Theory of Thin Aerofoil in Incompressible Flow

Exercise 7 (part b) - Page 109

The methods used in this note are useful in analysing two-dimensional inviscid incompressible fluid flow.

• Question 7. (a) The flow past a corner shown in Figure 2.12 is just a deflected version of problem 6. Since we are dealing with a steady incompressible inviscid flow the solution reminds the same. As before this problem could be solved by using the pressure definition

$$p_1 = \frac{1}{\pi} \int_0^b \frac{Y'_+(\zeta)}{(\zeta - x)} d\zeta.$$

however, we could determine pressure without using the definition above. Instead we seek the pressure in the form

$$p_1 + iv_1 = (C_r + iC_i) \ln z + iD,$$

which leads to

$$p_1 = C_r \ln r - C_i \theta,$$

$$v_1 = C_r \theta + C_i \ln r + D,$$

where $z = re^{i\theta}$. The flow is inviscid thus we can use impermeability conditions. First, if we consider the upstream of the flow where $\theta = 0$ and second condition is found when we go downstream of the flow which takes $\theta = -\pi$. Note that v_1 is the vertical velocity and at $\theta = 0$ it simply is zero.

$$v_1 = (C_r \theta + C_i \ln r + D) \Big|_{\theta=0} = 0,$$

which leads to $C_i = D = 0$.

$$\frac{v}{u} = \frac{\epsilon v_1}{1 + \epsilon u_1} = \epsilon \theta_0,$$

$$v_1 = \theta_0.$$

Now considering the second condition

$$v_1 = C_r \theta \bigg|_{\theta = -\pi} = \theta_0,$$

this gives $C_r = -\frac{\theta_0}{\pi}$. This leads to

$$p_1 = -\frac{\theta_0}{\pi} \ln r.$$

(b) Considering this analysis using conformal mapping method with no restriction on the deflection angle θ shows

$$\bar{V}(z) = -p - iv = \frac{\theta}{\epsilon \pi} \ln z + \dots,$$

$$\bar{V}(z) = u - iv = Az^{\frac{\theta}{\pi - \theta}} + \dots$$
 as $z \to 0$.

where A-real constant. We can show that

$$\ln(\bar{V}(z)) = \ln A - \frac{\theta \ln z}{\theta - \pi}, \text{ where } \theta \ln z \ll 1,$$

 $\ln(\bar{V}(z)) = \ln A$

which means $\bar{V}(z)$ is constant and we could let this constant to be

$$\bar{V}(z) = \frac{1}{\pi},$$

this shows that under the assumption $\theta \ln z \ll 1$, the equation for velocity obtained using the conformal mapping method reduces to the solution obtained from pressure.

• Question 8. Let $z = re^{i\theta}$

$$p_1 + iv_1 = (A_r + iA_i) + (B_r + iB_i)r^{\lambda}e^{i\lambda\theta}$$

separating the real and imaginary parts such

$$p_1 = A_r + r^{\lambda} (B_r \cos \lambda \theta - B_i \sin \lambda \theta),$$

$$v_1 = A_i + r^{\lambda} (B_i \cos \lambda \theta + B_r \sin \lambda \theta),$$

Using the boundary condition we are able to find A and B. At $\theta = 0$ we have $v_1 = 0$ which means

$$v_1 = A_i + r^{\lambda} (B_i \cos \lambda \theta + B_r \sin \lambda \theta) = 0,$$

which means $A_i = B_i = 0$. At $\theta = \pi$

$$v_1 = (r^{\lambda} B_r \sin \lambda \theta) \bigg|_{\theta = \pi} = \theta_0,$$

this gives $B_r = \frac{\theta_0 r^{-\lambda}}{\sin{(\lambda \pi)}}$. Therefore we find

$$p_1 = A_r + \frac{\theta_0}{\sin(\lambda \pi)} \cos(\theta \lambda) = A_r + \frac{\theta_0 \Re\{z\}}{\sin(\lambda \pi)}.$$

