes_head = True, lw =3)
pl.arrow(0,0,isd[-1],isq[-1], fc = 'crimson', ec = 'crimson', head_width = 0.05\
, length_includes_head = True, lw =3)
pl.arrow(0,0,usd[-2],usq[-2], fc = 'navy', ec = 'navy', head_width = 0.06\
, length_includes_head = True, lw =3)
# pl.arrow(0,0,usd[index1],usq[index1], fc = 'blue', ec = 'blue', head_width = 0.05\
# , length_includes_head = True, lw =3)

# pl.axis('equal')
pl.xlabel('d-axis')
pl.ylabel('q-axis')
pl.xlim(-0.5,1.01)
# pl.savefig(dirfig + "PMSMdqXYwCC.pdf", bbox_inches = 'tight', transparent = True)
pl.minorticks_on()
pl.grid(which = 'major', lw =0.5, c = 'k')
pl.grid(which = 'minor', lw =0.4, color = 'skyblue')
pl.show()
print(psirm)

```





0. 9