

## Theory of Thin Aerofoil in Incompressible Flow

### Exercise 7 (part b) - Page 109

The methods used in this note are useful in analysing two-dimensional inviscid incompressible fluid flow.

- Question 7. (a) The flow past a corner shown in Figure 2.12 is just a deflected version of problem 6. Since we are dealing with a steady incompressible inviscid flow the solution reminds the same. As before this problem could be solved by using the pressure definition

$$p_1 = \frac{1}{\pi} \int_0^b \frac{Y'_+(\zeta)}{(\zeta - x)} d\zeta.$$

however, we could determine pressure without using the definition above. Instead we seek the pressure in the form

$$p_1 + iv_1 = (C_r + iC_i) \ln z + iD,$$

which leads to

$$\begin{aligned} p_1 &= C_r \ln r - C_i \theta, \\ v_1 &= C_r \theta + C_i \ln r + D, \end{aligned}$$

where  $z = re^{i\theta}$ . The flow is inviscid thus we can use impermeability conditions. First, if we consider the upstream of the flow where  $\theta = 0$  and second condition is found when we go downstream of the flow which takes  $\theta = -\pi$ . Note that  $v_1$  is the vertical velocity and at  $\theta = 0$  it simply is zero.

$$v_1 = (C_r \theta + C_i \ln r + D) \Big|_{\theta=0} = 0,$$

which leads to  $C_i = D = 0$ .

$$\begin{aligned} \frac{v}{u} &= \frac{\epsilon v_1}{1 + \epsilon u_1} = \epsilon \theta_0, \\ v_1 &= \theta_0, \end{aligned}$$

Now considering the second condition

$$v_1 = C_r \theta \Big|_{\theta=-\pi} = \theta_0,$$

this gives  $C_r = -\frac{\theta_0}{\pi}$ . This leads to

$$p_1 = -\frac{\theta_0}{\pi} \ln r.$$

(b) Considering this analysis using conformal mapping method with no restriction on the deflection angle  $\theta$  shows

$$\bar{V}(z) = u - iv = Az^{\frac{\theta}{\pi-\theta}} + \dots \quad \text{as } z \rightarrow 0.$$

where  $A$ -real constant. Our task is to show  $\bar{V}(z)$  reduces to

$$p_1 = -\frac{\theta_0}{\pi} \ln z + \dots,$$

for the case where  $\theta \ln z \ll 1$ . Remember that  $p = \epsilon p_1$

$$p = \frac{\epsilon \theta}{\pi} \ln z + \dots$$

We can show that

$$\ln(\bar{V}(z)) = \ln A - \frac{\theta \ln z}{\theta - \pi},$$

We separate the real and imaginary part and using the fact that  $u_1 = -p_1$ . We find that

$$\begin{aligned} \ln(\bar{V}(z)) &= \ln|\bar{V}(z)| + i\phi, \\ \ln(\bar{V}(z)) &= u - iv = 1 + \epsilon u_1 - iv_1, \end{aligned}$$

the real part could be expressed as

$$\ln|\bar{V}(z)| = \sqrt{(1 + \epsilon u_1)^2 + (\epsilon v_1)^2} \approx 1 + \epsilon u_1,$$

thus

$$1 + \epsilon u_1 = \ln A - \frac{\epsilon \theta_0}{\epsilon \theta_0 - \pi} \ln z,$$

we can choose  $\ln A = 1$ .

$$\epsilon u_1 = -\frac{\epsilon \theta_0}{\epsilon \theta_0 - \pi} \ln z \approx \frac{\epsilon \theta_0}{\pi} \ln z,$$

Assuming  $\theta \ln z \ll 1$ , the equation for velocity obtained using the conformal mapping method reduces to the solution obtained from the solution for pressure.

- Question 8. Let  $z = re^{i\theta}$

$$p_1 + iv_1 = (A_r + iA_i) + (B_r + iB_i)r^\lambda e^{i\lambda\theta},$$

separating the real and imaginary parts such

$$\begin{aligned} p_1 &= A_r + r^\lambda (B_r \cos \lambda\theta - B_i \sin \lambda\theta), \\ v_1 &= A_i + r^\lambda (B_i \cos \lambda\theta + B_r \sin \lambda\theta), \end{aligned}$$

