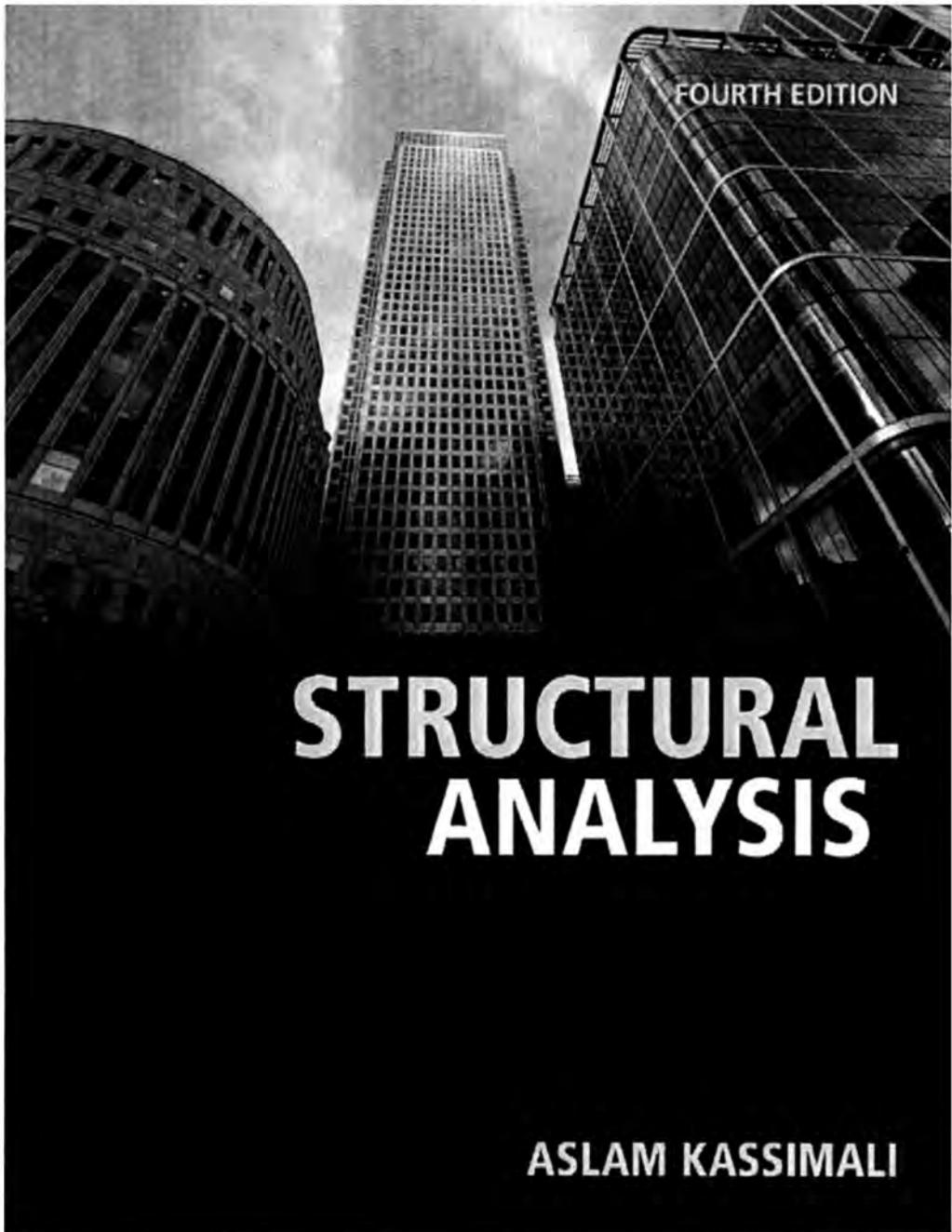


An Instructor's Solutions Manual to Accompany
Structural Analysis, 4th Edition

Aslam Kassimali



ISBN-13: 978-0-495-29566-2
ISBN-10: 0-495-29566-3



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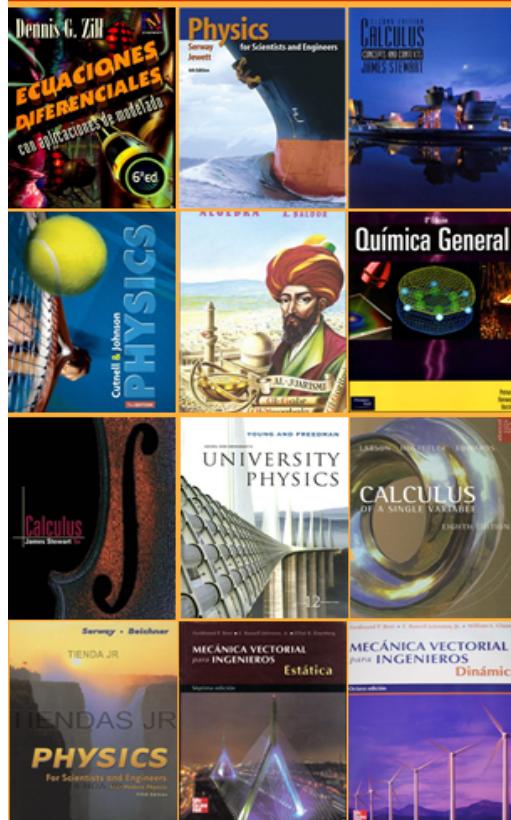
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An Instructor's Solutions Manual

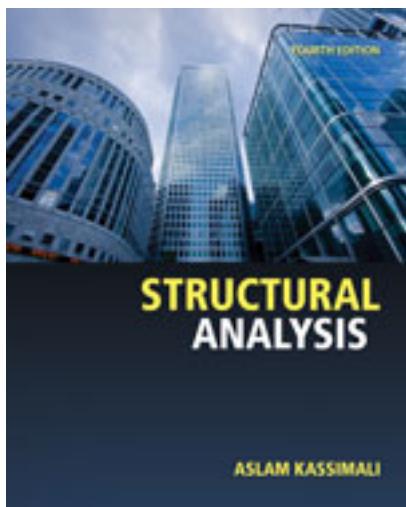
For

Structural Analysis

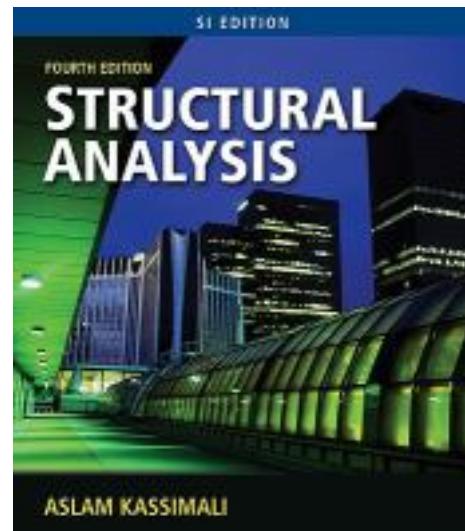
Fourth Edition

Aslam Kassimali

Southern Illinois University Carbondale



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SI



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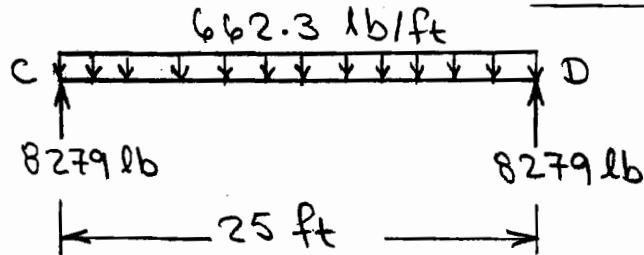
Chapter Two

Loads on Structures

CHAPTER 2

2.1 Beam CD

$$\begin{aligned}\text{Uniformly distributed load} &= 150(12)\left(\frac{4}{12}\right) + 490\left(\frac{18.3}{144}\right) \\ &= \underline{662.3 \text{ lb/ft}}\end{aligned}$$



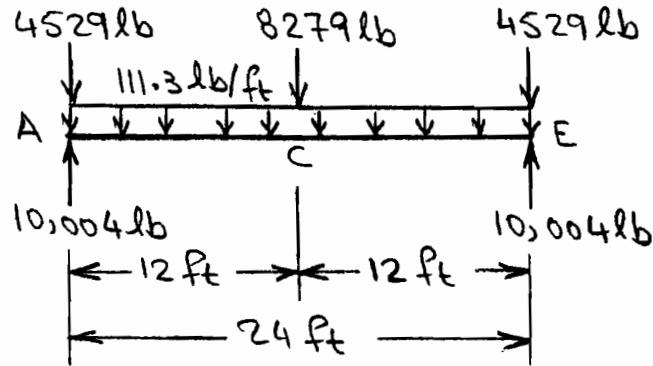
Girder AE

$$\text{Uniformly distributed load} = 490\left(\frac{32.7}{144}\right) = \underline{111.3 \text{ lb/ft}}$$

$$\text{Concentrated load at C} = \underline{8279 \text{ lb}}$$

Concentrated loads at A and E

$$= [150(6)\left(\frac{4}{12}\right) + 490\left(\frac{18.3}{144}\right)]\left(\frac{25}{2}\right) = \underline{4529 \text{ lb}}$$



2.2 See solution of Problem 2.1

Beam CD Uniformly distributed load

$$= 662.3 + 120 \left(\frac{6}{12}\right)(7) = 662.3 + 420 = \underline{1082.3 \text{ lb/ft}}$$

Girder AE Uniformly distributed load = 111.3 lb/ft

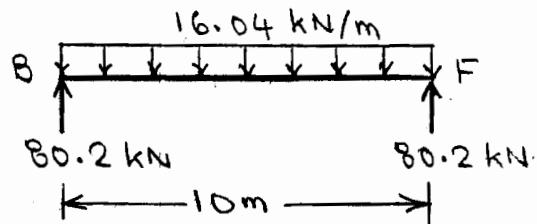
$$\text{Concentrated load at C} = 8279 + 420 \left(\frac{25}{2}\right) = \underline{13,529 \text{ lb}}$$

$$\text{Concentrated loads at A and E} = \underline{4529 \text{ lb}}$$

2.3 Beam BF

Uniformly distributed load

$$= 23.6(5)\left(\frac{130}{1000}\right) + 77\left(\frac{9100}{10^6}\right) = \underline{16.04 \text{ kN/m}}$$



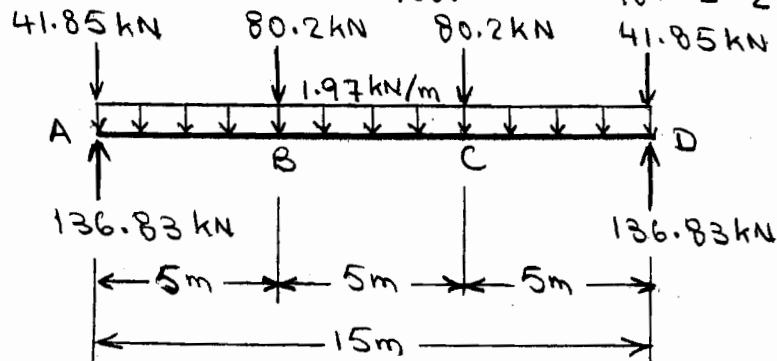
Girder AD

$$\text{Uniformly distributed load} = 77\left(\frac{25600}{10^6}\right) = \underline{1.97 \text{ kN/m}}$$

$$\text{Concentrated loads at B and C} = \underline{80.2 \text{ kN}}$$

$$\text{Concentrated loads at A and D}$$

$$= \left[23.6(2.5)\left(\frac{130}{1000}\right) + 77\left(\frac{9100}{10^6}\right) \right] \frac{10}{2} = \underline{41.85 \text{ kN}}$$



2.4

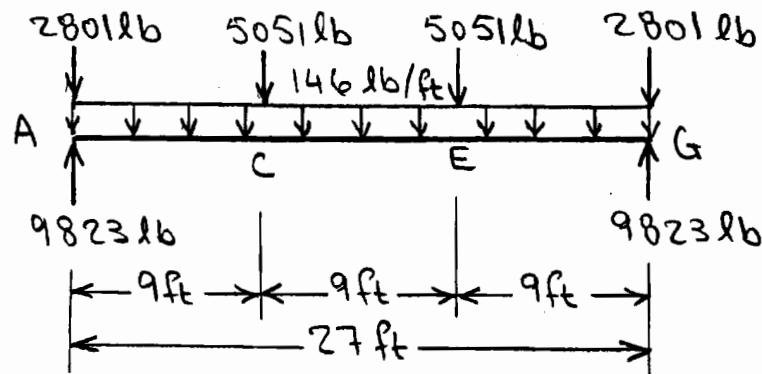
Uniformly distributed load = $490 \left(\frac{42.9}{144} \right) = 146 \text{ lb/ft}$

Concentrated loads at A and G

$$= \left[150(4.5)\left(\frac{4}{12}\right) + 490\left(\frac{16.2}{144}\right) \right] \left(\frac{20}{2}\right) = 2801 \text{ lb}$$

Concentrated loads at C and E

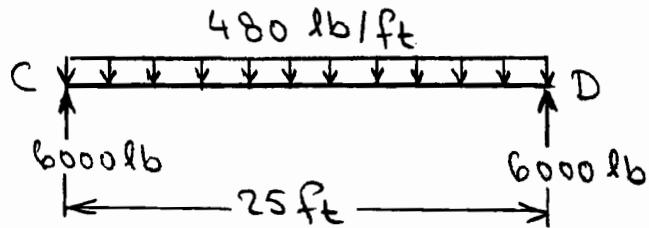
$$= \left[150(9)\left(\frac{4}{12}\right) + 490\left(\frac{16.2}{144}\right) \right] \left(\frac{20}{2}\right) = 5051 \text{ lb}$$



2-5 Live load = 40 psf

Beam CD

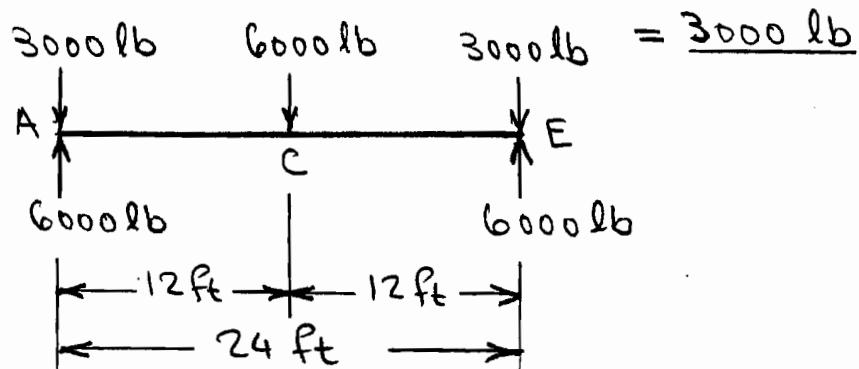
Uniformly distributed load = $40(12) = \underline{480 \text{ lb/ft}}$



Girder AE

Concentrated load at C = 6000 lb

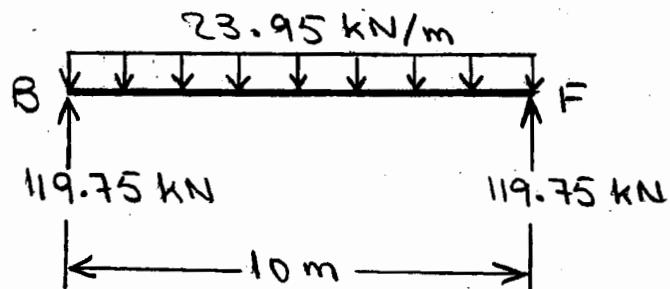
Concentrated loads at A and E = $[40(6)]\left(\frac{25}{2}\right)$



2.6 Live load = $4.79 \text{ kPa} = 4.79 \text{ kN/m}^2$

Beam BF

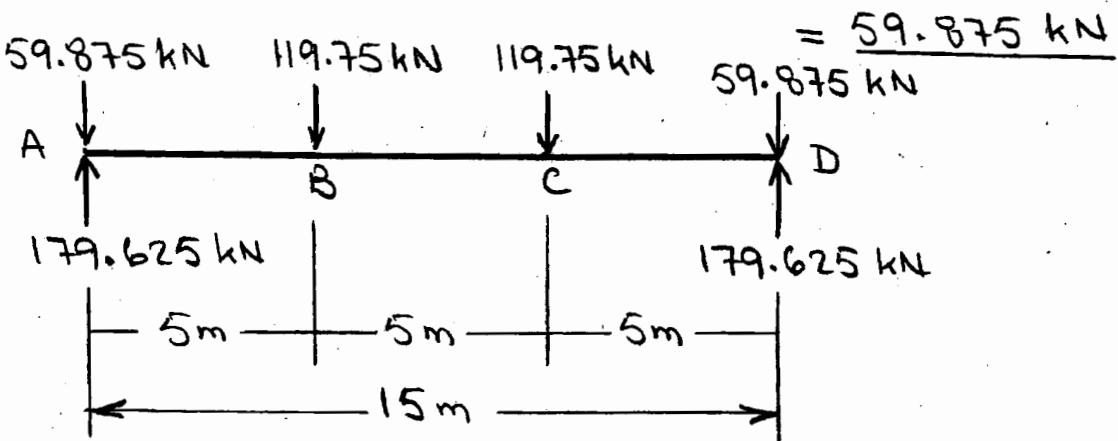
Uniformly distributed load = $4.79(5) = \underline{23.95 \text{ kN/m}}$



Girder AD

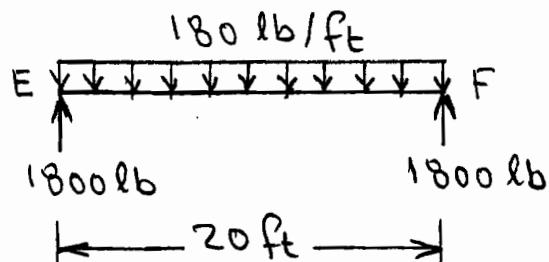
Concentrated loads at B and C = $\underline{119.75 \text{ kN}}$

Concentrated loads at A and D = $[4.79(2.5)] \frac{10}{2} = 59.875 \text{ kN}$



2.7 Beam EF

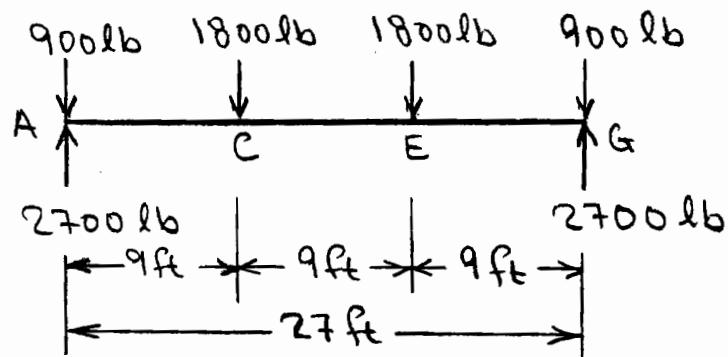
Uniformly distributed load = $20(9) = 180 \text{ lb/ft}$



Girder AG

Concentrated loads at C and E = 1800 lb

Concentrated loads at A and G = $1800/2 = 900 \text{ lb}$



Column A Concentrated load = 2700 lb

2.8 $V = 85 \text{ mph}$, $h = 40 + (15/2) = 47.5 \text{ ft}$,
 $I = 1.0$, $z_g = 1200 \text{ ft}$, $\alpha = 7.0$, $K_{zg} = 1$
and $K_d = 1$

 $k_h = 2.01 \left(\frac{47.5}{1200} \right)^{2/7} = 0.8$
 $q_h = 0.00256 (0.8)(1)(1)(85)^2 (1) = 14.8 \text{ psf}$

$G = 0.85$

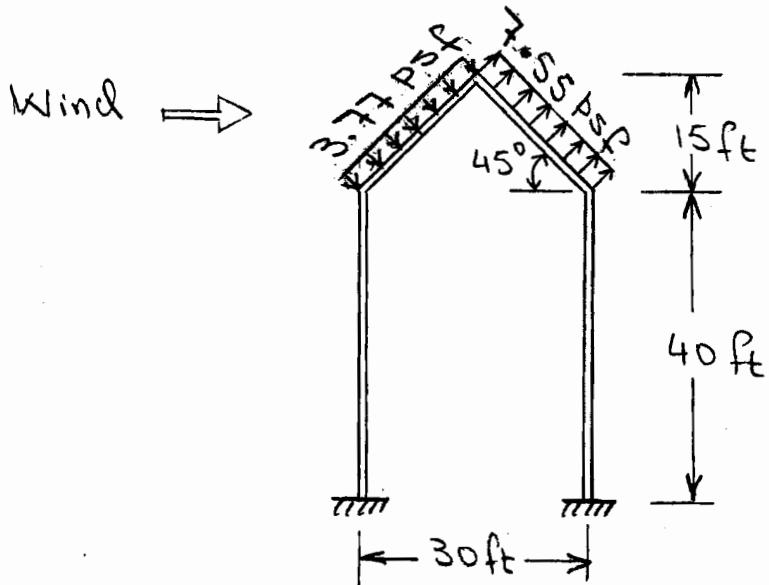
For $\Theta = 45^\circ$ and $h/L = 47.5/30 = 1.58$:

$C_p = 0.3 \quad \text{for windward side}$

$C_p = -0.6 \quad \text{for leeward side}$

Thus, the wind pressures are:

$P_h = 14.8 (0.85)(0.3) = \underline{3.77 \text{ psf}} \quad \text{for windward side}$
 $P_h = 14.8 (0.85)(-0.6) = \underline{-7.55 \text{ psf}} \quad \text{for leeward side}$



2.9 $V = 40 \text{ m/s}, h = 12 + \frac{5}{2} = 14.5 \text{ m}$

$I = 1.15, z_g = 366 \text{ m}, \alpha = 7.0, K_{gt} = 1$
and $K_d = 1$

$$K_h = 2.01 \left(\frac{14.5}{366} \right)^{2/7} = 0.8$$

$$q_h = 0.613(0.8)(1)(1)(40)^2(1.15) = 902.34 \text{ N/m}^2$$

$$G = 0.85$$

Roof slope: $\theta = \tan^{-1}(5/6) = 39.8^\circ$

$$\frac{h}{L} = \frac{14.5}{12} = 1.21$$

$C_p = -0.1$ and 0.25 for windward side

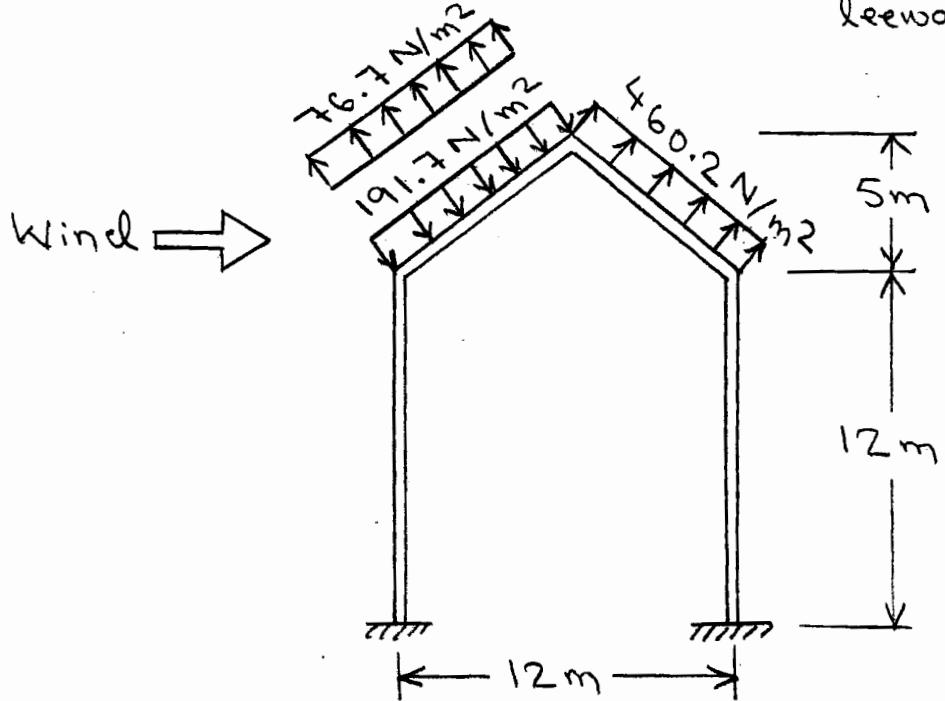
$C_p = -0.6$ for leeward side

Thus, the wind pressures are:

$$p_h = (902.34)(0.85)(-0.1) = \underline{-76.7 \text{ N/m}^2} \quad \text{for windward side}$$

$$p_h = (902.34)(0.85)(0.25) = \underline{191.7 \text{ N/m}^2}$$

$$p_h = (902.34)(0.85)(-0.6) = \underline{-460.2 \text{ N/m}^2 \text{ for leeward side}}$$



2-10

$$V = 90 \text{ mph}, h = 30 + \frac{11}{2} = 35.5 \text{ ft}$$

$$I = 1.15, \gamma_g = 900 \text{ ft}, \alpha = 9.5, k_{\gamma g} = 1$$

and $k_d = 1$

$$K_h = 2.01 \left(\frac{35.5}{900} \right)^{2/9.5} = 1.02$$

$$q_h = 0.00256 (1.02)(1)(1)(90)^2 (1.15) = 24.32 \text{ psf}$$

$$G = 0.85$$

$$\text{Roof Slope: } \theta = \tan^{-1}(11/20) = 28.8^\circ$$

$$\frac{h}{L} = \frac{35.5}{40} = 0.89$$

$C_p = -0.3$ and 0.2 for windward side

$C_p = -0.6$ for leeward side

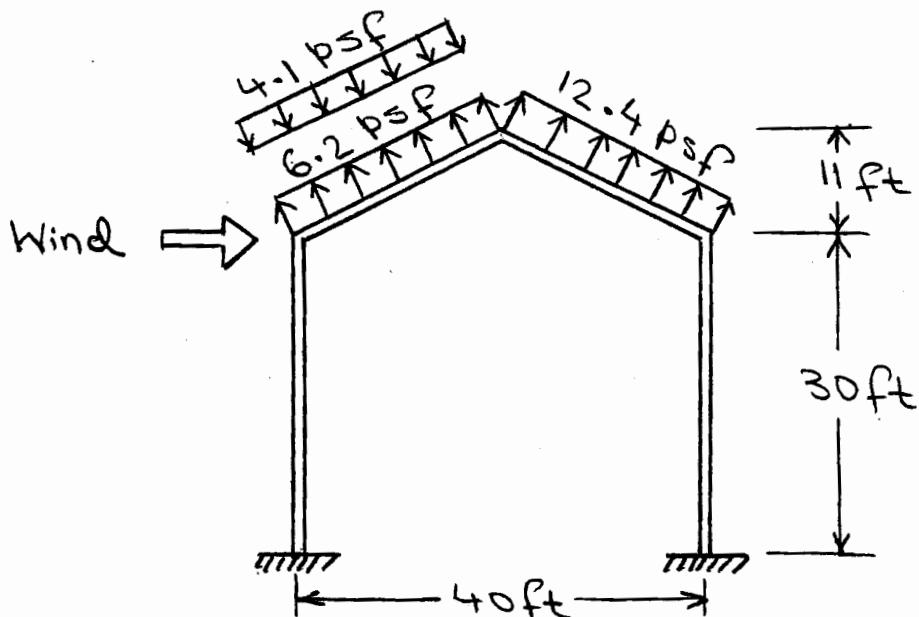
Thus, the wind pressures are:

$$p_h = 24.32 (0.85)(-0.3) = \underline{-6.2 \text{ psf}}$$

$$p_h = 24.32 (0.85)(0.2) = \underline{4.1 \text{ psf}}$$

$$p_h = 24.32 (0.85)(-0.6) = \underline{-12.4 \text{ psf}}$$

for leeward side



2-11 $V = 90 \text{ mph}$, $\Sigma = 1.15$, $\bar{z}_g = 900 \text{ ft}$, $\alpha = 9.5$

From the solution of Problem 2-10:

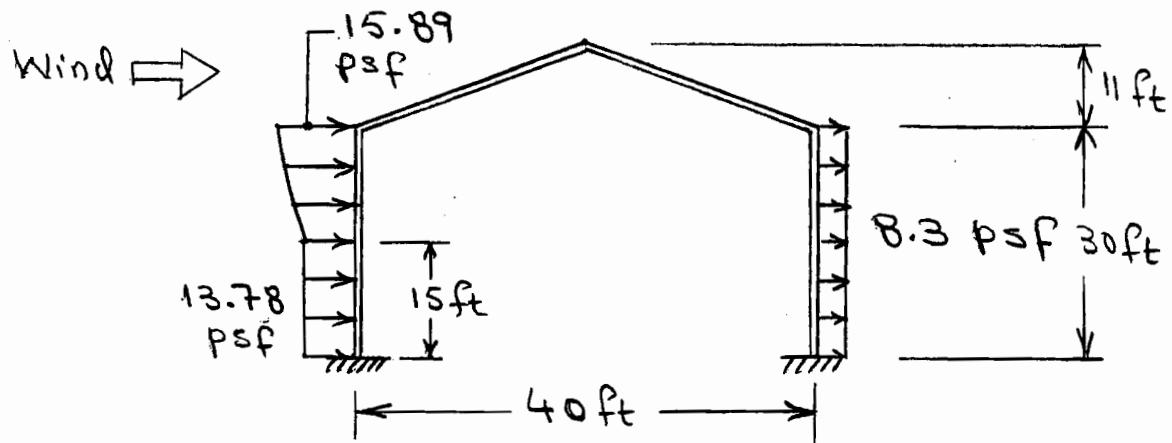
$$q_h = 24.32 \text{ psf} \quad \text{and} \quad G = 0.85$$

Leeward wall: For $L/B = 40/30 = 1.33$, $C_p = -0.4$

$$\begin{aligned} \text{Thus, the wind pressure, } p_h &= 24.32(0.85)(-0.4) \\ &= -8.3 \text{ psf} \end{aligned}$$

Windward wall: $C_p = 0.8$

$\bar{z} (\text{ft})$	$K_{\bar{z}}$	$q_{\bar{z}} (\text{psf})$	$p_{\bar{z}} (\text{psf})$
30	0.98	23.37	15.89
25	0.95	22.65	15.4
20	0.90	21.46	14.59
15	0.85	20.27	13.78



2.12 $p_g = 20 \text{ psf}$, $C_e = 1$, $C_t = 1$, $I = 1.2$

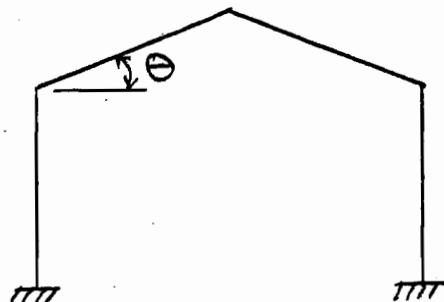
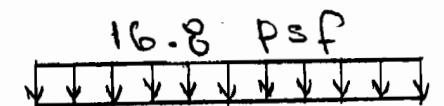
$$p_f = 0.7 C_e C_t I p_g = 0.7 (1)(1)(1.2)(20) = 16.8 \text{ psf}$$

$$\Theta = \tan^{-1}(11/20) = 28.8^\circ, \frac{70}{W} + 0.5 = \frac{70}{20} + 0.5 = 4^\circ$$

Therefore, the minimum values of p_f need not be considered.

$$C_s = 1$$

$$\text{Balanced load} = p_s = C_s p_f = 1(16.8) = \underline{\underline{16.8 \text{ psf}}}$$



Balanced
Snow Load

2.13 $P_g = 1.2 \text{ kN/m}^2$, $C_e = 1$, $C_t = 1$, $I = 1.1$

$$P_f = 0.7 C_e C_t I P_g = 0.7(1)(1)(1.1)(1.2) = 0.92 \text{ kN/m}^2$$

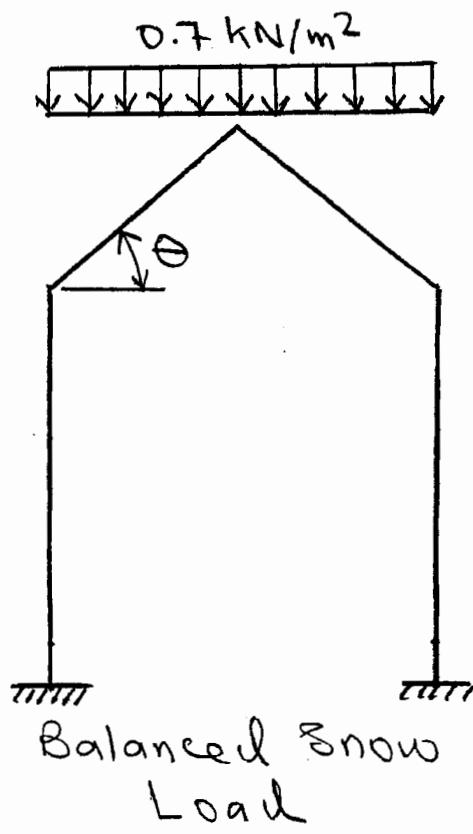
$$\Theta = \tan^{-1}(5/6) = 39.8^\circ, W = 6m = 19.7 \text{ ft}$$

$$\frac{70}{W} + 0.5 = \frac{70}{19.7} + 0.5 = 4.1^\circ$$

Therefore, the minimum values of P_f need not be considered.

$$C_s = 1 - \frac{\Theta - 30^\circ}{40^\circ} = 0.76$$

$$\text{Balanced Load} = P_s = C_s P_f = 0.76(0.92) = \underline{0.7 \text{ kN/m}^2}$$



Chapter Three

Equilibrium and Support Reactions

CHAPTER 3

3.1 (a) Internally stable, with $r=3$. It is statically determinate

(b) Internally stable, with $r=5$. It is statically indeterminate. $i_e = 5-3 = \underline{2}$

(c) Internally unstable, with $r=6$ and $e_c=2$.
As $r > 3+e_c$, the structure is
statically indeterminate. $i_e = 6-(3+2) = \underline{1}$

(d) Internally unstable with $r=3$ and $e_c=1$.
As $r < 3+e_c$, the beam is statically unstable.

3.2 (a) Internally stable, with $r=5$. It is statically indeterminate. $i_e = 5-3 = \underline{2}$.

(b) Internally unstable, with $r=5$ and $e_c=2$.
As $r = 3+e_c$, the beam is statically determinate.

(c) Internally stable, with $r=6$. It is statically indeterminate. $i_e = 6-3 = \underline{3}$

(d) Internally unstable, with $r=6$ and $e_c=1$.
As $r > 3+e_c$, the arch is statically indeterminate. $i_e = 6-(3+1) = \underline{2}$.

(e) Internally unstable, with $r=5$ and $e_c=2$.
As $r = 3+e_c$, the frame is statically determinate.

3.3 (a) Internally unstable, with $r=5$ and $e_c=3$.

As $r < 3 + e_c$, the beam is statically unstable.

(b) Internally unstable, with $r=4$ and $e_c=2$.

As $r < 3 + e_c$, the beam is statically unstable.

(c) Internally stable, with $r=4$. It is
statically indeterminate. i.e. $4 - 3 = 1$.

(d) Internally unstable, with $r=4$ and $e_c=2$.

As $r < 3 + e_c$, the frame is statically unstable.

3.4 (a) Internally stable, with $r = 4$.

It is statically indeterminate.

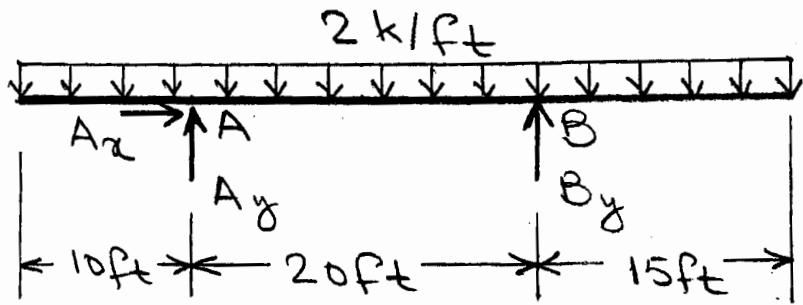
$$i_e = 4 - 3 = \underline{1}.$$

(b) Internally unstable, with $r = 4$ and $e_c = 2$. As $r < 3 + e_c$, the frame is statically unstable.

(c) Internally stable, with $r = 6$. It is statically indeterminate. $i_e = 6 - 3 = \underline{3}$.

(d) Internally unstable, with $r = 6$ and $e_c = 3$. As $r = 3 + e_c$, the frame is statically determinate.

3.5



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

$$+\leftarrow \sum M_B = 0$$

$$-A_y(20) + 2(45)(7.5) = 0$$

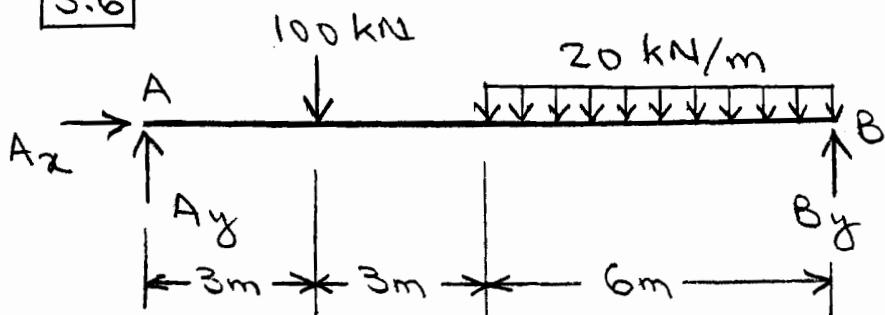
$$\underline{A_y = 33.75 \text{ k} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$33.75 - 2(45) + B_y = 0$$

$$\underline{B_y = 56.25 \text{ k} \uparrow}$$

3.6



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

$$+\leftarrow \sum M_B = 0$$

$$-A_y(12) + 100(9) + 20(6)3 = 0$$

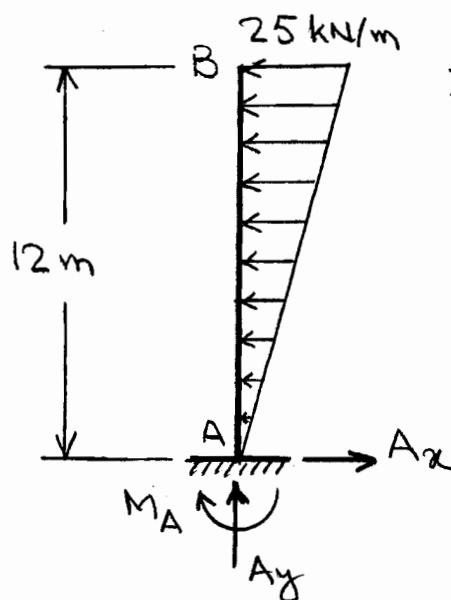
$$\underline{A_y = 105 \text{ kN} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$105 - 100 - 20(6) + B_y = 0$$

$$\underline{B_y = 115 \text{ kN} \uparrow}$$

3.7



$$\xrightarrow{+} \sum F_x = 0$$

$$-\frac{1}{2}(25)(12) + A_x = 0$$

$$\underline{A_x = 150 \text{ kN} \rightarrow}$$

$$\sum F_y = 0$$

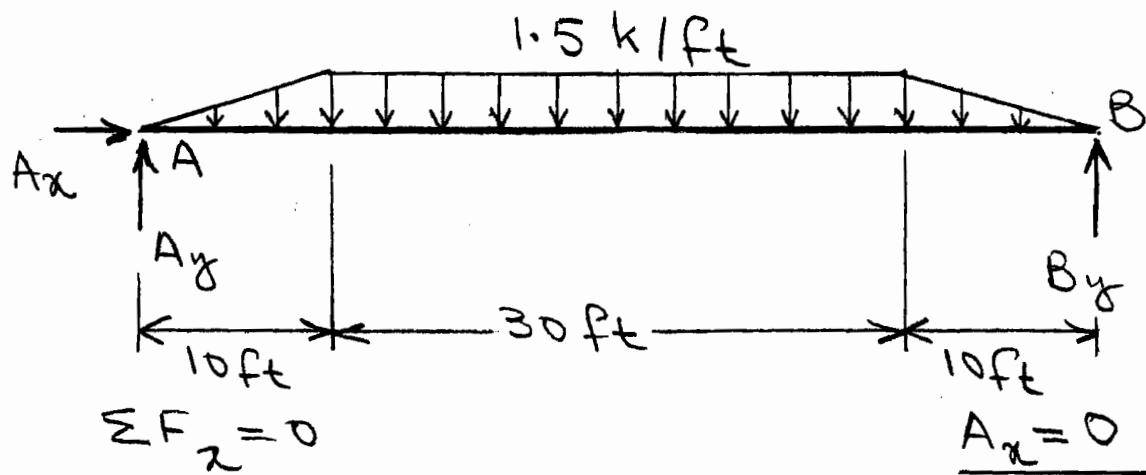
$$\underline{A_y = 0}$$

$$+\zeta \sum M_A = 0$$

$$-M_A + \frac{1}{2}(25)(12)(8) = 0$$

$$\underline{M_A = 1200 \text{ kN.m}}$$

3.8



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

$$+\zeta \sum M_B = 0$$

$$-A_y(50) + \frac{1}{2}(1.5)10\left(\frac{10}{3} + 40\right) + 1.5(30)(25)$$

$$+ \frac{1}{2}(1.5)10\left(\frac{2}{3}\right)(10) = 0$$

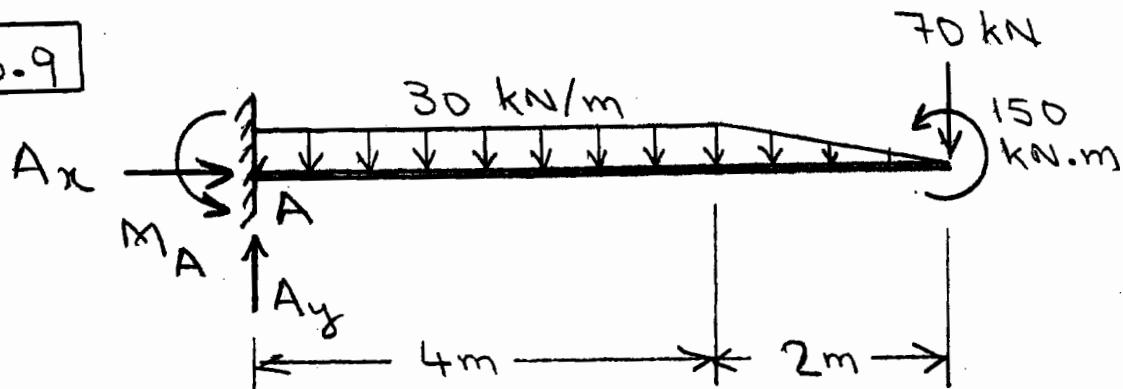
$$\underline{A_y = 30 \text{ k} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$30 - \frac{1}{2}(1.5)(10)(2) - 1.5(30) + B_y = 0$$

$$\underline{B_y = 30 \text{ k} \uparrow}$$

3.9



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

$$+\uparrow \sum F_y = 0$$

$$A_y - 30(4) - \frac{1}{2}(30)(2) - 70 = 0$$

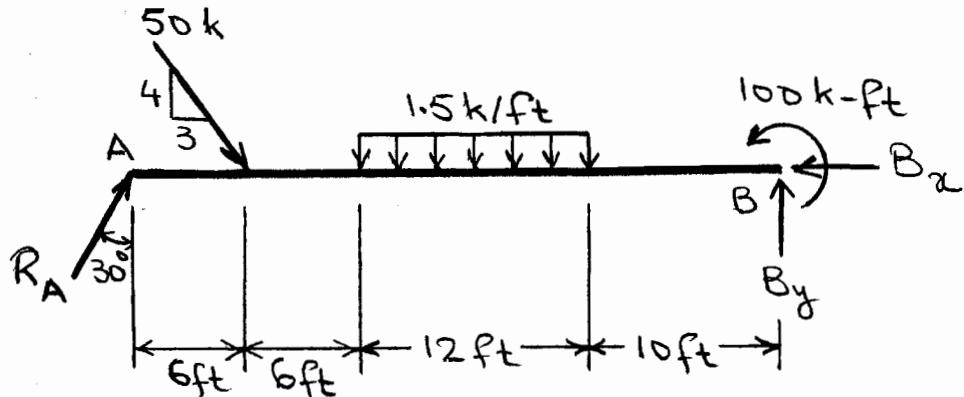
$$\underline{A_y = 220 \text{ kN} \uparrow}$$

$$+\zeta \sum M_A = 0$$

$$M_A - 30(4)(2) - \frac{1}{2}(30)(2)(4 + \frac{2}{3}) - 70(6) + 150 = 0$$

$$\underline{M_A = 650 \text{ kN.m} \swarrow}$$

3.10



$$+\zeta \sum M_B = 0$$

$$-R_A (\cos 30^\circ)(34) + \frac{4}{5}(50)(28) + 1.5(12)(16) + 100 = 0$$

$$\underline{R_A = 51.2 \text{ k} \uparrow}$$

$$\stackrel{\rightarrow}{\sum F_x} = 0$$

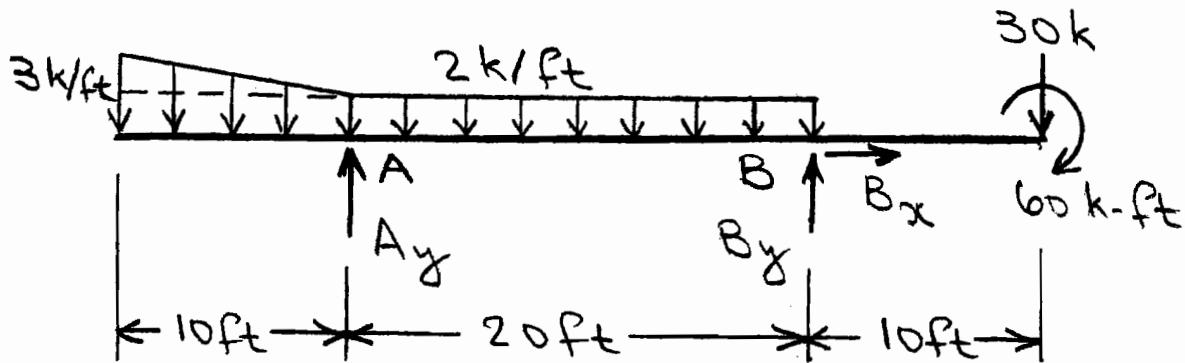
$$51.2(\sin 30^\circ) + \frac{3}{5}(50) - B_x = 0 \quad \underline{B_x = 55.6 \text{ k} \leftarrow}$$

$$+\uparrow \sum F_y = 0$$

$$51.2(\cos 30^\circ) - \frac{4}{5}(50) - 1.5(12) + B_y = 0$$

$$\underline{B_y = 13.65 \text{ k} \uparrow}$$

3.11



$$\sum F_x = 0$$

$$\underline{B_x = 0}$$

$$+\zeta \sum M_A = 0$$

$$\frac{1}{2}(1)(10)\left(\frac{20}{3}\right) - 2(30)(5) + B_y(20) - 30(30) - 60 = 0$$

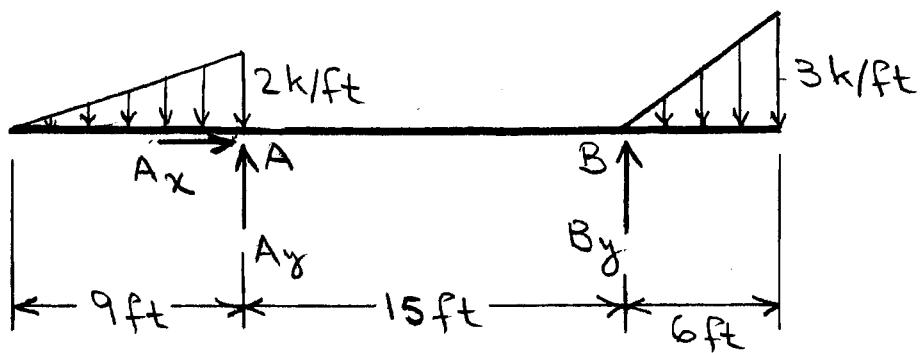
$$\underline{B_y = 61.33 \text{ k} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$-\frac{1}{2}(1)(10) - 2(30) + A_y + 61.33 - 30 = 0$$

$$\underline{A_y = 33.67 \text{ k} \uparrow}$$

3.12



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

$$+\leftarrow \sum M_B = 0$$

$$\frac{1}{2}(2)(9)\left[\frac{1}{3}(9)+15\right] - A_y(15) - \frac{1}{2}(3)(6)(4) = 0$$

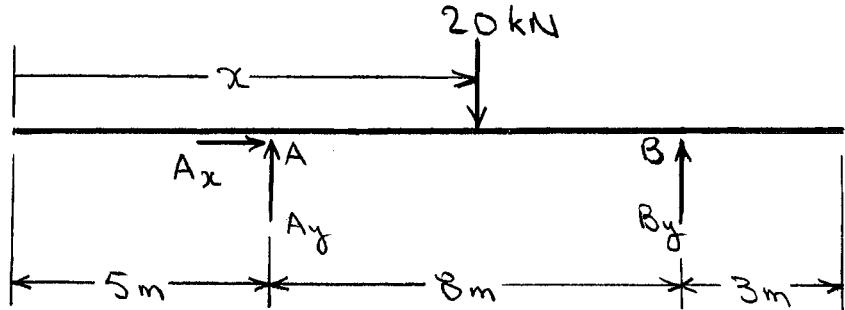
$$\underline{A_y = 8.4 k \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$-\frac{1}{2}(2)(9) + 8.4 + B_y - \frac{1}{2}(3)(6) = 0$$

$$\underline{B_y = 9.6 k \uparrow}$$

3.14



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

$$+\uparrow \sum M_B = 0$$

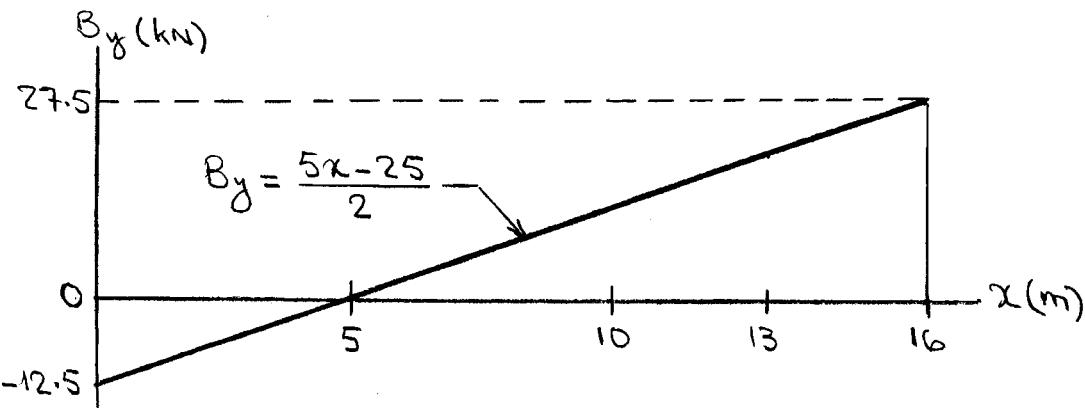
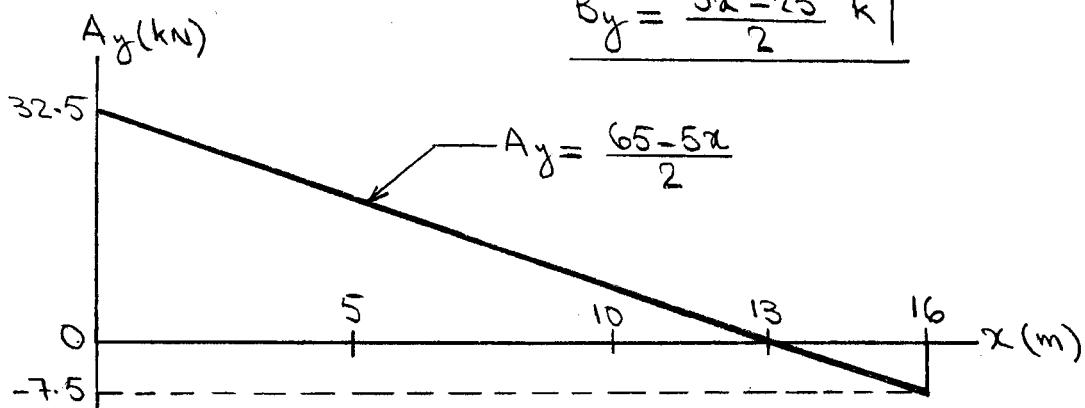
$$-A_y(8) + 20(13-x) = 0$$

$$\underline{A_y = \frac{65-5x}{2} \text{ k}\uparrow}$$

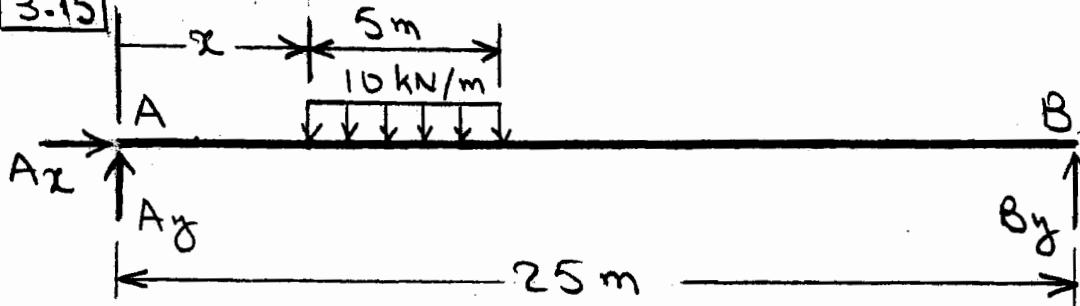
$$+\uparrow \sum F_y = 0$$

$$\underline{\frac{65-5x}{2} - 20 + B_y = 0}$$

$$\underline{B_y = \frac{5x-25}{2} \text{ k}\uparrow}$$



3.15



$$\sum F_x = 0$$

$$A_x = 0$$

$0 \leq x \leq 20 \text{ m}:$

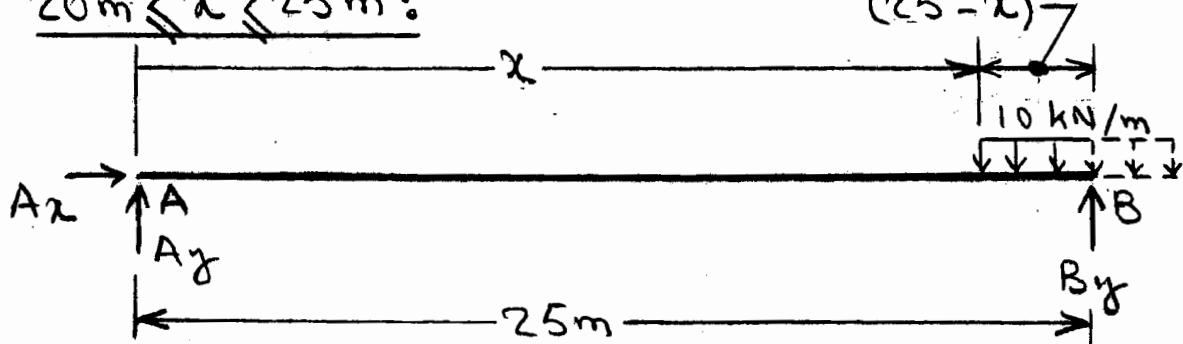
$$+\zeta \sum M_B = 0 \quad -A_y(25) + 10(5)[25 - (x+25)] = 0$$

$$A_y = 45 - 2x \text{ kN} \uparrow$$

$$+\uparrow \sum F_y = 0 \quad (45 - 2x) - 10(5) + B_y = 0$$

$$B_y = 5 + 2x \text{ kN} \uparrow$$

$20 \text{ m} \leq x \leq 25 \text{ m}:$



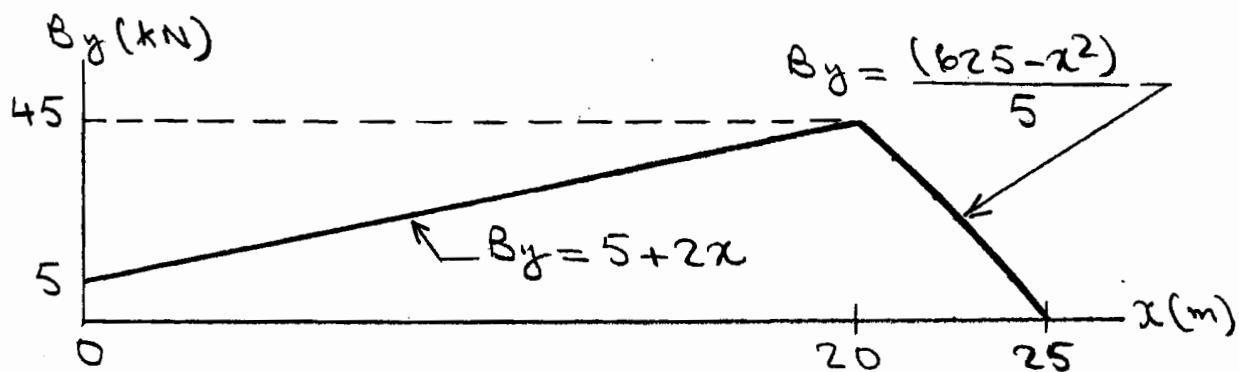
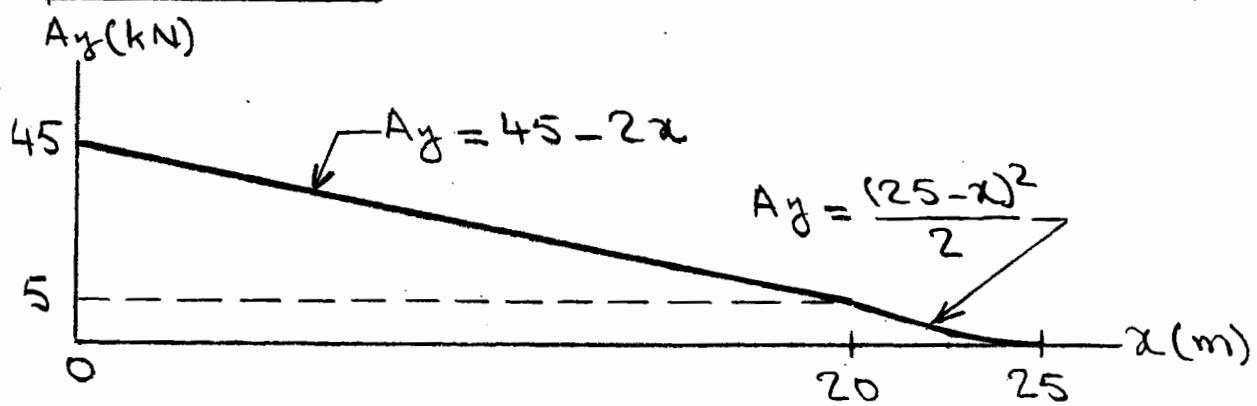
$$+\zeta \sum M_B = 0$$

$$-A_y(25) + 10(25-x)\left(\frac{25-x}{2}\right) = 0 \quad A_y = \frac{(25-x)^2}{5} \text{ kN} \uparrow$$

$$+\uparrow \sum F_y = 0 \quad \frac{(25-x)^2}{5} - 10(25-x) + B_y = 0$$

$$B_y = \frac{(625-x^2)}{5} \text{ kN} \uparrow$$

13.15 (contd.)

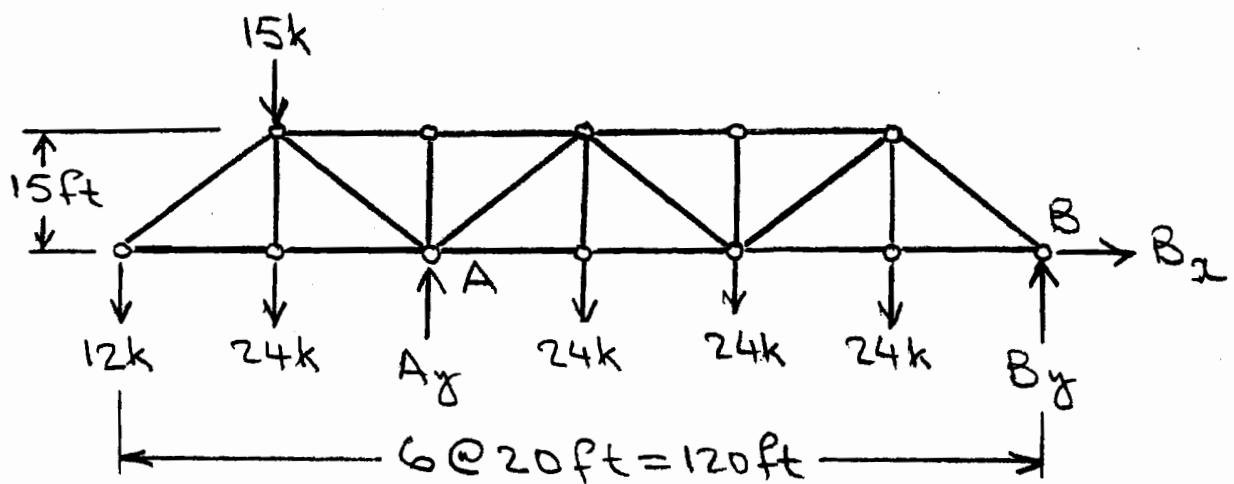


3.16

$$\underline{B_x = 0}$$

$$A_y = B_y = \frac{2(50) + 70}{2} = \underline{85 \text{ kN} \uparrow}$$

3.17



$$\sum F_x = 0$$

$$\underline{B_x = 0}$$

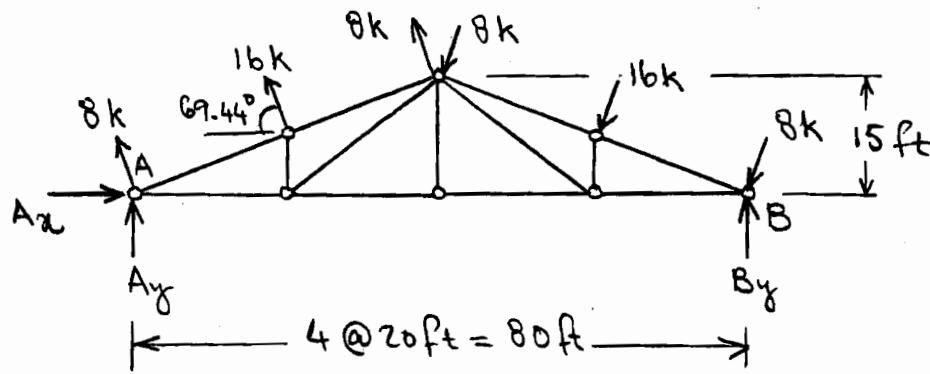
$$A_y = 12 \left(\frac{b}{4} \right) + (24 + 15) \left(\frac{5}{4} \right) + 24 \left(\frac{3}{4} + \frac{1}{2} + \frac{1}{4} \right)$$

$$\underline{A_y = 102.75 \text{ k} \uparrow}$$

$$B_y = 12 + 15 + 4(24) - 102.75$$

$$\underline{B_y = 20.25 \text{ k} \uparrow}$$

3.18



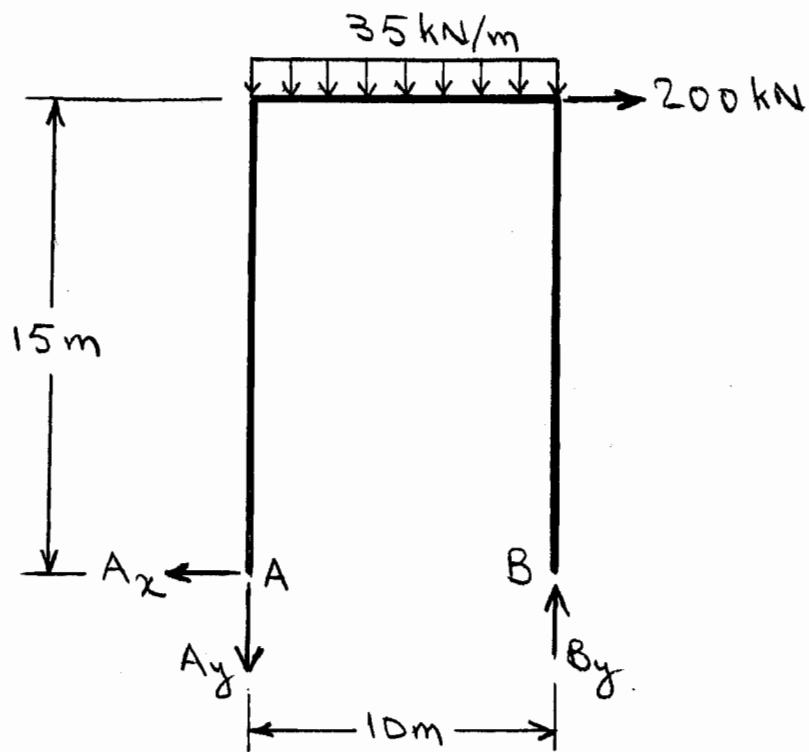
$$\rightarrow \sum F_x = 0 \quad A_x - (8 + 16 + 8 + 8 + 16 + 8) \cos 69.44^\circ = 0 \\ \underline{A_x = 22.48 k \rightarrow}$$

$$+G \sum M_B = 0 \quad -A_y(80) + 2[16(7.5) + 8(15)] \cos 69.44^\circ \\ + [-8(80) - 16(60) + 16(20)] \sin 69.44^\circ = 0$$

$$A_y = -12.87 k \quad \underline{A_y = 12.87 k \downarrow}$$

$$+\uparrow \sum F_y = 0 \quad -12.87 + (8 + 16 + 8 - 8 - 16 - 8) \sin 69.44^\circ + B_y = 0 \\ \underline{B_y = 12.87 k \uparrow}$$

3.19



$$+\rightarrow \sum F_x = 0$$

$$\underline{A_x = 200 \text{ kN} \leftarrow}$$

$$+\circlearrowleft \sum M_B = 0$$

$$A_y(10) + 35(10)(5) - 200(15) = 0$$

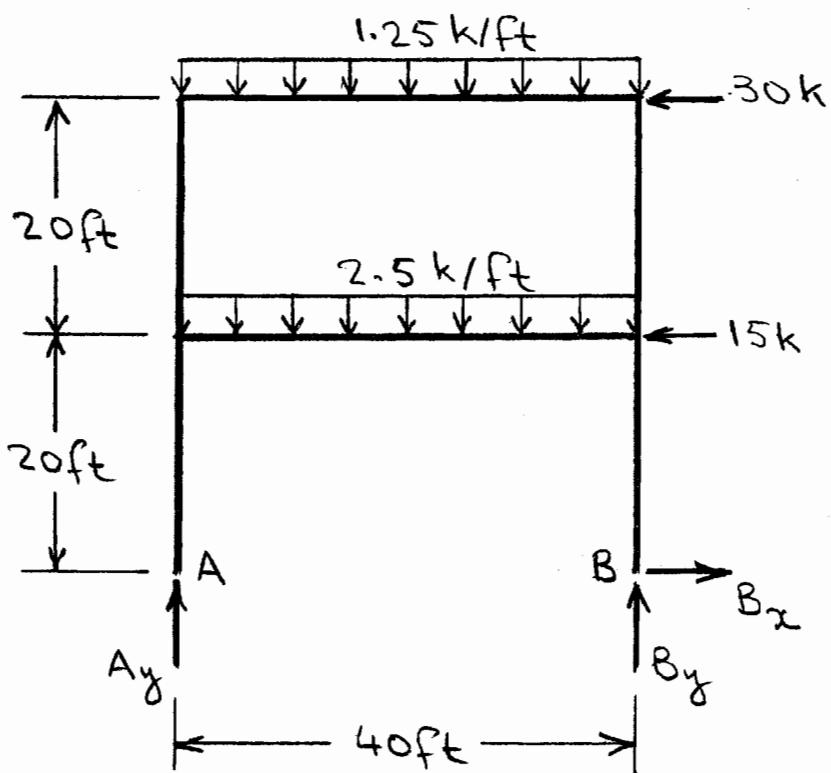
$$\underline{A_y = 125 \text{ kN} \downarrow}$$

$$+\uparrow \sum F_y = 0$$

$$-125 - 35(10) + B_y = 0$$

$$\underline{B_y = 475 \text{ kN} \uparrow}$$

3.20



$$\rightarrow \sum F_x = 0 \quad B_x - 15 - 30 = 0 \quad \underline{B_x = 45 k \rightarrow}$$

$$+ \zeta \sum M_B = 0$$

$$-A_y(40) + 2.5(40)(20) + 15(20) + 1.25(40)(20) + 30(40) = 0$$

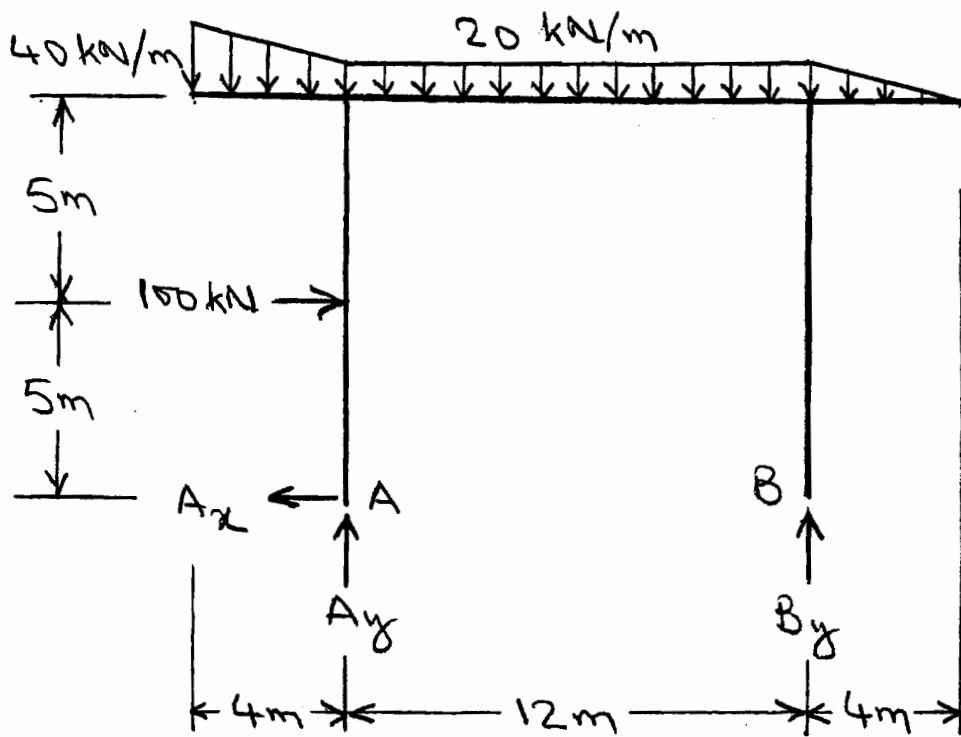
$$A_y = 112.5 k \uparrow$$

$$+\uparrow \sum F_y = 0$$

$$112.5 - 2.5(40) - 1.25(40) + B_y = 0$$

$$\underline{B_y = 37.5 k \uparrow}$$

3.21



$$\rightarrow \sum F_x = 0 \quad 100 - A_x = 0 \quad \underline{A_x = 100 \text{ kN} \leftarrow}$$

$$+\downarrow \sum M_A = 0$$

$$-100(5) + \frac{1}{2}(20)(4)\left(\frac{2}{3}\right)4 - 20(16)4 + B_y(12)$$

$$-\frac{1}{2}(20)4\left(12 + \frac{4}{3}\right) = 0$$

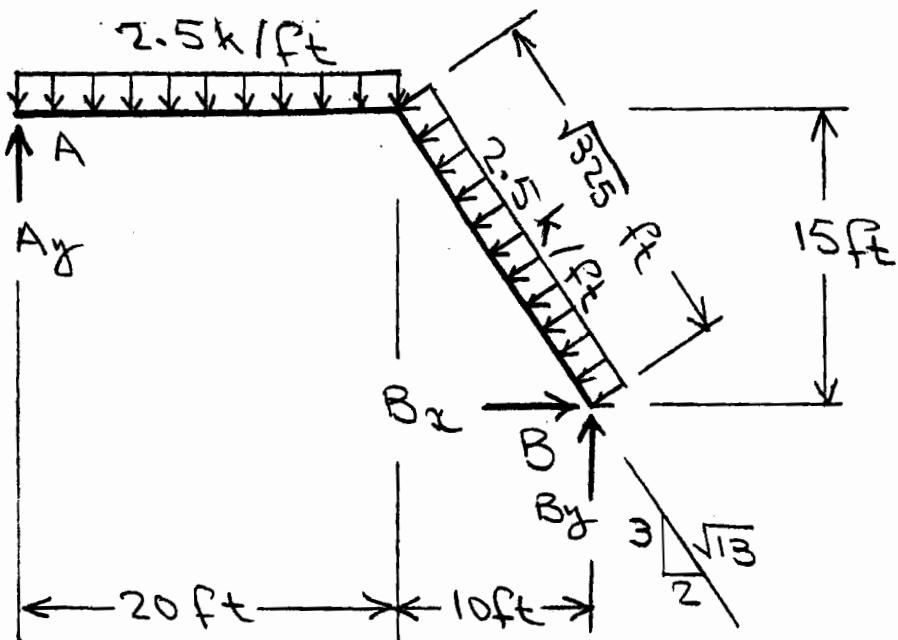
$$\underline{B_y = 183.89 \text{ kN} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$A_y - \frac{1}{2}(20)4 - 20(16) - \frac{1}{2}(20)4 + 183.89 = 0$$

$$\underline{A_y = 216.11 \text{ kN} \uparrow}$$

3.22



$$\rightarrow \sum F_x = 0 \quad B_x - 2.5(\sqrt{325})\left(\frac{3}{\sqrt{13}}\right) = 0$$

$$\underline{B_x = 37.5 k \rightarrow}$$

$$+\zeta \sum M_B = 0$$

$$-A_y(30) + 2.5(20)(20) + 2.5(\sqrt{325})\left(\frac{\sqrt{325}}{2}\right) = 0$$

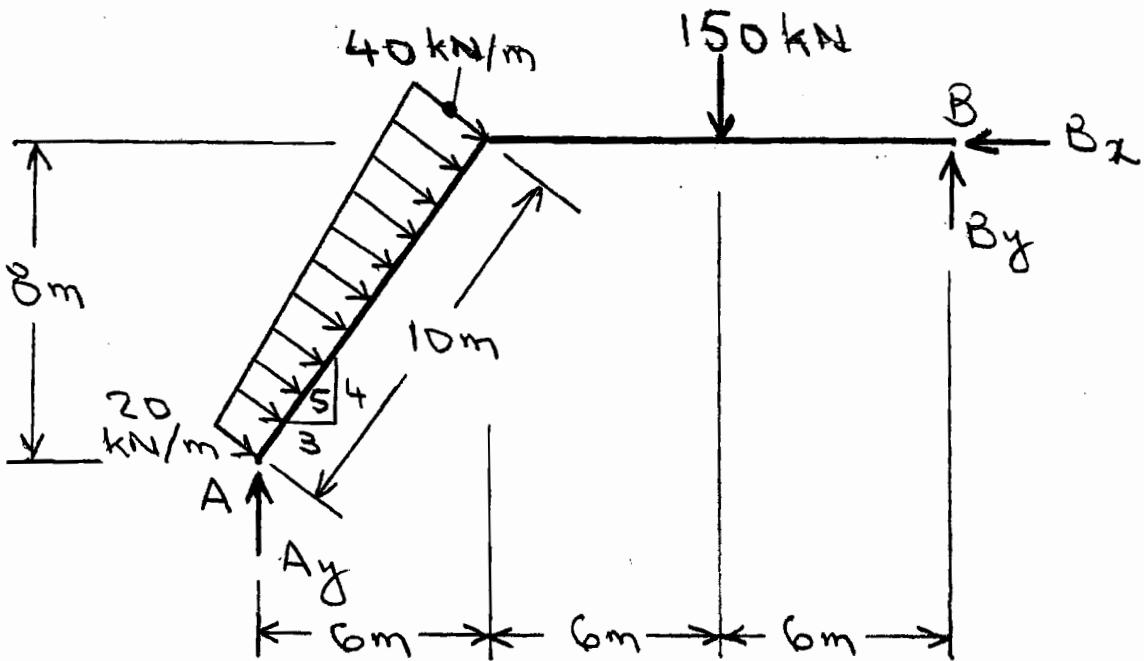
$$\underline{A_y = 46.875 k \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$46.875 - 2.5(20) - 2.5(\sqrt{325})\left(\frac{2}{\sqrt{13}}\right) + B_y = 0$$

$$\underline{B_y = 28.125 k \uparrow}$$

3.23



$$\stackrel{+}{\rightarrow} \sum F_x = 0 \quad \left(\frac{20+40}{2} \right) 10 \left(\frac{4}{5} \right) - B_x = 0$$

$$\underline{B_x = 240 \text{ kN} \leftarrow}$$

$$+\zeta \sum M_A = 0$$

$$-20(10)5 - \frac{1}{2}(20)10\left(\frac{20}{3}\right) - 150(12) + 240(8)$$

$$+ B_y (18) = 0$$

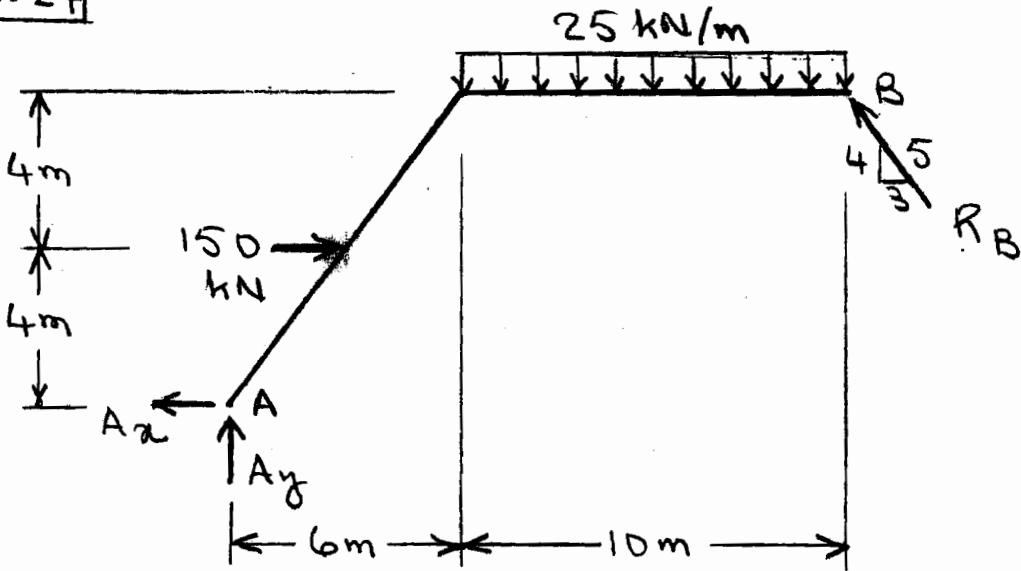
$$\underline{B_y = 85.93 \text{ kN} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$A_y - \left(\frac{20+40}{2} \right) 10 \left(\frac{3}{5} \right) - 150 + 85.93 = 0$$

$$\underline{A_y = 244.07 \text{ kN} \uparrow}$$

3.24



$$+\text{C} \sum M_A = 0$$

$$-150(4) - 25(10)(11) + \frac{3}{5}R_B(8) + \frac{4}{5}R_B(16) = 0$$

$$\underline{R_B = 190.3 \text{ kN} \uparrow}$$

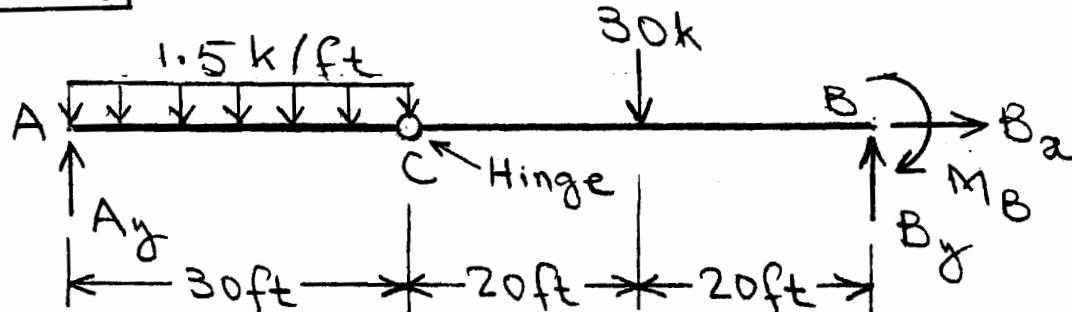
$$\rightarrow \sum F_x = 0 \quad -A_x + 150 - \frac{3}{5}(190.3) = 0$$

$$\underline{A_x = 35.8 \text{ kN} \leftarrow}$$

$$+\uparrow \sum F_y = 0 \quad A_y - 25(10) + \frac{4}{5}(190.3) = 0$$

$$\underline{A_y = 97.7 \text{ kN} \uparrow}$$

3.25



$$\sum F_x = 0$$

$$\underline{B_x = 0}$$

$$+ \text{C} \sum M_C^{AC} = 0$$

$$-A_y(30) + 1.5(30)15 = 0$$

$$\underline{A_y = 22.5 \text{ k} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$22.5 - 1.5(30) - 30 + B_y = 0$$

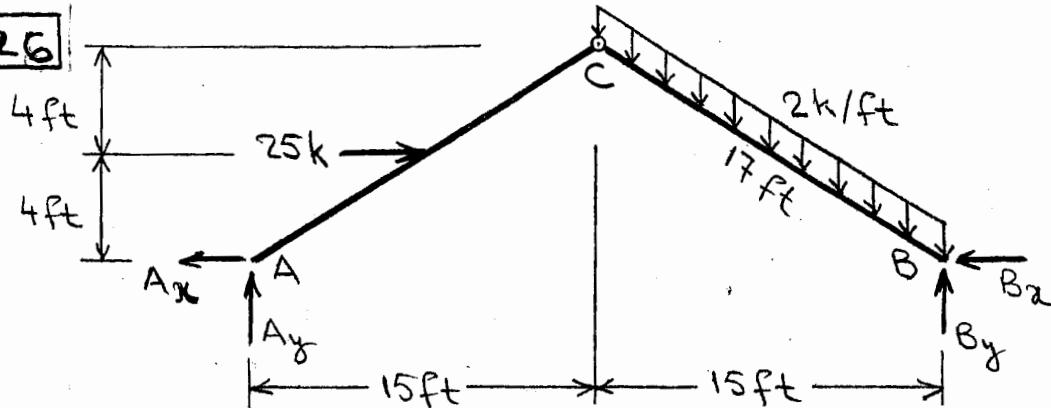
$$\underline{B_y = 52.5 \text{ k} \uparrow}$$

$$+\text{C} \sum M_A = 0$$

$$-1.5(30)15 - 30(50) + 52.5(70) - M_B = 0$$

$$\underline{M_B = 1500 \text{ k-ft}}$$

3.26



$$+\zeta \sum M_B = 0$$

$$-A_y(30) - 25(4) + 2\left(\frac{15}{2}\right) = 0 \quad \underline{A_y = 5.17 \text{ k} \uparrow}$$

$$+\zeta \sum M_C^{AC} = 0$$

$$-5.17(15) - A_x(8) + 25(4) = 0 \quad \underline{A_x = 2.81 \text{ k} \leftarrow}$$

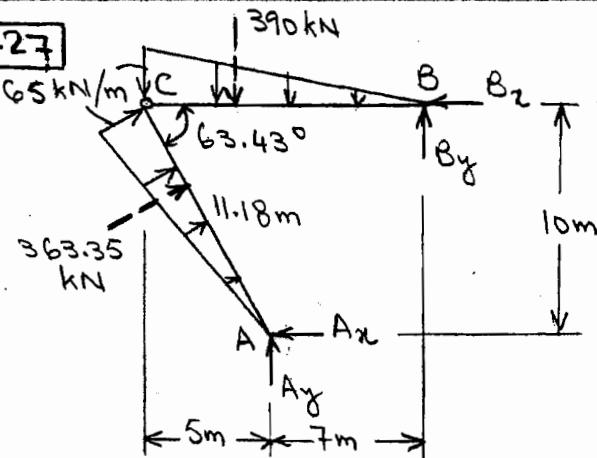
$$\rightarrow \sum F_x = 0$$

$$-2.81 + 25 - B_x = 0 \quad \underline{B_x = 22.19 \text{ k} \leftarrow}$$

$$+\uparrow \sum F_y = 0$$

$$5.17 - 2(17) + B_y = 0 \quad \underline{B_y = 28.83 \text{ k} \uparrow}$$

3.27



$$+\zeta \sum M_C^{BC} = 0$$

$$-390\left(\frac{12}{3}\right) + B_y(12) = 0$$

$$\underline{B_y = 130 \text{ kN} \uparrow}$$

$$+\zeta \sum M_A = 0$$

$$B_x(10) + 130(7) + 390(1)$$

$$-363.35\left(\frac{2}{3}\right)11.18 = 0$$

$$\underline{B_x = 140.82 \text{ kN} \leftarrow}$$

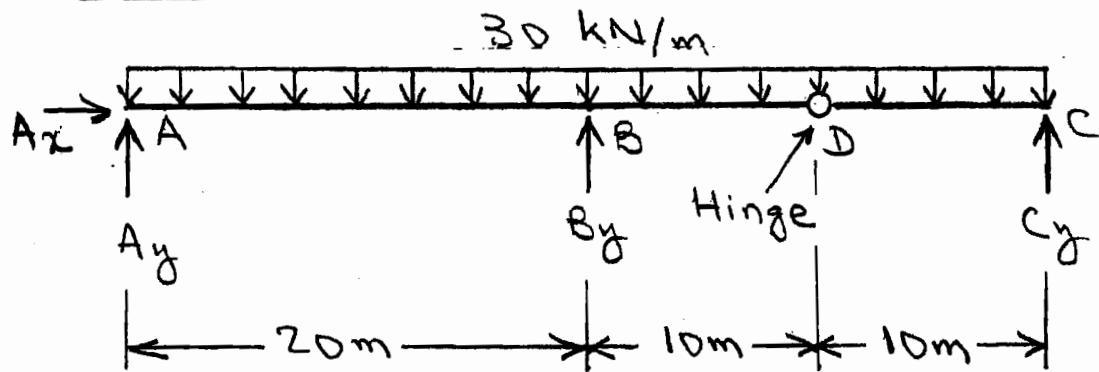
$$\rightarrow \sum F_x = 0 \quad 363.35 \sin 63.43^\circ - A_x - 140.82 = 0$$

$$\underline{A_x = 184.16 \text{ kN} \leftarrow}$$

$$+\uparrow \sum F_y = 0 \quad A_y + 363.35 \cos 63.43^\circ - 390 + 130 = 0$$

$$\underline{A_y = 97.48 \text{ kN} \uparrow}$$

3.28



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

$$+\zeta \sum M_D^{CD} = 0$$

$$C_y(10) - 30(10)5 = 0$$

$$\underline{C_y = 150 \text{ kN} \uparrow}$$

$$+\zeta \sum M_A = 0$$

$$-30(40)20 + B_y(20) + 150(40) = 0$$

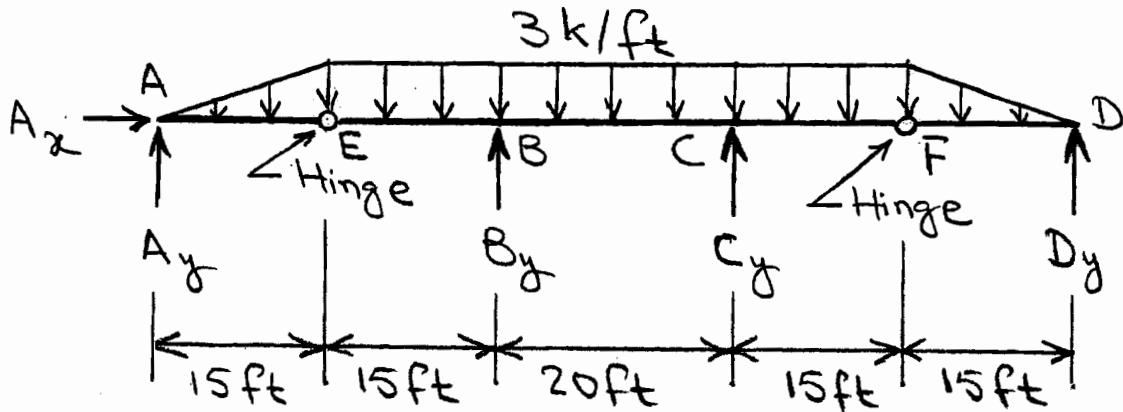
$$\underline{B_y = 900 \text{ kN} \uparrow}$$

$$+\uparrow \sum F_y = 0$$

$$A_y - 30(40) + 900 + 150 = 0$$

$$\underline{A_y = 150 \text{ kN} \uparrow}$$

3.29

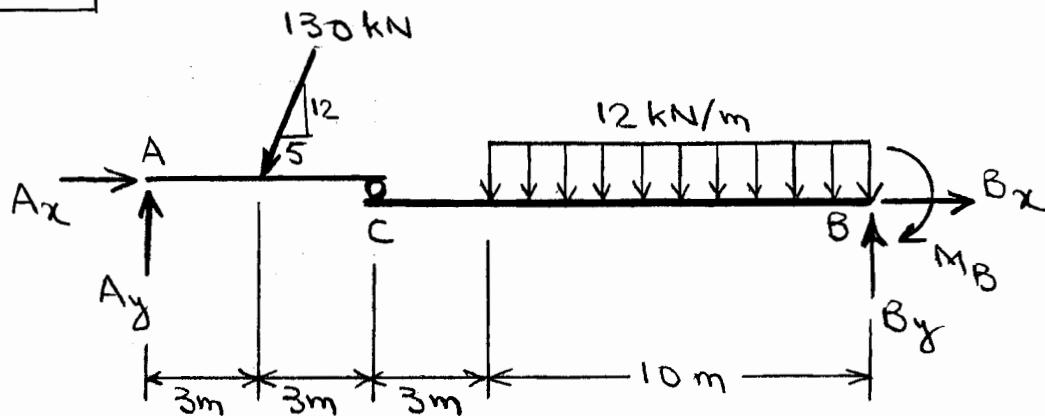


$$\begin{aligned} \sum F_x &= 0 & A_x &= 0 \\ +G \sum M_E^{AE} &= 0 & -A_y(15) + \frac{1}{2}(3)15(5) &= 0 \\ && A_y &= 7.5 \text{ k}\uparrow \\ +G \sum M_F^{DF} &= 0 & D_y(15) - \frac{1}{2}(3)15(5) &= 0 \\ && D_y &= 7.5 \text{ k}\uparrow \end{aligned}$$

$$\begin{aligned} +G \sum M_C &= 0 \\ -7.5(50) + \frac{1}{2}(3)15(40) + 3(50)10 - B_y(20) & \\ -\frac{1}{2}(3)15(20) + 7.5(30) &= 0 \\ B_y &= 90 \text{ k}\uparrow \end{aligned}$$

$$\begin{aligned} +\uparrow \sum F_y &= 0 \\ 7.5 - \frac{1}{2}(3)15 - 3(50) - \frac{1}{2}(3)15 + 90 + C_y + 7.5 &= 0 \\ C_y &= 90 \text{ k}\uparrow \end{aligned}$$

3.30



$$\rightarrow \sum F_x^{\text{AC}} = 0 \quad A_x - \frac{5}{13}(130) = 0 \quad \underline{A_x = 50 \text{ kN} \rightarrow}$$

$$+\zeta \sum M_C^{\text{AC}} = 0 \quad -Ay(6) + \frac{12}{13}(130)(3) = 0 \quad \underline{Ay = 60 \text{ kN} \uparrow}$$

$$\rightarrow \sum F_x = 0 \quad 50 - \frac{5}{13}(130) + B_x = 0 \quad \underline{B_x = 0}$$

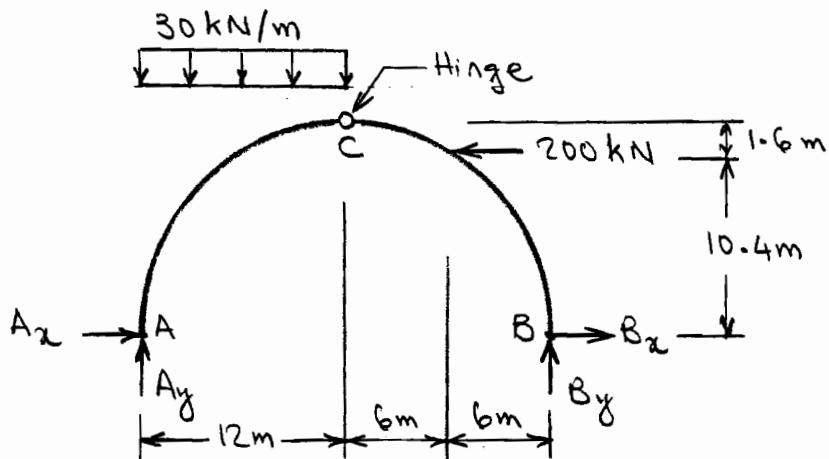
$$+\uparrow \sum F_y = 0 \quad 60 - \frac{12}{13}(130) - 12(10) + B_y = 0 \quad B_y = 180 \text{ kN} \uparrow$$

$$+\zeta \sum M_A = 0$$

$$-\frac{12}{13}(130)(3) - 12(10)(9+5) + 180(19) - M_B = 0$$

$$\underline{M_B = 1380 \text{ kN} \cdot \text{m} \swarrow}$$

3.31



$$+\text{C} \sum M_B = 0 \quad -A_y(24) + 30(12)(18) + 200(10.4) = 0$$

$$\underline{A_y = 356.67 \text{ kN} \uparrow}$$

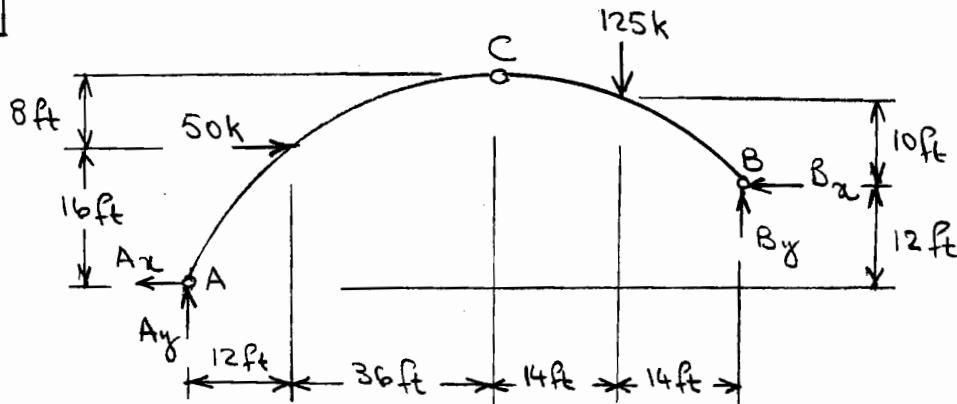
$$+\text{C} \sum M_C^{AC} = 0 \quad A_x(12) - 356.67(12) + 30(12)(6) = 0$$

$$\underline{A_x = 176.67 \text{ kN} \rightarrow}$$

$$\Sigma F_x = 0 \quad 176.67 - 200 + B_x = 0 \quad \underline{B_x = 23.33 \text{ kN} \rightarrow}$$

$$\Sigma F_y = 0 \quad 356.67 - 30(12) + B_y = 0 \quad \underline{B_y = 3.33 \text{ kN} \uparrow}$$

3.32



$$+\text{C} \sum M_C^{AC} = 0 \quad -A_x(24) - A_y(48) + 50(8) = 0$$

$$3A_x + 6A_y = 50 \quad \text{--- (1)}$$

$$+\text{C} \sum M_B = 0 \quad -A_x(12) - A_y(76) - 50(4) + 125(14) = 0$$

$$6A_x + 38A_y = 775 \quad \text{--- (2)}$$

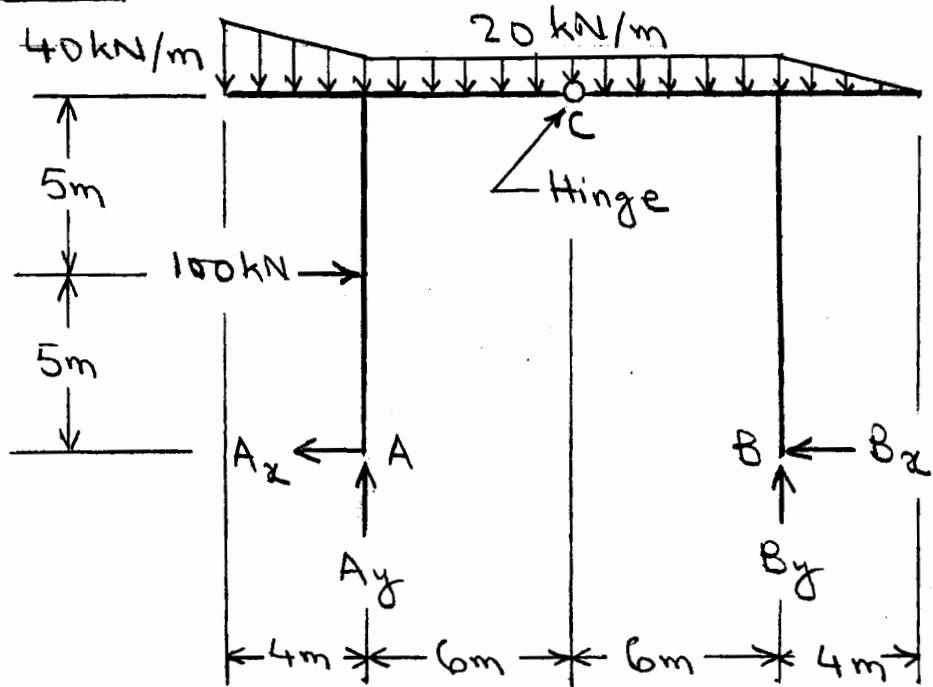
Solving (1) and (2) simultaneously, we obtain

$$A_x = -35.25 \text{ k} = \underline{35.25 \text{ k} \rightarrow} \quad \underline{A_y = 25.96 \text{ k} \uparrow}$$

$$\rightarrow \Sigma F_x = 0 \quad 35.25 + 50 - B_x = 0 \quad \underline{B_x = 85.25 \text{ k} \leftarrow}$$

$$\uparrow \Sigma F_y = 0 \quad 25.96 - 125 + B_y = 0 \quad \underline{B_y = 99.04 \text{ k} \uparrow}$$

3.33



$$+ \text{G} \sum M_A = 0$$

$$-100(5) + \frac{1}{2}(20)(4)\left(\frac{8}{3}\right) - 20(16)4 - \frac{1}{2}(20)4\left(12 + \frac{4}{3}\right)$$

$$+ B_y(12) = 0$$

$$\underline{B_y = 183.89 \text{ kN} \uparrow}$$

$$+ \text{G} \sum M_C^{BC} = 0$$

$$-20(6)3 - \frac{1}{2}(20)4\left(6 + \frac{4}{3}\right) + 183.89(6) - B_x(10) = 0$$

$$\underline{B_x = 45 \text{ kN} \leftarrow}$$

$$\rightarrow \sum F_x = 0 \quad 100 - A_x - 45 = 0$$

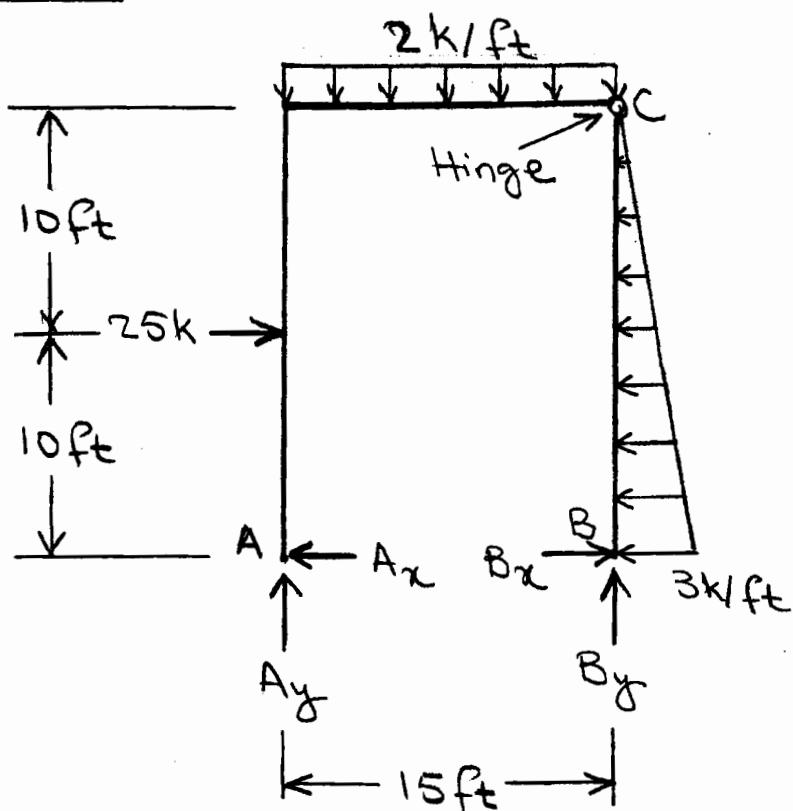
$$\underline{A_x = 55 \text{ kN} \leftarrow}$$

$$+\uparrow \sum F_y = 0 \quad A_y - \frac{1}{2}(20)4 - 20(16) - \frac{1}{2}(20)4$$

$$+ 183.89 = 0$$

$$\underline{A_y = 216.11 \text{ kN} \uparrow}$$

3.34



$$+\zeta \sum M_A = 0$$

$$-25(10) - 2(15)7.5 + \frac{1}{2}(3)20\left(\frac{20}{3}\right) + B_y(15) = 0$$

$$\underline{B_y = 18.33 \text{ k} \uparrow}$$

$$+\sum M_C = 0$$

$$-\frac{1}{2}(3)20\left(\frac{40}{3}\right) + B_x(20) = 0$$

$$\underline{B_x = 20 \text{ k} \rightarrow}$$

$$\stackrel{+}{\rightarrow} \sum F_x = 0$$

$$-A_x + 25 - \frac{1}{2}(3)20 + 20 = 0$$

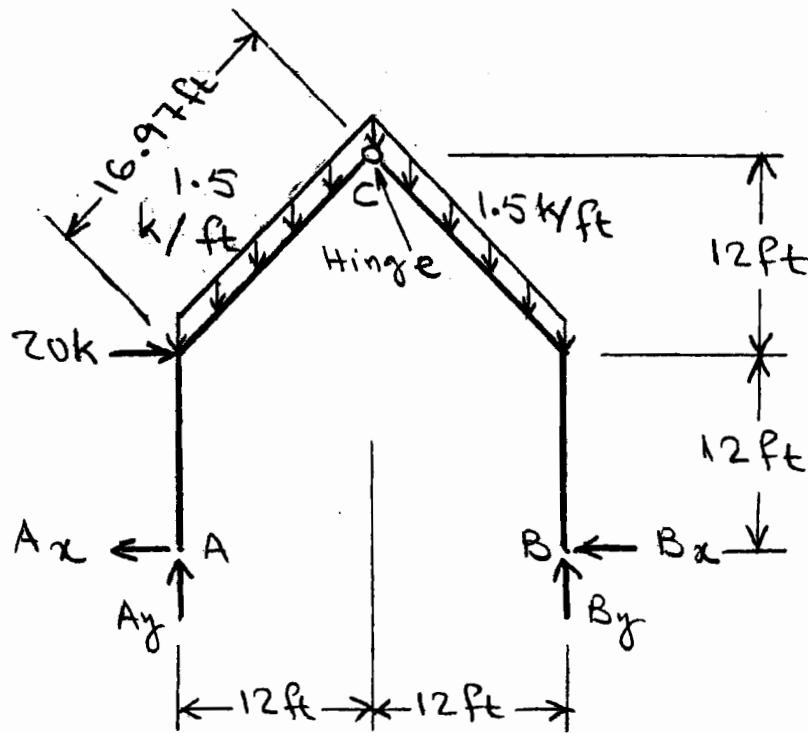
$$\underline{A_x = 15 \text{ k} \leftarrow}$$

$$+\uparrow \sum F_y = 0$$

$$A_y - 2(15) + 18.33 = 0$$

$$\underline{A_y = 11.67 \text{ k} \uparrow}$$

3.35



$$+\zeta \sum M_B = 0$$

$$-A_y(24) - 20(12) + 1.5(16.97)(18) + 1.5(16.97)(6) = 0$$

$$\underline{A_y = 15.46 \text{ k} \uparrow}$$

$$+\zeta \sum M_C^{AC} = 0$$

$$-A_x(24) - 15.46(12) + 20(12) + 1.5(16.97)(6) = 0$$

$$\underline{A_x = 8.63 \text{ k} \leftarrow}$$

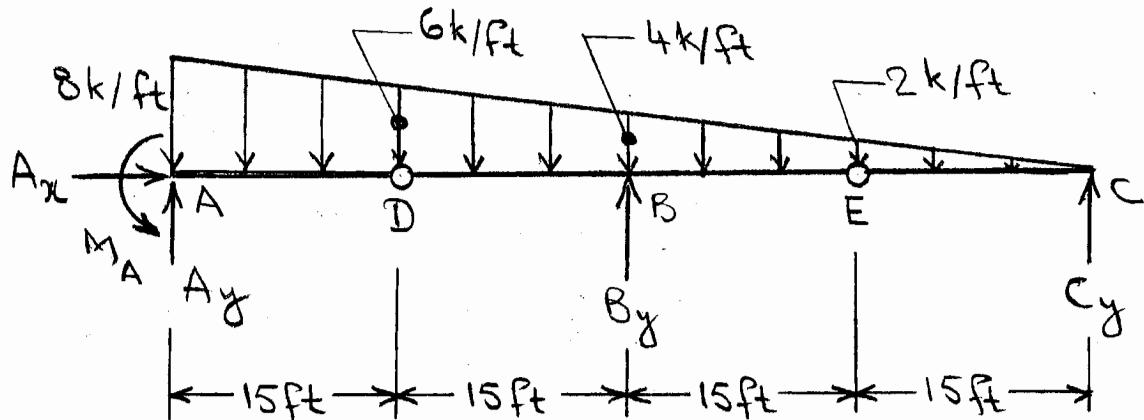
$$\Rightarrow \sum F_x = 0 \quad -8.63 + 20 - B_x = 0$$

$$\underline{B_x = 11.37 \text{ k} \leftarrow}$$

$$+\uparrow \sum F_y = 0 \quad 15.46 - 2(1.5)(16.97) + B_y = 0$$

$$\underline{B_y = 35.45 \text{ k} \uparrow}$$

3.36



$$\sum F_x = 0$$

$$A_x = 0$$

$$+\leftarrow \sum M_E^{CE} = 0 \quad C_y(15) - \frac{1}{2}(2)(15)(5) = 0$$

$$\underline{C_y = 5 k \uparrow}$$

$$+\zeta \sum M_D^{CD} = 0 \quad 5(45) - \frac{1}{2}(6)(45)(15) + B_y(15) = 0$$

$$\underline{B_y = 120 k \uparrow}$$

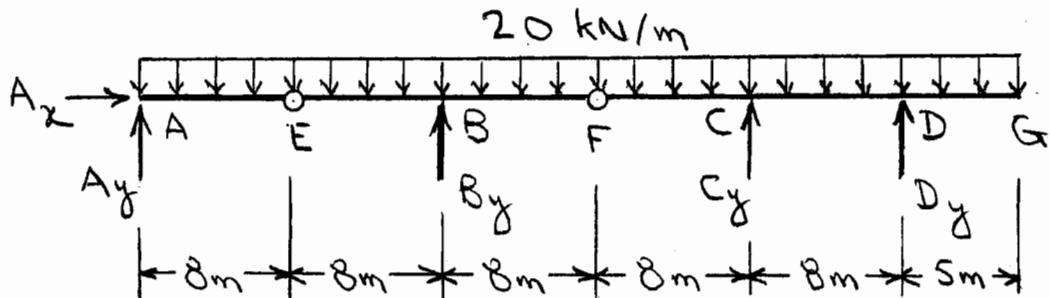
$$+\zeta \sum M_A = 0 \quad M_A - \frac{1}{2}(8)(60)(20) + 120(30) + 5(60) = 0$$

$$\underline{M_A = 900 k\cdot ft \curvearrowright}$$

$$+\uparrow \sum F_y = 0 \quad A_y - \frac{1}{2}(8)(60) + 120 + 5 = 0$$

$$\underline{A_y = 115 k \uparrow}$$

3.37



$$\sum F_x = 0$$

$$\underline{A_x = 0}$$

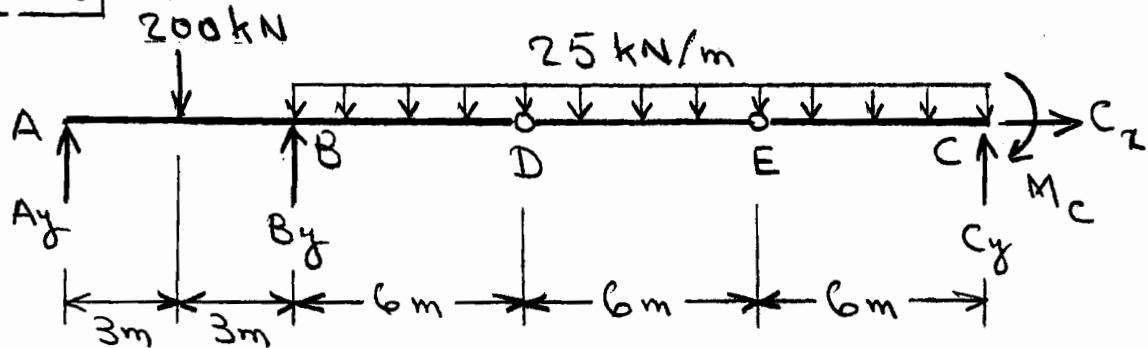
$$+\leftarrow \sum M_E^{AE} = 0 \quad -Ay(8) + 20(8)(4) = 0 \quad \underline{Ay = 80 \text{ kN} \uparrow}$$

$$+\leftarrow \sum M_F^{AF} = 0 \quad -80(24) + 20(24)(12) - By(8) = 0 \quad \underline{By = 480 \text{ kN} \uparrow}$$

$$+\leftarrow \sum M_D = 0 \quad -80(40) - 480(24) - Cy(8) + 20(45)(17.5) = 0 \quad \underline{Cy = 128.75 \text{ kN} \uparrow}$$

$$+\uparrow \sum F_y = 0 \quad 80 + 480 + 128.75 + Dy - 20(45) = 0 \quad \underline{Dy = 211.25 \text{ kN} \uparrow}$$

3.38



$$\sum F_x = 0$$

$$C_x = 0$$

$$+\sum M_D^{AD} = 0 \quad -Ay(12) + 200(9) - By(6) + 25(6)(3) = 0$$

$$2Ay + By = 375 \quad (1)$$

$$+\sum M_E^{AE} = 0 \quad -Ay(18) + 200(15) - By(12) + 25(12)(6) = 0$$

$$1.5Ay + By = 400 \quad (2)$$

Solving Eqs. (1) and (2) simultaneously, we obtain

$$Ay = -50 \text{ kN} = \underline{\underline{50 \text{ kN} \downarrow}}$$

$$By = \underline{\underline{475 \text{ kN} \uparrow}}$$

$$+\uparrow \sum F_y = 0 \quad -50 - 200 + 475 - 25(18) + Cy = 0$$

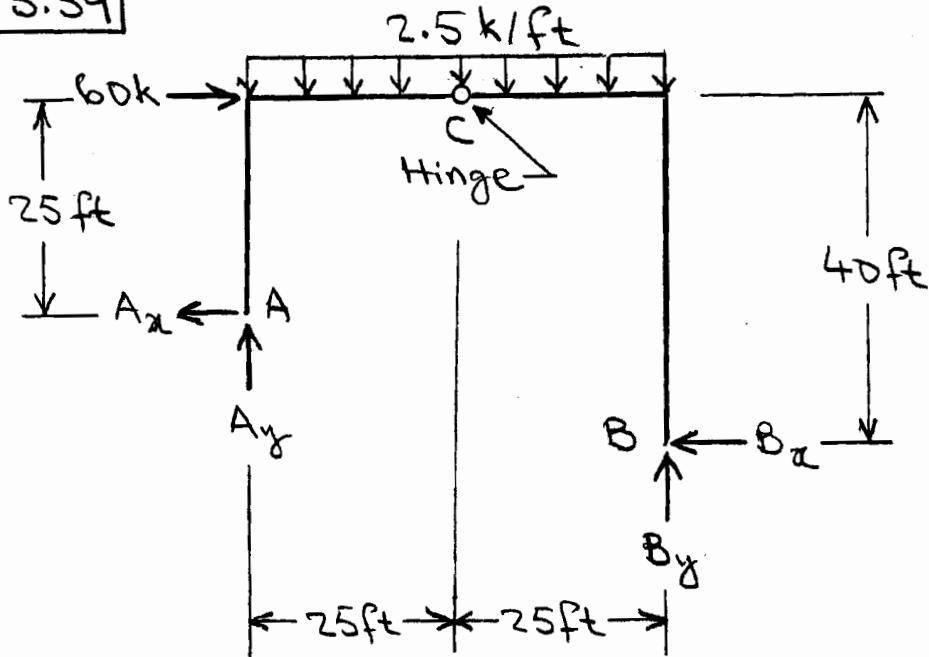
$$Cy = \underline{\underline{225 \text{ kN} \uparrow}}$$

$$+\sum M_C = 0$$

$$50(24) + 200(21) - 475(18) + 25(18)(9) - M_C = 0$$

$$M_C = \underline{\underline{900 \text{ kN.m} \leftarrow}}$$

3.39



$$+G \sum M_A = 0$$

$$-60(25) - 2.5(50)25 - B_x(15) + B_y(50) = 0$$

$$3B_x - 10B_y = -925 \quad (1)$$

$$+G \sum M_C = 0$$

$$-2.5(25)(12.5) - B_x(40) + B_y(25) = 0$$

$$8B_x - 5B_y = -156.25 \quad (2)$$

Solving Eqs. (1) and (2) simultaneously, we obtain

$$\underline{B_x = 47.12 \text{ k} \leftarrow}$$

$$\underline{B_y = 106.63 \text{ k} \uparrow}$$

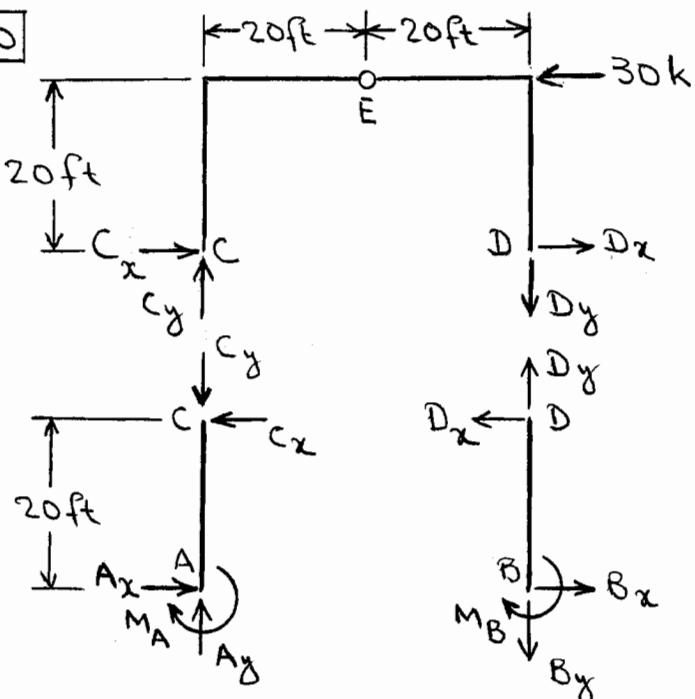
$$\stackrel{\rightarrow}{+} \sum F_x = 0 \quad -A_x + 60 - 47.12 = 0$$

$$\underline{A_x = 12.88 \text{ k} \leftarrow}$$

$$+\uparrow \sum F_y = 0 \quad A_y - 2.5(50) + 106.63 = 0$$

$$\underline{A_y = 18.37 \text{ k} \uparrow}$$

3.40



Considering the free body of portion CED, we determine the forces at the internal hinges C and D to be

$$+\zeta \sum M_D = 0 \quad -C_y(40) + 30(20) = 0 \quad C_y = 15 \text{ k}$$

$$+\zeta \sum M_E = 0 \quad C_x(20) - 15(20) = 0 \quad C_x = 15 \text{ k}$$

$$\rightarrow \sum F_x = 0 \quad 15 - 30 + D_x = 0 \quad D_x = 15 \text{ k}$$

$$+\uparrow \sum F_y = 0 \quad 15 - D_y = 0 \quad D_y = 15 \text{ k}$$

Considering the equilibrium of portion AC, we compute the reactions at support A to be

$$\rightarrow \sum F_x = 0 \quad A_x - 15 = 0 \quad \underline{A_x = 15 \text{ k} \rightarrow}$$

$$+\uparrow \sum F_y = 0 \quad A_y - 15 = 0 \quad \underline{A_y = 15 \text{ k} \uparrow}$$

$$+\zeta \sum M_A = 0 \quad 15(20) - M_A = 0 \quad \underline{M_A = 300 \text{ k-ft} \swarrow}$$

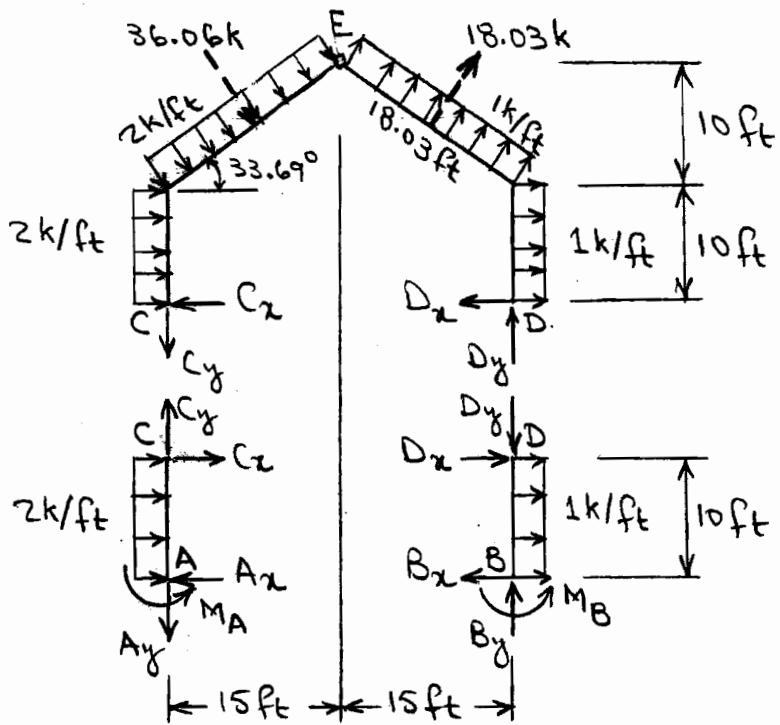
Similarly, considering the equilibrium of portion BD:

$$\rightarrow \sum F_x = 0 \quad B_x - 15 = 0 \quad \underline{B_x = 15 \text{ k} \rightarrow}$$

$$+\uparrow \sum F_y = 0 \quad -B_y + 15 = 0 \quad \underline{B_y = 15 \text{ k} \downarrow}$$

$$+\zeta \sum M_B = 0 \quad 15(20) - M_B = 0 \quad \underline{M_B = 300 \text{ k-ft} \swarrow}$$

3.41



Considering the free body of portion CED:

$$+G \sum M_D = 0 \\ Cy(30) - (2+1)(10)5 - (36.06 + 18.03) \sin 33.69^\circ (15) \\ + 36.06 \cos 33.69^\circ (22.5) - 18.03 \cos 33.69^\circ (7.5) = 0$$

$$Cy = 1.25 k$$

$$+G \sum M_E^{CE} = 0 \\ -Cx(20) + 1.25(15) + 2(10)(15) + 36.06 \left(\frac{18.03}{2}\right) = 0$$

$$Cx = 32.19 k$$

$$\therefore \sum F_x = 0 \\ -32.19 + (2+1)10 + (36.06 + 18.03) \sin 33.69^\circ \\ - Dx = 0 \\ Dx = 27.81 k$$

$$+\uparrow \sum F_y = 0 \\ -1.25 - 36.06 \cos 33.69^\circ + 18.03 \cos 33.69^\circ + Dy = 0 \\ Dy = 16.25 k$$

Free body of portion AC:

$$\therefore \sum F_x = 0 \\ 32.19 + 2(10) - Ax = 0 \\ \underline{\underline{Ax = 52.19 k \leftarrow}}$$

$$+\uparrow \sum F_y = 0 \\ -Ay + 1.25 = 0 \\ \underline{\underline{Ay = 1.25 k \downarrow}} \\ +G \sum M_A = 0 \\ Ma - 32.19(10) - 2(10)5 = 0 \\ \underline{\underline{Ma = 421.9 k-ft \uparrow}}$$

Free body of portion BD:

$$\therefore \sum F_x = 0 \\ 27.81 + 1(10) - Bx = 0 \\ \underline{\underline{Bx = 37.81 k \leftarrow}}$$

$$+\uparrow \sum F_y = 0 \\ By - 16.25 = 0 \\ \underline{\underline{By = 16.25 k \uparrow}}$$

$$+G \sum M_B = 0 \\ Mb - 27.81(10) - 1(10)5 = 0 \\ \underline{\underline{Mb = 328.1 k-ft \uparrow}}$$

Chapter Four

Plane and Space Trusses

CHAPTER 4

4.1 (a) $m=2, r=3, j=3; m+r < 2j$; Unstable

(b) $m=3, r=3, j=3; m+r = 2j$

Statically determinate

(c) $m=2, r=4, j=3; m+r = 2j$

Statically determinate

(d) $m=3, r=4, j=4; m+r < 2j$; Unstable

4.2 (a) $m=6, r=3, j=4; m+r > 2j$

Statically indeterminate; $i = (6+3)-2(4) = \underline{1}$

(b) $m=9, r=3, j=6; m+r = 2j$

Statically determinate

(c) $m=9, r=3, j=6; m+r = 2j$

Unstable, because two rigid portions are connected by an internal hinge and three reactions are not sufficient to prevent relative rotation of one rigid part with respect to the other.

(d) $m=7, r=4, j=5; m+r > 2j$

Statically indeterminate; $i = (7+4)-2(5) = \underline{1}$

4.3 (a) $m=17, r=3, j=9; m+r > 2j$

Statically indeterminate; $i = (17+3)-2(9) = \underline{2}$

(b) $m=12, r=3, j=7; m+r > 2j$

Statically indeterminate; $i = (12+3)-2(7) = \underline{1}$

(c) $m=25, r=4, j=14; m+r > 2j$

Statically indeterminate; $i = (25+4)-2(14) = \underline{1}$

(d) $m=24, r=6, j=15; m+r = 2j$

Statically determinate

4.4 (a) $m=22, r=4, j=13; m+r = 2j$

Statically determinate

(b) $m=14, r=3, j=9; m+r < 2j$; Unstable

(c) $m=11, r=3, j=7; m+r = 2j$

Statically determinate

(d) $m=23, r=4, j=13; m+r > 2j$

Statically indeterminate; $i = (23+4)-2(13) = \underline{1}$

4.5 (a) $m=14, r=4, j=9; m+r = 2j$

Unstable, because two rigid portions are connected by a hinge.

