

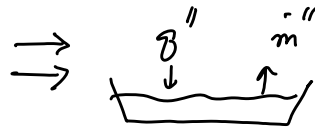
10/25/2010

This Lecture

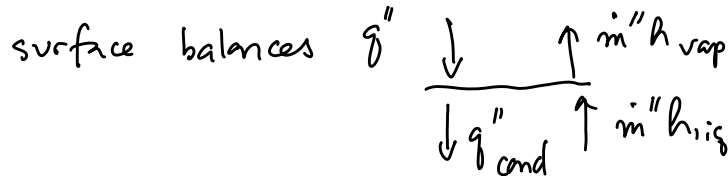
- * Review of Liquid Evaporation/Ignition
- * Introduce Solid Ignition

SUMMARY OF LIQUID EVAP & IGNITION

Review



surface balances & also control vol. balances.



$$q'' = \underbrace{m'' h_{fg}}_{\substack{\uparrow \\ \text{external} \\ \text{h.t.}}} + \underbrace{q''_{\text{cond}}}_{\substack{\uparrow \\ \text{mass} \\ \text{transfer}}} + \underbrace{q''_{\text{cond}}}_{\substack{\uparrow \\ \text{heating} \\ \text{of} \\ \text{liquid}}}$$

species balance.

$$m'' = h_m (Y_{\text{surf}} - Y_{\infty})$$

diffusive transport.

We know that unknowns are Y_{surf} & T_{surf} .

Combination of Newton's Law cooling & mass transfer eqn.

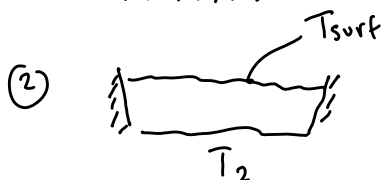
$$(a) \Rightarrow \underline{h(T_{\infty} - T_{\text{surf}}) = h_m(Y_{\text{surf}} - Y_{\infty}) h_{fg} + \underbrace{q''_{\text{cond}}}_{\text{deep pools.}}}$$

Three Limiting Cases.

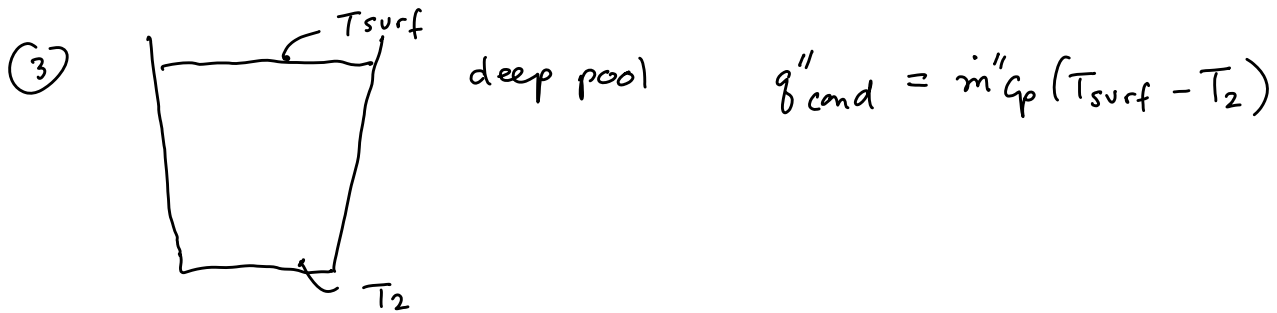


shallow well insulated pool.

