

# MIET2419: Mechanics and Materials 1

## Stress & Strain

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# Introduction

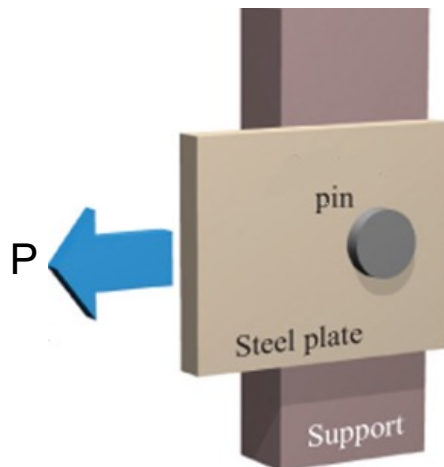
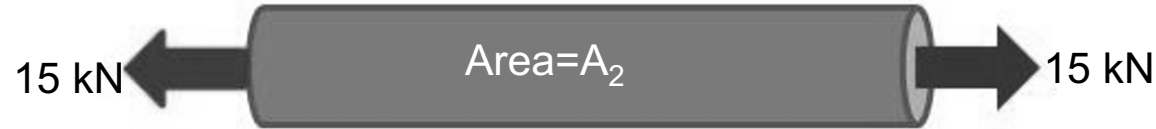
- How to compare two loaded rods under different external loads?



**Which rod will fail?**

What information do we need to find the answer?

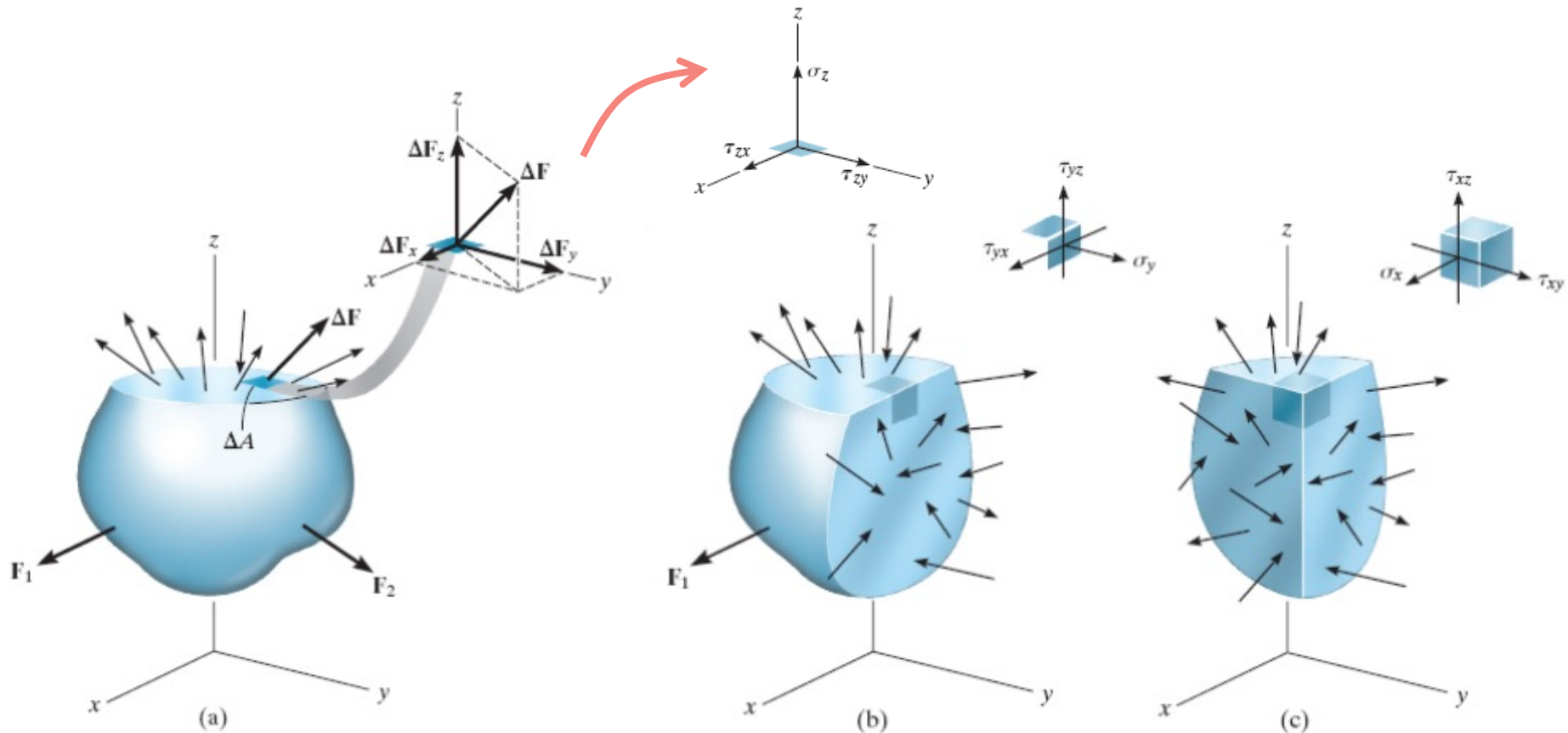
How can we define comparable parameters?



