

Circular and non-circular nearly horizon-skimming orbits in Kerr spacetimes

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We have performed a detailed analysis of orbital motion in the vicinity of a nearly extremal Kerr black hole. For very rapidly rotating black holes — spin parameter $a \equiv J/M > 0.9524M$ — we have found a class of very strong field eccentric orbits whose orbital angular momentum L_z increases with the orbit's inclination with respect to the equatorial plane, while keeping latus rectum and eccentricity fixed. This behavior is in contrast with Newtonian intuition, and is in fact opposite to the “normal” behavior of black hole orbits. Such behavior was noted previously for circular orbits; since it only applies to orbits very close to the black hole, they were named “nearly horizon-skimming orbits”. Our current analysis generalizes this result, mapping out the full generic (inclined and eccentric) family of nearly horizon-skimming orbits. The earlier work on circular orbits reported that, under gravitational radiation emission, nearly horizon-skimming orbits exhibit unusual inspiral, tending to evolve to smaller orbit inclination, toward prograde equatorial configuration. Normal orbits, by contrast, always demonstrate slowly *growing* orbit inclination — orbits evolve toward the retrograde equatorial configuration. Using up-to-date Teukolsky-based fluxes, we have concluded that the earlier result was incorrect — all orbits, *including nearly horizon skimming ones*, exhibit growing orbit inclination under radiative backreaction. We also find that the inclination change is, in any case, very small. As such, we conclude that these orbits are *not* likely to have a clear and peculiar imprint on the gravitational waveforms expected to be measured by the space-based detector LISA.

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I. INTRODUCTION

The space-based gravitational-wave detector LISA [1] will be a unique tool to probe the nature of supermassive black holes (SMBHs), making it possible to map in detail their spacetimes. This goal is expected to be achieved by observing gravitational waves emitted by compact stars or black holes with masses $\mu \approx 1 - 100 M_\odot$ spiraling into the SMBHs which reside in the center of galaxies [2] (particularly the low end of the galactic center black hole mass function, $M \approx 10^5 - 10^7 M_\odot$). Such events are known as *extreme mass ratio inspirals* (EMRIs). Current wisdom suggests that several tens to perhaps of order a thousand such events could be measured per year by LISA [3].

Though the distribution of spins for observed astrophysical black holes is not very well understood at present, very rapid spin is certainly plausible, as accretion tends to spin-up SMBHs [4]. Most QPO models suggest this is indeed the case in all low-mass x-ray binaries for which data is available [5]. On the other hand, continuum spectral fitting of some high-mass x-ray binaries indicates that modest spins (spin parameter $a/M \equiv J/M^2 \sim 0.6 - 0.8$) are likewise plausible [6]. The continuum fit technique *does* find extremely high spin for the galactic “microquasar” GRS 1915+105 — $a/M \gtrsim 0.98$ [7].

This argues for a wide variety of possible spins, depending on the detailed birth and growth history of a given black hole.

In the mass range corresponding to black holes in galactic centers, measurements of the broad iron $K\alpha$ emission line in active galactic nuclei suggest that SMBHs can be very rapidly rotating (see Ref. [8] for a recent review). For instance, in the case of MCG-6-30-15, for which highly accurate observations are available, a has been found to be larger than $0.987M$ at 90% confidence [9]. Because gravitational waves from EMRIs are expected to yield a very precise determination of the spins of SMBHs [10], it is interesting to investigate whether EMRIs around very rapidly rotating black holes may possess peculiar features which would be observable by LISA. Should such features exist, they would provide unambiguous information on the spin of SMBHs and thus on the mechanisms leading to their formation [11].

For extremal Kerr black holes ($a = M$), the existence of a special class of “circular” orbits was pointed out long ago by Wilkins [12], who named them “horizon-skimming” orbits. (“Circular” here means that the orbits are of constant Boyer-Lindquist coordinate radius r .) These orbits have varying inclination angle with respect to the equatorial plane and have the same coordinate radius as the horizon, $r = M$. Despite this seemingly hazardous location, it can be shown that all these $r = M$ orbits have finite separation from one another and from the event horizon [13]. Their somewhat pathological description is due to a singularity in the Boyer-Lindquist coordinates, which collapses a finite span of the spacetime into

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$r = M$.

Besides being circular and “horizon-skimming”, these orbits also show peculiar behavior in their relation of angular momentum to inclination. In Newtonian gravity, a generic orbit has $L_z = |L| \cos \iota$, where ι is the inclination angle relative to the equatorial plane (going from $\iota = 0$ for equatorial prograde orbits to $\iota = \pi$ for equatorial retrograde orbits, passing through $\iota = \pi/2$ for polar orbits), and L is the orbital angular momentum vector. As a result, $\partial L_z(r, \iota)/\partial \iota < 0$, and the angular momentum in the z -direction always decreases with increasing inclination if the orbit’s radius is kept constant. This intuitively reasonable decrease of L_z with ι is seen for almost all black hole orbits as well. Horizon-skimming orbits, by contrast, exhibit exactly the opposite behavior: L_z increases with inclination angle.

Reference [14] asked whether the behavior $\partial L_z/\partial \iota > 0$ could be extended to a broader class of circular orbits than just those at the radius $r = M$ for the spin value $a = M$. It was found that this condition is indeed more general, and extended over a range of radius from the “innermost stable circular orbit” to $r \simeq 1.8M$ for black holes with $a > 0.9524M$. Orbits that show this property have been named “nearly horizon-skimming”. The Newtonian behavior $\partial L_z(r, \iota)/\partial \iota < 0$ is recovered for all orbits at $r \gtrsim 1.8M$ [14].

A qualitative understanding of this behavior comes from recalling that very close to the black hole all physical processes become “locked” to the hole’s event horizon [15], with the orbital motion of point particles coupling to the horizon’s spin. This locking dominates the “Keplerian” tendency of an orbit to move more quickly at smaller radii, forcing an orbiting particle to slow down in the innermost orbits. Locking is particularly strong for the most-bound (equatorial) orbits; the least-bound orbits (which have the largest inclination) do not strongly lock to the black hole’s spin until they have very nearly reached the innermost orbit [14]. The property $\partial L_z(r, \iota)/\partial \iota > 0$ reflects the different efficiency of nearly horizon-skimming orbits to lock with the horizon.

Reference [14] argued that this behavior could have observational consequences. It is well-known that the inclination angle of an inspiraling body generally increases due to gravitational-wave emission [16, 17]. Since $dL_z/dt < 0$ because of the positive angular momentum carried away by the gravitational waves, and since “normal” orbits have $\partial L_z/\partial \iota < 0$, one would expect $d\iota/dt > 0$. However, if during an evolution $\partial L_z/\partial \iota$ switches sign, then $d\iota/dt$ might switch sign as well: An inspiraling body could evolve towards an equatorial orbit, signalling the presence of an “almost-extremal” Kerr black hole.

It should be emphasized that this argument is not rigorous. In particular, one needs to consider the joint evolution of orbital radius and inclination angle; and, one must include the dependence of these two quantities on orbital energy as well as angular momentum¹. As such, $d\iota/dt$ depends not only on

dL_z/dt and $\partial L_z/\partial \iota$, but also on dE/dt , $\partial E/\partial \iota$, $\partial E/\partial r$ and $\partial L_z/\partial r$.

In this sense, the argument made in Ref. [14] should be seen as claiming that the contribution coming from dL_z/dt and $\partial L_z/\partial \iota$ are simply the dominant ones. Using the numerical code described in [17] to compute the fluxes dL_z/dt and dE/dt , it was then found that a test-particle on a circular orbit passing through the nearly horizon-skimming region of a Kerr black hole with $a = 0.998M$ (the value at which a hole’s spin tends to be buffered due to photon capture from thin disk accretion [19]) had its inclination angle decreased by $\delta \iota \approx 1^\circ - 2^\circ$ [14]. It should be noted at this point that the rate of change of inclination angle, $d\iota/dt$, appears as the *difference* of two relatively small and expensive to compute rates of change [cf. Eq. (3.8) of Ref. [17]]. As such, small relative errors in those rates of change can lead to large relative errors in $d\iota/dt$. Finally, in Ref. [14] it was speculated that the decrease could be even larger for eccentric orbits satisfying the condition $\partial L_z/\partial \iota > 0$, possibly leading to an observable imprint on EMRI gravitational waveforms.

The main purpose of this paper is to extend Ref. [14]’s analysis of nearly horizon-skimming orbits to include the effect of orbital eccentricity, and to thereby test the speculation that there may be an observable imprint on EMRI waveforms of nearly horizon-skimming behavior. In doing so, we have revisited all the calculations of Ref. [14] using a more accurate Teukolsky solver (the details of which will be presented elsewhere [20]) which serves as the engine for the analysis presented in Ref. [21].

We have found that the critical spin value for circular nearly horizon-skimming orbits, $a > 0.9524M$, also delineates a family of eccentric orbits for which the condition $\partial L_z(p, e, \iota)/\partial \iota > 0$ holds. (More precisely, we consider variation with respect to an angle θ_{inc} that is easier to work with in the extreme strong field, but that is easily related to ι .) The parameters p and e are the orbit’s latus rectum and eccentricity, defined precisely in Sec. II. These generic nearly horizon-skimming orbits all have $p \lesssim 2M$, deep in the black hole’s extreme strong field.

We next study the evolution of these orbits under gravitational-wave emission. We first revisited the evolution of circular, nearly horizon-skimming orbits using the improved Teukolsky solver which was used for the analysis of Ref. [21]. The results of this analysis were somewhat surprising: Just as for “normal” orbits, we found that orbital inclination *always* increases during inspiral, even in the nearly horizon-skimming regime. This is in stark contrast to the claims of Ref. [14]. As noted above, the inclination’s rate of change depends on the difference of two expensive and difficult to compute numbers, and thus can be strongly impacted by small relative errors in those numbers. A primary result of this paper is thus to retract the claim of Ref. [14] that an important dynamic signature of the nearly horizon-skimming region

¹ In the general case, one must also include the dependence on “Carter’s constant” Q [18], the third integral of black hole orbits (described more

carefully in Sec. II). For circular orbits, $Q = Q(E, L_z)$: knowledge of E and L_z completely determines Q .

is a reversal in the sign of inclination angle evolution: The inclination *always* grows under gravitational radiation emission.

We next extended this analysis to study the evolution of generic nearly-horizon skimming orbits. The Teukolsky code to which we have direct access can, at this point, only compute the radiated fluxes of energy E and angular momentum L_z ; results for the evolution of the Carter constant Q are just now beginning to be understood [22], and have not yet been incorporated into this code. We instead use “kludge” expressions for dE/dt , dL_z/dt , and dQ/dt which were inspired by Refs. [23, 24]. These expressions are based on post-Newtonian flux formulas, modified in such a way that they fit strong-field radiation reaction results obtained from a Teukolsky integrator; see Ref. [24] for further discussion. Our analysis indicates that, just as in the circular limit, the result $dl/dt > 0$ holds for generic nearly-horizon skimming orbits. Furthermore, and contrary to the speculation of Ref. [14], we do *not* find a large amplification of dl/dt as orbits are made more eccentric.

Our conclusion is that the nearly horizon-skimming regime, though an interesting curiosity of strong-field orbits of nearly extremal black holes, will *not* imprint any peculiar observational signature on EMRI waveforms.

The remainder of this paper is organized as follows. In Sec. II, we review the properties of bound stable orbits in Kerr, providing expressions for the constants of motion which we will use in Sec. III to generalize nearly horizon-skimming orbits to the non-circular case. In Sec. IV, we study the evolution of the inclination angle for circular nearly horizon-skimming orbits using Teukolsky-based fluxes; in Sec. V we do the same for non-circular orbits and using kludge fluxes. We present and discuss our detailed conclusions in Sec. VI. The fits and post-Newtonian fluxes used for the kludge fluxes are presented in the Appendix. Throughout the paper we have used units in which $G = c = 1$.