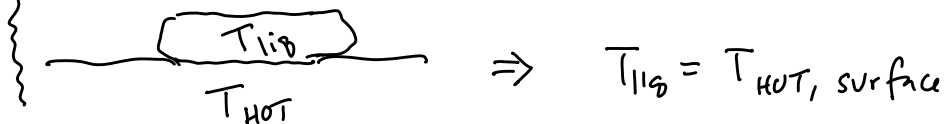
$$q''_{\text{cond}} = 0$$



$$q''_{\text{cond}} = h_{Liq} (T_{\text{surf}} - T_2)$$

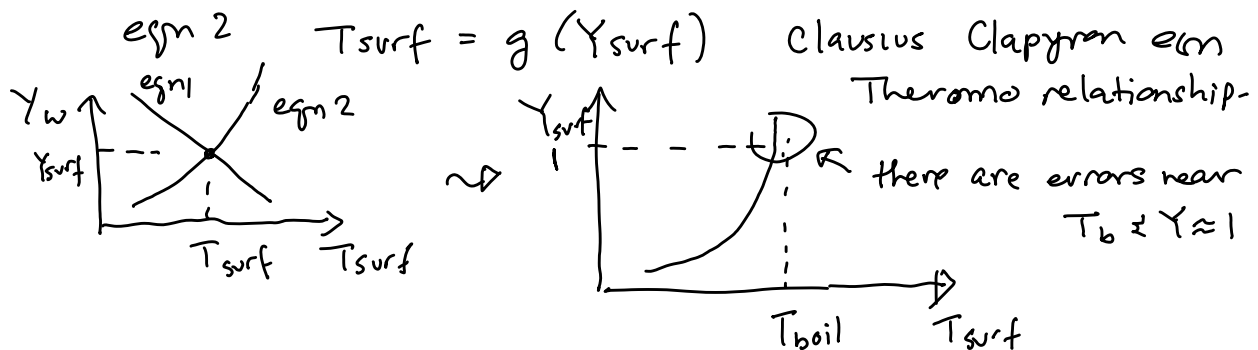


The trivial case occurs when you assume that T_{liquid} is known.

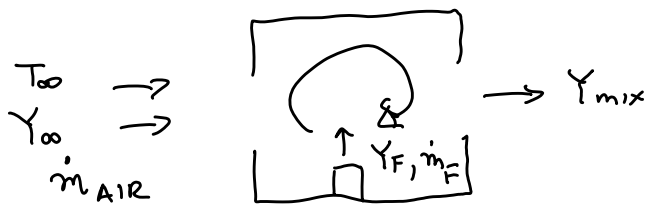


We need to understand the more complex cases.

eqn. 1 $T_{\text{surf}} = f(Y_{\text{surf}})$ abstraction of eqn (A)



2 Hard Problems



You can use chapter 3 analysis to compute or characterize Y_{mix}

couple evaporation analysis to conservation eqns.

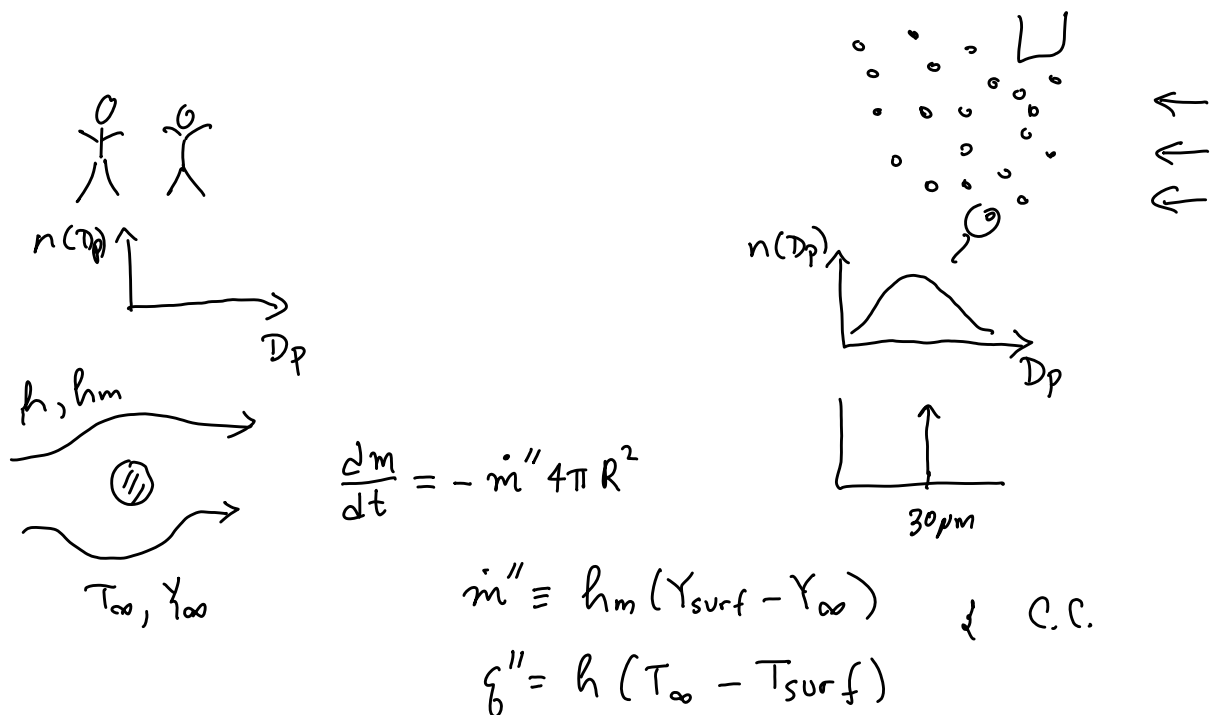
$$\frac{dm_i}{dt} = \dot{m}_f Y_{i,o} - \dot{m}_T Y_{i,mix} \quad w/ \quad m_i = m Y_{i,mix}$$

