# Vectorized Matlab Codes for Linear Two-Dimensional Elasticity

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#### Abstract

A vectorized Matlab implementation for the  $P_1$  finite element is provided for the twodimensional linear elasticity with mixed boundary conditions. Vectorization means that there is no loop over triangles. Numerical experiments show that our implementation is more efficient than the standard implementation with a loop over all triangles.

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## 1 Introduction

Matlab is nowadays a widely used tool in education, engineering and research and becomes a standard tool in many areas. But Matlab is a matrix language, that is, Matlab is designed for vector and matrix operations. For best performance in extensive problems, one should take advantage of this.

We propose a Matlab implementation of the  $P_1$  finite element for the numerical solutions of two-dimensional linear elasticity problems. Instead of mixing finite element types (e.g. [2, 1]), we propose a  $P_1$ -triangle vectorized code able to treat medium size meshes in "acceptable" time. Vectorization means that there is no loop over triangles nor nodes. Our implementation needs only Matlab basic distribution functions and can be easily modified and refined.

The paper is organized as follows. The model problem is described in Section 2, followed by a finite element discretization in Section 3. The data representation used in Matlab programs is given in Section 4. The heart of the paper is the assembling functions of the stiffness matrix in Section 5 and the right-hand side in Section 6. Numerical experiments are carried out in Section 7 where post-processing functions are given. Matlab programs used for numerical experiments are given in the appendix.

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## 2 Model problem

We consider an elastic body which occupies, in its reference configuration, a bounded domain  $\Omega$  in  $\mathbb{R}^2$  with a boundary  $\Gamma$ . Let  $\{\Gamma_D, \Gamma_N\}$  be a partition of  $\Gamma$  with  $\Gamma_N$  possibly empty. We assume Dirichlet conditions on  $\Gamma_D$  and Neumann conditions on  $\Gamma_N$ . Let  $u = (u_1, u_2)$  be the two-dimensional displacement field of the elastic body. Under the small deformations assumption, constitutive equations are

$$\sigma_{ij}(u) = 2\mu\epsilon_{ij}(u) + \lambda \operatorname{tr}(\epsilon(u))\mathbb{I}_2, \quad i, j = 1, 2,$$

$$\epsilon(u) = (\nabla u + \nabla u^T)/2,$$
(2.1)

where  $\lambda$  and  $\mu$  denote Lamé (positive) constants. These coefficients are related (in plane deformations) to the Young modulus E and the Poisson coefficient  $\nu$  by

$$\lambda = \frac{\nu E}{(1+\nu)(1-2\nu)}, \quad \mu = \frac{E}{2(1+\nu)}.$$

Given  $f = (f_1, f_2) \in L^2(\Omega)$ ,  $g = (g_1, g_2) \in L^2(\Gamma)$  and  $u_D$ , the problem studied in this paper can be formulated as follows

$$-\operatorname{div}\sigma(u) = f, \quad \text{in } \Omega, \tag{2.2}$$

$$\sigma(u) \cdot n = g, \quad \text{on } \Gamma_N,$$
 (2.3)

$$u = u_D, \text{ on } \Gamma_D.$$
 (2.4)

Let us introduce subspaces

$$V = \left\{ v \in H^1(\Omega) : v = 0 \text{ on } \Gamma_D \right\}, \quad \text{zero boundary}$$
 
$$V^D = \left\{ v \in H^1(\Omega) : v = u_D \text{ on } \Gamma_D \right\}.$$

The variational formulation of (2.2)-(2.3) is

Find  $u \in V^D$  such that

$$\int_{\Omega} \epsilon(u) : \mathbb{C}\epsilon(v) dx = \int_{\Omega} f \cdot v dx + \int_{\Gamma_N} g \cdot v ds, \quad \forall v \in V.$$
 (2.5)

In (2.5),  $\mathbb{C} = (\mathbb{C}_{ijkl})$  is the fourth-order elastic moduli tensor corresponding to (2.1), i.e.

$$\sigma_{ij}(u) = \sum_{k,l=1}^{2} \mathbb{C}_{ijkl} \epsilon_{ij}(u), \quad i, j = 1, 2,$$

where

$$\mathbb{C}_{ijkl} = \lambda \delta_{ij} \delta_{kl} + \mu (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk}), \quad 1 \le i, j, k, l \le 2,$$

 $\delta_{ij}$  being the Kronecker delta.

## 3 Finite element discretization

Let  $\mathcal{T}_h$  be a regular (in the sense of Ciarlet [3]) triangulation of  $\Omega$ . Spaces V and  $V^D$  are then replaced by their discrete approximations  $V_h$  and  $V_h^D$  defined by

$$V_h = \left\{ v_h \in \mathcal{C}^0(\bar{\Omega}); \ v_{|T} \in \mathbb{P}_1(T), \ \forall T \in \mathcal{T}_h; \ v_{h|\Gamma_D} = 0 \right\},$$

$$V_h^D = \left\{ v_h \in \mathcal{C}^0(\bar{\Omega}); \ v_{|T} \in \mathbb{P}_1(T), \ \forall T \in \mathcal{T}_h; \ v_{h|\Gamma_D} = u_D \right\}.$$

where  $\mathbb{P}_1(T)$  is the space of polynomials of degree less or equals to 1 on the triangle T. The discrete version of Eq. (2.5) is then

