# Scilab Textbook Companion for Thermal Engineering by K. K. Ramalingam<sup>1</sup>

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May 28, 2016

<sup>&</sup>lt;sup>1</sup>Funded by a grant from the National Mission on Education through ICT, http://spoken-tutorial.org/NMEICT-Intro. This Textbook Companion and Scilab codes written in it can be downloaded from the "Textbook Companion Project" section at the website http://scilab.in

# **Book Description**

Title: Thermal Engineering

Author: K. K. Ramalingam

Publisher: Scitech Publications, Chennai

Edition: 1

**Year:** 2009

**ISBN:** 9788183711982

Scilab numbering policy used in this document and the relation to the above book.

Exa Example (Solved example)

Eqn Equation (Particular equation of the above book)

**AP** Appendix to Example(Scilab Code that is an Appednix to a particular Example of the above book)

For example, Exa 3.51 means solved example 3.51 of this book. Sec 2.3 means a scilab code whose theory is explained in Section 2.3 of the book.

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# Chapter 1

## Gas power cycles

#### Scilab code Exa 1.1 The pressures

```
1 clc
2 clear
3 //Input data
4 V1=0.5; // Initial Volume before the commencement of
      compression in m<sup>3</sup>
5 P1=1; // Initial pressure before the commencement of
      compression in bar
6 T1=300; // Initial temperature in K
7 P2=12; // Final pressure at the end of compression
      stroke in bar
8 Q=220; // Heat added during the constant volume
      process in kJ
9 r=1.4; // Isentropic constant for air
10 R=0.287; // Characteristic Gas constant in kJ/kg K
11 Cv=0.718; // Specific heat of mixture in kJ/kg K
12
13 // Calculations
14 r1=(P2/P1)^(1/r);//Compression ratio
15 T2=T1*(r1)^(r-1); // Final temperature after the end
      of compression stroke in K
16 V2=(P1*T2*V1)/(P2*T1);//Final volume after the end
```

```
of compression stroke in m<sup>3</sup>
17 m = (P1*10^5*V1)/(R*T1*1000); //Mass of air flowing in
  T3=(Q/(m*Cv))+T2;//Temperature after constant volume
18
        heat addition in K
19 P3=(P2*T3)/T2;//Pressure after constant volume heat
       addition in K
20 V3=V2; //Volume at 3
21 P4=P3*(1/r1)^(r);//Pressure after isentropic
       expansion in bar
22 V4=V1; //Volume after isentropic expansion in m<sup>3</sup>
23 T4=T3*(1/r1)^(r-1); //Temperature at the end of
       isentropic expansion in K
24
25 //Output
26 printf('(a)The pressures at 1 is \%3.0 \, \text{fbar} \, \text{n} (b)
       Pressure at 2 is \%3.0 fbar\n (c) Pressure at 3 is
       \%3.2 \, \text{fbar} \setminus n \, (d) \, \text{Pressure at 4 is } \%3.2 \, \text{fbar} \setminus n \, (e)
       Temperature at 1 is \%3.1 \text{ fK} \setminus n (f) Temperature at 2
       is \%3.1\,\mathrm{fK}\n (g) Temperature at 3 is \%3.0\,\mathrm{fK}\n (h)
       Temperature at 4 is %3.0fK\n (i) Volume at 1 is %3
       .0 \, \text{fm}^3 \, \text{n} (j) Volume at 2 is \%3.5 \, \text{fm}^3 \, \text{n} (k) Volume
       at 3 is \%3.5 \,\mathrm{fm}^3 \, (1) Volume at 4 is \%3.0 \,\mathrm{fm}^3, P1
       , P2 , P3 , P4 , T1 , T2 , T3 , T4 , V1 , V2 , V3 , V4 )
```

#### Scilab code Exa 1.2 Compression ratio

```
1 clc
2 clear
3 //Input data
4 r1=6; //Initial compression ratio
5 r2=7; //Final compression ratio
6 r=1.4; //Isentropic coefficient of air
7
8 //Calculations
```

```
9 nr1=(1-(1/r1)^(r-1))*100;//Otto cycle efficiency
    when compression ratio is 6 in percentage
10 nr2=(1-(1/r2)^(r-1))*100;//Otto cycle efficiency
    when compression ratio is 7 in percentage
11 n=nr2-nr1;//Increase in efficiency in percentage
12
13 //Output
14 printf('The increase in efficiency due to change in
    compression ratio from 6 to 7 is %3.1 fpercent',n)
```

## Scilab code Exa 1.3 Air standard efficiency

```
1 clc
2 clear
3 //Input data
4 T1=315; // Temperature at the beginning of isentropic
      compression in K
5 T2=600; // Temperature at the end of isentropic
      compression in K
6 r=1.4; // Isentropic constant of air
8 // Calculations
9 r1=(T2/T1)^(1/(r-1));//Compression ratio
10 n=(1-(1/r1^{(r-1)}))*100; //Efficiency of Otto cycle in
       percent
11
12 //Output
13 printf('(a) The compression ratio is \%3.2 \text{ f} \setminus \text{n} (b)
      Efficiency of the Otto cycle is %3.1f percent',r1
      ,n)
```

Scilab code Exa 1.4 Air standard efficiency

```
1 clc
2 clear
3 //Input data
4 D=0.1; //Diameter of the cylinder in m
5 L=0.15; //Stroke length in m
6 Vc=0.295*10^-3; // Clearance volume in m^3
7 r=1.4; // Isentropic constant of air
9 // Calculations
10 Vs = (3.14/4) * (D^2*L); //Swept volume in m^3
11 r1=(Vc+Vs)/Vc;//Compression ratio
12 n = (1 - (1/r1)^{(r-1)}) *100; //Otto cycle efficiency in
      percentage
13
14 //Output
15 printf ('The air standard efficiency of air is %3.2 f
      percent',n)
```

#### Scilab code Exa 1.5 Mean effective pressure

```
14 // Calculations
15 Rc=Ru/M; // Characteristic gas constant in kJ/kg K
16 Cp=Rc+Cv; // Specific heat at constant pressure in kJ/
      kg K
17 r=Cp/Cv; // Isentropic gas constant
18 r1=(P2/P1)^(1/r);//Compression ratio
19 na=(1-(1/r1)^(r-1))*100; // Air standard efficiency in
       percentage
20 T2=T1*(P2/P1)^{((r-1)/r)};//Temperature at the end of
      isentropic compression process in K
21 T3=(P3/P2)*T2;//Temperature at the end of constant
      volume heat addition in K
22 Q=m*Cv*(T3-T2); //Heat supplied in kJ/kg
23 V1=(m*Rc*T1*1000)/(P1*10^5);//Initial volume before
      compression in m<sup>3</sup>
24 V2=V1/r1;//Volume at the end of compression stroke
      in m<sup>3</sup>
25 Vs=V1-V2; //Stroke volume in m<sup>3</sup>
26 MEP=(W/Vs)/100;//Mean effective pressure in bar
27
28 //Output
29 printf('(a) Compression ratio is \%3.2 \text{ f} \setminus \text{n} (b) The air
      standard efficiency is \%3.1f percent\n (c) Mean
      effective pressure is \%3.2f bar',r1,na,MEP)
```

#### Scilab code Exa 1.6 Compression ratio

```
bar
8 Q=210; //Heat added during constant heat process in
9 r=1.4; // Isentropic constant of air
10
11 // Calculations
12 r1=(P2/P1)^(1/r);//Compression ratio
13 V2=V1/r1; // Clearance volume in m<sup>3</sup>
14 C=(V2/(V1-V2))*100; // Percentage clearance in percent
15 na=(1-(1/r1)^(r-1))*100; // Air standard efficiency in
       percent
16 W=Q*(na/100);//Work done per cycle in kJ
17
18 //Output
19 printf('(a) Clearance volume as percentage of stroke
      volume is \%3.2 f percent\n (b) Compression ratio is
       \%3.2 \text{ f} \setminus \text{n} (c) Air standard efficiency is \%3.1 \text{ f}
      percent\n (d) Work done per cycle is %3.2 f kJ', C,
      r1, na, W)
```

#### Scilab code Exa 1.7 Ideal power

```
1 clc
2 clear
3 //Input data
4 r=5.5;//Compression ratio of an engine working on the otto cycle
5 Q=250;//Heat supplied during constant volume in kJ
6 N=500;//Engine operating speed in rpm
7 r1=1.4;//Isentropic ratio
8
9 //Calculations
10 n=(1-(1/r)^(r1-1))*100;//Otto cycle efficiency in percent
11 W=Q*(n/100);//Work done per cycle in kJ
```

#### Scilab code Exa 1.8 Mean effective pressure

```
1 clc
2 clear
3 //Input data
4 V1=0.53; //Volume of cylinder of an engine working on
       Otto cycle in m<sup>3</sup>
5 V2=0.1; // Clearance volume in m<sup>3</sup>
6 Q=210; // Heat supplied during constant volume in kJ
7 r=1.4; // Isentropic ratio
9 // Calculations
10 r1=V1/V2; // Compression ratio
11 n = (1 - (1/r1)^{(r-1)})*100; //Otto cycle efficiency in
      percentage
12 W=Q*(n/100); //Work done per cycle in kJ
13 P=W/((V1-V2)*100);//Mean effective pressure in bar
14
15 //Output data
16 printf ('Mean effective pressure is %3.3f bar',P)
```

## Scilab code Exa 1.10 Maximum theoretical power

```
1 clc
2 clear
3 //Input data
```

```
4 T3=1500; //Upper temperature limit of a otto cycle in
    K
5 T1=300; //Lower temperature limit in K
6 a=0.4; //Rate of flow of air through the cycle in kg/
    min
7 Cv=0.718; //
8
9 //Calculations
10 T2=(T1*T3)^(1/2); //Temperature at point 2 in K
11 T4=T2; //Temperature at point 4 in K
12 W=Cv*((T3-T2)-(T4-T1)); //Work done per cycle in kJ/
    kg
13 P=W*(a/60); //Maximum power developed by the engine
    in kW
14
15 //Output
16 printf('Maximum power developed by the engine is %3
    .3 f kW',P)
```

#### Scilab code Exa 1.11 Efficiencies for cut off ratio

```
Thermal efficiency of the diesel cycle for cut
off ratio 1.50

13 n3=(1-((1/rc^(r-1)*(p3^r-1)/(r*(p3-1)))))*100;//
Thermal efficiency of the diesel cycle for cut
off ratio 2.00

14
15 //Output
16 printf('(a)Thermal efficiency when cut off ratio is
1.25 is %3.2f percent\n (b)Thermal efficiency
when cut off ratio is 1.50 is %3.0f percent\n (c)
Thermal efficiency when cut off ratio is 2.00 is
%3.1f percent\n',n1,n2,n3)
```

#### Scilab code Exa 1.12 Air standard efficiency

```
clc
clear
r=15;//Compression ratio of a diesel engine
Q=5;//Heat supplied upto 5 percent of the stroke
r1=1.4;//Isentropic ratio

//Calculations
p=1+(Q/100)*(r-1);//Cut off ratio
n=(1-((1/r^(r1-1)*(p^r1-1)/(r1*(p-1)))))*100;//
Efficiency of diesel cycle in percent

//Output
printf('Air standard efficiency of the diesel cycle is %3.2 f percent',n)
```

#### Scilab code Exa 1.13 Efficiency

1 clc

```
2 clear
3 //Input data
4 r=17; //Compression ratio of a diesel engine
5 e=13.5; //Expansion ratio
6 r1=1.4; //Isentropic ratio
7
8 //Calculations
9 p=r/e; //Cut off ratio
10 n=(1-((1/r^(r1-1)*(p^r1-1)/(r1*(p-1)))))*100; //Air standard efficiency in percent
11
12 //Output
13 printf('Air standard efficiency is %3.1f percent',n)
```

#### Scilab code Exa 1.14 Compression ratio

```
1 clc
2 clear
3 //Input data
4 T1=300; // Temperature at the beggining of compression
       stroke in K
5 T2=873; //Temperature at the end of compression
      stroke in K
6 T3=2173; // Temperature at the beggining of expansion
      stroke in K
7 T4=1123; // Temperature at the end of expansion stroke
       in K
8 r1=1.4; // Isentropic ratio
10 // Calculations
11 r = (T2/T1)^{(1/(r1-1))}; // Compression ratio
12 rho=T3/T2;//Cut off ratio
13 n=(1-((1/r1)*((T4-T1)/(T3-T2))))*100; // Efficiency of
       diesel cycle in percent
14
```

```
15 //Output data
16 printf('(a) Compression ratio is %3.2 f \n (b) Cut off
    ratio is %3.2 f \n (c) Ideal efficiency of the
    diesel cycle is %3.2 f percent',r,rho,n)
```

#### Scilab code Exa 1.15 Pressure

```
1 clc
2 clear
3 //Input data
4 r=18; // Compression ratio of diesel cycle
5 Q=2000; //Heat added in kJ/kg
6 T1=300; //Lowest temperature in the cycle in K
7 p1=1; //Lowest pressure in the cycle in bar
8 Cp=1; // Specific heat of air at constant pressure in
      kJ/kg K
9 Cv=0.714; // Specific heat of air at constant volume
      in kJ/kg K
10
11 // Calculations
12 r1=Cp/Cv;//Isentropic ratio
13 v1 = ((1 - Cv) * T1) / (p1 * 10^5); / Initial volume at point 1
       in the graph in m<sup>3</sup>/kg
14 v2=v1/r; //Volume at point 2 in m^3/kg
15 p2=p1*(v1/v2)^(r1); //Pressure at point 2 in bar
16 T2=T1*(v1/v2)^(r1-1);//Temperature at point 2 in K
17 T3=(Q/Cp)+T2; // Temperature at point 3 in K
18 v3=v2*(T3/T2);//Volume at point 3 in K
19 v4=v1; //Since Constant volume heat rejection in m<sup>3</sup>/
      kg
20 T4=T3/(v4/v3)^{(r1-1)}; // Temperature at point 4 in K
      for isentropic expansion
21 p4=p1*(T4/T1); // Pressure at point 4 in bar
22
23 //Output
```

#### Scilab code Exa 1.16 Thermal efficiency

```
1 clc
2 clear
3 //Input data
4 r=16; // Compression ratio for the air standard diesel
       cycle
5 Q1=2200; // Heat added in kJ/kg
6 T4=1500; // Temperature at the end of isentropic
      expansion in K
7 T1=310; //Lowest temperature in the cycle in K
8 m=0.3; // Air flow rate in kg/sec
9 Cv=0.714; // Specific heat at constant volume in kJ/kg
      K
10
11 // Calculations
12 Q2=Cv*(T4-T1); //Heat rejected in kJ/kg
13 n=((Q1-Q2)/Q1)*100; // Efficiency in percent
14 P=m*(Q1-Q2);//Power developed in kW
15
16 // Output
17 printf('(a) Thermal efficiency is \%3.2 f percent\n (b)
     Power developed is %3.0 f kW',n,P)
```

#### Scilab code Exa 1.17 Air standard efficiency

```
1 clc
2 clear
3 //Input data
4 T1=303; // Temperature at the beginning of compression
       in K
5 T2=823; // Temperature at the end of compression in K
6 T3=3123; // Temperature at the end of heat addition in
  T4=1723; // Temperature at the end of isentropic
      expansion in K
8 r=1.4; // Isentropic ratio
10 // Calculations
11 n=(1-((T4-T1)/(r*(T3-T2))))*100; // Efficiency of the
      cycle in percent
12
13 //Output
14 printf ('Air standard efficiency of the cycle is %3.1
      f percent',n)
```