rate of mass flow of $Y_{i,0}$ is taken from

$$\dot{m}_f'' \equiv h_m (Y_{\text{surf}} - Y_{i,\text{mix}})$$

use the coupled h & m transfer problem to add to the analysis.

Applications of this type of Analysis.



Small droplets & particles are covered in a part of fluid mechanics called low Re flow.

$N_{\text{ud}} = 2$ When you solve the eqn you get

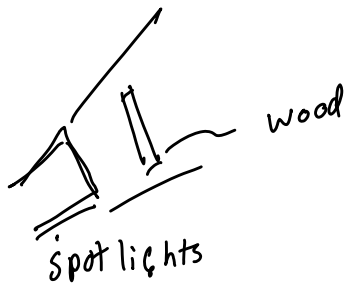
the D^2 law. $\frac{d(\frac{4}{3}\pi R^3 \rho_l)}{dt} = -\dot{m}'' 4\pi R^2$

$$\frac{dR}{dt} \propto \frac{C}{R}$$

$$R^2(t) \sim Ct$$

$$R^2 = R^2(0) - Kt$$

TRANSITION TO SOLID IGNITION



What types or levels (magnitudes) of heat flux will promote solid ignition.

Solid ignition heat fluxes are in the range

$$10 \frac{\text{kW}}{\text{m}^2} \rightarrow 20 \frac{\text{kW}}{\text{m}^2} .$$

Typical Heat Fluxes:

Solar irradiance 1 kW/m^2

Ignition cellulosic mat'ls 10 kW/m^2 (long time) }

Fast ignition 20 kW/m^2

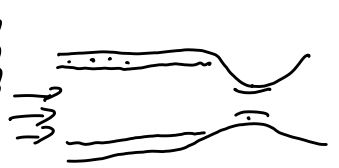
Fully involved room on fire $75 - 150 \text{ kW/m}^2$

Flame quenching 500 kW/m^2

In-cylinder SI engine $500 \frac{\text{kW}}{\text{m}^2} - 5000 \frac{\text{kW}}{\text{m}^2}$

Thermal Protective System $10 \frac{\text{MW}}{\text{m}^2} - 20 \frac{\text{MW}}{\text{m}^2}$ }

Rocket Nozzles (solid propellant) $20 - 50 \frac{\text{MW}}{\text{m}^2}$ }



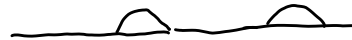
$$\dot{q}'' \approx \epsilon \sigma (T^4 - T_{\text{sur}}^4) \quad \text{w/} \quad \dot{q}'' \approx 10^4 \text{ W/m}^2, \quad \sigma = 5.67 \times 10^{-8} \frac{\text{W}}{\text{m}^2 \text{K}^4}$$

$\epsilon = 1$
 \therefore find T ?

Two types of externally driven solid ignition.

piloted
 hot gas phase
 source

glowing



intermediate
 step
 of hot
 spots on
 the solid.