

Final Exam Long

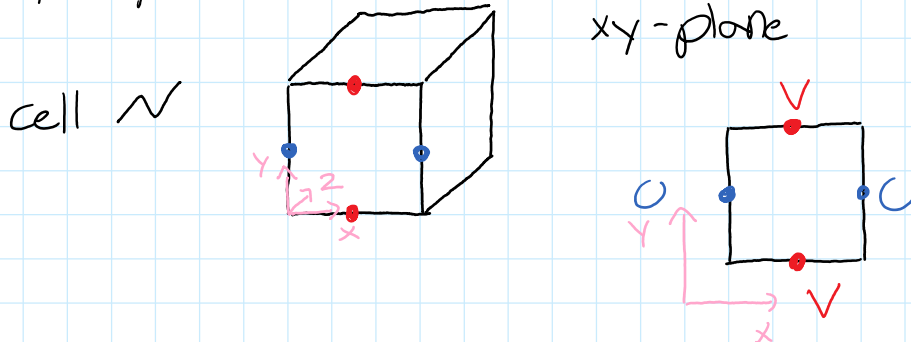
Honor Code
Wir Mitge

#1) IVS in 3D

a) Show where you need store velocity & fluxes.

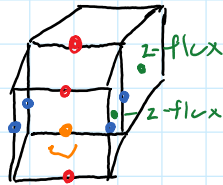
Assume a uniform cube for our mesh cells

Velocity components

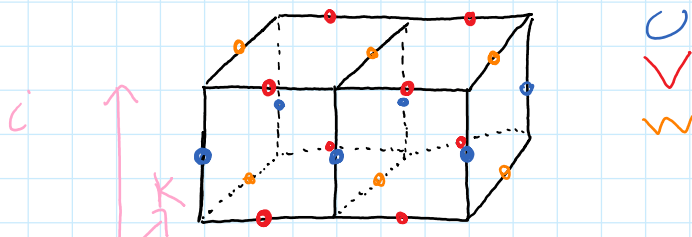


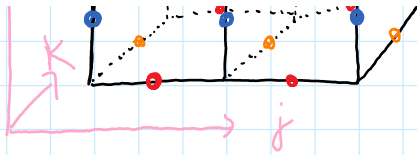
So \vec{u} is stored \perp to x-direction & \vec{v} is stored \perp to y-direction
 \hookrightarrow vertical edge \hookrightarrow horizontal edge

z-direction is \perp to both, so put it in the center of each cell

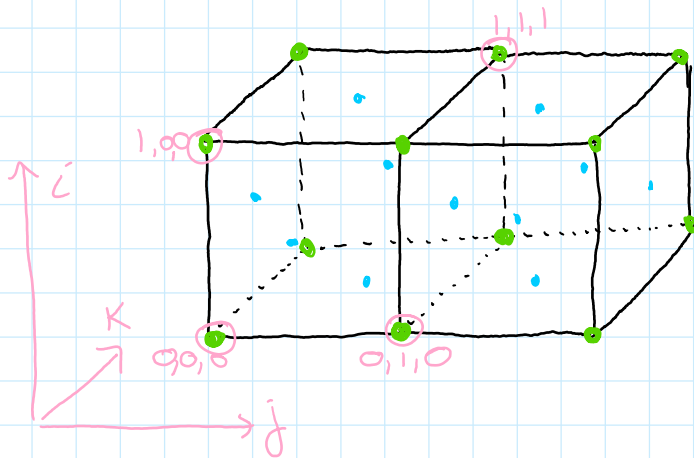


Each cube cell has 4 u & v components, want matching # of w-components for scheme & accuracy, put them on depth edge





The x-y stuff is the same, the z-component is a little different.
All z-forces are at vertices



