

In [1]:

```
import numpy as np
from scipy.integrate import odeint
from scipy import signal
import matplotlib.pyplot as plt
%matplotlib inline
```

In [2]:

```
#Defining Machine parameters
mach_mc = {"rs": 0.1729, "lq": 0.6986, "ld": 0.4347, "tmech":50.5, "psi_rm":0.9}
# mach_mc = {"rs": 0.009, "ld": 4.14, "lq": 4.21, "tmech":509.6, "psi_rm": 0.5}

# mach_mb = {"rs": 0.0185, "rr": 0.0132, "lh": 3.81, "ls": 3.9, "lr": 3.9, "tmech":397.31}

# mach_ma = {"rs": 0.015, "rr": 0.04, "lh": 2.31, "ls": 2.35, "lr": 2.35, "tmech":596.9}

# mach_BM = {"rs": 0.0426, "rr": 0.02113, "lh": 2.252, "ls": 2.252+0.078, "lr": 2.252+0.1052, "tmech":596.9}
```

The Machine Parameters

We use machine dictionaries to calculate the various parameters used in space vector based machine model
This function is called mach_para(dict)

In [3]:

```
def mach_para(mach):
    """Takes in the dictionary containing machine parameters (normalized)
    and returns various parameters that are used in space vector equations

    Takes argument: machine dictionary
    """
    rs = mach["rs"]
    ld = mach['ld']
    lq = mach['lq']
    # ws = mach['ws']
    # sig= 1-(lh*lh)/(lr*ls)
    # kr=lh/lr
    # sigls=sig*ls
    # tr=lr/rr
    # rk=(rs+(kr)*(kr)*rr)
    # tk=sigls/(rs+(kr)*(kr)*rr)
    tmech = mach['tmech']
    psi_rm = mach['psi_rm']
    # print("lh = {0:1.3f}".format(lh))
    # return rs, rr, lh, ls, lr, sig, kr, tr, rk, tk, tmech
    return rs, ld, lq, tmech, psi_rm
```

In [4]:

```
mach_para(mach_mc)
```

Out[4]:

```
(0.1729, 0.4347, 0.6986, 50.5, 0.9)
```

PMSM Dynamics

Building the Dynamic Model of a PMSM in the d-q Coordinate

To model a PMSM, two state variables are chosen, which are i_s and ω_s . The i_s can be further decomposed to i_{sd} , i_{sq} in the permanent magnetic field since it is a complex space vector. The differential equations of these vectors are given as:

$$\begin{aligned}\frac{di_{sd}}{d\tau} &= -\frac{r_s}{l_d}i_{sd} + \omega_s \frac{l_q}{l_d}i_{sq} + \frac{v_{sd}}{l_d} \\ \frac{di_{sq}}{d\tau} &= -\omega_s \frac{l_d}{l_q}i_{sd} - \frac{r_s}{l_q}i_{sq} - \omega_s \frac{\psi_{r,m}}{l_q} + \frac{v_{sq}}{l_q} \\ \frac{d\omega_s}{d\tau} &= \frac{1}{\tau_{mech}}[m_e - m_L], m_e = \psi_{r,m}i_{sq} + (l_d - l_q)i_{sd}i_{sq}\end{aligned}$$