Find  $u_h \in V_h^D$  such that

$$\int_{\Omega} \epsilon(u_h) : \mathbb{C}\epsilon(v_h) dx = \int_{\Omega} f \cdot v_h dx, \quad \forall v_h \in V_h.$$
 (3.1)

Let  $\{\phi^j\}$  be the system of piecewise global basis functions of  $V_h$ , i.e. for all  $u_h = (u_{1h}, u_{2h}) \in V_h$ 

$$u_{\alpha h} = \sum_{j=1}^{N} \phi^{j}(x) u_{\alpha}^{j}, \quad \alpha = 1, 2.$$

We set  $U = (u_1^1 \ u_2^1 \ u_1^2 \ u_2^2 \ \cdots \ u_1^N \ u_2^N)$ , where  $u_{\alpha}^j$  are nodal values of  $u_h$ , i.e.  $u_{\alpha h}(x_j) = u_{\alpha}^j$ . Applying the standard Galerkin method to (3.1) yields

$$\sum_{i=1}^{N} \left[ \int_{\Omega} \epsilon(\phi_i) : \mathbb{C}\epsilon(\phi_j) dx \right] \underline{U_j} = \int_{\Omega} f \cdot \phi_i dx + \int_{\Gamma_N} g \cdot \phi_i ds, \quad i = 1, \dots, N.$$
 (3.2)

The stiffness matrix  $A = (A_{ij})$  and the right-hand side  $b = b_i$  are then given by

$$A_{ij} = \int_{\Omega} \epsilon(\phi_i) : \mathbb{C}\epsilon(\phi_j) dx,$$

$$b_i = \int_{\Omega} f \cdot \phi_i dx + \int_{\Gamma_N} g \cdot \phi_i ds.$$

The stiffness matrix A is sparse, symmetric and positive semi-definite before incorporating the boundary condition  $u_{|\Gamma_D} = u_D$ .

In practice, integrals in (3.1) are computed as sums of integrals over all triangles, using the fact that  $\bar{\Omega} = \bigcup_{T \in \mathcal{T}_h} T$ 

$$\sum_{T \in \mathcal{T}_h} \int_T \epsilon(u_h) : \mathbb{C}\epsilon(v_h) dx = \sum_{T \in \mathcal{T}_h} \int_T f \cdot v_h dx + \sum_{E \subset \Gamma_N} \int_{\Gamma_N} g \cdot v_h ds, \quad \forall v_h \in V_h.$$

If we set

$$A_{ij}^{(T)} = \int_{T} \epsilon(\varphi_i) : \mathbb{C}\epsilon(\varphi_j) dx$$
 (3.3)

$$b_i^{(T)} = \int_T f \cdot \varphi_i \, dx, \tag{3.4}$$

$$b_i^{(E)} = \int_E g \cdot \varphi_i \, ds, \tag{3.5}$$

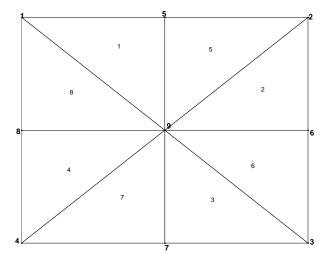


Figure 1: Plot of a triangulation

where  $\{\varphi_i\}$  are linear basis functions of the triangle T or the edge E. Then assembling operations consist of

$$\begin{aligned} A_{ij} &=& \sum_{T \in \mathcal{T}_h} A_{ij}^{(T)} \\ b_i &=& \sum_{T \in \mathcal{T}_h} b_i^{(T)} + \sum_{E \subset \Gamma_N} b_i^{(E)}. \end{aligned}$$

# 4 Data representation of the triangulation

For the mesh, we adopt the data representation used in Matlab PDE Toolbox [4]. Nodes coordinates and triangle vertices are stored in two arrays p(1:2,1:np) and t(1:3,1:nt), where np is the number of nodes and nt the number of triangles. The array t contains for each element the node numbers of the vertices numbered anti-clockwise. For the triangulation of Figure 1, the nodes array p is (np=9)

```
0.0000 1.0000 1.0000 0. 0.5000 1.0000 0.5000 0.0000 0.5000
1.0000 1.0000 0.0000 0. 1.0000 0.5000 0.0000 0.5000
```

and the elements array t is (nt=8)

5	6	7	8	2	3	4	1
1	2	3	4	5	6	7	8
9	9	9	9	9	9	9	9

Neumann boundary nodes are provided by an array ibcneum(1:2,1:nbcn) containing the two node numbers which bound the corresponding edge on the boundary. Then, a sum

over all edges E results in a loop over all entries of ibcneum. Dirichlet boundary conditions are provided by a list of nodes and a list of the corresponding prescribed boundary values.

Note that Matlab supports reading data from files in ASCII format (Matlab function load) and there exists good mesh generators written in Matlab, see e.g. [6].