(b) $m=24, r=3, j=14; m+r < 2j$; Unstable

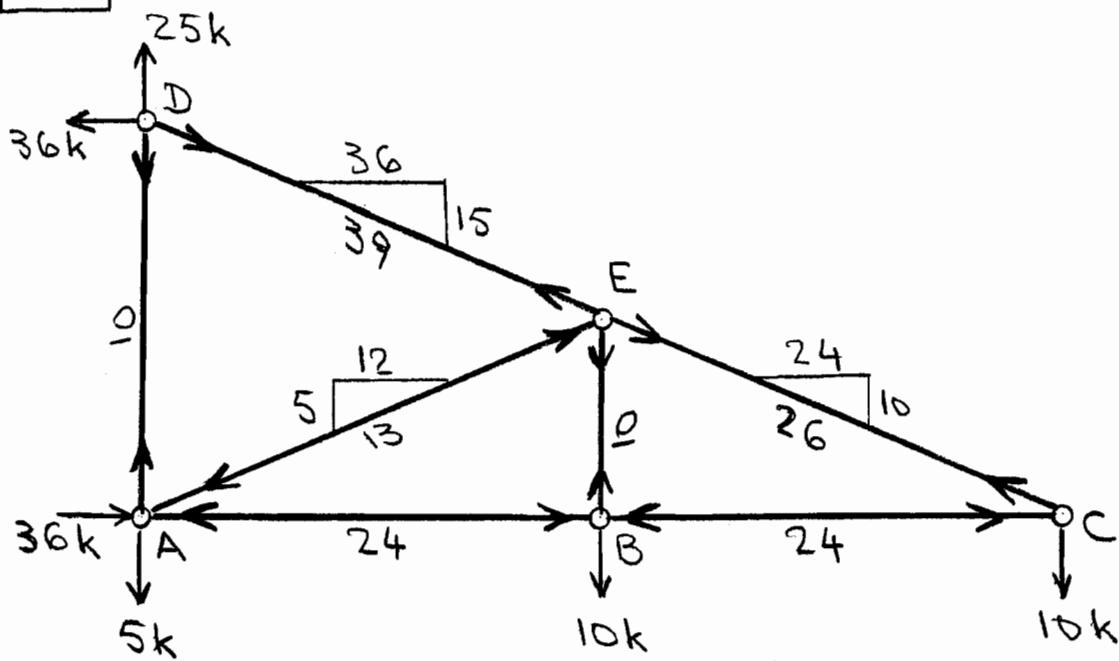
(c) $m=9, r=3, j=6; m+r = 2j$

Statically determinate

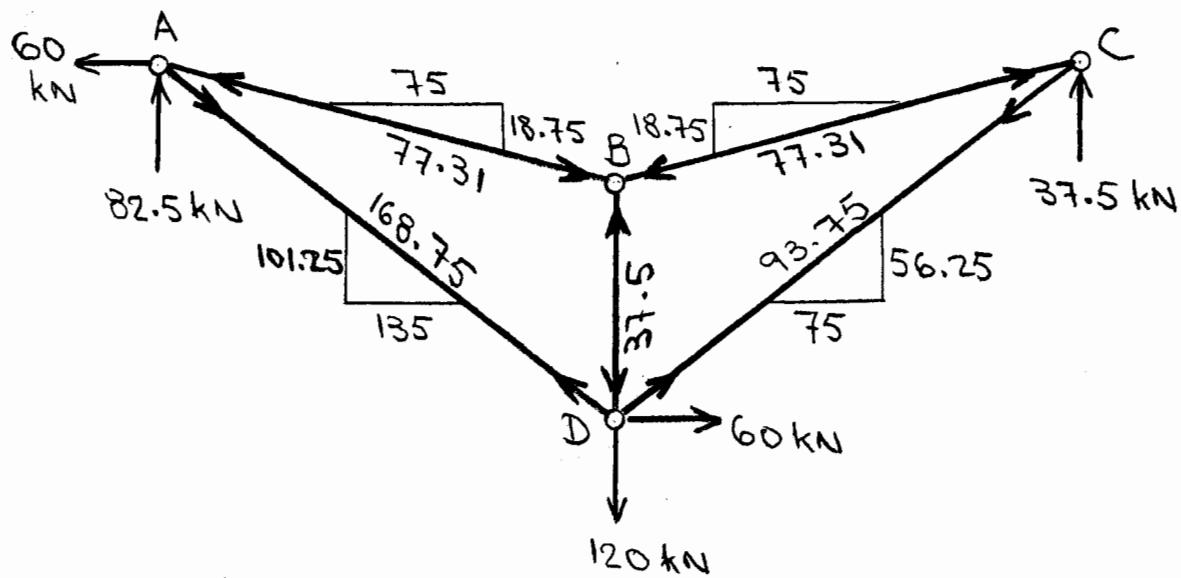
(d) $m=25, r=3, j=14; m+r = 2j$

Unstable, because two rigid portions are connected by three parallel members

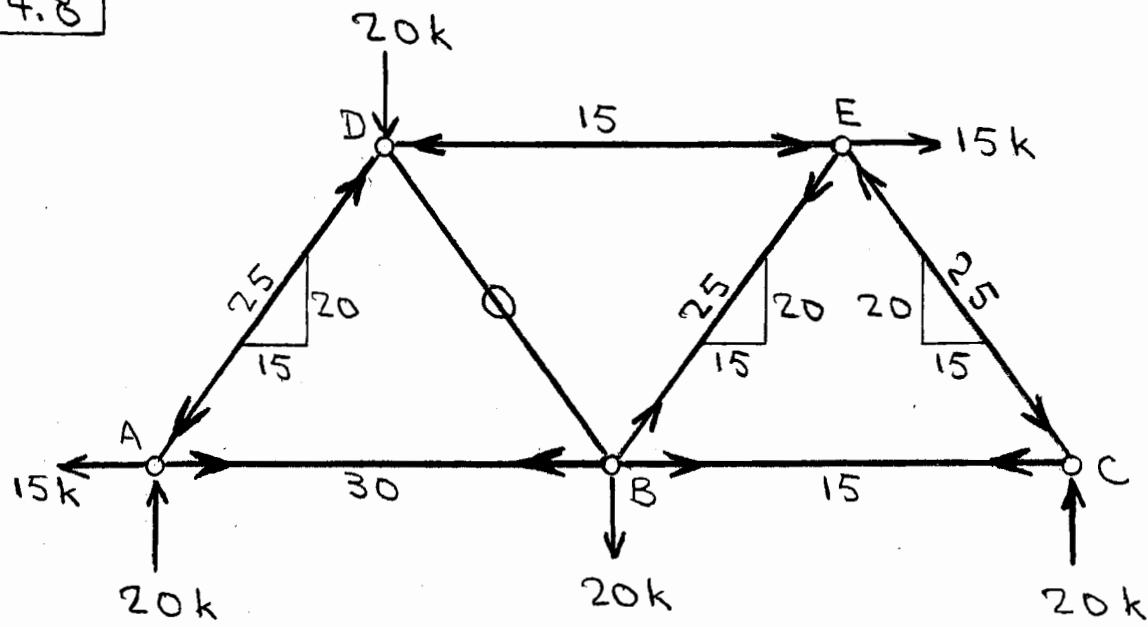
4.6



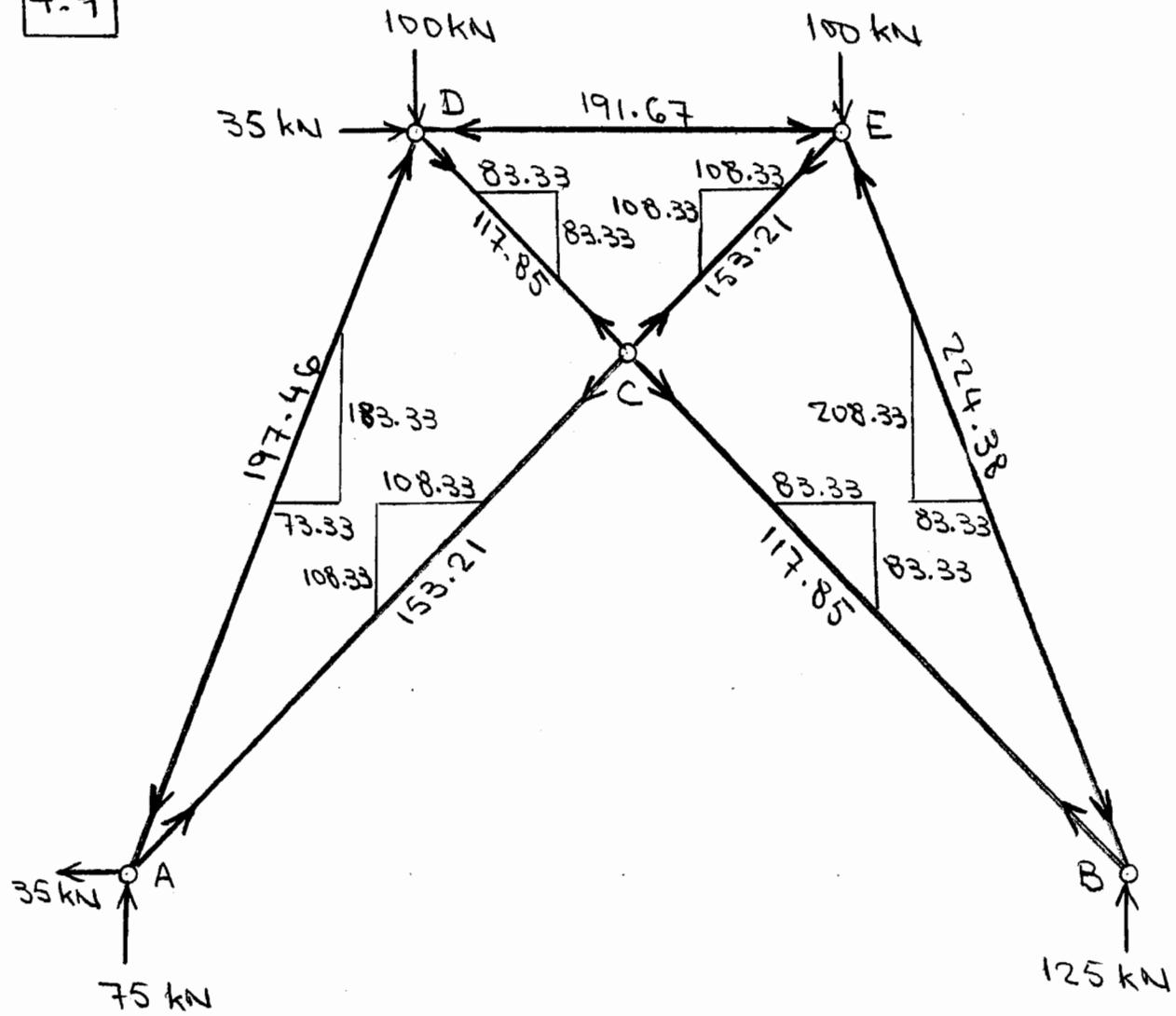
4.7

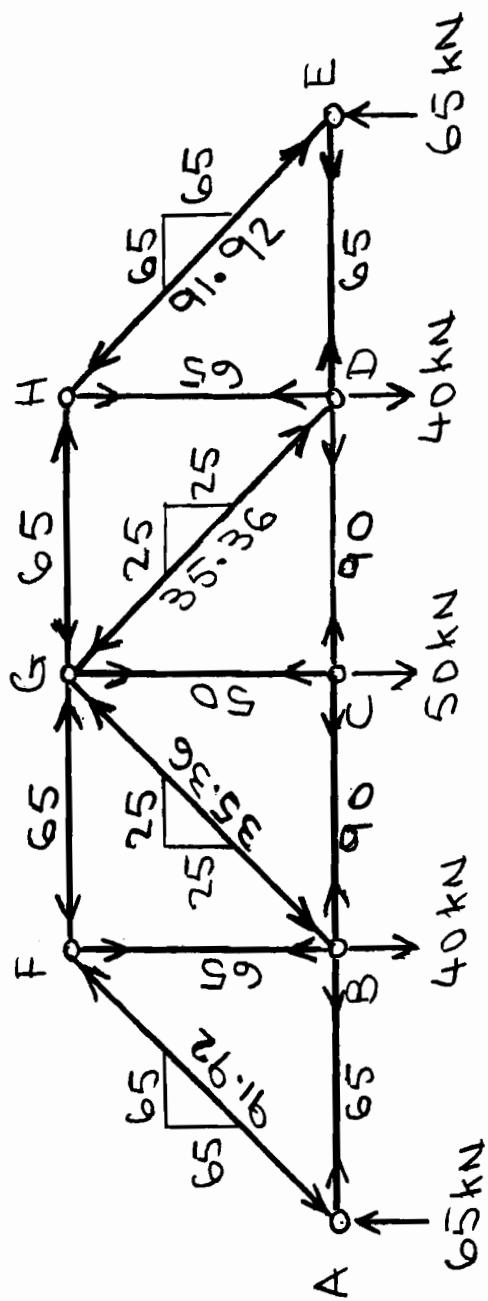


4.8

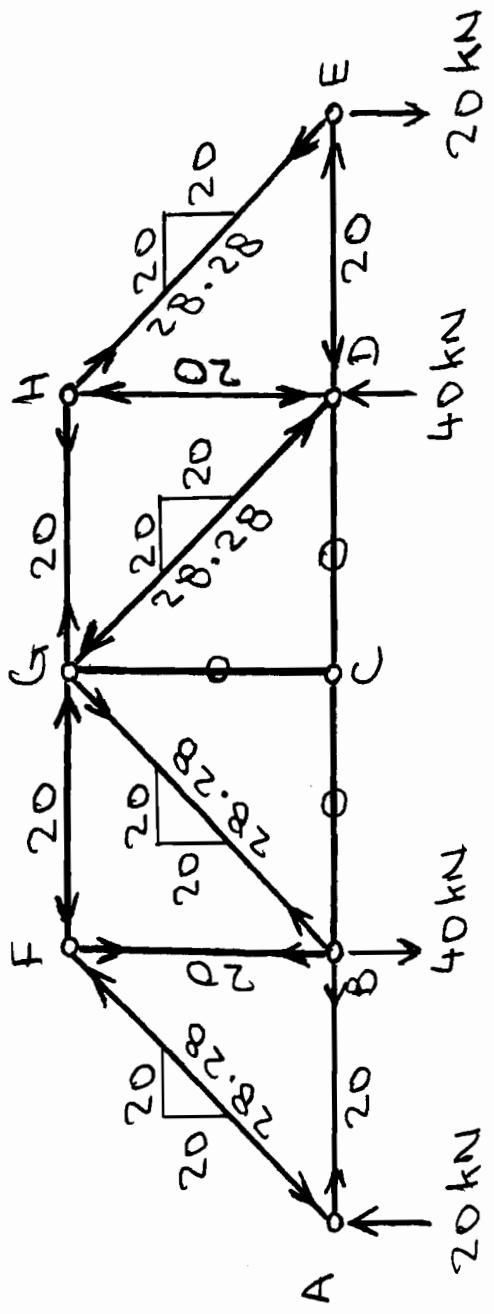


4.9



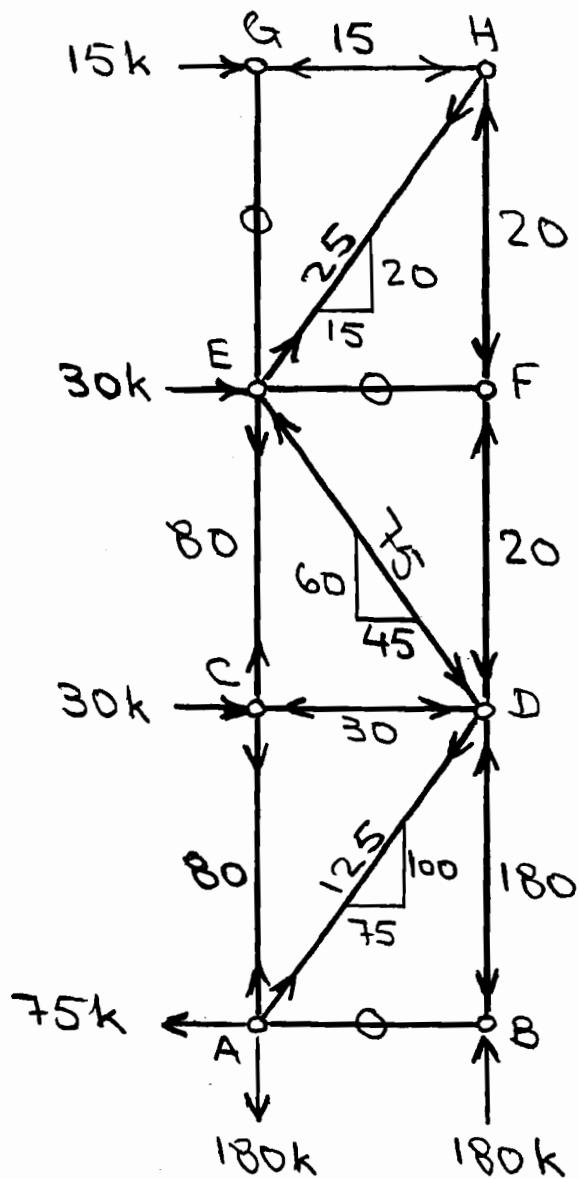


4.10

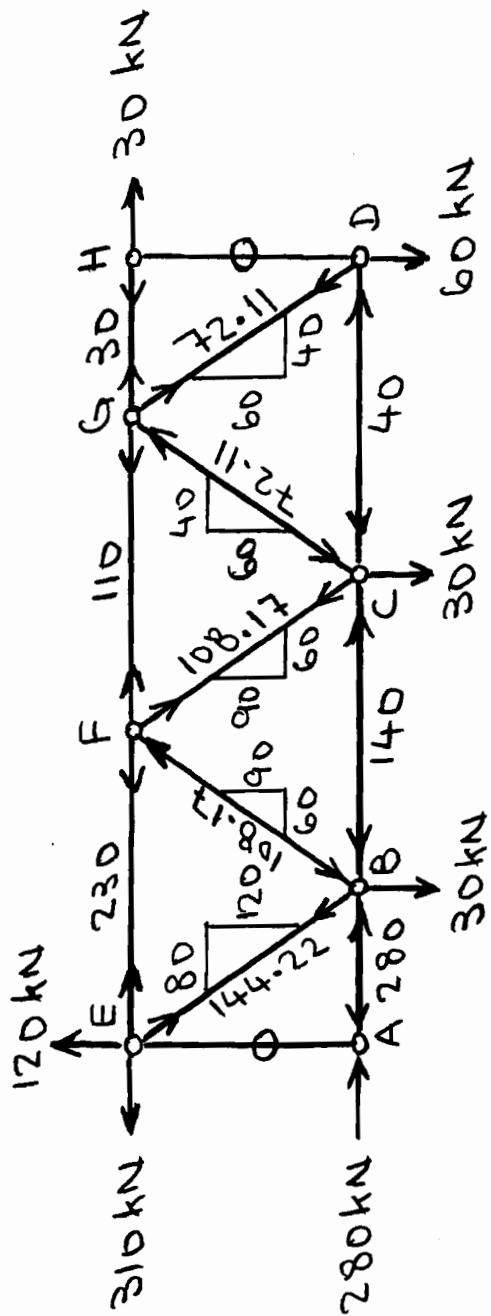


4.11

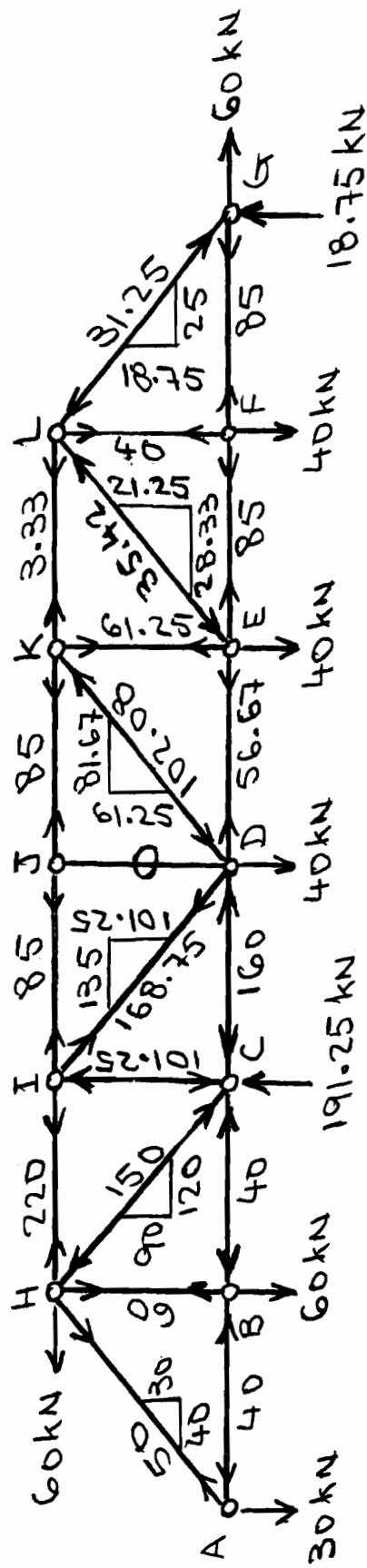
4.12



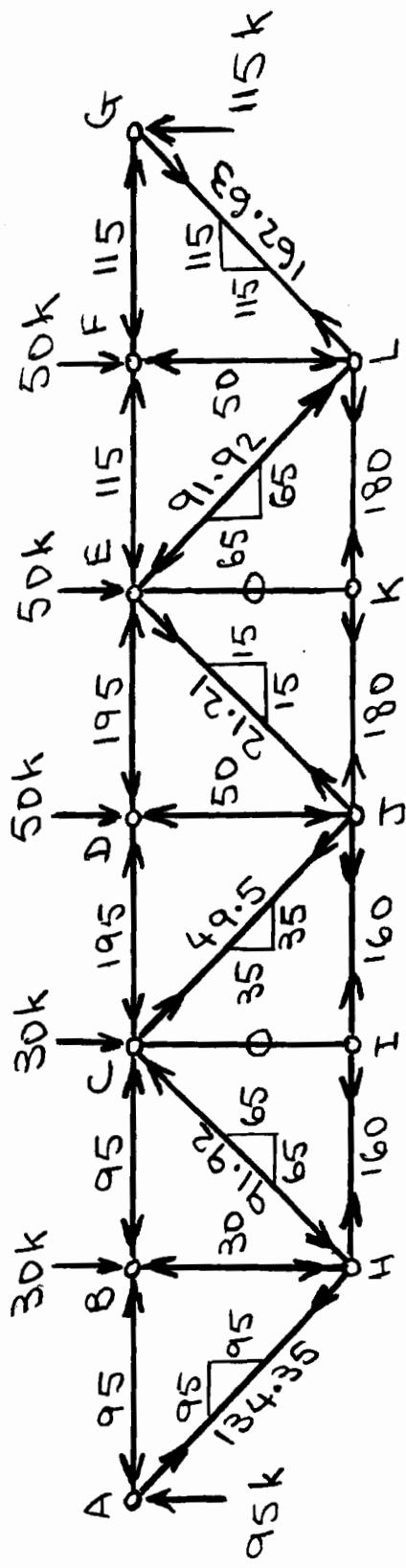
4.13



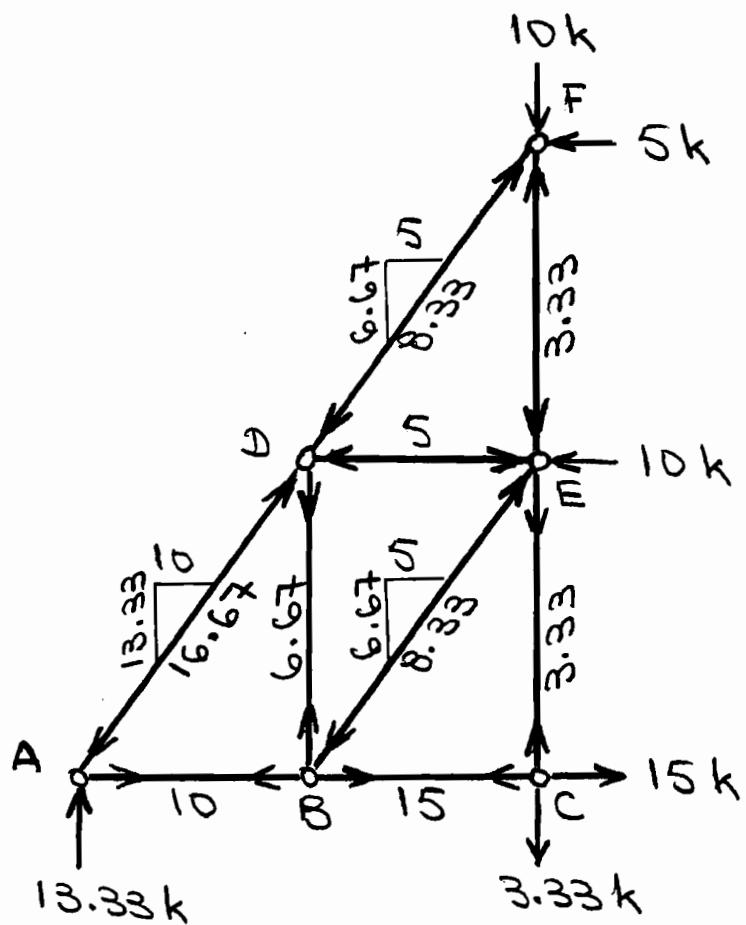
4.14

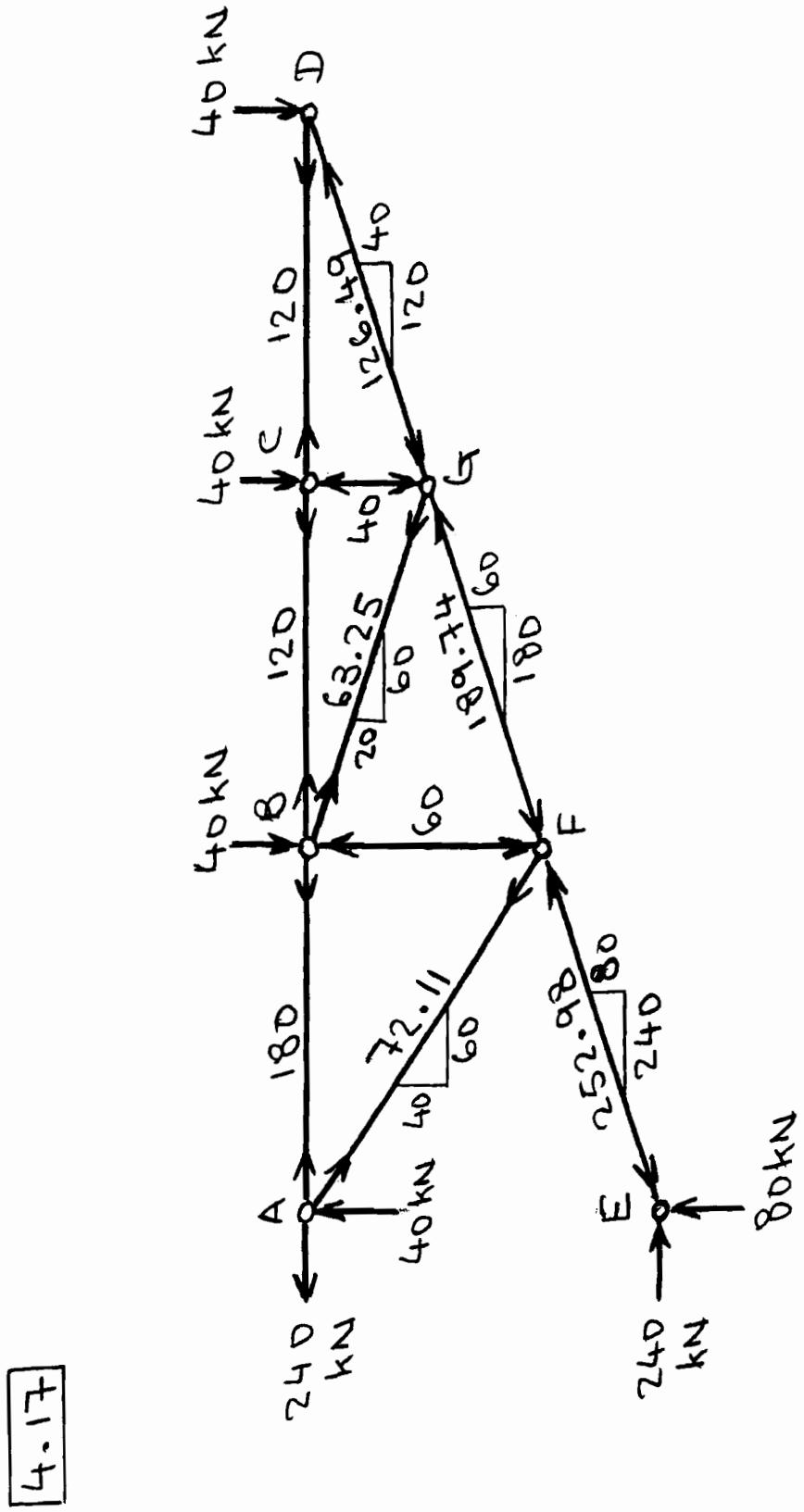


4.15

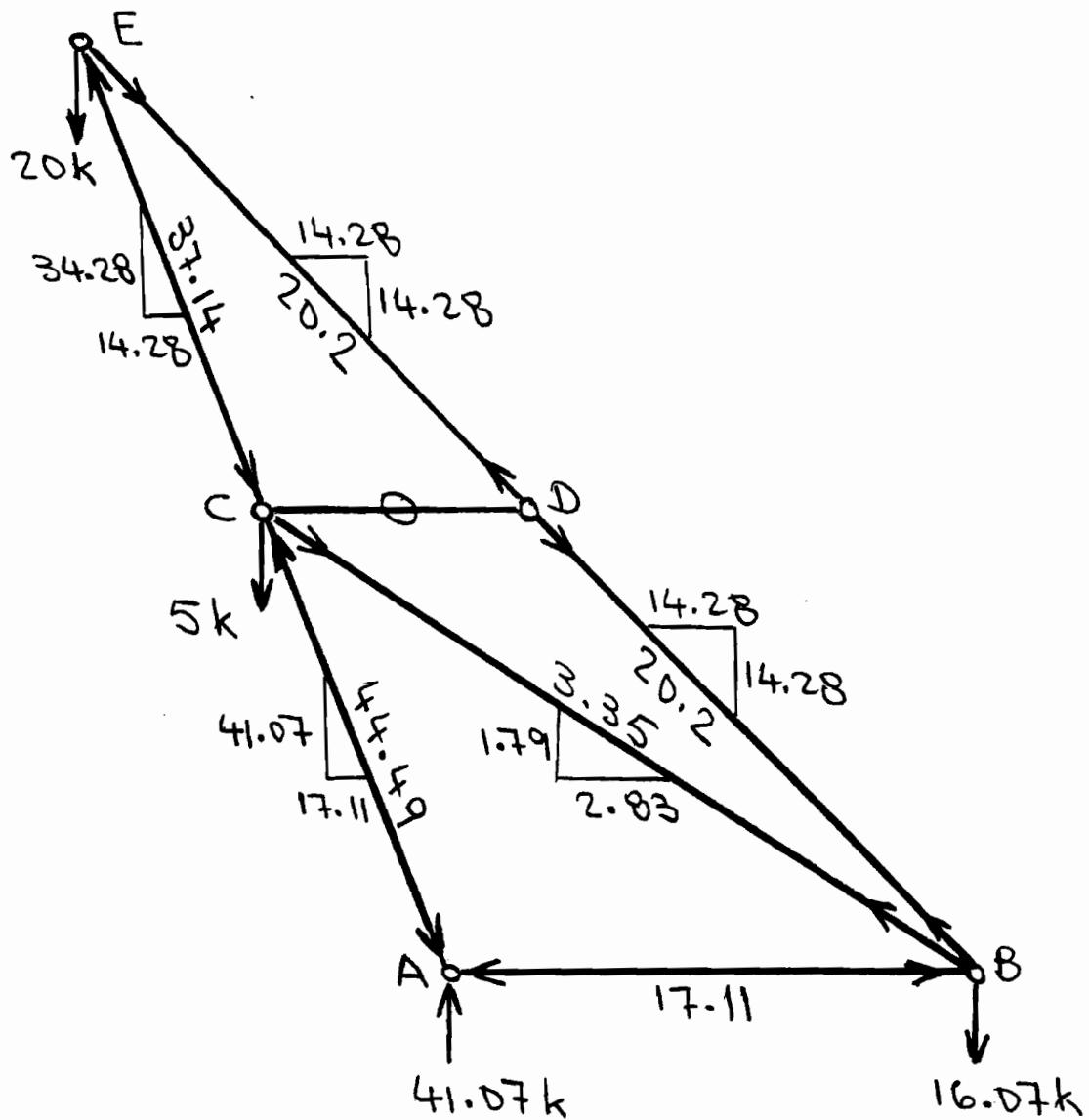


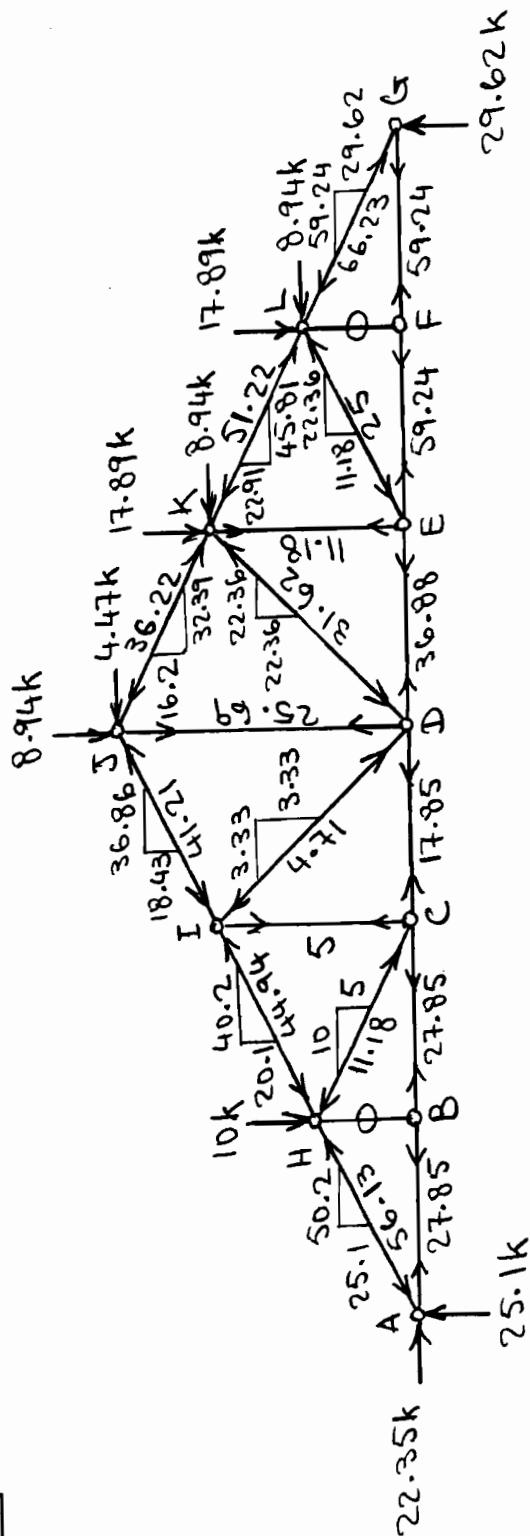
4-16



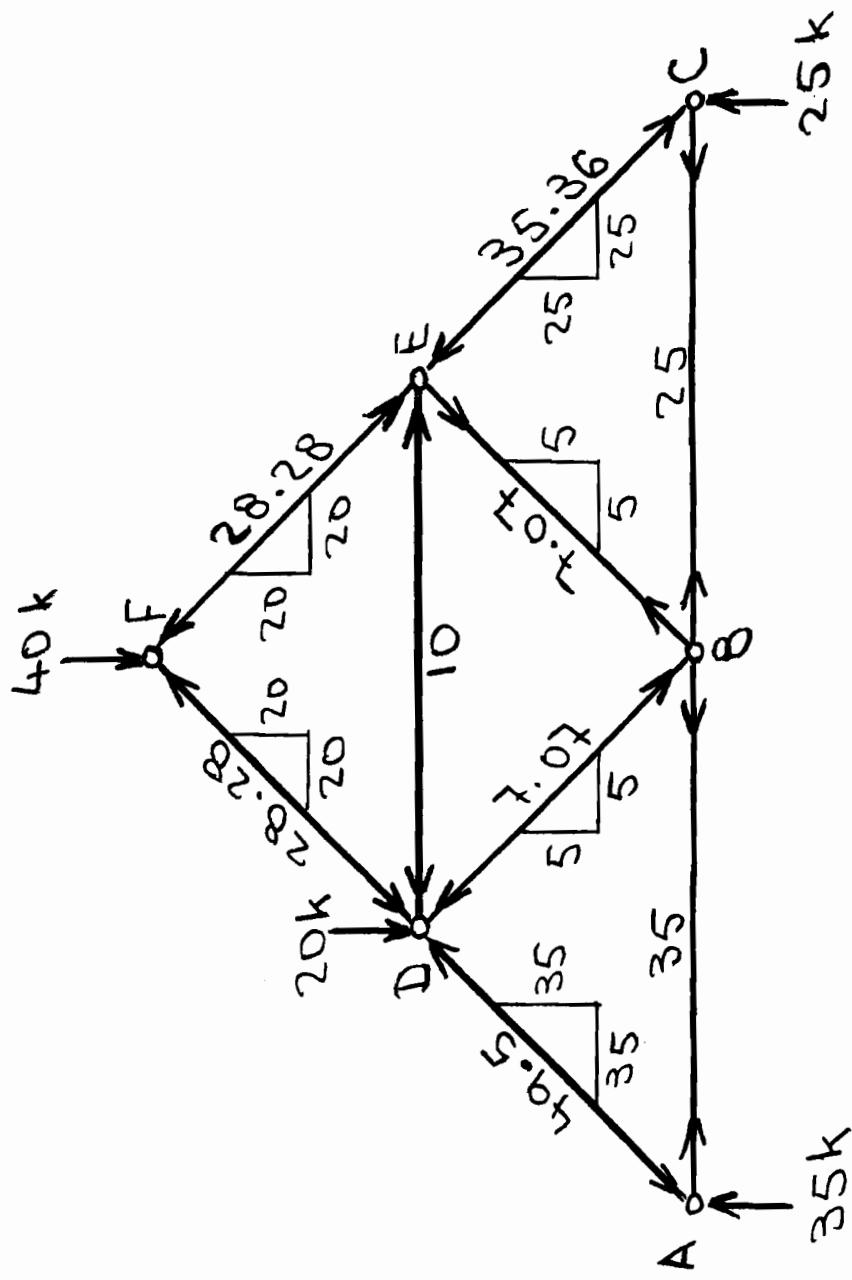


4.18



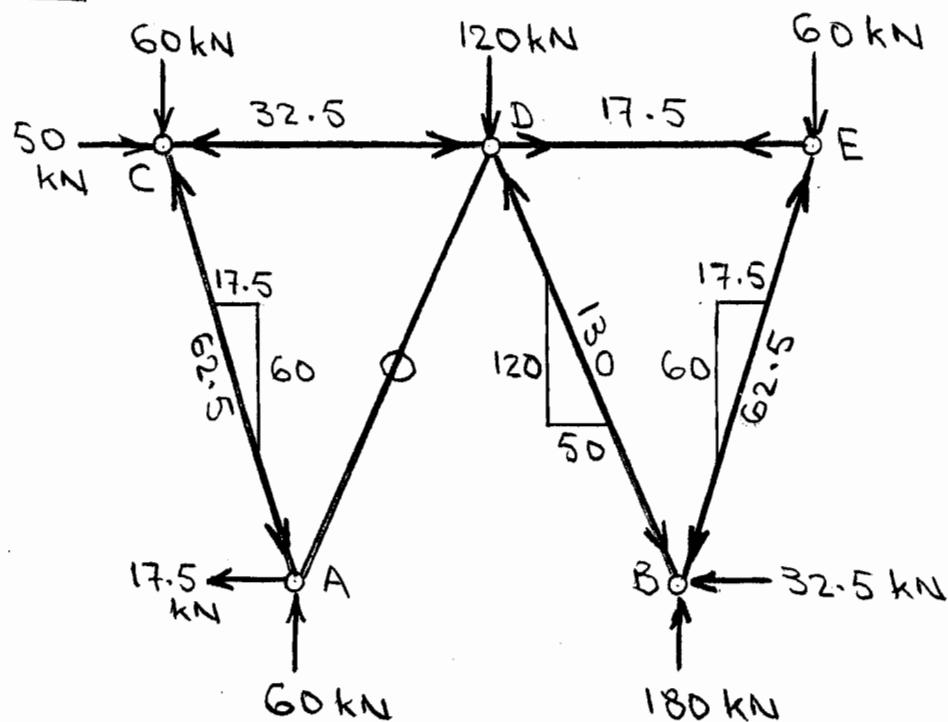


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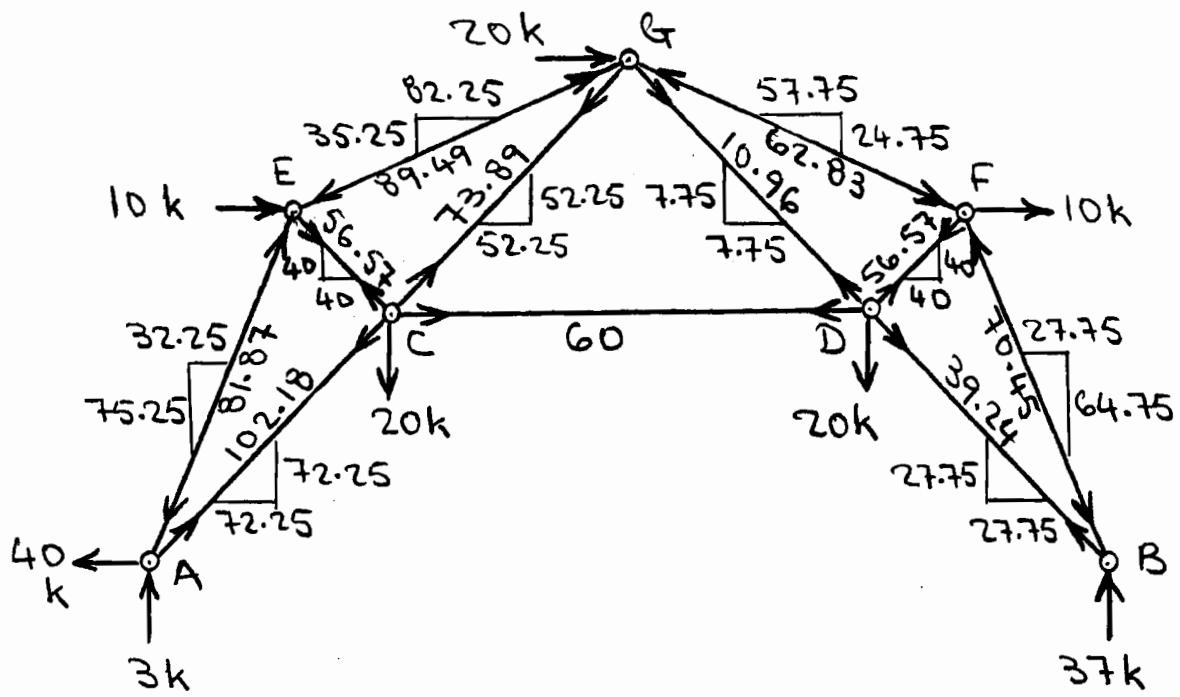


4.20

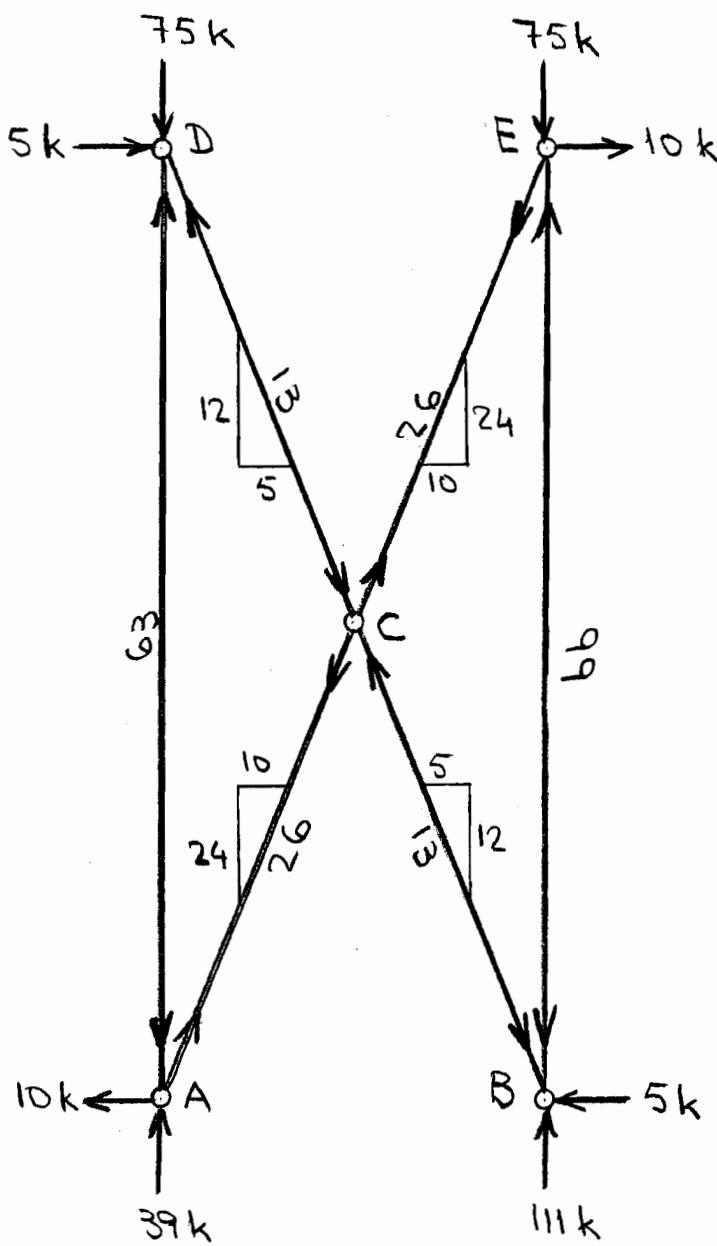
4.21



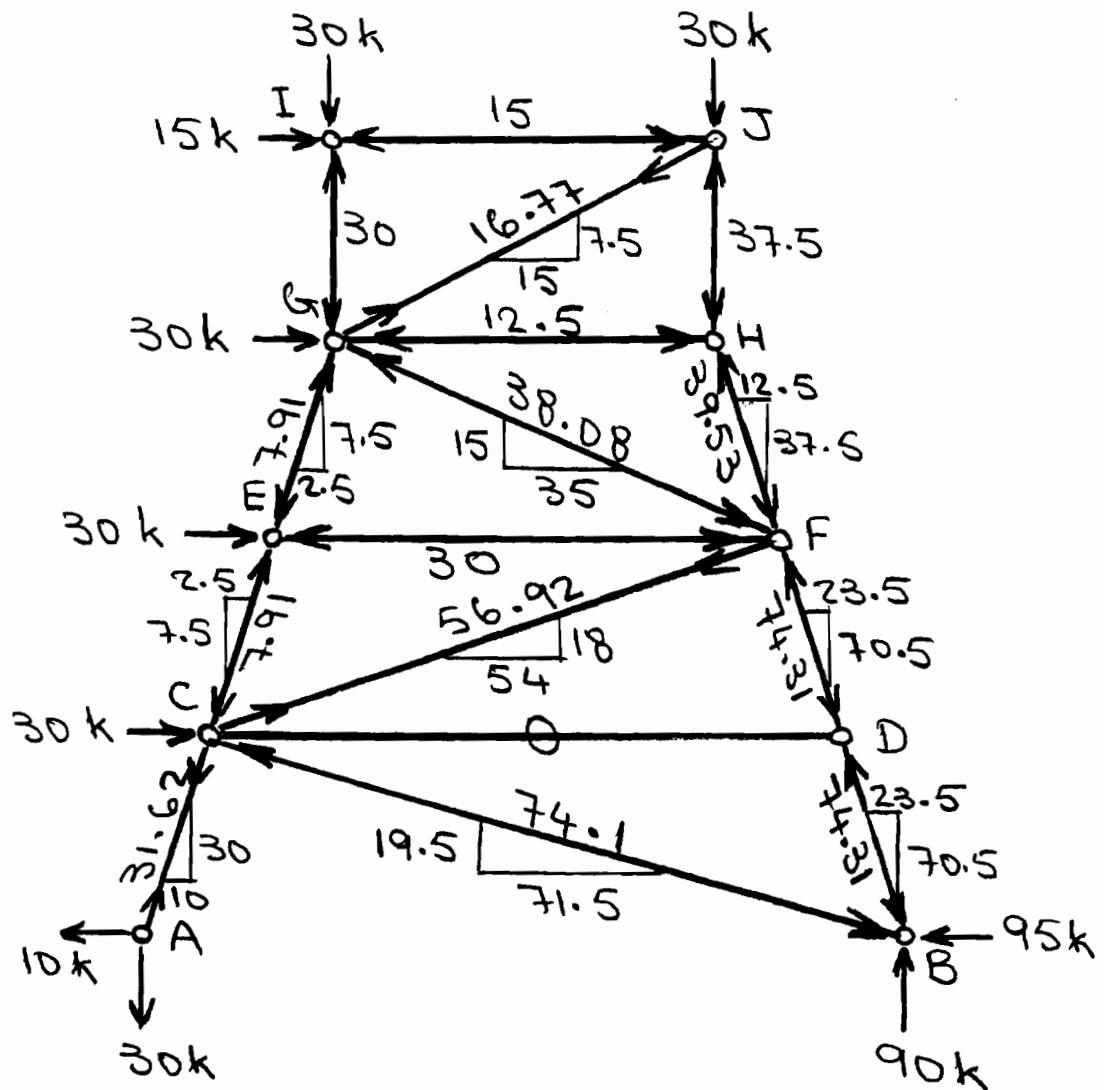
4.22



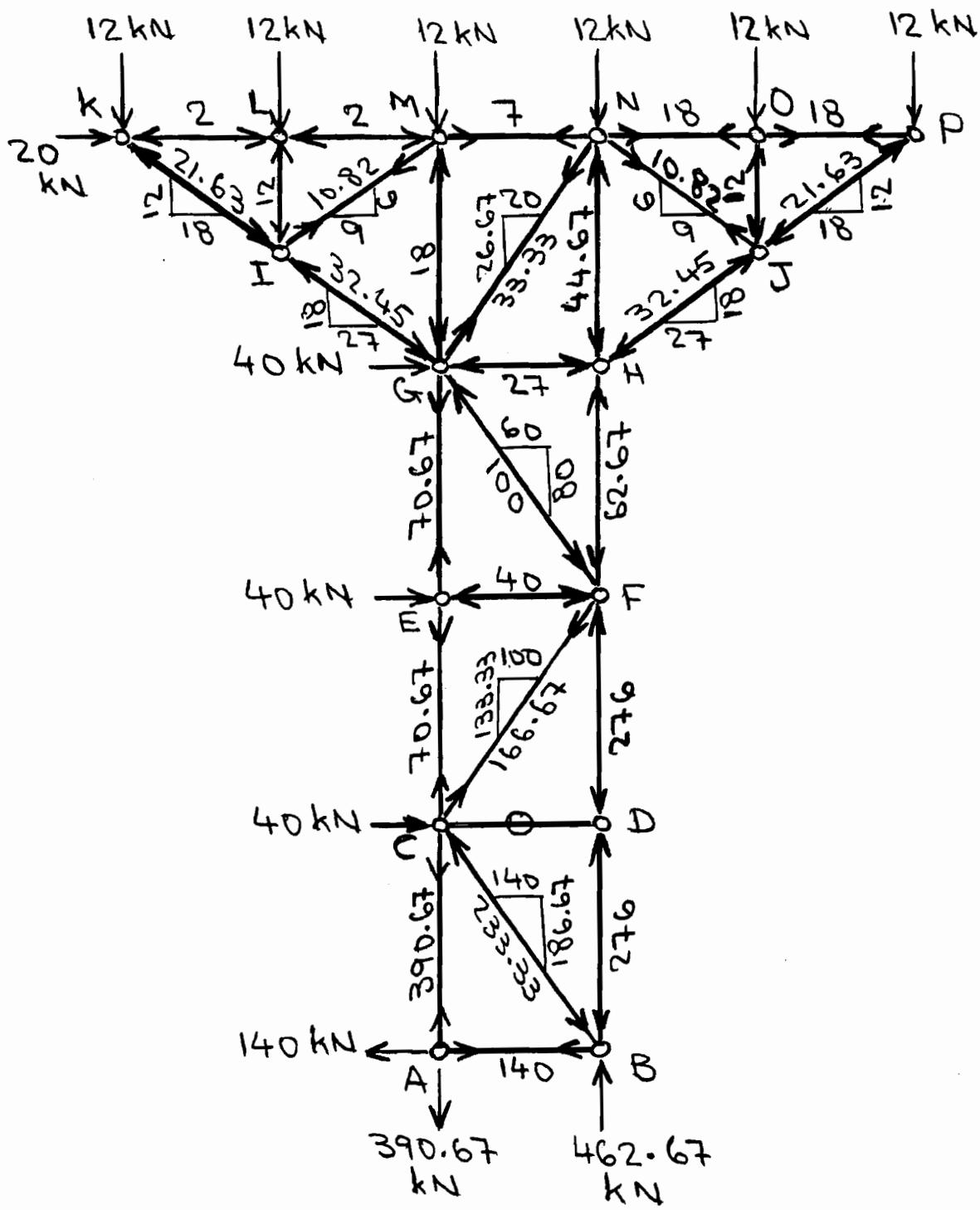
4.23



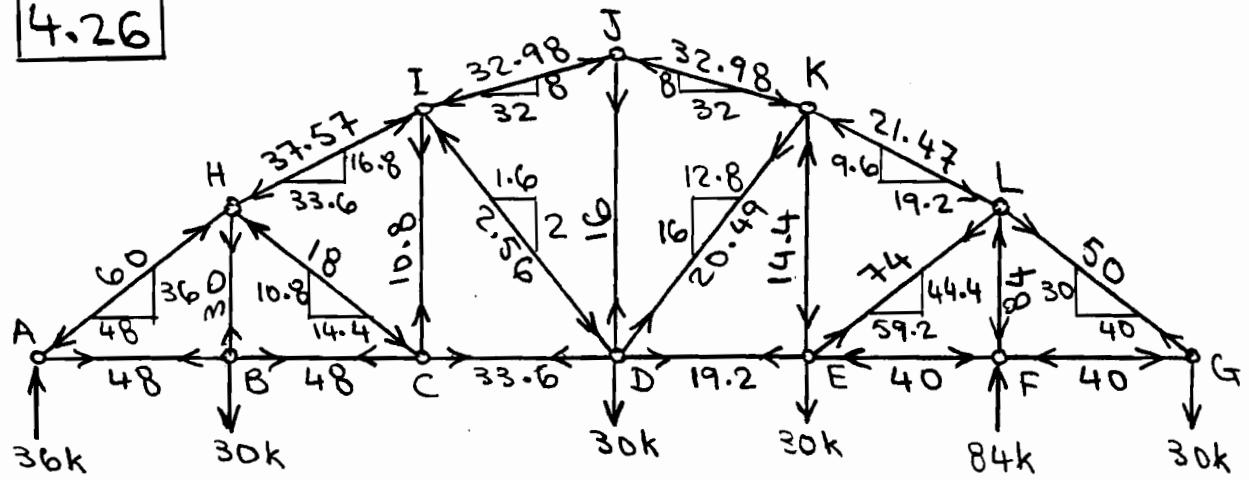
4.24



4.25



4.26



4.27 Reactions:

$$\rightarrow \sum F_x = 0 \quad A_x + 50 = 0 \quad A_x = -50 \quad \underline{A_x = 50 \text{ kN} \leftarrow}$$

$$+\zeta \sum M_C^{AC} = 0 \quad -A_y(8) + 120(4) = 0 \quad \underline{A_y = 60 \text{ kN} \uparrow}$$

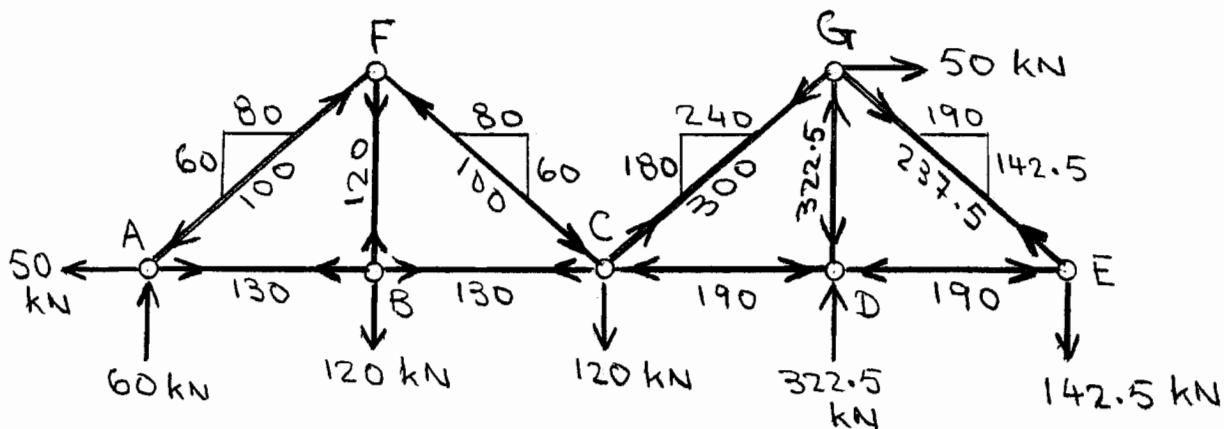
$$+\zeta \sum M_E = 0 \quad -60(16) + 120(12) + 120(8) - 50(3) - D_y(4) = 0$$

$$\underline{D_y = 322.5 \text{ kN} \uparrow}$$

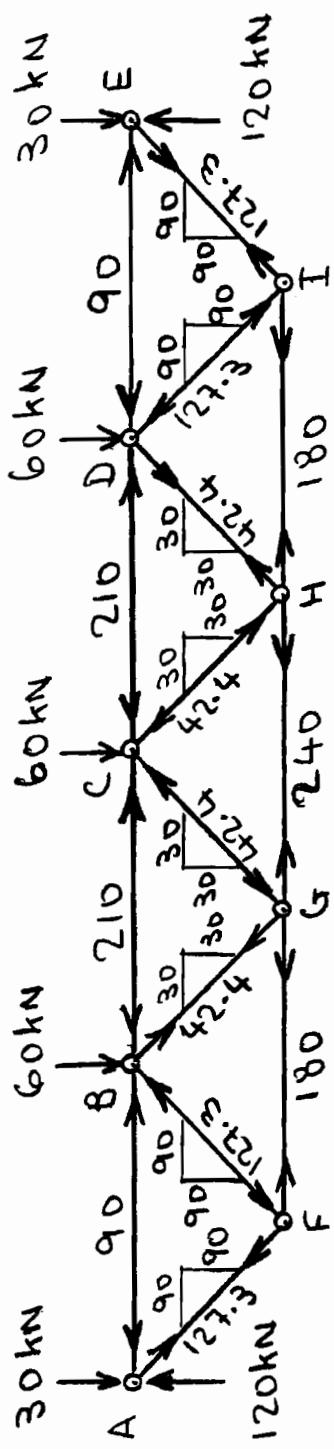
$$+\uparrow \sum F_y = 0 \quad 60 - 120 - 120 + 322.5 + E_y = 0$$

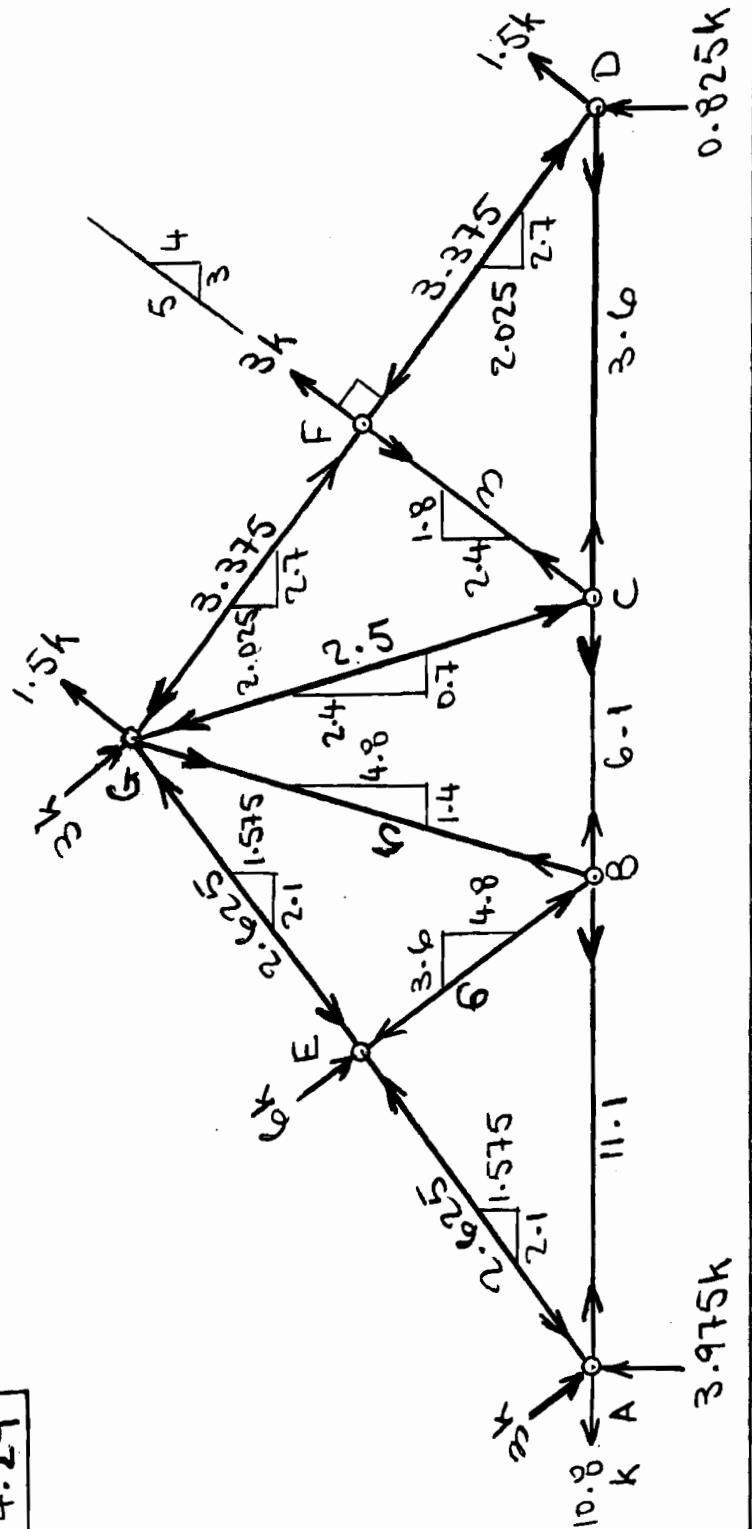
$$\underline{E_y = -142.5 \text{ kN}} \quad \underline{E_y = 142.5 \text{ kN} \downarrow}$$

Member Forces:



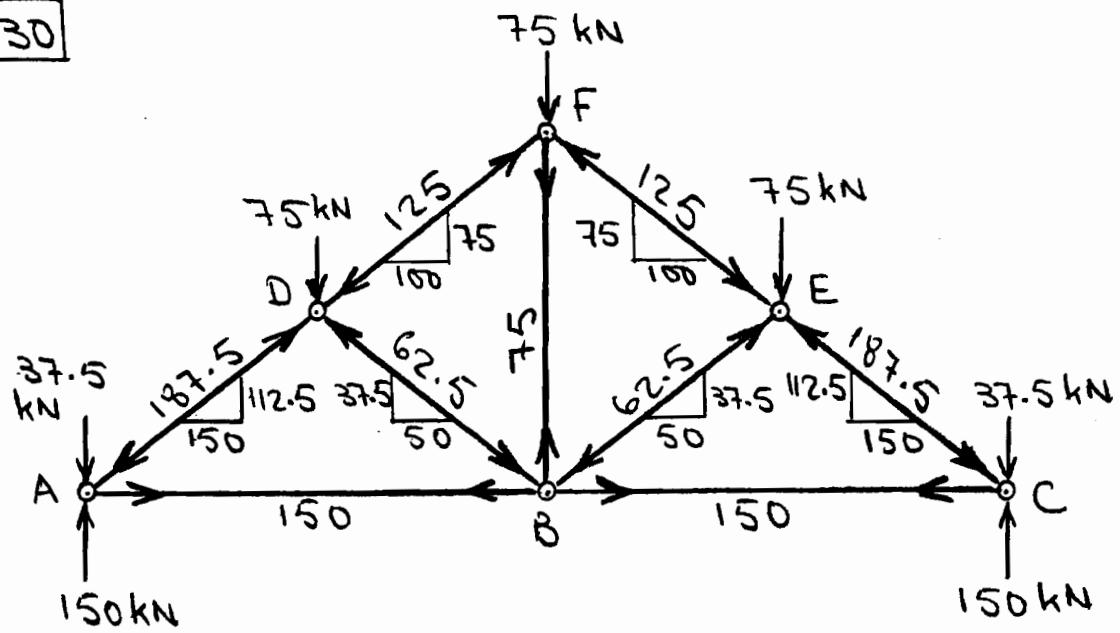
4.28



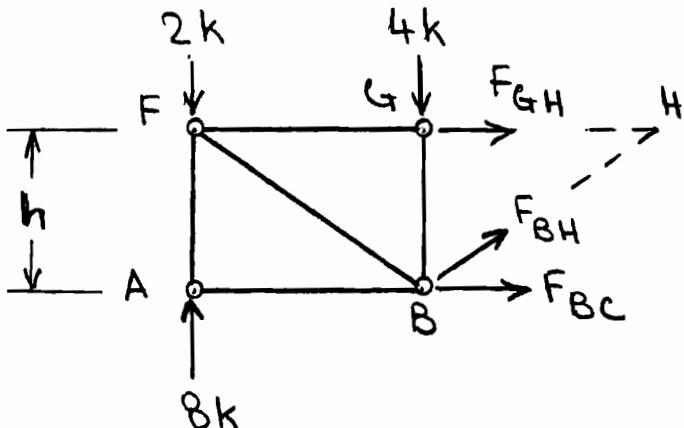


4.29

4.30



4.31 Section through members BC, BH and GH.



$$+ G \sum M_B = 0 \quad -8(b) + 2(b) - F_{GH}(h) = 0$$

$$F_{GH} = -\frac{3b}{h} \quad (1)$$

$$+\zeta \sum M_H = 0 \quad -8(12) + 2(12) + 4(6) + F_{BC}(h) = 0$$

$$F_{BC} = \frac{48}{h} \quad (2)$$

Equations (1) and (2) indicate that the magnitudes of F_{GH} and F_{BC} are inversely proportional to the truss height h .

$$\underline{\text{For } h = 3 \text{ ft:}} \quad F_{GH} = -\frac{3b}{2} = -12k = \underline{12k \text{ (c)}}$$

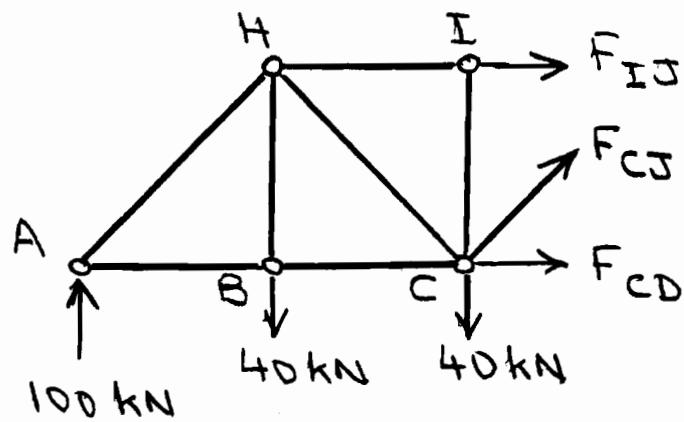
$$F_{BC} = \frac{40}{w} = \underline{16k(T)}$$

$$\text{For } h=6 \text{ ft: } F_{GH} = -\frac{3b}{r} = -6k = 6k(c)$$

$$F_{BC} = \frac{48}{6} = \underline{8 \text{ k(T)}}$$

4.32

Section through members CD, CJ and IJ:



$$+\uparrow \sum F_y = 0$$

$$100 - 2(40) + \left(\frac{1}{\sqrt{2}}\right) F_{CJ} = 0$$

$$F_{CJ} = -28.28 \text{ kN}$$

$$\underline{F_{CJ} = 28.28 \text{ kN (C)}}$$

$$+\rightarrow \sum M_C = 0$$

$$-100(10) + 40(5) - F_{IJ}(5) = 0$$

$$F_{IJ} = -160 \text{ kN}$$

$$\underline{F_{IJ} = 160 \text{ kN (C)}}$$

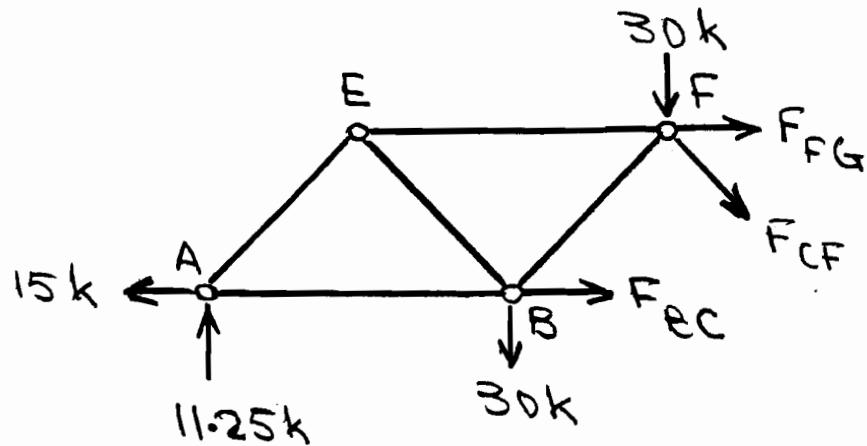
$$\rightarrow \sum F_x = 0$$

$$F_{CD} - \left(\frac{1}{\sqrt{2}}\right) 28.28 - 160 = 0$$

$$\underline{F_{CD} = 180 \text{ kN (T)}}$$

4.33

Section through members BC, CF and FG:



$$+\uparrow \sum F_y = 0$$

$$11.25 - 30 - 30 - (\frac{1}{\sqrt{2}}) F_{CF} = 0$$

$$F_{CF} = -68.94 \text{ k}$$

$$\underline{F_{CF} = 68.94 \text{ k (C)}}$$

$$+\leftarrow \sum M_F = 0$$

$$-15(10) - 11.25(30) + 30(10) + F_{BC}(10) = 0$$

$$\underline{F_{BC} = 18.75 \text{ k (T)}}$$

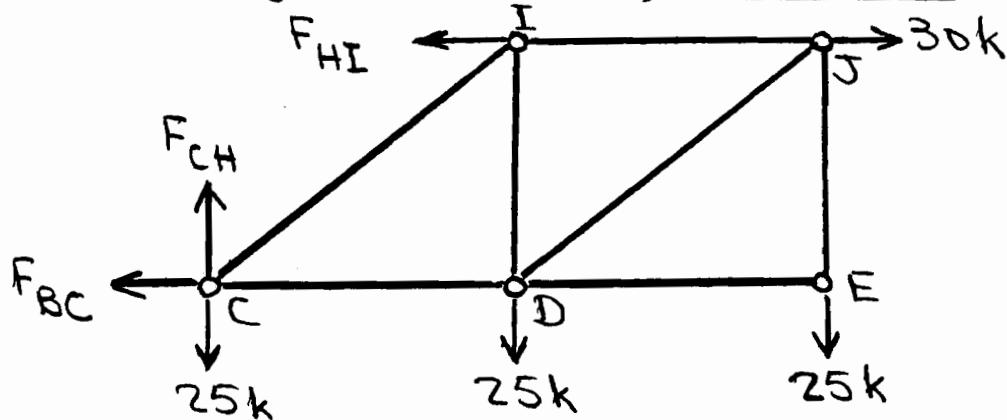
$$\rightarrow \sum F_x = 0$$

$$-15 + 18.75 - (\frac{1}{\sqrt{2}}) 68.94 + F_{FG} = 0$$

$$\underline{F_{FG} = 45 \text{ k (T)}}$$

4.34

Section through members BC, CH and HI:



$$+\uparrow \sum F_y = 0 \quad F_{CH} - 3(25) = 0$$

$$\underline{F_{CH} = 75 \text{ k (T)}}$$

$$+\zeta \sum M_C = 0$$

$$F_{HI}(15) - 30(15) - 25(20) - 25(40) = 0$$

$$\underline{F_{HI} = 130 \text{ k (T)}}$$

$$\dot{\rightarrow} \sum F_x = 0$$

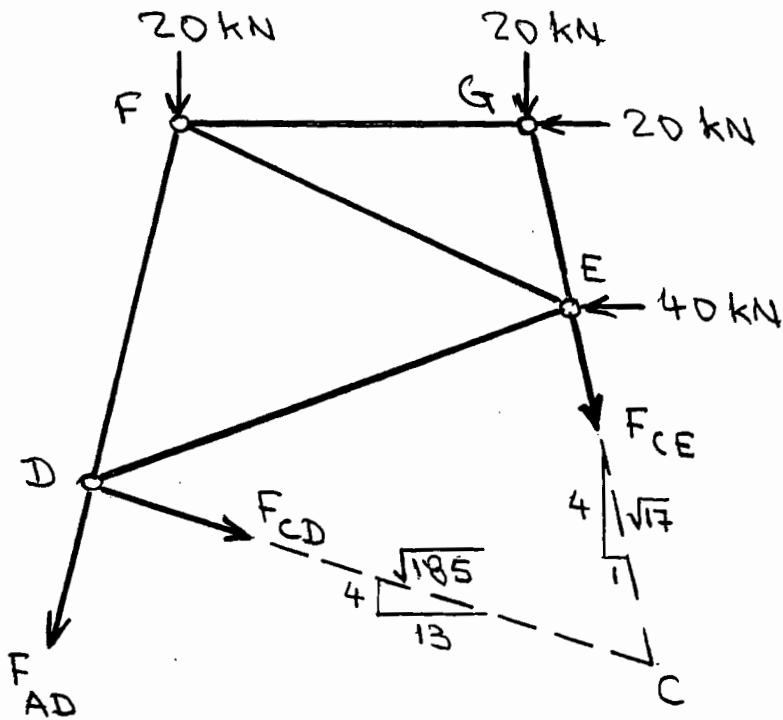
$$-F_{BC} - 130 + 30 = 0$$

$$F_{BC} = -100 \text{ k}$$

$$\underline{F_{BC} = 100 \text{ k (C)}}$$

4.35

Section through members AD, CD and CE:



$$+ G \sum M_D = 0$$

$$-20(1.5) - 20(7.5) + 20(6) + 40(3)$$

$$-\frac{1}{\sqrt{t}} F_{CE}(3) - \frac{4}{\sqrt{t}} F_{CE}(8.25) = 0$$

$$\underline{F_{CE} = 6.87 \text{ kN (T)}}$$

$$+ \sum C_i = 0$$

$$20(8.25) + 20(2.25) + 20(9) + 40(6)$$

$$+ \frac{1}{\sqrt{17}} F_{AD}(3) + \frac{4}{\sqrt{17}} F_{AD}(9.75) = 0$$

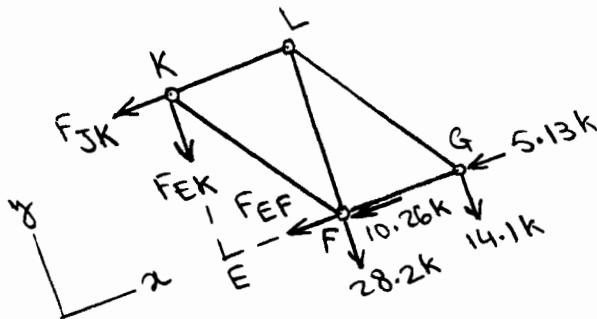
$$F_{AD} = -61.85 \text{ kN} \quad F_{AD} = 61.85 \text{ kN (C)}$$

$$+\uparrow \sum F_x = 0$$

$$\frac{4}{\sqrt{17}} (61.85) - 20 - 20 - \frac{4}{\sqrt{17}} (6.87) - \frac{4}{\sqrt{185}} F_{CD} = 0$$

$$F_{CD} = 45.34 \text{ kN (T)}$$

4.36 Section through members EF, EK and JK:



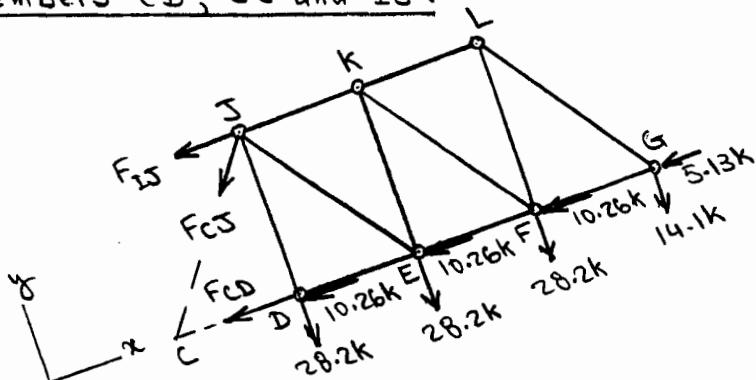
$$\sum F_y = 0 \quad -F_{EK} - 28.2 - 14.1 = 0 \quad F_{EK} = -42.3 \text{ k} = \underline{42.3 \text{ k (C)}}$$

$$+G \sum M_K = 0 \quad -F_{EF}(20) - 28.2(15) - 14.1(30) \\ - (5.13 + 10.26)(20) = 0$$

$$F_{EF} = -57.69 \text{ k} = \underline{57.69 \text{ k (C)}}$$

$$+G \sum M_E = 0 \quad F_{JK}(20) - 28.2(15) - 14.1(30) = 0 \quad F_{JK} = 42.3 \text{ k (T)}$$

Section through members CD, CJ and IJ:



$$\sum F_y = 0 \quad -\left(\frac{4}{5}\right)F_{CJ} - 3(28.2) - 14.1 = 0 \quad F_{CJ} = -123.38 \text{ k} = \underline{123.38 \text{ k (C)}}$$

$$+G \sum M_J = 0 \quad -F_{CD}(20) - [3(10.26) + 5.13](20) - 28.2(15 + 30) \\ - 14.1(45) = 0$$

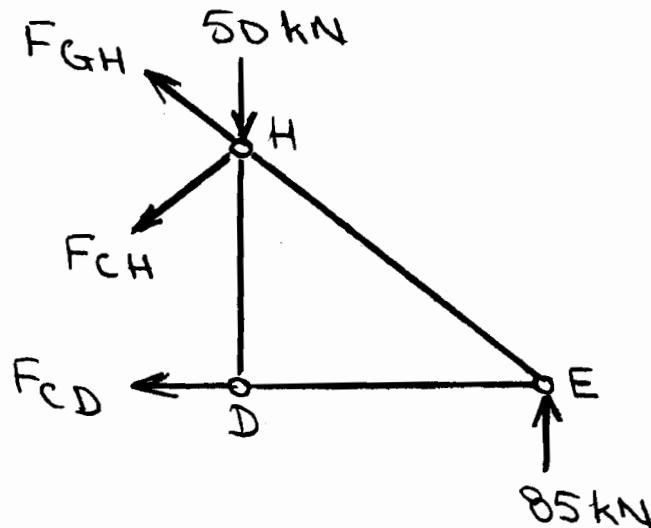
$$F_{CD} = -131.09 \text{ k} = \underline{131.09 \text{ k (C)}}$$

$$+G \sum M_C = 0 \quad F_{IJ}(20) - 28.2(15 + 30 + 45) - 14.1(60) = 0$$

$$F_{IJ} = 169.2 \text{ k (T)}$$

4.37

Section through members CD, CH and GH:



$$+\zeta \sum M_E = 0$$

$$50(4) + \frac{4}{5} F_{CH}(3) + \frac{3}{5} F_{CH}(4) = 0$$

$$F_{CH} = -41.67 \text{ kN}$$

$$\underline{F_{CH} = 41.67 \text{ kN (C)}}$$

$$+\uparrow \sum F_y = 0$$

$$\frac{3}{5} F_{GH} - 50 + \frac{3}{5} (41.67) + 85 = 0$$

$$F_{GH} = -100 \text{ kN}$$

$$\underline{F_{GH} = 100 \text{ kN (C)}}$$

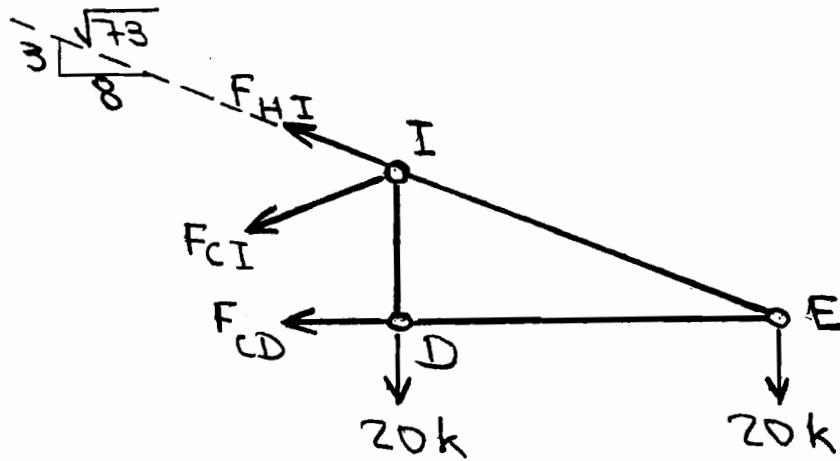
$$+\zeta \sum M_H = 0$$

$$-F_{CD}(3) + 85(4) = 0$$

$$\underline{F_{CD} = 113.33 \text{ kN (T)}}$$

4.38

Section through members CD, CI and HI:



$$+\text{C} \sum M_I = 0 \quad -F_{CD}(3.75) - 20(10) = 0$$

$$F_{CD} = -53.33 \text{ k} \quad \underline{F_{CD} = 53.33 \text{ k (c)}}$$

$$+\text{C} \sum M_E = 0$$

$$20(10) + \frac{8}{\sqrt{73}} F_{CI}(3.75) + \frac{3}{\sqrt{73}} F_{CI}(10) = 0$$

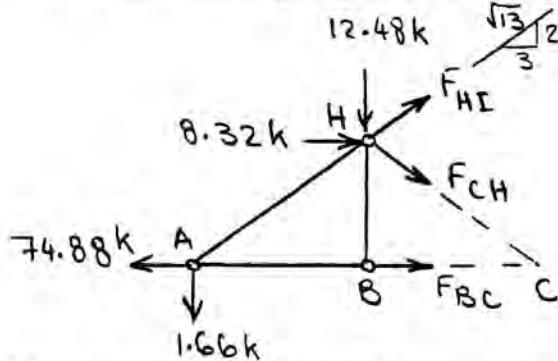
$$F_{CI} = -28.48 \text{ k} \quad \underline{F_{CI} = 28.48 \text{ k (c)}}$$

$$+\uparrow \sum F_y = 0$$

$$\frac{3}{\sqrt{73}} F_{HI} + \frac{3}{\sqrt{73}} (28.48) - 20 - 20 = 0$$

$$\underline{F_{HI} = 85.44 \text{ k (T)}}$$

4.39 Section through members BC, CH and HI:



$$+G \sum M_H = 0 \quad -74.88 \left(\frac{40}{3}\right) + 1.66(20) + F_{BC} \left(\frac{40}{3}\right) = 0$$

$$\underline{F_{BC} = 72.39 \text{ k (T)}}$$

$$+G \sum M_C = 0 \quad 1.66(40) + 12.48(20) - 8.32 \left(\frac{40}{3}\right)$$

$$- \frac{3}{\sqrt{3}} F_{HI} \left(\frac{40}{3}\right) - \frac{2}{\sqrt{3}} F_{HI} (20) = 0$$

$$\underline{F_{HI} = 9.24 \text{ k (T)}}$$

$$+G \sum M_A = 0 \quad -8.32 \left(\frac{40}{3}\right) - 12.48(20) - \frac{3}{\sqrt{3}} F_{CH} \left(\frac{40}{3}\right)$$

$$- \frac{2}{\sqrt{3}} F_{CH} (20) = 0$$

$$\underline{F_{CH} = -16.25 \text{ k} = 16.25 \text{ k (C)}}$$

Section through members DE, EK and KL:

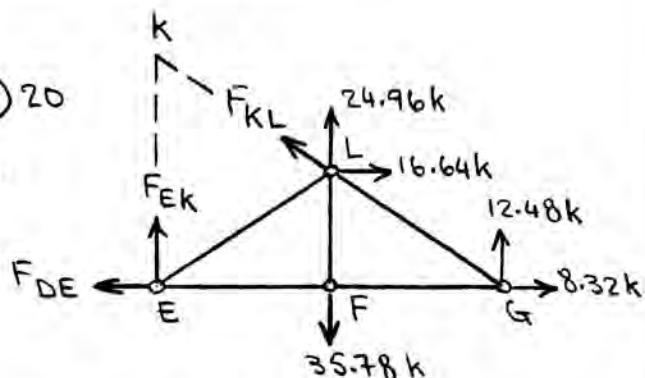
$$+G \sum M_K = 0$$

$$-F_{DE} \left(\frac{80}{3}\right) + (24.96 - 35.78) 20$$

$$+ 12.48(40) + 16.64 \left(\frac{40}{3}\right)$$

$$+ 8.32 \left(\frac{80}{3}\right) = 0$$

$$\underline{F_{DE} = 27.25 \text{ k (T)}}$$



$$+G \sum M_G = 0$$

$$-F_{EK}(40) + (35.78 - 24.96) 20 - 16.64 \left(\frac{40}{3}\right) = 0$$

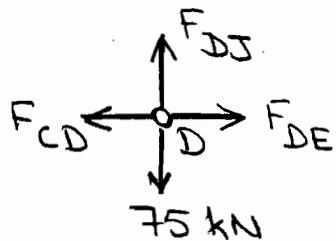
$$\underline{F_{EK} = -0.14 \text{ k} = 0.14 \text{ k (C)}}$$

$$+G \sum M_E = 0 \quad (24.96 - 35.78) 20 + 12.48(40) - 16.64 \left(\frac{40}{3}\right)$$

$$+ \frac{3}{\sqrt{3}} F_{KL} \left(\frac{40}{3}\right) + \frac{2}{\sqrt{3}} F_{KL} (20) = 0$$

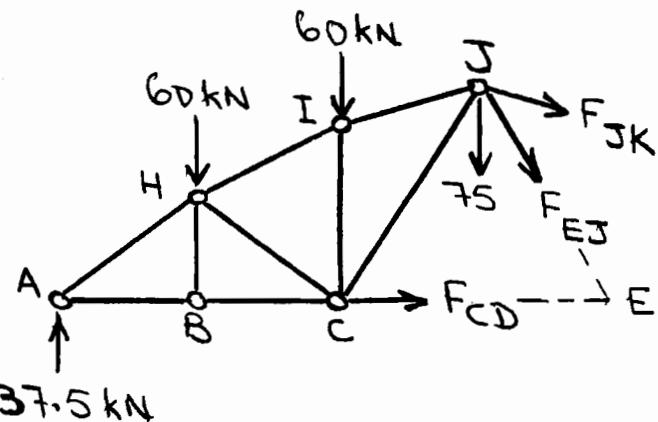
$$\underline{F_{KL} = -2.75 \text{ k} = 2.75 \text{ k (C)}}$$

4.40 Joint D:



$$\sum F_y = 0 \quad F_{DJ} = 75 \text{ kN (T)}$$

Section through members CD, DJ, EJ and JK:



$$+\zeta \sum M_J = 0$$

$$-37.5(12) + 60(8) + 60(4) + F_{CD}(6) = 0$$

$$F_{CD} = -45 \text{ kN} \quad \underline{F_{CD} = 45 \text{ kN (C)}}$$

$$+\zeta \sum M_E = 0$$

$$-37.5(16) + 60(12) + 60(8) + 75(4)$$

$$-\frac{4}{\sqrt{17}} F_{JK}(6) + \frac{1}{\sqrt{17}} F_{JK}(4) = 0$$

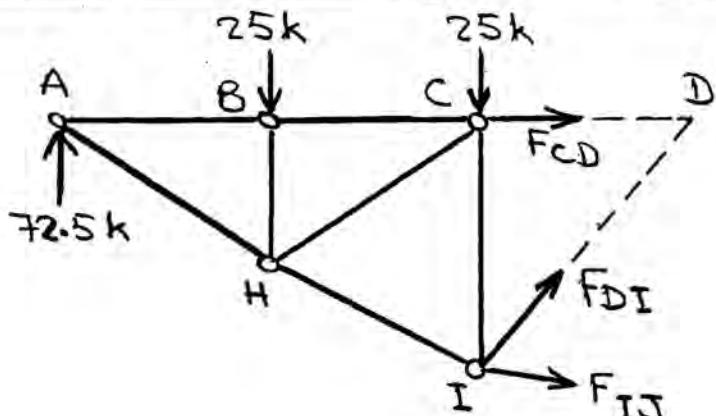
$$\underline{F_{JK} = 185.54 \text{ kN (T)}}$$

$$+\uparrow \sum F_y = 0$$

$$37.5 - 60 - 60 - 75 - \frac{3}{\sqrt{13}} F_{EJ} - \frac{1}{\sqrt{17}} (185.54) = 0$$

$$\underline{F_{EJ} = -243.37 \text{ kN} \quad F_{EJ} = 243.37 \text{ kN (C)}}$$

4.41 Section through members CD, DI and IJ:



$$+\text{G} \sum M_I = 0 \quad -72.5(60) + 25(30) - F_{CD}(35) = 0$$

$$F_{CD} = -102.86 \text{ k} \quad \underline{F_{CD} = 102.86 \text{ k (C)}}$$

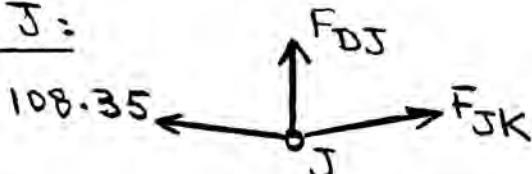
$$+\text{G} \sum M_D = 0 \quad -72.5(90) + 25(60) + 25(30) + \frac{6}{\sqrt{37}} F_{IJ}(35)$$

$$+ \frac{1}{\sqrt{37}} F_{IJ}(30) = 0 \quad \underline{F_{IJ} = 108.35 \text{ k (T)}}$$

$$+\uparrow \sum F_y = 0 \quad 72.5 - 25 - 25 - \frac{1}{\sqrt{37}} (108.35) + \frac{7}{\sqrt{85}} F_{DI} = 0$$

$$F_{DI} = -6.17 \text{ k} \quad \underline{F_{DI} = 6.17 \text{ k (C)}}$$

Joint J:



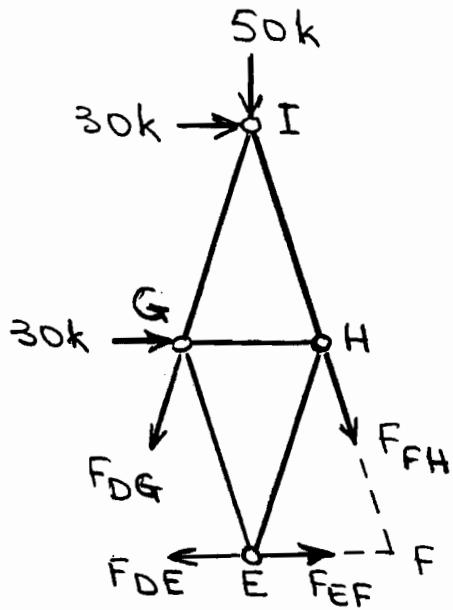
$$+\rightarrow \sum F_x = 0 \quad -\frac{6}{\sqrt{37}} (108.35) + \frac{6}{\sqrt{37}} F_{JK} = 0$$

$$F_{JK} = 108.35 \text{ k (T)}$$

$$+\uparrow \sum F_y = 0 \quad 2\left(\frac{1}{\sqrt{37}}\right) 108.35 + F_{DJ} = 0$$

$$F_{DJ} = -35.63 \text{ k} \quad \underline{F_{DJ} = 35.63 \text{ k (C)}}$$

4.42 Section through members DG, DE, EF and FH:



$$+\zeta \sum M_F = 0$$

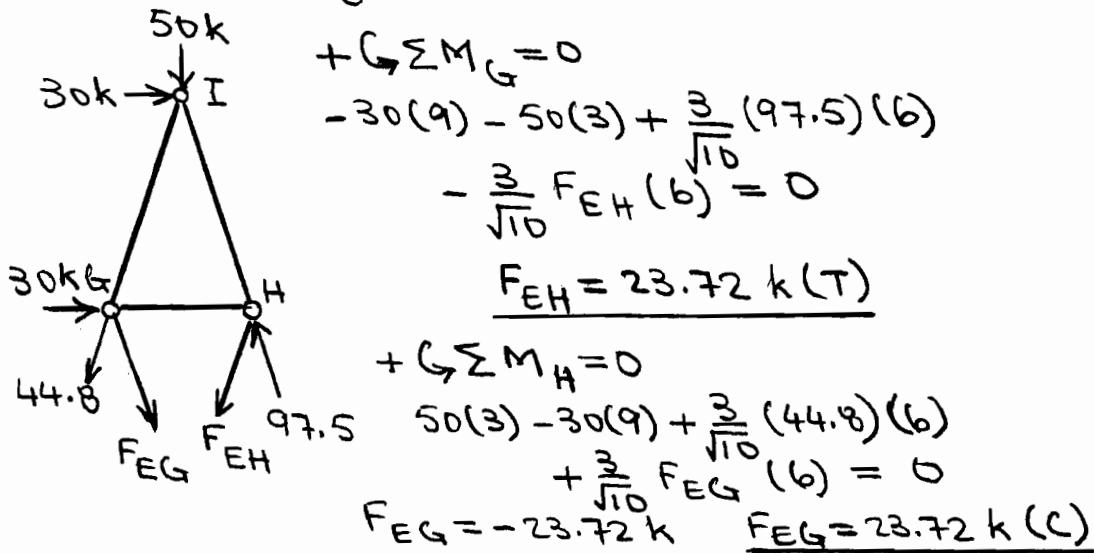
$$50(6) - 30(18) - 30(9) + \frac{1}{\sqrt{10}} F_{DG}(9) + \frac{3}{\sqrt{10}} F_{DG}(9) = 0$$

$$\underline{F_{DG} = 44.8 \text{ k (T)}}$$

$$+\uparrow \sum F_y = 0 \quad -\frac{3}{\sqrt{10}}(44.8) - 50 - \frac{3}{\sqrt{10}} F_{FH} = 0$$

$$\underline{F_{FH} = -97.5 \text{ k} = 97.5 \text{ k (C)}}$$

Section through members DG, EG, EH and FH:



$$+\zeta \sum M_G = 0$$

$$-30(9) - 50(3) + \frac{3}{\sqrt{10}}(97.5)(6) - \frac{3}{\sqrt{10}} F_{EH}(6) = 0$$

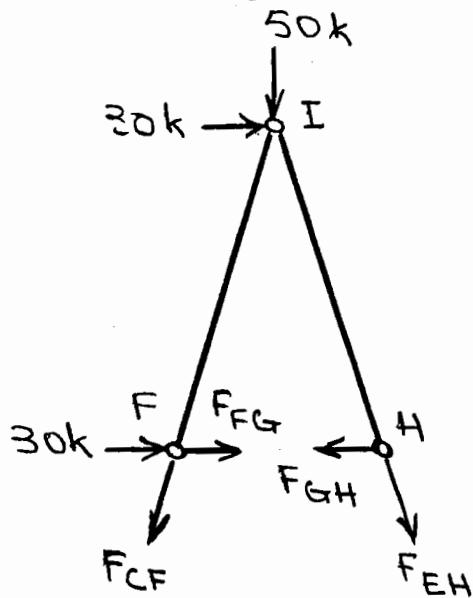
$$\underline{F_{EH} = 23.72 \text{ k (T)}}$$

$$+\zeta \sum M_H = 0$$

$$50(3) - 30(9) + \frac{3}{\sqrt{10}}(44.8)(6) + \frac{3}{\sqrt{10}} F_{EG}(6) = 0$$

$$\underline{F_{EG} = -23.72 \text{ k} \quad F_{EG} = 23.72 \text{ k (C)}}$$

4.43 Section through members CF, FG, GH and EH:



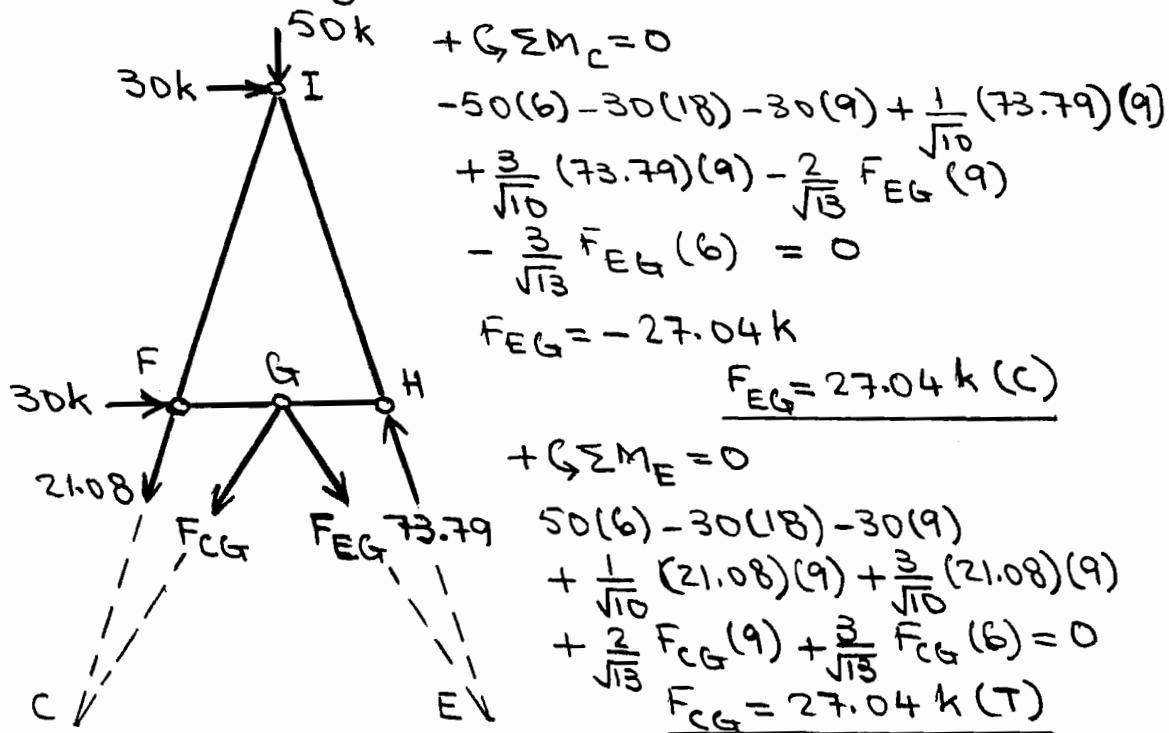
$$+\zeta \sum M_H = 0 \quad 50(3) - 30(9) + \frac{3}{\sqrt{10}} F_{CF}(6) = 0$$

$$\underline{F_{CF} = 21.08 \text{ k (T)}}$$

$$+\zeta \sum M_F = 0 \quad -50(3) - 30(9) - \frac{3}{\sqrt{10}} F_{EH}(6) = 0$$

$$\underline{F_{EH} = -73.79 \text{ k} = 73.79 \text{ k (C)}}$$

Section through members CF, CG, EG and EH:



$$+\zeta \sum M_C = 0$$

$$-50(6) - 30(18) - 30(9) + \frac{1}{\sqrt{10}} (73.79)(9)$$

$$+ \frac{3}{\sqrt{10}} (73.79)(9) - \frac{2}{\sqrt{13}} F_{EG}(9)$$

$$- \frac{3}{\sqrt{13}} F_{EH}(6) = 0$$

$$\underline{F_{EG} = -27.04 \text{ k}}$$

$$\underline{F_{EG} = 27.04 \text{ k (C)}}$$

$$+\zeta \sum M_E = 0$$

$$50(6) - 30(18) - 30(9)$$

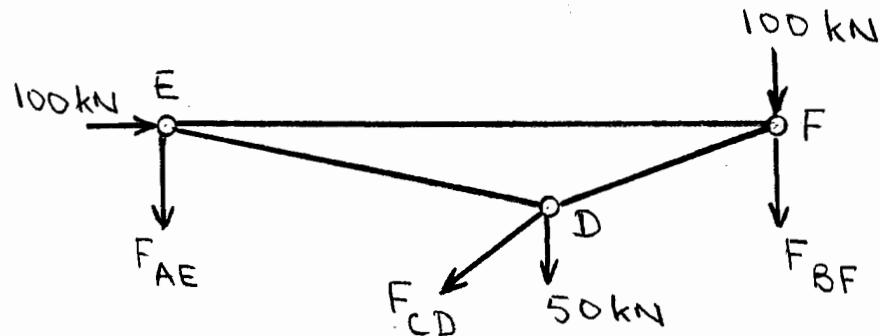
$$+ \frac{1}{\sqrt{10}} (21.08)(9) + \frac{3}{\sqrt{10}} (21.08)(9)$$

$$+ \frac{2}{\sqrt{13}} F_{CG}(9) + \frac{3}{\sqrt{15}} F_{CG}(6) = 0$$

$$\underline{F_{CG} = 27.04 \text{ k (T)}}$$

4.44

Section through members AE, CD and BF :



$$\rightarrow \sum F_x = 0 \quad 100 - \left(\frac{4}{5}\right) F_{CD} = 0 \quad \underline{F_{CD} = 125 \text{ kN (T)}}$$

$$+ \circlearrowleft \sum M_F = 0 \quad F_{AE}(16) - \frac{4}{5}(125)(2) + \frac{3}{5}(125)(6) + 50(6) = 0 \\ F_{AE} = -34.38 \text{ kN} = \underline{34.38 \text{ kN (C)}}$$

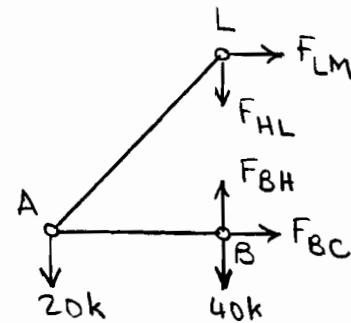
$$+ \uparrow \sum F_y = 0 \quad 34.38 - \frac{3}{5}(125) - 50 - 100 - F_{BF} = 0 \\ F_{BF} = -190.62 \text{ kN} = \underline{190.62 \text{ kN (C)}}$$

4.45 Section through members BC, BH, HL and LM:

$$+G \sum M_B = 0$$

$$20(30) - F_{LM}(30) = 0$$

$$\underline{F_{LM} = 20 \text{ k (T)}}$$



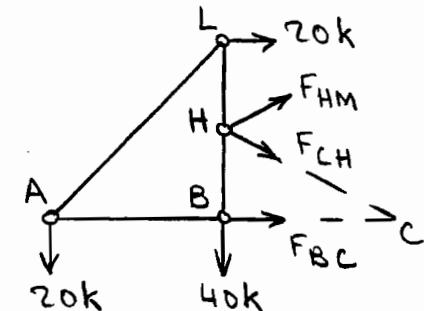
Section through members BC, CH, HM and LM:

$$+G \sum M_C = 0$$

$$20(60) + 40(30) - 20(30)$$

$$- \frac{2}{\sqrt{5}} F_{HM}(15) - \frac{1}{\sqrt{5}} F_{HM}(30) = 0$$

$$\underline{F_{HM} = 67.08 \text{ k (T)}}$$



$$+\uparrow \sum F_y = 0 \quad -20 - 40 + \frac{1}{\sqrt{5}} (67.08) - \frac{1}{\sqrt{5}} F_{CH} = 0$$

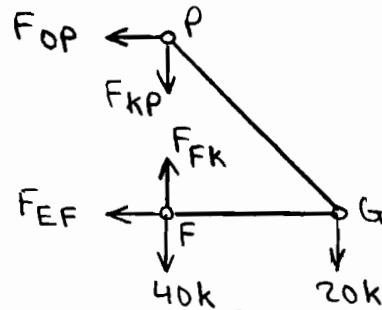
$$F_{CH} = -67.08 \text{ k} = \underline{67.08 \text{ k (C)}}$$

Section through members EF, FK, KP, and OP:

$$+G \sum M_P = 0$$

$$-F_{EF}(30) - 20(30) = 0$$

$$\underline{F_{EF} = -20 \text{ k} = 20 \text{ k (C)}}$$



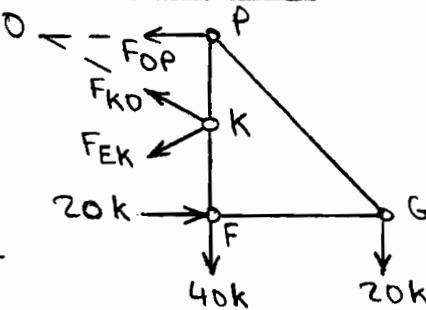
Section through members EF, EK, KO and OP:

$$+G \sum M_O = 0$$

$$-20(60) - 40(30) + 20(30)$$

$$- \frac{2}{\sqrt{5}} F_{EK}(15) - \frac{1}{\sqrt{5}} F_{EK}(30) = 0$$

$$\underline{F_{EK} = -67.08 \text{ k} = 67.08 \text{ k (C)}}$$

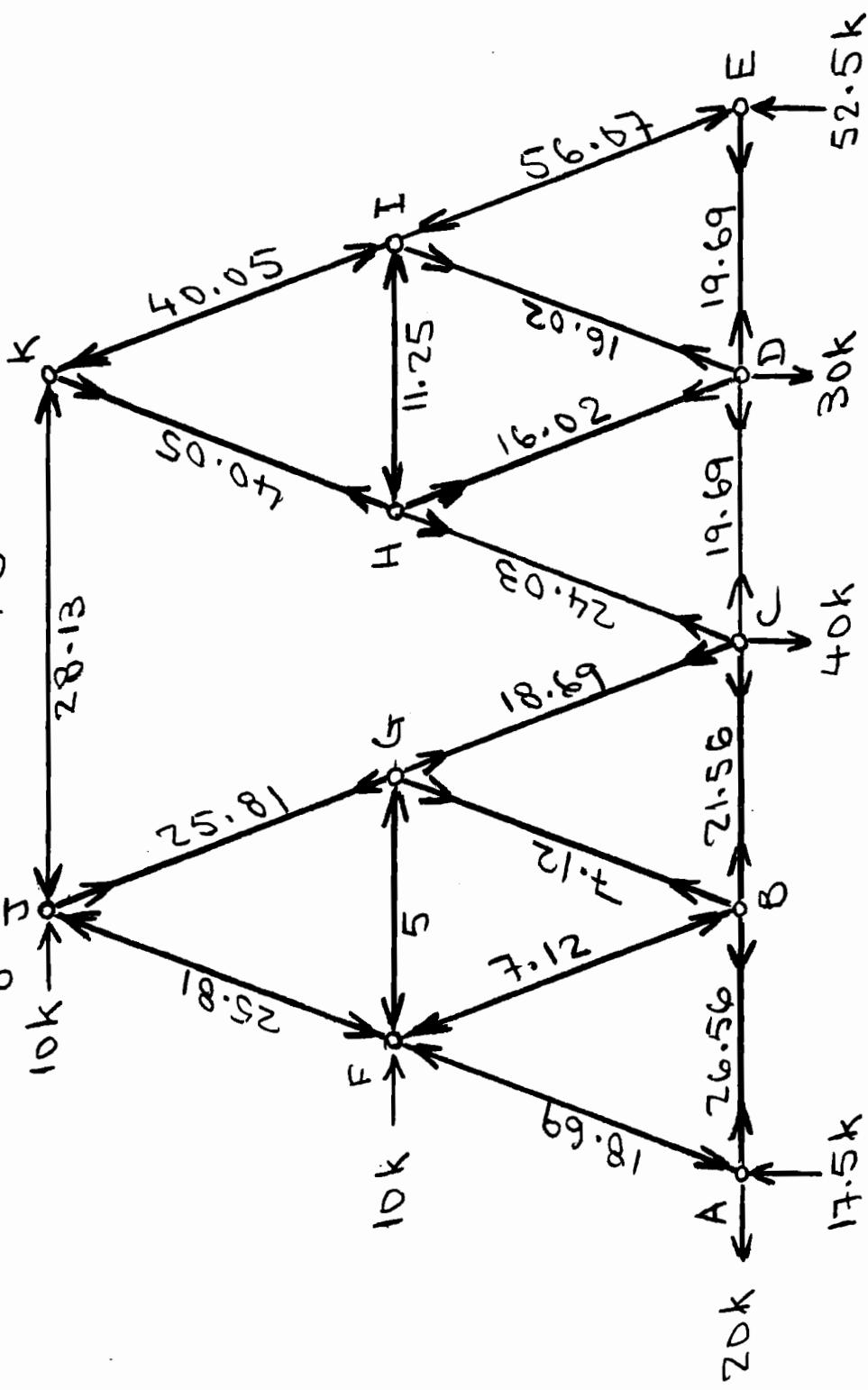


$$+\uparrow \sum F_y = 0$$

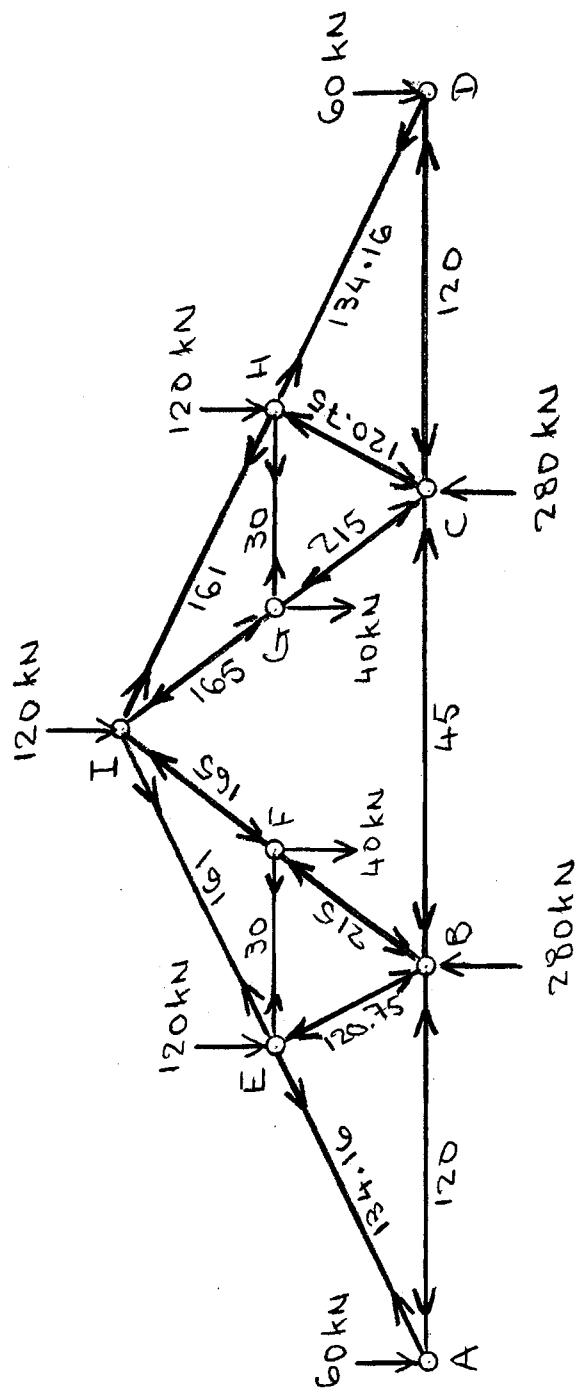
$$\frac{1}{\sqrt{5}} (67.08) + \frac{1}{\sqrt{5}} F_{KO} - 40 - 20 = 0$$

$$\underline{F_{KO} = 67.08 \text{ k (T)}}$$

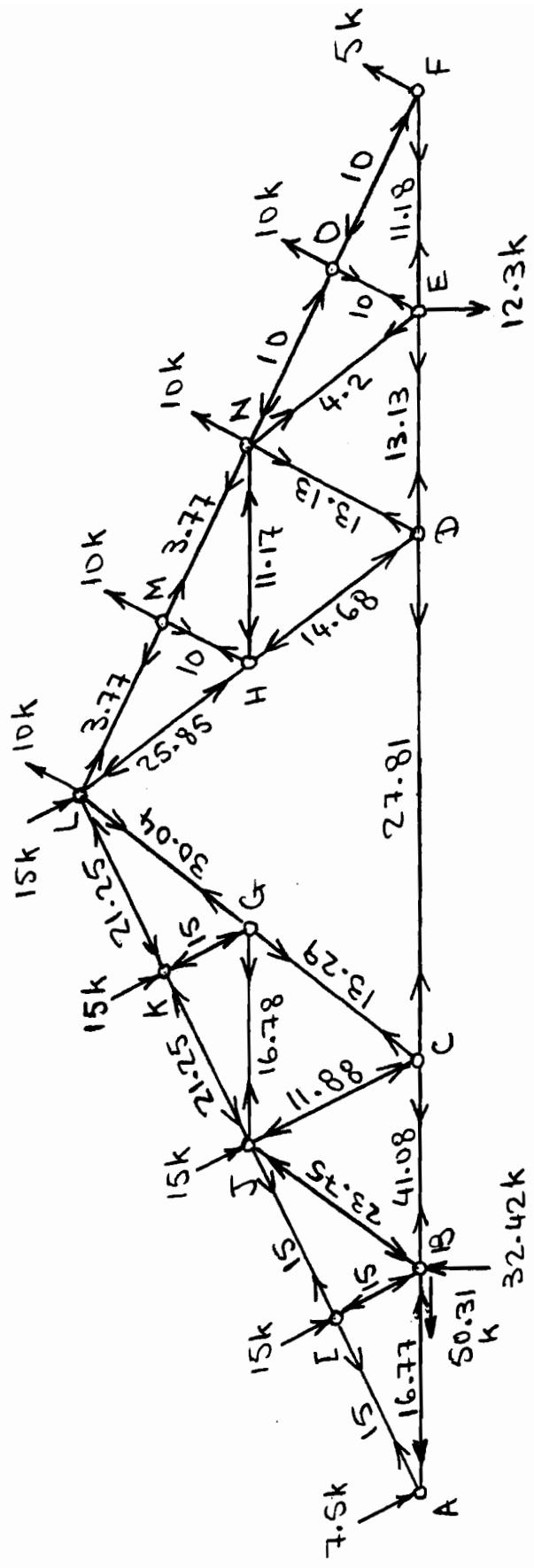
4.46 Considering a section through members BC, CG and JK, compute the force in member JK: $F_{JK} = 28.13 \text{ k}$ (C). Then, determine the remaining member forces by the method of joints.



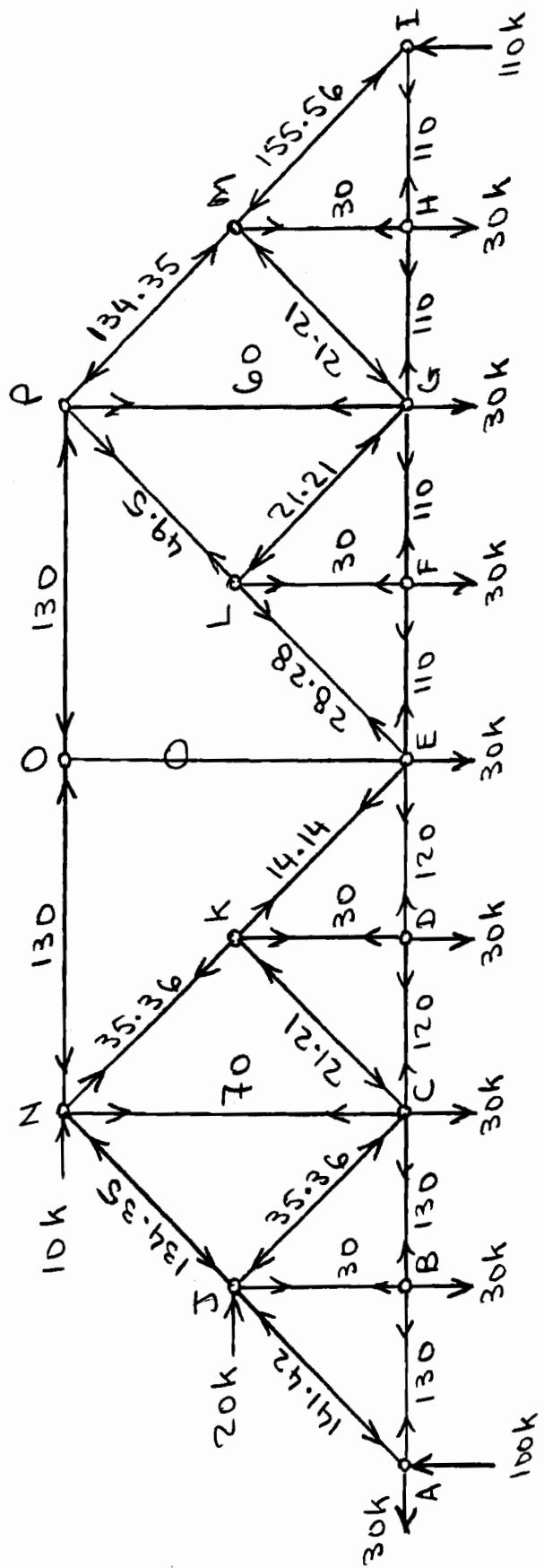
4.47 Considering a section through members BC, FI and EI, compute the force in member BC: $F_{BC} = 45 \text{ kN (c)}$. Then, determine the remaining member forces by the method of joints.



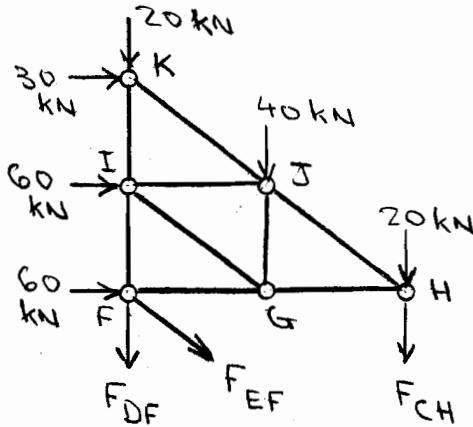
4.48 Considering a section through members CD, GL and KL, compute the force in member CD: $F_{CD} = 27.81 \text{ k (T)}$. Then, determine the remaining member forces by the method of joints.



4.49 Considering a section through members DE, EK and NO, compute the force in member NO: $F_{NO} = \underline{130 \text{ kcc}}$. Then, determine the remaining member forces by the method of joints.