In [5]:

```

#Induction machine 5x5 dynamic model

def IM_dynstep(X, t, params):
    """Defines a function to calculate the derivatives for a 5x5 Induction motor dynamic
    1. Uses state variables stator current  $\vec{i}_s$  and rotor flux space vector  $\vec{\psi}_r$ 
    2. Is non-linear and uses the state variable  $\omega$  rotor angular velocity to describe
        rotor dynamics
    3. calls mach_im function to calculate the system equation coefficients
    4. mach_im will in turn call mach_para function.
    5. uses normalized time  $2\pi \cdot 50 \cdot t$ 
    6. This is to be used in a for loop for the time span
    """
    x0 = X[0] #isd
    x1 = X[1] #isq
    ws = X[2] # w
    # x2 = X[2] #psiralpha
    # x3 = X[3] #psirbeta
    # x4 = X[4] #w
    # wx = X[5]
    # u1,u2,m_l,w_s,psi_rm, l_d, l_q = params
    u1,u2,m_l,w_s = params
    rs,l_d,l_q, tmech, psi_rm = mach_para(mach_mc)
    # print("u1:")
    # print(u1)

    A,B,C,D,tm = mach_imstep(mach_mc, params)
    # print("B[0]:")
    # print(B[0])
    m_diff = (psi_rm*x0+(l_d-l_q)*x0*x1) - m_l
    dx0_dt = (A[0][0]*x0 + A[0][1]*x1 + A[0][2]*ws + B[0][0]*u1)
    dx1_dt = (A[0][0]*x0 + A[0][1]*x1 + A[0][2]*ws + B[1][0]*(-w_s*psi_rm+u2))
    dx2_dt = (A[0][0]*x0 + A[0][1]*x1 + A[0][2]*ws + B[2][0]*m_diff)
    # print(A[1][0],A[1][1],A[1][2],A[1][3],B[4][0])
    # dx0dt = (A[0][0]*x0 + A[0][1]*x1 + A[0][2]*x2 + A[0][3]*x3*x4 + A[0][4]*x4 + B[0][0]*u1)
    # dx1dt = (A[1][0]*x0 + A[1][1]*x1 + A[1][2]*x2*x4 + A[1][3]*x3 + A[1][4]*x4 + B[1][0]*u2)
    # dx4dt = (A[4][0]*x0 + A[4][1]*x1 + A[4][2]*x2 + A[4][3]*x3 + A[4][4]*x4 + B[4][0]*mdiff)
    # dx2dt = (A[2][0]*x0 + A[2][1]*x1 + A[2][2]*x2 + A[2][3]*x3*x4 + A[2][4]*x4 + 0)
    # dx3dt = (A[3][0]*x0 + A[3][1]*x1 + A[3][2]*x2*x4 + A[3][3]*x3 + A[3][4]*x4 + 0)
    # dx4dt = (A[4][0]*x0 + A[4][1]*x1 + A[4][2]*x2 + A[4][3]*x3 + A[4][4]*x4 + B[4][0]*mdiff)
    # dwxdt = rate
    return [dx0_dt, dx1_dt, dx2_dt]

def mach_imstep(mc_dict, params):
    """
    Define the induction model that is a 3x3 matrix.
    1. The current components are  $i_{s\alpha} + j i_{s\beta}$ .
    2. The rotor flux components are  $\psi_{r\alpha} + j \psi_{r\beta}$  and  $\omega$ 
    3. Needs machine dictionary as input
    4. does not use  $\omega_s$  in this function, it will be multiplied in the derivative function
    """
    u1,u2,m_l,ws = params
    # rs,rr,ld,lq,sig,kr,tr,rk,tk,tmech = mach_para(mc_dict)
    rs,ld,lq, tmech, psi_rm = mach_para(mc_dict)
    a11 = -(rs/ld)
    a12 = ((ws*lq)/ld)
    a13 = 0
    # a14 = (kr/(sig*ls)) #has to be multiplied by omega
    # a15 = 0

```

```

a21 = -((ld*ws)/lq)
a22 = -rs/lq
a23 = 0
# a24 = a13
# a25 = 0

a31 = 0
a32 = 0
a33 = 0

b11 = 1/(ld)
b21 = 1/(lq)
b31 = 1/(tmech)
# b41 = 0
# b51 = 1/(tmech)

c11 = 1.0
c22 = 1.0
c33 = 1.0
# c44 = 1.0
# c55 = 1.0

A = ([a11, a12, a13], [a21, a22, a23], [a31, a32, a33])
B = ([b11], [b21], [b31])
C = ([c11, c22, c33])
D= ([0, 0], [0, 0], [0, 0], [0, 0], [0, 0])
# C=([c11, 0, 0], [0, c22, 0], [0, 0, c33])
# D= ([0, 0], [0, 0], [0, 0], [0, 0], [0, 0])
# A=([a11, a12, a13, a14, a15], [a21, a22, a23, a24, a25], [a31, a32, a33, a34, a35], [a41, a42, a43, a44, a45], [a51, a52, a53, a54, a55])
# B=([b11, 0], [b21, 0], [b31, 0], [b41, 0], [b51, 0])
# C=([c11, 0, 0, 0, 0], [0, c22, 0, 0, 0], [0, 0, c33, 0, 0], [0, 0, 0, c44, 0], [0, 0, 0, 0, a55])
# D= ([0, 0], [0, 0], [0, 0], [0, 0], [0, 0])

return A, B, C, D, tmech

```

In [6]:

```

#Rotor dynamics as mechanics
def rotor_dyndq(X, t, params):
    w = X[0]
    gamma = X[1] #rotor angle

    me, mL, tmech = params
    dwdt = (me-mL)/tmech
    dgammat = w
    return [dwdt, dgammat]

```

In [7]:

```
#Function for rotor angle integration
# d\delta/dt = w

def rotorangle(X, t, params):
    del0 = X
    w = params
    ddel0dt = w
    return ddel0dt
```

In [8]:

```
def PIcon(xe, y, dt, Kparams):
    """Pass on xe = [error[k], error[k-1]]
    Pass on y = [y[k-1]]
    Pass on Parameters = [kp, Ti]
    """
    kp, Ti = Kparams
    xeo = xe[1]
    xen = xe[0]
    # print(xeo, xen, y, kp, Ti)
    y1 = y
    y2 = y1 + kp*(xen - xeo) + (kp/Ti)*(xen)
    return y2

def PIconwithLim(xe, y, dt, Kparams):
    """Pass on xe = [error[k], error[k-1]]
    Pass on y = [y[k-1]]
    Pass on Parameters = [kp, Ti]
    """
    kp, Ti, uplim, dwnlim = Kparams
    xeo = xe[1]
    xen = xe[0]
    # print(xeo, xen, y, kp, Ti)
    # y2 = y1 + kp*(xen - xeo) + (kp/Ti)*(xen)
    # print(xeo, xen, y, kp, Ti)
    y1 = y
    yx = y1 + kp*(xen - xeo) + (kp/Ti)*(xen)
    if (yx > uplim):
        yx = uplim
    elif (yx < dwnlim):
        yx = dwnlim
    y2 = yx
    return y2
```