# 5 Assembling the stiffness matrix

As in [5, 2] we adopt Voigt's representation of the linear strain tensor

$$\gamma(u) = \begin{bmatrix} \partial u_1/\partial x \\ \partial u_2/\partial y \\ \partial u_1/\partial x + \partial u_2/\partial y \end{bmatrix} = \begin{bmatrix} \epsilon_{11}(u) \\ \epsilon_{22}(u) \\ 2\epsilon_{12}(u) \end{bmatrix}.$$

Using this representation, the constitutive equation  $\sigma(u_h) = \mathbb{C}\epsilon(u_h)$  becomes

$$\begin{bmatrix} \sigma_{11}(u_h) \\ \sigma_{22}(u_h) \\ \sigma_{12}(u_h) \end{bmatrix} = C\gamma(u_h), \tag{5.1}$$

where

$$C = \left[ \begin{array}{ccc} \lambda + 2\mu & \lambda & 0 \\ \lambda & \lambda + 2\mu & 0 \\ 0 & 0 & \mu \end{array} \right].$$

Let  $\mathbf{u} = (u_1 \ u_2 \ \cdots \ u_6)^t$  be the vector of nodal values of  $u_h$  on a triangle T. Elementary calculations provide, on a triangle T,

$$\gamma(u_h) = \frac{1}{2|T|} \begin{bmatrix} \varphi_{1,x} & 0 & \varphi_{2,x} & 0 & \varphi_{3,x} & 0 \\ 0 & \varphi_{1,y} & 0 & \varphi_{2,y} & 0 & \varphi_{3,y} \\ \varphi_{1,y} & \varphi_{1,x} & \varphi_{2,y} & \varphi_{2,x} & \varphi_{3,y} & \varphi_{3,x} \end{bmatrix} \begin{bmatrix} u_1 \\ \vdots \\ u_6 \end{bmatrix} =: \frac{1}{2|T|} R\mathbf{u}, \tag{5.2}$$

R similar to B matrix ...

where |T| is the area of the triangle T. From (5.1) and (5.2), it follows that

$$\epsilon(v_h) : \mathbb{C}\epsilon(u_h) = \gamma(v_h)^t C \gamma(u_h) = \frac{1}{4|T|^2} \mathbf{v}^t R^t C R \mathbf{u}$$

The element stiffness matrix is therefore

$$A^{(T)} = \frac{1}{4|T|} R^t C R. (5.3)$$

The element stiffness matrix (5.3) can be computed simultaneously for all indices using the fact that

$$\begin{bmatrix} \nabla \varphi_1^t \\ \nabla \varphi_2^t \\ \nabla \varphi_3^t \end{bmatrix} = \begin{bmatrix} 1 & 1 & 1 \\ x_1 & x_2 & x_3 \\ y_1 & y_2 & y_3 \end{bmatrix}^{-1} \begin{bmatrix} 0 & 0 \\ 1 & 0 \\ 0 & 1 \end{bmatrix} = \frac{1}{2|T|} \begin{bmatrix} y_2 - y_3 & x_3 - x_2 \\ y_3 - y_1 & x_1 - x_3 \\ y_1 - y_2 & x_2 - x_1 \end{bmatrix}.$$

D^-1: cartesian -> barycentric

Function 5.1 Assembly of the stiffness matrix

```
function A=elas2dmat1(p,t,Young,nu)
% Two-dimensional linear elasticity
% Assembly of the stiffness matrix
n=size(p,2); nt=size(t,2); nn=2*n; #dof is 2*#V
% Lame constants
lam=Young*nu/((1+nu)*(1-2*nu)); mu=.5*Young/(1+nu);
% Constant matrices
C=mu*[2,0,0;0,2,0;0,0,1]+lam*[1,1,0;1,1,0;0,0,0];
Zh=[zeros(1,2);eve(2)];
% Loop over all triangles
A=sparse(nn,nn);
for ih=1:nt loop over triangles
    it=t(1:3,ih);
    itt=2*t([1,1,2,2,3,3],ih)-[1;0;1;0;1;0]; indexing scheme into A
   xy=zeros(3,2); xy(:,1)=p(1,it)'; xy(:,2)=p(2,it)';
   D=[1,1,1;xy']; D^-1: cartesian -> barycentric transformation
    gradphi=D\Zh;
   R=zeros(3,6);
   R([1,3],[1,3,5])=gradphi';
   R([3,2],[2,4,6])=gradphi';
    A(itt,itt)=A(itt,itt)+0.5*det(D)*R'*C*R;
end
```

Matlab implementation given in [2, 5] (in a more modular form) can be summarized by the matlab Function 5.1. This implementation, directly derived from compiled languages as Fortran or C/C++, produces a very slow code in Matlab for large size meshes due to the presence of the loop for. Our aim is to remove this loop by reorganizing calculations. Let us introduce the following notations

$$x_{ij} = x_i - x_j, \quad y_{ij} = y_i - y_j, \quad i, j = 1, 2, 3,$$
 (5.4)

so that, from (5.2)

$$R = \left[ \begin{array}{ccccc} y_{23} & 0 & y_{31} & 0 & y_{12} & 0 \\ 0 & x_{32} & 0 & x_{13} & 0 & x_{21} \\ x_{32} & y_{23} & x_{13} & y_{31} & x_{21} & y_{12} \end{array} \right].$$