II. BOUND STABLE ORBITS IN KERR SPACETIMES

The line element of a Kerr spacetime, written in Boyer-Lindquist coordinates reads [25]

$$ds^2 = -\left(1 - \frac{2Mr}{\Sigma}\right) dt^2 + \frac{\Sigma}{\Delta} dr^2 + \Sigma d\theta^2 + \left(r^2 + a^2 + \frac{2Ma^2r}{\Sigma} \sin^2 \theta\right) \sin^2 \theta d\phi^2 - \frac{4Mar}{\Sigma} \sin^2 \theta dt d\phi, \quad (1)$$

where

$$\Sigma \equiv r^2 + a^2 \cos^2 \theta, \quad \Delta \equiv r^2 - 2Mr + a^2. \quad (2)$$

Up to initial conditions, geodesics can then be labelled by four constants of motion: the mass μ of the test particle, its energy E and angular momentum L_z as measured by an observer at infinity and the Carter constant Q [18]. The presence of these four conserved quantities makes the geodesic equations separable in Boyer-Lindquist coordinates. Introducing the Carter

time λ , defined by

$$\frac{d\tau}{d\lambda} \equiv \Sigma, \quad (3)$$

the geodesic equations become

$$\begin{aligned} \left(\frac{dr}{d\lambda}\right)^2 &= V_r(r), & \frac{dt}{d\lambda} &= V_t(r, \theta), \\ \left(\frac{d\theta}{d\lambda}\right)^2 &= V_\theta(\theta), & \frac{d\phi}{d\lambda} &= V_\phi(r, \theta), \end{aligned} \quad (4)$$

with

$$V_t(r, \theta) \equiv E \left(\frac{\varpi^4}{\Delta} - a^2 \sin^2 \theta \right) + aL_z \left(1 - \frac{\varpi^2}{\Delta} \right), \quad (5a)$$

$$V_r(r) \equiv (E\varpi^2 - aL_z)^2 - \Delta [\mu^2 r^2 + (L_z - aE)^2 + Q], \quad (5b)$$

$$V_\theta(\theta) \equiv Q - L_z^2 \cot^2 \theta - a^2 (\mu^2 - E^2) \cos^2 \theta, \quad (5c)$$

$$V_\phi(r, \theta) \equiv L_z \csc^2 \theta + aE \left(\frac{\varpi^2}{\Delta} - 1 \right) - \frac{a^2 L_z}{\Delta}, \quad (5d)$$

where we have defined

$$\varpi^2 \equiv r^2 + a^2. \quad (6)$$

The conserved parameters E , L_z , and Q can be remapped to other parameters that describe the geometry of the orbit. We have found it useful to describe the orbit in terms of an angle θ_{\min} — the minimum polar angle reached by the orbit — as well as the latus rectum p and the eccentricity e . In the weak-field limit, p and e correspond exactly to the latus rectum and eccentricity used to describe orbits in Newtonian gravity; in the strong field, they are essentially just a convenient remapping of the orbit’s apoastron and periastron:

$$r_{\text{ap}} \equiv \frac{p}{1-e}, \quad r_{\text{peri}} \equiv \frac{p}{1+e}. \quad (7)$$

Finally, in much of our analysis, it is useful to refer to

$$z_- \equiv \cos^2 \theta_{\min}, \quad (8)$$

rather than to θ_{\min} directly.

To map (E, L_z, Q) to (p, e, z_-) , use Eq. (4) to impose $dr/d\lambda = 0$ at $r = r_{\text{ap}}$ and $r = r_{\text{peri}}$, and to impose $d\theta/d\lambda = 0$ at $\theta = \theta_{\min}$. (Note that for a circular orbit, $r_{\text{ap}} = r_{\text{peri}} = r_0$. In this case, one must apply the rules $dr/d\lambda = 0$ and $d^2r/d\lambda^2 = 0$ at $r = r_0$.) Following this approach, Schmidt [26] was able to derive explicit expressions for E , L_z and Q in terms of p , e and z_- . We now briefly review Schmidt’s results.

Let us first introduce the dimensionless quantities

$$\tilde{E} \equiv E/\mu, \quad \tilde{L}_z \equiv L_z/(\mu M), \quad \tilde{Q} \equiv Q/(\mu M)^2, \quad (9)$$

$$\tilde{a} \equiv a/M, \quad \tilde{r} \equiv r/M, \quad \tilde{\Delta} \equiv \Delta/M^2, \quad (10)$$

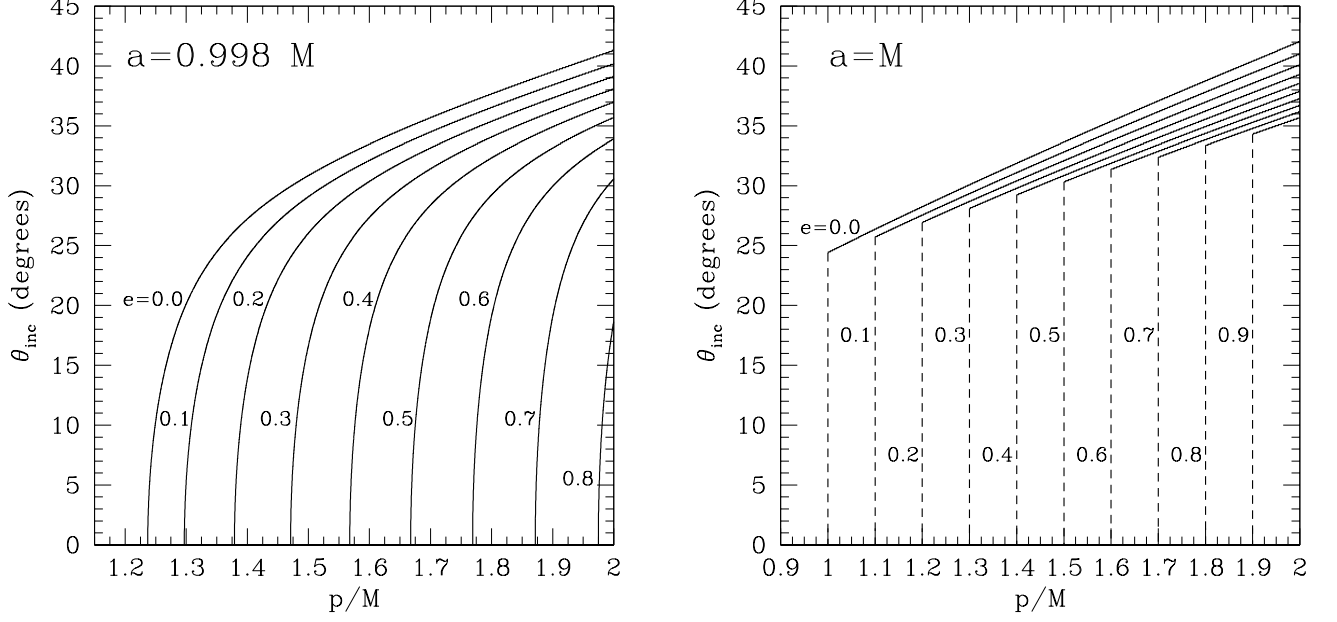


Figure 1: *Left panel:* Inclination angles θ_{inc} for which bound stable orbits exist for a black hole with spin $a = 0.998 M$. The allowed range for θ_{inc} goes from $\theta_{\text{inc}} = 0$ to the curve corresponding to the eccentricity under consideration, $\theta_{\text{inc}} = \theta_{\text{inc}}^{\text{max}}$. *Right panel:* Same as left but for an extremal black hole, $a = M$. Note that in this case $\theta_{\text{inc}}^{\text{max}}$ never reaches zero.

and the functions

$$f(\tilde{r}) \equiv \tilde{r}^4 + \tilde{a}^2 [\tilde{r}(\tilde{r} + 2) + z_- \tilde{\Delta}] , \quad (11)$$

$$g(\tilde{r}) \equiv 2 \tilde{a} \tilde{r} , \quad (12)$$

$$h(\tilde{r}) \equiv \tilde{r}(\tilde{r} - 2) + \frac{z_-}{1 - z_-} \tilde{\Delta} , \quad (13)$$

$$d(\tilde{r}) \equiv (\tilde{r}^2 + \tilde{a}^2 z_-) \tilde{\Delta} . \quad (14)$$

Let us further define the set of functions

$$(f_1, g_1, h_1, d_1) \equiv \begin{cases} (f(\tilde{r}_p), g(\tilde{r}_p), h(\tilde{r}_p), d(\tilde{r}_p)) & \text{if } e > 0 , \\ (f(\tilde{r}_0), g(\tilde{r}_0), h(\tilde{r}_0), d(\tilde{r}_0)) & \text{if } e = 0 , \end{cases} \quad (15)$$

$$(f_2, g_2, h_2, d_2) \equiv \begin{cases} (f(\tilde{r}_a), g(\tilde{r}_a), h(\tilde{r}_a), d(\tilde{r}_a)) & \text{if } e > 0 , \\ (f'(\tilde{r}_0), g'(\tilde{r}_0), h'(\tilde{r}_0), d'(\tilde{r}_0)) & \text{if } e = 0 , \end{cases} \quad (16)$$

and the determinants

$$\kappa \equiv d_1 h_2 - d_2 h_1 , \quad (17)$$

$$\varepsilon \equiv d_1 g_2 - d_2 g_1 , \quad (18)$$

$$\rho \equiv f_1 h_2 - f_2 h_1 , \quad (19)$$

$$\eta \equiv f_1 g_2 - f_2 g_1 , \quad (20)$$

$$\sigma \equiv g_1 h_2 - g_2 h_1 . \quad (21)$$

The energy of the particle can then be written

$$\tilde{E} = \sqrt{\frac{\kappa \rho + 2 \varepsilon \sigma - 2 D \sqrt{\sigma(\sigma \varepsilon^2 + \rho \varepsilon \kappa - \eta \kappa^2)}}{\rho^2 + 4 \eta \sigma}} . \quad (22)$$

The parameter D takes the values ± 1 . The angular momentum is a solution of the system

$$f_1 \tilde{E}^2 - 2 g_1 \tilde{E} \tilde{L}_z - h_1 \tilde{L}_z^2 - d_1 = 0 , \quad (23)$$

$$f_2 \tilde{E}^2 - 2 g_2 \tilde{E} \tilde{L}_z - h_2 \tilde{L}_z^2 - d_2 = 0 . \quad (24)$$

By eliminating the \tilde{L}_z^2 terms in these equations, one finds the solution

$$\tilde{L}_z = \frac{\rho \tilde{E}^2 - \kappa}{2 \tilde{E} \sigma} \quad (25)$$

for the angular momentum. Using $d\theta/d\lambda = 0$ at $\theta = \theta_{\text{min}}$, the Carter constant can be written

$$\tilde{Q} = z_- \left[\tilde{a}^2 (1 - \tilde{E}^2) + \frac{\tilde{L}_z^2}{1 - z_-} \right] . \quad (26)$$

Additional constraints on p , e , z_- are needed for the orbits to be stable. Inspection of Eq. (4) shows that an eccentric orbit is stable only if

$$\frac{\partial V_r}{\partial r}(r_{\text{peri}}) > 0 . \quad (27)$$

It is marginally stable if $\partial V_r / \partial r = 0$ at $r = r_{\text{peri}}$. Similarly, the stability condition for circular orbits is

$$\frac{\partial^2 V_r}{\partial r^2}(r_0) < 0 ; \quad (28)$$

marginally stable orbits are set by $\partial^2 V_r / \partial r^2 = 0$ at $r = r_0$.

Finally, we note that one can massage the above solutions for the conserved orbital quantities of bound stable orbits to rewrite the solution for \tilde{L}_z as

$$\tilde{L}_z = -\frac{g_1 \tilde{E}}{h_1} + \frac{D}{h_1} \sqrt{g_1^2 \tilde{E}^2 + (f_1 \tilde{E}^2 - d_1) h_1} . \quad (29)$$

From this solution, we see that it is quite natural to refer to orbits with $D = 1$ as *prograde* and to orbits with $D = -1$ as *retrograde*. Note also that Eq. (29) is a more useful form than the corresponding expression, Eq. (A4), of Ref. [21]. In that expression, the factor $1/h_1$ has been squared and moved inside the square root. This obscures the fact that h_1 changes sign for very strong field orbits. Differences between Eq. (29) and Eq. (A4) of [21] are apparent for $a \gtrsim 0.835$, although only for orbits close to the separatrix (*i.e.*, the surface in the space of parameters (p, e, ι) where marginally stable bound orbits lie).

III. NON-CIRCULAR NEARLY HORIZON-SKIMMING ORBITS

With explicit expressions for E , L_z and Q as functions of p , e and z_- , we now examine how to generalize the condition $\partial L_z(r, \iota)/\partial \iota > 0$, which defined circular nearly horizon-skimming orbits in Ref. [14], to encompass the non-circular case. We recall that the inclination angle ι is defined as [14]

$$\cos \iota = \frac{L_z}{\sqrt{Q + L_z^2}} . \quad (30)$$

Such a definition is not always easy to handle in the case of eccentric orbits. In addition, ι does not have an obvious physical interpretation (even in the circular limit), but rather was introduced essentially to generalize (at least formally) the definition of inclination for Schwarzschild black hole orbits. In that case, one has $Q = L_x^2 + L_y^2$ and therefore $L_z = |L| \cos \iota$.

A more useful definition for the inclination angle in Kerr was introduced in Ref. [21]:

$$\theta_{\text{inc}} = \frac{\pi}{2} - D \theta_{\text{min}} , \quad (31)$$

where θ_{min} is the minimum reached by θ during the orbital motion. This angle is trivially related to z_- ($z_- = \sin^2 \theta_{\text{inc}}$) and ranges from 0 to $\pi/2$ for prograde orbits and from $\pi/2$ to π for retrograde orbits. It is a simple numerical calculation to convert between ι and θ_{inc} ; doing so shows that the differences between ι and θ_{inc} are very small, with the two coinciding for $a = 0$, and with a difference that is less than 2.6° for $a = M$ and circular orbits with $r = M$.