4.50 Section through members DF, EF and CH:



$$+G \sum M_F = 0 \quad -30(6) - 60(3) - 40(4) - 20(8) - F_{CH}(8) = 0$$

$$F_{CH} = -85 \text{ kN} = \underline{85 \text{ kN (C)}}$$

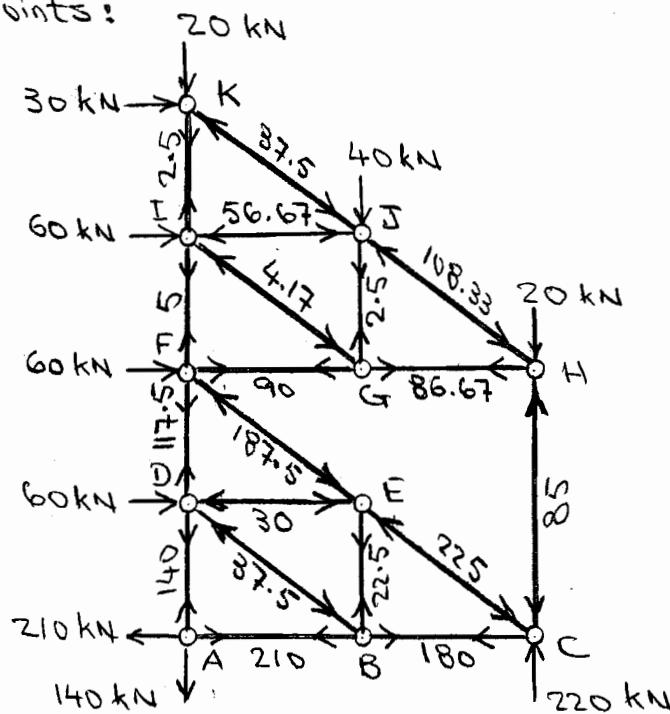
$$\rightarrow \sum F_x = 0 \quad 30 + 60 + 60 + \left(\frac{4}{5}\right)F_{EF} = 0$$

$$F_{EF} = -187.5 \text{ kN} = \underline{187.5 \text{ kN (C)}}$$

$$+\uparrow \sum F_y = 0 \quad -20 - 40 - 20 - F_{DF} + 85 + \left(\frac{3}{5}\right)(187.5) = 0$$

$$F_{DF} = 117.5 \text{ (T)}$$

The remaining member forces can now be determined by the method of joints:



4.51

Member	Projection			Length (ft)
	x (ft)	y (ft)	z (ft)	
AD	3	10	5	11.58
BD	8	10	7	14.59
CD	2	10	4	10.95

Joint D:

$$\sum F_x = 0 \quad -\left(\frac{3}{11.58}\right)F_{AD} + \left(\frac{8}{14.59}\right)F_{BD} + \left(\frac{2}{10.95}\right)F_{CD} + 6 = 0$$

$$\sum F_y = 0 \quad -\left(\frac{10}{11.58}\right)F_{AD} - \left(\frac{10}{14.59}\right)F_{BD} - \left(\frac{10}{10.95}\right)F_{CD} - 12 = 0$$

$$\sum F_z = 0 \quad \left(\frac{5}{11.58}\right)F_{AD} + \left(\frac{7}{14.59}\right)F_{BD} - \left(\frac{4}{10.95}\right)F_{CD} = 0$$

Solving these simultaneous equations, we obtain:

$$F_{AD} = 1.12 \text{ k (T)}$$

$$F_{BD} = -7.56 \text{ k} = \underline{\underline{7.56 \text{ k (C)}}}$$

$$F_{CD} = -8.54 \text{ k} = \underline{\underline{8.54 \text{ k (C)}}}$$

4.52

Member	Projection			Length (m)
	x (m)	y (m)	z (m)	
AB	0	0	3	3
AC	4	0	3	5
BC	4	0	0	4
AD	0	7	3	7.62
BD	0	7	0	7
CD	4	7	0	8.06

Joint D:

$$\sum F_x = 0 \quad \left(\frac{4}{8.06}\right) F_{CD} = 0 \quad \underline{F_{CD} = 0}$$

$$\sum F_y = 0 \quad \left(\frac{3}{7.62}\right) F_{AD} + 40 = 0 \quad F_{AD} = -101.6 \text{ kN}$$

$$\underline{F_{AD} = 101.6 \text{ kN (C)}}$$

$$\sum F_z = 0 \quad \left(\frac{7}{7.62}\right) 101.6 - F_{BD} - 60 = 0$$

$$\underline{F_{BD} = 33.33 \text{ kN (T)}}$$

Joint B:

$$\sum F_x = 0 \quad \underline{F_{BC} = 0}$$

$$\sum F_y = 0 \quad \underline{F_{AB} = 0}$$

Joint C:

$$\sum F_y = 0 \quad \left(\frac{3}{5}\right) F_{AC} = 0 \quad \underline{F_{AC} = 0}$$

4.53

Member	Projection			Length (ft)
	x (ft)	y (ft)	z (ft)	
AB	20	0	0	20
AC	14	0	10	17.2
BC	6	0	10	11.66
AD	10	20	5	22.91
BD	10	20	5	22.91
CD	4	20	5	21

Joint D:

$$\sum F_x = 0 \quad -\left(\frac{10}{22.91}\right)F_{AD} + \left(\frac{10}{22.91}\right)F_{BD} + \left(\frac{4}{21}\right)F_{CD} + 25 = 0$$

$$\sum F_y = 0 \quad -\left(\frac{20}{22.91}\right)F_{AD} - \left(\frac{20}{22.91}\right)F_{BD} - \left(\frac{20}{21}\right)F_{CD} - 30 = 0$$

$$\sum F_z = 0 \quad \left(\frac{5}{22.91}\right)F_{AD} + \left(\frac{5}{22.91}\right)F_{BD} - \left(\frac{5}{21}\right)F_{CD} = 0$$

Solving these equations, we obtain:

$$F_{AD} = 16.63 \text{ k (T)}$$

$$F_{BD} = -33.84 \text{ k} = \underline{33.84 \text{ k (C)}}$$

$$F_{CD} = -15.74 \text{ k} = \underline{15.74 \text{ k (C)}}$$

Joint A:

$$\sum F_y = 0 \quad -\left(\frac{5}{22.91}\right)16.63 - \left(\frac{10}{17.2}\right)F_{AC} = 0$$

$$F_{AC} = -6.24 \text{ k} = \underline{6.24 \text{ k (C)}}$$

$$\sum F_x = 0 \quad \left(\frac{10}{22.91}\right)16.63 - \left(\frac{14}{17.2}\right)6.24 + F_{AB} = 0$$

$$F_{AB} = -2.18 \text{ k} = \underline{2.18 \text{ k (C)}}$$

Joint B:

$$\sum F_z = 0 \quad \left(\frac{5}{22.91}\right)33.84 - \left(\frac{10}{11.66}\right)F_{BC} = 0$$

$$F_{BC} = 8.61 \text{ k (T)}$$

4.54

Member	Projection			Length (m)
	x (m)	y (m)	z (m)	
AB	8	0	0	8
CD	8	0	0	8
BC	0	0	4	4
AD	0	0	4	4
BD	8	0	4	8.94
AE	4	8	2	9.17
BE	4	8	2	9.17
CE	4	8	2	9.17
DE	4	8	2	9.17

Joint C: $\sum F_y = 0 \quad (\frac{8}{9.17}) F_{CE} = 0 \quad \underline{F_{CE} = 0}$

$$\sum F_z = 0 \quad \underline{F_{BC} = 0}$$

Joint E: $\sum F_x = 0 \quad \frac{4}{9.17} (-F_{AE} + F_{BE} - F_{DE}) + 30 = 0$

$$\sum F_y = 0 \quad -\frac{8}{9.17} (F_{AE} + F_{BE} + F_{DE}) - 60 = 0$$

$$\sum F_z = 0 \quad \frac{2}{9.17} (F_{AE} + F_{BE} - F_{DE}) + 40 = 0$$

Solving these equations, we obtain:

$$F_{AE} = -57.31 \text{ kN} = \underline{57.31 \text{ kN (C)}};$$

$$F_{BE} = -68.78 \text{ kN} = \underline{68.78 \text{ kN (C)}} \text{ and } F_{DE} = 57.31 \text{ kN (T)}$$

Joint A:

$$\sum F_x = 0 \quad -(\frac{4}{9.17}) 57.31 + F_{AB} = 0 \quad \underline{F_{AB} = 25 \text{ kN (T)}}$$

$$\sum F_z = 0 \quad (\frac{2}{9.17}) 57.31 - F_{AD} = 0 \quad \underline{F_{AD} = 12.5 \text{ kN (T)}}$$

Joint D:

$$\sum F_z = 0 \quad (\frac{2}{9.17}) 57.31 + 12.5 + (\frac{4}{8.94}) F_{BD} = 0$$

$$F_{BD} = -55.87 \text{ kN} = \underline{55.87 \text{ kN (C)}}$$

$$\sum F_x = 0 \quad (\frac{4}{9.17}) 57.31 - (\frac{8}{8.94}) 55.87 + F_{CD} = 0$$

$$F_{CD} = 25 \text{ kN (T)}$$

4.55

Member	Force (k)
AB	29.17(T)
CD	15.83(C)
BC	7.22(T)
AD	0
AC	0.5(T)
EF	28.33(T)
GH	10.0(C)
FG	5.55(C)
EH	5.55(C)
FH	2.0(T)
AE	4.12(T)
BF	8.25(C)
CG	80.41(C)
DH	39.17(C)
BE	52.25(C)
CF	8.83(T)
DG	45.12(T)
AH	2.21(C)

Chapter Five

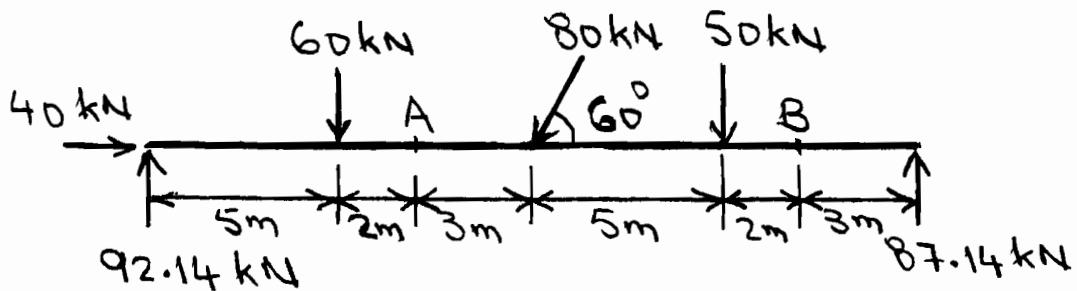
Beams and Frames:

Shear and Bending

Movement

CHAPTER 5

5.1



$$Q_A = \underline{-40 \text{ kN}}$$

$$S_A = 92.14 - 60 = \underline{32.14 \text{ kN}}$$

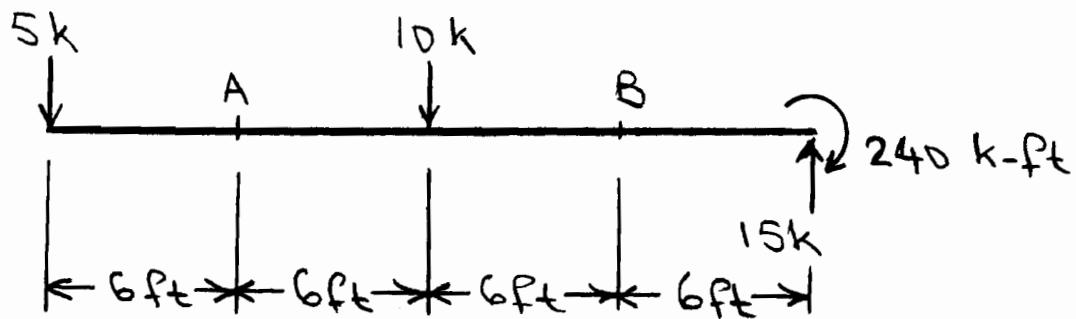
$$M_A = 92.14(7) - 60(2) = \underline{524.98 \text{ kN.m}}$$

$$Q_B = \underline{0}$$

$$S_B = \underline{-87.14 \text{ kN}}$$

$$M_B = 87.14(3) = \underline{261.42 \text{ kN.m}}$$

5.2



$$Q_A = 0$$

$$\beta_A = \underline{-5 \text{ k}}$$

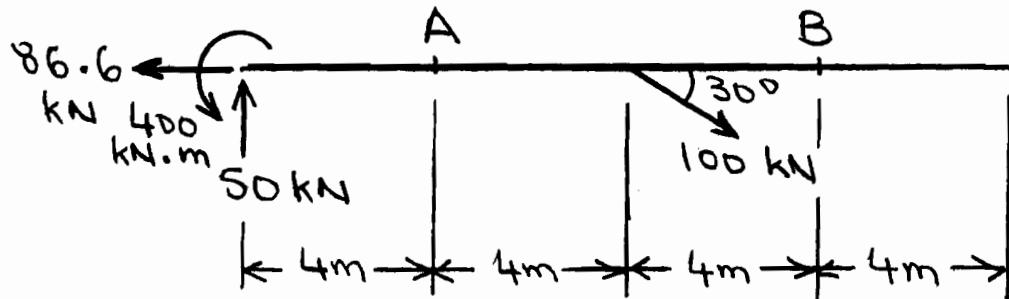
$$M_A = -5(6) = \underline{-30 \text{ k-ft}}$$

$$Q_B = 0$$

$$\beta_B = -5 - 10 = \underline{-15 \text{ k}}$$

$$M_B = -5(18) - 10(6) = \underline{-150 \text{ k-ft}}$$

5.3



$$Q_A = 100 \cos 30^\circ = \underline{86.6 \text{ kN}}$$

$$S_A = 100 \sin 30^\circ = \underline{50 \text{ kN}}$$

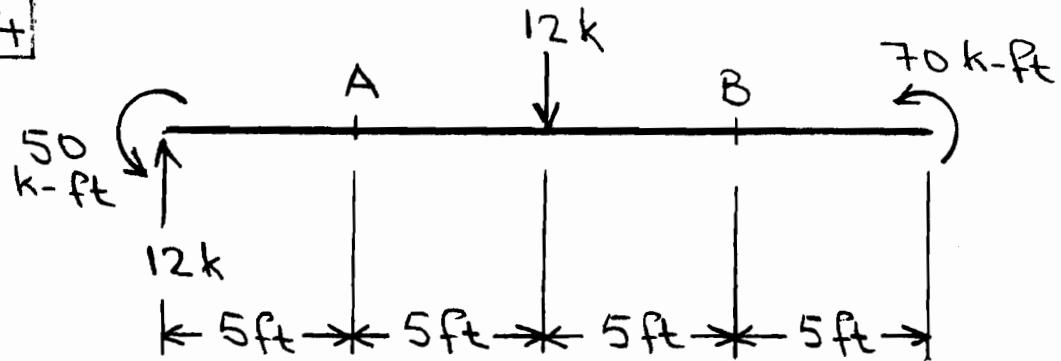
$$M_A = -100 \sin 30^\circ (4) = \underline{-200 \text{ kN.m}}$$

$$Q_B = \underline{0}$$

$$S_B = \underline{0}$$

$$M_B = \underline{0}$$

5.4



$$Q_A = 0$$

$$S_A = \underline{12 \text{ k}}$$

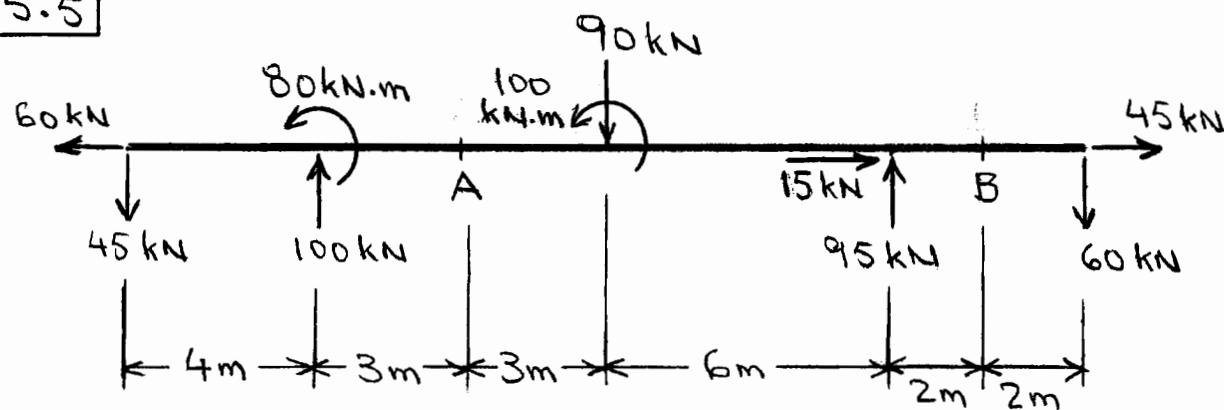
$$M_A = 12(5) - 50 = \underline{10 \text{ k-ft}}$$

$$Q_B = 0$$

$$S_B = \underline{0}$$

$$M_B = \underline{70 \text{ k-ft}}$$

5.5



$$Q_A = \underline{60\text{ kN}}$$

$$S_A = -45 + 100 = \underline{55\text{ kN}}$$

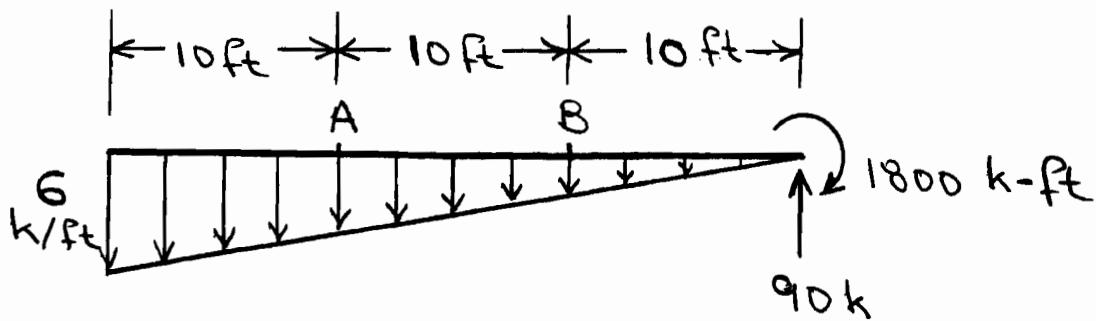
$$M_A = -45(7) - 80 + 100(3) = \underline{-95\text{ kN.m}}$$

$$Q_B = \underline{45\text{ kN}}$$

$$S_B = \underline{60\text{ kN}}$$

$$M_B = -60(2) = \underline{-120\text{ kN.m}}$$

5.6



$$Q_A = 0$$

$$S_A = -\frac{(6+4)}{2} (10) = \underline{-50 \text{ k}}$$

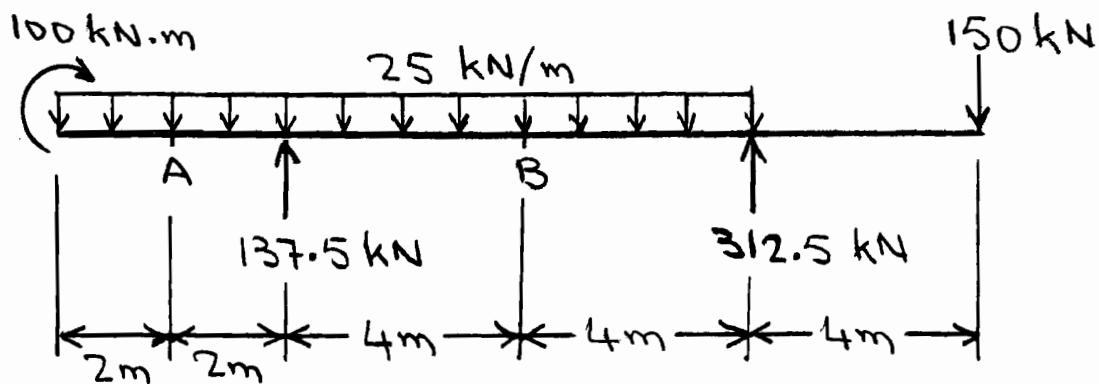
$$M_A = -4(10)5 - \frac{1}{2}(2)(10)\left(\frac{20}{3}\right) = \underline{-266.67 \text{ k-ft}}$$

$$Q_B = 0$$

$$S_B = -\frac{(6+2)}{2} (20) = \underline{-80 \text{ k}}$$

$$M_B = -2(20)10 - \frac{1}{2}(4)(20)\left(\frac{40}{3}\right) = \underline{-933.33 \text{ k-ft}}$$

5.7



$$Q_A = 0$$

$$S_A = -25(2) = -50 \text{ kN}$$

$$M_A = 100 - 25(2)(1) = 50 \text{ kN} \cdot \text{m}$$

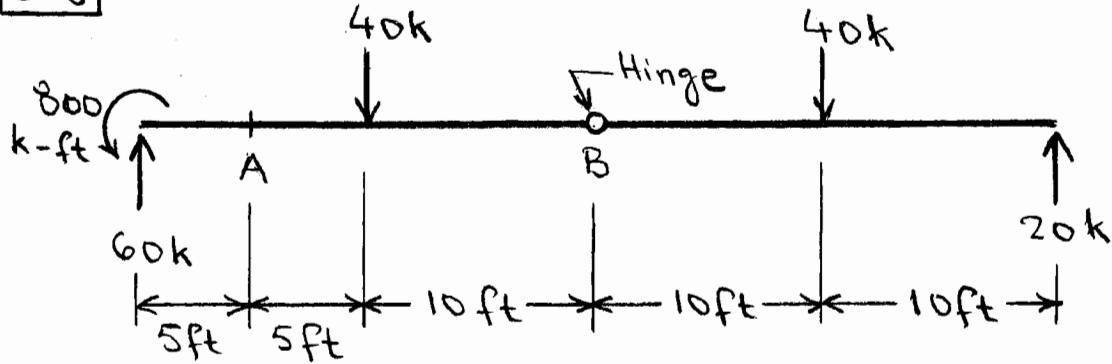
$$Q_B = 0$$

$$S_B = 150 - 312.5 + 25(4) = -62.5 \text{ kN}$$

$$M_B = -150(8) + 312.5(4) - 25(4)(2)$$

$$M_B = -150 \text{ kN} \cdot \text{m}$$

5.8



$$Q_A = 0$$

$$S_A = \underline{60 \text{ k}}$$

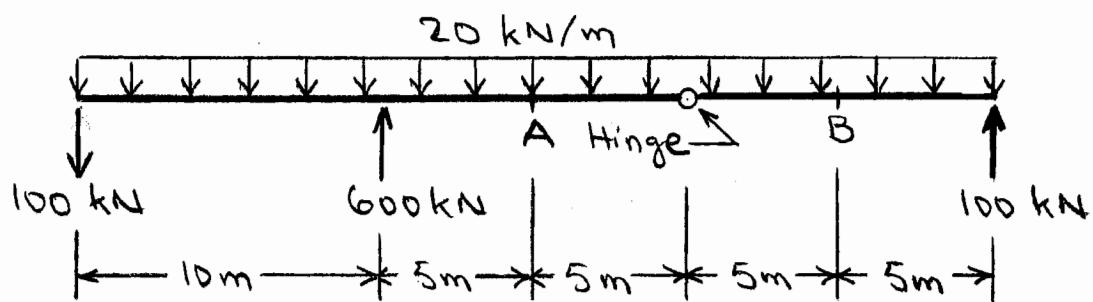
$$M_A = -800 + 60(5) = \underline{-500 \text{ k-ft}}$$

$$Q_B = 0$$

$$S_B = -20 + 40 = \underline{20 \text{ k}}$$

$$M_B = 20(20) - 40(10) = \underline{0}$$

5.9



$$Q_A = 0$$

$$S_A = -100 - 20(15) + 600 = \underline{200 \text{ kN}}$$

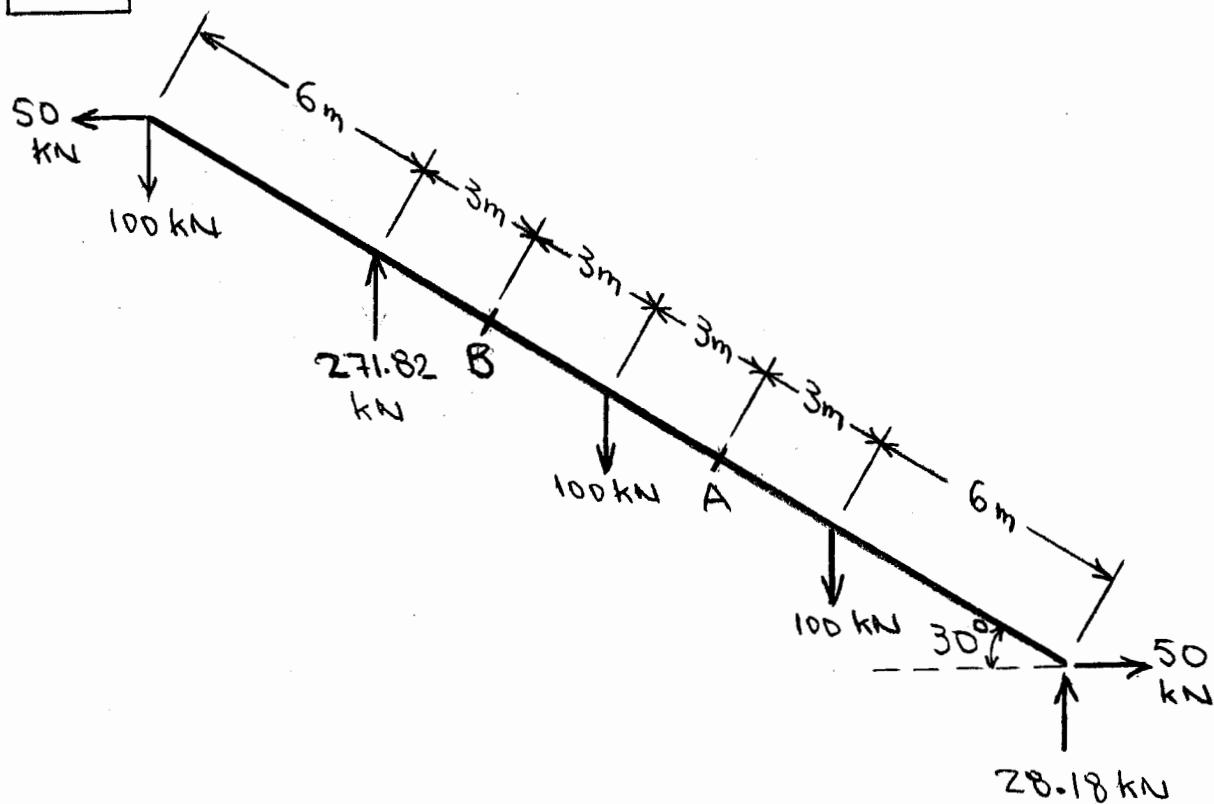
$$M_A = -100(15) - 20(15)(7.5) + 600(5) = \underline{-750 \text{ kN.m}}$$

$$Q_B = 0$$

$$S_B = -100 + 20(5) = \underline{0}$$

$$M_B = 100(5) - 20(5)(2.5) = \underline{250 \text{ kN.m}}$$

5.10



$$Q_A = 50(\cos 30^\circ) - 28.18(\sin 30^\circ) + 100(\sin 30^\circ) = \underline{79.2 \text{ kN}}$$

$$S_A = -50(\sin 30^\circ) - 28.18(\cos 30^\circ) + 100(\cos 30^\circ) = \underline{37.2 \text{ kN}}$$

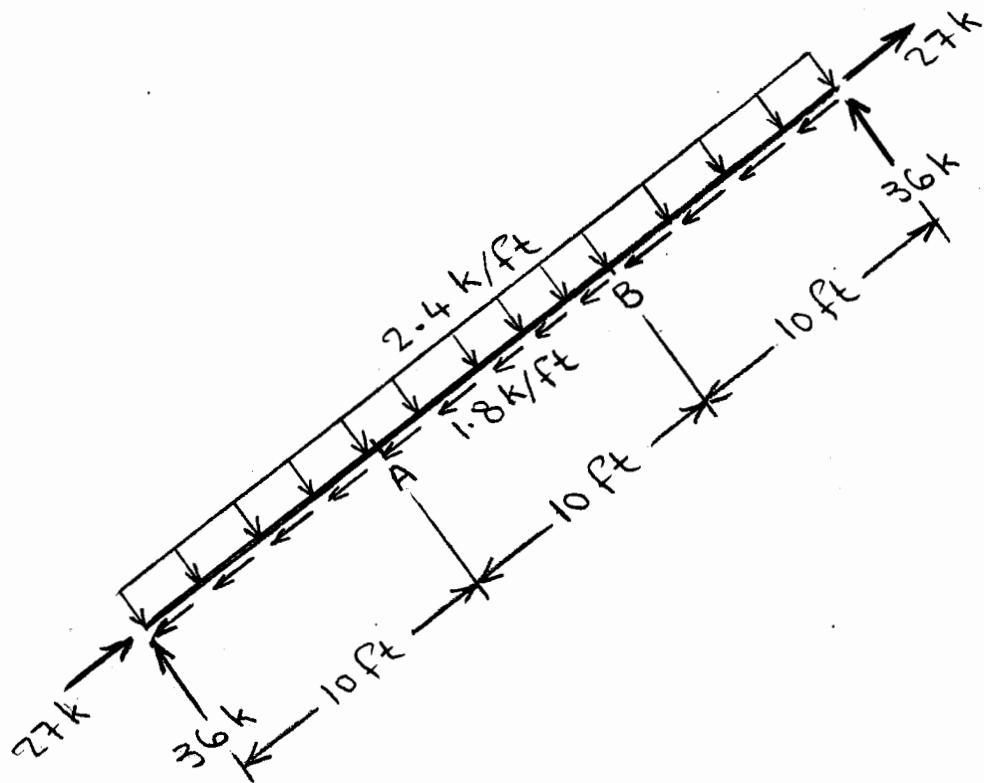
$$\begin{aligned} M_A &= 50(\sin 30^\circ)(9) + 28.18(\cos 30^\circ)(9) - 100(\cos 30^\circ)(3) \\ &= \underline{184.8 \text{ kN.m}} \end{aligned}$$

$$Q_B = 50(\cos 30^\circ) - 100(\sin 30^\circ) + 271.82(\sin 30^\circ) = \underline{129.2 \text{ kN}}$$

$$S_B = -50(\sin 30^\circ) - 100(\cos 30^\circ) + 271.82(\cos 30^\circ) = \underline{123.8 \text{ kN}}$$

$$\begin{aligned} M_B &= -50(\sin 30^\circ)(9) - 100(\cos 30^\circ)(9) + 271.82(\cos 30^\circ)(3) \\ &= \underline{-298.2 \text{ kN.m}} \end{aligned}$$

5.11



$$Q_A = -27 + 1.8(10) = \underline{-9 \text{ k}}$$

$$S_A = 36 - 2.4(10) = \underline{12 \text{ k}}$$

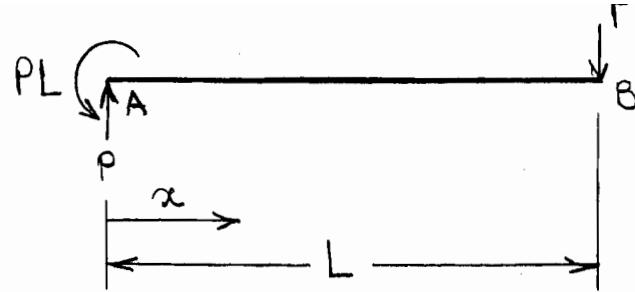
$$M_A = 36(10) - 2.4(10)(5) = \underline{240 \text{ k-ft}}$$

$$Q_B = 27 - 1.8(10) = \underline{9 \text{ k}}$$

$$S_B = -36 + 2.4(10) = \underline{-12 \text{ k}}$$

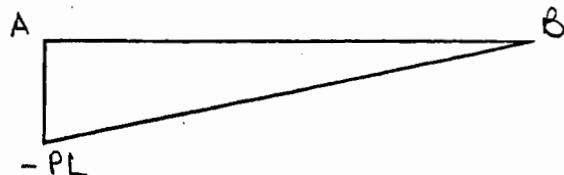
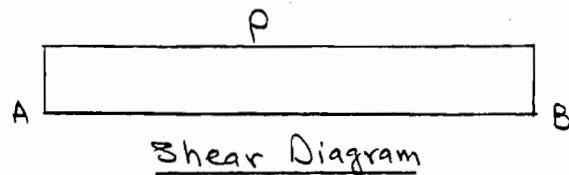
$$M_B = 36(10) - 2.4(10)(5) = \underline{240 \text{ k-ft}}$$

5.12

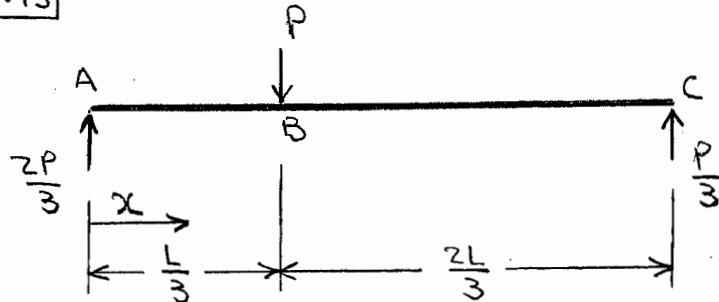


$$S = \frac{P}{x} \quad \text{for } 0 < x < L$$

$$M = -PL + Px = \frac{P(x-L)}{2} \quad \text{for } 0 < x \leq L$$

Bending Moment Diagram

5.13



$$0 < x < \frac{L}{3} :$$

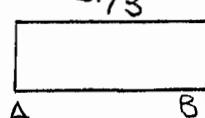
$$S = \frac{2P}{3}$$

$$M = \frac{2P}{3}x$$

$$\frac{L}{3} < x < L :$$

$$S = \frac{2P}{3} - P = -\frac{P}{3}$$

$$M = \frac{2P}{3}x - P(x - \frac{L}{3}) \\ = \frac{P}{3}(L - x)$$

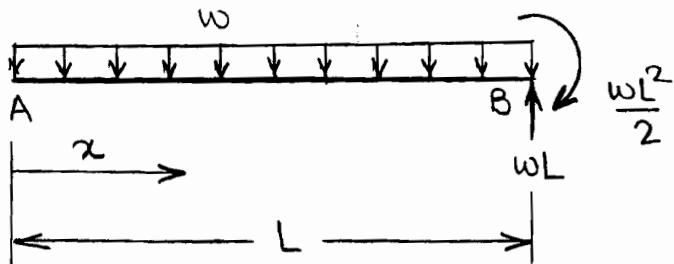
Shear Diagram

$$2PL/9$$

$$-P/3$$

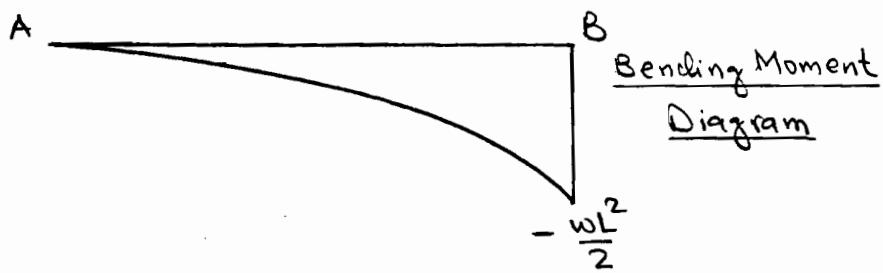
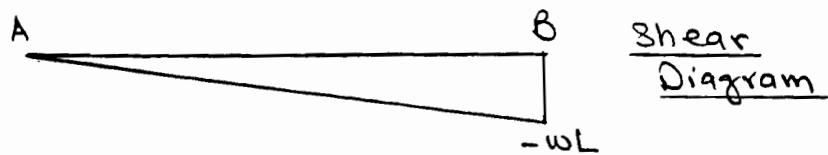
Bending Moment Diagram

5.14

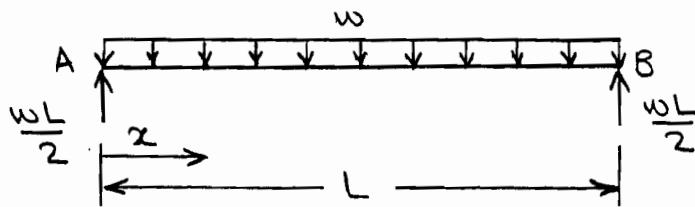


$$S = \frac{-wx}{2} \quad \text{for } 0 \leq x < L$$

$$M = \frac{-wx^2}{2} \quad \text{for } 0 \leq x < L$$

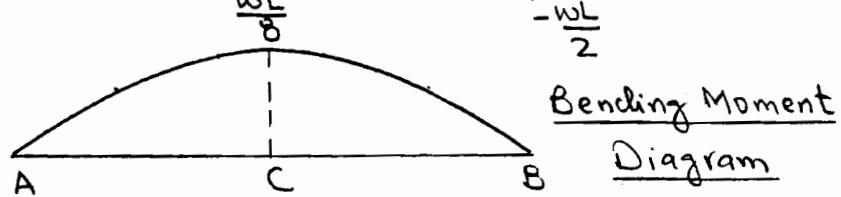
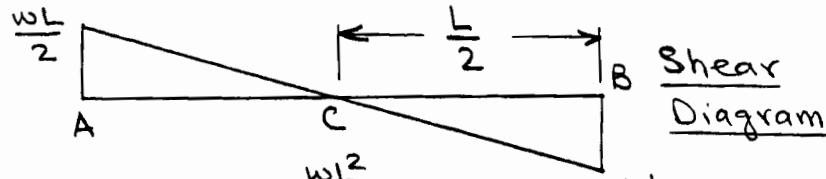


5.15

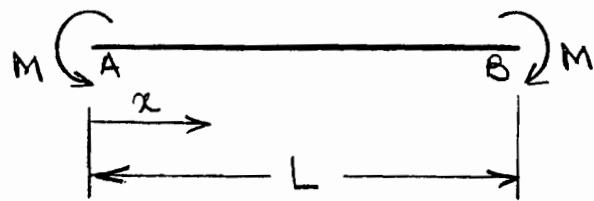


$$S = \frac{wL}{2}x - wx = w\left(\frac{L}{2} - x\right) \quad \text{for } 0 < x < L$$

$$M = \frac{wL}{2}x - \frac{wx^2}{2} = \frac{wx}{2}(L-x) \quad \text{for } 0 \leq x \leq L$$

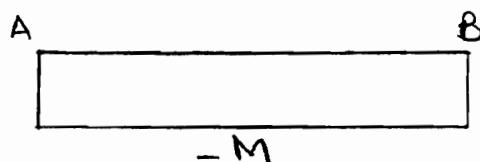


5.16



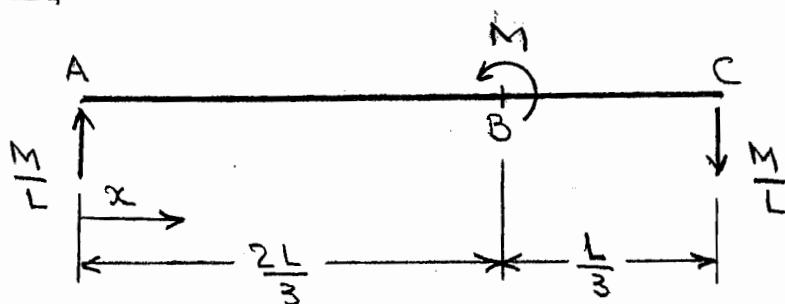
$$\delta = 0$$

$$M_x = -M \quad \text{for } 0 < x < L$$



Bending Moment Diagram

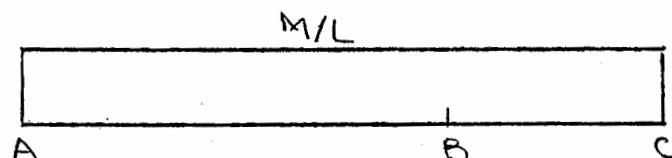
5.17



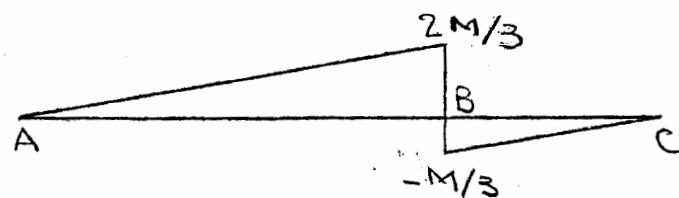
$$\delta = \frac{M}{L}x \quad \text{for } 0 < x < L$$

$$\text{B.M.} = \frac{Mx}{L} \quad \text{for } 0 \leq x \leq \frac{2L}{3}$$

$$\text{B.M.} = \frac{Mx}{L} - M = \frac{M}{L}(x-L) \quad \text{for } \frac{2L}{3} \leq x \leq L$$

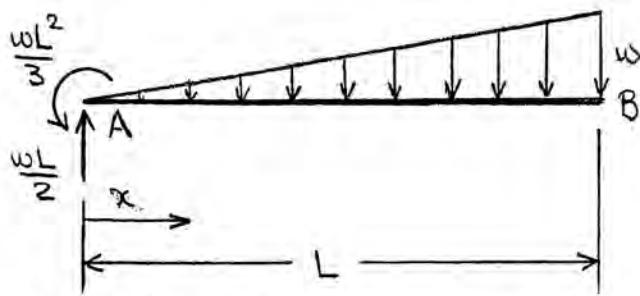


Shear Diagram



Bending Moment Diagram

5.18



$$S = \frac{wL}{2} - \frac{1}{2}(x)\left(\frac{wx}{L}\right) = \frac{wL}{2} \left[1 - \left(\frac{x}{L}\right)^2\right] \quad 0 < x \leq L$$

$$M = -\frac{wL^2}{3} + \left(\frac{wL}{2}\right)x - \left(\frac{wx^2}{2L}\right)\left(\frac{x}{3}\right) = -\frac{wL^2}{3} \left[1 - \frac{3}{2}\left(\frac{x}{L}\right) + \frac{1}{2}\left(\frac{x}{L}\right)^3\right]$$



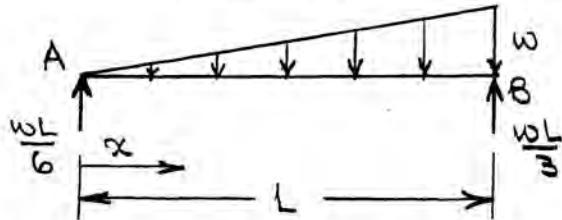
Shear Diagram



Bending Moment Diagram

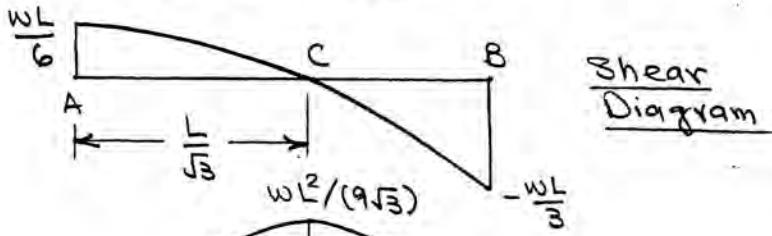


5.19

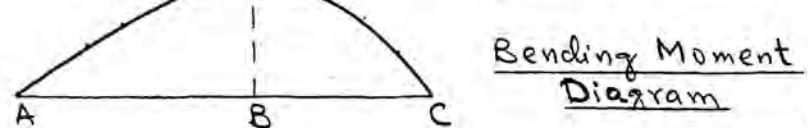


$$S = \frac{wL}{6} - \frac{1}{2}\left(\frac{wx}{L}\right)x = \frac{w}{6L}(L^2 - 3x^2) \quad \text{for } 0 < x < L$$

$$M = \frac{wL}{6}x - \frac{wx^2}{2L}\left(\frac{x}{3}\right) = \frac{wx}{6L}(L^2 - x^2) \quad \text{for } 0 < x < L$$

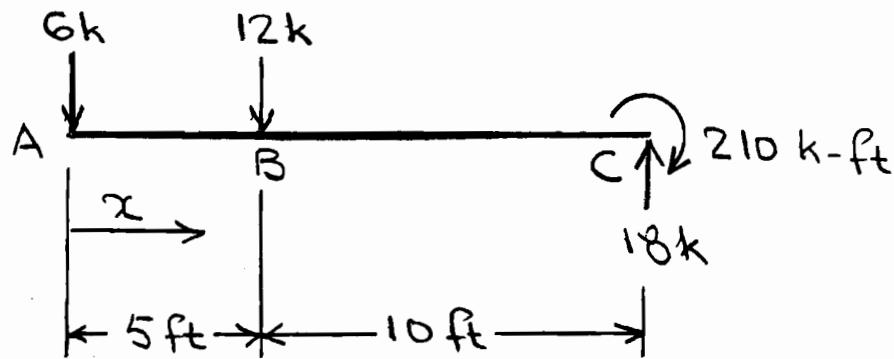


Shear Diagram



Bending Moment Diagram

5.20

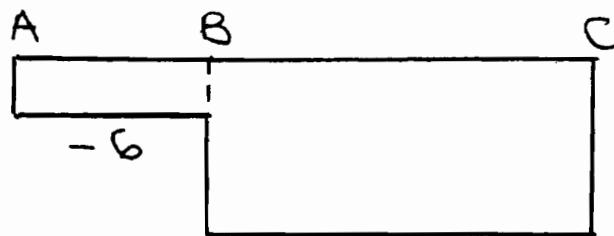


$$0 < x < 5': \quad S = -6$$

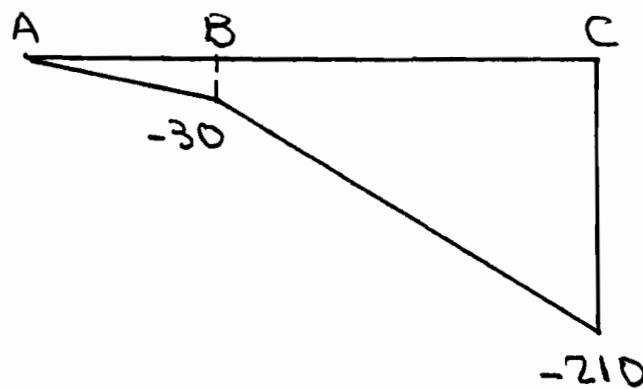
$$M = -6x$$

$$5' < x < 15': \quad S = -6 - 12 = -18$$

$$M = -6x - 12(x-5) = 60 - 18x$$

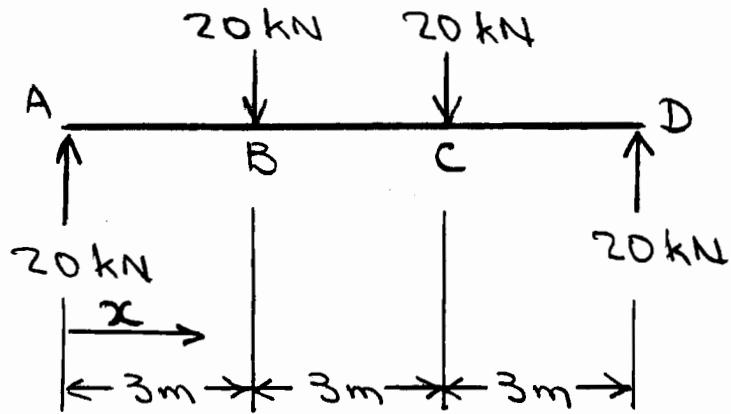


Shear Diagram (k)



Bending Moment Diagram (k-ft)

5-21



$$0 < x < 3 \text{ m:} \quad S = 20$$

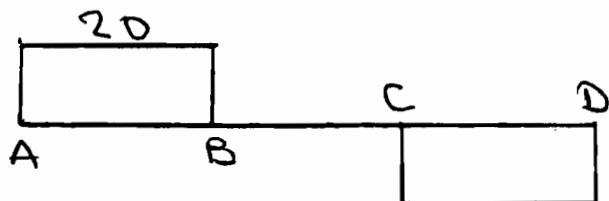
$$M = 20x$$

$$3 \text{ m} < x < 6 \text{ m:} \quad S = 20 - 20 = 0$$

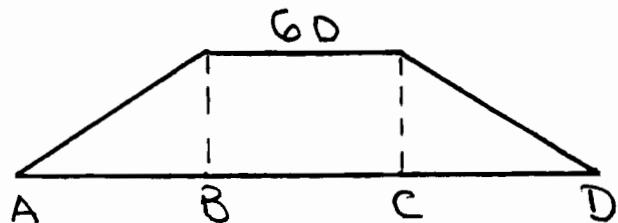
$$M = 20x - 20(x-3) = 60$$

$$6 \text{ m} < x < 9 \text{ m:} \quad S = 20 - 20 - 20 = -20$$

$$M = 20x - 20(x-3) - 20(x-6) \\ = 180 - 20x$$

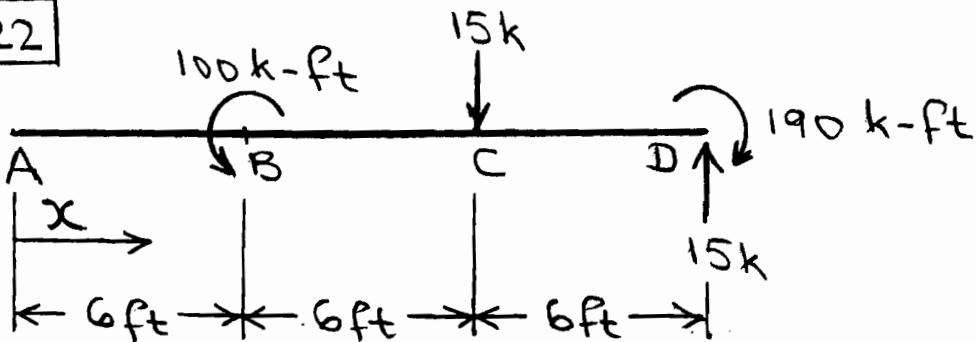


Shear Diagram (kN)



Bending Moment Diagram (kN.m)

5.22



$$0^\circ \leq \alpha < 6': S = 0$$

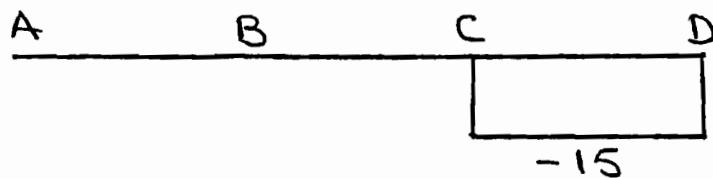
$$M = 0$$

$$6' < \alpha < 12': S = 0$$

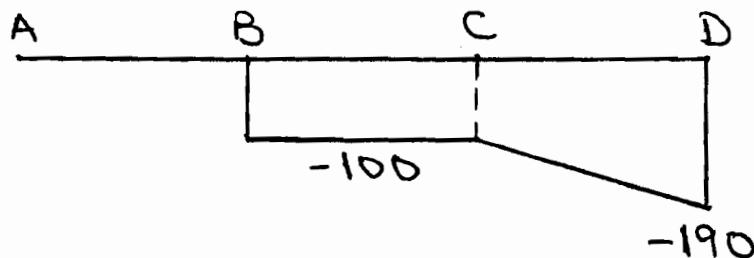
$$M = -100$$

$$12' < \alpha < 18': S = -15$$

$$M = -100 - 15(\alpha - 12) = 80 - 15\alpha$$

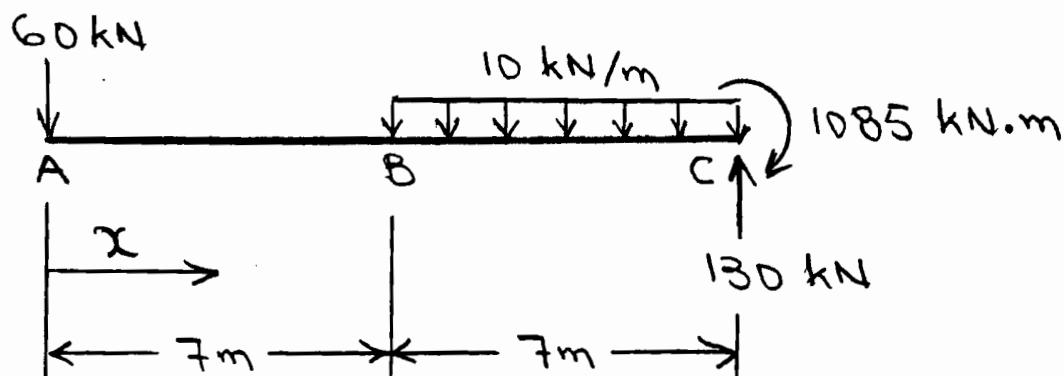


Shear Diagram (k)



Bending Moment Diagram (k-ft)

5.23



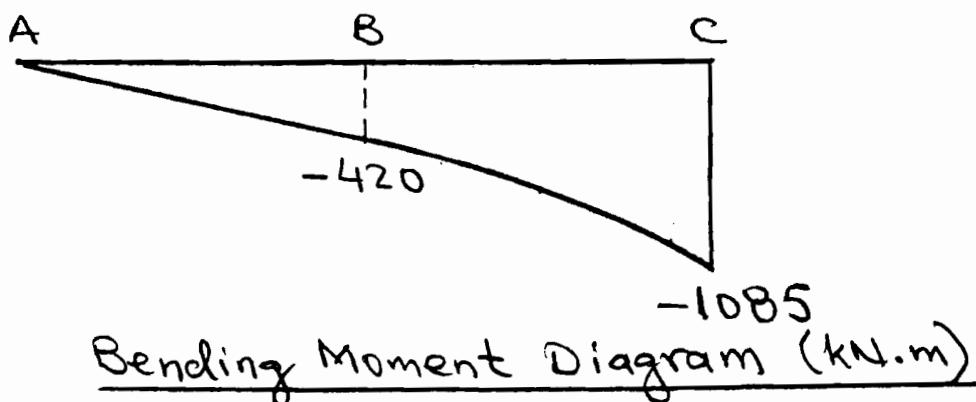
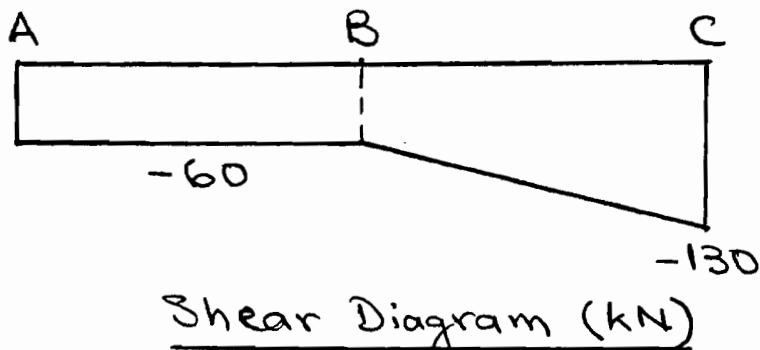
$$0 < x \leq 7\text{m}: \quad S = -60$$

$$M = -60x$$

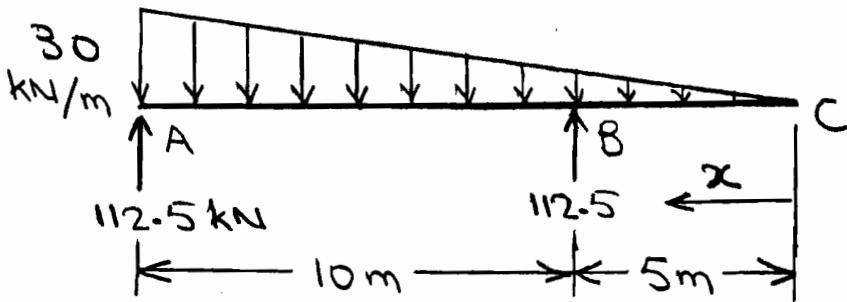
$$7\text{m} \leq x < 14\text{m}: \quad S = -60 - 10(x-7) = 10(1-x)$$

$$M = -60x - \frac{10(x-7)^2}{2}$$

$$= -5x^2 + 10x - 245$$



5.24



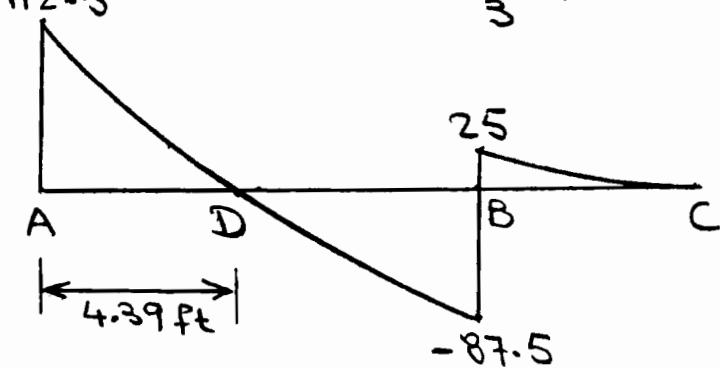
$$0 \leq x < 5 \text{ m}: \quad S = \frac{1}{2}(2x)x = x^2$$

$$M = -\frac{1}{2}(2x)x\left(\frac{x}{3}\right) = -\frac{x^3}{3}$$

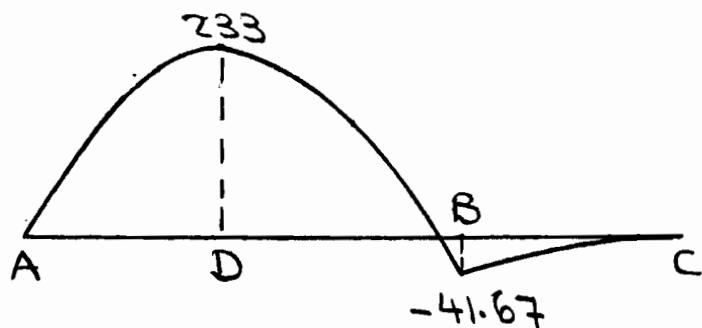
$$5 \text{ m} < x < 10 \text{ m}: \quad S = x^2 - 112.5$$

$$M = -\frac{x^3}{3} + 112.5(x-5)$$

$$= -\frac{x^3}{3} + 112.5x - 562.5$$

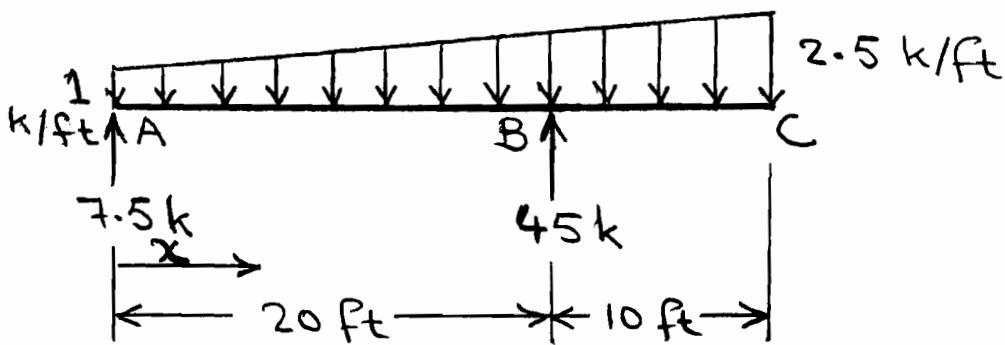


Shear Diagram (kN)



Bending Moment Diagram (kN.m)

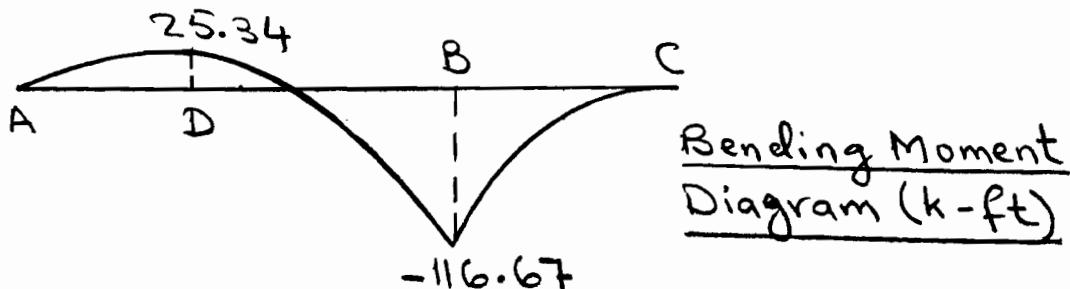
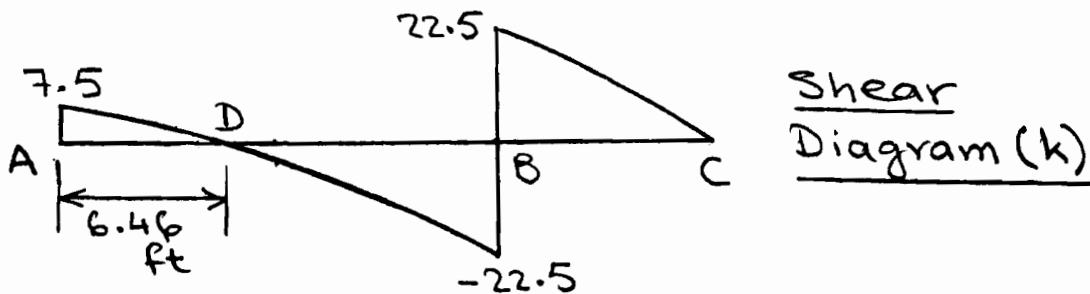
5.25



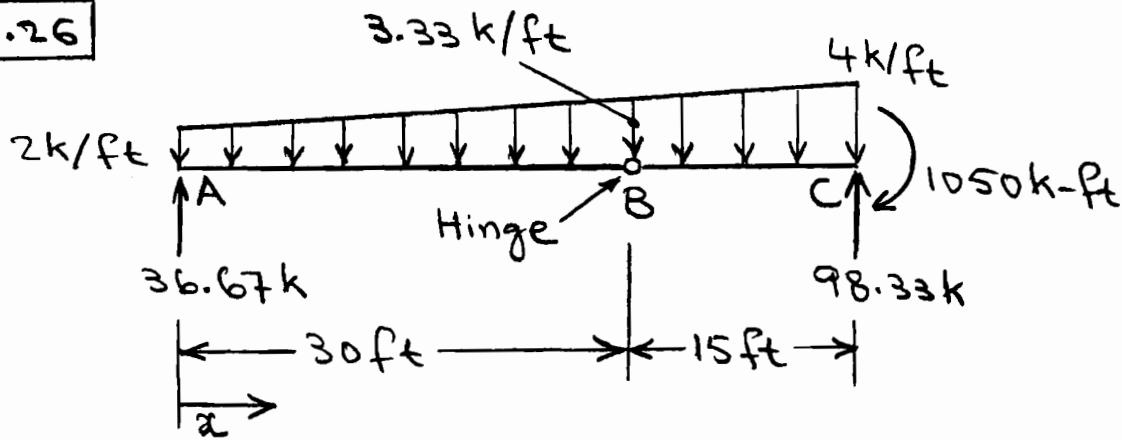
$$0 < x < 20': \quad S = 7.5 - 1(x) - \frac{1}{2} \left(\frac{x}{20} \right) x \\ = -\frac{x^2}{40} - x + 7.5$$

$$M = 7.5x - 1 \left(\frac{x^2}{2} \right) - \frac{1}{2} \left(\frac{x}{20} \right) x \left(\frac{x}{3} \right) \\ = -\frac{x^3}{120} - \frac{x^2}{2} + 7.5x$$

$$20' < x < 30': \quad S = 7.5 - 1(x) - \frac{1}{2} \left(\frac{x}{20} \right) x + 45 \\ = -\frac{x^2}{40} - x + 52.5 \\ M = 7.5x - 1 \left(\frac{x^2}{2} \right) - \frac{x^3}{120} + 45(x-20) \\ = -\frac{x^3}{120} - \frac{x^2}{2} + 52.5x - 900$$



5.26



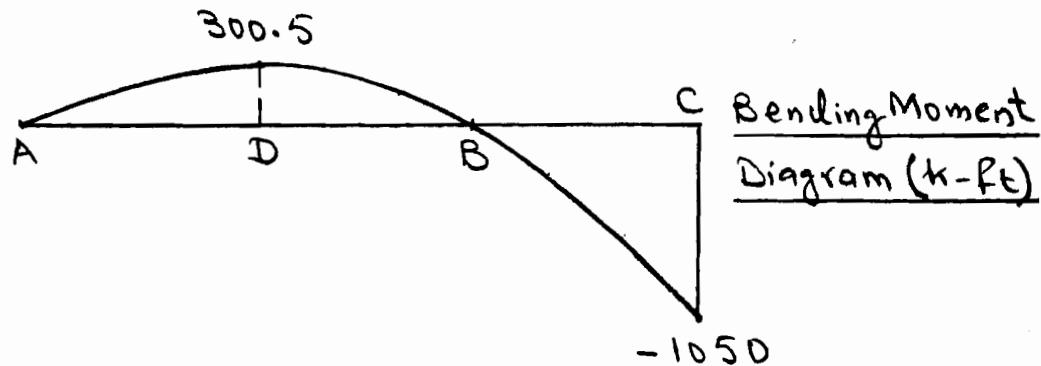
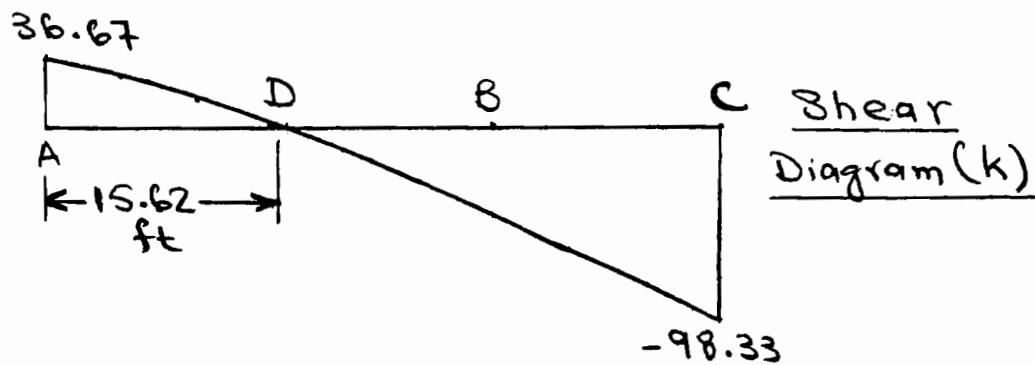
$$0 \leq x \leq 45':$$

$$S = 36.67 - 2x - \frac{1}{2} \left(\frac{2}{45} x \right) x$$

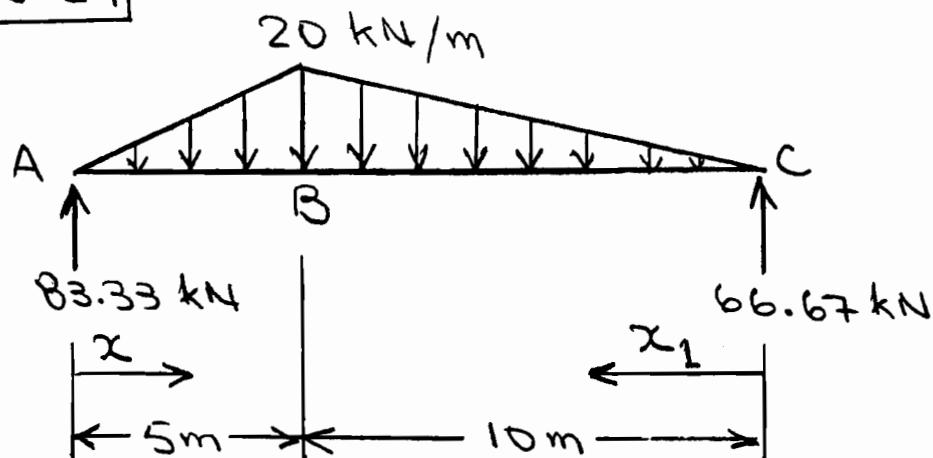
$$= 36.67 - 2x - \frac{x^2}{45}$$

$$M = 36.67x - 2\left(\frac{x}{2}\right)x - \frac{1}{2} \left(\frac{2}{45} x \right) x \left(\frac{x}{3} \right)$$

$$= 36.67x - x^2 - \frac{x^3}{135}$$



5.27



$0 < x \leq 5\text{m}:$

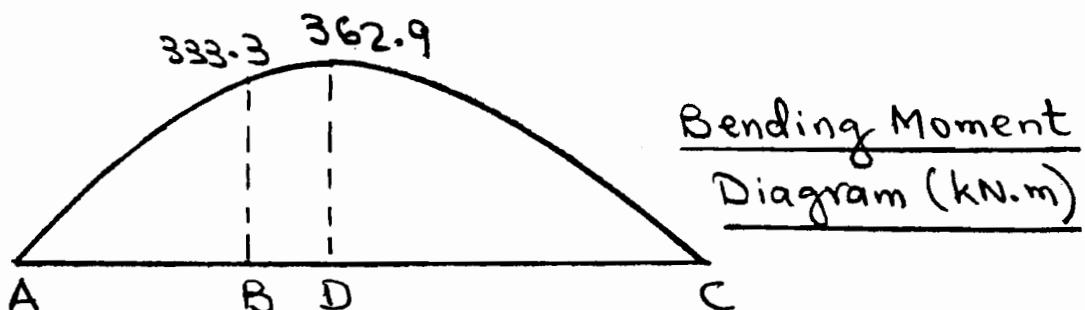
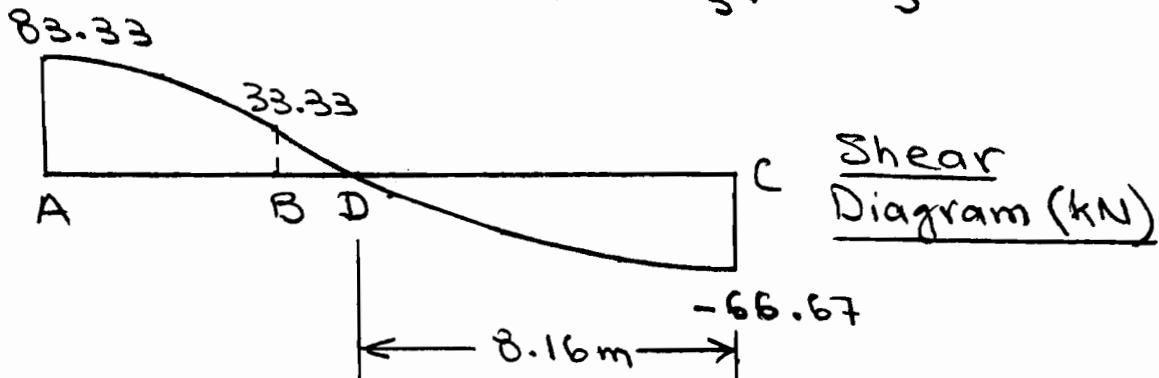
$$S = 83.33 - \frac{1}{2} \left(\frac{20x}{5} \right) x = -2x^2 + 83.33$$

$$M = 83.33x - \frac{1}{2} \left(\frac{20x}{5} \right) x \left(\frac{x}{3} \right) = -\frac{2x^3}{3} + 83.33x.$$

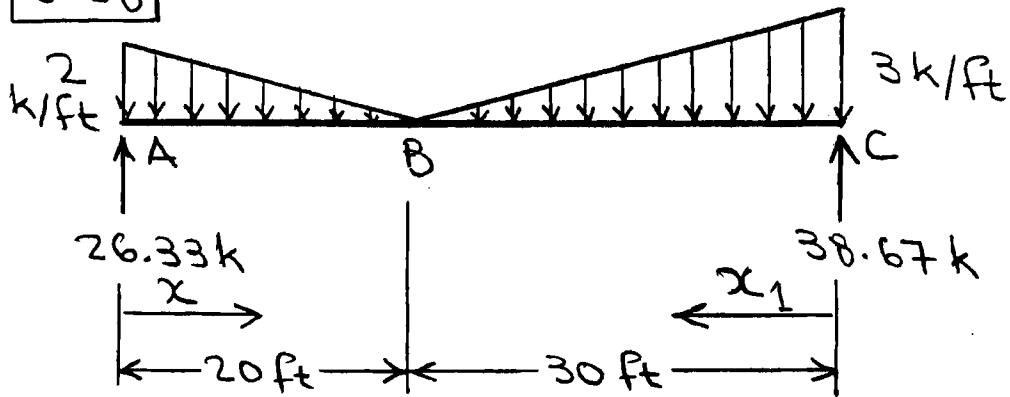
$0 < x_1 \leq 10\text{m}:$

$$S = -66.67 + \frac{1}{2} (2x_1)x_1 = x_1^2 - 66.67$$

$$M = 66.67x_1 - \frac{1}{2} (2x_1)x_1 \left(\frac{x_1}{3} \right) = -\frac{x_1^3}{3} + 66.67x_1$$



5.28



$0 < x \leq 20'$:

$$S = 26.33 - 2x + \frac{1}{2} \left(\frac{x}{10} \right) x = \frac{x^2}{20} - 2x + 26.33$$

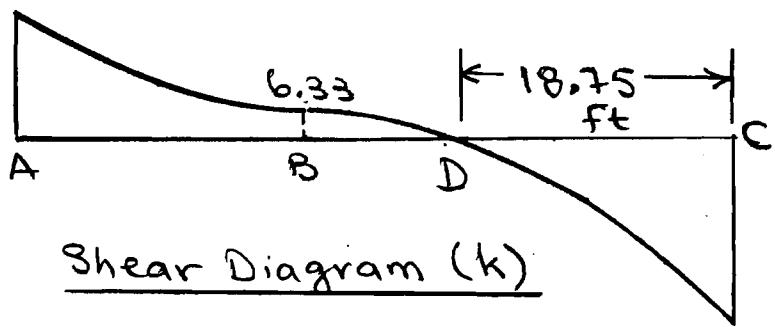
$$M = 26.33x - x^2 + \frac{1}{2} \left(\frac{x}{10} \right) x \left(\frac{x}{3} \right) = \frac{x^3}{60} - x^2 + 26.33x$$

$0 < x_1 \leq 30'$:

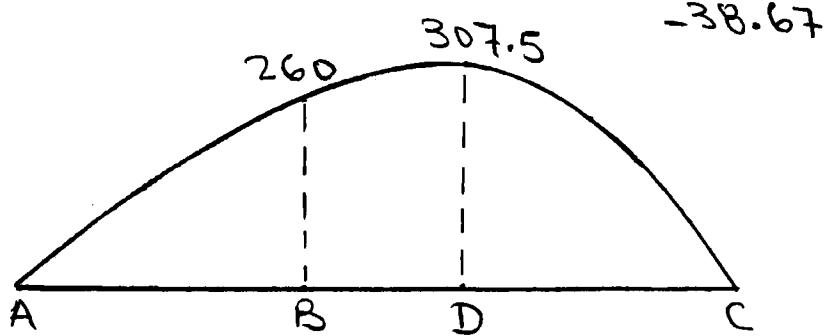
$$S = -38.67 + 3x_1 - \frac{1}{2} \left(\frac{x_1}{10} \right) x_1 = -\frac{x_1^2}{20} + 3x_1 - 38.67$$

$$\begin{aligned} M &= 38.67x_1 - \frac{3x_1^2}{2} + \frac{1}{2} \left(\frac{x_1}{10} \right) x_1 \left(\frac{x_1}{3} \right) \\ &= \frac{x_1^3}{60} - \frac{3x_1^2}{2} + 38.67x_1 \end{aligned}$$

26.33

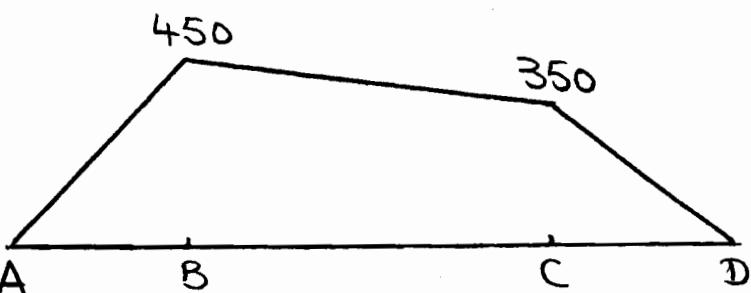
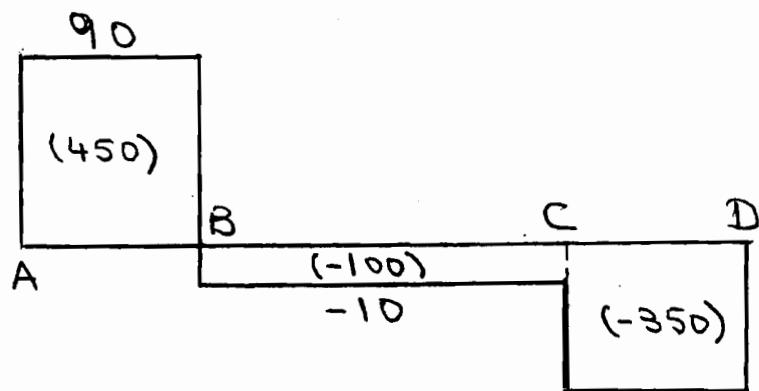
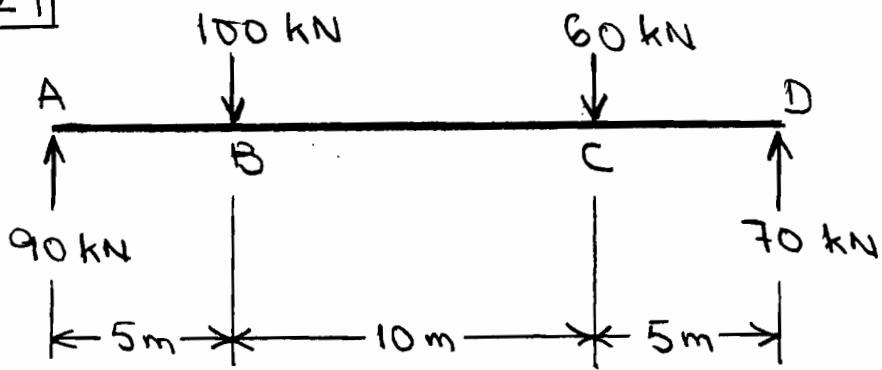


Shear Diagram (k)

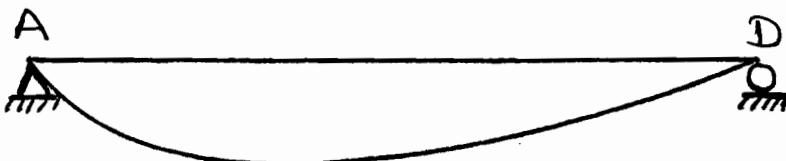


Bending Moment Diagram (k-ft)

5.29

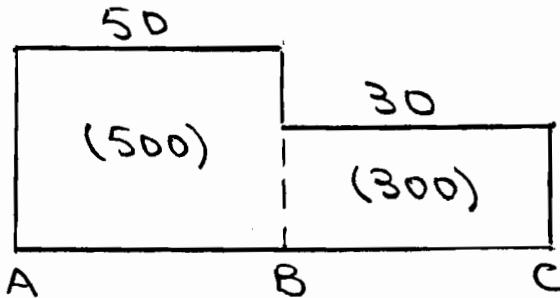
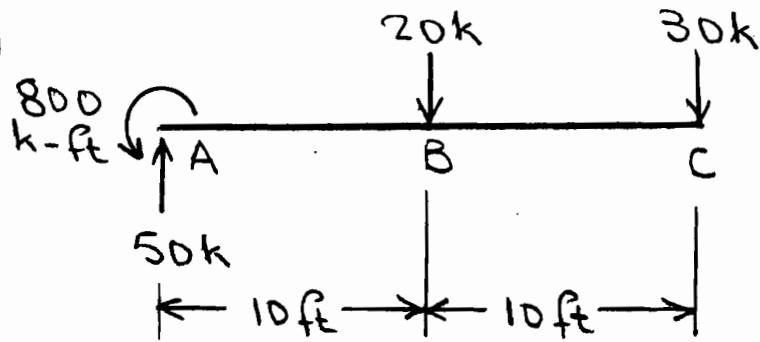


Bending Moment Diagram (kN.m)

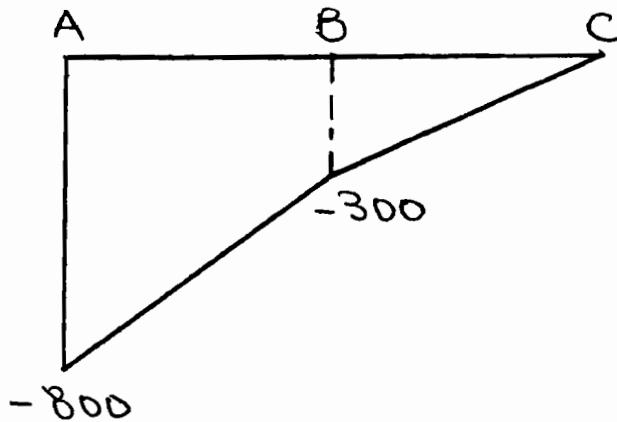


Qualitative Deflected Shape

5.30



Shear Diagram (k)

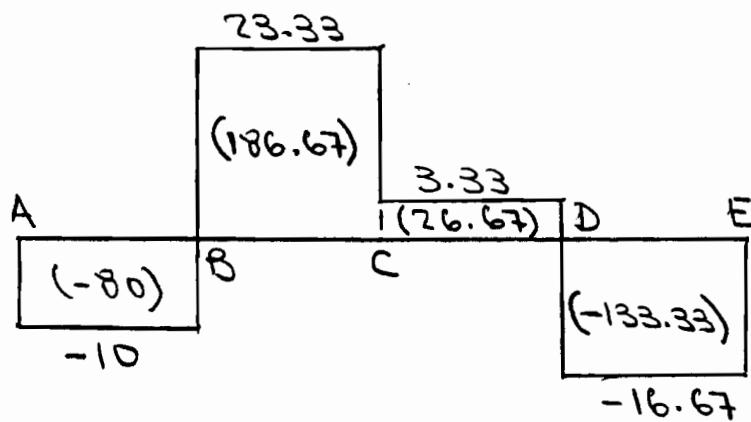
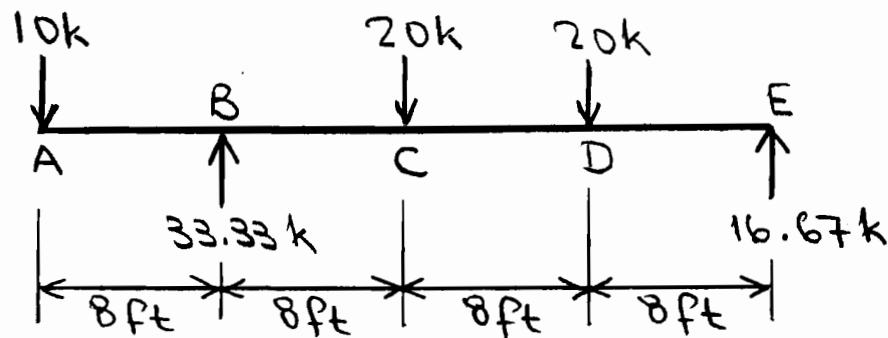


Bending Moment Diagram (k-ft)

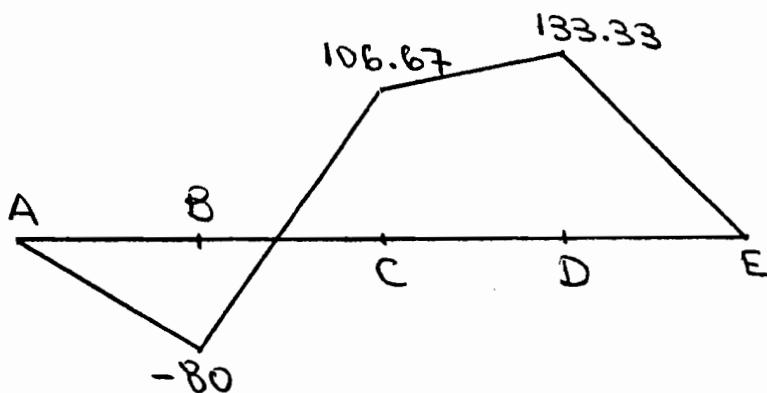


Qualitative Deflected Shape

5.31



Shear Diagram (k)

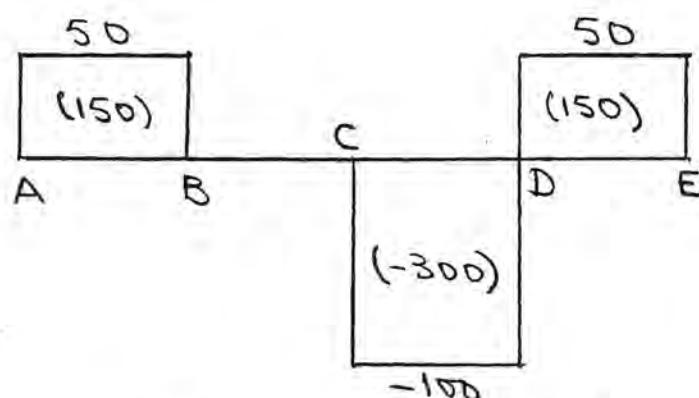
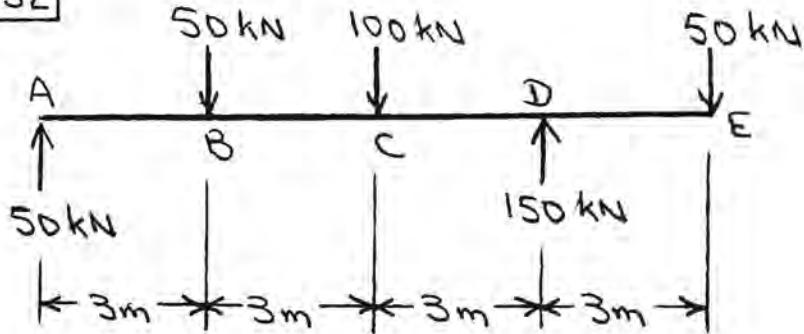


Bending Moment Diagram (k-ft)

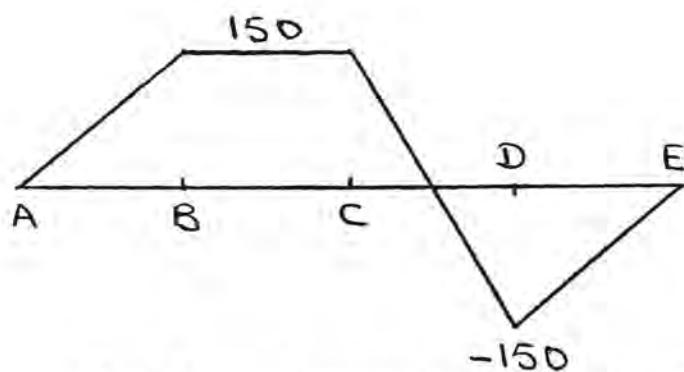


Qualitative Deflected Shape

5.32



Shear Diagram (kN)

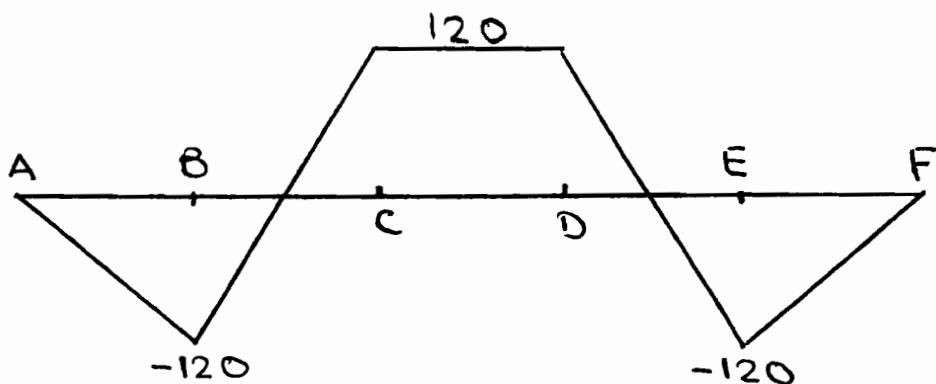
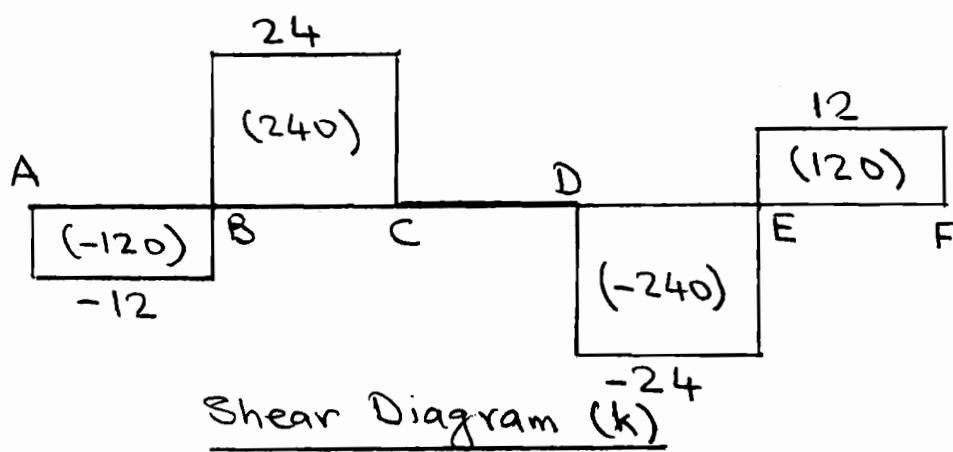
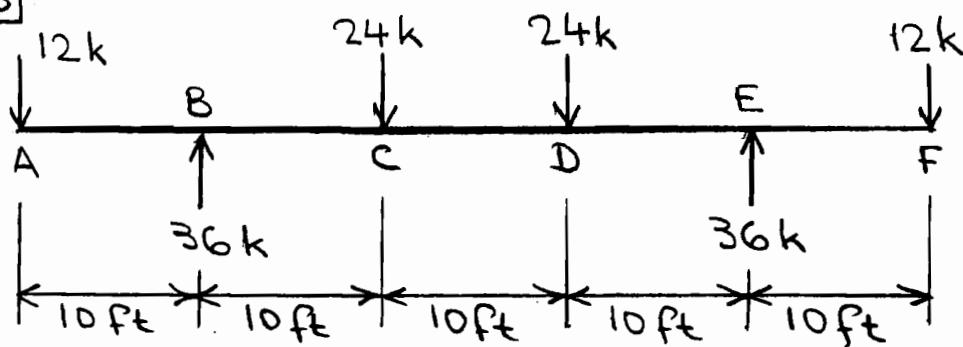
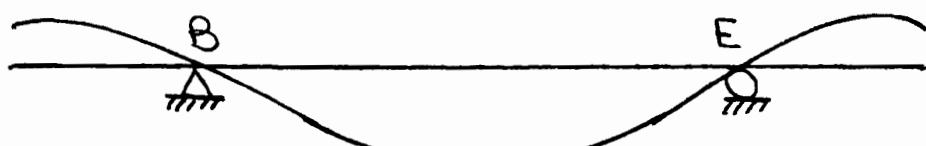


Bending Moment Diagram (kN.m)

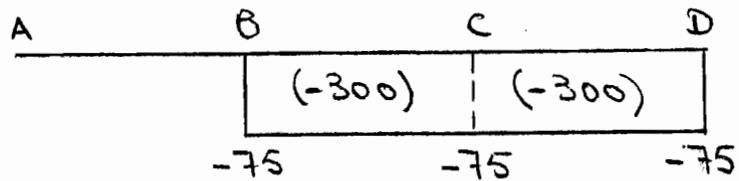
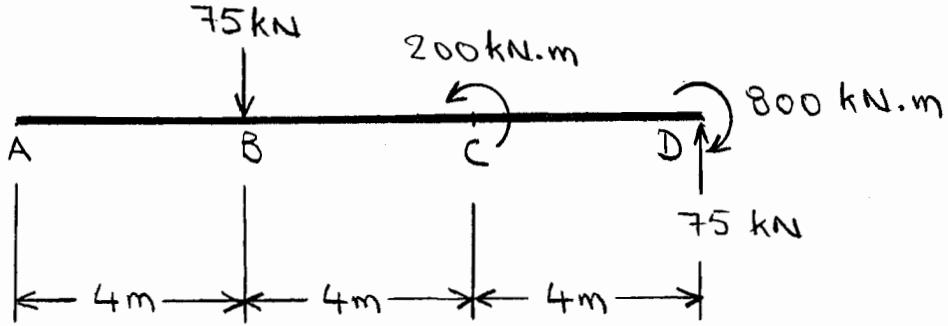


Qualitative Deflected Shape

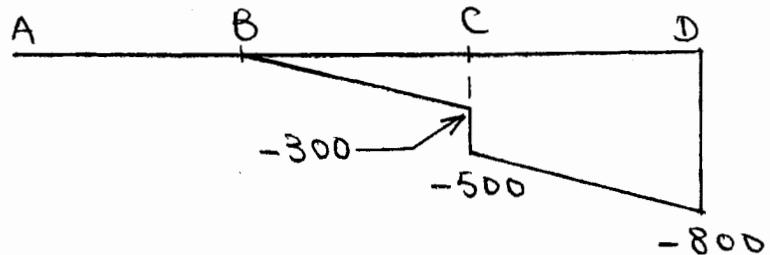
5.33

Bending Moment Diagram (k-ft)Qualitative Deflected Shape

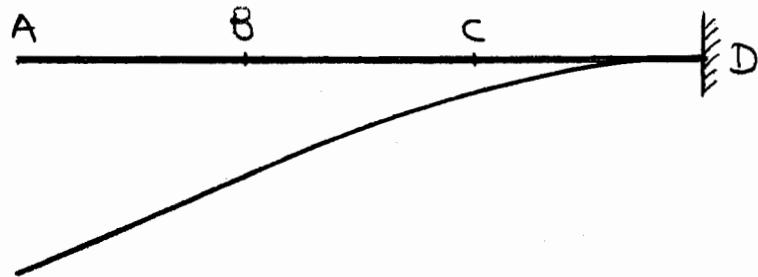
5.34



Shear Diagram (kN)

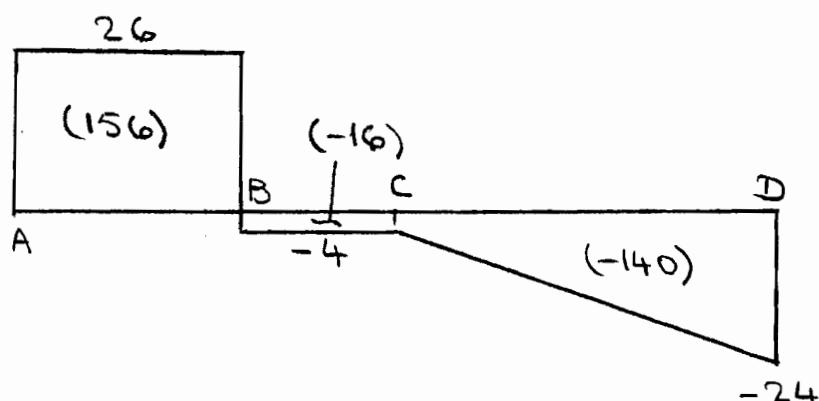
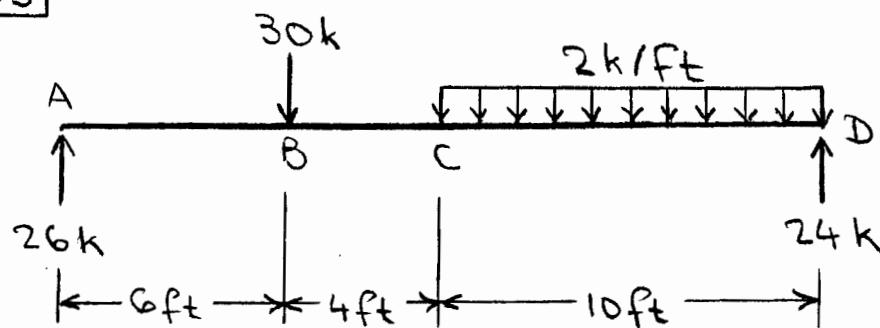


Bending Moment Diagram (kN.m)

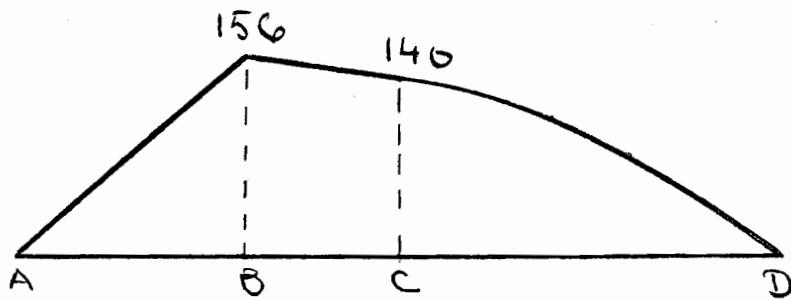


Qualitative Deflected Shape

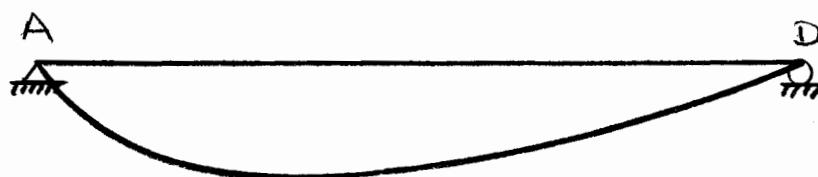
5.35



Shear Diagram (k)

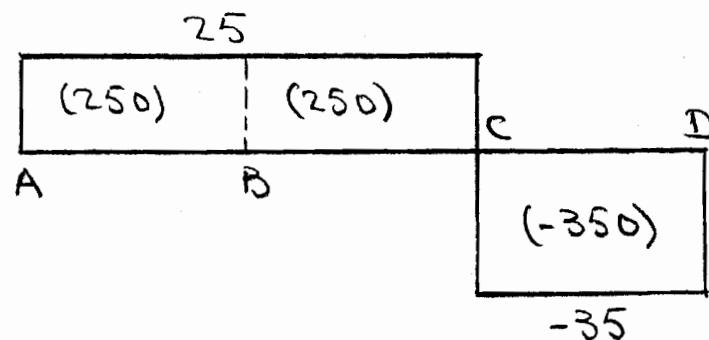
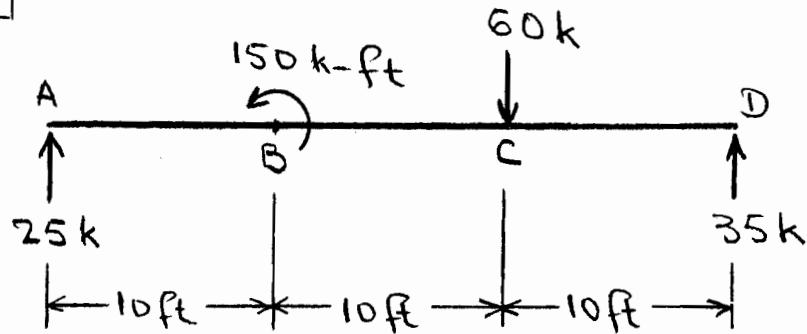


Bending Moment Diagram (k-ft)

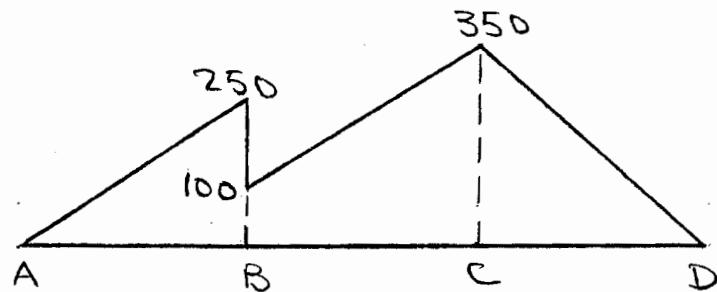


Qualitative Deflected Shape

5.36



Shear Diagram (k)

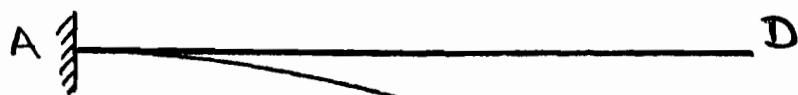
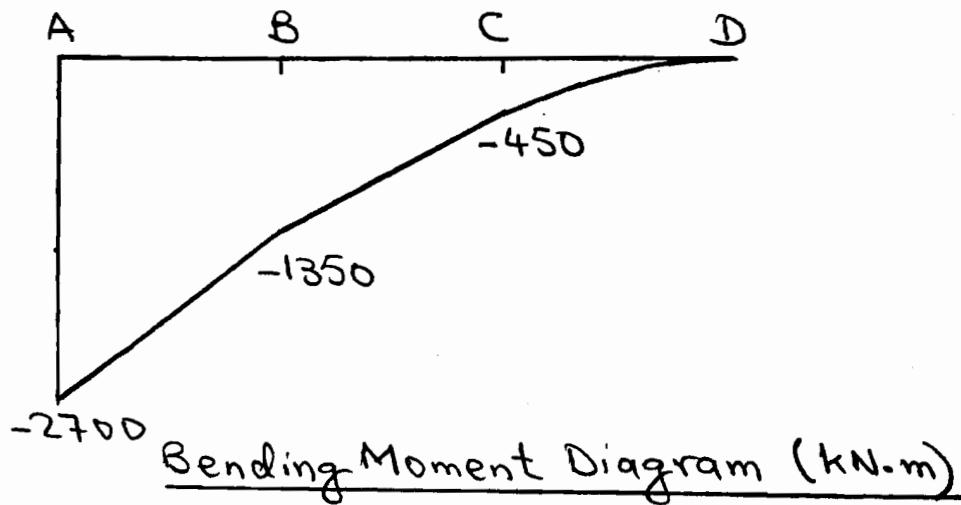
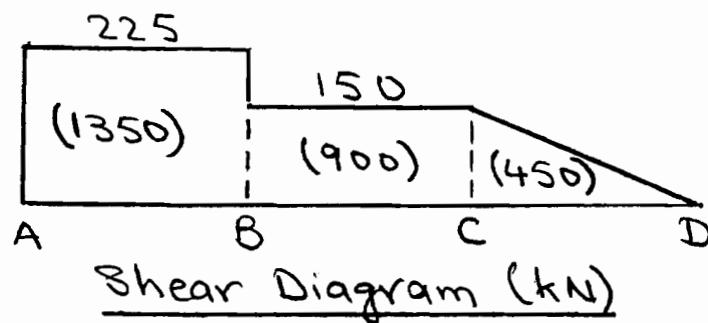
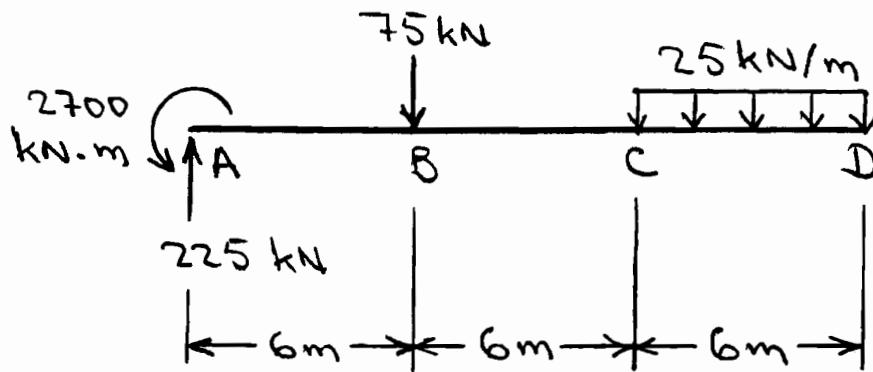


Bending Moment Diagram (k-ft)



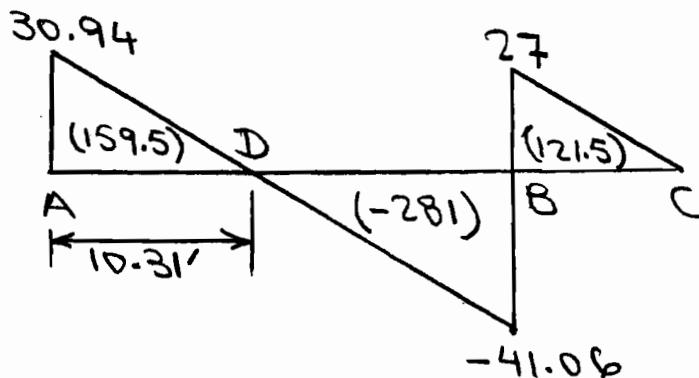
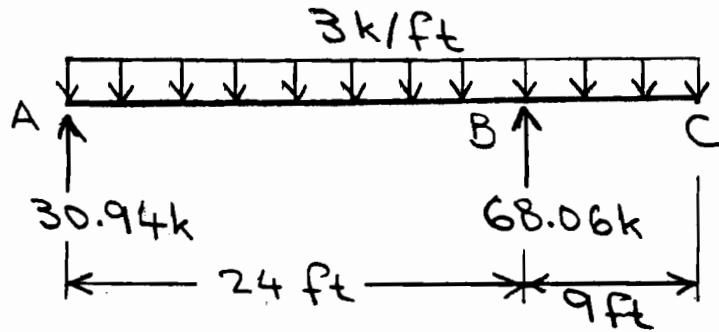
Qualitative Deflected Shape

5.37

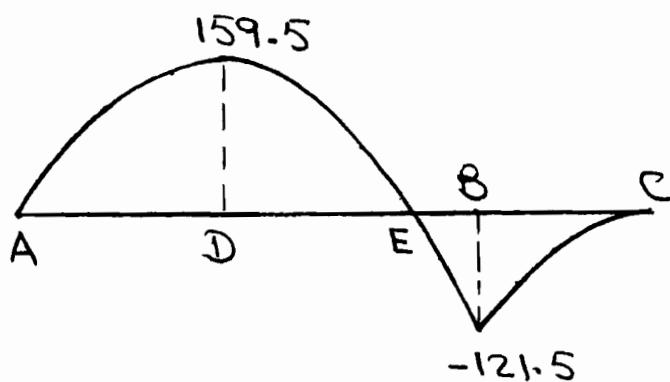


Qualitative Deflected Shape

5.38



Shear Diagram (k)

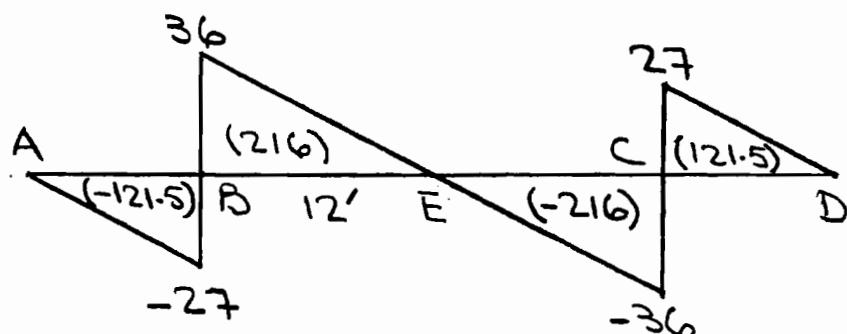
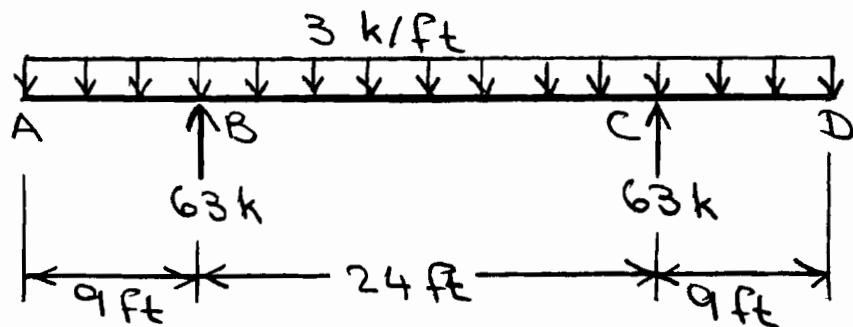


Bending Moment Diagram (k-ft)

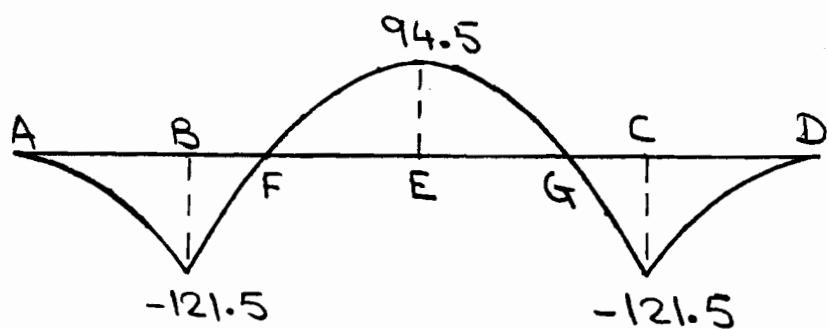


Qualitative Deflected Shape

5.39



Shear Diagram (k)

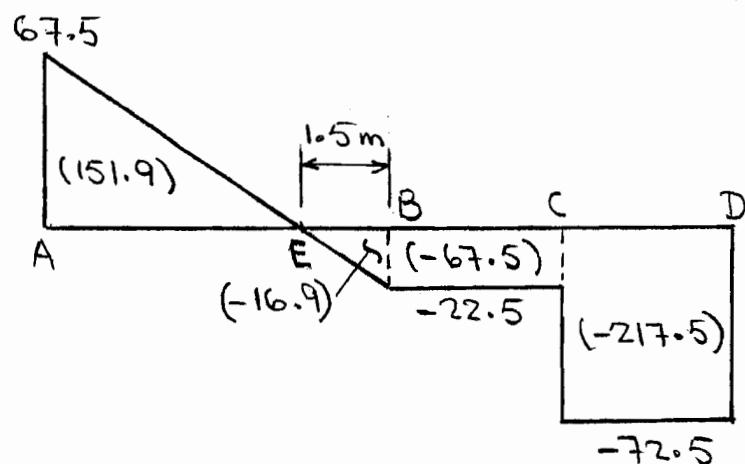
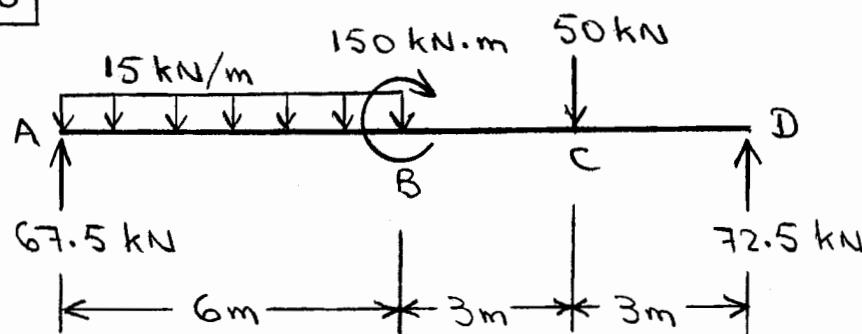


Bending Moment Diagram (k-ft)

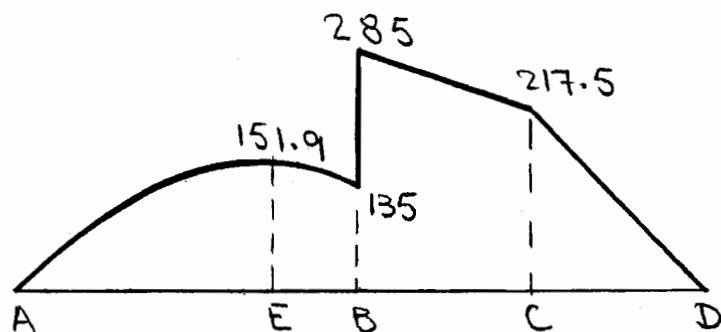


Qualitative Deflected Shape

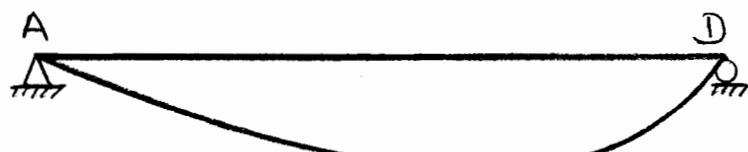
5.40



Shear Diagram (kN)

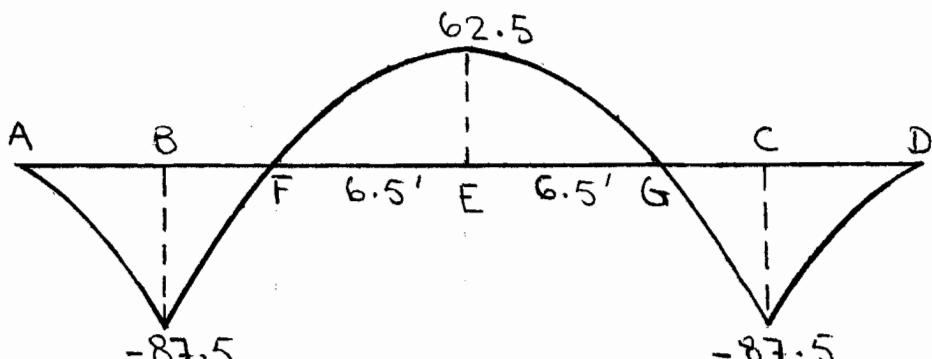
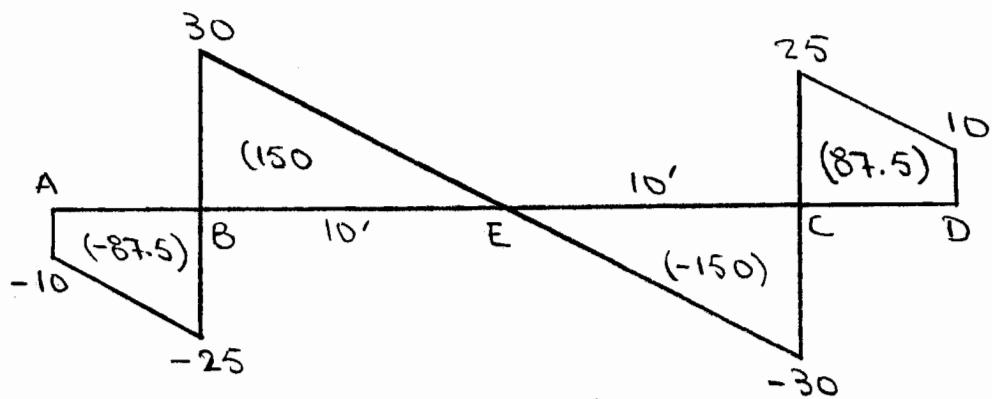
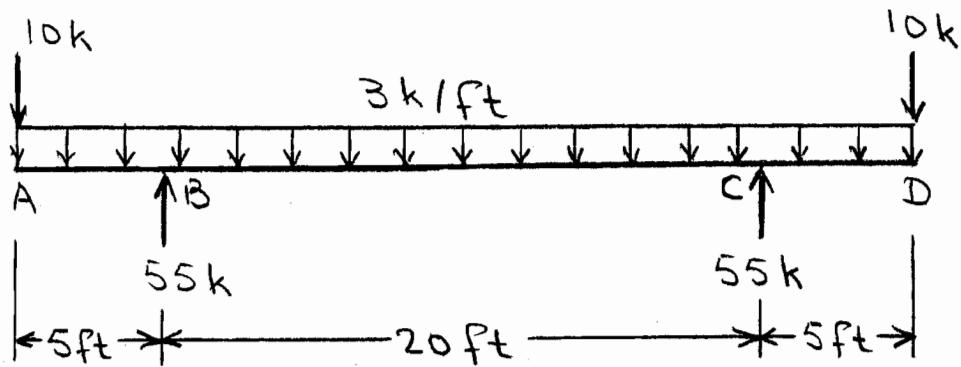


Bending Moment Diagram (kN.m)



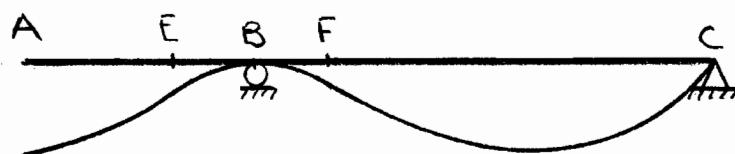
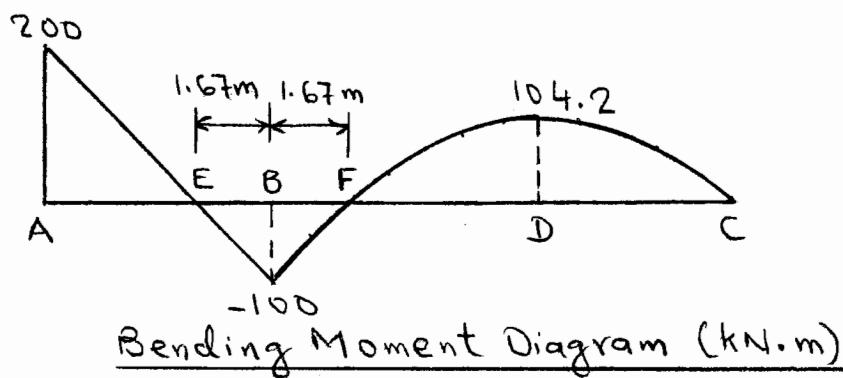
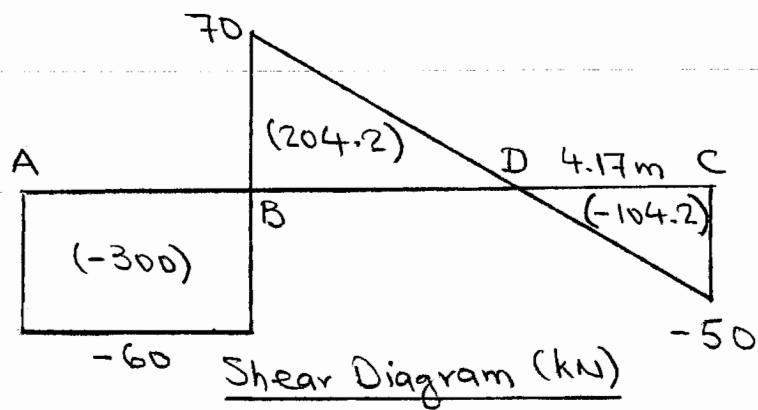
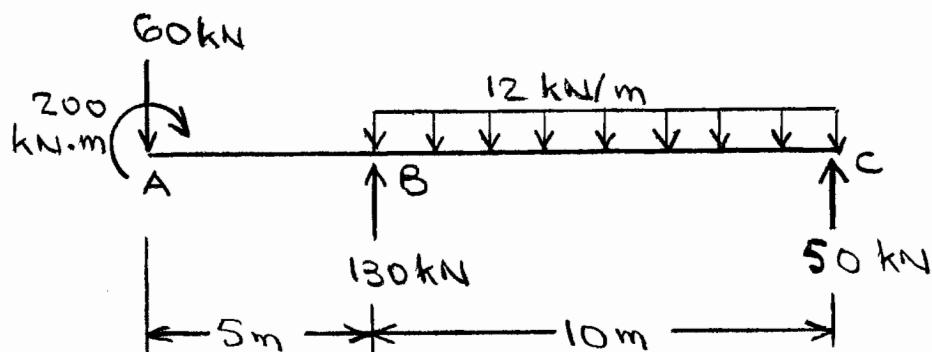
Qualitative Deflected Shape

5.41



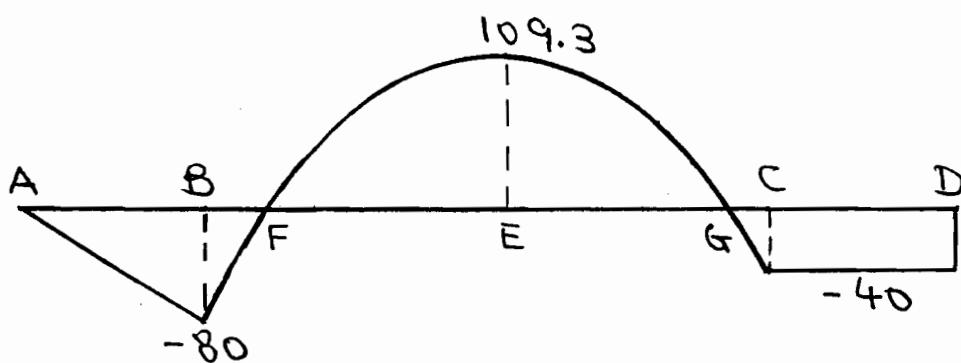
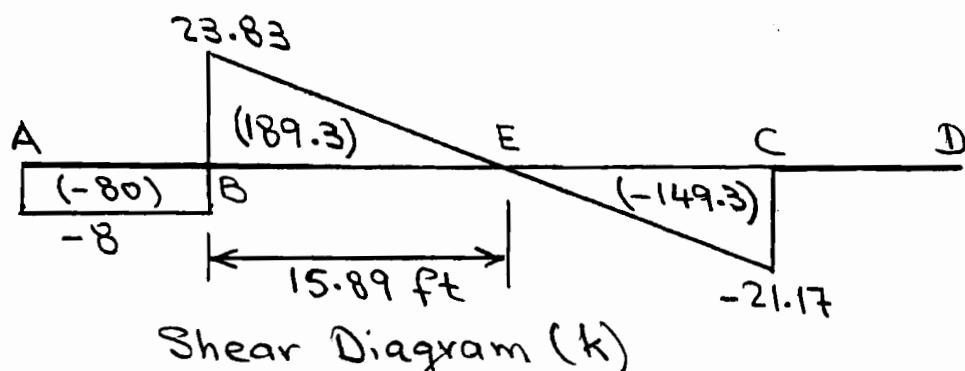
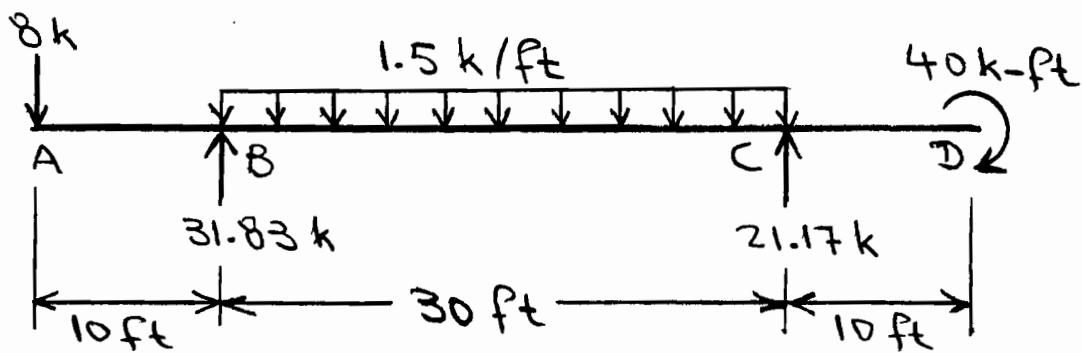
Qualitative Deflected Shape

5.42

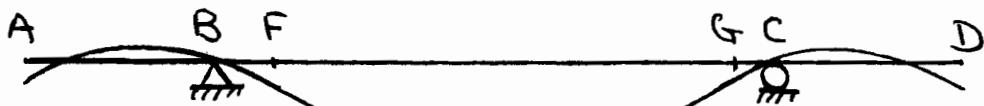


Qualitative Deflected Shape

5.43

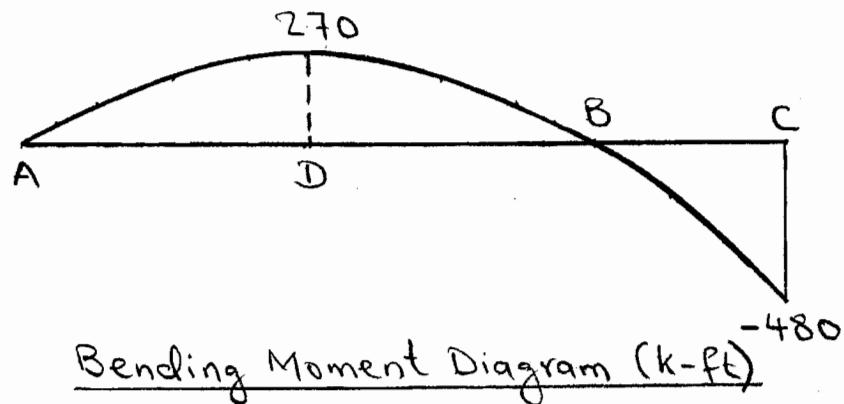
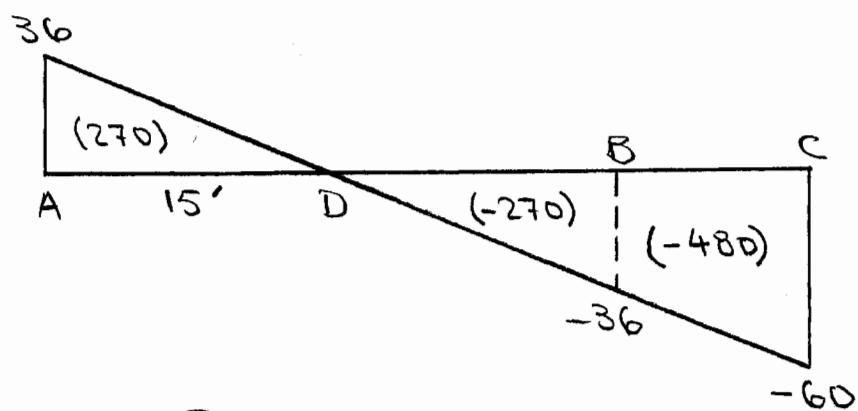
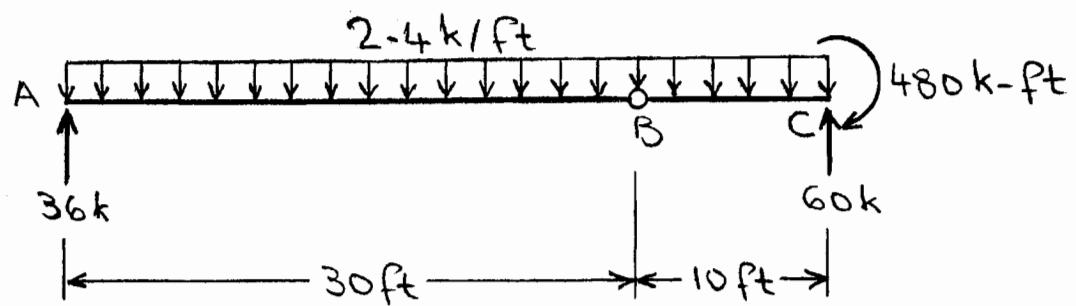


Bending Moment Diagram (k-ft)



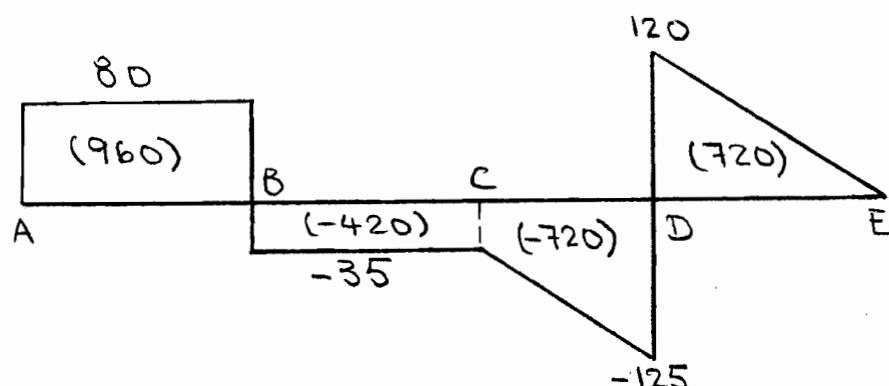
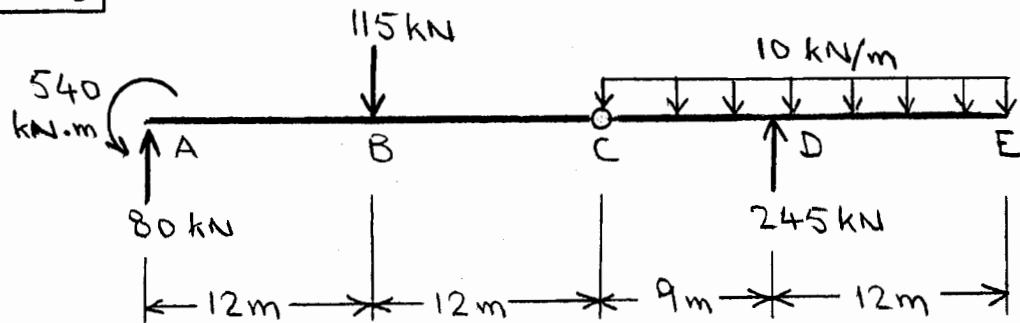
Qualitative Deflected Shape

5.44

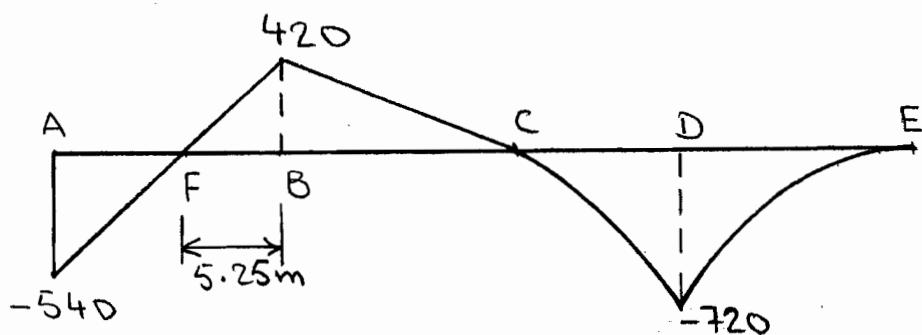


Qualitative Deflected Shape

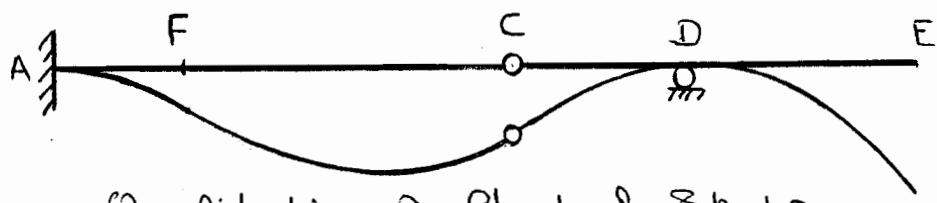
5.45



Shear Diagram (kN)

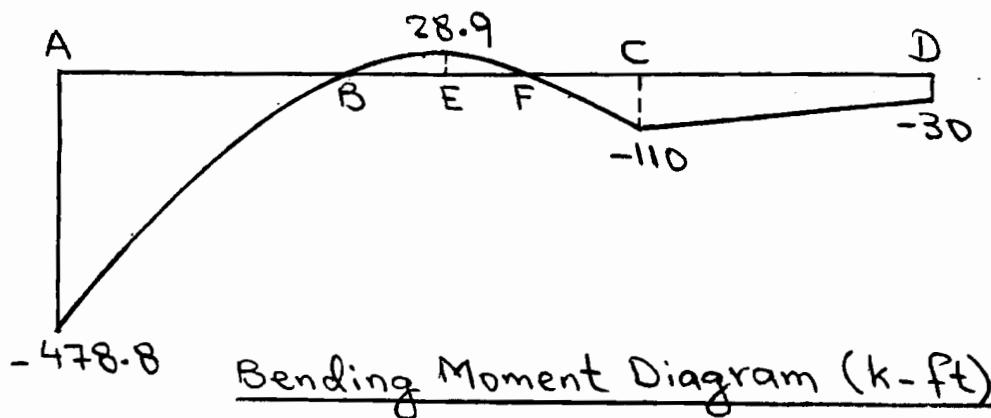
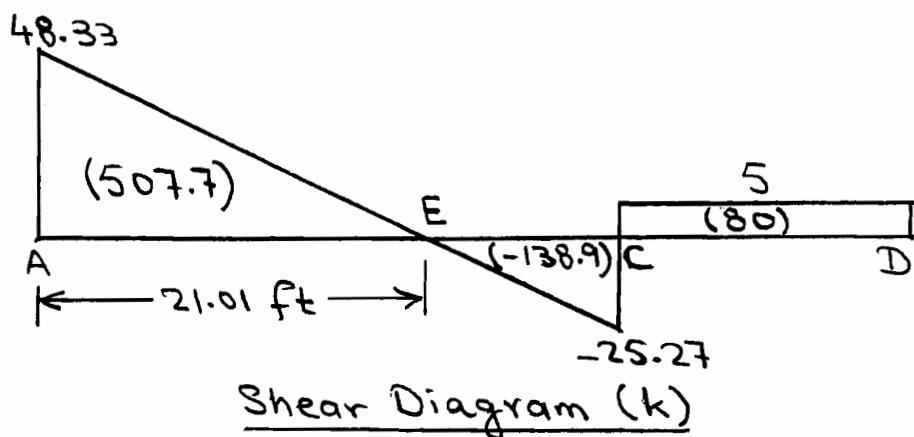
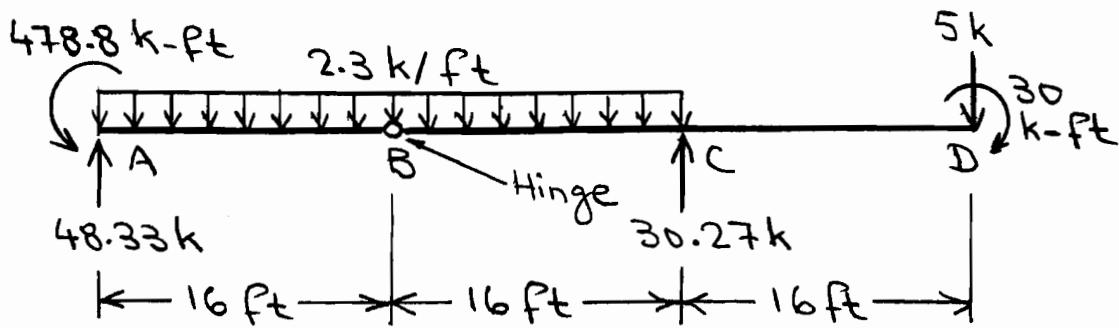


Bending Moment Diagram (kN.m)

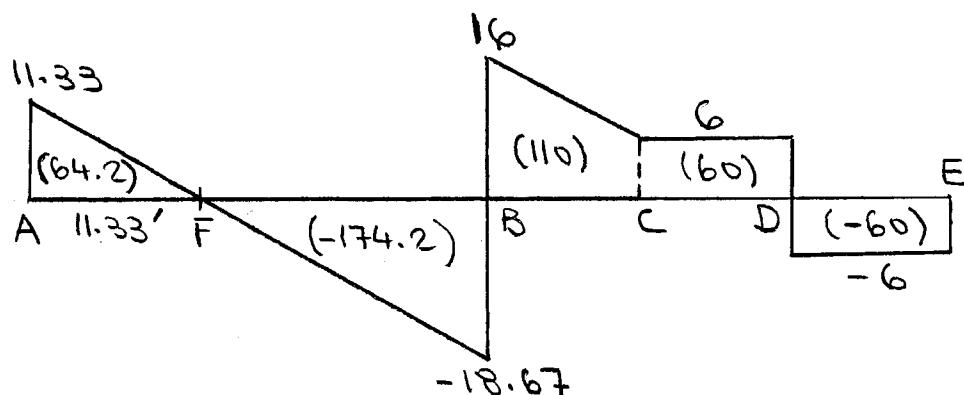
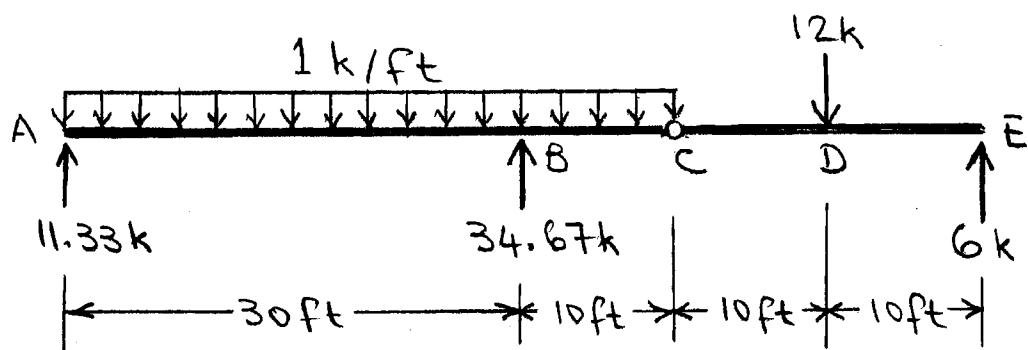


Qualitative Deflected Shape

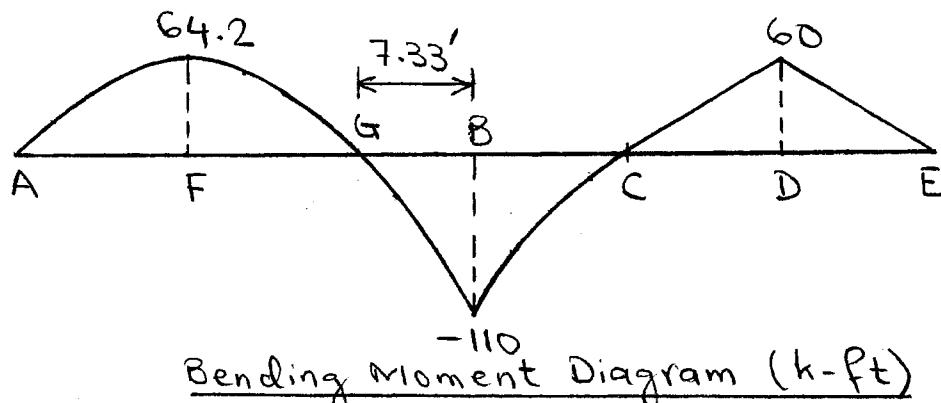
5.46



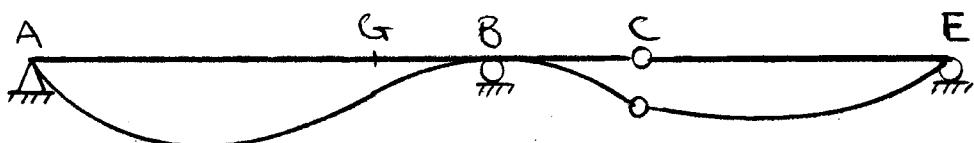
5.47



Shear Diagram (k)

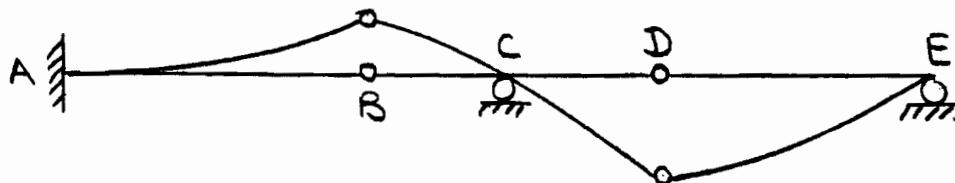
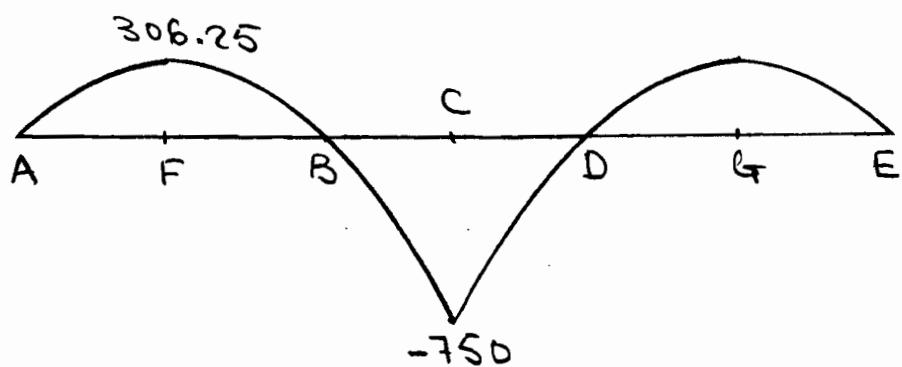
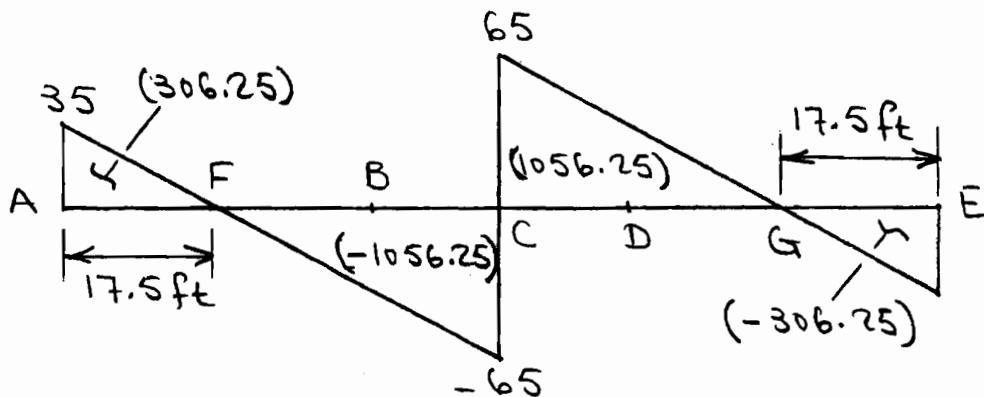
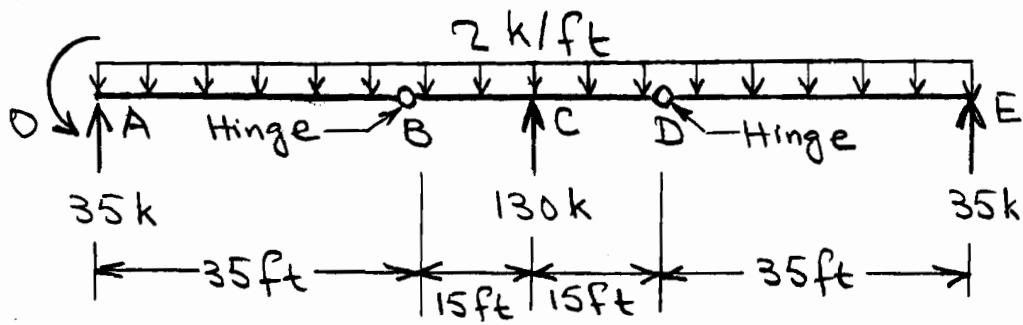