In [9]:

```
# # testing block:
# TA, TB, TC, TD, tm = mach_imstep(mach_mc, params)
# print(TA, TB, TC, TD, tm)
# # print('kr = {0:3.3f}'.format(kr))
```

1. System Dynamics without Control

In [24]:

```

#Setting up the simulaton for rotor model
# We will use for loop for simulation
rs, ld, lq, tmech, psi_rm = mach_para(mach_mc)
# ODE solver parameters
abserr = 1.0e-8
relerr = 1.0e-3

tend = 20*2*np.pi
tstart = 0.0
delta_t = 0.01
n = 100000
n2 = int(100*2*np.pi/0.01)
ws = -0.5
mL = 0.0
Tta = np.arange(tstart, tend, delta_t)
# Tta = np.linspace(tstart, tend, n2)
#delta_t = Tta[1] - Tta[0]
usd = np.zeros(len(Tta))
usq = np.zeros(len(Tta))
sol1 = np.zeros((len(Tta), 3))
# sol2 = np.zeros((len(Tta), 2))
# FCangle = np.zeros(len(Tta))
# Field coordinate currents
cosdelta = np.zeros(len(Tta))
sindelta = np.zeros(len(Tta))
isd = np.zeros(len(Tta))
isq = np.zeros(len(Tta))
eid = np.zeros(len(Tta))
eiq = np.zeros(len(Tta))
w = np.zeros(len(Tta))

# psird = np.zeros(len(Tta))
# psirq = np.zeros(len(Tta))

# # Initializing reference values for isd and isq:
# isdrefval = -0.1
# isdref = np.zeros(len(Tta))
# isqref = np.zeros(len(Tta))

# isqref1 = 0.0
# isqref2 = 0.8
# isqref3 = -0.35
# for ii in range(len(Tta)):
#     isdref[ii] = 0.0
#     if (Tta[ii]>=10*2*np.pi):
#         isqref[ii] = isqref3
#         isdref[ii] = -isdrefval
#     elif(Tta[ii]>=1*2*np.pi):
#         isqref[ii]=isqref2
#         isdref[ii] = isdrefval
#     else:
#         isqref[ii]=isqref1

me = np.zeros(len(Tta))
x0 = [0, 0, ws] # isd, isq, ws
y0 = [1.0, 0]

kpd = ld*1
Tid = 25.0e1

```

```

kpq = lq*2
Tiq = 25.0e1

Kparamsd = [kpd,Tid]
Kparamsq = [kpq,Tiq]

for ii in range(len(Tta)):
#   #Start controller after first step
#   if ii>=0:
#       eid[ii] = isdref[ii-1] - isd[ii-1]
#       PIed = [eid[ii],eid[ii-1]]
#       usd[ii] = PIcon(PIed, usd[ii-1], delta_t,Kparamsd)
#       eiq[ii] = isqref[ii-1] - isq[ii-1]
#       PIEq = [eiq[ii],eiq[ii-1]]
#       usq[ii] = PIcon(PIeq, usq[ii-1], delta_t,Kparamsq)
#       usd[ii] = 0
#       usq[ii] = .0001
    if ws>=1.0:
        a = 1.0
    else:
        a = ws
#   usa[ii] = a*np.cos(ws*Tta[ii])
#   usb[ii] = a*np.sin(ws*Tta[ii])
#   usa[ii] = usd[ii-1]*cosdelta[ii-1] - usq[ii-1]*sindelta[ii-1]
#   usb[ii] = usq[ii-1]*cosdelta[ii-1] + usd[ii-1]*sindelta[ii-1]

    params = [usd[ii], usq[ii],w[ii], ws]
    sol1a = odeint(IM_dynstep,x0,[0,delta_t], args = (params,),atol = abserr, rtol= relerr)
    sol1[ii]= sol1a[-1]
#   display(sol1a[-1])
    x0 = sol1a[-1]
#   me[ii] = 1.0*sol1[ii][1] + (ld -lq)*sol1[ii][0]*sol1[ii][1]
#   paramsw = [me[ii],0.0,tmech]
#   sol2a = odeint(rotor_dyndq, y0, [0,delta_t], args = (paramsw,), atol = abserr, rtol= relerr)
#   y0 = sol2a[-1]
#   sol2[ii] = sol2a[-1]
#   #Rotor angle using internal angle
#   FCangle[ii] = sol2[ii][1]
#   gamma[ii] = sol2[ii][1]
#   w[ii] = sol2[ii][0]
#   #using resolver output
#   # FCangle[ii] = sol2[ii]
#   cosdelta[ii] = np.cos(FCangle[ii])
#   sindelta[ii] = np.sin(FCangle[ii])
#Coordinate transformation
    isd[ii] = sol1[ii][0]
    isq[ii] = sol1[ii][1]
#   #Convert stator coordinate current to field coordinates
#   isa[ii] = sol1[ii][0]*cosdelta[ii] - sol1[ii][1]*sindelta[ii]
#   isb[ii] = sol1[ii][1]*cosdelta[ii] + sol1[ii][0]*sindelta[ii]
#Rotor flux in field coordinates Using estimator output as in practice
#actual flux will not be available for measurement
#   psird[ii] = 1.0
#   psirq[ii]= 0.0

# for ii in range(len(Tta)):
#   if ws>=1.0:
#       a = 1.0
#   else:
#       a = ws