Bearing all this in mind, the condition we have adopted to generalize nearly horizon-skimming orbits is

$$\frac{\partial L_z(p, e, \theta_{\text{inc}})}{\partial \theta_{\text{inc}}} > 0 . \quad (32)$$

We have found that certain parts of this calculation, particular the analysis of strong-field geodesic orbits, are best done using the angle θ_{inc} ; other parts are more simply done using the

angle ι , particularly the “kludge” computation of fluxes described in Sec. V. (This is because the kludge fluxes are based on an extension of post-Newtonian formulas to the strong-field regime, and these formulas use ι for inclination angle.) Accordingly, we often switch back and forth between these two notions of inclination, and in fact present our final results for inclination evolution using both $d\iota/dt$ and $d\theta_{\text{inc}}/dt$.

Before mapping out the region corresponding to nearly horizon-skimming orbits, it is useful to examine stable orbits more generally in the strong field of rapidly rotating black holes. We first fix a value for a , and then discretize the space of parameters $(p, e, \theta_{\text{inc}})$. We next identify the points in this space corresponding to bound stable geodesic orbits. Sufficiently close to the horizon, the bound stable orbits with specified values of p and e have an inclination angle θ_{inc} ranging from 0 (equatorial orbit) to a maximum value $\theta_{\text{inc}}^{\text{max}}$. For given p and e , $\theta_{\text{inc}}^{\text{max}}$ defines the *separatrix* between stable and unstable orbits.

Example separatrices are shown in Fig. 1 for $a = 0.998M$ and $a = M$. This figure shows the behavior of $\theta_{\text{inc}}^{\text{max}}$ as a function of the latus rectum for the different values of the eccentricity indicated by the labels. Note that for $a = 0.998M$ the angle $\theta_{\text{inc}}^{\text{max}}$ eventually goes to zero. This is the general behavior for $a < M$. On the other hand, for an extremal black hole, $a = M$, $\theta_{\text{inc}}^{\text{max}}$ never goes to zero. The orbits which reside at $r = M$ (the circular limit) are the “horizon-skimming orbits” identified by Wilkins [12]; the $a = M$ separatrix has a similar shape even for eccentric orbits. As expected, we find that for given latus rectum and eccentricity the orbit with $\theta_{\text{inc}} = 0$ is the one with the lowest energy E (and hence is the most-bound orbit), whereas the orbit with $\theta_{\text{inc}} = \theta_{\text{inc}}^{\text{max}}$ has the highest E (and is least bound).

Having mapped out stable orbits in $(p, e, \theta_{\text{inc}})$ space, we then computed the partial derivative $\partial L_z(p, e, \theta_{\text{inc}})/\partial \theta_{\text{inc}}$ and identified the following three overlapping regions:

- *Region A*: The portion of the (p, e) plane for which $\partial L_z(p, e, \theta_{\text{inc}})/\partial \theta_{\text{inc}} > 0$ for $0 \leq \theta_{\text{inc}} \leq \theta_{\text{inc}}^{\text{max}}$. This region is illustrated in Fig. 2 as the area under the heavy solid line and to the left of the dot-dashed line (green in the color version).
- *Region B*: The portion of the (p, e) plane for which $(L_z)_{\text{most bound}}(p, e)$ is smaller than $(L_z)_{\text{least bound}}(p, e)$. In other words,

$$L_z(p, e, 0) < L_z(p, e, \theta_{\text{inc}}^{\text{max}}) \quad (33)$$

in Region *B*. Note that Region *B* contains Region *A*. It is illustrated in Fig. 2 as the area under the heavy solid line and to the left of the dotted line (red in the color version).

- *Region C*: The portion of the (p, e) plane for which $\partial L_z(p, e, \theta_{\text{inc}})/\partial \theta_{\text{inc}} > 0$ for at least one angle θ_{inc} between 0 and $\theta_{\text{inc}}^{\text{max}}$. Region *C* contains Region *B*, and is illustrated in Fig. 2 as the area under the heavy solid line and to the left of the dashed line (blue in the color version).

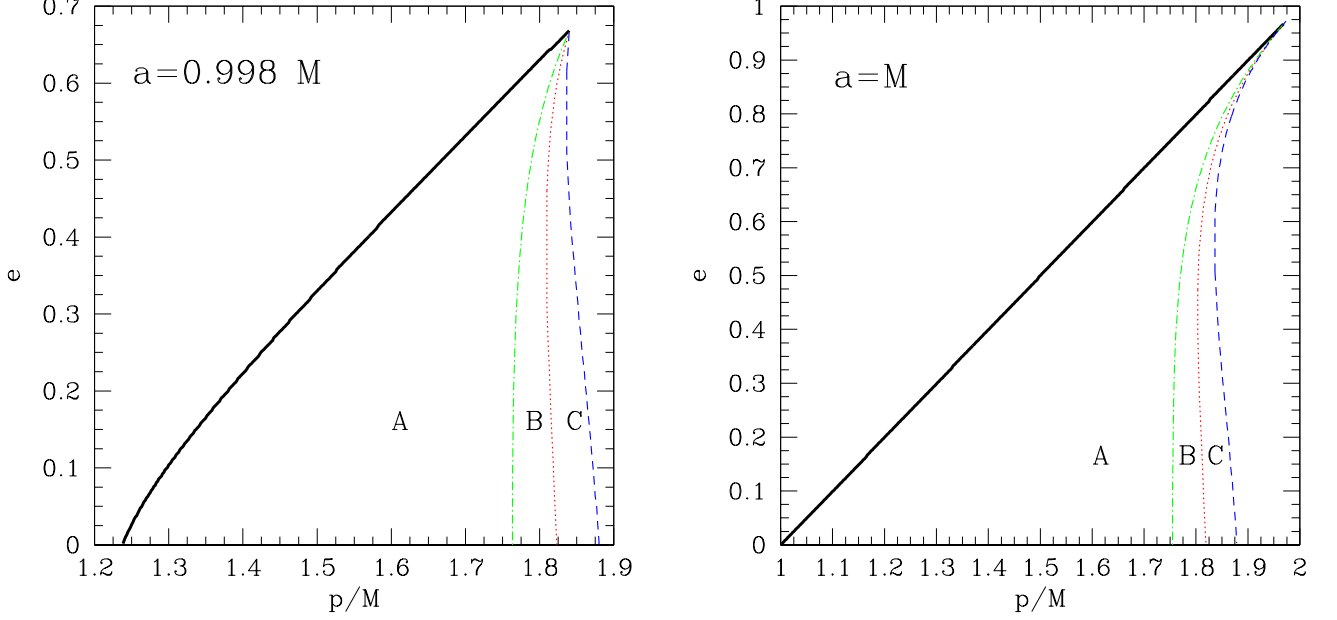


Figure 2: *Left panel*: Non-circular nearly horizon-skimming orbits for $a = 0.998M$. The heavy solid line indicates the separatrix between stable and unstable orbits for equatorial orbits ($\iota = \theta_{\text{inc}} = 0$). All orbits above and to the left of this line are unstable. The dot-dashed line (green in the color version) bounds the region of the (p, e) -plane where $\partial L_z / \partial \theta_{\text{inc}} > 0$ for all allowed inclination angles (“Region A”). All orbits between this line and the separatrix belong to Region A. The dotted line (red in the color version) bounds the region $(L_z)_{\text{most bound}} < (L_z)_{\text{least bound}}$ (“Region B”). Note that B includes A. The dashed line (blue in the color version) bounds the region where $\partial L_z / \partial \theta_{\text{inc}} > 0$ for at least one inclination angle (“Region C”); note that C includes B. All three of these regions are candidate generalizations of the notion of nearly horizon-skimming orbits. *Right panel*: Same as the left panel, but for the extreme spin case, $a = M$. In this case the separatrix between stable and unstable equatorial orbits is given by the line $p/M = 1 + e$.

Orbits in any of these three regions give possible generalizations of the nearly horizon-skimming circular orbits presented in Ref. [14]. Notice, as illustrated in Fig. 2, that the size of these regions depends rather strongly on the spin of the black hole. All three regions disappear altogether for $a < 0.9524M$ (in agreement with [14]); their sizes grow with a , reaching maximal extent for $a = M$. These regions never extend beyond $p \simeq 2M$.

As we shall see, the difference between these three regions is not terribly important for assessing whether there is a strong signature of the nearly-horizon skimming regime on the inspiral dynamics. As such, it is perhaps most useful to use Region C as our definition, since it is the most inclusive.

IV. EVOLUTION OF θ_{inc} : CIRCULAR ORBITS

To ascertain whether nearly horizon-skimming orbits can affect an EMRI in such a way as to leave a clear imprint in the gravitational-wave signal, we have studied the time evolution of the inclination angle θ_{inc} . This requires us to compute the rates of change dE/dt , dL_z/dt and dQ/dt due to gravitational-wave emission (“radiation reaction”). As discussed in Sec. II, E , L_z and Q can be expressed in terms of p , e , and θ_{inc} . Given rates of change of the orbital constants, it is then straightforward [23] to calculate dp/dt , de/dt , and $d\theta_{\text{inc}}/dt$ (or $d\iota/dt$).

Accurate calculation of dE/dt and dL_z/dt involves solving the Teukolsky and Sasaki-Nakamura equations [27, 28]. For generic orbits this has been done for the first time in Ref. [21]. The calculation of dQ/dt for generic orbits is more involved. A formula for dQ/dt has been recently derived [22], but has not yet been implemented (at least in a code to which we have access).

On the other hand, it is well-known that a circular orbit will remain circular under radiation reaction [29, 30, 31]. This constraint means that Teukolsky-based fluxes for E and L_z are sufficient to compute dQ/dt . Considering this limit, the rate of change dQ/dt can be expressed in terms of dE/dt and dL_z/dt as

$$\left(\frac{dQ}{dt}\right)_{\text{circ}} = -\frac{N_1(p, \iota)}{N_5(p, \iota)} \left(\frac{dE}{dt}\right)_{\text{circ}} - \frac{N_4(p, \iota)}{N_5(p, \iota)} \left(\frac{dL_z}{dt}\right)_{\text{circ}} \quad (34)$$

where

$$N_1(p, \iota) \equiv E(p, \iota) p^4 + a^2 E(p, \iota) p^2 - 2 a M (L_z(p, \iota) - a E(p, \iota)) p, \quad (35)$$

$$N_4(p, \iota) \equiv (2 M p - p^2) L_z(p, \iota) - 2 M a E(p, \iota) p, \quad (36)$$

$$N_5(p, \iota) \equiv (2 M p - p^2 - a^2)/2. \quad (37)$$

(These quantities are for a circular orbit of radius p .) Using this, it is simple to compute $d\theta_{\text{inc}}/dt$ (or $d\iota/dt$).

This procedure was followed in Ref. [14], using the code presented in Ref. [17], to determine the evolution of ι ; this analysis indicated that $d\iota/dt < 0$ for circular nearly horizon-skimming orbits. As a first step to our more general analysis, we have repeated this calculation but using the improved Sasaki-Nakamura-Teukolsky code presented in Ref. [21]; we focused on the case $a = 0.998M$.

Rather to our surprise, we discovered that the fluxes dE/dt and dL_z/dt computed with this more accurate code indicate that $d\iota/dt > 0$ (and $d\theta_{\text{inc}}/dt > 0$) for *all* circular nearly horizon-skimming orbits — in stark contrast with what was found in Ref. [14]. As mentioned in the introduction, the rate of change of inclination angle appears as the difference of two quantities. These quantities nearly cancel (and indeed cancel exactly in the limit $a = 0$); as such, small relative errors in their values can lead to large relative error in the inferred inclination evolution. Values for dE/dt , dL_z/dt , $d\iota/dt$, and $d\theta_{\text{inc}}/dt$ computed using the present code are shown in Table I in the columns with the header “Teukolsky”.

V. EVOLUTION OF θ_{inc} : NON-CIRCULAR ORBITS

The corrected behavior of circular nearly horizon-skimming orbits has naturally led us to investigate the evolution of non-circular nearly horizon-skimming orbits. Since our code cannot be used to compute dQ/dt , we have resorted to a “kludge” approach, based on those described in Refs. [23, 24]. In particular, we mostly follow the procedure developed by Gair & Glampedakis [24], though (as described below) importantly modified.