• Question 9. We know that $u_1 = -p_1$ and pressure by definition is

$$u_1 = -p_1 = \frac{-1}{2\pi} \int_0^1 \frac{Y'_+(\zeta) - Y'_-(\zeta)}{\zeta - x} d\zeta \pm \sqrt{\frac{1 - x}{x}} \left[\alpha_* - \frac{1}{2\pi} \int_0^1 \sqrt{\frac{\zeta}{\zeta - 1}} \frac{Y'_+(\zeta) + Y'_-(\zeta)}{\zeta - x} d\zeta \right].$$

Since we are dealing with a parabolic shape the first term becomes zero since $Y'_{+}(\zeta) = -Y'_{-}(\zeta)$.

$$u_1 = \pm \sqrt{\frac{1-x}{x}} \left[\alpha_* - \frac{1}{2\pi} \int_0^1 \sqrt{\frac{\zeta}{\zeta-1}} \frac{Y'_+(\zeta) + Y'_-(\zeta)}{\zeta - x} d\zeta \right].$$

Note that $(1-x)^{1/2} = 1 - \frac{1}{2}x + \dots$ as $x \to 0_+$. Also the integrand may be written as

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$$\sqrt{\frac{\zeta}{\zeta - 1}} [Y'_{+}(\zeta) + Y'_{-}(\zeta)](\zeta - x)^{-1} \approx \frac{1}{\sqrt{\zeta(\zeta - 1)}} [Y'_{+}(\zeta) + Y'_{-}(\zeta)] \quad \text{as} \quad x \to 0_{+},$$

which leads to

$$u_1 = \pm \frac{\alpha_*}{\sqrt{x}} \mp \frac{1}{2\pi\sqrt{x}} \int_0^1 \frac{Y'_+(\zeta) + Y'_-(\zeta)}{\sqrt{\zeta(\zeta - 1)}} d\zeta.$$

The surface on the upper side is defined as $Y = \sqrt{2X}$ and by substituting this expression into

$$\hat{V} = V_{\infty} \frac{Y + k}{\sqrt{Y^2 + 1}},$$

we find that

$$\hat{V} = V_{\infty} \frac{\sqrt{2X} + k}{\sqrt{2X} + 1} = \frac{1 + \frac{k}{2X}}{\sqrt{1 + (2X)^{-1}}} = 1 + \frac{k}{\sqrt{2X}} + O(x^{-1}),$$

Now using the definition

$$\frac{\hat{V}}{V_{\infty}} = \frac{\sqrt{\hat{u}^2 + \hat{v}^2}}{V_{\infty}},$$

$$= \frac{\sqrt{V_{\infty}^2 (1 + \epsilon u_1)^2 + (V_{\infty} \epsilon v_1)^2}}{V_{\infty}},$$

$$= \sqrt{1 + 2\epsilon u_1 + \dots},$$

$$\approx 1 + \epsilon u_1 + \dots$$

Our task is to find k and to do this we substitute $u_1(x,0) = \frac{\chi}{\sqrt{x}}$ into $\frac{\hat{V}}{V_{\infty}} = 1 + \epsilon u_1$ which leads to

$$\frac{\hat{V}}{V_{\infty}} = 1 \pm \epsilon \frac{\chi}{\sqrt{x}} + O(\epsilon^2),$$

comparing this expression with

$$\frac{\hat{V}}{V_{\infty}} = 1 + \epsilon \frac{k}{\sqrt{2X}} + O(X^{-1}),$$

given that $X = \frac{x}{\epsilon^2}$ we find k to be

$$k = \sqrt{2} \left[\alpha_* - \frac{1}{2\pi} \int_0^1 \left[\frac{Y'_+(\zeta) + Y'_-(\zeta)}{\sqrt{\zeta(\zeta - 1)}} d\zeta \right] \right].$$

Questions

The questions are found in Ruban (2015).

References

Ruban, A. I. (2015), 'Fluid dynamics. part 2, asymptotic problems of fluid dynamics'.