```



```

#     usa[ii] = a*np.cos(ws*Tta[ii])
#     usb[ii] = a*np.sin(ws*Tta[ii])
# #     me[ii] = kr*(sol1[ii-1][2]*sol1[ii-1][1] - sol1[ii-1][3]*sol1[ii-1][0]) - mL
#     params = [usa[ii], usb[ii], mL, ws]
#     sol1a = odeint(IM_dynstep, x0, [0, delta_t], args = (params,), atol = abserr, rtol= relerr)
#     sol1[ii]= sol1a[-1]
# #     display(sol1a[-1])
#     x0 = sol1a[-1]
# #     params2 = [sol1[ii][0], sol1[ii][1], sol1[ii][4]] # isa, isb, w
# #     sol2a = odeint(psiest_vr, y0, [0, delta_t], args = (params2,), atol = abserr, rtol= relerr)
# #     sol2[ii] = sol2a[-1]
# #     psircomp = np.complex(sol2[ii][0], sol2[ii][1])
# #     FCangle[ii] = np.arccos(sol2[ii][0]/np.abs(psircomp))
# #     cosdelta[ii] = sol2[ii][0]/np.abs(psircomp)
# #     sindelta[ii] = sol2[ii][1]/np.abs(psircomp)
# #     #Coordinate transformation
# #     #Convert stator coordinate current to field coordinates
# #     isd[ii] = sol1[ii][0]*cosdelta[ii] + sol1[ii][1]*sindelta[ii]
# #     isq[ii] = sol1[ii][1]*cosdelta[ii] - sol1[ii][0]*sindelta[ii]
# #     #Rotor flux in field coordinates
# #     psird[ii] = sol2[ii][0]*cosdelta[ii] + sol2[ii][1]*sindelta[ii]
# #     psirq[ii] = sol2[ii][1]*cosdelta[ii] - sol2[ii][0]*sindelta[ii]
# #     y0 = sol2a[-1]

# isd = sol1[:, 0]
# isq = sol1[:, 1]
# w = sol1[:, 2]
# psra = sol1[:, 2]
# psrb = sol1[:, 3]
# w = sol1[:, 4]
# psiradash = sol2[:, 0]
# psirbdash = sol2[:, 1]

#rs, rr, lh, ls, lr, sig, kr, tr, rk, tk, tmech = mach_para(mach_mb)
rs, l_d, l_q, tmech, psi_rm = mach_para(mach_mc)
me = psi_rm*isq + (l_d-l_q)*isd*isq

```

In [25]:

```

pl.figure(531, figsize = (6,6))
pl.rc('font', size = 16)
pl.subplot(2,1,1)
pl.plot(Tta, isd, 'purple', Tta, isdref, 'skyblue', lw=3)
pl.plot(Tta, isqref, 'tomato', Tta, isq, 'crimson', lw =3)
pl.xlim(0,tend)
pl.ylabel(r'$i_{sd}, i_{sq}$' )
pl.xticks(np.linspace(0, tend, 5))
ax1 = pl.subplot(2,1,2)
ax1.axhline(0)
ax1.plot(Tta, me, 'crimson', lw =3)
ax1.set_yticks(np.linspace(-1.0, 1.0, 5))
ax1.set_ylabel('me [p.u]')
# t1w = np.pi*2*45
# t2w = np.pi*2*60
# pl.xlim(t1w, t2w)
# # pl.xlim(0, tend)
# pl.axhline(0.75)
# pl.axhline(-0.75)
# pl.ylabel(r'$i_{s\alpha}, i_{s\beta}$' )
# pl.xticks(np.linspace(t1w, t2w, 5))
# pl.yticks(np.linspace(-1.0, 1.0, 5))
# pl.ylim(-1.0, 1.0)
# ax1 = pl.subplot(2,1,2)
# ax1.plot(Tta, isd, 'tomato', lw =3, label = r'$i_{sd}$')
# ax1.plot(t1w+1, 0.26, 'o', c = 'tomato')
# pl.text(t1w+2, 0.28, 'i = {0:1.3f}'.format(isd[-1]), fontsize = 12)
# ax1.plot(Tta, isq, 'purple', lw =3, label = r'$i_{sq}$')
# ax1.plot(Tta, psird, 'magenta', lw=2)
# ax1.set_ylabel(r'$i_{sd}, i_{sq}$' )
# ax1.set_xlabel(r'$\omega t$')
# ax2 = ax1.twinx()
# ax2.plot(Tta, FCangle, 'olive', lw =2)
# pl.rcParams['legend.fontsize']=12
# ax1.legend(loc = 'upper left')

#pl.axhline(1.0)
#pl.axhline(-1.0)

# t1w = np.pi*2*45
# t2w = np.pi*2*60
# ax1.set_xlim(t1w, t2w)
# ax1.set_xticks(np.linspace(t1w, t2w, 5))
# ax1.set_yticks(np.linspace(0, 1.0, 6))
# ax1.set_ylim(-0.1, 1.0)
# ax2.set_ylim(0, 3.14)
# pl.savefig(dirfig + "F0Currentsvst.pdf", bbox_inches = 'tight', transparent = True)

# pl.figure(533, figsize = (4,4))
# pl.rc('font', size = 16)
# pl.rcParams['axes.titlesize']=14
# pl.plot(isd, isq, 'crimson', lw =2, label = 'Field currents')
# # pl.plot(psird, psirq, 'blue', lw =4, label = "actual")
# # pl.axhline(0)
# pl.axvline(0)
# pl.arrow(0, 0, psird[-1], psirq[-1], fc = 'magenta', ec = 'magenta', head_width = 0.05\
# , length_includes_head = True, lw =3)
# pl.arrow(0, 0, isd[-1], isq[-1], fc = 'crimson', ec = 'crimson', head_width = 0.05\
# , length_includes_head = True, lw =3)