#### Scilab code Exa 1.18 Mean effective pressure

```
1 clc
2 clear
3 //Input data
4 r=15;//Compression Ratio of a diesel engine
5 P1=1;//Operating Pressure of a diesel engine in bar
6 r1=1.4;//Isentropic constant
7 V1=15;//Volume at the start of compression stroke in m^3
8 V3=1.8;//Volume at the end of constant Pressure heat addition in m^3
9 V4=V1;//Volume at the end of Isentropic expansion
```

```
stroke in m<sup>3</sup>
10 V2=1; // Volume at the end of isentropic compression
      stroke in m<sup>3</sup>
11 Vs=V1-V2; //Swept volume in m<sup>3</sup>
12
13 // Calculations
14 P2=P1*(r)^r1; // Pressure at the end of Isentropic
      compression of air
15 P3=P2; // Pressure at the end of constant pressure
      heat addition in bar
16 P4=P3*(V3/V4)^r1; // Pressure at the end of Isentropic
       expansion stroke in bar
17 Pm = (V2/Vs) * (P2 * ((V3/V2) - 1) + (P3 * (V3/V2) - P4 * (V4/V2)))/(
      r1-1)-(P2-P1*(V1/V2))/(r1-1));/Mean effective
      pressure in bar
18
19 //Output
20 printf ('Mean effective pressure of the cycle is \%3.2
      f bar', Pm)
```

#### Scilab code Exa 1.19 Compression ratio

```
1 clc
2 clear
3 //Input data
4 P1=1.5; // Pressure at the 7/8th stroke of compression in bar
5 P2=16; // Pressure at the 1/8th stroke of compression in bar
6 n=1.4; // Polytropic index
7 c=8; // Cutoff occurs at 8% of the stroke in percentage
8
9 // Calculations
10 R1=(P2/P1)^(1/n); // Ratio of volumes
```

#### Scilab code Exa 1.20 Loss in efficiency

```
1 clc
2 clear
3 //Input data
4 r=16; // Compression ratio of diesel engine
5 r1=1.4; // Isentropic ratio
6
7 // Calculations
8 rho1=1+(r-1)*(6/100); // Cutoff ratio at 6\% of stroke
9 rho2=1+(r-1)*(9/100); //Cutoff ratio at 9\% of stroke
10 n1=(1-(1/r^{(r1-1)})*(1/r1)*(rho1^r1-1)/(rho1-1))*100;
     // Efficiency of the cycle at 6% of the stroke in
      percent
11 n2=(1-(1/r^{(r1-1)})*(1/r1)*(rho2^{r1-1})/(rho2-1))*100;
     // Efficiency of the cycle at 9% of the stroke in
      percent
12 L=n1-n2; //The loss in efficiency in percent
13
14 //Output
15 printf ('The loss in efficiency is %3.2f percent',L)
```

#### Scilab code Exa 1.21 Compression ratio

```
1 clc
2 clear
3 //Input data
4 P1=1.03; // Pressure at the beginning of compression
      stroke in bar
5 T1=303; // Initial temperature in K
6 P2=40; //Maximum pressure in the cycle in bar
7 Q=550; //The heat supplied during the cycle in kJ/kg
8 r=1.4; // Isentropic compression ratio
9 Cp=1.004; // Specific heat at constant pressure in kJ/
      kg K
10
11 // Calculations
12 r1=(P2/P1)^(1/r);//Compression ratio
13 T2=(P2/P1)^{(r-1)/r}*T1;//Temperature at the end of
      compression stroke in K
14 T3=(Q/Cp)+T2; // Temperature at the end of heat
      addition in K
15 rho=T3/T2;//Cut off ratio
16 n = (1-(1/r1^{(r-1)})*(1/r)*(rho^{r-1})/(rho^{-1}))*100; //Air
       standard efficiency in percentage
17
18 //Output\n
19 printf('(a) Compression ratio is \%3.2 \text{ f} \setminus n (b)
      Temperature at the end of compression is \%3.1 f K
      n (c) Temperature at the end of comstant pressure
      heat addition is %3.0 f K \n (d) Air standard
      efficiency is %3.2f percent',r1,T2,T3,n)
```

Scilab code Exa 1.22 Air standard efficiency

```
1 clc
2 clear
3 //Input data
4 r=12; // Compression ratio of an oil engine, working
     on the combustion cycle
5 r1=1.4; // Isentropic ratio
6 P1=1; // Pressure at the
7 P3=35; // Pressure at the end of constant volume heat
      addition in bar
9 // Calculations
10 rho=1+(1/10)*(r-1);//Cut off ratio at 1/10th of the
      stroke
11 P2=P1*(r)^r1; // Pressure at the end of isentropic
      compression in bar
12 a=P3/P2; // Pressure ratio
13 n=(1-(1/r^{(r_1-1)})*(a*rho^{r_1-1})/((a-1)+(r_1*a*(rho-1)))
     ))*100; // Air standard efficiency in percent
14
15 //Output
16 printf ('The air standard efficiency of an oil engine
       working on the combustion cycle is \%3.2f percent
      ',n)
```

#### Scilab code Exa 1.23 Cut off ratio

```
1 clc
2 clear
3 //Input data
4 P1=1; //Pressure at the beginning of compression
        stroke of an oil engine working on a air standard
        dual cycle in bar
5 T1=303; //Temperature at the beginning of compression
        stroke in K
6 P3=40; //The maximum pressure reached in bar
```

```
7 T4=1673; //Maximum temperature reached in K
8 P4=P3; // Pressure at the start of constant pressure
      heat addition in bar
9 Cp=1.004; // Specific heat at constant pressure in kJ/
10 Cv=0.717; // Specific heat at constant volume in kJ/kg
11 r1=10; // Compression ratio
12
13 // Calculations
14 r=Cp/Cv;//Isentropic ratio
15 T2=T1*r1^(r-1); // Temperature at the end of
      compression stroke in K
16 P2=P1*r1^r; // Pressure at the end of compression
      stroke in bar
17 T3=T2*(P3/P2); // Temperature at the end of constant
      volume heat addition in K
18 rho=T4/T3;//Cut off ratio
19
20 //Output
21 printf('(a) Temperature at the end of constant volume
       heat addition is \%3.1f K\n (b) Cut off ratio is
     \%3.3 \, f', T3, rho)
```

#### Scilab code Exa 1.24 Work done

```
7 T4=1573; // Temperature at the end of constant volume
      heat addition in K
8 r=9.5; // Compression ratio
9 Cp=1.004; // Specific heat of air at constant pressure
10 Cv=0.717; // Specific heat of air at constant volume
11
12 // Calculations
13 r1=Cp/Cv;//Isentropic ratio
14 T2=T1*r^{(r1-1)}; // Temperature at the end of
      compression stroke in K
15 P2=P1*r^r1; // Pressure at the end of compression
      stroke in bar
16
  T3=T2*(P3/P2); // Temperature at the end of constant
      volume heat addition in K
17 rho=T4/T3;//Cut off ratio
18 T5=T4*(rho/r)^(r1-1);//Temperature at the end of
      expansion stroke in K
19 Qs=Cv*(T3-T2)+Cp*(T4-T3); // Heat supplied per kg in
20 Qr=Cv*(T5-T1); //Heat rejected per kg in kJ
21 W=Qs-Qr; //Work done per kg of air in kJ
22 n=(W/Qs)*100; // Efficiency of the air standard dual
      cycle in percent
23
24 //Output
25 printf('(a) The work done per kg of air is \%3.1 \text{ f kJ} \setminus \text{n}
       (b) Cycle efficiency is %3.2f percent', W,n)
```

### Scilab code Exa 1.25 Cycle efficiency

```
1 clc
2 clear
3 //Input data
4 r=10.5;//Compression ratio
5 P3=65;//Maximum pressure in bar
```

```
6 P4=P3; // Pressure at the end of constant volume heat
      addition in bar
7 qs=1650; //Heat supplied in kJ/kg
8 P1=1; // Pressure at the beginning of compression
      stroke in bar
9 T1=368; // Temperature at the beginning of compression
       stroke in K
10 Cp=1.004; // Specific heat of air at constant pressure
      in kJ/kg K
11 Cv=0.717; // Specific heat of air at constant volume
     in kJ/kg K
12
13 // Calculations
14 r1=Cp/Cv;//Compression ratio
15 P2=P1*r^r1; // Pressure at the end of compression
      stroke in bar
16 T2=T1*r^(r1-1); // Temperature at the end of
      compression stroke in K
17 T3=T2*(P3/P2); // Temperature at the end of constant
     volume heat addition in K
18 qv=Cv*(T3-T2); //Heat supplied at constant volume in
     kJ/kg
  qp=qs-qv;//Heat supplied at constant pressure in kJ/
19
     kg
20 T4=(qp/Cp)+T3; //Temperature at the end of constant
     volume heat addition in K
21 rho=T4/T3;//Cut off ratio
22 T5=T4*(rho/r)^(r1-1); //Temperature at the end of
      expansion stroke in K
23 P5=P4*(rho/r)^r1;//Pressure at the end of expansion
      stroke in K
24 q=Cv*(T5-T1); //Heat rejected in kJ/kg
25 n=((qs-q)/qs)*100;//Efficiency of the cycle in
     percent
26
27 //Output
28 printf('(a) Pressure at the end of compression stroke
       is \%3.1 f bar\n (b) Temperature at the end of
```

compression stroke is  $\%3.1\,\mathrm{f}$  K\n (c) Temperature at the end of constant volume heat addition is  $\%3.1\,\mathrm{f}$  K\n (d) Temperature at the end of constant pressure heat addition is  $\%3.2\,\mathrm{f}$  K\n (e) Temperature at the end of expansion stroke is  $\%3.2\,\mathrm{f}$  K\n (e) Pressure at the end of expansion stroke is  $\%3.2\,\mathrm{f}$  bar\n (f) Efficiency of the cycle is  $\%3.2\,\mathrm{f}$  percent',P2,T2,T3,T4,T5,P5,n)

#### Scilab code Exa 1.26 Air standard efficiency

```
1 clc
2 clear
3 //Input data
4 r=8.5; // Compression ratio
5 e=5.5; //Expansion ratio
6 P1=1; // Pressure at the beginning of compression
      stroke in bar
  T1=313; // Temperature at the beginning of compression
       stroke in K
8 n=1.3; // polytropic constant
9 Cp=1.004; // Specific heat of air at constant pressure
       in kJ/kg K
10 Cv=0.717; // Specific heat of air at constant volume
     in kJ/kg K
11
12 // Calculations
13 rho=r/e;//Cut off ratio
14 T2=T1*r^{(n-1)}; // Temperature at the end of
      compression stroke in K
15 T3=(2*Cv*T2)/(2*Cv-Cp*rho+1); // Temperature at the
     end of constant volume heat addition in K
16 T4=rho*T3; // Temperature at the end of constant
      pressure heat addition in K
17 a=T3/T2; // Pressure ratio i.e., P3/P2
```

```
18 n1=(1-(1/r^(n-1))*(a*rho^n-1)/((a-1)+(n*a*(rho-1))))
          *100;//Air standard efficiency in percent
19
20 //Output
21 printf('The air standard efficiency is %3.2f percent
',n1)
```

### Scilab code Exa 1.27 Ideal thermal efficiency

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial pressure in a compression engine
      working on a dual combustion engine in bar
5 T1=300; // Initial Temperature in K
6 P2=25; // Pressure at the end of compression stroke in
      bar
7 Q=400; //Heat supplied per kg of air during constant
      volume heating in kJ/kg
8 P5=2.6; // Pressure at the end of isentropic expansion
      in bar
9 Cp=1.005; // Specific heat of air at constant pressure
      in kJ/kg K
10 Cv=0.715; // Specific heat of air at constant volume
     in kJ/kg K
11
12 // Calculations
13 r=Cp/Cv; // Isentropic index
14 r1=(P2/P1)^(1/r);//Compression ratio
15 T2=T1*(r1)^(r-1); //Temperature at the end of
      compression stroke in K
16 T3=(Q/Cv)+T2;//Temperature at the end of constant
      volume heat addition in K
17 a=T3/T2; // Pressure ratio
18 P3=a*P2; // Pressure ratio at the end of constant
```

```
volume heat addition in bar

19 P4=P3; // Pressure at the end of constant pressure heat addition in bar

20 x=(P5/P4)^(1/r); // Ratio of volume at the end of constant pressure heat addition to the volume at the end of isentropic expansion

21 rho=x*(r1); // Cut off ratio

22 n=(1-(1/r1^(r-1))*(a*rho^r-1)/((a-1)+(r*a*(rho-1))))
    *100; // Air standard efficiency in percent of a dual combustion engine

23

24 // Output

25 printf('The ideal thermal efficiency is %3.1f percent',n)
```

### Scilab code Exa 1.28 Temperature

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial pressure of an enfine working on a
      dual combustion cycle in bar
5 T1=318; // Initial temperature before compression in K
6 r1=14; // Compression ratio
7 r=1.4; // Isentropic index
8 a=2; // Pressure ratio in the compression process
9 rho=2;//Cut off ratio
10
11 // Calculations
12 T2=T1*r1^(r-1); // Temperature at the end of
      compression stroke in K
13 T3=T2*a; // Temperature at the end of constant volume
     heat addition in K
14 T4=rho*T3; // Temperature at the end of constant
      pressure heat addition in K
```

```
15 T5=T4*(rho/r1)^(r-1);//Temperature at the end of
    isentropic compression in K
16 n=(1-((T5-T1)/(r*(T4-T3)+(T3-T2))))*100;//Efficiency
    of an engine working on a dual combustion cycle
    in percent
17
18 //Output
19 printf('(a)Temperature at the end of compression
    stroke is %3.0 f K\n (b)Temperature at the end of
    constant volume heat addition is %3.0 f K\n (c)
    Temperature at the end of constant pressure heat
    addition is %3.0 f K\n (d)Temperature at the end
    of isentropic expansion process is %3.0 f K\n (e)
    Efficiency of the cycle is %3.2 f percent', T2, T3,
    T4, T5,n)
```

#### Scilab code Exa 1.29 Pressure ratio

```
1 clc
2 clear
3 //Input data
4 r=15; // Compression ratio
5 Vs=0.01; //Stroke volume in m<sup>3</sup>
6 P1=1; // Initial pressure in bar
7 T1=310; // Initial temperature in K
8 P3=65; // Pressure in constant pressure heat addition
      stroke in bar
9 Cp=1; // Specific heat of air at constant pressure in
      kJ/kg K
10 Cv=0.714; // Specific heat of air at constant volume
      in kJ/kg K
11 R=287; // Molar gas constant
12
13 // Calculations
14 r1=Cp/Cv; // Isentropic index
```

```
15 P2=P1*(r)^r1;//Pressure at the end of compression
      stroke in bar
16 a=P3/P2; // Pressure ratio
17 rho=1+((5/100)*(r-1))
18 V2=Vs/(r-1); //Volume at the end of compression
      stroke in m<sup>3</sup>
19 V1=Vs+V2; // Initial volume in m<sup>3</sup>
20 m=P1*10^5*V1/(R*T1);//Mass of air contained in the
      cylinder in kg
  T2=T1*r^(r1-1); // Temperature at the end of
21
      compression stroke in K
22 a=P3/P2;//Pressure ratio
23 T3=T2*a; // Temperature at the end of constant volume
      heat addition in K
24 T4=T3*rho; // Temperature at the end of constant
      pressure heat addition in K
  T5=T4/(r/rho)^{(r1-1)}; // Temperature at the end of
25
      isentropic expansion in K
26 Qs=(Cv*(T3-T2)+Cp*(T4-T3))*m;//Heat supplied in kJ
27 Qr=m*Cv*(T5-T1);//Heat rejected in kJ
28 W=Qs-Qr; //Work done per cycle in kJ
29 n=(W/Qs)*100; // Efficiency of the cycle in percent
30 Mep=(W/Vs)/100;//Mean effective pressure in bar
31
32 // Output
33 printf('(1) Pressure ratio is \%3.3 \text{ f} \setminus \text{n} (2) Cut off
      ratio is \%3.2 \text{ f} \setminus \text{n} (3) Heat supplied per cycle is \%3
      .0 f kJ\n (4) Heat rejected per cycle is \%3.2 f kJ\n
       (5) Work done per cycle is \%3.2 \,\mathrm{f} kJ\n (6) Thermal
      efficiency of the cycle is \%3.0 \,\mathrm{f} percent\n (7)
      Mass of air contained in the cylinder is \%3.4 f kg
      \n (8) Mean effective pressure is \%3.2f bar',a,rho
      ,Qs,Qr,W,n,m,Mep)
```