- By increasing  $P$ , what types of failure may happen in the structure?

# Stress

- **Distribution** of internal loading is important in mechanics of materials.
- We will consider the material to be **continuous**.
- The **intensity** of **internal force** at a point is called **stress** = Force/Area

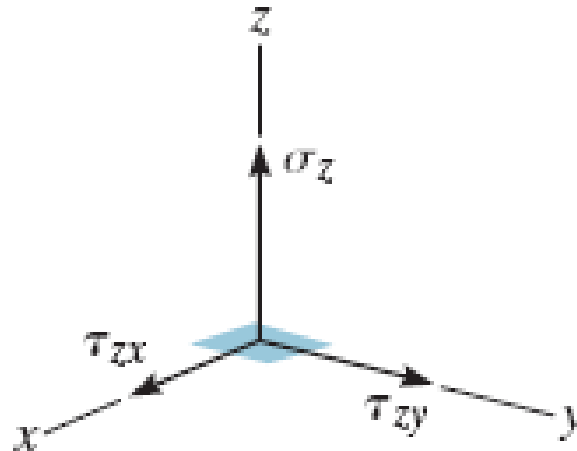


# Stress

- **Normal Stress  $\sigma$ :**

- Force per unit area acting normal to  $\Delta A$

$$\sigma_z = \lim_{\Delta A \rightarrow 0} \frac{\Delta F_z}{\Delta A}$$



## What's the unit of stress?

Keep the units consistent:

- N, mm<sup>2</sup>, MPa
- N, m<sup>2</sup>, Pa

- **Shear Stress  $\tau$  :**

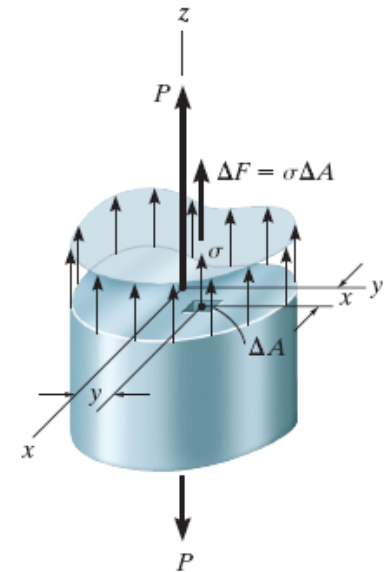
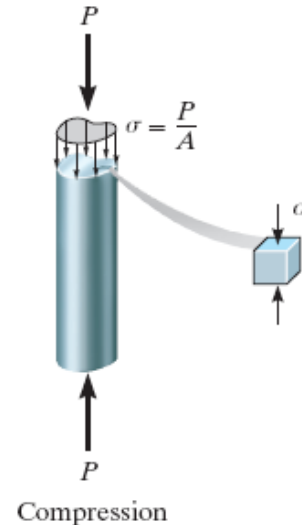
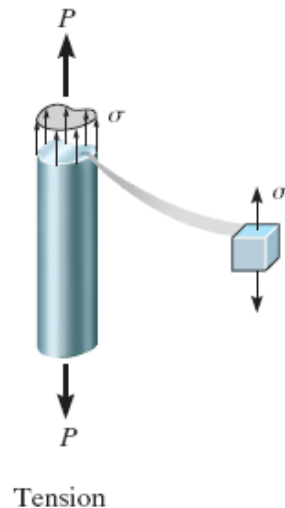
- Force per unit area acting tangent to  $\Delta A$