```

```

# pl.arrow(0,0,isd[-1],0, fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(isd[-1],0,0,isq[-1], fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.text(psiird[-1]-0.05,psirq[-1]+0.08, r'$\vec{\psi}_r$', fontsize=16 )
# pl.xlabel(r'$d$')
# pl.ylabel(r'$q$')
# pl.xlim(-0.01,1.0)
# pl.ylim(-0.02,1.0)
# # pl.xticks(np.linspace(-1.0,1.0,5))
# # pl.axis('equal')
# pl.rcParams['legend.fontsize']=9
# # pl.legend(loc = 'lower left', bbox_to_anchor = (0.8,0))
# pl.title('Field Coordinates')
# pl.savefig(dirfig + "FOXYvectors.pdf", bbox_inches = 'tight', transparent = True)

# pl.figure(536, figsize = (4,4))
# pl.rc('font', size = 16)
# pl.rcParams['axes.titlesize']=14
# pl.plot(isd,isq,'crimson', lw =2, label = 'Field currents')
# # pl.plot(psiird, psirq, 'blue', lw =4 , label = "actual")
# # pl.axhline(0)
# # pl.axvline(0)
# pl.arrow(0,0,psiird[-1],psirq[-1], fc = 'magenta', ec = 'magenta', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(0,0,isd[-1],isq[-1], fc = 'crimson', ec = 'crimson', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(0,0,isd[-1],0, fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(isd[-1],0,0,isq[-1], fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)

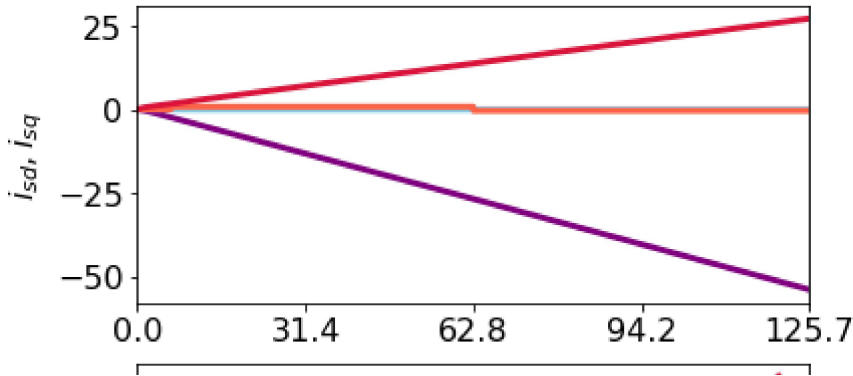
# pl.xlabel(r'$d$')
# pl.ylabel(r'$q$')
# # pl.xlim(0,1.0)
# # pl.ylim(-0.02,1.0)
# # pl.xticks(np.linspace(-1.0,1.0,5))
# pl.axis('equal')
# pl.rcParams['legend.fontsize']=9
# # pl.legend(loc = 'lower left', bbox_to_anchor = (0.8,0))
# pl.title('Field Coordinates')
# # pl.savefig(dirfig + "FOXYvectorsFull.pdf", bbox_inches = 'tight', transparent = True)
# pl.show()
# print(isd[-1], isq[-1])

```

Out[25]:

Text(0, 0.5, 'me [p.u]')