Bending Moment Diagram (k-ft)

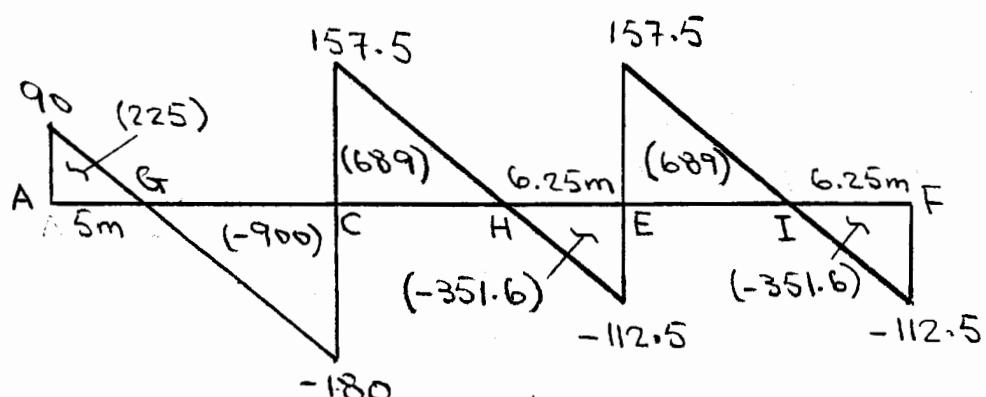
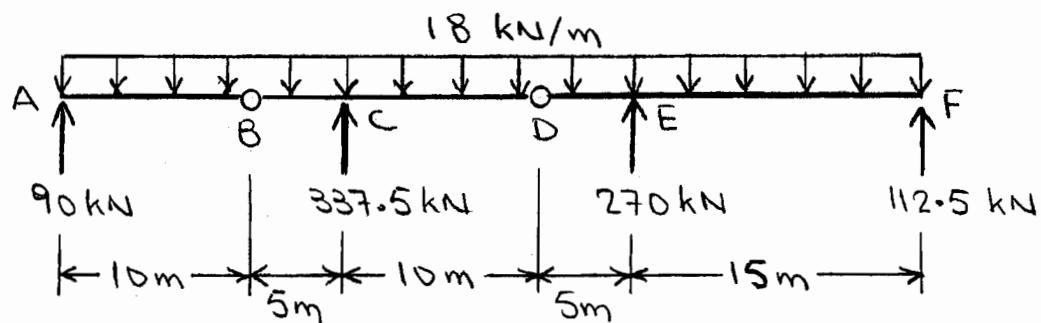


Qualitative Deflected Shape

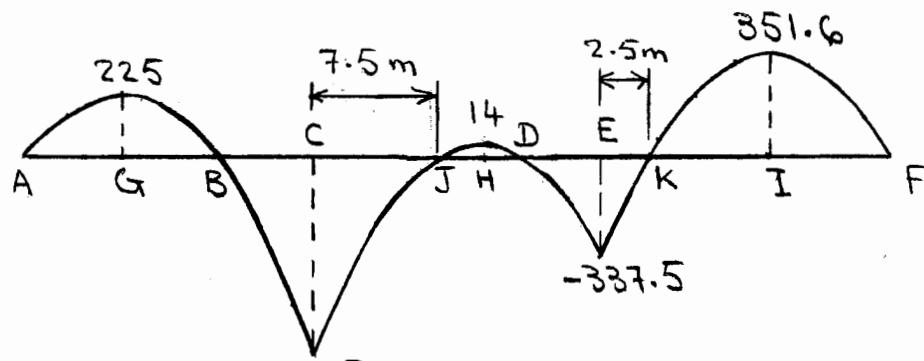
5.48



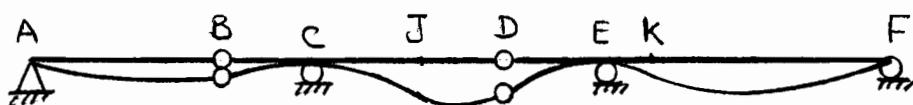
5.49



Shear Diagram (kN)

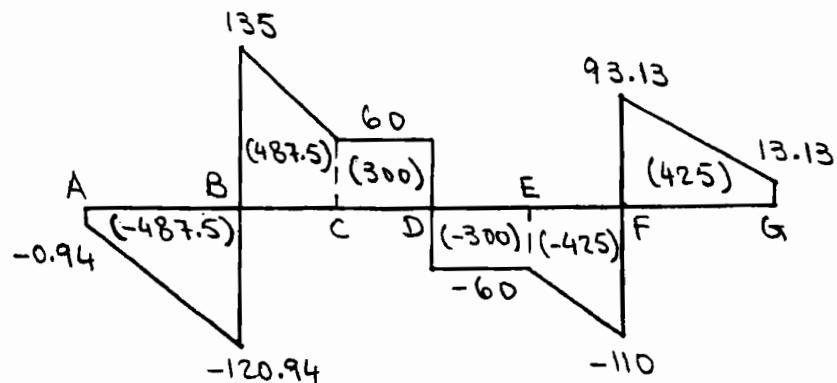
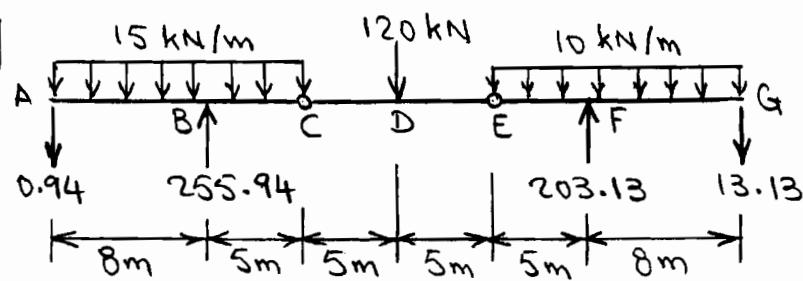


Bending Moment Diagram (kN.m)

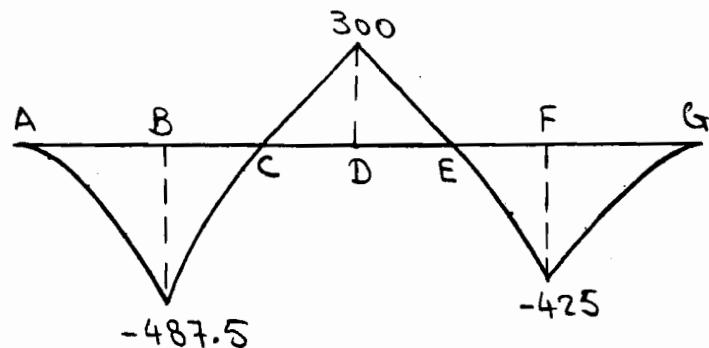


Qualitative Deflected Shape

5-50



Shear Diagram (kN)

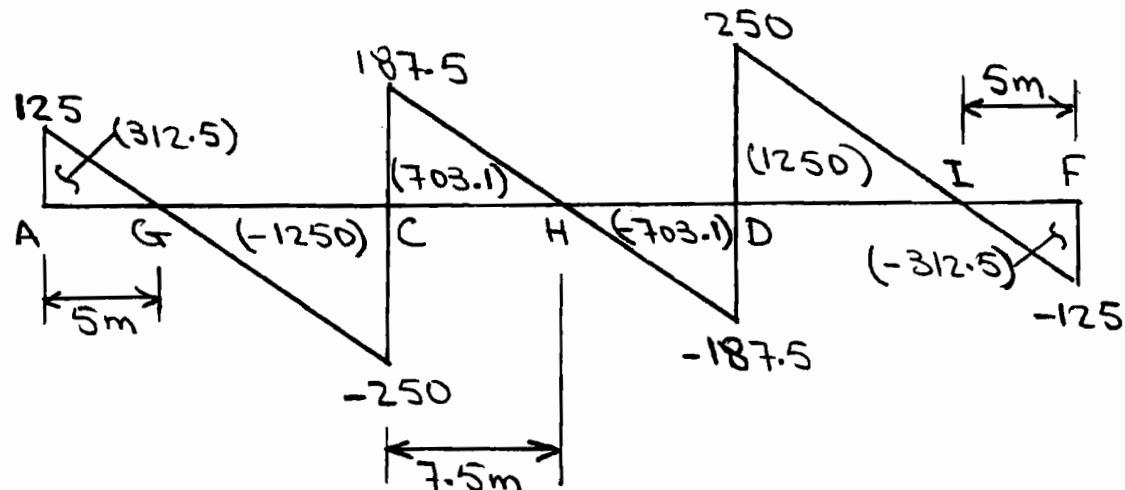
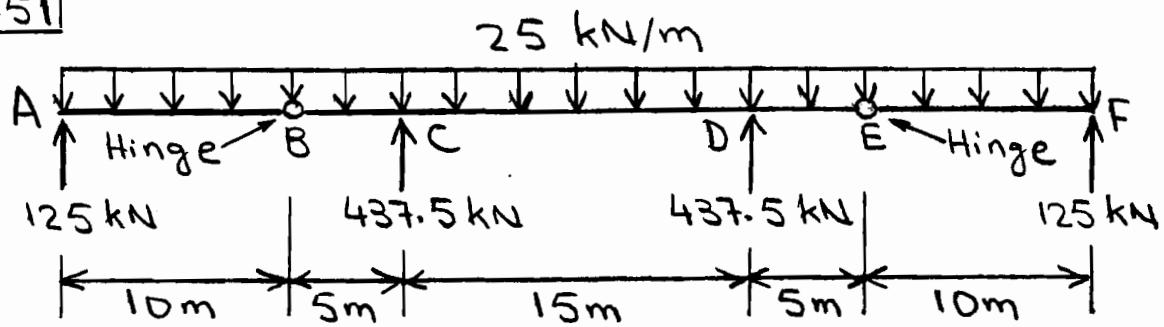


Bending Moment Diagram (kN-m)

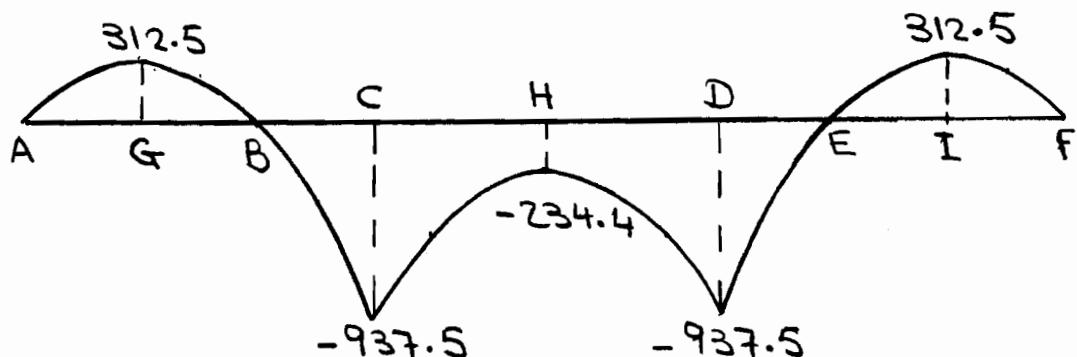


Qualitative Deflected Shape

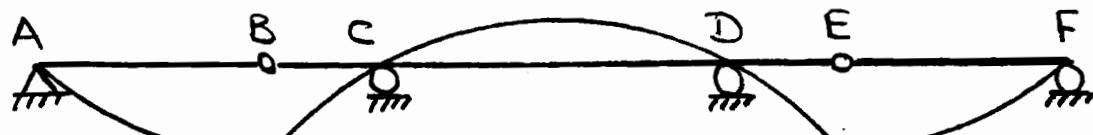
5.51



Shear Diagram (kN)

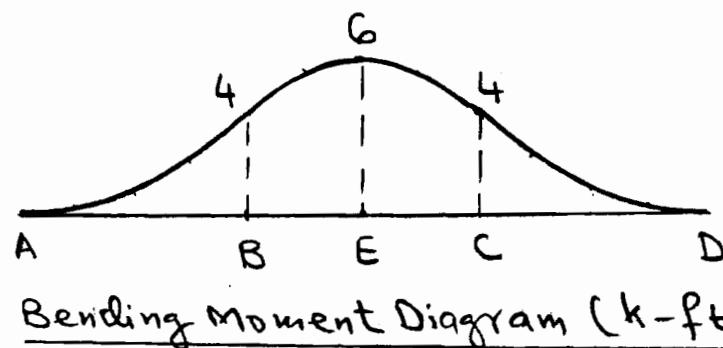
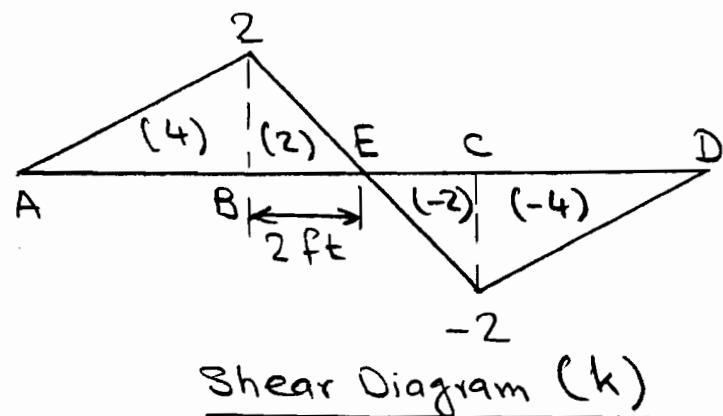
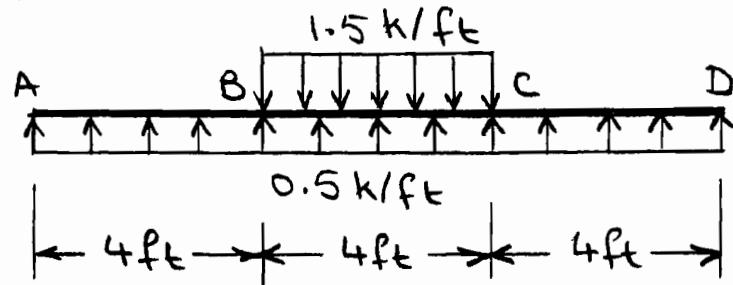


Bending Moment Diagram (kN.m)

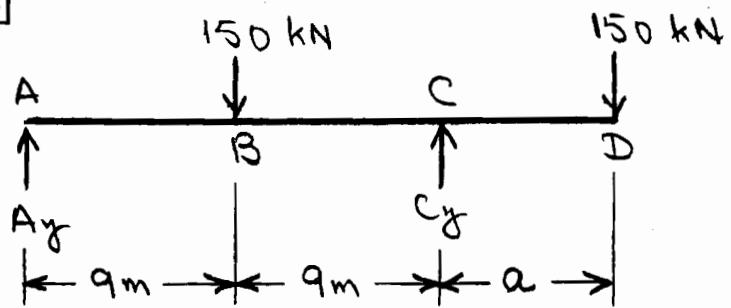


Qualitative Deflected Shape

5.52



5.53



$$+C \sum M_C = 0 \quad -A_y(18) + 150(9) - 150a = 0$$

$$A_y = 75 - \frac{25}{3}a$$

Maximum positive bending moment occurs at B:

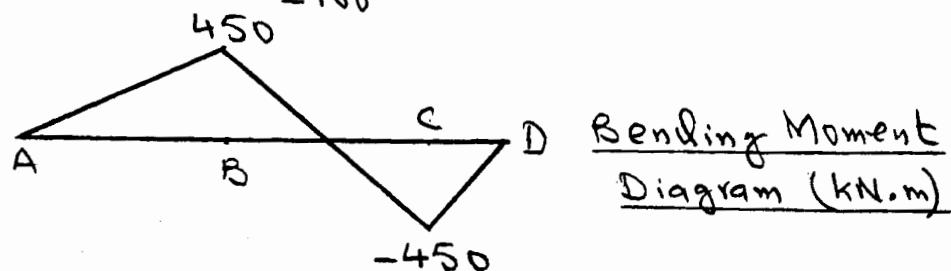
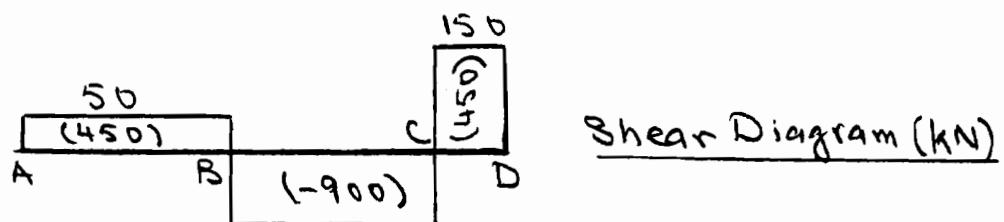
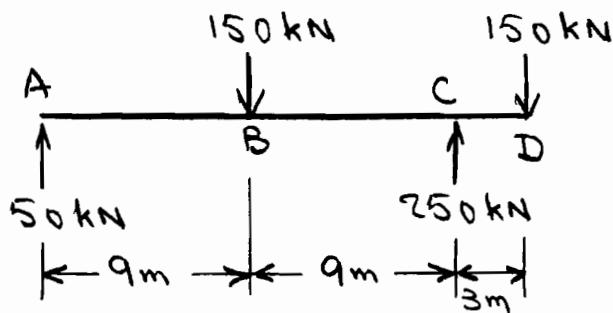
$$M_B = A_y(9) = 675 - 75a \quad (1)$$

Maximum negative bending moment occurs at C:

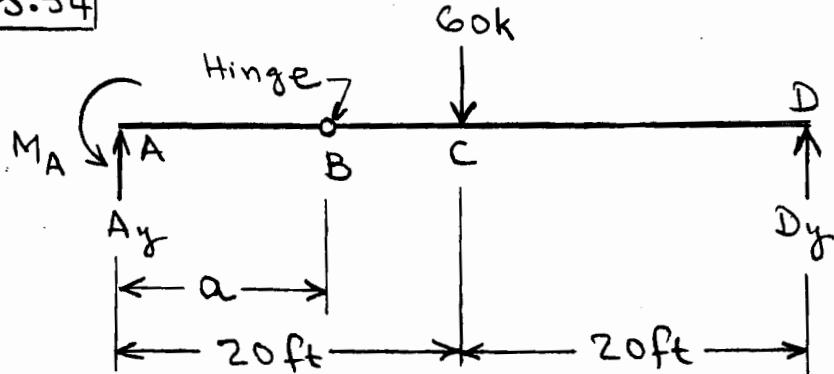
$$M_C = 150a \quad (2)$$

Equating Eqs. (1) and (2):

$$675 - 75a = 150a \quad a = 3m$$



5.54



$$+\zeta \sum M_B^D = 0 \quad -60(20-a) + D_y(40-a) = 0$$

$$D_y = \frac{60(20-a)}{40-a}$$

Maximum positive bending moment occurs at C:

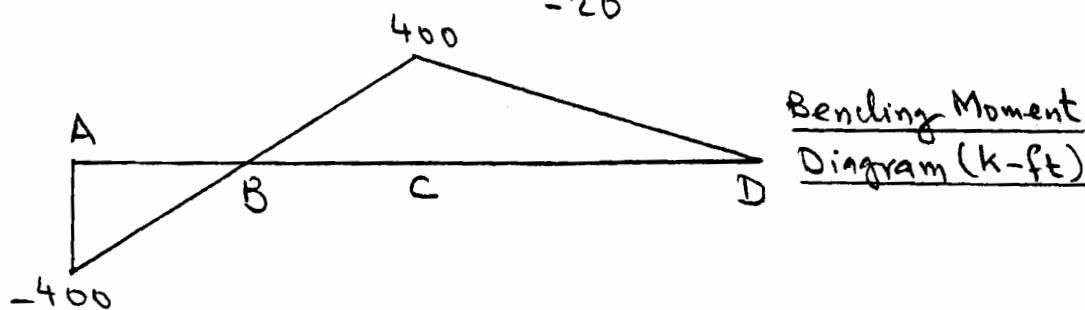
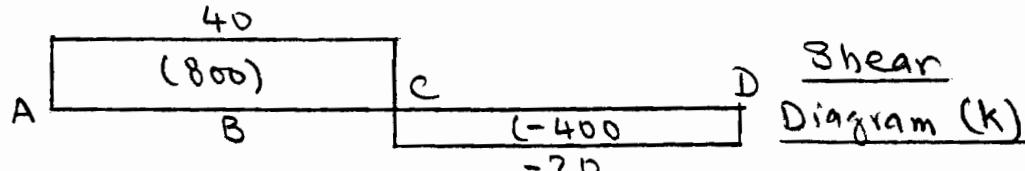
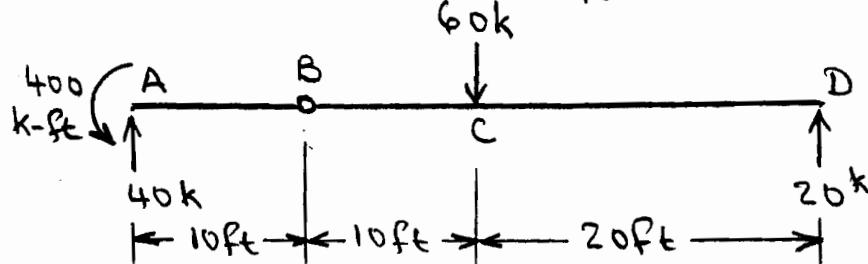
$$M_C = D_y(20) = \frac{1200(20-a)}{40-a} \quad (1)$$

Maximum negative bending moment occurs at A:

$$M_A = 60(20) - D_y(40) = 1200 - \frac{2400(20-a)}{40-a} \quad (2)$$

Equating Eqs. (1) and (2):

$$\frac{1200(20-a)}{40-a} = 1200 - \frac{2400(20-a)}{40-a} \quad a = 10 \text{ ft}$$



5.55

(a) $m = 4, j = 5, r = 3, e_c = 0$

$$3m+r = 3j+e_c; \text{ Statically determinate}$$

(b) $m = 6, j = 7, r = 3, e_c = 1$

$$3m+r < 3j+e_c; \text{ Unstable}$$

(c) $m = 7, j = 6, r = 3, e_c = 0$

$$3m+r > 3j+e_c; \text{ Statically indeterminate}$$

$$i = (21+3) - 18 = \underline{6}$$

(d) $m = 7, j = 7, r = 6, e_c = 1$

$$3m+r > 3j+e_c; \text{ Statically indeterminate}$$

$$i = (21+6) - (21+1) = \underline{5}$$

5.56

(a) $m = 6, j = 6, r = 4, e_c = 2$

$$3m+r > 3j+e_c; \text{ Statically indeterminate}$$

$$i = (18+4) - (18+2) = \underline{2}$$

(b) $m = 6, j = 7, r = 6, e_c = 3$

$$3m+r = 3j+e_c; \text{ Statically determinate}$$

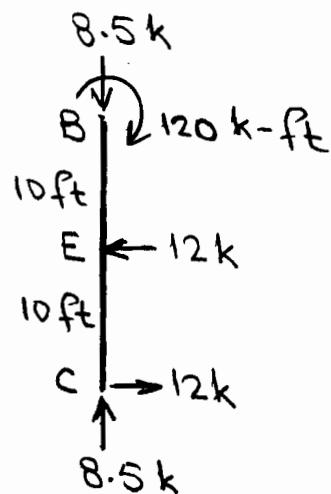
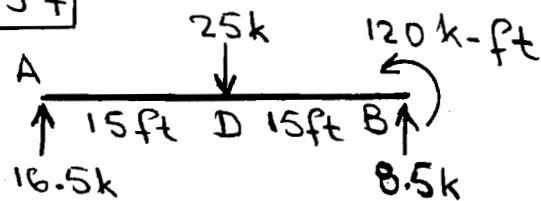
(c) $m = 5, j = 6, r = 4, e_c = 2$

$$3m+r < 3j+e_c; \text{ Unstable}$$

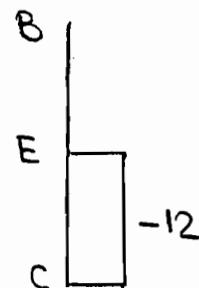
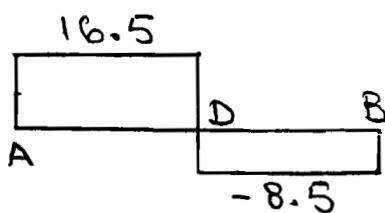
(d) Statically indeterminate

$$i = 3(8) = \underline{24}$$

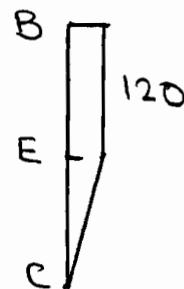
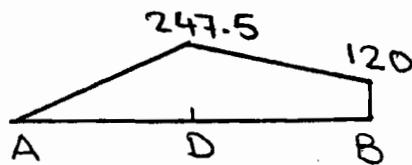
5.57



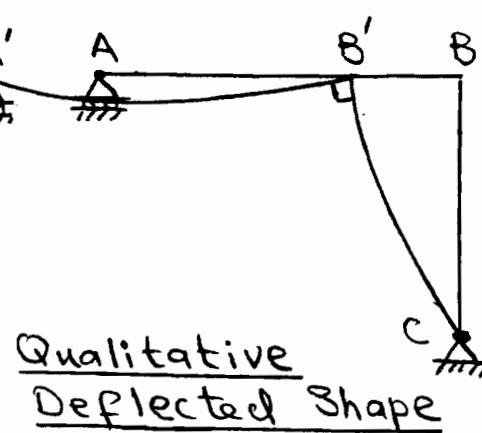
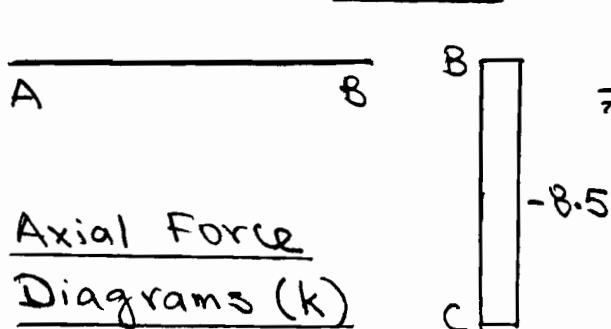
Member End Forces



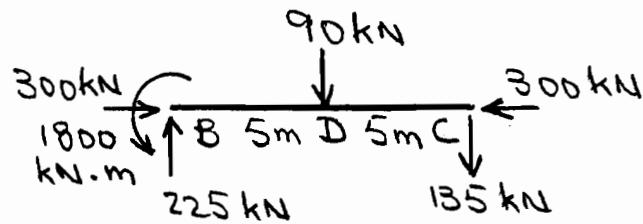
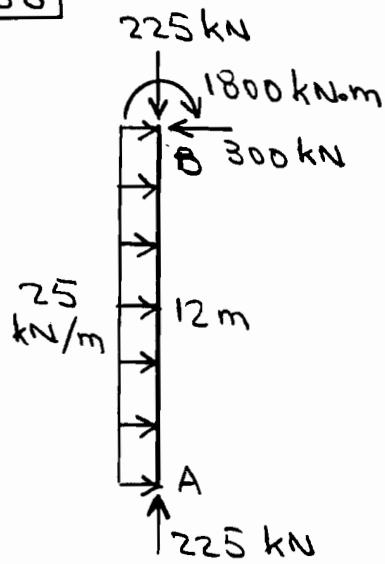
Shear Diagrams (k)



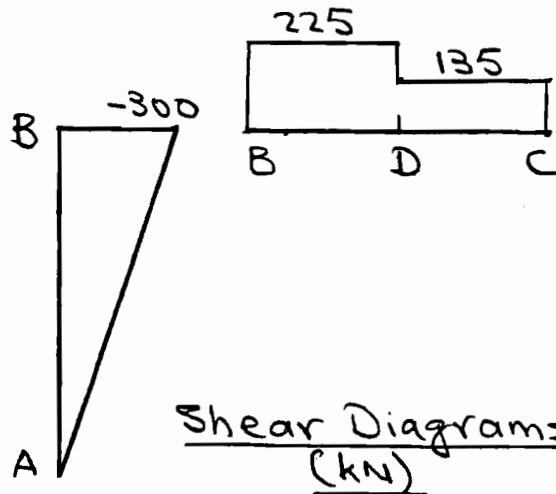
Bending Moment Diagrams (k-ft)



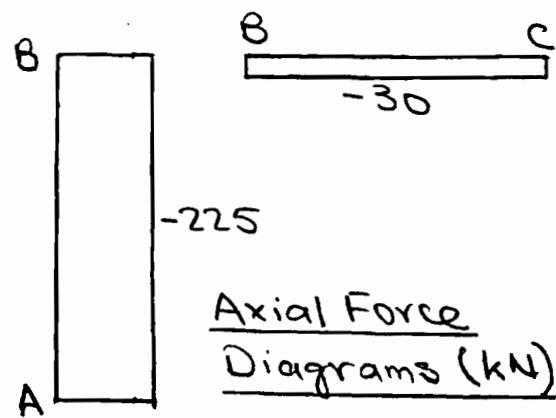
5.58



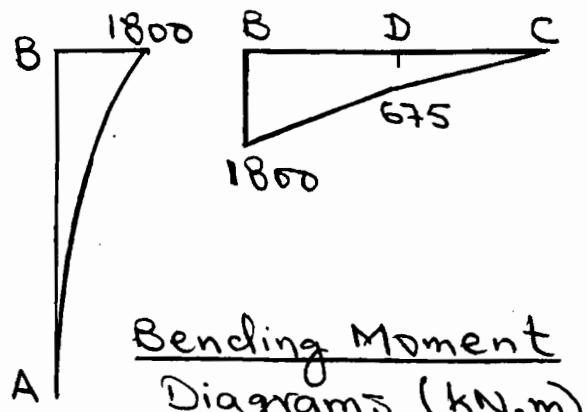
Member End Forces



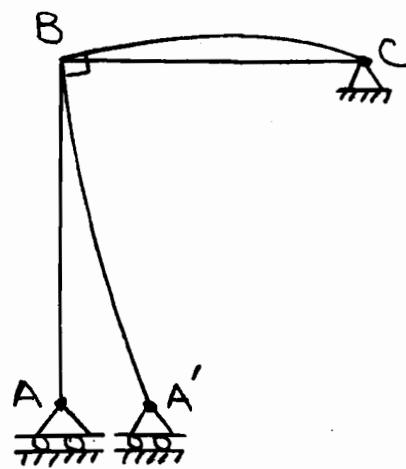
Shear Diagrams (kN)



Axial Force Diagrams (kN)

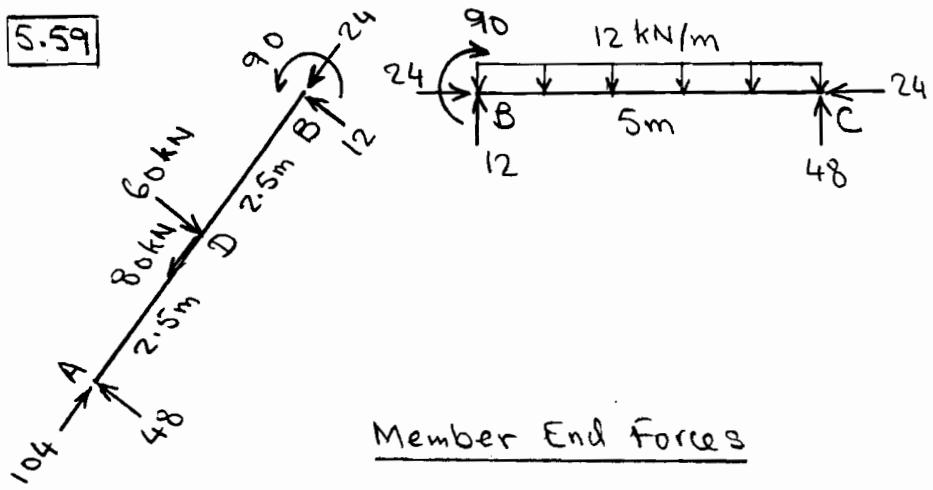


Bending Moment Diagrams (kN.m)

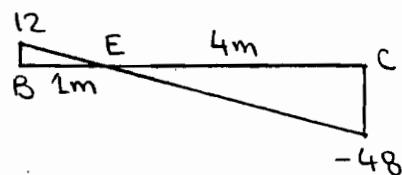
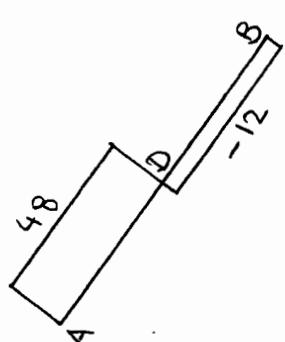


Qualitative Deflected Shape

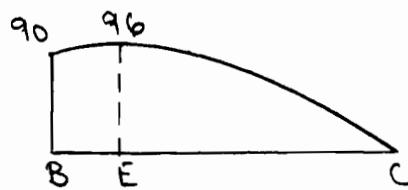
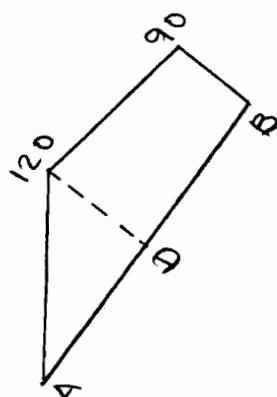
5.59



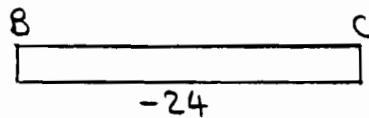
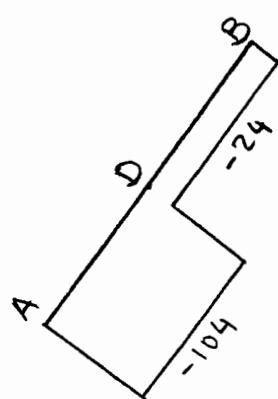
Member End Forces



Shear Diagrams (kN)

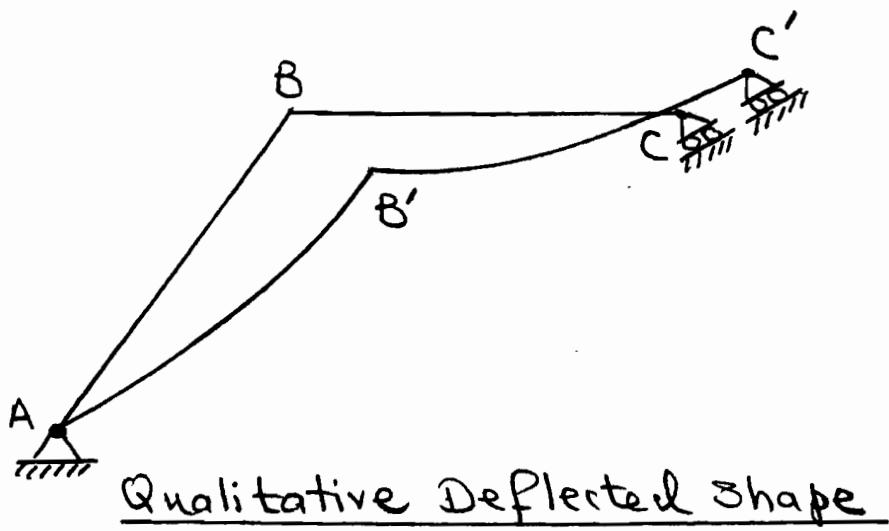


Bending Moment Diagrams (kNm)

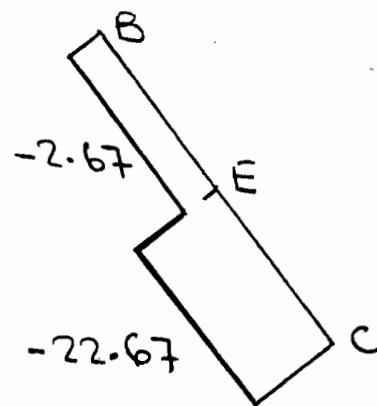
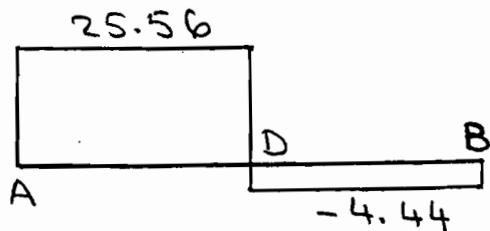
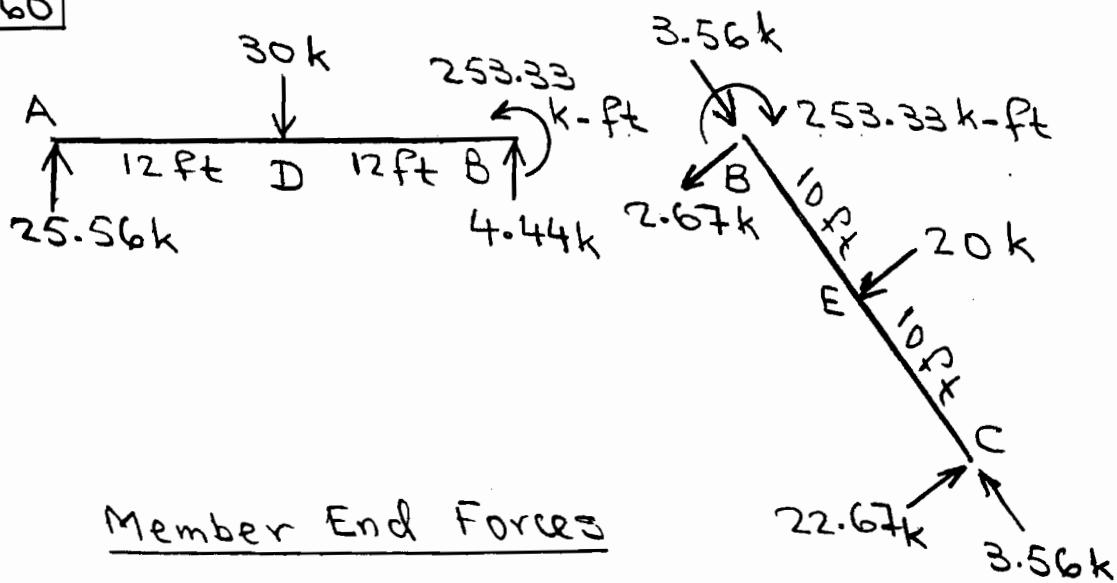


Axial Force Diagrams (kN)

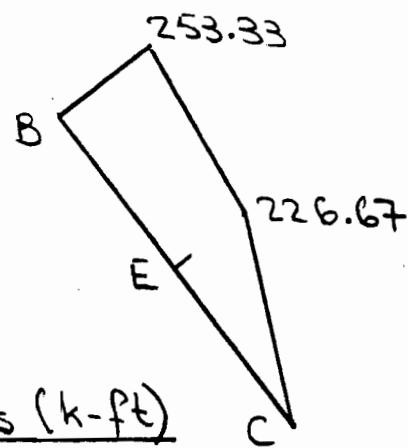
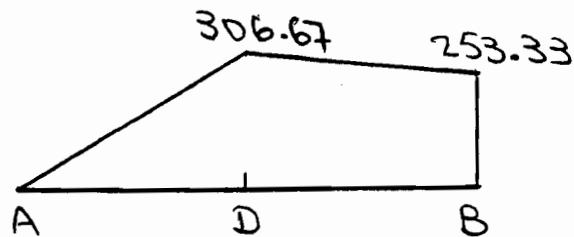
5.59 (contd.)



5.60

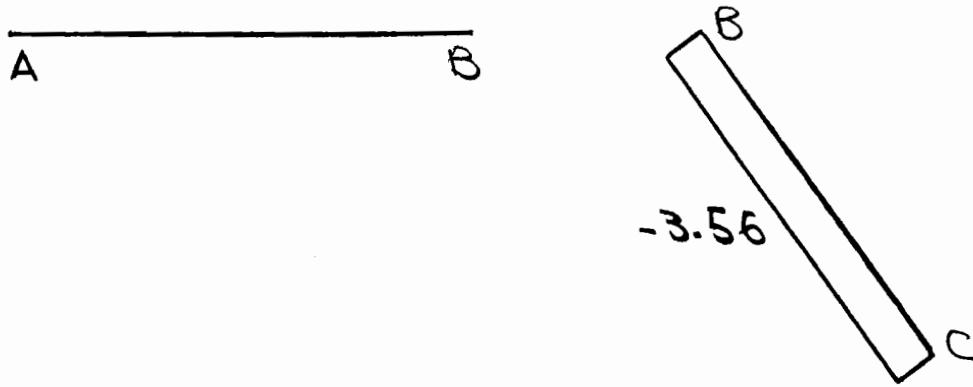


Shear Diagrams (k)

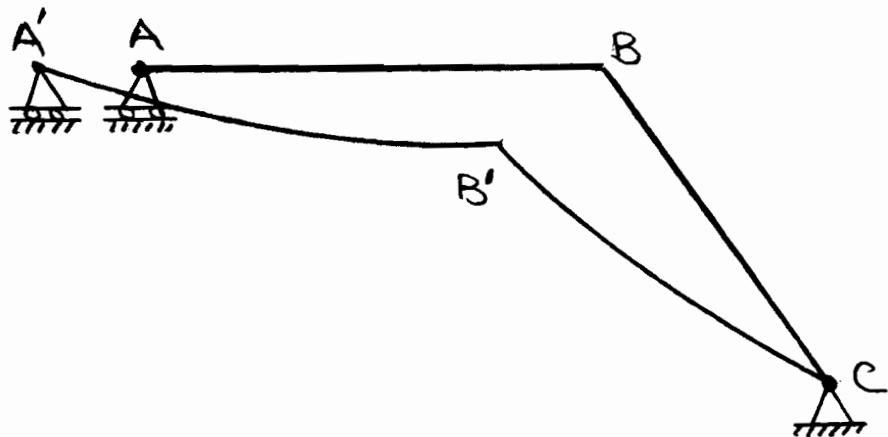


Bending Moment Diagrams (k-ft)

5.60 (contd.)

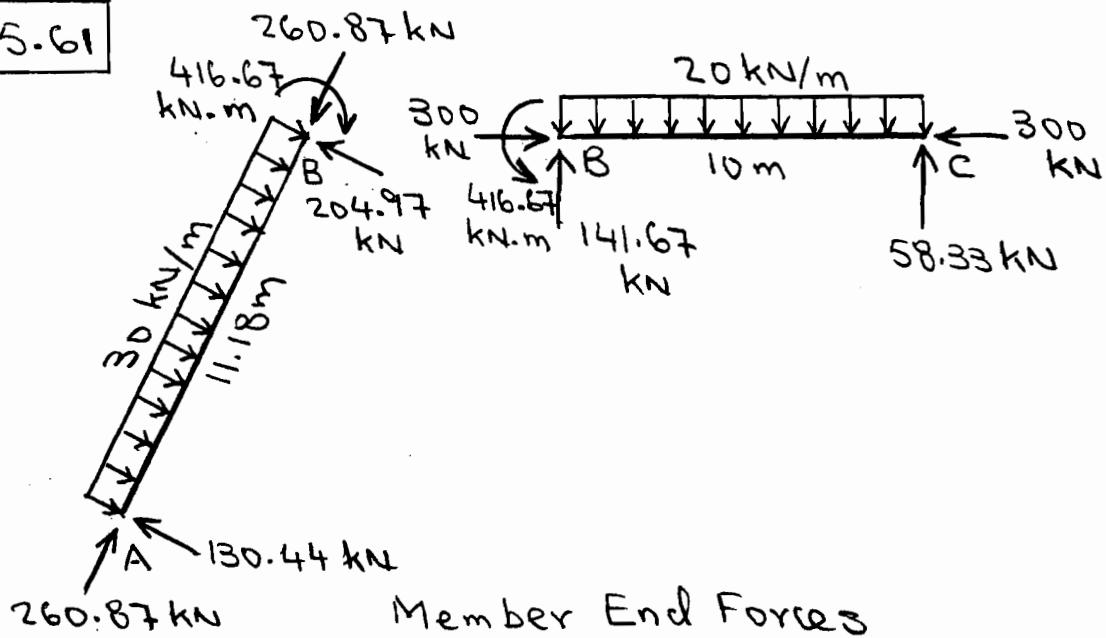


Axial Force Diagrams (k)

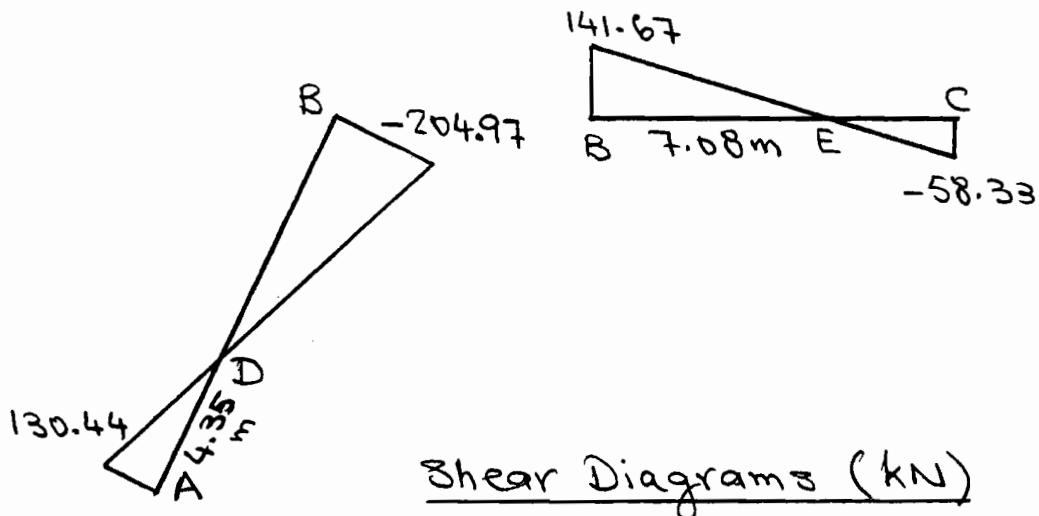


Qualitative Deflected Shape

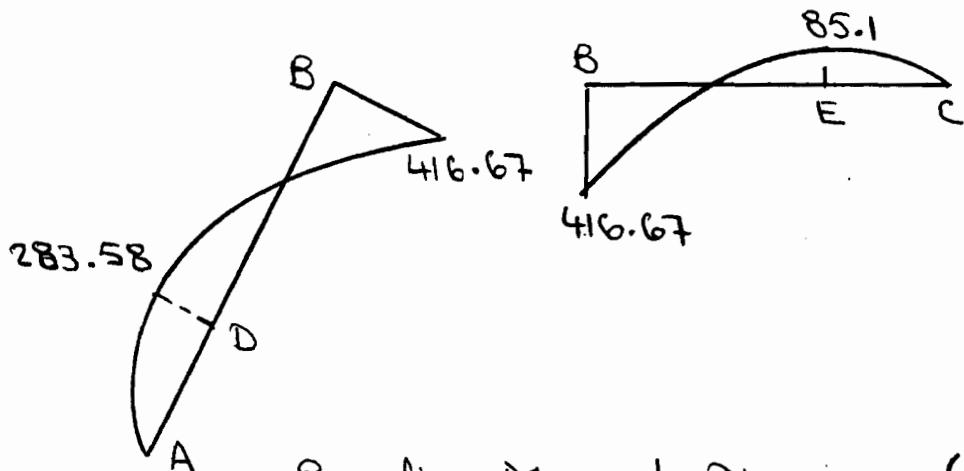
5.61



Member End Forces

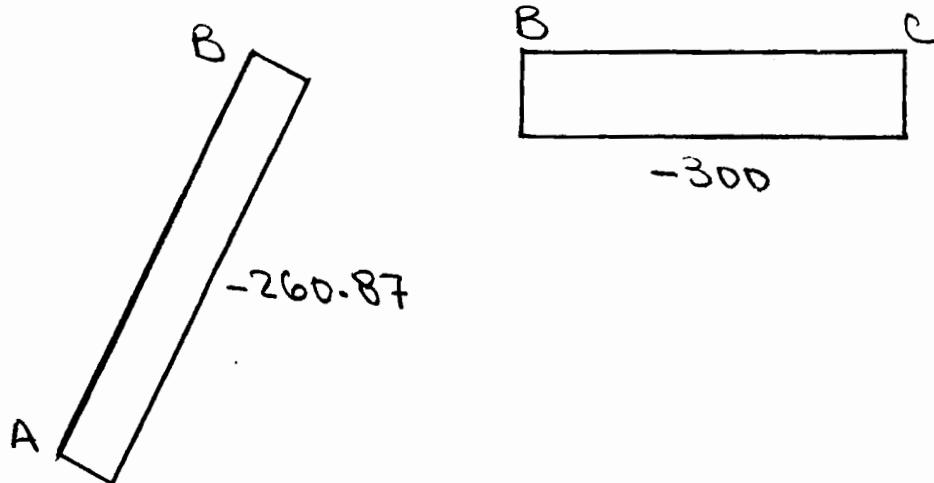


Shear Diagrams (kN)

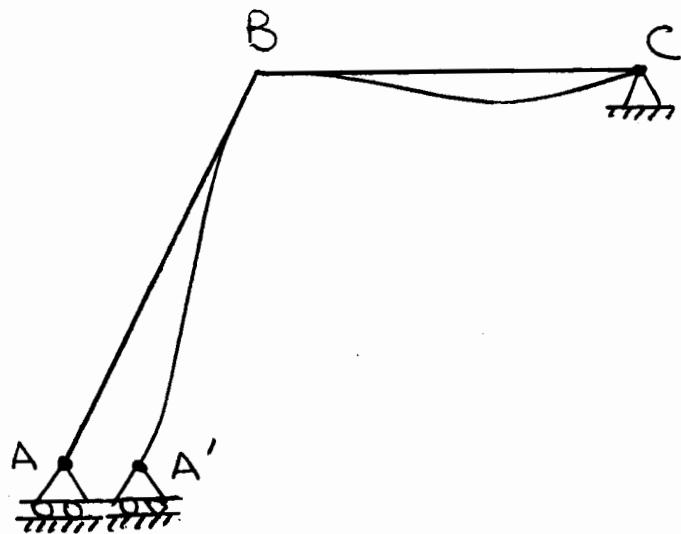


Bending Moment Diagrams (kN.m)

5.61 (cont'd.)

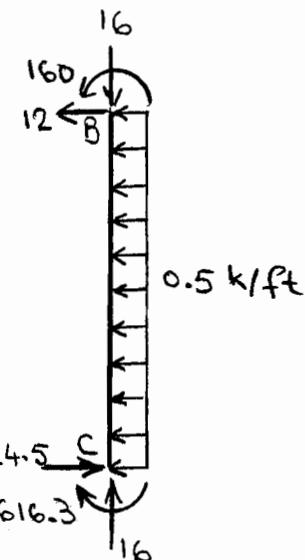
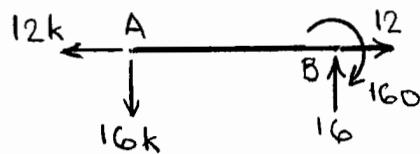


Axial Force Diagrams (kN)

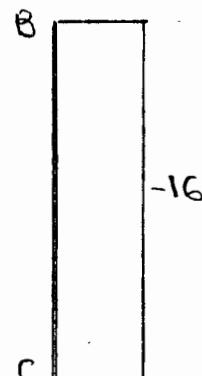
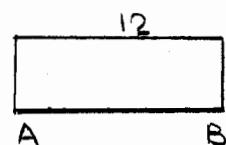
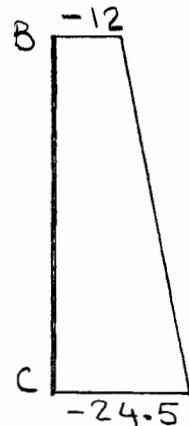
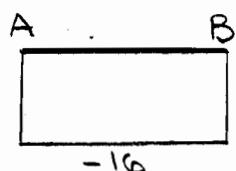


Qualitative Deflected Shape

5.62

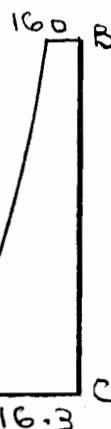
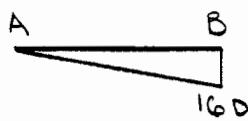


Member End Forces

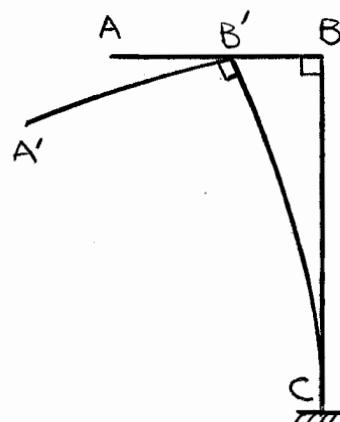


Shear Diagrams
(k)

Axial Force
Diagrams (k)

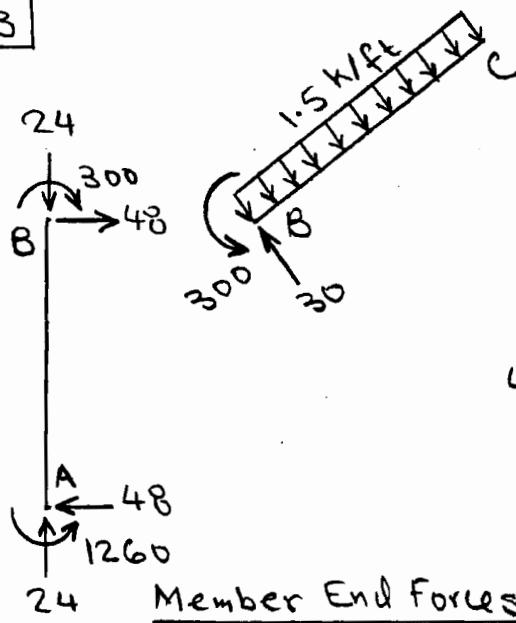


Bending Moment
Diagrams (k-ft)

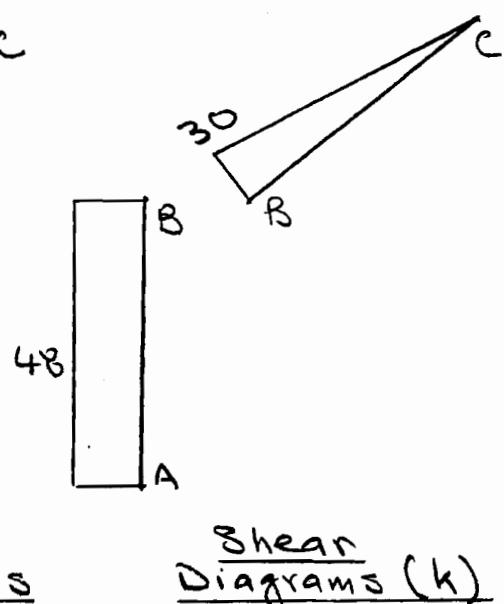


Qualitative Deflected
Shape

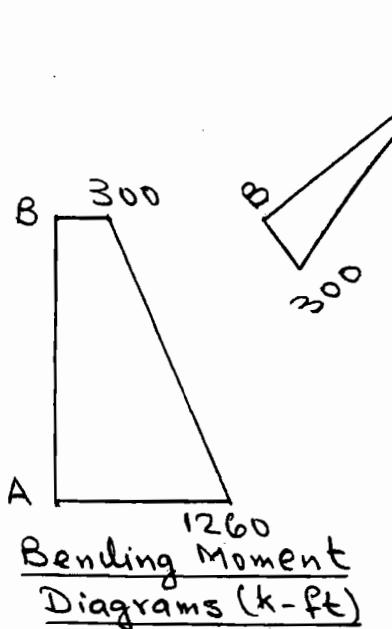
5.63



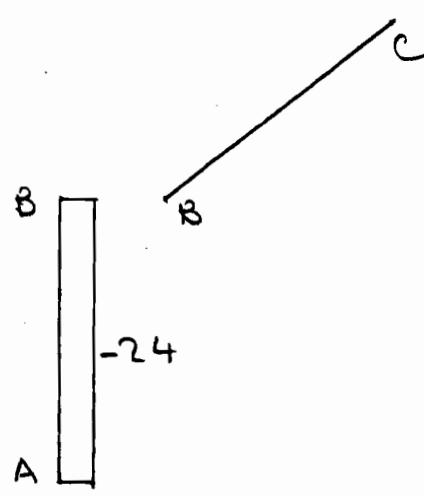
Member End Forces



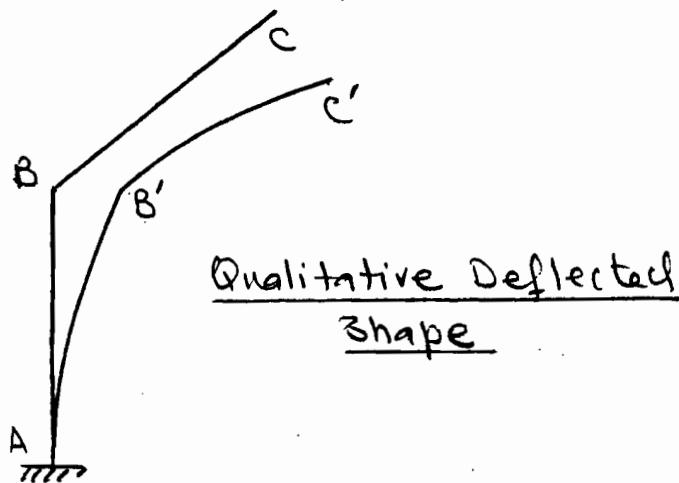
Shear Diagrams (k)



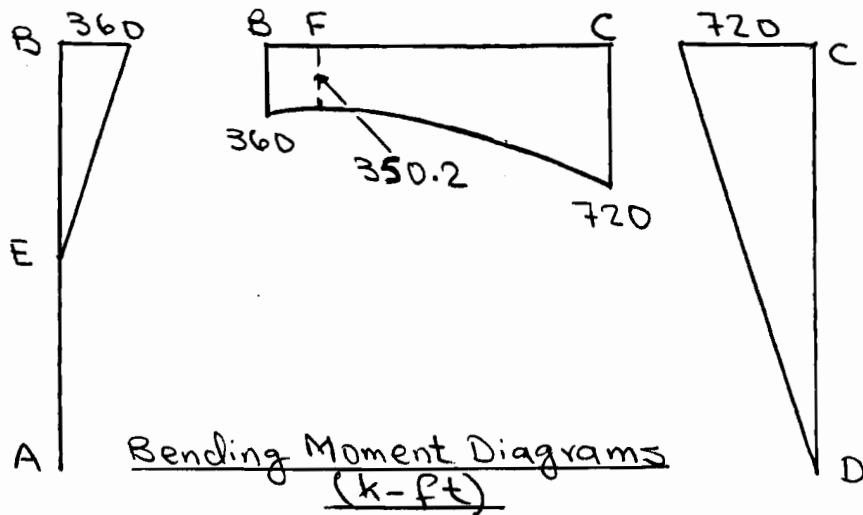
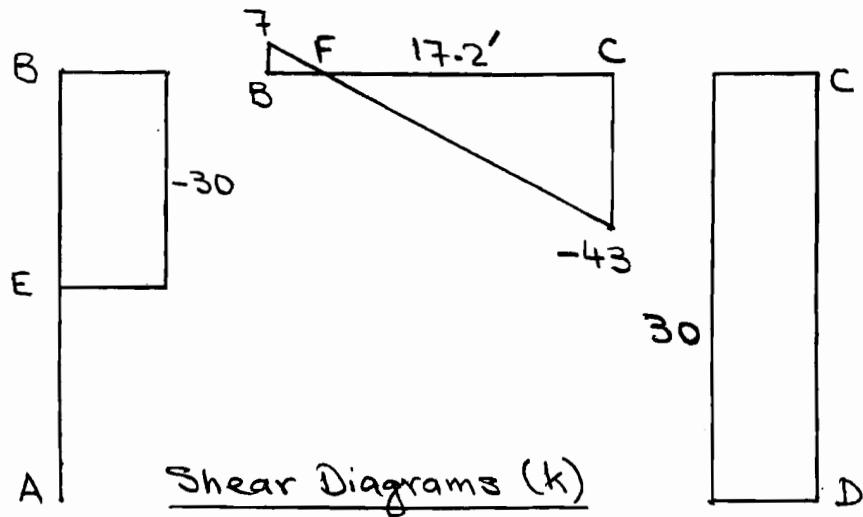
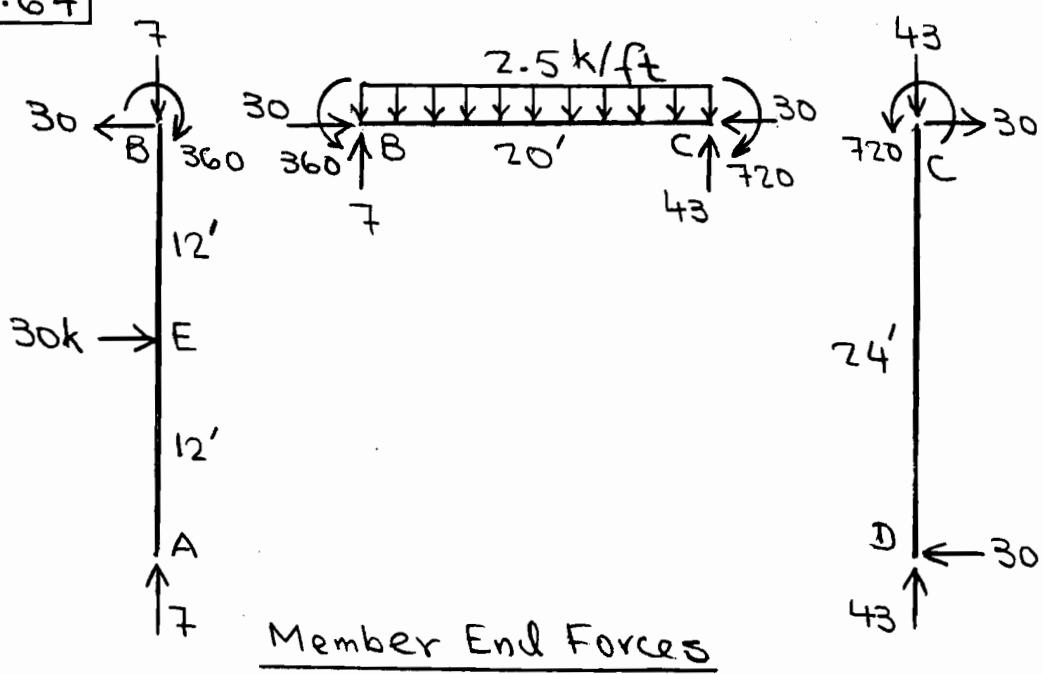
Bending Moment Diagrams (k-ft)



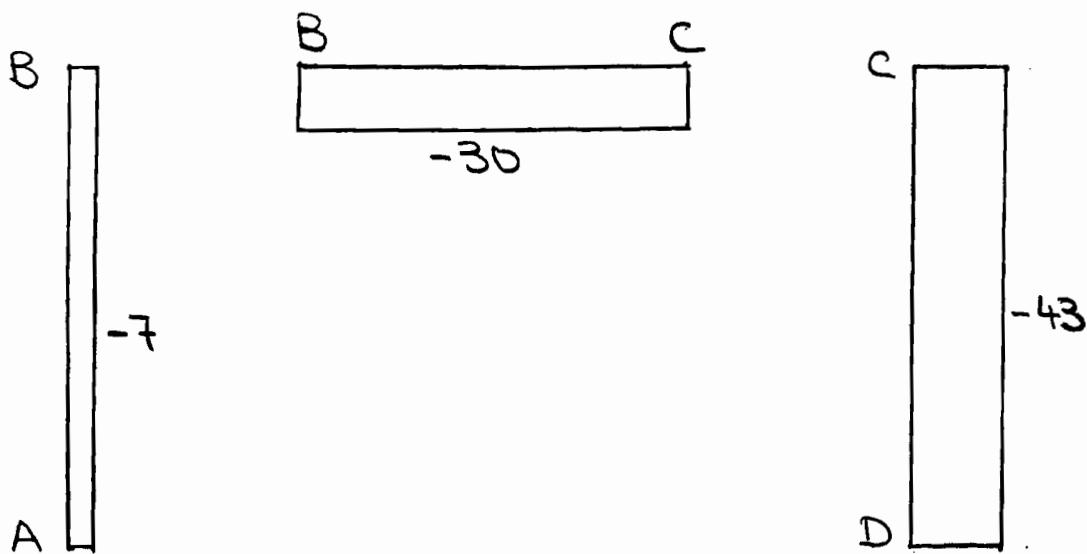
Axial Force Diagrams (kN)



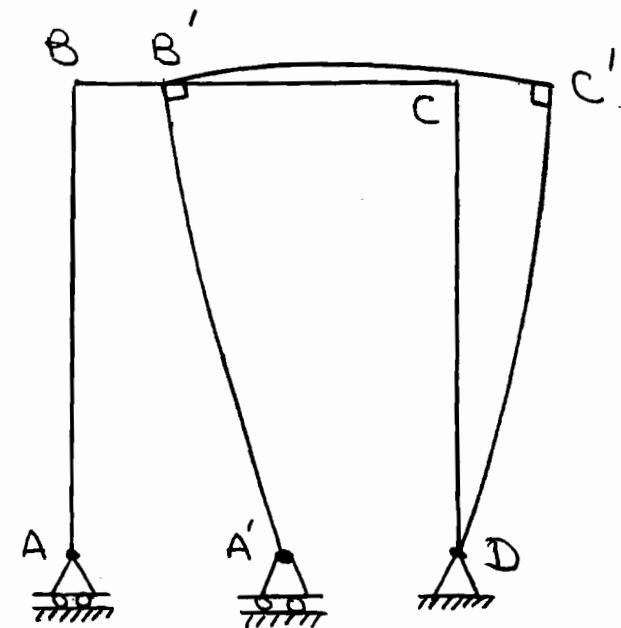
5.64



5.64 (cont'd.)

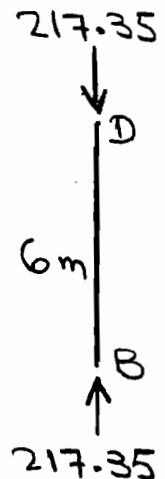
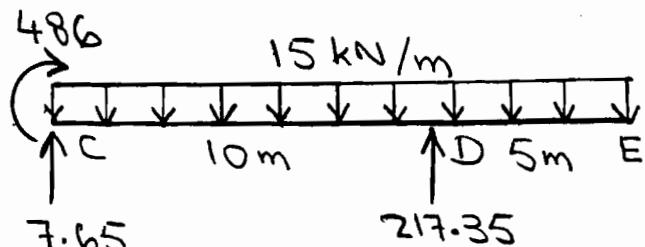
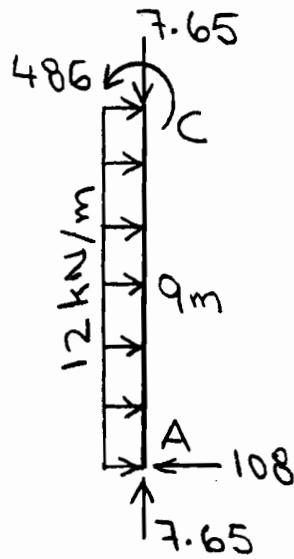


Axial Force Diagrams (k)

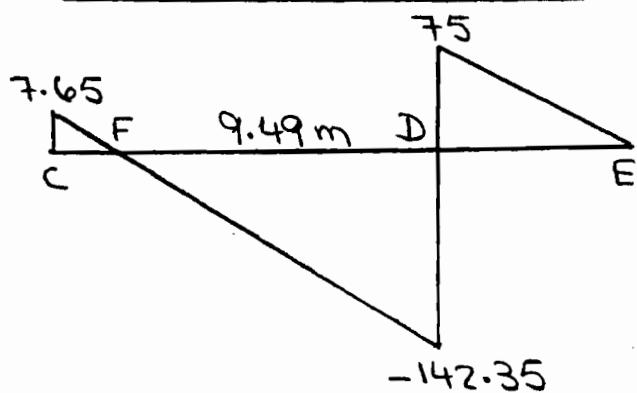


Qualitative Deflected Shape

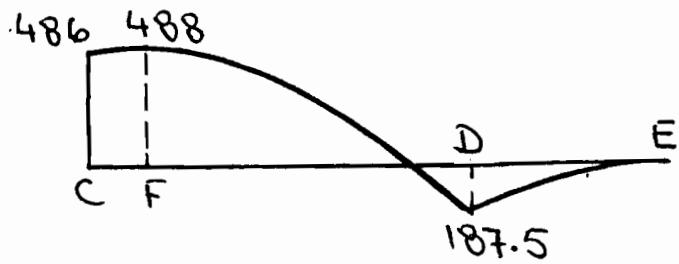
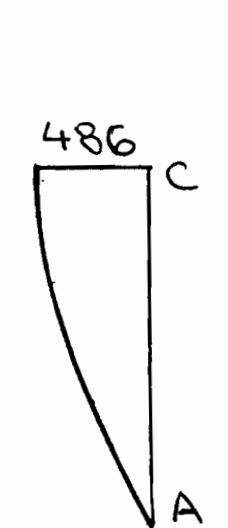
5.65



Member End Forces

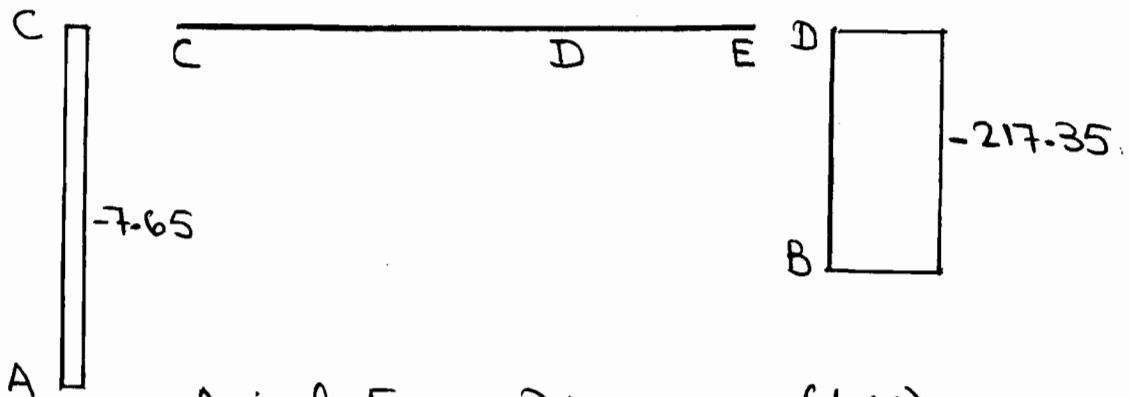


Shear Diagrams (kN)

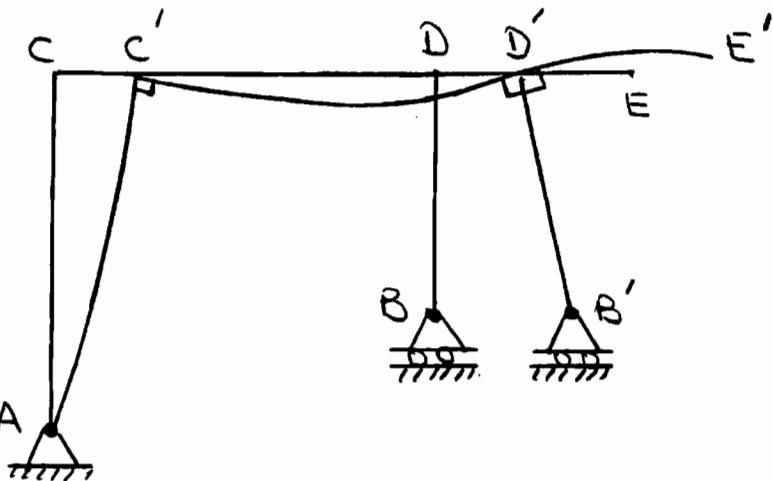


Bending Moment Diagrams (kN.m)

5.65 (contd.)

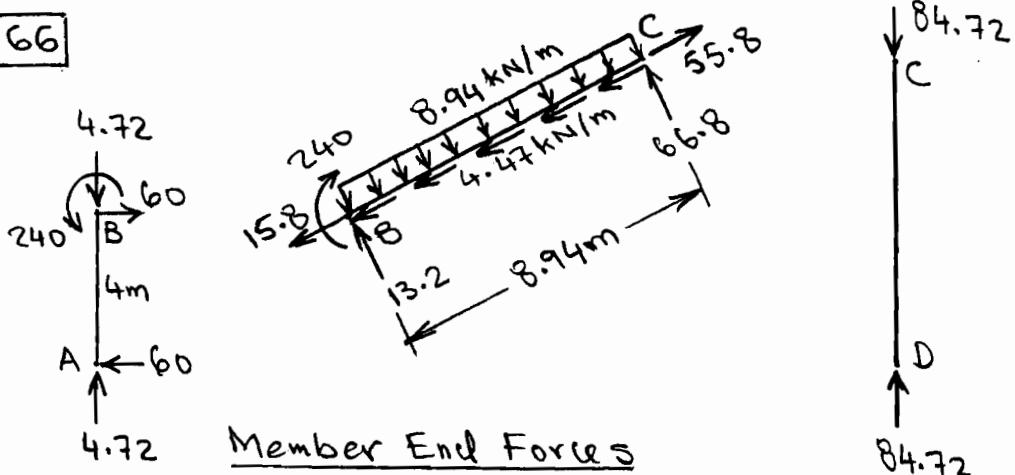


Axial Force Diagrams (kN)

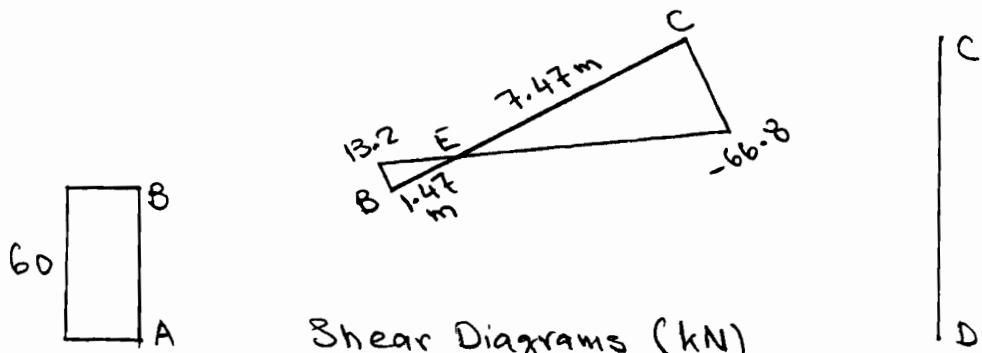


Qualitative Deflected Shape

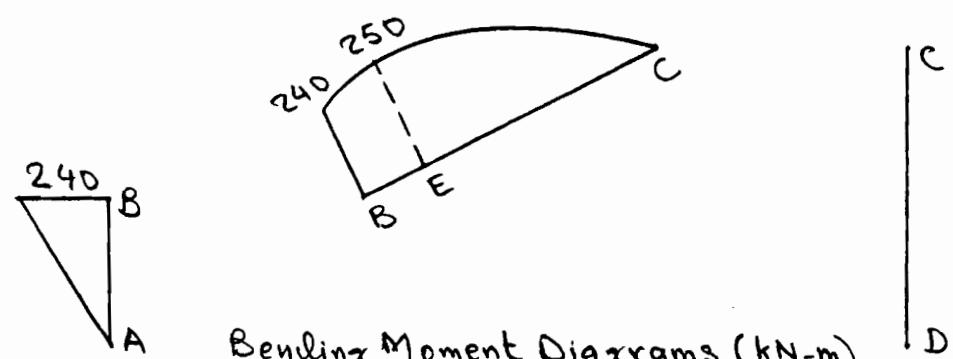
5.66



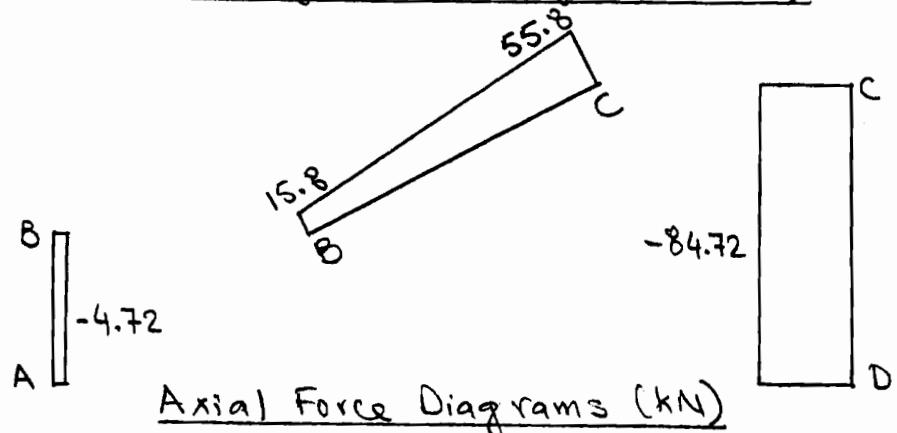
Member End Forces



Shear Diagrams (kN)

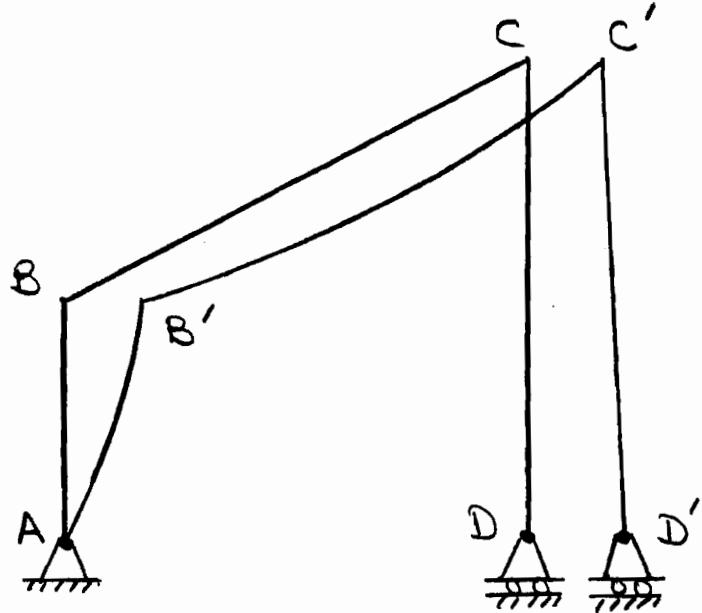


Bending Moment Diagrams (kN-m)



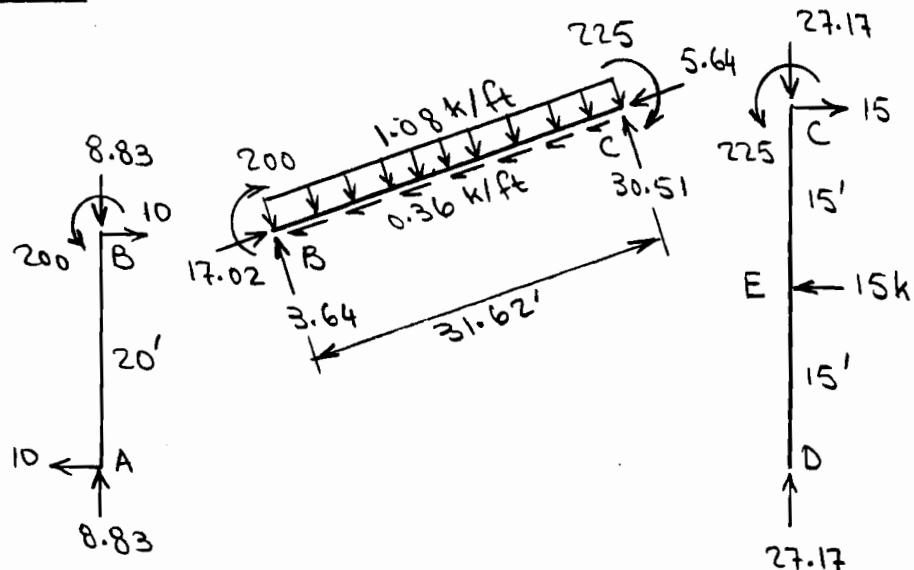
Axial Force Diagrams (kN)

5-66 (contd.)

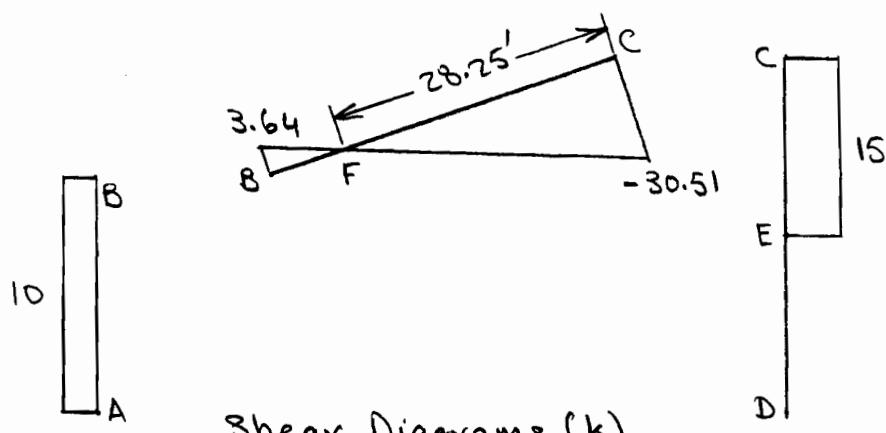


Qualitative Deflected Shape

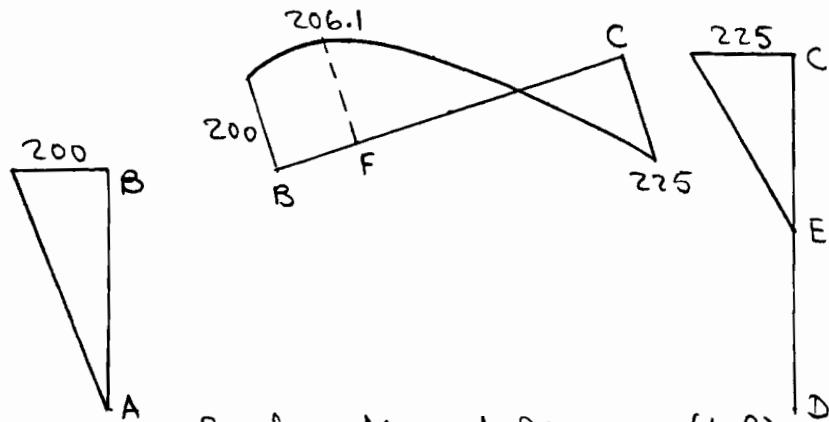
5.67



Member End Forces

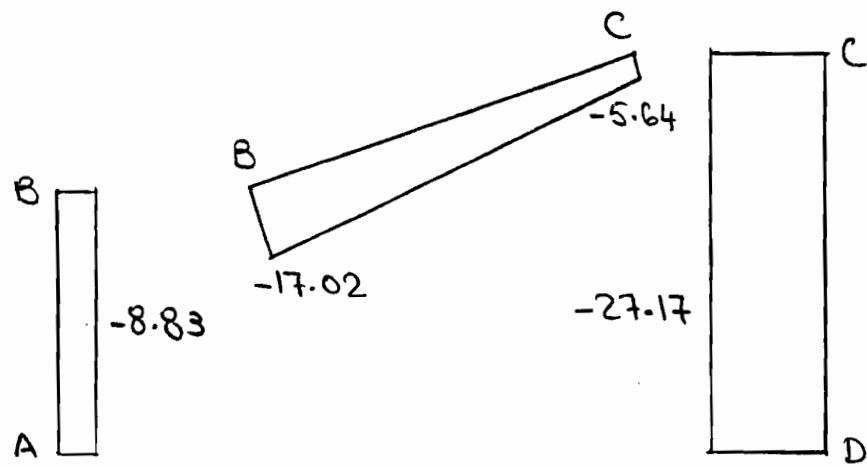


Shear Diagrams (k)

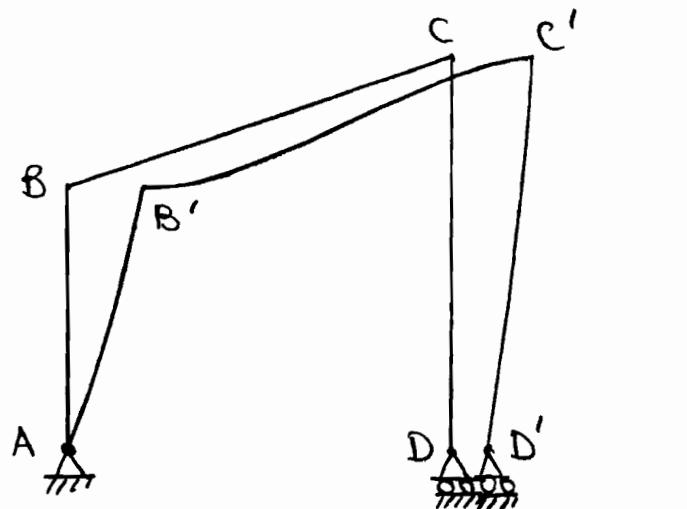


Bending Moment Diagrams (k-ft)

5.67 (contd.)

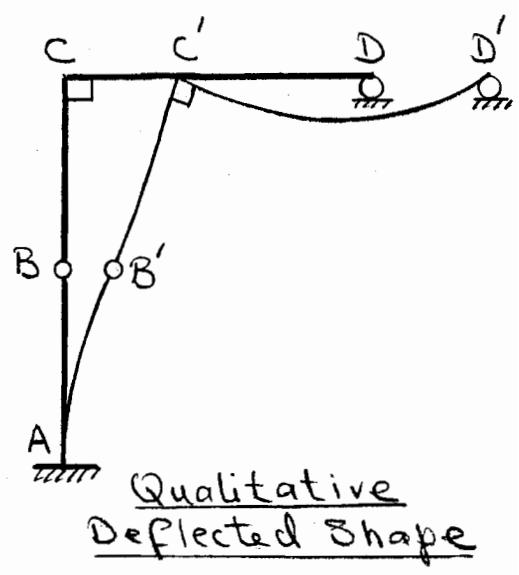
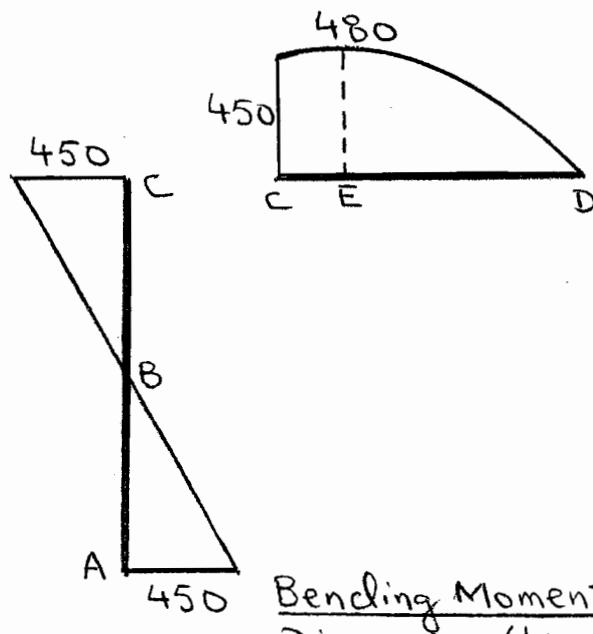
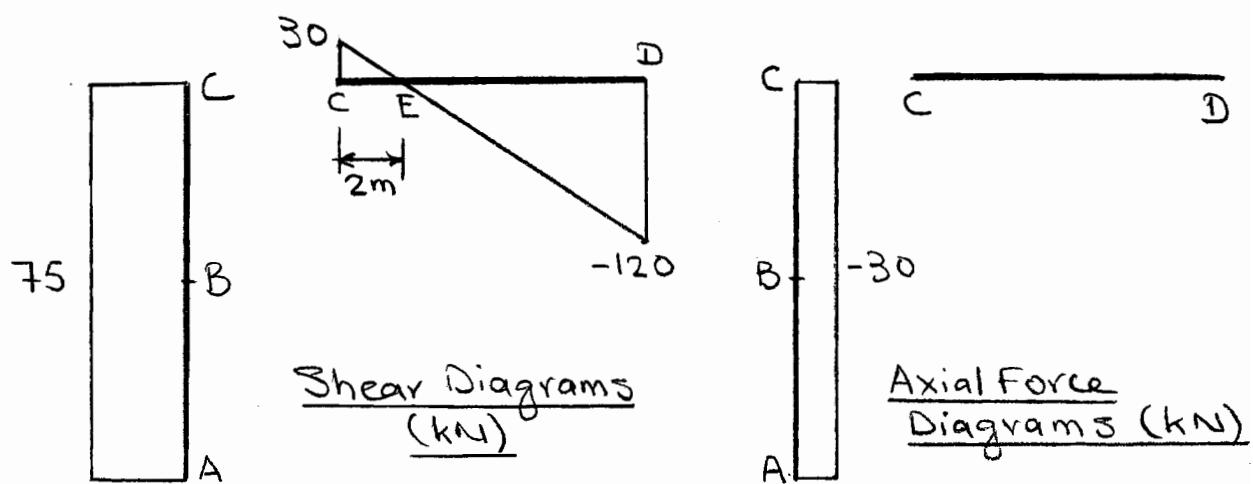
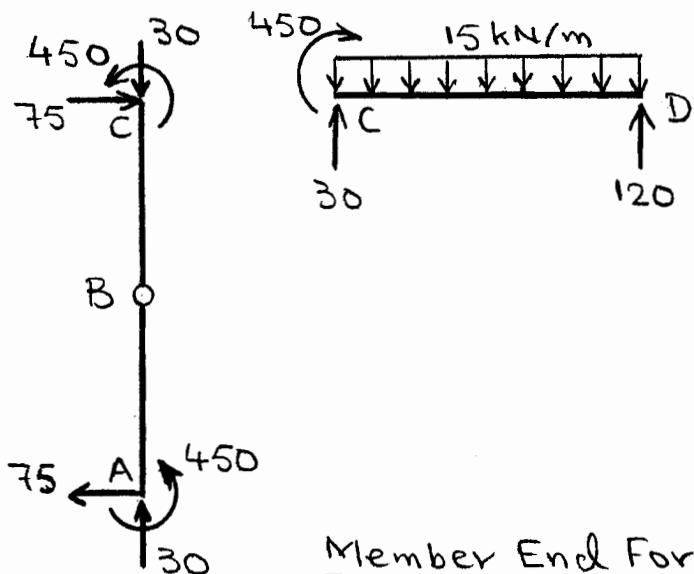


Axial Force Diagrams (k)

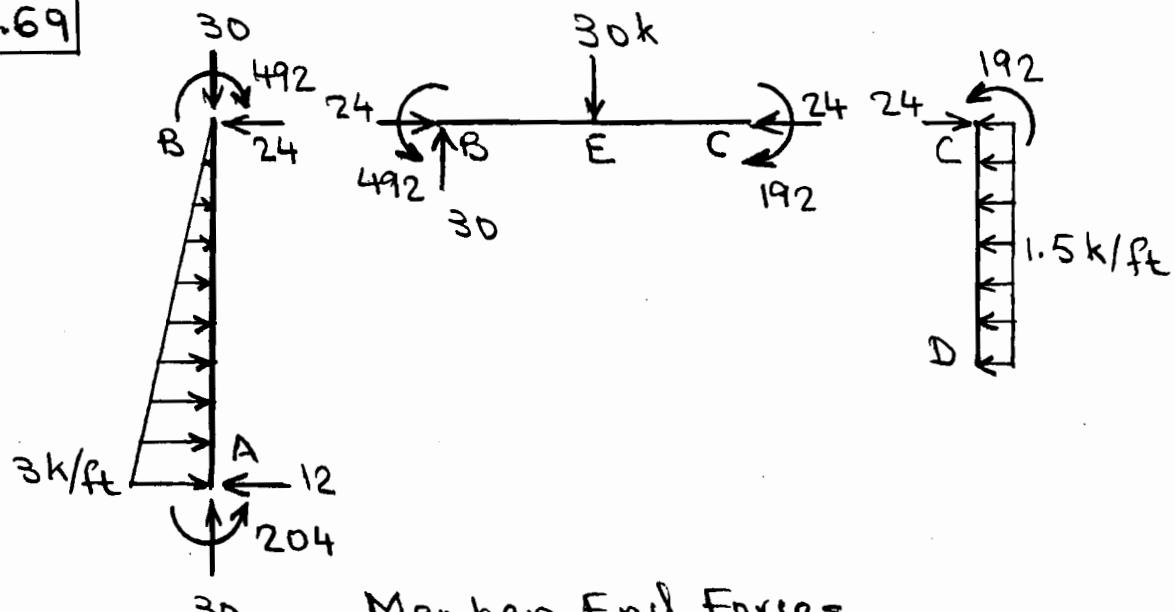


Qualitative Deflected Shape

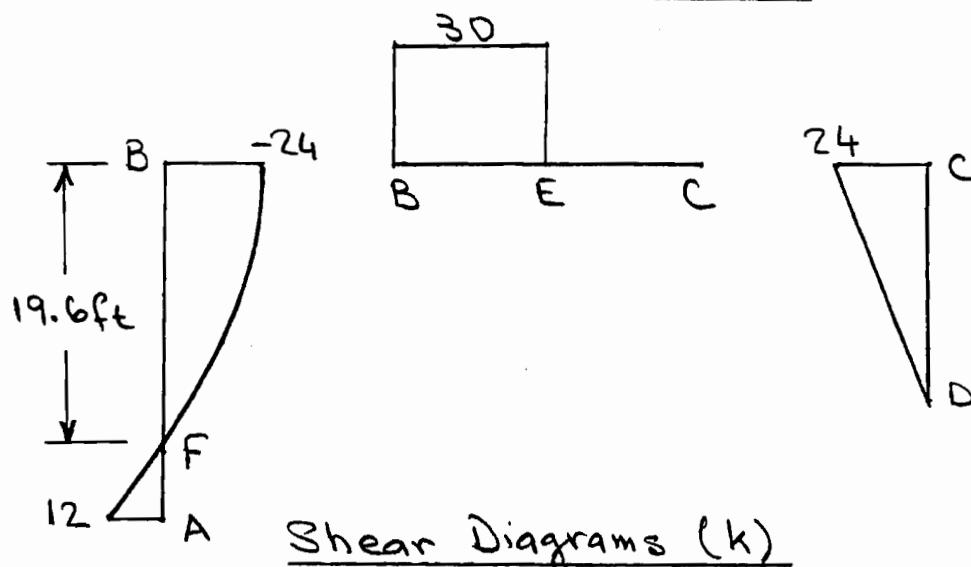
5.68



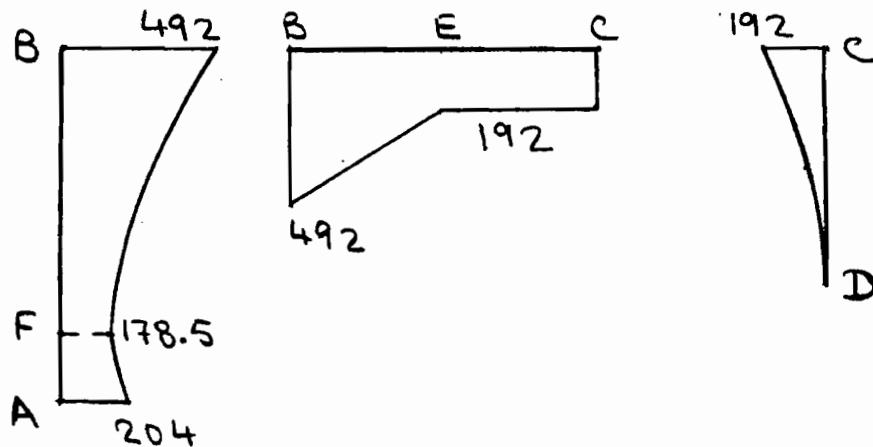
5.69



Member End Forces

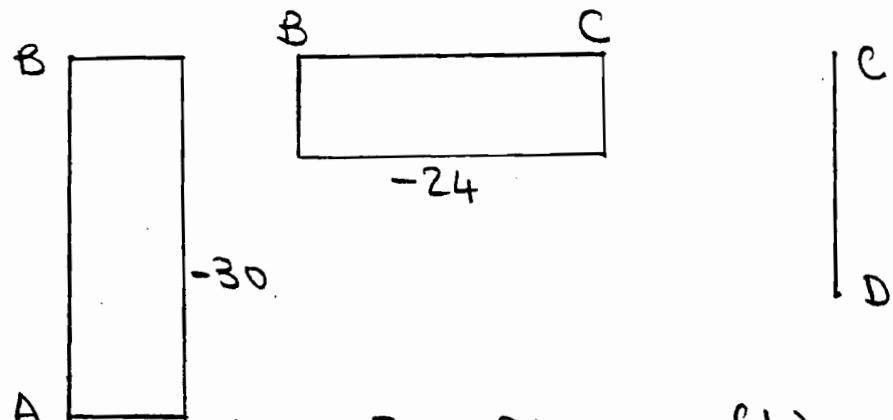


Shear Diagrams (k)

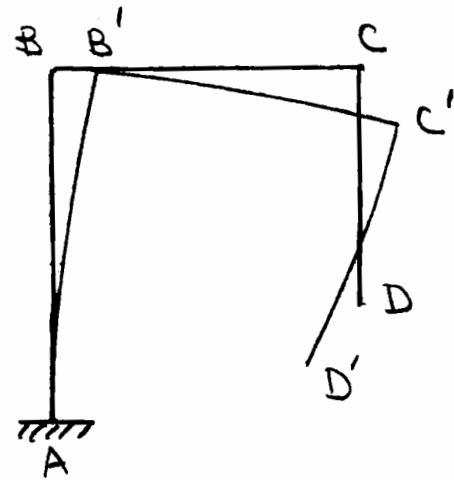


Bending Moment Diagrams (k-ft)

5.69 (contd.)

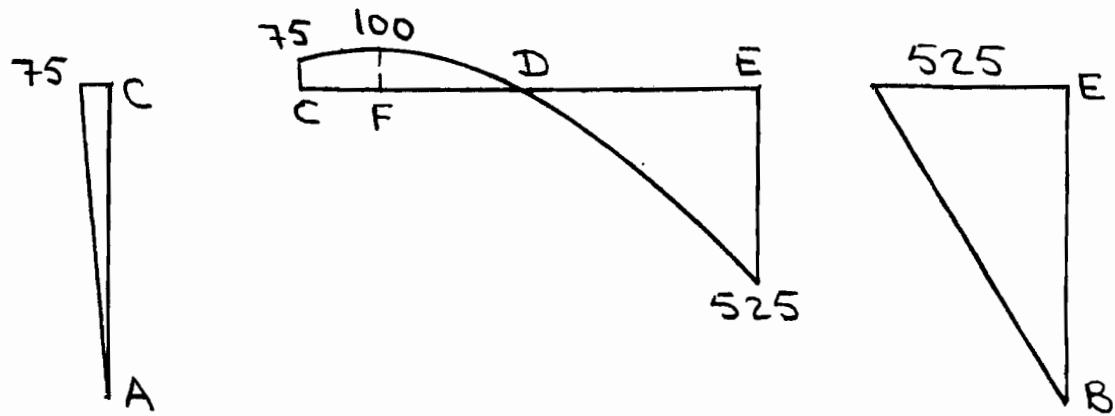
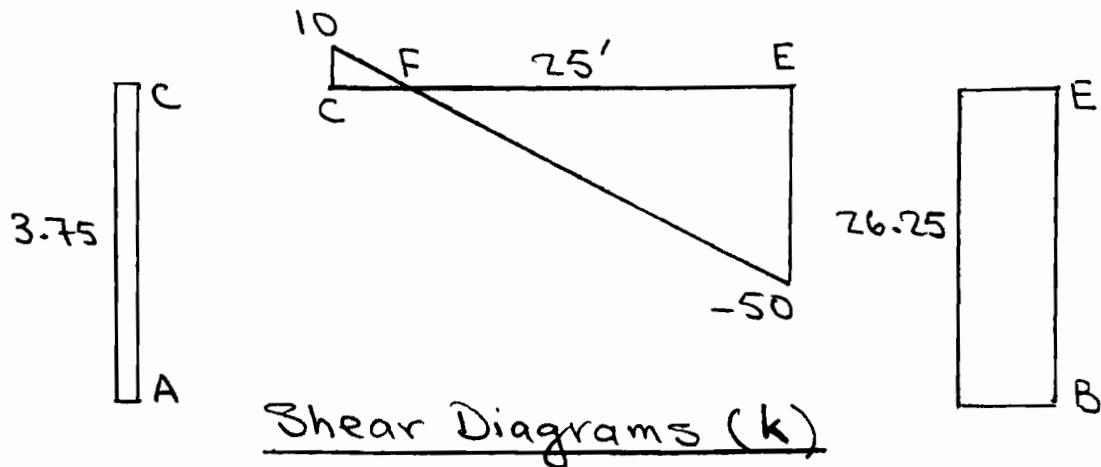
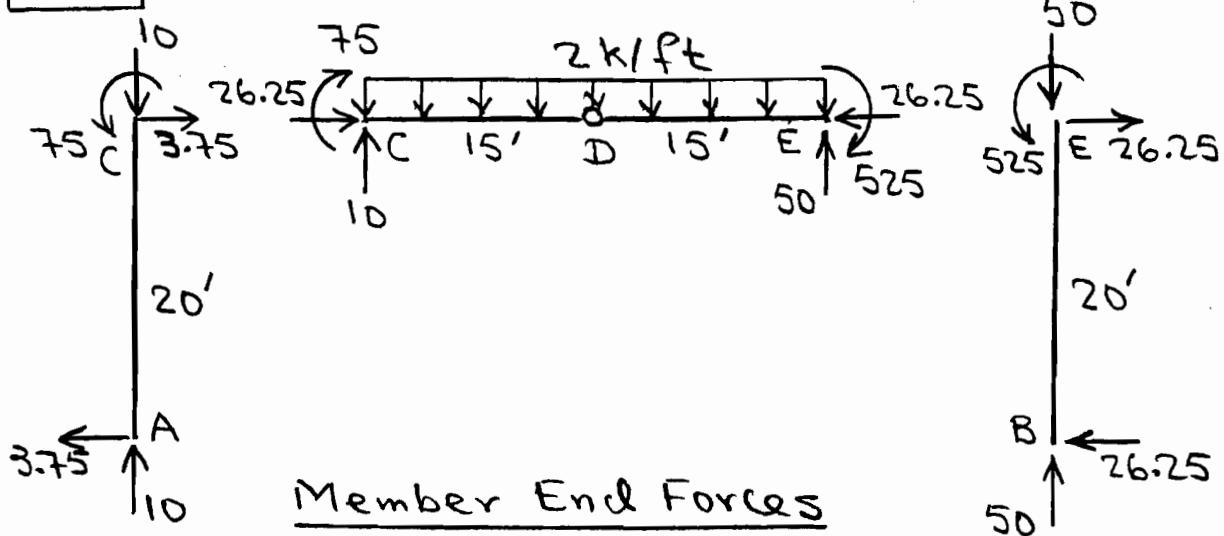


Axial Force Diagrams (k)

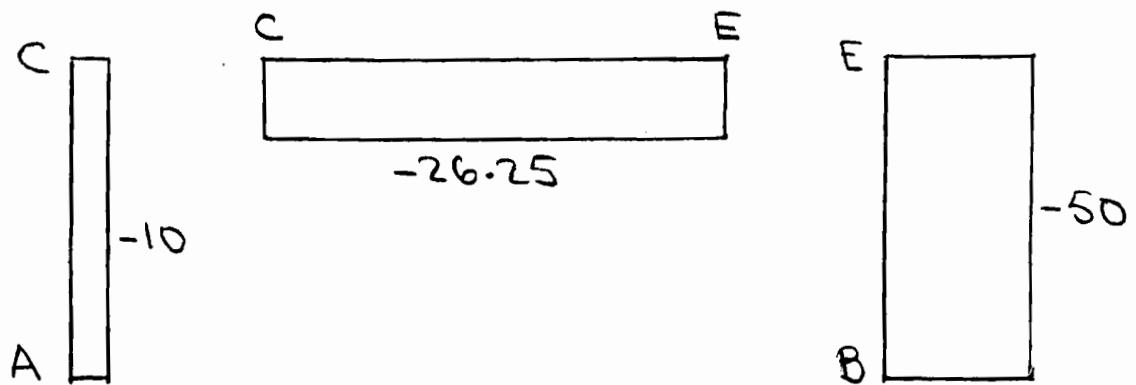


Qualitative Deflected Shape

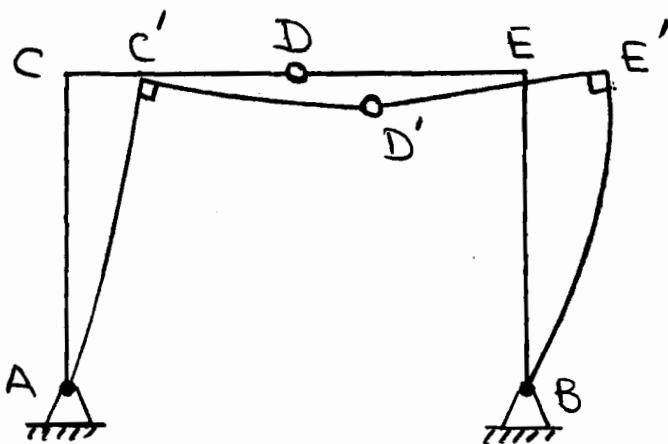
5.70



5.70 (contd.)

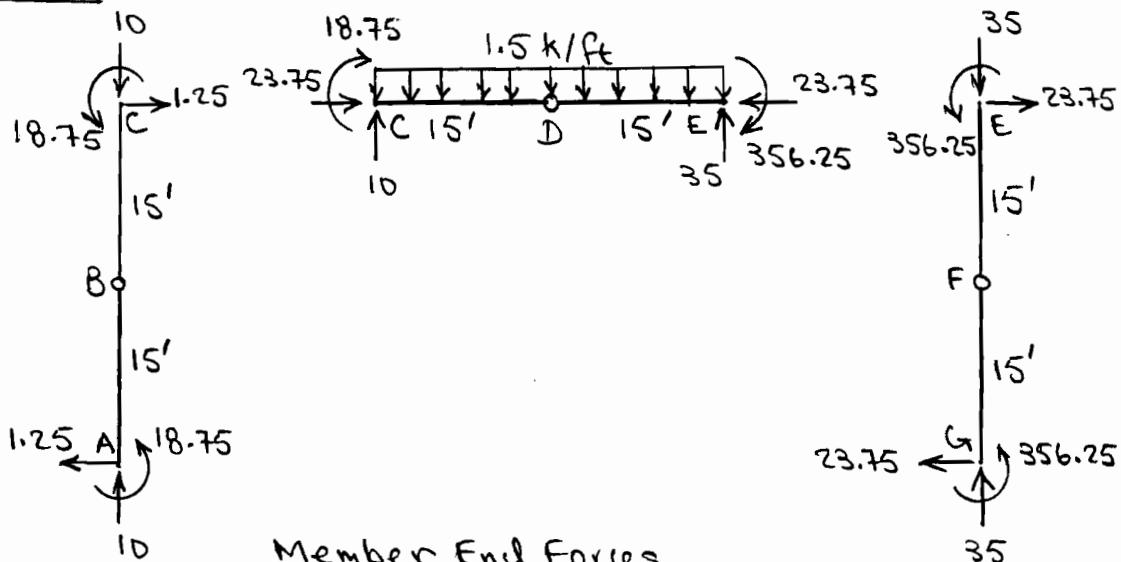


Axial Force Diagrams (k)

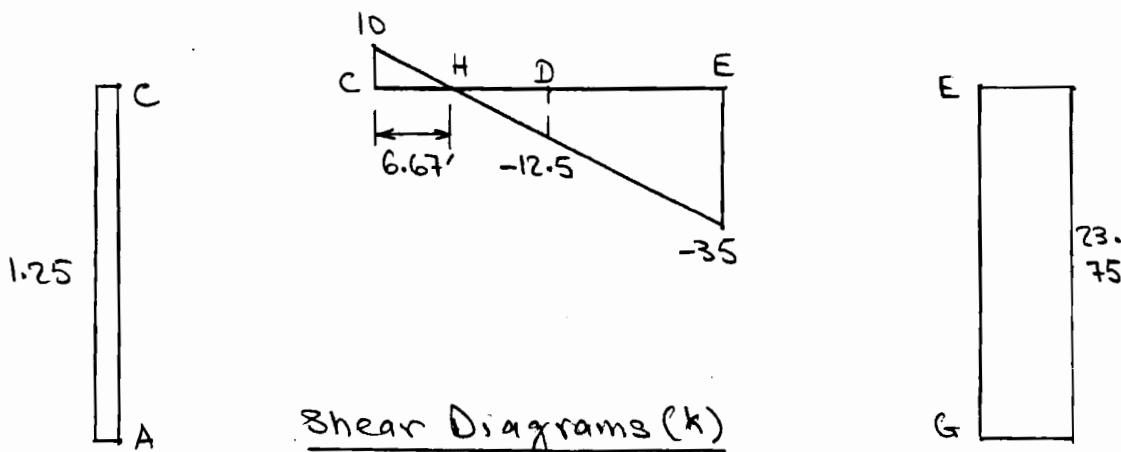


Qualitative Deflected Shape

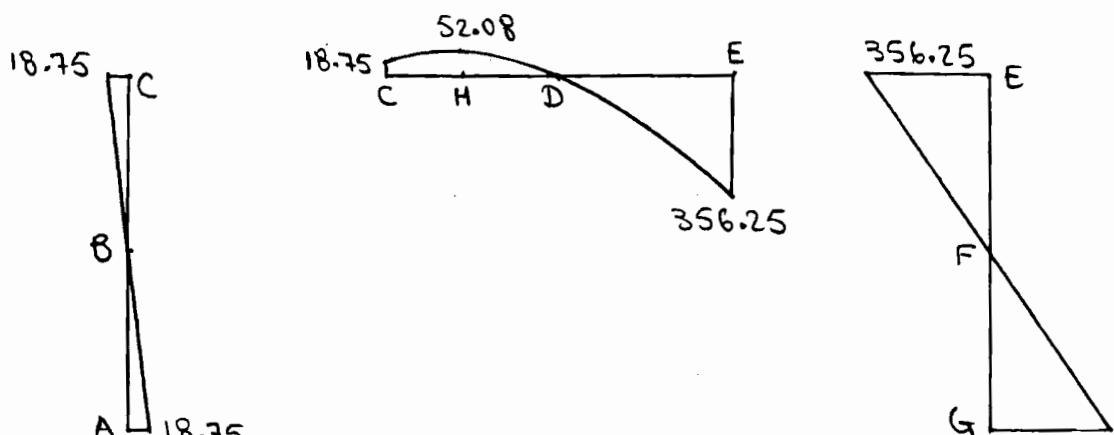
5.71



Member End Forces

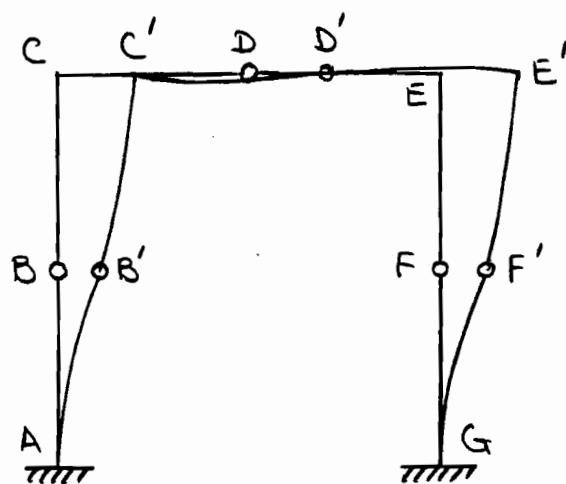
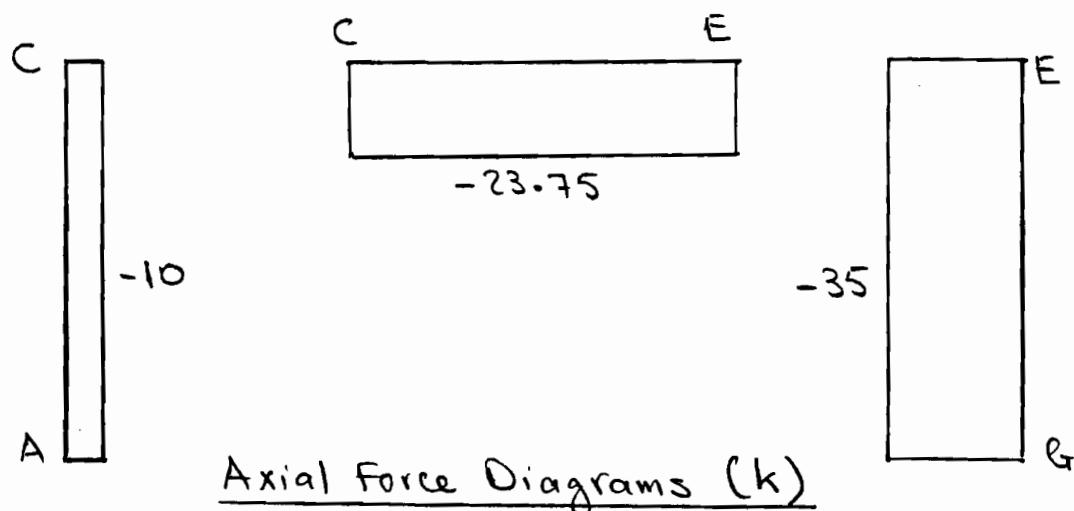


Shear Diagrams (k)



Bending Moment Diagrams (k-ft)

5.71 (contd.)



Qualitative Deflected Shape

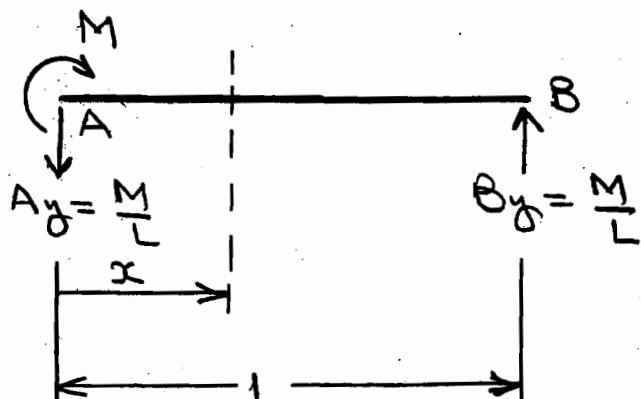
Chapter Six

Deflections of Beams:

Geometric Methods

CHAPTER 6

6.1



$$M_x = M - \frac{M}{L}(x) = \frac{M}{L}(L-x)$$

$$\frac{d^2y}{dx^2} = \frac{M}{EI} (L-x)$$

$$\Theta = \frac{dy}{dx} = \int \frac{M}{EI} (L-x) dx = \frac{M}{EI} \left(Lx - \frac{x^2}{2} \right) + C_1$$

$$\begin{aligned} y &= \int \left[\frac{M}{EI} \left(Lx - \frac{x^2}{2} \right) + C_1 \right] dx \\ &= \frac{M}{EI} \left(\frac{Lx^2}{2} - \frac{x^3}{6} \right) + C_1 x + C_2 \end{aligned}$$

Boundary Conditions:

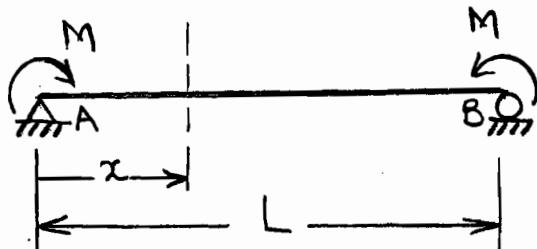
$$\text{at } x=0, y=0 \quad \therefore C_2 = 0$$

$$\text{at } x=L, y=0 \quad \therefore C_1 = -\frac{ML}{3EI}$$

$$\Theta = -\frac{M}{6EI} (3x^2 - 6Lx + 2L^2)$$

$$y = -\frac{M}{6EI} (x^3 - 3Lx^2 + 2L^2x)$$

6.2



$$M_x = M$$

$$\frac{d^2y}{dx^2} = \frac{M}{EI}$$

$$\Theta = \frac{dy}{dx} = \int \frac{M}{EI} dx = \frac{Mx}{EI} + C_1$$

$$y = \int \left(\frac{Mx}{EI} + C_1 \right) dx = \frac{Mx^2}{2EI} + C_1 x + C_2$$

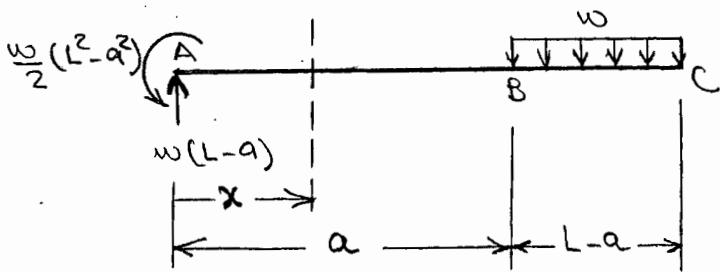
Boundary conditions:

$$\text{at } x=0, y=0 \quad \therefore C_2 = 0$$

$$\text{at } x=L, y=0 \quad \therefore C_1 = -\frac{ML}{2EI}$$

$$\underline{\Theta = \frac{M}{EI} \left(x - \frac{L}{2} \right)} ; \quad \underline{y = \frac{Mx}{2EI} (x-L)}$$

6.3



Segment AB: $0 \leq x \leq a$

$$M = -\frac{w}{2} (L^2 - a^2) + w(L-a)x$$

$$\frac{d^2y}{dx^2} = \frac{1}{EI} \left[-\frac{w}{2} (L^2 - a^2) + w(L-a)x \right]$$

$$EI\Theta = -\frac{w}{2} (L^2 - a^2)x + w(L-a)\frac{x^2}{2} + C_1$$

$$EIy = -\frac{w}{2} (L^2 - a^2) \frac{x^2}{2} + w(L-a) \frac{x^3}{6} + C_1x + C_2$$

Using the boundary conditions, $\Theta = 0$ and $y = 0$ at $x = 0$, we obtain $C_1 = 0$ and $C_2 = 0$

Thus,

$$\begin{aligned}\Theta &= \frac{w x}{2EI} [a^2 - L^2 + (L-a)x] \\ y &= \frac{w x^2}{2EI} \left[\frac{a^2 - L^2}{2} + \frac{(L-a)x}{3} \right]\end{aligned}$$

Segment BC: $a \leq x \leq L$

$$M = -\frac{w}{2} (L^2 - a^2) + w(L-a)x - \frac{w}{2} (x-a)^2$$

$$EI\Theta = -\frac{w}{2} (L^2 - a^2)x + w(L-a)\frac{x^2}{2} - \frac{w}{2} \left(\frac{x^3}{3} + a^2x - ax^2 \right) + C_3$$

$$\begin{aligned}EIy &= -\frac{w}{2} (L^2 - a^2) \frac{x^2}{2} + w(L-a) \frac{x^3}{6} - \frac{w}{2} \left(\frac{x^4}{12} + ax^2 - a\frac{x^3}{3} \right) \\ &\quad + C_3x + C_4\end{aligned}$$

By using the conditions that at $x=a$, $\Theta_{B,\text{Left}} = \Theta_{B,\text{Right}}$

and $y_{B,\text{Left}} = y_{B,\text{Right}}$, we obtain:

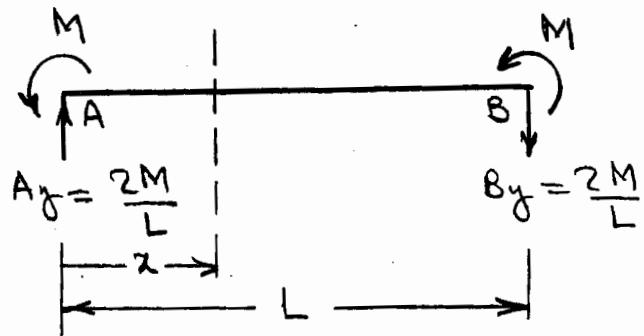
$$C_3 = \frac{w a^3}{6} \text{ and } C_4 = -\frac{w a^4}{24}$$

Thus

$$\Theta = \frac{w}{2EI} [xL(x-L) - \frac{x^3}{3} + \frac{a^3}{3}]$$

$$y = \frac{w}{2EI} [x^2 L (\frac{x}{3} - \frac{L}{2}) - \frac{x^4}{12} - \frac{a^4}{12} + \frac{a^3 x}{3}]$$

6.4



$$M_x = \frac{2M}{L}x - M$$

$$\frac{d^2y}{dx^2} = \frac{M}{EI} \left(\frac{2x}{L} - 1 \right)$$

$$\Theta = \frac{dy}{dx} = \int \frac{M}{EI} \left(\frac{2x}{L} - 1 \right) dx = \frac{Mx}{EI} \left(\frac{x}{L} - 1 \right) + C_1$$

$$y = \int \left[\frac{Mx}{EI} \left(\frac{x}{L} - 1 \right) + C_1 \right] dx = \frac{Mx^2}{EI} \left(\frac{x}{3L} - \frac{1}{2} \right) + C_1 x + C_2$$

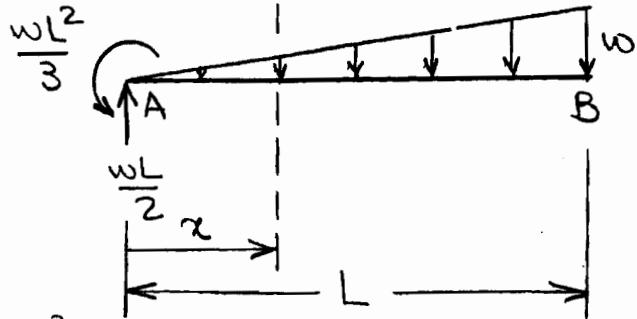
Boundary conditions:

$$\text{at } x=0, y=0 \quad \therefore C_2 = 0$$

$$\text{at } x=L, y=0 \quad \therefore C_1 = \frac{ML}{6EI}$$

$$\underline{\Theta = \frac{M}{6EI} (6x^2 - 6Lx + L^2)} ; \underline{y = \frac{Mx}{6EI} (2x^2 - 3Lx + L^2)}$$

6.5



$$M = -\frac{wL^2}{3} + \frac{wL}{2}x - \frac{1}{2}x\left(\frac{wL}{L}\right)\frac{x}{3}$$

$$EI \frac{d^2y}{dx^2} = -\frac{wL^2}{3} + \frac{wL}{2}x - \frac{wL^3}{6L}$$

$$EI\Theta = -\frac{wL^2}{3}x + \frac{wL}{2}\left(\frac{x^2}{2}\right) - \frac{w}{6L}\left(\frac{x^4}{4}\right) + C_1$$

$$EIy = -\frac{wL^2}{3}\left(\frac{x^2}{2}\right) + \frac{wL}{2}\left(\frac{x^3}{6}\right) + \frac{w}{6L}\left(\frac{x^5}{20}\right) + C_1x + C_2$$

Using the boundary conditions that at $x=0$,

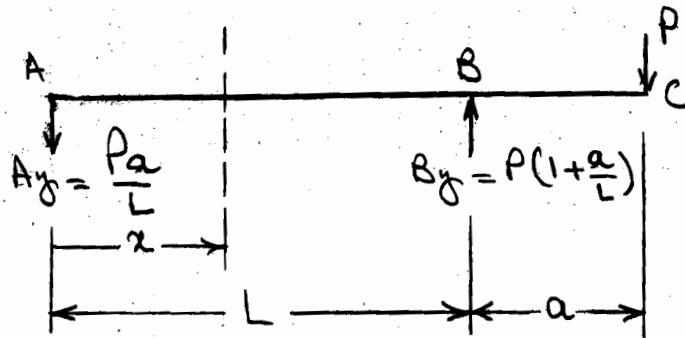
$\Theta=0$ and $y=0$, we obtain $C_1=0$ and $C_2=0$.

Thus,

$$\Theta = \frac{wx}{24EI} (-8L^3 + 6L^2x - x^3)$$

$$y = \frac{wx^2}{120EI} (-20L^3 + 10L^2x - x^3)$$

6.6



Segment AB: $0 \leq x \leq L$

$$M = -\frac{Pax}{L}$$

$$\frac{d^2y}{dx^2} = -\frac{Pax}{EI_L}$$

$$\Theta = -\frac{Pa}{EI_L} \left(\frac{x^2}{2}\right) + C_1$$

$$y = -\frac{Pa}{2EI_L} \left(\frac{x^3}{3}\right) + C_1x + C_2$$

Applying the boundary conditions, $y=0$ at $x=0$

and $y=0$ at $x=L$, we obtain

$$C_2 = 0 \quad C_1 = \frac{Pal}{6EI}$$

Thus,

$$\Theta = \frac{Pal}{6EI} \left(1 - 3\frac{x^2}{L^2}\right); \quad y = \frac{Palx}{6EI} \left(1 - \frac{x^2}{L^2}\right)$$

Segment BC: $L \leq x \leq (L+a)$

$$M = -P(L+a-x)$$

$$\frac{d^2y}{dx^2} = -\frac{P}{EI} (L+a-x)$$

$$\Theta = -\frac{P}{EI} \left(Lx + ax - \frac{x^2}{2}\right) + C_3$$

$$y = -\frac{P}{EI} \left(\frac{Lx^2}{2} + \frac{ax^2}{2} - \frac{x^3}{6}\right) + C_3x + C_4$$

6.6 (contd.)