Scilab code Exa 1.30 Thermal efficiency

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial pressure of air received by gas
      turbine plant in bar
5 T1=310; // Initial tamperature in K
6 P2=5.5; // Pressure at the end of compression in bar
7 r=1.4; //isentropic index
9 // Calculations
10 rp=P2/P1;//pressure ratio
11 n=(1-(1/rp)^{((r-1)/r)})*100; //Thermal efficiency of
      the turbine in percent
12
13 //Output data
14 printf ('Thermal efficiency of the turbine unit is \%3
      .2 f percent',n)
```

#### Scilab code Exa 1.31 Power developed

```
// Calculations
// Calculations
T2=T1*(P2/P1)^((r-1)/r); // Temperature at the end of process 1-2 in K

T4=T3*(P4/P3)^((r-1)/r); // Temperature at the end of process 3-4 in K

Wt=Cp*(T3-T4); // Work done by the turbine in kJ/kg
Wc=Cp*(T2-T1); // Work required by the compressor in kJ/kg
W=Wt-Wc; // Net work done by the turbine in kJ/kg
P=1*W; // Power developed by the turbine assembly per kg per second in kW

// Output
printf('Power developed by the turbine assembly per kg of air supplied per second is %3.2 f kW', P)
```

#### Scilab code Exa 1.32 Maximum temperature

```
1 clc
2 clear
3 //Input data
4 P1=1; //The pressure of air entering the compressor
     of a gas turbine plant operating on Brayton cycle
      in bar
5 T1=293; // Initial temperature in K
6 r=6.5; // Pressure ratio of the cycle
7 r1=1.4; // Isentropic ratio
9 // Calculations
10 T2=T1*(r)^((r1-1)/r1); //Temperature at the end of
      compression in K
11 T4=2.3*(T2-T1)/0.708;//Temperature at point 4 in K
12 T3=T4*(r)^((r1-1)/r1);//Maximum temperature in K
13 n=(1-((T4-T1)/(T3-T2)))*100; //Turbine plant
      efficiency in percent
```

#### Scilab code Exa 1.33 Air fuel ratio

```
1 clc
2 clear
3 //Input data
4 P1=1; // Pressure in an oil gas turbine installation
     in bar
5 T1=298; // Initial Temperature in K
6 P2=4; // Pressure after compression in bar
7 CV=42100; // Calorific value of oil in kJ/kg
8 T3=813; //The temperature reached after compression
      in K
9 m=1.2; // Air flow rate in kg/s
10 Cp=1.05; // Specific heat of air at constant pressure
     in kJ/kg K
11 r=1.4; // Isentropic ratio
12
13 // Calculations
14 r1=P2/P1; // Pressure ratio
15 T2=(r1)^{((r-1)/r)*T1}; // Temperature at the end of
      compression stroke in K
  T4=T3/(r1)^{(r-1)/r}; // Temperature at the end of
      isentropic expansion in K
17 Wt=m*Cp*(T3-T4); //Work done by the turbine in kJ/s
      or kW
  Wc=m*Cp*(T2-T1);//Work to be supplied to the
      compressor in kJ/s or kW
19 Wn=Wt-Wc; //Net work done by the turbine unit in kW
20 qs=m*Cp*(T3-T2);//Heat supplied by the oil in kJ/s
```

```
21 M=qs/CV; // Mass of fuel burnt per second in kg/s
22 a=m/M; // Air fuel ratio
23
24 // Output
25 printf('(a) The net power output of the installation
    is %3.2 f kW\n (b) Air fuel ratio is %3.1 f', Wn, a)
```

#### Scilab code Exa 1.34 Net power

```
1 clc
2 clear
3 //Input data
4 T1=300; //Minimum temperature of the plant containing
       a two stage compressor with perfect intercooling
       and a single stage turbine in K
5 T5=1100; //Maximum temperature of the plant in K
6 P1=1; // Initial Pressure in bar
7 P5=15; // Final pressure in bar
8 Cp=1.05; // Specific heat of air in kJ/kg K
9 r=1.4; // Isentropic ratio
10 P6=P1; // Pressure at 6 in bar
11
12 // Calculations
13 P3=(P1*P5)^(1/2); //The intermediate pressure for
      cooling in bar
14 P2=P3; // Pressure at point 2 in bar
15 T2=T1*(P2/P1)^{((r-1)/r)};//Temperature at the end of
      process 1-2
16 T3=T1; // Intermediate temperature in K
17 T4=1.473*T3; // Temperature at point 4 in K
18 T6=T5/(P5/P6)^{(r-1)/r}; //Temperature at point 6 in
     k
19 Wt=Cp*(T5-T6); //Work done by the turbine per kg of
      air in kJ/s
20 Wc=Cp*(T4-T3)+Cp*(T2-T1);//Work done by the
```

```
compressor per kg of air in kJ/s
21 Wn=Wt-Wc;//Net work done in kJ/s
22 Pn=Wn;//Net power developed in kW
23
24 //Output
25 printf('The net power of the plant per kg of air/s is %3.2 f kW', Pn)
```

# Scilab code Exa 1.35 Maximum power

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial Pressure of a gas turbine power plant
      in bar
5 P2=8; // Final pressure in bar
6 T1=300; // Initial temperature in K
7 T5=850; // Temperature of air expanded in the turbine
     in K
8 m=1.8; //Mass of air circulated per second in kg
9 Cp=1.05; // Specific heat of air at constant pressure
     in kJ/kg K
10 r=1.4; // Ratio of specific heat
11
12 // Calculations
13 P4=(P1*P2)^(0.5); // Pressure for maximum power output
       in bar
14 P3=P2; // Pressure after the constant pressure process
      in bar
15 T3=T5; //For reheating condition Temperature in K
16 T2=T1*(P2/P1)^{((r-1)/r)};//Temperature at the end of
      constant entropy process in K
17 T4=T3/((P3/P4)^{((r-1)/r))}; // Temperature after the
      process 3-4 in K
18 T6=T4; // Temperature at the end of process 5-6 in K
```

```
19 Wt=m*Cp*((T3-T4)+(T5-T6));//Work done by the turbine
        in kJ/s
20 Wc=m*Cp*(T2-T1);//Work absorbed by the compressor in
        kJ/s
21 P=Wt-Wc;//Power that can be obtained from gas
        turbine installation in kW
22
23 //Output
24 printf('The maximum power that can be obtained from
        turbine installation is %3.0 f kW',P)
```

#### Scilab code Exa 1.36 Mass of fluid

```
1 clc
2 clear
3 //Input data
4 P1=1.5; // Pressure at the inlet of the low pressure
     compressor in bar
  T1=300; // Temperature at the inlet of the low
     pressure compressor in K
6 P5=9; //Maximum pressure in bar
7 T5=1000; //Maximum temperature in K
8 P=400; //Net power developed by the turbine in kW
9 Cp=1.0; // Specific heat of air at constant pressure
     in kJ/kg K
10 r=1.4; // Ratio of specific heat
11
12 // Calculations
13 P8=P1; //For perfect intercooling and perfect
     reheating in bar
14 P4=P5; //For perfect intercooling and perfect
     reheating in bar
15 P2=(P1*P4)^0.5; // Pressure at the end of Isentropic
      compression in LP compressor in bar
16 P6=P2; // Pressure at the end of process 5-6 in bar
```

```
17 T2=T1*(P2/P1)^{((r-1)/r)};//Temperature at the end of
      isentropic compression in K
18 T3=T1; //For perfect intercooling in K
19 T4=T2; //For perfect intercooling in K
20 T6=T5/(P5/P6)^{(r-1)/r};//Temperature at the end of
      process 5-6 in K
21 T7=T5; // Temperature in K
22 T8=T6; // Temperature in K
23 Wt=Cp*((T5-T6)+(T7-T8)); //Work done by the turbine
      in kg/s
24 Wc=Cp*((T2-T1)+(T4-T3)); //Work absorbed by the
      compressor in kJ/s
25 Wn=Wt-Wc; //Net work output in kJ/s
26 m=P/Wn;//Mass of fluid flow per second in kg/s
27 qs=m*Cp*((T5-T4)+(T7-T6)); //Heat supplied from the
      external source in kJ/s
28
29 // Output
30 printf('(a) Mass of fluid to be circulated in the
      turbine is \%3.3 \text{ f kg/s/n} (b) The amount of heat
      supplied per second from the external source is
      \%3.1 \, \text{f kJ/s}, m, qs)
```

#### Scilab code Exa 1.37 Mass of air

```
9 r=1.4; // Isentropic ratio
10
11 // Calculations
12 T2=T1*a^{((r-1)/r)}; // Temperature after the isentropic
       compression stroke in K
13 T4=T3/a^{((r-1)/r)}; // Temperature after the isentropic
       expansion process in K
14 Wt=Cp*(T3-T4); //Work done by the turbine per kg of
      air per second in kJ
15 Wc=Cp*(T2-T1); //Work absorbed by the compressor per
      kg of air per second in kJ
16 Wn=Wt-Wc; //Net work output in kJ/s
17 m=P/Wn; //Mass of fluid circulated per second in kg/s
18 Q=m*Cp*(T3-T2); // Heat supplied by the heating
      chamber in kJ/s
19
20 //Output
21 printf('(a) Mass of air circulating in the
      installation is \%3.2 f \text{ kg/s/n} (b) Heat supplied by
      the heating chamber is \%3.1 \, \text{f kJ/s',m,Q}
```

# Scilab code Exa 1.38 Overall efficiency

```
clear
clear
//Input data
a=6;//Pressure ratio of a gas turbine plant
T1=293;//Inlet temperature of air in K
T3=923;//Maximum temperature of the cycle in K
P=2000;//Power developed in the cycle in kW
nc=85;//Efficiency of the compressor in percentage
nt=85;//Efficiency of the turbine in percentage
Cp=1;//Specific heat of gas at constant pressure in kJ/kg K
Cv=0.714;//Specific heat of gas at constant volume
```

```
in kJ/kg K
12
13 // Calculations
14 r=Cp/Cv; // Ratio of specific heats
15 T2a=a^{(r-1)/r}*T1; // Temperature at 2' in K
16 T2=((T2a-T1)/(nc/100))+T1;//Temperature at point 2
      in K
17 T4a=T3/a^{((r-1)/r)}; // Temperature at the point 4' in
  T4=T3-((T3-T4a)*(nt/100)); //Temperature at the point
      4 in K
19 Wt=Cp*(T3-T4); //Work done by the turbine per kg of
      air in kJ
20 Wc=Cp*(T2-T1); //Work done by the compressor per kg
      of air in kJ
  Wn=Wt-Wc; // Net work output of the turbine per kg of
      air in kJ
22 qA=Cp*(T3-T2);//Heat supplied per kg of air in kJ
23 n=(Wn/qA)*100; // Overall efficiency of the turbine
      plant in percentage
24 m=P/Wn; //Mass of air circulated per second in kg
25
26 //Output
27 printf('(1) Overall efficiency of the turbine is \%3.0
      f percentage\n (2) Mass of air circulated by the
      turbine is %3.2 f kg',n,m)
```

#### Scilab code Exa 1.39 Isentropic efficiency

```
1 clc
2 clear
3 //Input data
4 T1=293; //Initial temperature of a gas turbine plant
    in K
5 P1=1; //Initial pressure in bar
```

```
6 P2=4.5; // Pressure after the compression in bar
7 nc=80; // Isentropic efficiency of a compressor in
      percentage
8 T3=923; // Temperature of the gas whose properties may
      be assumed to resemble with those of air in the
      combustion chamber in K
9 deltaP=0.1; // Pressure drop in a combustion chamber
      in bar
10 nt=20; //Thermal efficiency of the plant in
      percentage
11 r=1.4; // Isentropic index
12 P4=1; // Pressure at point 4 in bar
13
14 // Calculations
15 P3=P2-deltaP;//Pressure at point 3 in bar
16 T21=T1*(P2/P1)^((r-1)/r); // Temperature after the
      compression process in K
  T2=(T21-T1)/(nc/100)+T1;//Temperature at the point 2
17
18 T41=T3/(P3/P4)^{((r-1)/r)}; // Temperature at the end of
       expansion process in K
19 Ac=T2-T1; //Work done by the compressor per kg of air
      per specific heat at constant pressure Ac=Wc/Cp
20 At=T3; //Work done by the turbine per kg of air per
      specific heat at constant pressure At=Wt/Cp
21 An=At-Ac; //Net work done per kg of air
22 Bs=T3-T2; //Heat supplied per kg of air per specific
     heat at constant pressure Bs=qs/Cp;qs=heat
      supplied
23 T4=An-((nt/100)*Bs);//Temperature at point 4 in K
24 \text{ nT} = ((T3-T4)/(T3-T41))*100; //Isentropic efficiency of
      the turbine in percentage
25
26 //Output
27 printf ('The isentropic efficiency of the turbine is
     \%3.2 f percent', nT)
```

# Scilab code Exa 1.40 Overall efficiency

```
1 clc
2 clear
3 //Input data
4 P1=1; // Pressure of air received by the gas turbine
      plant in bar
5 T1=300; // Initial Temperature in K
6 P2=5; // Pressure of air after compression in bar
7 T3=850; // Temperature of air after the compression in
8 nc=80; // Efficiency of the compressor in percent
9 nt=85; // Efficiency of the turbine in percent
10 r=1.4; // Isentropic index of gas
11 P3=P2; //Since 2-3 is constant pressure process in
     bar
12 P41=1; // Pressure at the point 41 in bar
13 Cp=1.05; // Specific heat of the gas at constant
      pressure in kJ/kg K
14
15 // Calculations
16 T21=T1*(P2/P1)^((r-1)/r); // Temperature at the point
      21 on the curve in K
17 T2=(T21-T1)/(nc/100)+T1;//Temperature at the point 2
      in K
18 T41=T3/(P3/P41)^{((r-1)/r)}; // Temperature at the point
      41 in K
19 T4=T3-((nt/100)*(T3-T41)); //Temperature of gas at
      the point 4 in K
20 Wt=Cp*(T3-T4); //work done by the turbine in kJ/kg of
21 Wc=Cp*(T2-T1); //Work done by the compressor in kJ/kg
       of air
22 Wn=Wt-Wc; //Net work done by the plant in kJ
```