$$\tau_{zx} = \lim_{\Delta A \rightarrow 0} \frac{\Delta F_x}{\Delta A}$$

$$\tau_{zy} = \lim_{\Delta A \rightarrow 0} \frac{\Delta F_y}{\Delta A}$$

# Average Normal Stress in an Axially Loaded Bar

- When a **cross-sectional area** bar is subjected to axial force through the centroid, it is only subjected to normal stress.
- Stress is assumed to be averaged over the area.



$$\int dF = \int_A \sigma dA$$

$$P = \sigma A$$

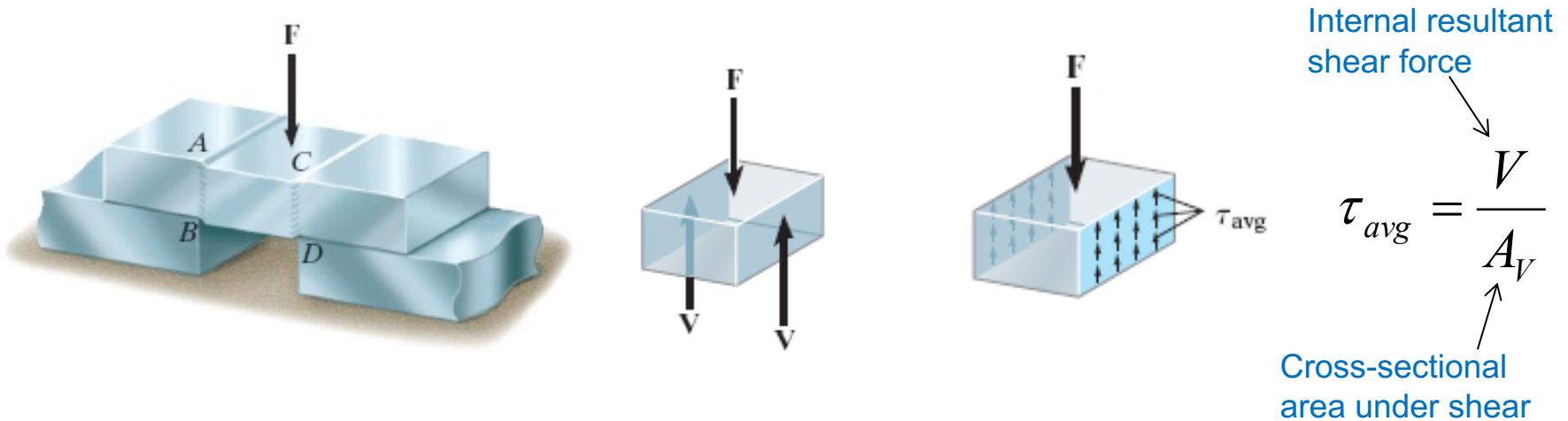
Force

Average Stress  $\rightarrow \sigma = \frac{P}{A}$  Area

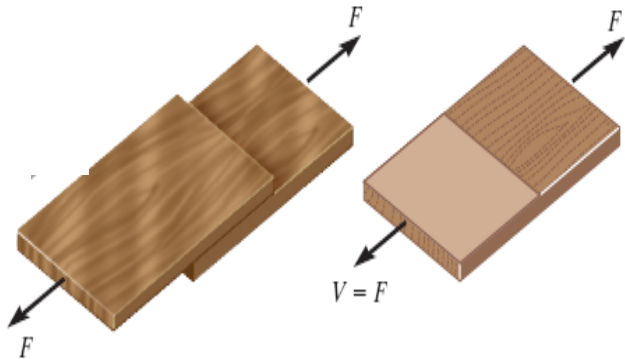
- Force distribution diagrams

# Average Shear Stress

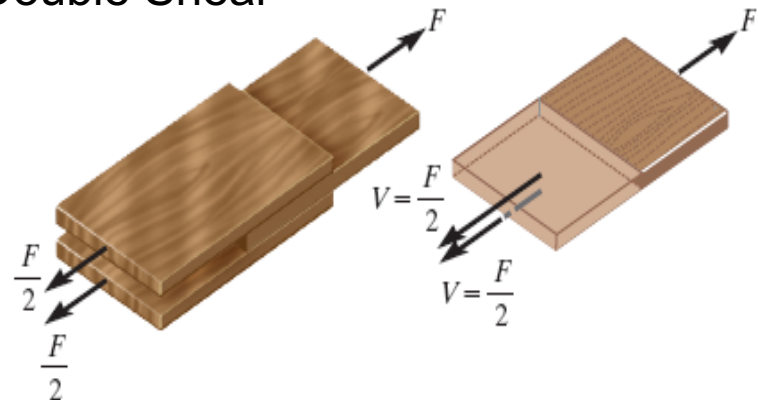
- The **average shear stress** distributed over each sectioned area that develops a shear force.



a) Single Shear



b) Double Shear



# Allowable Stress

- Many unknown factors that influence the actual stress in a member.
- A *factor of safety* is needed to obtain the allowable load.
- The ***factor of safety*** (F.S.) is a ratio of the failure load divided by the allowable load:

$$F.S. = \frac{F_{fail}}{F_{allow}}$$

$$F.S. = \frac{\sigma_{fail}}{\sigma_{allow}}$$

$$F.S. = \frac{\tau_{fail}}{\tau_{allow}}$$

**F.S. is always greater than 1 to avoid the potential failure.**

- To size the cross-sectional area of a simple connection:

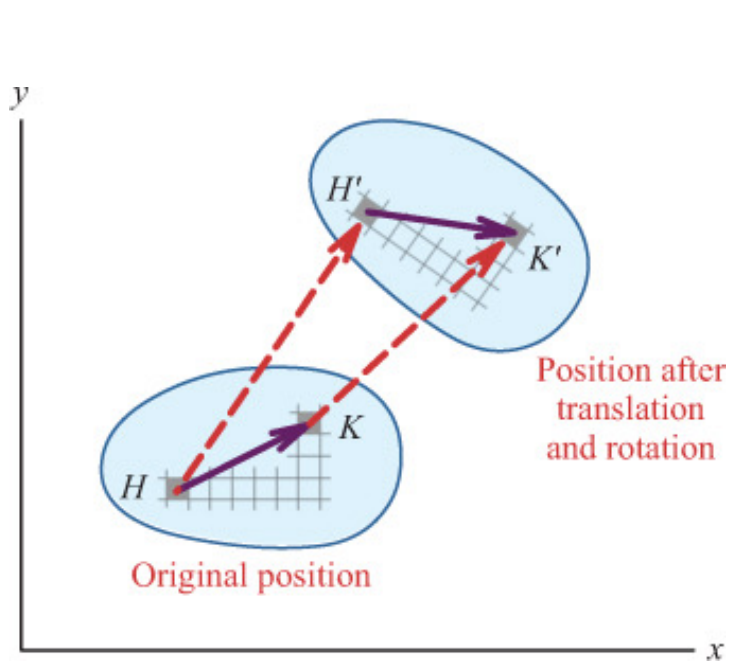
$$\text{Assume a } F.S. \Rightarrow \sigma_{allow} = \frac{\sigma_{fail}}{F.S.} \Rightarrow A = \frac{P}{\sigma_{allow}}$$

- How to estimate F.S.?
- How to determine  $\sigma_{fail}$ ?