By using the condition that at $x = L$,

$\Theta_{B,\text{Left}} = \Theta_{B,\text{Right}}$, we obtain

$$C_3 = \frac{PL}{EI} \left(\frac{L}{2} + \frac{2a}{3} \right)$$

Next, by applying the condition that at $x = L$, $y = 0$, we obtain

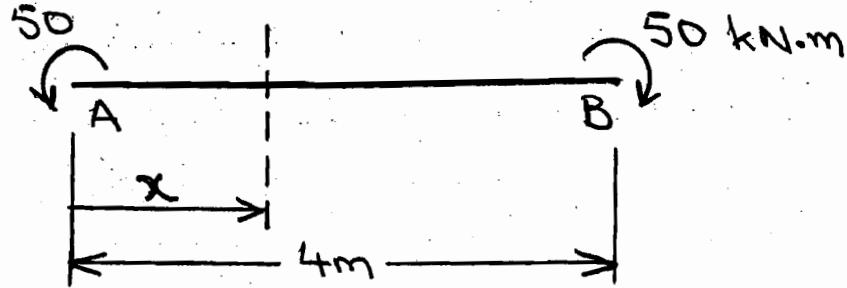
$$C_4 = -\frac{PL^2}{6EI} (L+a)$$

Thus,

$$\Theta = \frac{PL}{EI} \left[\frac{x^2}{2L} - x \left(1 + \frac{a}{L} \right) + \frac{L}{2} + \frac{2a}{3} \right]$$

$$y = \frac{PL}{EI} \left[\frac{x^3}{6L} - \frac{a^2}{2} \left(1 + \frac{a}{L} \right) + x \left(\frac{L}{2} + \frac{2a}{3} \right) - \frac{L}{6} (L+a) \right]$$

6.7



$$M_x = -50$$

$$\frac{d^2y}{dx^2} = -\frac{50}{EI}$$

$$\Theta = \frac{dy}{dx} = -\frac{50}{EI}x + C_1$$

$$y = -\frac{25}{EI}x^2 + C_1x + C_2$$

Boundary Conditions:

$$\text{at } x=0, \Theta=0 \therefore C_1=0$$

$$\text{at } x=0, y=0 \therefore C_2=0$$

Thus,

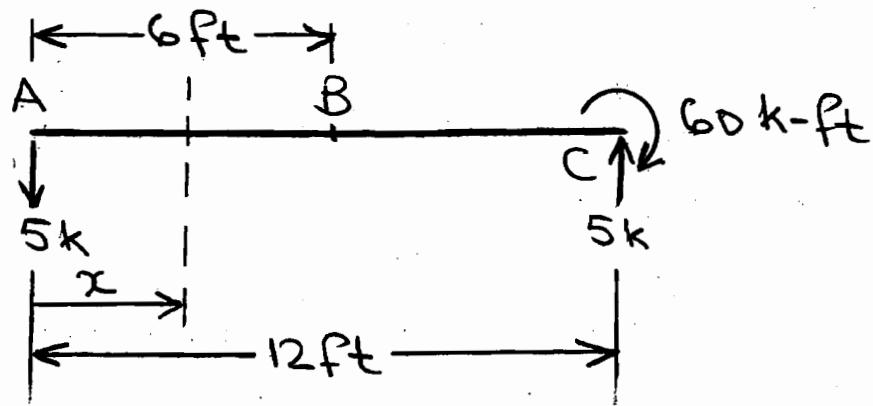
$$\Theta = -\frac{50x}{EI}; y = -\frac{25x^2}{EI}$$

At $x=4m$:

$$\Theta = -\frac{50(4)}{70(164)} = -0.0174 \text{ rad} = \underline{\underline{0.0174 \text{ rad}}} \quad \checkmark$$

$$y = -\frac{25(4)^2}{70(164)} = -0.0348 \text{ m} = \underline{\underline{34.8 \text{ mm} \downarrow}}$$

6.8



$$M_x = -5x$$

$$\frac{d^2y}{dx^2} = -\frac{5x}{EI}$$

$$\Theta = \frac{dy}{dx} = -\frac{5x^2}{2EI} + C_1$$

$$y = -\frac{5x^3}{6EI} + C_1 x + C_2$$

Boundary Conditions:

$$\text{at } x=0, y=0 \quad \therefore C_2=0$$

$$\text{at } x=12', y=0 \quad \therefore C_1 = \frac{120}{EI}$$

Thus,

$$\Theta = \frac{5}{2EI} (-x^2 + 48)$$

$$y = \frac{5x}{6EI} (-x^2 + 144)$$

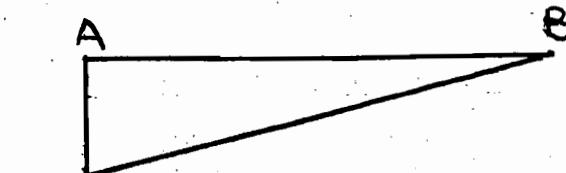
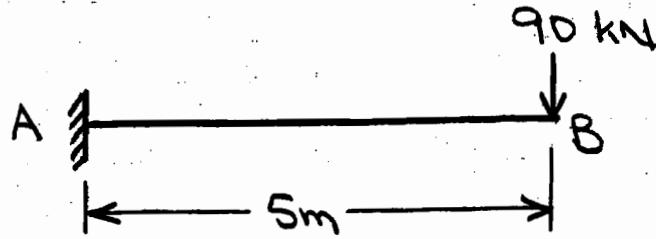
By substituting $x=6 \text{ ft}$, $E=10000(144) \text{ ksi}$

and $I = \frac{800}{(12)^4} \text{ ft}^4$, we obtain

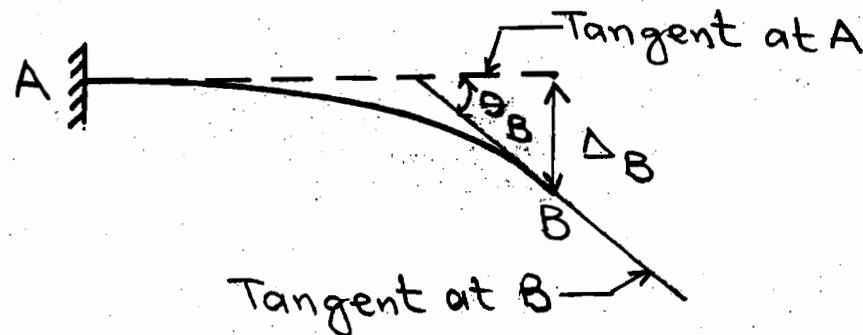
$$\Theta = 0.00054 \text{ rad } \leftarrow$$

$$y = 0.00972 \text{ ft} = 0.117 \text{ in. } \uparrow$$

6.9



$\frac{450}{EI}$ $\frac{M}{EI}$ Diagram ($\frac{kN.m}{EI}$)



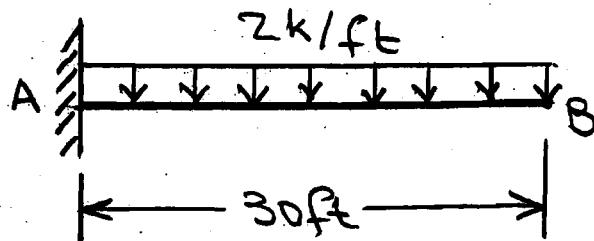
$$\Theta_B = \Theta_{BA} = \frac{1}{2}(5)\left(\frac{450}{EI}\right) = \frac{1125 \text{ kN.m}^2}{EI}$$

$$= \frac{1125}{200(800)} = 0.00703 \text{ rad } \checkmark$$

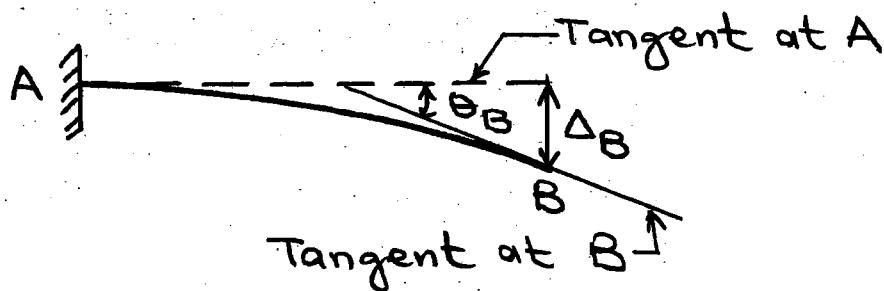
$$\Delta_B = \Delta_{BA} = \frac{1125}{EI} \left(\frac{10}{3}\right) = \frac{3750 \text{ kN.m}^3}{EI}$$

$$= \frac{3750}{200(800)} = 0.0234 \text{ m} = 23.4 \text{ mm } \downarrow$$

6.10



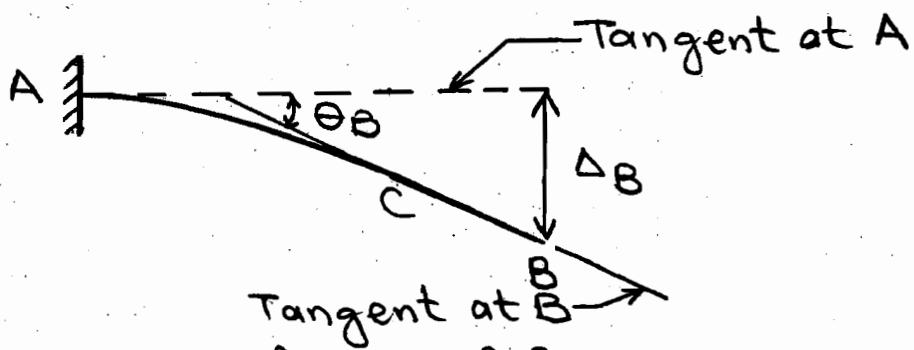
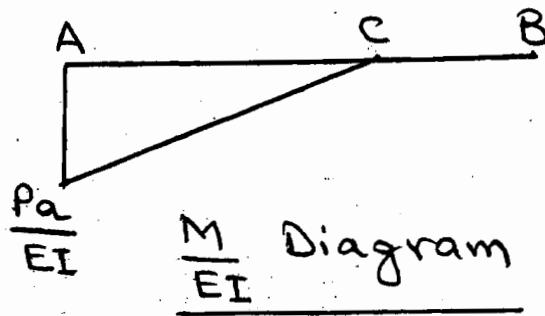
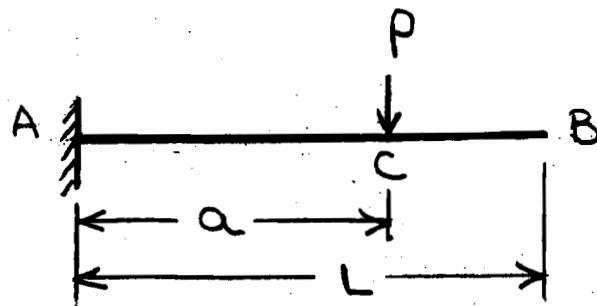
$$\frac{900}{EI} \quad \frac{M}{EI} \text{ Diagram } \left(\frac{k \cdot ft}{EI} \right)$$



$$\begin{aligned} \Theta_B &= \Theta_{BA} = \frac{1}{3}(30)\left(\frac{900}{EI}\right) = \frac{9000 \text{ k-ft}^2}{EI} \\ &= \frac{9000 (12)^2}{29000 (3000)} = 0.0149 \text{ rad} \end{aligned}$$

$$\begin{aligned} \Delta_B &= \Delta_{BA} = \frac{9000}{EI} \left(\frac{90}{4}\right) = \frac{202500 \text{ k-ft}^3}{EI} \\ &= \frac{202500 (12)^3}{29000 (3000)} = 4.022 \text{ in.} \downarrow \end{aligned}$$

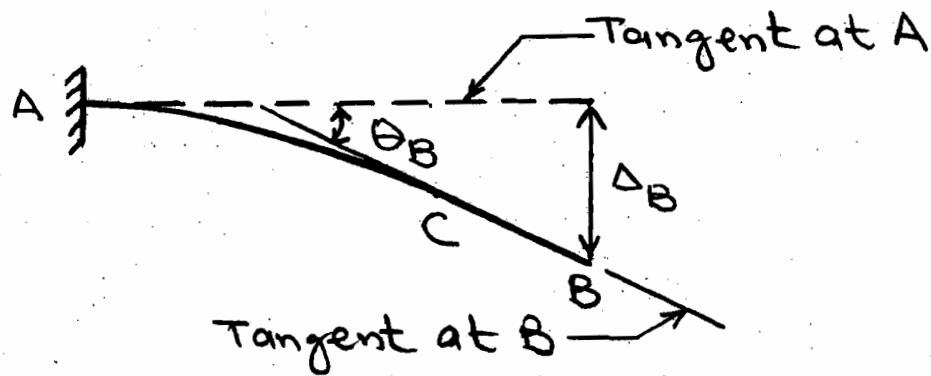
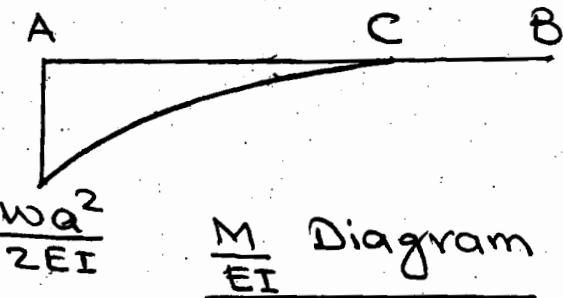
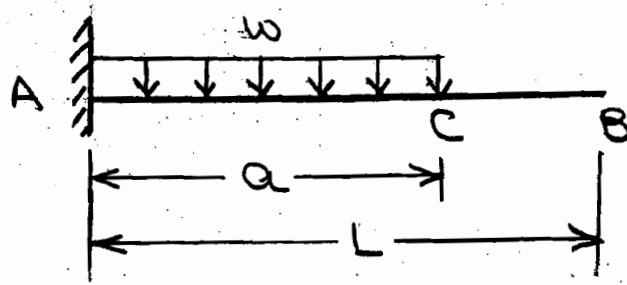
6.11



$$\Theta_B = \Theta_{BA} = \frac{1}{2}(a) \frac{Pa}{EI} = \frac{Pa^2}{2EI}$$

$$\Delta_B = \Delta_{BA} = \frac{Pa^2}{2EI} \left(L - \frac{a}{3}\right) = \frac{Pa^2}{6EI} (3L - a)$$

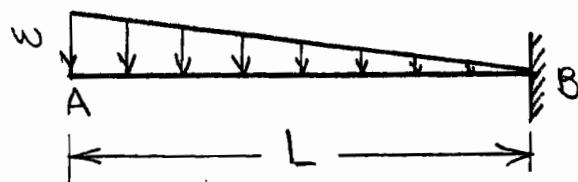
6.12



$$\Theta_B = \Theta_{BA} = \frac{1}{3}(a) \left(\frac{wq^2}{2EI} \right) = \frac{wqa^3}{6EI}$$

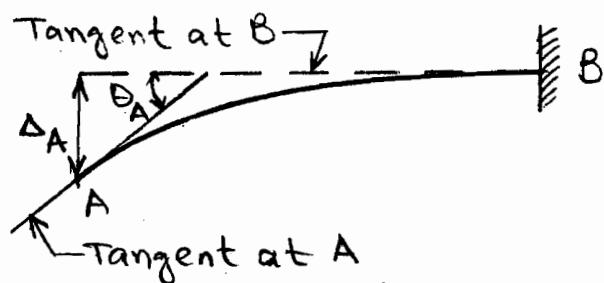
$$\Delta_B = \Delta_{BA} = \frac{wqa^3}{6EI} \left(L - \frac{a}{4} \right) = \frac{wqa^3}{24EI} (4L - a)$$

6.13



$$\frac{M}{EI} = \frac{wx^2}{2EI} \left(1 - \frac{x}{\frac{3L}{2}}\right)$$

$\frac{wL^2}{3EI}$

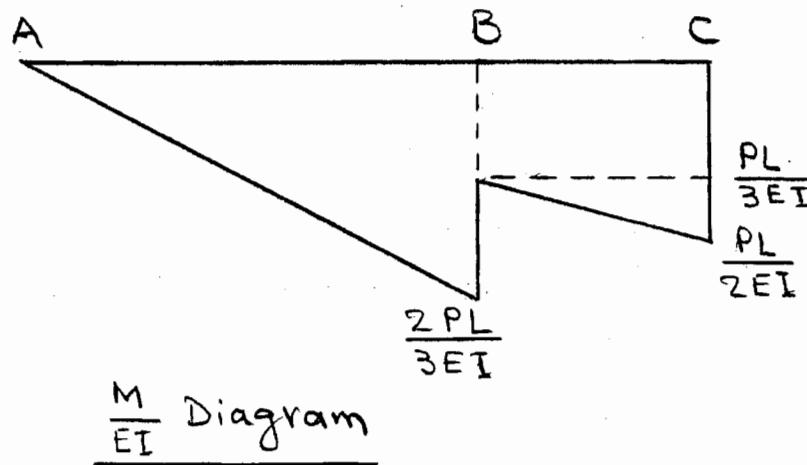
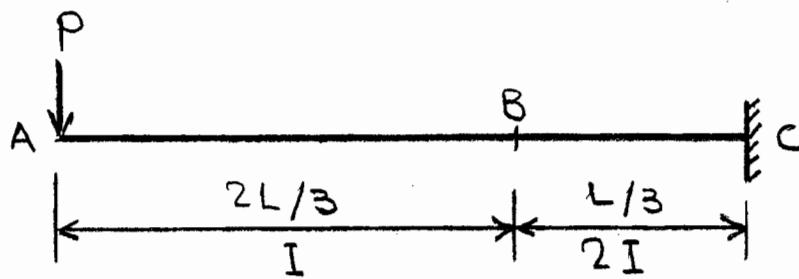


$$\Theta_A = \Theta_{AB} = \int_0^L \frac{M}{EI} dx = \int_0^L \frac{wx^2}{2EI} \left(1 - \frac{x}{\frac{3L}{2}}\right) dx = \frac{wL^3}{8EI}$$

$$\Delta_A = \Delta_{AB} = \int_0^L \frac{M}{EI}(x) dx = \int_0^L \frac{wx^2}{2EI} \left(1 - \frac{x}{\frac{3L}{2}}\right)x dx$$

$$= \frac{11}{120} \frac{wL^4}{EI}$$

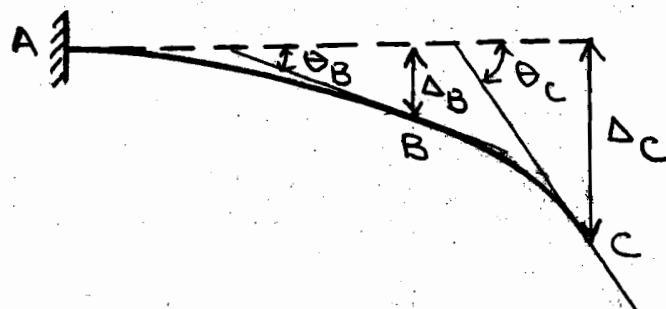
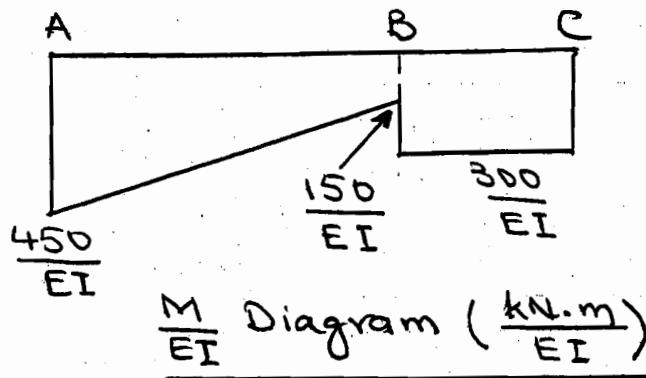
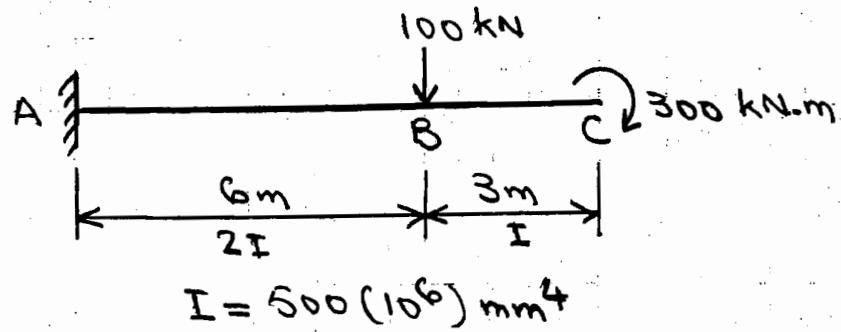
6.14



$$\Theta_A = \Theta_{Ac} = \frac{1}{2} \left(\frac{2L}{3} \right) \left(\frac{2PL}{3EI} \right) + \frac{L}{3} \left(\frac{PL}{3EI} \right) + \frac{1}{2} \left(\frac{L}{3} \right) \left(\frac{PL}{6EI} \right) = \underline{\underline{\frac{13PL^2}{36EI}}}$$

$$\Delta_A = \Delta_{Ac} = \frac{2PL^2}{9EI} \left(\frac{4L}{9} \right) + \frac{PL^2}{9EI} \left(\frac{5L}{6} \right) + \frac{PL^2}{36EI} \left(\frac{8L}{9} \right) = \underline{\underline{\frac{35PL^3}{162EI}}}$$

6.15



$$\Theta_B = \Theta_{BA} = \frac{1}{EI} [150(6) + \frac{1}{2}(300)(6)] = \frac{1800 \text{ kN}\cdot\text{m}^2}{EI}$$

$$= \frac{1800}{70(500)} = 0.0514 \text{ rad } \checkmark$$

$$\Delta_B = \Delta_{BA} = \frac{1}{EI} [150(6)3 + \frac{1}{2}(300)(6)4] = \frac{6300 \text{ kN}\cdot\text{m}^3}{EI}$$

$$= \frac{6300}{70(500)} = 0.18 \text{ m} = 180 \text{ mm } \downarrow$$

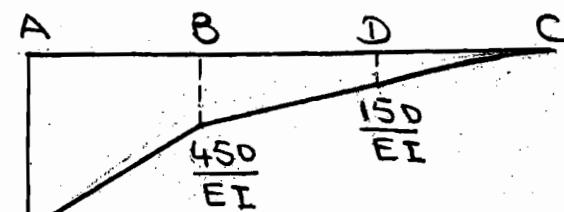
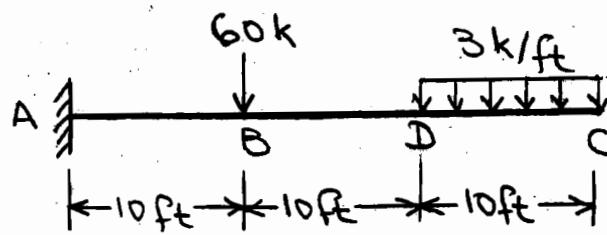
$$\Theta_C = \Theta_{CA} = \frac{1}{EI} [1800 + 300(3)] = \frac{2700 \text{ kN}\cdot\text{m}^2}{EI}$$

$$= \frac{2700}{70(500)} = 0.0771 \text{ rad } \checkmark$$

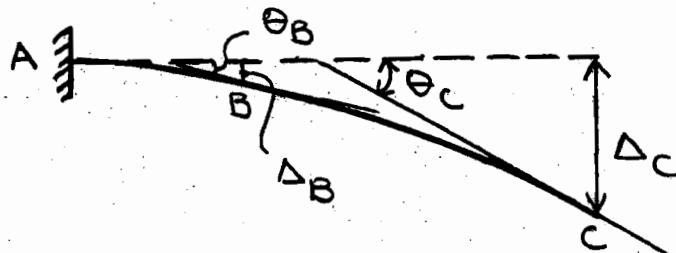
$$\Delta_C = \Delta_{CA} = \frac{1}{EI} [150(6)6 + \frac{1}{2}(300)(6)7 + 300(3)(1.5)]$$

$$= \frac{13050 \text{ kN}\cdot\text{m}^3}{EI} = \frac{13050}{70(500)} = 0.373 \text{ m} = 373 \text{ mm } \downarrow$$

6.16



M/EI Diagram ($\frac{k \cdot ft}{EI}$)



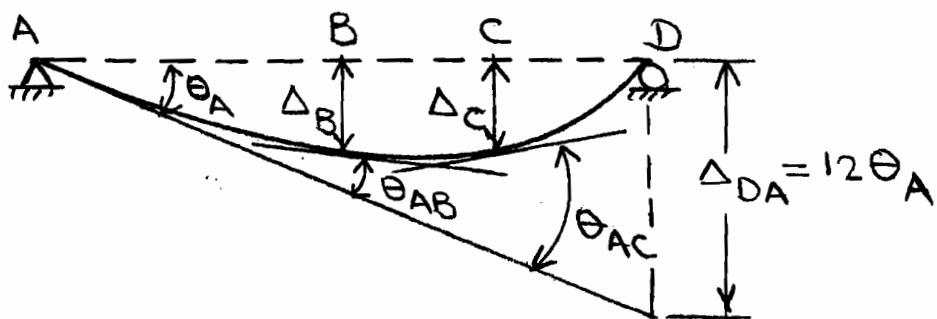
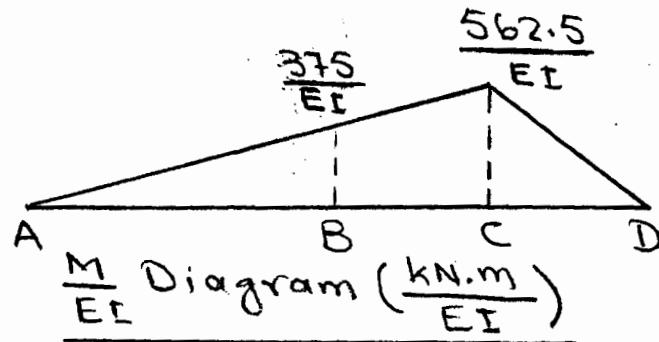
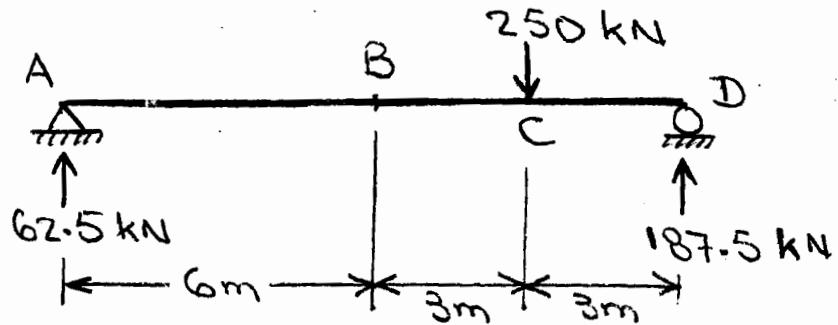
$$\begin{aligned}\theta_B &= \theta_{BA} = \frac{1}{EI} [450(10) + \frac{1}{2}(900)10] = \frac{9000 k \cdot ft^2}{EI} \\ &= \frac{9000 (12)^2}{29000 (4000)} = 0.0112 \text{ rad} \downarrow\end{aligned}$$

$$\begin{aligned}\Delta_B &= \Delta_{BA} = \frac{1}{EI} [450(10)5 + \frac{1}{2}(900)(10)(\frac{20}{3})] = \frac{52500 k \cdot ft^3}{EI} \\ &= \frac{52500 (12)^3}{29000 (4000)} = 0.782 \text{ in.} \downarrow\end{aligned}$$

$$\begin{aligned}\theta_C &= \theta_{CA} = \frac{1}{EI} [9000 + 150(10) + \frac{1}{2}(300)10) + \frac{1}{3}(150)10] \\ &= \frac{12500 k \cdot ft^2}{EI} = \frac{12500 (12)^2}{29000 (4000)} = 0.0155 \text{ rad} \downarrow\end{aligned}$$

$$\begin{aligned}\Delta_C &= \Delta_{CA} = \frac{1}{EI} [450(10)25 + \frac{1}{2}(900)10(\frac{20}{3} + 20) \\ &\quad + 150(10)15 + \frac{1}{2}(300)10(\frac{20}{3} + 10) + \frac{1}{3}(150)10(\frac{30}{4})] \\ &= \frac{283750 k \cdot ft^3}{EI} = \frac{283750 (12)^3}{29000 (4000)} = 4.227 \text{ in.} \downarrow\end{aligned}$$

6.17



$$\Delta_{DA} = \frac{1}{EI} \left[\frac{1}{2} (562.5) 9(6) + \frac{1}{2} (562.5) 3(2) \right] = \frac{16875}{EI}$$

$$\theta_A = \frac{\Delta_{DA}}{12} = \frac{1406.25}{EI}$$

$$\begin{aligned}\theta_B &= \theta_A - \theta_{AB} = \frac{1}{EI} \left[1406.25 - \frac{1}{2} (375) 6 \right] \\ &= \frac{281.25 \text{ kN.m}^2}{EI} = \frac{281.25}{200(462)} = \underline{\underline{0.00304 \text{ rad}}} \end{aligned}$$

$$\begin{aligned}\Delta_B &= 6\theta_A - \Delta_{BA} = \frac{1}{EI} \left[6(1406.25) - \frac{1}{2} (375) 6(2) \right] \\ &= \frac{6187.5 \text{ kN.m}^3}{EI} = \frac{6187.5}{200(462)} = 0.067 \text{ m} \end{aligned}$$

$$\underline{\underline{\Delta_B = 67 \text{ mm} \downarrow}}$$

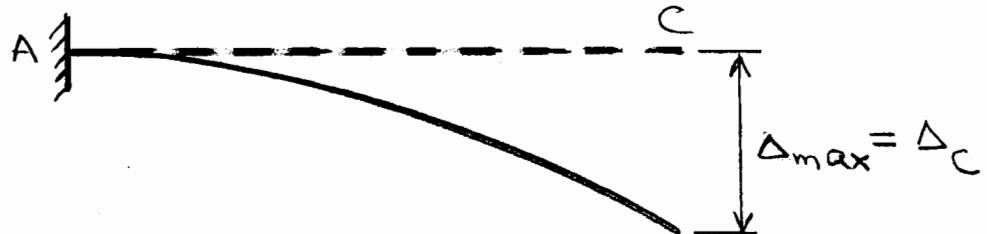
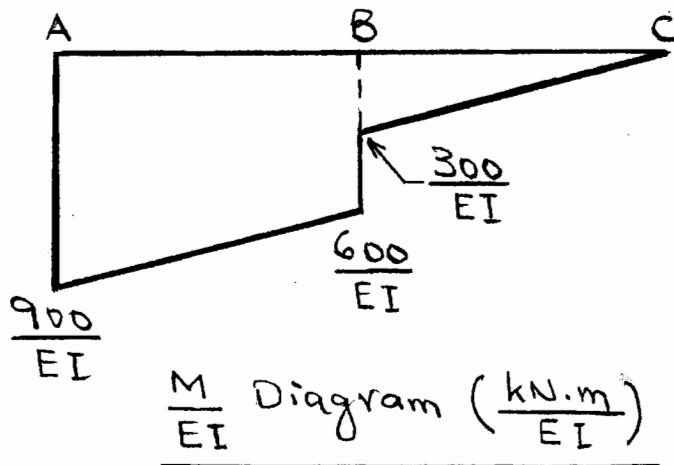
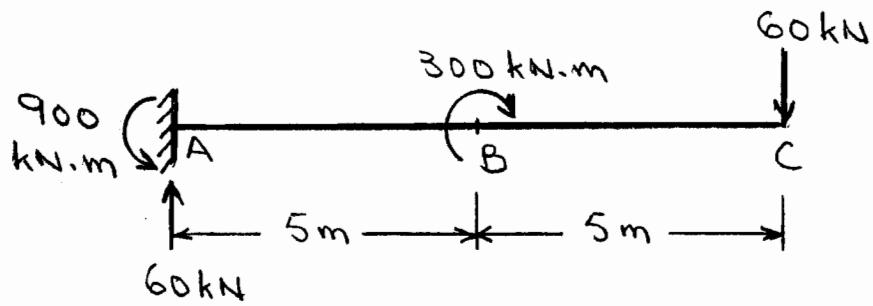
6.17 (contd.)

$$\Theta_C = \Theta_{AC} - \Theta_A = \frac{1}{EI} \left[\frac{1}{2} (562.5) 9 - 1406.25 \right]$$
$$= \frac{1125 \text{ kN}\cdot\text{m}^2}{EI} = \frac{1125}{200(462)} = \underline{0.0122 \text{ rad}} \quad \checkmark$$

$$\Delta_C = 9\Theta_A - \Delta_{CA} = \frac{1}{EI} \left[9(1406.25) - \frac{1}{2} (562.5) 9(3) \right]$$
$$= \frac{5062.5 \text{ kN}\cdot\text{m}^3}{EI} = \frac{5062.5}{200(462)} = \underline{0.0548 \text{ m}}$$

$$\underline{\Delta_C = 54.8 \text{ mm} \downarrow}$$

6.18



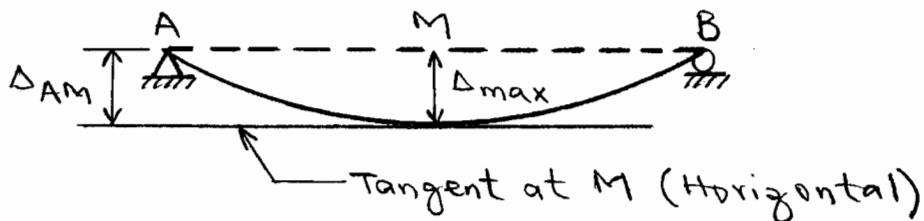
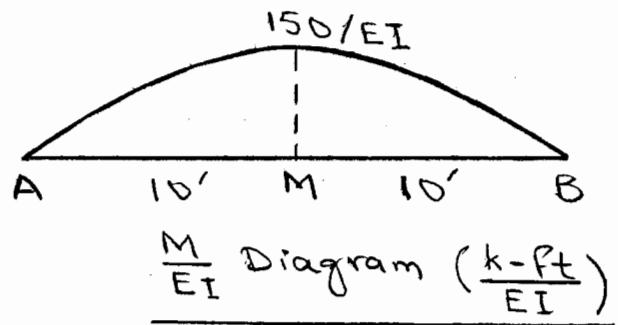
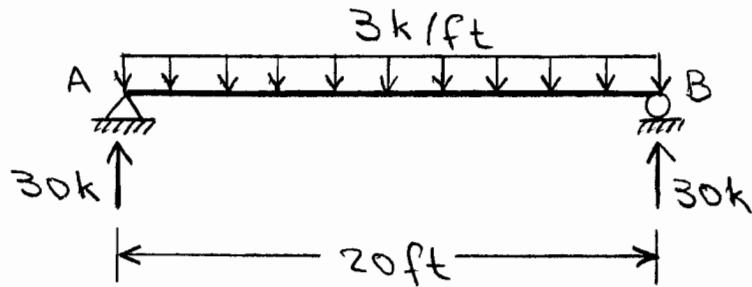
$$\Delta_{max} = \Delta_C = \Delta_{CA} = \frac{1}{EI} \left[600(5)(7.5) + \frac{1}{2}(300)(5)\left(\frac{25}{3}\right) + \frac{1}{2}(300)(5)\left(\frac{10}{3}\right) \right] = \frac{31250 \text{ kN} \cdot \text{m}^3}{EI}$$

$$\Delta_{max} = \frac{L}{360}$$

$$\frac{31250}{200(10^6)I} = \frac{10}{360}$$

$$\text{From which, } I = 5625(10^{-6}) \text{ m}^4 = \underline{\underline{5625(10^6) \text{ mm}^4}}$$

6.19



The maximum deflection occurs at the beam midspan, M, where the tangent is horizontal.

$$\begin{aligned}\Delta_{\max} &= \Delta_{AM} = \frac{1}{EI} \left[\frac{2}{3} (150) (10) \left(\frac{50}{8} \right) \right] \\ &= \frac{6250 \text{ k-ft}^3}{EI}\end{aligned}$$

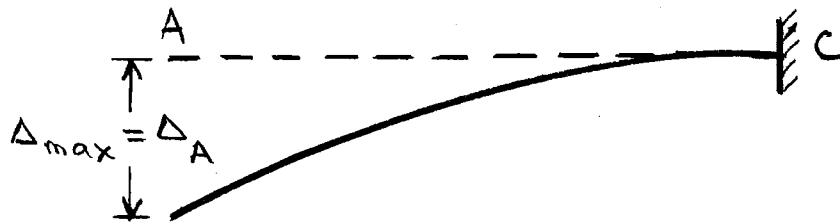
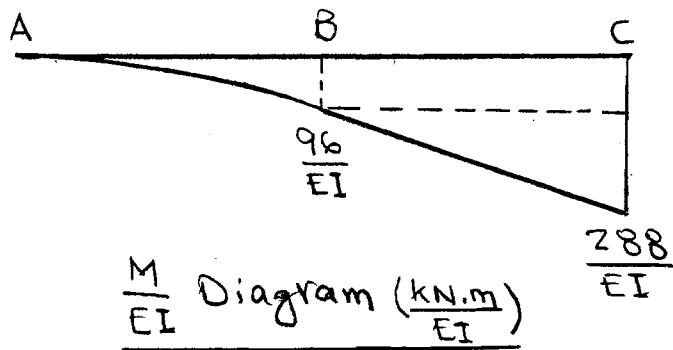
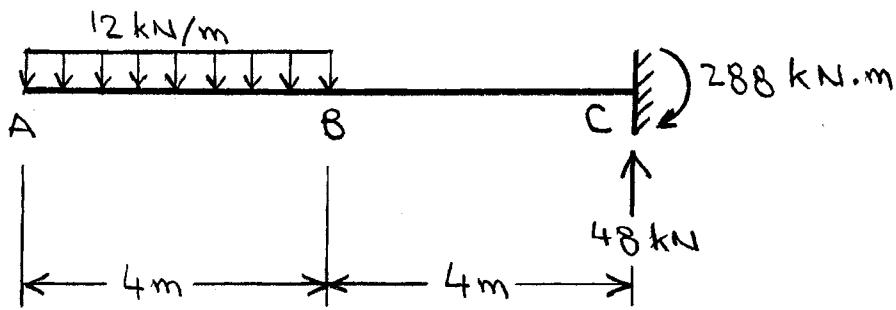
$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{6250(12)^3}{29000(I)} = \frac{20(12)}{360}$$

from which,

$$I = 559 \text{ in}^4$$

6.20



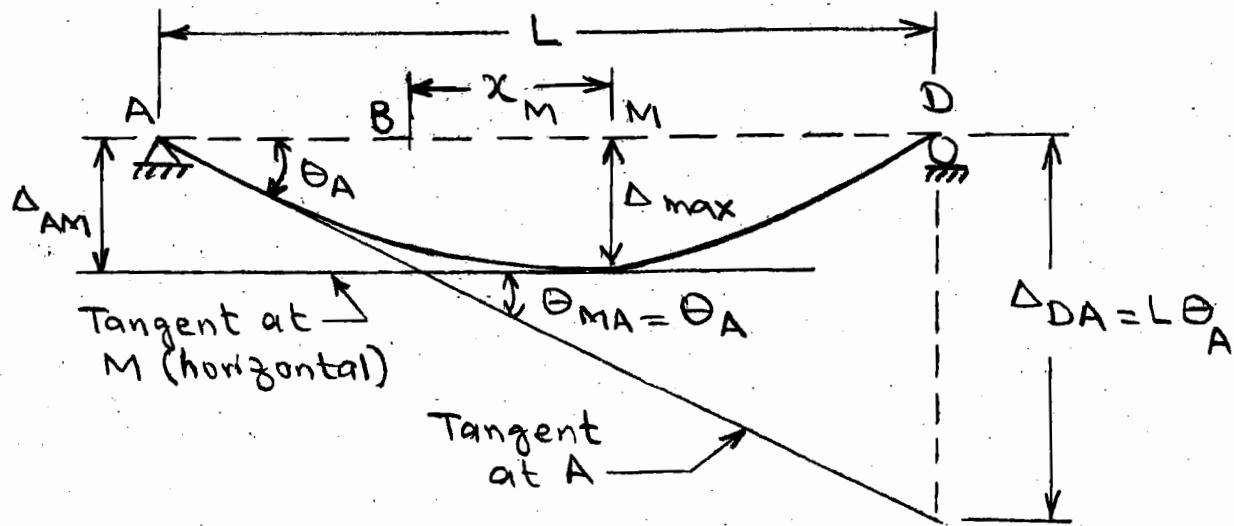
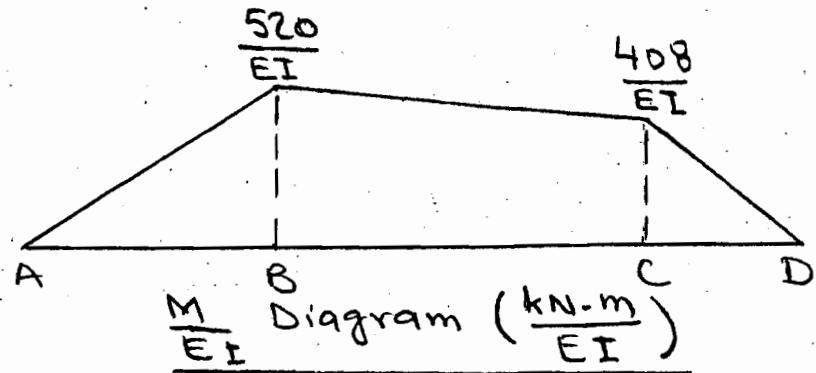
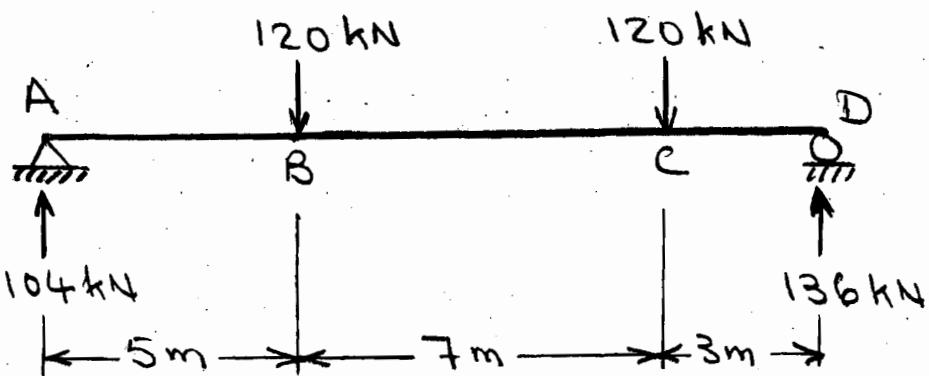
$$\Delta_{\max} = \Delta_A = \Delta_{AC} = \frac{1}{EI} \left[\frac{1}{3} (96)(4)(3) + 96(4)(6) + \frac{1}{2} (192)4 \left(\frac{8}{3} + 4 \right) \right] = \frac{5248 \text{ kN} \cdot \text{m}^3}{EI}$$

$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{5248}{70(10^6)(I)} = \frac{8}{360}$$

from which, $I = 3374(10^6) \text{ m}^4 = \underline{\underline{3374(10^6) \text{ mm}^4}}$

6.21



$$\begin{aligned}\Delta_{DA} &= \frac{1}{EI} \left[\frac{1}{2} (520)(5) \left(\frac{5}{3} + 10 \right) + (408)(7)(6.5) \right. \\ &\quad \left. + \frac{1}{2} (112)(7) \left(\frac{14}{3} + 3 \right) + \frac{1}{2} (408)(3)(2) \right] \\ &= \frac{37960 \text{ kN.m}^3}{EI}\end{aligned}$$

$$\Theta_A = \frac{\Delta_{DA}}{L} = \frac{37960}{15(EI)} = \frac{2530.67 \text{ kN.m}^2}{EI}$$

6.21 (contd.)

If the maximum deflection occurs at M, then

$$\theta_{MA} = \theta_A$$

$$\frac{1}{EI} \left[\frac{1}{2}(520)(5) + 520x_M - \frac{1}{2} \left(\frac{112}{7} x_M \right) x_M \right] = \frac{2530.67}{EI}$$

$$8x_M^2 - 520x_M + 1230.67 = 0$$

$$x_M = 2.46 \text{ m}$$

$$\Delta_{\max} = \Delta_{AM} = \frac{1}{EI} \left[\frac{1}{2}(520)(5) \left(\frac{10}{3} \right) \right]$$

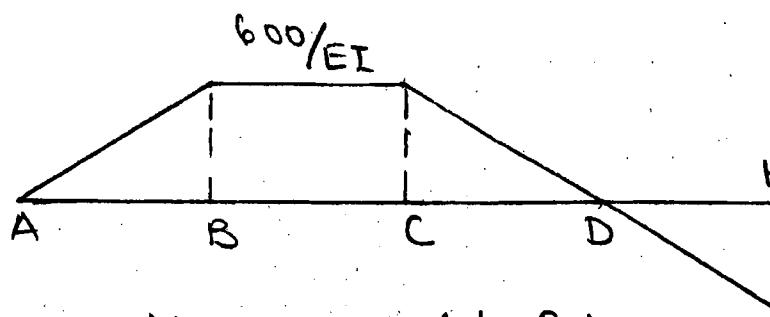
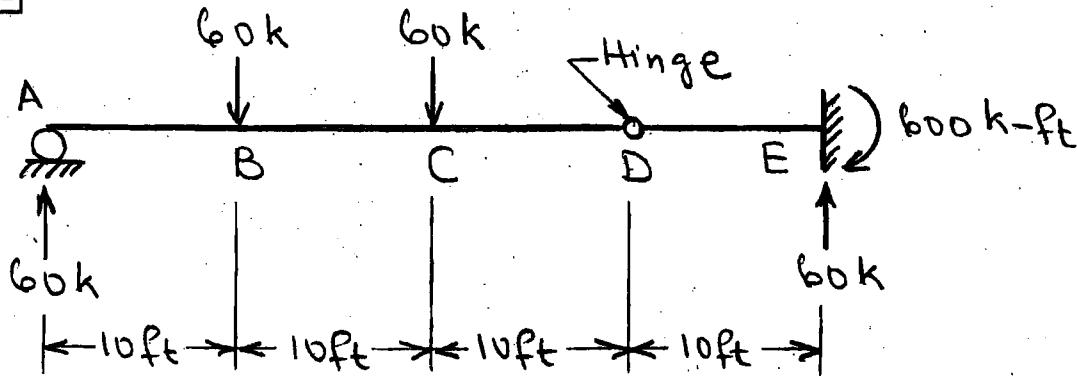
$$+ 520(2.46) \left(\frac{2.46}{2} + 5 \right) - \frac{1}{2} \left(\frac{112}{7} \right) (2.46)(2.46)(6.64) \right] \\ = \frac{11981.29 \text{ kN.m}^3}{EI}$$

$$\Delta_{\max.} = \frac{L}{360}$$

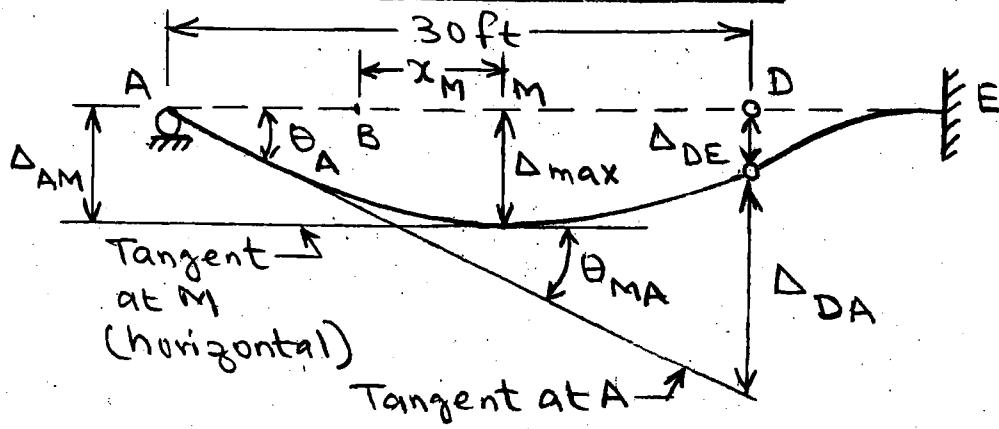
$$\frac{11981.29}{30(10^6)I} = \frac{15}{360}$$

$$\text{from which } I = 9585(10^6) \text{ m}^4 = \underline{\underline{9585(10^6) \text{ mm}^4}}$$

6.22



$\frac{M}{EI}$ Diagram ($\frac{k \cdot ft}{EI}$)



$$\Delta_{DA} = \frac{1}{EI} \left[\frac{1}{2} (600)(10) \left(\frac{10}{3} + 20 \right) + 600(10)(15) \right]$$

$$+ \frac{1}{2} (600)(10) \left(\frac{20}{3} \right) \right] = \frac{180000 \text{ k-ft}^3}{EI}$$

$$\Delta_{DE} = \frac{1}{EI} \left[\frac{1}{2} (600)(10) \left(\frac{20}{3} \right) \right] = \frac{20000 \text{ k-ft}^3}{EI}$$

$$\Theta_A = \frac{\Delta_{DA} + \Delta_{DE}}{30} = \frac{180000 + 20000}{30} = \frac{6666.67}{EI}$$

6.22 (contd.)

If the maximum deflection occurs at M, then

$$\Theta_{MA} = \Theta_A$$

$$\frac{L}{EI} \left[\frac{1}{2} (600)(10) + 600 x_M \right] = \frac{6666.67}{EI}$$

$$x_M = 6.11 \text{ ft}$$

$$\begin{aligned} \Delta_{max} &= \Delta_{AM} = \frac{L}{EI} \left[\frac{1}{2} (600)(10) \left(\frac{20}{3} \right) \right. \\ &\quad \left. + (600)(6.11)(13.06) \right] = \frac{67878 \text{ k-ft}^3}{EI} \end{aligned}$$

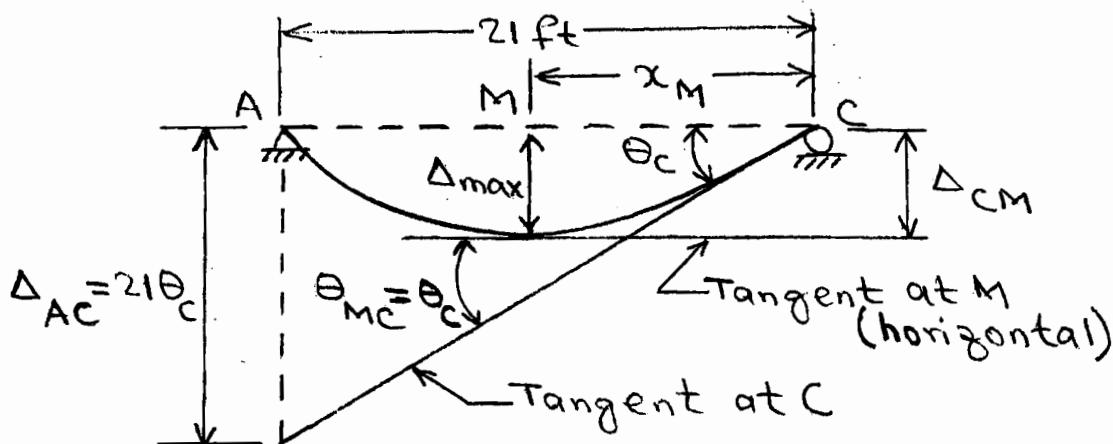
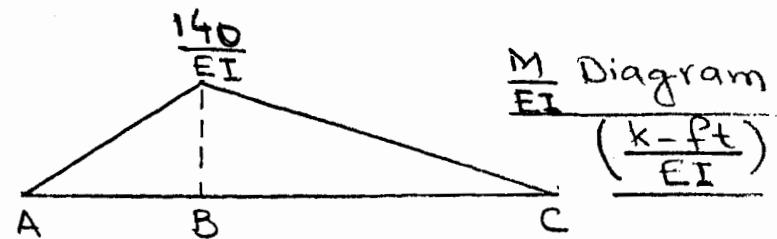
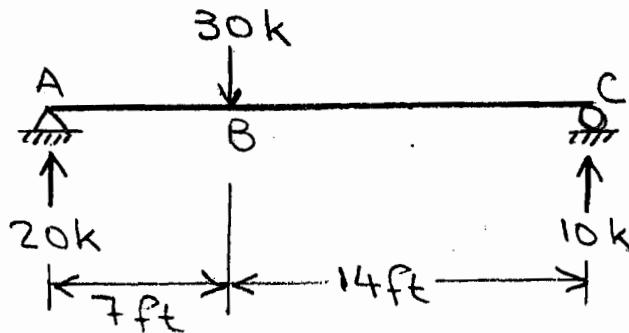
$$\Delta_{max} = \frac{L}{360}$$

$$\frac{67878 (12)^3}{29000 (I)} = \frac{40(12)}{360}$$

from which

$$I = 3033 \text{ in}^4$$

6.23



$$\Delta_{AC} = \frac{1}{EI} \left[\frac{1}{2} (140) 7 + \left(\frac{14}{3} \right) + \frac{1}{2} (140) 14 \left(7 + \frac{14}{3} \right) \right] = \frac{13720}{EI}$$

$$\theta_c = \frac{\Delta_{AC}}{21} = \frac{653.33}{EI}$$

$$\theta_{MC} = \theta_c$$

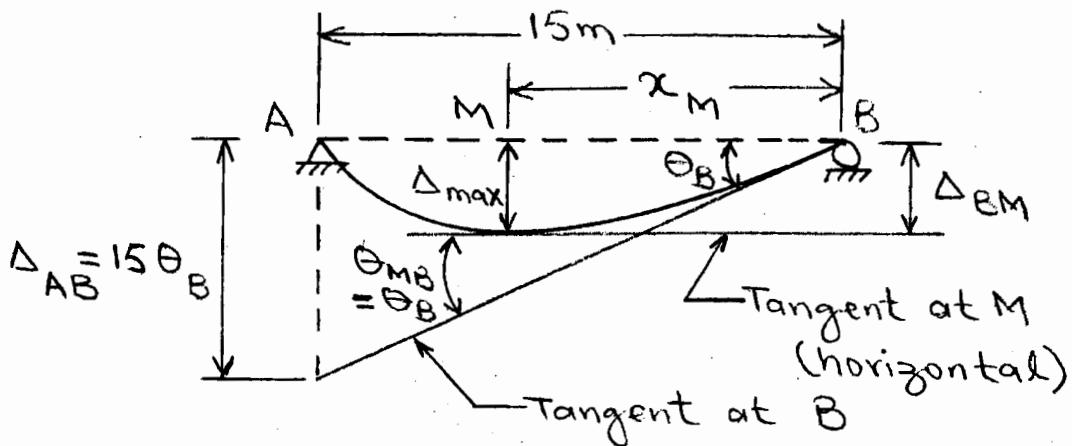
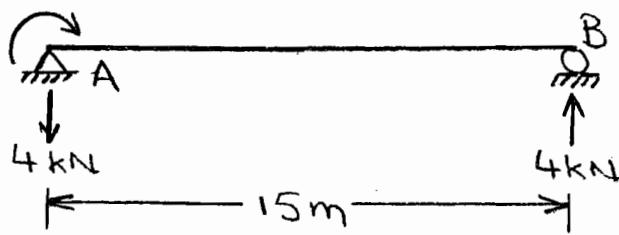
$$\frac{1}{EI} \left[\frac{1}{2} \left(\frac{140}{14} x_M \right) x_M \right] = \frac{653.33}{EI}$$

from which, $x_M = 11.43 \text{ ft}$

$$\begin{aligned} \Delta_{\max} &= \Delta_{CM} = \frac{1}{EI} \left[\frac{1}{2} (11.43) (11.43) \frac{2}{3} (11.43) \right] \\ &= \frac{4978.81 \text{ k-ft}^3}{EI} = \frac{4978.81 (12)^3}{10000 (500)} = \underline{1.72 \text{ in.} \downarrow} \end{aligned}$$

6.24

60 kN.m



$$\Delta_{AB} = \frac{1}{EI} \left[\frac{1}{2} (60) 15 (5) \right] = \frac{2250}{EI}$$

$$\theta_B = \frac{\Delta_{AB}}{15} = \frac{150}{EI}$$

$$\theta_{MB} = \theta_B$$

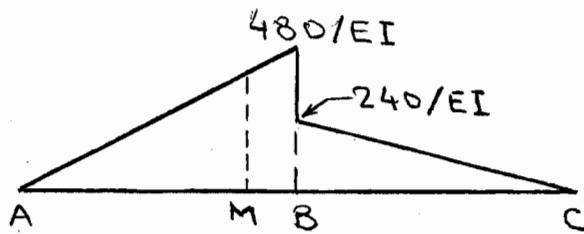
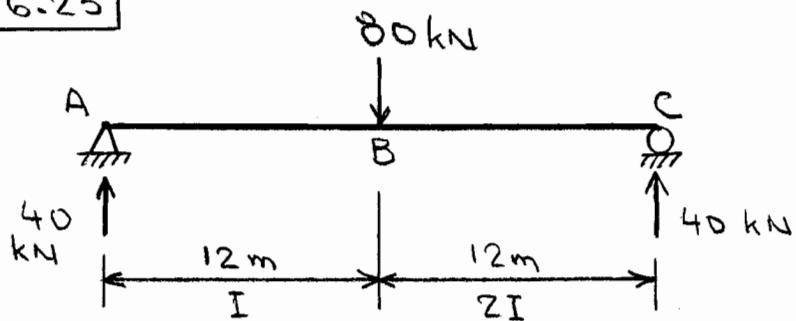
$$\frac{1}{EI} \left[\frac{1}{2} \left(\frac{60}{15} x_M \right) x_M \right] = \frac{150}{EI}$$

from which $x_M = 8.66 \text{ m}$

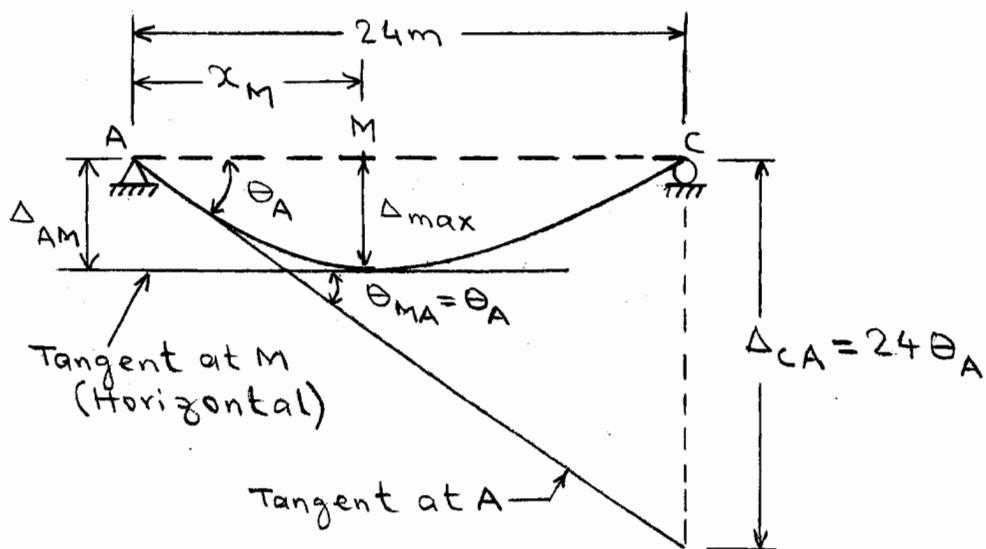
$$\begin{aligned} \Delta_{\max} &= \Delta_{BM} = \frac{1}{EI} \left[\frac{1}{2} (34.64)(8.66) \frac{2}{3} (8.66) \right] \\ &= \frac{866 \text{ kN} \cdot \text{m}^3}{EI} = \frac{866}{70(712)} = 0.0174 \text{ m} \end{aligned}$$

$$\underline{\Delta_{\max} = 17.4 \text{ mm} \downarrow}$$

6.25



M/EI Diagram ($\frac{\text{kN}\cdot\text{m}}{\text{EI}}$)



$$\Delta_{CA} = \frac{1}{EI} \left[\frac{1}{2} (480)(12)(16) + \frac{1}{2} (240)(12)(8) \right] = \frac{57600 \text{ kN}\cdot\text{m}^3}{EI}$$

$$\Theta_A = \frac{\Delta_{CA}}{24} = \frac{57600}{24(EI)} = \frac{2400 \text{ kN}\cdot\text{m}^2}{EI}$$

If the maximum deflection occurs at M, then

$$\Theta_{MA} = \Theta_A$$

6.25 (contd.)

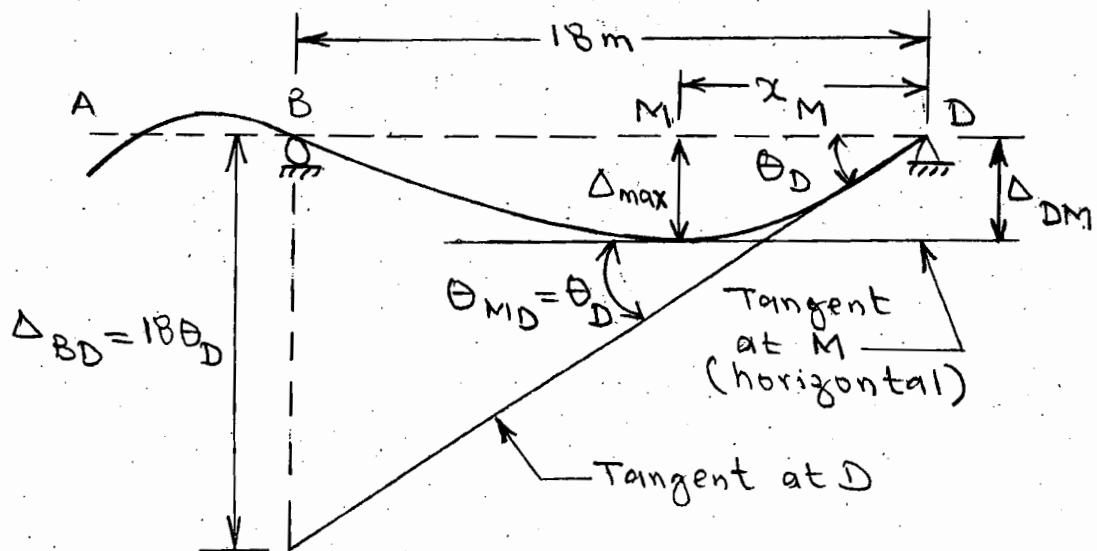
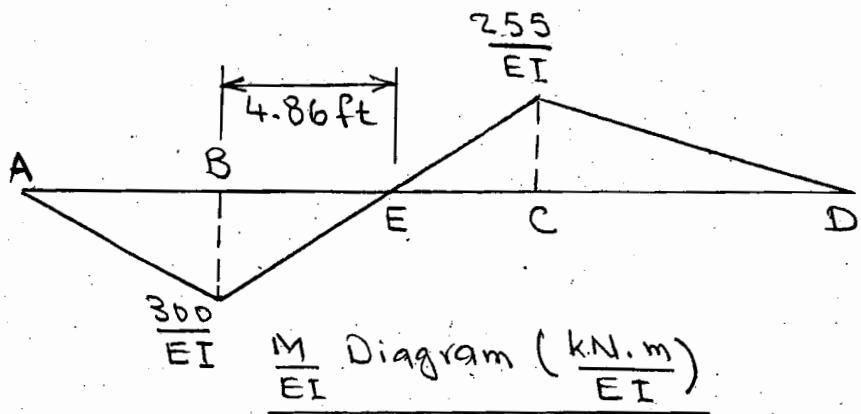
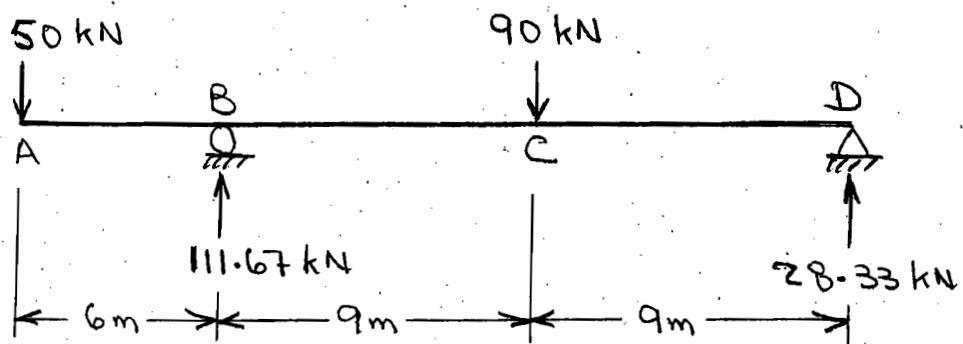
$$\frac{1}{EI} \left[\frac{1}{2} (40x_M) x_M \right] = \frac{2400}{EI}$$

From which, $x_M = 10.95 \text{ m}$

$$\begin{aligned}\Delta_{\max} &= \Delta_{AM} = \frac{1}{EI} \left[\frac{1}{2} (438) (10.95) \frac{2}{3} (10.95) \right] \\ &= \frac{17506 \text{ kN.m}^3}{EI} = \frac{17506}{200(600)} = 0.146 \text{ m}\end{aligned}$$

$$\underline{\Delta_{\max} = 146 \text{ mm} \downarrow}$$

6.26



$$\begin{aligned}\Delta_{BD} &= \frac{1}{EI} \left[-\frac{1}{2} (300) (4.86) \left(\frac{4.86}{3} \right) + \frac{1}{2} (255) (4.14) (7.62) \right. \\ &\quad \left. + \frac{1}{2} (255) (9) (12) \right] = \frac{16611.24 \text{ kN.m}^3}{EI}\end{aligned}$$

$$\theta_D = \frac{\Delta_{BD}}{18} = \frac{16611.24}{18(EI)} = \frac{922.85 \text{ kN.m}^2}{EI}$$

6.26 (contd.)

$$\Theta_{MD} = \Theta_D$$

$$\frac{1}{EI} \left[\frac{1}{2} \left(\frac{255}{9} x_M \right) x_M \right] = \frac{922.85}{EI}$$

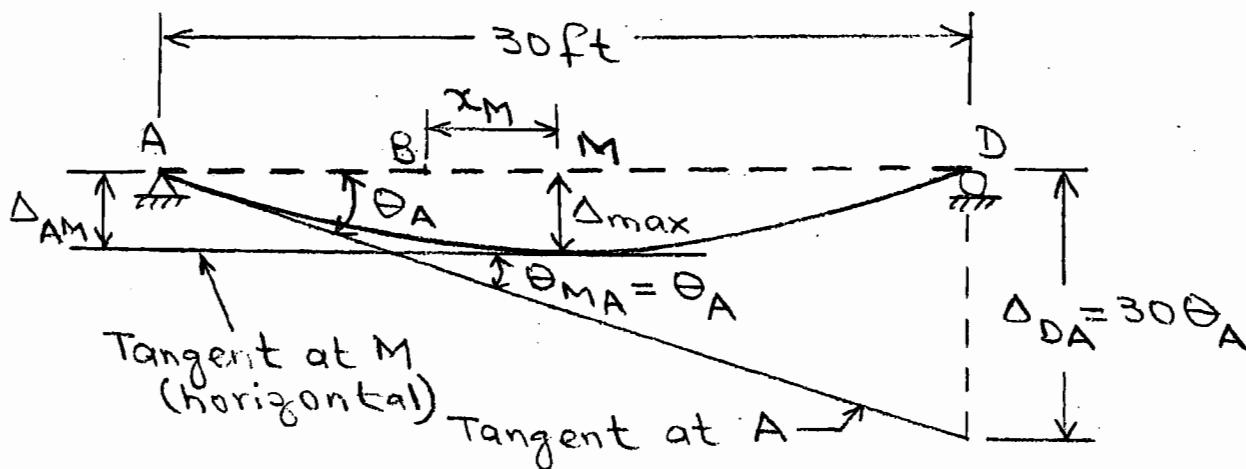
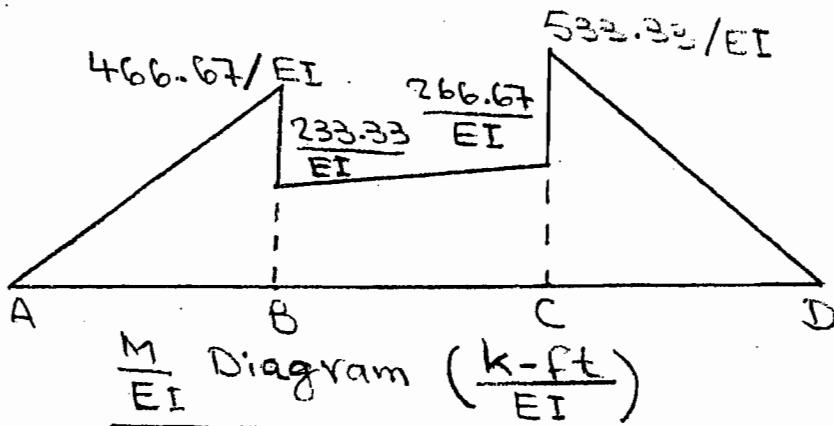
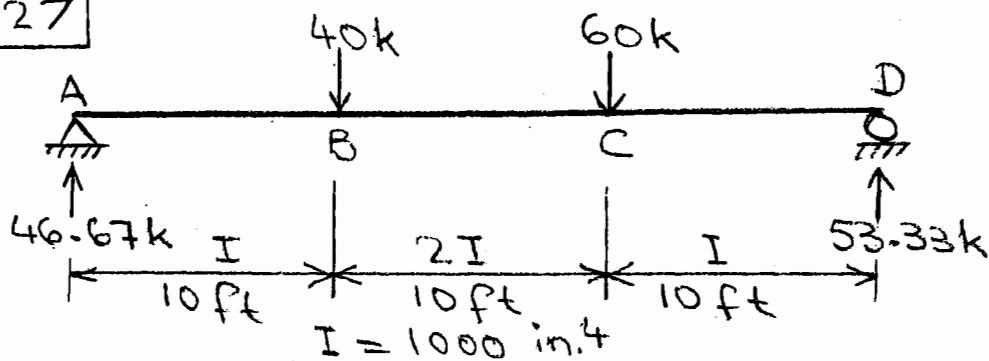
From which, $x_M = 8.07 \text{ m}$

$$\Delta_{max} = \Delta_{DM} = \frac{1}{EI} \left[\frac{1}{2} (228.65)(8.07) \frac{2}{3} (8.07) \right]$$

$$= \frac{4963.6 \text{ kN.m}^3}{EI} = \frac{4963.6}{70(95)} = 0.746 \text{ m}$$

$$\underline{\Delta_{max} = 746 \text{ mm} \downarrow}$$

6.27



$$\begin{aligned}\Delta_{DA} &= \frac{1}{EI} \left[\frac{1}{2} (466.67) 10 \left(\frac{10}{3} + 20 \right) + \frac{1}{2} (33.33) 10 \left(\frac{10}{3} + 10 \right) \right. \\ &\quad \left. + 233.33 (10) 15 + \frac{1}{2} (533.33) 10 \left(\frac{20}{3} \right) \right] \\ &= \frac{109444.4 \text{ k-ft}^3}{EI}\end{aligned}$$

$$\theta_A = \frac{\Delta_{DA}}{30} = \frac{3648.15 \text{ k-ft}^2}{EI}$$

6.27 (contd.)

If the maximum deflection occurs at M, then

$$\theta_{MA} = \theta_A$$

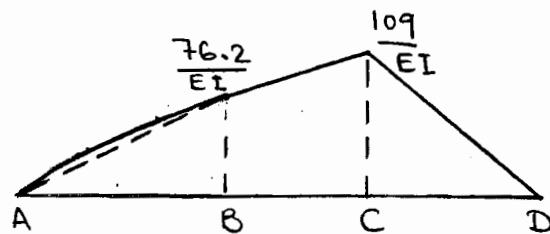
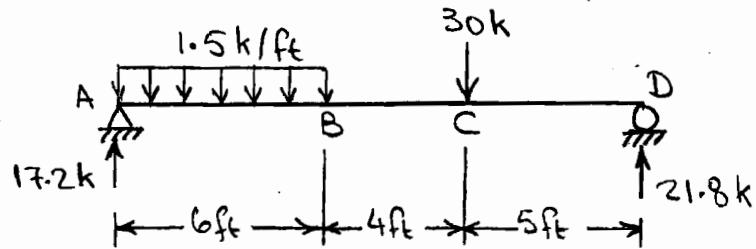
$$\frac{1}{EI} \left[\frac{1}{2} (466.67) 10 + 233.33 x_M + \frac{1}{2} \left(\frac{33.33}{10} x_M \right) x_M \right] = \frac{3648.15}{EI}$$

$$1.67 x_M^2 + 233.33 x_M - 1314.8 = 0$$

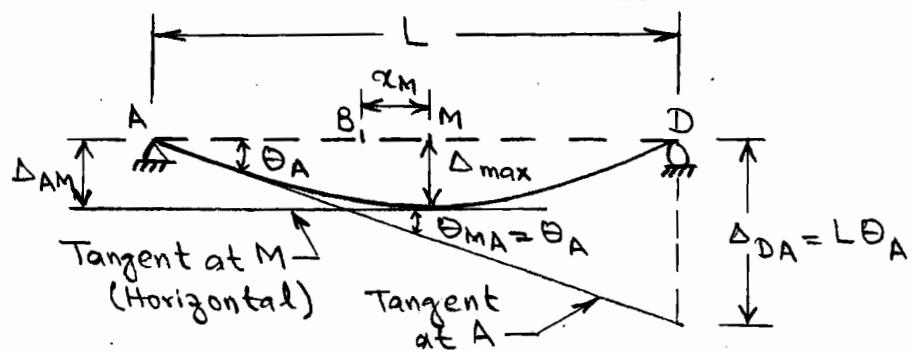
$$x_M = 5.42 \text{ ft}$$

$$\begin{aligned}\Delta_{max} &= \Delta_{AM} = \frac{1}{EI} \left[\frac{1}{2} (466.67) 10 \left(\frac{20}{3} \right) \right. \\ &\quad \left. + 233.33 (5.42) \left(\frac{5.42}{2} + 10 \right) \right. \\ &\quad \left. + \frac{1}{2} (18.06) (5.42) (3.61 + 10) \right] \\ &= \frac{32296 \text{ k-ft}^3}{EI} \\ &= \frac{32296 (12)^3}{29990 (1000)} = 1.92 \text{ in.} \downarrow\end{aligned}$$

6.28



$\frac{M}{EI}$ Diagram ($\frac{k \cdot ft}{EI}$)



$$\begin{aligned}\Delta_{DA} &= \frac{L}{EI} \left[\frac{1}{2} (76.2)(6)(11) + \frac{2}{3} \frac{(1.5)(6)^2}{8} (6)(12) \right. \\ &\quad \left. + (76.2)(4)(7) + \frac{1}{2} (32.8)(4) \left(\frac{4}{3} + 5 \right) + \frac{1}{2} (109)(5) \left(\frac{10}{3} \right) \right] \\ &= \frac{62946 \text{ k-ft}^3}{EI}\end{aligned}$$

$$\Theta_A = \frac{62946}{15EI} = \frac{419.73 \text{ k-ft}^2}{EI}$$

By setting $\Theta_{MA} = \Theta_A$, we write

$$\begin{aligned}\frac{1}{EI} \left[\frac{1}{2} (76.2)(6) + \frac{2}{3} \frac{(1.5)(6)^2}{8} (6) + \frac{1}{2} (152.4 + 8.2\alpha_M) \alpha_M \right] \\ = \frac{419.73}{EI}\end{aligned}$$

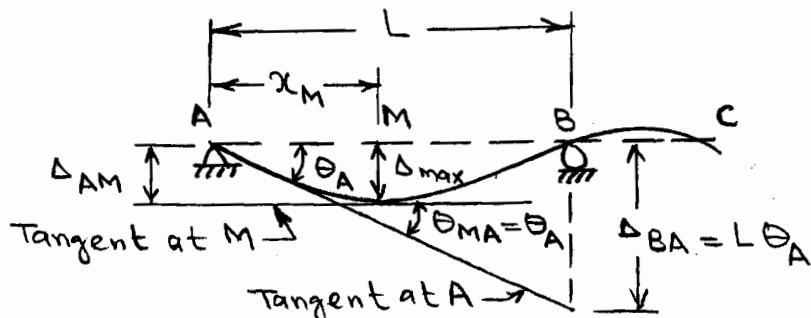
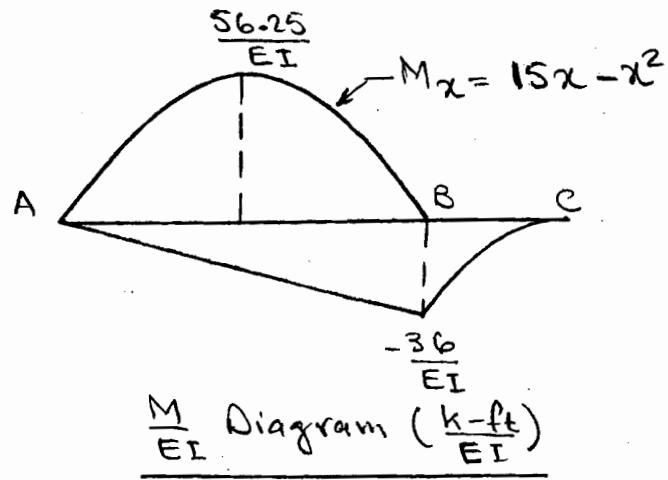
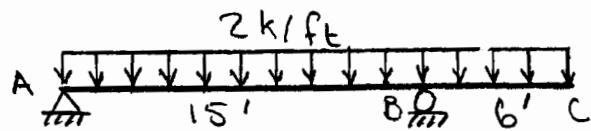
$$4.1 \alpha_M^2 + 76.2 \alpha_M - 164.13 = 0$$

$$\alpha_M = 1.95 \text{ ft}$$

6.28 (Cont'd.)

$$\Delta_{\max} = \Delta_{AM} = \frac{1}{EI} \left[\frac{1}{2} (76.2)(6)(4) + \frac{2}{3} \frac{(1.5)(6)^3}{8} (3) \right. \\ \left. + 76.2 (1.95) \left(\frac{1.95}{2} + 6 \right) + \frac{1}{2} (16)(1.95)(1.3 + 6) \right] \\ = \frac{2145.7 \text{ k-ft}^3}{EI} = \frac{2145.7 (12)^3}{(1500)(20000)} = 0.124 \text{ in.} \downarrow$$

6.29



$$\begin{aligned}\Delta_{BA} &= \frac{1}{EI} \left[\frac{2}{3} (56.25) 15 (7.5) - \frac{1}{2} (36) (15) 5 \right] \\ &= \frac{2868.75 \text{ k-ft}^3}{EI}\end{aligned}$$

$$\theta_A = \frac{2868.75}{15 EI} = \frac{191.25 \text{ k-ft}^2}{EI}$$

$$\theta_{MA} = \theta_A$$

$$\begin{aligned}\frac{1}{EI} \left[\frac{1}{2} (15x_M - x_M^2)x_M + \frac{2}{3} \left(\frac{2x_M^2}{8} \right)x_M \right. \\ \left. - \frac{1}{2} (2.4x_M)x_M \right] &= \frac{191.25}{EI}\end{aligned}$$

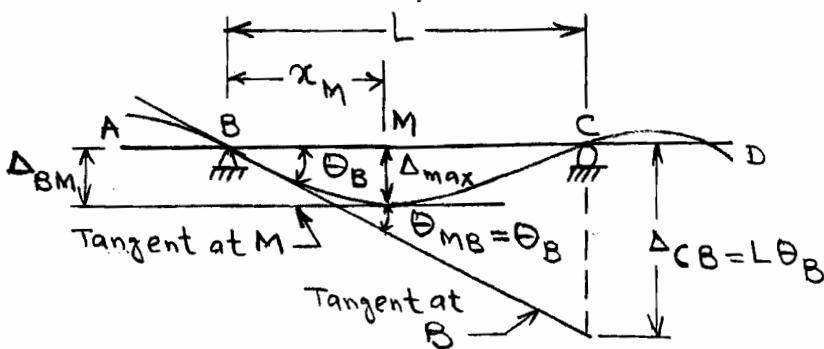
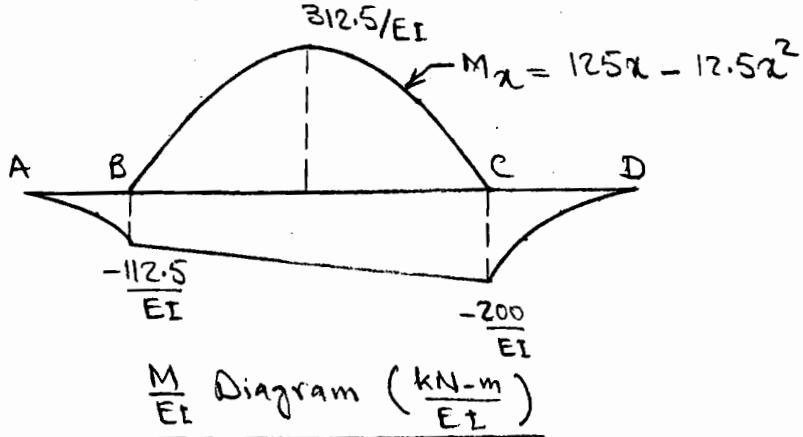
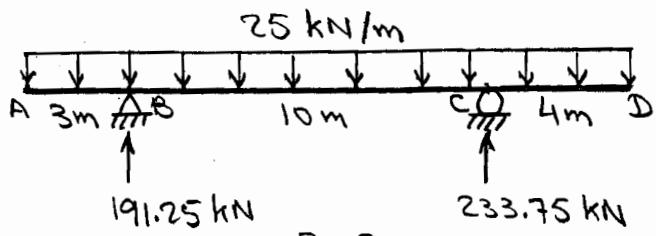
$$-\frac{x_M^3}{3} + 6.3x_M^2 = 191.25$$

from which $x_M = 6.92 \text{ ft}$

6.29 (contd.)

$$\begin{aligned}\Delta_{\max} &= \Delta_{AM} = \frac{1}{EI} \left[\frac{1}{2} (55.91) (6.92)(4.61) \right. \\ &\quad \left. + \frac{2}{3} (11.97) \frac{(6.92)^2}{2} - \frac{1}{2} (16.61) (6.92) \frac{2}{3} (6.92) \right] \\ &= \frac{817.73 \text{ k-ft}^3}{EI} = \frac{817.73 (12)^3}{29000 (3500)} \\ &= \underline{\underline{0.0139 \text{ in.} \downarrow}}\end{aligned}$$

6.30



$$\begin{aligned}\Delta_{CB} &= \frac{1}{EI} \left[\frac{2}{3} (312.5)(10)(5) - (112.5)(10)(5) - \frac{1}{2} (87.5)(10)(\frac{10}{3}) \right] \\ &= \frac{3333.33}{EI} \text{ kN-m}^3\end{aligned}$$

$$\Theta_B = \frac{3333.33}{10EI} = \frac{333.33}{EI} \text{ kN-m}^2$$

$$\Theta_{MB} = \Theta_A$$

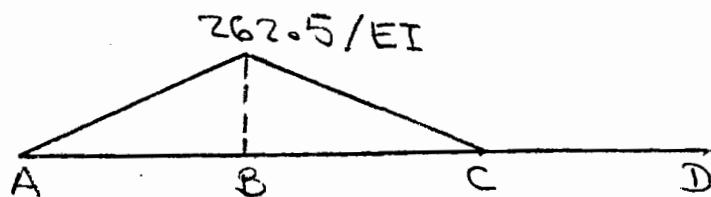
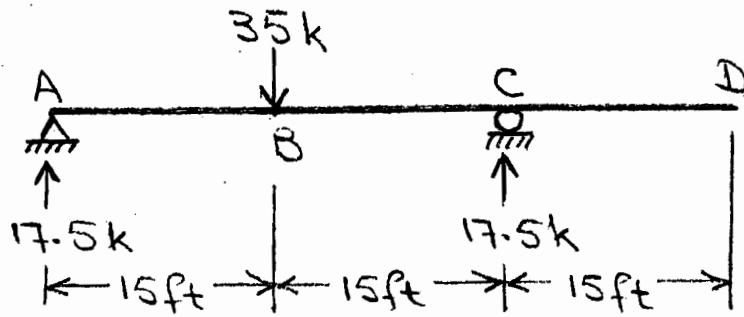
$$\begin{aligned}\frac{1}{EI} \left[\frac{1}{2} (125x_M - 12.5x_M^2)x_M + \frac{2}{3} \left(\frac{25x_M^2}{8} \right)x_M - 112.5x_M \right. \\ \left. - \frac{1}{2} (87.5x_M)x_M \right] &= \frac{333.33}{EI} \\ -4.167x_M^3 + 58.125x_M^2 - 112.5x_M &= 333.33\end{aligned}$$

From which $x_M = 4.77 \text{ m}$

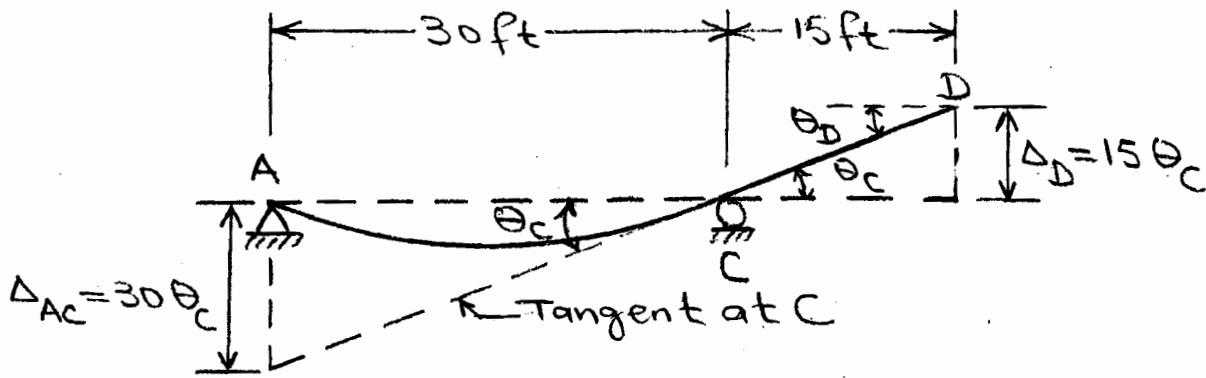
6.30 (Cont'd.)

$$\begin{aligned}\Delta_{\max} &= \Delta_{BM} = \frac{1}{EI} \left[\frac{1}{2} (311.84) (4.77) (3.18) + \frac{2}{3} (71.1) \right. \\ &\quad \times \left. \frac{(4.77)^2}{2} - (112.5) \frac{(4.77)^2}{2} - \frac{1}{2} (41.74) (4.77) (3.18) \right] \\ &= \frac{1307.91 \text{ kN-m}^3}{EI} = \frac{1307.91}{(250)(500)} = 0.0131 \text{ m} \\ \underline{\Delta_{\max}} &= 13.1 \text{ mm } \downarrow\end{aligned}$$

6.31



$\frac{M}{EI}$ Diagram ($\frac{k \cdot ft}{EI}$)



$$\Delta_{AC} = \frac{1}{EI} \left[\frac{1}{2} (262.5) 30 (15) \right] = \frac{59062.5}{EI}$$

$$\theta_c = \frac{\Delta_{AC}}{30} = \frac{1968.75}{EI}$$

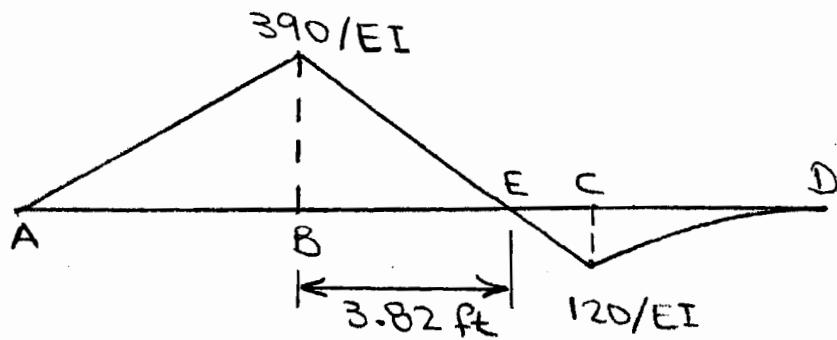
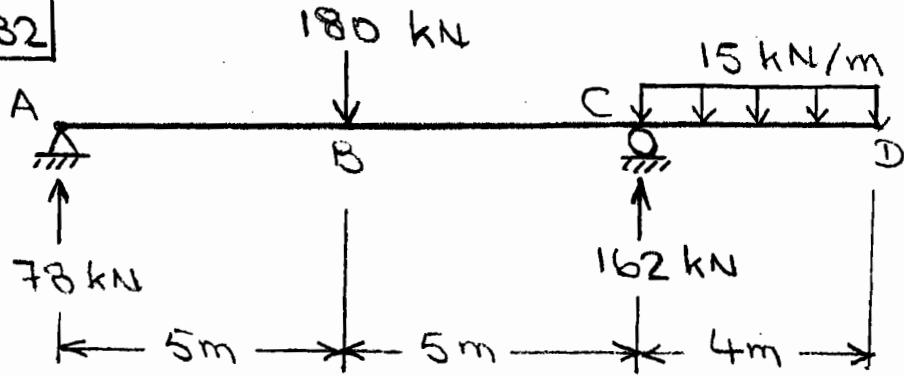
$$\theta_D = \theta_c = \frac{1968.75 \text{ k-ft}^2}{EI} = \frac{1968.75 (12)^2}{10000 (2500)}$$

$$\theta_D = 0.01134 \text{ rad} \checkmark$$

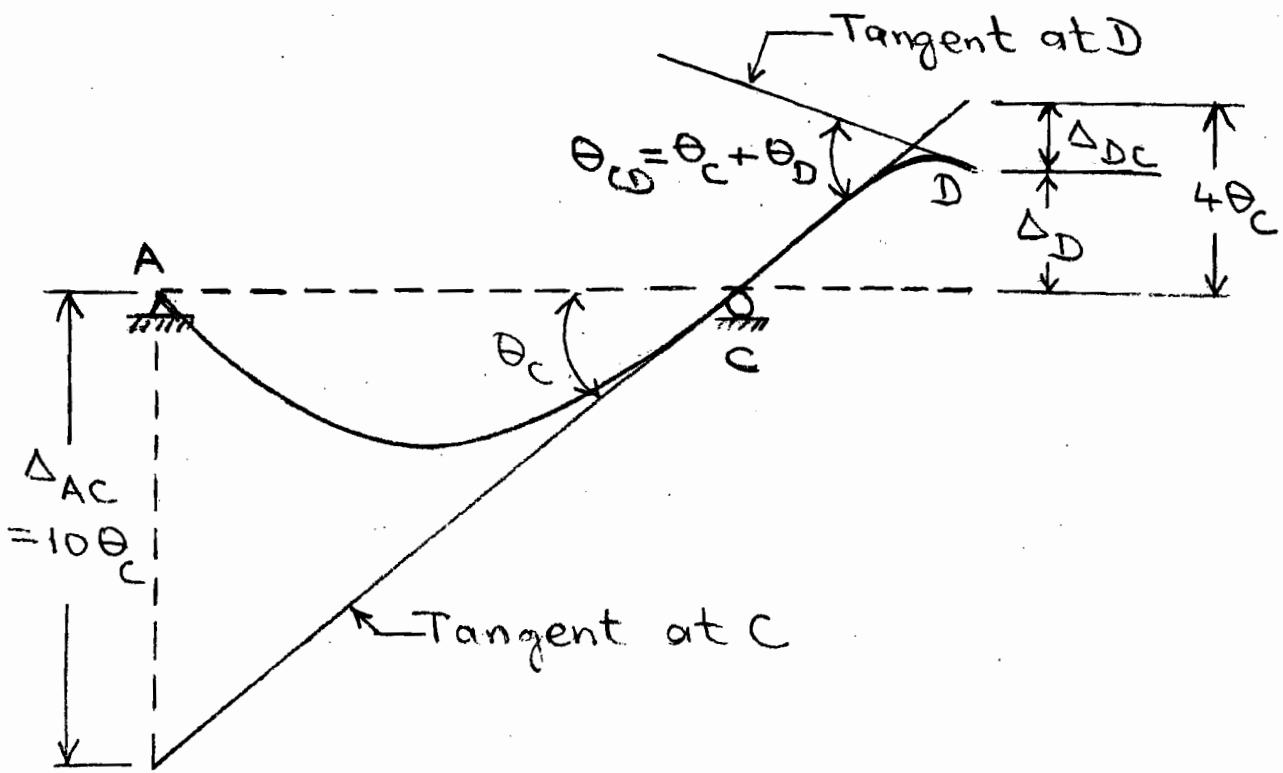
$$\Delta_D = 15\theta_c = \frac{29531.25 \text{ k-ft}^3}{EI} = \frac{29531.25 (12)^3}{10000 (2500)}$$

$$\Delta_D = 2.04 \text{ in. } \uparrow$$

6.32



$\frac{M}{EI}$ Diagram ($\frac{kN \cdot m}{EI}$)



6.32 (contd.)

$$\Delta_{AC} = \frac{1}{EI} \left[\frac{1}{2} (390) 5 \left(\frac{10}{3} \right) + \frac{1}{2} (390) 3.82 \left(\frac{3.82}{3} + 5 \right) - \frac{1}{2} (120) (1.18) (9.607) \right] = \frac{724.3}{EI}$$

$$\Theta_C = \frac{\Delta_{AC}}{10} = \frac{724.3}{EI}$$

$$\Theta_{CD} = \frac{1}{EI} \left[\frac{1}{3} (120) 4 \right] = \frac{160}{EI}$$

$$\begin{aligned}\Theta_D &= \Theta_{CD} - \Theta_C = \frac{1}{EI} (160 - 724.3) = -\frac{564.3 \text{ kN.m}^2}{EI} \\ &= -\frac{564.3}{70(2340)} = -0.00345 \text{ rad}\end{aligned}$$

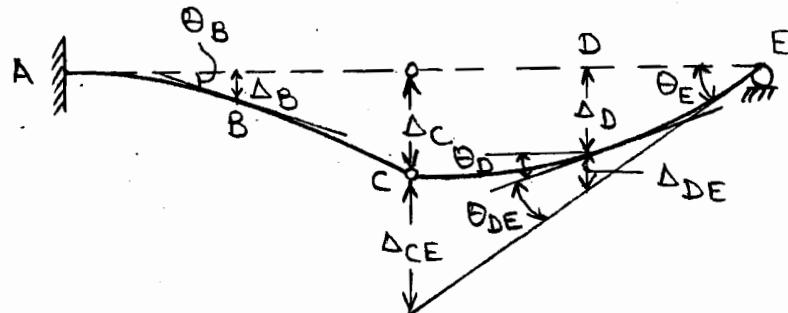
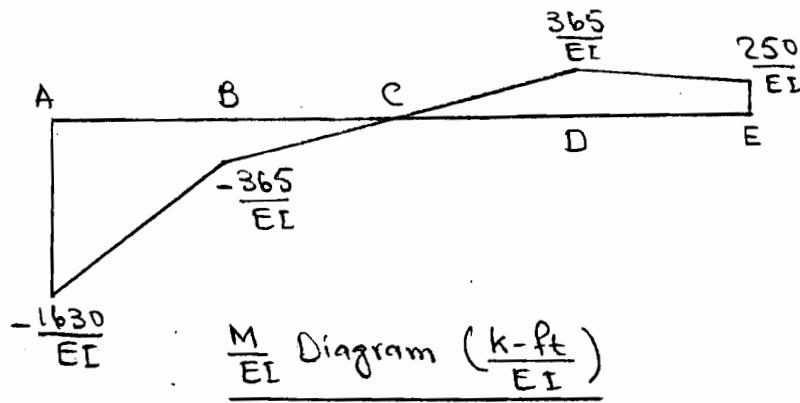
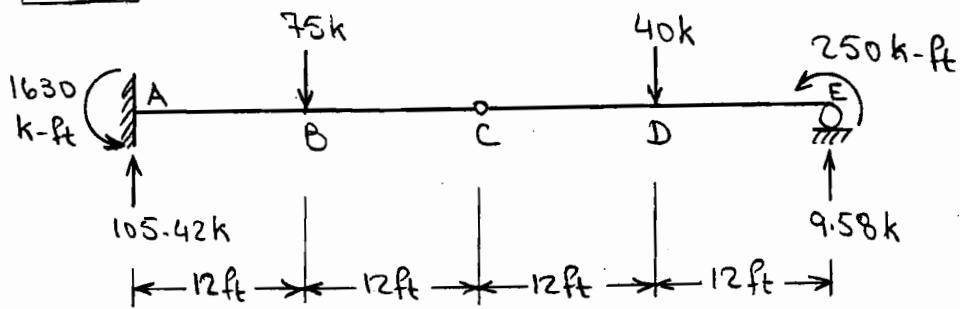
$$\underline{\Theta_D = 0.00345 \text{ rad } \uparrow}$$

$$\Delta_{DC} = \frac{1}{EI} [160(3)] = \frac{480}{EI}$$

$$\begin{aligned}\Delta_D &= 4\Theta_C - \Delta_{DC} = \frac{1}{EI} [4(724.3) - 480] \\ &= \frac{2417.2 \text{ kN.m}^3}{EI} = \frac{2417.2}{70(2340)} = 0.01476 \text{ m}\end{aligned}$$

$$\underline{\Delta_D = 14.76 \text{ mm} \uparrow}$$

6.33



$$\theta_B = \theta_{BA} = \frac{1}{EI} \left[\left(\frac{1630 + 365}{2} \right) 12 \right] = \frac{11970 \text{ k-ft}^2}{EI} = 0.0099 \text{ rad. } \checkmark$$

$$\Delta_B = \Delta_{BA} = \frac{1}{EI} \left[(365) 12 (4) + \frac{1}{2} (1265) (12) (8) \right] = \frac{87000}{EI} = 0.86 \text{ in. } \downarrow$$

$$\begin{aligned} \Delta_C &= \Delta_{CA} = \frac{1}{EI} \left[365 (12) (18) + \frac{1}{2} (1265) (12) (20) + \frac{1}{2} (365) \right. \\ &\quad \left. \times (12) 8 \right] = \frac{248160 \text{ k-ft}^3}{EI} \end{aligned}$$

$$\begin{aligned} \Delta_{CE} &= \frac{1}{EI} \left[\frac{1}{2} (365) (12) (8) + (250) (12) (18) + \frac{1}{2} (115) (12) (16) \right] \\ &= \frac{82560 \text{ k-ft}^3}{EI} \end{aligned}$$

$$\theta_E = \frac{\Delta_C + \Delta_{CE}}{24} = \frac{248160 + 82560}{24 EI} = \frac{13780 \text{ k-ft}^2}{EI}$$

6.33 (cont'd.)

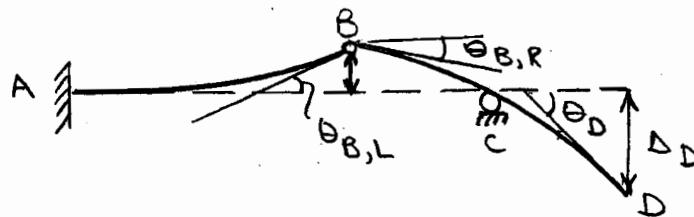
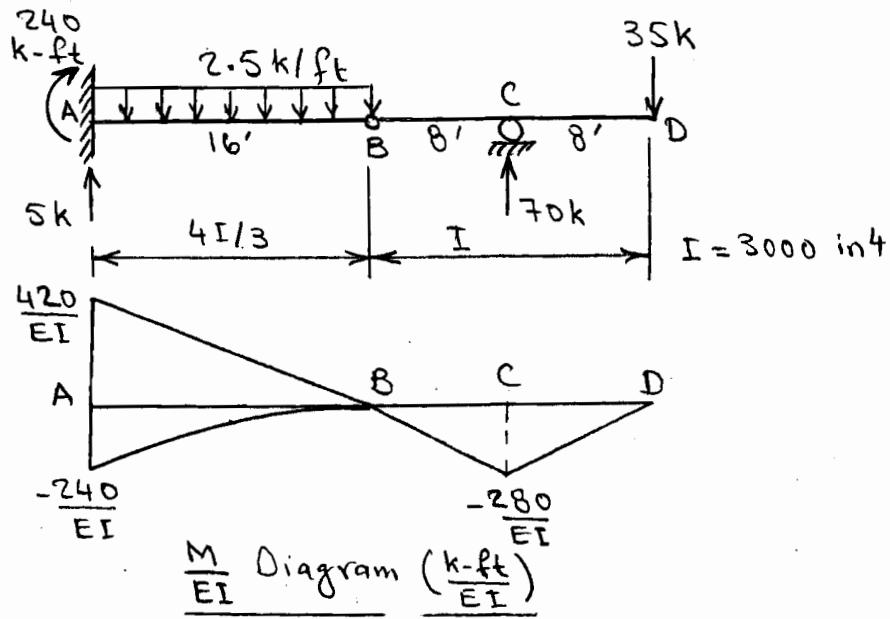
$$\Theta_{DE} = \frac{1}{EI} \left[\left(\frac{365 + 250}{2} \right) 12 \right] = \frac{3690 \text{ k-ft}^2}{EI}$$

$$\Theta_D = \Theta_E - \Theta_{DE} = \frac{13780 - 3690}{EI} = \frac{10090 \text{ k-ft}^2}{EI} = 0.0084 \text{ rad. } \cancel{\downarrow}$$

$$\Delta_{DE} = \frac{1}{EI} \left[(250)(12)(6) + \frac{1}{2}(115)(12)(4) \right] = \frac{20760 \text{ k-ft}^3}{EI}$$

$$\Delta_D = 12\Theta_E - \Delta_{DE} = \frac{1}{EI} [12(13780) - 20760] = \frac{144600 \text{ k-ft}^3}{EI}$$
$$\underline{\Delta_D = 1.44 \text{ in. } \downarrow}$$

6.34



$$\begin{aligned}\Theta_{B,\text{Left}} &= \Theta_{AB} = \frac{1}{EI} \left[\frac{1}{2}(420)16 - \frac{1}{3}(240)16 \right] \\ &= \frac{2080 \text{ k-ft}^2}{EI} = \frac{2080 (12)^2}{30000 (3000)} = 0.0033 \text{ rad.} \quad \triangle\end{aligned}$$

$$\begin{aligned}\Delta_B &= \Delta_{BA} = \frac{1}{EI} \left[\frac{1}{2}(420)16 (16.67) - \frac{1}{3}(240)16 (12) \right] \\ &= \frac{20480 \text{ k-ft}^3}{EI} = \frac{20480 (12)^3}{30000 (3000)} = 0.39 \text{ in } \uparrow\end{aligned}$$

$$\Delta_{BC} = \frac{1}{EI} \left[\frac{1}{2}(280)8 \left(\frac{16}{3} \right) \right] = \frac{5973.33}{EI}$$

$$\Delta_{BC} + \Delta_B = 8 \Theta_C$$

$$\Theta_C = \frac{1}{8EI} (5973.33 + 20480) = \frac{3306.67}{EI}$$

$$\Theta_{BC} = \frac{1}{EI} \left[\frac{1}{2}(280)8 \right] = \frac{1120}{EI}$$

$$\begin{aligned}\Theta_{B,\text{Right}} &= \Theta_C - \Theta_{BC} = \frac{1}{EI} (3306.67 - 1120) \\ &= \frac{2186.67 \text{ k-ft}^2}{EI} = 0.0035 \text{ rad.} \quad \triangle\end{aligned}$$

6.34 (contd.)

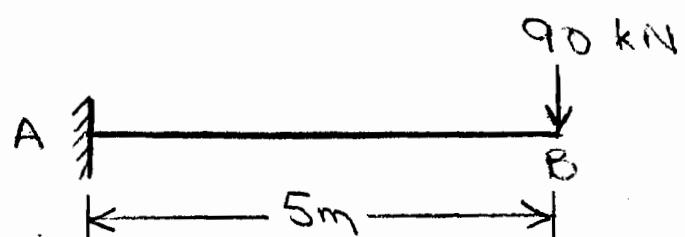
$$\Theta_{CD} = \frac{1}{EI} \left[\frac{1}{2} (280) 8 \right] = \frac{1120}{EI}$$

$$\begin{aligned}\Theta_D &= \Theta_C + \Theta_{CD} = \frac{1}{EI} (3306.67 + 1120) \\ &= \frac{4426.67 \text{ k-ft}^2}{EI} = \underline{\underline{0.0071 \text{ rad.}}}\end{aligned}$$

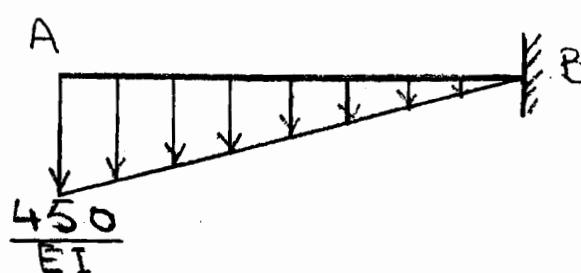
$$\Delta_{DC} = \frac{1}{EI} \left[1120 \left(\frac{16}{3} \right) \right] = \frac{5973.33}{EI}$$

$$\begin{aligned}\Delta_D &= 8\Theta_C + \Delta_{DC} = \frac{1}{EI} [8(3306.67) + 5973.33] \\ &= \frac{32426.67 \text{ k-ft}^3}{EI} = \underline{\underline{0.62 \text{ in.} \downarrow}}\end{aligned}$$

6.35



Real

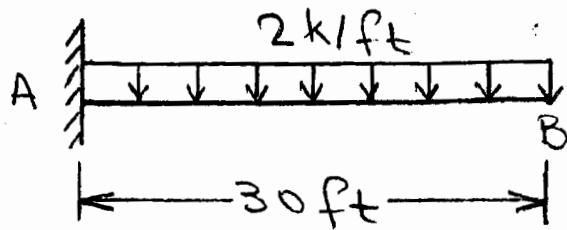


Conjugate

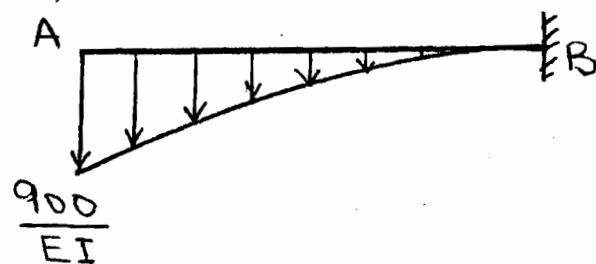
$$\Theta_B = -\frac{1}{2}(5)\left(\frac{450}{EI}\right) = -\frac{1125 \text{ kN.m}^2}{EI} = -\frac{1125}{200(800)} \\ = -0.00703 \text{ rad} = \underline{\underline{0.00703 \text{ rad}}} \quad \checkmark$$

$$\Delta_B = -\frac{1}{2}(5)\left(\frac{450}{EI}\right)\left(\frac{10}{3}\right) = -\frac{3750 \text{ kN.m}^3}{EI} \\ = -\frac{3750}{200(800)} = -0.0234 \text{ m} = \underline{\underline{23.4 \text{ mm}}} \downarrow$$

6.36



Real



Conjugate

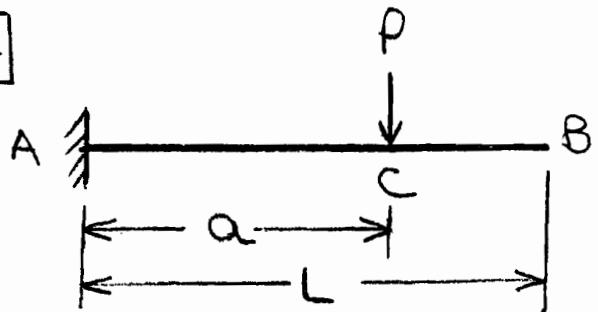
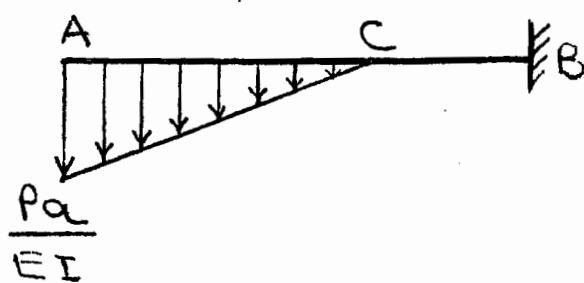
$$\begin{aligned}\Theta_B &= -\frac{1}{3} \left(\frac{900}{EI} \right) 30 = -\frac{9000 \text{ k-ft}^2}{EI} \\ &= -\frac{9000 (12)^2}{29000 (3000)} = -0.0149 \text{ rad}\end{aligned}$$

$$\underline{\Theta_B = 0.0149 \text{ rad } \checkmark}$$

$$\begin{aligned}\Delta_B &= -\frac{9000}{EI} \left(\frac{3}{4} \right) 30 = -\frac{202500 \text{ k-ft}^3}{EI} \\ &= -\frac{202500 (12)^3}{29000 (3000)} = -4.022 \text{ in.}\end{aligned}$$

$$\underline{\Delta_B = 4.022 \text{ in. } \downarrow}$$

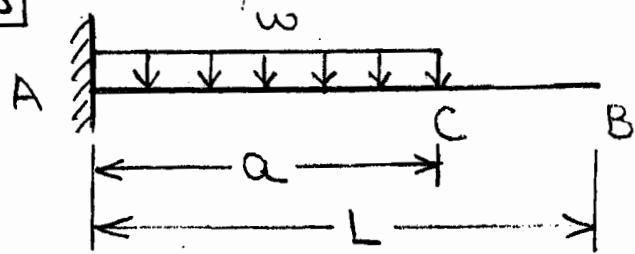
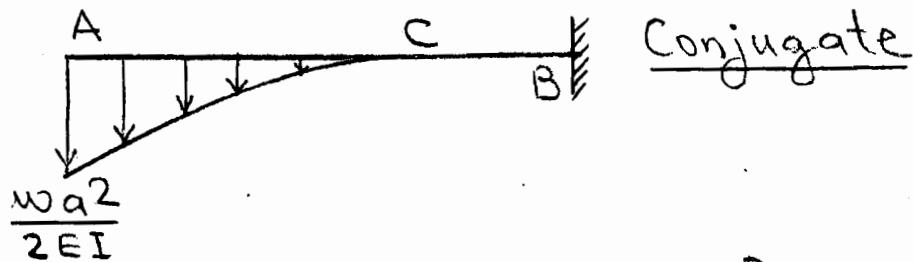
6.37

RealConjugate

$$\Theta_B = -\frac{1}{2} \left(\frac{Pa}{EI} \right) a = -\frac{Pa^2}{2EI} = \frac{Pa^2}{EI} \quad \nabla$$

$$\begin{aligned} \Delta_B &= -\frac{Pa^2}{2EI} \left(L - \frac{a}{3} \right) = -\frac{Pa^2}{6EI} (3L - a) \\ &= \frac{Pa^2}{6EI} (3L - a) \downarrow \end{aligned}$$

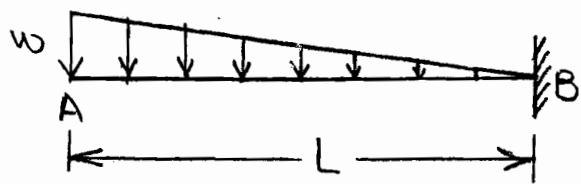
6.38

RealConjugate

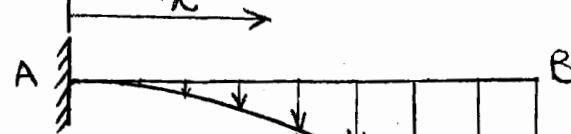
$$\Theta_B = -\frac{1}{3} \left(\frac{w a^2}{2 EI} \right) a = -\frac{w a^3}{6 EI} = \underline{\underline{\frac{w a^3}{6 EI}}}$$

$$\begin{aligned} \Delta_B &= -\frac{w a^3}{6 EI} \left(L - \frac{a}{4} \right) = -\frac{w a^3}{24 EI} (4L - a) \\ &= \underline{\underline{\frac{w a^3}{24 EI} (4L - a)}} \end{aligned}$$

6.39



Real



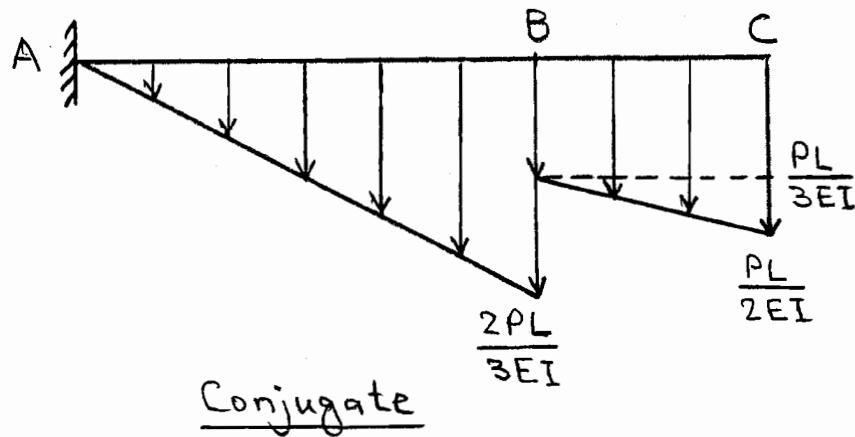
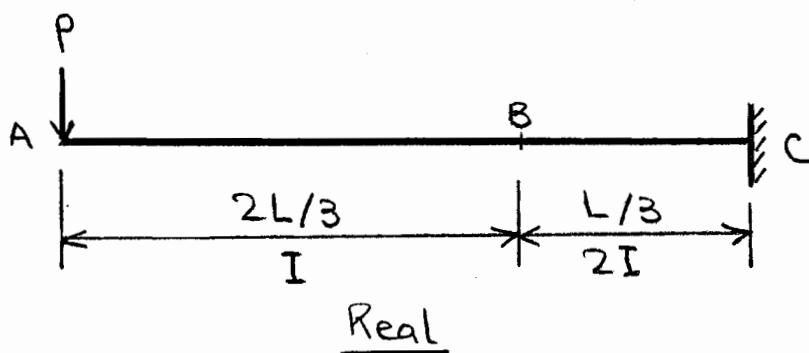
conjugate

$$\frac{wx^2}{2EI} \left(1 - \frac{x}{3L}\right) \quad \frac{wL^2}{3EI}$$

$$\Theta_A = \int_0^L \frac{wx^2}{2EI} \left(1 - \frac{x}{3L}\right) dx = \frac{wL^3}{9EI}$$

$$\Delta_A = \int_0^L \frac{wx^2}{2EI} \left(1 - \frac{x}{3L}\right) x dx = -\frac{11}{120} \frac{wL^4}{EI} = \frac{11}{120} \frac{wL^4}{EI}$$

6.40

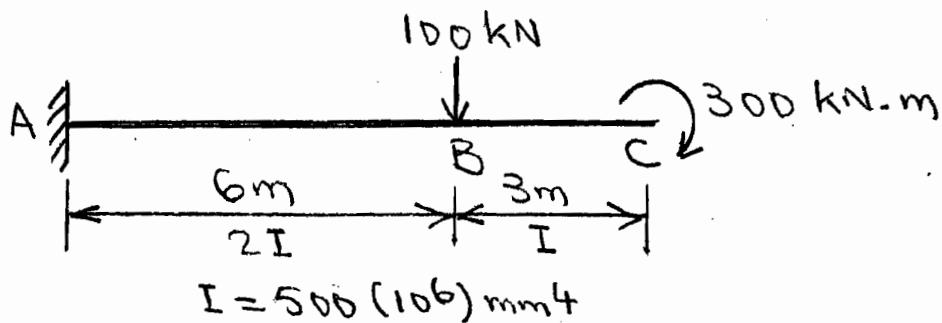


$$\Theta_A = \frac{1}{2} \left(\frac{2L}{3} \right) \left(\frac{2PL}{3EI} \right) + \frac{L}{3} \left(\frac{PL}{3EI} \right) + \frac{1}{2} \left(\frac{L}{3} \right) \left(\frac{PL}{6EI} \right) = \frac{13PL^2}{36EI}$$

$$\begin{aligned} \Delta_A &= -\frac{2PL^2}{9EI} \left(\frac{4L}{9} \right) - \frac{PL^2}{9EI} \left(\frac{5L}{6} \right) - \frac{PL^2}{36EI} \left(\frac{8L}{9} \right) = -\frac{35PL^3}{162EI} \\ &= \underline{\underline{\frac{35PL^3}{162EI}}} \end{aligned}$$

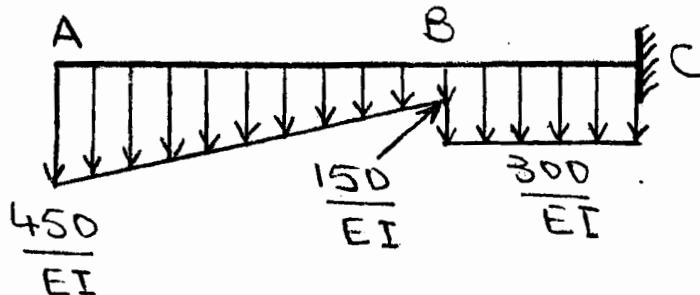
6.41

Real



$$I = 500 (10^6) \text{ mm}^4$$

Conjugate



$$\Theta_B = \frac{1}{EI} \left[-150(6) - \frac{1}{2}(300)6 \right] = -\frac{1800 \text{ kN.m}^2}{EI}$$

$$= -\frac{1800}{70(500)} = -0.0514 \text{ rad} = \underline{\underline{0.0514 \text{ rad}}} \quad \checkmark$$

$$\Delta_B = \frac{1}{EI} \left[-150(6)3 - \frac{1}{2}(300)6(4) \right] = -\frac{6300 \text{ kN.m}^3}{EI}$$

$$= -\frac{6300}{70(500)} = -0.18 \text{ m} = \underline{\underline{180 \text{ mm} \downarrow}}$$

$$\Theta_C = \frac{1}{EI} \left[-1800 - 300(3) \right] = -\frac{2700 \text{ kN.m}^2}{EI}$$

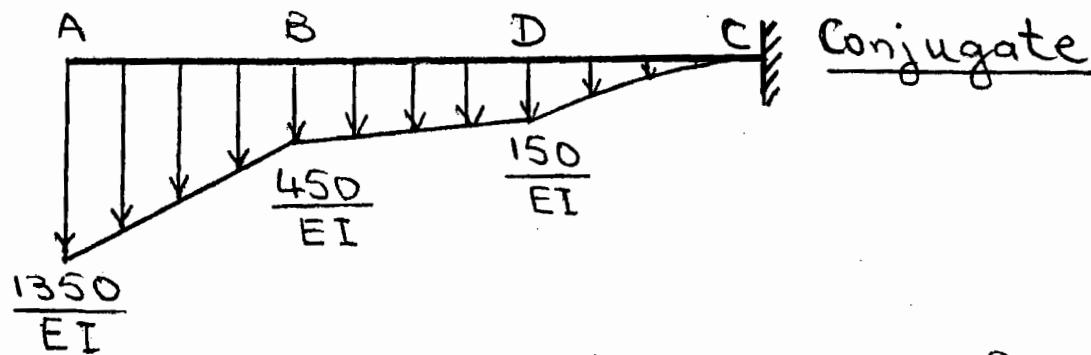
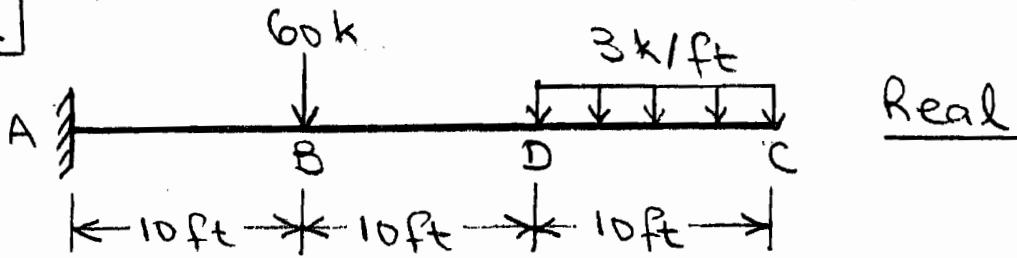
$$= -\frac{2700}{70(500)} = -0.0771 \text{ rad} = \underline{\underline{0.0771 \text{ rad}}} \quad \checkmark$$

$$\Delta_C = \frac{1}{EI} \left[-150(6)(6) - \frac{1}{2}(300)6(7) - 300(3)(1.5) \right]$$

$$= -\frac{13050 \text{ kN.m}^3}{EI} = -\frac{13050}{70(500)} = -0.373 \text{ m}$$

$$= \underline{\underline{373 \text{ mm} \downarrow}}$$

6.42



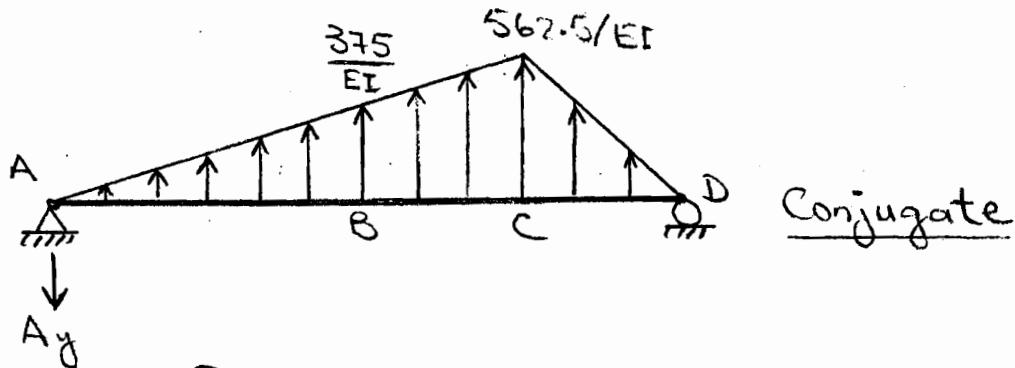
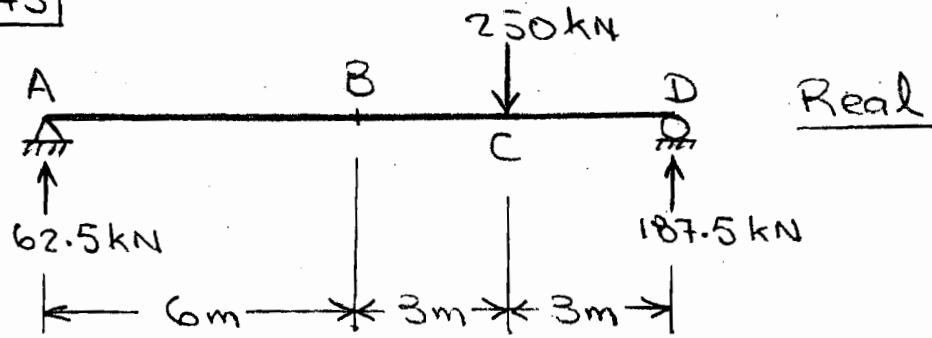
$$\begin{aligned}\Theta_B &= \frac{1}{EI} \left[-450(10) - \frac{1}{2}(900)10 \right] = -\frac{9000 \text{ k-ft}^2}{EI} \\ &= -\frac{9000 (12)^2}{29000 (4000)} = -0.0112 \text{ rad} = \underline{0.0112 \text{ rad}} \quad \square\end{aligned}$$

$$\begin{aligned}\Delta_B &= \frac{1}{EI} \left[-450(10)5 - \frac{1}{2}(900)10\left(\frac{20}{3}\right) \right] = -\frac{52500 \text{ k-ft}^3}{EI} \\ &= -\frac{52500 (12)^3}{29000 (4000)} = -0.782 \text{ in.} = \underline{0.782 \text{ in.} \downarrow}\end{aligned}$$

$$\begin{aligned}\Theta_C &= \frac{1}{EI} \left[-9000 - 150(10) - \frac{1}{2}(300)10 - \frac{1}{3}(150)10 \right] \\ &= -\frac{12500 \text{ k-ft}^2}{EI} = -\frac{12500 (12)^2}{29000 (4000)} = -0.0155 \text{ rad} \\ &= \underline{0.0155 \text{ rad}} \quad \square\end{aligned}$$

$$\begin{aligned}\Delta_C &= \frac{1}{EI} \left[-450(10)25 - \frac{1}{2}(900)10\left(\frac{20}{3} + 20\right) \right. \\ &\quad \left. - 150(10)15 - \frac{1}{2}(300)10\left(\frac{20}{3} + 10\right) - \frac{1}{3}(150)10\left(\frac{30}{4}\right) \right] \\ &= -\frac{283750 \text{ k-ft}^3}{EI} = -\frac{283750 (12)^3}{29000 (4000)} \\ &= -4.227 \text{ in.} = \underline{4.227 \text{ in.} \downarrow}\end{aligned}$$

6.43



Reaction for conjugate beam:

$$+G \sum M_D = 0$$

$$Ay(12) - \frac{1}{2} \left(\frac{562.5}{EI} \right) 9(6) - \frac{1}{2} \left(\frac{562.5}{EI} \right) 3(2) = 0$$

$$Ay = \frac{1406.25}{EI} \text{ kN.m}^2$$

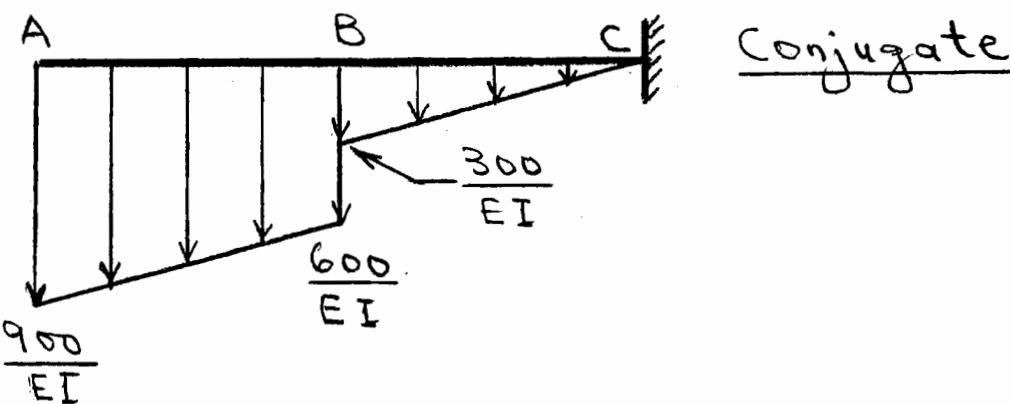
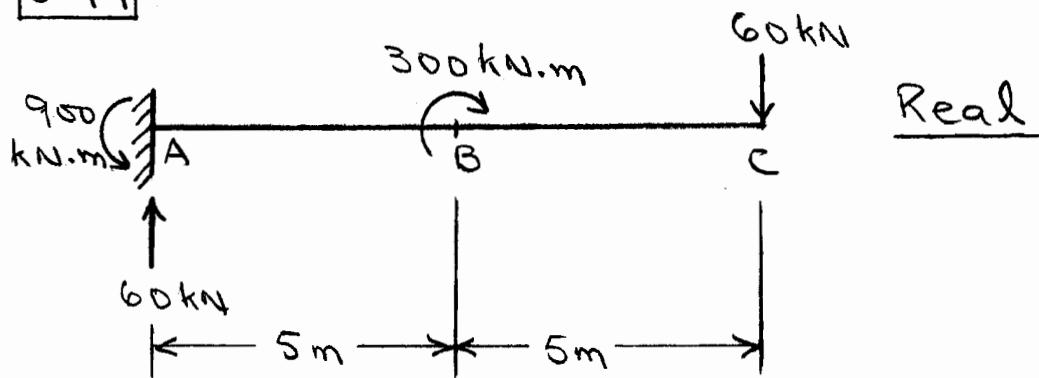
$$\begin{aligned} \Theta_B &= \frac{1}{EI} \left[-1406.25 + \frac{1}{2} (375) 6 \right] = -\frac{281.25}{EI} \text{ kN.m}^2 \\ &= -\frac{281.25}{200(462)} = -0.00304 \text{ rad} = \underline{\underline{0.00304 \text{ rad}}} \end{aligned}$$

$$\begin{aligned} \Delta_B &= \frac{1}{EI} \left[-1406.25(6) + \frac{1}{2} (375) 6(2) \right] = -\frac{6187.5}{EI} \text{ kN.m}^3 \\ &= -\frac{6187.5}{200(462)} = -0.067 \text{ m} = \underline{\underline{67 \text{ mm}}} \downarrow \end{aligned}$$

$$\begin{aligned} \Theta_C &= \frac{1}{EI} \left[-1406.25 + \frac{1}{2} (562.5) 9 \right] = \frac{1125}{EI} \text{ kN.m}^2 \\ &= \frac{1125}{200(462)} = \underline{\underline{0.0122 \text{ rad}}} \end{aligned}$$

$$\begin{aligned} \Delta_C &= \frac{1}{EI} \left[-1406.25(9) + \frac{1}{2} (562.5) 9(3) \right] = -\frac{5062.5}{EI} \text{ kN.m}^3 \\ &= -\frac{5062.5}{200(462)} = -0.0548 \text{ m} = \underline{\underline{54.8 \text{ mm}}} \downarrow \end{aligned}$$

6.44



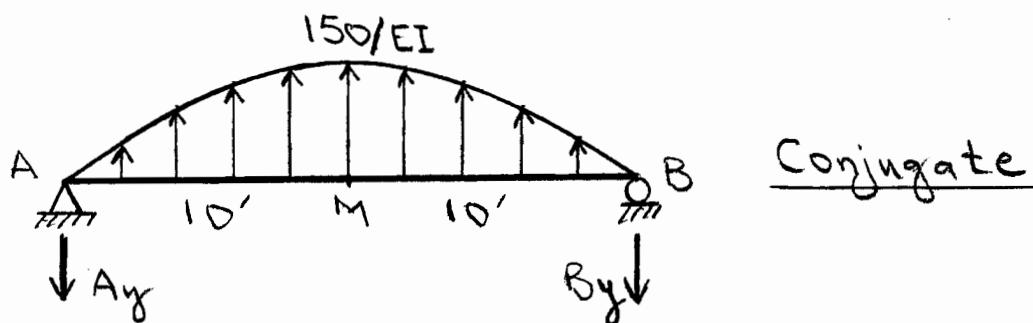
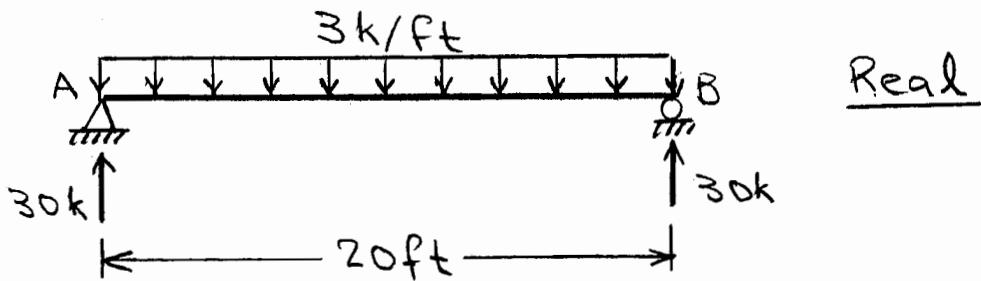
$$\begin{aligned}\Delta_{\max} = \Delta_C &= -\frac{1}{EI} \left[600(5)(7.5) + \frac{1}{2}(300)(5)\left(\frac{25}{3}\right) \right. \\ &\quad \left. + \frac{1}{2}(300)(5)\left(\frac{10}{3}\right) \right] = \frac{-31250 \text{ kN.m}^3}{EI} \\ &= \frac{31250 \text{ kN.m}^3}{EI} \downarrow\end{aligned}$$

$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{31250}{200(10^6)I} = \frac{10}{360}$$

$$\text{From which, } I = 5625(10^6) \text{ m}^4 = \underline{\underline{5625(10^6) \text{ mm}^4}}$$

6.45



Reactions for conjugate beam:

$$A_y = B_y = \frac{1}{2} \left(\frac{2}{3} \right) \left(\frac{150}{\text{EI}} \right) (20) = \frac{1000 \text{ k-ft}^2}{\text{EI}}$$

$$\begin{aligned} \Delta_{\max} &= \Delta_M = \frac{1}{\text{EI}} \left[-1000(10) + \frac{2}{3}(150)(10)\left(\frac{30}{8}\right) \right] \\ &= -\frac{6250 \text{ k-ft}^3}{\text{EI}} = \frac{6250 \text{ k-ft}^3}{\text{EI}} \downarrow \end{aligned}$$

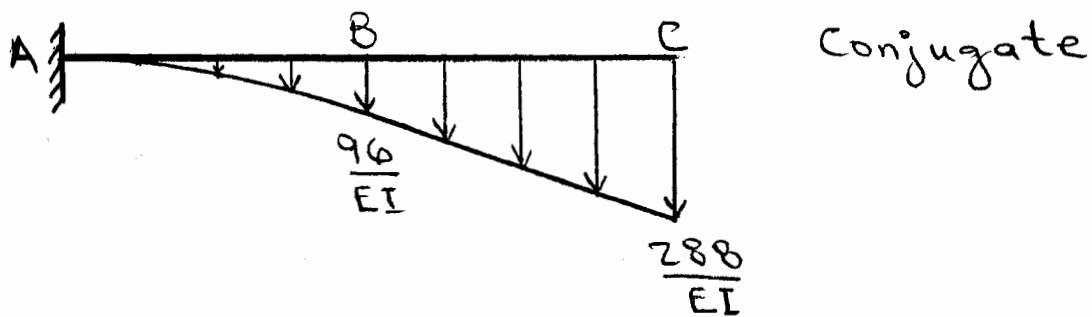
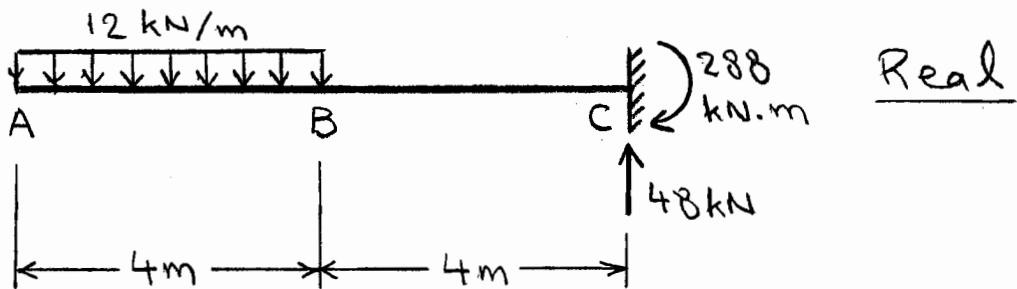
$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{6250(12)^3}{29000(I)} = \frac{20(12)}{360}$$

from which,

$$I = 559 \text{ in}^4$$

6.46



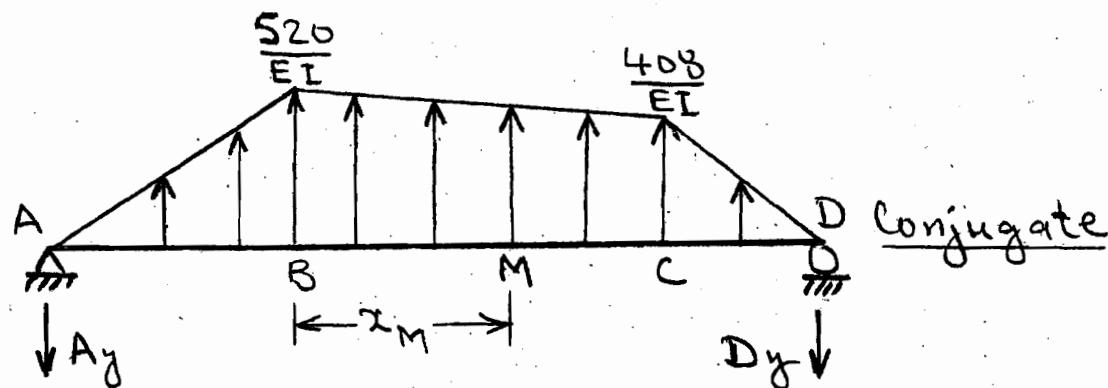
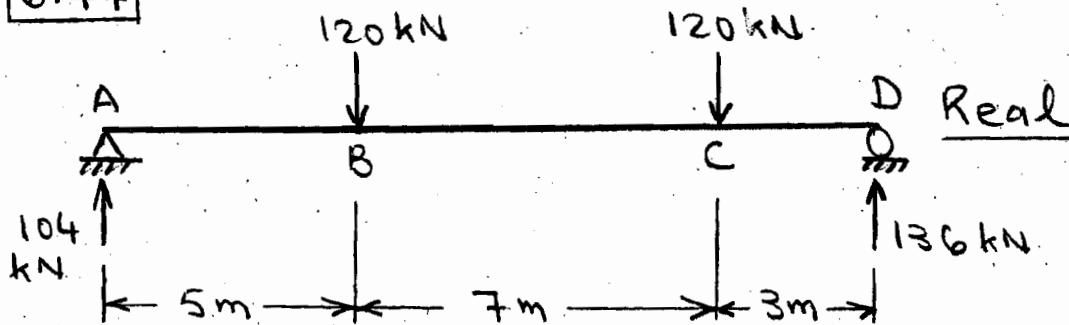
$$\begin{aligned}\Delta_{\max} &= \Delta_A = \frac{1}{EI} \left[-\frac{1}{3} (96)(4)(3) - 96(4)(6) \right. \\ &\quad \left. - \frac{1}{2} (192)(4) \left(\frac{8}{3} + 4 \right) \right] = -\frac{5248 \text{ kN.m}^3}{EI} \\ &= \frac{5248 \text{ kN.m}^3}{EI} \downarrow\end{aligned}$$

$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{5248}{70(10^6)I} = \frac{8}{360}$$

from which, $I = 3374(10^6) \text{ m}^4 = \underline{\underline{3374(10^6) \text{ mm}^4}}$

6.47



$$+ \zeta \sum M_D = 0$$

$$A_y(15) - \frac{1}{2} \left(\frac{520}{EI} \right) 5 \left(\frac{5}{3} + 10 \right) - \left(\frac{408}{EI} \right) (7) \left(\frac{7}{2} + 3 \right)$$

$$- \frac{1}{2} \left(\frac{112}{EI} \right) 7 \left(\frac{14}{3} + 3 \right) - \frac{1}{2} \left(\frac{408}{EI} \right) 3 (2) = 0$$

$$A_y = \frac{2530.67 \text{ kN}\cdot\text{m}}{EI}$$

Let the maximum bending moment in the conjugate beam occur at point M, at a distance x_M from point B. Then the shear at M must be zero. Thus,

$$S_M = \frac{1}{EI} \left[-2530.67 + \frac{1}{2} (520)(5) + (520)x_M - \frac{1}{2} \left(\frac{112}{7} x_M \right) x_M \right] = 0$$

$$8x_M^2 - 520x_M + 1230.67 = 0$$

$$x_M = 2.46 \text{ m}$$

6.47 (contd.)

$$\Delta_{\max} = \Delta_M = \frac{1}{EI} \left[-2530.67(7.46) + \frac{1}{2}(520)(5) \right]$$

$$\left(\frac{5}{3} + 2.46 \right) + (520)(2.46) \left(\frac{2.46}{2} \right) - \frac{1}{2}(39.36)$$

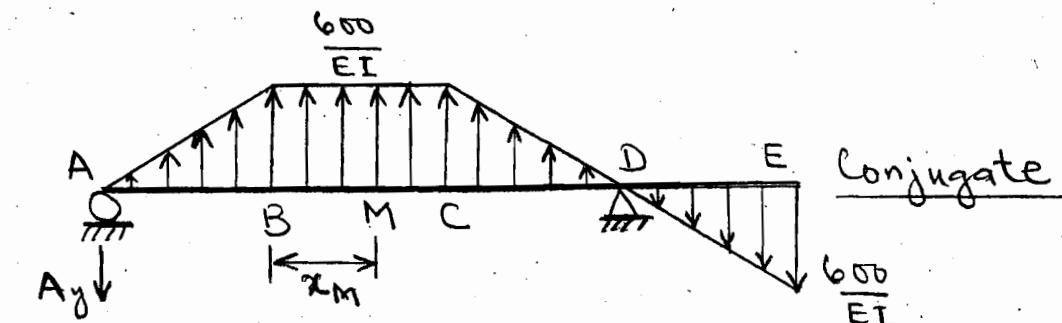
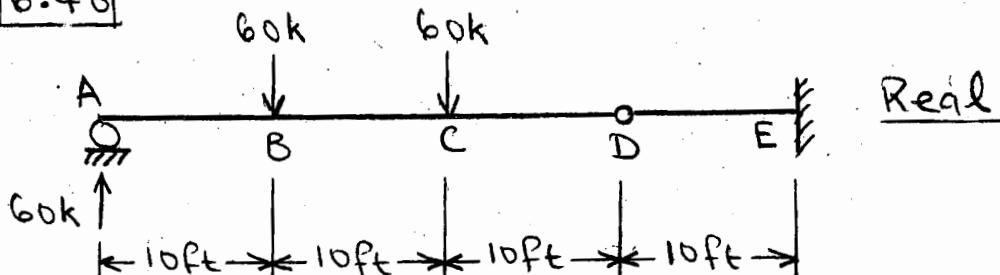
$$\left(2.46 \right) \left(\frac{2.46}{3} \right) \] = -\frac{11980.41}{EI} = \frac{11980.41 \text{ kN}\cdot\text{m}^3}{EI} \downarrow$$

$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{11980.41}{30(10^6)I} = \frac{15}{360}$$

$$\text{From which, } I = 9584(10^6) \text{ m}^4 = \underline{\underline{9584(10^6) \text{ mm}^4}}$$

6.48



$$+G \sum M_D = 0$$

$$A_y(30) - \frac{1}{2} \left(\frac{600}{EI}\right) 10 \left(\frac{10}{3} + 20\right) - \left(\frac{600}{EI}\right)(10)(15)$$

$$- \frac{1}{2} \left(\frac{600}{EI}\right)(10)\left(\frac{20}{3}\right) - \frac{1}{2} \left(\frac{600}{EI}\right) 10 \left(\frac{20}{3}\right) = 0$$

$$A_y = \frac{6666.67}{EI} k \cdot ft^2$$

Let the maximum bending moment in the conjugate beam occur at point M, at a distance x_M from point B. Then the shear at M must be zero. Thus,

$$S_M = \frac{1}{EI} \left[-6666.67 + \frac{1}{2} (600) 10 + (600)x_M \right] = 0$$

$$x_M = 6.11 \text{ ft}$$

$$\Delta_{max} = \Delta_M = \frac{1}{EI} \left[-6666.67(6.11) + \frac{1}{2} (600)(10) \right]$$

$$\left(\frac{10}{3} + 6.11 \right) + (600)(6.11)\left(\frac{6.11}{2}\right)$$

$$= -\frac{67870}{EI} = \frac{67870 \text{ k-ft}^3}{EI}$$

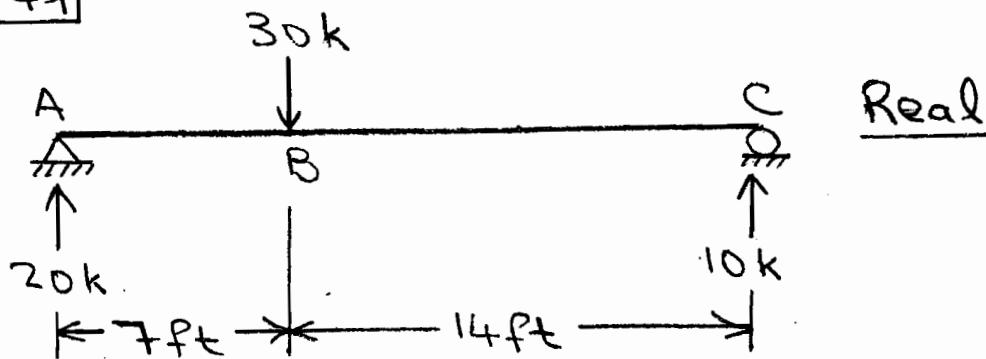
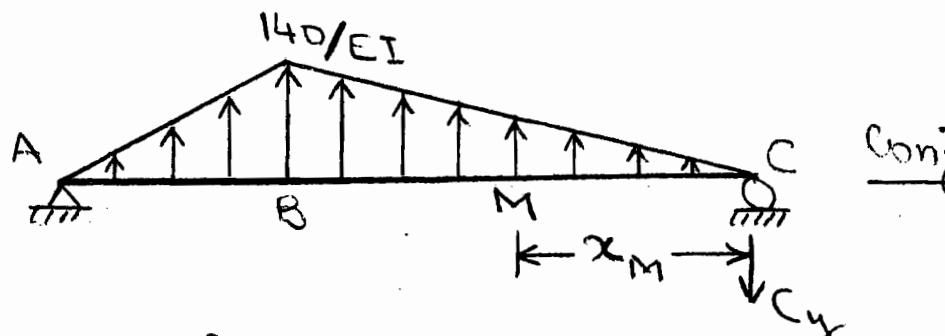
$$\Delta_{max} = \frac{L}{360}$$

$$\frac{67870(12)^3}{29000(I)} = \frac{40(12)}{360}$$

from which

$$I = 3033 \text{ in}^4$$

6.49

RealConjugate

Reaction for conjugate beam:

$$+G \sum M_A = 0$$

$$-C_y(21) + \frac{1}{2} \left(\frac{140}{EI}\right) 14 \left(\frac{14}{3} + 7\right) + \frac{1}{2} \left(\frac{140}{EI}\right) 7 \left(\frac{14}{3}\right) = 0$$

$$C_y = \frac{653.33 \text{ k-ft}^2}{EI}$$

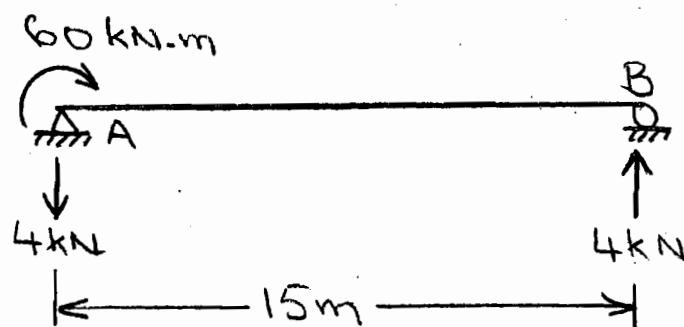
Let the maximum bending moment in the conjugate beam occur at point M, at a distance x_M from point C. Then, the shear at M must be zero. Thus,

$$S_M = \frac{1}{EI} \left[653.33 - \frac{1}{2} (10x_M)x_M \right] = 0$$

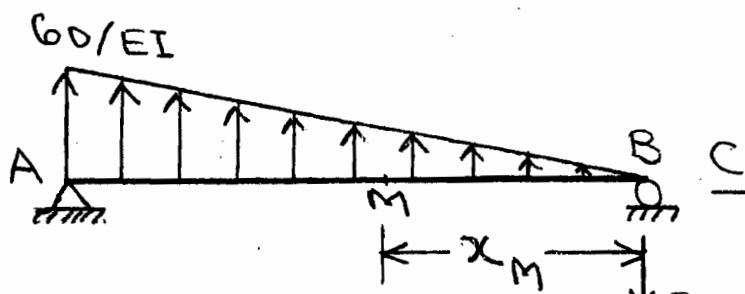
$$x_M = 11.43 \text{ ft}$$

$$\begin{aligned} \Delta_{max} &= \Delta_M = \frac{1}{EI} \left[-653.33 (11.43) + \frac{1}{2} (114.3) 11.43 \left(\frac{11.43}{3}\right)^3 \right] \\ &= -\frac{4978.81 \text{ k-ft}^3}{EI} = -\frac{4978.81 (12)^3}{10000 (500)} \\ &= -1.72 \text{ in.} = \underline{1.72 \text{ in.} \downarrow} \end{aligned}$$

6.50



Real



Conjugate

Reaction for conjugate beam:

$$+\zeta \sum M_A = 0$$

$$-B_y(15) + \frac{1}{2} \left(\frac{60}{EI}\right) 15 (5) = 0$$

$$B_y = \frac{150 \text{ kN} \cdot \text{m}^2}{EI}$$

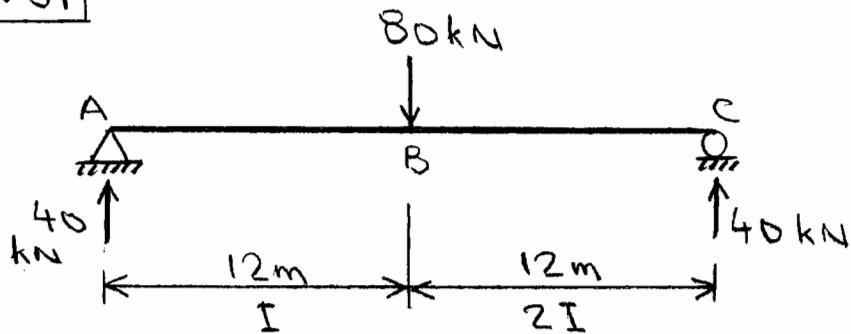
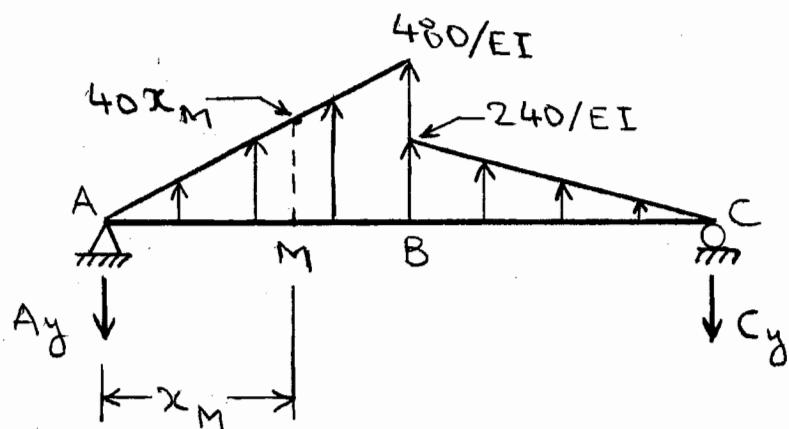
Let the maximum bending moment in the conjugate beam occur at point M, at a distance x_M from point B. Then the shear at M must be zero. Thus,

$$S_M = \frac{1}{EI} [150 - \frac{1}{2} (4x_M)x_M] = 0$$

$$x_M = 8.66 \text{ m}$$

$$\begin{aligned} \Delta_{\max} &= \Delta_M = \frac{1}{EI} \left[-150(8.66) + \frac{1}{2} (34.64)(8.66)\left(\frac{8.66}{3}\right) \right] \\ &= -\frac{866 \text{ kN} \cdot \text{m}^3}{EI} = -\frac{866}{70(712)} = -0.0174 \text{ m} \\ &= \underline{\underline{17.4 \text{ mm} \downarrow}} \end{aligned}$$

6.51

RealConjugate

$$+ \text{G} \sum M_C = 0$$

$$Ay(24) - \frac{1}{2} \left(\frac{480}{EI} \right) (12)(16) - \frac{1}{2} \left(\frac{240}{EI} \right) (12)(8) = 0$$

$$Ay = \frac{2400 \text{ kN} \cdot \text{m}^2}{EI}$$

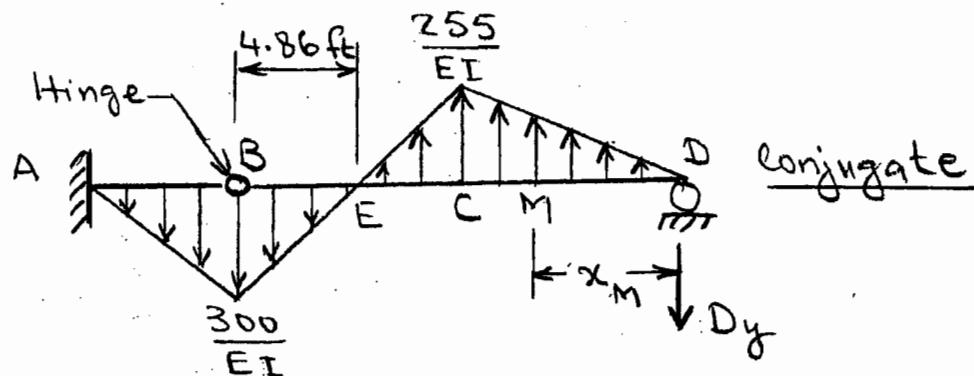
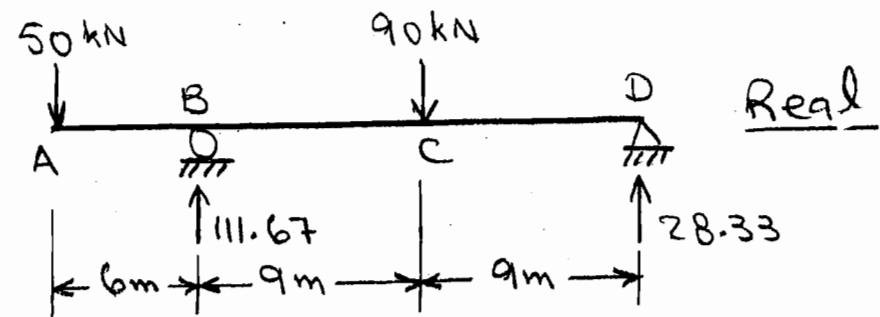
$$S_M = 0$$

$$\frac{1}{EI} \left[-2400 + \frac{1}{2} (40x_M)x_M \right] = 0 \quad x_M = 10.95 \text{ m}$$

$$\begin{aligned} \Delta_{\max} &= \Delta_M = \frac{1}{EI} \left[-2400 (10.95) + \frac{1}{2} (438)(10.95) \left(\frac{10.95}{3} \right) \right] \\ &= -\frac{17527 \text{ kN} \cdot \text{m}^3}{EI} = -\frac{17527}{200(600)} = -0.146 \text{ m} \end{aligned}$$

$$\underline{\Delta_{\max} = 146 \text{ mm} \downarrow}$$

6.52



$$+\sum_M \frac{B_D}{B} = 0$$

$$-Dy(18) + \frac{1}{2} \left(\frac{255}{EI}\right)(9)(12) + \frac{1}{2} \left(\frac{255}{EI}\right)(4.14)(7.62)$$

$$-\frac{1}{2} \left(\frac{300}{EI}\right)(4.86)\left(\frac{4.86}{3}\right) = 0$$

$$Dy = \frac{922.85 \text{ kN.m}^2}{EI}$$

Let the maximum bending moment in the conjugate beam occur at point M, at a distance x_M from point D. Then the shear at M must be zero. Thus,

$$S_M = \frac{1}{EI} \left[922.85 - \frac{1}{2} \left(\frac{255}{9}\right)x_M x_M \right] = 0$$

$$x_M = 8.07 \text{ m}$$

$$\Delta_{max} = \Delta_M = \frac{1}{EI} \left[-922.85(8.07) + \frac{1}{2} \left(\frac{255}{9}\right)(8.07) \cdot \right.$$

$$\left. (8.07) \left(\frac{8.07}{3}\right) \right] = -\frac{4965 \text{ kN.m}^3}{EI}$$

$$= -\frac{4965}{70(9.5)} = -0.746 \text{ m} = \underline{\underline{746 \text{ mm}}}$$

6.52 (cont'd.)

$$1.67 x_M^2 + 233.33 x_M - 1314.8 = 0$$

$$x_M = 5.42 \text{ ft}$$

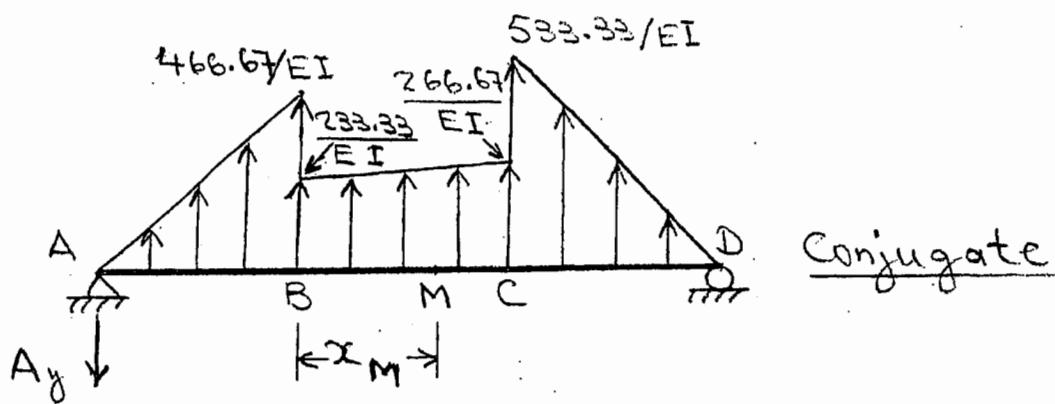
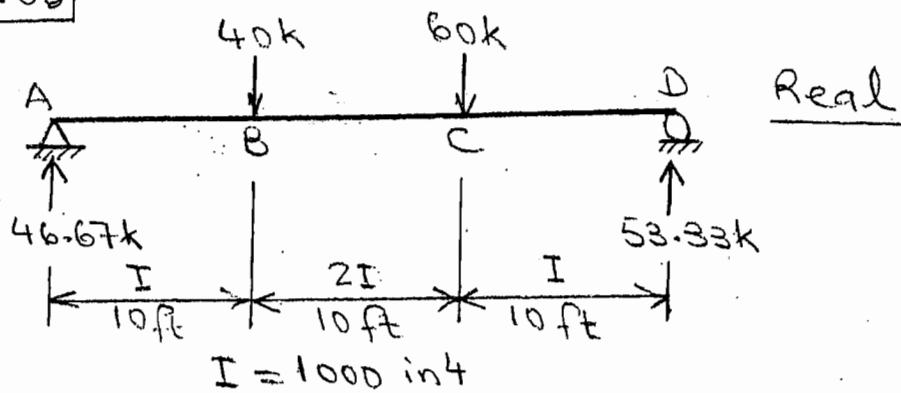
$$\Delta_{\max} = \Delta_M = \frac{1}{EI} \left[-3648.15 (15.42) \right]$$

$$+ \frac{1}{2} (466.67) 10 \left(\frac{10}{3} + 5.42 \right) + 233.33 (5.42) \left(\frac{5.42}{2} \right)$$

$$+ \frac{1}{2} (18.06) 5.42 \left(\frac{5.42}{3} \right) \right] = -\frac{32314}{EI} \text{ k-ft}^3$$

$$= -\frac{32314 (12)^3}{29000 (1000)} = -1.92 \text{ in.} = \underline{1.92 \text{ in.} \downarrow}$$

6.53



6.53 (cont d.)

Reaction for conjugate beam:

$$+ \zeta \sum M_D = 0$$

$$A_y(30) - \frac{1}{EI} \left[\frac{1}{2} (466.67) 10 \left(\frac{10}{3} + 20 \right) + 233.33 (10) 15 \right.$$

$$\left. + \frac{1}{2} (33.33) 10 \left(\frac{10}{3} + 10 \right) + \frac{1}{2} (533.33) 10 \left(\frac{20}{3} \right) \right] = 0$$

$$A_y = \frac{3648.15 \text{ k-ft}^2}{EI}$$

Let the maximum bending moment in the conjugate beam occur at point M, at a distance x_M from point B. Then the shear at M must be zero. Thus,

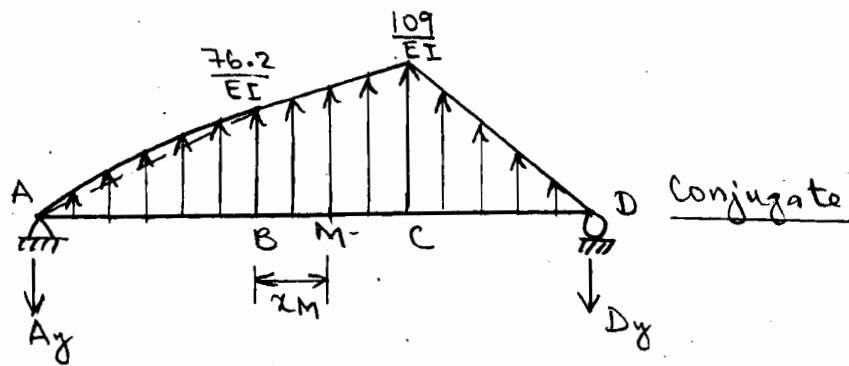
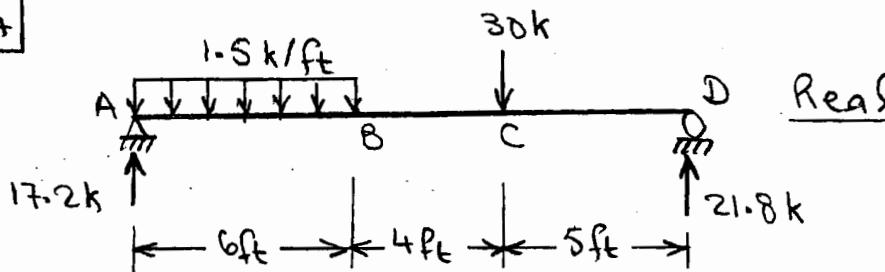
$$S_M = \frac{1}{EI} \left[-3648.15 + \frac{1}{2} (466.67) 10 + 233.33 x_M + \frac{1}{2} (33.33 x_M) x_M \right] = 0$$

$$1.67 x_M^2 + 233.33 x_M - 1314.8 = 0$$

$$x_M = 5.42 \text{ ft}$$

$$\begin{aligned} \Delta_{\max} &= \Delta_M = \frac{1}{EI} \left[-3648.15 (15.42) \right. \\ &+ \frac{1}{2} (466.67) 10 \left(\frac{10}{3} + 5.42 \right) + 233.33 (5.42) \left(\frac{5.42}{2} \right) \\ &\left. + \frac{1}{2} (33.33) 5.42 \left(\frac{5.42}{3} \right) \right] = -\frac{32314 \text{ k-ft}^3}{EI} \\ &= -\frac{32314 (12)^3}{29000 (1000)} = -1.92 \text{ in.} = \underline{\underline{1.92 \text{ in.}} \downarrow} \end{aligned}$$

6.54



$$+G \sum M_D = 0$$

$$Ay(15) - \frac{1}{EI} \left[\frac{1}{2}(76.2)6(11) + \frac{2}{3} \frac{(1.5)(6)^2}{8}(6)(12) + 76.2(4)7 + \frac{1}{2}(32.8)(4)\left(\frac{4}{3} + 5\right) + \frac{1}{2}(109)5\left(\frac{10}{3}\right) \right] = 0$$

$$Ay = \frac{419.73 \text{ k-ft}^2}{EI}$$

Let the maximum bending moment in the conjugate beam occur at point M. Thus,

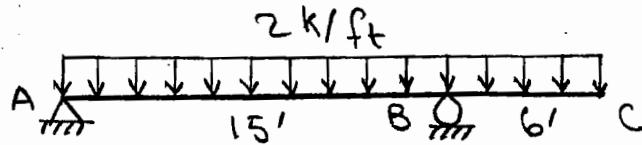
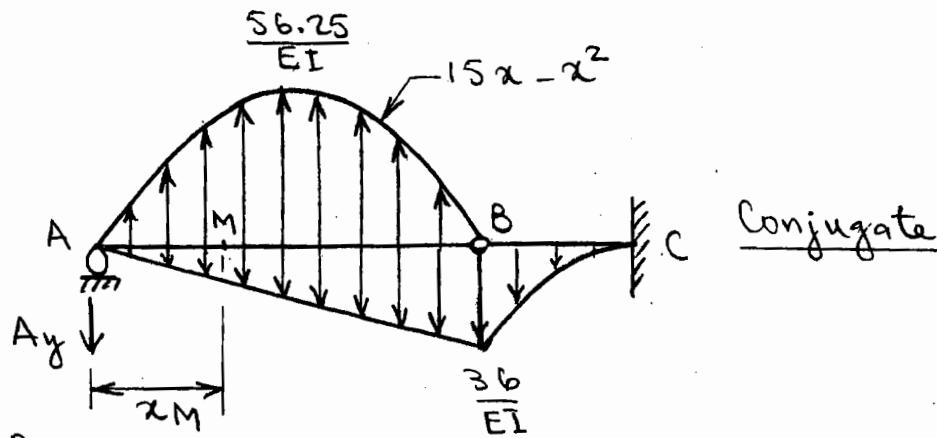
$$3M = \frac{1}{EI} \left[-419.73 + \frac{1}{2}(76.2)6 + \frac{2}{3} \frac{(1.5)(6)^2}{8}(6) + 76.2x_M + \frac{1}{2}(8.2x_M)x_M \right] = 0$$

$$4.1x_M^2 + 76.2x_M - 164.13 = 0$$

$$x_M = 1.95 \text{ ft}$$

$$\begin{aligned} \Delta_{max} &= \Delta_M = \frac{1}{EI} \left[-419.73(7.95) + \frac{1}{2}(76.2)6(3.95) + \frac{2}{3} \frac{(1.5)(6)^2}{8}(6)(4.95) + 76.2 \frac{(1.95)^2}{2} + \frac{1}{2}(15.99)(1.95)\left(\frac{1.95}{3}\right) \right] = -\frac{2145.2 \text{ k-ft}^3}{EI} \\ &= -\frac{2145.2(12)^3}{1500(20000)} = -0.124 \text{ in} = \underline{0.124 \text{ in. } \downarrow} \end{aligned}$$

6.55

Real

$$+\zeta \sum M_B^A = 0$$

$$A_y(15) - \frac{1}{EI} \left[\frac{2}{3} (56.25) 15 (7.5) - \frac{1}{2} (36) 15 (5) \right] = 0$$

$$A_y = \frac{191.25 \text{ k-ft}^2}{EI}$$

$$\delta_M = 0$$

$$\frac{1}{EI} \left[-191.25 + \frac{1}{2} (15x_M - x_M^2)x_M + \frac{2}{3} \left(\frac{2x_M^2}{8} \right) x_M - \frac{1}{2} (2.4x_M)x_M \right] = 0$$

$$-\frac{x_M^3}{3} + 6.3x_M^2 - 191.25 = 0$$

$$\text{from which } x_M = 6.92 \text{ ft}$$

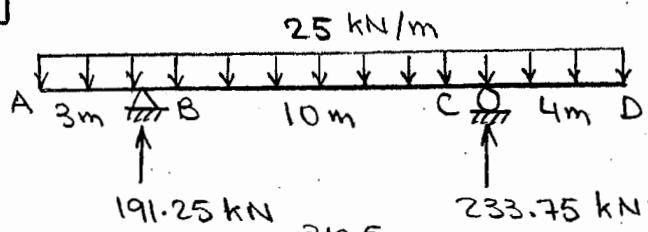
$$\Delta_{max} = \Delta_M = \frac{1}{EI} \left[-191.25 (6.92) + \frac{1}{2} (55.91) \frac{(6.92)^2}{3} \right]$$

$$+ \frac{2}{3} (11.97) \frac{(6.92)^2}{2} - \frac{1}{2} (16.61) 6.92 \left(\frac{6.92}{3} \right)$$

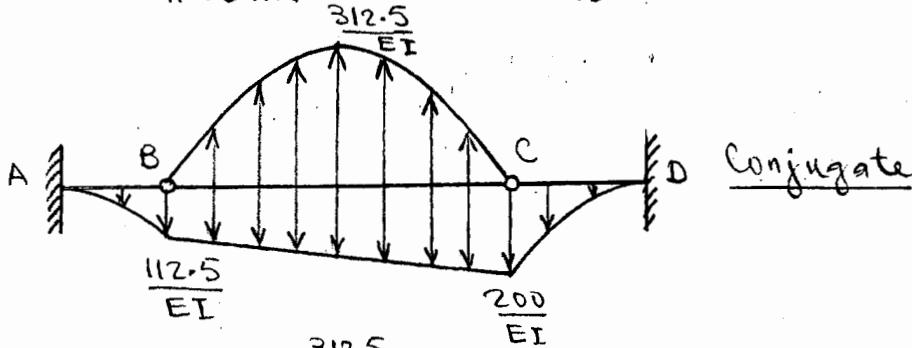
$$= -\frac{818.73 \text{ k-ft}^3}{EI} = -0.0139 \text{ in}$$

$$= \underline{0.0139 \text{ in.} \downarrow}$$

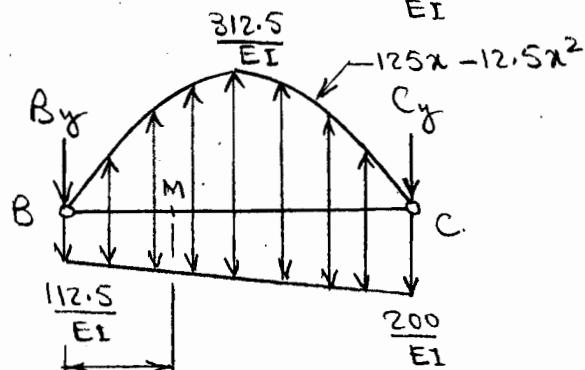
6.56



Real



Conjugate



$$+C \sum M_{C}^{BC} = 0$$

$$B_y(10) - \frac{1}{EI} \left[\frac{2}{3} (312.5) 10(5) - 112.5(10)5 \right]$$

$$- \frac{1}{2} (87.5) 10 \left(\frac{10}{3} \right) = 0$$

$$B_y = \frac{333.33 \text{ kN-m}^2}{EI}$$

$$\delta_M = 0$$

$$\frac{1}{EI} \left[-333.33 + \frac{1}{2} (125x_M - 12.5x_M^2)x_M \right]$$

$$+ \frac{2}{3} \left(\frac{25x_M^2}{8} \right) x_M - 112.5 x_M - \frac{1}{2} (8.75x_M) x_M \right] = 0$$

$$-4.167 x_M^3 + 58.125 x_M^2 - 112.5 x_M - 333.33 = 0$$

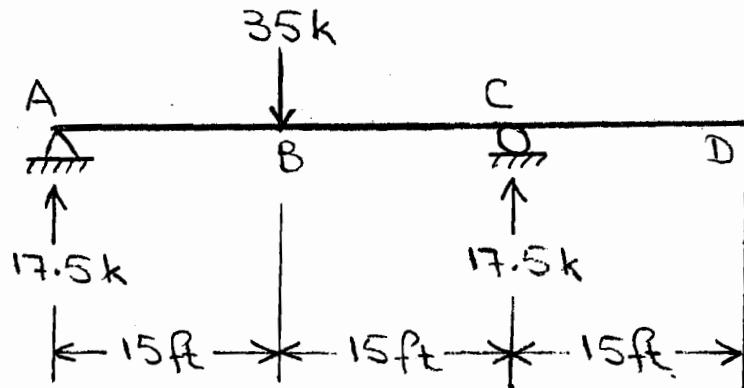
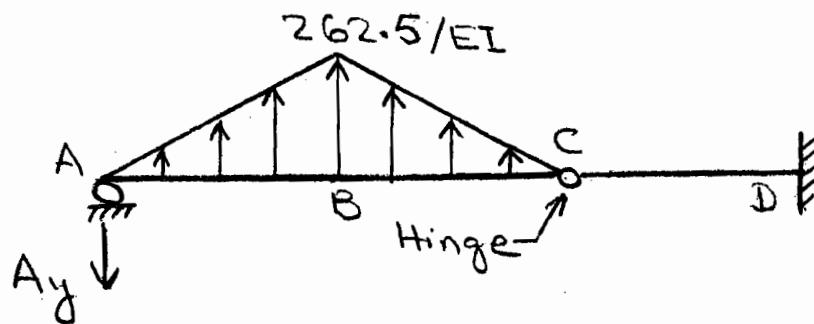
from which $x_M = 4.77 \text{ m}$

$$\Delta_{\max} = \Delta_M = \frac{1}{EI} \left[-333.33(4.77) + \frac{1}{2} (311.84) \frac{(4.77)^2}{3} \right]$$

$$+ \frac{2}{3} \left(71.1 \right) \frac{(4.77)^2}{2} - 112.5 \frac{(4.77)^2}{2} - \frac{1}{2} (41.74) \frac{(4.77)^2}{3} \right]$$

$$= - \frac{1306.3 \text{ kN-m}^3}{EI} = - 0.0131 \text{ m} = \underline{13.1 \text{ mm} \downarrow}$$

6.57

RealConjugate

$$+G \sum M_C^{AC} = 0$$

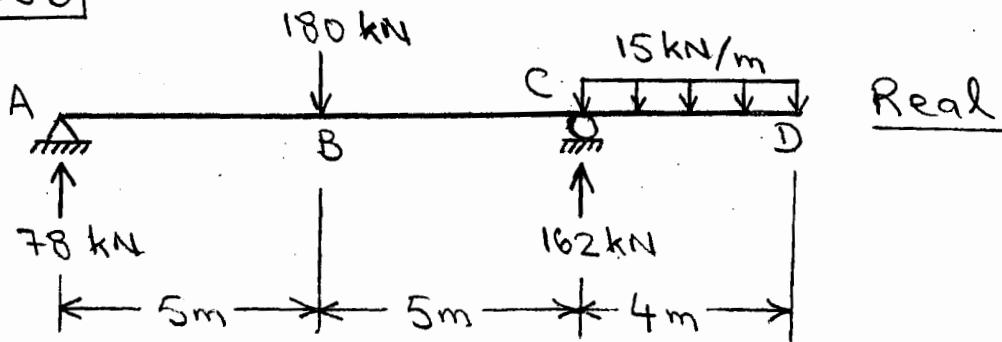
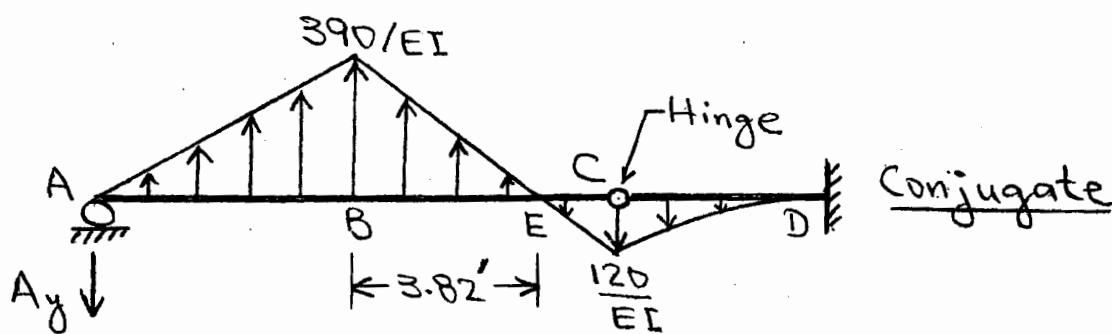
$$A_y(30) - \frac{1}{2} \left(\frac{262.5}{EI} \right) 30(15) = 0$$

$$A_y = \frac{1968.75 \text{ k-ft}^2}{EI}$$

$$\begin{aligned} \Theta_D &= \frac{1}{EI} \left[-1968.75 + \frac{1}{2} (262.5) 30 \right] = \frac{1968.75 \text{ k-ft}^2}{EI} \\ &= \frac{1968.75 (12)^2}{10000 (2500)} = \underline{\underline{0.01134 \text{ rad}}} \end{aligned}$$

$$\begin{aligned} \Delta_D &= \frac{1}{EI} \left[-1968.75 (45) + \frac{1}{2} (262.5) 30 (30) \right] \\ &= \frac{29531.25 \text{ k-ft}^3}{EI} = \frac{29531.25 (12)^3}{10000 (2500)} \\ &= \underline{\underline{2.04 \text{ in. up}}} \end{aligned}$$

6.58

RealConjugate

$$+\zeta \sum M_C^{AC} = 0$$

$$A_y(1.0) - \frac{1}{EI} \left[\frac{1}{2}(390)5\left(\frac{5}{3} + 5\right) + \frac{1}{2}(390)(3.82)(3.73) - \frac{1}{2}(120)1.18\left(\frac{1.18}{3}\right) \right] = 0$$

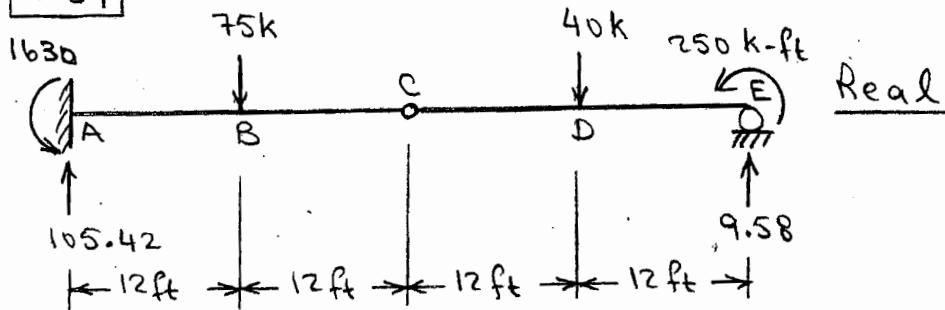
$$A_y = \frac{925.1 \text{ kN} \cdot \text{m}^2}{EI}$$

$$\Theta_D = \frac{1}{EI} \left[-925.1 + \frac{1}{2}(390)8.82 - \frac{1}{2}(120)1.18 - \frac{1}{3}(120)4 \right]$$

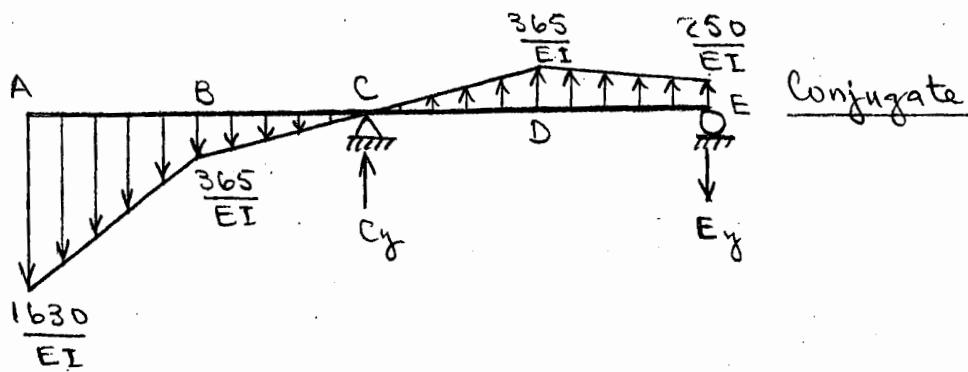
$$= \frac{564 \text{ kN} \cdot \text{m}^2}{EI} = \frac{564}{70(2340)} = \underline{0.00344 \text{ rad}}$$

$$\begin{aligned} \Delta_D &= \frac{1}{EI} \left[-925.1(14) + \frac{1}{2}(390)5(10.67) + \frac{1}{2}(390)(3.82)(7.727) - \frac{1}{2}(120)1.18(4.393) - \frac{1}{3}(120)4(3) \right] = \frac{2417 \text{ kN} \cdot \text{m}^3}{EI} \\ &= \frac{2417}{70(2340)} = 0.01476 \text{ m} = \underline{14.76 \text{ mm}} \uparrow \end{aligned}$$

6.59



Real



Conjugate

$$+\curvearrowleft \sum M_E = 0$$

$$\begin{aligned} \frac{1}{EI} & \left[\frac{1}{2} (1265) 12 (44) + 365 (12) 42 + \frac{1}{2} (365) 12 (32) \right. \\ & \quad \left. - \frac{1}{2} (365) 12 (16) - \frac{1}{2} (115) 12 (8) - 250 (12) 6 \right] \\ & - C_y (24) = 0 \end{aligned}$$

$$C_y = \frac{22060 \text{ k-ft}^2}{EI} \uparrow$$

$$+\uparrow \sum F_y = 0$$

$$\frac{1}{EI} \left[-\frac{1}{2} (1630 + 365) 12 + \frac{1}{2} (365 + 250) 12 + 22060 \right]$$

$$- E_y = 0$$

$$E_y = \frac{13780 \text{ k-ft}^2}{EI} \downarrow$$

$$\Theta_B = -\frac{1}{EI} \left[\frac{1}{2} (1630 + 365) 12 \right] = -\frac{11970 \text{ k-ft}^2}{EI} = 0.0099 \text{ rad} \times$$

$$\Delta_B = -\frac{1}{EI} \left[365 (12) 6 + \frac{1}{2} (1265) 12 (8) \right] = -\frac{87000 \text{ k-ft}^3}{EI}$$

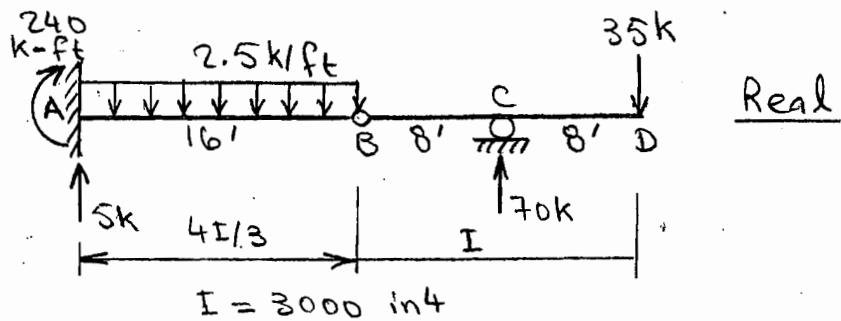
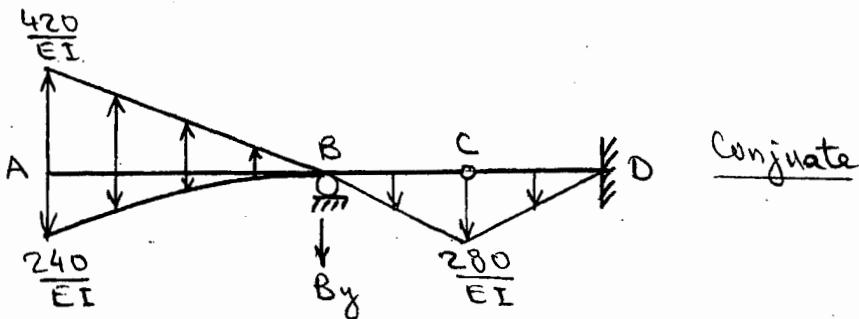
$$\Delta_B = 0.86 \text{ in} \downarrow$$

$$\Theta_D = \frac{1}{EI} \left[13780 - \frac{1}{2} (365 + 250) 12 \right] = \frac{10090}{EI} = 0.0084 \text{ rad} \triangle$$

$$\Delta_D = \frac{1}{EI} \left[-13780 (12) + 250 (12) 6 + \frac{1}{2} (115) 12 (4) \right]$$

$$= -144600 \text{ k-ft}^3 = 1.44 \text{ in} \downarrow$$

6.60

RealConjugate

$$+C \sum M_C = 0$$

$$\frac{1}{EI} \left[\frac{1}{3} (240) 16(20) - \frac{1}{2} (420) 16(18.67) + \frac{1}{2} (280) 8\left(\frac{8}{3}\right) \right]$$

$$+By(8) = 0$$

$$By = \frac{4266.67 \text{ k-ft}^2}{EI} \downarrow$$

$$\begin{aligned} \Theta_{B,\text{Left}} &= \frac{1}{EI} \left[\frac{1}{2} (420) 16 - \frac{1}{3} (240) 16 \right] = \frac{2080 \text{ k-ft}^2}{EI} \\ &= \frac{2080 (12)^2}{30000 (3000)} = \underline{0.0033 \text{ rad. } \uparrow} \end{aligned}$$

$$\begin{aligned} \Theta_{B,\text{Right}} &= \frac{1}{EI} (2080 - 4266.67) = -\frac{2186.67 \text{ k-ft}^2}{EI} \\ &= \underline{-0.0035 \text{ rad. } \downarrow} \end{aligned}$$

$$\begin{aligned} \Delta_B &= \frac{1}{EI} \left[\frac{1}{2} (420) 16(10.67) - \frac{1}{3} (240) 16(12) \right] \\ &= \frac{20480 \text{ k-ft}^3}{EI} = \frac{20480 (12)^3}{30000 (3000)} = \underline{0.39 \text{ in. } \uparrow} \end{aligned}$$

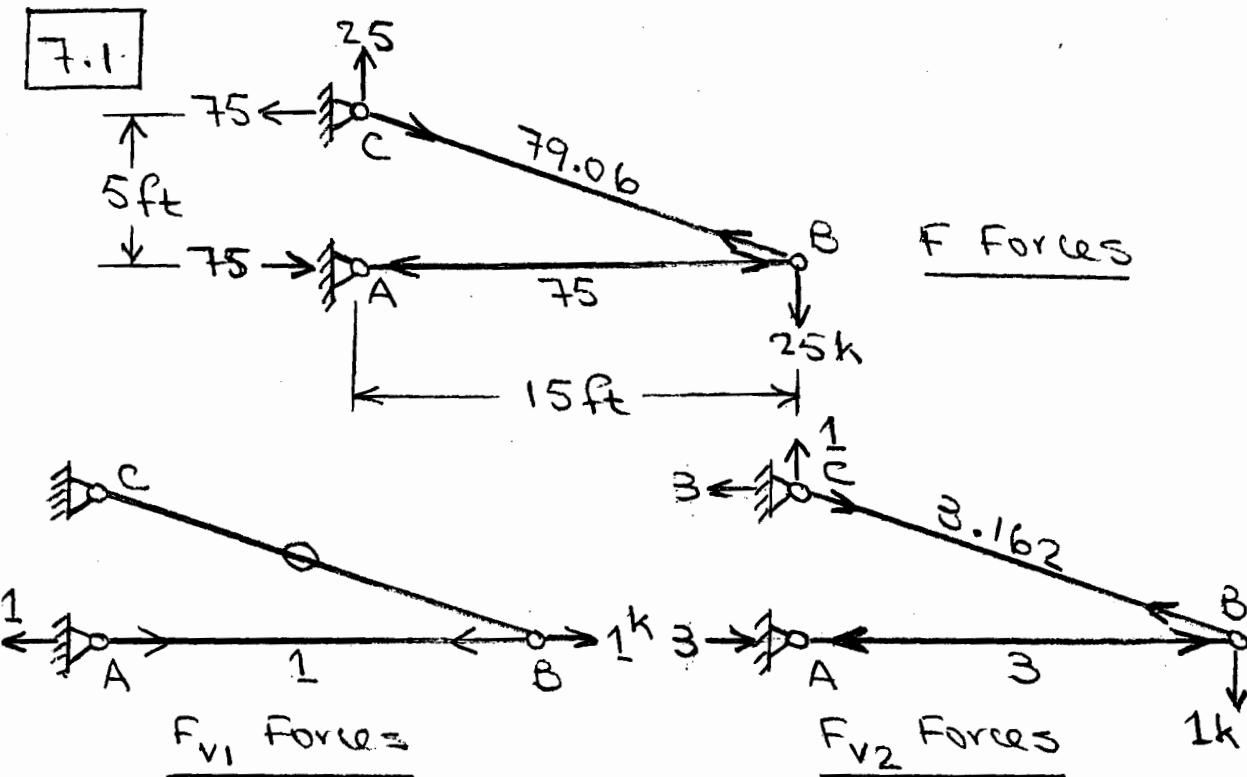
$$\begin{aligned} \Theta_D &= \frac{1}{EI} \left[\frac{1}{2} (420) 16 - \frac{1}{3} (240) 16 - 4266.67 - \frac{1}{2} (280) 16 \right] \\ &= -\frac{4426.67 \text{ k-ft}^2}{EI} = -0.0071 \text{ rad. } = \underline{0.0071 \text{ rad. } \downarrow} \end{aligned}$$

$$\begin{aligned} \Delta_D &= \frac{1}{EI} \left[\frac{1}{2} (420) 16(26.67) - \frac{1}{3} (240) 16(28) - 4266.67(16) \right. \\ &\quad \left. - \frac{1}{2} (280) 16(8) \right] = -\frac{32426.67}{EI} = \underline{0.62 \text{ in. } \downarrow} \end{aligned}$$

Chapter Seven

Deflections of Trusses, Beams, and Frames: Work-Energy Methods

CHAPTER 7



Member	L (in)	F (k)	F_{V1} (k)	$F_{V1}(FL)$ (k^2 -in)	F_{V2} (k)	$F_{V2}(FL)$ (k^2 -in)
AB	180	-75	1	-13500	-3	40500
BC	189.74	79.06	0	0	3.162	47432.67
\sum				-13500		87932.67

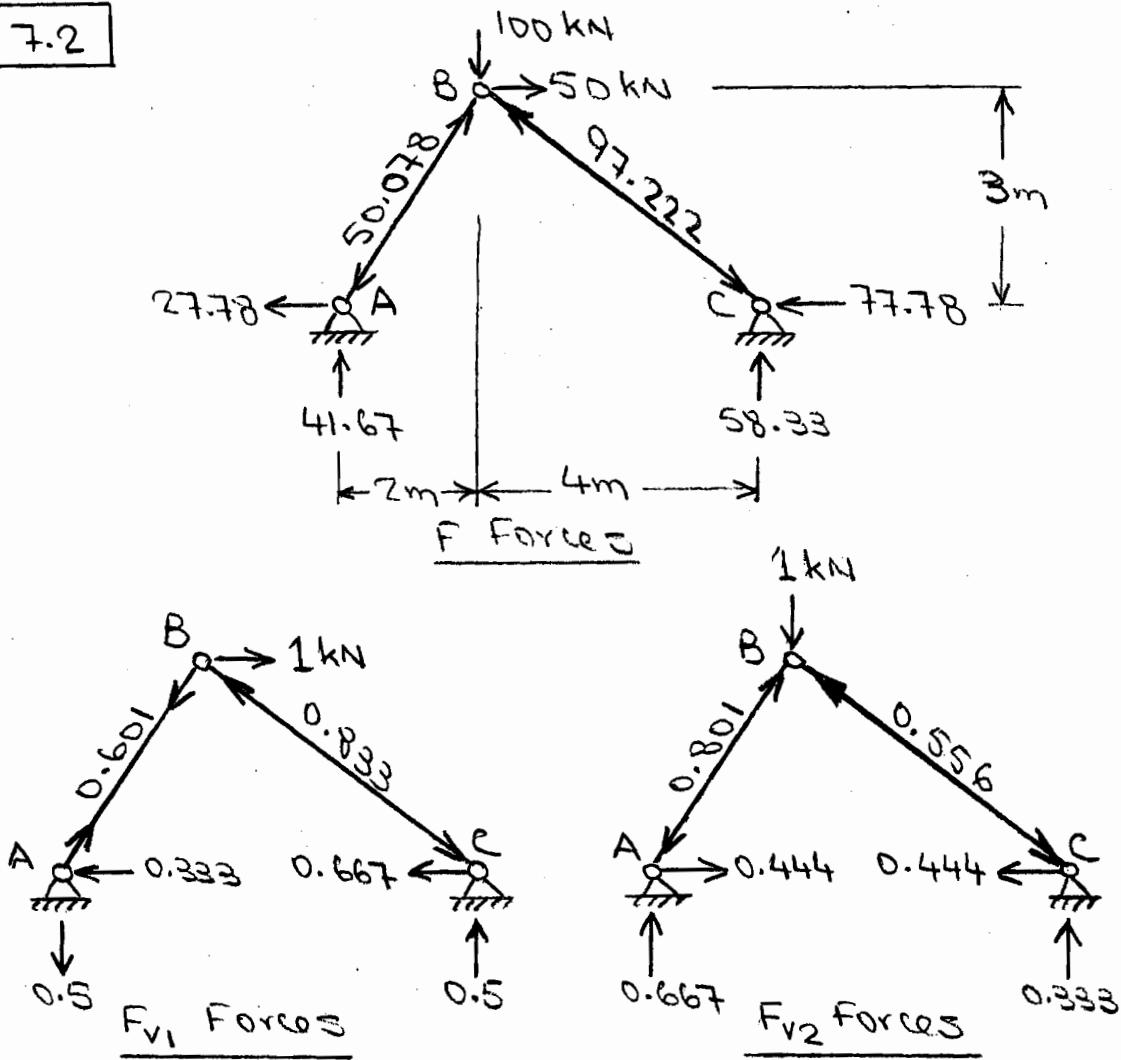
$$(1k) \Delta_{EH} = -\frac{13500}{10000(6)} = -0.225 \text{ k-in.}$$

$$\Delta_{EH} = -0.225 \text{ in.} = \underline{0.225 \text{ in.} \leftarrow}$$

$$(1k) \Delta_{BV} = \frac{87932.67}{10000(6)} = 1.466 \text{ k-in.}$$

$$\Delta_{BV} = 1.466 \text{ in.} \downarrow$$

7.2



Member	L (m)	F (kN)	F_{V1} (kN)	$F_{V1}(FL)$ ($kN^2 \cdot m$)	F_{V2} (kN)	$F_{V2}(FL)$ ($kN^2 \cdot m$)
AB	3.606	-50.078	0.601	-108.53	-0.801	144.65
BC	5	-97.222	-0.833	405.09	-0.556	270.06
			\sum	296.56		414.71

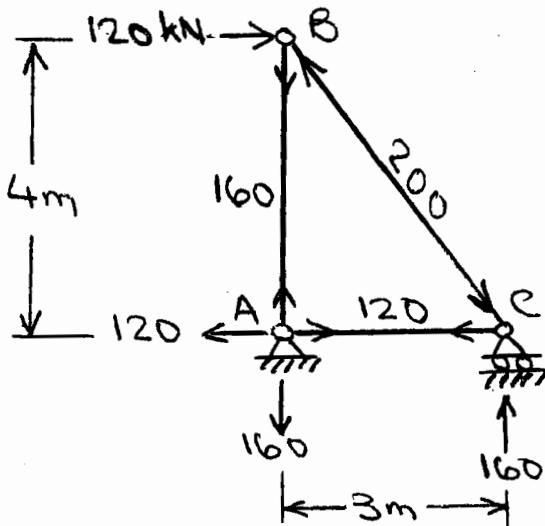
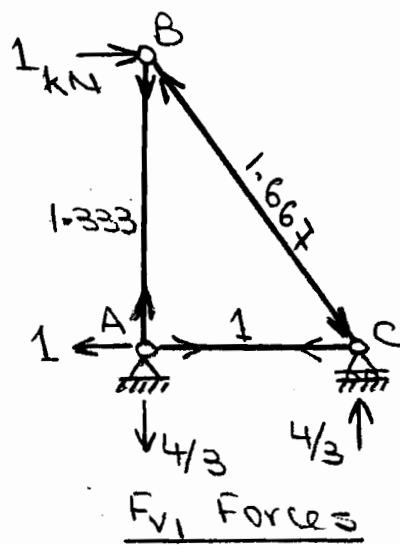
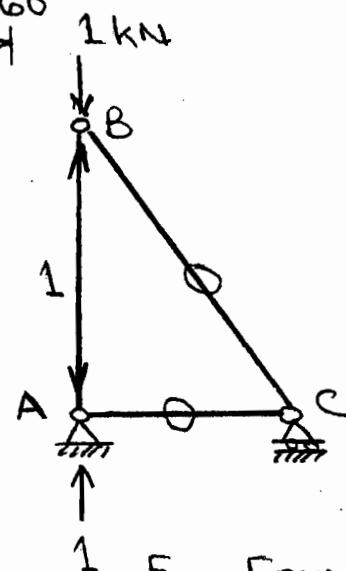
$$(1\text{ kN}) \Delta_{BH} = \frac{296.56}{70(10^6)(0.001)} = 0.00424 \text{ kN} \cdot \text{m}$$

$$\Delta_{BH} = 0.00424 \text{ m} = 4.24 \text{ mm} \rightarrow$$

$$(1\text{ kN}) \Delta_{BV} = \frac{414.71}{70(10^6)(0.001)} = 0.00592 \text{ kN} \cdot \text{m}$$

$$\Delta_{BV} = 0.00592 \text{ m} = 5.92 \text{ mm} \downarrow$$

7.3

F ForcesFv1 ForcesFv2 Forces

Member	L (m)	F (kN)	Fv1 (kN)	Fv1(FL) (kN ² -m)	Fv2 (kN)	Fv2(FL) (kN ² -m)
AC	3	120	1	360	0	0
AB	4	160	1.333	853.33	-1	-640
BC	5	-200	-1.667	1666.67	0	0
			\sum	2880		-640

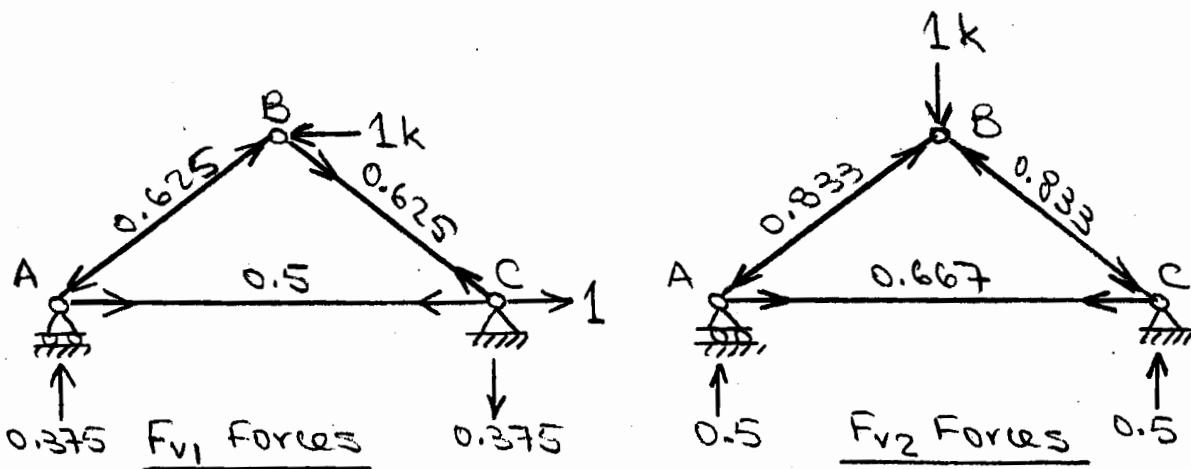
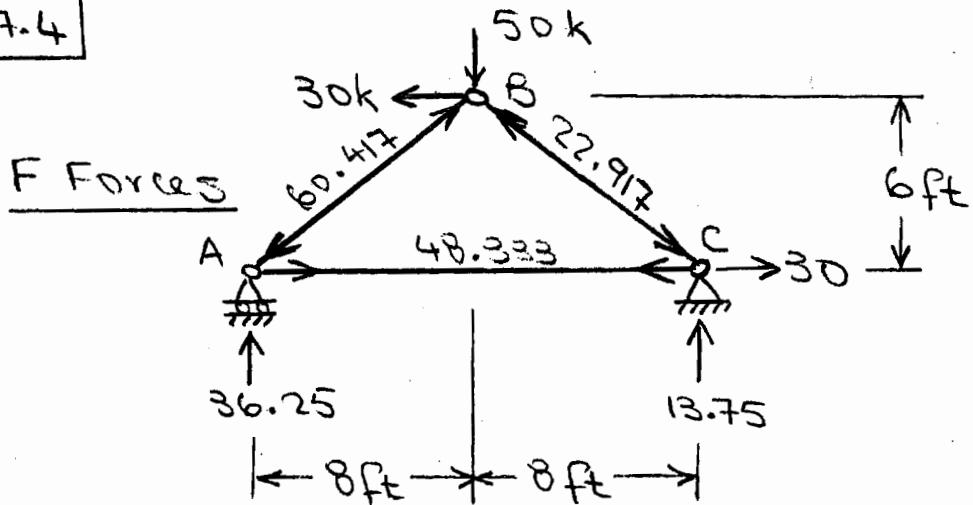
$$(1 \text{ kN}) \Delta_{BH} = \frac{2880}{200(10^6)(0.0015)} = 0.0096 \text{ kN.m}$$

$$\Delta_{BH} = 0.0096 \text{ m} = 9.6 \text{ mm} \rightarrow$$

$$(1 \text{ kN}) \Delta_{BV} = \frac{-640}{200(10^6)(0.0015)} = -0.00213 \text{ kN.m}$$

$$\Delta_{BV} = -0.00213 \text{ m} = 2.13 \text{ mm} \uparrow$$

7.4



Member	L (in.)	F (k)	F _{v1} (k)	F _{v1} (FL) (k ² -in.)	F _{v2} (k)	F _{v2} (FL) (k ² -in.)
AC	192	48.333	0.5	4640	0.667	6186.67
AB	120	-60.417	-0.625	4531.25	-0.833	6041.67
BC	120	-22.917	0.625	-1718.75	-0.833	2291.67
\sum				7452.5		14520

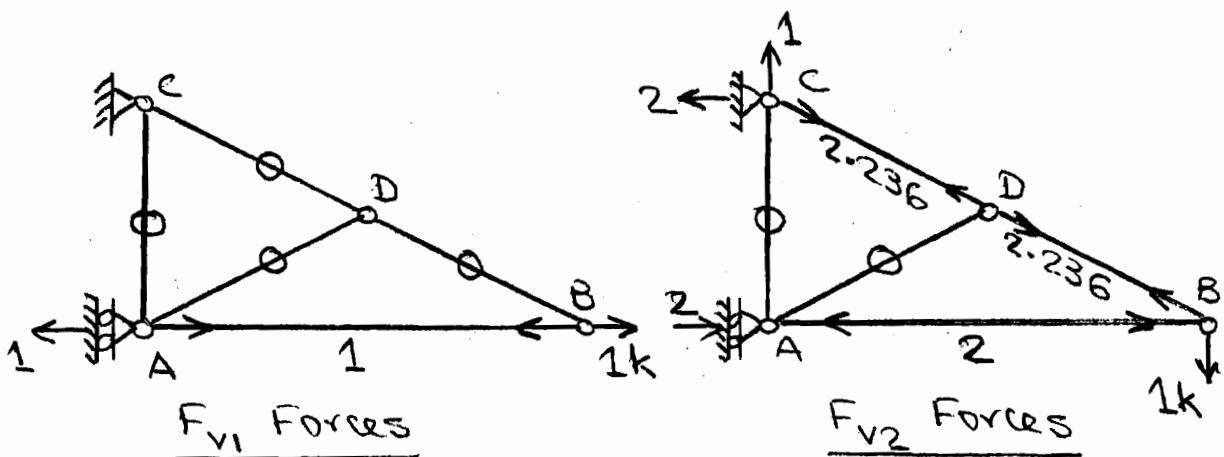
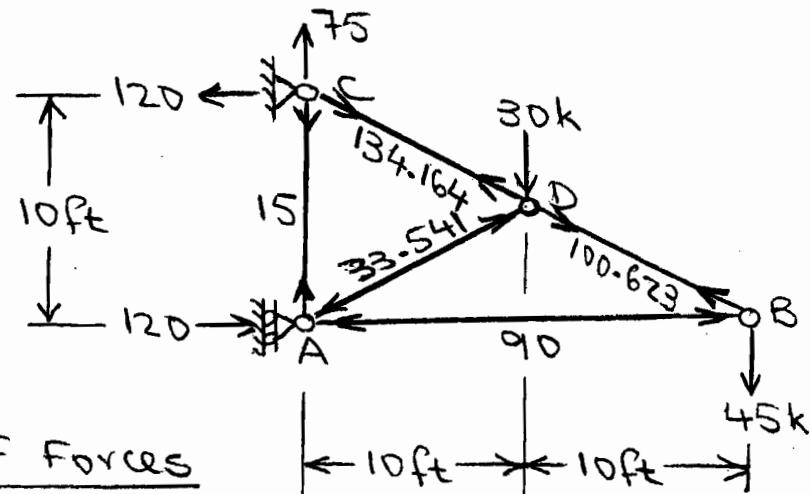
$$(1k) \Delta_{BH} = \frac{7452.5}{29000(3)} = 0.0857 \text{ k-in.}$$

$$\underline{\Delta_{BH} = 0.0857 \text{ in.} \leftarrow}$$

$$(1k) \Delta_{BV} = \frac{14520}{29000(3)} = 0.167 \text{ k-in.}$$

$$\underline{\Delta_{BV} = 0.167 \text{ in.} \downarrow}$$

7.5



Member	L (in.)	A (in ²)	F (k)	F _{v1} (k)	F _{v1} (FL/A) (k ² /in.)	F _{v2} (k)	F _{v2} (FL/A) (k ² /in.)
AB	240	6	-90	1	-3600	-2	7200
AC	120	4	15	0	0	0	0
AD	134.16	4	-33.541	0	0	0	0
CD	134.16	6	134.164	0	0	2.236	6707.99
BD	134.16	6	100.623	0	0	2.236	5030.99
				Σ	-3600		18938.98

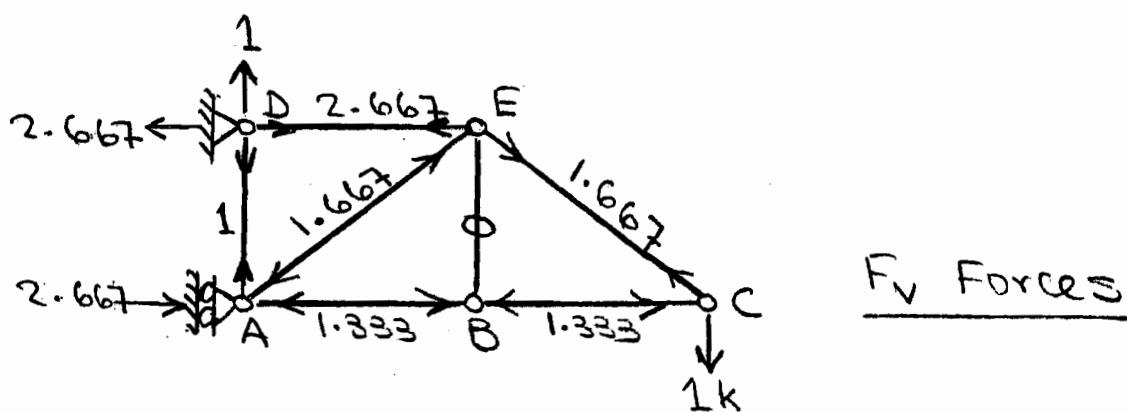
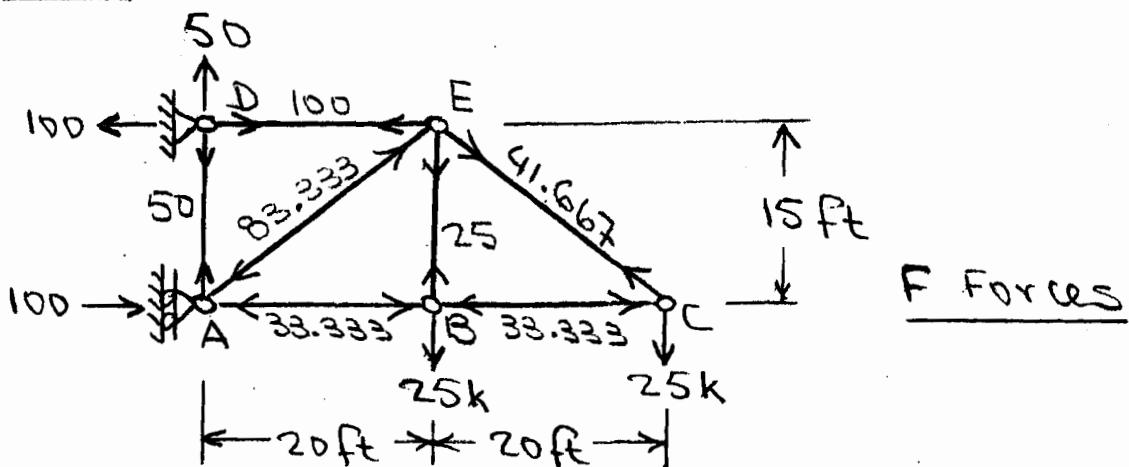
$$(1k) \Delta_{RH} = \frac{-3600}{10000} = -0.36 \text{ k-in}$$

$$\underline{\Delta_{RH} = 0.36 \text{ in.} \leftarrow}$$

$$(1k) \Delta_{BV} = \frac{18938.99}{10000} = 1.894 \text{ k-in.}$$

$$\underline{\Delta_{BV} = 1.894 \text{ in.} \downarrow}$$

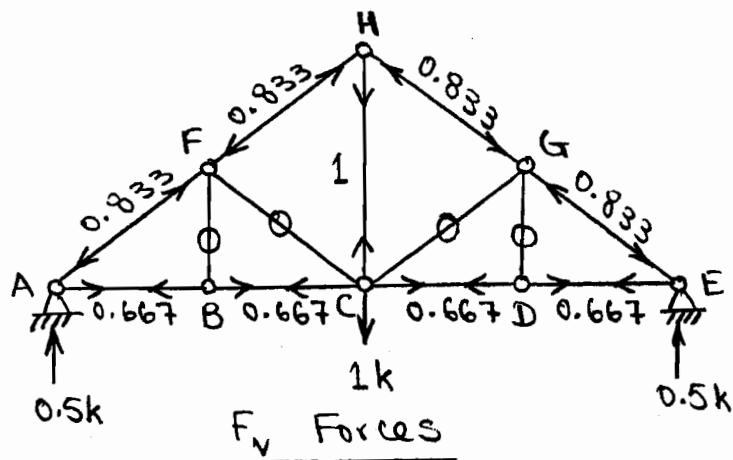
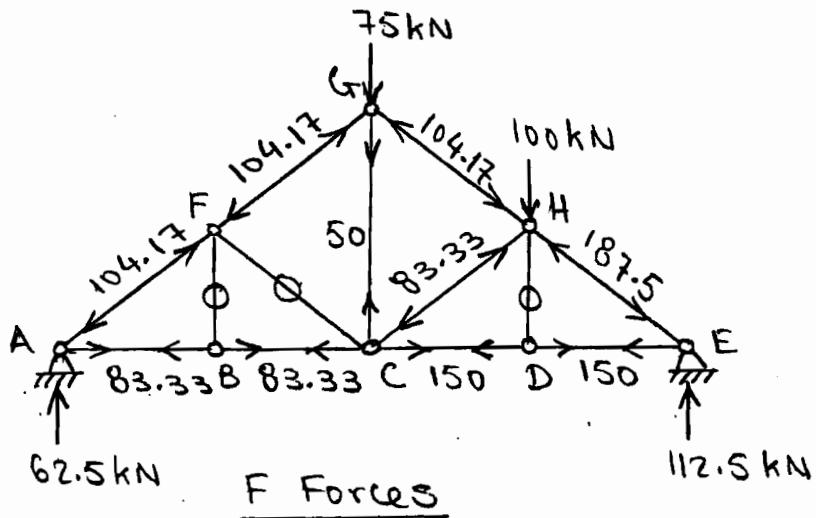
7.6



Member	L (in.)	A (in. ²)	F (k)	F _v (k)	F _v (FL/A) (k ² /in.)
AB	240	4	-33.333	-1.333	2666.67
BC	240	4	-33.333	-1.333	2666.67
DE	240	4	100	2.667	16000
AD	180	3	50	1	3000
BE	180	3	25	0	0
AE	300	3	-83.333	-1.667	13888.89
CE	300	4	41.667	1.667	5208.33
					$\sum 43430.56$

$$\Delta_C = \frac{43430.56}{29000} = 1.498 \text{ in. } \downarrow$$

7.7

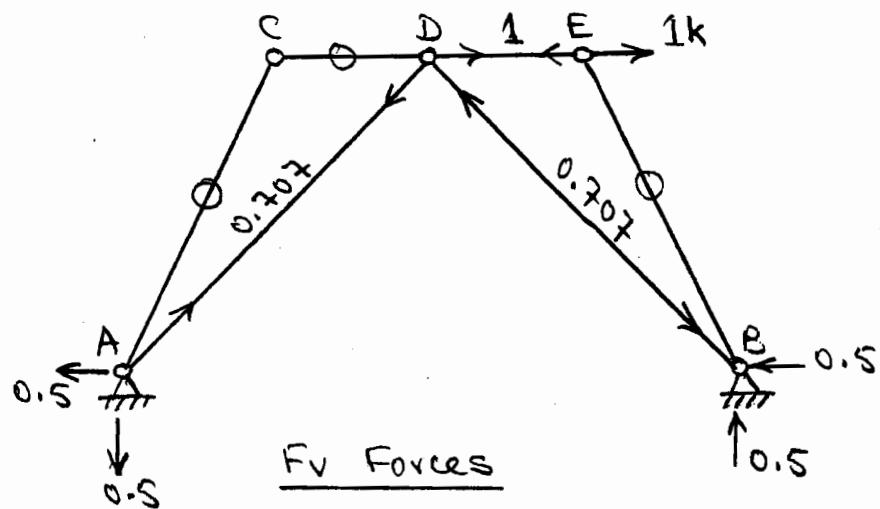
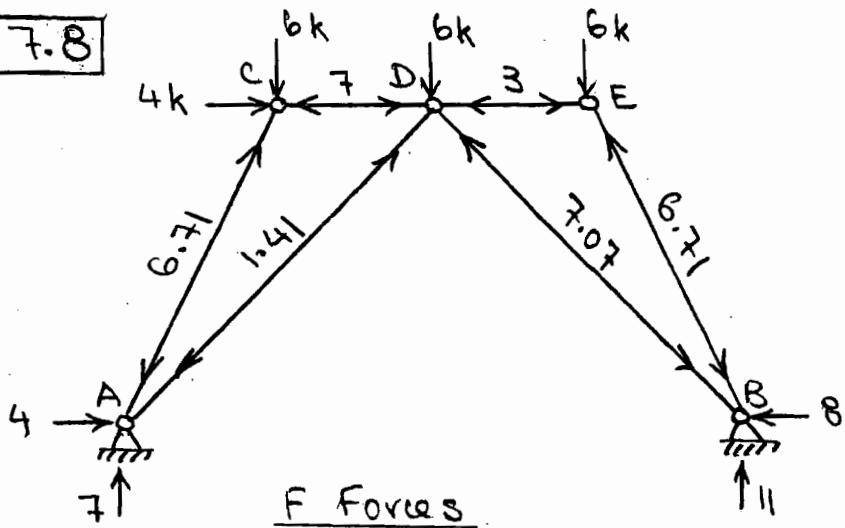


Member	L (m)	F (kN)	F _v (kN)	F _v (FL) (kN ² -m)
AB	6	83.33	0.667	333.5
BC	6	83.33	0.667	333.5
CD	6	150	0.667	600.3
DE	6	150	0.667	600.3
AF	7.5	-104.17	-0.833	650.8
FG	7.5	-104.17	-0.833	650.8
GH	7.5	-104.17	-0.833	650.8
EH	7.5	-187.5	-0.833	1171.4
CG	9	50	1	450
			Σ	5441.4

$$(1 \text{ kN}) \Delta_C = \frac{5441.4}{200(10^6)(0.003)} = 0.0091 \text{ kN-m}$$

$$\Delta_C = 0.0091 \text{ m} = \underline{9.1 \text{ mm}} \downarrow$$

7.8

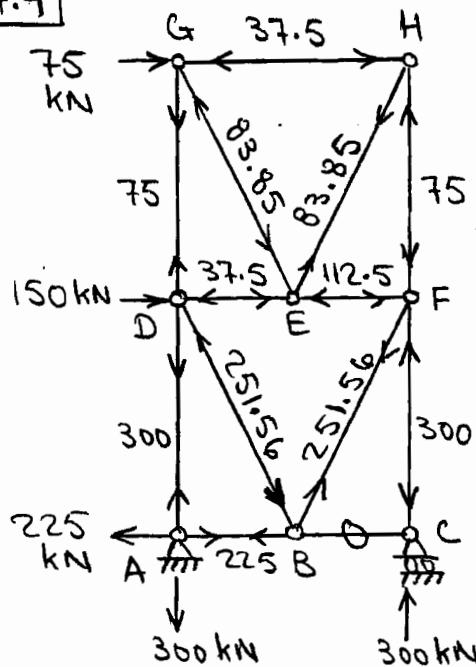
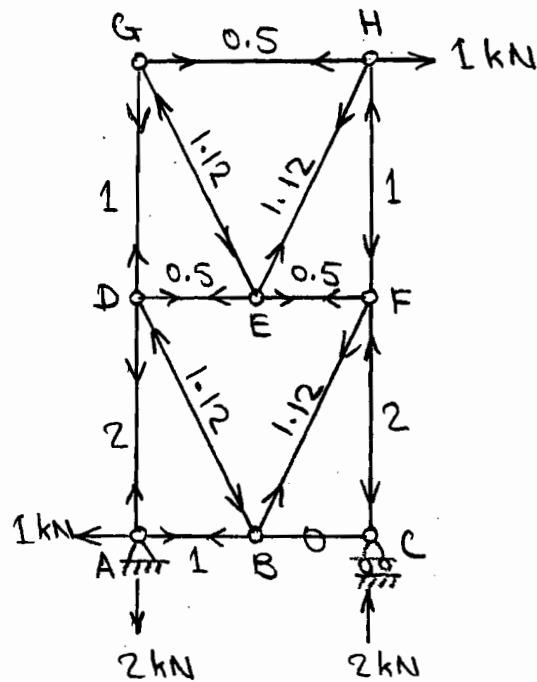


Member	L (ft)	F (k)	Fv (k)	Fv(FL) (k ² -ft)
AD	14.14	-1.41	0.707	-14.1
BD	14.14	-7.07	-0.707	70.68
DE	5	-3	1	-15
Σ			41.58	

$$\Delta_{EH} = \frac{41.58}{(29000)6} = 0.000239 \text{ ft.}$$

$$= 0.0029 \text{ in. } \rightarrow$$

7.9

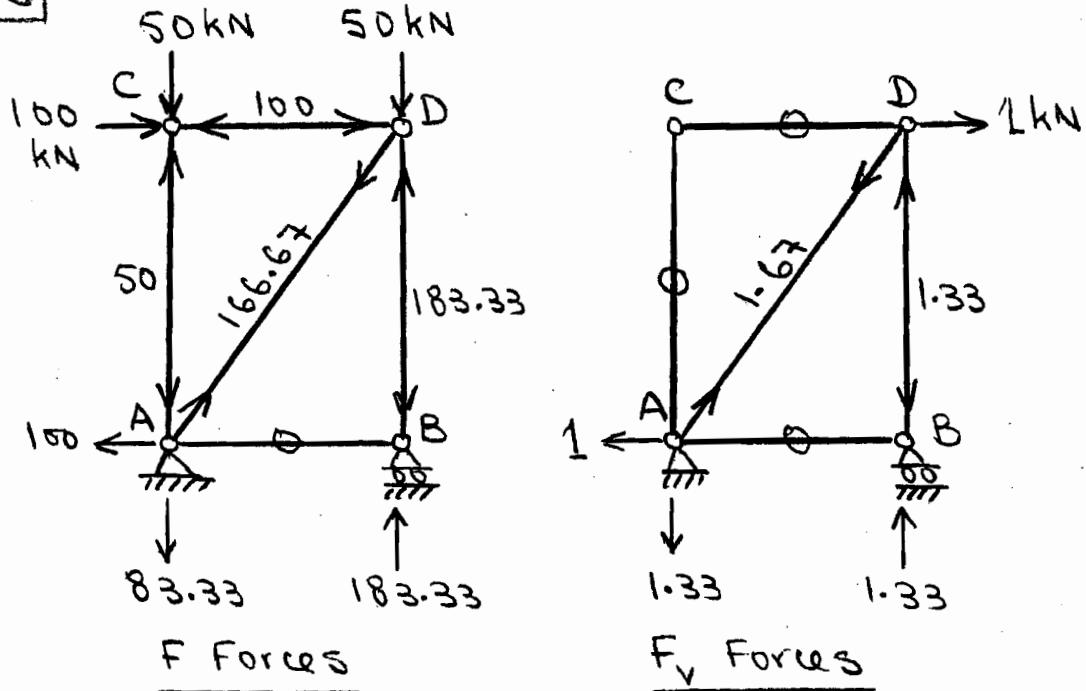
F ForcesF_v Forces

Member	L (m)	A (m ²)	F (kN)	F _v (kN)	F _v (FL/A) (kN ² /m)
AD	4	0.0025	300	2	960 000
CF	4	0.0025	-300	-2	960 000
DG	4	0.0025	75	1	120 000
FH	4	0.0025	-75	-1	120 000
GH	4	0.0015	-37.5	0.5	-50 000
AB	2	0.0015	225	1	300 000
DE	2	0.0015	-37.5	0.5	-25 000
EF	2	0.0015	-112.5	0.5	-75 000
BD	4.47	0.0015	-251.56	-1.12	839607
BF	4.47	0.0015	251.56	1.12	839607
EG	4.47	0.0015	-83.85	-1.12	279858
EH	4.47	0.0015	83.85	1.12	279858
Σ					4548930

$$(1 \text{ kN}) \Delta_H = \frac{4548930}{200(10^6)} = 0.023 \text{ kN}\cdot\text{m}$$

$$\Delta_H = 0.023 \text{ m} = \underline{23 \text{ mm}} \rightarrow$$

7.10



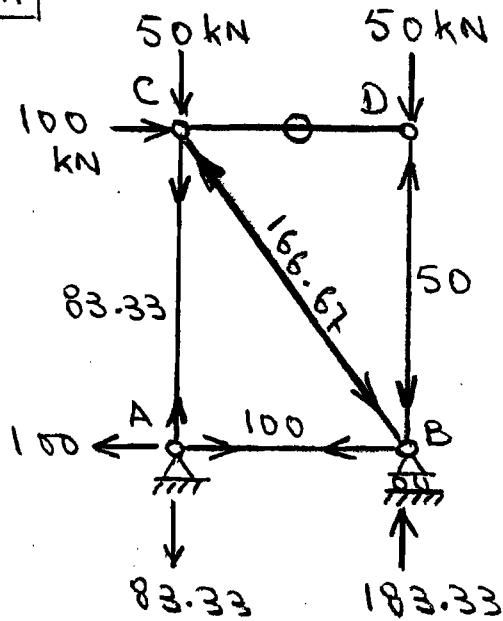
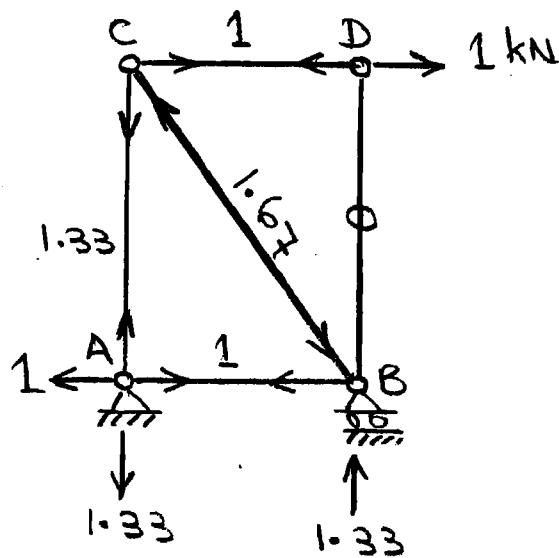
Member	L (m)	F (kN)	F _v (kN)	F _v (FL) (kN ² .m)
AB	3	0	0	0
CD	3	-100	0	0
AC	4	-50	0	0
BD	4	-183.33	-1.33	975.32
AD	5	166.67	1.67	1391.69
		Σ	2367	

$$(1 \text{ kN}) \Delta_D = \frac{2367 \text{ kN}^2 \cdot \text{m}}{EA}$$

$$\Delta_D = \frac{2367 \text{ kN} \cdot \text{m}}{EA} = \frac{2367}{70(10^6)A} = 0.01 \text{ m}$$

from which, $A = 0.003381 \text{ m}^2 = \underline{\underline{3381 \text{ mm}^2}}$

7.11

F Forces F_v Forces

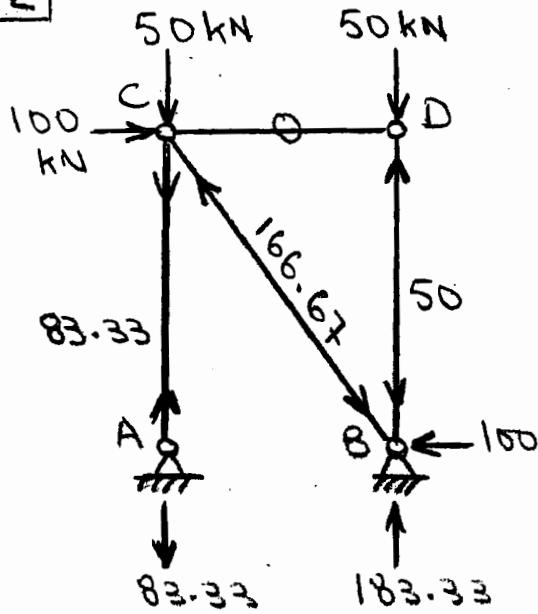
Member	L (m)	F (kN)	F_v (kN)	$F_v(FL)$ (kN \cdot m)
AB	3	100	1	300
CD	3	0	1	0
AC	4	83.33	1.33	443.32
BD	4	-50	0	0
BC	5	-166.67	-1.67	1391.69
		Σ	2135	

$$(1 \text{ kN}) \Delta_D = \frac{2135 \text{ kN}^2 \cdot \text{m}}{EA}$$

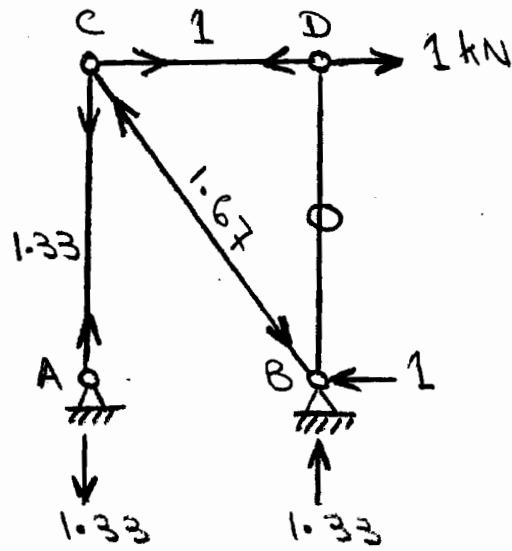
$$\Delta_D = \frac{2135 \text{ kN} \cdot \text{m}}{EA} = \frac{2135}{70(10^6)A} = 0.01 \text{ m}$$

from which, $A = 0.00305 \text{ m}^2 = \underline{\underline{3050 \text{ mm}^2}}$

7.12



F Forces



F_v Forces

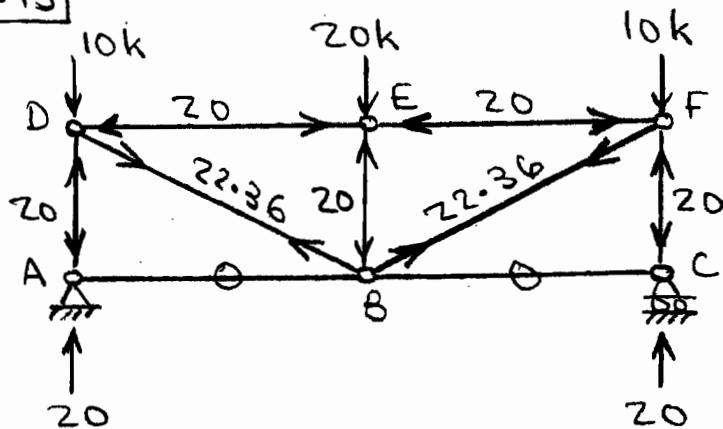
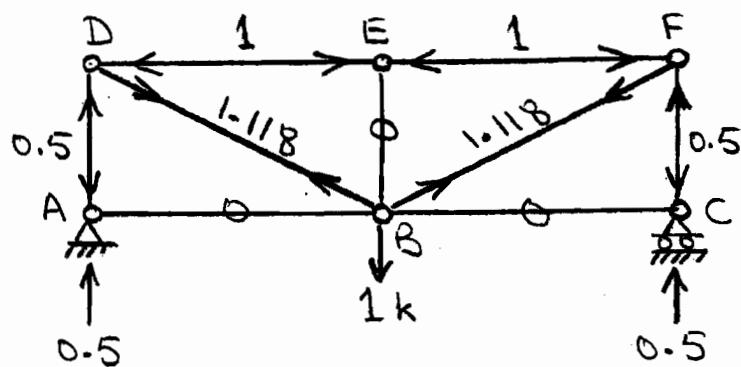
Member	L (m)	F (kN)	F _v (kN)	F _v (FL) (kN ² .m)
CD	3	0	1	0
AC	4	83.33	1.33	443.32
BD	4	-50	0	0
BC	5	-166.67	-1.67	1391.69
			Σ	1835

$$(1 \text{ kN}) \Delta_D = \frac{1835 \text{ kN}^2 \cdot \text{m}}{\text{EA}}$$

$$\Delta_D = \frac{1835 \text{ kN} \cdot \text{m}}{\text{EA}} = \frac{1835}{70(10^6)A} = 0.01 \text{ m}$$

$$\text{From which, } A = 0.002621 \text{ m}^2 = \underline{2621 \text{ mm}^2}$$

7.13

F Forces F_v Forces

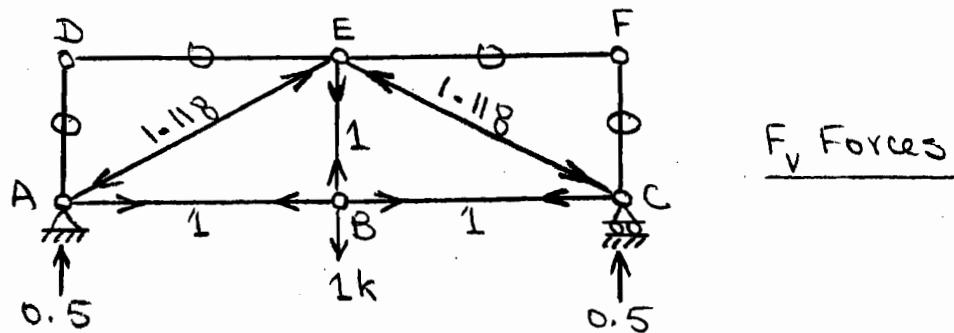
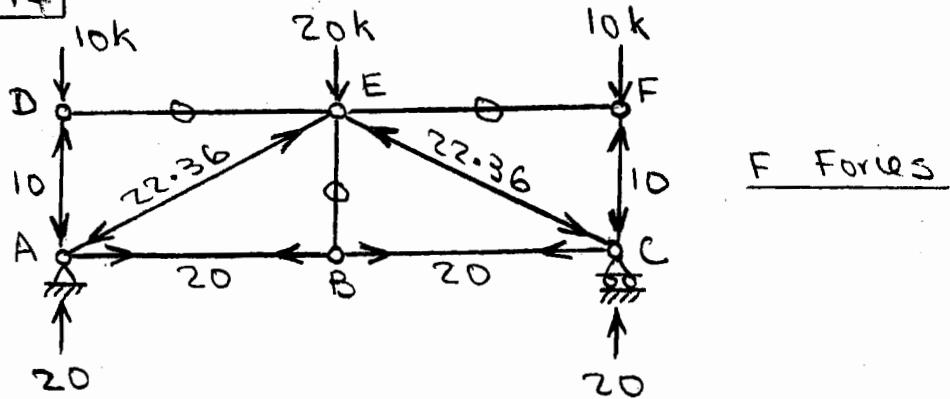
Member	L (in)	F (k)	F_v (k)	$F_v(FL)$ (k^2 -in)
AB	72	0	0	0
BC	72	0	0	0
DE	72	-20	-1	1440
EF	72	-20	-1	1440
AD	36	-20	-0.5	360
BE	36	-20	0	0
CF	36	-20	-0.5	360
BD	80.5	22.36	1.118	2012.4
BF	80.5	22.36	1.118	2012.4
Σ				7624.8

$$(1k)\Delta_B = \frac{7624.8 \text{ k}^2\text{-in}}{\text{EA}}$$

$$\Delta_B = \frac{7624.8 \text{ k-in}}{\text{EA}} = \frac{7624.8}{1600 (\text{A})} = 0.4 \text{ in}$$

from which, $A = 11.91 \text{ in}^2$

7-14



Member	L (in.)	F (k)	F_v (k)	$F_v(FL)$ (k^2 -in)
AB	72	20	1	1440
BC	72	20	1	1440
DE	72	0	0	0
EF	72	0	0	0
AD	36	-10	0	0
BE	36	0	1	0
CF	36	-10	0	0
AE	80.5	-22.36	-1.118	2012.4
CE	80.5	-22.36	-1.118	2012.4
				$\sum 6904.8$

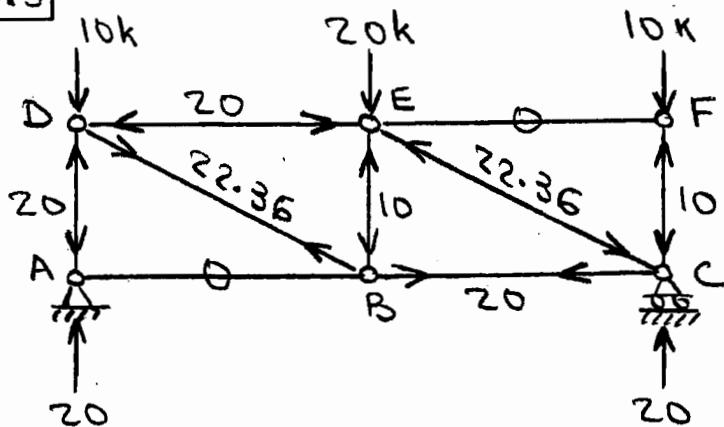
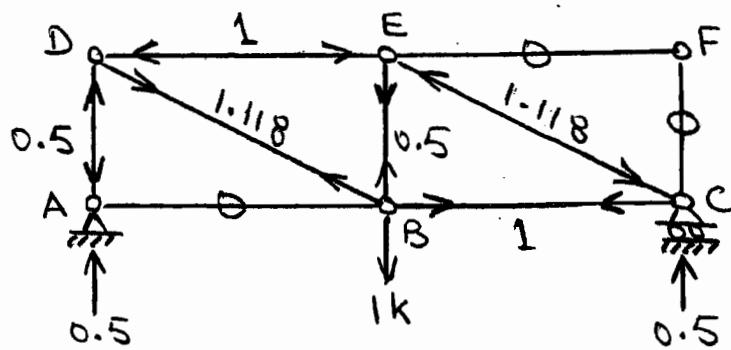
$$(1k) \Delta_B = \frac{6904.8 \text{ k}^2\text{-in}}{EA}$$

$$\Delta_B = \frac{6904.8 \text{ k-in}}{EA} = \frac{6904.8}{1600(\text{A})} = 0.4 \text{ in}$$

From which,

$$A = 10.79 \text{ in}^2$$

7.15

F Forces F_v Forces

Member	L (in.)	F (k)	F_v (k)	$F_v(FL)$ ($k^2 \cdot \text{in.}$)
AB	72	0	0	0
BC	72	20	1	1440
DE	72	-20	-1	1440
EF	72	0	0	0
AD	36	-20	-0.5	360
BE	36	-10	0.5	-180
CF	36	-10	0	0
BD	80.5	22.36	1.118	2012.4
CE	80.5	-22.36	-1.118	2012.4
\sum				7084.8

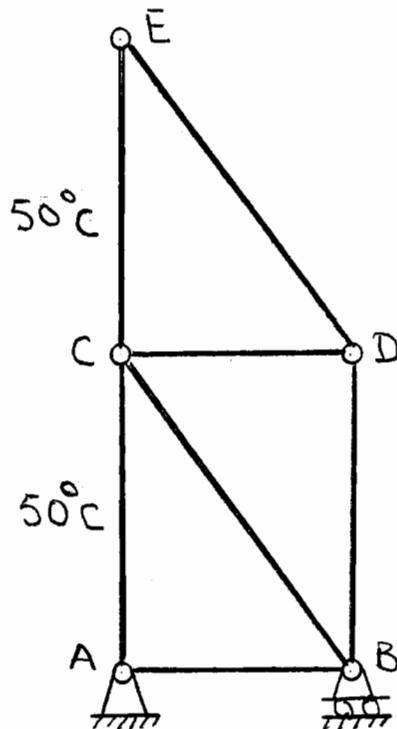
$$(1k) \Delta_B = \frac{7084.8 \text{ k}^2 \cdot \text{in.}}{EA}$$

$$\Delta_B = \frac{7084.8 \text{ k} \cdot \text{in.}}{EA} = \frac{7084.8}{1600 (\text{A})} = 0.4 \text{ in}$$

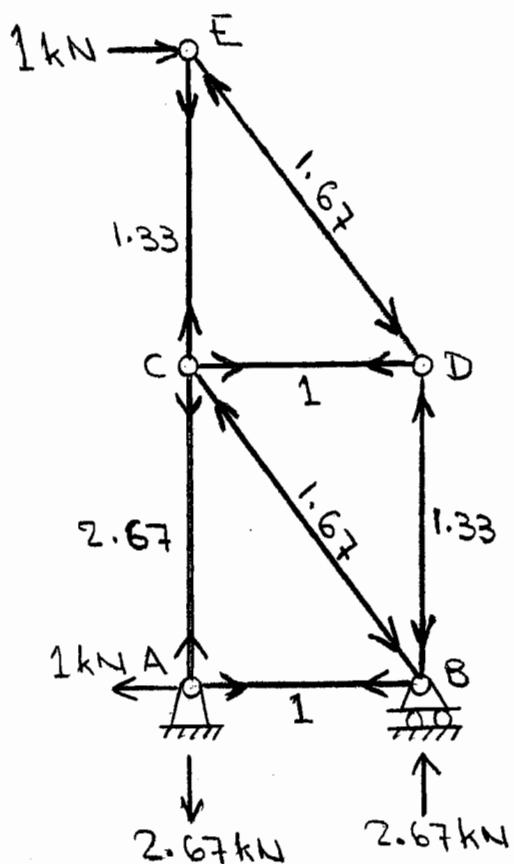
From which,

$$A = 11.07 \text{ in}^2$$

7.16



Real System - ΔT



Virtual System -

F_v Forces

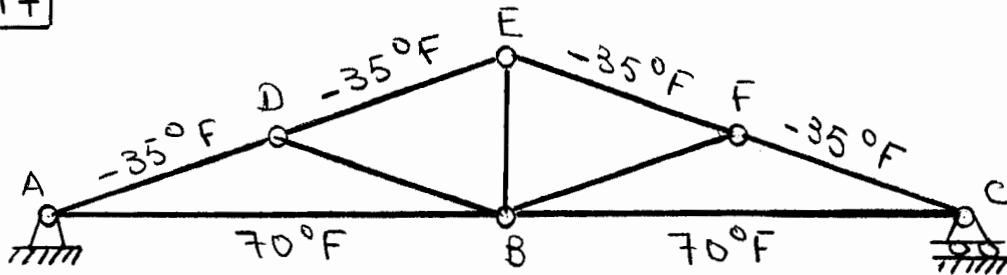
Member	F_v (kN)
AC	2.67
CE	1.33
Σ	4

$$(1 \text{ kN}) \Delta_E = \alpha (\Delta T) L \sum F_v = 1.2 \times 10^{-5} (50) (4) (4)$$

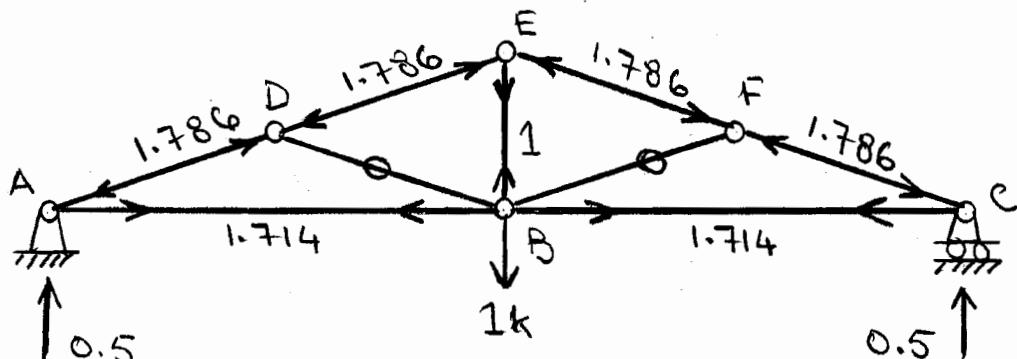
$$= 0.0096 \text{ kN.m}$$

$$\Delta_E = 0.0096 \text{ m} = \underline{9.6 \text{ mm}} \rightarrow$$

7.17



Real System - ΔT



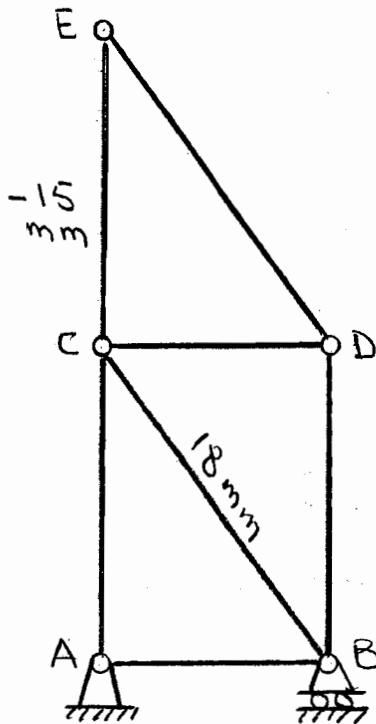
Virtual System - F_v Forces

Member	L (in.)	ΔT (°F)	F_v (k)	$F_v(\Delta T)L$ (k-in-°F)
AB	288	70	1.714	34560
BC	288	70	1.714	34560
AD	150	-35	-1.786	9375
DE	150	-35	-1.786	9375
EF	150	-35	-1.786	9375
CF	150	-35	-1.786	9375
Σ				106620

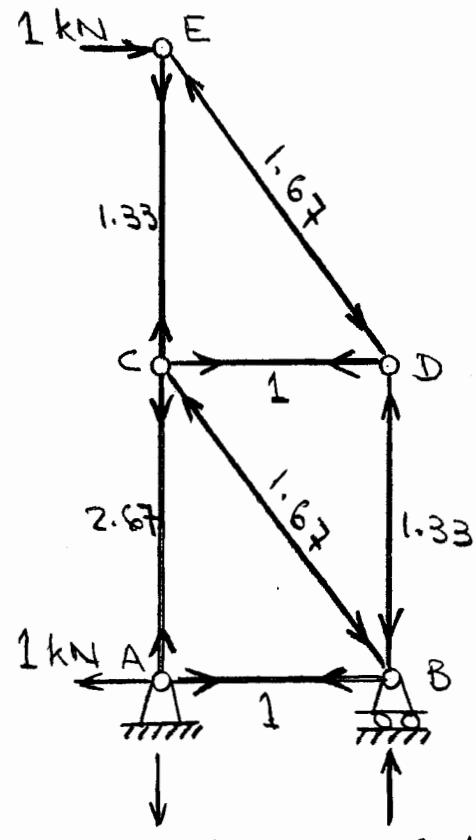
$$(1k) \Delta_B = \alpha \sum F_v(\Delta T)L = 6.5(10^{-6})(106620)$$

$$\Delta_B = 0.693 \text{ in.} \downarrow$$

7.18



Real System - S



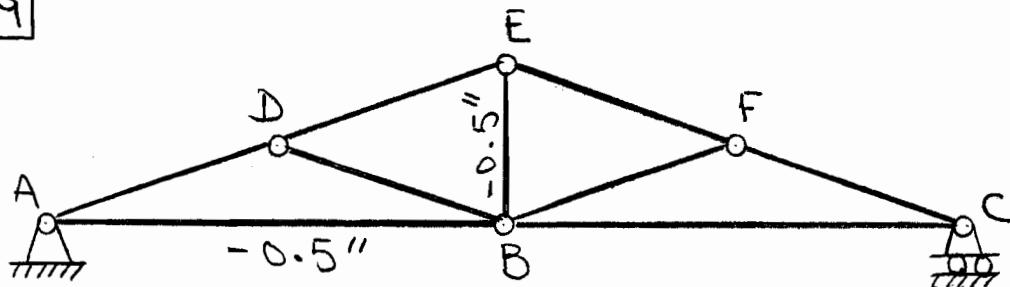
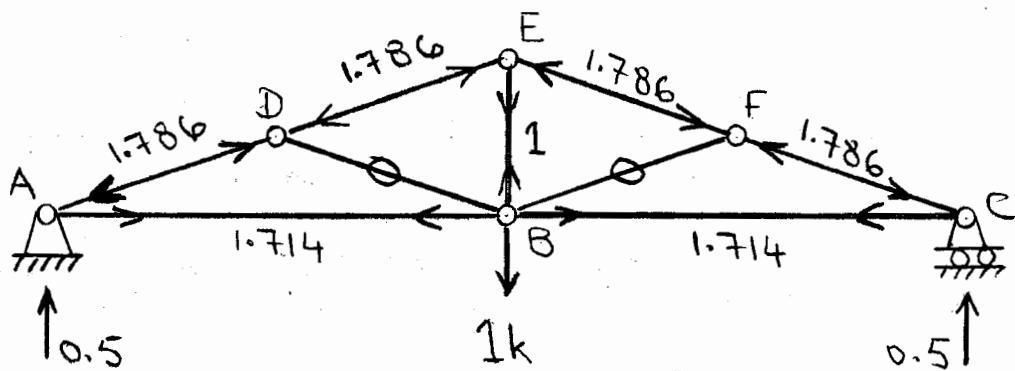
Virtual System - Fv Forces

Member	δ (mm)	F_v (kN)	$F_v(\delta)$ (kN·mm)
BC	18	-1.67	-30
CE	-15	1.33	-20
Σ			-50

$$(1 \text{ kN}) \Delta_E = \sum F_v(\delta) = -50 \text{ kN} \cdot \text{mm}$$

$$\Delta_E = -50 \text{ mm} = \underline{\underline{50 \text{ mm}}} \leftarrow$$

7.19

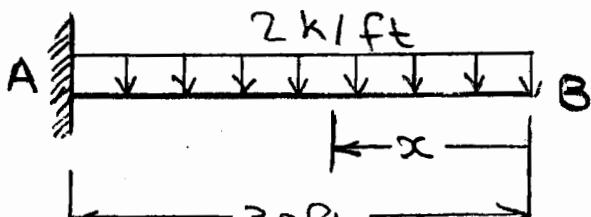
Real System - δ Virtual System - F_v Forces

Member	F_v (k)
AB	1.714
BE	1
Σ	2.714

$$(1k) \Delta_B = \delta \sum F_v = (-0.5) 2.714 = -1.357 \text{ k-in.}$$

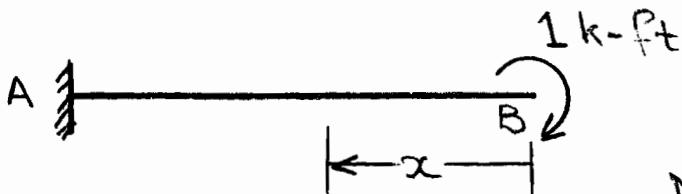
$$\Delta_B = -1.357 \text{ in.} = \underline{\underline{1.357 \text{ in.}}}$$

7.20



$$M = -\frac{2x^2}{2} = -x^2$$

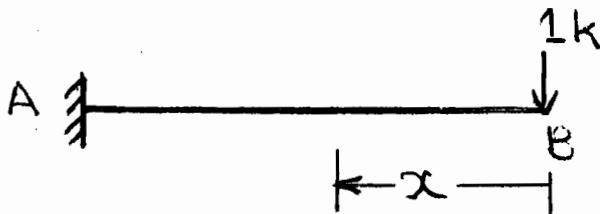
Real System - M



$$M_{V1} = -1$$

Virtual System - M_{V1}

$$\begin{aligned}\Theta_B &= \frac{1}{EI} \int_0^{30} -1(-x^2) dx = \frac{9000 \text{ k-ft}^2}{EI} \\ &= \frac{9000 (12)^2}{29000 (3000)} = \underline{0.0149 \text{ rad}} \quad \nabla\end{aligned}$$

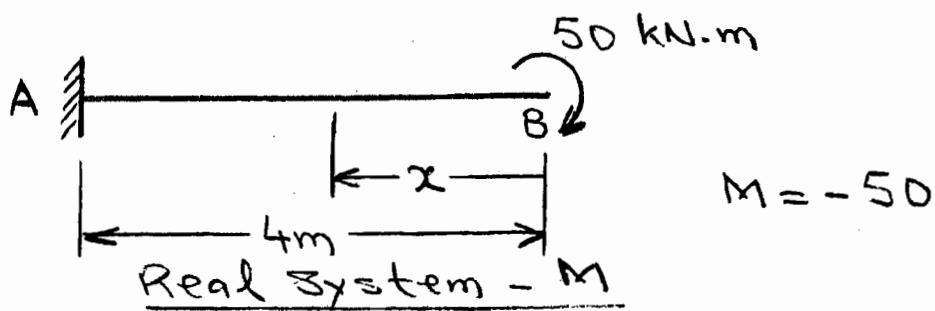


$$M_{V2} = -1(x) = -x$$

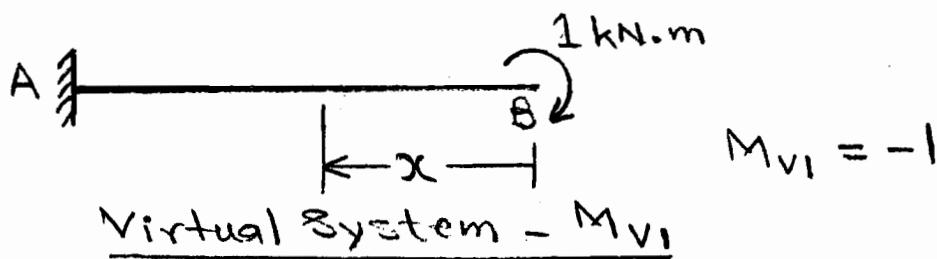
Virtual System M_{V2}

$$\begin{aligned}\Delta_B &= \frac{1}{EI} \int_0^{30} (-x)(-x^2) dx = \frac{202500 \text{ k-ft}^3}{EI} \\ &= \frac{202500 (12)^3}{29000 (3000)} = \underline{4.022 \text{ in.}} \downarrow\end{aligned}$$

7.21

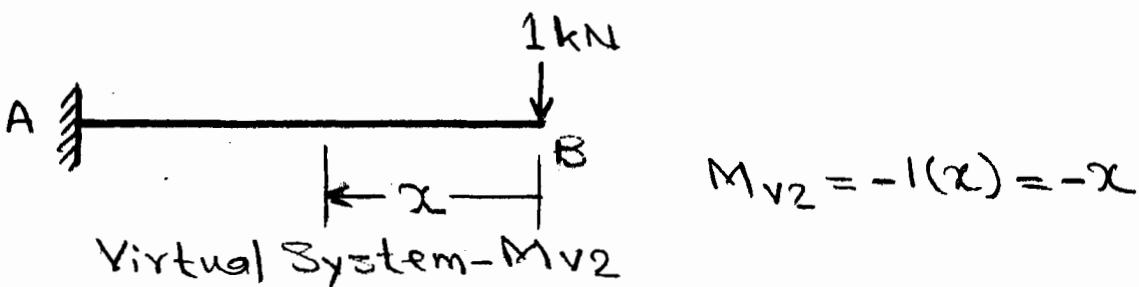


$$M = -50$$



$$M_{V1} = -1$$

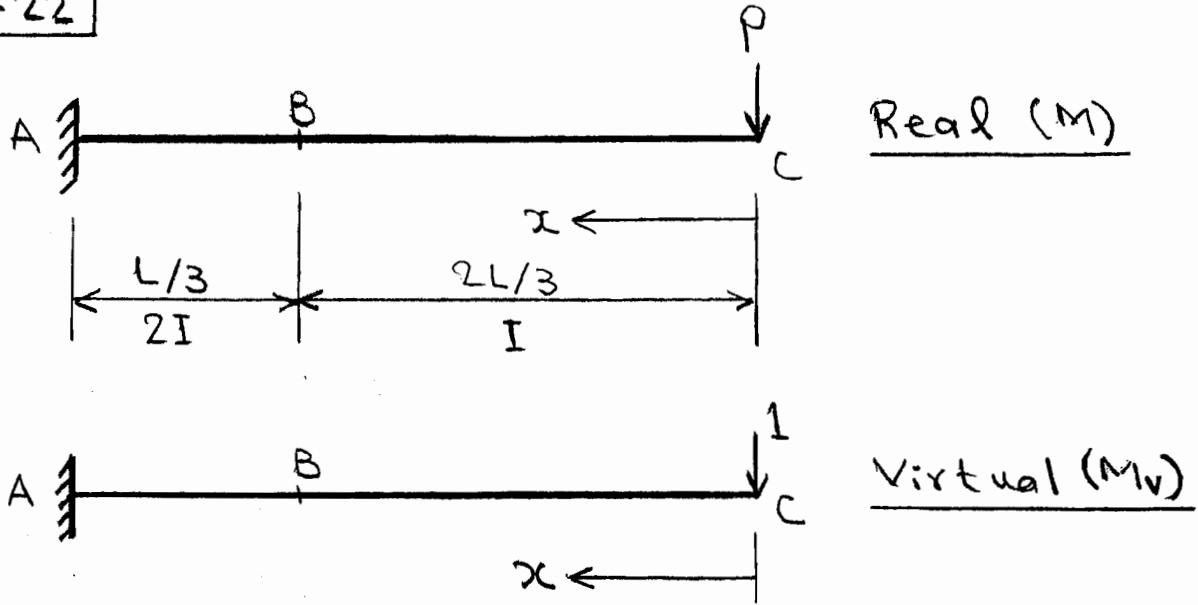
$$\begin{aligned}\Theta_B &= \frac{1}{EI} \int_0^4 (-1)(-50) dx = \frac{200 \text{ kN}\cdot\text{m}^2}{EI} \\ &= \frac{200}{70(164)} = 0.0174 \text{ rad} \quad \square\end{aligned}$$



$$M_{V2} = -1(x) = -x$$

$$\begin{aligned}\Delta_B &= \frac{1}{EI} \int_0^4 (-x)(-50) dx = \frac{400 \text{ kN}\cdot\text{m}^3}{EI} \\ &= \frac{400}{70(164)} = 0.0348 \text{ m} = 34.8 \text{ mm} \downarrow\end{aligned}$$

7.22

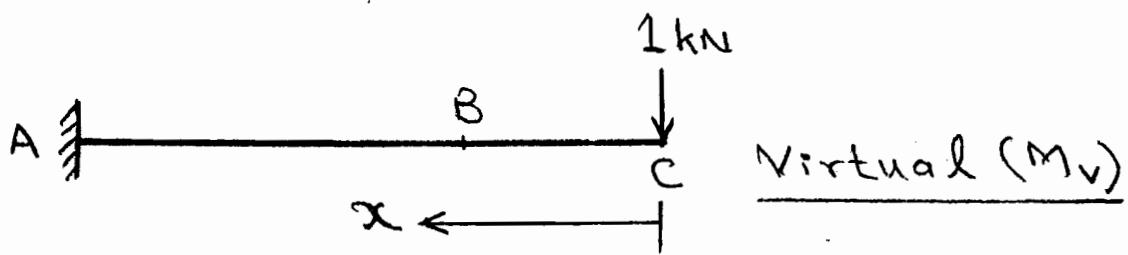
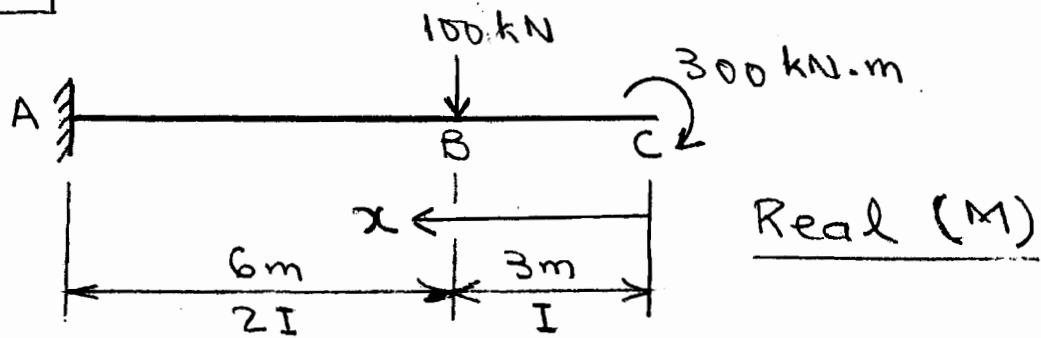


Segment	x Coordinate		M	M_V
	Origin	Limits		
CB	C	$0 - \frac{2L}{3}$	$-Px$	$-Ix$
BA	C	$\frac{2L}{3} - L$	$-Px$	$-Ix$

$$\Delta_C = \frac{1}{EI} \left[\int_0^{\frac{2L}{3}} (-x)(-Px) dx + \frac{1}{2} \int_{\frac{2L}{3}}^L (-x)(-Px) dx \right]$$

$$= \frac{35PL^3}{162EI} \downarrow$$

7.23



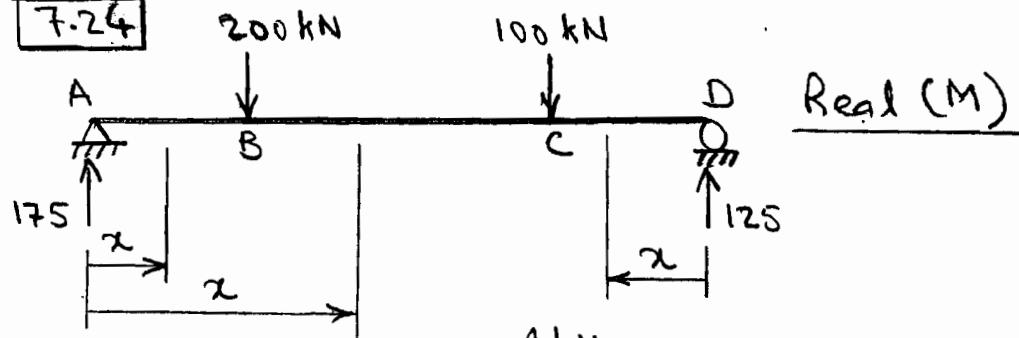
Segment	x Coordinate		M (kN·m)	M_V (kN·m)
	Origin	Limits (m)		
CB	C	0-3	-300	-x
BA	C	3-9	$\frac{-300}{-150(x-3)}$	-x

$$\Delta_C = \frac{1}{EI} \left[\int_0^3 (-x)(-300) dx + \frac{1}{2} \int_3^9 (-x)(-100x) dx \right]$$