2. System Dynamics with Control

In [19]:

```

#Setting up the simulaton for rotor model
# We will use for loop for simulation
rs, ld, lq, tmech, psi_rm = mach_para(mach_mc)
# ODE solver parameters
abserr = 1.0e-8
relerr = 1.0e-3

tend = 20*2*np.pi
tstart = 0.0
delta_t = 0.01
n = 100000
n2 = int(100*2*np.pi/0.01)
ws = -0.5
mL = 0.0
Tta = np.arange(tstart, tend, delta_t)
# Tta = np.linspace(tstart, tend, n2)
#delta_t = Tta[1] - Tta[0]
usd = np.zeros(len(Tta))
usq = np.zeros(len(Tta))
sol1 = np.zeros((len(Tta), 3))
# sol2 = np.zeros((len(Tta), 2))
# FCangle = np.zeros(len(Tta))
# Field coordinate currents
cosdelta = np.zeros(len(Tta))
sindelta = np.zeros(len(Tta))
isd = np.zeros(len(Tta))
isq = np.zeros(len(Tta))
eid = np.zeros(len(Tta))
eiq = np.zeros(len(Tta))
w = np.zeros(len(Tta))

# psird = np.zeros(len(Tta))
# psirq = np.zeros(len(Tta))

# # Initializing reference values for isd and isq:
isdrefval = -0.1
isdref = np.zeros(len(Tta))
isqref = np.zeros(len(Tta))

isqref1 = 0.0
isqref2 = 0.8
isqref3 = -0.35
for ii in range(len(Tta)):
    isdref[ii] = 0.0
    if (Tta[ii]>=10*2*np.pi):
        isqref[ii] = isqref3
        isdref[ii] = -isdrefval
    elif (Tta[ii]>=1*2*np.pi):
        isqref[ii]=isqref2
        isdref[ii] = isdrefval
    else:
        isqref[ii]=isqref1

me = np.zeros(len(Tta))
x0 = [0, 0, ws] # isd, isq, ws
y0 = [1.0, 0]

kpd = ld*1
Tid = 25.0e1

```

```

kpq = lq*2
Tiq = 25.0e1

Kparamsd = [kpd,Tid]
Kparamsq = [kpq,Tiq]

for ii in range(len(Tta)):
    #Start controller after first step
    if ii>=0:
        eid[ii] = isdref[ii-1] - isd[ii-1]
        PIed = [eid[ii],eid[ii-1]]
        usd[ii] = PIcon(PIed, usd[ii-1], delta_t,Kparamsd)
        eiq[ii] = isqref[ii-1] - isq[ii-1]
        PIeq = [eiq[ii],eiq[ii-1]]
        usq[ii] = PIcon(PIeq, usq[ii-1], delta_t,Kparamsq)
#         usd[ii] = 0
#         usq[ii] = .0001
    if ws>=1.0:
        a = 1.0
    else:
        a = ws
#     usa[ii] = a*np.cos(ws*Tta[ii])
#     usb[ii] = a*np.sin(ws*Tta[ii])
#     usa[ii] = usd[ii-1]*cosdelta[ii-1] - usq[ii-1]*sindelta[ii-1]
#     usb[ii] = usq[ii-1]*cosdelta[ii-1] + usd[ii-1]*sindelta[ii-1]

    params = [usd[ii], usq[ii],w[ii], ws]
    sol1a = odeint(IM_dynstep,x0,[0,delta_t], args = (params,),atol = abserr, rtol= relerr)
    sol1[ii]= sol1a[-1]
#     display(sol1a[-1])
    x0 = sol1a[-1]
#     me[ii] = 1.0*sol1[ii][1] + (ld -lq)*sol1[ii][0]*sol1[ii][1]
#     paramsw = [me[ii],0.0,tmech]
#     sol2a = odeint(rotor_dyndq, y0, [0,delta_t], args = (paramsw,), atol = abserr, rtol= relerr)
#     y0 = sol2a[-1]
#     sol2[ii] = sol2a[-1]
#     #Rotor angle using internal angle
#     FCangle[ii] = sol2[ii][1]
#     gamma[ii] = sol2[ii][1]
#     w[ii] = sol2[ii][0]
#     #using resolver output
# #     FCangle[ii] = sol2[ii]
#     cosdelta[ii] = np.cos(FCangle[ii])
#     sindelta[ii] = np.sin(FCangle[ii])
#Coordinate transformation
    isd[ii] = sol1[ii][0]
    isq[ii] = sol1[ii][1]
#     #Convert stator coordinate current to field coordinates
#     isa[ii] = sol1[ii][0]*cosdelta[ii] - sol1[ii][1]*sindelta[ii]
#     isb[ii] = sol1[ii][1]*cosdelta[ii] + sol1[ii][0]*sindelta[ii]
#Rotor flux in field coordinates Using estimator output as in practice
#actual flux will not be available for measurement
#     psird[ii] = 1.0
#     psirq[ii]= 0.0

# for ii in range(len(Tta)):
#     if ws>=1.0:
#         a = 1.0
#     else:
#         a = ws

