

# AAE 451: Group 4

## Design, Build, Fly

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### Design & Analysis

#### Requirements

**Size & Weight:** Fits within a 75x75x150cm storage volume, max 150cm wingspan, and MTOW  $\leq$  6kg.

**Performance & Handling:** Takeoff within 25m, soft-field landing, stable and controllable with or without payload, and easy for an external pilot to fly.

**Payload Requirements:** Must carry at least one 518g steel bar with dimensions 2.75x0.25x6in.

**Setup & Operations:** Transition from storage to flight in under 5 min, payload and battery installed in under 2 min, and battery connects without removing components.

**Power and Budget:** Propulsion system limited to 1000W max electrical power, total budget is \$400.

#### Design Parameters

**Wing Airfoil:** Eppler E423

**Wing Area:** 0.5m<sup>2</sup>

**Aspect Ratio:** 4.5

**MAC:** 0.34m

**Taper Ratio:** 0.6

**Root Chord:** 0.42m

**Tip Chord:** 0.25m

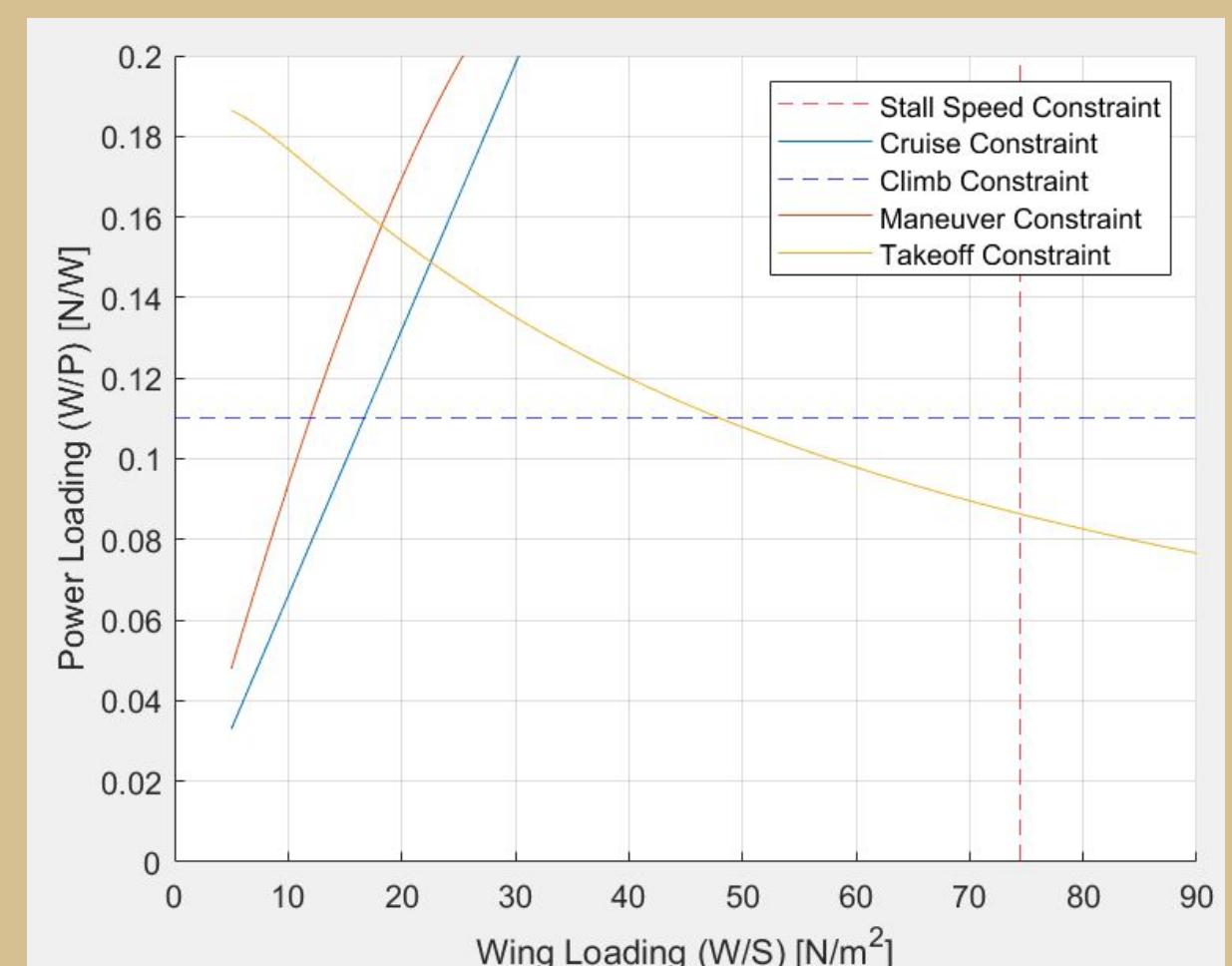
**Wing Sweep:** 0°

**C<sub>D</sub> Trimmed:** 0.03

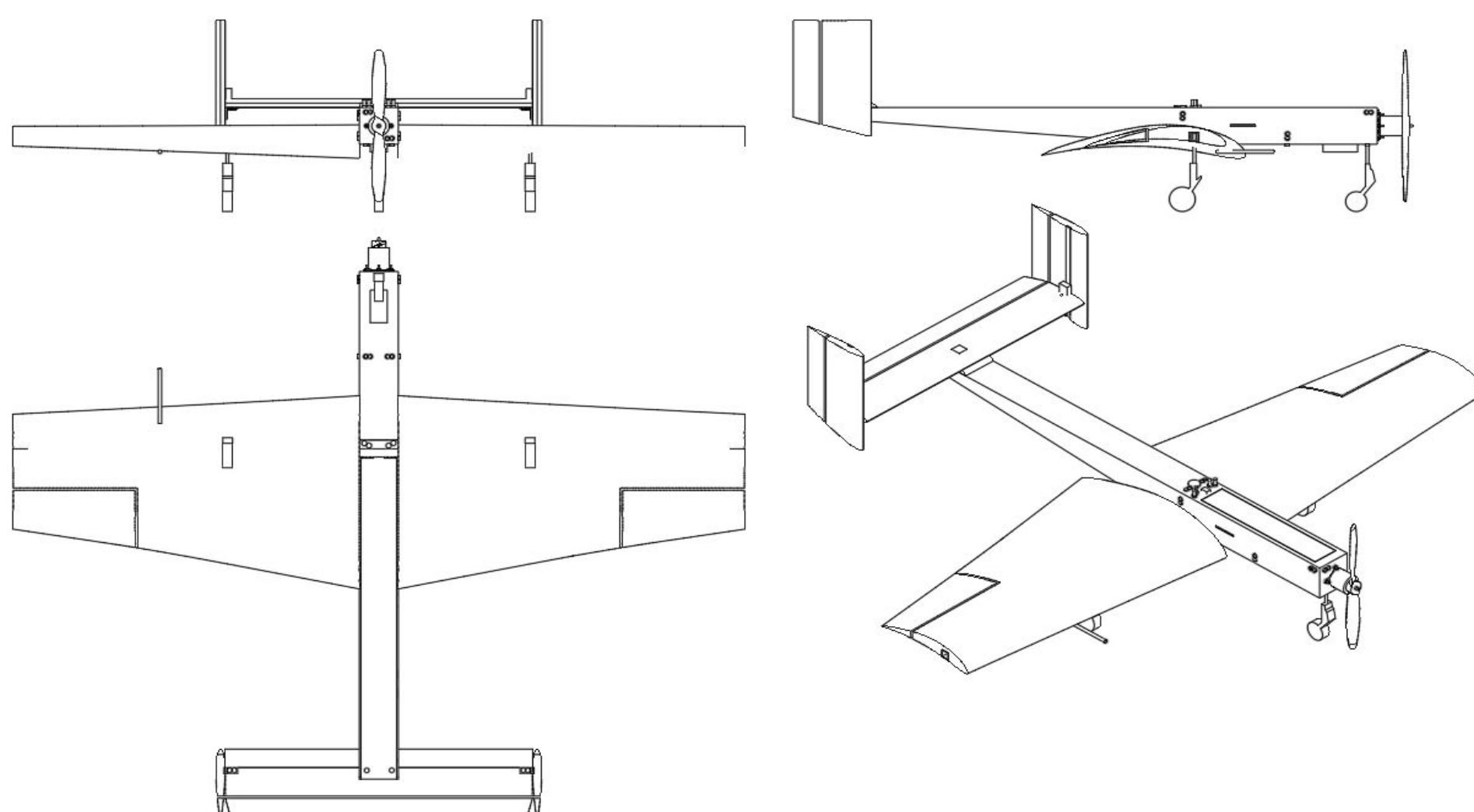
**C<sub>D</sub> Untrimmed:** 0.024

**Tail Airfoil:** NACA 0012

**H-Tail Configuration**



### Aircraft & Manufacturing



#### Weight Table

Parameter	Predicted (kg)	Actual (kg)
Empty Weight	2.875	2.483
Battery Weight	.452	.399
Total Weight	3.845	3.4

#### Propulsion System

**Motor:** Cobra C-3520/14

Brushless Motor

**ESC:** RC Electric Parts Brushless

ESC Classic 4S 60A

**Battery:** Turnigy 4000mAh 4S

30C LiPo Pack

**Propeller:** APC 13x6.5E

#### Manufacturing Changes

- Smaller nuts & bolts
- Full box structure for tail boom
- Tail servos in the center back of tail boom
- ESC moved to the top of the fuselage