The basic idea of the “kludge” procedure is to use the functional form of 2PN fluxes E , L_z and Q , but to correct the circular part of these fluxes using fits to circular Teukolsky data. As developed in Ref. [24], the fluxes are written

$$\left(\frac{dE}{dt}\right)_{\text{GG}} = (1 - e^2)^{3/2} \left[(1 - e^2)^{-3/2} \left(\frac{dE}{dt}\right)_{2\text{PN}}(p, e, \iota) - \left(\frac{dE}{dt}\right)_{2\text{PN}}(p, 0, \iota) + \left(\frac{dE}{dt}\right)_{\text{fit circ}}(p, \iota) \right], \quad (38)$$

$$\left(\frac{dL_z}{dt}\right)_{\text{GG}} = (1 - e^2)^{3/2} \left[(1 - e^2)^{-3/2} \left(\frac{dL_z}{dt}\right)_{2\text{PN}}(p, e, \iota) - \left(\frac{dL_z}{dt}\right)_{2\text{PN}}(p, 0, \iota) + \left(\frac{dL_z}{dt}\right)_{\text{fit circ}}(p, \iota) \right], \quad (39)$$

$$\left(\frac{dQ}{dt}\right)_{\text{GG}} = (1 - e^2)^{3/2} \sqrt{Q(p, e, \iota)} \times \left[(1 - e^2)^{-3/2} \left(\frac{dQ/dt}{\sqrt{Q}}\right)_{2\text{PN}}(p, e, \iota) - \left(\frac{dQ/dt}{\sqrt{Q}}\right)_{2\text{PN}}(p, 0, \iota) + \left(\frac{dQ/dt}{\sqrt{Q}}\right)_{\text{fit circ}}(p, \iota) \right]. \quad (40)$$

The post-Newtonian fluxes $(dE/dt)_{2\text{PN}}$, $(dL_z/dt)_{2\text{PN}}$ and $(dQ/dt)_{2\text{PN}}$ are given in the Appendix [particularly Eqs. (A.45), (A.46), and (A.47)].

Since for circular orbits the fluxes dE/dt , dL_z/dt and dQ/dt are related through Eq. (34), only two fits to circular Teukolsky data are needed. One possible choice is to fit dL_z/dt and $d\iota/dt$, and then use the circularity constraint to obtain² [24]

$$\left(\frac{dQ/dt}{\sqrt{Q}}\right)_{\text{fit circ}}(p, \iota) = 2 \tan \iota \left[\left(\frac{dL_z}{dt}\right)_{\text{fit circ}} + \frac{\sqrt{Q(p, 0, \iota)}}{\sin^2 \iota} \left(\frac{d\iota}{dt}\right)_{\text{fit circ}} \right], \quad (41)$$

$$\left(\frac{dE}{dt}\right)_{\text{fit circ}}(p, \iota) = -\frac{N_4(p, \iota)}{N_1(p, \iota)} \left(\frac{dL_z}{dt}\right)_{\text{fit circ}}(p, \iota) - \frac{N_5(p, \iota)}{N_1(p, \iota)} \sqrt{Q(p, 0, \iota)} \left(\frac{dQ/dt}{\sqrt{Q}}\right)_{\text{fit circ}}(p, \iota). \quad (42)$$

As stressed in Ref. [24], one does not expect these fluxes to work well in the strong field, both because the post-Newtonian approximation breaks down close to the black hole, and because the circular Teukolsky data used for the fits in Ref. [24] was computed for $3M \leq p \leq 30M$. As a first attempt to improve their behavior in the nearly horizon-skimming region, we have made fits using circular Teukolsky data for orbits with $M < p \leq 2M$. In particular, for a black hole with $a = 0.998M$, we computed the circular Teukolsky-based fluxes dL_z/dt and $d\iota/dt$ listed in Table I (columns 8 and 10). These results were fit (with error $\lesssim 0.2\%$); see Eqs. (A.48) and (A.49) in the Appendix.

Despite using strong-field Teukolsky fluxes for our fit, we found fairly poor behavior of these rates of change, particularly as a function of eccentricity. To compensate for this, we introduced a kludge-type fit to correct the equatorial part of the flux, in addition to the circular part. We fit, as a function of p and e , Teukolsky-based fluxes for dE/dt and dL_z/dt for orbits in the equatorial plane, and then introduce the following kludge fluxes for E and L_z :

$$\frac{dE}{dt}(p, e, \iota) = \left(\frac{dE}{dt}\right)_{\text{GG}}(p, e, \iota) - \left(\frac{dE}{dt}\right)_{\text{GG}}(p, e, 0) + \left(\frac{dE}{dt}\right)_{\text{fit eq}}(p, e) \quad (43)$$

$$\frac{dL_z}{dt}(p, e, \iota) = \left(\frac{dL_z}{dt}\right)_{\text{GG}}(p, e, \iota) - \left(\frac{dL_z}{dt}\right)_{\text{GG}}(p, e, 0) + \left(\frac{dL_z}{dt}\right)_{\text{fit eq}}(p, e). \quad (44)$$

² This choice might seem more involved than fitting directly dL_z/dt and dQ/dt , but, as noted by Gair & Glampedakis, ensures more sensible results for the evolution of the inclination angle. This generates more physically realistic inspirals [24].

[Note that Eq. (40) for dQ/dt is not modified by this procedure since $dQ/dt = 0$ for equatorial orbits.] Using equatorial non-circular Teukolsky data provided by Drasco [21, 32] for $a = 0.998$ and $M < p \leq 2M$ (the $\iota = 0$ “Teukolsky” data in Tables II, III and IV), we found fits (with error $\lesssim 1.5\%$); see Eqs. (A.52) and (A.53). Note that the fits for equatorial fluxes are significantly less accurate than the fits for circular fluxes. This appears to be due to the fact that, close to the black hole, *many* harmonics are needed in order for the Teukolsky-based fluxes to converge, especially for eccentric orbits (cf. Figs. 2 and 3 of Ref. [21], noting the number of radial harmonics that have significant contribution to the flux). Truncation of these sums is likely a source of some error in the fluxes themselves, making it difficult to make a fit of as high quality as we could in the circular case.

These fits were then finally used in Eqs. (43) and (44) to calculate the kludge fluxes dE/dt and dL_z/dt for generic orbits. This kludge reproduces to high accuracy our fits to the Teukolsky-based fluxes for circular orbits ($e = 0$) or equatorial orbits ($\iota = 0$). Some residual error remains because the $\iota = 0$ limit of the circular fits do not precisely equal the $e = 0$ limit of the equatorial fits.

Table I compares our kludge to Teukolsky-based fluxes for circular orbits; the two methods agree to several digits. Tables II, III and IV compare our kludge to the generic Teukolsky-based fluxes provided by Drasco [21, 32]. In all cases, the kludge fluxes dE/dt and dL_z/dt have the correct qualitative behavior, being negative for all the orbital parameters under consideration ($a = 0.998M$, $1 < p/M \leq 2$, $0 \leq e \leq 0.5$ and $0^\circ \leq \iota \leq 41^\circ$). The relative difference between the kludge and Teukolsky fluxes is always less than 25% for $e = 0$ and $e = 0.1$ (even for orbits very close to separatrix). The accuracy remains good at larger eccentricity, though it degrades somewhat as orbits come close to the separatrix.

Tables I, II, III and IV also present the kludge values of the fluxes $d\iota/dt$ and $d\theta_{\text{inc}}/dt$ as computed using Eqs. (43) and (44) for dE/dt and dL_z/dt , plus Eq. (40) for dQ/dt . Though certainly not the last word on inclination evolution (pending rigorous computation of dQ/dt), these rates of change probably represent a better approximation than results published to date in the literature. (Indeed, prior work has often used the crude approximation $d\iota/dt = 0$ [21] to estimate dQ/dt given dE/dt and dL_z/dt .)

Most significantly, we find that $(d\iota/dt)_{\text{kludge}} > 0$ and $(d\theta_{\text{inc}}/dt)_{\text{kludge}} > 0$ for all of the orbital parameters we consider. In other words, we find that $d\iota/dt$ and $d\theta_{\text{inc}}/dt$ do not ever change sign.

Finally, in Table V we compute the changes in θ_{inc} and ι for the inspiral with mass ratio $\mu/M = 10^{-6}$. In all cases, we start at $p/M = 1.9$. The small body then inspirals through the nearly horizon-skimming region until it reaches the separatrix; at this point, the small body will fall into the large black hole on a dynamical timescale $\sim M$, so we terminate the calculation. The evolution of circular orbits is computed using our fits to the circular-Teukolsky fluxes of E and L_z ; for eccentric orbits we use the kludge fluxes (40), (43) and (44). As this exercise demonstrates, the change in inclination during inspiral is never larger than a few degrees. Not only is

there no unique sign change in the nearly horizon-skimming region, but the magnitude of the inclination change remains puny. This leaves little room for the possibility that this class of orbits may have a clear observational imprint on the EMRI-waveforms to be detected by LISA.

VI. CONCLUSIONS

We have performed a detailed analysis of the orbital motion near the horizon of near-extremal Kerr black holes. We have demonstrated the existence of a class of orbits, which we have named “non-circular nearly horizon-skimming orbits”, for which the angular momentum L_z increases with the orbit’s inclination, while keeping latus rectum and eccentricity fixed. This behavior, in stark contrast to that of Newtonian orbits, generalizes earlier results for circular orbits [14].

Furthermore, to assess whether this class of orbits can produce a unique imprint on EMRI waveforms (an important source for future LISA observations), we have studied the radiative evolution of inclination angle for a small body orbiting in the nearly horizon-skimming region. For circular orbits, we have re-examined the analysis of Ref. [14] using an improved code for computing Teukolsky-based fluxes of the energy and angular momentum. Significantly correcting Ref. [14]’s results, we found *no* decrease in the orbit’s inclination angle. Inclination always increases during inspiral.

We next carried out such an analysis for eccentric nearly horizon-skimming orbits. In this case, we used “kludge” fluxes to evolve the constants of motion E , L_z and Q [24]. We find that these fluxes are fairly accurate when compared with the available Teukolsky-based fluxes, indicating that they are likely to provide reliable information regarding inclination evolution. As for circular orbits, we find that the orbit’s inclination never decreases. For both circular and non-circular configurations, we find that the magnitude of the inclination change is quite paltry — only a few degrees at most.

Quite generically, therefore, we found that the inclination angle of both circular and eccentric nearly horizon-skimming orbits never decreases during the inspiral. Revising the results obtained in Ref. [14], we thus conclude that such orbits are not likely to yield a peculiar, unique imprint on the EMRI-waveforms detectable by LISA.

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In this Appendix we report the expressions for the post-Newtonian fluxes and the fits to the Teukolsky data necessary

to compute the kludge fluxes introduced in Sec. V. In particular the 2PN fluxes are given by [24]

$$\left(\frac{dE}{dt}\right)_{2\text{PN}} = -\frac{32}{5} \frac{\mu^2}{M^2} \left(\frac{M}{p}\right)^5 (1-e^2)^{3/2} \left[g_1(e) - \tilde{a} \left(\frac{M}{p}\right)^{3/2} g_2(e) \cos \iota - \left(\frac{M}{p}\right) g_3(e) + \pi \left(\frac{M}{p}\right)^{3/2} g_4(e) - \left(\frac{M}{p}\right)^2 g_5(e) + \tilde{a}^2 \left(\frac{M}{p}\right)^2 g_6(e) - \frac{527}{96} \tilde{a}^2 \left(\frac{M}{p}\right)^2 \sin^2 \iota \right], \quad (\text{A.45})$$

$$\left(\frac{dL_z}{dt}\right)_{2\text{PN}} = -\frac{32}{5} \frac{\mu^2}{M} \left(\frac{M}{p}\right)^{7/2} (1-e^2)^{3/2} \left[g_9(e) \cos \iota + \tilde{a} \left(\frac{M}{p}\right)^{3/2} (g_{10}^a(e) - \cos^2 \iota g_{10}^b(e)) - \left(\frac{M}{p}\right) g_{11}(e) \cos \iota + \pi \left(\frac{M}{p}\right)^{3/2} g_{12}(e) \cos \iota - \left(\frac{M}{p}\right)^2 g_{13}(e) \cos \iota + \tilde{a}^2 \left(\frac{M}{p}\right)^2 \cos \iota \left(g_{14}(e) - \frac{45}{8} \sin^2 \iota \right) \right], \quad (\text{A.46})$$

$$\left(\frac{dQ}{dt}\right)_{2\text{PN}} = -\frac{64}{5} \frac{\mu^2}{M} \left(\frac{M}{p}\right)^{7/2} \sqrt{Q} \sin \iota (1-e^2)^{3/2} \left[g_9(e) - \tilde{a} \left(\frac{M}{p}\right)^{3/2} \cos \iota g_{10}^b(e) - \left(\frac{M}{p}\right) g_{11}(e) + \pi \left(\frac{M}{p}\right)^{3/2} g_{12}(e) - \left(\frac{M}{p}\right)^2 g_{13}(e) + \tilde{a}^2 \left(\frac{M}{p}\right)^2 \times \left(g_{14}(e) - \frac{45}{8} \sin^2 \iota \right) \right], \quad (\text{A.47})$$

where μ is the mass of the infalling body and where the various e -dependent coefficients are

$$\begin{aligned} g_1(e) &\equiv 1 + \frac{73}{24} e^2 + \frac{37}{96} e^4, & g_2(e) &\equiv \frac{73}{12} + \frac{823}{24} e^2 + \frac{949}{32} e^4 + \frac{491}{192} e^6, & g_3(e) &\equiv \frac{1247}{336} + \frac{9181}{672} e^2, \\ g_4(e) &\equiv 4 + \frac{1375}{48} e^2, & g_5(e) &\equiv \frac{44711}{9072} + \frac{172157}{2592} e^2, & g_6(e) &\equiv \frac{33}{16} + \frac{359}{32} e^2, & g_9(e) &\equiv 1 + \frac{7}{8} e^2, \\ g_{10}^a(e) &\equiv \frac{61}{24} + \frac{63}{8} e^2 + \frac{95}{64} e^4, & g_{10}^b(e) &\equiv \frac{61}{8} + \frac{91}{4} e^2 + \frac{461}{64} e^4, & g_{11}(e) &\equiv \frac{1247}{336} + \frac{425}{336} e^2, \\ g_{12}(e) &\equiv 4 + \frac{97}{8} e^2, & g_{13}(e) &\equiv \frac{44711}{9072} + \frac{302893}{6048} e^2, & g_{14}(e) &\equiv \frac{33}{16} + \frac{95}{16} e^2, \end{aligned}$$