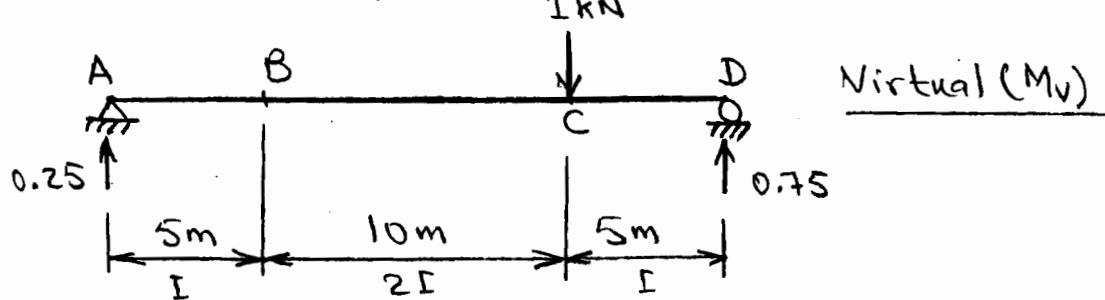
$$= \frac{13050 \text{ kN.m}^3}{EI} = \frac{13050}{70(500)} = 0.373 \text{ m}$$

$$\underline{\Delta_C = 373 \text{ mm} \downarrow}$$

7.24



Real (M)

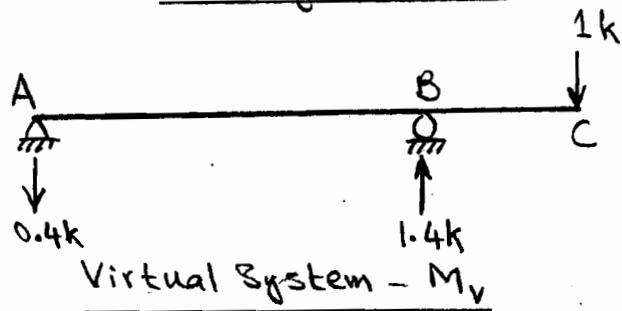
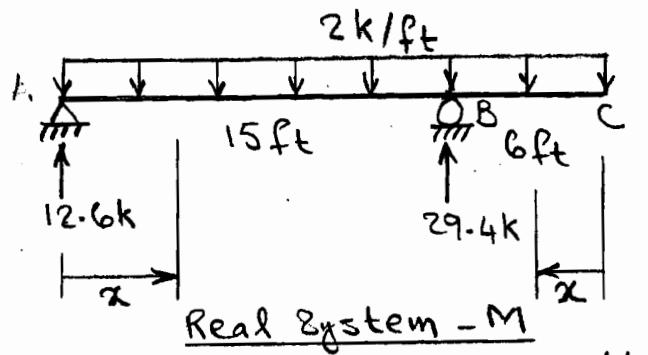


Virtual (Mv)

Segment	x coordinate		M (kN-m)	M _v (kN-m)
	Origin	Limits (m)		
AB	A	0 - 5	175x	0.25x
BC	A	5 - 15	175x - 200(x-5)	0.25x
DC	D	0 - 5	125x	0.75x

$$\begin{aligned}
 \Delta_C &= \frac{1}{EI} \left[\int_0^5 (0.25x)(175x) dx \right. \\
 &\quad \left. + \frac{1}{2} \int_5^{15} 0.25x(-25x + 1000) dx + \int_0^5 0.75x(125x) dx \right] \\
 &= \frac{14843.75 \text{ kN-m}^3}{EI} = \frac{14843.75}{250(600)} \\
 &= 0.099 \text{ m} = \underline{99 \text{ mm}}
 \end{aligned}$$

7.25



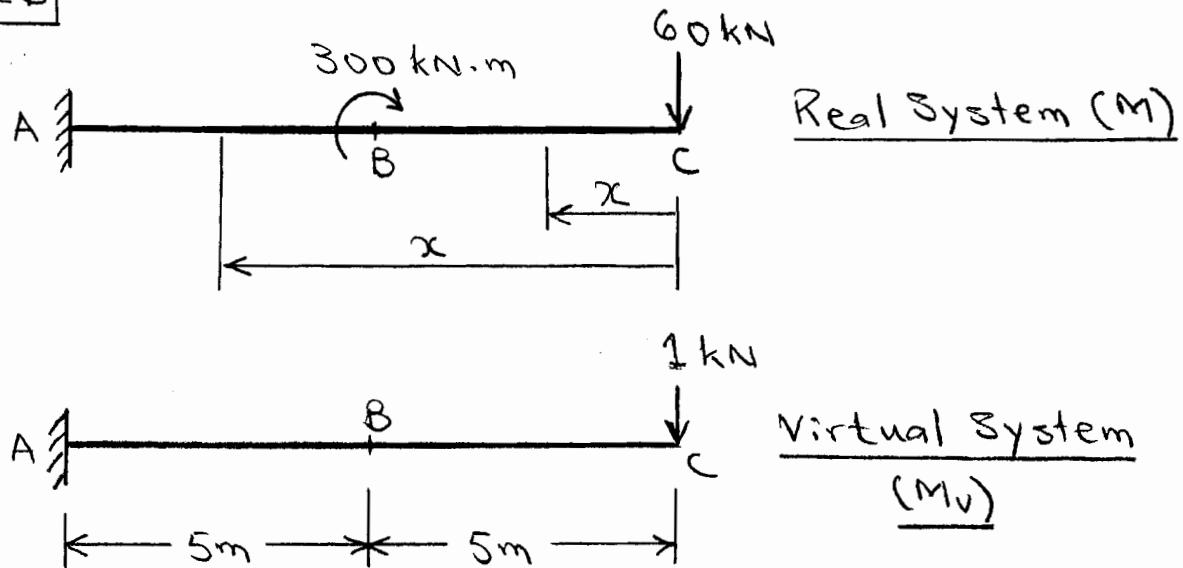
Segment	x Coordinate		M (k-ft)	M_v (k-ft)
	Origin	Limits (ft)		
AB	A	0-15	$12.6x - x^2$	$-0.4x$
CB	C	0-6	$-x^2$	$-x$

$$(1k)\Delta_C = \frac{1}{EI} \left[\int_0^{15} -0.4x(12.6x - x^2) dx + \int_0^6 -x(-x^2) dx \right]$$

$$= -\frac{283.5 k^2 \cdot ft^3}{EI}$$

$$\Delta_C = -\frac{283.5 (12)}{(29000)(3500)} = -0.0048 \text{ in.} = \underline{\underline{0.0048 \text{ in.} \uparrow}}$$

7.26



Segment	x coordinate		M (kN·m)	M_v (kN·m)
	Origin	Limits (m)		
CB	C	0-5	-60x	-1x
BA	C	5-10	-60x - 300	-1x

$$\Delta_{\max} = \Delta_C = \frac{1}{EI} \left[\int_0^5 (-1x)(-60x) dx + \int_5^{10} (-1x)(-60x - 300) dx \right]$$

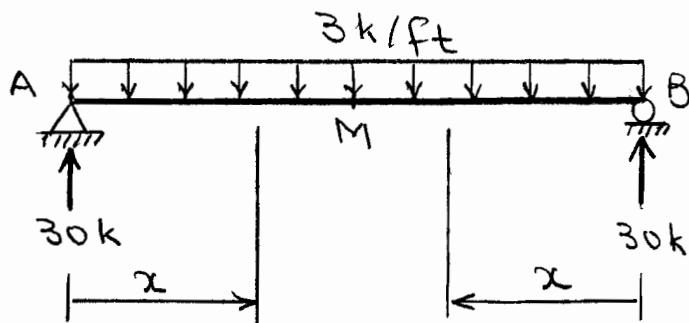
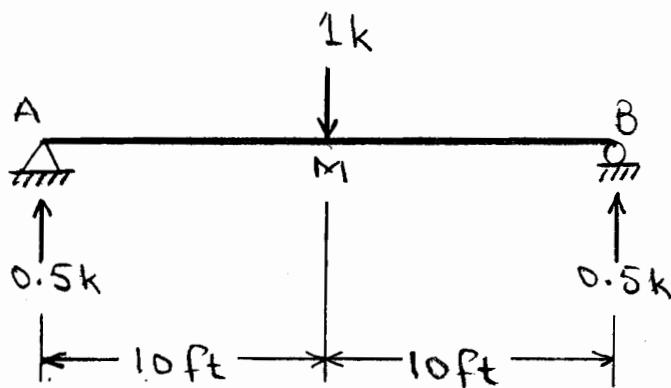
$$= \frac{31250 \text{ kN}\cdot\text{m}^3}{EI}$$

$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{31250}{200(10^6)I} = \frac{10}{360}$$

$$\text{from which, } I = 5625(10^6) \text{ m}^4 = \underline{\underline{5625(10^6) \text{ mm}^4}}$$

7.27

Real System (M)Virtual System
(M_v)

Segment	x Coordinate		M (k-ft)	M_v (k-ft)
	Origin	Limits (ft)		
AM	A	0 - 10	$30x - \frac{3x^2}{2}$	$0.5x$
BM	B	0 - 10	$30x - 3\frac{x^2}{2}$	$0.5x$

$$\Delta_{max} = \Delta_M = \frac{1}{EI} \left[2 \int_0^{10} (0.5x)(30x - \frac{3x^2}{2}) dx \right]$$

$$= \frac{6250 \text{ k-ft}^3}{EI}$$

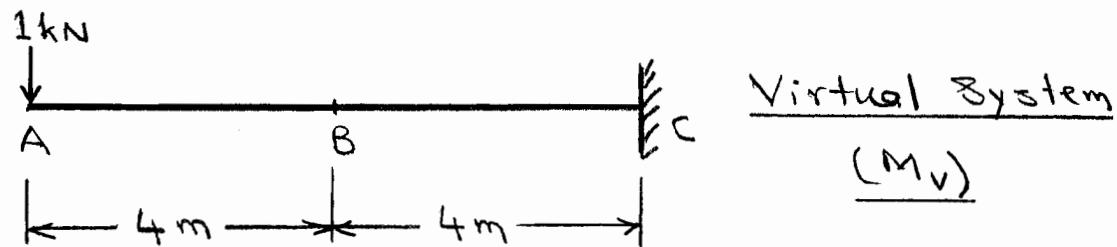
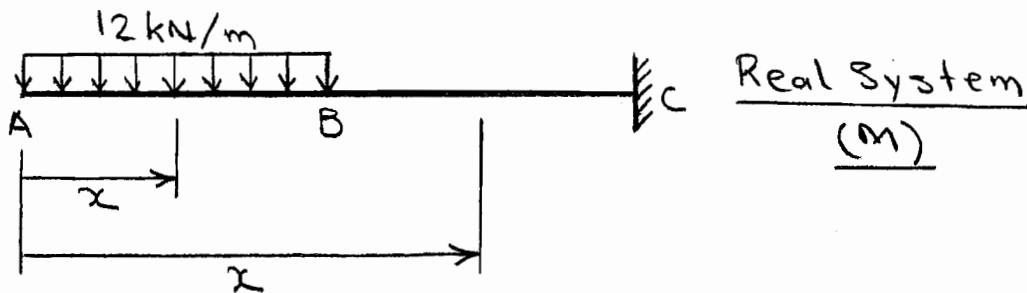
$$\Delta_{max} = \frac{L}{360}$$

$$\frac{6250(12)^3}{29000(I)} = \frac{20(12)}{360}$$

from which,

$$I = 559 \text{ in}^4$$

7.28



Segment	x Coordinate		M (kN.m)	M_V (kN.m)
	Origin	Limits (m)		
AB	A	0 - 4	$-12x^2/2$	$-1x$
BC	A	4 - 8	$-12(4)(x-2)$	$-1x$

$$\Delta_{\max} = \Delta_A = \frac{1}{EI} \left[\int_0^4 (-1x)(-6x^2) dx + \int_4^8 (-1x)(-48)(x-2) dx \right]$$

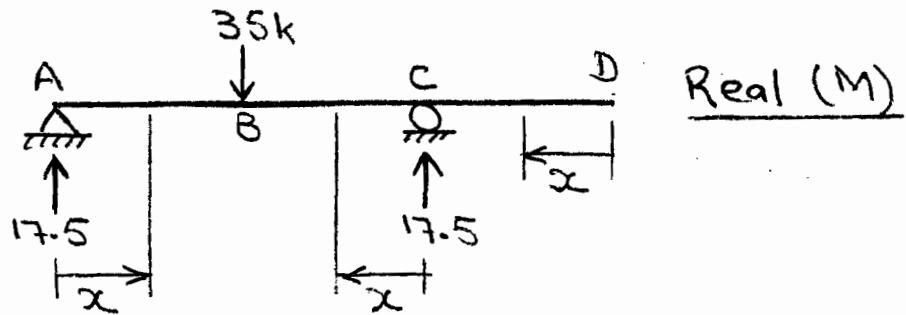
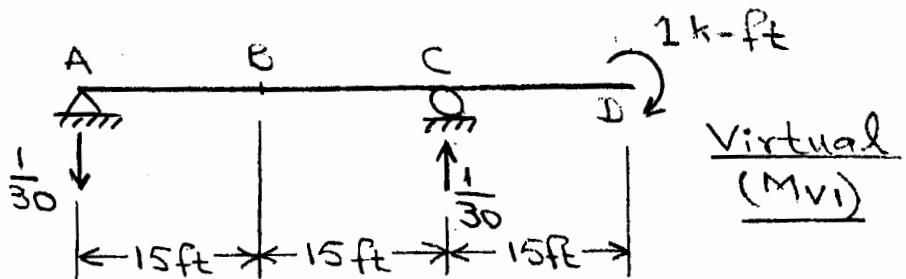
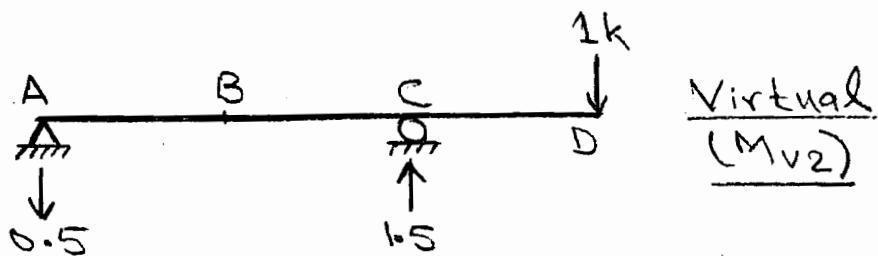
$$= \frac{5248 \text{ kN.m}^3}{EI}$$

$$\Delta_{\max} = \frac{L}{360}$$

$$\frac{5248}{70(10^6)I} = \frac{8}{360}$$

$$\text{from which, } I = 3374 (10^6) \text{ m}^4 = \underline{\underline{3374 (10^6) \text{ mm}^4}}$$

7.29

Real (M)Virtual (Mv1)Virtual (Mv2)

Segment	x coordinate		M (k-ft)	Mv1 (k-ft)	Mv2 (k-ft)
	Origin	Limits (ft)			
AB	A	0-15	17.5x	-x/30	-0.5x
CB	C	0-15	17.5x	-1+(x/30)	-1(x+15)+1.5x
DC	D	0-15	0	-1	-1x

$$\Theta_D = \frac{1}{EI} \left[\int_0^{15} -\frac{x}{30} (17.5x) dx + \int_0^{15} \left(-1 + \frac{x}{30} \right) (17.5x) dx \right]$$

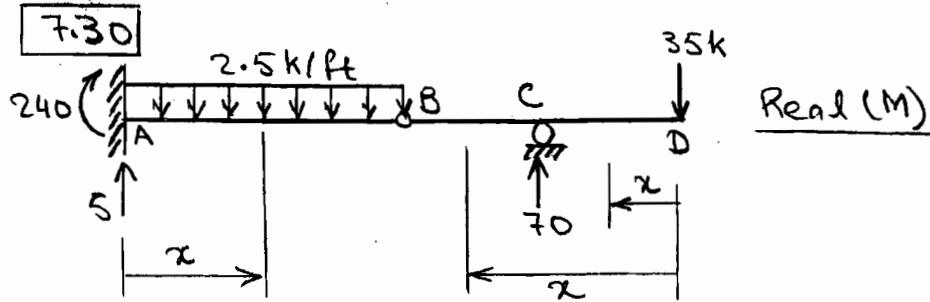
$$= - \frac{1968.75 \text{ k-ft}^2}{EI} = - \frac{1968.75 (12)^2}{10000 (2500)} = -0.01134 \text{ rad}$$

$$\underline{\Theta_D = 0.01134 \text{ rad}}$$

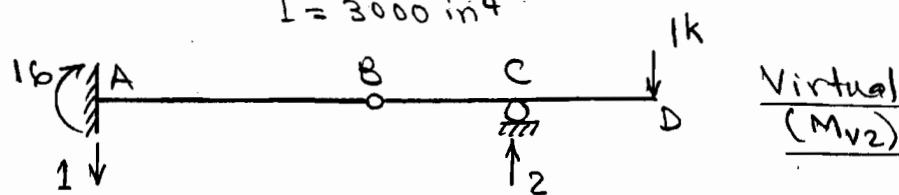
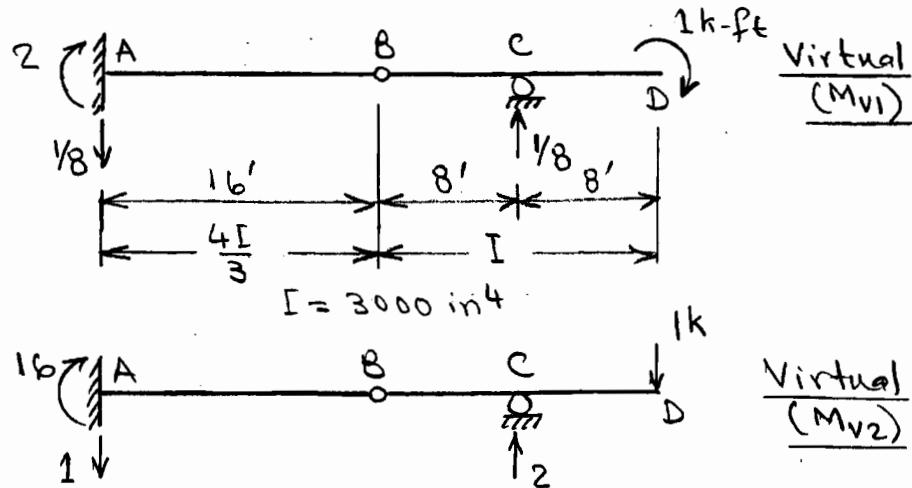
$$\Delta_D = \frac{1}{EI} \left[\int_0^{15} (-0.5x)(17.5x) dx + \int_0^{15} (0.5x-15)(17.5x) dx \right]$$

$$= - \frac{29531.25 \text{ k-ft}^3}{EI} = - \frac{29531.25 (12)^3}{10000 (2500)}$$

$$= -2.04 \text{ in.} = \underline{2.04 \text{ in. up}}$$



Real (M)



Segment	x coordinate		M (k-ft)	Mv1 (k-ft)	Mv2 (k-ft)
	Origin	Limits (ft)			
DC	D	0 - 8	-35x	-1	-12
CB	D	8 - 16	-35x + 70(2-x)	$-1 + \frac{1}{8}(x-8)$	$x - 16$
AB	A	0 - 16	$240 + 5x - 1.25x^2$	$2 - \frac{x}{8}$	$16 - x$

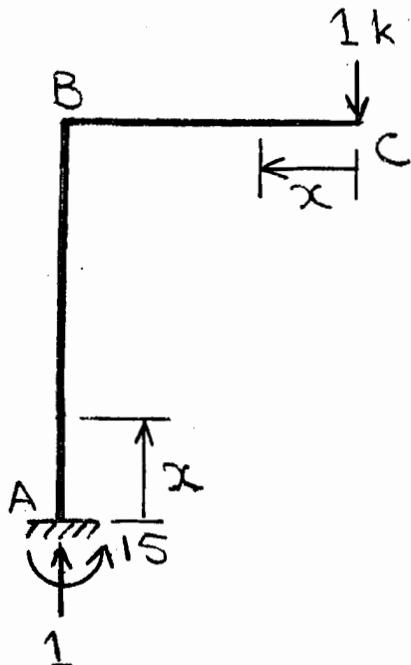
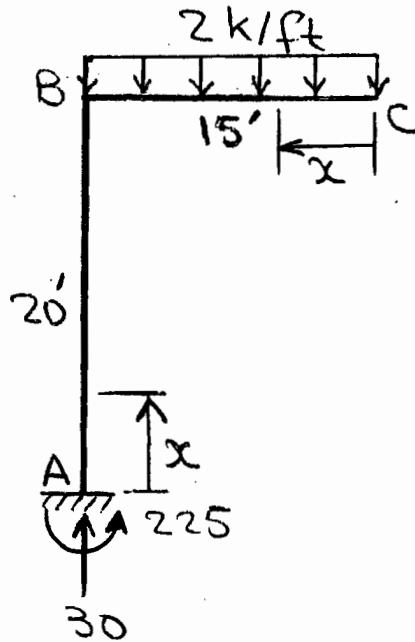
$$\Theta_D = \frac{1}{EI} \left[\int_0^8 -1(-35x) dx + \int_8^{16} \left(\frac{x}{8} - 2 \right) (35x - 560) dx + \frac{3}{4} \int_0^{16} \left(2 - \frac{x}{8} \right) (240 + 5x - 1.25x^2) dx \right]$$

$$= \frac{4426.67 \text{ k-ft}^2}{EI} = \frac{4426.67 (12)^2}{30000 (3000)} = 0.0071 \text{ rad. } \checkmark$$

$$\Delta_D = \frac{1}{EI} \left[\int_0^8 -x(-35x) dx + \int_8^{16} (x-16)(35x - 560) dx + \frac{3}{4} \int_0^{16} (16-x)(240 + 5x - 1.25x^2) dx \right]$$

$$= \frac{32426.67 \text{ k-ft}^3}{EI} = \frac{32426.67 (12)^3}{30000 (3000)} = 0.62 \text{ in. } \downarrow$$

7.31

Real System - M Virtual System - M_v

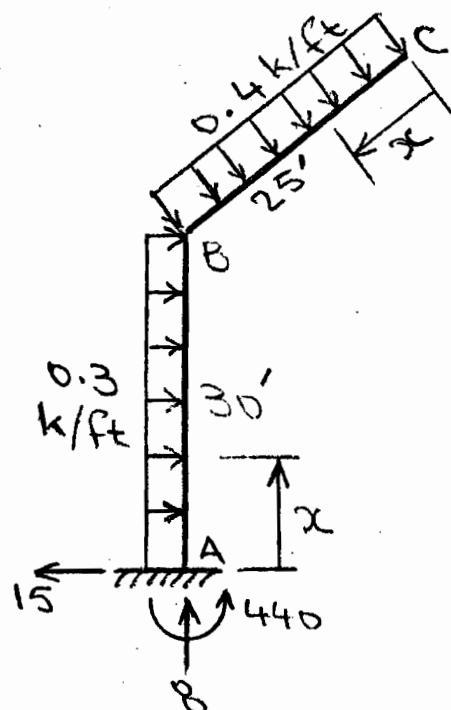
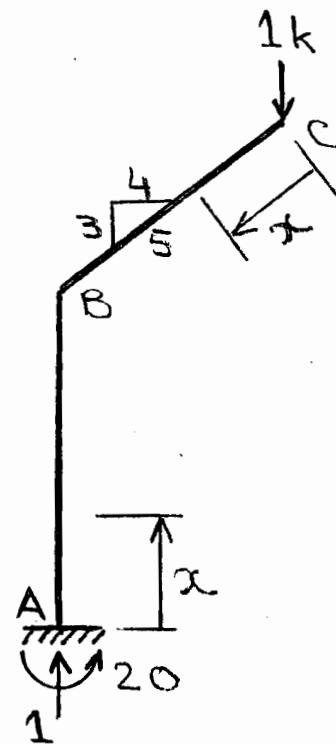
Segment	x Coordinate		M (k-ft)	M_v (k-ft)
	Origin	Limits (ft.)		
AB	A	0-20	-225	-15
CB	C	0-15	$-x^2$	$-x$

$$(1k) \Delta_c = \frac{1}{EI} \left[\int_0^{20} (-15)(-225) dx + \int_0^{15} (-x)(-x^2) dx \right]$$

$$= \frac{80156.25 k^2 \cdot ft^3}{EI}$$

$$\Delta_c = \frac{80156.25 k \cdot ft^3}{EI} = \frac{80156.25 (12)^3}{29000 (2000)} \\ = \underline{\underline{2.388 \text{ in.} \downarrow}}$$

7.32

Real System - M Virtual System - M_v

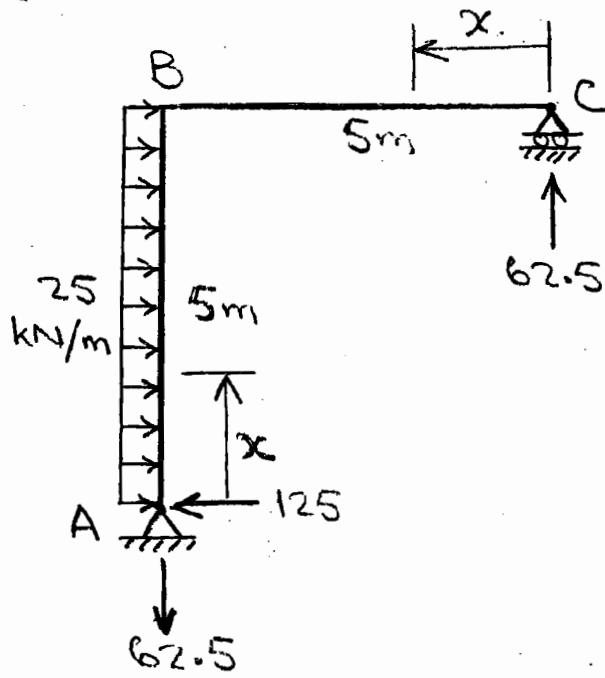
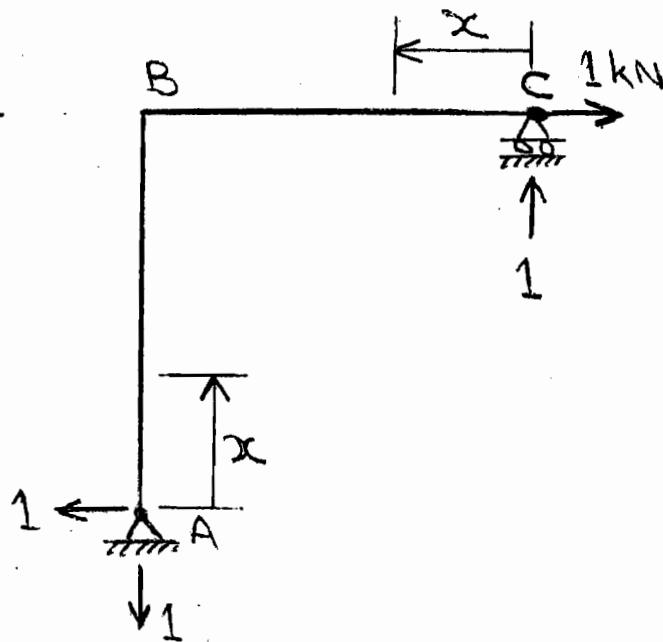
Segment	x Coordinate		M (k-ft)	M_v (k-ft)
	Origin	Limits (ft)		
AB	A	0-30	$-440 + 15x - 0.15x^2$	-20
CB	C	0-25	$-0.2x^2$	$-\frac{4}{5}x$

$$(1k) \Delta_C = \frac{1}{EI} \left[\int_0^{30} (-20)(-440 + 15x - 0.15x^2) dx + \int_0^{25} \left(-\frac{4}{5}x\right)(-0.2x^2) dx \right]$$

$$= \frac{171625 k^2 \cdot ft^3}{EI}$$

$$\Delta_C = \frac{171625 k \cdot ft^3}{EI} = \frac{171625 (12)^3}{10000(8160)} = 3.63 \text{ in.}$$

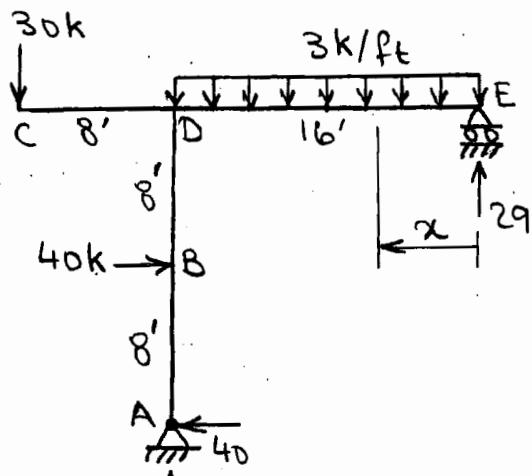
7.33

Real System - M Virtual System - M_V

Segment	x Coordinate		M (kN.m)	M_V (kN.m)
	Origin	Limits (m)		
AB	A	0-5	$125x - 12.5x^2$	$1x$
CB	C	0-5	$62.5x$	$1x$

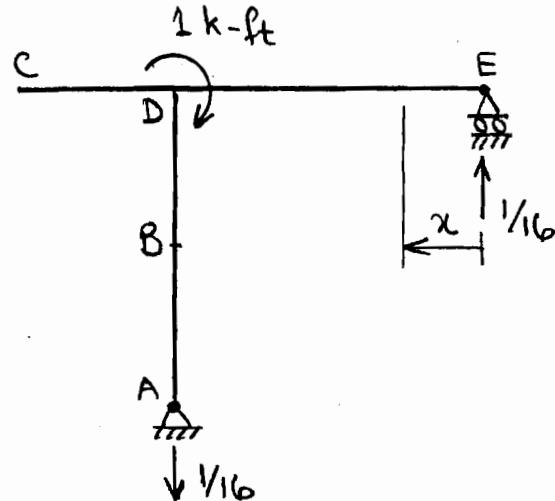
$$\begin{aligned}\Delta_C &= \frac{1}{EI} \left[\int_0^5 x(125x - 12.5x^2) dx + \int_0^5 x(62.5x) dx \right] \\ &= \frac{5859.375 \text{ kN.m}^3}{EI} = \frac{5859.375}{70(10^3)} \\ &= 0.0813 \text{ m} = \underline{\underline{81.3 \text{ mm}}} \rightarrow\end{aligned}$$

7-34



Real System - M

$$M = 29x - 1.5x^2$$

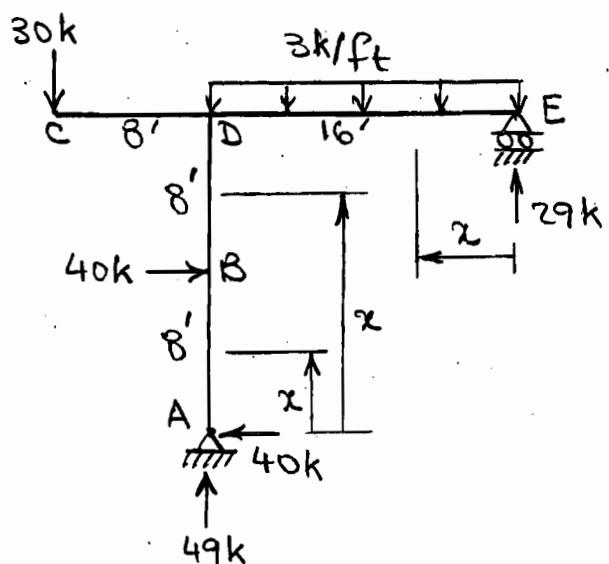


Virtual System - M_v

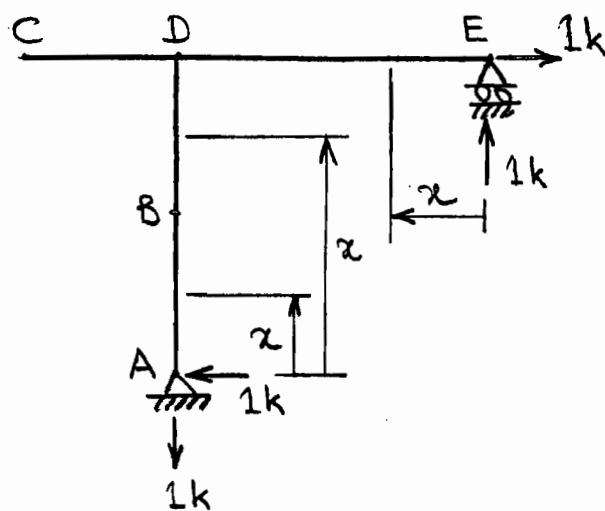
$$M_v = x/16$$

$$\begin{aligned} \Theta_D &= \frac{1}{2EI} \int_0^{16} \left(\frac{x}{16}\right) (29x - 1.5x^2) dx = \frac{469.33 \text{ k-ft}^2}{EI} \\ &= \frac{469.33 (12)^2}{2000(10000)} = \underline{\underline{0.0034 \text{ rad}}} \end{aligned}$$

7.35



Real System - M



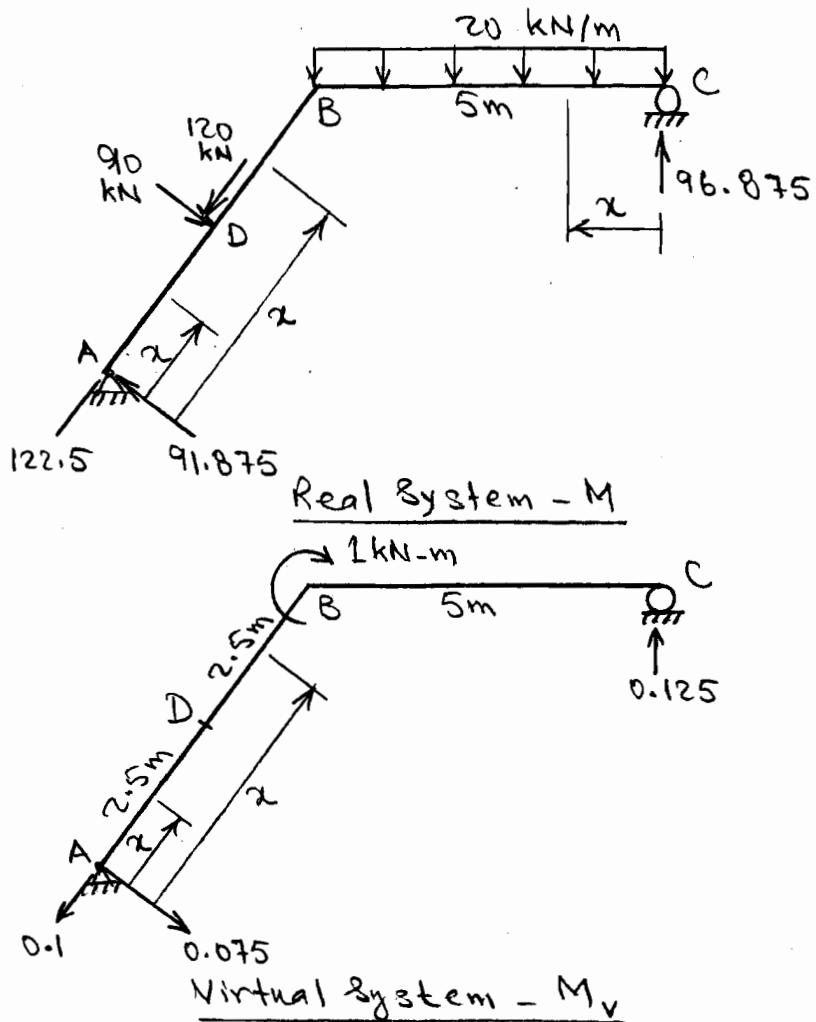
Virtual System - M_v

Segment	x coordinate		M (k-ft)	M_v (k-ft)
	Origin	Limits (ft)		
AB	A	0-8	40x	1x
BD	A	8-16	320	1x
ED	E	0-16	$29x - \frac{3}{2}x^2$	1x

$$(1k) \Delta_E = \frac{1}{2EI} \left[\int_0^8 x(40x) dx + \int_8^{16} x(320) dx + \int_0^{16} x(29x - \frac{3}{2}x^2) dx \right] = \frac{26282.67 k^2 \cdot ft^3}{EI}$$

$$\Delta_E = \frac{26282.67 (12)^3}{(2000)(10000)} = 2.27 \text{ in.} \rightarrow$$

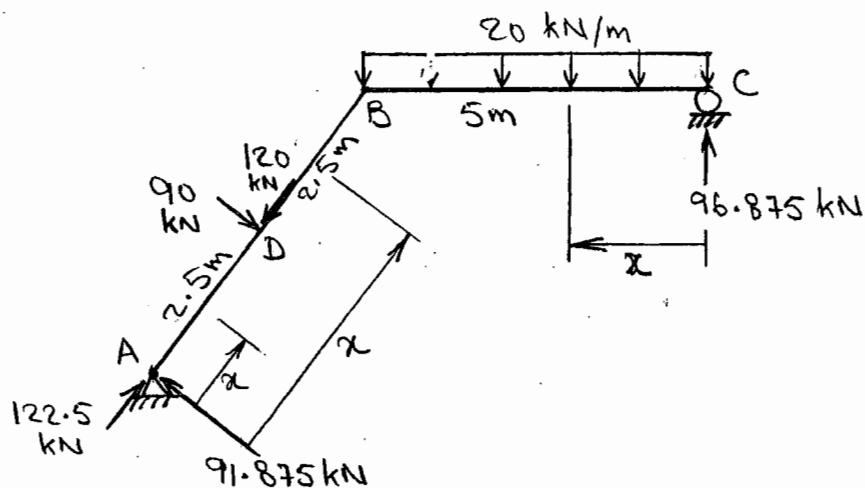
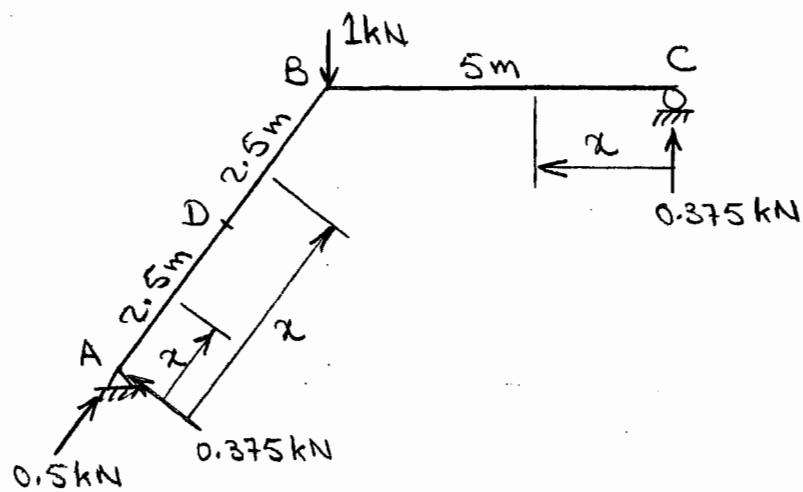
7.36



Segment	x Coordinate		M ($\text{kN}\cdot\text{m}$)	M_v ($\text{kN}\cdot\text{m}$)
	Origin	Limits (m)		
AD	A	0 - 2.5	$91.875x$	-0.075x
DB	A	2.5 - 5	$1.875x + 225$	-0.075x
CB	C	0 - 5	$96.875x - 10x^2$	$0.125x$

$$\begin{aligned}
 \Theta_B &= \frac{1}{EI} \left[\int_0^{2.5} -0.075x (91.875x) dx \right. \\
 &\quad + \int_{2.5}^5 -0.075x (1.875x + 225) dx \\
 &\quad \left. + \int_0^5 0.125x (96.875x - 10x^2) dx \right] \\
 &= \frac{110.01 \text{ kN}\cdot\text{m}^2}{EI} = \frac{110.01}{200(500)} = 0.0011 \text{ rad. } \checkmark
 \end{aligned}$$

7.37

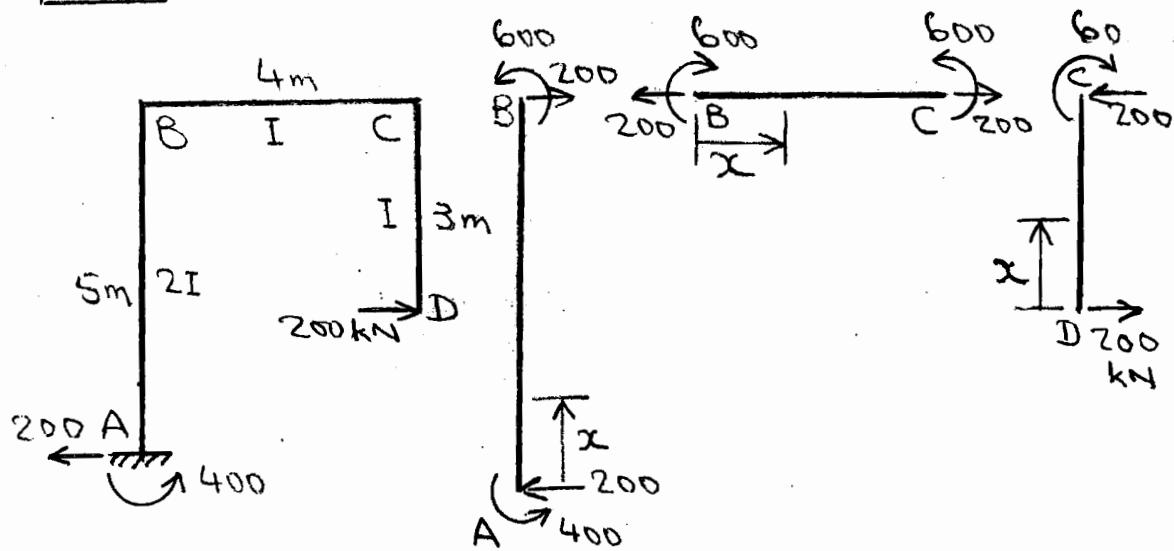
Real System - MVirtual System - M_v

Segment	x Coordinate		M (kN.m)	M _v (kN.m)
	Origin	Limits (m)		
AD	A	0-2.5	91.875x	0.375x
DB	A	2.5-5	1.875x + 225	0.375x
CB	C	0-5	96.875x - 10x ²	0.375x

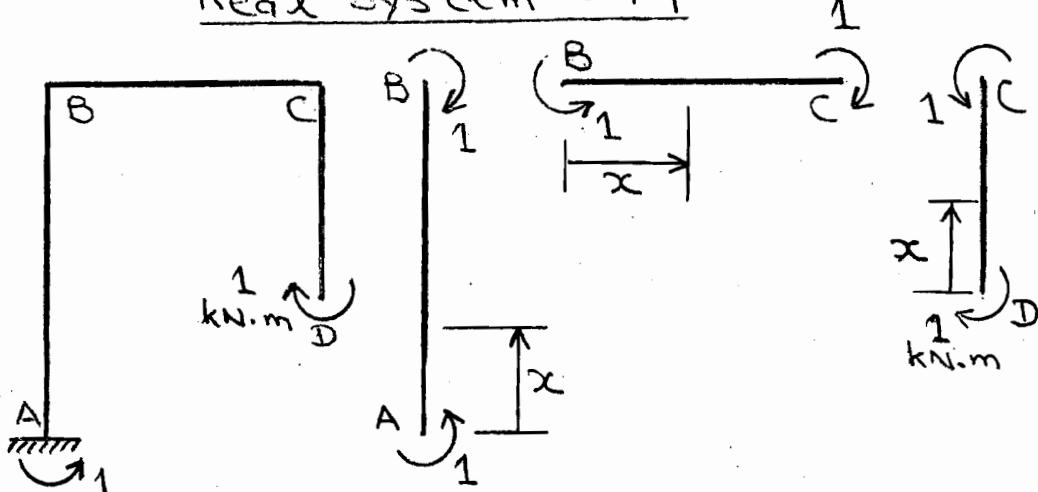
$$(1 \text{ kN}) \Delta_B = \frac{1}{EI} \left[\int_0^{2.5} 0.375x (91.875x) dx + \int_{2.5}^5 0.375x (1.875x + 225) dx + \int_0^5 0.375x (96.875x - 10x^2) dx \right] \\ = \frac{1923.83 \text{ kN} \cdot \text{m}^3}{EI}$$

$$\Delta_B = \frac{1923.83}{200(500)} = 0.0192 \text{ m} = 19.2 \text{ mm} \downarrow$$

7.38



Real System - M

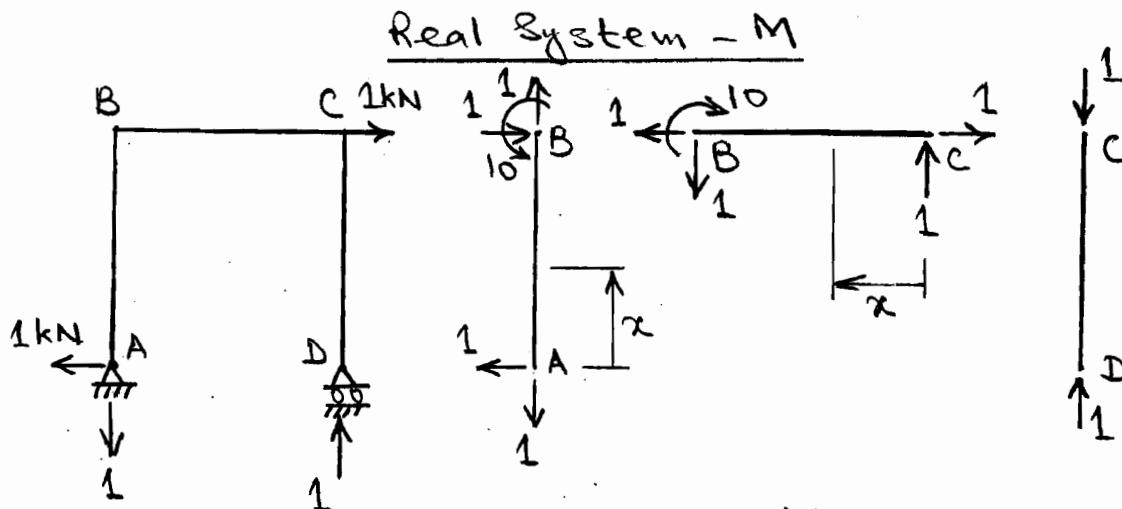
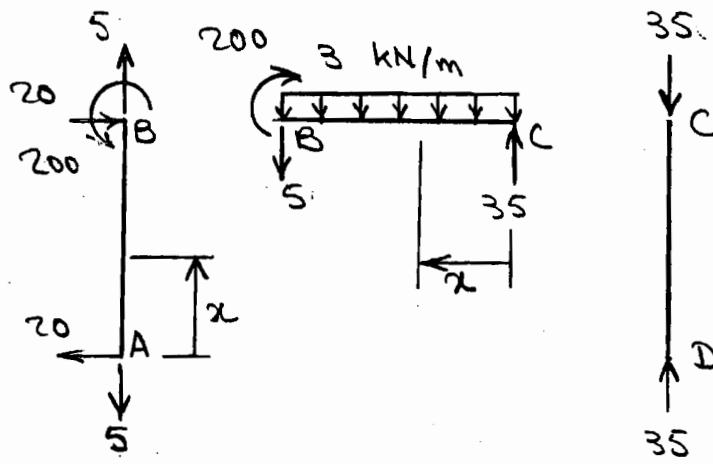
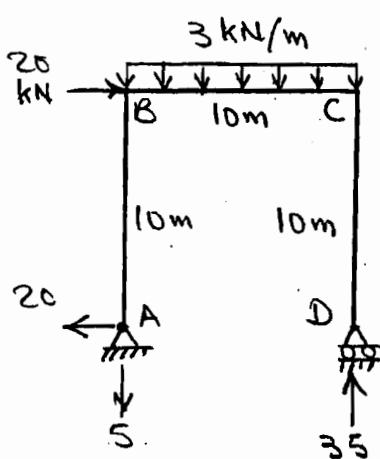


Virtual System - M_y

Segment	x Coordinate		M (kN.m)	M_V (kN.m)
	Origin,	Limits (m)		
AB	A	0-5	-400 + 200x	-1
BC	B	0-4	600	-1
DC	D	0-3	-200x	1

$$\begin{aligned}\Theta_D &= \frac{1}{EI} \left[\frac{1}{2} \int_0^5 -1(-400 + 200x) dx + \int_0^4 -1(600) dx \right. \\ &\quad \left. + \int_0^3 1(-200x) dx \right] \\ &= -\frac{3550 \text{ kN}\cdot\text{m}^2}{EI} = -\frac{3550}{70(1290)} = -0.0393 \text{ rad} \\ &\quad = 0.0393 \text{ rad} \end{aligned}$$

7.39



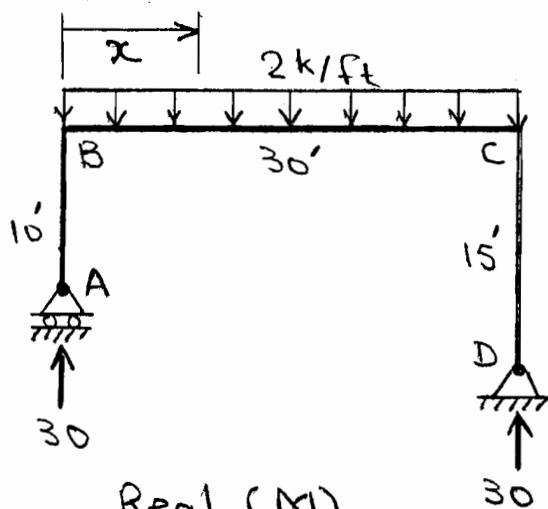
Segment	x coordinate		M (kN.m)	M _v (kN.m)
	Origin	Limits (m)		
AB	A	0-10	20x	1x
CB	C	0-10	$35x - 1.5x^2$	$1x + 10$

$$(1\text{kN}) \Delta_C = \frac{1}{EI} \left[\int_0^{10} x(20x) dx + \int_0^{10} x(35x - 1.5x^2) dx \right]$$

$$= \frac{14583.33}{EI} \text{ kN}^2 \cdot \text{m}^3$$

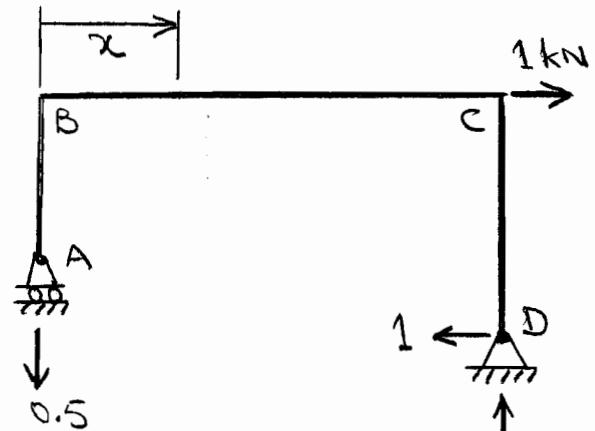
$$\Delta_C = \frac{14583.33}{200(400)} = \underline{0.182 \text{ m} \rightarrow}$$

7.40



Real (M)

$$M = 30x - \frac{2x^2}{2}$$

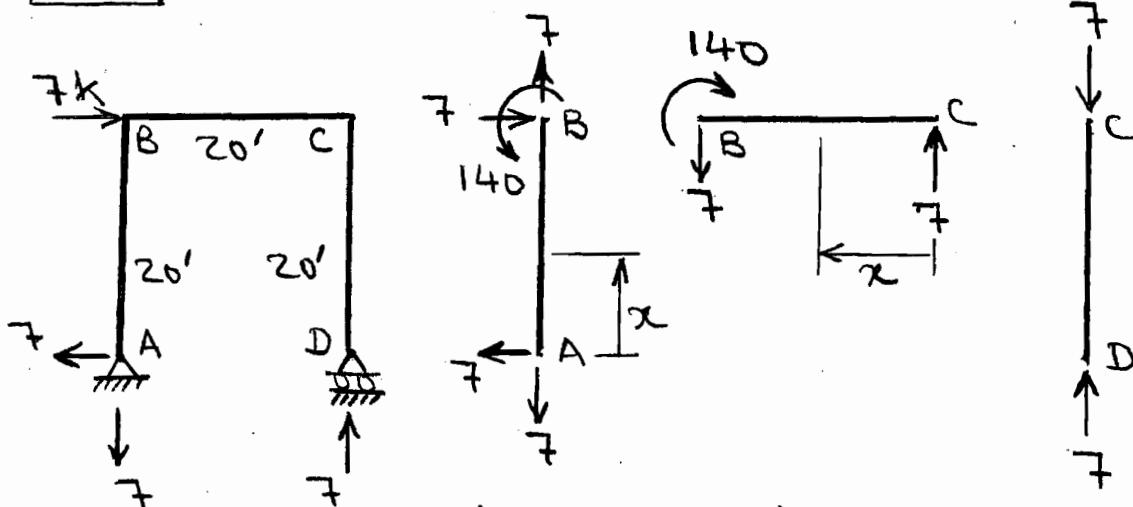


Virtual (M_v) 0.5

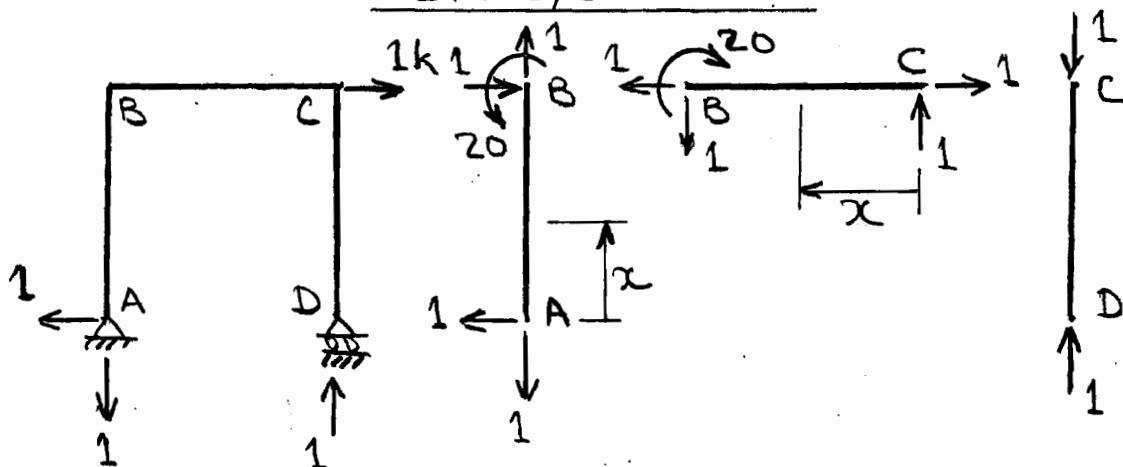
$$M_v = -0.5x$$

$$\begin{aligned}\Delta_C &= \frac{1}{EI} \int_0^{30} (-0.5x)(30x - x^2) dx = -\frac{33750 \text{ k-ft}^3}{EI} \\ &= -\frac{33750 (12)^3}{29000 (1500)} = -1.34 \text{ in.} = \underline{\underline{1.34 \text{ in. } \leftarrow}}\end{aligned}$$

7.41



Real System - M



Virtual System - M_v

Segment	x coordinate		M (k-ft)	M_v (k-ft)
	Origin	Limits (ft)		
AB	A	0-20	7x	1x
CB	C	0-20	7x	1x

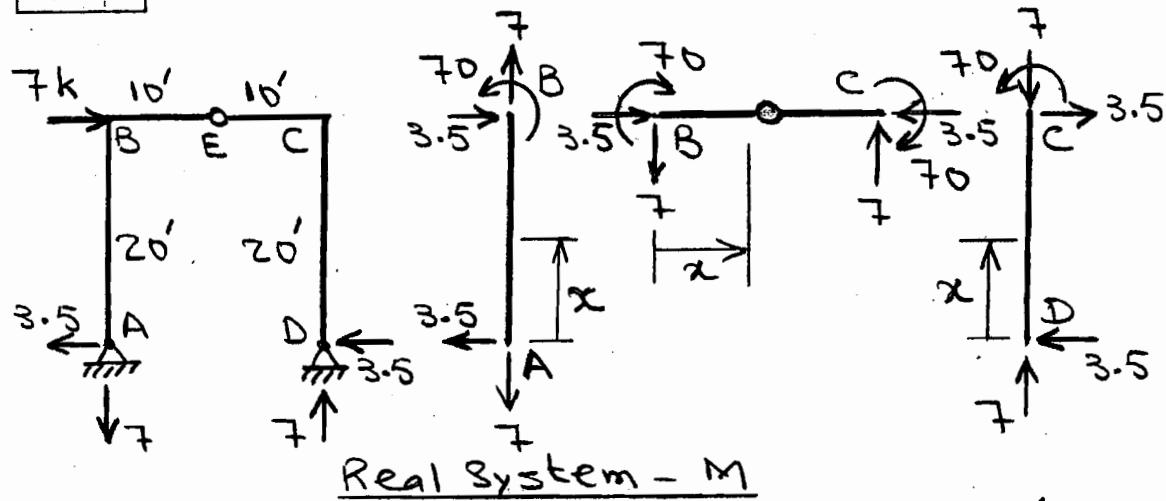
$$(1k) \Delta_C = \frac{1}{EI} \left[\int_0^{20} (1x)(7x) dx + \int_0^{20} (1x)(7k) dx \right]$$

$$= \frac{37333 k^2 \cdot ft^3}{EI}$$

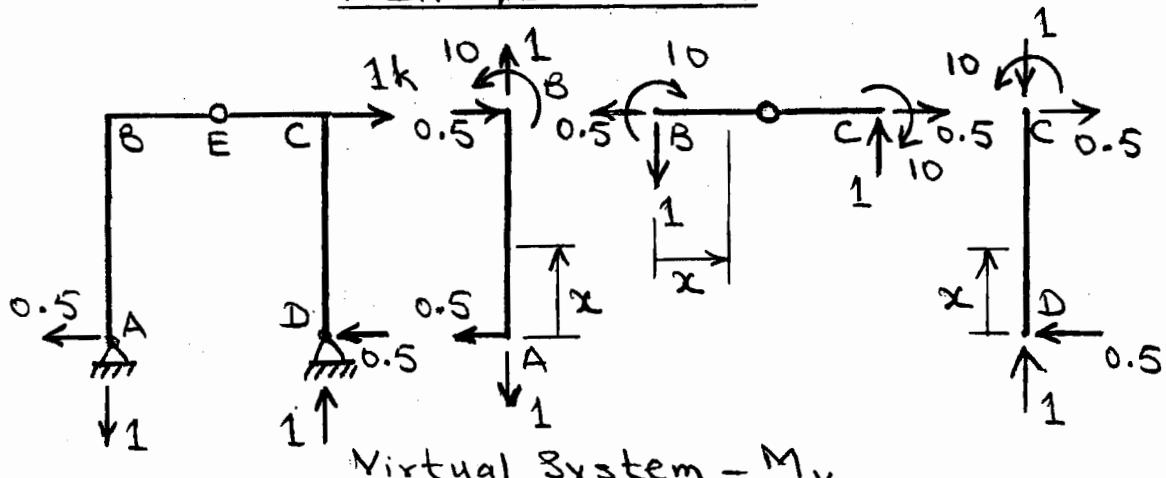
$$\Delta_C = \frac{37333 k \cdot ft^3}{EI} = \frac{37333 (12)^3}{29000(I)} = 1 \text{ in.}$$

from which, $I = 2225 \text{ in}^4$

7.42



Real System - M



Virtual System - M_V

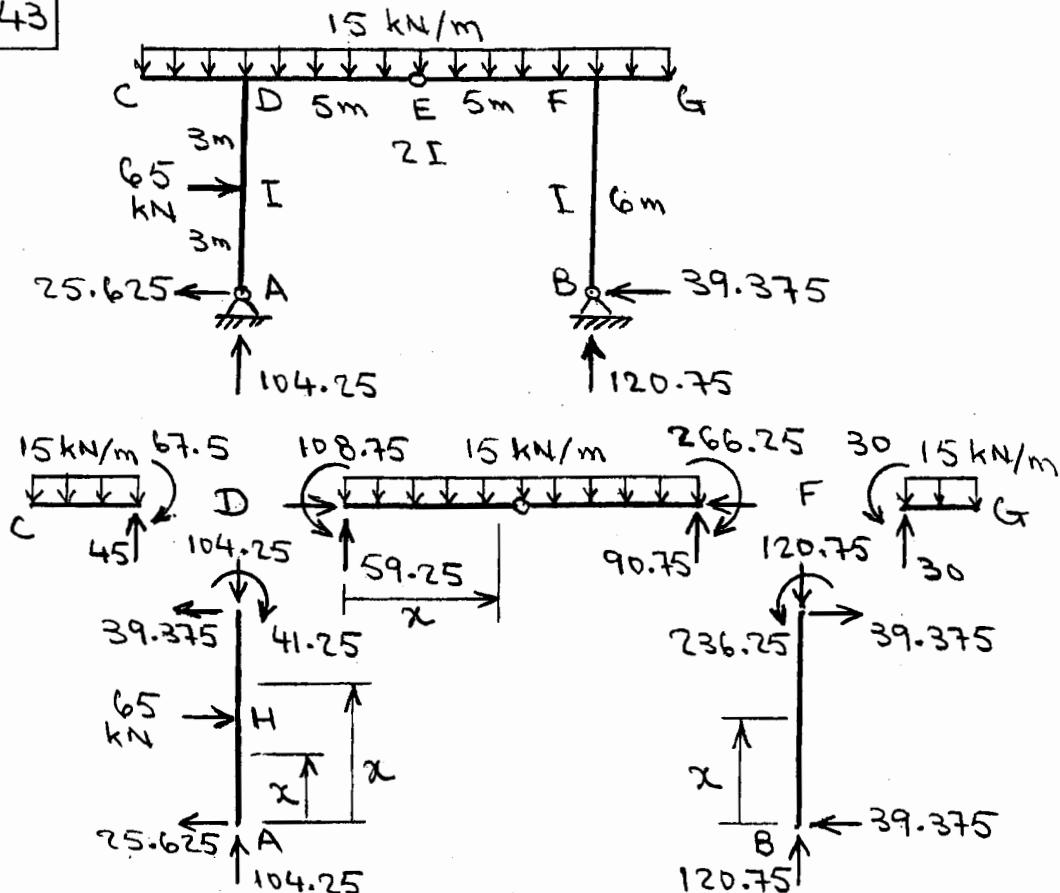
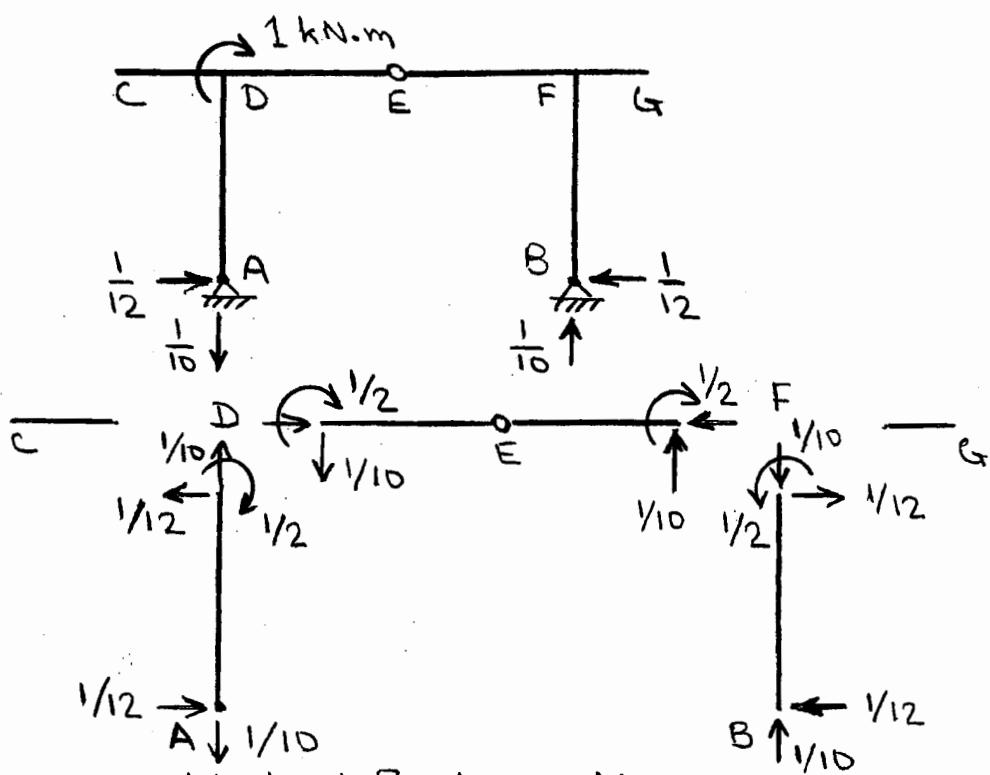
Segment	x coordinate		M (k-ft)	M_V (k-ft)
	Origin	Limits (ft)		
AB	A	0-20	$3.5x$	$0.5x$
BC	B	0-20	$70 - 7x$	$10 - 1x$
DC	D	0-20	$3.5x$	$0.5x$

$$(1k) \Delta_C = \frac{1}{EI} \left[2 \int_0^{20} (0.5x)(3.5x) dx + \int_0^{20} (10-x)(70-7x) dx \right] = \frac{14000 k^2 ft^3}{EI}$$

$$\Delta_C = \frac{14000 k \cdot ft^3}{EI} = \frac{14000 (12)^3}{29000 (I)} = 1 \text{ in.}$$

From which, $I = 834 \text{ in}^4$

7.43

Real System - M

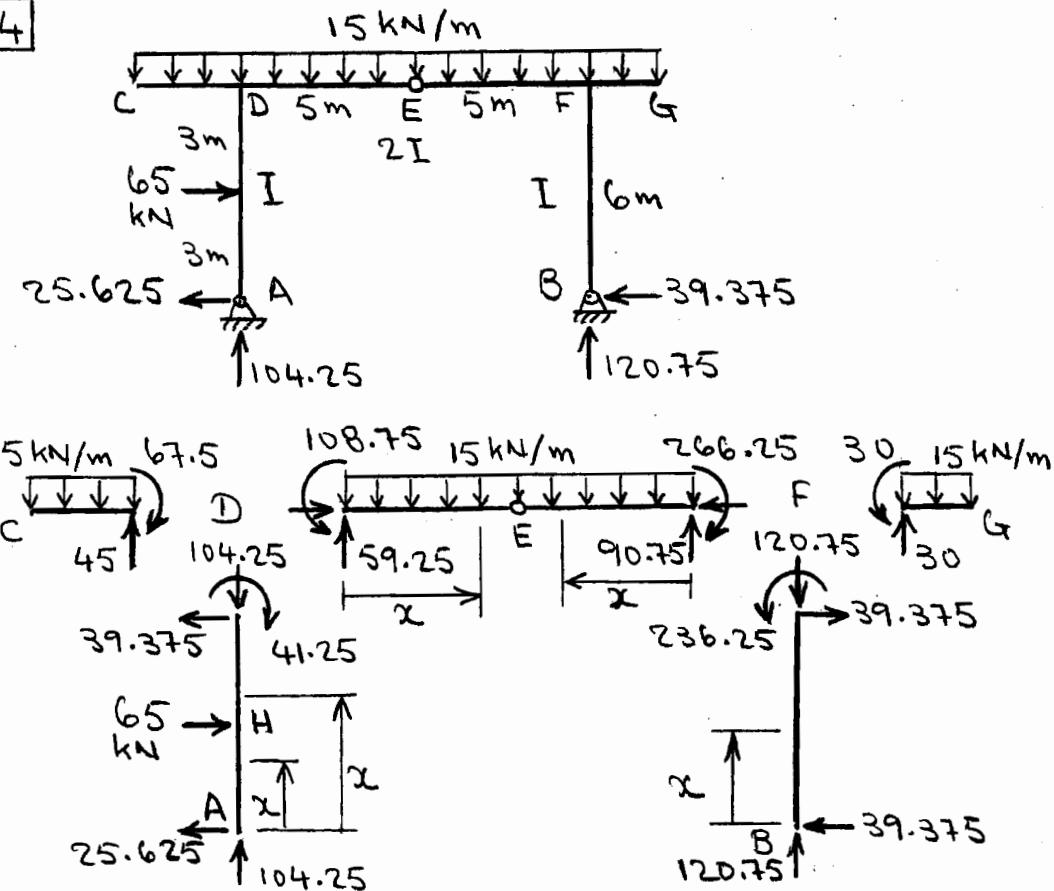
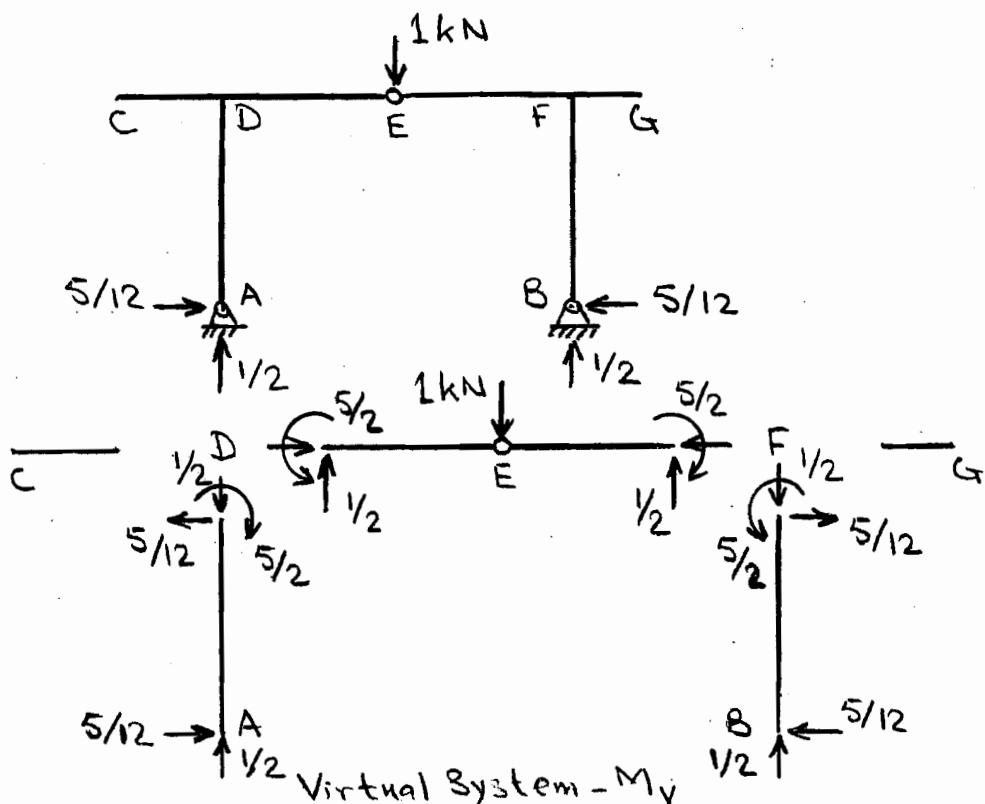
7.43 (contd.)

Segment	x Coordinate		M (kN.m)	M_v (kN.m)
	Origin	Limits (m)		
AH	A	0-3	$25.625x$	$-x/12$
HD	A	3-6	$25.625x - 65(x-3)$	$-x/12$
BF	B	0-6	$39.375x$	$x/12$
DF	D	0-10	$59.25x - 108.75$ $- 7.5x^2$	$\frac{1}{2} - \frac{x}{10}$

$$(1 \text{ kN.m}) \Theta_D = \frac{1}{EI} \left[\int_0^3 \left(-\frac{x}{12}\right) (25.625x) dx + \right. \\ \left. \int_3^6 \left(-\frac{x}{12}\right) (25.625x - 65x + 195) dx + \int_0^6 \frac{x}{12} (39.375x) dx \right. \\ \left. + \frac{1}{2} \int_0^{10} \left(\frac{1}{2} - \frac{x}{10}\right) (59.25x - 108.75 - 7.5x^2) dx \right] \\ = \frac{270 \text{ kN}^2 \cdot \text{m}^3}{EI}$$

$$\Theta_D = \frac{270}{200(350)} = \underline{\underline{0.00386 \text{ rad.}}} \quad \nabla$$

7.44

Real System - MVirtual System - M_V

7.44 (contd.)

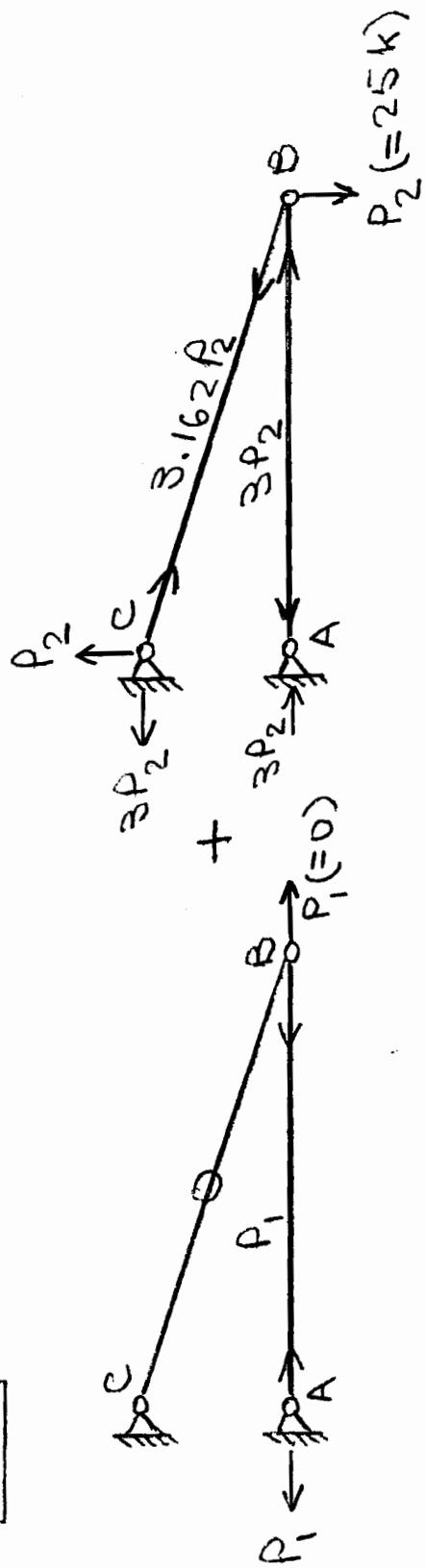
Segment	x coordinate		M (kN.m)	M _v (kN.m)
	Origin	Limits (m)		
AH	A	0 - 3	$25.625x$	$-5x/12$
HD	A	3 - 6	$25.625x - 65(x-3)$	$-5x/12$
BF	B	0 - 6	$39.375x$	$5x/12$
DE	D	0 - 5	$59.25x - 108.75$ $-7.5x^2$	$\frac{x}{2} - \frac{5}{2}$
FE	F	0 - 5	$90.75x - 266.25$ $-7.5x^2$	$\frac{x}{2} - \frac{5}{2}$

$$(1\text{ kN}) \Delta_E = \frac{1}{EI} \left[\int_0^3 \left(-\frac{5x}{12}\right) (25.625x) dx \right. \\ + \int_3^6 \left(-\frac{5x}{12}\right) (25.625x - 65x + 195) dx \\ + \int_0^6 \frac{5x}{12} (39.375x) dx + \frac{1}{2} \int_0^5 \left(\frac{x}{2} - \frac{5}{2}\right) (59.25x - 108.75 \\ \left. - 7.5x^2) dx + \frac{1}{2} \int_0^5 \left(\frac{x}{2} - \frac{5}{2}\right) (90.75x - 266.25 - 7.5x^2) dx \right] \\ = \frac{1607.8125 \text{ kN}^2 \cdot \text{m}^3}{EI}$$

$$\Delta_E = \frac{1607.8125}{200(350)} = 0.023 \text{ m } \downarrow$$

$$\underline{\Delta_E = 23 \text{ mm } \downarrow}$$

7.45

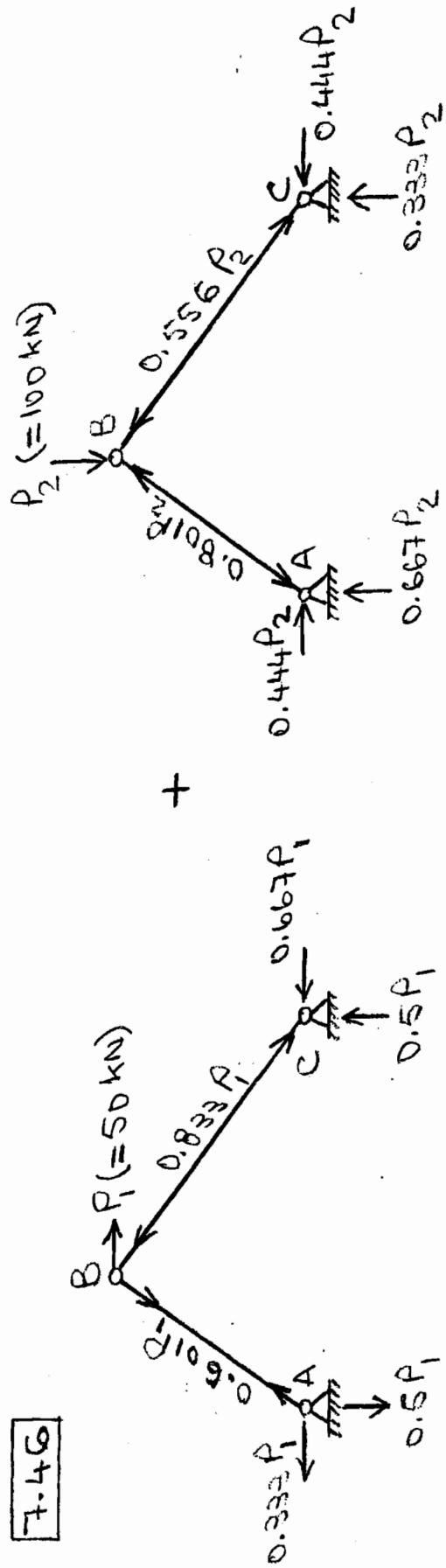


Member (in.)	L	F	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	$\frac{P_1 = 0, P_2 = 25 \text{ kN}}{\sum F_L}$
			$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	$\frac{\partial F}{\partial P_1}$
AB	180	$P_1 - 3P_2$	-1	-3	-13500
BC	189.74	$3.162 P_2$	0	3.162	0
					$\sum -13500 \quad 87932.67$

$$\Delta_{BA} = -\frac{13500}{10000(6)} = -0.225 \text{ in.} = 0.225 \text{ in.} \rightarrow$$

$$\Delta_{BV} = \frac{87932.67}{10000(6)} = 1.466 \text{ in.} \downarrow$$

7.46

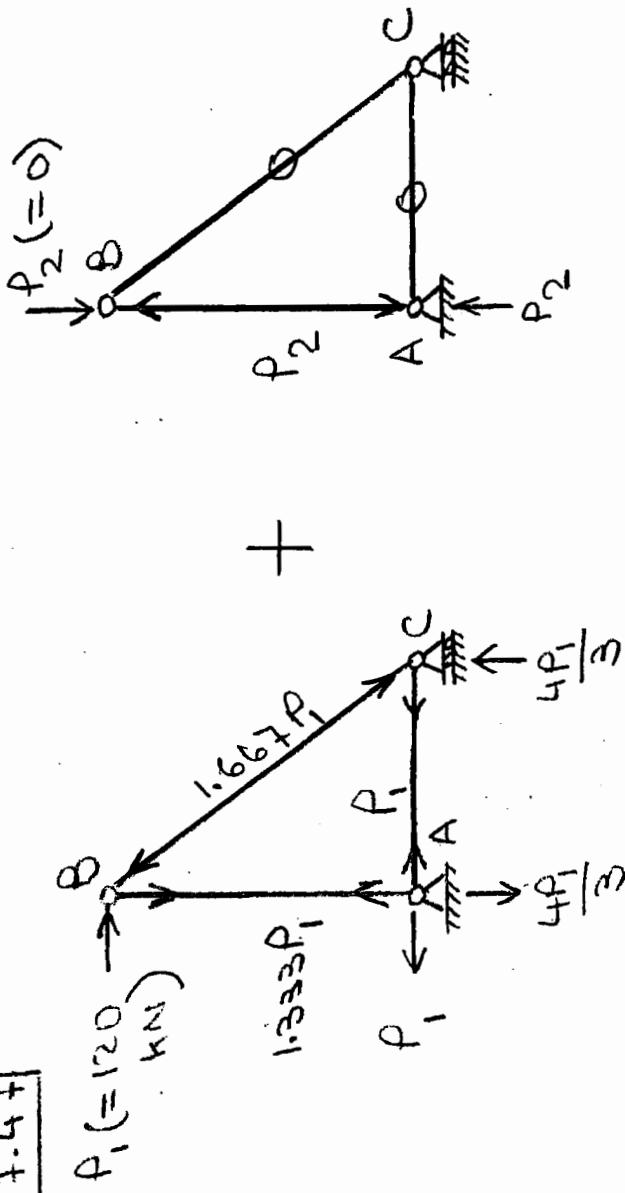


Member (m)	F	$\frac{\partial F}{\partial P_1}$		$\frac{\partial F}{\partial P_2}$		\sum
		$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	
AB	3.606	0.601P ₁ - 0.801P ₂	0.601	-0.801	-0.801	144.65
BC	5	-0.833P ₁ - 0.556P ₂	-0.833	-0.556	405.09	270.06

$$\Delta_{BH} = \frac{296.56}{70(10^6)(0.001)} = 0.00424 \text{ m} = 4.24 \text{ mm} \rightarrow$$

$$\Delta_{BV} = \frac{414.71}{70(10^6)(0.001)} = 0.00592 \text{ m} = 5.92 \text{ mm} \downarrow$$

7.47

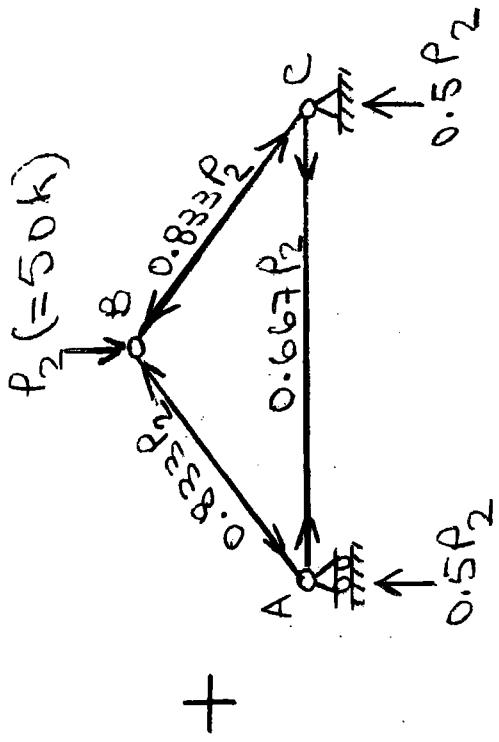
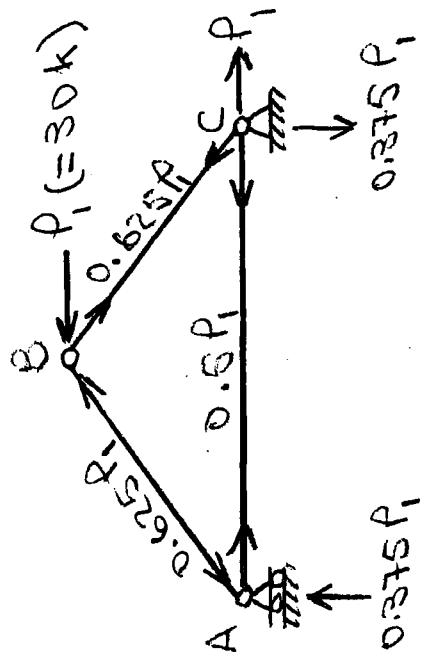


Member	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	\sum
AC	-1.333	-1.667	0	0	-640
AB	0	0	0.333	0.667	2880
BC	0	0	0.333	0.667	2880

$$\Delta_{BA} = \frac{2880}{200(10^6)(0.0015)} = 0.00096 \text{ m} = 9.6 \text{ mm} \rightarrow$$

$$\Delta_{AV} = -\frac{640}{288(10^6)(0.0015)} = -0.00213 \text{ m} = 2.13 \text{ mm} \downarrow$$

-7.48

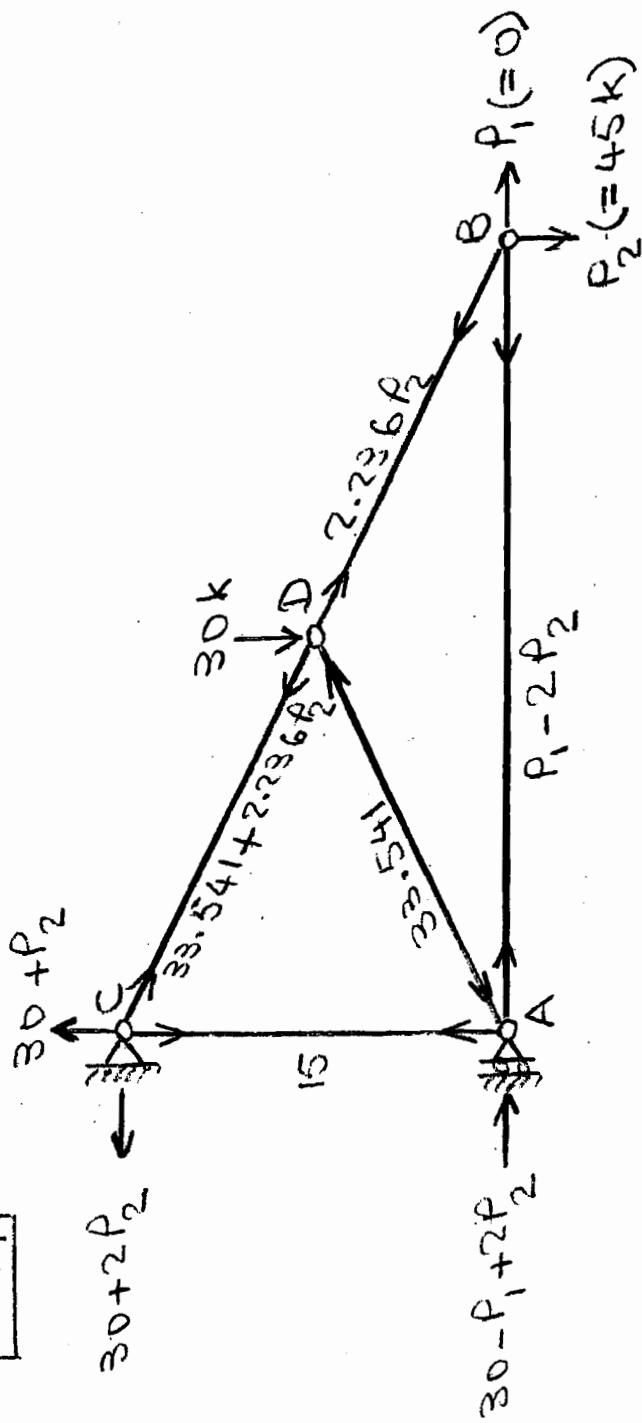


Member	L (in.)	F	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	$P_1 = 30 \text{ k}, P_2 = 50 \text{ k}$
A-C	192	$0.5 P_1 + 0.667 P_2$	0.5	0.667	6186.67
A-B	120	$-0.625 P_1 - 0.833 P_2$	-0.625	-0.833	4531.25
B-C	120	$0.625 P_1 - 0.833 P_2$	0.625	-0.833	-1718.75
					$\Sigma 2291.67$
					$\Sigma 7452.5$
					14520

$$\Delta_{EH} = \frac{7452.5}{29000(3)} = 0.0857 \text{ in. } \downarrow$$

$$\Delta_{EV} = \frac{14520}{29000(3)} = 0.167 \text{ in. } \downarrow$$

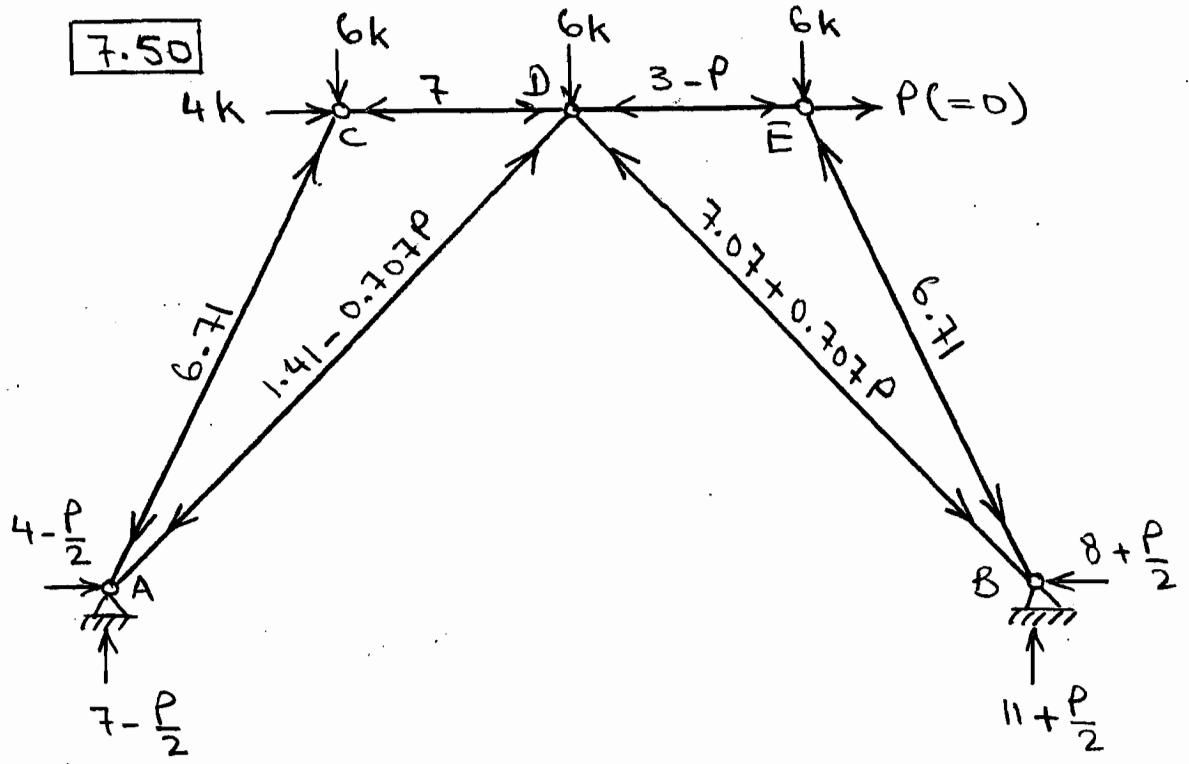
7.49



Member	L (in.)	A (in. ²)	F	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	$\frac{\partial F}{\partial P_1}$	$\frac{\partial F}{\partial P_2}$	\sum
AB	240	6.4	0	-2P ₂	P ₁ - 2P ₂	-	0	-3600	7200	-3600
AC	120	6.4	0	15	-	0	0	0	0	0
AD	134.16	6.4	0	-33.541	-	0	0	0	0	0
BC	134.16	6.4	0	33.541 + 2.236P ₂	2.236P ₂	2.236P ₂	2.236P ₂	2.236P ₂	2.236P ₂	18938.98
CD	134.16	6.4	0	30 - 2.236P ₂	2.236P ₂	2.236P ₂	2.236P ₂	2.236P ₂	2.236P ₂	18938.98

$$\Delta_{BH} = \frac{-3600}{10000} = -0.36 \text{ in.} = 0.36 \text{ in.} \downarrow$$

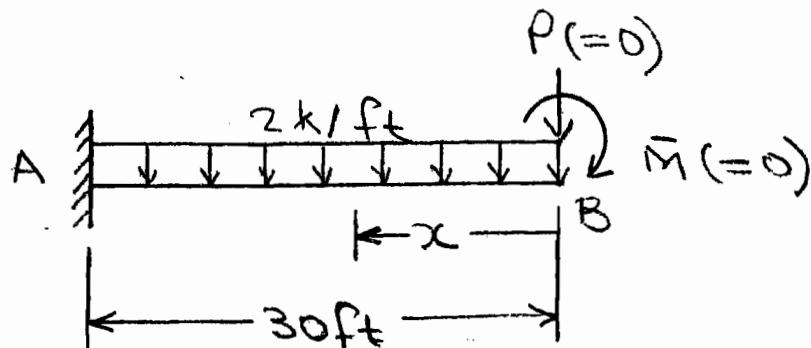
$$\Delta_{BV} = \frac{18938.98}{10000} = 1.894 \text{ in.} \downarrow$$



Member	L (in.)	F	$\frac{\partial F}{\partial P}$	$P=0$
			$\frac{\partial F}{\partial P}$ (FL)	
AC	134.16	-6.71	0	0
BE	134.16	-6.71	0	0
AD	169.71	$-1.41 + 0.707P$	0.707	-169.18
BD	169.71	$-7.07 - 0.707P$	-0.707	848.29
CD	60	-7	0	0
DE	60	$-3 + P$	1	-180
Σ				499.11

$$\Delta_{EH} = \frac{499.11}{29000(6)} = \underline{0.0029 \text{ in.}} \rightarrow$$

7.51



$$M = -x^2 - \bar{M} - Px$$

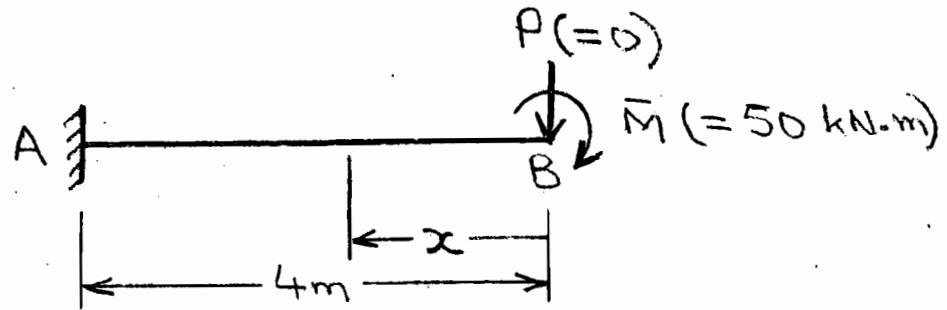
$$\frac{\partial M}{\partial \bar{M}} = -1$$

$$\begin{aligned}\Theta_B &= \frac{1}{EI} \left[\int_0^{30} (-1)(-x^2) dx \right] = \frac{9000 \text{ k-ft}^2}{EI} \\ &= \frac{9000 (12)^2}{29000 (3000)} = \underline{0.0149 \text{ rad}} \quad \checkmark\end{aligned}$$

$$\frac{\partial M}{\partial P} = -x$$

$$\begin{aligned}\Delta_B &= \frac{1}{EI} \left[\int_0^{30} (-x)(-x^2) dx \right] = \frac{202500 \text{ k-ft}^3}{EI} \\ &= \frac{202500 (12)^3}{29000 (3000)} = \underline{4.022 \text{ in.} \downarrow}\end{aligned}$$

7.52



$$M = -\bar{M} - Px$$

$$\frac{\partial M}{\partial M} = -1$$

$$\Theta_B = \frac{1}{EI} \left[\int_0^4 (-1)(-50) dx \right] = \frac{200 \text{ kN}\cdot\text{m}^2}{EI}$$

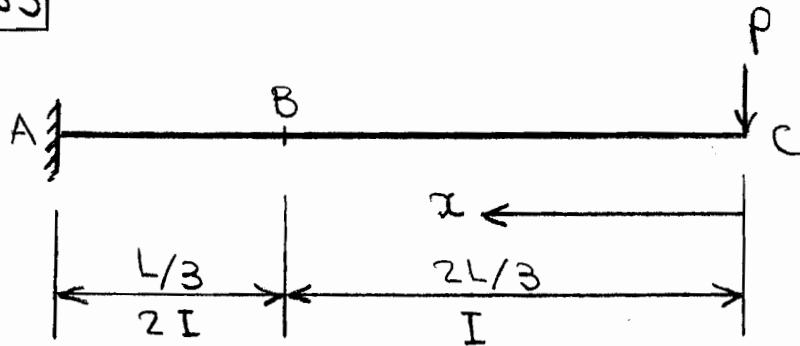
$$= \frac{200}{70(164)} = \underline{0.0174 \text{ rad}} \quad \checkmark$$

$$\frac{\partial M}{\partial P} = -x$$

$$\Delta_B = \frac{1}{EI} \left[\int_0^4 (-x)(-50) dx \right] = \frac{400 \text{ kN}\cdot\text{m}^3}{EI}$$

$$= \frac{400}{70(164)} = 0.0348 \text{ m} = \underline{34.8 \text{ mm}} \quad \downarrow$$

7.53

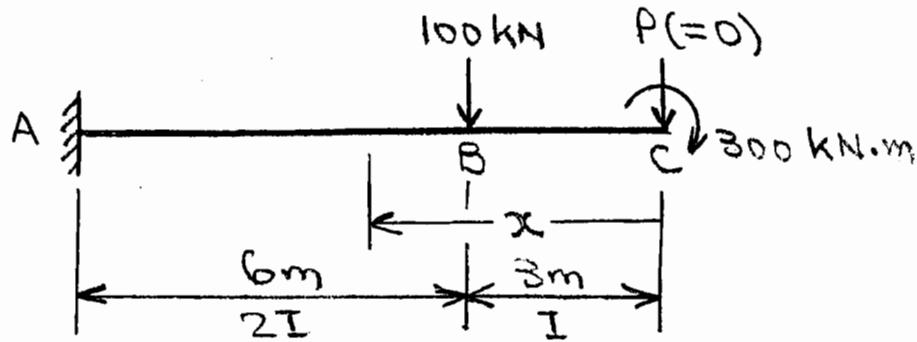


Segment	x Coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits		
CB	C	$0 - \frac{2L}{3}$	$-Px$	$-x$
BA	C	$\frac{2L}{3} - L$	$-Px$	$-x$

$$\Delta_C = \frac{1}{EI} \left[\int_0^{\frac{2L}{3}} (-x)(-Px) dx + \frac{1}{2} \int_{\frac{2L}{3}}^L (-x)(-Px) dx \right]$$

$$= \frac{35PL^3}{162EI}$$

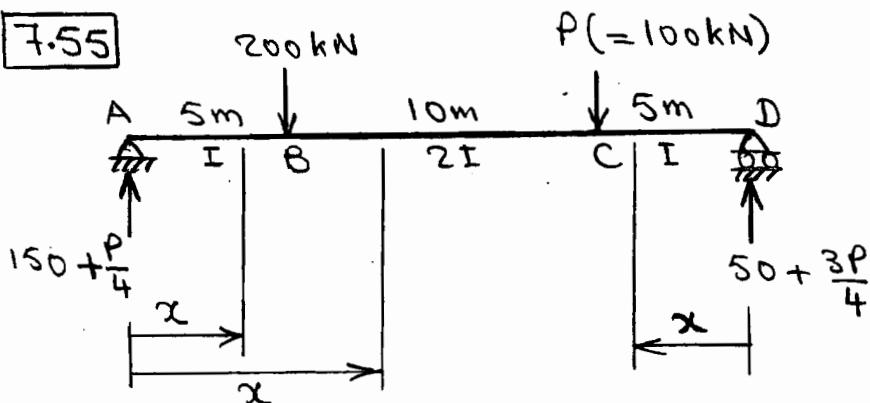
7.54



Segment	x Coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits (m)		
CB	C	0-3	$-300 - Px$	$-x$
BA	C	3-9	$-300 - Px$ $-100(x-3)$	$-x$

$$\begin{aligned}\Delta_C &= \frac{1}{EI} \left[\int_0^3 (-x)(-300) dx + \frac{1}{2} \int_3^9 (-x)(-100x) dx \right] \\ &= \frac{13050 \text{ kN}\cdot\text{m}^3}{EI} = \frac{13050}{70(500)} \\ &= 0.373 \text{ m} = \underline{373 \text{ mm}} \downarrow\end{aligned}$$

7.55



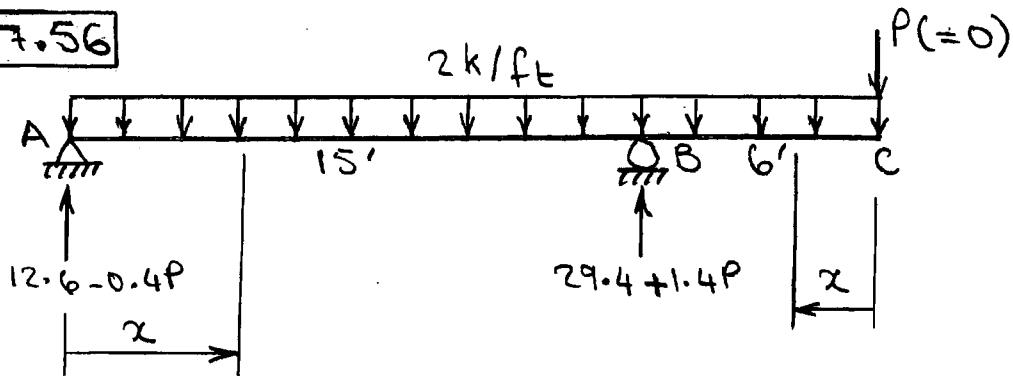
Segment	x coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits		
AB	A	0 - 5	$(150 + P/4)x$	0.25x
BC	A	5 - 15	$(150 + P/4)x - 200(x-5)$	0.25x
DC	D	0 - 5	$(50 + 3/4P)x$	0.75x

Substituting $P = 100 \text{ kN}$ and integrating, we obtain

$$\begin{aligned}\Delta_c &= \frac{1}{EI} \left[\int_0^5 (0.25x)(175x) dx \right. \\ &\quad + \frac{1}{2} \int_5^{15} 0.25x(-25x + 1000) dx \\ &\quad \left. + \int_0^5 0.75x(125x) dx \right] \\ &= \frac{14843.75 \text{ kN-m}^3}{EI} = \frac{14843.75}{250(600)} = 0.099 \text{ m}\end{aligned}$$

$$\underline{\Delta_c = 99 \text{ mm } \downarrow}$$

7.56

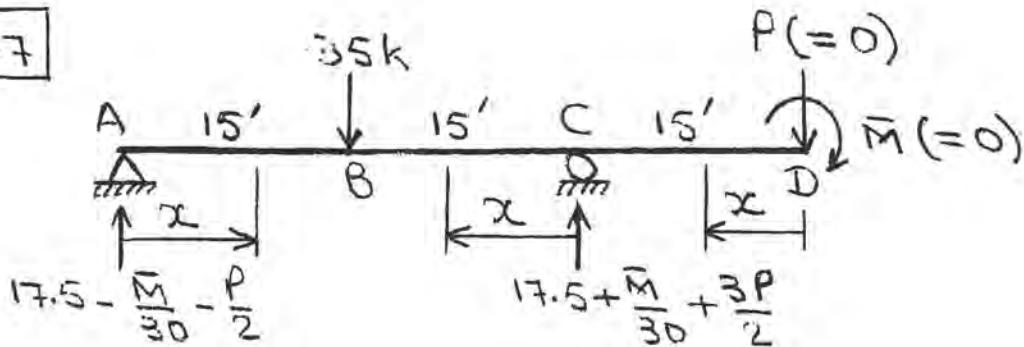


Segment	x Coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits		
AB	A	0 - 15	$(12.6 - 0.4P)x - x^2$	-0.4x
CD	C	0 - 6	$-Px - x^2$	-x

Substituting $P=0$ and integrating, we obtain

$$\begin{aligned}
 \Delta_C &= \frac{1}{EI} \left[\int_0^{15} (-0.4x)(12.6x - x^2) dx \right. \\
 &\quad \left. + \int_0^6 -x(-x^2) dx \right] \\
 &= - \frac{283.5 k \cdot ft^3}{EI} = - \frac{283.5 (12)^3}{29000 (3500)} \\
 &= -0.0048 \text{ in.} = \underline{\underline{0.0048 \text{ in.} \uparrow}}
 \end{aligned}$$

7.57



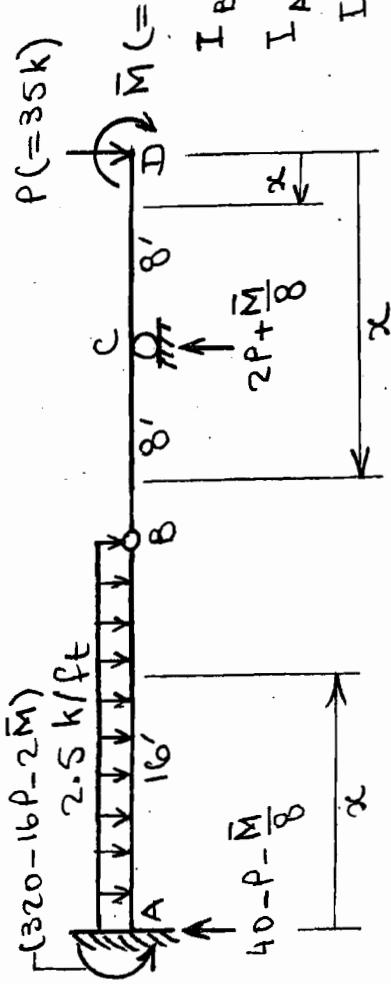
Segment	x Coordinate		M	$\frac{\partial M}{\partial M}$	$\frac{\partial M}{\partial P}$
	Origin	Limits (ft)			
AB	A	0-15	$(17.5 - \frac{M}{30} - \frac{P}{2})x$	$-x/30$	$-x/2$
CB	C	0-15	$(17.5 + \frac{M}{30} + \frac{3P}{2})x - M - P(x+15)$	$\frac{x}{30} - 1$	$\frac{x}{2} - 15$
DC	D	0-15	$-M - Px$	-1	$-x$

Substituting $M = P = 0$, and integrating, we obtain

$$\begin{aligned}\Theta_D &= \frac{1}{EI} \left[\int_0^{15} -\frac{x}{30} (17.5x) dx + \int_0^{15} \left(\frac{x}{30} - 1\right) (17.5x) dx \right] \\ &= -\frac{1968.75 k \cdot ft^2}{EI} = -\frac{1968.75 (12)^2}{10000 (2500)} \\ &= -0.01134 \text{ rad} = 0.01134 \text{ rad } \uparrow\end{aligned}$$

$$\begin{aligned}\Delta_D &= \frac{1}{EI} \left[\int_0^{15} -\frac{x}{2} (17.5x) dx + \int_0^{15} \left(\frac{x}{2} - 15\right) (17.5x) dx \right] \\ &= -\frac{29531.25 k \cdot ft^3}{EI} = -\frac{29531.25 (12)^3}{10000 (2500)} \\ &= -2.04 \text{ in.} = 2.04 \text{ in. } \uparrow\end{aligned}$$

7.58



$$\begin{aligned} I_{BC} &= I_{CD} = I \\ I_{AB} &= \frac{4}{3} I \\ I &= 3000 \text{ in.}^4 \end{aligned}$$

Segment	x Coordinate Origin	Limits	M	$\frac{\partial M}{\partial x}$	$\frac{\partial M}{\partial P}$
DC	D	0 - 8	$-Px - \bar{M}$	-1	-x
CB	D	8 - 16	$-Px - \bar{M} + (2P + \frac{\bar{M}}{8})(x - 8)$	$\frac{x}{8} - 2$	$x - 16$
AB	A	0 - 16	$-(320 - 16P - 2\bar{M}) + (40 - P - \frac{\bar{M}}{8})x - 1.25x^2$	$2 - \frac{x}{8}$	$16 - x$

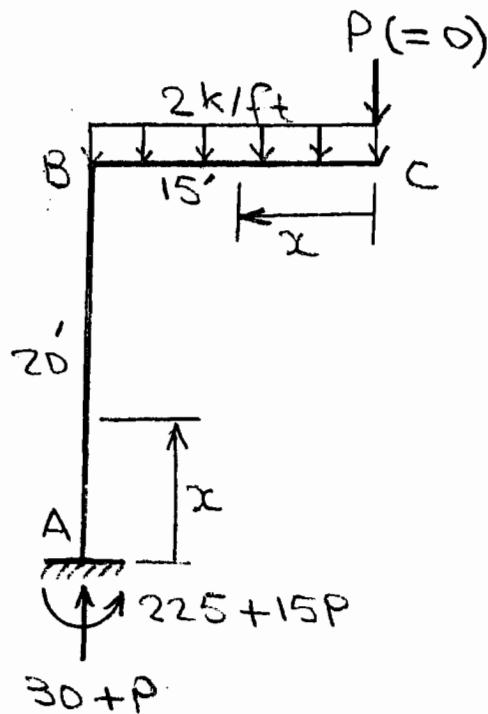
Substituting $\bar{M} = 0$ and $P = 35 \text{ k}$, and integrating χ , we obtain

$$\theta_D = \frac{1}{EI} \left[\int_0^8 -1(-35x) dx + \int_8^{16} \left(\frac{x}{8} - 2 \right) (35x - 560) dx \right]$$

$$+ \frac{3}{4} \int_0^{16} \left(2 - \frac{x}{8} \right) (240 + 5x - 1.25x^2) dx = \frac{4426.67 \text{ k-ft}^2}{EI} = 0.0071 \text{ rad.}$$

$$\Delta D = \frac{1}{EI} \left[\int_0^8 -x(-35x) dx + \int_8^{16} (x - 16)(35x - 560) dx \right] = \frac{32426.67 \text{ k-ft}^3}{EI} = 0.62 \text{ in.}$$

7.59

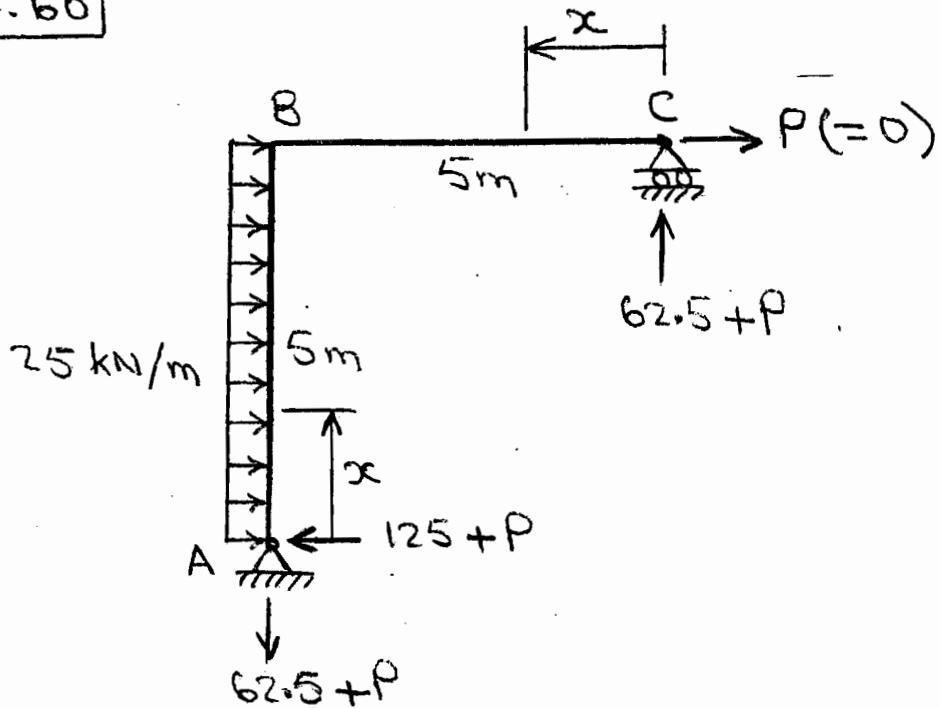


Segment	x Coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits (ft)		
AB	A	0-20	$-(225 + 15P)$	-15
CB	C	0-15	$-x^2 - Px$	$-x$

Substituting $P = 0$ and integrating, we obtain

$$\begin{aligned}\Delta_C &= \frac{1}{EI} \left[\int_0^{20} (-15)(-225) dx + \int_0^{15} (-x)(-x^2) dx \right] \\ &= \frac{80156.25 \text{ k-ft}^3}{EI} = \frac{80156.25 (12)^3}{29000 (2000)} \\ &= \underline{2.388 \text{ in.} \downarrow}\end{aligned}$$

7.60

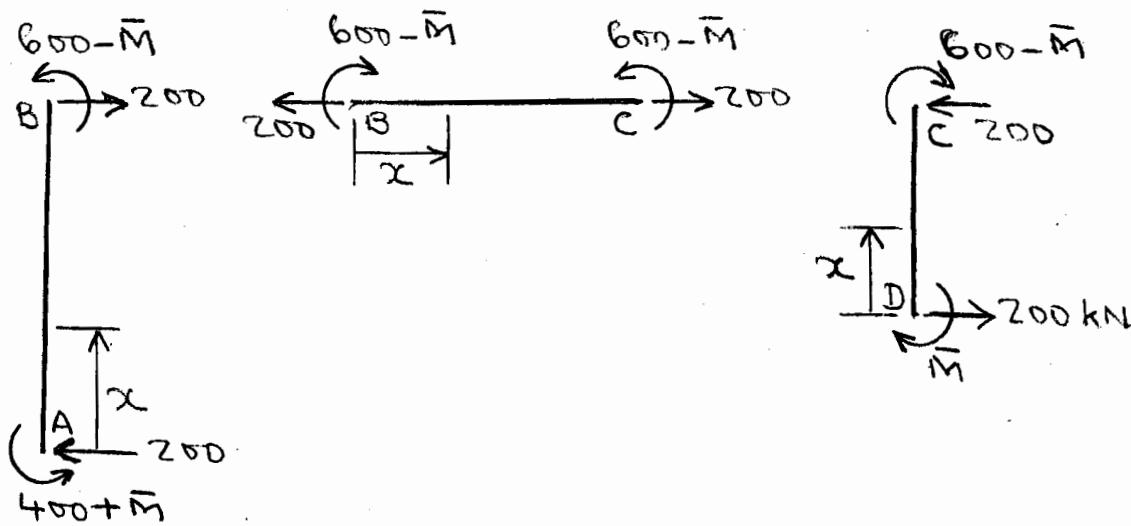
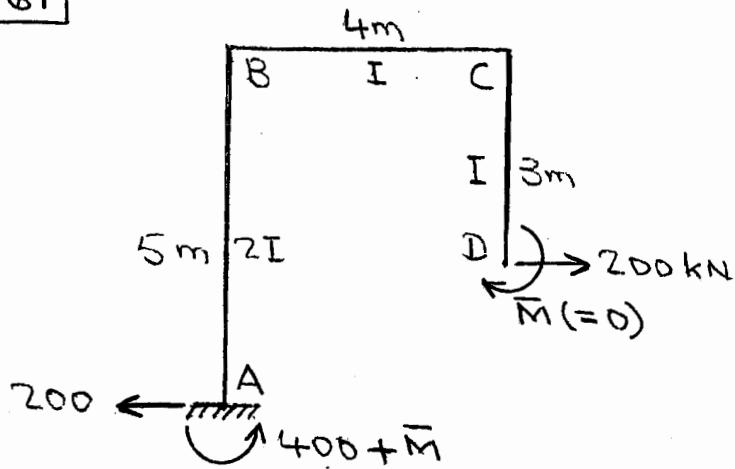


Segment	x Coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits (m)		
AB	A	0-5	$(125+P)x - 12.5x^2$	x
CB	C	0-5	$(62.5+P)x$	x

Substituting $P = 0$ and integrating, we obtain

$$\begin{aligned}\Delta_C &= \frac{1}{EI} \left[\int_0^5 x(125x - 12.5x^2) dx + \int_0^5 x(62.5x) dx \right] \\ &= \frac{5859.375 \text{ kN.m}^3}{EI} = \frac{5859.375}{70(1030)} \\ &= 0.0813 \text{ m} = \underline{81.3 \text{ mm}} \rightarrow\end{aligned}$$

7.61

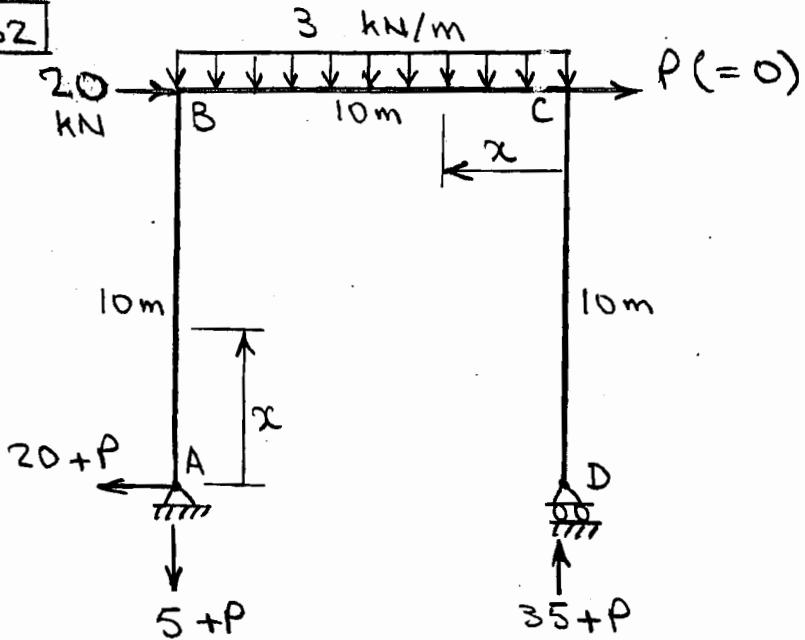


Segment	x Coordinate		M	$\frac{\partial M}{\partial M}$
	Origin	Limits (m)		
AB	A	0 - 5	$-(400 + M) + 200x$	-1
BC	B	0 - 4	$600 - M$	-1
DC	D	0 - 3	$-200x + M$	-1

Substituting $M=0$ and integrating, we obtain

$$\begin{aligned} \Theta_D &= \frac{1}{EI} \left[\frac{1}{2} \int_0^5 -1(-400 + 200x) dx + \int_0^4 -1(600) dx \right. \\ &\quad \left. + \int_0^3 1(-200x) dx \right] \\ &= -\frac{3550 \text{ kN.m}^2}{EI} = -\frac{3550}{70(1290)} = -0.0393 \text{ rad} \\ &= 0.0393 \text{ rad} \end{aligned}$$

7.62



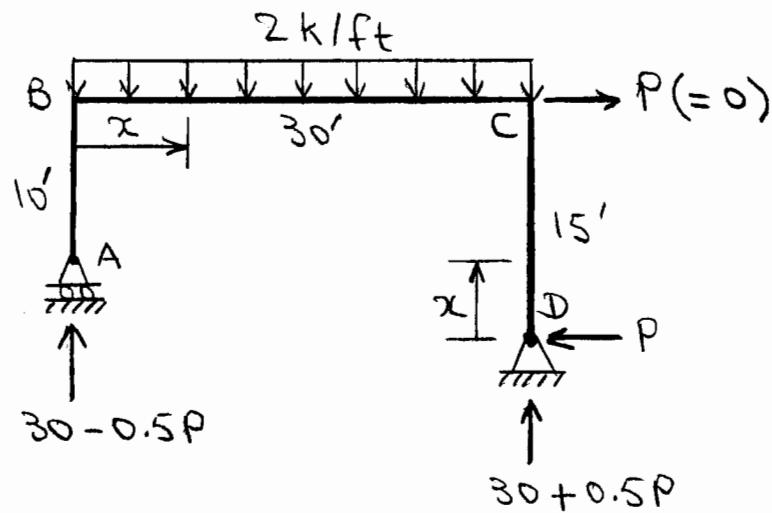
Segment	x Coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits		
AB	A	0-10	$(20+P)x$	x
CB	C	0-10	$(35+P)x - 1.5x^2$	x

Substituting $P=0$ and integrating, we obtain:

$$\Delta_C = \frac{1}{EI} \left[\int_0^{10} x(20x) dx + \int_0^{10} x(35x - 1.5x^2) dx \right]$$

$$= \frac{14583.33 \text{ kN-m}^3}{EI} = 0.182 \text{ m} \rightarrow$$

7.63



Segment	x Coordinate		M	$\frac{\partial M}{\partial P}$
	Origin	Limits		
BC	B	0 - 30	$(30 - 0.5P)x - x^2$	-0.5x
DC	D	0 - 15	Px	x

Substituting $P = 0$ and integrating, we obtain

$$\Delta_C = \frac{1}{EI} \int_0^{30} (-0.5x)(30x - x^2) dx = -\frac{33750 \text{ k-ft}^3}{EI}$$

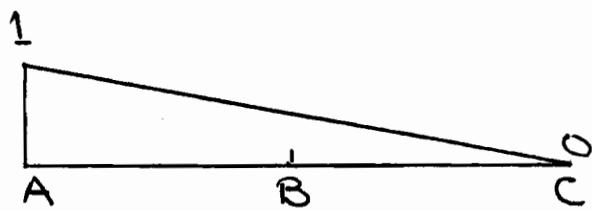
$$= -\frac{33750 (12)^3}{29000 (1500)} = -1.34 \text{ in.} = \underline{1.34 \text{ in.} \leftarrow}$$

Chapter Eight

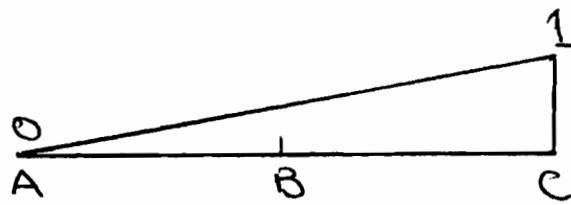
Influence Lines

CHAPTER 8

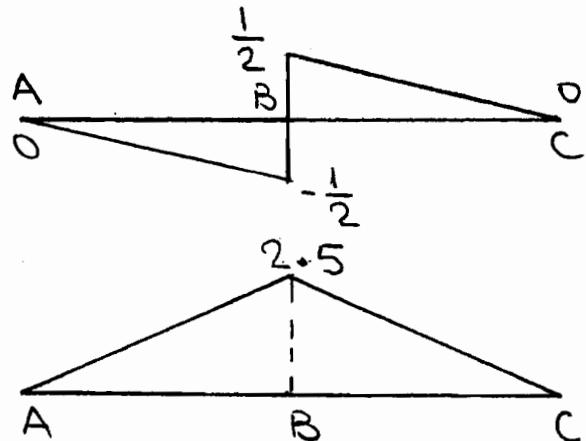
8-1



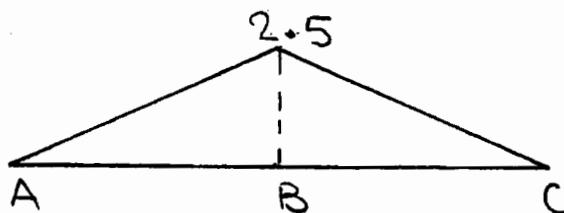
A_y



C_y

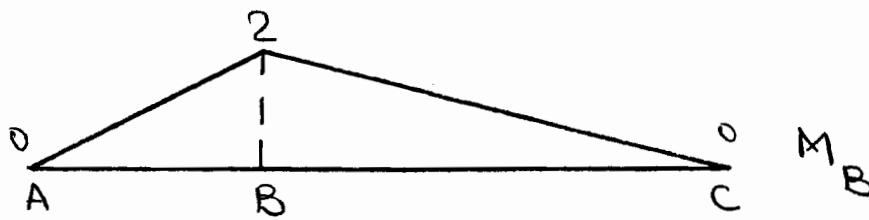
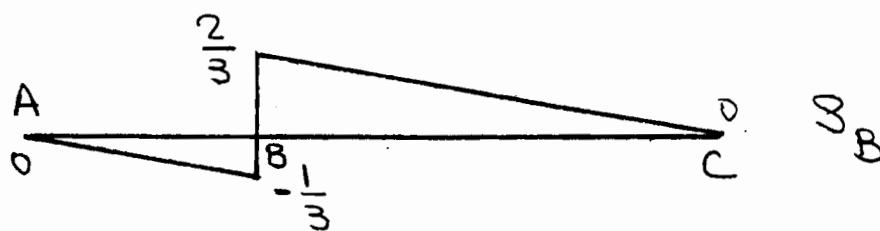
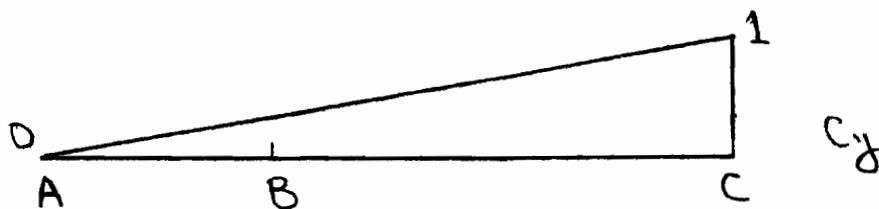
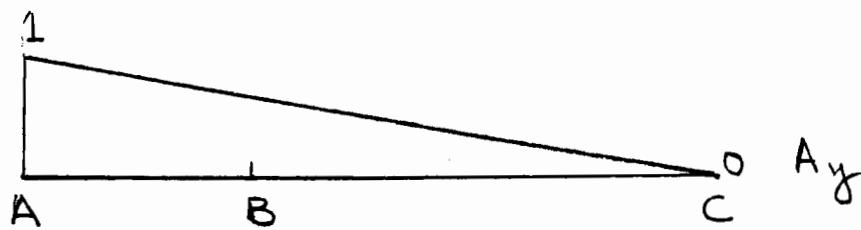