### Flight Performance & Results

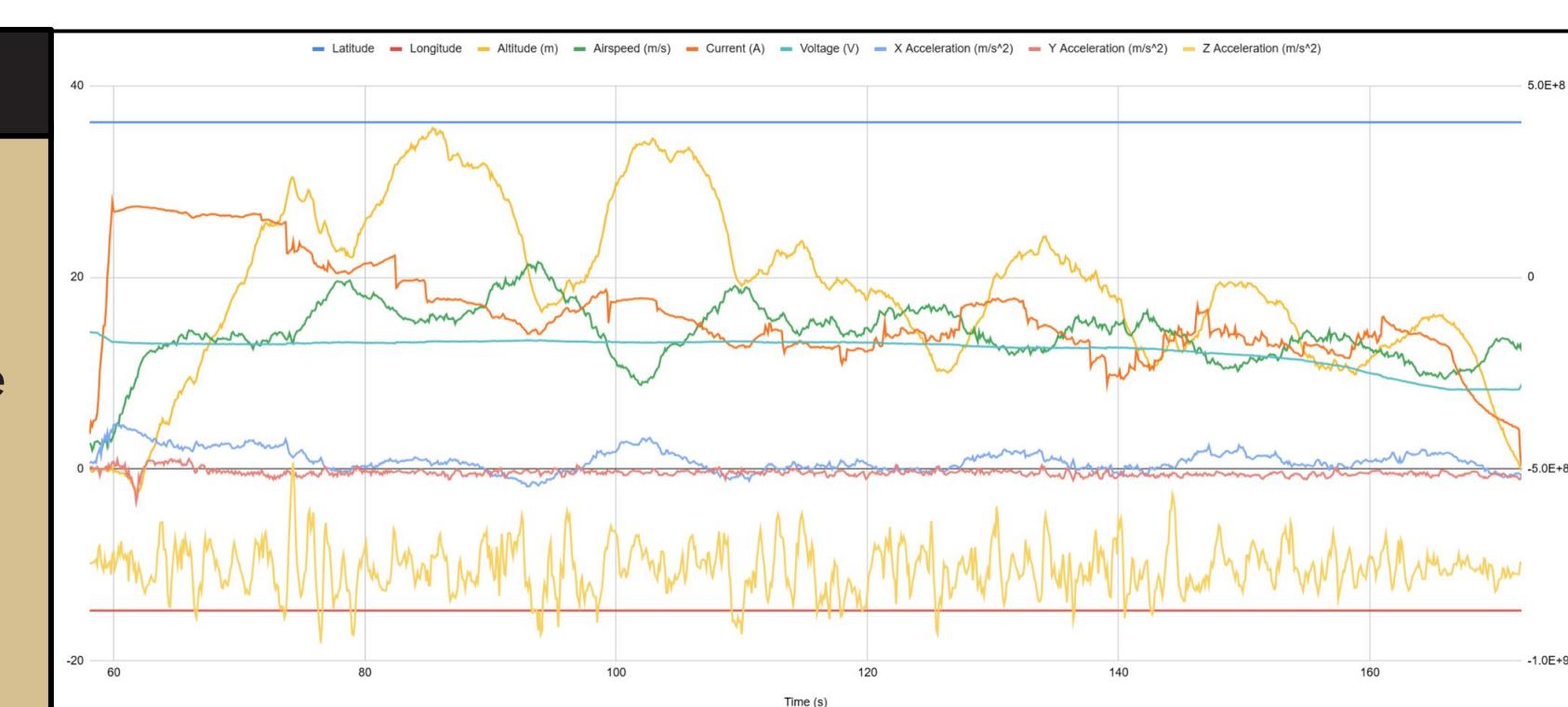
#### Flight Test Issues

##### Tail Flutter:

At high power settings the horizontal & vertical stabilizers would flutter due to insufficiently rigid attachment to fuselage

##### Crash Landing:

An insufficiently charged battery caused a loss of power, resulting in an uncontrolled descent and crash landing



#### Lessons Learned

**Avoid Sharp Internal Corners** in notched components, as they create major stress concentrations and weaken the structure.  
**Wing-spar was over-engineered;** reducing its cross-section would significantly cut weight and improve its fit within the thin wing-tip profile.



#### RFP Requirements Verification

Parameter	PASS or FAIL
Can be stored within 1.5 x 0.75 x 0.75 m box	FAIL
Wingspan under 1.5 m	FAIL (1.51 m)
MTOW under 6 kg	PASS (3.4 kg)
Takeoff in under 25 m	PASS (15.95 m)
Climb to 30 m	PASS (35.6 m)
Successful Soft Field Landing	FAIL (Structural Failure)
Avoid Property Damage	PASS
Under 1000 W of Power Draw	PASS (368.6 W)
Connect Battery in Safe Manner	PASS
Assemble Without Battery / Payload in 5 Minutes	PASS
Install Battery / Payload in Under 2 Minutes	PASS
Stable / Controllable with & without Payload	PASS
Easy to Fly by External Pilot	PASS
Part Cost Under \$400	PASS