The fits to the circular-Teukolsky data of Table I are instead given by

$$\begin{aligned} \left(\frac{dL_z}{dt}\right)_{\text{fit circ}}(p, \iota) &= -\frac{32}{5} \frac{\mu^2}{M} \left(\frac{M}{p}\right)^{7/2} \left\{ \cos \iota + \left(\frac{M}{p}\right)^{3/2} \left(\frac{61}{24} - \frac{61}{8} \cos^2 \iota + 4\pi \cos \iota \right) - \frac{1247}{336} \left(\frac{M}{p}\right) \cos \iota \right. \\ &\quad + \left(\frac{M}{p}\right)^2 \cos \iota \left(-\frac{1625}{567} - \frac{45}{8} \sin^2 \iota \right) + \left(\frac{M}{p}\right)^{5/2} \left[\tilde{d}_1(p/M) + \tilde{d}_2(p/M) \cos \iota + \tilde{d}_3(p/M) \cos^2 \iota \right. \\ &\quad \left. \left. + \tilde{d}_4(p/M) \cos^3 \iota + \tilde{d}_5(p/M) \cos^4 \iota + \tilde{d}_6(p/M) \cos^5 \iota + \cos \iota \left(\frac{M}{p}\right)^{3/2} (A + B \cos^2 \iota) \right] \right\}, \quad (\text{A.48}) \end{aligned}$$

$$\begin{aligned} \left(\frac{d\iota}{dt}\right)_{\text{fit circ}}(p, \iota) &= \frac{32}{5} \frac{\mu^2}{M} \frac{\sin^2 \iota}{\sqrt{Q(p, 0, \iota)}} \left(\frac{M}{p}\right)^5 \left\{ \frac{61}{24} + \left(\frac{M}{p}\right) \tilde{d}_1(p/M) + \cos \iota \left(\frac{M}{p}\right)^{1/2} \times \right. \\ &\quad \left[a_d^7 + b_d^7 \left(\frac{M}{p}\right) + c_d^7 \left(\frac{M}{p}\right)^{3/2} \right] + \cos^2 \iota \left(\frac{M}{p}\right) \tilde{d}_8(p/M) + \cos \iota \left(\frac{M}{p}\right)^{5/2} \left[\tilde{h}_1(p/M) + \cos^2 \iota \tilde{h}_2(p/M) \right] \right\}, \quad (\text{A.49}) \end{aligned}$$

where

$$\tilde{d}_i(x) \equiv a_d^i + b_d^i x^{-1/2} + c_d^i x^{-1}, \quad i = 1, \dots, 8, \quad \tilde{h}_i(x) \equiv a_h^i + b_h^i x^{-1/2}, \quad i = 1, 2 \quad (\text{A.50})$$

and the numerical coefficients are given by

$$a_h^1 = -278.9387, \quad b_h^1 = 84.1414, \quad a_h^2 = 8.6679, \quad b_h^2 = -9.2401, \quad A = -18.3362, \quad B = 24.9034, \quad (\text{A.51})$$

and by the following table

i	1	2	3	4	5	6	7	8
a_d^i	15.8363	445.4418	-2027.7797	3089.1709	-2045.2248	498.6411	-8.7220	50.8345
b_d^i	-55.6777	-1333.2461	5940.4831	-9103.4472	6113.1165	-1515.8506	-50.8950	-131.6422
c_d^i	38.6405	1049.5637	-4513.0879	6926.3191	-4714.9633	1183.5875	251.4025	83.0834

Note that the functional form of these fits was obtained from Eqs. (57) and (58) of Ref. [24] by setting \tilde{a} (*i.e.*, q in their notation) to 1. Finally, we give expressions for the fits to the equatorial Teukolsky data of tables II, III and IV (data with $\iota = 0$, columns with header “Teukolsky”):

$$\left(\frac{dE}{dt}\right)_{\text{fit eq}}(p, e) = \left(\frac{dE}{dt}\right)_{2PN}(p, e, 0) - \frac{32}{5} \left(\frac{\mu}{M}\right)^2 \left(\frac{M}{p}\right)^5 (1 - e^2)^{3/2} \left[\tilde{g}_1(e) + \tilde{g}_2(e) \left(\frac{M}{p}\right)^{1/2} + \tilde{g}_3(e) \left(\frac{M}{p}\right) + \tilde{g}_4(e) \left(\frac{M}{p}\right)^{3/2} + \tilde{g}_5(e) \left(\frac{M}{p}\right)^2 \right], \quad (\text{A.52})$$

$$\left(\frac{L_z}{dt}\right)_{\text{fit eq}}(p, e) = \left(\frac{L_z}{dt}\right)_{2PN}(p, e, 0) - \frac{32}{5} \frac{\mu^2}{M} \left(\frac{M}{p}\right)^{7/2} (1 - e^2)^{3/2} \left[\tilde{f}_1(e) + \tilde{f}_2(e) \left(\frac{M}{p}\right)^{1/2} + \tilde{f}_3(e) \left(\frac{M}{p}\right) + \tilde{f}_4(e) \left(\frac{M}{p}\right)^{3/2} + \tilde{f}_5(e) \left(\frac{M}{p}\right)^2 \right], \quad (\text{A.53})$$

where

$$\tilde{g}_i(e) \equiv a_g^i + b_g^i e^2 + c_g^i e^4 + d_g^i e^6, \quad \tilde{f}_i(e) \equiv a_f^i + b_f^i e^2 + c_f^i e^4 + d_f^i e^6, \quad i = 1, \dots, 5 \quad (\text{A.54})$$

and the numerical coefficients are given by the following table

i	a_g^i	b_g^i	c_g^i	d_g^i	a_f^i	b_f^i	c_f^i	d_f^i
1	6.4590	-2038.7301	6639.9843	227709.2187	5.4577	-3116.4034	4711.7065	214332.2907
2	-31.2215	10390.6778	-27505.7295	-1224376.5294	-26.6519	15958.6191	-16390.4868	-1147201.4687
3	57.1208	-19800.4891	39527.8397	2463977.3622	50.4374	-30579.3129	15749.9411	2296989.5466
4	-49.7051	16684.4629	-21714.7941	-2199231.9494	-46.7816	25968.8743	656.3460	-2038650.9838
5	16.4697	-5234.2077	2936.2391	734454.5696	15.6660	-8226.3892	-4903.9260	676553.2755

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$\frac{p}{M}$	e	θ_{inc} (deg.)	ι (deg.)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (kludge)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (Teukolsky)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{d\iota}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{d\iota}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)
1.3	0	0	0	-9.108×10^{-2}	-9.109×10^{-2}	-2.258×10^{-1}	-2.259×10^{-1}	0	0	0	0
1.3	0	10.4870	11.6773	-9.328×10^{-2}	-9.332×10^{-2}	-2.304×10^{-1}	-2.306×10^{-1}	1.837×10^{-2}	1.839×10^{-2}	6.462×10^{-3}	6.475×10^{-3}
1.3	0	14.6406	16.1303	-9.588×10^{-2}	-9.588×10^{-2}	-2.359×10^{-1}	-2.360×10^{-1}	2.397×10^{-2}	2.400×10^{-2}	8.645×10^{-3}	8.667×10^{-3}
1.3	0	17.7000	19.3172	-9.875×10^{-2}	-9.876×10^{-2}	-2.420×10^{-1}	-2.421×10^{-1}	2.728×10^{-2}	2.731×10^{-2}	1.007×10^{-2}	1.010×10^{-2}
1.3	0	20.1636	21.8210	-1.019×10^{-1}	-1.019×10^{-1}	-2.486×10^{-1}	-2.488×10^{-1}	2.943×10^{-2}	2.950×10^{-2}	1.111×10^{-2}	1.117×10^{-2}
1.4	0	0	0	-8.700×10^{-2}	-8.709×10^{-2}	-2.311×10^{-1}	-2.312×10^{-1}	0	0	0	0
1.4	0	14.5992	16.0005	-9.062×10^{-2}	-9.070×10^{-2}	-2.386×10^{-1}	-2.386×10^{-1}	2.316×10^{-2}	2.319×10^{-2}	8.823×10^{-3}	8.848×10^{-3}
1.4	0	20.1756	21.7815	-9.520×10^{-2}	-9.526×10^{-2}	-2.482×10^{-1}	-2.482×10^{-1}	2.875×10^{-2}	2.877×10^{-2}	1.141×10^{-2}	1.143×10^{-2}
1.4	0	24.1503	25.7517	-1.006×10^{-1}	-1.007×10^{-1}	-2.595×10^{-1}	-2.596×10^{-1}	3.140×10^{-2}	3.141×10^{-2}	1.289×10^{-2}	1.288×10^{-2}
1.4	0	27.2489	28.7604	-1.067×10^{-1}	-1.068×10^{-1}	-2.725×10^{-1}	-2.725×10^{-1}	3.274×10^{-2}	3.275×10^{-2}	1.378×10^{-2}	1.377×10^{-2}
1.5	0	0	0	-8.009×10^{-2}	-7.989×10^{-2}	-2.270×10^{-1}	-2.265×10^{-1}	0	0	0	0
1.5	0	16.7836	18.1857	-8.401×10^{-2}	-8.383×10^{-2}	-2.348×10^{-1}	-2.343×10^{-1}	2.360×10^{-2}	2.351×10^{-2}	9.602×10^{-3}	9.545×10^{-3}
1.5	0	23.0755	24.6167	-8.917×10^{-2}	-8.897×10^{-2}	-2.454×10^{-1}	-2.449×10^{-1}	2.872×10^{-2}	2.863×10^{-2}	1.228×10^{-2}	1.222×10^{-2}
1.5	0	27.4892	28.9670	-9.537×10^{-2}	-9.516×10^{-2}	-2.583×10^{-1}	-2.579×10^{-1}	3.091×10^{-2}	3.082×10^{-2}	1.372×10^{-2}	1.367×10^{-2}
1.5	0	30.8795	32.2231	-1.025×10^{-1}	-1.023×10^{-1}	-2.733×10^{-1}	-2.728×10^{-1}	3.184×10^{-2}	3.173×10^{-2}	1.452×10^{-2}	1.443×10^{-2}
1.6	0	0	0	-7.181×10^{-2}	-7.156×10^{-2}	-2.168×10^{-1}	-2.162×10^{-1}	0	0	0	0
1.6	0	18.3669	19.7220	-7.568×10^{-2}	-7.545×10^{-2}	-2.242×10^{-1}	-2.237×10^{-1}	2.240×10^{-2}	2.229×10^{-2}	9.600×10^{-3}	9.515×10^{-3}
1.6	0	25.1720	26.6245	-8.084×10^{-2}	-8.062×10^{-2}	-2.346×10^{-1}	-2.341×10^{-1}	2.701×10^{-2}	2.685×10^{-2}	1.223×10^{-2}	1.210×10^{-2}
1.6	0	29.9014	31.2625	-8.708×10^{-2}	-8.687×10^{-2}	-2.474×10^{-1}	-2.470×10^{-1}	2.889×10^{-2}	2.872×10^{-2}	1.363×10^{-2}	1.349×10^{-2}
1.6	0	33.5053	34.7164	-9.425×10^{-2}	-9.399×10^{-2}	-2.622×10^{-1}	-2.616×10^{-1}	2.964×10^{-2}	2.951×10^{-2}	1.441×10^{-2}	1.432×10^{-2}
1.7	0	0	0	-6.332×10^{-2}	-6.317×10^{-2}	-2.034×10^{-1}	-2.031×10^{-1}	0	0	0	0
1.7	0	19.6910	20.9859	-6.702×10^{-2}	-6.687×10^{-2}	-2.101×10^{-1}	-2.098×10^{-1}	2.057×10^{-2}	2.052×10^{-2}	9.202×10^{-3}	9.171×10^{-3}
1.7	0	26.9252	28.2884	-7.197×10^{-2}	-7.184×10^{-2}	-2.199×10^{-1}	-2.196×10^{-1}	2.467×10^{-2}	2.456×10^{-2}	1.170×10^{-2}	1.162×10^{-2}
1.7	0	31.9218	33.1786	-7.794×10^{-2}	-7.782×10^{-2}	-2.319×10^{-1}	-2.316×10^{-1}	2.632×10^{-2}	2.620×10^{-2}	1.306×10^{-2}	1.296×10^{-2}
1.7	0	35.7100	36.8118	-8.475×10^{-2}	-8.465×10^{-2}	-2.457×10^{-1}	-2.455×10^{-1}	2.698×10^{-2}	2.686×10^{-2}	1.384×10^{-2}	1.373×10^{-2}
1.8	0	0	0	-5.531×10^{-2}	-5.528×10^{-2}	-1.888×10^{-1}	-1.887×10^{-1}	0	0	0	0
1.8	0	20.8804	22.1128	-5.879×10^{-2}	-5.874×10^{-2}	-1.948×10^{-1}	-1.946×10^{-1}	1.858×10^{-2}	1.858×10^{-2}	8.635×10^{-3}	8.639×10^{-3}
1.8	0	28.5007	29.7791	-6.343×10^{-2}	-6.336×10^{-2}	-2.036×10^{-1}	-2.035×10^{-1}	2.221×10^{-2}	2.223×10^{-2}	1.098×10^{-2}	1.101×10^{-2}
1.8	0	33.7400	34.9034	-6.901×10^{-2}	-6.894×10^{-2}	-2.146×10^{-1}	-2.144×10^{-1}	2.368×10^{-2}	2.371×10^{-2}	1.228×10^{-2}	1.232×10^{-2}
1.8	0	37.6985	38.7065	-7.533×10^{-2}	-7.533×10^{-2}	-2.271×10^{-1}	-2.271×10^{-1}	2.429×10^{-2}	2.427×10^{-2}	1.306×10^{-2}	1.303×10^{-2}
1.9	0	0	0	-4.809×10^{-2}	-4.811×10^{-2}	-1.740×10^{-1}	-1.740×10^{-1}	0	0	0	0
1.9	0	21.9900	23.1615	-5.132×10^{-2}	-5.134×10^{-2}	-1.792×10^{-1}	-1.793×10^{-1}	1.666×10^{-2}	1.664×10^{-2}	8.022×10^{-3}	8.007×10^{-3}
1.9	0	29.9708	31.1702	-5.562×10^{-2}	-5.564×10^{-2}	-1.872×10^{-1}	-1.872×10^{-1}	1.986×10^{-2}	1.987×10^{-2}	1.019×10^{-2}	1.020×10^{-2}
1.9	0	35.4385	36.5176	-6.078×10^{-2}	-6.077×10^{-2}	-1.971×10^{-1}	-1.970×10^{-1}	2.118×10^{-2}	2.122×10^{-2}	1.143×10^{-2}	1.148×10^{-2}
1.9	0	39.5592	40.4847	-6.659×10^{-2}	-6.658×10^{-2}	-2.082×10^{-1}	-2.082×10^{-1}	2.177×10^{-2}	2.182×10^{-2}	1.222×10^{-2}	1.228×10^{-2}
2.0	0	0	0	-4.174×10^{-2}	-4.175×10^{-2}	-1.598×10^{-1}	-1.598×10^{-1}	0	0	0	0
2.0	0	23.0471	24.1605	-4.471×10^{-2}	-4.472×10^{-2}	-1.643×10^{-1}	-1.643×10^{-1}	1.489×10^{-2}	1.489×10^{-2}	7.425×10^{-3}	7.424×10^{-3}
2.0	0	31.3715	32.4978	-4.867×10^{-2}	-4.871×10^{-2}	-1.713×10^{-1}	-1.714×10^{-1}	1.773×10^{-2}	1.770×10^{-2}	9.436×10^{-3}	9.411×10^{-3}
2.0	0	37.0583	38.0608	-5.341×10^{-2}	-5.345×10^{-2}	-1.801×10^{-1}	-1.801×10^{-1}	1.893×10^{-2}	1.889×10^{-2}	1.062×10^{-2}	1.057×10^{-2}
2.0	0	41.3358	42.1876	-5.873×10^{-2}	-5.875×10^{-2}	-1.900×10^{-1}	-1.900×10^{-1}	1.950×10^{-2}	1.948×10^{-2}	1.141×10^{-2}	1.138×10^{-2}