If we rewrite **u** in the following non standard form  $\mathbf{u} = (u_1 \ u_3 \ u_5 \ u_2 \ u_4 \ u_6)^t$  then the element stiffness matrix can be rewritten as

$$A^{(T)} = \begin{bmatrix} A_{11}^{(T)} & A_{12}^{(T)} \\ A_{21}^{(T)} & A_{22}^{(T)} \end{bmatrix}$$

where  $A_{11}^{(T)}$  and  $A_{22}^{(T)}$  are symmetric and  $A_{21}^{(T)}=(A_{21}^{(T)})^t$ . After laborious (but elementary)

calculations, we get

$$A_{11}^{(T)} = \begin{bmatrix} \widetilde{\lambda} y_{23}^2 + \mu x_{32}^2 & \widetilde{\lambda} y_{23} y_{31} + \mu x_{32} x_{13} & \widetilde{\lambda} y_{23} y_{12} + \mu x_{32} x_{21} \\ \widetilde{\lambda} y_{31} y_{23} + \mu x_{13} x_{32} & \widetilde{\lambda} y_{31}^2 + \mu x_{13}^2 & \widetilde{\lambda} y_{31} y_{12} + \mu x_{13} x_{21} \\ \widetilde{\lambda} y_{12} y_{23} + \mu x_{21} x_{32} & \widetilde{\lambda} y_{12} y_{31} + \mu x_{21} x_{13} & \widetilde{\lambda} y_{12}^2 + \mu x_{21}^2 \end{bmatrix}$$

$$A_{22}^{(T)} = \begin{bmatrix} \widetilde{\lambda}x_{32}^2 + \mu y_{23}^2 & \widetilde{\lambda}x_{32}x_{13} + \mu y_{23}y_{31} & \widetilde{\lambda}x_{32}x_{21} + \mu y_{23}y_{12} \\ \widetilde{\lambda}x_{13}x_{32} + \mu y_{31}y_{23} & \widetilde{\lambda}x_{13}^2 + \mu y_{31}^2 & \widetilde{\lambda}x_{13}x_{21} + \mu y_{31}y_{12} \\ \widetilde{\lambda}x_{21}x_{32} + \mu y_{12}y_{23} & \widetilde{\lambda}x_{21}x_{13} + \mu y_{12}y_{31} & \widetilde{\lambda}x_{21}^2 + \mu y_{12}^2 \end{bmatrix}$$

where  $\widetilde{\lambda} = \lambda + 2\mu$ ; and

$$A_{12}^{(T)} = \begin{bmatrix} \lambda y_{23} x_{32} + \mu x_{32} y_{23} & \lambda y_{23} x_{13} + \mu x_{32} y_{31} & \lambda y_{23} x_{21} + \mu x_{32} y_{12} \\ \lambda y_{31} x_{32} + \mu x_{13} y_{23} & \lambda y_{31} x_{13} + \mu x_{13} y_{31} & \lambda y_{31} x_{21} + \mu x_{13} y_{12} \\ \lambda y_{12} x_{32} + \mu x_{21} y_{23} & \lambda y_{12} x_{13} + \mu x_{21} y_{31} & \lambda y_{12} x_{21} + \mu x_{21} y_{12} \end{bmatrix}.$$

Let us introduce the following vectors

$$\mathbf{x} = \begin{bmatrix} x_{32} \\ x_{13} \\ x_{21} \end{bmatrix} \quad \mathbf{y} = \begin{bmatrix} y_{23} \\ y_{31} \\ y_{12} \end{bmatrix}.$$

One can verify that matrices  $A_{11}^{(T)}$ ,  $A_{22}^{(T)}$  and  $A_{12}^{(T)}$  can be rewritten in a simple form, using  $\mathbf{x}$  and  $\mathbf{y}$ , as

$$A_{11}^{(T)} = (\lambda + 2\mu)\mathbf{y}\mathbf{y}^t + \mu\mathbf{x}\mathbf{x}^t$$

$$A_{22}^{(T)} = (\lambda + 2\mu)\mathbf{x}\mathbf{x}^t + \mu\mathbf{y}\mathbf{y}^t$$

$$A_{12}^{(T)} = \lambda\mathbf{y}\mathbf{x}^t + \mu\mathbf{x}\mathbf{y}^t$$

that is, for  $1 \le i, j \le 3$ 

$$A_{11ij}^{(T)} = (\lambda + 2\mu)\mathbf{y}_i\mathbf{y}_j + \mu\mathbf{x}_i\mathbf{x}_j,$$

$$A_{22ij}^{(T)} = (\lambda + 2\mu)\mathbf{x}_i\mathbf{x}_j + \mu\mathbf{y}_i\mathbf{y}_j,$$

$$(5.5)$$

$$A_{22ij}^{(T)} = (\lambda + 2\mu)\mathbf{x}_i\mathbf{x}_j + \mu\mathbf{y}_i\mathbf{y}_j, \tag{5.6}$$

$$A_{12ij}^{(T)} = \lambda \mathbf{y}_i \mathbf{x}_j + \mu \mathbf{x}_i \mathbf{y}_j \tag{5.7}$$

With Matlab,  $\mathbf{x}_i$  and  $\mathbf{y}_i$  can be computed in a fast way for all triangles using vectorization. Then assembling the stiffness matrix reduces to two *constant* loops for computing  $\mathbf{x}\mathbf{x}^t$ ,  $\mathbf{y}\mathbf{y}^t$  and  $\mathbf{x}\mathbf{y}^t$ . We do not need to assemble separately  $A_{11}^{(T)}$ ,  $A_{22}^{(T)}$  and  $A_{12}^{(T)}$ . Sub matrices (5.5)-(5.7) are directly assembled in the global stiffness matrix, in its standard form, since we know their locations. Matlab vectorized implementation of the Assembly of the stiffness matrix is presented in Function 5.2.