# Scilab code Exa 1.41 Overall efficiency

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial pressure of a gas turbine plant in
     bar
5 T1=310; // Initial temperature in K
6 P2=4; // Pressure of air after compressing in a rotary
       compressor in bar
7 P3=P2; // Constant pressure process
8 P41=P1; // Since 1-41 is a constant pressure process
      in bar
9 T3=900; // Temperature of air at the point 3 in
      constant process in K
10 nc=80; // Efficiency of the compressor in percentage
11 nt=85; // Efficiency of the turbine in percentage
12 E=70; // Effectiveness of the plant in percentage
13 r=1.4; // Isentropic index
14 Cp=1; // Specific heat of air at constant pressure in
     kJ/kg K
15
16 // Calculations
17 T21=T1*(P2/P1)^((r-1)/r); // Temperature at the point
      21 in the temperature versus entropy graph in K
18 T2=T1+((T21-T1)/(nc/100)); // Temperature of air after
       the compression process in K
19 T41=T3/((P3/P41)^{((r-1)/r)}); //Temperature at the
```

```
point 41 after the isentropic expansion process
20 T4=T3-((T3-T41)*(nt/100)); // Temperature at the point
      4 in K
21 Wt=Cp*(T3-T4); //Work done by the turbine in kJ
22 Wc=Cp*(T2-T1);//Work done by the compressor in kJ
23 Wn=Wt-Wc; //Net work done in kJ
24 qs=Cp*(T3-T2);//Heat supplied in kJ
25 qa=Cp*(T4-T2);//Heat available in the exhaust gases
     in kJ
26 H=qa*(E/100); // Actual heat recovered from the
     exhaust gases in the heat exchanger in kJ
27
  Hs=qs-(H); // Heat supplied by the combustion chamber
     in kJ
28 nt=(Wn/Hs)*100; //Thermal efficiency of the gas
      turbine plant with heat exchanger in percent
29
30 //Output
31 printf ('The overall efficiency of the plant is \%3.1f
       percent', nt)
```

# Chapter 7

# Performance of IC engines

# Scilab code Exa 7.1 Brake torque

```
1 clc
2 clear
3 //Input data
4 N=1500; //Engine speed in rpm
5 p=110;//Load on brakes in kg
6 L=900; //Length of brake arm in mm
7 g=9.81; // Gravitational force in N/m<sup>2</sup>
8 pi=3.14; // Mathematical constant
9
10 // Calculations
11 T = ((p*g)*(L/1000)); //Braking torque in Nm
12 P = ((T/1000) * ((2*3.14*N)/60)); //Power available at
      the brakes of the engine in kW
13
14 // Output
15 printf('(a) Brake torque is %3.1 f Nm \n (b) Power
      available at the brakes of the engine is \%3.2 f kW
      ',T,P)
```

#### Scilab code Exa 7.2 Power available at brakes

```
1 clc
2 clear
3 //Input data
4 N=700; // Engine speed in rpm
5 D=0.6; // Diameter of brake drum in m
6 d=0.05; // Diameter of rope in m
7 W=35; //Dead load on the brake drum in kg
8 S=4.5; //Spring balance reading in kg
9 g=9.81; // Gravitational constant in N/m<sup>2</sup>
10 pi=3.14; // Mathematical constant
11
12 // Calculations
13 P = (((W-S)*g*pi*(D+d))/1000)*(N/60); //Power in kW
14
15 //Output
16 printf ('The power available at the brakes is %3.3 f
     kW',P)
```

# Scilab code Exa 7.3 Brake thermal efficiency

```
1 clc
2 clear
3 //Input data
4 W=950; //Load on hydraulic dynamometer in N
5 C=7500; //Dynamometer constant
6 f=10.5; //Fuel used per hour in kg
7 h=50000; // Calorific value of fuel in kJ/kg
8 N=400; //Engine speed in rpm
9
10 //Calculations
11 P=(W*N)/C; //Power available at the brakes in kW
12 H=P*60; //Heat equivalent of power at brakes in kJ/min
```

```
13 Hf=(f*h)/60;//Heat supplied by fuel per minute in kJ
    /min
14 n=(H/Hf)*100;//Brake thermal efficiency in
        percentage
15
16 //Output
17 printf(' Brake thermal efficiency of the engine is
        %3.2f percent',n)
```

# Scilab code Exa 7.4 Specific fuel consumption

```
1 clc
2 clear
3 //Input data
4 n1=50.5; // Air standard efficiency in percentage
5 n2=50; //Brake thermal efficiency in percentage
6 N=3000; //Engine speed in rpm
7 H=10500; // Heating value of fuel in kcal/kg
8 T=7.2; //Torque developed in kgf*m
9 B=6.3; //Bore diameter in cm
10 S=0.095; // stroke in m
11
12 // Calculations
13 nbt=(n1/100)*(n2/100);//Brake thermal efficiency in
      percentage
14 B1=(2*(22/7)*N*T)/4500;//Brake horse power in kW
15 B2=B1/4; //Brake horse power per cylinder in kW
16 Bsf=(4500*60)/(H*427*nbt);//Brake specific fuel
      consumption in kg/BHP hr
17 bmep=(B2*4500)/(S*(3.14*B^2/4)*(N/2));//Brake mean
      effective pressure in kgf/cm<sup>2</sup>
18
19 // Output
20 printf('(a) Specific fuel consumption is \%3.3 f kg/BHP
       hr\n (b)Brake mean effective pressure is \%3.3 f
```

# Scilab code Exa 7.5 Mechanical efficiency

```
1 clc
2 clear
3 //Input data
4 W=30; //The net dynamometer load in kg
5 R=0.5; // Radius in m
6 N=2400; //Speed in rpm
7 FHP=6.5; //Engine power in hp
9 // Calculations
10 BHP=(2*3.14*R*N*W)/4500;//Brake horse power in kW
11 IHP=BHP+FHP; // Indicated horse power in kW
12 nm=(BHP/IHP)*100; // Mechanical efficiency in
      percentage
13
14 //Output
15 printf ('Mechanical efficiency of the engine is \%3.2 f
       percent', nm)
```

#### Scilab code Exa 7.6 IHP

```
1 clc
2 clear
3 //Input data
4 d=25;//Diameter of cylinder in cm
5 1=0.4;//Stroke of piston in m
6 N=200;//Speed in rpm
7 m=10;//Misfires per minute
8 M=6.2;//Mean effective pressure in kgf/cm^2
9 nm=0.8;//Mechanical efficiency in percent
```

# Scilab code Exa 7.7 Average piston speed

```
1 clc
2 clear
3 //Input data
4 I=5; //Indicated power developed by single cylinder
     of 2 stroke petrol engine
5 M=6.5; //Mean effective pressure in bar
6 d=0.1; //Diameter of piston in m
7
8 // Calculations
9 A=(3.14*d^2)/4; //Area of the cylinder
10 LN=(I*1000*60)/(M*10^5*A);//Product of length of
     stroke and engine speed
11 S=2*LN; // Average piston speed in m/s
12
13 //Output
14 printf('The average piston speed is %3.2 f m/s',S)
```

Scilab code Exa 7.8 Dimensions of cylinder

```
1 clc
2 clear
3 //Input data
4 P=60; //Power developed by oil engine in kW
5 M=6.5; //Mean effective pressure in kgf/cm<sup>2</sup>
6 N=85; // Number of explosions per minute
7 r=1.75; // Ratio of stroke to bore diameter
8 nm=0.8; // Mechanical efficiency
9
10 // Calculations
11 I=P/nm; //Indicated horse power
12 d = ((I*100*4*4500)/(M*r*3.14*N))^(1/3);//Bore
      diameter in cm
13 l=r*d; //Stroke length in cm
14
15 // Output
16 printf('(a) Diameter of the bore is \%3.2 \text{ f cm } \setminus \text{n} (b)
      Stroke length of the piston is %3.2 f cm',d,1)
```

# Scilab code Exa 7.9 Bore and stroke of piston

```
12 l=1.3*d;//Stroke length in cm
13
14 //Output
15 printf('(a)The bore diameter of the cylinder is %3.2
    f cm\n (b)Stroke length of the piston is %3.2 f cm
    ',d,1)
```

# Scilab code Exa 7.10 Volumetric efficiency

```
1 clc
2 clear
3 //Input data
4 d=6; // Diameter of the bore in cm
5 1=9; //Length of the stroke in cm
6 m=0.00025; // Mass of charge admitted in each suction
      stroke
7 R=29.27; //Gas constant Kgfm/kg K
8 p=1;//Normal pressure in kgf/cm<sup>2</sup>
9 T=273; // Temperature in K
10
11 // Calculations
12 V=(m*R*T)*10^6/(p*10^4); //Volume of charge admitted
      in each cycle in m<sup>3</sup>
13 Vs = (3.14*d^2*1)/4; //Swept volume of the cylinder
14 nv=(V/Vs)*100;//Volumetric efficiency in percentage
15
16 // Output
17 printf ('The volumetric efficiency is %3.1f percent',
      nv)
```

# Scilab code Exa 7.11 Volumetric efficiency

```
1 clc
```

```
2 clear
3 //Input data
4 d=0.12; // Diameter of the bore in m
5 1=0.13; //Length of stroke in m
6 N=2500; //Speed of the engine in rpm
7 d1=0.06; //Diameter of the orifice in m
8 Cd=0.70; // Discharge coefficient of orifice
9 hw=33;//Heat causing air flow through orifice in cm
      of water
10 p=760; //Barometric reading in mm of Hg
11 T1=298; // Ambient temperature in degree K
12 p1=1.013; // Pressure of air at the end of suction in
13 T2=22; // Temperature of air at the end of suction in
      degree C
14 R=0.287; // Universal gas constant
15 n=6; //Number of cylinders in the engine
16 n1=1250; //Number of strokes per minute for a four
      stroke engine operating at 2500 rpm
17
18 // Calculations
19 V=(3.14*d^2*1)/4; //Swept volume of piston in m<sup>3</sup>
20 Ao=(3.14*d1^2)/4; // Area of the orifice in m<sup>2</sup>
21 rho=p1*10^5/((R*T1)*1000);//Density of air at 1.013
      bar and 22 degrees C
22 \text{ Va}=840*\text{Cd}*\text{Ao}*(\text{hw/rho})^{(1/2)};//\text{Volume of air passing}
      through the orifice in m<sup>3</sup>/min
23 V1=8.734/n; // Actual volume of air per cylinder in m
      ^3/\min
24 As=V1/n1; // Air supplied per cycle per cylinder in m
25
  nv=(As/V)*100; // Volumetric efficiency of the engine
      in percentage
26
27 // Output
28 printf ('The volumetric efficiency of the engine is
      \%3.2 f percent', nv)
```

# Scilab code Exa 7.12 Air standard efficiency

```
1 clc
2 clear
3 //Input data
4 d=0.15; // Diameter of the piston in m
5 l=0.19; //Length of the stroke in m
6 V=0.00091; // Clearance volume in m<sup>3</sup>
7 N=250; //Speed of the engine in rpm
8 M=6.5; //Indicated mean effective pressure in bar
9 c=6.3; //Gas consumption in m<sup>3</sup>/hr
10 H=16000; // Calorific value of the has in kJ/m<sup>3</sup>
11 r1=1.4; // Polytropic index
12
13 // Calculations
14 Vs = (3.14*d^2*1)/4; //Swept volume in m^3
15 Vt=Vs+V; // Total cylinder volume in m<sup>3</sup>
16 r=Vt/V; // Compression ratio
17 na=(1-(1/r^{(r1-1))})*100; // Air standard efficiency in
       percent
18 A = (3.14*d^2)/4; // Area of the bore in m
19 I = (M*10^5*1*A*N)/(1000*60); //Indicated power in kW
20 Hs=(c*H)/(60*60); // Heat supplied per second
21 nt=(I/Hs)*100; //Indicated thermal efficiency in
      percent
22
23 //Output
24 printf('(a)The air standard efficiency is \%3.1f
      percent\n (b) Indicated power is \%3.3 \text{ f kW} \setminus \text{n} (c)
      Indicated thermal efficiency is %3.1f percent', na
      , I , nt)
```

#### Scilab code Exa 7.13 Diameter of venturi

```
1 clc
2 clear
3 //Input data
4 ma=6; // Air supplied per minute by a single jet
      carburetor in kg/min
5 mf=0.44; //Mass flow rate of petrol in kg/min
6 s=0.74; // Specific gravity of petrol in kg/m<sup>3</sup>
7 p1=1; // Initial pressure of air in bar
8 T1=300; //Initial temperature of air in K
9 Ci=1.35; // Isentropic coefficient of air
10 V=90; //Speed of air in the venturi in m/s
11 Vc=0.85; // Velocity coefficient of the venturi in m/s
12 Cf=0.66; // Coefficient of discharge for the jet
13 Cp=1005; // Coefficient of pressure in J/kg K
14 n=1.35; // Isentropic coefficient of air
15 R=0.281; // Real gas constant in Nm/kg K
16 rhof=740; // Density of fuel in mm of Hg
17
18 // Calculations
19 p2=(1-((V/Vc)^{2})/(2*T1*Cp)))^{(n)/(n-1)};//Pressure
       at the venturi in bar
20 V1=((R*T1)/(p1*10^5))*1000;//Initial volume in m^3/
21 V2=V1*((p1/p2)^(0.741)); //Final volume in m^3/kg
22 A2=((ma*V2)/(V*60))*10^4;//Throat area of venturi in
       cm^2
23 d=((A2*4)/3.14)^(0.5);//Diameter of venturi in cm
24 deltaPa=1-p2; // Pressure drop causing air flow in bar
25 deltaPf=0.8*deltaPa; // Pressure drop causing fuel
      flow in bar
26 \text{ Af} = (\text{mf}/60) * (10^4) / ((\text{Cf}) * (2*\text{rhof} * \text{deltaPf} * 10^5)^(1/2))
      ;//Area through which fuel flows in cm<sup>2</sup>
  df = ((Af*(4/3.14))^(1/2))*10; //Diameter of fuel jet
27
      in mm
28
29 printf('(a)The diameter of the venturi of the
```

venturi if the air speed is 90 m/s is %3.2 f cm\n (b) The diameter of the jet if the pressure drop at the jet is 0.8 times the pressure drop at the venturi is %3.4 f mm',d,df)

## Scilab code Exa 7.14 Fuel supplied

```
1 clc
2 clear
3 //Input data
4 r=14; //The compression ratio of a diesel engine
5 Vc=1; // Clearance volume in m<sup>3</sup>
6 c=0.08; //Fuel supply cut off point
7 nr=0.55; // Relative efficiency
8 H=10000; // Calorific value of fuel in kcal/kg
9 r1=1.4; //Ratio of specific heat of air
10 Vs=13; //Stroke volume in m<sup>3</sup>
11
12 // Calculations
13 rho=Vc+(c*Vs);//Cut off ratio
14 na=1-(1*(rho^r1-1)/((r^(r1-1)*r1)*(rho-1)));//Air
      standard efficiency of diesel cycle in percent
15 In=(na*nr); // Indicated thermal efficiency in percent
16 H1=(4500*60)/(In*427);//Heat in fuel supplied/1HP hr
17 W=H1/10^4; //Weight of fuel required/1HP hr
18
19 //Output
20 printf ('The weight of fuel required per 1HP hr is \%3
      .4 f kg', W)
```

#### Scilab code Exa 7.15 Fuel to be injected