F^{zz}
 F^{yz} F^{zy}
 F^{xz} F^{zx}
 F^{xy} F^{yx} vertex points

P F^{xx} F^{yy} - center of faces

b) Write update for $w^{n+1/2}$

$$w_t + F_x^{zx} + F_y^{zy} + F_z^{zz} + \cancel{P_z} = 0 \quad \text{ignore pressure}$$

$$\frac{w_{i,j,k+1/2}^{n+1/2} - w_{i,j,k+1/2}^n}{\Delta t} = -F_x^{zx} - F_y^{zy} - F_z^{zz}$$

$$w_{i,j,k+1/2}^{n+1/2} = w_{i,j,k+1/2}^n - \Delta t (F_x^{zx} - F_y^{zy} + F_z^{zz})$$

$$F_z^{zz} = \frac{F_{i,j,k+1}^{zz} - F_{i,j,k}^{zz}}{\Delta z}$$

$$F_x^{zx} = \partial_z - \nabla w_x$$

$$F_y^{zy} = \partial_z - \nabla w_y$$

$$F_x^{zx} = \frac{F_{i,j+1,k}^{zx} - F_{i,j-1,k}^{zx}}{2\Delta x}$$

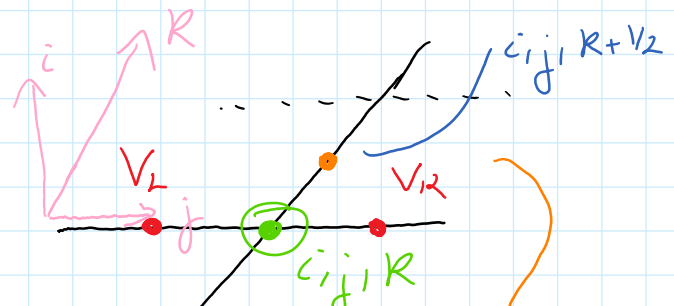
$$F_y^{zy} = \frac{F_{i+1,j,k}^{zy} - F_{i-1,j,k}^{zy}}{2\Delta y}$$

c) Show smart F^{zy} for $q > 0$

$$F^{zy} = \partial_z - \nabla w_y \quad \text{below}$$

$$z = \text{SMART}(\phi_{j-1}, \phi_j, \phi_{j+1})$$

$v > 0$, index towards 0



$v \sim$, index towards v)

$$z = \frac{V_L + V_R}{2} > 0$$

$$\phi_{j-1} = 2c_{ij} R^{-3/2}$$

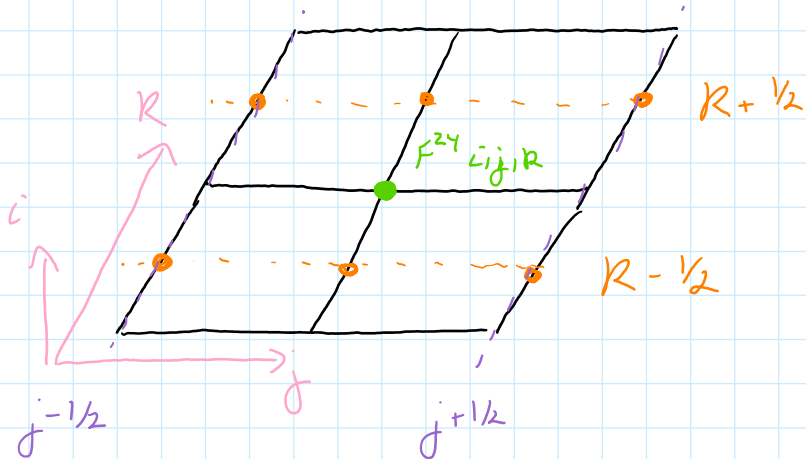
$$\phi_j = 2c_{ij} R^{-1/2}$$

$$\phi_{j+1} = 2c_{ij} R^{+1/2}$$

plug into w/ actual values

3 w nodes

For w_y I don't have points to line up, so I can either Taylor series, or average 2 nearby w_y 's