```

```

#     usa[ii] = a*np.cos(ws*Tta[ii])
#     usb[ii] = a*np.sin(ws*Tta[ii])
# #     me[ii] = kr*(sol1[ii-1][2]*sol1[ii-1][1] - sol1[ii-1][3]*sol1[ii-1][0]) - mL
#     params = [usa[ii], usb[ii], mL, ws]
#     sol1a = odeint(IM_dynstep, x0, [0, delta_t], args = (params,), atol = abserr, rtol= relerr)
#     sol1[ii]= sol1a[-1]
# #     display(sol1a[-1])
#     x0 = sol1a[-1]
# #     params2 = [sol1[ii][0], sol1[ii][1], sol1[ii][4]] # isa, isb, w
# #     sol2a = odeint(psiest_vr, y0, [0, delta_t], args = (params2,), atol = abserr, rtol= relerr)
# #     sol2[ii] = sol2a[-1]
# #     psircomp = np.complex(sol2[ii][0], sol2[ii][1])
# #     FCangle[ii] = np.arccos(sol2[ii][0]/np.abs(psircomp))
# #     cosdelta[ii] = sol2[ii][0]/np.abs(psircomp)
# #     sindelta[ii] = sol2[ii][1]/np.abs(psircomp)
# #     #Coordinate transformation
# #     #Convert stator coordinate current to field coordinates
# #     isd[ii] = sol1[ii][0]*cosdelta[ii] + sol1[ii][1]*sindelta[ii]
# #     isq[ii] = sol1[ii][1]*cosdelta[ii] - sol1[ii][0]*sindelta[ii]
# #     #Rotor flux in field coordinates
# #     psird[ii] = sol2[ii][0]*cosdelta[ii] + sol2[ii][1]*sindelta[ii]
# #     psirq[ii] = sol2[ii][1]*cosdelta[ii] - sol2[ii][0]*sindelta[ii]
# #     y0 = sol2a[-1]

# isd = sol1[:, 0]
# isq = sol1[:, 1]
# w = sol1[:, 2]
# psra = sol1[:, 2]
# psrb = sol1[:, 3]
# w = sol1[:, 4]
# psiradash = sol2[:, 0]
# psirbdash = sol2[:, 1]

#rs, rr, lh, ls, lr, sig, kr, tr, rk, tk, tmech = mach_para(mach_mb)
rs, l_d, l_q, tmech, psi_rm = mach_para(mach_mc)
me = psi_rm*isq + (l_d-l_q)*isd*isq

```

In [20]:

```

print(isd)
print(isq)
print(me)

```

```

[0.00100436 0.00097342 0.00089229 ... 0.10000011 0.10000011 0.10000011]
[-0.00058619 0.00581419 0.01203562 ... -0.34999994 -0.34999994
 -0.34999994]
[-0.00052742 0.00523128 0.01082922 ... -0.30576344 -0.30576344
 -0.30576344]

```

In [21]:

```

pl.figure(531, figsize = (6,6))
pl.rc('font', size = 16)
pl.subplot(2,1,1)
pl.plot(Tta, isd, 'purple', Tta, isdref, 'skyblue', lw=3)
pl.plot(Tta, isqref, 'tomato', Tta, isq, 'crimson', lw =3)
pl.xlim(0, tend)
pl.ylabel(r'$i_{sd}, i_{sq}$' )
pl.xticks(np.linspace(0, tend, 5))
ax1 = pl.subplot(2,1,2)
ax1.axhline(0)
ax1.plot(Tta, me, 'crimson', lw =3)
ax1.set_yticks(np.linspace(-1.0, 1.0, 5))
ax1.set_ylabel('me [p.u]')
# t1w = np.pi*2*45
# t2w = np.pi*2*60
# pl.xlim(t1w, t2w)
# # pl.xlim(0, tend)
# pl.axhline(0.75)
# pl.axhline(-0.75)
# pl.ylabel(r'$i_{s\alpha}, i_{s\beta}$' )
# pl.xticks(np.linspace(t1w, t2w, 5))
# pl.yticks(np.linspace(-1.0, 1.0, 5))
# pl.ylim(-1.0, 1.0)
# ax1 = pl.subplot(2,1,2)
# ax1.plot(Tta, isd, 'tomato', lw =3, label = r'$i_{sd}$')
# ax1.plot(t1w+1, 0.26, 'o', c = 'tomato')
# pl.text(t1w+2, 0.28, 'i = {0:1.3f}'.format(isd[-1]), fontsize = 12)
# ax1.plot(Tta, isq, 'purple', lw =3, label = r'$i_{sq}$')
# ax1.plot(Tta, psird, 'magenta', lw=2)
# ax1.set_ylabel(r'$i_{sd}, i_{sq}$' )
# ax1.set_xlabel(r'$\omega t$')
# ax2 = ax1.twinx()
# ax2.plot(Tta, FCangle, 'olive', lw =2)
# pl.rcParams['legend.fontsize']=12
# ax1.legend(loc = 'upper left')

#pl.axhline(1.0)
#pl.axhline(-1.0)

# t1w = np.pi*2*45
# t2w = np.pi*2*60
# ax1.set_xlim(t1w, t2w)
# ax1.set_xticks(np.linspace(t1w, t2w, 5))
# ax1.set_yticks(np.linspace(0, 1.0, 6))
# ax1.set_ylim(-0.1, 1.0)
# ax2.set_ylim(0, 3.14)
# pl.savefig(dirfig + "F0Currentsvst.pdf", bbox_inches = 'tight', transparent = True)

# pl.figure(533, figsize = (4,4))
# pl.rc('font', size = 16)
# pl.rcParams['axes.titlesize']=14
# pl.plot(isd, isq, 'crimson', lw =2, label = 'Field currents')
# # pl.plot(psird, psirq, 'blue', lw =4, label = "actual")
# # pl.axhline(0)
# pl.axvline(0)
# pl.arrow(0, 0, psird[-1], psirq[-1], fc = 'magenta', ec = 'magenta', head_width = 0.05\
# , length_includes_head = True, lw =3)
# pl.arrow(0, 0, isd[-1], isq[-1], fc = 'crimson', ec = 'crimson', head_width = 0.05\
# , length_includes_head = True, lw =3)