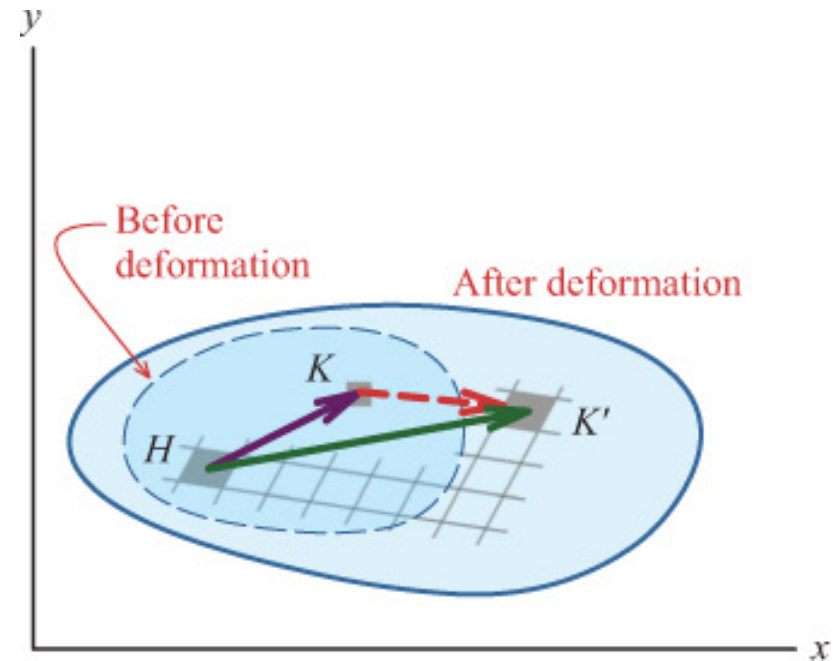


# Displacement, Deformation & Strain

Rigid-body displacement: no changes in the size and shape of a body



Deformation: changes in the size and shape of a body due to load- or temperature-induced displacement



Strain: intensity of a deformation



# Normal Strain

The elongation / contraction of a line segment per unit of length is referred to as **normal strain**.

- Average normal strain: 
$$\varepsilon_{avg} = \frac{l_f - l_o}{l_o} = \frac{e}{l_o}$$

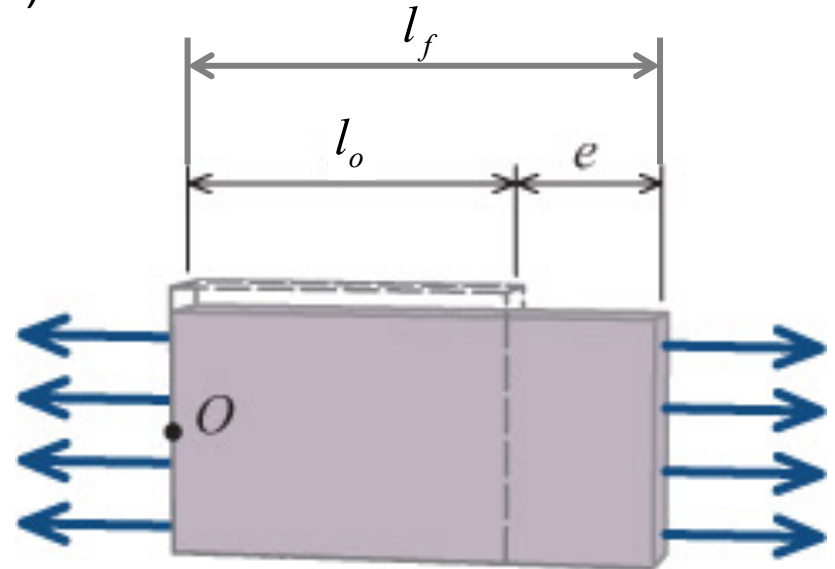
- If the normal strain is known, then the approximate final length is:  $l_f \approx (1 + \varepsilon)l_o$

- Normal strain unit: dimensionless (mm/mm)