Using the boundary condition we are able to find  $A$  and  $B$ . At  $\theta = 0$  we have  $v_1 = 0$  which means

$$v_1 = A_i + r^\lambda (B_i \cos \lambda\theta + B_r \sin \lambda\theta) = 0,$$

which means  $A_i = B_i = 0$ . At  $\theta = \pi$

$$v_1 = (r^\lambda B_r \sin \lambda\theta) \Big|_{\theta=\pi} = \theta_0,$$

this gives  $B_r = \frac{\theta_0 r^{-\lambda}}{\sin(\lambda\pi)}$ . Therefore we find

$$p_1 = A_r + \frac{\theta_0}{\sin(\lambda\pi)} \cos(\theta\lambda) = A_r + \frac{\theta_0 \Re\{z\}}{\sin(\lambda\pi)}.$$

- Question 9. We know that  $u_1 = -p_1$  and pressure by definition is

$$u_1 = -p_1 = \frac{-1}{2\pi} \int_0^1 \frac{Y'_+(\zeta) - Y'_-(\zeta)}{\zeta - x} d\zeta \pm \sqrt{\frac{1-x}{x}} \left[ \alpha_* - \frac{1}{2\pi} \int_0^1 \sqrt{\frac{\zeta}{\zeta-1}} \frac{Y'_+(\zeta) + Y'_-(\zeta)}{\zeta - x} d\zeta \right].$$

Since we are dealing with a parabolic shape the first term becomes zero since  $Y'_+(\zeta) = -Y'_-(\zeta)$ .

$$u_1 = \pm \sqrt{\frac{1-x}{x}} \left[ \alpha_* - \frac{1}{2\pi} \int_0^1 \sqrt{\frac{\zeta}{\zeta-1}} \frac{Y'_+(\zeta) + Y'_-(\zeta)}{\zeta - x} d\zeta \right].$$

Note that  $(1-x)^{1/2} = 1 - \frac{1}{2}x + \dots$  as  $x \rightarrow 0_+$ . Also the integrand may be written as

$$\sqrt{\frac{\zeta}{\zeta-1}} [Y'_+(\zeta) + Y'_-(\zeta)] (\zeta - x)^{-1} \approx \frac{1}{\sqrt{\zeta(\zeta-1)}} [Y'_+(\zeta) + Y'_-(\zeta)] \quad \text{as } x \rightarrow 0_+,$$

which leads to

$$u_1 = \pm \frac{\alpha_*}{\sqrt{x}} \mp \frac{1}{2\pi\sqrt{x}} \int_0^1 \frac{Y'_+(\zeta) + Y'_-(\zeta)}{\sqrt{\zeta(\zeta-1)}} d\zeta.$$

The surface on the upper side is defined as  $Y = \sqrt{2X}$  and by substituting this expression into

$$\hat{V} = V_\infty \frac{Y + k}{\sqrt{Y^2 + 1}},$$

we find that

$$\hat{V} = V_\infty \frac{\sqrt{2X} + k}{\sqrt{2X + 1}} = \frac{1 + \frac{k}{\sqrt{2X}}}{\sqrt{1 + (2X)^{-1}}} = 1 + \frac{k}{\sqrt{2X}} + O(x^{-1}),$$

Now using the definition

$$\begin{aligned} \frac{\hat{V}}{V_\infty} &= \frac{\sqrt{\hat{u}^2 + \hat{v}^2}}{V_\infty}, \\ &= \frac{\sqrt{V_\infty^2 (1 + \epsilon u_1)^2 + (V_\infty \epsilon v_1)^2}}{V_\infty}, \\ &= \sqrt{1 + 2\epsilon u_1 + \dots}, \\ &\approx 1 + \epsilon u_1 + \dots \end{aligned}$$

Our task is to find  $k$  and to do this we substitute  $u_1(x, 0) = \frac{\chi}{\sqrt{x}}$  into  $\frac{\hat{V}}{V_\infty} = 1 + \epsilon u_1$  which leads to

$$\frac{\hat{V}}{V_\infty} = 1 \pm \epsilon \frac{\chi}{\sqrt{x}} + O(\epsilon^2),$$

comparing this expression with

$$\frac{\hat{V}}{V_\infty} = 1 + \epsilon \frac{k}{\sqrt{2X}} + O(X^{-1}),$$

given that  $X = \frac{x}{\epsilon^2}$  we find  $k$  to be

$$k = \sqrt{2} \left[ \alpha_* - \frac{1}{2\pi} \int_0^1 \frac{Y'_+(\zeta) + Y'_-(\zeta)}{\sqrt{\zeta(\zeta-1)}} d\zeta \right].$$

## Questions

The questions are found in Ruban (2015).

## References

Ruban, A. I. (2015), ‘Fluid dynamics. part 2, asymptotic problems of fluid dynamics’.