Function 5.2 Assembly of the stiffness matrix: vectorized code

```
function K=elas2dmat2(p,t,Young,nu)
% Two-dimensional finite element method for the Lame Problem
% Assembly of the stiffness matrix
n=size(p,2); nn=2*n;
% Lame constants
lam=Young*nu/((1+nu)*(1-2*nu)); mu=.5*Young/(1+nu);
% area of triangles
it1=t(1,:); it2=t(2,:); it3=t(3,:);
x21=(p(1,it2)-p(1,it1)); x32=(p(1,it3)-p(1,it2)); x31=(p(1,it3)-p(1,it1));
y21=(p(2,it2)-p(2,it1))'; y32=(p(2,it3)-p(2,it2))'; y31=(p(2,it3)-p(2,it1))';
ar=.5*(x21.*y31-x31.*y21);
muh=mu./(4*ar);
lamh=lam./(4*ar);
lamuh=(lam+2*mu)./(4*ar);
clear it1 it2 it3
x=[x32-x31 x21]; y=[-y32 y31-y21];
it1=2*t'-1; it2=2*t';
% Assembly
K=sparse(nn,nn);
for i=1:3
for j=1:3
  K=K+sparse(it1(:,i),it2(:,j),lamh.*y(:,i).*x(:,j)+muh.*x(:,i).*y(:,j),nn,nn);
  K=K+sparse(it2(:,j),it1(:,i),lamh.*y(:,i).*x(:,j)+muh.*x(:,i).*y(:,j),nn,nn);
  K=K+sparse(it1(:,i),it1(:,j),lamuh.*y(:,i).*y(:,j)+muh.*x(:,i).*x(:,j),nn,nn);
  K=K+sparse(it2(:,i),it2(:,j),lamuh.*x(:,i).*x(:,j)+muh.*y(:,i).*y(:,j),nn,nn);
end
end
```

# 6 Assembling the right-hand side

We assume that the volume forces  $f = (f_1, f_2)$  are provided at mesh nodes. The integral (3.5) is approximated as follows

$$\int_T f \cdot \phi_i dx \approx \frac{1}{6} \det \left| \begin{array}{cc} x_2 - x_1 & x_3 - x_1 \\ y_2 - y_1 & y_3 - y_1 \end{array} \right| f_j(x_c, y_c), \quad j = \operatorname{mod}(i - 1, 2) + 1$$

where  $(x_c, y_c)$  is the center of mass of the triangle T. With the assumption on f,

$$f_j(\mathbf{x}_s, \mathbf{y}_s) = (f_j(\mathbf{x}_1, \mathbf{y}_1) + f_j(\mathbf{x}_2, \mathbf{y}_2) + f_j(\mathbf{x}_3, \mathbf{y}_3))/3$$
 (6.1)

where  $\{(x_i, y_i)\}_{i=1,3}$  are vertices of the triangle T. Using the notation convention (5.4), we have

$$\begin{vmatrix} x_2 - x_1 & x_3 - x_1 \\ y_2 - y_1 & y_3 - y_1 \end{vmatrix} = x_{21}y_{31} - x_{31}y_{12}.$$
 (6.2)

We can compute (6.1) and (6.2) over all triangles using vectorization techniques, and assemble the result with the Matlab function sparse. Matlab implementation of the assembly of the volume forces is presented in Function 6.1.

#### Function 6.1 Assembly of the right-hand side: body forces

```
function f=elas2drhs1(p,t,f1,f2)
%-----
% Two-dimensional linear elasticity
% Assembly of the right-hand side 1: body forces
n=size(p,2); nn=2*n;
% triangle vertices
it1=t(1,:); it2=t(2,:); it3=t(3,:);
% edge vectors
a21=p(:,it2)-p(:,it1); a31=p(:,it3)-p(:,it1);
% area of triangles
area=abs(a21(1,:).*a31(2,:)-a21(2,:).*a31(1,:))/2;
% assembly
f1h=(f1(it1)+f1(it2)+f1(it3)).*area'/9;
f2h=(f2(it1)+f2(it2)+f2(it3)).*area'/9;
ff1=sparse(it1,1,f1h,n,1)+sparse(it2,1,f1h,n,1)+sparse(it3,1,f1h,n,1);
ff2=sparse(it2,1,f2h,n,1)+sparse(it2,1,f2h,n,1)+sparse(it3,1,f2h,n,1);
% right-hand side
f=zeros(nn,1);
f(1:2:nn)=full(ff1);
f(2:2:nn)=full(ff2);
```

Integrals involving Neumann conditions are approximated using the value of g at the center  $(x_c, y_c)$  of the edge E

$$\int_{E} g \cdot \phi_i dx \approx \frac{1}{2} |E| g_j(x_c, y_c), \quad j = \text{mod}(i - 1, 2) + 1$$

As for volume forces, |E| and  $g_j(x_c, y_c)$  are computed over all triangles using vectorization techniques and assembled using Matlab function sparse, Function 6.2.

# 7 Numerical experiments

In elasticity, it is usual to display undeformed and deformed meshes. In the graphical representation of the deformed mesh, a magnification factor is used for the displacement.

Function 6.2 Assembly of the right-hand side: Neumann boundary conditions

```
function g=elas2drhs2(p,ineum,g1,g2)
% Two-dimensional linear elasticity
% Assembly of the right-hand side 2: Neumann condition
n=size(p,2); nn=2*n;
ie1=ineum(1,:); ie2=ineum(2,:);
ibcn=union(ie1,ie2);
ne=length(ibcn);
% edge lengths
exy=p(:,ie2)-p(:,ie1);
le=sqrt(sum(exy.^2,1));
% g1,g2 at the center of mass
g1h=(g1(ie1)+g1(ie2)).*le/4;
g2h=(g2(ie1)+g2(ie2)).*le/4;
% assembly
gg1=sparse(ne,1);
gg1=sparse(ie1,1,g1h,ne,1)+sparse(ie2,1,g1h,ne,1);
gg2=sparse(ne,1);
gg2=sparse(ie1,1,g2h,ne,1)+sparse(ie2,1,g2h,ne,1);
% right-hand side
g=zeros(nn,1);
g(2*ibcn-1)=full(gg1);
g(2*ibcn) =full(gg2);
```

The Matlab function 7.1 displays the deformed mesh with an additional function sf issued from post-processing. It is directly derived from the Matlab function Show given in [2]. The additional function can be stresses, strains, potential energy,... evaluated at mesh nodes.

In our numerical experiments, the additional function used in the function 7.1 is either the Von Mises effective stress or the shear energy density  $|\text{dev}\sigma_h|^2/(4\mu)$ , where

$$|\text{dev}\sigma_h|^2 = \left(\frac{1}{2} + \frac{\mu^2}{6(\mu + \lambda)}\right) (\sigma_{h11} + \sigma_{h22})^2 + 2(\sigma_{h12}^2 - \sigma_{h11}\sigma_{h22}).$$

The stress tensor  $\sigma_h$  is computed at every node as the mean value of the stress on the corresponding patch. Matlab function 7.2 calculates approximate shear energy density and Von mises effective stress.

In all examples, Dirichlet boundary conditions are taken into account by using large spring constants (i.e. penalization).

#### Function 7.1 Visualization function

```
function elas2dshow(p,t,u,magnify,sf)
%
%-----
% Two-dimensional linear elasticity
% Visualization
%-----
%
np=size(p,2);
uu=reshape(u,2,np)'; 2x#V vertex-wise displacement
pu=p(1:2,:)'+magnify*uu;
colormap(1-gray)
trisurf(t(1:3,:)',pu(:,1),pu(:,2),zeros(np,1),sf,'facecolor','interp')
view(2)
```

### 7.1 L-shape

The L-shape test problem is a common benchmark problem for which the exact solution is known. The domain  $\Omega$  is described by the polygon

$$(-1,-1), (0,-2), (2,0), (0,2), (-1,-1), (0,0)$$

The exact solution is known in polar coordinates  $(r, \theta)$ ,

$$u_r(r,\theta) = \frac{1}{2\mu} r^{\alpha} \left[ (c_2 - \alpha - 1)c_1 \cos((\alpha - 1)\theta) - (\alpha + 1)\cos((\alpha + 1)\theta) \right],$$
 (7.1)

$$u_{\theta}(r,\theta) = \frac{1}{2\mu} r^{\alpha} \left[ (\alpha+1)\sin((\alpha+1)\theta) + (c_2 + \alpha - 1)c_1\sin((\alpha-1)\theta) \right].$$
 (7.2)

The exponent  $\alpha$  is the solution of the equation

$$\alpha \sin(2\omega) + \sin(2\omega\alpha) = 0$$

with  $\omega = 3\pi/4$  and  $c_1 = -\cos((\alpha + 1)\omega)/\cos((\alpha - 1)\omega)$ ,  $c_2 = 2(\lambda + 2\mu)/(\lambda + \mu)$ . The displacement field (7.1)-(7.2) solves (2.2)-(2.3) with f = 0 and  $u_D = (u_r, u_\theta)$  on  $\Gamma = \Gamma_D$ . Numerical experiments are carried out with Young's modulus E = 100000 and Poisson's coefficient  $\nu = 0.3$ .