Table I: Teukolsky-based fluxes and kludge fluxes [computed using Eqs. (40), (43) and (44)] for circular orbits about a hole with $a = 0.998M$; μ represents the mass of the infalling body.

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$\frac{p}{M}$	e	θ_{inc} (deg.)	ι (deg.)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (kludge)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (Teukolsky)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{d\iota}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{d\iota}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)
1.3	0.1	0	0	-8.804×10^{-2}	-8.804×10^{-2}	-2.098×10^{-1}	-2.098×10^{-1}	0	0	0	0
1.4	0.1	0	0	-8.728×10^{-2}	-8.719×10^{-2}	-2.274×10^{-1}	-2.275×10^{-1}	0	0	0	0
1.4	0.1	8	8.8664	-9.110×10^{-2}	-8.736×10^{-2}	-2.355×10^{-1}	-2.273×10^{-1}	4.066×10^{-2}	0	2.938×10^{-2}	1.436×10^{-2}
1.4	0.1	16	17.4519	-1.030×10^{-1}	-8.958×10^{-2}	-2.602×10^{-1}	-2.309×10^{-1}	7.428×10^{-2}	0	5.475×10^{-2}	2.880×10^{-2}
1.4	0.1	24	25.5784	-1.243×10^{-1}	-9.771×10^{-2}	-3.037×10^{-1}	-2.415×10^{-1}	9.663×10^{-2}	0	7.316×10^{-2}	4.342×10^{-2}
1.5	0.1	0	0	-8.069×10^{-2}	-8.095×10^{-2}	-2.255×10^{-1}	-2.260×10^{-1}	0	0	0	0
1.5	0.1	8	8.7910	-8.323×10^{-2}	-8.133×10^{-2}	-2.310×10^{-1}	-2.264×10^{-1}	2.996×10^{-2}	0	2.070×10^{-2}	1.439×10^{-2}
1.5	0.1	16	17.3490	-9.121×10^{-2}	-8.395×10^{-2}	-2.483×10^{-1}	-2.314×10^{-1}	5.512×10^{-2}	0	3.888×10^{-2}	2.902×10^{-2}
1.5	0.1	24	25.5197	-1.059×10^{-1}	-8.980×10^{-2}	-2.792×10^{-1}	-2.423×10^{-1}	7.255×10^{-2}	0	5.264×10^{-2}	4.407×10^{-2}
1.6	0.1	0	0	-7.255×10^{-2}	-7.281×10^{-2}	-2.161×10^{-1}	-2.168×10^{-1}	0	0	0	0
1.6	0.1	8	8.7195	-7.430×10^{-2}	-7.321×10^{-2}	-2.201×10^{-1}	-2.173×10^{-1}	2.258×10^{-2}	0	1.502×10^{-2}	1.385×10^{-2}
1.6	0.1	16	17.2437	-7.986×10^{-2}	-7.533×10^{-2}	-2.323×10^{-1}	-2.212×10^{-1}	4.179×10^{-2}	0	2.839×10^{-2}	2.780×10^{-2}
1.6	0.1	24	25.4388	-9.025×10^{-2}	-8.040×10^{-2}	-2.547×10^{-1}	-2.309×10^{-1}	5.554×10^{-2}	0	3.886×10^{-2}	4.215×10^{-2}
1.6	0.1	32	33.2683	-1.082×10^{-1}	-9.435×10^{-2}	-2.920×10^{-1}	-2.551×10^{-1}	6.316×10^{-2}	0	4.559×10^{-2}	5.799×10^{-2}
1.7	0.1	0	0	-6.427×10^{-2}	-6.440×10^{-2}	-2.036×10^{-1}	-2.040×10^{-1}	0	0	0	0
1.7	0.1	8	8.6555	-6.552×10^{-2}	-6.478×10^{-2}	-2.065×10^{-1}	-2.045×10^{-1}	1.742×10^{-2}	0	1.124×10^{-2}	1.305×10^{-2}
1.7	0.1	16	17.1454	-6.953×10^{-2}	-6.651×10^{-2}	-2.154×10^{-1}	-2.075×10^{-1}	3.240×10^{-2}	0	2.134×10^{-2}	2.609×10^{-2}
1.7	0.1	24	25.3531	-7.707×10^{-2}	-7.052×10^{-2}	-2.317×10^{-1}	-2.150×10^{-1}	4.342×10^{-2}	0	2.948×10^{-2}	3.928×10^{-2}
1.7	0.1	32	33.2416	-9.009×10^{-2}	-7.959×10^{-2}	-2.590×10^{-1}	-2.324×10^{-1}	4.998×10^{-2}	0	3.512×10^{-2}	5.325×10^{-2}
1.8	0.1	0	0	-5.640×10^{-2}	-5.640×10^{-2}	-1.897×10^{-1}	-1.897×10^{-1}	0	0	0	0
1.8	0.1	8	8.5991	-5.732×10^{-2}	-5.676×10^{-2}	-1.918×10^{-1}	-1.902×10^{-1}	1.371×10^{-2}	0	8.640×10^{-3}	1.212×10^{-2}
1.8	0.1	16	17.0562	-6.028×10^{-2}	-5.817×10^{-2}	-1.984×10^{-1}	-1.925×10^{-1}	2.562×10^{-2}	0	1.647×10^{-2}	2.417×10^{-2}
1.8	0.1	24	25.2693	-6.588×10^{-2}	-6.139×10^{-2}	-2.105×10^{-1}	-1.983×10^{-1}	3.456×10^{-2}	0	2.291×10^{-2}	3.619×10^{-2}
1.8	0.1	32	33.2018	-7.555×10^{-2}	-6.849×10^{-2}	-2.307×10^{-1}	-2.120×10^{-1}	4.020×10^{-2}	0	2.765×10^{-2}	4.871×10^{-2}
1.9	0.1	0	0	-4.915×10^{-2}	-4.911×10^{-2}	-1.753×10^{-1}	-1.751×10^{-1}	0	0	0	0
1.9	0.1	8	8.5494	-4.985×10^{-2}	-4.945×10^{-2}	-1.768×10^{-1}	-1.755×10^{-1}	1.097×10^{-2}	0	6.791×10^{-3}	1.116×10^{-2}
1.9	0.1	16	16.9760	-5.208×10^{-2}	-5.064×10^{-2}	-1.817×10^{-1}	-1.774×10^{-1}	2.055×10^{-2}	0	1.298×10^{-2}	2.221×10^{-2}
1.9	0.1	24	25.1898	-5.633×10^{-2}	-5.328×10^{-2}	-1.908×10^{-1}	-1.819×10^{-1}	2.788×10^{-2}	0	1.816×10^{-2}	3.314×10^{-2}
1.9	0.1	32	33.1555	-6.364×10^{-2}	-5.870×10^{-2}	-2.059×10^{-1}	-1.920×10^{-1}	3.272×10^{-2}	0	2.214×10^{-2}	4.415×10^{-2}
2.0	0.1	0	0	-4.263×10^{-2}	-4.264×10^{-2}	-1.607×10^{-1}	-1.608×10^{-1}	0	0	0	0
2.0	0.1	8	8.5057	-4.316×10^{-2}	-4.292×10^{-2}	-1.619×10^{-1}	-1.611×10^{-1}	8.862×10^{-3}	0	5.424×10^{-3}	1.021×10^{-2}
2.0	0.1	16	16.9042	-4.488×10^{-2}	-4.390×10^{-2}	-1.656×10^{-1}	-1.625×10^{-1}	1.666×10^{-2}	0	1.039×10^{-2}	2.028×10^{-2}
2.0	0.1	24	25.1156	-4.815×10^{-2}	-4.604×10^{-2}	-1.724×10^{-1}	-1.660×10^{-1}	2.271×10^{-2}	0	1.459×10^{-2}	3.014×10^{-2}
2.0	0.1	32	33.1064	-5.376×10^{-2}	-5.031×10^{-2}	-1.838×10^{-1}	-1.736×10^{-1}	2.684×10^{-2}	0	1.793×10^{-2}	3.989×10^{-2}
2.0	0.1	40	40.8954	-6.339×10^{-2}	-6.236×10^{-2}	-2.027×10^{-1}	-1.967×10^{-1}	2.917×10^{-2}	0	2.036×10^{-2}	5.185×10^{-2}

Table II: As in Table I but for non-circular orbits. No Teukolsky-based code (that we are aware of) presently has the capability to compute inclination angle evolution for generic orbits; we have instead inserted the constraint $d\iota/dt = 0$ to determine dQ/dt given Teukolsky-based fluxes dE/dt and dL_z/dt . The Teukolsky-based value of $d\theta_{\text{inc}}/dt$ is then computed from that inferred value of dQ/dt . As such, both “Teukolsky” values for the inclination change should be considered highly unreliable; the kludge values are almost certainly more accurate in this case.

