## MAE/CEE 5060 Homework Assignment #6 (60 Points) Due 5:00PM, Friday, Mar 26, 2021 Please follow HW Guidelines ©

\*\*Use composite material properties listed on pages 105/225 of the notes\*\*

## (IMPORTANT)

- <u>DO NOT</u> hand in any printed output from your CLT code that will lead <u>NO</u> credits.
- Turn in <u>a summary page</u> showing important values with comments <u>ONLY</u>.
- 1) **(20 Points)** Determine *i*) the total number of plies, *ii*) the number of plies in each ply group, and *iii*) factor of safety (FS) for each given stress resultant for a AS/3501 carbon/epoxy laminate. The laminate must be designed to satisfy the following requirements:
  - a. The laminate should be able to carry  $N_{xx} = 4000$  lb/in and  $N_{xy} = 800$  lb/in
  - b. The laminate should not exceed strains of  $\varepsilon_{xx} = 0.004$  in/in and  $\gamma_{xy} = 0.005$  in/in.
  - c. The ply thickness is  $t_p = 0.006$  in.
  - d. The final layup should be balanced and symmetric.
  - e. The laminate consist of 0°, ±45°, and 90° plies.
  - f. There must be at least 10% of the plies in each of the three major directions.
- 2) (20 Points) <u>Determine</u> the critical value of the normal strain for each of the following cases. Draw a conclusion about the results (meaning that you MUST comment on your results). The following are design requirements:
  - a. Using Scotchply uni-tape with a ply thickness of 0.005 inches
  - b. Use first-ply-failure and the Tsai-Wu failure criterion for strength ratio (R) calculation.

Lay-up	[0/45/90]%	Nxx (lb/in)	E <sub>XX</sub>	R	<b>E</b> xxC
[0 <sub>16</sub> /45/-45/90 <sub>2</sub> ] <sub>S</sub>		1000			
[0 <sub>12</sub> /[45/-45] <sub>3</sub> /90 <sub>2</sub> ]s		1000			
$[0_{10}/[45/-45]_4/90_2]_S$		1000			
[0 <sub>8</sub> /[45/-45] <sub>5</sub> /90 <sub>2</sub> ] <sub>S</sub>		1000			
$[0_6/[45/-45]_5/90_4]_S$		1000			
[0 <sub>2</sub> /[45/-45] <sub>8</sub> /90 <sub>2</sub> ]s		1000			

3) **(20 Points)** A balanced and symmetric laminate must carry  $N_{xx} = 5000$  lb/in and  $N_{xy} = 900$  lb/in. The material is carbon/epoxy uni-tape. Refer to the carpet plots shown below. The allowable strains are  $\varepsilon_{xx} = 0.003$  in/in and  $\gamma_{xy} = 0.005$  in/in. The laminate consists of 0°, ±45°, and 90° plies. There must be at least 8% of the plies in each of the three major directions. Determine the least number of plies (and the best distribution of 0's, 45's, and 90's) needed for the laminate to carry the given loads without exceeding the allowable strains. The ply thickness is 0.005 inches.