We first compare the performances of Matlab functions elas2dmat1 and elas2dmat2 for assembling the stiffness matrix of the L-shape problem. To this end, various meshes of the L-shape are generated and we use Matlab commands tic and toc to compute the elapsed time. Table 1 shows the performances of assembling functions 5.1-5.2. One can notice that the saving of computational time, with the vectorized function elas2dmat2, is considerable. Figure 2 shows the deformed mesh (321 nodes and 400 triangles) of the L-shape with a displacement field multiply by a factor 3000. The grey tones visualize the approximate shear energy density

Mesh	CPU time (in Sec.)			
Triangles/Nodes	elas2dmat1	elas2dmat2		
100/66	0.0236	0.0182		
400/231	0.1133	0.0357		
1600/861	0.7863	0.1791		
6400/3321	25.4558	0.7303		
25600/13041	607.643	3.5527		
102400/51681	$1.1183 \times 10^4$	13.1036		

Table 1: Performances of assembling functions  ${\tt elas2dmat1}$  and  ${\tt elas2dmat2}$ 

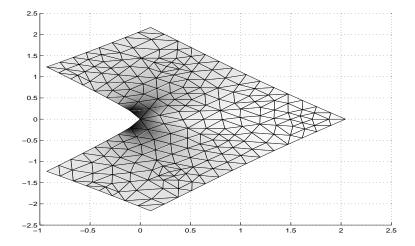


Figure 2: Deformed mesh for the L-shape problem

Function 7.2 Matlab function for computing the shear energy density Sed and the Von Mises effective stress Vms

```
function [Sed, Vms] = elas2dsvms(p,t,uu,E,nu)
%-----
% Two-dimensional linear elasticity
% Sed Shear energy density
% Vms Von Mises effective stress
%______
n=size(p,2); nn=2*n;
% Lame constant
lam=E*nu/((1+nu)*(1-2*nu)); mu=E/(2*(1+nu));
% area of traingles
it1=t(1,:); it2=t(2,:); it3=t(3,:);
x21=(p(1,it2)-p(1,it1)); x32=(p(1,it3)-p(1,it2)); x31=(p(1,it3)-p(1,it1));
y21=(p(2,it2)-p(2,it1))'; y32=(p(2,it3)-p(2,it2))'; y31=(p(2,it3)-p(2,it1))';
ar=.5*(x21.*y31-x31.*y21);
% gradient of scalar basis functions
phi1=[-y32./(2*ar) x32./(2*ar)];
phi2=[ y31./(2*ar) -x31./(2*ar)];
phi3=[-y21./(2*ar) x21./(2*ar)];
% displacements
u=uu(1:2:end); v=uu(2:2:end);
uh=[u(it1) u(it2) u(it3)]; vh=[v(it1) v(it2) v(it3)];
% strains
e11=uh(:,1).*phi1(:,1)+uh(:,2).*phi2(:,1)+uh(:,3).*phi3(:,1);
e22=vh(:,1).*phi1(:,2)+vh(:,2).*phi2(:,2)+vh(:,3).*phi3(:,2);
e12=uh(:,1).*phi1(:,2)+uh(:,2).*phi2(:,2)+uh(:,3).*phi3(:,2)...
   +vh(:,1).*phi1(:,1)+vh(:,2).*phi2(:,1)+vh(:,3).*phi3(:,1);
clear uh vh
% stresses
sig11=(lam+2*mu)*e11+lam*e22; sig22=lam*e11+(lam+2*mu)*e22; sig12=mu*e12;
clear e11 e22 e12
% area of patches
arp=full(sparse(it1,1,ar,n,1)+sparse(it2,1,ar,n,1)+sparse(it3,1,ar,n,1));
% mean value of stresses on patches
sm1=ar.*sig11; sm2=ar.*sig22; sm12=ar.*sig12;
s1=full(sparse(it1,1,sm1,n,1)+sparse(it2,1,sm1,n,1)+sparse(it3,1,sm1,n,1));
s2=full(sparse(it1,1,sm2,n,1)+sparse(it2,1,sm2,n,1)+sparse(it3,1,sm2,n,1));
s12=full(sparse(it1,1,sm12,n,1)+sparse(it2,1,sm12,n,1)+sparse(it3,1,sm12,n,1));
s1=s1./arp; s2=s2./arp; s12=s12./arp;
% Shear energy density
Sed=((.5+mu*mu/(6*(mu+lam)^2))*(s1+s2).^2+2*(s12.^2-s1.*s2))/(4*mu);
% Von Mises effective stress
delta=sqrt((s1-s2).^2+4*s12.^2); sp1=s1+s2+delta; sp2=s1+s2-delta;
Vms=sqrt(sp1.^2+sp2.^2-sp1.*sp2);
```

#### 7.2 A membrane problem

We consider a two-dimensional membrane  $\Omega$  described by the polygon

$$(0,-5), (35,-2), (35,2), (0,5)$$

with E=30000 and  $\nu=0.4$ . The membrane is clamped at x=0 and subjected to a a volume force f=(0,-.75) and shearing load g=(0,10) at x=35. Figures 3-4 show initial (undeformed) and deformed configurations of the membrane, for a mesh with 995 nodes (i.e. 1990 degrees of freedom). The magnification factor, for the displacement field, is 20. The grey tones visualize the approximate Von Mises effective stress. The deformed configuration shows peak stresses at corners (0,-5) and (0,5) as expected.

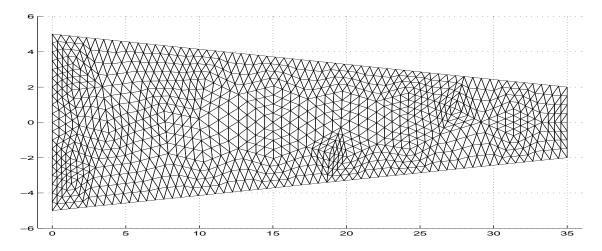


Figure 3: Elastic Membrane: initial configuration

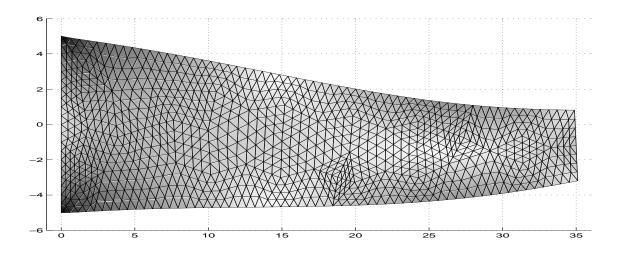


Figure 4: Elastic membrane: deformed configuration

# **Appendix**

## Main program for the L-shape problem

```
%----- Two-Dimensional linear elasticity
      Example 1: L-Shape problem
%-----
% Elastic constants
E=1e5; nu=0.3;
lam=E*nu/((1+nu)*(1-2*nu)); mu=E/(2*(1+nu));
Penalty=10^20;
% Mesh
load lshape231 p t ibcd
n=size(p,2); nn=2*n;
% Exact solution
ue=zeros(nn,1);
[th,r]=cart2pol(p(1,:),p(2,:));
alpha=0.544483737; omg=3*pi/4; ralpha=r.^alpha/(2*mu);
C2=2*(lam+2*mu)/(lam+mu); C1=-cos((alpha+1)*omg)/cos((alpha-1)*omg);
ur=ralpha.*(-(alpha+1)*cos((alpha+1)*th)+(C2-alpha-1)*C1*cos((alpha-1)*th));\\
ut=ralpha.*( (alpha+1)*sin((alpha+1)*th)+(C2+alpha-1)*C1*sin((alpha-1)*th));
ue(1:2:end)=ur.*cos(th)-ut.*sin(th);
ue(2:2:end)=ur.*sin(th)+ut.*cos(th);
% Boundary conditions
nnb=2*length(ibcd);
ibc=zeros(nnb,1); ibc(1:2:end)=2*ibcd-1; ibc(2:2:end)=2*ibcd;
ubc=zeros(nnb,1); ubc(1:2:end)=ue(2*ibcd-1); ubc(2:2:end)=ue(2*ibcd);
% Assembly of the Stiffness matrix
K=elas2dmat1(p,t,E,nu);
% Right-hand side
f=zeros(nn,1);
% Penalization of Stiffness matrix and right-hand sides
K(ibc,ibc)=K(ibc,ibc)+Penalty*speye(nnb);
f(ibc)=Penalty*ubc;
% Solution by Gaussian elimination
\% Shear energy density & Von Mises effective stress
[Sh,Vms]=elas2dsvms(p,t,u,E,nu);
% Show the deformed mesh and shear energy density
elas2dshow(p,t,u,3000,Sh)
```

#### Main program for the membrane problem

```
%----- Two-Dimensional linear elasticity
% Example 2: Elastic membrane
%-----
% Elastic constants
E=30000; nu=0.4;
%
```

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```
% Mesh
load exple2mesh995 p t ibcd ibcneum
n=size(p,2); nn=2*n;
% Boundary conditions
nnb=2*length(ibcd);
ibc=zeros(nnb,1); ibc(1:2:end)=2*ibcd-1; ibc(2:2:end)=2*ibcd;
% Assembly of the Stiffness matrix
K=elas2dmat2(p,t,E,nu);
% Right-hand side: body forces
f1=zeros(n,1); f2=-.75*ones(n,1);
f=elas2drhs1(p,t,f1,f2);
% Right-hand side: Neumann forces
ibcn1=ibcneum(1,:); ibcn2=ibcneum(2,:); ibcn=union(ibcn1,ibcn2);
g1=zeros(n,1); g2=zeros(n,1); g2(ibcn)=10;
g=elas2drhs2(p,ibcneum,g1,g2);
clear ibcn ibcn1 ibcn2
% Penalization of Stiffness matrix and right-hand sides
K(ibc,ibc)=K(ibc,ibc)+Penalty*speye(nnb);
b=f+g; b(ibc)=0;
% Solution by Gaussian elimination
u=K\setminus b;
% Shear energy density & Von Mises effective stress
[Sh,Vms]=elas2dsvms(p,t,u,E,nu);
% Show the deformed mesh and shear energy density
elas2dshow(p,t,u,20,Vms)
```

#### References

Penalty=10^15;

- [1] Alberty J., Carstensen C. and Funken S.A. Remarks around 50 lines of matlab: short finite element implementation. *Numer. Algorithms*, 20:117–137, 1999.
- [2] Alberty J., Carstensen C., Funken S.A. and Klose R. Matlab implementation of the finite element method in elasticity. *Computing*, 69:239–263, 2002.
- [3] CIARLET P.-G. The Finite Element Method for Elliptic Problems. North-Holland, Amsterdam, 1979.
- [4] Langemyr L. et al. Partial Differential Equations Toolbox User's Guide. The Math Works, Inc., 1995.
- [5] KWON Y.W. and Bang H. *The Finite Element Method Using MATLAB*. CRC Press, New York, 2000.
- [6] Persson P.-O. and Strang G. A simple mesh generator in Matlab. *SIAM Rev.*, 42:329–345, 2004.