- Micro-strain:  $1 \mu\varepsilon = 1 \times 10^{-6} \text{ m/m}$

$+\varepsilon \rightarrow$  line elongate

$-\varepsilon \rightarrow$  line contracts

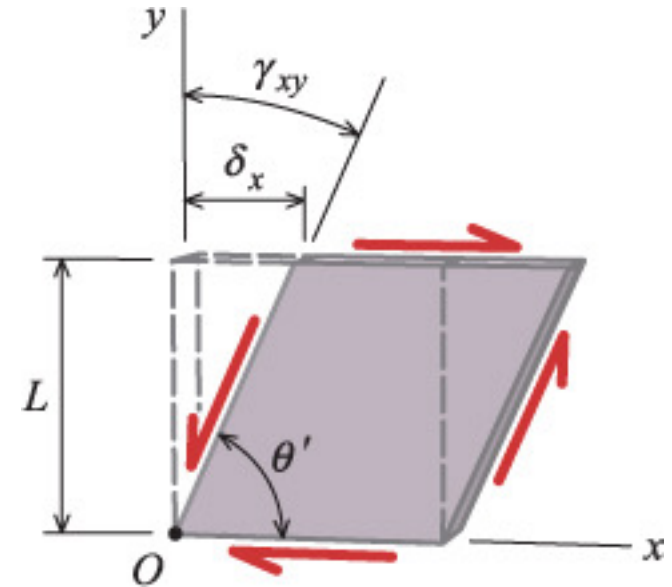


# Shear strain

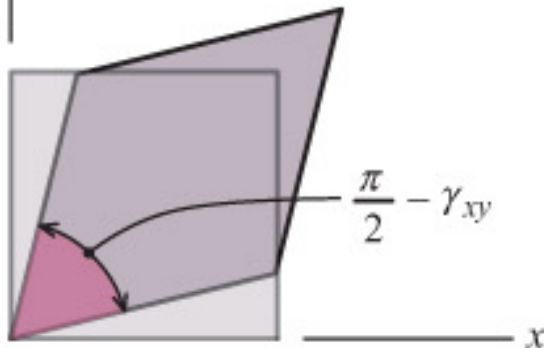
- Change in angle between 2 line segments that were *perpendicular* to one another

$$\gamma_{avg} = \frac{\delta_x}{L} \quad \gamma_{xy}(O) = \frac{\pi}{2} - \theta'$$

- Shear strain units: dimensionless (rad or  $\mu\text{rad}$ )

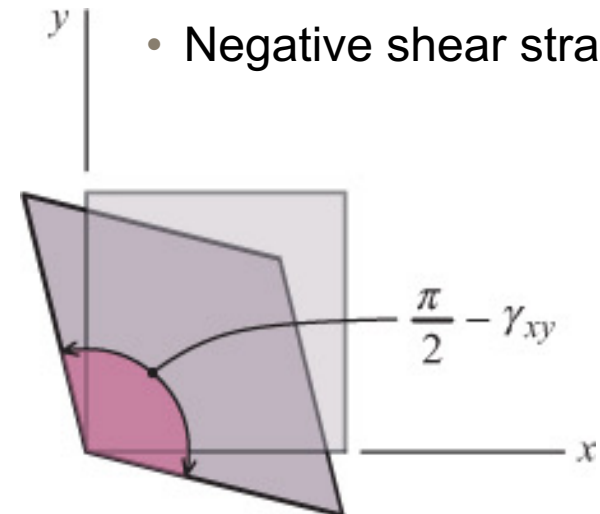


- Positive shear strain



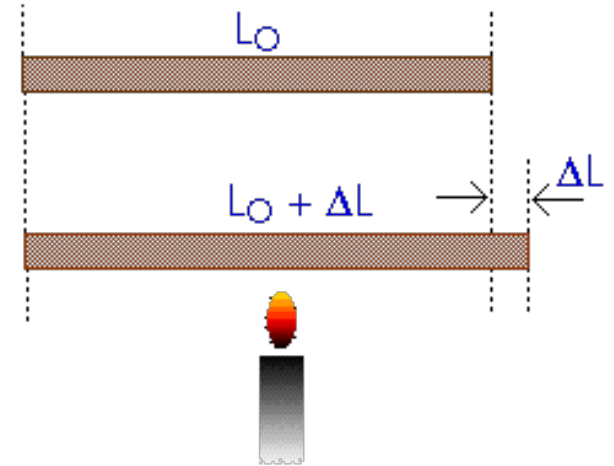
$\theta' < 90 \rightarrow +\text{shear strain}$   
 $\theta' > 90 \rightarrow -\text{shear strain}$

- Negative shear strain



# Thermal strain

- Coefficient of thermal expansion:
- $\alpha$ : The amount of **expansion** (or contraction) per unit length of a material resulting from one degree change in temperature
- Behaviour of material subjected to temperature change ( $\Delta T$ )
- Thermal strain:  $\epsilon_T = \alpha \Delta T$
- Total strain:  $\epsilon_{\text{total}} = \epsilon_{\sigma} + \epsilon_T$ 
  - Strain due to applied load
  - Strain due to change in temperature



Further reading: Chapter 7