```



```

# pl.arrow(0,0,isd[-1],0, fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(isd[-1],0,0,isq[-1], fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.text(psiird[-1]-0.05,psirq[-1]+0.08, r'$\vec{\psi}_r$', fontsize=16 )
# pl.xlabel(r'$d$')
# pl.ylabel(r'$q$')
# pl.xlim(-0.01,1.0)
# pl.ylim(-0.02,1.0)
# # pl.xticks(np.linspace(-1.0,1.0,5))
# # pl.axis('equal')
# pl.rcParams['legend.fontsize']=9
# # pl.legend(loc = 'lower left', bbox_to_anchor = (0.8,0))
# pl.title('Field Coordinates')
# pl.savefig(dirfig + "FOXYvectors.pdf", bbox_inches = 'tight', transparent = True)

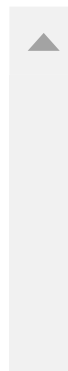
# pl.figure(536, figsize = (4,4))
# pl.rc('font', size = 16)
# pl.rcParams['axes.titlesize']=14
# pl.plot(isd,isq,'crimson', lw =2, label = 'Field currents')
# # pl.plot(psiird, psirq, 'blue', lw =4 , label = "actual")
# # pl.axhline(0)
# # pl.axvline(0)
# pl.arrow(0,0,psiird[-1],psirq[-1], fc = 'magenta', ec = 'magenta', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(0,0,isd[-1],isq[-1], fc = 'crimson', ec = 'crimson', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(0,0,isd[-1],0, fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)
# pl.arrow(isd[-1],0,0,isq[-1], fc = 'coral', ec = 'coral', head_width = 0.05\
#           ,length_includes_head = True, lw =3)

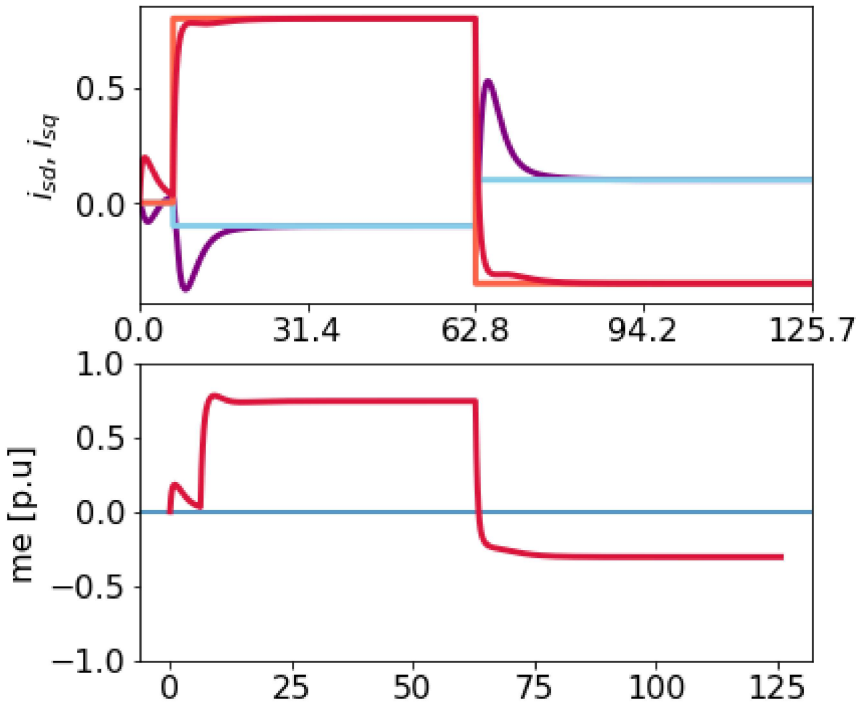
# pl.xlabel(r'$d$')
# pl.ylabel(r'$q$')
# # pl.xlim(0,1.0)
# # pl.ylim(-0.02,1.0)
# # pl.xticks(np.linspace(-1.0,1.0,5))
# pl.axis('equal')
# pl.rcParams['legend.fontsize']=9
# # pl.legend(loc = 'lower left', bbox_to_anchor = (0.8,0))
# pl.title('Field Coordinates')
# # pl.savefig(dirfig + "FOXYvectorsFull.pdf", bbox_inches = 'tight', transparent = True)
# pl.show()
# print(isd[-1], isq[-1])

```

Out[21]:

Text(0, 0.5, 'me [p.u]')





In []:

In []: