## Scilab Textbook Companion for Fundamental of Thermodynamics by Moran and Shapiro<sup>1</sup>

Created by
Jatin Pavagadhi
MCA
Computer Engineering
Changa Institute, Gujarat
College Teacher
None
Cross-Checked by
Harpreeth Singh

February 9, 2015

<sup>1</sup>Funded by a grant from the National Mission on Education through ICT, http://spoken-tutorial.org/NMEICT-Intro. This Textbook Companion and Scilab codes written in it can be downloaded from the "Textbook Companion Project" section at the website http://scilab.in

# **Book Description**

Title: Fundamental of Thermodynamics

Author: Moran and Shapiro

Publisher: John Wiley, Southern Gate

Edition: 5

**Year:** 2006

**ISBN:** 9780470030370

Scilab numbering policy used in this document and the relation to the above book.

Exa Example (Solved example)

**Eqn** Equation (Particular equation of the above book)

**AP** Appendix to Example(Scilab Code that is an Appednix to a particular Example of the above book)

For example, Exa 3.51 means solved example 3.51 of this book. Sec 2.3 means a scilab code whose theory is explained in Section 2.3 of the book.

## Contents

Lis	st of Scilab Codes	4
2	Energy and the First Law of Thermodynamics	8
3	Evaluating Properties	16
4	Control Volume Analysis Using Energy	27
5	The Second Law of Thermodynamics	42
6	Using Entropy	45
7	Exergy Analysis	61
8	Vapor Power Systems	82
9	Gas Power Systems	105
10	Refrigeration and Heat Pump Systems	142
11	Thermodynamic Relations	152
<b>12</b>	Ideal Gas Mixtures and Psychrometrics Applications	167
13	Reacting Mixtures and Combustion	197
14	Chemical and Phase Equilibrium	221

# List of Scilab Codes

Exa 2.1	Example 1												8
Exa 2.2	Example .												9
Exa 2.3	Example .												10
Exa 2.4	Example .												12
Exa 2.5	Example .												13
Exa 2.6	Example .												13
Exa 3.1	Example .												16
Exa 3.2	Example .												18
Exa 3.3	Example .												18
Exa 3.4	Example .												19
Exa 3.6	Example .												21
Exa 3.7	Example .												22
Exa 3.8	Example .												24
Exa 3.9	Example .												25
Exa 3.11	Example .												26
Exa 4.1	Example .												27
Exa 4.3	Example .												28
Exa 4.4	Example .												29
Exa 4.5	Example .												30
Exa 4.6	Example .												32
Exa 4.7	Example .												33
Exa 4.8	Example .												34
Exa 4.9	Example .												36
Exa 4.10	Example .												37
Exa 4.11	Example .												38
Exa 4.12	Example .												39
Exa 5.1	Example .												42
Exa 5.2	Example .												43

Exa 5.3	Example												43
Exa 6.1	Example												45
Exa 6.2	Example												46
Exa 6.3	Example												46
Exa 6.4	Example												47
Exa~6.5	Example												48
Exa 6.6	Example												49
Exa 6.7	Example												50
Exa 6.8	Example												51
Exa 6.9	Example												54
Exa 6.10	Example												55
Exa 6.11	Example												56
Exa 6.12	Example												57
Exa 6.13	Example												58
Exa 6.14	Example												59
Exa 6.15	Example												60
Exa 7.1	Example												61
Exa 7.2	Example												62
Exa 7.3	Example												64
Exa 7.4	Example												66
Exa 7.5	Example												67
Exa 7.6	Example												69
Exa 7.7	Example												72
Exa 7.8	Example												74
Exa 7.9	Example												77
Exa 7.10	Example												78
Exa 8.1	Example												82
Exa 8.2	Example												85
Exa 8.3	Example												87
Exa 8.4	Example												89
Exa 8.5	Example												90
Exa 8.6	Example												93
Exa 8.7	Example												98
Exa 8.8	Example												100
Exa 8.9	Example												102
Exa 9.1	Example												105
Exa 9.2	Example												108
Exa 9.3	Example					-		-					111

Exa 9.4	Example													113
Exa 9.6	Example													115
Exa 9.7	Example													118
Exa 9.8	Example										 			119
Exa 9.9	Example													121
Exa 9.11	Example													123
Exa 9.12	Example										 			126
Exa 9.13	Example													129
Exa 9.14	Example													134
Exa 9.15	Example													136
Exa 10.1	Example													142
Exa 10.2	Example													144
Exa 10.3	Example													145
Exa 10.4	Example													148
Exa 10.5	Example													150
Exa 11.1	Example													152
Exa 11.3	Example													154
Exa 11.4	Example													156
Exa 11.6	Example													158
Exa 11.8	Example													159
Exa 11.9	Example													161
Exa 11.10	Example													163
Exa 12.1	Example													167
Exa 12.2	Example													169
Exa 12.3	Example													170
Exa 12.4	Example													172
Exa 12.5	Example													175
Exa 12.6	Example													177
Exa 12.7	Example													179
Exa 12.8	Example													181
Exa 12.9	Example													184
Exa 12.10	Example													185
Exa 12.11	Example													188
Exa 12.12	Example										 			190
Exa 12.13	Example										 			191
Exa 12.14	Example													193
Exa 12.15	Example													195
Exa 13 1	Example													197

Exa 13.2	Example													198
Exa 13.3	Example													200
Exa 13.4	Example													202
Exa 13.5	Example													203
Exa 13.6	Example													204
Exa 13.7	Example													205
Exa 13.8	Example													208
Exa 13.9	Example													209
Exa 13.10	Example													211
Exa 13.11	Example													213
Exa 13.12	Example													214
Exa 13.13	Example													215
Exa 13.14	Example													216
Exa 13.15	Example													218
Exa 13.16	Example													219
Exa 14.1	Example													221
Exa 14.2	Example													224
Exa 14.3	Example													225
Exa 14.4	Example													226
Exa 14.5	Example													227
Exa 14.8	Example													228
Eva 14 10	Example													229

## Chapter 2

# Energy and the First Law of Thermodynamics

#### Scilab code Exa 2.1 Example 1

```
1 // Given:-
2 p1 = 3*(10**5)
                                                     //
      initial pressure of gas in pascal
                                                     //
3 v1 = 0.1
      initial volumme of gas in meter<sup>3</sup>
4 v2 = 0.2
                                                     //
      final volume of gas in meter 3
6 // calculations
7 // Part (a) i.e. n=1.5
8 // constant = p1*(v1**n)
                                                       // p
      *(v^n) = constant
9 constant1 = p1*(v1**1.5)
10 \quad constant2 = p1*(v1**1)
11 constant3 = p1*(v1**0)
12 // function p
13 function v = p1(v)
14
       v = constant1/(v^1.5)
15 endfunction
```

```
16
17 function v = p2(v)
        v = constant2/(v^1)
18
19 endfunction
20
21 function v = p3(v)
       v = constant3/(v^0)
23 endfunction
24
25 \text{ work1} = intg(v1, v2, p1)
       integrating pdv from initial to final volume
26 \text{ w1} = \text{work1}(1)/1000
                                                            //
       divided by 1000 to convert to KJ
27 printf ('The work done for n=1.5 in KJ is \%.2 \, f', w1)
28
\frac{29}{\text{part}(b)} i.e. n = 1
30 \text{ work2} = intg(v1, v2, p2)
31 \text{ w2} = \text{work2}(1)/1000
32 printf ('The work done for n=1 in KJ is %.2f', w2)
33
34 / part(c) i.e. n=0
35 \text{ work3} = intg(v1, v2, p3)
36 \text{ w3} = \text{work3}(1)/1000
37 printf ('The work done for n=0 in KJ is %.2f', w3)
```

#### Scilab code Exa 2.2 Example

```
mass of the gas in kg
6 \text{ deltau} = -4.6
      change in specific internal energy in KJ/Kg
  // Calculations
10 constant = p1*(v1**1.5)
                                                          //
       p*(v^n) = constant
11
12 function v = p(v)
       v = constant/(v**1.5)
13
                                            // expressing
          pressure as function of volume
14 endfunction
15
16 \text{ work} = intg(v1, v2, p)
      integrating pdv from initial to final volume
17 \text{ w=work}(1)/1000
      divided by 1000 to convert to KJ
18
19 deltaU = m*deltau
      change in internal energy in KJ
20 \ Q = deltaU + w
      neglecting kinetic and potential energy changes
21
22 // Result
23 printf ('net heat transfer for the process in KJ %.2
      f',Q)
```

#### Scilab code Exa 2.3 Example

```
1 // Given:-
2 clc;
3 patm = 10**5  //
    atmospheric pressure in pascal.
```

```
4 \text{ mp} = 45.0
                                                   // mass
      of piston in Kg
5 A = 0.09
                                                   // face
      area of piston in m2
6 \text{ deltaV} = 0.045
      increment of the volume of air in m3
7 m = 0.27
                                                   // mass
      of air in kg
8 \text{ deltau} = 42.0
      specific internal energy increase of air in kJ/kg
9 g = 9.81
                                                   // local
      acceleration of gravity
10
11
12 // Part (a) i.e. air is system
13 // Calculations
14 p = (mp*g)/A + patm
                                                    //
      constant pressure of air obtained from
      equilibrium of piston
15 \text{ w} = (p*deltaV)/1000
                                                    // work
      done in KJ
16 deltaU = m*deltau
      internal energy change of air in KJ
17 Q = w + deltaU
                                                    //
      applying first with air as system
18 // Result
19 printf ('\nheat transfer from resistor to air in KJ
      for air alone as system is: %.2f',Q)
20
21 // The answer given in book is incorrect. deltaU is
      incorrect in book.
22
23 // Part(b) i.e. (air+piston) is system
24 // Calculations
25 \text{ wd} = (\text{patm*deltaV})/1000
                                                     // work
       done in KJ
26 \text{ deltaz} = (\text{deltaV})/A
      change in elevation of piston
```

#### Scilab code Exa 2.4 Example

```
1 // Given:-
2 \text{ w1dot} = -60.0
                                   // input work rate in
     KW
3 h = 0.171
                                   // heat transfer
      coefficient, unit in KW/m2
                                   .K
4 A = 1.0
                                   // outer surface area
      of gearbox, unit in m2
5 \text{ Tb} = 300.0
                                   // outer surface
      temperature in kelvin
                                   // temperature of the
6 \text{ Tf} = 293.0
      sorrounding
8 // Calculations
9 Qdot = -h*A*(Tb-Tf);
                                   // rate of energy
      transfer by heat
10 \text{ wdot} = Qdot;
                                   // steady state energy
      equation
11 w2dot = wdot-w1dot;
12
13 // Results
14 printf ('The heat transfer rate in KW is:\ln tQdot =
       %f', Qdot)
```

15 printf( 'The power delivered through output shaft in KW is: =%f', w2dot);

#### Scilab code Exa 2.5 Example

```
1 // Given:-
2 s=5*(10**-3)
                                         // measurement on
      a side in meter
3 \text{ wdot} = -0.225
                                           // power input
      in watt
4 \text{ Tf} = 293.0
                                           // coolant
     temprature in kelvin
5 h = 150.0
                                           // heat
     transfer coefficient in w/m2 k
6 A = s**2
                                          // surface area
8 // Calculation
9 Tb = ((-wdot/(h*A)) + Tf - 273)
                                          // surface
      temperature in degree
10
11 // Result
12 printf ('The surface temperature of the chip in
      degree celcius is: %f ',Tb);
```

#### Scilab code Exa 2.6 Example

```
6 Wshaftdot = (tau*omega)/1000
                                         //shaft work rate
      in KW
                                         //net work rate in
7 Wdot = Welecdot + Wshaftdot
       KW
9 //function [Qdot] = f(t)
10 // \text{Qdot} = (-0.2) * [1 - 2 * * (-0.05 * t)]
11
12
                                              //function for
13 / function [Edot] = f1(t)
       rate of change of energy
14 / Edot = (-0.2) * [1 - 2 * * (-0.05 * t)] - Wdot
15
16 //function [deltaE] = f2(t)
                                              //function for
       change in energy
17
18 t = linspace(0, 120, 100);
19 \text{ for } i = 1:100
20
       Qd(i) = i
21
       Wd(i) = i
22
       dltaE(i) = i
23
       Qd(i) = (-0.2*(1-%e^{(-0.05*t(i))}))
       Wd(i) = Wdot
24
       dltaE(i) = 4*(1 - %e^{(-0.05*t(i))})
25
26 \text{ end}
27
28 subplot (2,2,1)
29 plot(t,Qd)
30 \text{ xlabel}("Time (s)")
31 ylabel("Qdot (KW)")
32
33 subplot(2,2,2)
34 plot(t, Wd)
35 xlabel("Time (s)")
36 ylabel("Wdot (KW)")
37
38 subplot (2,2,3)
```

```
39 plot(t,dltaE)
40 xlabel("Time (s)")
41 ylabel("deltaE (KJ)")
```

## Chapter 3

## **Evaluating Properties**

#### Scilab code Exa 3.1 Example

```
1 // Given:-
2 // Those with 1 are of state 1 and 2 are with state
4 // State 1
                                    // initial pressure
5 p1 = 10**5
     in pascal
6 \times 1 = 0.5
                                    // initial quality
8 T1 = 99.63
                                    // temperature in
      degree celcius, from table A-3
                                    // volume of
9 v = 0.5
     container in m3
10 \text{ vf1} = 1.0432*(10**(-3))
                                    // specific volume of
      fluid in state 1 in m3/Kg(from table A-3)
11 \text{ vg1} = 1.694
                                    // specific volume of
       gas in state 1 in m3/kg (from table A-3)
12
13 // State 2
14 p2 = 1.5*(10**5)
                                    // pressure after
     heating in pascal
```

```
15
16 T2 = 111.4
                                     // temperature in
      degree celcius in state 2, from A-3
17 \text{ vf2} = 1.0582*(10**(-3))
                                   // specific volume of
       fluid in state 2 in m3/Kg, from A-3
18 \text{ vg2} = 1.159
                                     // specific volume of
       gas in state 2 in m3/Kg, from A-3
19
20 // Calculations
21
22 v1 = vf1 + x1*(vg1-vf1)
                                     // specific volume in
       state 1 in m3/Kg
23 v2 = v1
                                     // specific volume in
       state 2 in m3/Kg
24 m = v/v1
                                     // total mass in Kg
                                     // mass of vapour in
25 \text{ mg1} = \text{x1*m}
      state 1 in Kg
26
27 	 x2 = (v1-vf2)/(vg2-vf2)
                                     // quality in state 2
28 \text{ mg2} = x2*m
                                     // mass of vapor in
      state 2 in Kg
29
30 // State 3
31 p3 = 2.11
                                     // pressure in state
      3 from table A-3
32
33 // Results
34 printf(' The temperature in state 1 is %f degree
      celcius.',T1)
35 printf ('The temperature in state 2 is %f degree
      celcius.',T2)
36 printf (' The mass of vapour in state 1 is \%.2 \,\mathrm{f} kg.'
      ,mg1)
37 printf ('The mass of vapour in state 2 is \%.2 \,\mathrm{f} kg.'
38 printf (' The pressure corresponding to state 3 is \%
      .2 f bar', p3)
```

#### Scilab code Exa 3.2 Example

```
1 // Given:-
2 m = 0.05
                                     // mass of ammonia in
      kg
3 p1 = 1.5*(10**5)
                                     // initial pressure
      of ammonia in pascal
                                     // specific volume in
4 v1 = 0.7787
       state 1 in m3/kg from table A-14
5 v2 = 0.9553
                                     // specific volume in
       state 2 in m3/kg from table A-15
6 T2 = 25.0
                                     // final temperature
      in degree celcius
8 // Calculations
10 \ V1 = m * v1
                                     // volume occupied by
       ammonia in state 1 in m3
                                     // volume occupied by
11 \quad V2 = m * v2
       ammonia in state 2 in m3
12 \text{ w} = (p1*(V2-V1))/1000
                                     // work in KJ
13
14 // Results
15 printf ('The volume occupied by ammonia in state 1
      is \%.2 \, \text{f m} \, ^{\circ}3.
16 printf ('The volume occupied by ammonia in state 2
      is %.2 f m^3', V2)
17 printf(' The work done for the process is %.2 f KJ',
      w)
```

#### Scilab code Exa 3.3 Example

```
1 // Given:-
2 V = 0.25
                        // volume of tank in m3
                        // specific volume in m3/kg
3 v = 1.673
     obtained using table A-2
5 // State 1
6 T1 = 100.0
                        // initial temperature in
     degree celcius
7 u1 = 2506.5
                         // specific internal energy in
     state 1 in KJ/Kg obtained from table A-2
9 // State 2
10 p2 = 1.5
                         // final pressure in bars
11 T2 = 273.0
                         // temperature in state 2 in
     degree celcius obtained from table A-4
12 u2 = 2767.8
                        // specific internal energy in
      state 2 in KJ/Kg obtained from table A-4
13
14 // Calculations
15 \text{ m} = \text{V/v}
                        // mass of the system in kg
16 DeltaU = m*(u2-u1) // change in internal energy
     in KJ
17 W = - DeltaU
                        // from energy balance
18
19 // Results
20 printf (' The temperature at the final state in is \%
     .2f degree celcius.',T2)
21 printf ('The work during the process is %f KJ.',W);
```

#### Scilab code Exa 3.4 Example

```
4 T1 = 400.0
                                               // initial
      temperature in degree celcius
5 v1 = 0.3066
                                               // specific
      volume in state 1 in m3/kg obtained from table A
      -4
6 \text{ u1} = 2957.3
                                               // specific
      internal energy in state 1 in KJ/Kg obtained from
       table A-4
8 // State 2
10 \quad v2 = 0.1944
                                               // specific
      volume in state 2 in m3/kg obtained from table A
11 \text{ w2to3} = 0
                                               // work in
      process 2-3
12
13
14 // State 3
15 v3 = v2
16 \text{ vf3} = 1.0905*(10**(-3))
                                               // specific
      volume of fluid in state 3 from table A-2
17 \text{ vg3} = 0.3928
                                               // specific
      volume of gas in state 3 from table A-2
18 \text{ uf3} = 631.68
                                              // specific
      internal energy for fluid in state 3 from table A
19 \text{ ug3} = 2559.5
                                               // specific
      internal energy for gas in state 3 from table A-2
20
21 // Calculations
22 \text{ w1to2} = (P1*(v2-v1))/1000
                                               // work in KJ
      /Kg in process 1-2
23 W = w1to2 + w2to3
                                               // net work
      in KJ/kg
24 \times 3 = (v3 - vf3)/(vg3 - vf3)
25 	 u3 = uf3+x3*(ug3-uf3)
                                               // specific
      internal energy in state 3 in Kj/Kg
```

#### Scilab code Exa 3.6 Example

```
1 // Given:-
2 // State 1
3 p1 = 20.0
                                     // initial pressure
      in MPa
4 T1 = 520.0
                                     // initial
      temperature in degree celcius
5 \ Z1 = 0.83
                                     // compressibility
      factor
6 R = 8.314
                                     // universal gas
     constant in SI unit
7 n = 1000.0/18.02
                                     // number of moles in
       a kg of water
9 // State 2
10 T2 = 400.0
                                     // final temperature
      in degree celcius
11
12 // From table A-1
13 \text{ Tc} = 647.3
                                     // critical
      temperature in kelvin
14 \text{ pc} = 22.09
                                     // critical pressure
      in MPa
15
16 // Calculations
```

```
// reduced
17 \text{ Tr} = (T1+273)/Tc
      temperature
                                       // reduced pressure
18 \text{ Pr} = p1/pc
19 v1 = (Z1*n*R*(T1+273))/(p1*(10**6))
20 \text{ vr} = v1*(pc*(10**6))/(n*R*Tc)
21 \text{ Tr}2 = (T2+273)/Tc
22 \text{ PR} = 0.69
                                         // at above vr and
      Tr2
23 P2 = pc*PR
24
25 // Results
26 printf( ' The specific volume in state1 is \%f m3/kg
      and the corresponding value obtained from table A
      -4 is .01551 m<sup>3</sup>/Kg<sup>3</sup>,v1)
27 printf ('The pressure in MPa in the final state is
      %f MPa and the corresponding value from the table
       is 15.16Mpa', P2);
```

#### Scilab code Exa 3.7 Example

```
1 // Given:-
2 T1 = 300.00
                                                  //
      temperature in state 1 in kelvin
3 P1 = 1.00
                                                  //
     pressure in state 1 in bar
4 P2 = 2.00
      pressure in state 2 in bar
5 R = 287.00
                                                  //gas
      constant of air in SI units
7 // Calculations
8 v1 = (R*T1)/(P1*10**5)
                                                //specific
      volume in state 1
9 P = linspace(1,2,50)
10 \text{ for } i = 1:50
```

```
11
        v(i) = v1
12 end
13
14
15 T2 = (P2*10**5*v1)/R
16 \text{ v3} = (R*T2)/(P1*10**5)
17 \text{ vv} = linspace(v1, v3, 50)
18 \text{ for } i = 1:50
19
        Pa(i) = P1
20 end
21
22 //function [out] = f(inp)
23 //out = (R*T2)/(inp)
24
25 \text{ VV} = linspace(v1, v3, 50)
26 \text{ for } j = 1:50
        pp(j) = (R*T2)/VV(j)/(10**5)
27
28 end
29 vcommon = cat(1,v,VV')
30 \text{ pcommon} = [P \text{ pp'}]
31 size (vcommon)
32 size (pcommon)
33 / subplot (211)
34 plot (vcommon, pcommon)
35 \text{ xlabel}('v')
36 ylabel('p(bar)')
37
38 / \text{subplot}(212)
39 plot(vv,Pa)
40 xlabel('v')
41 ylabel('p(bar)')
42
43 //The two steps are shown in one graph and the other
        on is shown in the other graph"""
44
45 printf ('The temperature in kelvin in state 2 is T2
      = \%f, T2)
46 printf ('The specific volume in state 3 in m<sup>3</sup>/kg is
```

#### Scilab code Exa 3.8 Example

```
1 // Given:-
2 // State 1
3 m = 0.9
                                           // mass of air
      in kg
4 T1 = 300.0
                                           // initial
     temperature in kelvin
5 P1 = 1.0
                                           // initial
     pressure in bar
7 // State 2
8 T2 = 470.0
                                           // final
     temperature in kelvin
9 P2 = 6.0
                                              final
    pressure in bar
10 \ Q = -20.0
                                           // heat
     transfer in kj
11
12 // From table A-22
13 u1 = 214.07
                                           // in KJ/kg
                                           // in KJ/Kg
14 \ u2 = 337.32
15
16 // Calculations
17 deltaU = m*(u2-u1)
                                           // change in
     internal energy in kj
18 W = Q - deltaU
                                           // in KJ/kg
19
20 // Results
21
22 printf(' The work during the process is %f KJ.',W);
```

#### Scilab code Exa 3.9 Example

```
1 // Given:-
2 // State 1
3 \text{ m1} = 2.0
                                        // initial mass of
      gas in tank 1 in kg
4 T1 = 350.0
                                        // initial
      temperature in kelvin in tank1
5 p1 = 0.7
                                        // initial
      pressure in bar in tank 1
6
7 // State 2
8 m2 = 8.0
                                        // initial mass of
      gas in tank 2 in kg
9 T2 = 300.0
                                        // initial
     temperature in kelvin in tank 2
                                        // initial
10 p2 = 1.2
      pressure in bar in tank 2
11 \text{ Tf} = 315.0
                                        // final
      equilibrium temperature in kelvin
12
13 // From table A-20
14 \text{ Cv} = 0.745
                                        // in KJ/Kg.k
15
16 // Calculations
17 pf = ((m1+m2)*Tf)/((m1*T1/p1)+(m2*T2/p2))
18 Ui = (m1*Cv*T1)+(m2*Cv*T2)
19 Uf = (m1+m2)*Cv*Tf
20 deltaU = Uf-Ui
21 Q = deltaU
22
23 // Results
24 printf ('The final equilibrium pressure is %f bar.'
      ,pf);
```

#### Scilab code Exa 3.11 Example

```
1 // Given:-
2 p1 = 1.0
                                             // initial
     pressure in bar
3 T1 = 295.0
                                             // initial
     temperature in kelvin
4 p2 = 5.0
                                             // final
      pressure in bar
5 n = 1.3
                                             // polytropic
      constant
6 R = 8314/28.97
                                             // gas
     constant for air in SI units
8 // From table A-22
9 u2 = 306.53
10 \text{ u1} = 210.49
11
12 // Calculations
13 T2 = T1*(p2/p1)**((n-1)/n)
14 w = R*(T2-T1)/(1-n)
15 Q = u2-u1+w/1000
16
17 // Results
18 printf ('The work done per unit mass is %f KJ/kg.',
     w/1000)
19 {\tt printf} ( ' The heat transfer per unit mass is %f KJ/
     kg.',Q);
```

## Chapter 4

# Control Volume Analysis Using Energy

#### Scilab code Exa 4.1 Example

```
1 // Given:-
2 // At inlet 1:-
                                                          //
3 p1 = 7.0
      pressure in bar
4 T2 = 200.0
      temperature in degree celcius
5 \text{ m1dot} = 40.0
                                                          //
      mass flow rate in kg/s
7 // At inlet 2:-
                                                          //
8 p2 = 7.0
      pressure in bar
9 T2 = 40.0
                                                          //
      temperature in degree celcius
10 \quad A2 = 25.0
                                                          //
      area in cm<sup>2</sup>
11
12 // At exit:-
13 p3 = 7.0
                                                          //
```

```
pressure in bar
14 \text{ AV3} = 0.06
       Volumetric flow rate through wxir in m<sup>3</sup>/s
15
16 // From table A-3
17 \quad v3 = (1.108)*(10**(-3))
       specific volume at the exit in m<sup>3</sup>/kg
18
19 // from table A-2
20 \text{ v2} = (1.0078)*(10**(-3))
                                                                 //
       specific volume in state 2 in m<sup>3</sup>/kg
21
22 // Calculation:-
23 \text{ m3dot} = \text{AV3/v3}
                                                                 //
       mass flow rate at exit
24 \text{ m2dot} = \text{m3dot} - \text{m1dot}
                                                                 //
       mass flow rate at inlet 2
25 V2= (m2dot*v2)/(A2*(10**(-4)))
26
27 // Results:-
28 printf (' The mass flow rate at the inlet 2 is \%.2\,\mathrm{f}
       \,\mathrm{kg}/\,\mathrm{s}\,. ',m2dot)
29 printf ('The mass flow rate at the exit is %.2 f kg/
       s.', m3dot)
30 printf( 'The velocity at the inlet is \%.2 \, \mathrm{f} \, \mathrm{m/s.}',
       V2)
```

#### Scilab code Exa 4.3 Example

```
velocity m/s
5
6 // At exit:-
                                                       //
7 p2 = 10.0
      pressure in bar
8 V2 = 665.0
      velocity in m/s
9 \text{ mdot} = 2.0
                                                        // mass
       flow rate in kg/s
10
11 // From table A-4
12 h1= 3213.6
      snpecific enthalpy in kJ/kg
13 \quad v2 = 0.1627
      specific volume at the exit in m<sup>3</sup>/kg
14
15 // Calculation:-
16 \text{ h2} = \text{h1} + ((V1**2-V2**2)/2)/1000
      snpecific enthalpy in kJ/kg
17 A2 = (mdot * v2) / V2
                                                        // Exit
       area
18
19 // Results:-
20 printf ('The exit Area of the nozzle is %.4f m^2',
      A2)
```

#### Scilab code Exa 4.4 Example

```
//
5 T1 = 400.0
      temperature in degree celc
                                                         //
6 V1 = 10.0
       velocity in m/s
8 // At exit:-
9 p2 = 0.10
                                                         //
       pressure in bar
                                                         //
10 q2 = 0.90
       quality
11 \quad V2 = 50.0
                                                         //
       velocity in m/s
12
13 // From table A-2 and A-3:-
14 h1= 3177.2
       specific enthalpy at inlet in kJ/kg
15 hf2= 191.83
16 hg2= 2584.63
17
18 // Calculation:-
                                                         //
19 h2 = hf2+q2*(hg2-hf2)
       specific enthalpy at exit in kJ/kg
20 Qcvdot = Wcvdot + m1dot*((h2-h1)+(V2**2-V1**2)
      /(2*1000))/3600
21
22 // Results:-
23 printf ('The rate of heat transfer between the
      turbine and surroundings is %.2 f kW', Qcvdot)
```

#### Scilab code Exa 4.5 Example

```
temperature in kelvin
                                                           //
4 \quad A1 = 0.1
      area in m<sup>2</sup>
5 V1 = 6.00
                                                           //
      velocity in m/s
7 // At exit:-
                                                           //
9 p2=7.00
      pressure in bar
10 t2 = 450.00
      temperature in kelvin
11 V2= 2.00
      velocity in m/s
12 \; Qcvdot = -180.0
                                                           //
      heat transfer rate in kJ/min
13 R= 8.314
                                                           //
      universal gas constant in SI units
14
15 // from table A-22
16
                                                           //
17 h1= 290.16
      specific enthalpy in kJ/kg
18 h2= 451.8
                                                           //
      specific enthalpy in kJ/kg
19
20 // Calculations:-
21
22 \text{ v1} = (R*1000*t1)/(28.97*p1*10**5)
                                                           //
      specific volume
23 \text{ mdot} = (A1 * V1) / V1
                                                           //
      mass flow rate
24 \text{ Wcvdot} = \text{Qcvdot}/60 + \text{mdot}*((h1-h2)+(V1**2-V2**2)
      /(2*1000))
25
26 // Results:-
27
28 printf ('The power input to the compressor is %.2 f
```

#### Scilab code Exa 4.6 Example

```
1 // Given:-
2 // At Entry:=
3 t1=20.0
                                                             //
      Temperatue in deg celcius
4 p1=1.0
                                                              //
      pressure in atm
5 \text{ AV1} = 0.1
      volumetric flow rate in litre/s
6 D1 = 2.5
                                                             //
      Diameter of th hose in cm
8 // At Exit:=
9 t2=23.0
                                                             //
      temperatuer in deg celcius
10 p2=1.0
      pressure in atm
11 \quad V2 = 50.0
                                                              //
      Velocity in m/s
12 \quad Z2=5.0
                                                              //
      elevation in m
13 \text{ g} = 9.8
                                                             //
      acceleration due to gravity in m/s<sup>2</sup>
14
15 // from table A-2 and A-19:-
16
                                                             //
17 v = (1.0018)*((10.0)**(-3))
      specific volume in m<sup>3</sup>/kg
18 c = 4.18
19
20 // Calculation:-
                                                             //
21 \text{ mdot} = (AV1/1000)/v
```

```
mass flow rate in kg/s
22 V1= (AV1/1000)/(3.14*(D1/(2*100))**2) //
        Entry velocity in m/s
23 deltah = c*(t2-t1)+v*(p2-p1)
24 Wcvdot= ((mdot*10)/9)*(-deltah+(V1**2-V2**2)
        /(2*1000)+g*(0-Z2)/1000)
25
26 // Results:-
27 printf( 'The power input to the motor is %.2 f kw',
        Wcvdot)
```

#### Scilab code Exa 4.7 Example

```
1 // Given:-
2 // Entering:-
                                                    //
3 p1=0.1
     pressure in bar
4 x1 = 0.95
                                                    //
     Quality
5 p2 = 0.1
     pressure in bar
6 t2 = 45.0
      temperature in deg celcius
7 t3 = 20.0
      temperature of cooling entry in deg cel
8 t4=35.0
                                                      //
     temperature of cooling exit
10 // From table A-3
11 hf = 191.53
     Enthalpy in KJ/kg
12 hg= 2584.7
                                                    //
     Enthalpy in KJ/kg
13 h2=188.45
     Assumption at states 2,3 and 4, h is approx equal
```

```
to hf(T), in kJ/kg
14 \text{ deltah4}_3 = 62.7
                                                       //
      Assumption 4, in kJ/kg
15
16
17 // Calculations:-
18 \text{ h1} = \text{hf} + \text{x1}*(\text{hg-hf})
19 ratio= (h1-h2)/(deltah4_3)
                                                       //
20 QRate= (h2-h1)
      Part B
21
22 // Results:-
23 printf ('The rate of the mass flow rate of the
      cooling water to the mass flow rate of the
      condenstaing stream is (m3dot/m1dot) %.2f ',ratio
24 printf ('The rate of energy transfer from the
      condensing steam to the cooling water of the
                     through the condenser is \%.2 f kJ/
      steam passing
      kg.', QRate)
```

#### Scilab code Exa 4.8 Example

```
6 \text{ pec} = -80.0
      power received by electronic components in watt
7 \text{ Pf} = -18.0
                                                     //
      Power received by fan in watt
8 R = 8.314
                                                      //
      Universal gas constant
9 M = 28.97*(10**(-3))
                                          // Molar mass
      of air in kg
10 Qcvdot=0
      Heat transfer from the outer surface of the
      electronics enclosure to the surroundings is
      negligible.
11 Cp= 1.005*(10**3)
                                            // in j/kg*k
12
13
14 // Calculations:-
16 Wcvdot = pec +Pf
      electric power provided to electronic components
      and fan in watt
17 mdotmin= (-Wcvdot)/(Cp*(T2max-T1))
                         // minimum mass flow rate
18 v1= ((R/M)*T1)/P1
                                            // specific
      volume
19 A1min = (mdotmin*v1)/V1max
20 D1min = (4*A1min/(\%pi))**(0.5)
21
22 // Results:-
23 printf ('The smallest fan inlet diameter is %.2 f cm
      ',D1min*100)
```

#### Scilab code Exa 4.9 Example

```
1 // Given:-
2 P1 = 20.0
                                             // pressure in
       supply line in bars
                                             // exhaust
3 P2 = 1.0
      pressure in bar
4 T2 = 120.0
                                             // exhaust
      temperature in degree celcius
6 // from table A-3 at 20 bars
7 \text{ hf1} = 908.79
                                           // Enthalpy in
      kj/kg
8 \text{ hg1} = 2799.5
                                           // Enthalpy in
      kj/kg
10 // from table A-4, at 1 bar and 120 degree celcius
11 h2 = 2766.6
                                           // in kj/kg
                                           // from
12 \ h1 = h2
      throttling process assumption
13
14
15 // Calculations:-
16 	 x1 = (h1-hf1)/(hg1-hf1)
17
18 // Results:-
19 printf( 'The quality of the steam in the supply
      line is \%.2 \,\mathrm{f}, x1)
20
21
22 // Note: rounding off error. please check manually.
```

#### Scilab code Exa 4.10 Example

```
1 // Given:-
2 P1 = 1.0
                                    // pressure of
      industrial discharge in bar
3 T1 = 478.0
                                     // temperature of
      industrial discharge in kelvin
                                     // mass flow rate of
4 \text{ m1dot} = 69.78
      industrial discharge in kg/s
5 T2 = 400.0
                                     // temperature of exit
       products from steam generator in kelvin
6 P2 = 1.0
                                     // pressure of exit
      products from steam generator in bar
7 P3 = 0.275
                                     // pressure of water
      stream entering the generator in Mpa
                                     // temperature of
8 T3 = 38.9
      water stream entering the generator in degree
      celcius
9 \text{ m3dot} = 2.079
                                     // mass flow rate of
      water stream entering in kg/s
10 P5 = 0.07
                                     // exit pressure of
      the turbine in bars
11 \times 5 = 0.93
                                     // quality of turbine
      exit
12
13 // Part (a)
14 \text{ m2dot} = \text{m1dot}
                                     // since gas and water
       streams do not mix
15 \text{ m5dot} = \text{m3dot}
                                     // ---DO
16
17 // from table A-22, A-2 and A-3:
                                     // in kj/kg
18 \text{ h1} = 480.3
19 h2 = 400.98
                                     // in Kj/kg
20 \text{ h3} = 162.9
                                     // assumption: h3 = hf
      (T3), units in Kj/kg
21 \text{ hf5} = 161.0
                                     // in kj/kg
22 \text{ hg5} = 2571.72
                                     // in kj/kg
23
```

```
24 // Part (b)
25 P4 = P3
                                       // from the assumption
        that there is no pressure drop for water flowing
       through the steam generator
26 \text{ T4} = 180
                                       // in degree celcius
27
28 // Calculations:-
29 h5 = hf5 + x5*(hg5-hf5)
30 \text{ Wcvdot} = \text{m1dot*h1} + \text{m3dot*h3} - \text{m2dot*h2} - \text{m5dot*h5}
31 \text{ h4} = \text{h3} + (\text{m1dot/m3dot})*(\text{h1} -\text{h2}) // from steady
      state energy rate balance
                                               // interpolating
32
                                                   in table A
                                                  -4, with
                                                  these P4 and
33 // Results:-
34 printf( ' The power developed by the turbine is \%.2\,\mathrm{f}
       kJ/s.', Wcvdot)
35 printf( 'Turbine inlet temperature is %.2f degree
       celcius.', T4)
```

#### Scilab code Exa 4.11 Example

```
9
10 \text{ vf1} = 1.2755e-3
                                                      // in m^3/kg
11 \text{ vg1} = 0.04221
                                                      // in m^3/kg
12
13
14
15 // for final state, from table A-2,
                                                       // units in
16 u2 = 2599.0
      KJ/kg
17 \quad v2 = 42.21e-3
                                                       // units in
      m^3/Kg
18 he = 2796.6
                                                       // units in
      KJ/kg
19
20 // Calculations:-
21 	 u1 = uf1 + X1*(ug1-uf1)
                                                       // in kj/kg
22 \text{ v1} = \text{vf1} + \text{X1}*(\text{vg1}-\text{vf1})
                                                       // in m^3/kg
23 \text{ m1} = \text{V/v1}
                                                       // initial
       mass in kg
24 \text{ m2} = \text{V/v2}
                                                       // final
       mass in kg
25 \ U2 = m2*u2
                                                       // final
      internal energy in KJ
26 \text{ U1} = \text{m1}*\text{u1}
                                                       // initial
       internal energy in KJ
27 \text{ Qcv} = (U2-U1) - \text{he}*(m2-m1)
28
29 // Results:-
30 printf ('The amount of heat transfer is %.2 f KJ.',
       Qcv)
```

## Scilab code Exa 4.12 Example

```
1 // Given:-
2 Pv = 15.0
```

```
//
      pressure in the vessel in bar
3 \text{ Tv} = 320.0
      temperature in the vessel in degree celcius
4 \text{ Vt} = 0.6
                                                         //
      volume of a tank in m<sup>3</sup>
5 \text{ Tt} = 400.0
      temperature in the tank in degree celcius when
      the tank is full
6
7 // Since the tank is initially empty:-
8 m1 = 0
9 u1 = 0
10
11 // From table A-4, at 15bar and 400 degree celcius:-
12 v2 = 0.203
                                                       //
      Volume in m<sup>3</sup>/kg
13 \text{ m2} = Vt/v2
                                                       // mass
       within the tank at the end of the process in kg
14 \text{ hi} = 3081.9
                                                      // in kj
      /kg
15 \quad u2 = 2951.3
                                                      // in kj
      /kg
16
17 // Calculations:-
18 \text{ deltaUcv} = m2*u2-m1*u1
19 Wcv = hi*(m2-m1)-deltaUcv
20
21 // Results:-
22 printf (' The amount of work developed by the
      turbine is %.2 f kJ.', Wcv)
```

# Chapter 5

# The Second Law of Thermodynamics

## Scilab code Exa 5.1 Example

```
1 // Given :-
2 W = 410.00
                                                // net work
     output in kj claimed
3 Q = 1000.00
                                                 // energy
      input by heat transfer in kj
4 \text{ Tc} = 300.00
      temperature of cold reservoir in kelvin
  TH = 500.00
      temperature of hot reservoir in kelvin
7 // Calculations
                                                 // thermal
8 \text{ eta} = W/Q
      efficiency
9 \text{ etamax} = 1 - (Tc/TH)
10
11 // Results
12 printf ( ' Eta = \%.4 \,\mathrm{f}', eta)
13 printf( 'Etamax = \%.4 \,\mathrm{f}', etamax)
14 printf(' Since eta is more than etamax, the claim
```

## Scilab code Exa 5.2 Example

```
1 // Given :-
2 \text{ Qcdot} = 8000.00
                                                        // in
       kj/h
3 \text{ Wcycledot} = 3200.00
                                                        // in
       kj/h
4 \text{ Tc} = 268.00
      temperature of compartment in kelvin
  TH = 295.00
      temperature of the surrounding air in kelvin
7 // Calculations
8 beta = Qcdot/Wcycledot
                                                        //
      coefficient of performance
9 betamax = Tc/(TH-Tc)
      reversible coefficient of performance
10
11 // Results
12 printf (' Coefficient of performance is \%.3 \, \mathrm{f}', beta)
13 printf ( 'Coefficient of performance of a reversible
       cycle is \%.3 f, betamax)
```

## Scilab code Exa 5.3 Example

# Chapter 6

# Using Entropy

# Scilab code Exa 6.1 Example

```
1 // Given:-
2 T = 373.15
                                                     //
      temperature in kelvin
4 // From table A-2
6 p = 1.014*(10**5)
                                                     //
     pressure in pascal
7 \text{ vg} = 1.673
8 \text{ vf} = 1.0435e-3
9 \text{ sg} = 7.3549
10 \text{ sf} = 1.3069
11
12 // Calculations
13 w = p*(vg-vf)*(10**(-3))
14 Q = T*(sg-sf)
15
16 // Results
17 printf( 'The work per unit mass is \%.3 \, f \, KJ/Kg', w)
18 printf( 'The heat transfer per unit mass is \%.2\,\mathrm{f} kj
      /kg',Q)
```

## Scilab code Exa 6.2 Example

```
1 // Given:-
2 // Assumptions:
4 // From table A-2 at 100 degree celcius
5 \text{ ug} = 2506.5
                                                       // in
      kj/kg
6 \text{ uf} = 418.94
                                                       // in
      kj/kg
  sg = 7.3549
8 \text{ sf} = 1.3069
9
10
11 // Calculations:-
12 // From energy balance
13 W = -(ug-uf)
14 // From entropy balance
15 \text{ sigmabym} = (sg-sf)
16
17 // Results
18 printf ('The net work per unit mass is %.2 f KJ/kg.
      ',W)
19 printf (' The amount of entropy produced per unit
      mass is \%.2 f \text{ KJ/kg.}, sigmabym)
```

# Scilab code Exa 6.3 Example

```
3 P2 = 0.7*(10**6)
                                                // final
      pressure in pascal
5 // \text{ From table A-10},
6 u1 = 227.06
                                                // in kj/
     kg
8 // minimum theoretical work corresponds to state of
     isentropic compression
9 // From table A-12,
10 \text{ u2s} = 244.32
                                                // in kj/
     kg
11
12 // Calculations
13 Wmin = u2s-u1
14
15 // Results
16 printf (' The minimum theoretical work input
      required per unit mass of refrigerant is: %.2f kJ
      /kg', Wmin)
```

#### Scilab code Exa 6.4 Example

```
13 sigmadt = -Qdot/Tf
14
15 // Results
16 printf( 'The rate of entropy production with
        gearbox as system is %f kw/k', sigmadot)
17 printf( 'The rate of entropy production with
        gearbox + sorrounding as system is %f kw/k',
        sigmadt)
```

#### Scilab code Exa 6.5 Example

```
1 // GIven:-
2 \text{ Tmi} = 1200.0
                                                        //
      initial temperature of metal in kelvin
3 \text{ cm} = 0.42
      specific heat of metal in KJ/kg.k
4 \text{ mm} = 0.3
      mass of metal in kg
  Twi = 300.0
      initial temperature of water in kelvin
6 \text{ cw} = 4.2
                                                        //
      specific heat of water in KJ/Kg.k
7 \text{ mw} = 9.0
                                                        //
      mass of water in kg
8
9 // Calculations
10 // Part(a)
11 // Solving energy balance equation yields
12 Tf = (mw*(cw/cm)*Twi+mm*Tmi)/(mw*(cw/cm)+mm)
13
14 // Part (b)
15 // Solving entropy balance equation yields
16 sigma = mw*cw*log(Tf/Twi)+mm*cm*log(Tf/Tmi)
17
18 // Results
```

```
19 printf( 'The final equilibrium temperature of the
      metal bar and the water is %.2f kelvin.', Tf)
20 printf( 'The amount of entropy produced is: %.2f kJ
      /k.', sigma)
```

#### Scilab code Exa 6.6 Example

```
1 // Given:-
2 P1 = 30.0
     pressure of steam entering the turbine in bar
3 T1 = 400.0
                                                  //
     temperature of steam entering the turbine in
     degree celcius
4 V1 = 160.0
     velocity of steam entering the turbine in m/s
5 T2 = 100.0
     temperature of steam exiting in degree celcius
6 V2 = 100.0
                                                  //
     velocity of steam exiting in m/s
7 \text{ Wcvdot} = 540.0
                                              // work
     produced by turbine in kJ/kg of steam
8 \text{ Tb} = 350.0
                                                  //
     temperature of the boundary in kelvin
10 // From table A-4 and table A-2
11 h1 = 3230.9
                                                 //
      specific enthalpy at entry in Kj/kg
```

```
12 h2 = 2676.1
                                                  //
      specific enthalpy at exit in kj/kg
13
14 // Calculations
15
16 // Reduction in mass and energy balance equations
      results in
17 Qcvdot = Wcvdot + (h2 - h1)+ (V2**2-V1**2)
      /(2*(10**3)) // heat transfer rate
18
19 // From table A-2
20 	 s2 = 7.3549
                                                     // in
       kj/kg.k
21 // From table A-4
22 \text{ s1} = 6.9212
                                                     // in
       kj/kg.k
23
24 // From entropy and mass balance equations
25 \text{ sigmadot} = -(Qcvdot/Tb) + (s2-s1)
26
27 // Results
28 printf ('The rate at which entropy is produced
      within the turbine per kg of steam flowing is %
      .2 f kJ/kg.k', sigmadot)
```

# Scilab code Exa 6.7 Example

```
4 T2 = 352.0
                                            // exit
      temperature of hot stream in kelvin
                                            // exit
5 P2 = 1.0
      pressure of hot stream in bars
6 T3 = 255.0
                                            // exit
      temperature of cold stream in kelvin
7 P3 = 1.0
                                            // exit
      pressure of cold stream in bars
8 \text{ cp} = 1.0
                                            // in kj/kg.k
10 // Calculations
11 R = 8.314/28.97
12 se = 0.4*(cp*log((T2)/(T1))-R*log(P2/P1)) + 0.6*(cp*log(T2)/(T1))
      log((T3)/(T1))-R*log(P3/P1))
13
                                            // specific
                                               entropy in
                                               kj/kg.k
14
15
16 // Results
17 printf ('Specific entropy in kj/kg.k = \%.3 f KJ/kg.
18 printf ('Since se > 0, the claim of the writer is
      true');
```

#### Scilab code Exa 6.8 Example

		pressure of refrigerant entering the condenser
_	πО	in bars
9	12	= 348.0 //
		temperature of refrigerant entering the condenser in kelvin
6	DЗ	= 14.0 //
U	13	pressure of refrigerant exiting the condenser in
		bars
7	ТЗ	= 301.0 //
		temperature of refrigerant exiting the condenser
0	D 4	in kelvin
8	Ρ4	= 3.5 //
		pressure of refrigerant after passing through
0	DE	expansion valve in bars = 1.0 //
9	P5	
		pressure of indoor return air entering the condenser in bars
10	ΤБ	= 293.0 //
10	10	temperature of indoor return air entering the
		condenser in kelvin
11	A V 5	5 = 0.42 //
		volumetric flow rate of indoor return air
		entering the condenser in m <sup>3</sup> /s
12	Р6	= 1.0
		pressure of return air exiting the condenser in
		bar
13	Т6	= 323.0 //
		temperature of return air exiting the condenser
		in kelvin
14		
15	//	Part (a)
16	, ,	
17		From table A-9
18	s1	= 0.9572 //
1.0	, ,	in kj/kg.k
19	//	
20	S Z	= $0.98225$ // in kj/kg.k
		III KJ/Kg.K

```
21 \quad h2 = 294.17
                                                                    //
        in kj/kg
22 // From table A-7
23 	 s3 = 0.2936
        in kj/kg.k
24 \text{ h3} = 79.05
                                                                    //
        in kj/kg
25
26 \text{ h4} = \text{h3}
                                                                    //
        since expansion through valve is throttling
       process
27
\frac{28}{\sqrt{From table A-8}}
29 \text{ hf4} = 33.09
                                                                    //
        in kj/kg
30 \text{ hg4} = 246.00
                                                                     //
        in kj/kg
31 \text{ sf4} = 0.1328
                                                                    //
        in kj/kg.k
32 \text{ sg4} = 0.9431
                                                                    //
        in kj/kg.k
33 \text{ cp} = 1.005
                                                                    //
        in kj/kg.k
34
35 // Calculations
36
37 \times 4 = (h4-hf4)/(hg4-hf4)
                                                                    //
        quality at state 4
   s4 = sf4 + x4*(sg4-sf4)
                                                                     //
38
        specific entropy at state 4
39
40 // CONDENSER!!
                                                                    //
41 \text{ v5} = ((8314/28.97)*T5)/(P5*(10**5))
        specific volume at state 5
42 \text{ mairdot} = AV5/v5
43 \text{ h6} = \text{cp}*T6
44 h5 = cp*T5
45 \text{ mrefdot} = \text{mairdot}*(h6-h5)/(h2-h3)
```

```
46 deltaS65 = cp*log(T6/T5) - (8.314/28.97)*log(P6/P5) //
       change in specific entropy
47 sigmacond = (mrefdot*(s3-s2)) + (mairdot*(deltaS65))
48
49 // COMPRESSOR!!
50 \text{ sigmacomp} = \text{mrefdot}*(s2-s1)
51
52 // VALVE!!
53 sigmavalve = mrefdot *(s4-s3)
54
55 // Results
56 printf(' The rates of entropy production for
      control volume enclosing the condenser is %f kw/
      k', sigmacond);
57 printf( 'The rates of entropy production for
      control volume enclosing the compressor is %f kW
      /K.', sigmacomp);
58 printf (' The rates of entropy production for
      control volume enclosing the expansion valve
      \%f kW/K ',sigmavalve)
```

#### Scilab code Exa 6.9 Example

# Scilab code Exa 6.10 Example

```
1 // Given:-
2 m1 = 5.00
     // initial mass in kg
3 P1 = 5.00
    // initial pressure in bar
4 \text{ T1} = 500.00
     // initial temperature in kelvin
5 P2 = 1.00
      // final pressure in bar
7 // From table A-22
8 \text{ pr1} = 8.411
9
10
11
12 // Using this value of pr2 and interpolation in
      table A-22
13 T2 = 317.00
      // in kelvin
14
15 // Calculations
```

```
16 \text{ pr2} = (P2/P1)*pr1
17 \text{ m2} = (P2/P1)*(T1/T2)*m1
18
19 // Results
20 printf ('The amount of mass remaining in the tank is
       %f kg', m2)
21 printf('and its temperature is %f kelvin.',T2);
   Scilab code Exa 6.11 Example
1 // Given:-
2 P1 = 1.00
                                                          //
      inlet pressure in bar
3 T1 = 593.00
      inlet temperature in kelvin
4 P2 = 1.00
      exit pressure in bar
5 \text{ eta} = 0.75
      turbine efficiency
7 // From table A-4
8 \text{ h1} = 3105.6
                                                        // in
      Kj/kg
9 \text{ s1} = 7.5308
                                                        // in
      kj/kg.k
10 // From table A-4 at 1 bar
11 \text{ h2s} = 2743.00
                                                      // in kj
```

/kg

```
12
13 // Calculations
14 w = eta*(h1 - h2s)
15
16 // Result
17 printf( 'The work developed per unit mass of steam flowing through is %f kJ/kg.',w);
```

# Scilab code Exa 6.12 Example

```
1 // Given:-
2 P1 = 3.00
      pressure of air entering in bar
3 T1 = 390.00
                                                     //
      temperature of air entering in kelvin
4 P2 = 1.00
      pressure of exit air
5 \text{ Wcvdot} = 74.00
                                                     // work
      developed in kj/kg
7 // From table A-22, at 390k
8 \text{ h1} = 390.88
                                                     // in kj/
      kg
9 \text{ pr1} = 3.481
11 // From interpolation table A-22
                                                     // in kj/
12 \text{ h2s} = 285.27
      kg
13
14 // calculations
15 \text{ pr2} = (P2/P1)*pr1
16 Wcvdots = h1 - h2s
17 eta = Wcvdot/Wcvdots
18
19 // Result
```

#### Scilab code Exa 6.13 Example

```
1 // Given:-
2 P1 = 1.00
                                                       //
      pressure of entering steam in Mpa
                                                       //
3 T1 = 593.00
      temperature of entering steam in kelvin
4 V1 = 30.00
                                                       //
      velocity of entering steam in m/s
5 P2 = 0.3
                                                       //
      pressure of exit steam in Mpa
6 	ext{ T2} = 453.00
                                                       //
      temperature of exit steam in kelvin
8 // From table A-4, at T1 = 593 kelvin and P1 = 1 Mpa
9 // and at T2 = 453 kelvin and P2 = .3 Mpa
10 \text{ h1} = 3093.9
                                                       //
     in kj/kg
11 	 s1 = 7.1962
                                                       //
      in kj/kg.k
12 h2 = 2823.9
                                                       //
      in kj/kg
13
14
15 // Interpolating in table A-4
16 \text{ h2s} = 2813.3
                                                       //
      in kj/kg
17
18 // Calculations
19 V2squareby2 = h1 - h2 + (V1**2)/2000
20 V2squareby2s = h1 - h2s + (V1**2)/2000
21 eta = V2squareby2/V2squareby2s
```

```
22
23 // Results
24 printf(' The nozzle efficiency is %.4f',eta)
```

## Scilab code Exa 6.14 Example

```
1 // Given:-
2 // From table A-9
3 \text{ h1} = 249.75
                                                            // in
       kj/kg
4 h2 = 294.17
                                                            // in
        kj/kg
5 \text{ mdot} = 0.07
                                                            // in
       kg/s
7 // From table A-9
8 \text{ s1} = 0.9572
                                                            // in
       Kj/Kg.k
9 \text{ h2s} = 285.58
                                                           // in
      kj/kg
10
11 // Calculations
12 \text{ wcvdot} = -(\text{mdot}*(\text{h2}-\text{h1}))
13 eta = (h2s-h1)/(h2-h1)
14
15 // Results
16 printf( 'The power in is %f kw', wcvdot);
17 printf(' The isentropic efficiency is %.3f', eta)
```

#### Scilab code Exa 6.15 Example

```
1 // Given:-
2 P1 = 1.00
                                               // pressure
     of entering air in bar
3 T1 = 293.00
      temperature of entering air in kelvin
4 P2 = 5.00
                                               // pressure
      of exit air in bar
5 n = 1.3
6 R = 8.314/28.97
8 // From table A-22
9 \text{ h1} = 293.17
                                               // in kj/kg
                                               // in kj/kg
10 \text{ h2} = 426.35
11
12 // Calculations
13 T2 = T1*((P2/P1)**((n-1)/n))
                                               // in
      kelvin
14 wcvdot = ((n*R)/(n-1))*(T1-T2)
                                               // in kj/kg
                                               // in kj/kg
15 Qcvdot= wcvdot + (h2-h1)
16
17 // Results
18 printf(' The work per unit mass passing through the
       device is %.2 f kJ/kg', wcvdot)
19 \texttt{printf}( ' The heat transfer per unit mass is \%.2\,\text{f} kJ
     /kg. ',Qcvdot)
```

# Chapter 7

# **Exergy Analysis**

# Scilab code Exa 7.1 Example

```
1 // Given:-
2 v = 2450.00
      volume of gaseous products in cm<sup>3</sup>
                                                           //
3 P = 7.00
      pressure of gaseous product in bar
4 T = 867.00
      temperature of gaseous product in degree celcius
5 \text{ TO} = 300.00
                                                           // in
       kelvin
                                                           // in
6 \text{ PO} = 1.013
       bar
8 // From table A-22
9 u = 880.35
                                                           // in
       kj/kg
10 \ u0 = 214.07
                                                           // in
       kj/kg
11 \text{ sOT} = 3.11883
                                                           // in
       kj/kg.k
12 \text{ s0T0} = 1.70203
                                                           // in
       kj/kg.k
```

## Scilab code Exa 7.2 Example

```
1 // Given:-
2 \text{ mR} = 1.11
                                                   // mass of
      the refrigerant in kg
3 \text{ T1} = -28.00
                                                   // initial
      temperature of the saturated vapor in degree
      celcius
4 P2 = 1.4
                                                   // final
      pressure of the refrigerant in bar
5 \text{ TO} = 293.00
                                                   // in kelvin
6 P0 = 1.00
                                                   // in bar
8 // Part (a)
9 // From table A-10
10 \ u1 = 211.29
                                                   // in kj/kg
                                                   // in m<sup>3</sup>/kg
11 v1 = 0.2052
12 \text{ s1} = 0.9411
                                                   // in kj/kg.
      k
13 // From table A-12
14 \ u0 = 246.67
                                                   // in kj/kg
15 \text{ v0} = 0.23349
                                                   // in m^3/kg
16 \text{ s0} = 1.0829
                                                   // in kj/kg.
      k
```

```
17
18 // From table A-12
19 u2 = 300.16
                                             // in kj/kg
                                             // in kj/kg.
20 	 s2 = 1.2369
     k
21 v2 = v1
22
23 // Calculations
24 	 E1 = mR*((u1-u0) + P0*(10**5)*(v1-v0)*(10**(-3))-T0
      *(s1-s0))
25 E2 = mR*((u2-u0) + P0*(10**5)*(v2-v0)*(10**(-3))-T0
      *(s2-s0))
26
27 // Results for Part A
28 printf (' Part (a) The initial exergy is %.2 f kJ.', E1
29 printf ('The final exergy is %.2 f kJ.', E2)
30 printf ('The change in exergy of the refrigerant is
      \%.2 \, f \, kj', E2-E1)
31
32
33 // Part (b)
34 // Calculations
35 \text{ deltaU} = mR*(u2-u1)
36 // From energy balance
37 deltaPE = -deltaU
38 // With the assumption::The only significant changes
       of state are experienced by the refrigerant and
      the suspended mass. For the refrigerant,
39 // there is no change in kinetic or potential energy
      . For the suspended mass, there is no change in
      kinetic or internal energy. Elevation is
40 // the only intensive property of the suspended mass
       that changes
41 deltaE = deltaPE
42
43 // Results for part b
44 printf(' Part(b) The change in exergy of the
```

```
suspended mass is %.3f kJ',deltaE)

45
46
47 // Part(c)
48 // Calculations
49 deltaEiso = (E2-E1) + deltaE
50
51 // Results
52 printf( 'Part(c)The change in exergy of an isolated system of the vessel and pulley mass assembly is %.2f kJ',deltaEiso)
```

# Scilab code Exa 7.3 Example

```
1 // Given :-
2 T = 373.15
       initial temperature of saturated liquid in kelvin
3 \text{ TO} = 293.15
                                                              // in
        kelvin
4 \text{ PO} = 1.014
                                                               //
      in bar
6 // Part(a)
7 // From table A-2
8 \text{ ug} = 2506.5
                                                              // in
        kj/kg
9 \text{ uf} = 418.94
                                                              // in
        kj/kg
10 \text{ vg} = 1.673
                                                               //
```

```
in m^3/kg
11 vf = 1.0435*(10**(-3))
                                            // in m^3/kg
12 \text{ sg} = 7.3549
                                                         // in
       kj/kg.k
13 \text{ sf} = 1.3069
                                                         // in
       kj/kg.k
14
15
16 // Calculations
17 // Energy transfer accompanying work
18 \text{ etaw} = 0
       since p = p0
19 // Exergy transfer accompanying heat
20 \ Q = 2257
                                                             //
       in kj/kg, obtained from example 6.1
21 \text{ etah} = (1-(T0/T))*Q
22
23 // Exergy destruction
24 \text{ ed} = 0
      // since the process is accomplished without any
      irreversibilities
25 deltae = ug-uf + P0*(10**5)*(vg-vf)/(10**3)-T0*(sg-
      sf)
26
27 // Results
28 printf (' Part (a) the change in exergy is \%.2 \, \text{f kJ/kg}.
      ', deltae)
29 printf (' The exergy transfer accompanying work is \%
      .2 f kJ/kg., etaw)
30 printf (' The exergy transfer accompanying heat is \%
      .2\,\mathrm{f} \mathrm{kJ/kg}, etah)
31 printf ('The exergy destruction is \%.2 f kJ/kg.', ed)
```

```
32
33
34 // Part(b)
35 Deltae = deltae
                                                     // since
       the end states are same
36 Etah = 0
      // since process is adiabatic
37 // Exergy transfer along work
38 W = -2087.56
                                                         // in
       kj/kg from example 6.2
39 Etaw = W- P0*(10**5)*(vg-vf)/(10**3)
40 // Exergy destruction
41 \text{ Ed} = -(\text{Deltae+Etaw})
42
43 // Results
44 printf (' Part (b) the change in exergy is \%.2 \, \text{f kJ/kg}.
      ', Deltae)
45 printf (' The exergy transfer accompanying work is \%
      .2 f kJ/kg., Etaw)
46 printf (' The exergy transfer accompanying heat is \%
      .2 f kJ/kg., Etah)
47 printf(' The exergy destruction is \%.2 \, \mathrm{f} \, \mathrm{kJ/kg}.',Ed)
```

#### Scilab code Exa 7.4 Example

```
gearbox in kelvin from example 6.4a
5 \text{ sigmadot} = 0.004
     // rate of entropy production in KW/k from
     example 6.4a
7 // Calculations
8 R = -(1-T0/Tb)*Qdot
                                         // time rate of
      exergy transfer accompanying heat
9 Eddot = T0*sigmadot
     // rate of exergy destruction
10
11 // Results
12 printf( ' Balance sheet');
13 printf ('\n Rate of exergy in high speed shaft 60Kw'
14 printf ('\n Disposition of the exergy: Rate of
     exergy out low-speed shaft \%.1\,\mathrm{f} Kw',58.8 )
15 printf( '\n Heat transfer is %.3 f kw.', R)
16 printf ('\n Rate of exergy destruction is %.3 f kw',
     Eddot)
```

#### Scilab code Exa 7.5 Example

```
// in
       degree celcius
6 p0 = 1.0
                                                           //
      in atm
8 // From table A-4
9 \text{ h1} = 3043.4
                                                       // in
      kj/kg
10 \text{ s1} = 6.6245
                                                       // in
      kj/kg.k
11 \ h2 = h1
      from reduction of the steady-state mass and
      energy rate balances
12 	 s2 = 7.4223
      Interpolating at a pressure of 0.5~\mathrm{MPa} with \mathrm{h2}=
      h1, units in kj/kg.k
13
14 // From table A-2
15 \text{ h0} = 104.89
                                                       // in
      kj/kg
16 \text{ s0} = 0.3674
                                                       // in
      kj/kg.k
17
18 // Calculations
19 ef1 = h1-h0-(T0+273)*(s1-s0)
                                   // flow exergy at the
      inlet
20 \text{ ef2} = h2-h0-(T0+273)*(s2-s0)
                                   // flow exergy at the
      exit
21 // From the steady-state form of the exergy rate
```

```
balance

22 Ed = ef1-ef2

// the
exergy destruction per unit of mass flowing is

23

24 // Results

25 printf(' The specific flow exergy at the inlet is %
.2 f kJ/kg.',ef1)

26 printf(' The specific flow exergy at the exit is %
.2 f kJ/kg.',ef2)

27 printf(' The exergy destruction per unit of mass flowing is %.2 f kJ/kg.',Ed)
```

#### Scilab code Exa 7.6 Example

```
1 // Given:-
2 T1 = 610.0
      temperature of the air entering heat exchanger
     in kelvin
3 p1 = 10.0
     // pressure of the air entering heat exchanger in
     bar
4 T2 = 860.0
                                                     //
      temperature of the air exiting the heat
     exchanger in kelvin
5 p2 = 9.70
     // pressure of the air exiting the heat exchanger
     in bar
6 T3 = 1020.0
     temperature of entering hot combustion gas in
```

```
kelvin
 7 p3 = 1.10
      // pressure of entering hot combustion gas in
      bar
8 p4 = 1.0
      // pressure of exiting hot combustion gas in bar
 9 \text{ mdot} = 90.0
                                                             //
      mass flow rate in kg/s
10 \text{ TO} = 300.0
                                                              //
      in kelvin
11 p0 = 1.0
      // in bar
12
13 // Part (a)
14 // From table A-22
15 \text{ h1} = 617.53
                                                             //
      in kj/kg
16 \text{ h2} = 888.27
      in kj/kg
17 h3 = 1068.89
      in kj/kg
18
19 // Calculations
20 \text{ h4} = \text{h3+h1-h2}
21
22 // Using interpolation in table A-22 gives
23 \text{ T4} = 778
      // in kelvin
24
```

```
25 // Results
26 printf ('The exit temperature of the combustion gas
         is %f kelvin., T4);
27
28 // Part(b)
29 // From table A-22
30 \text{ s2} = 2.79783
                                                          //
      in kj/kg.k
31 	 s1 = 2.42644
                                                         //
      in kj/kg.k
32 \text{ s4} = 2.68769
                                                          //
      in kj/kg.k
33 	 s3 = 2.99034
      in kj/kg.k
34
35 // Calculations for part b
36
37 \text{ deltaR} = (\text{mdot}*((\text{h2-h1})-\text{T0}*(\text{s2-s1-}(8.314/28.97)*\log(
      p2/p1))))/1000
38 deltRc = mdot*((h4-h3)-T0*(s4-s3-(8.314/28.97)*log(
      p4/p3)))/1000
39
40 // Results for part b
41 printf (' The net change in the flow exergy rate
      from inlet to exit of compressed gas is %.3 f MW
      .', deltaR)
42\  \, {\tt printf} \, ( ' The net change in the flow exergy rate
      from inlet to exit of hot combustion gas is %.3
      f MW. ', deltRc)
43
44 // Part(c)
45 //From an exergy rate balance
46 Eddot = -deltaR-deltRc
47
```

```
48 // Results
49 printf( 'The rate exergy destroyed, is %.3 f MW.'
,Eddot)
```

#### Scilab code Exa 7.7 Example

```
1 // Given:-
2 p1 = 30.0
                                                     //
     pressure of entering steam in bar
3 t1 = 400.0
     temperature of entering steam in degree celcius
4 v1 = 160.0
                                                    //
      velocity of entering steam in m/s
5 t2 = 100.0
     temperature of exiting saturated vapor in degree
      celcius
6 v2 = 100.0
      velocity of exiting saturated vapor in m/s
7 W = 540.0
     rate of work developed in kj per kg of steam
8 \text{ Tb} = 350.0
      temperature on the boundary where heat transfer
     occurs in kelvin
9 \text{ TO} = 25.0
                                                     // in
       degree celcius
10 p0 = 1.0
```

```
in atm
11
12 // From table A-4
13 \text{ h1} = 3230.9
                                                      // in
      kj/kg
14 \text{ s1} = 6.9212
                                                      // in
      kj/kg.k
15 // From table A-2
16 \text{ h2} = 2676.1
                                                      // in
      kj/kg
17 	 s2 = 7.3549
                                                      // in
      kj/kg.k
18 // From example 6.6
19 \ Q = -22.6
                                                        // in
       kj/kg
20
21 // Calculations
22 DELTAef = (h1-h2)-(T0+273)*(s1-s2)+(v1**2-v2**2)
      /(2*1000)
23 // The net exergy carried in per unit mass of steam
      flowing in kj/kg
24 \text{ Eq} = (1-(T0+273)/Tb)*(Q)
                                       // exergy transfer
      accompanying heat in kj/kg
25 \text{ Ed} = ((1-(T0+273)/Tb)*(Q))-W+(DELTAef)
                       // The exergy destruction
      determined by rearranging the steady-state form
      of the exergy
                                                                //
26
                                                                   rate
                                                                   balance
```

# Scilab code Exa 7.8 Example

```
1 // Given:-
2 clc;
3 \text{ m1dot} = 69.78
                                                    // in
     kg/s
4 p1 = 1.0
   // in bar
5 T1 = 478.0
     in kelvin
6 T2 = 400.0
   in kelvin
7 p2 = 1.0
   // in bar
8 p3 = 0.275
                                                       //
     in Mpa
9 T3 = 38.9
```

```
// in degree celcius
10 \text{ m3dot} = 2.08
 in kg/s
11 \quad T4 = 180.0
 in degree celcius
12 p4 = 0.275
 in Mpa
13 p5 = 0.07
// in bar
14 \times 5 = 0.93
15 \text{ Wcvdot} = 876.8
                                                     // in
  \mathrm{kW}
16 \text{ TO} = 298.0
                                                         //
 in kelvin
17
18
19 // Part(a)
20 // From table A-22
21 \text{ h1} = 480.35
                                                        //
in kj/kg
22 \text{ h2} = 400.97
                                                        //
in kj/kg
23 	 s1 = 2.173
 in kj/kg
24 	 s2 = 1.992
      in kj/kg
25
```

```
\frac{26}{\sqrt{\text{From table A}-2E}}
27 \text{ h3} = 162.82
                                                         //
      in kj/kg
28 	 s3 = 0.5598
      in kj/kg.k
29 // Using saturation data at 0.07 bars from Table A-3
30 \text{ h5} = 2403.27
      in kj/kg
31 \text{ s5} = 7.739
                                                           //
       in kj/kg.k
32 //The net rate exergy carried out by the water
      stream
33
34 // From table A-4
35 \text{ h4} = 2825.0
                                                          //
      in kj/kg
36 \text{ s4} = 7.2196
      in kj/kg.k
37 // Calculations
38 netRE = m1dot*(h1-h2-T0*(s1-s2-(8.314/28.97)*log(p1/
      p2))) // the net rate exergy carried into the
      control volume
39 \quad netREout = m3dot*(h5-h3-T0*(s5-s3))
40 // From an exergy rate balance applied to a control
      volume enclosing the steam generator
41 Eddot = netRE + m3dot*(h3-h4-T0*(s3-s4))
                         // the rate exergy is destroyed
      in the heat-recovery steam generator
42
43 // From an exergy rate balance applied to a control
      volume enclosing the turbine
44 EdDot = -Wcvdot + m3dot*(h4-h5-T0*(s4-s5))
```

```
// the rate exergy is destroyed in
      the tpurbine
45
46 // Results
47 printf( '\n balance sheet')
48 printf( '\n- Net rate of exergy in: \%f kJ/kg.',netRE
49 printf( '\n Disposition of the exergy:')
50 printf('\n Rate of exergy out')
51 printf( '\n power developed %f kJ/kg.',netRE-
     netREout -Eddot -EdDot)
52 printf( '\n water stream \%f', netREout)
53 printf('\n Rate of exergy destruction')
54 printf( '\n heat-recovery steam generator %f kJ/kg',
     Eddot)
55 printf( '\n turbine %f', EdDot)
56
57 // note : answer is slightly different because of
     rounding off error.
```

#### Scilab code Exa 7.9 Example

```
8 \text{ sigmadotcond} = 7.95e-4
                                             // in kw/k
10 // Calculations
11 // The rates of exergy destruction
12 EddotComp = T0*sigmadotComp
                                       // in kw
13 EddotValve = T0*sigmadotValve
                                      // in kw
14 Eddotcond = T0*sigmadotcond
                                        // in kw
15
16 \text{ mCP} = 3.11
     // From the solution to Example 6.14, the
      magnitude of the compressor power in kW
17
18 // Results
19 printf( 'Daily cost in dollars of exergy
      destruction due to compressor irreversibilities =
      \%.3 f', EddotComp*pricerate*24)
20 printf( 'Daily cost in dollars of exergy
      destruction due to irreversibilities in the
      throttling valve = \%.3 \, f', EddotValve*pricerate*24)
21 printf ( 'Daily cost in dollars of exergy
      destruction due to irreversibilities in the
      condenser = \%.3 f', Eddotcond*pricerate*24)
22 printf ('Daily cost in dollars of electricity to
      operate compressor = \%.3 \, f', mCP*pricerate*24)
```

#### Scilab code Exa 7.10 Example

```
1 // Given:-
2 EfFdot = 100.00
```

//

```
exergy rate of fuel entering the boiler in MW
3 \text{ cF} = 1.44
      // unit cost of fuel in cents per kw.h
4 \text{ Zbdot} = 1080.00
      the cost of owning and operating boiler in
      dollars per hour
5 \text{ Ef1dot} = 35.00
      exergy rate of exiting steam from the boiler in
     MW
6 p1 = 50.00
      // pressure of exiting steam from the boiler in
      bar
7 T1 = 466.00
      // temperature of exiting steam from the boiler
      in degree celcius
8 \text{ Ztdot} = 92.00
                                                        //
      the cost of owning and operating turbine in
      dollars per hour
9 p2 = 5.00
      // pressure of exiting steam from the turbine in
      bars
10 T2 = 205.00
      // temperature of exiting steam from the turbine
      in degree celcius
11 \text{ m2dot} = 26.15
      mass flow rate of exiting steam from the turbine
      in kg/s
12 \text{ TO} = 298.00
```

```
// in kelvin
13
14
15 // Part(a)
16 // From table A-4,
17 \text{ h1} = 3353.54
       in kj/kg
18 \text{ h2} = 2865.96
       in kj/kg
19 \text{ s1} = 6.8773
      // in kj/kg.k
20 \text{ s2} = 7.0806
      // in kj/kg.k
21
22 // Calculations
23 // From assumption, For each control volume, Qcvdot =
      0 and kinetic and potential energy effects are
      negligible, the mass and energy rate
24 // balances for a control volume enclosing the
      turbine reduce at steady state to give
25 \text{ Wedot} = m2dot *(h1-h2)/1000
                                          // power in MW
26 	ext{ Ef2dot} = 	ext{Ef1dot} + m2dot * (h2-h1-T0*(s2-s1))/1000
                      // the rate exergy exits with the
      steam in MW
27
28 // Results
29 printf ('For the turbine, the power is %.2 f MW.',
      Wedot)
30 printf (' For the turbine, the rate exergy exits with
       the steam is \%.2 f MW.', Ef2dot)
31
32 // Part (b)
33 // Calculations
```

```
34 	 c1 = cF*(EfFdot/Ef1dot) + ((Zbdot/Ef1dot)/10**3)*100
               // unit cost of exiting steam from
      boiler in cents/Kw.h
35 c2 = c1
      // Assigning the same unit cost to the steam
      entering and exiting the turbine
36 \text{ ce} = c1*((Ef1dot-Ef2dot)/Wedot) + ((Ztdot/Wedot))
      /10**3)*100 // unit cost of power in cents/kw.h
37
38 // Results
39 printf ('The unit costs of the steam exiting the
      boiler of exergy is: \%.2 f cents per kw.h.',c1)
40 printf ('The unit costs of the steam exiting the
      turbine of exergy is: %.2f cents per kw.h.',c2)
41 printf('Unit cost of power is: %f cents per kw.h.',
      ce)
42
43 // Part(c)
44 \text{ C2dot} = (c2*Ef2dot*10**3)/100
                                       // cost rate for
      low-pressure steam in dollars per hour
45 \text{ Cedot} = (ce*Wedot*10**3)/100
                                        // cost rate for
      power in dollars per hour
46
47 // Results
48 printf ('The cost rate of the steam exiting the
      turbine is: %.2f dollars per hour.', C2dot)
49 printf (' The cost rate of the power is: \%.2 \,\mathrm{f}
      dollars per hour.', Cedot)
```

# Chapter 8

# Vapor Power Systems

#### Scilab code Exa 8.1 Example

```
1 // Given:-
2 p1 = 8.0
      // pressure of saturated vapor entering the
      turbine in MPa
3 p3 = 0.008
      // pressure of saturated liquid exiting the
      condenser in MPa
4 \text{ Wcycledot} = 100.00
      // the net power output of the cycle in MW
6 // Analysis
7 // From table A-3
8 \text{ h1} = 2758.0
      // in kj/kg
9 \text{ s1} = 5.7432
      // in kj/kg.k
10 \text{ s2} = \text{s1}
11 \text{ sf} = 0.5926
      // in kj/kg.k
12 \text{ sg} = 8.2287
      // in kj/kg.k
```

```
13 \text{ hf} = 173.88
      // in kj/kg
14 \text{ hfg} = 2403.1
      // in kj/kg
15 \text{ v3} = 1.0084e-3
      // in m^3/kg
16
17 // State 3 is saturated liquid at 0.008 MPa, so
18 h3 = 173.88
      // in kj/kg
19
20 // Calculations
21 \quad x2 = (s2-sf)/(sg-sf)
      // quality at state 2
22 \quad h2 = hf + x2*hfg
23 p4 = p1
24 \text{ h4} = \text{h3} + \text{v3}*(\text{p4}-\text{p3})*10**6*10**-3
      // in kj/kg
25
26 // Part(a)
27 //Mass and energy rate balances for control volumes
      around the turbine and pump give, respectively
28 wtdot = h1 - h2
29 \text{ wpdot} = h4-h3
30
31 // The rate of heat transfer to the working fluid as
       it passes through the boiler is determined using
       mass and energy rate balances as
32 \text{ qindot} = h1-h4
33
34 eta = (wtdot-wpdot)/qindot
                                    // thermal efficiency)
35
36 // Result for part a
37 printf( 'The thermal efficiency for the cycle is \%
      .2 f', eta)
38
39 // Part (b)
```

```
40 bwr = wpdot/wtdot
                                             // back work
      ratio
41
42 // Result
43 printf(' The back work ratio is %f',bwr)
44
45 // Part(c)
46 \text{ mdot} = (\text{Wcycledot}*10**3*3600)/((h1-h2)-(h4-h3))
            // mass flow rate in kg/h
47
48 // Result
49 printf ('The mass flow rate of the steam is %.2 f kg
      /h . ', mdot)
50
51 // Part (d)
52 Qindot = mdot*qindot/(3600*10**3)
                            // in MW
53
54 // Results
55 printf('The rate of heat transfer, Qindot, into the
      working fluid as it passes through the boiler, is
       \%.2 f MW., Qindot)
56
57 // Part (e)
58 \text{ Qoutdot} = \text{mdot}*(h2-h3)/(3600*10**3)
                          // in MW
59
60 // Results
61 printf (' The rate of heat transfer, Qoutdot from the
       condensing steam as it passes through the
      condenser, is %.2 f MW.', Qoutdot)
62
63 // Part (f)
64 // From table A-2
65 hcwout= 146.68
                                                // in kj/kg
66 \text{ hcwin} = 62.99
```

# Scilab code Exa 8.2 Example

```
1 // Given:-
2 \text{ etat} = .85
                                                     // given
       that the turbine and the pump each have an
      isentropic efficiency of 85%
3 // Analysis
4 // State 1 is the same as in Example 8.1, so
5 h1 = 2758.0
                                                     // in kj
      /kg
6 	 s1 = 5.7432
                                                     // in kj
     /kg.k
7 // From example 8.1
8 \text{ h1} = 2758.0
                                                     // in kj
     /kg
9 \text{ h2s} = 1794.8
                                                     // in kj
      /kg
10 // State 3 is the same as in Example 8.1, so
11 h3 = 173.88
                                                     // in kj
      /kg
12
13 // Calculations
14 h2 = h1 - etat*(h1-h2s)
                                                     // in kj
      /kg
15 \text{ wpdot} = 8.06/\text{etat}
                                                     // where
       the value 8.06 is obtained from example 8.1
```

```
16
17 \text{ h4} = \text{h3} + \text{wpdot}
18
19 // Part (a)
20 eta = ((h1-h2)-(h4-h3))/(h1-h4)
                                                   //
      thermal efficiency
21
22 // Result for part (a)
23 printf(' Thermal efficiency is: %.3f',eta)
24
25 // Part (b)
                                                   // given
26 Wcycledot = 100
      , a net power output of 100 MW
27 // Calculations
28 mdot = (Wcycledot*(10**3)*3600)/((h1-h2)-(h4-h3))
29 // Result for part (b)
30 printf ('The mass flow rate of steam, in kg/h, for
      a net power output of 100 MW is \%.3 f kg/h. ',
      mdot)
31
32 // Part(c)
33 Qindot = mdot*(h1-h4)/(3600 * 10**3)
34 // Result
35 printf ('The rate of heat transfer Qindot into the
      working fluid as it passes through the boiler, is
      \%.3 f MW. ', Qindot)
36
37 // Part (d)
38 \text{ Qoutdot} = \text{mdot}*(h2-h3)/(3600*10**3)
39 // Result
40 printf ('The rate of heat transfer Qoutdotfrom the
       condensing steam as it passes through the
      condenser, is %.3 f MW., Qoutdot)
41
42 // Part (e)
43 // From table A-2
44 \text{ hcwout} = 146.68
                                                   // in kj
      /kg
```

# Scilab code Exa 8.3 Example

```
1 // Given:-
2 clc;
                                         // temperature of
3 T1 = 480.0
      steam entering the first stage turbine in degree
      celcius
4 p1 = 8.0
                                         // pressure of
      steam entering the first stage turbine in MPa
5 p2 = 0.7
                                         // pressure of
      steam exiting the first stage turbine in MPa
6 T3 = 440.0
                                         // temperature of
      steam before entering the second stage turbine
7 \text{ Pcond} = 0.008
                                         // condenser
      pressure in MPa
                                          // the net power
8 \text{ Wcycledot} = 100.0
      output in MW
9
10 // Analysis
11 // From table A-4
12 \text{ h1} = 3348.4
                                          // in kj/kg
13 \text{ s1} = 6.6586
                                          // in kj/kg.k
                                          // isentropic
14 \ s2 = s1
      expansion through the first-stage turbine
15 // From table A-3
16 \text{ sf} = 1.9922
                                         // in kj/kg.k
17 \text{ sg} = 6.708
                                          // in kj/kg.k
18 \text{ hf} = 697.22
                                          // in kj/kg
```

```
// in kj/kg
19 \text{ hfg} = 2066.3
20
21 // Calculations
22 	 x2 = (s2-sf)/(sg-sf)
23 h2 = hf + x2*hfg
24 // State 3 is superheated vapor with p3 = 0.7 MPa
      and T3=440C, so from Table A-4
25 \text{ h3} = 3353.3
                                           // in kj/kg
                                           // in kj/kg.k
26 	 s3 = 7.7571
                                           // isentropic
27 \text{ s4} = \text{s3}
      expansion through the second-stage turbine
28 // For determing quality at state 4, from table A-3
29 \text{ sf} = 0.5926
                                           // in kj/kg.k
30 \text{ sg} = 8.2287
                                           // in kj/kg.k
                                           // in kj/kg
31 \text{ hf} = 173.88
32 \text{ hfg} = 2403.1
                                           // in kj/kg
33
34 // Calculations
35 \times 4 = (s4-sf)/(sg-sf)
36 \text{ h4} = \text{hf} + \text{x4*hfg}
37
38 // State 5 is saturated liquid at 0.008 MPa, so
39 \text{ h5} = 173.88
40 // The state at the pump exit is the same as in
      Example 8.1, so
41 \quad h6 = 181.94
42
43 // Part (a)
44 \text{ eta} = ((h1-h2)+(h3-h4)-(h6-h5))/((h1-h6)+(h3-h2))
45 // Result
46 printf ('\n The thermal efficiency of the cycle is:
      \%.2 f', eta)
47
48 // Part(b)
49 mdot = (Wcycledot*3600*10**3)/((h1-h2)+(h3-h4)-(h6-h4))
      h5))
50 printf ('\n The mass flow rate of steam, is: %.2 f kg
      /h.', mdot)
```

```
51
52  // Part(c)
53  Qoutdot = (mdot*(h4-h5))/(3600*10**3)
54  printf('\nThe rate of heat transfer Qoutdot from the condensing steam as it passes through the condenser, is %.2 f MW', Qoutdot)
```

# Scilab code Exa 8.4 Example

```
1 // Given :-
2 // Part (a)
3 \text{ etat} = 0.85
      // given efficiency
4 // From the solution to Example 8.3, the following
      specific enthalpy values are known, in kJ/kg
5 h1 = 3348.4
6 \text{ h2s} = 2741.8
7 h3 = 3353.3
8 \text{ h4s} = 2428.5
9 h5 = 173.88
10 \text{ h6} = 181.94
11
12
13 // Calculations
14 h2 = h1 - etat*(h1 - h2s)
      // The specific enthalpy at the exit of the first
      -stage turbine in kj/kg
15 \text{ h4} = \text{h3} - \text{etat*(h3-h4s)}
      // The specific enthalpy at the exit of the
      second-stage turbine in kj/kg
16 eta = ((h1-h2)+(h3-h4)-(h6-h5))/((h1-h6)+(h3-h2))
17
```

```
18 // Result
19 printf(' The thermal efficiency is: %f',eta)
20
21 // Part (b)
22 x = linspace(0.85, 1, 50)
23 \text{ for } i = 1:50
24
       h2(i) = h1 - x(i)*(h1 - h2s)
                                                  // The
          specific enthalpy at the exit of the first-
          stage turbine in kj/kg
       h4(i) = h3 - x(i)*(h3-h4s)
25
                                                    // The
           specific enthalpy at the exit of the second-
          stage turbine in kj/kg
       y(i) = ((h1-h2(i))+(h3-h4(i))-(h6-h5))/((h1-h6))
26
          +(h3-h2(i))
27 end
28
29 \text{ plot}(x,y)
30 xlabel('isentropic turbine efficiency')
31 ylabel ('cycle thermal efficiency')
```

## Scilab code Exa 8.5 Example

```
1 // Given:-
2 T1 = 480.0

    // temperature of steam entering the turbine in
    degree celcius
3 p1 = 8.0

    // pressure of steam entering the turbine in MPa
4 Pcond = 0.008

    // condenser pressure in MPa
```

```
5 \text{ etat} = 0.85
      // turbine efficiency
6 \text{ Wcycledot} = 100.0
      // net power output of the cycle
7
8
9 // Analysis
10 // With the help of steam tables
11 \ h1 = 3348.4
     // in kj/kg
12 h2 = 2832.8
     // in kj/kg
13 \text{ s2} = 6.8606
      // in kj/kg.k
14 \text{ h4} = 173.88
      // in kj/kg
15 // With s3s = s2, the quality at state 3s is x3s =
      0.8208; using this, we get
16 \text{ h3s} = 2146.3
      // in kj/kg
17
18 // Calculations
19 // The specific enthalpy at state 3 can be
      determined using the efficiency of the second-
      stage turbine
20 h3 = h2 - etat*(h2-h3s)
21
22 // State 6 is saturated liquid at 0.7 MPa. Thus,
23 \text{ h6} = 697.22
      // in kj/kg
```

```
24 // For determining specific enthalpies at states 5
      and 7, we have
25 p5 = 0.7
      // in MPa
26 p4 = 0.008
      // in MPa
27 p7 = 8.0
      // in MPa
28 p6 = 0.7
      // in MPa
29 \text{ v4} = 1.0084 \text{e} - 3
      // units in m<sup>3</sup>/kg, obtained from steam tables
30 \text{ v6} = 1.1080e-3
      // units in m<sup>3</sup>/kg, obtained from steam tables
31
32 // Calculations
33 h5 = h4 + v4*(p5-p4)*10**6*10**-3
      // in kj/kg
34 \text{ h7} = \text{h6} + \text{v6}*(\text{p7}-\text{p6})*10**3
      // in kj/kg
35
36 // Applying mass and energy rate balances to a
      control volume enclosing the open heater, we find
       the fraction y of the flow extracted at state 2
      from
37 \text{ y} = (h6-h5)/(h2-h5)
39 // Part(a)
40 \text{ wtdot} = (h1-h2) + (1-y)*(h2-h3)
```

```
// the total turbine work output, units in KJ/Kg
wpdot = (h7-h6) + (1-y)*(h5-h4)

// The total pump work per unit of mass passing through the first-stage turbine, in KJ/kg
42 qindot = h1 - h7

// in kj/kg
43 eta = (wtdot-wpdot)/qindot
44
45 // Results
46 printf( 'The thermal efficiency is: %.2f',eta)
47
48 // Part(b)
49 m1dot = (Wcycledot*3600*10**3)/(wtdot-wpdot)
50
51 // Results
52 printf( 'The mass flow rate of steam entering the first turbine stage, is: %.2f kg/h.',m1dot)
```

#### Scilab code Exa 8.6 Example

```
1
2 // Given:-
3 // Analysis
4 // State 1 is the same as in Example 8.3, so
5 h1 = 3348.4

    // in kj/kg
6 s1 = 6.6586

    // in kj/kg.k
7 // State 2 is fixed by p2 2.0 MPa and the specific entropy s2, which is the same as that of state 1. Interpolating in Table A-4, we get
```

```
8 h2 = 2963.5
      // in kj/kg
9 // The state at the exit of the first turbine is the
       same as at the exit of the first turbine of
      Example 8.3, so
10 \text{ h3} = 2741.8
      // in kj/kg
11 // State 4 is superheated vapor at 0.7 MPa, 440C.
      From Table A-4,
12 \text{ h4} = 3353.3
      // in kj/kg
13 \text{ s4} = 7.7571
      // in kj/kg.k
14 // Interpolating in table A-4 at p5 = .3MPa and s5 =
       s4, the enthalpy at state 5 is
15 h5 = 3101.5
      // in kj/kg
16 // Using s6 = s4, the quality at state 6 is found to
       be
17 \times 6 = 0.9382
18 // Using steam tables, for state 6
19 \text{ hf} = 173.88
      // in kj/kg
20 \text{ hfg} = 2403.1
      // in kj/kg
21
22 \text{ h6} = \text{hf} + \text{x6*hfg}
24 // At the condenser exit, we have
25 h7 = 173.88
```

```
// in kj/kg
26 \text{ v7} = 1.0084e-3
      // in m^3/kg
27 p8 = 0.3
      // in MPa
28 p7 = 0.008
      // in MPa
29
30 \text{ h8} = \text{h7} + \text{v7}*(\text{p8-p7})*10**6*10**-3
      The specific enthalpy at the exit of the first
      pump in kj/kg
31 // The liquid leaving the open feedwater heater at
      state 9 is saturated liquid at 0.3 MPa. The
      specific enthalpy is
32 \text{ h9} = 561.47
      // in kj/kg
33
34 // For the exit of the second pump,
35 \text{ v9} = 1.0732e-3
      // in m^3/kg
36 p10 = 8.0
     // in MPa
37 p9 = 0.3
     // in MPa
38 \text{ h10} = \text{h9} + \text{v9*(p10-p9)*10**6*10**-3}
      The specific enthalpy at the exit of the second
      pump in kj/kg
39 // The condensate leaving the closed heater is
      saturated at 2 MPa. From Table A-3,
```

```
40 \text{ h12} = 908.79
       // in kj/kg
41 \text{ h} 13 = \text{h} 12
       // since The fluid passing through the trap
       undergoes a throttling process
42 // For the feedwater exiting the closed heater
43 hf = 875.1
       // in kj/kg
44 \text{ vf} = 1.1646 e - 3
       // in m^3/kg
45 p11 = 8.0
       // in MPa
46 \text{ psat} = 1.73
       // in MPa
47 \text{ h11} = \text{hf} + \text{vf*(p11-psat)*}10**6*10**-3
                                                           // in
       kj/kg
48
49 ydash = (h11-h10)/(h2-h12)
       // the fraction of the total flow diverted to the
        closed heater
50 \text{ ydashdash} = ((1-\text{ydash})*\text{h8+ydash}*\text{h13-h9})/(\text{h8-h5})
                                               // the fraction
       of the total flow diverted to the open heater
51
52 // Part (a)
53 \text{ wt1dot} = (h1-h2) + (1-ydash)*(h2-h3)
                                                             // The
        work developed by the first turbine per unit of
       mass entering in kj/kg
54 \text{ wt2dot} = (1-\text{ydash})*(\text{h4-h5}) + (1-\text{ydash-ydashdash})*(\text{h5})
```

```
-h6)
                                      // The work developed
       by the second turbine per unit of mass in kj/kg
55 \text{ wpldot} = (1-ydash-ydashdash)*(h8-h7)
                                                       // The
       work for the first pump per unit of mass in kj/
      kg
56 \text{ wp2dot} = h10-h9
      // The work for the second pump per unit of mass
      in kj/kg
57 \text{ qindot} = (h1-h11) + (1-ydash)*(h4-h3)
      total heat added expressed on the basis of a unit
       of mass entering the first
58
59 eta = (wt1dot+wt2dot-wp1dot-wp2dot)/qindot
                                                // thermal
      efficiency
60
61 // Result
62 printf( 'The thermal efficiency is: \%.2 \, \mathrm{f}', eta)
63
64 // Part(b)
65 Wcycledot = 100.0
      // the net power output of the cycle in MW
66 \text{ m1dot} = (\text{Wcycledot}*3600*10**3)/(\text{wt1dot}+\text{wt2dot}-\text{wp1dot})
      -wp2dot)
67
68 // Result
69 printf (' The mass flow rate of the steam entering
      the first turbine, in kg/h is: \%.2 f', m1dot)
```

# Scilab code Exa 8.7 Example

```
2 // Given:-
3 // Analysis
4 // The solution to Example 8.2 gives
5 \text{ h1} = 2758
     // in kj/kg
6 \text{ h4} = 183.36
     // in kj/kg
7 // From table A-22
8 \text{ hi} = 1491.44
     // in kj/kg
9 \text{ he} = 843.98
      // in kj/kg
10 // Using the conservation of mass principle and
      energy rate balance, the ratio of mass flow rates
       of air and water is
11 madotbymdot = (h1-h4)/(hi-he)
12 // From example 8.2
13 \text{ mdot} = 4.449e5
      // in kg/h
14 \text{ madot} = \text{madotbymdot*mdot}
      // in kg/h
15
16 // Part(a)
17 T0 = 295
```

```
// in kj/kg.k
32 // Calculation
33 Rout = mdot*(h1-h4-T0*(s1-s4))/(3600*10**3)
                                          // in MW
34 // Result
35 printf ('The net rate at which exergy is carried
      from the heat exchanger by the water stream, is:
       \%.2 \text{ f MW} . , \text{Rout}
36
37 // Part(c)
38 Eddot = Rin-Rout
      // in MW
39 // Result
40 printf (' The rate of exergy destruction, in MW is:
       \%.2\,\mathrm{f} ', Eddot)
41
42 // Part (d)
43 epsilon = Rout/Rin
44 // Result
45 printf (' The exergetic efficiency is: \%.2 \,\mathrm{f}',
      epsilon)
```

#### Scilab code Exa 8.8 Example

```
1
2 // Given:-
3 T0 = 295.00

// in kelvin
4 P0 = 1.00

// in atm
5
6 // Analysis
```

```
7 // From table A-3
8 \text{ s1} = 5.7432
     // in kj/kg.k
9 \text{ s3} = 0.5926
     // in kj/kg.k
10
11 // Using h2 = 1939.3 \text{ kJ/kg} from the solution to
     Example 8.2, the value of s2 can be determined
      from Table A-3 as
12 	 s2 = 6.2021
     // in kj/kg.k
13 \text{ s4} = 0.5957
     // in kj/kg.k
14 \text{ mdot} = 4.449e5
     // in kg/h
15
16 // Calculations
17 Eddot = mdot*T0*(s2-s1)/(3600*10**3)
                                                 // the
      rate of exergy destruction for the turbine in MW
18 EddotP = mdot*T0*(s4-s3)/(3600*10**3)
                                                // the
      exergy destruction rate for the pump
19
20 // Results
21 printf (' The rate of exergy destruction for the
      turbine is: %.2 f MW.', Eddot)
22 // From the solution to Example 8.7, the net rate at
       which exergy is supplied by the cooling
      combustion gases is 231.28 MW
23 printf ('The turbine rate of exergy destruction
      expressed as a percentage is: %.f',(Eddot
      /231.28) *100)
```

```
24 // However, since only 69% of the entering fuel
      exergy remains after the stack loss and
      combustion exergy destruction are accounted for,
25 // it can be concluded that
26 printf (' Percentage of the exergy entering the
      plant with the fuel destroyed within the turbine
      is : \%.2 f', 0.69*(Eddot/231.28)*100)
27 printf ('The exergy destruction rate for the pump
      in MW is : \%.2 f ', EddotP)
28 printf ('and expressing this as a percentage of the
      exergy entering the plant as calculated above, we
      have \%.2 \, \text{f}, (EddotP/231.28) *69)
29 printf ('The net power output of the vapor power
      plant of Example 8.2 is 100 MW. Expressing this
      as a percentage of the rate at which exergy is ')
30 printf ('carried into the plant with the fuel, \%.2\,\mathrm{f}'
      ,(100/231.28)*69)
```

#### Scilab code Exa 8.9 Example

```
// in kj/kg
11 \text{ hi} = 62.99
     // in kj/kg
12 \text{ se} = 0.5053
      // in kj/kg.k
13 \text{ si} = 0.2245
      // in kj/kg.k
14 // Calculations
15 Rout = mcwdot*(he-hi-T0*(se-si))/(3600*10**3)
                                  // The net rate at
      which exergy is carried out of the condenser in
     MW
16 // Results
17 printf (' The net rate at which exergy is carried
      from the condenser by the cooling water, is: \%.2 f
      MW. ', Rout)
18 printf ( 'Expressing this as a percentage of the
      exergy entering the plant with the fuel, we get \%
      .2 f percent', (Rout/231.28) *69)
19
20 // Part(b)
21 // From table
22 	 s3 = 0.5926
     // in kj/kg.k
23 	 s2 = 6.2021
     // in kg/kg.k
24 \text{ mdot} = 4.449e5
      // in kg/h
25 // Calculations
26 Eddot = T0*(mdot*(s3-s2)+mcwdot*(se-si))
      /(3600*10**3)
                                          // the rate of
```

```
exergy destruction for the condenser in MW

27 // Results

28 printf( 'The rate of exergy destruction for the condenser is: %.2 f MW.', Eddot)

29 printf( 'Expressing this as a percentage of the exergy entering the plant with the fuel, we get, %.2 f percent', (Eddot/231.28)*69)
```

# Chapter 9

# Gas Power Systems

# Scilab code Exa 9.1 Example

```
9 // At T1 = 300k, table A-22 gives
10 \ u1 = 214.07
     // in kj/kg
11 \text{ vr1} = 621.2
12 // Interpolating with vr2 in Table A-22, we get
13 T2 = 673.00
     // in kelvin
14 u2 = 491.2
     // in kj/kg
15 // At T3 = 2000 K, Table A-22 gives
16 \text{ u3} = 1678.7
    // in kj/kg
17 \text{ vr3} = 2.776
18 // Interpolating in Table A-22 with vr4 gives
19 \quad T4 = 1043
    // in kelvin
20 \text{ u4} = 795.8
      // in kj/kg
21
22 // Calculations
23 // For the isentropic compression Process 1 2
24 \text{ vr2} = \text{vr1/r}
25 // With the ideal gas equation of state
26 p2 = p1*(T2/T1)*(r)
      // in bars
27 // Since Process 2 3 occurs at constant volume,
     the ideal gas equation of state gives
28 p3 = p2*(T3/T2)
      // in bars
29 // For the isentropic expansion process 3 4
```

```
30 \text{ vr4} = \text{vr3}*(r)
31 // The ideal gas equation of state applied at states
      1 and 4 gives
32 p4 = p1*(T4/T1)
     // in bars
33
34 // Results
35 printf( 'At state1, the pressure is: \%f bar.',p1)
36 printf ('At state1, the temperature is %f kelvin.',
      T1)
37 printf(' At state2, the pressure is: %.3f bar.',p2
38 printf (' At state2, the temperature is %f kelvin.',
     T2)
39 printf(' At state3, the pressure is: %.3f bar.',p3
40 printf ('At state3, the temperature is %f kelvin.',
41 printf(' At state4, the pressure is: %.4f bar.',p4
42 printf (' At state4, the temperature is %f kelvin.',
     T4)
43
44 // Part (b)
45 \text{ eta} = 1-(u4-u1)/(u3-u2)
     // thermal efficiency
46 // Result
47 printf(' The thermal efficiency is: %.2f',eta)
48
49 // Part(c)
50 R = 8.314
     // universal gas constant, in SI units
51 M = 28.97
     // molar mass of air in grams
```

#### Scilab code Exa 9.2 Example

```
1
2  // Given :-
3  clc;
4  r = 18.00

    // compression ratio
5  T1 = 300.00

    // temperature at the beginning of the compression process in kelvin
6  p1 = 0.1

    // pressure at the beginning of the compression process in MPa
7  rc = 2.00

    // cutoff ratio
8
9  // Part(a)
```

```
10 // With T1 = 300 K, Table A-22 gives
11 \ u1 = 214.07
     // in kj/kg
12 \text{ vr1} = 621.2
13 // Interpolating in Table A-22, we get
14 T2 = 898.3
    // in kelvin
15 \text{ h2} = 930.98
    // in kj/kg
16 // From Table A-22,
17 h3 = 1999.1
   // in kj/kg
18 \text{ vr3} = 3.97
20 // Interpolating in Table A-22 with vr4, we get
21 \quad u4 = 664.3
     // in kj/kg
22 \quad T4 = 887.7
      // in kelvin
23
24 // Calculations
25 // Since Process 2 3 occurs at constant pressure,
      the ideal gas equation of state gives
26 T3 = rc*T2
     // in kelvin
27 // With the ideal gas equation of state
28 p2 = p1*(T2/T1)*(r)
     // in MPa
29 p3 = p2
30 // For the isentropic compression process 1 2
```

```
31 \text{ vr2} = \text{vr1/r}
32 // For the isentropic expansion process 3
33 \text{ vr4} = (r/rc)*vr3
34 // The ideal gas equation of state applied at states
       1 and 4 gives
35 p4 = p1*(T4/T1)
      // in MPa
36
37 // Results
38 printf ('\n At state1, the pressure is: \%.2 f bar.',
39 printf( '\n At state1, the temperature is \%.2 \, \mathrm{f}
      kelvin.',T1)
40 printf( '\n At state2, the pressure in bar is : \%.2\,\mathrm{f}
       bar.',p2)
41 printf('\n At state2, the temperature is
                                                    \%.2 {\rm f}
      kelvin.,,T2)
42 printf ('\n At state3, the pressure in bar is: \%.2 f
       bar.',p3)
43 printf ('\n At state3, the temperature is \%.2 \, \mathrm{f}
      kelvin.',T3)
44 printf ('\n At state4, the pressure is: %.2 f MPa.',
45 printf( '\n At state4, the temperature is \%.2 \, \mathrm{f}
      kelvin.', T4)
46
47 // Part (b)
48 \text{ eta} = 1 - (u4 - u1)/(h3 - h2)
49 printf ('\n The thermal efficiency is: \%.2 \,\mathrm{f}', eta)
50
51 // Part(c)
52 R = 8.314
      // universal gas constant, in SI units
53 M = 28.97
      // molar mass of air in grams
```

#### Scilab code Exa 9.3 Example

```
1 // Given :-
2 T1 = 300.00

    // beginning temperature in kelvin
3 p1 = 0.1

    // beginning pressure in MPa
4 r = 18.00

    // compression ratio
5 pr = 1.5

    // The pressure ratio for the constant volume part of the heating process
6 vr = 1.2

    // The volume ratio for the constant pressure part of the heating process
```

```
8 // Analysis
9 // States 1 and 2 are the same as in Example 9.2, so
10 \ u1 = 214.07
   // in kj/kg
11 T2 = 898.3
    // in kelvin
12 u2 = 673.2
      // in kj/kg
13
14 // Interpolating in Table A-22, we get
15 \text{ h3} = 1452.6
      // in kj/kg
16 u3 = 1065.8
      // in kj/kg
17
18 // From Table A-22,
19 \text{ h4} = 1778.3
      // in kj/kg
20 \text{ vr4} = 5.609
21
22 // Interpolating in Table A-22, we get
23 \text{ u5} = 475.96
      // in kj/kg
24
25 // Calculations
26 // Since Process 2 3 occurs at constant volume,
      the ideal gas equation of state reduces to give
27 T3 = pr*T2
      // in kelvin
```

```
28 // Since Process 3 4 occurs at constant pressure,
      the ideal gas equation of state reduces to give
29 T4 = vr*T3
      // in kelvin
30 // Process 4 5 is an isentropic expansion, so
31 \text{ vr5} = \text{vr4*r/vr}
32
33 // Part(a)
34 \text{ eta} = 1-(u5-u1)/((u3-u2)+(h4-h3))
35 // Result
36 printf ('The thermal efficiency is: %.2f',eta)
37
38 // Part (b)
39 // The specific volume at state 1 is evaluated in
      Example 9.2 as
40 \text{ v1} = 0.861
      // in m^3/kg
41 mep = (((u3-u2)+(h4-h3)-(u5-u1))/(v1*(1-1/r)))
      *10**3*10**-6
                                            // in MPa
42
43 // Result
44 \tt printf( ' The mean effective pressure, is : \%.2\,f MPa
      .', mep)
```

#### Scilab code Exa 9.4 Example

```
4 p1 = 100.00
      // in kpa
 5 pr = 10.00
      // compressor pressure ratio
 6 T3 = 1400.00
      // turbine inlet temperature in kelvin
8 // Analysis
9 // At state 1, the temperature is 300 K. From Table
      A-22,
10 \text{ h1} = 300.19
     // in kj/kg
11 \text{ pr1} = 1.386
12
13
14 // Interpolating in Table A-22,
15 \text{ h2} = 579.9
     // in kj/kg
16 // From Table A-22
17 h3 = 1515.4
      // in kj/kg
18 \text{ pr3} = 450.5
19
20 // Interpolating in Table A-22, we get
21 \text{ h4} = 808.5
      // in kj/kg
22
23 // calculations
24 pr2 = pr*pr1
25 \text{ pr4} = \text{pr3*1/pr}
26
```

```
27
28 // Part(a)
29 eta = ((h3-h4)-(h2-h1))/(h3-h2)
     // thermal efficiency
30 // Result
31 printf(' The thermal efficiency is: %.4f', eta)
32
33 // Part(b)
34 \text{ bwr} = (h2-h1)/(h3-h4)
     // back work ratio
35 // Result
36 printf (' The back work ratio is: %.4f', bwr)
37
38 // Part(c)
39 R = 8.314
     // universal gas constant, in SI units
40 M = 28.97
      // molar mass of air in grams
41 // Calculations
42 \text{ mdot} = AV*p1/((R/M)*T1)
      // mass flow rate in kg/s
43 Wcycledot = mdot*((h3-h4)-(h2-h1))
                                                      //
      The net power developed
44 // Result
45 printf(' The net power developed, is: %.2 f kW.',
      Wcycledot)
```

Scilab code Exa 9.6 Example

```
1 // Given:-
2 T1 = 300.00
     // in kelvin
3 \text{ AV} = 5.00
      // volumetric flow rate in m<sup>3</sup>/s
4 p1 = 100.00
     // in kpa
5 pr = 10.00
     // compressor pressure ratio
6 T3 = 1400.00
      // turbine inlet temperature in kelvin
7 \text{ Wt_ms} = 706.9
      // kJ/kg
8 \text{ Wc_m} = 279.7
9 // Analysis
10 // At state 1, the temperature is 300 K. From Table
     A-22,
11 h1 = 300.19
      // in kj/kg
12 \text{ pr1} = 1.386
13
14
15 // Interpolating in Table A-22,
16 \text{ h2} = 579.9
      // in kj/kg
17 // From Table A-22
18 \text{ h3} = 1515.4
      // in kj/kg
19 \text{ pr3} = 450.5
```

```
20
21 // Interpolating in Table A-22, we get
22 \text{ h4} = 808.5
      // in kj/kg
23
24 // calculations
25 \text{ Wtbym} = 0.8*\text{Wt_ms}
26 \text{ Wcbym} = \text{Wc_m/0.8}
27 \text{ h2} = 300.19 + \text{Wcbym}
28
29 //pr2 = pr*pr1
30 / pr4 = pr3*1/pr
31
32
33 // Part(a)
34 // \text{eta} = ((h3-h4)-(h2-h1))/(h3-h2)
       // thermal efficiency
35 Qinbym = h3 - h2
36 n = (Wtbym - Wcbym) / Qinbym
37 // Result
38 printf( '\n The thermal efficiency is : \%.3 \, \text{f} ',n)
39
40 // Part (b)
41 / bwr = (h2-h1)/(h3-h4)
      // back work ratio
42 \text{ bwr} = \text{Wcbym/Wtbym}
43 // Result
44 printf( '\n The back work ratio is : \%.3f',bwr)
45
46 // Part(c)
47 R = 8.314
      // universal gas constant, in SI units
48 M = 28.97
```

```
// molar mass of air in grams
49 // Calculations
50 //mdot = AV*p1/((R/M)*T1)

    // mass flow rate in kg/s
51 Wcycledot = 5.807*(Wcbym-Wtbym)

    The net power developed
52 // Result
53 printf( '\n The net power developed, is : %.f kW .'
,-Wcycledot)
```

## Scilab code Exa 9.7 Example

```
// in kj/kg
12 eta = ((h3-h4)-(h2-h1))/(h3-hx)
      // thermal efficiency
13 // Result
14 printf ('The thermal efficiency is: %.2f', eta)
15
16 // Part (b)
17
18 etareg = linspace(0,0.8,50)
19 \text{ for } i = 1:50
20
       x(i) = (etareg(i)*(h4-h2))+h2
       eta(i) = ((h3-h4)-(h2-h1))/(h3-x(i))
22 \text{ end}
23
24 plot(etareg, eta)
25 xlabel('Regenerator effectiveness')
26 ylabel('Thermal efficiency')
```

### Scilab code Exa 9.8 Example

```
7 // The temperature at state b is the same as at
       state 3, so
8 \text{ hb} = \text{h3}
9
10 \text{ pa} = 300.00
      // in kpa
11 p3 = 1000.00
      // in kpa
12 // From table A-22
13 \text{ pr3} = 450.5
14
15 // Interpolating in Table A-22, we get
16 \text{ ha} = 1095.9
      // in kj/kg
17 p4 = 100.00
      // in kpa
18 \text{ pb} = 300.00
      // in kpa
19 // Interpolating in Table A-22, we obtain
20 \text{ h4} = 1127.6
       // in kj/kg
21
22 // Calculions
23 \text{ pra} = \text{pr3}*(\text{pa/p3})
24 \text{ prb} = \text{pra}
25 \text{ pr4} = \text{prb*}(\text{p4/pb})
26 // Since the regenerator effectiveness is 100%,
27 \text{ hx} = \text{h4}
28 eta = ((h3-ha)+(hb-h4)-(h2-h1))/((h3-hx)+(hb-ha))
                                             // thermal
       efficiency
29
```

```
30 // Result 31 printf( 'The thermal efficiency is : \%.2\,\mathrm{f} ', eta)
```

### Scilab code Exa 9.9 Example

```
1 // Given:-
2 T1 = 300.00
    // in kelvin
3 p1 = 100.00
    // in kpa
4 p2 = 1000.00
   // in kpa
5 p3 = p2
6 \text{ pc} = 300.00
     // in kpa
7 \text{ pd} = 300.00
    // in kpa
8 \text{ Td} = 300.00
      // in kelvin
9
10
11 // Part(a)
12 // From table A-22
13 \text{ prd} = 1.386
14 // Interpolating in Table A-22, we get
15 T2 = 422
      // in kelvin
16 \text{ h2} = 423.8
```

```
// in kj/kg
17 // Calculations
18 pr2 = prd*(p2/pd)
19 // Result
20 printf ('The temperature at the exit of the second
      compressor stage is: %.2f kelvin.',T2)
21
22 // Part (b)
23 // From Table A-22 at T1 = 300
24 \text{ h1} = 300.19
      // in kj/kg
25 // Since Td = T1,
26 \text{ hd} = 300.19
      // in kj/kg
27 // with pr data from Table A-22 together
28 \text{ pr1} = 1.386
29 // Interpolating in Table A-22, we obtain
30 \text{ hc} = 411.3
      // in kj/kg
31 // Calculations
32 \text{ prc} = \text{pr1}*(\text{pc/p1})
33 \text{ wcdot} = (hc-h1)+(h2-hd)
      // The total compressor work per unit of mass in
      kj/kg
34 // Result
35 printf (' The total compressor work input per unit
      of mass flow is : \%.2 \, \text{f kJ/kg'}, wcdot)
36
37 // Part(c)
38 // Interpolating in Table A-22, we get
39 \quad T3 = 574
      // in kelvin
```

```
// in kj/kg
// Calculations
// Calculations
// pr3 = pr1*(p3/p1)
wcdot = h3-h1

// The work input for a single stage of
compression in kj/kg
// Results
printf(' For a single stage of compression, the
temperature at the exit state is: %.2f kelvin',
T3)
printf(' For a single stage of compression, the
work input is: %.2f kJ.',wcdot)
```

### Scilab code Exa 9.11 Example

```
8 p5 = p4
9 p6 = p4
10 \quad T6 = 1400.00
   // in kelvin
11 T8 = T6
12 p7 = 300.00
// in kpa
13 p8 = p7
14 \text{ etac} = 0.8
     // isentropic efficiency of compressor
15 \text{ etat} = 0.8
      // isentropic efficiency of turbine
16 \text{ etareg} = 0.8
      // regenerator effectiveness
17 \ // \ Analysis
18 // From example 9.9
19 h1 = 300.19
 // in kj/kg
20 \text{ h3} = \text{h1}
  // in kj/kg
21 \text{ h2s} = 411.3
  // in kj/kg
22 \text{ h4s} = 423.8
      // in kj/kg
23 // From example 9.8
24 h6 = 1515.4
   // in kj/kg
25 \text{ h8} = \text{h6}
```

```
26 \text{ h7s} = 1095.9
      // in kj/kg
27 \text{ h9s} = 1127.6
      // in kj/kg
28
29 // Calculations
30 \text{ h4} = \text{h3} + (\text{h4s-h3})/\text{etac}
      // in kj/kg
31 h2 = h1 + (h2s-h1)/etac
      // in kj/kg
32 \text{ h9} = \text{h8-etat*(h8-h9s)}
      // in kj/kg
33 h7 = h6-etat*(h6-h7s)
      // in kj/kg
34 \text{ h5} = \text{h4+etareg*(h9-h4)}
      // in kj/kg
35
36 // Part(a)
37 // Calculations
38 \text{ wtdot} = (h6-h7)+(h8-h9)
      // The total turbine work per unit of mass flow
      in kj/kg
39 \text{ wcdot} = (h2-h1)+(h4-h3)
      // The total compressor work input per unit of
      mass flow in kj/kg
40 \text{ qindot} = (h6-h5)+(h8-h7)
      // The total heat added per unit of mass flow in
      kj/kg
```

```
41 eta = (wtdot-wcdot)/qindot
      // thermal efficiency
42 // Result
43 printf(' The thermal efficiency is: %.2f',eta)
44
45 // Part (b)
46 \text{ bwr} = \text{wcdot/wtdot}
     // back work ratio
47 // Result
48 printf( 'The back work ratio is: %.2f',bwr)
49
50 // Part(c)
51 Wcycledot = mdot*(wtdot-wcdot)
      // net power developed in kw
52 // Result
53 printf( 'The net power developed, is: \%.2 \, f \, kW.',
      Wcycledot)
```

### Scilab code Exa 9.12 Example

```
1 // Given:-
2 Ta = 240.00

// in kelvin
3 pa = 0.8

// in bar
4 Va = 278.00

// in m/s
5 PR = 8.00
```

```
// pressure ratio across the compressor
6 T3 = 1200.00
      // in kelvin
7 p5 = 0.8
      // in bar
9 // From table A-22
10 \text{ ha} = 240.02
     // in kj/kg
11 h1 = ha + ((Va**2)/2)*10**-3
     // in kj/kg
12 // Interpolating in Table A-22 gives
13 \text{ pr1} = 1.070
14 \text{ pra} = .6355
15
16 // Interpolating in Table A-22, we get
17 h2 = 505.5
      // in kj/kg
18 // At state 3 the temperature is given as T3 = 1200
      K. From Table A-22
19 \text{ h3} = 1277.79
      // in kj/kg
20
21
22 // Interpolating in Table A-22 with h4, gives
23 \text{ pr4} = 116.8
24 // pr data from table A-22 gives
25 \text{ pr4} = 116.00
26 \text{ pr3} = 238.00
27 // From table A-22
28 h5 = 621.3
```

```
// in kj/kg
29
30 // The expansion through the nozzle is isentropic to
31 p5 = .8
      // in bars
32
33 // Calculations
34 p1 = (pr1/pra)*pa
      // in bars
35 // With the help of assumption, 'The turbine work
      output equals the work required to drive the
      compressor.',
36 \text{ h4} = \text{h3+h1-h2}
      // in kj/kg
37 p2 = PR*p1
      // in bars
38 // Using assumption 'There is no pressure drop for
      flow through the combustor',
39 p3 = p2
40 p4 = p3*(pr4/pr3)
      // in bars
41 \text{ pr5} = \text{pr4}*(\text{p5/p4})
42 \quad V5 = ((2*(h4-h5)*10**3))**(0.5)
      // the velocity at the nozzle exit in m/s
43
44 // Results
45 printf(' The velocity at the nozzle exit in m/s is:
       \%.2\,\mathrm{f} ', V5)
46 printf(' pa in bars = \%.2 \,\mathrm{f}',pa)
47 printf (' p1 in bars = \%.2 \, \text{f}', p1)
48 printf(' p2 in bars = \%.2 \, \text{f',p2})
49 printf (' p3 in bars = \%.2 \, \text{f',p3})
```

```
50 printf( ' p4 in bars = \%.2 \, f',p4)
51 printf( ' p5 in bars = \%.2 \, f',p5)
```

#### Scilab code Exa 9.13 Example

```
1 // Given:-
2 Wnetdot = 45.00
   // in MW
3 T1 = 300.00
   // in kelvin
4 p1 = 100.00
   // in kpa
5 \text{ etac} = 0.84
    // The isentropic efficiency of the compressor
6 	 T3 = 1400.00
   // in kelvin
7 p2 = 1200.00
   // in kpa
8 p3 = p2
9 \text{ etat} = 0.88
   // isentropic efficiency of the turbine
10 \text{ T5} = 400.00
   // in kelvin
11 p4 = 100.00
 // in kpa
12 p5 = p4
```

```
13 \quad T7 = 400.00
     // in degree celcius
14 p7 = 8.00
      // in MPa
15 \text{ etatw } = 0.9
      // isentropic efficiency of turbine of the vapor
      cycle
16 p8 = 8.00
     // in kpa
17 p9 = p8
18 \text{ etap} = 0.8
      // isentropic efficiency of pump of the vapor
      cycle
19 \text{ TO} = 300.00
     // in kelvin
20 p0 = 100.00
      // -in kpa
21
22 // Analysis
23 // With procedure similar to that used in the
      examples of chapters 8 and 9, we can determine
      following property data
24 \text{ h1} = 300.19
     // in kj/kg
25 \text{ h2} = 669.79
     // in kj/kg
26 \text{ h3} = 1515.42
      // in kj/kg
```

```
27 \text{ h4} = 858.02
  // in kj/kg
28 \text{ h5} = 400.98
 // in kj/kg
29 h6 = 183.96
// in kj/kg 30 h7 = 3138.30
 // in kj/kg
31 h8 = 2104.74
 // in kj/kg
32 h9 = 173.88
 // in kj/kg
33 \text{ s1} = 1.7020
 // in kj/kg.k
34 	 s2 = 2.5088
 // in kj/kg.k
35 	 s3 = 3.3620
 // in kj/kg.k
36 	 s4 = 2.7620
 // in kj/kg.k
37 \text{ s5} = 1.9919
   // in kj/kg.k
38 	 s6 = 0.5975
   // in kj/kg.k
39 	 s7 = 6.3634
```

```
// in kj/kg.k
40 \text{ s8} = 6.7282
     // in kj/kg.k
41 	 s9 = 0.5926
     // in kj/kg.k
42
43 // Part(a)
44 // By applying mass and energy rate balances
45 // Calculations
46 mvdotbymgdot = (h4-h5)/(h7-h6)
      // ratio of mass flow rates of vapor and air
47 \text{ mgdot} = (Wnetdot*10**3)/(((h3-h4)-(h2-h1)) +
     mvdotbymgdot*((h7-h8)-(h6-h9))) // mass
      flow rate of air in kg/s
48 mvdot = mvdotbymgdot*mgdot
     // mass flow rate of vapor in kg/s
49 Wgasdot = mgdot*((h3-h4)-(h2-h1))*10**-3
                                                   // net
     power developed by gas turbine in MW
50 \text{ Wvapdot} = \text{mvdot}*((h7-h8)-(h6-h9))*10**-3
                                                   // net
     power developed by vapor cycle in MW
51
52 // Results
53 printf(' Mass flow rate of air is: %.2f kg/s.',
     mgdot)
54 printf (' Mass flow rate of vapor is: %.2 f kg/s.',
     mvdot)
55 printf ('Net power developed by gas turbine is:
      .2 f MW., Wgasdot)
56 printf ( 'Net power developed by vapor cycle is:
      .2 f MW. ', Wvapdot)
57
58
```

```
59 // Part (b)
60
61 // The net rate of exergy increase of the air
      passing through the combustor is
62 \text{ Edotf32} = \text{mgdot}*(h3-h2-T0*(s3-s2))*10**-3
                                                    // in MW
63 // The net rate exergy is carried out by the exhaust
       air stream at 5 is
64 \text{ Edotf51} = \text{mgdot}*(h5-h1-T0*(s5-s1))/10**3
                                                     // in
      MW
65 // The net rate exergy is carried out as the water
      passes through the condenser is
66 \text{ Edotf89} = \text{mvdot}*(\text{h8-h9-T0}*(\text{s8-s9}))*10**-3
                                                    // in MW
67 R = 8.314
      // universal gas constant, in SI units
68 M = 28.97
      // molar mass of air in grams
69 // The rate of exergy destruction for air turbine is
70 Eddott = mgdot*T0*(s4-s3-(R/M)*log(p4/p3))/10**3
                                       // in MW
71 // The rate of exergy destruction for compressor is
72 Eddotc = mgdot*T0*(s2-s1-(R/M)*log(p2/p1))/10**3
                                       // in MW
  // The rate of exergy destruction for steam turbine
74 \text{ Eddotst} = \text{mvdot}*T0*(s8-s7)/10**3
      // in MW
75 // The rate of exergy destruction for pump is
76 \text{ Eddotp} = \text{mvdot}*\text{T0}*(s6-s9)/10**3
      // in MW
77 // For heat exchanger
78 EddotHE = T0*(mgdot*(s5-s4)+mvdot*(s7-s6))/10**3
```

```
// in MW
79
80 // Results
81 printf( 'Balance sheet')
82 printf ('Net exergy increase of the gas passing')
83 printf( 'Through the combustor: %.2 f MW', Edotf32)
84 printf ('Disposition of the exergy:')
85 printf('
                Net power developed')
86 printf ('gas turbine cycle %.2 f MW', Wgasdot)
87 printf ('vapor cycle %.2 f MW', Wvapdot)
88 printf(' Net exergy lost')
89 printf ('with exhaust gas at state 5~\%.2 \,\mathrm{f} MW',
      Edotf51)
90 printf ('from water passing through condenser %.2 f
     MW', Edotf89)
91 printf('
                Exergy destruction')
92 printf ('air turbine %.2 f MW', Eddott)
93 printf ('compressor %.2 f MW', Eddotc)
94 printf ('steam turbine %.2 f MW', Eddotst)
95 printf ('pump %.2 f MW', Eddotp)
96 printf ('heat exchanger %.2 f MW', EddotHE)
```

#### Scilab code Exa 9.14 Example

```
6 k = 1.4
8 // Calculations
9 pstarbypnot = (1+(k-1)/2)**(k/(1-k))
10 pstar = pstarbypnot*pnot
11
12 // Part (a)
13 // Since back pressure of 500 kpa is less than
      critical pressure pstar (528kpa in this case)
      found above, the nozzle is choked
14 // At the exit
15 M = 1.00
16 p2 = pstar
     // in MPa
17 T2 = Tnot/(1+((k-1)/2)*(M**2))
      // exit temperature in kelvin
18 R = 8.314
     // universal gas constant, in SI units
19 \text{ Mwt} = 28.97
     // molar mass of air in grams
20 \text{ V2} = ((k*(R/Mwt)*T2*10**3)**0.5)
      // exit velocity in m/s
21 \text{ mdot} = (p2/((R/Mwt)*T2))*A2*V2*10**3
                                                         //
      mass flow rate in kg/s
22
23 // Results
24 printf (' The exit mach number for back pressure of
      500 \,\mathrm{kpa} is: \%.2 \,\mathrm{f}, M)
25 printf ('The mass flow rate in kg/s for back
      pressure of 500kpa is: %.2f', mdot)
26
27 // Part (b)
```

```
28 // Since the back pressure of 784kpa is greater than
       critical pressure of pstar determined above, the
      flow throughout the nozzle is subsonic and the
      exit pressure equals the back pressure,
29 p2 = 784.00
      // exit pressure in kpa
30 // Calculations
31 M2 = (((2.00)/(k-1))*(((pnot*10**3)/p2)**((k-1)/k)
                                           // exit mach
      number
32 \text{ T2} = \text{Tnot}/(1+((k-1)/2)*(M2**2))
      // exit temperature in kelvin
33 V2 = M2*((k*(R/Mwt)*10**3*T2)**0.5)
                                                           //
       exit velocity in m/s
34 \text{ mdot2} = (p2/((R/Mwt)*T2))*A2*V2
      // mass flow rate in kg/s
35 // Results
36 printf(' The mass flow rate at the exit for back
      pressure of 784 \,\mathrm{kpa} is: \%.2 \,\mathrm{f} kg/s.', mdot2)
37 printf ('The exit mach number for back pressure of
      784 \text{ kpa is: } \%.2 \text{ f', M2)}
```

#### Scilab code Exa 9.15 Example

```
// throat area in cm<sup>2</sup>
5 \text{ Ae} = 15.00
      // exit area in cm<sup>2</sup>
7 // The flow throughout the nozzle, including the
      exit, is subsonic. Accordingly, with this value
      for A2byAstar, Table 9.1 gives
8 M2 = 0.24
9 // \text{ For } M2 = 0.24
10 \quad T2byTnot = 0.988
11 p2bypnot = 0.959
12 k = 1.4
13 \text{ TO} = 280.00
      // in kelvin
14 \text{ pnot} = 6.8
      // in bars
15 // Calculations
16 // \text{ With Mt} = 0.7, \text{ Table } 9.1 \text{ gives}
17 AtbyAstar = 1.09437
18 A2byAstar = (Ae/At)*AtbyAstar
19 T2 = T2byTnot*T0
      // in kelvin
20 p2 = p2bypnot*pnot
      // in bars
21 \quad V2 = M2*((k*(8.314/28.97)*T2*10**3)**0.5)
      velocity at the exit in m/s
22 \text{ mdot} = (p2/((8.314/28.97)*T2))*Ae*V2*10**-2
                                                  // mass flow
       rate in kg/s
23 // Results
24 printf (' Part (a) the mass flow rate in kg/s is: \%
      .2 f', mdot)
```

```
25 printf ('The exit pressure in bars is: %.2f',p2)
26 printf(' The exit mach number is: %.2f',M2)
27
28 // Part(b)
29 \text{ Mt} = 1.00
     // mach number at the throat
30 // From table 9.1
31 \quad M2 = 0.26
32 \quad T2byTnot = 0.986
33 p2bypnot = 0.953
34
35 \text{ TO} = 280.00
     // in kelvin
36 \text{ pnot} = 6.8
     // in bars
37 // Calculations
38 T2 = T2byTnot*T0
     // in kelvin
39 p2 = p2bypnot*pnot
     // in bars
40 k = 1.4
41 \quad V2 = M2*((k*(8314/28.97)*T2)**0.5)
     // exit velocity in m/s
42 mdot = (p2/((8.314/28.97)*T2))*Ae*V2*10**-2
                                              // mass flow
       rate in kg/s
43 // Results
44 printf ('Part(b) the mass flow rate is: %.f kg/s
      . ', mdot)
45 printf (' The exit pressure is: %f bars.',p2)
46 printf(' The exit mach number is: %f', M2)
47
```

```
48 // Part (c)
49 // From part (b), the exit Mach number in the
      present part of the example is
50 M2 = 2.4
51 // Using this, Table 9.1 gives
52 p2bypnot = 0.0684
53 \text{ pnot} = 6.8
     // in bars
54 // Calculation
55 p2 = p2bypnot*pnot
      // in bars
56 // Results
57 // Since the nozzle is choked, the mass flow rate is
      the same as found in part (b).
58 printf(' Part(c) the mass flow rate is: %f kg/s.'
      , mdot)
59 printf (' The exit pressure is: %f bars.',p2)
60 printf ('The exit mach number is: %f', M2)
61
62 // Part(d)
63 // Since a normal shock stands at the exit and the
      flow upstream of the shock is isentropic, the
      Mach number Mx and the pressure px correspond to
      the values found in part (c),
64 \text{ Mx} = 2.4
65 px = 0.465
      // in bars
66 // Then, from Table 9.2
67 \text{ My} = 0.52
68 //py is the exit pressure
69 \text{ pybypx} = 6.5533
70 \text{ py = px*pybypx}
71
72 // The pressure downstream of the shock is thus
      3.047 bars. This is the exit pressure
```

```
73 // The mass flow is the same as found in part (b).
74 // Results
75 printf(' Part(d) the mass flow rate is: %f kg/s.'
      , mdot)
76 printf(' The exit pressure is: %.3f bars.',py)
77 printf (' The exit mach number is: %f', My)
78
79 // Part (e)
80 // A shock stands in the diverging portion where the
       area is
81 \text{ Ax} = 12.5
      // in cm<sup>2</sup>
82 // Since a shock occurs, the flow is sonic at the
      throat, so
83 \text{ Axstar} = 6.25
      // in cm<sup>2</sup>
84 At = Axstar
85 // The Mach number Mx can then be found from Table
      9.1, by using AxbyAxstar as
86 \text{ Mx} = 2.2
87
88 // Results
89 // With Mx = 2.2, the ratio of stagnation pressures
      is obtained from Table 9.2 as
90 \text{ pnotybypnotx} = 0.62812
91
92 // Using this ratio and noting that the flow is
      subsonic after the shock, Table 9.1 gives
93 \quad M2 = 0.43
94 // \text{ For } M2 = 0.43
95 p2bypnoty = 0.88
96 // Calculations
97 A2byAystar = (Ae/Axstar)*pnotybypnotx
98 p2 = p2bypnoty*pnotybypnotx*pnot
                                                          //
       in bars
```

# Chapter 10

# Refrigeration and Heat Pump Systems

# Scilab code Exa 10.1 Example

```
1 // Given:—
2 Tc = 273.00

    // temperature of cold region in kelvin
3 Th = 299.00

    // temperature of hot region in kelvin
4 mdot = 0.08

    // mass flow rate in kg/s
5 6 // Analysis
7 // At the inlet to the compressor, the refrigerant is a saturated vapor at 0C, so from Table A-10
8 h1 = 247.23

    // in kj/kg
9 s1 = 0.9190
```

```
// in kj/kg.k
10
11 // The pressure at state 2s is the saturation
      pressure corresponding to 26C, or
12 p2 = 6.853
      // in bars
13 // The refrigerant at state 2s is a superheated
      vapor with
14 \text{ h2s} = 264.7
      // in kj/kg
15 // State 3 is saturated liquid at 26C, so
16 \text{ h3} = 85.75
      // in kj/kg
17 \text{ h4} = \text{h3}
      // since The expansion through the valve is a
      throttling process
18
19 // Part (a)
20 \text{ Wcdot} = \text{mdot}*(h2s-h1)
      // The compressor work input in KW
21 printf ('The compressor power, in kW, is: %.2 f',
      Wcdot)
22
23 // Part(b)
24 \ Qindot = mdot*(h1-h4)*60/211
      // refrigeration capacity in ton
25 printf ( 'The refrigeration capacity in tons is: \%
      .2 f', Qindot)
26
27 // Part(c)
28 \text{ beta1} = (h1-h4)/(h2s-h1)
29 printf ('The coefficient of performance is: %.2f',
```

```
beta1)
30
31 // Part(d)
32 betamax = Tc/(Th-Tc)
33 printf( 'The coefficient of performance of a Carnot
    refrigeration cycle operating between warm and
    cold regions at 26 and 0C, respectively is: %.2f
    ',betamax);
```

### Scilab code Exa 10.2 Example

```
1 // Given:-
2 \text{ mdot} = 0.08
      // mass flow rate in kg/s
3 // Analysis
4 // At the inlet to the compressor, the refrigerant
     is a saturated vapor at 10C, so from Table A-10,
5 h1 = 241.35
      // in kj/kg
6 	 s1 = .9253
      // in kj/kg.k
7 // Interpolating in Table A-12 gives
8 \text{ h2s} = 272.39
      // in kj/kg.k
9 // State 3 is a saturated liquid at 9 bar, so
10 \text{ h3} = 99.56
      // in kj/kg
11 \text{ h4} = \text{h3}
      // since The expansion through the valve is a
```

```
throttling process
12
13 // Part (a)
14 \text{ Wcdot} = \text{mdot}*(h2s-h1)
      // The compressor power input in KW
15 // Result
16 printf(' \nThe compressor power in kw is: \%.2 \,\mathrm{f}',
      Wcdot)
17
18 // Part (b)
19 Qindot = mdot*(h1-h4)*60/211
      // refrigeration capacity in tons
20 // Result
21 printf( '\nThe refrigeration capacity in tons is:
      \%.2 f',Qindot)
22
23 // Part(c)
24 \text{ beta1} = (h1-h4)/(h2s-h1)
25 // Result
26 printf( '\nThe coefficient of performance is: \%.2\,\mathrm{f}
      ', beta1)
```

#### Scilab code Exa 10.3 Example

```
1 // Given:-
2 Tnot = 299

    //in kelvin
3 etac = .8

    //compressor efficiency of 80 percent
4 mdot = .08
```

```
//mass flow rate in kg/s
5 //analysis
6 //State 1 is the same as in Example 10.2, so
7 h1 = 241.35
      //in kj/kg
8 \text{ s1} = .9253
      //in kj/kg.k
9 //from example 10.2
10 \text{ h2s} = 272.39
      //in kj/kg
11 h2 = (h2s-h1)/etac + h1
                                                           //in
      kj/kg
12 //Interpolating in Table A-12,
13 \text{ s2} = .9497
      //in kj/kg.k
14 h3 = 91.49
      //in kj/kg
15 \text{ s3} = .3396
16 \text{ h4} = \text{h3}
      //since The expansion through the valve is a
      throttling process
17 //from data table
18 \text{ hf4} = 36.97
     //in kj/kg
19 \text{ hg4} = 241.36
      //in kj/kg
20 \text{ sf4} = .1486
      //in kj/kg.k
```

```
21 \text{ sg4} = .9253
      //in kj/kg.k
22 	 x4 = (h4-hf4)/(hg4-hf4)
                                                           //
       quality at state 4
23 	 s4 = sf4 + x4*(sg4-sf4)
                                                           //
       specific entropy at state 4 in kj/kg.k
24
25 // part (a)
26 \text{ Wcdot} = \text{mdot}*(h2-h1)
                                                                //
      compressor power in kw
27 printf ('The compressor power in kw is: \%.2 \,\mathrm{f} kW',
      Wcdot)
28
29 //part(b)
30 Qindot = mdot*(h1-h4)*60/211
                                                      //
       refrigeration capacity in ton
31 printf ('The refrigeration capacity in ton is: %.2 f
      ton', Qindot)
32
33 //part(c)
34 \text{ beta} = (h1-h4)/(h2-h1)
                                                              //
       coefficient of performance
35 printf ('The coefficient of performance is: \%.2\,\mathrm{f}',
      beta)
36
37 // part (d)
38 \quad \text{Eddotc} = \text{mdot}*\text{Tnot}*(s2-s1)
                                                          //in kw
39 \quad \text{Eddotv} = \text{mdot}*\text{Tnot}*(s4-s3)
                                                          //in kw
40 printf ('The rate of exergy destruction within the
       compressor is: %.2 f kW', Eddotc)
```

41 printf ('The rate of exergy destruction within the valve is: %.2f kw', Eddotv)

### Scilab code Exa 10.4 Example

```
1 // Given:-
2 p1 = 1.00
    // in bar
3 T1 = 270.00
     // in kelvin
4 \text{ AV} = 1.4
    // in m^3/s
5 r = 3.00
      // compressor pressure ratio
6 T3 = 300.00
      // turbine inlet temperature in kelvin
8 // Analysis
9 // From Table A-22,
10 \text{ h1} = 270.11
      // in kj/kg
11 \text{ pr1} = 0.9590
12 // Interpolating in Table A-22,
13 \text{ h2s} = 370.1
     // in kj/kg
14 // From Table A-22,
15 \text{ h3} = 300.19
```

```
// in kj/kg
16 \text{ pr3} = 1.3860
17 // Interpolating in Table A-22, we obtain
18 \text{ h4s} = 219.00
      // in kj/kg
19 // Calculations
20 \text{ pr2} = r*pr1
21 \text{ pr4} = \text{pr3/r}
23 // Part(a)
24 R = 8.314
      // universal gas constant, in SI units
25 M = 28.97
      // molar mass of air in grams
26
27 // Results
28 \text{ mdot} = (AV*p1)/((R/M)*T1)*10**2
      // mass flow rate in kg/s
29 Wcycledot = mdot*((h2s-h1)-(h3-h4s))
30 printf ('The net power input in kw is: %.2f',
      Wcycledot)
31
32 // Part (b)
33 Qindot = mdot*(h1-h4s)
      // refrigeration capacity in kw
34 printf ( ' The refregeration capacity in kw is: \%.2\,\mathrm{f}
      ',Qindot)
35
36 // Part(c)
37 beta = Qindot/Wcycledot
      // coefficient of performance
38 printf ('The coefficient of performance is: %.2f',
```

# Scilab code Exa 10.5 Example

```
1 // Given:-
2 // Part(a)
3 \text{ wcdots} = 99.99
      // work per unit mass for the isentropic
      compression determined with data from the
      solution in Example 10.4 in kj/kg
4 \text{ mdot} = 1.807
     // mass flow rate in kg/s from 10.4
5 \text{ etac} = 0.8
     // isentropic efficiency of compressor
6 Wcdot = (mdot*wcdots)/etac
      // The power input to the compressor in kw
8 // Using data form the solution to Example 10.4
      gives
9 wtdots =81.19
     // in kj/kg
10 \text{ etat} = 0.8
     // isentropic efficiency of turbine
11 // Calculations
12 Wtdot = mdot*etat*wtdots
     // actual turbine work in kw
13 Wdotcycle = Wcdot-Wtdot
```

```
// The net power input to the cycle in kw
14 // Result
15 printf( 'The net power input in kw is: \%.2\,\mathrm{f}',
      Wdotcycle)
16
17 // Part (b)
18 \text{ h3} = 300.19
      // in kj/kg
19 // From table A-22
20 \text{ h1} = 270.11
      // in kj/kg
21 // Calculations
22 \text{ h4} = \text{h3} - \text{Wtdot/mdot}
23 Qindot = mdot*(h1-h4)
      // refrigeration capacity in kw
24 // Result
25 printf ('The refrigeration capacity in kw is: %.2f
      ',Qindot)
26
27 // Part(c)
28 beta = Qindot/Wdotcycle
      // coefficient of performance
29 // Result
30 printf ('The coefficient of performance is: %.2f',
      beta)
```

# Chapter 11

# Thermodynamic Relations

#### Scilab code Exa 11.1 Example

```
// volume occupied by the gas in m<sup>3</sup>
12 vbar = M*(V/m)
      // The molar specific volume in m<sup>3</sup>/kmol
13
14 // Part (a)
15 // From Table A-1 for CO
16 \text{ Tc} = 133
      // in kelvin
17 \text{ Pc} = 35
      // in bar
18 \text{ Tr} = T/Tc
      // reduced temperature
19 \text{ Rbar} = 8314
      // universal gas constant in N.m/kmol.K
20 \ Z = 0.9
21 // Calculations
22 \text{ vrdash} = (\text{vbar}*\text{Pc}*10**5)/(\text{Rbar}*\text{Tc})
                                                            //
      pseudoreduced specific volume
23 p = (Z*Rbar*T/vbar)*10**-5
      // in bar
24 // Result
25 printf( '\n part(a) the pressure in bar is: %.2f bar
      ',p)
26
27 // Part(b)
28 // The ideal gas equation of state gives
29 // Calculations
30 p = (Rbar*T/vbar)/10**5
      // in bar
```

```
31 // Result
32 printf('\n Part(b) the pressure in bar is: %.2f bar'
      ,p)
33
34 // Part(c)
35 // For carbon monoxide, the van der Waals constants
     a and b can be read directly from Table A-24
36 \quad a = 1.474
     // in (m^3/kmol)^2
37 b = 0.0395
     // in m<sup>3</sup>/kmol
38 // Calculations
39 p = (Rbar*T/(vbar-b))/10**5 - a/vbar**2
40 // Result
41 printf( '\n Part(c) the pressure in bars is:
     bar',p)
42
43 // Part (d)
44 // For carbon monoxide, the Redlich Kwong
      constants can be read directly from Table A-24
45 \quad a = 17.22
     // in m^6*K^.5/kmol^2
46 b = 0.02737
     // in m^3/kmol
47 // Calculations
48 p = (Rbar*T/(vbar-b))/10**5 - a/(vbar*(vbar+b)*T
     **.5)
49 // Result
50 printf('\n Part(d)the pressure in bar is: %.2f bar
     ', p)
```

#### Scilab code Exa 11.3 Example

```
1 // Given:-
2 // Part(a)
3 v = 0.4646
      // specific volume in in m<sup>3</sup>/kg
4 M = 18.02
      // molar mass of water in kg/kmol
_{5} // At the specified state, the temperature is _{513} K
      and the specific volume on a molar basis is
6 \text{ vbar} = \text{v}*\text{M}
     // in m^3/kmol
7 // From Table A-24
8 a = 142.59
     // (m^3/kmol)^2 * K^.5
9 b = 0.0211
      // in m<sup>3</sup>/kmol
10
11 \text{ Rbar} = 8314.0
      // universal gas constant in N.m/kmol.K
12 T = 513.0
      // in kelvin
13 delpbydelT = (Rbar/(vbar-b) + a/(2*vbar*(vbar+b)*T
      **1.5)*10**5)/10**3
                                          // in kj/(m^3*K)
14
15 // By The Maxwell relation
16 delsbydelv = delpbydelT
17 // Result
18 printf ('The value of delpbydelT in kj/(m^3*K) is:
        \%.2 f', delpbydelT);
19
```

```
20 // Part (b)
21 // A value for (dels/delv)T can be estimated using a
       graphical approach with steam table data, as
      follows: At 240C, Table A-4 provides the values
      for specific entropy s and specific volume v
      tabulated below
22 T = 240.0
      // in degree celcius
23 \ // \ At \ p \ = 1, \ 1.5 \, , \ 3 \, , \ 5 \, , \ 7 \, , \ 10 \ bar \ respectively
24 \text{ y} = [7.994, 7.805, 7.477, 7.230, 7.064, 6.882]
25 \times = [2.359, 1.570, 0.781, 0.4646, 0.3292, 0.2275]
26 plot(x,y)
27 xlabel ("Specific volume")
28 ylabel ("Specific entropy")
29
30 // The pressure at the desired state is 5 bar. The
      corresponding slope is
31 delsbydelv = 1
      // in kj/m^3.K
32 printf ('From the data of the table, delsbydely = \%
      .2 f', delsbydelv);
```

#### Scilab code Exa 11.4 Example

```
6 \text{ sgf} = 6.048
      // in kj/kg.K
7 // Values
8 printf ('From table, hg-hf = \%.2 \,\mathrm{f}', hgf);
9 printf( 'From table, ug-uf = \%.2 \, \text{f',ugf});
10 printf( 'From table, sg-sf = \%.2f', sgf);
11
12 // Part (a)
13 T = 373.15
      // in kelvin
14 // If we plot a graph between temperature and
      saturation pressure using saturation
      pressure temperature data from the steam tables
      , the desired slope is:
15 delpbydelT = 3570.00
      // in N/(m^2.K)
16 \text{ vg} = 1.673
      // in m^3/kg
17 \text{ vf} = 1.0435e-3
      // in m^3/kg
18 // Calculations
19 // From the Clapeyron equation
20 hgf = T*(vg-vf)*delpbydelT*10**-3
                                                           //
      in kj/kg
21 // Result
22 printf ('\n Part(a) using Clapeyron equation, hg-hf =
       \%.2 f KJ/kg', hgf);
23
24 // Part (b)
25 \text{ psat} = 1.014e5
      // in N/m<sup>2</sup>
```

### Scilab code Exa 11.6 Example

```
1 // Given:-
2 // Part(a)
3 v = 1.00/998.21

    // specific volume of water in m^3/kg
4 T = 293.00

    // given temperature in kelvin
5 beta = 206.6e-6

    // volume expansivity in /K
6 k = 45.90e-6

    // isothermal compressibility in /bar
7 // Interpolating in Table A-19
8 cp = 4.188
```

```
// in kj/kg.k
9 // Calculations
10 cpv = (v*T*beta**2.00/k)*10**2
                                                         //
      in kj/kg.k
11 \text{ cv} = \text{cp-cpv}
      // in kj/kg.k
12 errorPercentage = 100*(cp-cv)/cv
13 // Result
14 printf(' The percentage error is: %.2f',
      errorPercentage)
15
16 // Part(b)
17 // Calculations
18 \text{ K} = \text{cp/cv}
     // specific heat ratio
19 c = ((K*v/k)*10**5)**0.5
      // velocity of sound in m/s
20 // Result
21 printf( 'The velocity of sound is: \%.2 \,\mathrm{f} m/s',c)
```

# Scilab code Exa 11.8 Example

```
// in bar
5 T2 = 245.00
      // in kelvin
7
8 // From table A-23
9 \text{ h1starbar} = 8723.00
      // in kj/kmol
10 \text{ h2starbar} = 7121.00
      // in kj/kmol
11 // From Tables A-1
12 \text{ Tc} = 126.00
      // critical temperature in kelvin
13 \text{ pc} = 33.9
     // critical pressure in bar
14 M = 28.00
     // molar mass in kg/kmol
15 \text{ Rbar} = 8.314
      // universal gas constant in kj/(kmol.K)
16 \text{ Term1} = 0.5
17 \text{ Term2} = 0.31
18
19 // Calculations
20 \text{ TR1} = \text{T1/Tc}
      // reduced temperature at the inlet
21 PR1 = p1/pc
      // reduced pressure at the inlet
22 \text{ TR2} = \text{T2/Tc}
```

```
// reduced temperature at the exit
23 PR2 = p2/pc

// reduced pressure at the exit
24 wcvdot = (1.00/M)*(h1starbar-h2starbar-Rbar*Tc*(
    Term1-Term2)) // in kj/kg
25
26 // Result
27 printf(' The work developed, in kJ per kg of nitrogen flowing is: %.2f',wcvdot)
```

# Scilab code Exa 11.9 Example

```
// in bar
11 \text{ Term1} = 0.21
12 \text{ Term2} = 0.14
13
14 // Calculations
15
16 S2StarBarMinusS1StarBar = sT2bar-sT1bar-Rbar*log(p2/
                                 // The change in specific
      p1)
      entropy in kj/(kmol.K)
17 sigmacvdot = (1.00/M)*(S2StarBarMinusS1StarBar-Rbar
      *(Term2-Term1))
18 // Result
19 printf (' the rate of entropy production in kj/kg.K
      is: \%.2 \, f', sigmacvdot)
20
21 // Part(b)
\frac{22}{\sqrt{\text{From Table A}-23}}
23 \text{ h2starbar} = 6654.00
      // in kj/kmol
24 \text{ h1starbar} = 8723.00
      // in kj/kmol
25 \text{ Tc} = 126.00
      // critical temperature in kelvin
26 \text{ Term2} = 0.36
27 \text{ Term1} = 0.5
28 \text{ wcvdot} = 50.1
      // from example 11.8
29
30 // Calculations
31 wcvdots = (1.00/M)*(h1starbar-h2starbar-Rbar*Tc*(
                                          // isentropic work
      Term1-Term2))
       in kj/kg
32 etat = wcvdot/wcvdots
```

```
// turbine efficiency
33
34 // Result
35 printf( 'The isentropic turbine efficiency is: %.2f
    ', etat)
```

### Scilab code Exa 11.10 Example

```
2 // Given:-
3 // Analysis
4 V = 0.241
     // volume of the mixture in m<sup>3</sup>
5 T = 511.00
     // temperature of the mixture in kelvin
6 \text{ n1} = 0.18
      // number of moles of methane in kmol
7 n2 = 0.274
      // number of moles of butane in kmol
8 \text{ Rbar} = 8314
      // universal gas constant in (N.m)/(kmol.K)
10 // Calculations
11 \quad n = n1 + n2
     // The total number of moles of mixture
12 y1 = n1/n
     // mole fraction of methane
13 y2 = n2/n
```

```
// mole fraction of butane
14 vbar = V/(n)
      // The specific volume of the mixture on a molar
      basis in m<sup>3</sup>/kmol
15
16 // Part (a)
17 p = (Rbar*T/vbar)*10**-5
      // in bar
18 // Result
19 printf (' The pressure in bar obtained using ideal
      gas equation is: %.2 f',p)
20
21 // Part(b)
22 // From table A-1
23 \text{ Tc1} = 191.00
      // critical temperature for methane in kelvin
24 \text{ Pc1} = 46.4
      // critical pressure for methane in bar
25 \text{ Tc2} = 425.00
      // critical temperature for butane in kelvin
26 \text{ Pc2} = 38.00
      // critical pressure for butane in bar
27 Z = 0.88
28
29
30 // Calculations
31 \text{ Tc} = y1*\text{Tc}1 + y2*\text{Tc}2
      // critical temperature in kelvin
32 \text{ Pc} = y1*Pc1 + y2*Pc2
```

```
// critical pressure in bar
33 \text{ TR} = T/Tc
      // reduced temperature of the mixture
34 vRdash= vbar*Pc/(Rbar*Tc)
35 p = ((Z*Rbar*T)/vbar)*10**-5
      // mixture pressure in bar
36 // Result
37 printf (' Pressure obtained using Kay s rule
      together with the generalized compressibility
      chart, is: \%.2 f, p)
38
39 // Part(c)
40 // Table A-24 gives the following van der Waals
      constants values for methane
41 \text{ a} 1 = 2.293
      // in (m^3/kmol)^2
42 \text{ b1} = 0.0428
      // in m<sup>3</sup>/kmol
43 // Table A-24 gives the following van der Waals
      constants values for butane
44 \text{ a} 2 = 13.86
      // in (m^3/kmol)^2
45 b2 = 0.1162
      // in m^3/kmol
47 a = (y1*a1**.5 + y2*a2**.5)**2
      // in bar * (m^3/kmol)^2
48 \ b = y1*b1+y2*b2
      // in m^3/kmol
49 // From van der Waals equation
```

```
50 p = ((Rbar*T)/(vbar-b))*10**-5 - a/(vbar**2)
51 printf ('The pressure in bar from van der Waals
      equation is: \%.2 f, p)
52
53 // Part (d)
54 // For methane
55 \text{ TR1} = \text{T/Tc1}
56 \text{ vR1dash} = (.241/.18)*10**5*Pc1/(Rbar*Tc1)
57 	 Z1 = 1.00
58 // For butane
59 \text{ TR2} = \text{T/Tc2}
60 \text{ vR2dash} = (.88*10**5*Pc2)/(Rbar*Tc2)
61 \quad Z2 = 0.8
62 Z = y1*Z1 + y2*Z2
63 // Accordingly, the same value for pressure as
      determined in part (b) using Kay s rule results
64 p = 70.4
65
66 // Result
67 printf ('The pressure in bar obtained using the
      rule of additive pressures employing the
      generalized compressibility chart is: %.2f',p)
```

# Chapter 12

# Ideal Gas Mixtures and Psychrometrics Applications

# Scilab code Exa 12.1 Example

```
// molar mass of H2O in kg/kmol
10 M3 = 32.0
     // molar mass of O2 in kg/kmol
11 \quad M4 = 28.0
      // molar mass of N2 in kg/kmol
12
13 // Calculations
14 M = M1*n1 + M2*n2 + M3*n3 + M4*n4
                                                         //
       in kg/kmol
15 // Result
16 printf ('The apparent molecular weight of the
      mixture in kg/kmol is: %f',M)
17
18 // Part (b)
19 mf1 = (M1*n1/M)*100.0
     // mass fraction of CO2 in percentage
20 \text{ mf2} = (M2*n2/M)*100.0
     // mass fraction of H2O in percentage
21 \text{ mf3} = (M3*n3/M)*100.0
      // mass fraction of O2 in percentage
22 \text{ mf4} = (M4*n4/M)*100.0
      // mass fraction of N2 in percentage
23
24 // Results
25 printf ('The mass fraction of CO2 in percentage is:
       %f', mf1)
26 printf ('The mass fraction of H2O in percentage is:
       %f',mf2)
27 printf ('The mass fraction of O2 in percentage is:
      %f', mf3)
```

28 printf ('The mass fraction of N2 in percentage is: %f',mf4)

### Scilab code Exa 12.2 Example

```
1 // Given:-
2 \text{ mf1} = 0.1
      // mass fractiion of H2
3 \text{ mf2} = 0.6
      // mass fraction of N2
4 \text{ mf3} = 0.3
      // mass fraction of CO2
6 // Part(a)
7 M1 = 2.0
      // molar mass of H2 in kg/kmol
8 M2 = 28.0
      // molar mass of N2 in kg/kmol
9 M3 = 44.0
      // molar mass of CO2 in kg/kmol
10
11 // Calculations
12 \text{ n1} = (\text{mf1/M1})/(\text{mf1/M1} + \text{mf2/M2} + \text{mf3/M3})
                                                           // mole
       fraction of H2
13 \text{ n2} = (\text{mf2/M2})/(\text{mf1/M1} + \text{mf2/M2} + \text{mf3/M3})
                                                           // mole
       fraction of N2
14 \text{ n3} = (mf3/M3)/(mf1/M1 + mf2/M2 + mf3/M3)
```

```
// mole
     fraction of CO2
15
16 // Results
17 printf ('The mole fraction of H2 in percentage is:
     %f',n1*100)
18 printf ('The mole fraction of N2 in percentage is:
     %f',n2*100)
19 printf ('The mole fraction of CO2 in percentage is:
      %f',n3*100)
20
21 // Part(b)
22 // Calculation
23 M = n1*M1 + n2*M2 + n3*M3
     // in kg/kmol
24 // Result
25 printf ('The apparent molecular weight of the
     mixture in kg/kmol is: %f',M);
```

# Scilab code Exa 12.3 Example

```
6 p2 = 3.0
     // in bar
7 n = 1.25
9 // Part(a)
10 // Calculation
11 T2 = T1*(p2/p1)**((n-1)/n)
     // in kelvin
12 // Result
13 printf ('The final temperature in Kelvin is: %f',T2
14
15 // Part (b)
16 \text{ Rbar} = 8.314
     // universal gas constant in SI units
17 // Calculations
18 M = (m1+m2)/(m1/44 + m2/28)
      // molar mass of mixture in kg/kmol
19 W = ((m1+m2)*(Rbar/M)*(T2-T1))/(1-n)
                                                     // in
       kј
20 // Result
21 printf( 'The work in kj is: \%f',W )
22
23 // Part(c)
24 // From table A-23
25 \text{ uCO2T1} = 6939.0
      // internal energy of CO2 on molar mass basis at
      temperature T1
26 \text{ uCO2T2} = 9198.0
      // internal energy of CO2 on molar mass basis at
      temperature T2
```

```
27 \text{ uN2T1} = 6229.0
      // internal energy of N2 on molar mass basis at
      temperature T1
28 \text{ uN2T2} = 7770.0
      // internal energy of N2 on molar mass basis at
      temperature T2
29 deltaU = (m1/44)*(uCO2T2-uCO2T1) + (m2/28)*(uN2T2-uCO2T1)
                                         // internal energy
       change of the mixture in KJ
30
31 // With assumption, The changes in kinetic and
      potential energy between the initial and final
      states can be ignored
32 Q = deltaU + W
33 // Result
34 printf ('The heat transfer in kj is: %f',Q);
35
36 // Part (d)
37 // From table A-23
38 \text{ sbarT2C02} = 222.475
39 \text{ sbarT1C02} = 213.915
40 \text{ sbarT2N2} = 198.105
41 \text{ sbarT1N2} = 191.682
42 \text{ Rbar} = 8.314
      // universal gas constant
43 // Calculation
44 \text{ deltaS} = (m1/44)*(sbarT2CO2-sbarT1CO2-Rbar*log(p2/p1)
      )) + (m2/28)*(sbarT2N2-sbarT1N2-Rbar*log(p2/p1))
45 // Result
46 printf ('The change in entropy of the mixture in kj/
      k is: %f',deltaS)
```

#### Scilab code Exa 12.4 Example

```
1 // Given:-
2 y1 = 0.8
     // mole fraction of CO2
3 y2 = 0.2
    // mole fraction of O2
4 T1 = 700.0
     // in kelvin
5 p1 = 5.0
    // in bars
6 V1 = 3.0
    // in m/s
7 p2 = 1.0
     // in bars
8
9
10 // Part(a)
11 // From table A-23
12 \text{ sO2barT1} = 231.358
13 \text{ sCO2barT1} = 250.663
14 // Calculations
16 RHS = y2*s02barT1 + y1*sC02barT1 + 8.314*log(p2/p1)
17 // Using table A-23
18 LHSat510K = y2*221.206 + y1*235.7
19 LHSat520K = y2*221.812 + y1*236.575
20 // Using linear interpolation,
21 T2 = 510 + ((520-510)/(LHSat520K-LHSat510K))*(RHS-
     LHSat510K)
22 // Result
23 printf ('The temperature at the nozzle exit in K is:
```

```
%f', T2);
24
25 // Part (b)
26 // From table A-23
27 \text{ sbar02T2} = 221.667
      // in kj/kmol.K
28 \text{ sbar}02T1 = 231.358
      // in kj/kmol.K
29 \text{ sbarCO2T2} = 236.365
      // in kj/kmol.K
30 \text{ sbarCO2T1} = 250.663
      // in kj/kmol.K
31 // Calculations
32 \text{ deltasbar02} = \text{sbar02T2-sbar02T1-8.314*} \log(p2/p1)
                                    // in kj/kmol.K
33 deltasbarCO2 = sbarCO2T2-sbarCO2T1-8.314*log(p2/p1)
                                // in kj/kmol.K
34 // Results
35 printf ('The entropy changes of the CO2 from inlet
      to exit, in KJ/Kmol.K is: %f',deltasbarCO2)
36 printf ('The entropy change of the O2 from inlet to
      the exit in kj/kmol.k is: %f',deltasbar02)
37
38 // Part(c)
39 // From table A-23, the molar specific enthalpies of
       O2 and CO2 are
40 \text{ h1bar02} = 21184.0
41 \text{ h}2\text{bar}02 = 15320.0
42 \text{ h1barCO2} = 27125.0
43 \text{ h2barCO2} = 18468.0
44 // Calculations
45 M = y1*44.0 + y2*32.0
      // apparent molecular weight of the mixture in kg
```