along $R+1/2$ $w_y = \frac{w_{i,j+1/2,R+1/2} - w_{i,j-1/2,R+1/2}}{2\Delta y}$

along $R-1/2$ $w_y = \frac{w_{i,j+1/2,R-1/2} - w_{i,j-1/2,R-1/2}}{2\Delta y}$

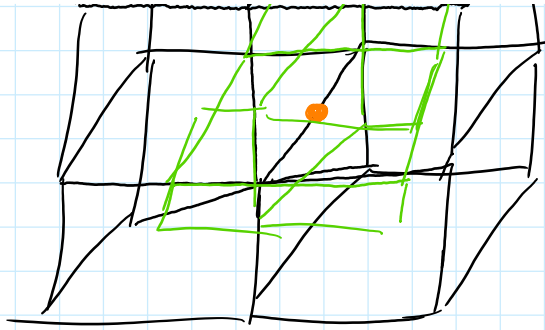
then average these to get w_y ,

$$w_y = \frac{w_{y,R+1/2} + w_{y,R-1/2}}{2}$$

this is for F^2y

d) Write formula & sketch z-momentum residual





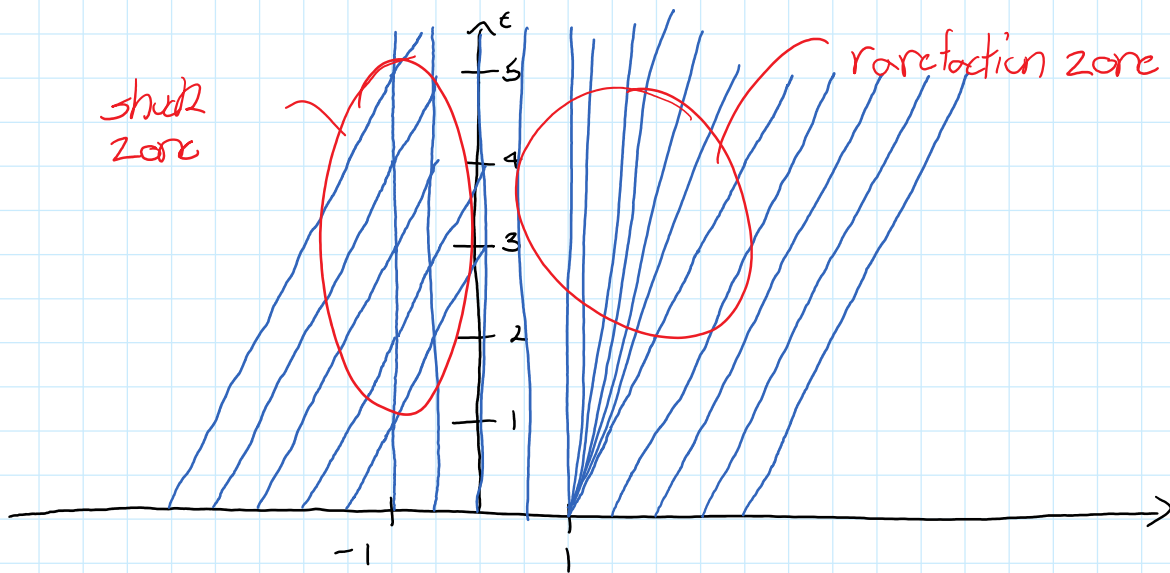
You have this weird pseudo-cell between all other cells for each w .

Need pressure terms, cross momentum & non

$$R_i = \frac{1}{\Delta} [P_{\text{pressure side 1}} - P_{\text{side 2}}] + \Delta F^{zz} + \Delta (\text{cross mom})$$