```
1 clc
```

```
2 clear
3 //Input data
4 P=120; //Power developed by a six cykinder four
      stroke diesel engine
5 N=2400; //Speed in rpm
6 f=0.2; //Brake specific fuel consumption in kg/kWh
7 s=0.85; // Specific gravity of fuel
9 // Calculations
10 F=f*P; // Fuel consumed per hour in kg
11 F1=F/6; // Fuel consumed per cylinder in kg/h
12 n=(N*60)/2; //Number of cycles per hour
13 F2=(F1/n)*10^3; //Fuel consumption per cycle in gm
14 V=F2/s; //Volume of fuel to be injected per cycle in
     cc
15
16 //Output
17 printf ('The quantity kof fuel to be injected per
      cycle per cylinder is %3.4 f cc', V)
```

#### Scilab code Exa 7.16 Diameter of orifice

```
clear
//Input data
P=20;//Power developed by a four stroke diesel
    engine per cylinder in kW

N=2000;//Operating speed of the diesel engine in rpm
s=0.25;//Specific fuel consumption in kh/kW
p1=180;//Pressure of fuel injected in bar
d=25;//Distance travelled by crank in degrees
p2=38;//Pressure in the combustion chamber in bar
Cd=0.85;//Coefficient of velocity
A=30;//API in degrees
```

```
// Calculations
// Calculations
T=d/(360*(N/60)); // Duration of fuel injection in s
SG=(141.5/(131.5+A))*10^3; // Specific gravity of fuel
V=Cd*(2*(p1-p2)*10^5/SG)^(1/2); // Velocity of fuel
injection in m/s
Vf=(s/60)*P/((N/2)*SG); // Volume of fuel injected per
cycle in m^3/cycle
Na=Vf/(V*T); // Nozzle orifice area in m^2
d=(((4*Na)/3.14)^(1/2))*10^3; // Diameter of the
orifice of the fuel injector in mm
// Output
printf('The diameter of the orifice is %3.4 f mm',d)
```

#### Scilab code Exa 7.17 Total orifice area

```
1 clc
2 clear
3 //Input data
4 P=200; //Power developed by a six cylinder diesel
      engine in kW
5 N=2000; // Operating speed of the engine in rpm
6 bs=0.2; //The brake specific fuel consumption in kg/
     kWh
7 p1=35; //The pressure of air in the cylinder at the
      beginning of injection in bar
8 p2=55; //Maximum cylinder pressure in bar
9 p3=180; // Initial injection pressure in bar
10 p4=520; //Maximum pressure at the injector in bar
11 Cd=0.75; // Coefficient of discharge
12 S=850; // Specific gravity of fuel
13 p5=1; // Atmospheric pressure in bar
14 a=16; //The crank angle over which injection takes
      place in degrees
15
```

```
16 // Calculations
17 Po=P/6; //Power output per cylinder in kW
18 F=(Po*bs)/60;//Fuel consumed per cylinder in kg/min
19 Fi=F/(N/2); //Fuel injected per cycle in kg
20 T=a/(360*(N/60)); //Duration of injection in s
21 deltaP1=p3-p1; // Pressure difference at the beginning
       of injection in bar
22 deltaP2=p4-p2; // Pressure difference at the end of
      injection in bar
  avP=(deltaP1+deltaP2)/2;//Average pressure
      difference in bar
V = Cd * (2*(avP*10^5)/S)^(1/2); // Velocity of injection
      of fuel jet in m/s
25 Vo=Fi/S; //Volume of fuel injected per cycle in m<sup>3</sup>/
      cycle
26 A=(Vo/(V*T))*10^6; //Area of fuel orifices in mm^2
27
28 //Output
29 printf ('The total orifice area required per injector
       if the injection takes place over 16 degree
      crank angle is %3.4 f mm<sup>2</sup>, A)
```

# Scilab code Exa 7.18 Indicated power

```
1 clc
2 clear
3 //Input data
4 A=450; //Area of indicator diagram in mm^2
5 l=60; //Length of indicator diagram in mm
6 s=1.1; //Spring number in bar/mm
7 d=0.1; //Diameter of piston in m
8 L=0.13; //Length of stroke in m
9 N=400; //Operating speed of the engine in rpm
10
11 //Calculations
```

#### Scilab code Exa 7.19 BHP

```
1 clc
2 clear
3 //Input data
4 d=25; // Diameter of the bore in cm
5 1=0.4; //Stroke length in m
6 N=300; // Operating speed of the engine in rpm
7 n=120; //Number of explosions per minute
8 pm=6.7; //Mean effective pressure in kgf/cm<sup>2</sup>
9 Tnet=90; //Net brake load in kg
10 R=0.75; //Radius of brake drum in m
11 f=0.22; //Fuel supplied per minute in m<sup>3</sup>
12 C=4500; // Calorific value of fuel in kcal/m<sup>3</sup>
13
14 // Calculations
15 BHP=(2*3.14*R*N*Tnet)/4500;//Brake horse power in kW
16 A=(3.14*d^2)/4; // Area of the cylinder in cm<sup>2</sup>
17 IHP=(pm*1*A*n)/4500;//Indicated horse power in kW
18 H=f*C; //Heat supplied by fuel per minute in kcal
19 nt1=((IHP*C)/(990*427))*100;//Thermal efficiency on
      IHP basis in percent
20 nt2=((BHP*C)/(990*427))*100; //Thermal efficiency on
```

```
BHP basis in percent

21
22 //Output
23 printf('(a)The brake horse power is %3.2 f kW\n (b)
    Indicated horse power is %3.3 f kW\n (c)Thermal
    efficiency on IHP basis is %3.2 f percent\n (d)
    Thermal efficiency on BHP basis is %3.2 f percent'
    ,BHP,IHP,nt1,nt2)
```

## Scilab code Exa 7.20 IHP

```
1 clc
2 clear
3 //Input data
4 D=0.6; //Brake wheel diameter of a constant speed
      compression ignition engine operating on four
      stroke cycle in m
5 t=0.01; // Thickness of brake band in m
6 N=500;//Operating speed of the engine in rpm
7 W=20; //Load on brake band in kgf
8 S=3; // Spring balance reading in kgf
9 1=6.25; //Length of indicator diagram in cm
10 A=4.35; // Area of indicator diagram in cm<sup>2</sup>
11 Sn=11; //Spring number in kgf/cm<sup>2</sup>/cm
12 d=10; // Diameter of the bore in cm
13 L=0.13; //Length of the stroke in m
14 F=0.23; // Specific fuel consumption in kg/BHP hr
15 CV=10000; // Heating value of fuel in kcal/kg
16
17 // Calculations
18 BHP=(3.14*(D+t)*N*(W-S))/4500; //Brake horse power in
19 MEP=(A*Sn)/1;//Mean effective pressure in kgf/cm<sup>2</sup>
20 Ar=(3.14*d^2)/4; //Area of the cylinder in cm<sup>2</sup>
21 np=N/2; //Number of explosions per minute
```

```
22 IHP=(MEP*L*Ar*np)/4500;//Indicated horse power in kW
23 nm=(BHP/IHP)*100; // Mechanical efficiency in
      percentage
24 Wf=F*BHP; //Fuel consumption per hr in kg/hr
25 nt=((IHP*4500*60)/(Wf*CV*427))*100;//Indicated
      thermal efficiency in percentage
26 nb=((BHP*4500*60)/(Wf*CV*427))*100;//Brake thermal
      efficiency in kW
27
28 //Output
29 printf('(a)The brake horse power is \%3.2 f kW\n (b)
     Indicated horse power is %3.3 f kW\n (c) Mechanical
       efficiency is %3.1f percent\n (d) Indicated
     thermal efficiency is %3.0f percent\n (e)Brake
     thermal efficiency is %3.1f percent', BHP, IHP, nm,
     nt, nb)
```

## Scilab code Exa 7.21 Indicated thermal efficiency

```
1 clc
2 clear
3 //Input data
4 N=1200; // Operating speed of a four cylinder engine
      in rpm
  BHP=25.3; //The brake horse power when all 4
      cylinders are operating in kW
6 T=10.5; //The average torque when one cylinder was
      cut out in mkgf
  CV=10000; // Calorific value of the fuel used in kcal/
     kg
8 f=0.25; //The amount of petrol used in engine per BHP
      hour
9 J=427; //
10
11 // Calculations
```

#### Scilab code Exa 7.22 IHP

```
1 clc
2 clear
3 //Input data
4 B=32; //Brake horse power in kW with all cylinders
      working
5 B1=21.6; //BHP with number 1 cylinder cut out in kW
6 B2=22.3; //BHP with number 2 cylinder cut out in kW
7 B3=22.5; //BHP with number 3 cylinder cut out in kW
8 B4=23; //BHP with number 4 cylinder cut out in kW
10 // Calculations
11 I1=B-B1; // Indicated horse power of number 1 cylinder
      in kW
12 I2=B-B2; //IHP of number 2 cylinder in kW
13 I3=B-B3; //IHP of number 3 cylinder in kW
14 I4=B-B4; //IHP of number 4 cylinder in kW
15 I=I1+I2+I3+I4; // Total IHP of the engine in kW
```

# Scilab code Exa 7.23 Compression ratio

```
1 clc
2 clear
3 //Input data
4 r=15; //The air fuel ratio by weight
5 CV=45000; // Calorific value of fuel in kJ/kg
6 nm=85; // Mechanical efficiency of 4 stroke 4 cylinder
       engine in percent
7 na=53; // Air standard efficiency of the engine in
      percent
8 nr=65; // Relative efficiency of the engine in percent
9 nv=80; // Volumetric efficiency of the engine in
      percent
10 r1=1.3; // Stroke to bore ratio
11 p1=1; // Suction pressure in bar
12 T=303; // Suction temperature in K
13 S=3000; //The operating speed of the engine in rpm
14 P=75; //Power at brakes in kW
15 r2=1.4; //Ratio of specific heats for air
16 R1=0.287; // Characteristic gas constant for air fuel
      mixture in kJ/kg K
17
18 // Calculations
19 R=(1/(1-(na/100)))^(1/(r2-1)); //Compression ratio of
       the engine
20 nti = ((na/100) * (nr/100)) * 100; //The indicated thermal
      efficiency in percent
21 Pi=P/(nm/100);//Indicated power in kW
```

```
22 F=Pi/((nti*CV)/100);//Fuel per second injected in kg
      /sec
23 B=F/P; //Brake specific fuel consumption in kg/kWsec
24 A=1+r; //Mass of fuel mixture entering the engine foe
       every one kg of fuel in kg
25 m=A*F; //Mass of air fuel mixture per second in kg
26 \text{ V} = (m*R1*T)/(p1*10^5/1000); //Volume of air fuel
      mixture supplied to the engine per sec
27 Vs=V/(nv/100);//Swept volume per second in m<sup>3</sup>/sec
28 d=((Vs*2*60*4)/(S*3.14*r1*4))^(1/3)*1000;//Diameter
      of the bore in mm
29 L=r1*d; //Stroke length in mm
30
31 // Output
32 printf('(a) Compression ratio is \%3.1 \,\mathrm{f} \, \ln \,(b)
      Indicated thermal efficiency is \%3.1f percent\n (
      c) Brake specific fuel consumption is \%3.7 f kg/kW
      sec\n (d) Bore diameter of the engine is \%3.1 f mm\
      n (e) Stroke length of the engine is %3.1 f mm', R,
      nti,B,d,L)
```

#### Scilab code Exa 7.24 Heat balance

```
clear
clear
//Input data
d=0.3;//Diameter of the bore in m
L=0.45;//Stroke length in m
N=220;//Operating speed of the engine in rpm
T=3600;//Duration of trial in sec
F=7;//Fuel consumption in kg per minute
CV=45000;//Calorific value of fuel in kJ/kg
A=320;//Area of indicator diagram in mm^2
1=60;//Length of indicator diagram in mm
2S=1.1;//Spring index in bar/mm
```

```
13 W=130; //Net load on brakes in kg
14 D=1.65; //Diameter of brake drum in m
15 W1=500; // Total weight of jacket cooling water in kg
16 t=40; //Temperature rise of jacket cooling water in
      degrees celsius
17 t1=300; // Temperature of exhaust gases in degrees
      celsius
18 ma=300; // Air consumption in kg
19 sg=1.004; // Specific heat of exhaust gas in kJ/kgK
20 sw=4.185; // Specific heat of water in kJ/kgK
21 t2=25; //Room temperature in degrees celsius
22 \text{ g=9.81;} // \text{gravity}
23
24 // Calculations
25 P = (W*g*3.14*D*N)/(1000*60); //Power available at
      brakes in kW
26 pm=(A*S)/1;//Mean effective pressure in bar
27 I = (pm*10^5*L*((3.14*d^2)/4)*N)/(1000*2*60); //
      Indicated power developed in kW
28 nm=(P/I)*100; // Mechanical efficiency in percent
29 nt=(P/((F/T)*CV))*100;//Brake thermal efficiency in
      percent
30 ni=(I/((F/T)*CV))*100; //Indicated thermal efficiency
       in percent
31 Hs=F*CV; // Heat supplied on one hour basis
32 Hp=P*T; //Heat equivalent of brake power in kJ
33 Hf=I-P; //Heat lost in friction in kJ
34 Hc=W1*t*sw; //Heat carried away by cooling water in
35 He=(ma+F)*(t1-t2)*sg; //Heat carried away by exhaust
      gas in kJ
36 Hu=Hs-(He+Hf+Hc+He); // Heat unaccounted in kJ
37 nb=(He/Hs)*100; //Heat equivalent of power at brakes
     in percent
38 nf=(Hf/Hs)*100;//Heat lost in friction in percent
39 nw=(Hc/Hs)*100; //Heat removed by jacket water in
      percent
40 ne=(He/Hs)*100; // Heat carried away by exhaust gases
```

#### Scilab code Exa 7.25 BHP

```
1 clc
2 clear
3 //Input data
4 d=25; //The bore diameter of a single cylinder 4
      stroke engine in cm
5 1=0.38; //Stroke length in m
6 t=3600; // Duration of test in sec
7 r=19710; // Total number of revolutions
8 F=6.25; // Fuel oil used in kg
9 A=5.7; // Area of indicator diagram in cm<sup>2</sup>
10 L=7.6; //Length of indicator diagram in cm
11 S=8.35; //Spring number in kgf/cm<sup>3</sup>
12 P=63.5; //Net load on brake drum in kg
13 R=1.2; // Radius of brake drum in m
14 Ww=5.7; //Rate of coolant flow in kg/min
15 deltaT=44; // Temperature rise of coolant in degrees
      celsius
16 T1=15.5; // Atmospheric temperature in degrees celsius
17 As=30; // Air supplied per kg of fuel
18 CV=10600; // Calorific value of fuel in kcal/kg
19 Te=390; //Exhaust gas temperature in degrees celsius
20 sm=0.25; //Mean specific heat of exhaust gas
```

```
21
22 // Calculations
23 Hs=(F*CV)/60; //Heat supplied by fuel per minute in
      kcal
24 pm=(A*S)/L;//Mean effective pressure in kgf/cm^2
25 I = (pm*1*(3.14*d^2)*r)/(4*60*2*4500); //Indicated
      horse power in kW
26 B=(P*R*2*3.14*r)/(4500*60);//Brake horse power in kW
27 Hei=(I*4500)/427; //Heat equivalent of IHP/min in
      kcal
28 Heb=(B*4500)/427; //Heat equivalent of BHP/min in
29 Hf=Hei-Heb; //Heat in friction per minute in kcal
30 Hc=Ww*deltaT; // Heat carried away by coolant in kcal
31 We=(F+(As*F))/60;//Weight of exhaust gases per
      minute
32 He=We*(Te-T1)*sm;//Heat carried away by exhaust
      gases in kcal
33
34 //Output
35 printf('(a) Indicated horse power is \%3.2 \,\mathrm{f} \,\mathrm{kcal} \,\mathrm{n} (b)
      Brake horse power developed is %3.2 f kcal\n (c)
      Heat equivalent of friction is \%3.1f kcal', I,B,Hf
      )
```

Scilab code Exa 7.26 Percentage of heat carried away by exhaust gas

```
1 clc
2 clear
3 //Input
4 F=10; // Quantity of fuel supplied during the trial of
        a diesel engine in kg/hr
5 CV=42500; // Calorific value of fuel in kJ/kg
6 r=20; // Air fuel ratio
7 T=20; // Ambient temperature in degrees celsius
```

```
8 mw=585; //Water circulated through the gas
      calorimeter in litres/hr
9 T1=35; // Temperature rise of water through the
      calorimeter in degrees celsius
10 T2=95; // Temperature of gases at exit from the
      calorimeter in degrees celsius
11 se=1.05; // Specific heat of exhaust gases in kJ/kgK
12 sw=4.186; // Specific heat of water in kJ/kgK
13
14 // Calculations
15 M=(F/60)*(r+1); //Mass of exhaust gases formed per
     minute
16 H = ((mw/60)*sw*T1) + (M*se*(T2-T)); // Heat carried away
     by the exhaust gases per minute in kJ/min
17 Hs=(F/60)*CV; //Heat supplied by fuel per minute in
     kJ/min
18 nh=(H/Hs)*100; // Percentage of heat carried away by
      the exhaust gas
19
20 //Output
21 printf ('Percentage of heat carried away by exhaust
      gas is %3.2f percent',nh)
```

### Scilab code Exa 7.27 Percentage of heat carried away by exhaust gases

```
1 clc
2 clear
3 //Input data
4 F=11; //Fuel used per hour observed during the trial
    of a single cylinder four stroke diesel engine in
        kg
5 mc=85; //Carbon present in the fuel in percent
6 mh=14; //Hydrogen present in the fuel in percent
7 mn=1; //Non combustibles present in the fuel in
        percent
```

- 8 CV=50000; // Calorific value of fuel in kJ/kg
- 9 Vc=8.5; // Percentage of carbon dioxide present in exhaust gas by Volumetric analysis
- 10 Vo=10; //Oxygen present in exhaust gases in percent
- 11 Vn=81.5; // Nitrogen present in exhaust gases in percent
- 12 Te=400; // Temperature of exhaust gases in degrees celsius
- 13 se=1.05; // Specific heat of exhaust gas in kJ/kg
- 14 Pp=0.030; // Partial pressure of steam in the exhaust in bar
- 15 Ta=20; // Ambient temperature in degrees celsius
- 16 hs=2545.6; //Enthalpy of saturated steam in kJ/kg
- 17 Tsa=24.1; // Saturation temperature from graph in degrees celcius
- 18 Cp=2.1; // Specific heat in kJ/kg K
- 19 hst=3335; //Enthalpy of super heated steam in kJ/kg 20
- 21 // Calculations
- 22 Ma=(Vn\*mc)/(33\*Vc);//Mass of air supplied per kg of fuel in kg
- 23 Me=Ma+1; // Mass of exhaust gases formed per kg of fuel in kg
- 24 me=(Me\*F)/60;//Mass of exhaust gases formed per minute in kg
- 25 ms=F\*(mh/100);//Mass of steam formed per kg of fuel in kg
- 26 ms1=(ms\*F)/60;//Mass of steam formed per minute in kg
- 27 mde=me-ms1;//Mass of dry exhaust gases formed per minute in kg
- 28 H=mde\*se\*(Te-Ta); // Heat carried away by the dry exhaust gases per minute in kJ/min
- 29 Es=hs+(Cp\*(Te-Tsa)); // Enthalpy of superheated steam in kJ/kg
- 30 He=ms1\*hst;//Heat carried away by steam in the exhaust gases in kJ/min
- 31 Hl=H+He; // Total heat lost through dry exhaust gases

```
and steam in kJ/min

32 Hf=(F/60)*CV;//Heat supplied by fuel per minute in kJ/min

33 nh=(H1/Hf)*100;//Percentage of heat carried away by exhaust gases

34 //Output

36 printf('Percentage of heat carried away by exhaust gases is %3.1f percent',nh)
```

Scilab code Exa 7.28 Increase in brake power of engine due to supercharging

```
1 clc
2 clear
3 //Input data
4 C=0.0033; //The capacity of a four stroke engine of
      compression ignition type
5 I=13; // Average indicated power developed in kW/m<sup>3</sup>
6 N=3500; // Operating speed of the engine
7 nv=80; // Volumetric efficiency in percentage
8 p1=1.013; // Initial pressure in bar
9 T1=298; // Initial temperature in K
10 r=1.75; // Pressure ratio of the engine
11 ni=75; //The isentropic efficiency in percentage
12 nm=80; //mechanical efficiency in percentage
13 r1=1.4; // Polytropic index
14
15 // Calculations
16 Vs = (N/2) *C; //Swept volume in m^3/min
17 Vi=Vs*(nv/100);//Unsupercharged engine inducted
     volume in m<sup>3</sup>/min
18 Pb=p1*r; //Blower delivery pressure in bar
19 T2s=((r)^{(r1-1)/r1})*T1;//Final temperature in K
20 T2=((T2s-T1)/(ni/100))+T1;//Blower delivery
```

```
temperature in K
21 Ve=((Pb*Vs)*T1)/(T2*p1); // Equivalent volume at 1.013
       bar and 298K in m<sup>3</sup>/min
  Vin=Ve-Vi; // Increase in inducted volume of air in m
22
      ^3/\min
23 Pin=Vin*I; // Increase in indicated power due to extra
       air inducted in kW
24 Pinp = ((Pb-p1)*Vs*100)/60; //Increase in indicated
     power due to increase in induction pressure in kW
25 Pt=Pin+Pinp; // Total increase in indicated power in
     kW
26 nb=Pt*(nm/100);//Total increase in brake power
      efficiency in kW
  ma=(Pb*Vs*100)/(60*0.287*T2);//Mass of air delivered
      by the blower in kg/s
  Wb=ma*1.005*(T2-T1); //Work input to air by blower in
      kW
  Pb1=Wb/(nv/100); //Power required to drive the blower
       in kW
30 Pb2=nb-Pb1; //Net increase in brake power in kW
31
32 // Output
```

33 printf ('The net increase in brake power is \%3.2 f kW'

,Pb2)

## Chapter 8

## Steam Nozzles and Turbines

#### Scilab code Exa 8.1 Final velocity of steam

```
1 clc
2 clear
3 //Input data
4 P1=12; // Pressure of Dry saturated steam entering a
     steam nozzle in bar
5 P2=1.5; // Discharge pressure of Dry saturated steam
     in bar
6 f=0.95; // Dryness fraction of the discharged steam
7 1=12; // Heat drop lost in friction in percentage
8 hg1=2784.8; // Specific enthalpy of steam at 12 bar
     from steam tables in kJ/kg
9 hg2=2582.3; // Specific enthalpy of 0.95 dry steam at
      1.5 bar from steam tables in kJ/kg
10
11 // Calculations
12 hd=hg1-hg2; //Heat drop in kJ/kg
13 V1=44.72*(hd)^(0.5);//Velocity of steam at discharge
      from the nozzle in m/s
14 n=1-(1/100); //Nozzle coefficient when 12 percent
     heat drop is lost in friction
15 V2=44.72*(n*hd)^(0.5); // Velocity of steam in m/s
```

```
16 percentV=((V1-V2)/V1)*100;//Percentage reduction in
         velocity
17
18 //Output
19 printf('(a) Final velocity of steam is %3.1 f m/s\n (b
     ) Percentage reduction in velocity is %3.2 f
     percent', V1, percentV)
```

#### Scilab code Exa 8.2 Mass of steam discharged

```
1 clc
2 clear
3 //Input data
4 P1=12; // Initial pressure of dry saturated steam
      expanded in a nozzle in bar
5 P2=0.95; // Final pressure of dry saturated steam
      expanded in a nozzle in bar
6 f=10; // Frictional loss in the nozzle of the total
      heat drop in percentage
7 d=12; //Exit diameter of the nozzle in mm
8 hd=437.1; //Heat drop in kJ/kg from steam tables
9 q=0.859; // Dryness fraction of steam at discharge
      pressure
10 vg=1.777; // Specific volume of dry saturated steam at
       0.95 bar
11
12 // Calculations
13 n=1-(f/100); // Nozzle coefficient from moiller chart
14 V2=44.72*(n*hd)^(0.5); // Velocity of steam at nozzle
      exit in m/s
15 A = (3.14/4) * (0.012)^(2); //Area of the nozzle at the
      exit in mm<sup>2</sup>
16 m = ((A*V2)/(q*vg))*3600; //Mass of steam discharged
      through the nozzle per hour in kg/hour
17
```

#### Scilab code Exa 8.3 Throat area

```
1 clc
2 clear
3 //Input data
4 P1=12; // Inlet pressure of steam nozzle in bar
5 T1=250; // Inlet temperature of steam nozzle in
      degrees celcius
6 P2=2; // Final pressure of the steam nozzle in bar
7 n=1.3; // Polytropic constant for superheated steam
8 St=6.831; //For isentropic expansion, entropy remains
       constant in kJ/kg
9 h1=2935.4//Enthalpy of steam at P1 from steam table
     in kJ/kg
10 ht=2860; //Enthalpy of steam at pt in kJ/kg
11 vt=0.325; // Specific volume of steam at the throat
      conditions in m<sup>3</sup>/kg
12 m=0.2; // Mass of steam discharged through the nozzle
     in kg/hour
13 q=0.947; //The dryness fraction of steam at exit from
       steam tables
14 hg=2589.6; //Enthalpy of steam at exit in kJ/kg
15 vs=0.8854; //Specific volume of saturated steam in m
      ^3/kg
16
17 // Calculations
18 pt=(P2/(n+1))^(n/(n-1))*P1;//Critical pressure ratio
      i.e., Throat pressure in bar
19 Vt = (2*1000*(h1-ht))^(0.5); // Velocity of steam at
     throat in m/s
```

#### Scilab code Exa 8.4 Final exit velocity of steam

```
1 clc
2 clear
3 //Input data
4 P1=10; // Pressure of steam in bar
5 f=0.9; //Dryness fraction of steam
6 At=350; // Throat area in mm<sup>2</sup>
7 Pb=1.4; //Back pressure in bar
8 h1=2574.8; //Enthalpy of steam at nozzle inlet from
      steam tables in kJ/kg
9 ft=0.87; //Dryness fraction of steam at throat
      pressure
10 fe=0.81; // Dryness fraction of steam at exit pressure
11 ht=2481; //Enthalpy of steam at throat pressure at ft
       in kJ/kg
12 vt=0.285; // Specific volume of steam at throat in m
      ^3/\mathrm{kg}
13 he=2266.2; //Enthalpy of steam at exit conditions in
      kJ/kg
14 ve=1.001; // Specific volume of steam at exit
      conditions in m<sup>3</sup>/kg
```

```
15
16 // Calculations
17 Pt=0.582*P1;//Steam pressure at the throat in bar
18 hd=h1-ht; //Enthalpy drop upto the throat in kJ/kg
19 Vt=44.7*(hd)^(0.5); // Velocity of steam at the throat
       in m/s
20 hde=h1-he; //Enthalpy drop from nozzle entrance to
      exit in kJ/kg
21 Ve=44.7*(hde)^(0.5); //Velocity of steam at nozzle
      exit in m/s
22 Ae=(At*Vt*ve)/(Ve*vt);//Exit area of nozzle from the
       mass rate of flow equation in mm<sup>2</sup>
23
24 // Output
25 printf('(a) Final exit velocity of steam is \%3.1 f m/s
      \n (b) Cross sectional area of the nozzle at exit
      for maximum discharge is %3.0 f mm<sup>2</sup>', Ve, Ae)
```

#### Scilab code Exa 8.5 Velocity of steam at the throat

```
1 clc
2 clear
3 //Input data
4 P1=7; //Inlet pressure of a convergent divergent
    steam nozzle in bar
5 T1=275; //Inlet temperature of the nozzle in degrees
    celcius
6 P2=1; // Discharge pressure of steam in bar
7 l=60; // Length of diverging portion of the nozzle in
    mm
8 dt=6; // Diameter of the throat in mm
9 f1=10; // Percent of total available enthalpy drop
    lost in friction in the diverging portion in
    percentage
10 h1=3006.9; // Enthalpy of steam at 7bar pressure and
```

```
275 degrees celcius in kJ/kg
11 ht=2865.9; //Enthalpy at the throat from Moiller
      chart in kJ/kg
12 he=2616.7; //Enthalpy at the exit from moiller chart
      in kJ/kg
13 vt=0.555; // Specific volume of steam at throat in m
      ^3/\mathrm{kg}
14 Tt=202.8; // Temperature of steam at throat in degrees
       celcius from moiller chart
15 ve=1.65; //Volume of steam at exit in m<sup>3</sup>/kg
16
17 // Calculations
18 Pt=0.546*P1; //The throat pressure for maximum
      discharge in bar
19 hd=h1-ht; //Enthalpy drop upto throat in kJ/kg
20 Vt=44.7*(hd)^(0.5);//Velocity of steam at throat in
     m/s
21 hid=h1-he; // Total isentropic drop from 7 bar, 275
      degrees celcius to 1 bar in kJ/kg
22 hda=(1-(f1/100))*(hid);//Actual heat drop in kJ/kg
23 Ve=44.7*(hda)^(0.5); // Velocity at exit in m/s
24 At = (3.14/4)*(6/1000)^(2); //Throat area of the nozzle
       in m<sup>2</sup>
25 m=(At*Vt)/vt;//Mass flow rate at nozzle throat in kg
26 Ae=((m*ve)/Ve)*10^4; //Exit area of the nozzle in cm
27 \text{ de} = (((Ae*4)/3.14)^{(0.5)})*10; //Diameter of the nozzle
       at exit in mm
28 alpha=atand((de-dt)/(2*60));//Half of the cone angle
       of the nozzle in degrees
29 alpha1=2*alpha; //Cone angle of the nozzle in degrees
30
31 //Output
32 printf('(a) Velocity of steam at throat is %3.0 f m/s
      n (b) Temperature of steam at the throat is \%3.1f
      degrees celcius \n (c) Cone angle of the divergent
      portion is %3.3 f degrees', Vt, Tt, alpha1)
```

## Chapter 9

# Air compressors

#### Scilab code Exa 9.1 Isothermal compression

```
1 clc
2 clear
3 //Input data
4 m=1; //Mass of air that has to be compressed in kg
5 P1=1; // Initial pressure of a single stage
      reciprocating air compressor in bar
6 P2=6; // Final pressure in bar
7 T1=303; // Initial temperature of air in K
8 n=1.2; // Polytropic index of air
9 R=287; //Gas constant for air in J/kg K
10 r=1.4; // Isentropic index
11
12 // Calculations
13 W1 = (m*R*T1*log(P2/P1))/1000; //Work required for
     compression in kJ/kg in Isothermal compression
     process
14 W2=((n/(n-1))*m*R*T1*((P2/P1)^((n-1)/n)-1))/1000;//
     Work required for compression in a polytropic
     compression process in kJ/kg
15 W3=((r/(r-1))*m*R*T1*((P2/P1)^((r-1)/r)-1))/1000;//
     Work required for compression in a Isentropic
```