# Scilab code Exa 12.5 Example

```
1 // Given:-
2 \text{ nN2} = 0.79
      // initial moles of nitrogen in kmol
3 pN2 = 2.0
      // initial pressure of nitrogen in bars
4 \text{ TN2} = 250.0
      // initial temperature of nitrogen in kelvin
5 \text{ nO2} = 0.21
     // initial moles of oxygen in kmol
6 p02 = 1.0
     // initial pressure of oxygen in bars
7 \text{ TO2} = 300.0
      // initial temperature of oxygen in kelvin
9 // Part(a)
10 \text{ MN2} = 28.01
     // molar mass of nitrogen in kg/kmol
11 \text{ MO2} = 32.0
```

```
// molar mass of oxygen in kg/kmol
12 // Calculations
13 // With the help of table A-20
14 \text{ cvbarN2} = MN2*0.743
      // in kj/kmol.K
15 \text{ cvbarO2} = MO2*0.656
      // in kj/kmol.K
16 \quad T2 = (nN2*cvbarN2*TN2+nO2*cvbarO2*TO2)/(nN2*cvbarN2+
      n02*cvbar02)
17 // Result
18 printf ('The final temperature of the mixture in
      kelvin is: \%f',T2);
19
20 // Part(b)
21 // Calculation
22 p2 = ((nN2+nO2)*T2)/(nN2*TN2/pN2 + nO2*TO2/pO2)
23 // Result
24 printf ('The final pressure of the mixture in bar is
      : %f',p2);
25
26 // Part(c)
27 \text{ Rbar} = 8.314
      // universal gas constant
28 // Calculations
29 cpbarN2 = cvbarN2 + Rbar
30 \text{ cpbar02} = \text{cvbar02} + \text{Rbar}
31 \quad yN2 = nN2/(nN2+nO2)
      // mole fraction of N2
32 \text{ yO2} = \text{nO2/(nN2+nO2)}
      // mole fraction of O2
33 sigma = nN2*(cpbarN2*log(T2/TN2)-Rbar*log(yN2*p2/pN2
      )) + n02*(cpbar02*log(T2/T02)-Rbar*log(y02*p2/p02)
      ))
```

```
34 // Result
35 printf( 'The amount of entropy produced in the
      mixing process, in kJ/K is: %f',sigma);
```

# Scilab code Exa 12.6 Example

```
1 // Given:-
2 T1 = 32.0
     // temperature of dry air in degree celcius
3 p1 = 1.0
     // pressure of dry air in bar
4 \text{ AV1} = 100.0
     // volume rate of dry air in m^3/min
5 T2 = 127.0
     // temperature of oxygen stream in degree celcius
6 p2 = 1.0
    // pressure of oxygen stream in bar
7 T3 = 47.0
    // temperature of mixed stream in degree celcius
8 p3 = 1.0
      // pressure of mixed stream in bar
10 // Part(a)
11 \text{ Rbar} = 8314.0
     // universal gas constant
12 \text{ Ma} = 28.97
```

```
// molar mass of air
13 \text{ Mo} = 32.0
      // molar mass of oxygen
14 // From table A-22 and A-23
15 \text{ haT3} = 320.29
      // in kj/kg
16 \text{ haT1} = 305.22
      // in kj/kg
17 \text{ hnotT2} = 11711.0
      // in kj/kmol
18 \text{ hnotT1} = 9325.0
      // in kj/kmol
19
20 // Calculations
21 \text{ val} = (Rbar/Ma)*(T1+273.0)/(p1*10**5)
                                                         //
      specific volume of air in m<sup>3</sup>/kg
22 \text{ maldot} = AV1/va1
      // mass flow rate of dry air in kg/min
23 modot = ma1dot*(haT3-haT1)/((1/Mo)*(hnotT2-hnotT1))
                                         // in kg/min
24 // Results
25 printf ('The mass flow rate of dry air in kg/min is:
        %f', maldot);
26 printf ('The mass flow rate of oxygen in kg/min is:
       %f', modot);
27
28 // Part(b)
29 nadot = maldot/Ma
      // molar flow rate of air in kmol/min
30 \mod ot = \mod ot/Mo
```

```
// molar flow rate of oxygen in kmol/min
31 ya = nadot/(nadot+nodot)
     // mole fraction of air
32 yo = nodot/(nadot+nodot)
      // mole fraction of oxygen
33 // Results
34 printf ('The mole fraction of dry air in the exiting
       mixture is: %f',ya)
35 printf ('The mole fraction of dry oxygen in the
      exiting mixture is: %f', yo)
36
37 // Part(c)
38 // With the help of tables A-22 and A-23
39 \text{ sanotT3} = 1.7669
     // in kj/kg.K
40 \text{ sanotT1} = 1.71865
     // in kj/kg.K
41 \text{ sbarT3} = 207.112
     // in kj/kmol.K
42 \text{ sbarT2} = 213.765
     // in kj/kmol.K
43 // Calculations
44 sigmadot = ma1dot*(sanotT3-sanotT1-(8.314/Ma)*log(ya
      ))+ (modot/Mo)*(sbarT3-sbarT2-8.314*log(yo))
45 // Result
46 printf ('The time rate of entropy production, in kJ/
     K \cdot \min is : \%f', sigmadot)
```

#### Scilab code Exa 12.7 Example

```
1 // Given:-
2 m = 1.0
     // mass of sample in kg
3 T1 = 21.0
     // initial temperature in degree celcius
4 \text{ psi1} = 0.7
     // initial relative humidity
5 T2 = 5.0
     // final temperature in degree celcius
7 // Part(a)
8 // From table A-2
9 \text{ pg} = 0.02487
     // in bar
10 // Calculations
11 \text{ pv1} = \text{psi1*pg}
     // partial pressure of water vapor in bar
12 omega1 = 0.622*(0.2542)/(14.7-0.2542)
13 // Result
14 printf ('the initial humidity ratio is: %f', omega1)
16 // Part (b)
17 // The dew point temperature is the saturation
     temperature corresponding to the partial pressure
     , pv1. Interpolation in Table A-2 gives
18 T = 15.3
     // the dew point temperature in degree celcius
19 // Result
20 printf ('The dew point temperature in degree celcius
```

```
is: %f',T)
21
22 // Part(c)
23 // The partial pressure of the water vapor remaining
       in the system at the final state is the
      saturation pressure corresponding to 5C:
24 // Calculations
25 \text{ mv1} = 1/((1/\text{omega1})+1)
      // initial amount of water vapor in the sample in
       kg
26 \text{ ma} = \text{m-mv1}
      // mass of dry air present in kg
27 \text{ pg} = 0.00872
      // in bar
28 \text{ omega2} = 0.622*(pg)/(1.01325-pg)
      // humidity ratio after cooling
29 \text{ mv2} = \text{omega2*ma}
      // The mass of the water vapor present at the
      final state
30 \quad mw = mv1 - mv2
31
32 // Result
33 printf ('The amount of water vapor that condenses,
      in kg. is: %f', mw)
```

#### Scilab code Exa 12.8 Example

```
1
2 // Given:-
3 V = 35.0
```

```
// volume of the vessel in m<sup>3</sup>
4 p1 = 1.5
     // in bar
5 T1 = 120.0
      // in degree celcius
6 \text{ psi1} = 0.1
7 T2 = 22.0
      // in degree celcius
8
9 // Part(a)
10 // The dew point temperature at the initial state is
      the saturation temperature corresponding to the
      partial pressure pv1. With the given relative
      humidity and the saturation pressure at 120C from
       Table A-2
11 \text{ pg1} = 1.985
12 // Interpolating in Table A-2 gives the dew point
      temperature as
13 T = 60.0
     // in degree celcius
14 // Calculation
15 \text{ pv1} = \text{psi1*pg1}
      // partial pressure in bar
16 // Result
17 printf ('The dew point temperature corresponding to
      the initial state, in degee celcius is: %f',T)
18
19 // Part(b)
20 \text{ Rbar} = 8314.0
      // universal gas constant
21 \text{ My} = 18.0
```

```
// molar mass of vapor in kj/kmol
22 // Interpolation in Table A-2
23 \text{ Tdash} = 56.0
      // in degrees
24 \text{ vv1} = ((Rbar/Mv)*(T1+273))/(pv1*10**5)
      the specific volume of the vapor at state 1 in m
       ^3/kg
25 // Result
26 printf ('The temperature at which condensation
      actually begins in degree celcius is: %f', Tdash)
27
28 // Part(c)
29 // From table
30 \text{ vf2} = 1.0022e-3
31 \text{ vg2} = 51.447
32 \text{ vv2} = \text{vv1}
      // specific volume at final state
33 // Calculations
34 \text{ mv1} = \text{V/vv1}
      // initial amount of water vapor present in kg
35 	ext{ x2} = (vv2-vf2)/(vg2-vf2)
      // quality
36 \text{ mv2} = \text{x2*mv1}
      // the mass of the water vapor contained in the
      system at the final state
37 \quad mw2 = mv1 - mv2
38 // Result
39 printf ('The amount of water condense in kg is:
      , mw2)
```

# Scilab code Exa 12.9 Example

```
2 // Given:-
3 V = 35.0
   // volume of vessel in m^3
4 p1 = 1.5
   // initial pressure in bar
5 T1 = 120.0
     // initial temperature in degree celcius
6 \text{ psi} = 0.1
7 T2 = 22.0
    // in degree celcius
8 \text{ Rbar} = 8314.0
    // universal gas constant
9 \text{ Ma} = 28.97
     // molar mass of air
10 \text{ pv1} = 0.1985
   // in bar, from example 12.8
11 \text{ mv2} = 0.681
   // in kg, from examples 12.8
12 \text{ mv1} = 3.827
    // in kg, from example 12.8
13 \text{ mw2} = 3.146
```

```
// in kg, from example 12.8
14 // evaluating internal energies of dry air and water
      from Tables A-22 and A-2, respectively
15 \text{ ua2} = 210.49
      // in kj/kg
16 \text{ ua1} = 281.1
    // in kj/kg
17 \text{ ug2} = 2405.7
     // in kj/kg
18 \text{ uf2} = 92.32
     // in kj/kg
19 \text{ ug1} = 2529.3
     // in kj/kg
20
21 // Calculations
22 ma = ((p1-pv1)*10**5)*V)/((Rbar/Ma)*(T1+273))
                                          // mass of dry
      air in kg
Q = ma*(ua2-ua1) + mv2*ug2 + mw2*uf2 - mv1*ug1
24
25 // Result
26 printf ('The heat transfer during the process, in kJ
            %f',Q)
       is:
```

#### Scilab code Exa 12.10 Example

```
1
2 // Given :-
3 AV1 = 150.0
```

```
// entry volumetric flow rate in m^3/min
4 T1 = 10.0
      // entry temperature in degree celcius
5 \text{ psi1} = 0.8
6 T2 = 30.0
      // exit temperature in degree celcius
7 p = 1.0
      // in bar
9 // Part(a)
10 \text{ Rbar} = 8314.0
      // universal gas constant
11 \text{ Ma} = 28.97
      // molar mass of air
12 // The specific enthalpies of the dry air are
      obtained from Table A-22 at the inlet and exit
      temperatures T1 and T2, respectively:
13 \text{ ha1} = 283.1
      // in kj/kg
14 \text{ ha2} = 303.2
      // in kj/kg
15 // The specific enthalpies of the water vapor are
      found using hv hg and data from Table A-2 at T1
      and T2, respectively:
16 \text{ hv1} = 2519.8
     // in kj/kg
17 \text{ hv2} = 2556.3
     // in kj/kg
18 // From table A-2
```

```
19 \text{ pg1} = 0.01228
      // in bar
20 // Calculations
21 \text{ pv1} = \text{psi1*pg1}
      // the partial pressure of the water vapor in bar
22 pa1 = p-pv1
23 va1 = (Rbar/Ma)*(T1+273)/(pa1*10**5)
                                                            //
       specific volume of the dry air in m<sup>3</sup>/kg
24 \text{ madot} = AV1/va1
      // mass flow rate of the dry air in kg/min
25 \text{ omega} = 0.622*(pv1/(p-pv1))
      // humidity ratio
26 \ Qcvdot = madot*((ha2-ha1)+omega*(hv2-hv1))
                                                     // in kj/
      min
27 // Result
28 printf ('Rate of heat transfer, in kJ/min is: \%.2 f'
       ,Qcvdot);
29
30 // Part(b)
31 // From Table A-2 at 30C
32 \text{ pg2} = 0.04246
      // in bar
33 // Calculations
34 \text{ pv2} = \text{pv1}
35 \text{ psi2} = \text{pv2/pg2}
      // relative humidity at the exit
36 // Result
37 printf ('The relative humidity at the exit is: \%.2 \,\mathrm{f}
      ',psi2);
```

# Scilab code Exa 12.11 Example

```
2 // Given:-
3 T1 = 30.0
   // in degree celcius
4 \text{ AV1} = 280.0
    // in m^3/min
5 \text{ psi1} = 0.5
     // relative humidity at the inlet
6 T2 = 10.0
     // in degree celcius
7 p = 1.013
     // pressure in bar
9 // Part(a)
10 // From table A-2
11 \text{ pg1} = 0.04246
   // in bar
12 \text{ Rbar} = 8314
    // universal gas constant
13 \text{ Ma} = 28.97
      // molar mass of air
14 // Calculations
15 \text{ pv1} = \text{psi1*pg1}
```

```
// in bar
16 pa1 = p-pv1
      // partial pressure of the dry air in bar
17 madot = AV1/((Rbar/Ma)*((T1+273)/(pa1*10**5)))
                                           // common mass
      flow rate of the dry air in kg/min
18 // Result
19 printf ('\n The mass flow rate of the dry air in kg/
      min is: \%.2 f, madot);
20
21 // Part (b)
22 // From table A-2
23 \text{ pv2} = 0.01228
      // in bar
24 // Calculations
25 \text{ omega1} = 0.622*(pv1/(p-pv1))
26 \text{ omega2} = 0.622*(pv2/(p-pv2))
27 mwdotbymadot = omega1-omega2
28 // Result
29 printf ('\n The rate at which water is condensed, in
       kg per kg of dry air flowing through the control
       volume is: %.4 f', mwdotbymadot);
30
31 // Part(c)
32 // From table A-2 and A-22
33 \text{ ha2} = 283.1
      // in kg/kj
34 \text{ ha1} = 303.2
      // in kg/kj
35 \text{ hg1} = 2556.3
      // in kg/kj
36 \text{ hg2} = 2519.8
```

# Scilab code Exa 12.12 Example

```
1
2 // Given:-
3 T1 = 22.0
     // entry temperature of moist air in degree
     celcius
4 \text{ Twb} = 9.0
     // wet-bulb temperature of entering moist air in
     degree celcius
5 \text{ madot} = 90.0
     // mass flow rate of dry air in kg/min
6 \text{ Tst} = 110.0
     // temperature of injected saturated water vapor
     in degree celcius
7 \text{ mstdot} = 52.0
     // mass flow rate of injected saturated water
     vapor in kg/h
8 p = 1.0
```

```
// pressure in bar
9
10 // Part(a)
11 // By inspection of the psychrometric chart
12 \text{ omega1} = 0.002
13 // Calculation
14 \text{ omega2} = \text{omega1} + \text{mstdot/(madot*60)}
15 // Result
16 printf ('The humidity ratio at the exit is: %.2f',
      omega2);
17
18 // Part (b)
19 // The steady-state form of the energy rate balance
      can be rearranged as
20 // (ha + omega*hg)2 = (ha + omega*hg)1 + (omega2-
      omega1)*hg3
21 // On putting values in the above equation from
      tables and figures, temperature at the exit can
      then be read directly from the chart
22 	ext{ T2} = 23.5
      // in degree celcius
23 // Result
24 printf ('The temperature at the exit in degree
      celcius is: %.2 f', T2)
```

#### Scilab code Exa 12.13 Example

```
1
2 // Given:-
3 T1 = 38.0

// temperature of entering air in degree celcius
4 psi1 = 0.1
```

```
// relative humidity of entering air
5 \text{ AV1} = 140.0
      // volumetric flow rate of entering air in m^3/
      min
6 \text{ Tw} = 21.0
      // temperature of added water in degree celcius
7 T2 = 21.0
      // temperature of exiting moist air in degree
      celcius
8 p = 1.0
      // pressure in atm
10 // Part(a)
11 // From table A-2
12 \text{ pg1} = 0.066
      // in bar
13 // The specific volume of the dry air can be
      evaluated from the ideal gas equation of state.
      The result is
14 \text{ val} = .887
      // in m^3/kg
15 \text{ cpa} = 1.005
16 // From table A-2
17 \text{ hf} = 88.14
18 \text{ hg1} = 2570.7
19 \text{ hg2} = 2539.94
20 // Calculations
21 \text{ pv1} = \text{psi1*pg1}
      // the partial pressure of the moist air entering
       the control volume in bar
```

```
20 \text{ omega1} = 0.622*(pv1/(p*1.01325-pv1))
23 omega2 = (cpa*(T1-T2)+omega1*(hg1-hf))/(hg2-hf)
24 \text{ madot} = AV1/va1
      // mass flow rate of the dry air in kg/min
25 mwdot = madot *60 * (omega2 - omega1)
      // in kg/h
26 // Result
27 printf ('\n The mass flow rate of the water to the
      soaked pad in is: \%.2 f kg(water)/h', mwdot);
28
29 // Part (b)
30 \text{ pv2} = (\text{omega2*p*1.01325})/(\text{omega2+0.622})
                                                       // in
      bars
31 // At 21C, the saturation pressure is
32 \text{ pg2} = 0.02487
33 \text{ psi2} = \text{pv2/pg2}
34 // Result
35 printf( '\n The relative humidity of the moist air
      at the exit to the evaporative cooler is: %.2f',
      psi2)
```

#### Scilab code Exa 12.14 Example

```
6 \text{ AV2} = 425.0
      // in m^3/min
7 T2 = 24.0
      // in degree celcius
8 \text{ psi2} = 0.5
9 p = 1.0
      // in bar
10
11
12 // Part (a)
13 // From the psychrometric chart, Fig. A-9.
14 \text{ val} = 0.79
      // in m^3/kg
15 \text{ va2} = 0.855
      // in m^3/kg
16 \text{ omega2} = 0.0094
17 // Calculations
18 \text{ maldot} = AV1/va1
      // in kg/min
19 \text{ ma2dot} = AV2 / va2
      // in kg/min
20 omega3 = (omega1*ma1dot+omega2*ma2dot)/(ma1dot +
      ma2dot)
21 // Result
22 printf( '\n The humidity ratio is: \%.4 \,\mathrm{f}', omega3);
23
24 // Part (b)
25 // Reduction of the energy rate balance gives
26 // (ha + omega*hv)3 = [ma1dot*(ha + omega*hv)1 +
      ma2dot*(ha + omega*hv)2]/(ma1dot+ma2dot)
27 // With (ha + omega*hv)1 = 10 \text{ kj/kg} and (ha + omega*
```

```
hv)2 = 47.8 kj/kg from figure A-9

28 LHS = (maldot*10+ma2dot*47.8)/(maldot + ma2dot)

29

30 // This value for the enthalpy of the moist air at the exit, together with the previously determined value for omega3, fixes the state of the exiting moist air. From inspection of Fig. A-9,

31 T3 = 19.0

// in degree celcius

32 // Result

33 printf( '\n The temperature of the exiting mixed stream in degree celcius T3 is: %.2 f', T3)
```

# Scilab code Exa 12.15 Example

```
1
2  // Given:-
3  T1 = 38.0

    // in degree celcius
4  m1dot = 4.5e7

    // in kg/h
5  T2 = 30.0

    // in degree celcius
6  m2dot = 4.5e7

    // in kg/h
7  T3 = 25.0

    // in degree celcius
8  psi3 = 0.35
9  T4 = 35.0
```

```
// in degree celcius
10 \text{ psi4} = 0.9
11 	ext{ T5} = 20.0
      // in degree celcius
12
13 // Analysis
14 // The humidity ratios omega3 and omega4 can be
      determined using the partial pressure of the
      water vapor obtained with the respective relative
       humidity
15 \text{ omega3} = 0.00688
16 \text{ omega4} = 0.0327
17 // From tables A-2 and A-22
18 \text{ hf1} = 159.21
19 \text{ hf2} = 125.79
20 \text{ ha4} = 308.2
21 \text{ ha3} = 298.2
22 \text{ hg4} = 2565.3
23 \text{ hg3} = 2547.2
24 \text{ hf5} = 83.96
25 // Calculations
26 madot = (m1dot*(hf1-hf2))/(ha4-ha3+omega4*hg4-omega3)
      *hg3-(omega4-omega3)*hf5)
                                      // in kg/h
27 m5dot = madot*(omega4-omega3)
      // in kg/h
28 // Results
29 printf ('The mass flow rate of dry air in kg/h is:
      \%.2 f', madot)
30 printf ('The mass flow rate of makeup water in kg/h
      is: \%.2 f, m5dot)
```

# Chapter 13

# Reacting Mixtures and Combustion

# Scilab code Exa 13.1 Example

```
1
2 // Given:-
3 // Part(a)
4 // The combustion equation can be written in the
      form of
\frac{5}{\sqrt{8H18 + a(O2 + 3.76N2)}} - b CO2 + c H2O + d N2
6 // Using conservation of mass principle
7 b = 8.00
8 c = 18.00/2.00
9 a = (2.00*b+c)/2.00
10 d = 3.76*a
11
12 // The air fuel ratio on a molar basis is
13 AFbar = a*(1+3.76)/1.00
14 \text{ Ma} = 28.97
      // molar mass of air
15 \text{ MC8H18} = 114.22
```

```
// molar mass of C8H18
16 // The air fuel ratio expressed on a mass basis is
17 AF = AFbar*(Ma/MC8H18)
18
19 // Result
20 printf(' The air fuel ratio on a molar basis is:
      %f', AFbar);
21 printf ('The air fuel ratio expressed on a mass
     basis is: %.2f',AF)
22
23 // Part(b)
24 // For 150\% theoretical air, the chemical equation
     for complete combustion takes the form
25 / c8H18 + 1.5*12.5*(O2 + 3.76N2) -- b CO2 + c H2O
     + d N2 + e O2
26 // Using conservation of mass
27 // Calculations
28 b = 8.00
29 c = 18.00/2.00
30 e = (1.5*12.5*2 - c -2*b)/2.00
31 d = 1.5*12.5*3.76
32 // The air fuel ratio on a molar basis is
33 AFbar = 1.5*12.5*(1+3.76)/1
34 // The air fuel ratio expressed on a mass basis is
35 \text{ AF} = \text{AFbar}*(\text{Ma/MC8H18})
36
37 // Results
38 printf ('The air fuel ratio on a molar basis is:
      \%f', AFbar)
39 printf ('The air fuel ratio expressed on a mass
     basis is: %.2f',AF)
```

#### Scilab code Exa 13.2 Example

```
2 // Given:-
3 // Part(a)
4 // The chemical equation
5 // a CH4 + b*(O2 + 3.76N2) -- 9.7CO2 + .5CO + 2.95
     O2 + 86.85 N2 + cH2O
6 // Calculations
7 // Applying conservation of mass
8 a = 9.7 + 0.5
9 c = 2.0*a
10 b = ((9.7)*(2.0)+(0.5)+((2.0)*(2.95))+c)/2.00
11 \text{ Ma} = 28.97
     // molar mass of air
12 \text{ MCH4} = 16.04
      // molar mass of methane
13 // On a molar basis, the air fuel ratio is
14 AFbar = (b*(1+3.76))/a
15 // On a mass basis
16 \text{ AF} = \text{AFbar}*(\text{Ma/MCH4})
17
18 // Results
19 printf ('The air-fuel ratio on a molar basis is:
      %f', AFbar)
20 printf ('The air-fuel ratio on a mass basis is: %
      .2 f ', AF)
21
22 // Part (b)
23 // The balanced chemical equation for the complete
      combustion of methane with the theoretical amount
       of air is
24 / CH4 + 2(O2 + 3.76N2) - CO2 + 2H2O + 7.52N2
25 // The theoretical air fuel ratio on a molar basis
      is
26 // Calculations
27 \text{ AFbartheo} = 2.00*(1+3.76)/1.0
28 // The percent theoretical air is
29 Ta = AFbar/AFbartheo
```

# Scilab code Exa 13.3 Example

```
1
2 // Given:-
3 // Part(a)
4 // The chemical equation
5 // (.8062 \text{CH4} + .0541 \text{C2H6} + .0187 \text{C3H8} + .0160 \text{C4H10} +
      .1050 \,\mathrm{N2}) + \mathrm{a}(\mathrm{O2} + 3.76 \,\mathrm{N2})
                                     --- b(.078CO2 + .002
      CO + .07O2 + .85N2) + c H2O
6 // Calculations
7 // Using mass conservation
8 b = (0.8062 + 2*.0541 + 3*.0187 + 4*.0160)/(.078 +
      .002)
9 c = (4*.8062 + 6*.0541 + 8*.0187 + 10*.0160)/2
10 a = (b*(2*.078+.002+2*.07) + c)/2
11 // The air fuel ratio on a molar basis is
12 AFbar = a*(1+3.76)/1
13 // Result
```

```
14 printf ( 'The air-fuel ratio on a molar mass basis
      i\,s:~\%.\,2\,f ',AFbar)
15
16 // Part(b)
17 p = 1.0
     // in bar
18 \ V = 100.0
     // in m<sup>3</sup>
19 \text{ Rbar} = 8314.0
     // in N.m/kmol.K
20 T = 300.0
      // in kelvin
21 // Calculations
22 // The amount of fuel in kmol
23 \text{ nF} = (p*10**5*V)/(Rbar*T)
24 // The amount of product mixture that would be
      formed from 100 m3 of fuel mixture is
25 \quad n = nF*(b+c)
26 // Result
27 printf ('The amount of products in kmol that would
      be formed from 100 m3 of fuel mixture at 300 K
      and 1 bar is: %.2 f', n)
28
29 // Part(c)
30 // The balanced chemical equation for the complete
      combustion of the fuel mixture with the
      theoretical amount of air is
31 // (10.8062 \text{CH4} + 0.0541 \text{C2H6} + 0.0187 \text{C3H8} + 0.0160)
      C4H10 + 0.1050N2) + 2(O2 + 3.76N2)
                                             --- 1.0345
      CO2 + 1.93H2O + 7.625N2
32 // Calculations
33 // The theoretical air fuel ratio on a molar basis
       is
34 \text{ AFbartheo} = 2*(1+3.76)/1
```