#2) Traffic Flow

a) Sketch until $t=5$ & include shocks & such

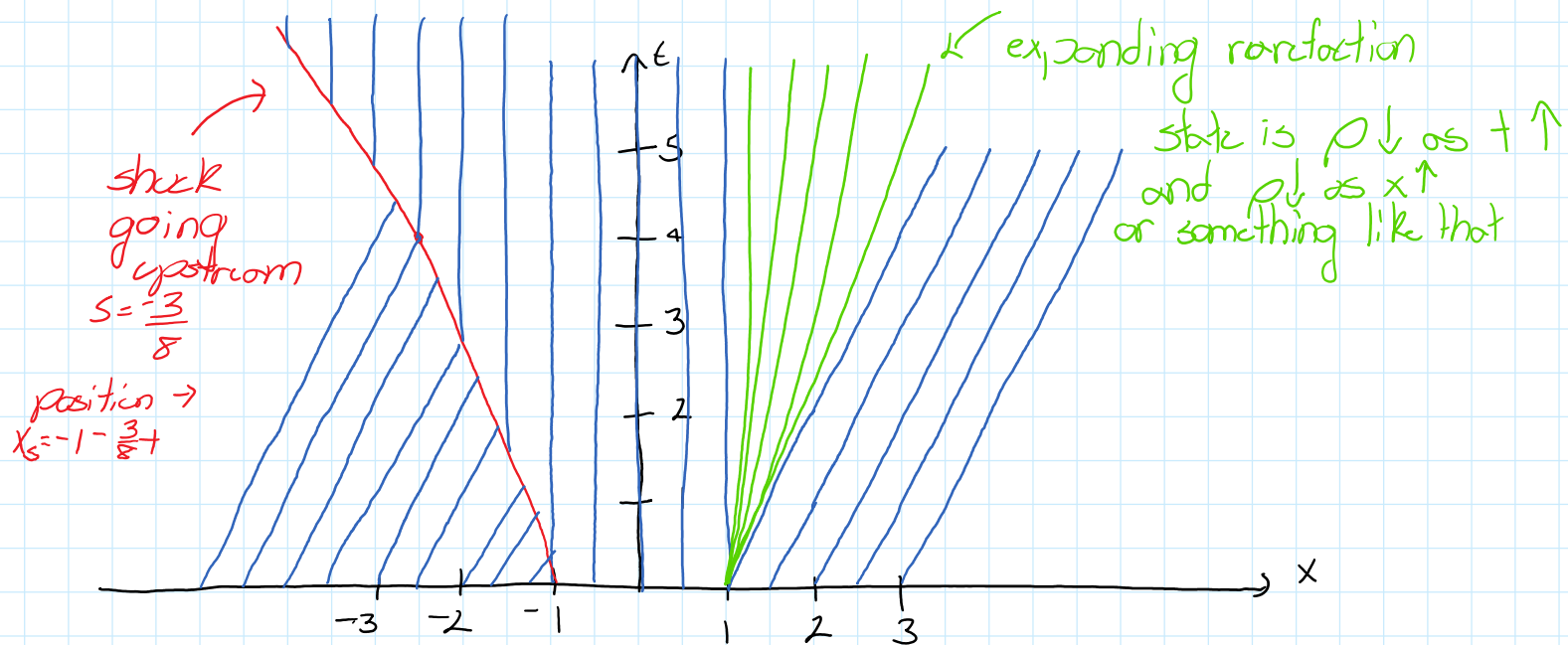


$$s = \frac{f_L - f_R}{u_L - u_R}$$

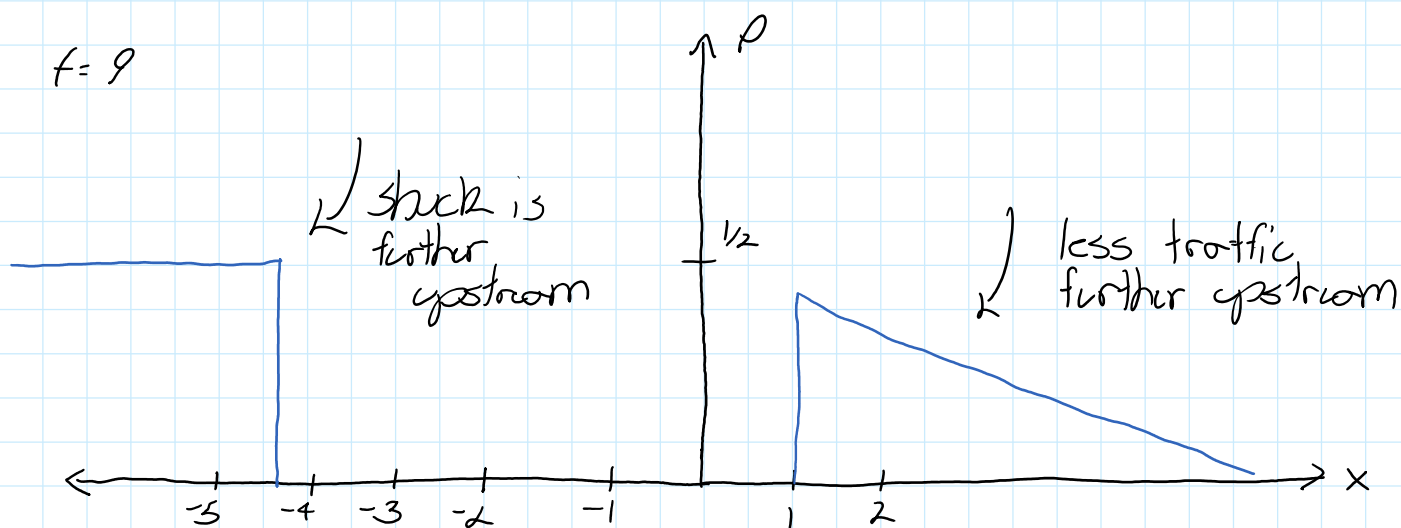
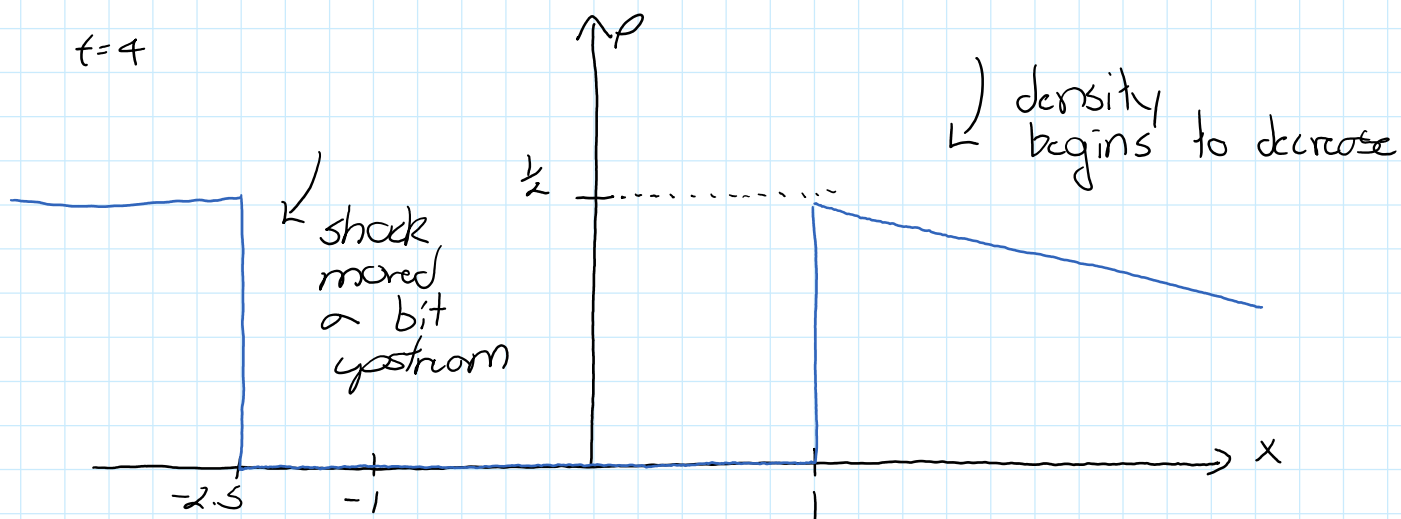
$$f_L = \frac{1}{2} u_{\max} (1 - \frac{1}{2}) = \frac{u_{\max}}{4} = \frac{1}{4}$$

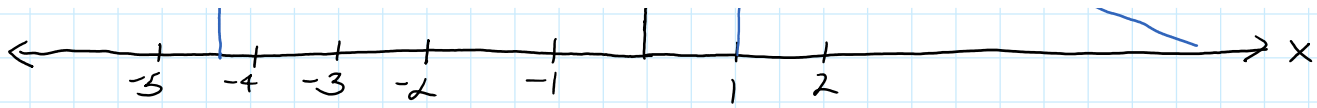
$$f_R = 1 \quad u_L = \frac{1}{2} \quad u_R = 0$$

$$s = \frac{\frac{1}{4} - 1}{\frac{1}{2} - 0} = \frac{-\frac{3}{4}}{\frac{1}{2}} = -\frac{3}{2} \quad \text{shock moves left w/ time}$$



b) Sketch at $t=4$ & $t=9$





Thanks for semester, still don't understand 523 MA, but
sort of did 520 MA

Sad we didn't do any supersonic flows

Hopefully, you teach 623 in the Fall