[32] Data available at <http://gmunu.mit.edu/sdrasco/snapshots/>

$\frac{p}{M}$	e	θ_{inc} (deg.)	ι (deg.)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (kludge)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (Teukolsky)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{dl}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{dl}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)
1.4	0.2	0	0	-8.636×10^{-2}	-8.642×10^{-2}	-2.119×10^{-1}	-2.121×10^{-1}	0	0	0	0
1.4	0.2	8	8.8215	-9.853×10^{-2}	-8.240×10^{-2}	-2.374×10^{-1}	-2.015×10^{-1}	1.148×10^{-1}	0	9.714×10^{-2}	1.250×10^{-2}
1.5	0.2	0	0	-8.362×10^{-2}	-8.349×10^{-2}	-2.236×10^{-1}	-2.230×10^{-1}	0	0	0	0
1.5	0.2	8	8.7595	-9.141×10^{-2}	-8.276×10^{-2}	-2.410×10^{-1}	-2.206×10^{-1}	7.893×10^{-2}	0	6.549×10^{-2}	1.386×10^{-2}
1.5	0.2	16	17.2957	-1.145×10^{-1}	-8.394×10^{-2}	-2.915×10^{-1}	-2.215×10^{-1}	1.466×10^{-1}	0	1.230×10^{-1}	2.745×10^{-2}
1.5	0.2	24	25.4608	-1.524×10^{-1}	-9.230×10^{-2}	-3.712×10^{-1}	-2.357×10^{-1}	1.952×10^{-1}	0	1.661×10^{-1}	4.226×10^{-2}
1.6	0.2	0	0	-7.596×10^{-2}	-7.616×10^{-2}	-2.171×10^{-1}	-2.176×10^{-1}	0	0	0	0
1.6	0.2	8	8.6935	-8.111×10^{-2}	-7.641×10^{-2}	-2.292×10^{-1}	-2.177×10^{-1}	5.520×10^{-2}	0	4.502×10^{-2}	1.374×10^{-2}
1.6	0.2	16	17.1994	-9.649×10^{-2}	-7.798×10^{-2}	-2.647×10^{-1}	-2.198×10^{-1}	1.032×10^{-1}	0	8.500×10^{-2}	2.736×10^{-2}
1.6	0.2	24	25.3891	-1.221×10^{-1}	-8.314×10^{-2}	-3.212×10^{-1}	-2.288×10^{-1}	1.388×10^{-1}	0	1.160×10^{-1}	4.131×10^{-2}
1.7	0.2	0	0	-6.765×10^{-2}	-6.799×10^{-2}	-2.057×10^{-1}	-2.068×10^{-1}	0	0	0	0
1.7	0.2	8	8.6329	-7.116×10^{-2}	-6.813×10^{-2}	-2.144×10^{-1}	-2.066×10^{-1}	3.963×10^{-2}	0	3.176×10^{-2}	1.306×10^{-2}
1.7	0.2	16	17.1064	-8.171×10^{-2}	-6.995×10^{-2}	-2.398×10^{-1}	-2.096×10^{-1}	7.441×10^{-2}	0	6.024×10^{-2}	2.612×10^{-2}
1.7	0.2	24	25.3085	-9.948×10^{-2}	-7.443×10^{-2}	-2.806×10^{-1}	-2.178×10^{-1}	1.009×10^{-1}	0	8.290×10^{-2}	3.940×10^{-2}
1.7	0.2	32	33.2037	-1.257×10^{-1}	-8.558×10^{-2}	-3.371×10^{-1}	-2.366×10^{-1}	1.175×10^{-1}	0	9.806×10^{-2}	5.357×10^{-2}
1.8	0.2	0	0	-5.965×10^{-2}	-5.962×10^{-2}	-1.927×10^{-1}	-1.926×10^{-1}	0	0	0	0
1.8	0.2	8	8.5789	-6.211×10^{-2}	-5.997×10^{-2}	-1.990×10^{-1}	-1.930×10^{-1}	2.919×10^{-2}	0	2.300×10^{-2}	1.220×10^{-2}
1.8	0.2	16	17.0211	-6.953×10^{-2}	-6.147×10^{-2}	-2.175×10^{-1}	-1.954×10^{-1}	5.504×10^{-2}	0	4.380×10^{-2}	2.432×10^{-2}
1.8	0.2	24	25.2283	-8.216×10^{-2}	-6.502×10^{-2}	-2.474×10^{-1}	-2.016×10^{-1}	7.515×10^{-2}	0	6.068×10^{-2}	3.648×10^{-2}
1.8	0.2	32	33.1656	-1.009×10^{-1}	-7.410×10^{-2}	-2.890×10^{-1}	-2.190×10^{-1}	8.839×10^{-2}	0	7.258×10^{-2}	4.982×10^{-2}
1.9	0.2	0	0	-5.218×10^{-2}	-5.210×10^{-2}	-1.786×10^{-1}	-1.783×10^{-1}	0	0	0	0
1.9	0.2	8	8.5312	-5.394×10^{-2}	-5.244×10^{-2}	-1.833×10^{-1}	-1.787×10^{-1}	2.197×10^{-2}	0	1.704×10^{-2}	1.127×10^{-2}
1.9	0.2	16	16.9441	-5.928×10^{-2}	-5.373×10^{-2}	-1.970×10^{-1}	-1.807×10^{-1}	4.156×10^{-2}	0	3.254×10^{-2}	2.245×10^{-2}
1.9	0.2	24	25.1518	-6.843×10^{-2}	-5.669×10^{-2}	-2.192×10^{-1}	-1.858×10^{-1}	5.706×10^{-2}	0	4.535×10^{-2}	3.356×10^{-2}
1.9	0.2	32	33.1207	-8.213×10^{-2}	-6.277×10^{-2}	-2.502×10^{-1}	-1.966×10^{-1}	6.767×10^{-2}	0	5.475×10^{-2}	4.481×10^{-2}
2.0	0.2	0	0	-4.528×10^{-2}	-4.530×10^{-2}	-1.637×10^{-1}	-1.638×10^{-1}	0	0	0	0
2.0	0.2	8	8.4891	-4.657×10^{-2}	-4.557×10^{-2}	-1.671×10^{-1}	-1.641×10^{-1}	1.679×10^{-2}	0	1.283×10^{-2}	1.032×10^{-2}
2.0	0.2	16	16.8749	-5.049×10^{-2}	-4.664×10^{-2}	-1.774×10^{-1}	-1.657×10^{-1}	3.184×10^{-2}	0	2.457×10^{-2}	2.052×10^{-2}
2.0	0.2	24	25.0802	-5.725×10^{-2}	-4.904×10^{-2}	-1.941×10^{-1}	-1.696×10^{-1}	4.391×10^{-2}	0	3.440×10^{-2}	3.056×10^{-2}
2.0	0.2	32	33.0730	-6.743×10^{-2}	-5.427×10^{-2}	-2.175×10^{-1}	-1.793×10^{-1}	5.243×10^{-2}	0	4.184×10^{-2}	4.086×10^{-2}
1.5	0.3	0	0	-8.481×10^{-2}	-8.478×10^{-2}	-2.094×10^{-1}	-2.094×10^{-1}	0	0	0	0
1.5	0.3	8	8.7037	-1.006×10^{-1}	-7.824×10^{-2}	-2.442×10^{-1}	-1.934×10^{-1}	1.484×10^{-1}	0	1.301×10^{-1}	1.189×10^{-2}
1.5	0.3	16	17.2003	-1.469×10^{-1}	-7.811×10^{-2}	-3.435×10^{-1}	-1.864×10^{-1}	2.766×10^{-1}	0	2.440×10^{-1}	2.246×10^{-2}
1.6	0.3	0	0	-8.144×10^{-2}	-8.123×10^{-2}	-2.183×10^{-1}	-2.178×10^{-1}	0	0	0	0
1.6	0.3	8	8.6498	-9.182×10^{-2}	-7.807×10^{-2}	-2.426×10^{-1}	-2.095×10^{-1}	1.028×10^{-1}	0	8.918×10^{-2}	1.302×10^{-2}
1.6	0.3	16	17.1246	-1.223×10^{-1}	-8.089×10^{-2}	-3.122×10^{-1}	-2.144×10^{-1}	1.928×10^{-1}	0	1.683×10^{-1}	2.624×10^{-2}
1.6	0.3	24	25.3046	-1.716×10^{-1}	-8.666×10^{-2}	-4.197×10^{-1}	-2.229×10^{-1}	2.607×10^{-1}	0	2.295×10^{-1}	3.950×10^{-2}
1.7	0.3	0	0	-7.362×10^{-2}	-7.314×10^{-2}	-2.104×10^{-1}	-2.095×10^{-1}	0	0	0	0
1.7	0.3	8	8.5953	-8.060×10^{-2}	-7.224×10^{-2}	-2.277×10^{-1}	-2.065×10^{-1}	7.240×10^{-2}	0	6.224×10^{-2}	1.287×10^{-2}
1.7	0.3	16	17.0415	-1.013×10^{-1}	-7.369×10^{-2}	-2.774×10^{-1}	-2.084×10^{-1}	1.365×10^{-1}	0	1.180×10^{-1}	2.559×10^{-2}
1.7	0.3	24	25.2339	-1.349×10^{-1}	-7.800×10^{-2}	-3.547×10^{-1}	-2.153×10^{-1}	1.861×10^{-1}	0	1.622×10^{-1}	3.836×10^{-2}
1.8	0.3	0	0	-6.488×10^{-2}	-6.484×10^{-2}	-1.973×10^{-1}	-1.972×10^{-1}	0	0	0	0
1.8	0.3	8	8.5454	-6.970×10^{-2}	-6.480×10^{-2}	-2.099×10^{-1}	-1.966×10^{-1}	5.206×10^{-2}	0	4.436×10^{-2}	1.226×10^{-2}
1.8	0.3	16	16.9628	-8.402×10^{-2}	-6.671×10^{-2}	-2.461×10^{-1}	-1.998×10^{-1}	9.857×10^{-2}	0	8.445×10^{-2}	2.454×10^{-2}
1.8	0.3	24	25.1601	-1.075×10^{-1}	-7.030×10^{-2}	-3.026×10^{-1}	-2.056×10^{-1}	1.353×10^{-1}	0	1.169×10^{-1}	3.667×10^{-2}
1.8	0.3	32	33.1047	-1.404×10^{-1}	-8.153×10^{-2}	-3.762×10^{-1}	-2.255×10^{-1}	1.600×10^{-1}	0	1.394×10^{-1}	5.050×10^{-2}
1.9	0.3	0	0	-5.669×10^{-2}	-5.690×10^{-2}	-1.829×10^{-1}	-1.832×10^{-1}	0	0	0	0
1.9	0.3	8	8.5010	-6.010×10^{-2}	-5.683×10^{-2}	-1.922×10^{-1}	-1.824×10^{-1}	3.823×10^{-2}	0	3.229×10^{-2}	1.137×10^{-2}
1.9	0.3	16	16.8911	-7.025×10^{-2}	-5.818×10^{-2}	-2.189×10^{-1}	-1.844×10^{-1}	7.263×10^{-2}	0	6.165×10^{-2}	2.263×10^{-2}
1.9	0.3	24	25.0887	-8.701×10^{-2}	-6.054×10^{-2}	-2.609×10^{-1}	-1.874×10^{-1}	1.003×10^{-1}	0	8.579×10^{-2}	3.341×10^{-2}
1.9	0.3	32	33.0624	-1.106×10^{-1}	-6.912×10^{-2}	-3.157×10^{-1}	-2.034×10^{-1}	1.195×10^{-1}	0	1.032×10^{-1}	4.570×10^{-2}
2.0	0.3	0	0	-4.953×10^{-2}	-4.946×10^{-2}	-1.683×10^{-1}	-1.683×10^{-1}	0	0	0	0
2.0	0.3	8	8.4616	-5.199×10^{-2}	-4.970×10^{-2}	-1.753×10^{-1}	-1.685×10^{-1}	2.862×10^{-2}	0	2.395×10^{-2}	1.047×10^{-2}
2.0	0.3	16	16.8262	-5.932×10^{-2}	-5.079×10^{-2}	-1.954×10^{-1}	-1.699×10^{-1}	5.452×10^{-2}	0	4.585×10^{-2}	2.079×10^{-2}
2.0	0.3	24	25.0215	-7.150×10^{-2}	-5.328×10^{-2}	-2.269×10^{-1}	-1.737×10^{-1}	7.564×10^{-2}	0	6.411×10^{-2}	3.090×10^{-2}
2.0	0.3	32	33.0172	-8.878×10^{-2}	-6.003×10^{-2}	-2.682×10^{-1}	-1.864×10^{-1}	9.077×10^{-2}	0	7.771×10^{-2}	4.191×10^{-2}

Table III: As in Table II, but for additional values of eccentricity e .