```
compression process in kJ/kg
17 // Output
18 printf('(a) Work required in a isothermal compression
       is \%3.3 \, f \, kJ/kg \, n(b) Work required in a
```

polytropic compression is  $\%3.3 \,\mathrm{f} \,\mathrm{kJ/kg} \,\mathrm{n(c)}$  Work required in a isentropic compression is \%3.3 f kJ/

## Scilab code Exa 9.2 Size of the cylinder

kg',W1,W2,W3)

16

```
1 clc
2 clear
3 //Input data
4 Pi=60000; //Indicated power of a double acting air
     compressor in W
5 P1=1; // Initial pressure in bar
6 T1=293; // Initial temperature in K
7 n=1.2; // Polytropic index of the process
8 P2=8; // Final pressure in bar
9 N=120; //Speed at which the cylinder operates in rpm
10 S=150; // Average piston speed in m/min
11
12 // Calculations
13 L=S/(2*N); //Length of the stroke in m
14 X=(3.14*L)/4; //X=V/D^2 i.e., Volume of air before
     compression/square of the diameter in m
15 Y = ((n/(n-1))*P1*10^5*X*(((P2/P1)^((n-1)/n))-1)); //Y=
     W/D<sup>2</sup> Work done by the compressor per cycle in N/
     m
16 Nw=2*N; //Number of working strokes per minute since
     it is a double acting cylinder
17 D=(((Pi*60)/(Y*Nw))^{(0.5)})*1000; //Diameter of the
      cylinder in mm
18
```

#### Scilab code Exa 9.3 Indicated power

```
1 clc
2 clear
3 //Input data
4 D=0.15; // Diameter of a cylinder of a single acting
      reciprocating air compressor in m
5 L=0.2; //Length of the stroke in m
6 P1=1; //The pressure at which compressor sucks air in
       bar
7 P2=10; // Final pressure in bar
8 T1=298; // Initial Temperature in K
9 N=150; // Operating speed of the compressor in rpm
10 n=1.3; // Polytropic index of the process
11
12 // Calculations
13 V1 = ((3.14*D^2*L)/4); //Volume of air before
      compression in m<sup>3</sup>
14 W = ((n/(n-1))*P1*10^5*V1*((P2/P1)^((n-1)/n)-1)); //
      Work done by the compressor for a polytropic
      compression of air in Nm
15 Pi = ((W*N)/60)/1000; //Indicated power of the
      compressor in kW
16
17 //Output
18 printf ('The indicated power of the compressor is \%3
      .3 f \text{ kW}', \text{Pi})
```

Scilab code Exa 9.4 Mass of air delivered per minute

```
1 clc
2 clear
3 //Input data
4 D=0.25; // Diameter of the cylinder of a single acting
       air compressor in m
5 L=0.4; //Length of the stroke in m
6 P1=1; // Initial Pressure of the compressor in bar
7 T1=303; // Initial temperature of the compressor in K
8 P2=6; // Pressure during running in bar
9 N=250; // Operating speed of the compressor in rpm
10 R=287; //Gas constant in J/kg K
11
12 // Calculations
13 V1=(3.14*D^2*L)/4; //Volume of air before compression
       in m<sup>3</sup>
14 m=(P1*10^5*V1)/(R*T1);//Mass of air delivered by the
       compressor per stroke in kg/stroke
15 Nw=N; // Since single acting cylinder number of
      working stroke is equal to Operating speed of the
       compressor in rpm
16 ma=m*Nw; // Mass of air delivered per minute in kg/min
17
18 //Output
19 printf ('Mass of air delivered per minute is %3.2 f kg
     /\min', ma)
```

#### Scilab code Exa 9.5 Temperature

```
7 T1=308; // Inlet air temperature in K
8 n=1.3; // Polytropic index
9
10 // Calculations
11 T2=T1*(P2/P1)^((n-1)/n); // Temperature of air delivered by the compressor in K
12
13 // Output
14 printf('Temperature of air delivered by the compressor is %3.2 f K', T2)
```

#### Scilab code Exa 9.6 Isentropic compression

```
1 clc
2 clear
3 //Input data
4 P1=1; // Pressure at which air is sucked by a
     compressor in bar
5 T1=293; // Initial temperature in K
6 P2=9; // Delivery pressure after compression in bar
7 r=1.41; // Isentropic index
8 n=1.3; // Polytropic index
9
10 // Calculations
11 T21=T1*((P2/P1)^((r-1)/r)); //Temperature at the end
      of isentropic compression process in K
12 T22=T1*((P2/P1)^((n-1)/n));//Temperature at the end
      of isentropic compression process in K
13 T23=T1; // Temperature at the end of isotropic
     compression process in K (Temperature remains
     constant)
14
15 // Output
16 printf('(a) Temperature at the end of isentropic
     compression is \%3.2 f \text{ K/n} (b) Temperature at the
```

end of polytropic compression is  $\%3.2\,\mathrm{f}$  K\n (c) Temperature at the end of isotropic compression is  $\%3.0\,\mathrm{f}$  K', T21, T22, T23)

#### Scilab code Exa 9.7 Work done by air during suction

```
1 clc
2 clear
3 //Input data
4 V1=0.07; // Displacement of the piston of a single
      stage single cylinder air compressor in m<sup>3</sup>
5 P1=1; // Initial pressure in bar
6 T1=308; // Initial temperature of air in K
7 P2=8.5; // Pressure after the compression process in
8 r=1.4; // Isentropic compression
9
10 // Calculations
11 V2=V1*((P1/P2)^(1/1.4)); // Final volume of the
      cylinder in m<sup>3</sup>
12 W1=P1*10^5*V1; //Work done by air during suction in
     Nm (or) J
13 W2 = (P1 * 10^5 * V1 * (1 - (P2/P1)^((r-1)/r)))/(r-1); //Work
      done by air during compression in Nm or J
14 Wa1=P2*10^5*V2; //Work done on air during delivery in
       Nm or J
15 Wa2 = ((-W2) + Wa1 - W1) / 1000; / Net work done on air
      during the cycle in kJ
16
17 // Output
18 printf('(a) Work done by air during suction is \%3.0 f
      J\n (b) Work done on air during compression is \%3
      .0 f J\n (c) Work done on air during delivery is \%3
      .0f J\n (d) Net work done on air during the cycle
      is \%3.3 \, \text{f} \, \text{kJ}, \mbox{W1,W2,Wa1,Wa2}
```

#### Scilab code Exa 9.8 Work done on air during delivery

```
1 clc
2 clear
3 //Input data
4 V1=0.05; // displacement of a piston of a single
      cylinder single stage reciprocating compressor in
      m^3
5 P1=1; // pressure of air sucked in the compressor in
     bar
6 T1=300; // Initial Temperature of air in K
7 P2=7; // Pressure after the compression process in bar
9 // Calculations
10 V2=(P1*V1)/P2;//Volume after the compression in m<sup>3</sup>
11 W1=P1*10^5*V1; //Work done by air during suction in
     Nm
12 W2=P1*10^5*V1*log(V2/V1);//Work done on sir during
      isothermal compression in Nm
13 H=-W2; // Heat transferred to the cylinder walls in Nm
       or J
14 W3=P1*10^5*V1; //Work done on air during delivery in
15 Wn=W1+(-W2)-W3; //Net work done during the cycke in N
16
17 //Output
18 printf('(a)Work done by air during suction is %3.0f
     Nm\n (b) Work done on air during Isothermal
      compression is %3.0 f Nm\n (c) Heat transferred
      during this process is \%3.0 f J\n (d) Work done on
      air during delivery is %3.0 f Nm\n (e) Net work
      done during the cycle is %3.0 f Nm', W1, W2, H, W3, Wn)
```

#### Scilab code Exa 9.9 Power required

```
1 clc
2 clear
3 //Input data
4 m=2;//Mass of air delivered per second in kg
5 P1=1; // Initial pressure of a single stage compressor
      in bar
6 T1=293; // Initial temperature in K
7 P2=7; // Final pressure in bar
8 n=1.4; // Polytropic index
9 R=287; //Gas constant in J/kg K
10
11 // Calculations
12 W=((n/(n-1))*m*R*T1*(((P2/P1)^((n-1)/n))-1))
      /(60*1000);//Work done by compressor in kW
13
14 //Output
15 printf ('Power required to compress and deliver 2kg
      of air per minute is %3.3 f kW', W)
```

#### Scilab code Exa 9.10 Work done by compressor

```
9 n=1.3; // Polytropic index
10
11 // Calculations
12 Vs = (3.14*D^2*L)/4; //Stroke volume of the compressor
      in m<sup>3</sup>
13 Vc=0.05*Vs;//Clearance volume in m<sup>3</sup>
14 V1=Vs+Vc; // Initial volume of air in m<sup>3</sup>
15 V4=Vc*(P2/P1)^(1/n); //The air in the clearance
      volume expands during suction stroke in m<sup>3</sup>
16 V=V1-V4; // Effective swept volume in m<sup>3</sup>
17 W=((n/(n-1))*P1*10^5*(V1-V4)*(((P2/P1)^((n-1)/n))-1)
      );//Work done by the compressor per cycle in Nm
18
19 //Output
20 printf ('Work done by the compressor per cycle is \%3
      .1 f Nm', W)
```

#### Scilab code Exa 9.11 Volume of free air

```
1 clc
2 clear
3 //Input data
4 D=0.1; // Diameter of the bore of a single acting
      compressor in m
5 L=0.1; //Length of the stroke in m
6 N=400; // Operating speed of the compressor in in rpm
7 Vc=0.00008; // Clearance volume in m<sup>3</sup>
8 n=1.2; // Polytropic index
9 T1=303; // Initial temperature in K
10 Tf=293; // Final temperature in K
11 P1=0.95; // Initial pressure in bar
12 P2=8; // Final pressure in bar
13 Pf=1.013; //Free air pressure in bar
14
15 // Calculations
```

#### Scilab code Exa 9.12 Power of the compressor

```
1 clc
2 clear
3 //Input data
4 P1=1; // Pressure of air drawn by a two stage single
      acting reciprocating air compressor in bar
5 T1=293; // Initial temperature in K
6 P3=60; //Final pressure after the compression in bar
7 P2=10; // Pressure after compression in the LP
      cylinder in bar
8 T2=303; // Temperature after cooling in K
9 D=0.16; //Diameter of a cylinder in m
10 L=0.2; //Stroke length of the cylinder in m
11 n=1.3; // Polytropic index
12 N=300; // Operating speed of the compressor in rpm
13 R=287; //Gas constant in J/kg K
14
15 // Calculations
```

```
16 V1=(3.14*D^2*L)/4; //Volume of the LP cylinder in m^3
17 V2=(P1*V1*T2)/(T1*P2); //Volume of the HP cylinder in m^3
18 W=(n/(n-1))*(P1*10^5*V1*(((P2/P1)^((n-1)/n))-1)+(P2 *10^5*V2*(((P3/P2)^((n-1)/n))-1))); //Work done by the compressor per working cycle in N m
19 P=(W*N)/(60*1000); //Power of the compressor in kW
20
21 //Output
22 printf('Power of the compressor when it runs at 300 rpm is %3.3 f kW',P)
```

#### Scilab code Exa 9.13 Percentage saving in work

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial pressure in bar
5 P3=9; // Final pressure in bar
6 n=1.3; // Compression index
8 // Calculations
9 W1=(n/(n-1))*(P1*10^5*(((P3/P1)^((n-1)/n))-1));//
     Work done in compression in a single stage per
     unit volume per kg of air in N m
10 P2=(P1*P3)^(0.5); // Intercooler pressure for perfect
     intercooling in bar
11 W2=2*(n/(n-1))*(P1*10^5*(((P2/P1)^((n-1)/n))-1));
     Work done in compression in a two stage
     compressor per unit volume per kg of air in N m
12 Wc=W1-W2; //Saving in work of compression in N m
13 nw=((W1-W2)/W1)*100;//Percentage saving in work of
     compression in percentage
14
15 // Output
```

```
16 printf('Percentage saving in the work of compression
      of air in two stages instead of single stage is
      %3.2 f percent',nw)
```

#### Scilab code Exa 9.14 Work required

```
1 clc
2 clear
3 //Input data
4 m=1; //Mass of air to be compressed in kg
5 P1=1; // Pressure of air before compression in bar
6 T1=303; // Initial temperature in K
7 P3=25; //Final pressure of air after compression in
     bar
8 n=1.3; // Polytropic index
9 R=287; //Gas constant in J/kg K
10
11 // Calculations
12 P2=(P1*P3)^(0.5);//Intermediate pressure in the case
       of perfect intercooling in bar
13 W=2*(n/(n-1))*(m*R*T1*(((P2/P1)^((n-1)/n))-1)); //
     Work done in compression in a two stage
     compressor per unit volume per kg of air in N m
14
15 //Output data
16 printf ('Minimum work required to compress 1kg of air
       for given conditions is \%3.0 f N m', W)
```

#### Scilab code Exa 9.15 Power required to drive compressor

```
1 clc
2 clear
3 //Input data
```

```
4 V1=3; //Volume of air sucked in by a two stage
      compressor in m<sup>3</sup>
5 P1=1.04; // Initial pressure in bar
6 T1=298; // Initial temperature in K
7 P2=9; // Delivery pressure in bar
8 n=1.25; // Polytropic index
9
10 // Calculations
11 P2=(P1*P2)^(0.5); // Intermediate pressure for perfect
       intercooling and for minimum work of compression
       in bar
12 W=2*(n/(n-1))*(P1*10^5*V1*(((P2/P1)^((n-1)/n))-1));
     //Work done in compression in a two stage
      compressor per unit volume per kg of air in Nm
13 P=W/(60*1000);//Power required to drive the
      compressor in kW
14
15 //Output
16 printf ('The minimum power required to drive the
      compressor is %3.3 f kW',P)
```

#### Scilab code Exa 9.16 Mass of water

```
11 Cp=1; // Specific heat of air in kJ/kg K
12 Cw=4.2; // Specific heat of water in kJ/kg K
13 ma=1; //Mass of air in the compressor in kg
14
15 // Calculations
16 P2=(P1*P3)^(0.5); // Intercooler pressure for complete
       intercooling and for minimum work of compression
      in bar
17 T2=T1*(P2/P1)^((n-1)/n); //Temperature after the
      compression process in K
18 mw = (ma*Cp*(T2-T3))/(Cw*(Tc)); //Mass of water to
      circulate in the intercooler per kg of air in kg
19
20 //Output
21 printf ('Mass of water to circulate in the
      intercooler for abstracting heat is \%3.3 f kg', mw)
```

#### Scilab code Exa 9.17 Volume ratio of LP to HP cylinders

#### Scilab code Exa 9.18 Ratio of cylinder diameters

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial pressure of air entering a two stage
      air compressor with complete intercooling in bar
5 P3=25; // Delivery pressure of air toe the mains in
6 T1=303; // Initial temperature in K
7 n=1.35; // Compression index
9 // Calculations
10 P2=(P1*P3)^(0.5);//Inter cooler pressure for perfect
       intercooling in bar
11 R=(P2/P1)^(0.5); // Ratio of cylindrical diameters
12
13 //Output
14 printf ('The ratio of cylinder diameters for the
      efficiency of compression to be maximum is \%3.3 f'
      , R)
```

Scilab code Exa 9.19 Number of stages

```
1 clc
2 clear
3 //Input data
4 P1=1; // Initial pressure of a multistage compression
      in bar
5 Pn1=120; // Final pressure in bar
6 r=4;//Permissible pressure ratios per stage
8 // Calculations
9 n = \log(Pn1/P1)/\log(r)
10 n1=4; //As n=3.45 say 4 stages
11 P5=Pn1; // Since number of stages is 4
12 P4=P5/(Pn1/P1)^(1/n1); //Pressure after the stage 3
      in bar
13 P3=P4/(Pn1/P1)^(1/n1); //Pressure after the stage 2
14 P2=P3/(Pn1/P1)^(1/n1); //Pressure after the stage 1
      in bar
15
16 //Output
17 printf('(a) Number of stages are \%3.0 \text{ f} \setminus n (b)
      Intermediate pressures are, P2 = \%3.2 \, f bar, P3 =
      \%3.2 \, \text{f} \, \text{bar}, \, P4 = \%3.2 \, \text{f} \, \text{bar}, \, \text{n1,P2,P3,P4}
```

#### Scilab code Exa 9.20 Intermediate pressures

```
9
10 // Calculations
11 W=((3*n)/(n-1))*P1*10^5*V1*(((P4/P1)^((n-1)/(3*n)))
      -1);//Work done by the compressor in kJ/min
12 P=W/(60*1000); //Power required to deliver 15 m<sup>3</sup>/min
       air in kW
13 P2=P1*(P4/P1)^(1/3);//Intermediate pressure after
      stage 1 in bar
14 P3=P2*(P4/P1)^(1/3);//Intermediate pressure after
      stage 2 in bar
15
16 //Output
17 printf('(a) Power required to deliver 15 m<sup>3</sup>/min air
      at suction condition is \%3.1 f kW\n (b)
      Intermediate pressures are P2 = \%3.3 \, f bar P3 = \%3
      .3 f bar', P, P2, P3)
```

#### Scilab code Exa 9.21 Heat rejected

```
1 clc
2 clear
3 //Input data
4 P1=1; // Atmospheric pressure in bar
5 P4=60; // Delivery pressure in bar
6 T1=303; // Initial temperature in K
7 n=1.3; //Index of compression
8 Cp=1.005; // Specific heat of air at constant pressure
      in kJ/kg K
9 S=3; //Number of stages
10
11 // Calculations
12 P2=P1*(P4/P1)^(1/3); //Intermediate pressure in bar
13 T2=T1*(P2/P1)^((n-1)/n);//Temperature of air
      entering the intercoolers in K
14 H=Cp*(T2-T1); //Heat rejected in each intercooler in
```

```
kJ
15
16 //Output
17 printf('Amount of heat rejected in each intercooler is %3.0 f kJ', H)
```

#### Scilab code Exa 9.22 Ratio of cylinder volumes

```
1 clc
2 clear
3 //Input data
4 P1=1; // Pressure at the end of suction stroke in LP
      cylinder of a 3 stage single acting reciprocating
       compressor in bar
5 T1=293; // Temperature at the end of suction stroke in
      LP cylinder in K
6 V=9; //Free air delivered by the compressor in m<sup>3</sup>
7 P4=65; // Pressure delivered by the compressor in bar
8 n=1.25; // Polytropic index
9
10 // Calculations
11 P2=P1*(P4/P1)^(1/3);//Intermediate pressure after
      stage 1 in bar
12 P3=P2*(P4/P1)^(1/3);//Intermediate pressure after
      stage 2 in bar
13 V3=1; //The volume of cylinder for the third stage in
      m^3
14 V2=V3*(P3/P2);//Volume of the cylinder for second
      stage in m<sup>3</sup>
15 V1=(P2/P1)*V2;//Volume of the cylinder for first
      stage in m<sup>3</sup>
16 W = (((3*n)/(n-1))*P1*10^5*V*(((P4/P1)^((n-1)/(3*n)))
      -1))/1000;//Work done by the compressor in kJ/min
17 Pi=W/60; //Indicated power in kW
18
```

### 19 // Output

20 printf('(a)L.P. and I.P. compressor delivery pressure is  $P2 = \%3.3\,f$  bar  $P3 = \%3.2\,f$  bar\n (b)Ratio of cylinder volumes is  $V1:V2:V3 = \%3.2\,f:\%3.3\,f:\%3.0\,f$ \n (c)Total indicated power is  $\%3.2\,f$  kW',P2,P3,V1,V2,V3,Pi)

# Chapter 10

# Refrigeration and air conditioning

#### Scilab code Exa 10.1 Power rating

```
1 clc
2 clear
3 //Input data
4 T1=273; //The temperature of ice in K
5 T2=298; //Temperature of water at room in K
6 COP=2.1; //Cop of the plant
7 ne=90; // Overall electrochemical efficiency in
     percentage
8 w=15; // Weight of ice produced per day in tonnes
9 cw=4.187; // Specific heat of water in kJ/kg degrees
      celcius
10 Li=335; // Latent heat of ice in kJ/kg
11 mi=1; //Mass of ice produced at 0 degrees celcius
12
13 // Calculations
14 m = (w*1000)/(24*60); //Mass of ice produced in kg/min
15 h=(mi*cw*(T2-T1))+Li;//Heat extracted from 1kg of
     water at 25 degrees celcius to produce 1kg of ice
       at 0 degrees celcius in kJ/kg
```

```
16 Q=m*h;//Total heat extracted in kJ
17 W=Q/COP;//Work done by the compressor in kJ/kg
18 P=W/(60*(ne/100));//Power of compressor in kW
19
20 //Output
21 printf('Power rating of the compressor-motor unit if the cop of the plant is 2.1 is %3.1 f kW',P)
```

#### Scilab code Exa 10.2 Refrigeration capacity

```
1 clc
2 clear
3 //Input data
4 m=400; //Mass of fruits supplied to a cold storage in
5 T1=293; // Temperature at which fruits are stored in K
6 T2=268; // Temperature of cold storage in K
7 t=8;//The time untill which fruits are cooled in
     hours
8 hfg=105; //Latent heat of freezing in kJ/kg
9 Cf=1.25; // Specific heat of fruit
10 TR=210; //One tonne refrigeration in kJ/min
11
12 // Calculations
13 Q1=m*Cf*(T1-T2); // Sensible heat in kJ
14 Q2=m*hfg;//Latent heat of freezing in kJ
15 Q=Q1+Q2; //Heat removed from fruits in 8 hrs
16 Th=(Q1+Q2)/(t*60); // Total heat removed in one minute
       in kJ/kg
17 Rc=Th/TR; // Refrigerating capacity of the plant in TR
18
19 //Output
20 printf ('The refrigeration capacity of the plant is
     \%3.3 \, f \, TR', Rc)
```

#### Scilab code Exa 10.3 COP of a heat pump

```
1 clc
2 clear
3 //Input data
4 T1=300; //The maximum temperature at which carnot
      cycle operates in K
  T2=250; //The minimum temperature at which carnot
      cycle operates in K
7 // Calculations
8 COPr=T2/(T1-T2);//COP of the refrigerating machine
9 COPh=T1/(T1-T2)//COP of heat pump
10 n=((T1-T2)/T1)*100;//COP or efficiency of the heat
      engine in percentage
11
12 //Output data
13 printf('(a)COP of the machine when it is operated as
       a refrigerating machine is \%3.2 \text{ f} \setminus n (b)COP when
      it is operated as heat pump is \%3.2 f\n (c)COP or
      efficiency of the Heat engine is %3.2f percent',
      COPr, COPh, n)
```

#### Scilab code Exa 10.4 Time taken to achieve cooling

```
1 clc
2 clear
3 //Input data
4 m=20000;//The storage capacity of fish in a storage plant in kg
5 T1=298;//Supplied temperature of fish in K
```

```
6 T2=263; // Temperature of cold storage in which fish
      are stored in K
7 T3=268; // Freezing point of fish in K
8 Caf=2.95; // Specific heat of fish above freezing
      point in kJ/kg K
  Cbf=1.25; // Specific heat of below freezing point in
      kJ/kg K
10 W=75; //Work required by the plant in kW
11 TR=210; //One tonne refrigeration in kJ/min
12 hfg=230; //Latent heat of fish in kJ/kg
13
14 // Calculations
15 COPr=T2/(T1-T2);//COP of reversed carnot cycle
16 COPa=0.3*COPr;//Given that actual COP is 0.3 times
      of reversed COP
17 Hr = (COPa*W)*60; // Heat removed by the plant in kJ/min
18 C=Hr/TR; // Capacity of the plant in TR
19 Q1=m*Caf*(T1-T3); //Heat removed from the fish above
      freezing point in kJ
20 Q2=m*Cbf*(T3-T2);//Heat removed from fish below
      freezing point in kJ
21 Q3=m*hfg;//Total latent heat of the fish in kJ
22 Q=Q1+Q2+Q3; //Total heat removed by the plant in kJ
23 T=(Q/Hr)/60; //Time taken to achieve cooling in hrs
24
25 //Output data
26 printf('(a) Capacity of the plant is \%3.2 \text{ f TR} \setminus \text{n} (b)
      Time taken to achieve cooling is \%3.2 f hours', C, T
      )
```

#### Scilab code Exa 10.5 Theoretical COP

```
1 clc
2 clear
3 //Input data
```

```
4 T2=298; //Maximum temperature at which CO2 machine
     works in K
5 T1=268; //Minimum temperature at which CO2 machine
     works in K
6 sf1=-0.042; //Liquid entropy at 268 K in kJ/kg K
7 hfg1=245.3; //Latent heat of gas at 268 K in kJ/kg
8 sf2=0.251; //Liquid entropy in kJ/kg K
9 hfg2=121.4; //Latent heat of gas at 298 K in kJ/kg
10 hf1=-7.54; //Liquid enthalpy at 268 K in kJ/kg
11 hf2=81.3; //Liquid enthalpy at 298 K in kJ/kg
12 hf3=81.3; //Enthalpy at point 3 in graph in kJ/kg
13
14 // Calculations
15 s2=sf2+(hfg2/T2); //Entropy at point 2 from the graph
      in kJ/kg K
16 x1=(s2-sf1)/(hfg1/T1);//Dryness fraction at point 1
17 h1=hf1+(x1*hfg1); //Enthalpy at point 1 in kJ/kg
18 h2=hf2+hfg2; //Enthalpy at point 2 in kJ/kg
19 COP=(h1-hf3)/(h2-h1);//Coefficient of performance
     for a CO2 machine working at given temperatures
20
21 //Output data
22 printf ('Theoretical COP for a CO2 machine working at
      given temperatures is \%3.2 f', COP)
```

#### Scilab code Exa 10.6 Capacity of refrigerator

```
7 sf1=0.5443; //Liquid entropy at 298 K in kJ/kg K
8 sf2=1.1242; //Liquid entropy at 263 K in kJ/kg K
9 hfg1=1297.68; // Latent heat at 298 K in kJ/kg
10 hfg2=1166.94; //Latent heat at 263 K in kJ/kg
11 hf1=135.37; //Liquid enthalpy at point 1 in graph in
     kJ/kg
12 hf2=298.9; //Liquid enthalpy at point 2 in graph in
     kJ/kg
13 TR=210; //One tonne refrigeration in TR
14
15 // Calculations
16 s2=sf2+(hfg2/T2); //Entropy at point 2 in kJ/kg
17 x1=(s2-sf1)/(hfg1/T1);//Dryness fraction at point 1
18 h1=hf1+(x1*hfg1); //Enthalpy at point 1 in kJ/kg
19 h=h1-hf2; //Heat extracted of refrigerating effect
     produced per kg of refrigerant in kJ/kg
20 ht=mf*h; // Total heat extracted at a fluid flow rate
     of 5 kg/min in kJ/min
21 C=ht/TR; // Capacity of refrigerating in TR
22
23 //Output
24 printf ('The capacity of refrigerator is \%3.0 f TR', C)
```

#### Scilab code Exa 10.7 Theoretical COP

```
1 clc
2 clear
3 //Input data
4 T1=263; //Minimum temperature at which ammonia
    refrigerating machine works in K
5 T2=303; //Maximum temperature at which ammonia
    refrigerating machine works in K
6 x1=0.6; // Dryness fraction of ammonia during suction
    stroke
7 sf1=0.5443; // Liquid entropy at 263 K in kJ/kg K
```

```
8 hfg1=1297.68; // Latent heat at 263 K in kJ/kg
9 sf2=1.2037; //Liquid entropy at 303 K in kJ/kg K
10 hfg2=1145.8; // Latent heat at 303 K in kJ/kg
11 hf1=135.37; //Liquid enthalpy at 263 K in kJ/kg
12 hf2=323.08; //Liquid enthalpy at 303 K in kJ/kg
13
14 // Calculations
15 s1=sf1+((x1*hfg1)/T1);//Entropy at point 1 in kJ/kg
16 x2=(s1-sf2)/(hfg2/T2);//Entropy at point 2 in kJ/kg
     K
17 h1=hf1+(x1*hfg1); //Enthalpy at point 1 in kJ/kg
18 h2=hf2+(x2*hfg2); //Enthalpy at point 2 in kJ/kg
19 COP=(h1-hf2)/(h2-h1);//Theoretical COP of ammonia
      refrigerating machine
20
21 //Output
22 printf('The theoretical COP of a ammonia
      refrigerating machine working between given
     temperatures is %3.2 f', COP)
```

#### Scilab code Exa 10.8 Ice produced

```
1 clc
2 clear
3 //Input data
4 T1=263; //Minimum temperature at which Vapour compression refrigerator using methyl chloride operates in K
5 T2=318; //Maximum temperature at which Vapour compression refrigerator using methyl chloride operates in K
6 sf1=0.183; //Entropy of the liquid in kJ/kg K
7 hfg1=460.7; //Enthalpy of the liquid in kJ/kg 8 sf2=0.485; //Entropy of the liquid in kJ/kg K
```

```
9 hfg2=483.6; //Enthalpy of the liquid in kJ/kg
10 x2=0.95; // Dryness fraction at point 2
11 hf3=133.0; //Enthalpy of the liquid in kJ/kg
12 W=3600; //Work to be spent corresponding to 1kW/hour
13 Cw=4.187; // Specific heat of water in kJ/kg degrees
      celcius
14 mi=1; //Mass of ice produced at 0 degrees celcius
15 Li=335; // Latent heat of ice in kJ/kg
16 hf1=45.4; //Enthalpy of liquid at 263 K in kJ/kg
17 hf2=133; //Enthalpy of liquid at 318 K in kJ/kg
18
19 // Calculations
20 s2=sf2+((x2*(hfg2-hf2))/T2);//Enthalpy at point 2 in
      kJ/kg
21 \times 1 = (s2-sf1)/((hfg1-hf1)/T1); //Dryness fraction at
22 h1=hf1+(x1*hfg1); //Enthalpy at point 1 in kJ/kg
23 h2=hf2+(x2*hfg2); //Enthalpy at point 2 in kJ/kg
24 COP=(h1-hf3)/(h2-h1);//Theoretical COP
25 COPa=0.6*COP; // Actual COP which is 60 percent of
      theoretical COP
26 H=W*COPa; //Heat extracted or refrigeration effect
      produced per kW hour in kJ
27 Hw=(mi*Cw*10)+Li;//Heat extracted from water at 10
      degrees celcius for the formation of 1 kg of ice
      at 0 degrees celcius
28 I=H/Hw; //Amount of ice produced in kg/kW hr
29
30 //Output
31 printf ('The amount of ice produced is \%3.2 f kg/kW hr
      ',I)
```