#### Scilab code Exa 13.4 Example

```
2 // Given:-
3 // The balanced chemical equation for complete
      combustion with the theoretical amount of air is
      obtained from the solution to Example 13.1 as
4 \ \ // \ \ \text{C8H18} \ \ +12.5\text{O2} \ + \ 47\text{N2} \ ----- \ \ 8\text{CO2} \ + \ 9\text{H2O} \ + \ 47\text{N2}
5 // From tabel A-25
6 \text{ hRbar} = -249910
      // in kj/kmol
7 \text{ mfdot} = 1.8e-3
      // mass flow rate of liquid octane in kg/s
8 M = 114.22
      // molar mass of octane
9 \text{ Wcvdot} = 37
      // power output of the engine in kw
10
11 // Calculations
12 // With enthalpy of formation values for CO2 and H2O
      (g) from Table A-25, and enthalpy values for N2,
      H2O, and CO2 from Table A-23
13 hpbar = 8*(-393520 + (36876 - 9364)) + 9*(-241820 +
       (31429 - 9904)) + 47*((26568 - 8669))
14 \text{ nFdot} = \text{mfdot/M}
```

# Scilab code Exa 13.5 Example

```
1
2 // Given:-
3 // When expressed on a per mole of fuel basis, the
      balanced chemical equation obtained in the
      solution to Example 13.2 takes the form
4 / CH4 + 2.265O2 + 8.515N2
                                ----- .951 \text{CO2} + .049 \text{CO}
     + .289O2 + 8.515N2 + 2H2O
5 \text{ cpbar} = 38.00
      // specific heat in KJ/kmol.K
6 // From table A-25
7 \text{ hfnotbar} = -74850.00
      // enthalpy of formation for methane
8 // From table A-23
9 \text{ deltahbar02} = 14770-8682
10 \text{ deltahbarN2} = 14581-8669
11
12 // Calculations
13 hRbar = hfnotbar + cpbar*(400-298) + 2.265*
      deltahbar02 + 8.515*deltahbarN2
                                           // in kj/
14 // With enthalpy of formation values for CO2, CO,
```

# Scilab code Exa 13.6 Example

```
// universal gas constant
9 // The chemical reaction equation for the complete
      combustion of methane with oxygen is
10 // CH4 + 2O2
                 --- CO2 + 2H2O
11
12 // Part (a)
13 // with enthalpy of formation values from table A-25
14 \text{ hfbarCO2} = -393520
15 \text{ hfbarH20} = -241820
16 \text{ hfbarCH4} = -74850
17 // Calculations
18 // with enthalpy values from table A-23
19 \text{ deltahbarCO2} = 37405 - 9364
20 \text{ deltahbarH20} = 31828-9904
Q = ((hfbarCO2 + deltahbarCO2) + 2*(hfbarH2O + 2)
      deltahbarH20) - hfbarCH4) + 3*Rbar*(T1+273-T2)
22 // Result
23 printf ( ' The amount of heat transfer in kJ is: \%.2
      f', Q)
24
25 // Part(b)
26 p2 = p1*(T2/(T1+273))
      // in atm
27 // Result
28 printf ('The final pressure in atm is: \%.2 \, \mathrm{f}',p2)
```

# Scilab code Exa 13.7 Example

```
1
2 // Given:-
3 // The combustion equation is
4 // CH4 + 2O2 + 7.52N2 --- CO2 + 2H2O + 7.52N2
5
```

```
6 // Part(a)
7 // With enthalpy of formation values from Table A-25
8 \text{ hfbarCO2} = -393520
      // in kj/kmol
9 \text{ hfbarH20} = -285830
      // in kj/kmol
10 \text{ hfbarCH4} = -74850
      // in kj/kmol
11 M = 16.04
      // molar mass of CH4 in kg/kmol
12 // Calculations
13 hRPbar = hfbarCO2 + 2*hfbarH2O - hfbarCH4
                                                 // in kj/
      kmol
14 \text{ hRP} = \text{hRPbar/M}
     // in kj/kg
15 // Result
16 printf(' Part(a) the enthalpy of combustion of
      gaseous methane, fuel is: %f kJ/kg.',hRP)
17
18 // Part (b)
19 \text{ hfbarCO2} = -393520
      // in kj/kmol
20 \text{ hfbarH20} = -241820
      // in kj/kmol
21 \text{ hfbarCH4} = -74850
      // in kj/kmol
22 // Calculations
23 hRPbar = hfbarCO2 + 2*hfbarH2O - hfbarCH4
                                                 // in kj/
```

```
kmol
24 \text{ hRP} = \text{hRPbar/M}
      // in kj/kg
25 // Result
26 printf ('Part (b) the enthalpy of combustion of
      gaseous methane, fuel is: \%f kJ/kg', hRP);
27
28 // Part(c)
29 // From table A-23
30 \text{ deltahbar02} = 31389-8682
      // in kj/kmol
31 \text{ deltahbarH20} = 35882-9904
      // in kj/kmol
32 \text{ deltahbarCO2} = 42769-9364
      // in kj/kmol
33
34 // Using table A-21
35 // Calculations
36 // function cpbar = f(T)
37 T=298
      // in kelvin
38
39 function T = cpbar(T)
       T = (3.826 - (3.979e-3)*T + 24.558e-6*T**2 -
           22.733e-9*T**3 + 6.963e-12*T**4)*8.314
41 endfunction
42
43 deltahbarCH4 = intg(298,1000,cpbar)
44 var = deltahbarCH4(1)
45
46 hRPbar = hRPbar + (deltahbarCO2 + 2*deltahbarH2O -
      var -2*deltahbar02)
47 \text{ hRP} = \text{hRPbar/M}
```

```
// in kj/kg
48 // Result
49 printf( ' Part(c) the enthalpy of combustion of
        gaseous methane, per kg of fuel is %.f kJ/kg',hRP
    );
```

# Scilab code Exa 13.8 Example

```
1
2 // Given:-
3 // Part(a)
4 // For combustion of liquid octane with the
      theoretical amount of air, the chemical equation
5 // C8H18(1) + 12.5 O2 + 47N2
                                    -----8 \text{ CO2} + 9 \text{ H2O}(
      g) + 47N2
6 // with enthalpy of formation data from Table A-25
7 \text{ hfbarC8H18} = -249910.0
      // in kj/kmol
8 \text{ hfbarCO2} = -393520.0
9 \text{ hfbarH20} = -241820.0
10
11 // Calculations
12 RHS = hfbarC8H18 - (8*hfbarC02 + 9*hfbarH20)
                                               // in kj/kmol
13 // at temperature 2400k
14 \text{ LHS1} = 5089337.0
      // in kj/kmol
15 // at temperature 2350 k
16 \text{ LHS2} = 4955163.0
      // in kj/kmol
```

```
17 // Interpolation between these temperatures gives
18 Tp = 2400.00 + ((2400.0 - 2350.0)/(LHS1 - LHS2))*(RHS -
     LHS1)
19 // Result
20 printf (' The temperature in kelvin with theoretical
      amount of air is: %.2 f', Tp)
21
22 // Part (b)
23 // For complete combustion of liquid octane with 400
     \% theoretical air, the chemical equation is
24 // C8H18(1) + 50O2 + 188N2 ---- 8CO2 + 9H2O +
     37.5O2 + 188N2
25
26 // Proceeding iteratively as part(a)
27 Tp = 962
     // in kelvin
28
29 // Result
30 printf (' The temperature in kelvin using 400
     percent theoretical air is: %.2f',Tp)
```

#### Scilab code Exa 13.9 Example

```
1
2 // Given:-
3
4 // Part(a)
5 Tp = 2395

    // in kelvin, from example 13.8
6 // For combustion of liquid octane with the theoretical amount of air, the chemical equation is
7 // C8H18(1) + 12.5O2 + 47N2 --- 8CO2 + 9H2O(g) +
```

```
47N2
9 // From table A-25
10 \text{ sFbar} = 360.79
      // absolute entropy of liquid octane in kj/kmol.K
11
12 // From table A-23
13 // For reactant side
14 \text{ sbar02atTref} = 205.03
      // in kj/kmol.K
15 \text{ sbarN2atTref} = 191.5
      // in kj/kmol.K
16 \text{ Rbar} = 8.314
      // universal gas constant in SI units
17 \text{ y}02 = 0.21
18 \text{ yN2} = 0.79
19 // For product side
20 \text{ yCO2} = 8.0/64.0
21 \text{ yH20} = 9.0/64.0
22 \text{ yN2p} = 47.0/64.0
23
24 // Calculations
25 \text{ sbar02} = \text{sbar02atTref} - \text{Rbar}*log(y02)
                                                    // in kj/
      kmol.K
26 sbarN2 = sbarN2atTref - Rbar*log(yN2)
                                                    // in kj/
      kmol.K
27 // With the help from table A-23
28 \text{ sbarCO2} = 320.173 - Rbar*log(yCO2)
29 sbarH20 = 273.986 - Rbar*log(yH20)
30 \quad sbarN2p = 258.503 - Rbar*log(yN2p)
31 \text{ sigmadot} = (8*\text{sbarCO2} + 9*\text{sbarH2O} + 47*\text{sbarN2p}) -
      sFbar - (12.5*sbar02 + 47*sbarN2)
```

```
32
33 // Result
34 printf ('The rate of entropy production, in kJ/K
      per kmol of fuel with theoretical amount of air
      is: \%.2 f', sigmadot)
35
36 // Part(b)
37 // The complete combustion of liquid octane with 400
      % theoretical air is described by the following
      chemical equation:
38 / C8H18(1) + 50 O2 + 188N2 ---- 8 CO2 + 9H2O(g) +
       37.5O2 + 188N2
39
40 // For product side
41 \text{ yCO2} = 8.0/242.5
42 \text{ yH20} = 9.0/242.5
43 \text{ y02} = 37.5/242.5
44 \text{ yN2p} = 188.0/242.5
45 // Calculations
46 // With help from table A-23
47 \text{ sbarCO2} = 267.12 - Rbar*log(yCO2)
48 \text{ sbarH20} = 231.01 - Rbar*log(yH20)
49 sbar02p = 242.12 - Rbar*log(y02)
50 \quad \text{sbarN2p} = 226.795 - \text{Rbar}*\log(yN2p)
51 \text{ sigmadot} = (8.0*\text{sbarCO2} + 9.0*\text{sbarH2O} + 37.5*\text{sbarO2p})
       +188.0*sbarN2p) -sFbar - (50.0*sbarO2 + 188.0*
      sbarN2)
52
53 // Result
54 printf (' The rate of entropy production, in kJ/K
      per kmol of fuel with 400 percent theoretical air
       i\,s:~\%.\,2\,f~ ', sigmadot)
```

Scilab code Exa 13.10 Example

```
2 // Given:-
3 \text{ Rbar} = 8.314
      // universal gas constant in SI units
4 // The chemical equation for the complete combustion
       of methane with oxygen is
5 // \text{CH4} + 2O2 --- \text{CO2} + 2H2O
6 \text{ yCH4} = 1.0/3.0
7 y02 = 2.0/3.0
8 \text{ yCO2} = 1.0/3.0
9 \text{ yH20} = 2.0/3.0
10 // From table A-25
11 \text{ sbarCH4atTref} = 186.16
      // in kj/kmol.K
12 \text{ sbarO2atTref} = 205.03
      // in kj/kmol.K
13 p2 = 3.02
      // in atm
14 \text{ pref} = 1.0
      // in atm
15
16 // Calculations
17 sbarCH4 = sbarCH4atTref - Rbar*log(yCH4)
18 sbar02 = sbar02atTref - Rbar*log(y02)
19 // With help from table A-23
20 sbarCO2 = 263.559 - Rbar*log(yCO2*p2/pref)
                                           // in kj/kmol.K
21 \text{ sbarH20} = 228.321 - \text{Rbar*log}(yH20*p2/pref)
                                           // in kj/kmol.K
22 deltaS = sbarCO2 + 2*sbarH2O - sbarCH4 -2*sbarO2
                                          // in kj/K
23
24 // Result
```

```
25 printf( 'The change in entropy of the system is: % .2 f kJ/K ',deltaS)
```

# Scilab code Exa 13.11 Example

```
1
2 // Given:-
3 // Methane is formed from carbon and hydrogen
      according to
4 // C + 2H2 ---- CH4
6 // In the present case, all substances are at the
      same temperature and pressure, 25C and 1 atm,
      which correspond to the standard reference state
      values
7 \text{ hCbar} = 0
8 \text{ hH2bar} = 0
9 \text{ gRbar} = 0
10 // With enthalpy of formation and absolute entropy
      data from Table A-25
11 \text{ hfbarCH4} = -74850
12 \text{ sbarCH4} = 186.16
13 \text{ sbarC} = 5.74
14 \text{ sbarH2} = 130.57
15 \text{ Tref} = 298.15
      // in kelvin
16
17 // Calculation
18 gfbarCH4 = hfbarCH4 -Tref*(sbarCH4-sbarC-2*sbarH2)
                                       // in kj/kmol
19
20 // Result
21 printf (' The gibbs function of formation of methane
       at the standard state is: %f kJ/mol',gfbarCH4)
```

#### Scilab code Exa 13.12 Example

```
1
2 // Given:-
3 // Complete combustion of liquid octane with O2 is
      described by
4 // C8H18(1) + 12.5O2
                           ----- 8CO2 + 9H2O
6 // Part(a)
7 \text{ Rbar} = 8.314
      // universal gas constant in SI units
8 \text{ Tnot} = 298.15
      // in kelvin
9 // From table A-25
10 \text{ gbarC8H18} = 6610.0
11 \text{ gbar02} = 0
12 \text{ gbarCO2} = -394380
13 \text{ gbarH20} = -228590
14 \text{ y}02 = 0.2035
15 \text{ yCO2} = 0.0003
16 \text{ yH} 20 = 0.0312
17 M = 114.22
      // molecular weight of liquid octane
18
19 // Calculations
20 \text{ ech} = ((gbarC8H18 + 12.5*gbarO2 -8*gbarCO2 -9*)
      gbarH20) + Rbar*Tnot*log(y02**12.5/(yC02**8*yH20
      **9
              )))/M
21 // Result
22 printf (' Part (a) the chemical exergy obtained on a
      unit mass basis is: \%.2 f kJ/K, ech)
```

# Scilab code Exa 13.13 Example

```
// in kj/kg
9 \text{ snot} = 0.3674
      // in kj/kg
10 // With data from Table A-25
11 gbarH20liq = -237180.0
12 \text{ gbarH20gas} = -228590.0
13 \text{ yeH20} = 0.0303
14 M = 18.0
      // molar mass of steam
15
16 // Calculations
17 ech = (1.0/M)*(gbarH2Oliq-gbarH2Ogas + Rbar*Tnot*log
      (1/yeH2O))
                   // in kj/kg
18 	ext{ ef = } h-hnot-Tnot*(s-snot) + ech
                                                       // in
       kj/kg
19
20 // Result
21 printf ('The flow exergy of the steam, in is: %.2f
       kJ/kg ',ef)
```

#### Scilab code Exa 13.14 Example

```
9 \text{ yN2p} = 10.53/(1.0+2.0+10.53+.8)
10 \text{ yO2p} = 0.8/(1.0+2.0+10.53+.8)
11
12 \text{ Rbar} = 8.314
      // universal gas constant in SI units
13 \text{ Tnot} = 298.15
      // in kelvin
14
15 \text{ yeN2} = 0.7567
16 \text{ yeO2} = 0.2035
17 \text{ yeH20} = 0.0303
18 \text{ yeCO2} = 0.0003
19
20 // Calculations
21
22 ebarch = Rbar*Tnot*(log(yCO2p/yeCO2) + 2*log(yH2Op/
      yeH20) + 10.53*log(yN2p/yeN2) + .8*log(y02p/ye02)
      )
23
24 // with data from tables A-23 at 480 and 1560 kelvin
      , the thermomechanical contribution to the flow
      exergy, per mole of fuel, is
25 \text{ contri480} = 17712.0
      // kJ per kmol of fuel
26 \quad contri1560 = 390853.0
      // kJ per kmol of fuel
27 \text{ efbar480} = \text{contri480} + \text{ebarch}
      // kJ per kmol of fuel
28 efbar1560 = contri1560 + ebarch
      // kJ per kmol of fuel
29
30 // Results
```

```
31 printf( 'At T= 480k, the flow exergy of the
      combustion products, in kJ per kmol of fuel is:
      %.2 f',efbar480)
32 printf( 'At T = 1560K, the flow exergy of the
      combustion products, in kJ per kmol of fuel is:
      %.2 f',efbar1560)
```

# Scilab code Exa 13.15 Example

```
1
2 // Given:-
3 \text{ mFdot} = 1.8e-3
     // fuel mass flow rate in kg/s
4 \text{ ech} = 47346.0
      // in kj/kg, from example 13.12(a)
5 \text{ Wcvdot} = 37.0
      // power developed by the engine in kw
7 // Calculations
8 Efdot = mFdot*ech
      // rate at which exergy enters with the fuel in
      kw
9 epsilon = Wcvdot/Efdot
      // exergetic efficiency
10
11 // Result
12 printf(' The exergetic efficiency is: \%.3 \, \mathrm{f}',
      epsilon)
```

# Scilab code Exa 13.16 Example

```
1
2 // Given:-
3 \text{ Tnot} = 298
     // in kelvin
5 // For the case of complete combustion with the
      theoretical amount of air
6 \text{ sigmadot} = 5404.0
      // rate of entropy production from example 13.9,
     in kj/kmol.K
7 \text{ Efdot} = 5407843.0
     // rate at which exergy enters with the fuel from
      example 13.12, in kj/kmol
8 // Calculations:-
9 Eddot = Tnot*sigmadot
     // in kj/kmol
10 epsilon = 1-Eddot/Efdot
11 // Result
12 printf ('The exergetic efficiency with theoretical
      amount of air is: %.3f', epsilon)
13
14 // For the case of combustion with 400\% theoretical
      air
15 sigmadot = 9754.0
      // rate of entropy production from example 13.9,
     in kj/kmol.K
16 // Calculations
```

# Chapter 14

# Chemical and Phase Equilibrium

# Scilab code Exa 14.1 Example

```
// in kj/kmol
12 \text{ deltahbarCO2} = 0
     // in kj/kmol
13 \text{ deltahbarCO} = 0
      // in kj/kmol
14 \text{ deltahbar02} = 0
      // in kj/kmol
15 \text{ sbarCO2} = 213.69
      // in kj/kmol.K
16 \text{ sbarCO} = 197.54
      // in kj/kmol.K
17 \text{ sbar02} = 205.03
      // in kj/kmol.K
18 // From table A-27
19 \log K table = 45.066
20 // Calculations
21 deltaG = (hfbarCO2-hfbarCO-.5*hfbarO2) + (
      deltahbarCO2-deltahbarCO-.5*deltahbarO2) - T*(
      sbarC02-sbarC0-.5*sbar02)
21 \ln K = -deltaG/(Rbar*T)
23 \log K = (1/\log(10))*\ln K
24 // Results
25 printf (' Part(a) the value of equilibrium constant
      expressed as log10K is: %f',logK);
26 printf ('The value of equilibrium constant
      expressed as log10K from table A-27 is: %f',
      logKtable);
27
28 // Part (b)
29 T = 2000.0
```

```
// in kelvin
30 // From table A-23
31 \text{ hfbarCO2} = -393520.0
      // in kj/kmol
32 \text{ hfbarCO} = -110530.0
      // in kj/kmol
33 \text{ hfbarO2} = 0
      // in kj/kmol
34 \text{ deltahbarCO2} = 100804 - 9364
      // in kj/kmol
35 \text{ deltahbarCO} = 65408 - 8669
      // in kj/kmol
36 \text{ deltahbar02} = 67881 - 8682
      // in kj/kmol
37 \text{ sbarCO2} = 309.210
      // in kj/kmol.K
38 \text{ sbarCO} = 258.6
      // in kj/kmol.K
39 \text{ sbar02} = 268.655
      // in kj/kmol.K
40 // Calculations
41 deltaG = (hfbarCO2-hfbarCO-.5*hfbarO2) + (
      deltahbarCO2-deltahbarCO-.5*deltahbarO2) - T*(
      sbarCO2-sbarCO-.5*sbarO2)
42 \ln K = -deltaG/(Rbar*T)
43 \log K = (1/\log(10))*lnK
44 // From table A-27
45 \log K table = 2.884
46 // Results
```

# Scilab code Exa 14.2 Example

```
1
2 // Given:-
3 // Applying conservation of mass, the overall
      balanced chemical reaction equation is
  // CO + .5O2
                               zCO + (z/2)O2 + (1-z)CO2
6 // At 2500 K, Table A-27 gives
7 \log 10K = -1.44
8 // Part(a)
9 p = 1.0
      // in atm
10 // Calculations
11 K = (10.0) **(log10K)
      // equilibrium constant
12 // Solving equation K = (z/(1-z))*(2/(2+z))^{.5} *(p)
     /1) \hat{} . 5 gives
13 z = 0.129
14 \text{ yCO} = 2.0*z/(2.0 + z)
15 \text{ y}02 = z/(2.0 + z)
16 \text{ yCO2} = 2.0*(1.0 - z)/(2.0 + z)
17
18 // Results
19 printf(' Part(a) mole fraction of CO is: %.3f', yCO
20 printf (' Mole fraction of O2 is: \%.3 f', yO2)
```

```
21 printf(' Mole fraction of CO2 is: %.3f',yCO2)
22
23 // Part(b)
24 p = 10.0
      // in atm
25 // Solving equation K = (z/(1-z))*(2/(2+z))^{.5} *(p)
      /1) \hat{} . 5 gives
26 z = 0.062
27 \text{ yCO} = 2.0*z/(2.0 + z)
28 \text{ yO2} = z/(2.0 + z)
29 \text{ yCO2} = 2.0*(1.0 - z)/(2.0 + z)
30
31 // Results
32 printf(' Part(b) mole fraction of CO is: %.3f',yCO
33 printf(' Mole fraction of O2 is: \%.3 f', yO2)
34 printf( ' Mole fraction of CO2 is: \%.3\,\mathrm{f} ',yCO2)
```

# Scilab code Exa 14.3 Example

```
11 z = 2*yCO/(2 - yCO)
12 K = (z/(1-z))*(z/(2 + z))**.5*(p/pref)**.5
13
14 // Result
15 printf( ' The temperature T of the mixture in kelvin is: %f',T);
```

# Scilab code Exa 14.4 Example

```
1
2 // Given:-
3 // For a complete reaction of CO with the
      theoretical amount of air
4 // CO + .5 O2 + 1.88N2 ---- CO2 + 1.88N2
5 // Accordingly, the reaction of CO with the
      theoretical amount of air to form CO2, CO, O2,
      and N2 is
6 // CO + .5O2 + 1.88N2 - zCO + z/2 O2 + (1-z)CO2 +
      1.88\,\mathrm{N2}
8 K = 0.0363
      // equilibrium constant the solution to Example
     14.2
9 p = 1.0
      // in atm
10 \text{ pref} = 1.0
      // in atm
11
12 // Calculations
13 // Solving K = (z*z^{5}.5/(1-z))*((p/pref)*2/(5.76+z))
      ^.5 gives
14 z = 0.175
```

```
15 yCO = 2.0*z/(5.76 + z)

16 yO2 = z/(5.76 + z)

17 yCO2 = 2.0*(1.0-z)/(5.76 + z)

18 yN2 = 3.76/(5.76 + z)

19

20 // Results

21 printf(' The mole fraction of CO is: %.3f',yCO)

22 printf(' The mole fraction of O2 is: %.3f',yO2)

23 printf(' The mole fraction of CO2 is: %.3f',yCO2)

24 printf(' The mole fraction of N2 is: %.3f',yN2)
```

# Scilab code Exa 14.5 Example

```
1
2 // Given:-
3 // Applying the conservation of mass principle, the
      overall dissociation reaction is described by
4 // CO2 --- zCO2 + (1-z)CO + ((1-z)/2)O2
6 p = 1.0
    // in atm
7 \text{ pref} = 1.0
     // in atm
8 // At 3200 K, Table A-27 gives
9 \log 10k = -.189
10 // Solving k = ((1-z)/2)*((1-z)/(3-z))^{.5} gives
11 z = 0.422
12
13 // Calculations
14 \quad k = 10**log10k
15 // From tables A-25 and A-23
16 \text{ hfbarCO2} = -393520.0
```

```
// in kj/kmol
17 \text{ deltahbarCO2} = 174695-9364
      // in kj/kmol
18 \text{ hfbarCO} = -110530.0
      // in kj/kmol
19 \text{ deltahbarCO} = 109667-8669
     // in kj/kmol
20 \text{ hfbarO2} = 0
     // in kj/kmol
21 \text{ deltahbar02} = 114809-8682
      // in kj/kmol
22 \text{ hfbarCO2r} = -393520.0
      // in kj/kmol
23 \text{ deltahbarCO2r} = 0
      // in kj/kmol
24
25 Qcvdot = 0.422*(hfbarCO2 + deltahbarCO2) + 0.578*(
      hfbarCO + deltahbarCO) + 0.289*(hfbarO2 +
      deltahbar02)- (hfbarC02r + deltahbarC02r)
26
27 // Result
28 printf ('The heat transfer to the reactor, in kJ
      per kmol of CO2 entering is: %f', Qcvdot);
```

### Scilab code Exa 14.8 Example

```
1
2 // Given:-
```

```
3 // The ionization of cesium to form a mixture of Cs,
      Cs+, and e- is described by
4 // Cs --- (1-z) Cs + z Cs + Ze -
6 K = 15.63
7 z = 0.95
8 \text{ pref} = 1
     // in atm
9 // Calculation
10 p = pref*K*((1-z**2)/z**2)
11
12 // Results
13 printf(' The pressure if the ionization of CS is 95
       percent complete is: %f atm',p);
14
15 x = linspace(0,10,100)
16 \text{ for } i = 1:100
       y(i) = 100*((1/(1+x(i)/K))**0.5)
17
18 end
19
20 plot(x,y)
21 xlabel("Pressure (atm)")
22 ylabel ("Ionization")
```

# Scilab code Exa 14.10 Example

```
// in bar
6 p = 1.0
     // in bar
7 T = 293.15
     // in kelvin
8 \text{ Rbar} = 8.314
     // universal gas constant in SI units
9 M = 18.02
     // molat mass of water in kg/kmol
10 e = 2.715
11
12 // Calculations
13 pvbypsat = e**(vf*(p-psat)*10**5/((1000*Rbar/M)*T))
14 \text{ percent} = (pvbypsat-1)*100
15
16 // Result
17 printf(' The departure, in percent, of the partial
      pressure of the water vapor from the saturation
      pressure of water at 20 is: %.3f',percent)
```