$\frac{p}{M}$	e	θ_{inc} (deg.)	ι (deg.)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (kludge)	$\frac{dE}{dt} \times \frac{M^2}{\mu^2}$ (Teukolsky)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{dL_z}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{dl}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{dl}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (kludge)	$\frac{d\theta_{\text{inc}}}{dt} \times \frac{M}{\mu^2}$ (Teukolsky)
1.6	0.4	0	0	-7.766×10^{-2}	-7.772×10^{-2}	-1.918×10^{-1}	-1.919×10^{-1}	0	0	0	0
1.6	0.4	8	8.5863	-9.433×10^{-2}	-7.645×10^{-2}	-2.297×10^{-1}	-1.881×10^{-1}	1.528×10^{-1}	0	1.370×10^{-1}	1.139×10^{-2}
1.6	0.4	16	17.0151	-1.432×10^{-1}	-7.651×10^{-2}	-3.382×10^{-1}	-1.837×10^{-1}	2.873×10^{-1}	0	2.584×10^{-1}	2.183×10^{-2}
1.7	0.4	0	0	-7.882×10^{-2}	-7.953×10^{-2}	-2.097×10^{-1}	-2.115×10^{-1}	0	0	0	0
1.7	0.4	8	8.5426	-9.002×10^{-2}	-7.408×10^{-2}	-2.367×10^{-1}	-1.978×10^{-1}	1.087×10^{-1}	0	9.656×10^{-2}	1.209×10^{-2}
1.7	0.4	16	16.9502	-1.229×10^{-1}	-7.682×10^{-2}	-3.143×10^{-1}	-2.025×10^{-1}	2.054×10^{-1}	0	1.830×10^{-1}	2.438×10^{-2}
1.7	0.4	24	25.1282	-1.760×10^{-1}	-8.090×10^{-2}	-4.336×10^{-1}	-2.075×10^{-1}	2.809×10^{-1}	0	2.514×10^{-1}	3.617×10^{-2}
1.8	0.4	0	0	-7.107×10^{-2}	-7.007×10^{-2}	-2.013×10^{-1}	-1.988×10^{-1}	0	0	0	0
1.8	0.4	8	8.4989	-7.877×10^{-2}	-7.001×10^{-2}	-2.209×10^{-1}	-1.981×10^{-1}	7.788×10^{-2}	0	6.879×10^{-2}	1.214×10^{-2}
1.8	0.4	16	16.8817	-1.015×10^{-1}	-7.009×10^{-2}	-2.774×10^{-1}	-1.965×10^{-1}	1.478×10^{-1}	0	1.309×10^{-1}	2.371×10^{-2}
1.8	0.4	24	25.0646	-1.383×10^{-1}	-7.314×10^{-2}	-3.646×10^{-1}	-2.003×10^{-1}	2.036×10^{-1}	0	1.810×10^{-1}	3.508×10^{-2}
1.8	0.4	32	33.0184	-1.887×10^{-1}	-9.193×10^{-2}	-4.755×10^{-1}	-2.319×10^{-1}	2.414×10^{-1}	0	2.156×10^{-1}	5.070×10^{-2}
1.9	0.4	0	0	-6.187×10^{-2}	-6.267×10^{-2}	-1.861×10^{-1}	-1.881×10^{-1}	0	0	0	0
1.9	0.4	8	8.4591	-6.728×10^{-2}	-6.216×10^{-2}	-2.006×10^{-1}	-1.861×10^{-1}	5.666×10^{-2}	0	4.980×10^{-2}	1.141×10^{-2}
1.9	0.4	16	16.8173	-8.328×10^{-2}	-6.222×10^{-2}	-2.424×10^{-1}	-1.844×10^{-1}	1.079×10^{-1}	0	9.506×10^{-2}	2.226×10^{-2}
1.9	0.4	24	25.0006	-1.094×10^{-1}	-6.486×10^{-2}	-3.071×10^{-1}	-1.878×10^{-1}	1.495×10^{-1}	0	1.322×10^{-1}	3.290×10^{-2}
1.9	0.4	32	32.9804	-1.452×10^{-1}	-7.884×10^{-2}	-3.896×10^{-1}	-2.158×10^{-1}	1.787×10^{-1}	0	1.588×10^{-1}	4.755×10^{-2}
2.0	0.4	0	0	-5.483×10^{-2}	-5.457×10^{-2}	-1.735×10^{-1}	-1.729×10^{-1}	0	0	0	0
2.0	0.4	8	8.4235	-5.871×10^{-2}	-5.445×10^{-2}	-1.844×10^{-1}	-1.720×10^{-1}	4.222×10^{-2}	0	3.686×10^{-2}	1.053×10^{-2}
2.0	0.4	16	16.7586	-7.020×10^{-2}	-5.555×10^{-2}	-2.158×10^{-1}	-1.733×10^{-1}	8.064×10^{-2}	0	7.057×10^{-2}	2.088×10^{-2}
2.0	0.4	24	24.9396	-8.902×10^{-2}	-5.844×10^{-2}	-2.645×10^{-1}	-1.778×10^{-1}	1.122×10^{-1}	0	9.860×10^{-2}	3.112×10^{-2}
2.0	0.4	32	32.9389	-1.150×10^{-1}	-6.536×10^{-2}	-3.267×10^{-1}	-1.896×10^{-1}	1.351×10^{-1}	0	1.193×10^{-1}	4.188×10^{-2}
1.7	0.5	0	0	-7.421×10^{-2}	-7.401×10^{-2}	-1.815×10^{-1}	-1.810×10^{-1}	0	0	0	0
1.7	0.5	8	8.4736	-8.957×10^{-2}	-7.168×10^{-2}	-2.173×10^{-1}	-1.750×10^{-1}	1.379×10^{-1}	0	1.256×10^{-1}	1.040×10^{-2}
1.7	0.5	16	16.8300	-1.347×10^{-1}	-6.999×10^{-2}	-3.201×10^{-1}	-1.676×10^{-1}	2.611×10^{-1}	0	2.378×10^{-1}	1.956×10^{-2}
1.8	0.5	0	0	-7.589×10^{-2}	-7.620×10^{-2}	-1.993×10^{-1}	-2.000×10^{-1}	0	0	0	0
1.8	0.5	8	8.4395	-8.644×10^{-2}	-6.929×10^{-2}	-2.254×10^{-1}	-1.829×10^{-1}	1.005×10^{-1}	0	9.076×10^{-2}	1.098×10^{-2}
1.8	0.5	16	16.7776	-1.175×10^{-1}	-7.210×10^{-2}	-3.004×10^{-1}	-1.880×10^{-1}	1.911×10^{-1}	0	1.726×10^{-1}	2.218×10^{-2}
1.8	0.5	24	24.9413	-1.678×10^{-1}	-7.395×10^{-2}	-4.158×10^{-1}	-1.881×10^{-1}	2.638×10^{-1}	0	2.385×10^{-1}	3.215×10^{-2}
1.9	0.5	0	0	-6.646×10^{-2}	-6.620×10^{-2}	-1.855×10^{-1}	-1.849×10^{-1}	0	0	0	0
1.9	0.5	8	8.4059	-7.386×10^{-2}	-6.320×10^{-2}	-2.048×10^{-1}	-1.768×10^{-1}	7.312×10^{-2}	0	6.579×10^{-2}	1.064×10^{-2}
1.9	0.5	16	16.7233	-9.572×10^{-2}	-6.551×10^{-2}	-2.603×10^{-1}	-1.809×10^{-1}	1.395×10^{-1}	0	1.255×10^{-1}	2.140×10^{-2}
1.9	0.5	24	24.8877	-1.312×10^{-1}	-7.087×10^{-2}	-3.461×10^{-1}	-1.909×10^{-1}	1.937×10^{-1}	0	1.744×10^{-1}	3.274×10^{-2}
1.9	0.5	32	32.8741	-1.795×10^{-1}	-8.247×10^{-2}	-4.544×10^{-1}	-2.091×10^{-1}	2.320×10^{-1}	0	2.092×10^{-1}	4.498×10^{-2}
2.0	0.5	0	0	-5.987×10^{-2}	-5.995×10^{-2}	-1.761×10^{-1}	-1.763×10^{-1}	0	0	0	0
2.0	0.5	8	8.3750	-6.516×10^{-2}	-5.918×10^{-2}	-1.906×10^{-1}	-1.738×10^{-1}	5.456×10^{-2}	0	4.882×10^{-2}	1.044×10^{-2}
2.0	0.5	16	16.6725	-8.081×10^{-2}	-5.817×10^{-2}	-2.324×10^{-1}	-1.694×10^{-1}	1.044×10^{-1}	0	9.343×10^{-2}	2.003×10^{-2}
2.0	0.5	24	24.8347	-1.063×10^{-1}	-6.254×10^{-2}	-2.970×10^{-1}	-1.776×10^{-1}	1.456×10^{-1}	0	1.304×10^{-1}	3.047×10^{-2}
2.0	0.5	32	32.8378	-1.412×10^{-1}	-6.993×10^{-2}	-3.787×10^{-1}	-1.893×10^{-1}	1.756×10^{-1}	0	1.576×10^{-1}	4.092×10^{-2}

Table IV: As in Tables II and III, but for different values of eccentricity e .

e	θ_{inc} (deg.)	ι (deg.)	$\Delta t/M$	$\Delta\theta_{\text{inc}}$ (deg.)	$\Delta\iota$ (deg.)
0	0	0	1.250×10^6	0	0
0	5	5.355510	1.217×10^6	1.949×10^{-1}	4.954×10^{-1}
0	10	10.679331	1.118×10^6	3.468×10^{-1}	8.631×10^{-1}
0	15	15.943192	9.574×10^5	4.236×10^{-1}	1.019
0	20	21.125167	7.446×10^5	4.109×10^{-1}	9.440×10^{-1}
0	25	26.211779	4.981×10^5	3.158×10^{-1}	6.860×10^{-1}
0	30	31.199048	2.528×10^5	1.732×10^{-1}	3.527×10^{-1}
0	35	36.092514	6.584×10^4	4.636×10^{-2}	8.806×10^{-2}
0.1	0	0	1.228×10^6	0	0
0.1	5	5.351602	1.198×10^6	4.517×10^{-1}	7.766×10^{-1}
0.1	10	10.671900	1.103×10^6	6.900×10^{-1}	1.236
0.1	15	15.932962	9.426×10^5	7.283×10^{-1}	1.344
0.1	20	21.113129	7.315×10^5	6.433×10^{-1}	1.187
0.1	25	26.199088	4.900×10^5	4.780×10^{-1}	8.547×10^{-1}
0.1	30	31.186915	2.513×10^5	2.730×10^{-1}	4.585×10^{-1}
0.1	35	36.082095	6.589×10^4	8.385×10^{-2}	1.279×10^{-1}
0.2	0	0	1.173×10^6	0	0
0.2	5	5.339916	1.150×10^6	1.204	1.598
0.2	10	10.649670	1.064×10^6	1.698	2.331
0.2	15	15.902348	9.043×10^5	1.618	2.293
0.2	20	21.077081	6.980×10^5	1.324	1.900
0.2	25	26.161046	4.693×10^5	9.545×10^{-1}	1.351
0.2	30	31.150481	2.486×10^5	5.674×10^{-1}	7.711×10^{-1}
0.2	35	36.050712	7.562×10^4	2.070×10^{-1}	2.648×10^{-1}
0.3	0	0	1.087×10^6	0	0
0.3	5	5.320559	1.069×10^6	2.307	2.788
0.3	10	10.612831	1.001×10^6	3.256	4.007
0.3	15	15.851572	8.454×10^5	2.984	3.741
0.3	20	21.017212	6.483×10^5	2.375	2.998
0.3	25	26.097732	4.408×10^5	1.700	2.129
0.3	30	31.089639	2.493×10^5	1.040	1.276
0.3	35	35.997987	1.108×10^5	4.626×10^{-1}	5.569×10^{-1}

Table V: Variation in the inclination angles ι and θ_{inc} as well as time needed to reach the separatrix for several inspirals through the nearly horizon-skimming regime. In all of these cases, the binary's mass ratio was fixed to $\mu/M = 10^{-6}$, the large black hole's spin was fixed to $a = 0.998M$, and the orbits were begun at $p = 1.9M$. The time interval Δt is the total accumulated time it takes for the inspiralling body to reach the separatrix (at which time it rapidly plunges into the black hole). The angles $\Delta\theta_{\text{inc}}$ and $\Delta\iota$ are the total integrated change in these inclination angles that we compute. For the $e = 0$ cases, inspirals are computed using fits to the circular-Teukolsky fluxes of E and L_z ; for eccentric orbits we use the kludge fluxes (40), (43) and (44). Notice that $\Delta\theta_{\text{inc}}$ and $\Delta\iota$ are always positive — the inclination angle always increases during the inspiral through the nearly horizon-skimming region. The magnitude of this increase never exceeds a few degrees.