S_B

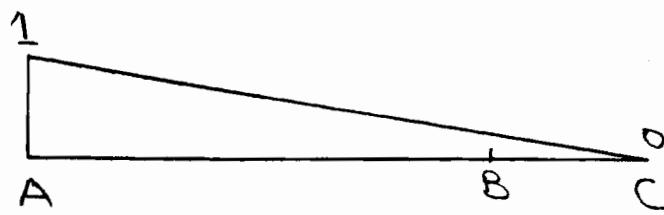


M_B

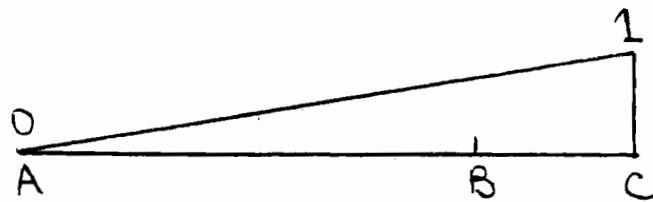
8.2



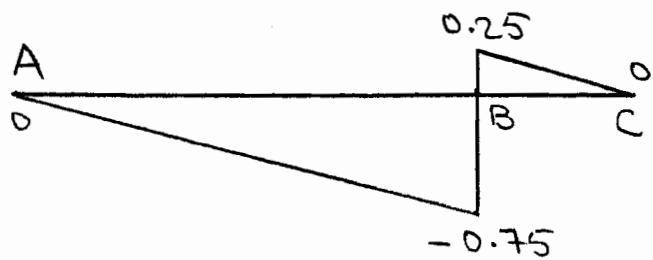
8.3



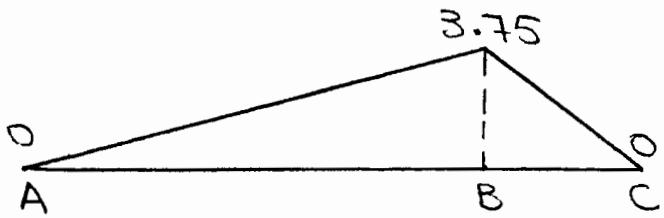
A_y



C_y

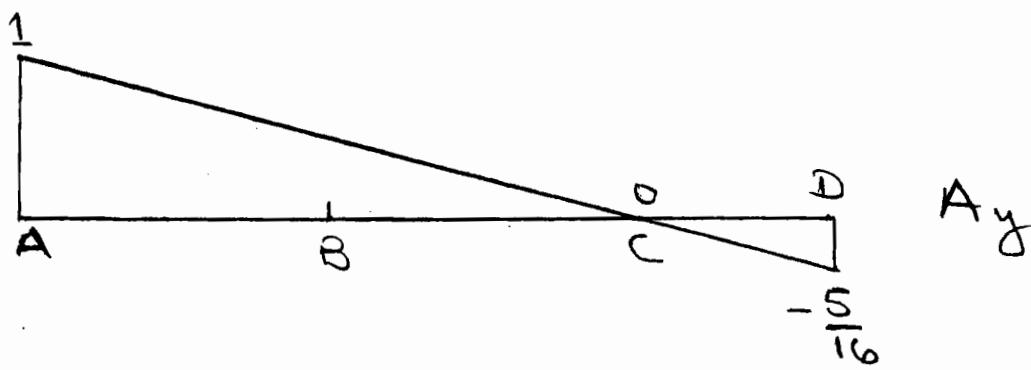


δ_B

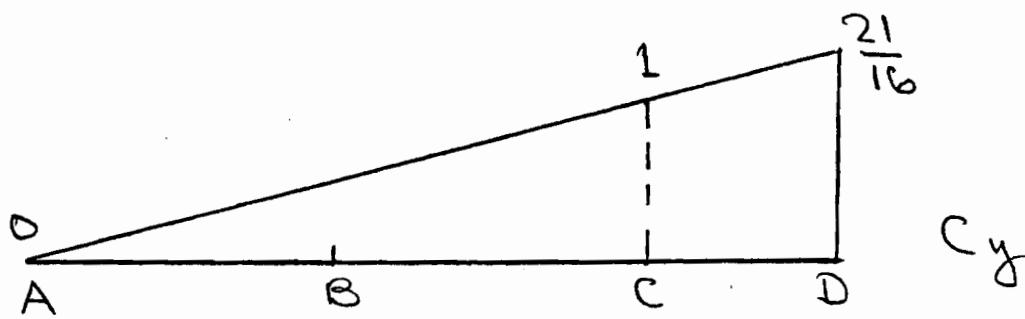


M_B

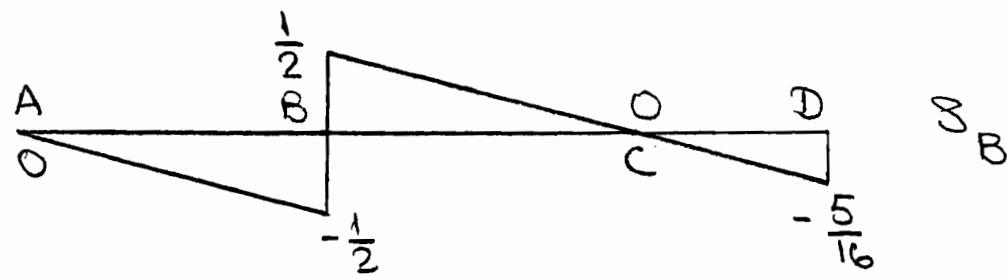
8.4



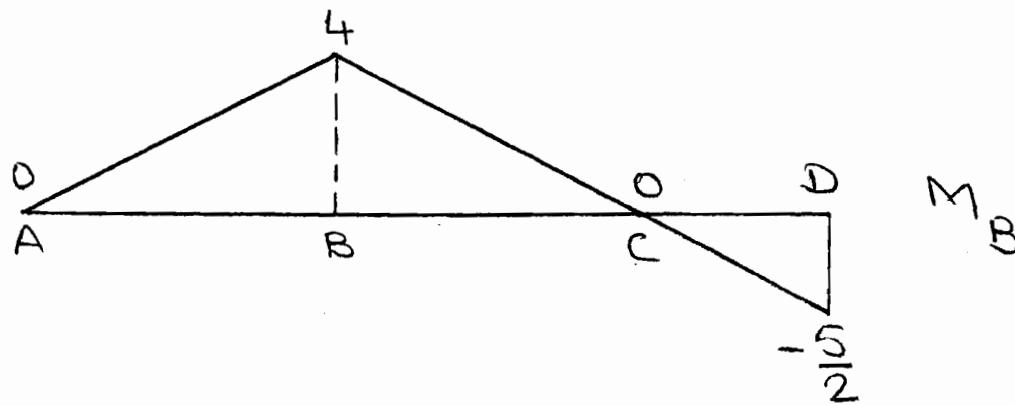
A_y



C_y

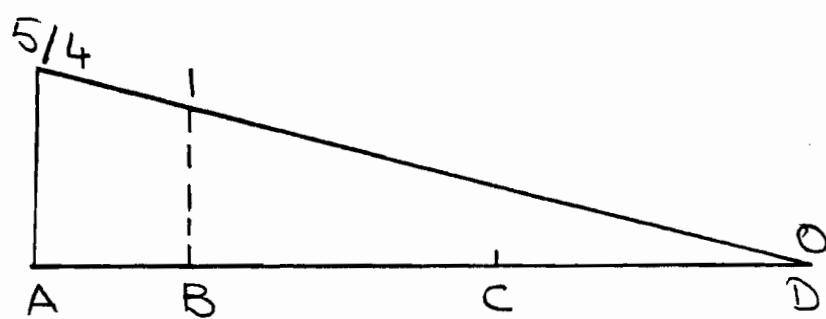


θ_B

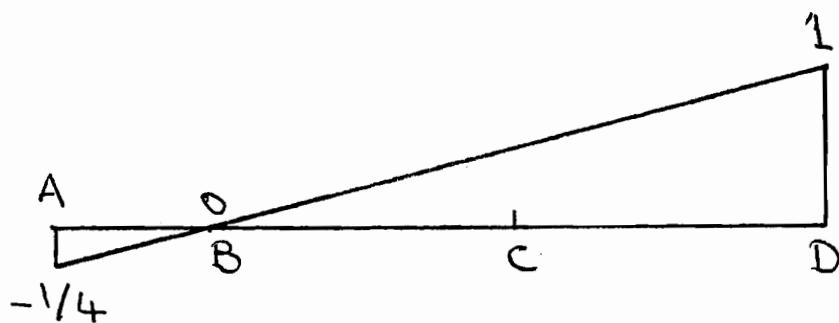


M_B

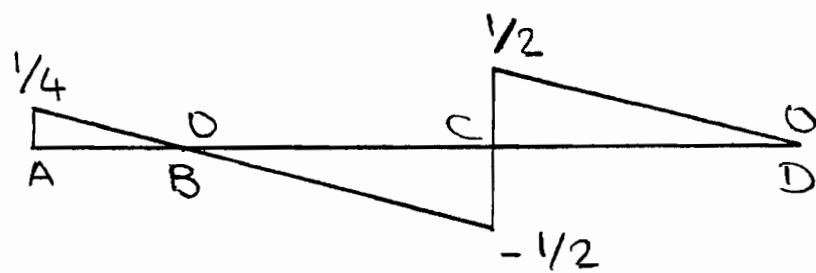
8.5



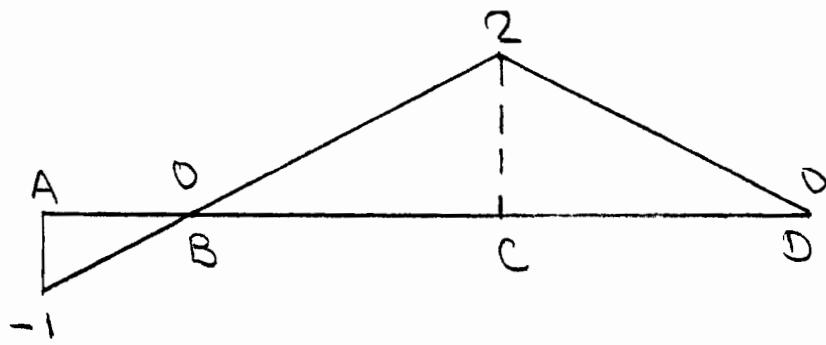
B_y



D_y

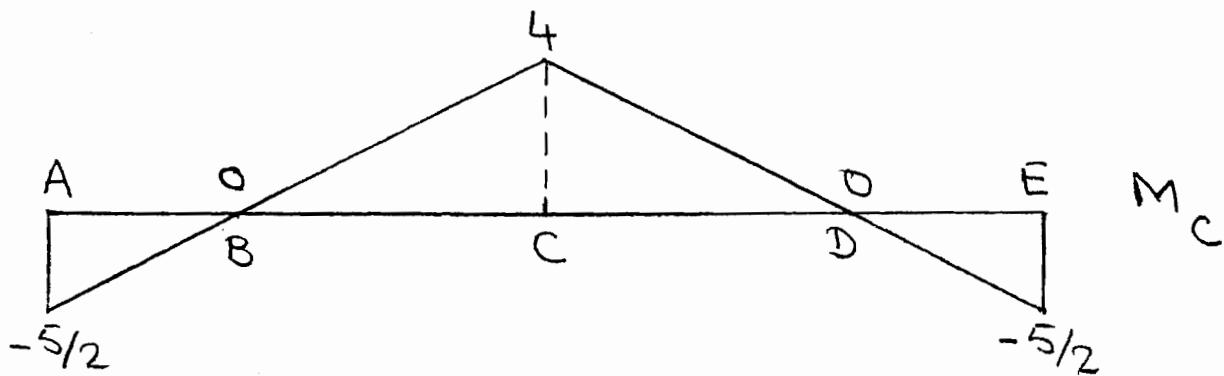
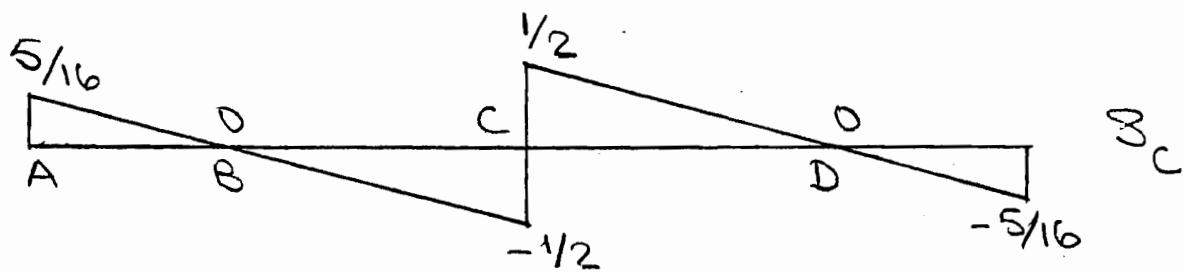
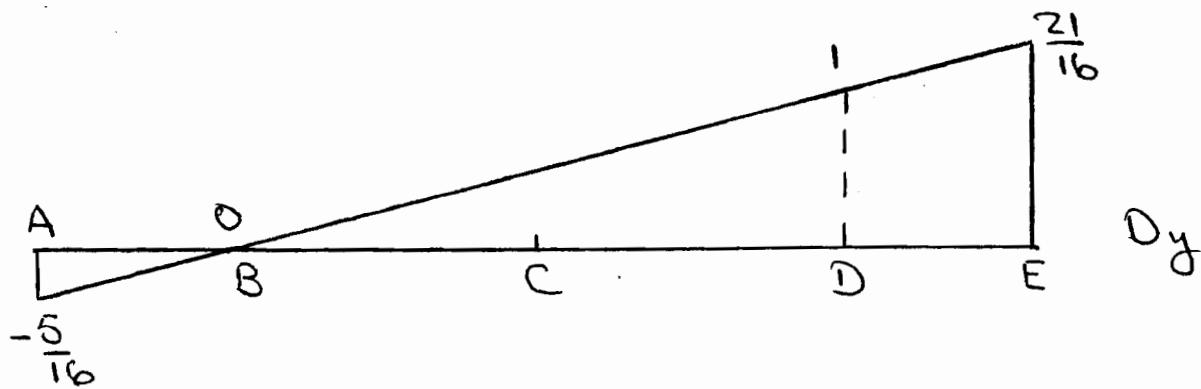
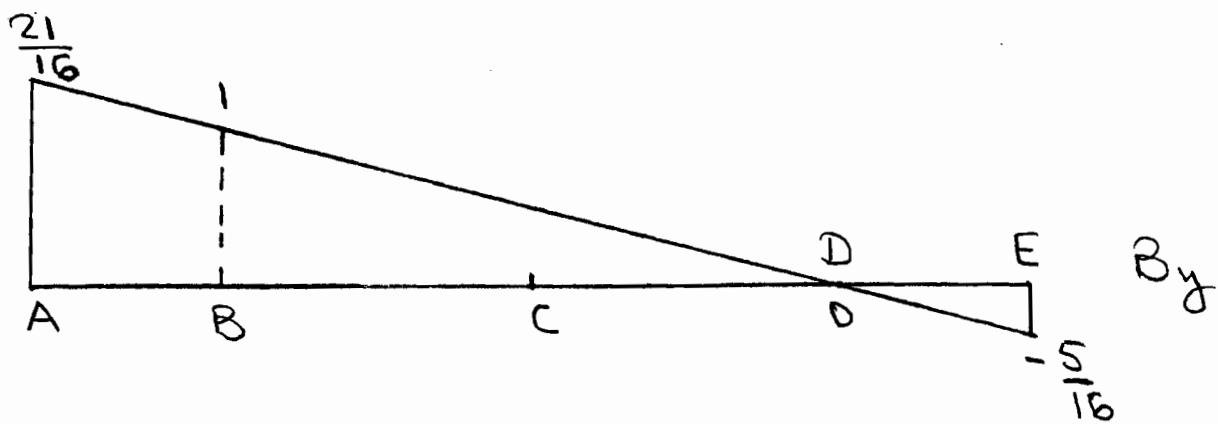


S_c

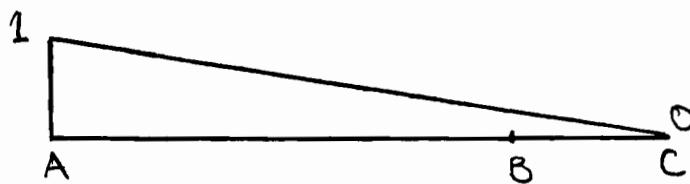


M_c

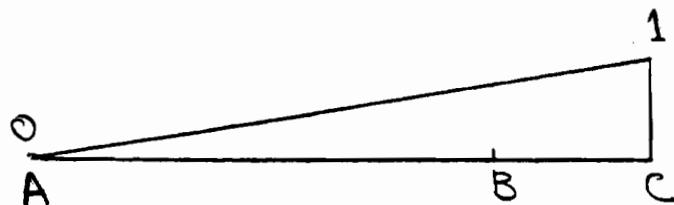
B.6



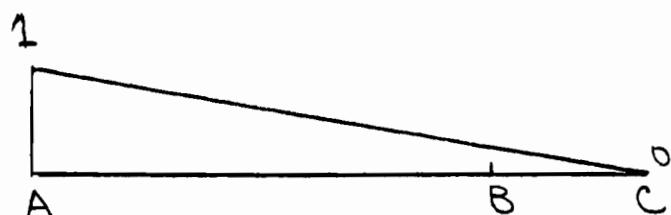
8.7



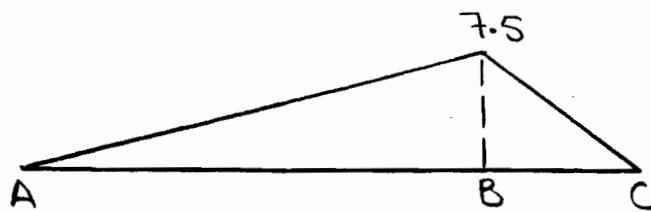
A_y



C_y

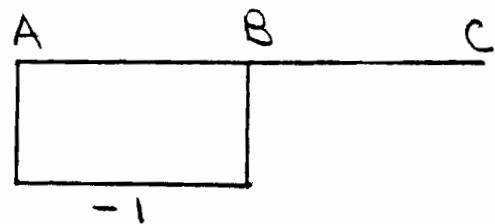


$S_{A,R}$

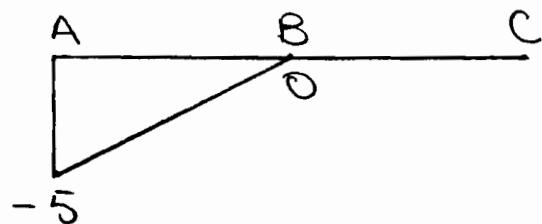


M_B

8.8

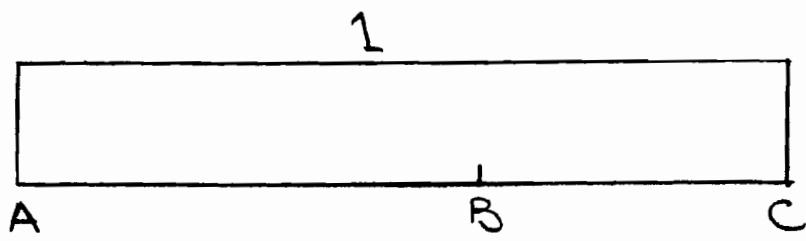


s_B

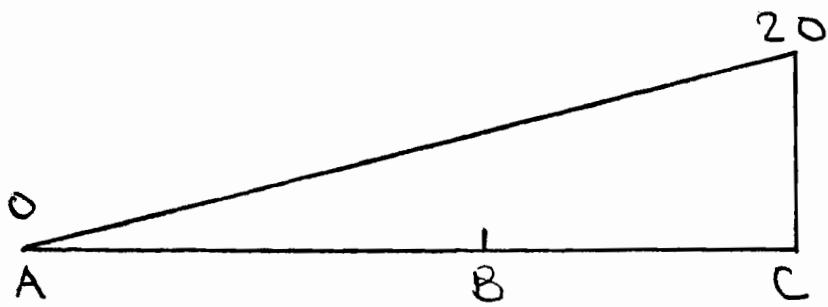


m_B

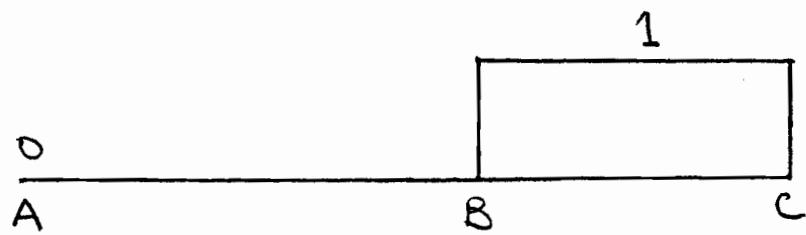
8.9



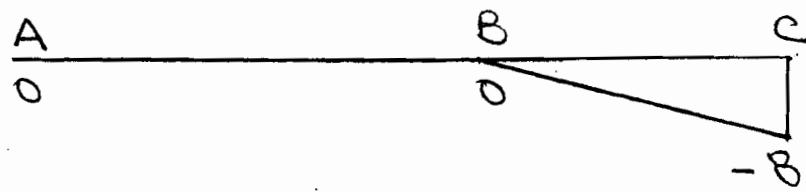
A_y



$M_A (+\zeta)$

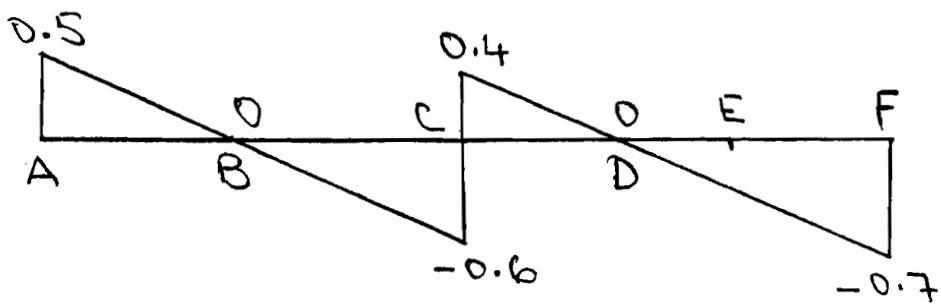


S_B

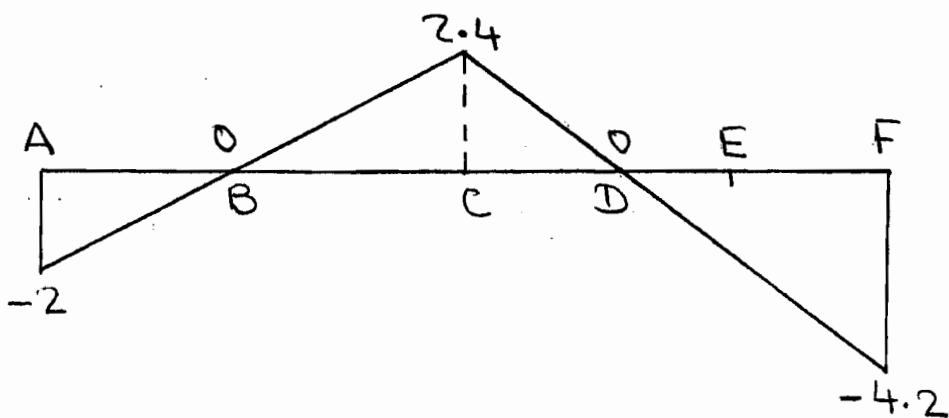


M_B

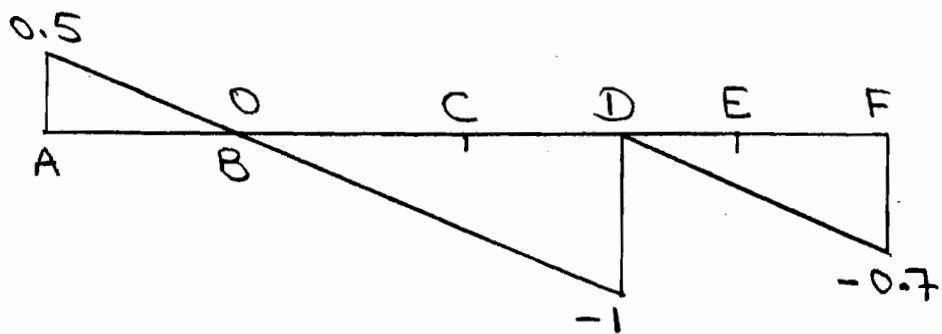
8.10



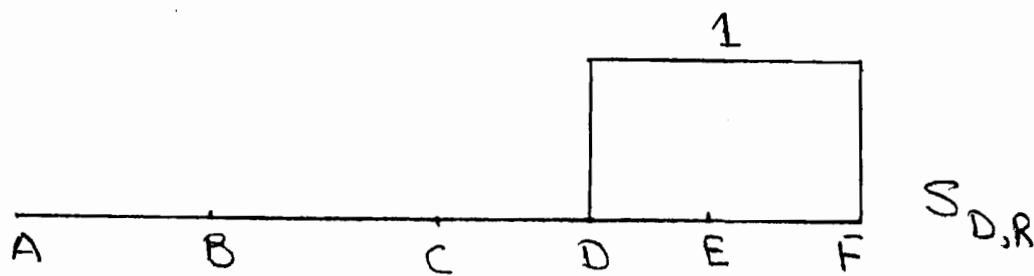
S_C



M_C

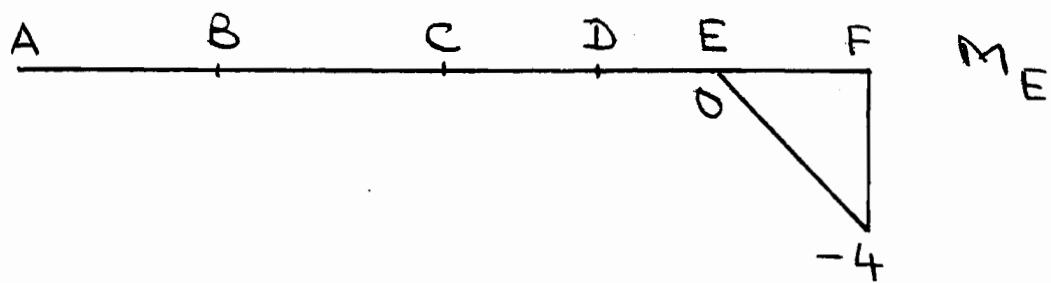
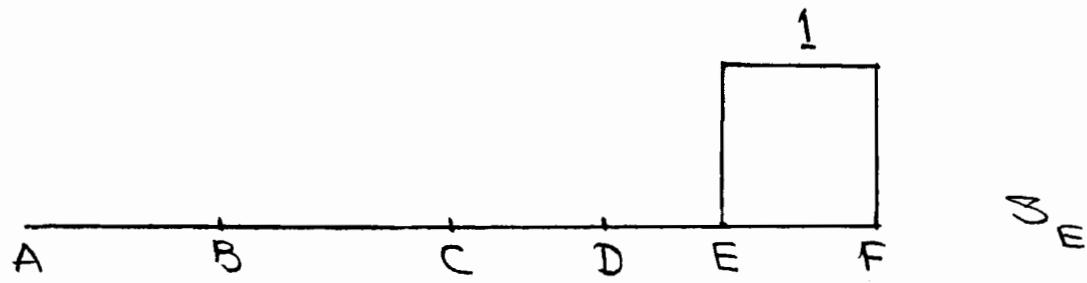


$S_{D,L}$

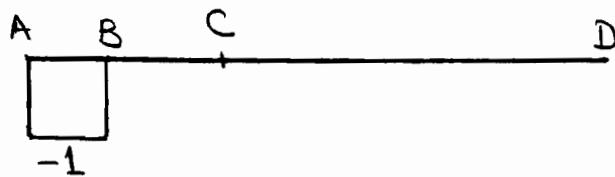


$S_{D,R}$

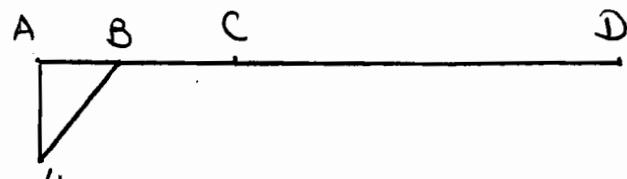
8.11



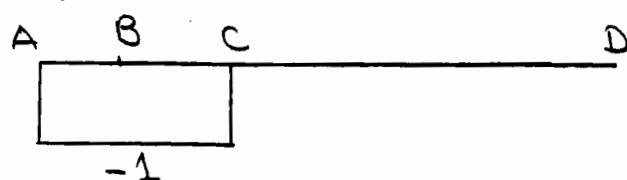
B.12



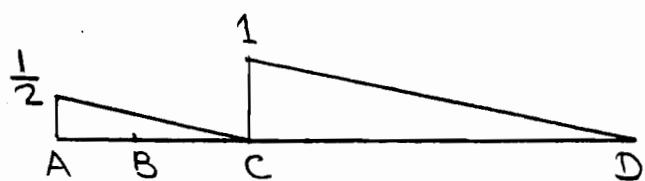
s_B



M_B

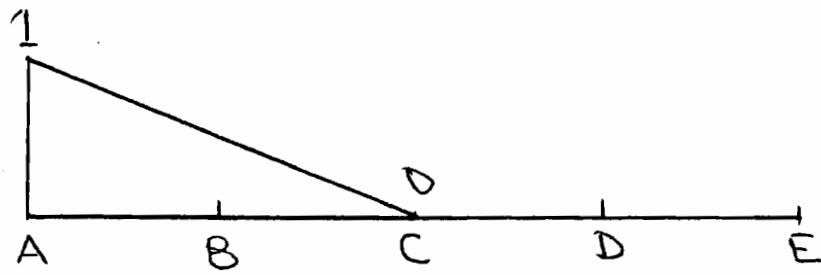


$s_{C,L}$

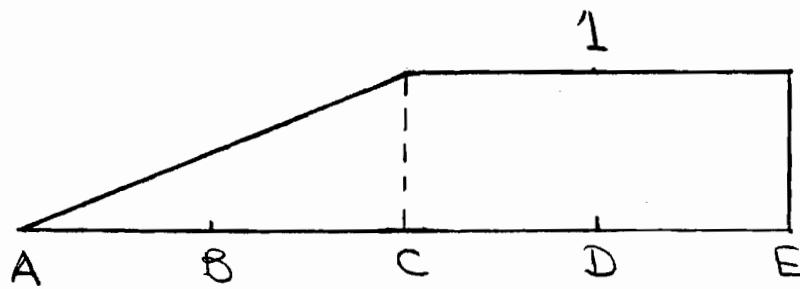


$s_{C,R}$

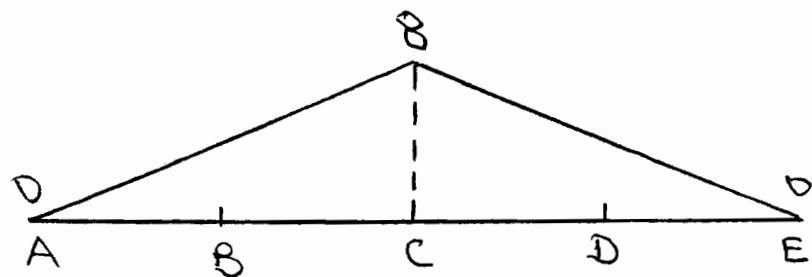
8.13



A_y

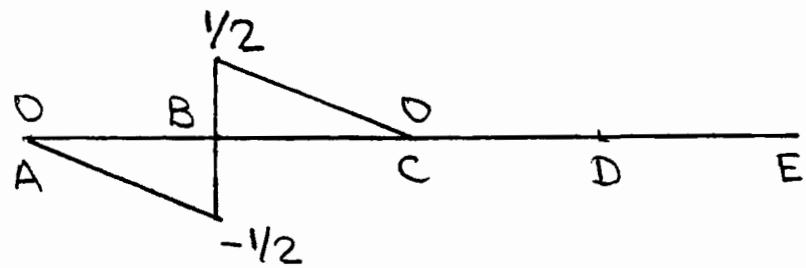


E_y

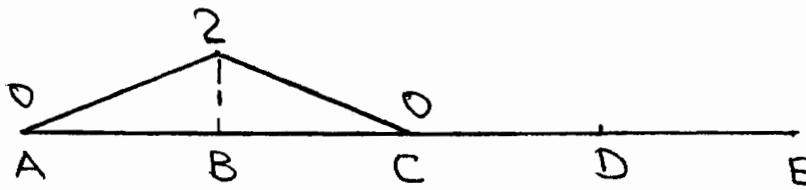


$M_E (+)$

8.14

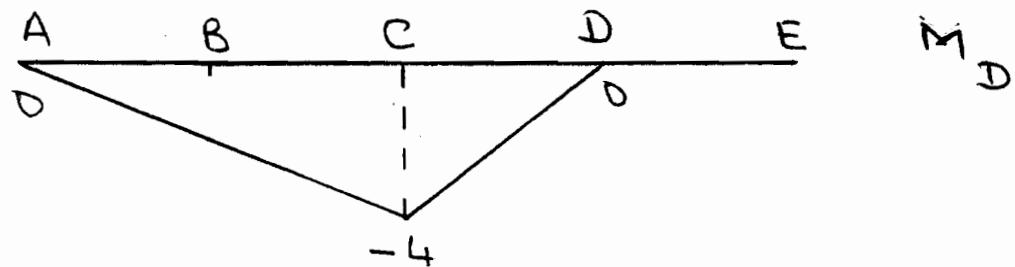
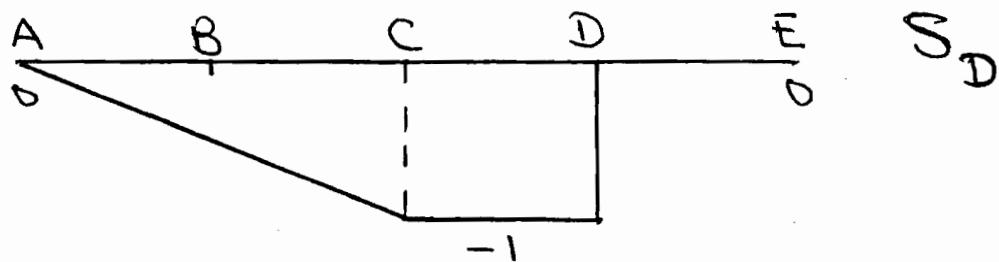


S_B

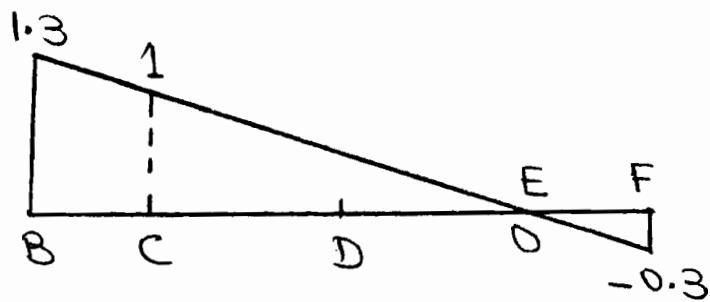


M_B

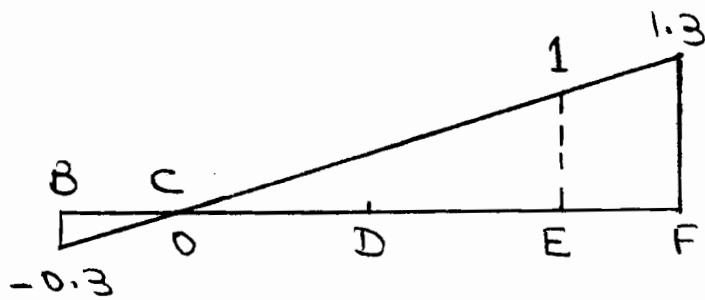
8.15



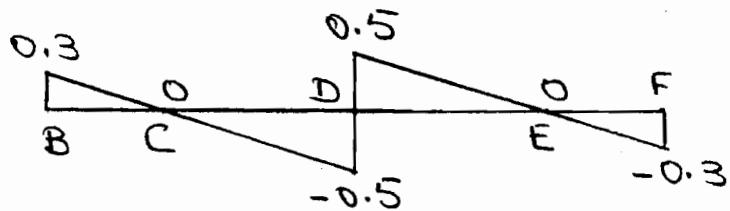
8.16



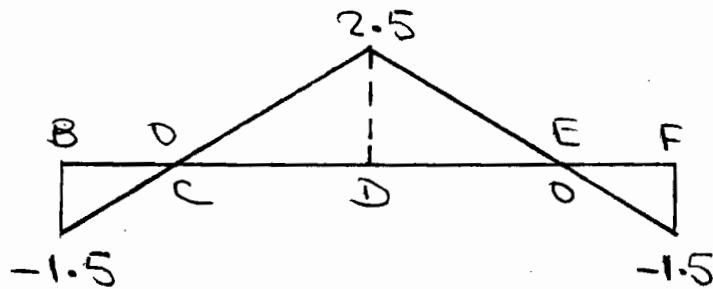
A_y



E_y

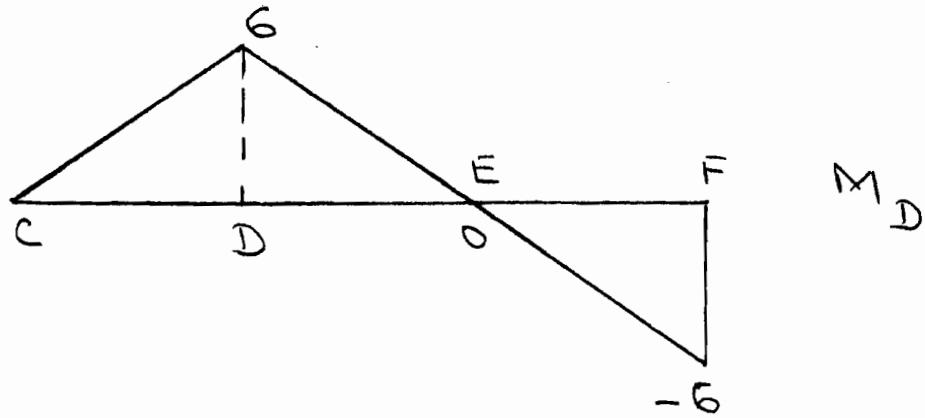
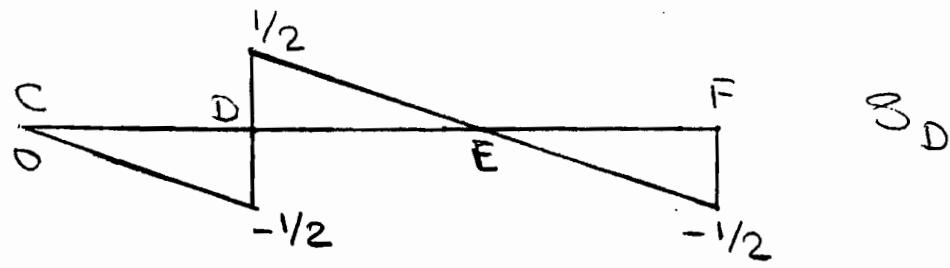
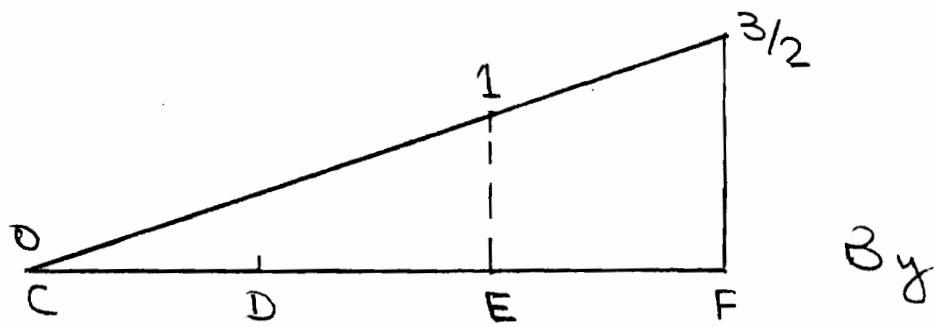
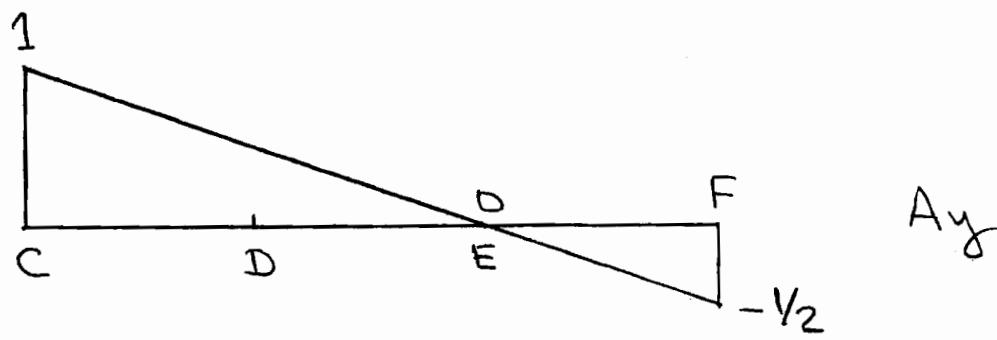


S_D



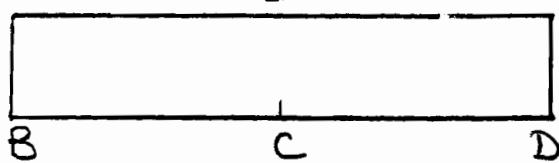
M_D

8.17

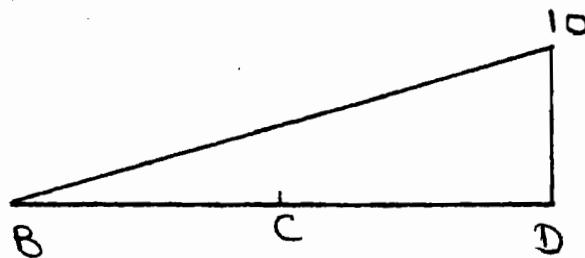


8.18

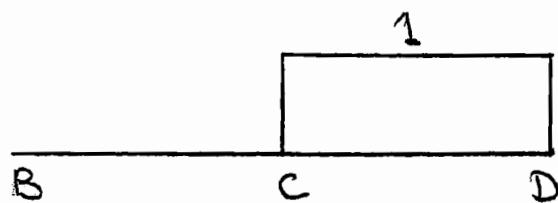
1



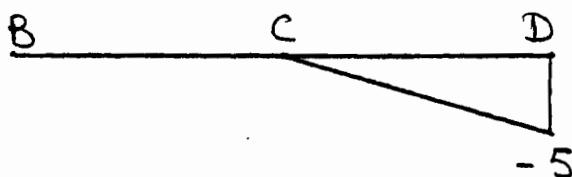
A_y



$M_A (+\curvearrowright)$

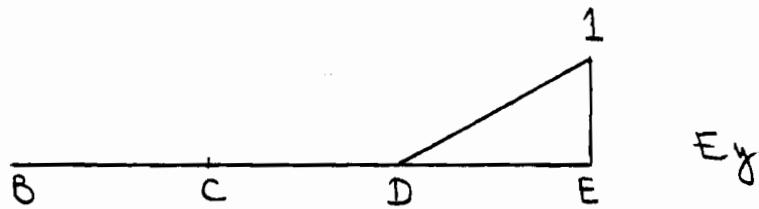


S_c

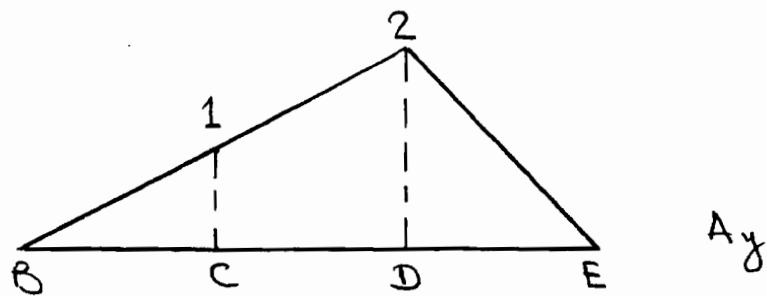


M_c

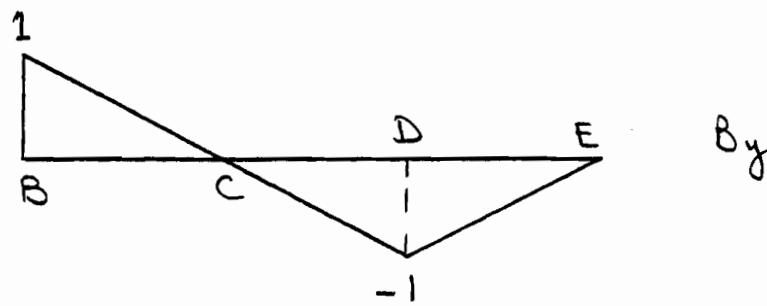
8.19



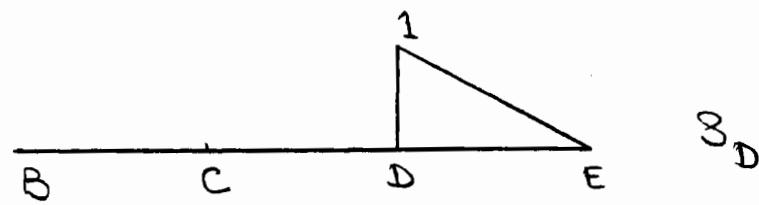
E_y



A_y

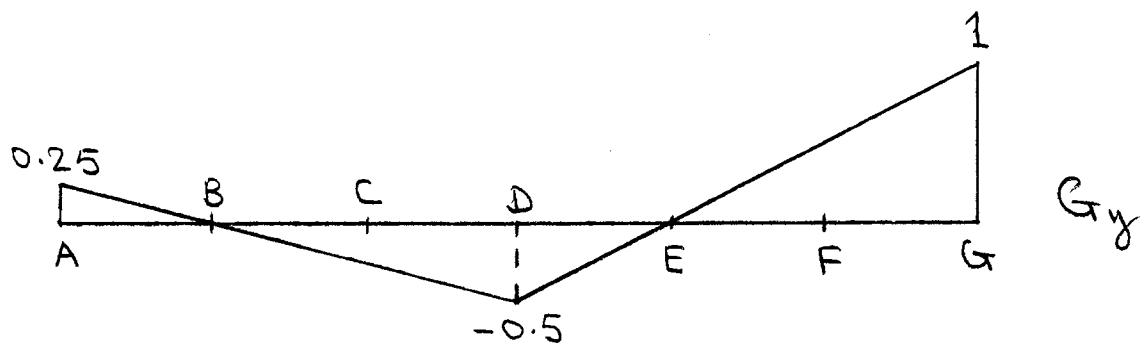
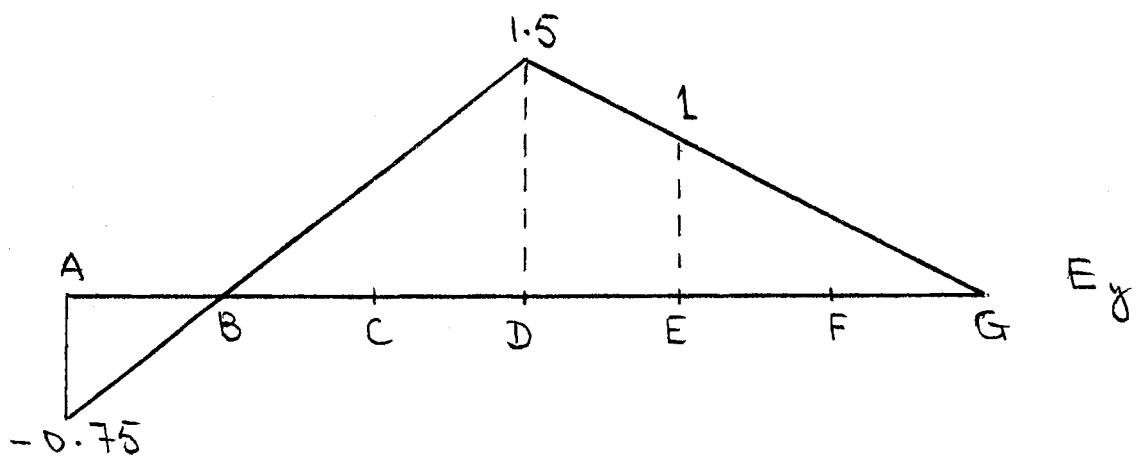
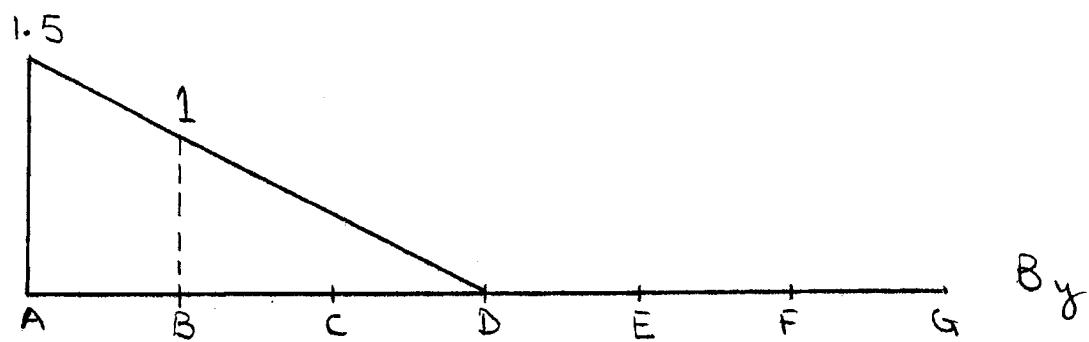


B_y

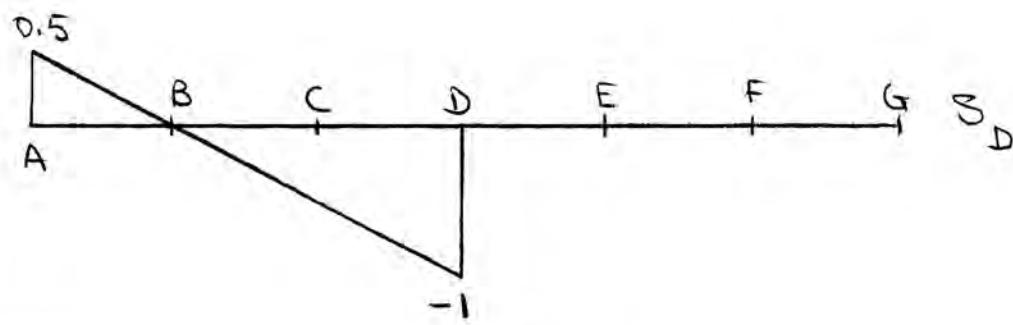
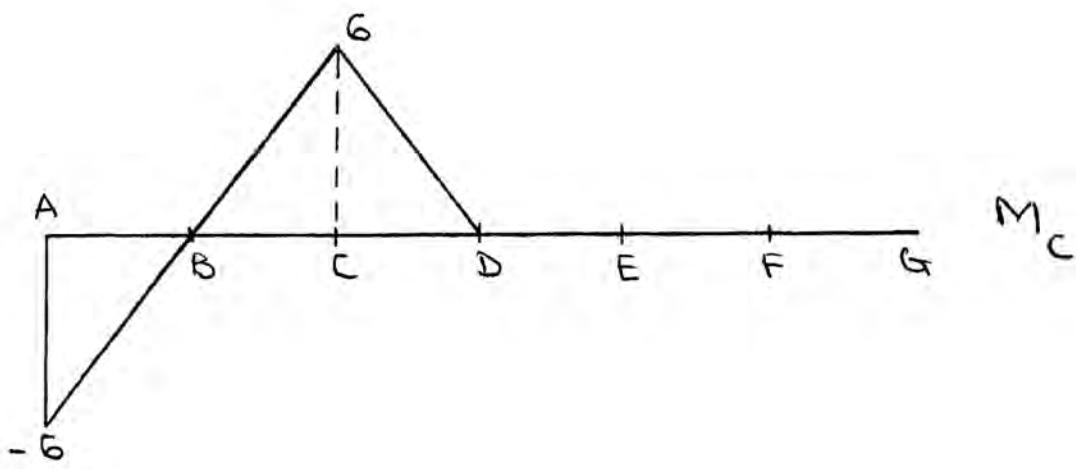
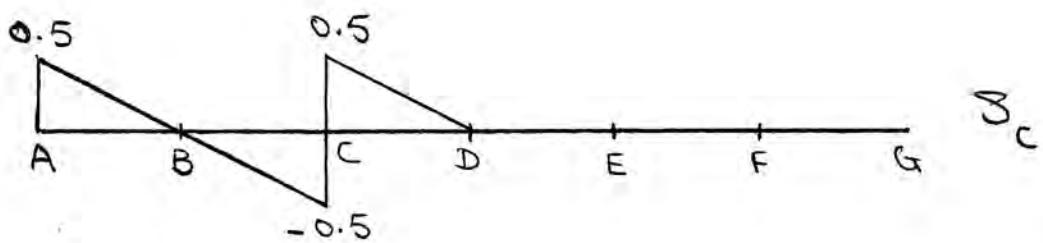


S_D

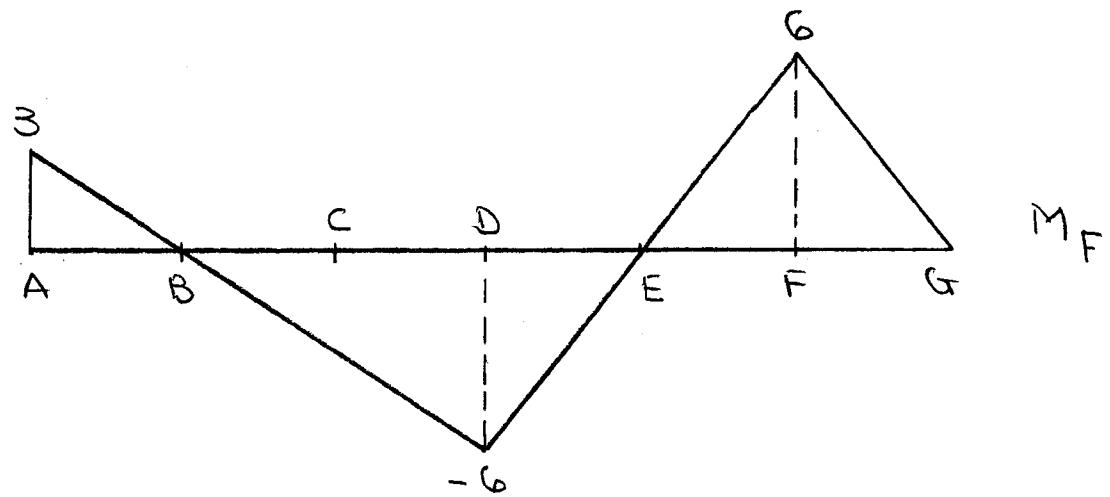
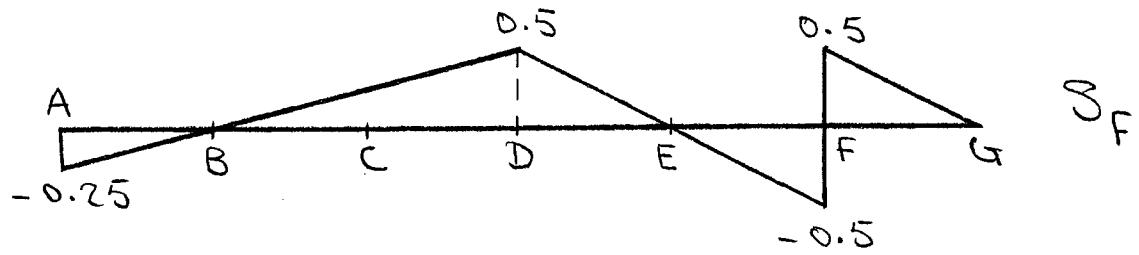
8.20



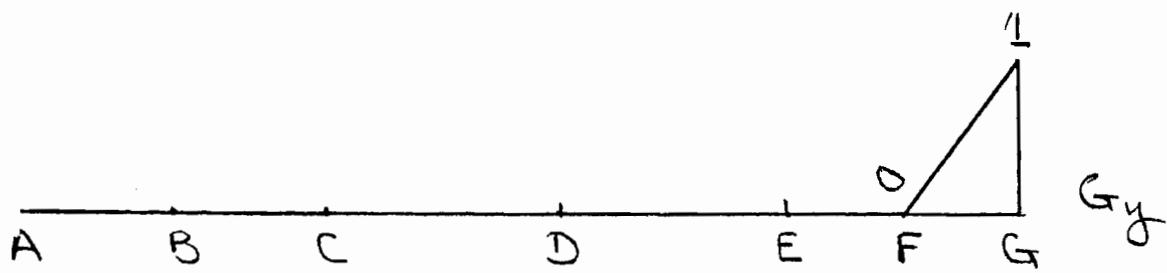
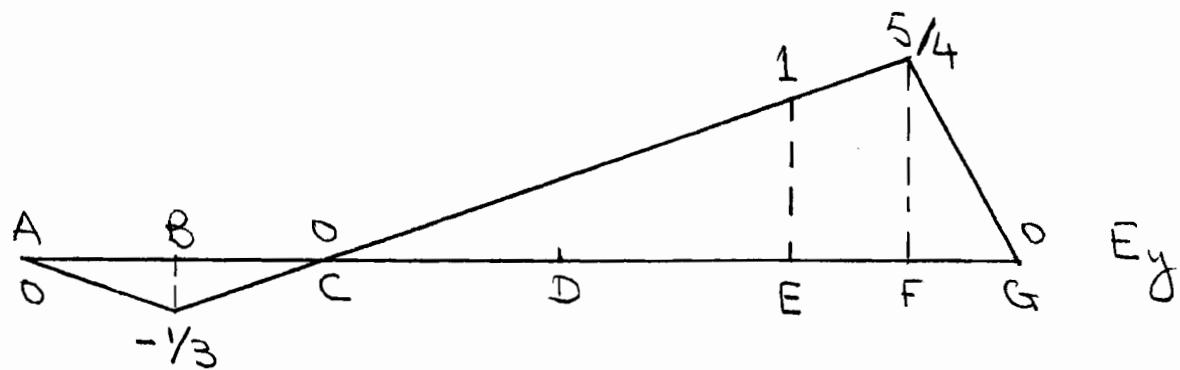
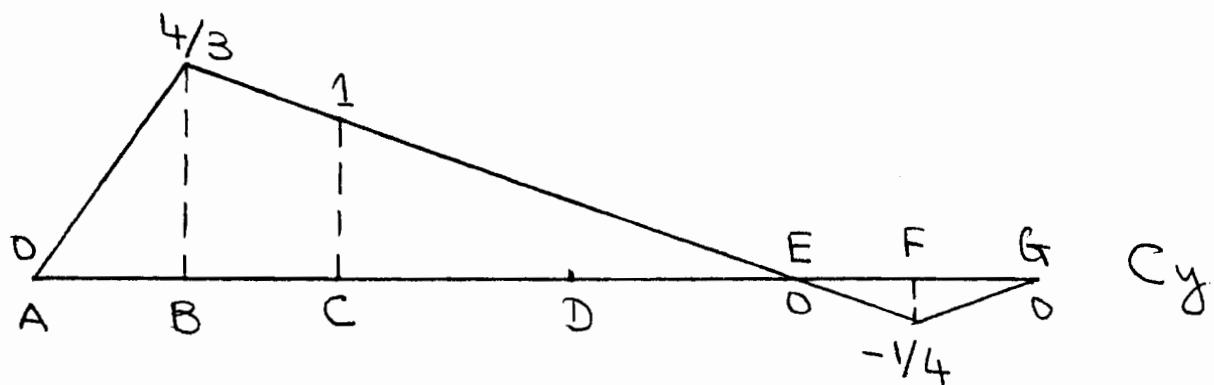
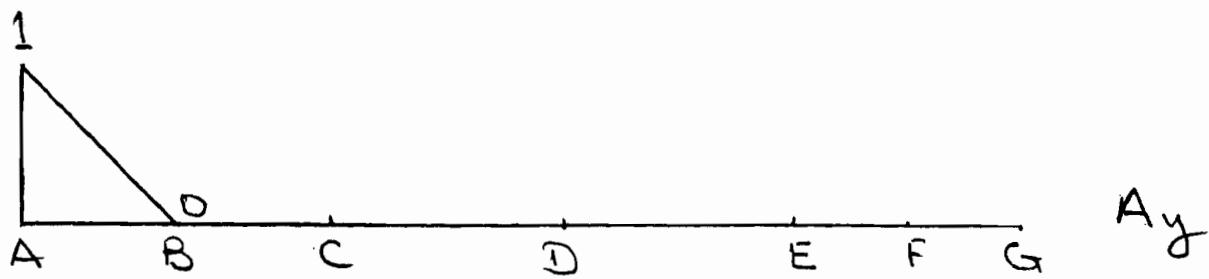
8.21



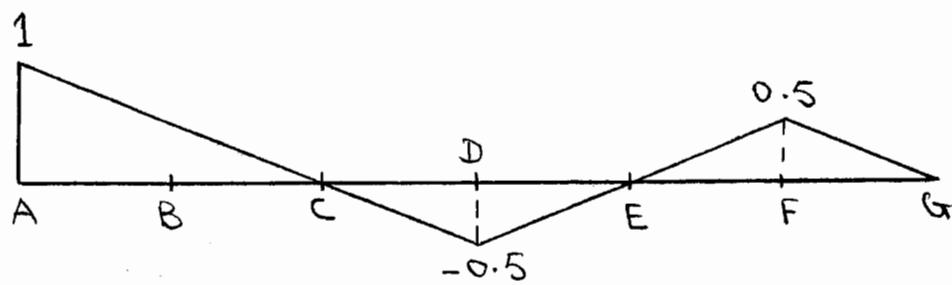
8.22



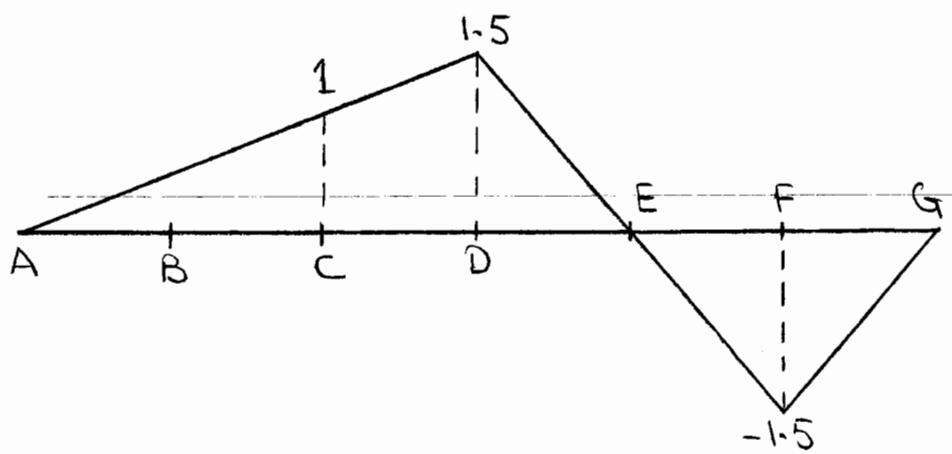
8.23



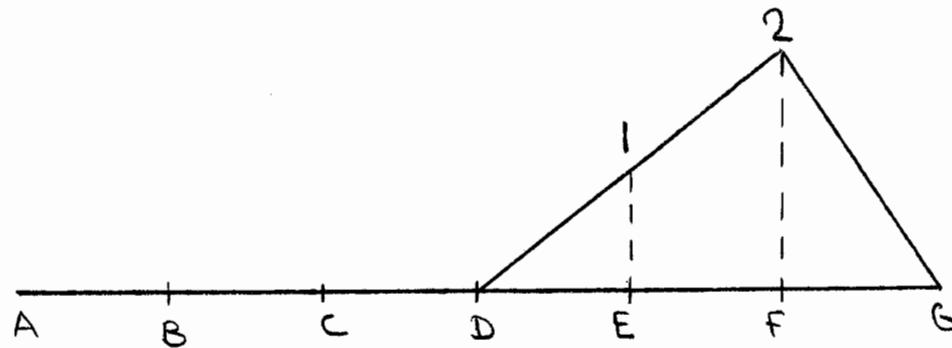
8.24



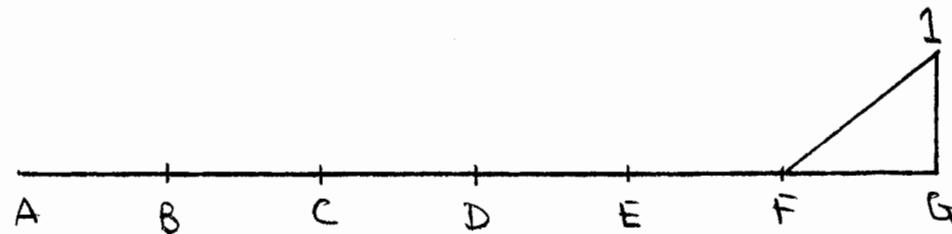
A_y



C_y

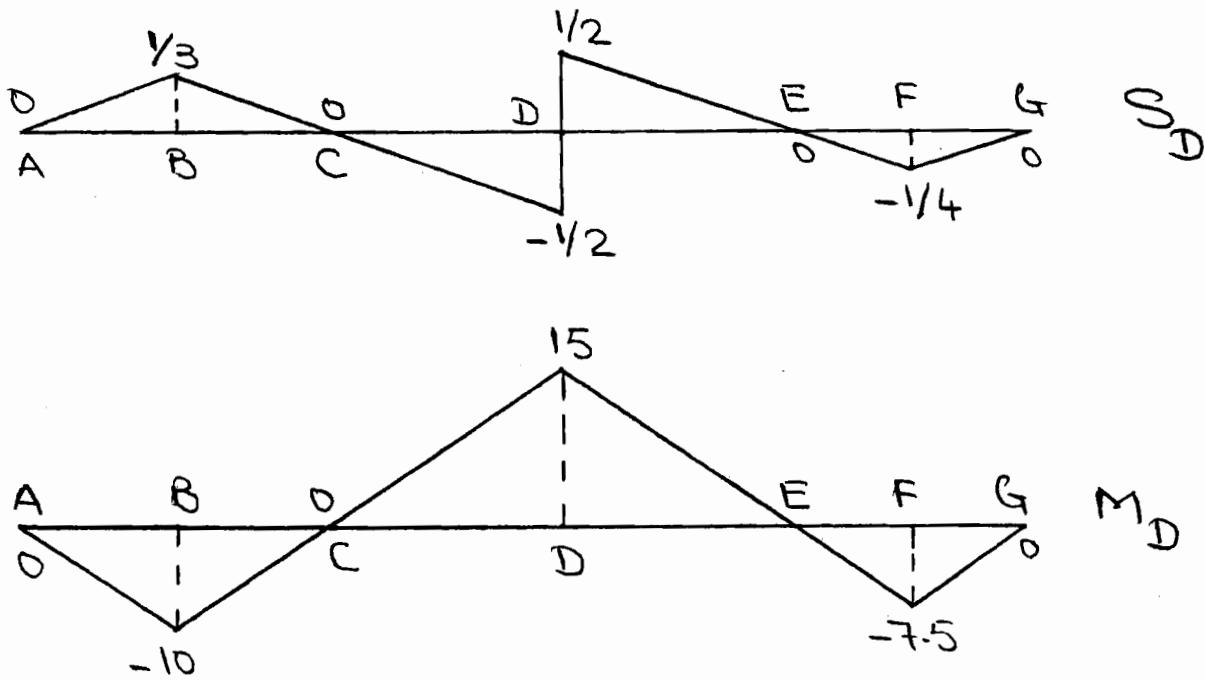


E_y

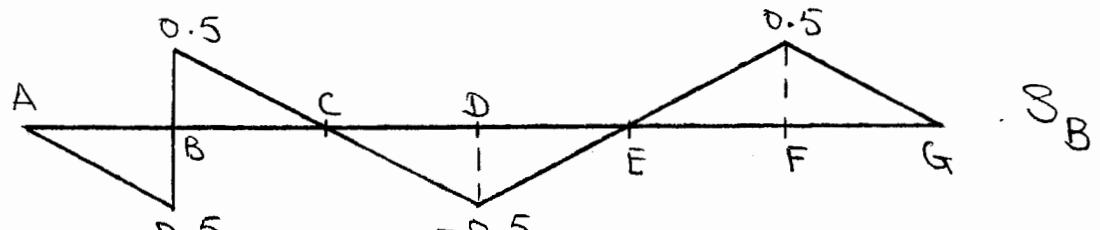


G_y

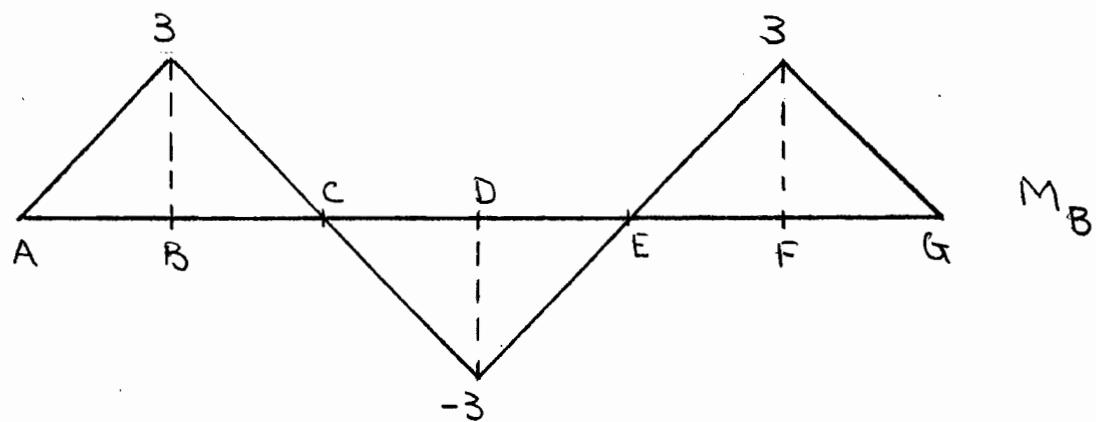
8.25



8.26

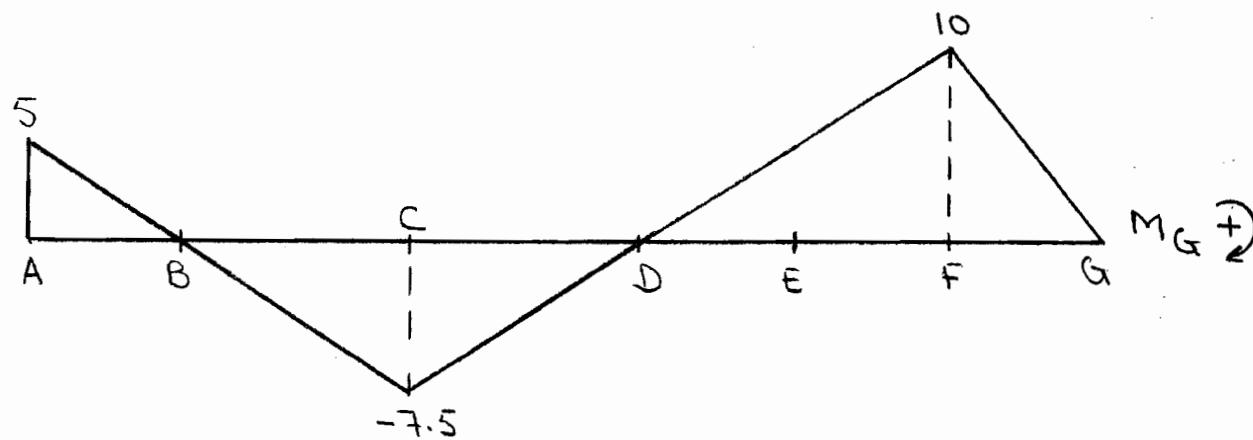
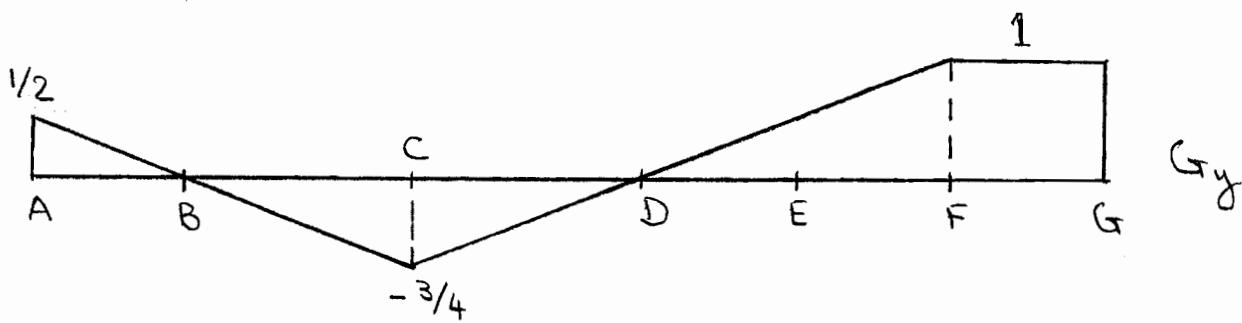
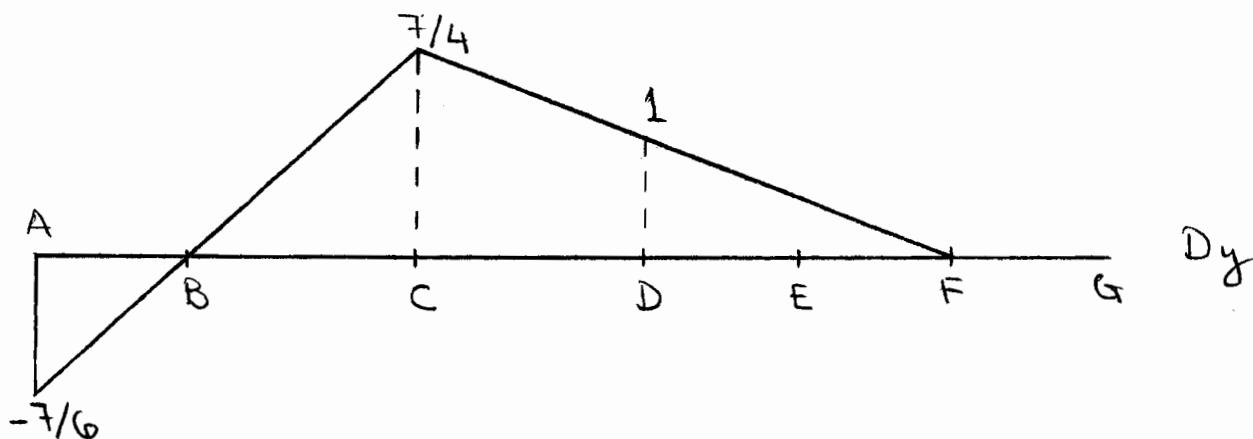
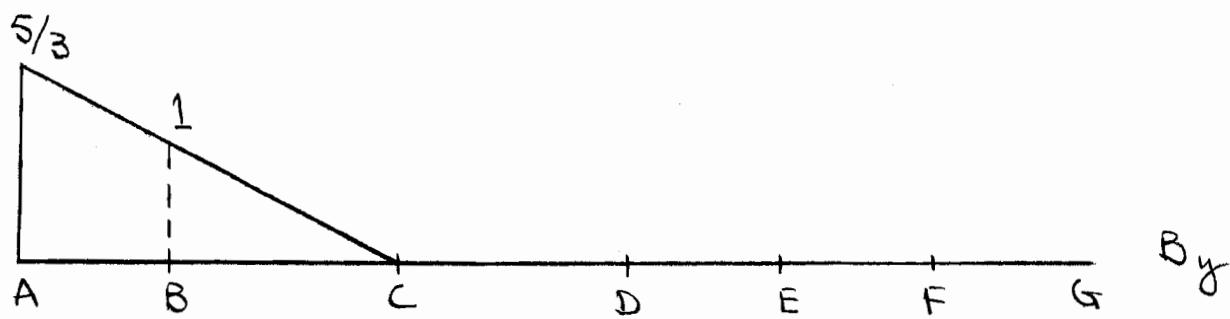


δ_B

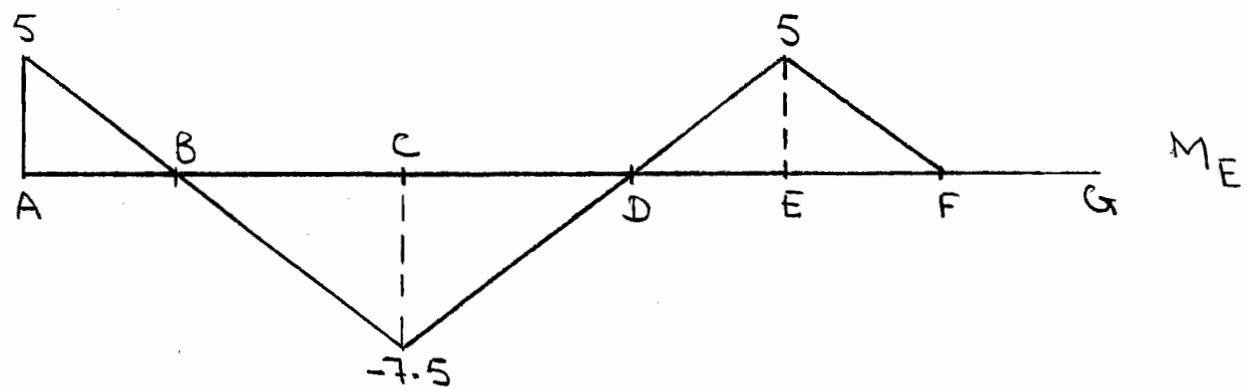
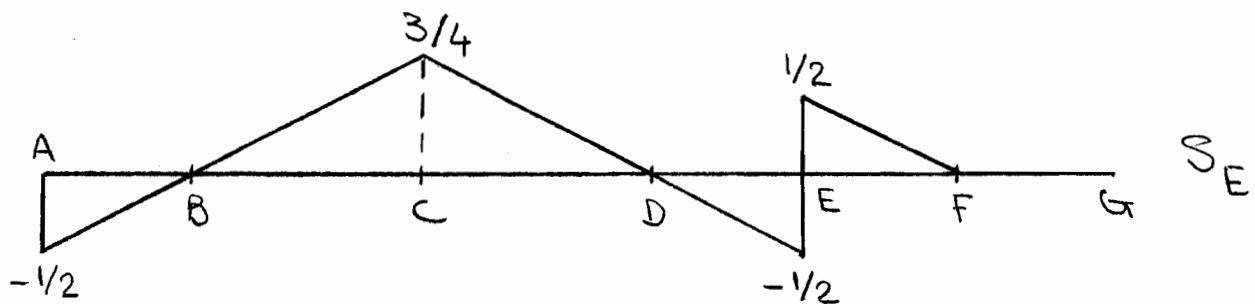


M_B

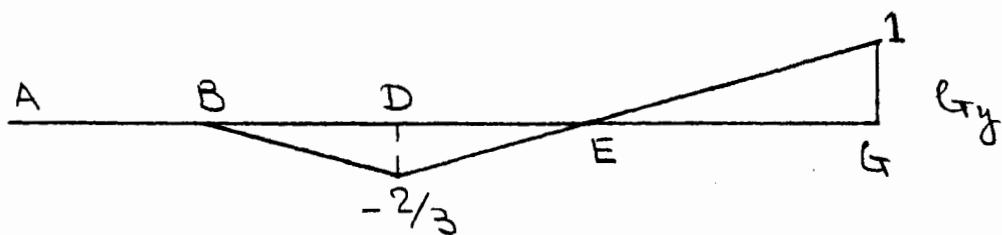
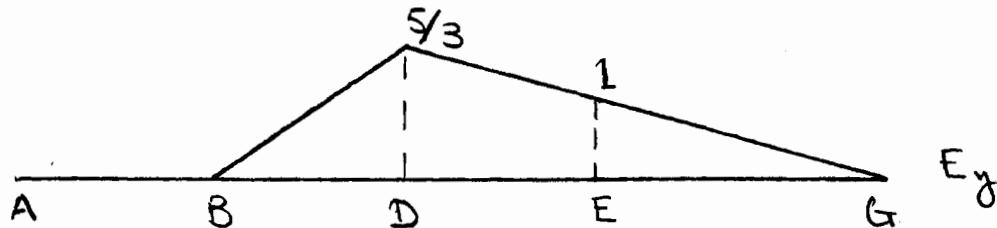
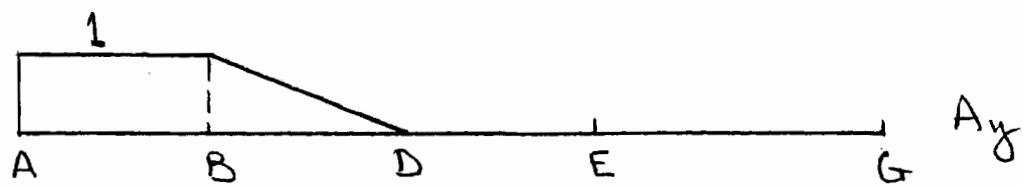
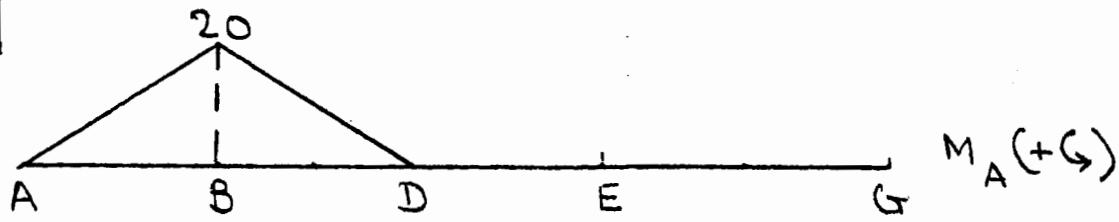
8.27



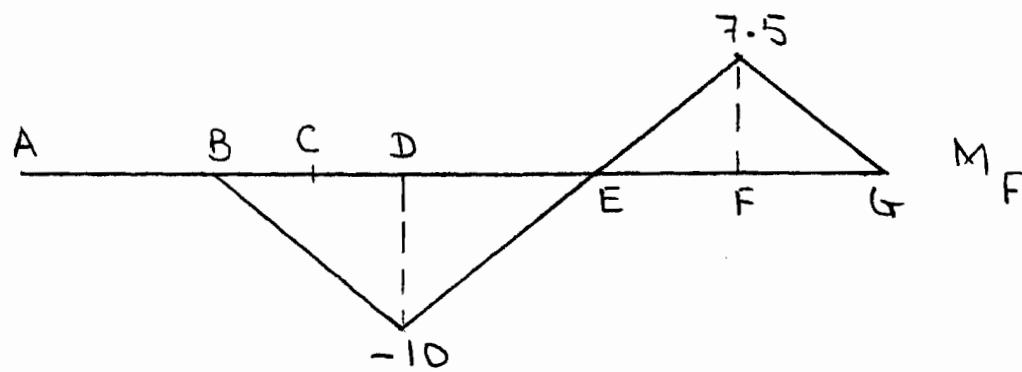
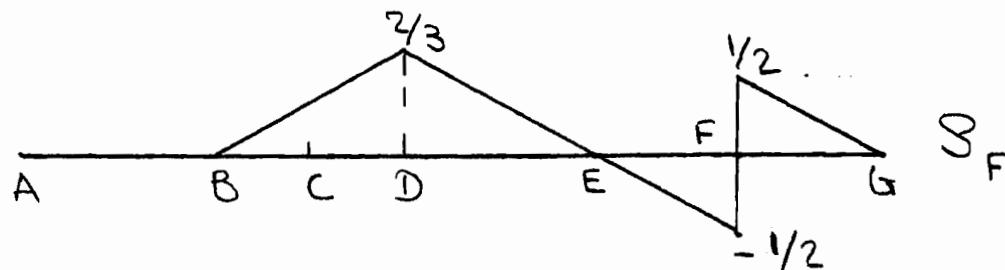
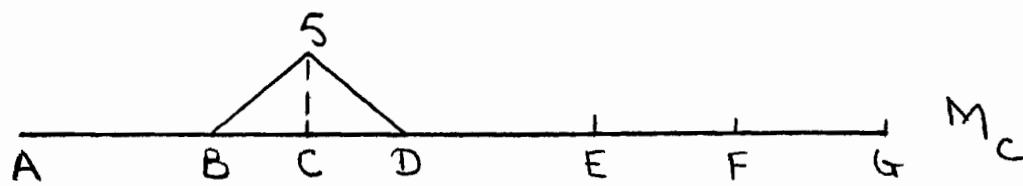
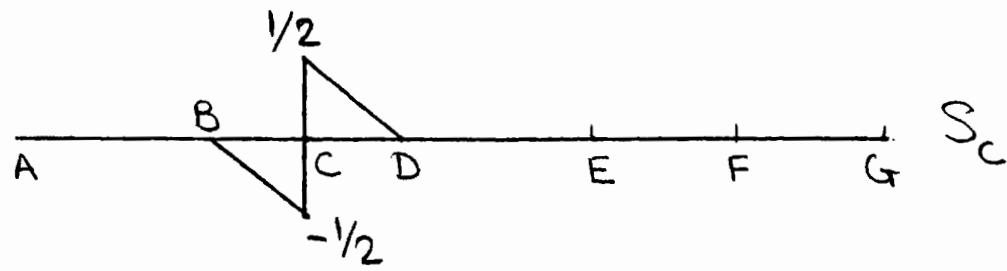
8.28



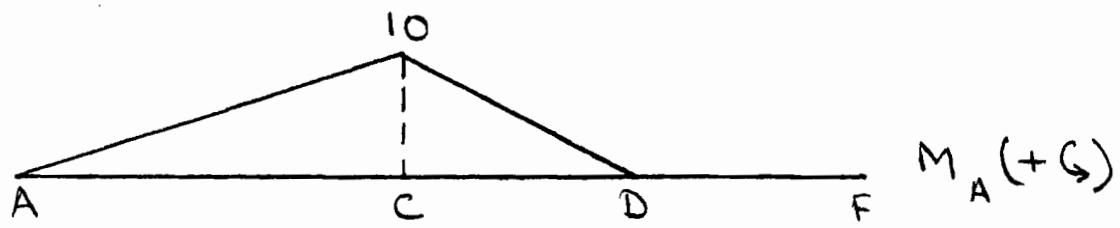
8-29



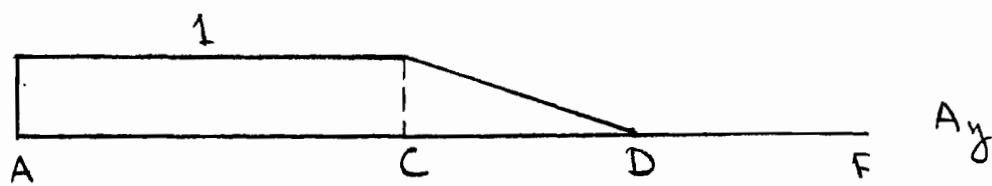
8.30



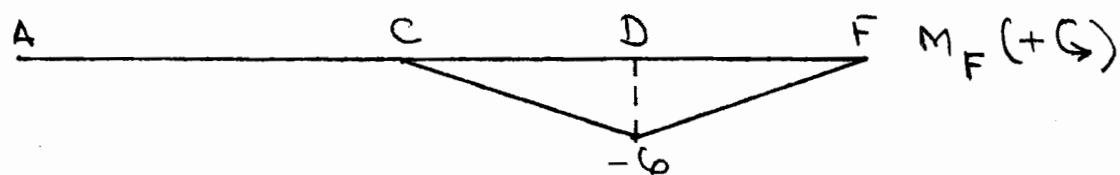
8.31



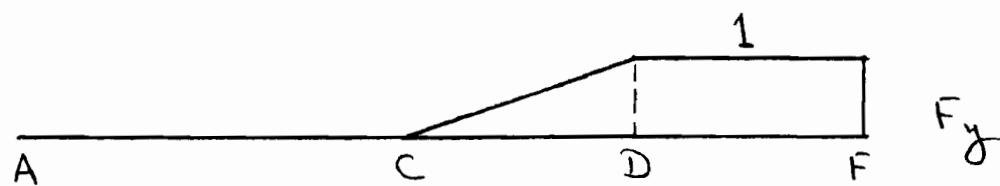
$M_A (+\leftarrow)$



A_y

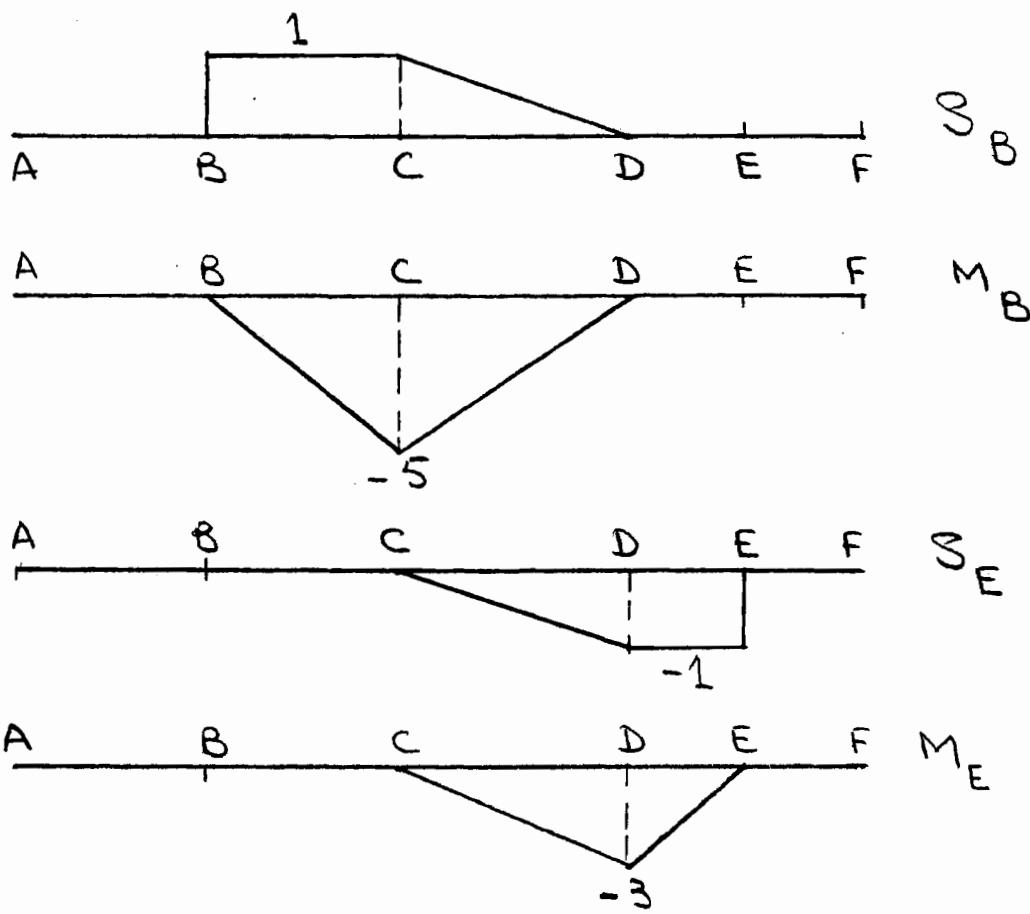


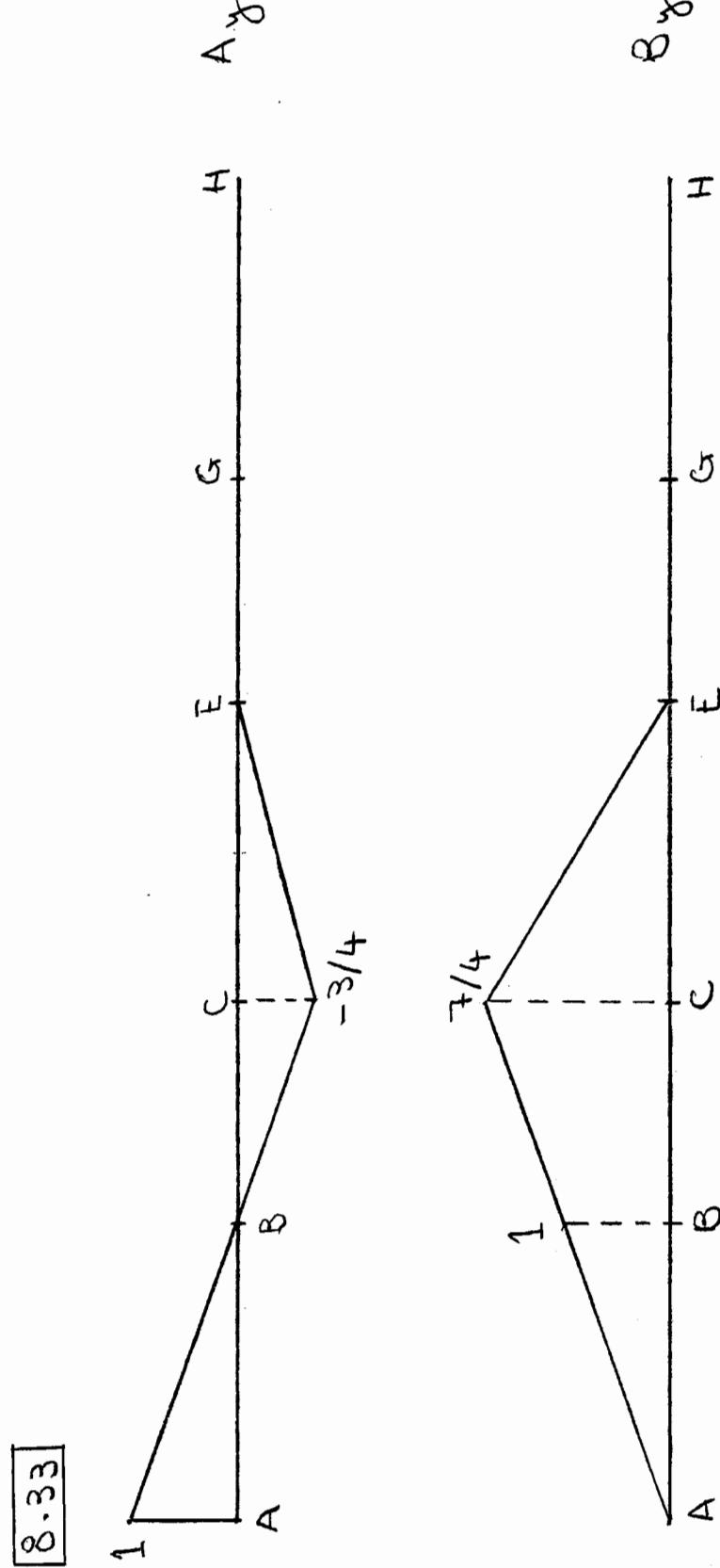
$M_F (+\leftarrow)$

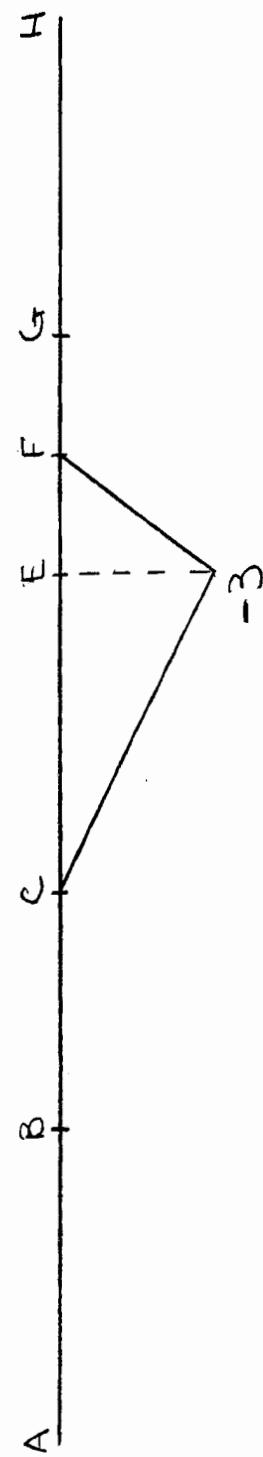
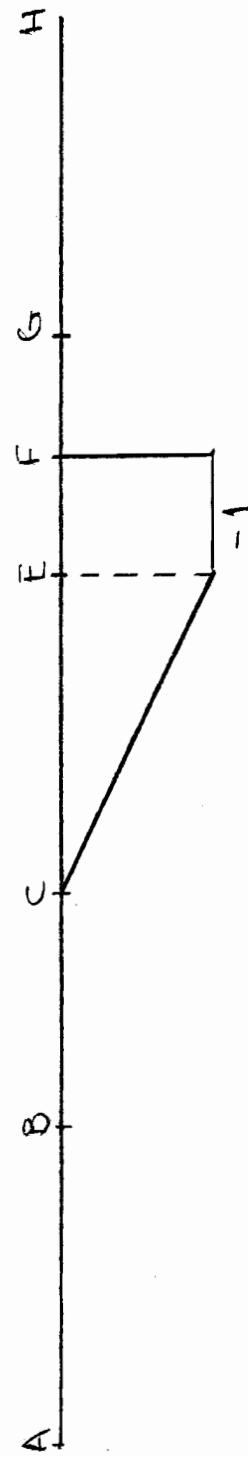
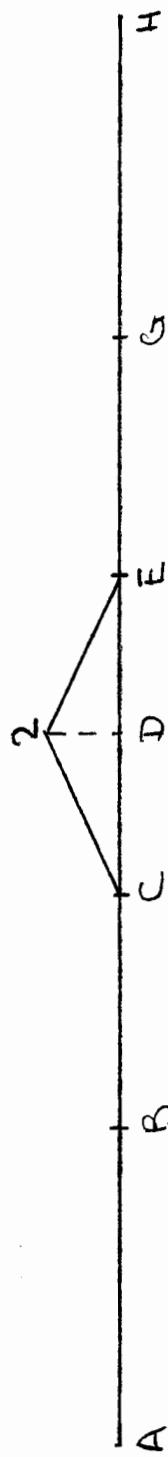
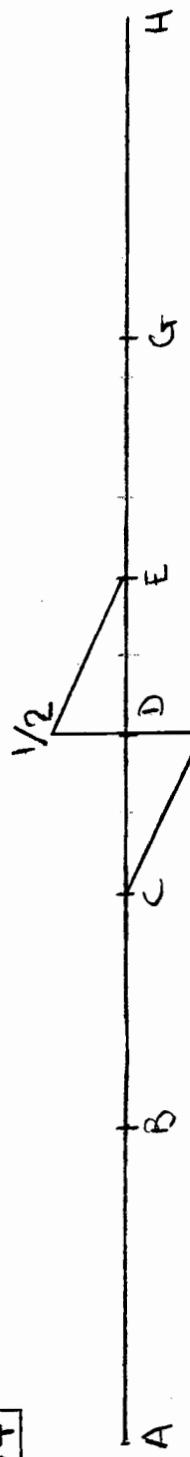


F_y

8.32

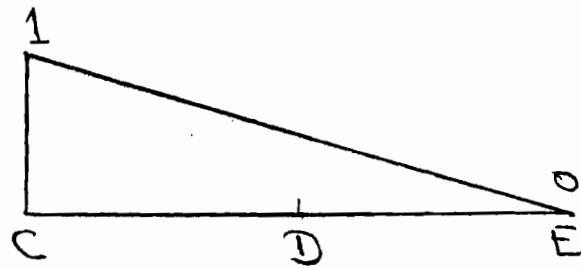




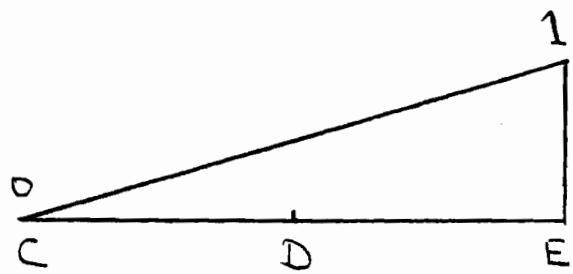
S^D Σ^D S^F Σ^F 

8.34

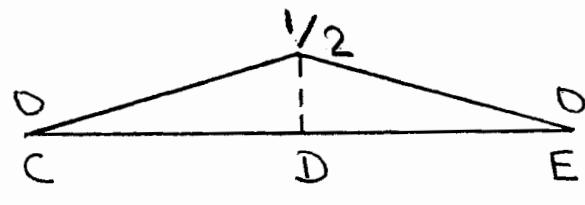
8.35



A_y

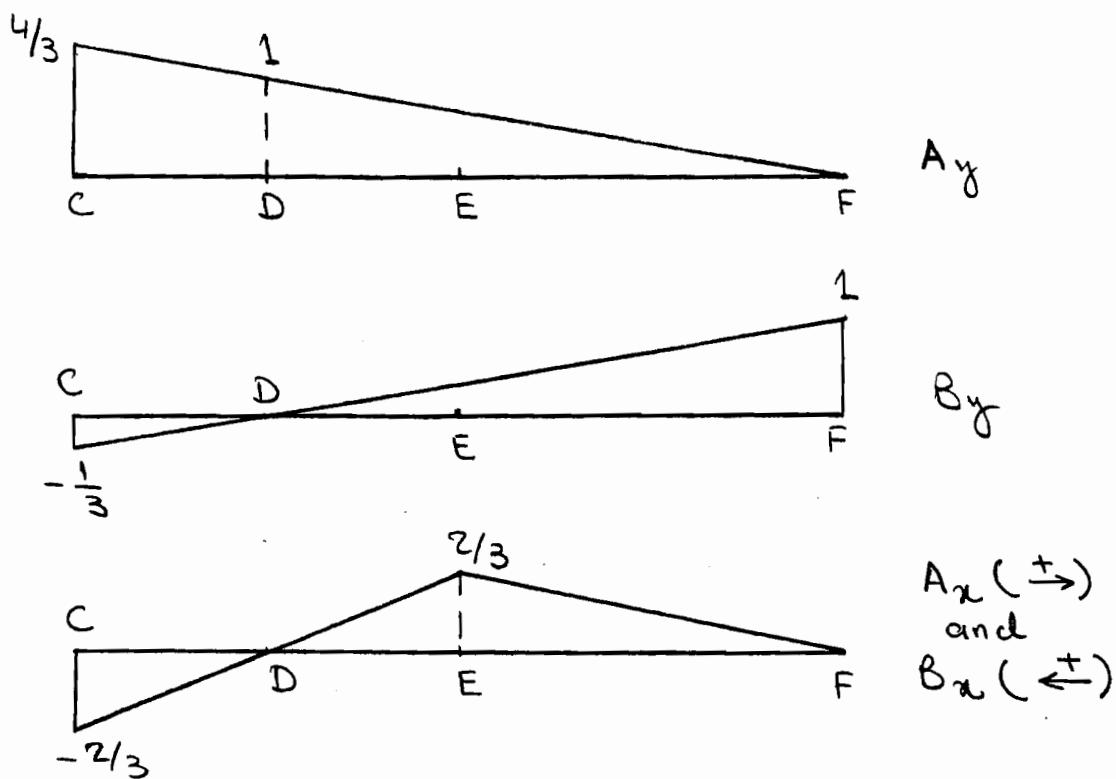


B_y

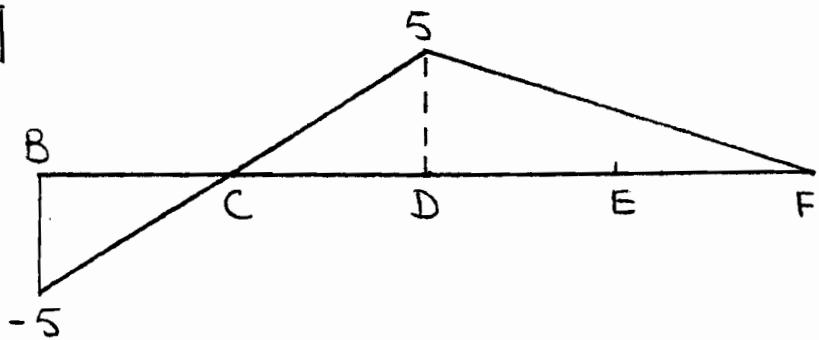


$A_x (+ \rightarrow)$
and
 $B_x (+ \leftarrow)$

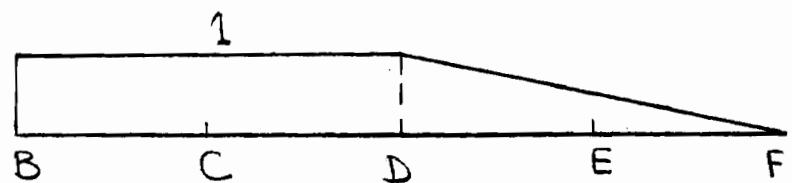
8.36



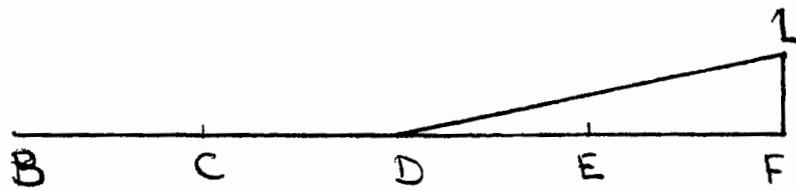
8.37



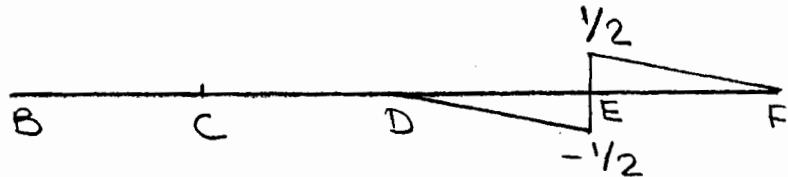
$M_A (+\rightarrow)$



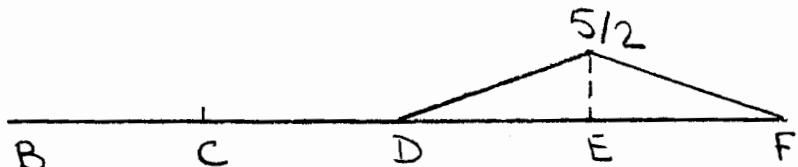
A_y



F_y

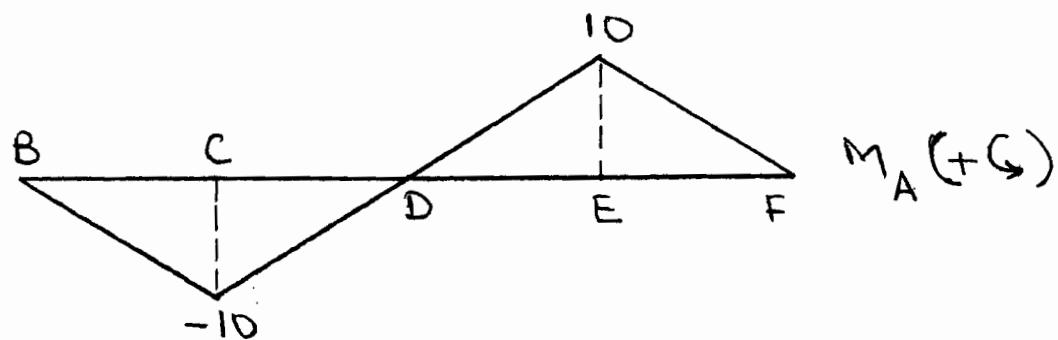


S_E

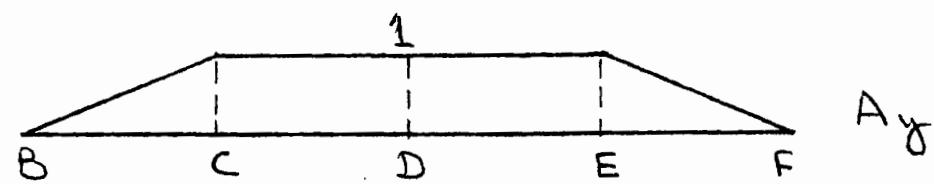


M_E

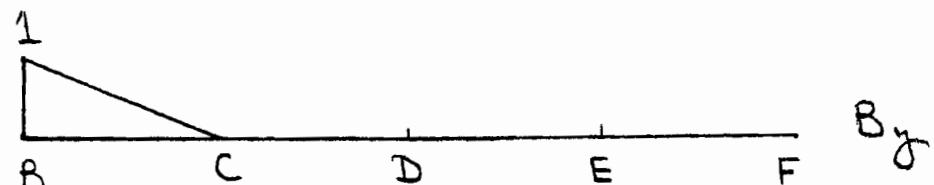
8.38



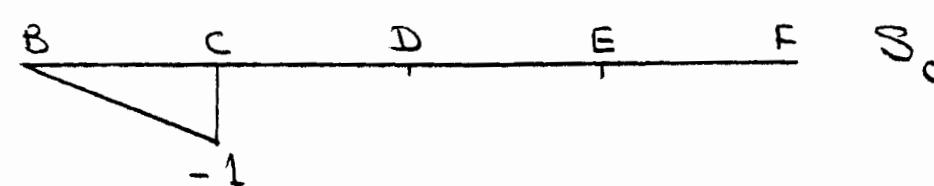
$M_A (+\leftarrow)$



A_y

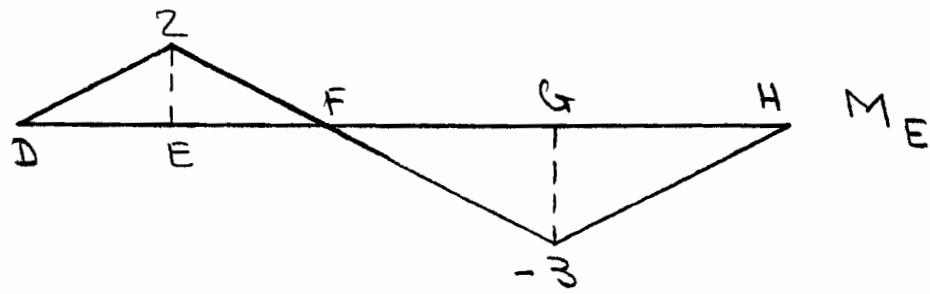
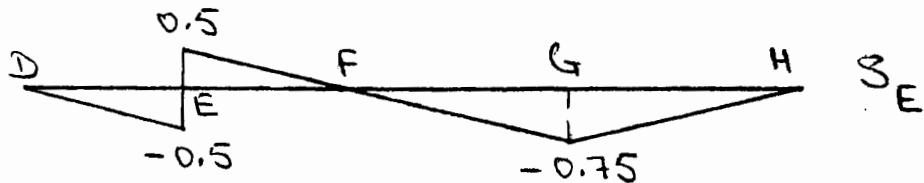
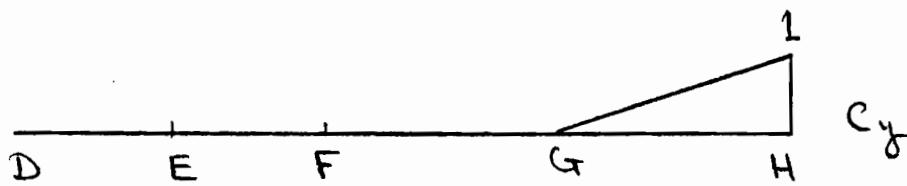
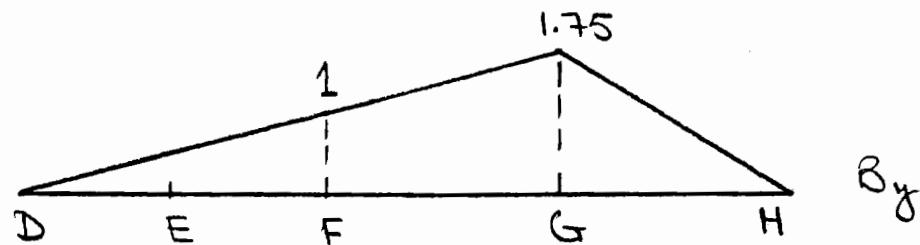
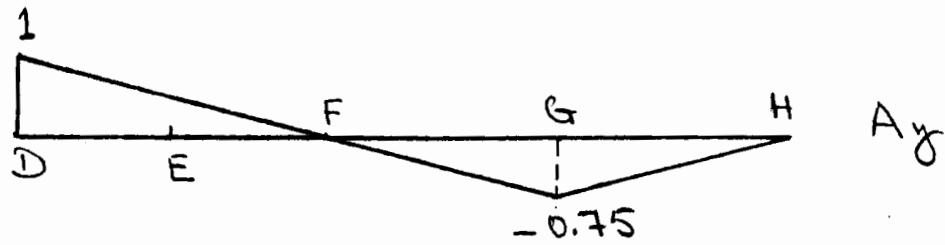


B_y

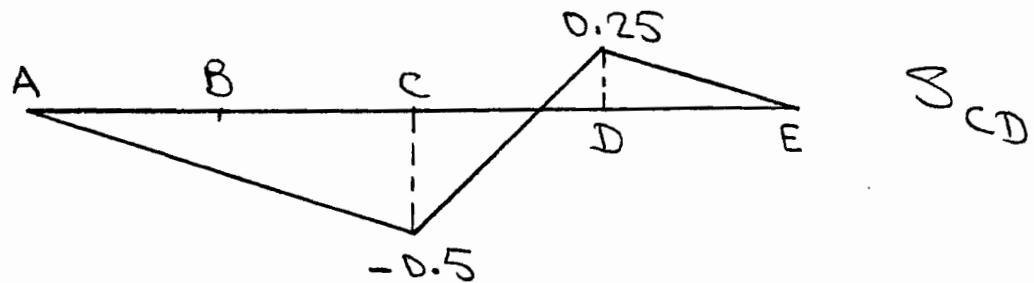


S_c

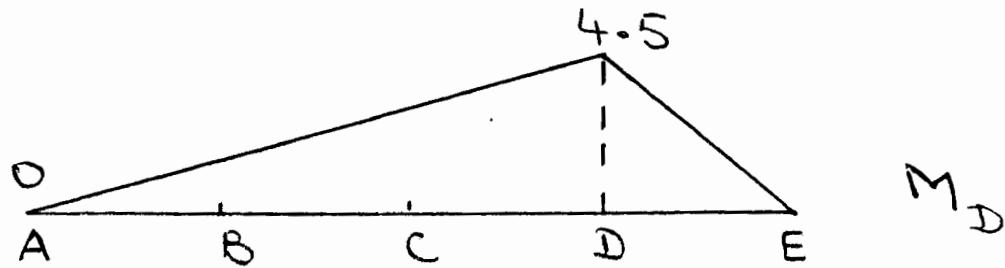
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8.40

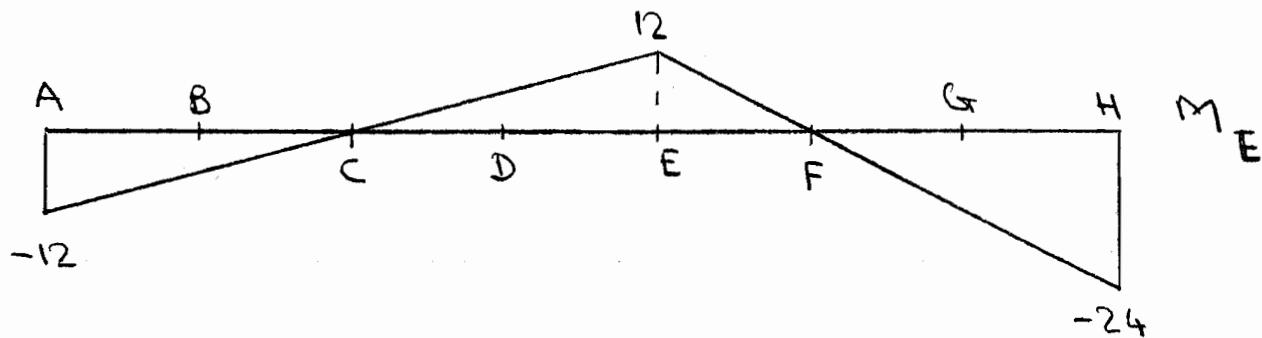
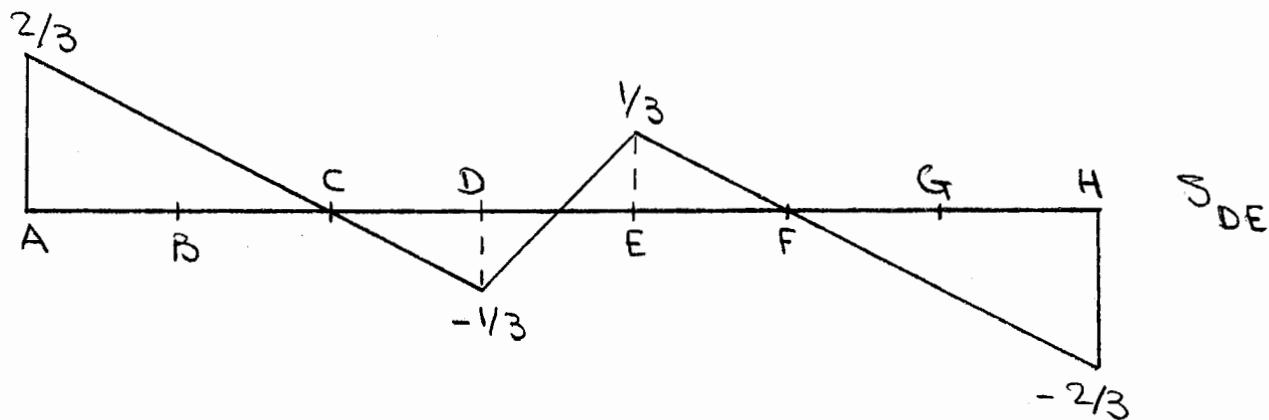


S_{CD}

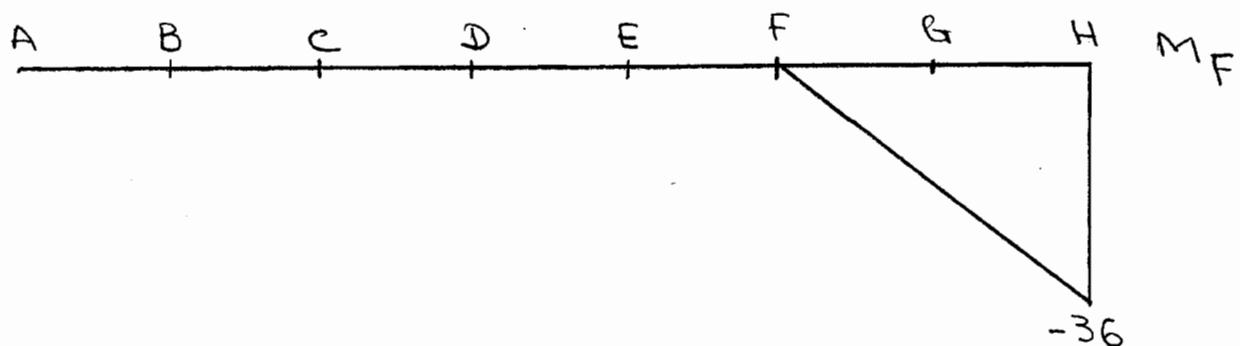
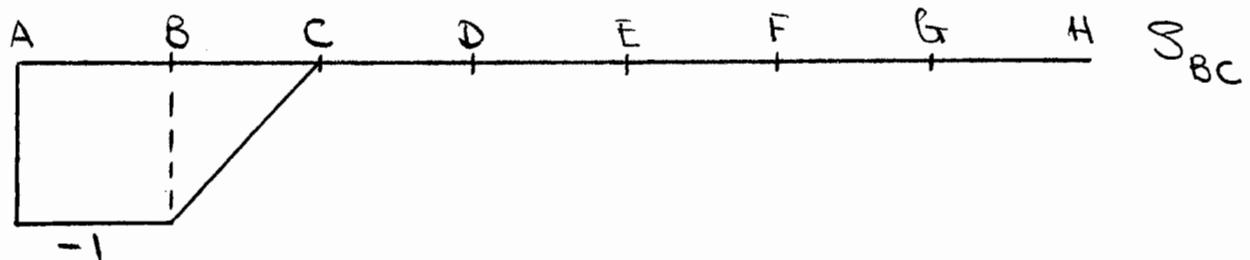


M_D

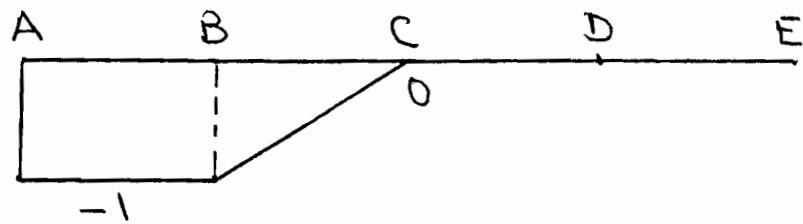
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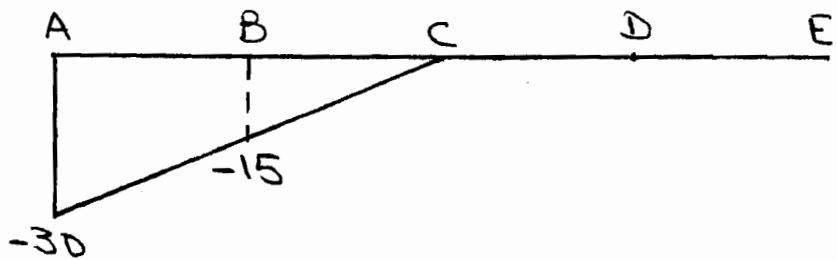
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8.43

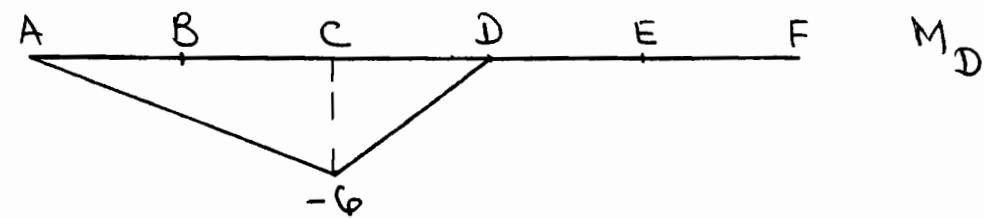
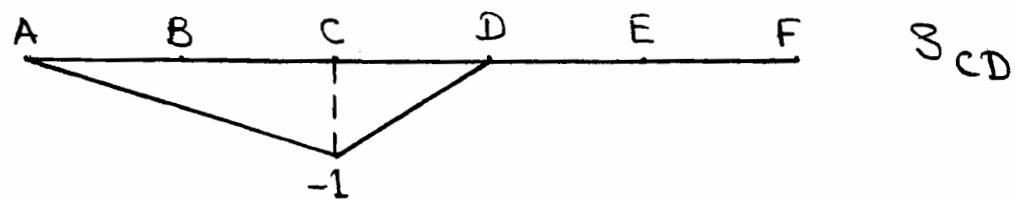


β_{BC}

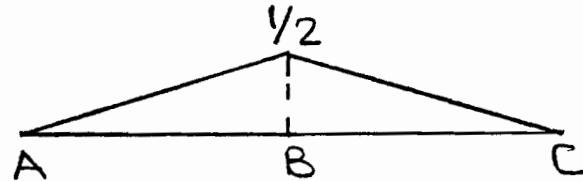


M_C

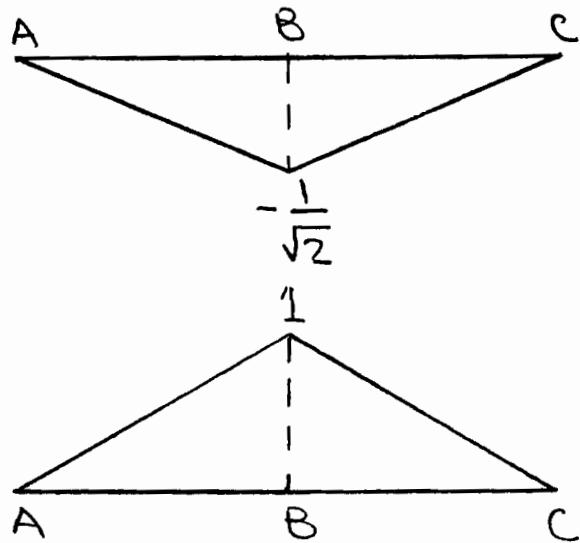
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8.45



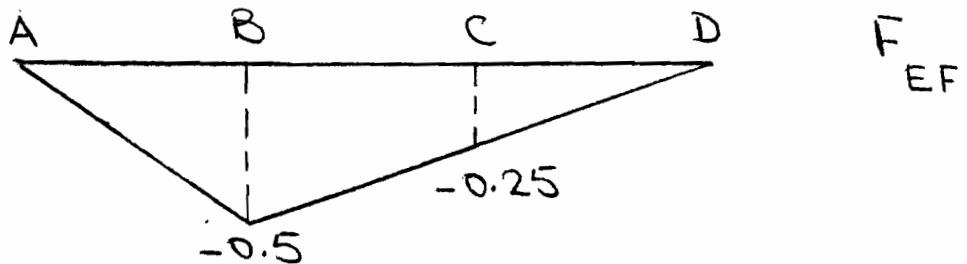
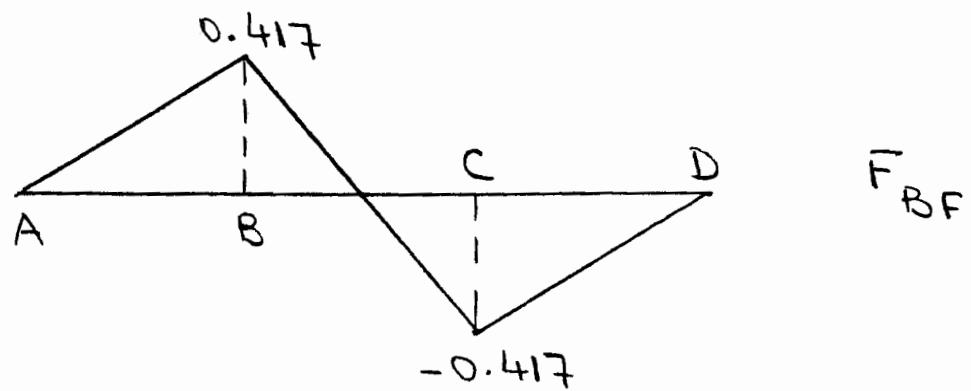
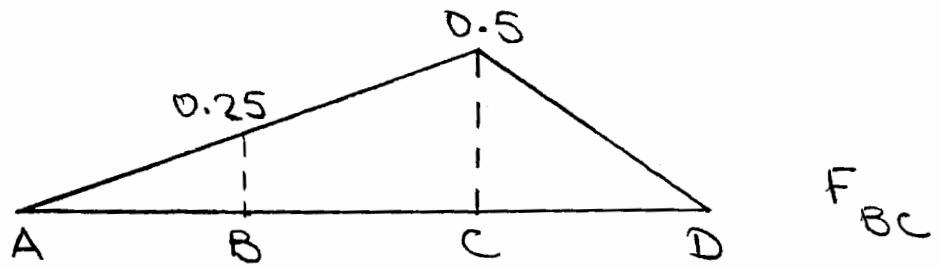
F_{AB}



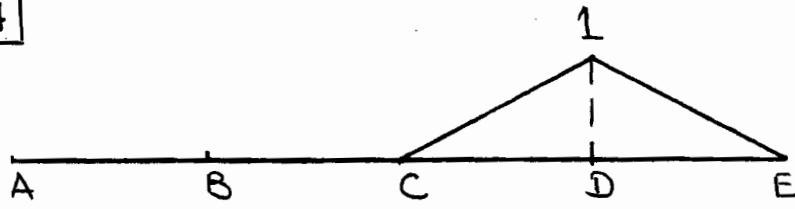
F_{AD}

F_{BD}

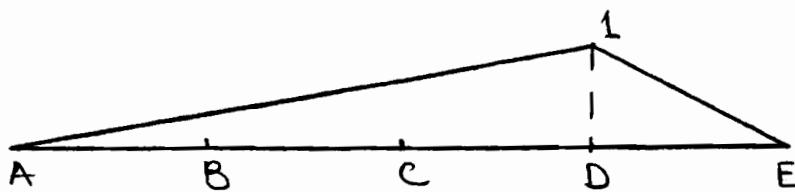
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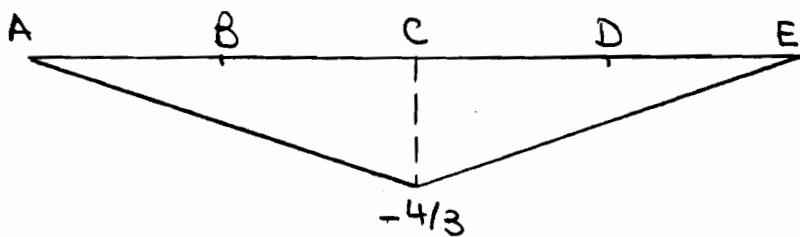
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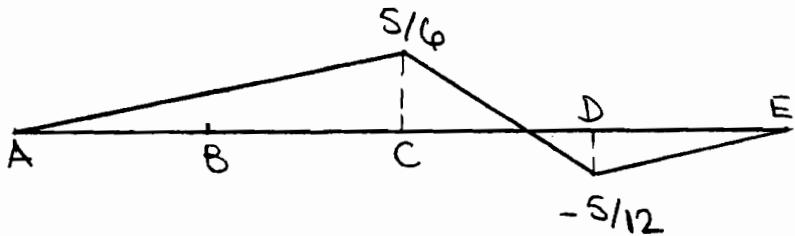
$$F_{DH}$$



$$F_{GD}$$

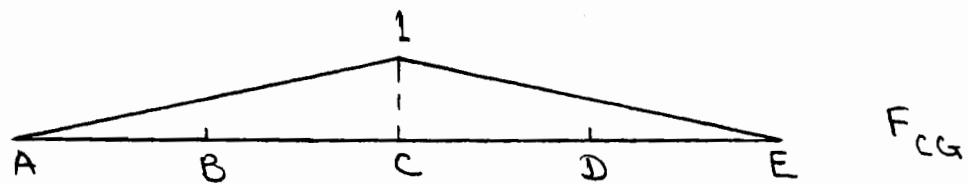
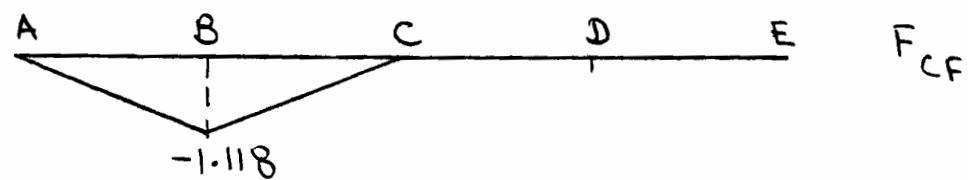
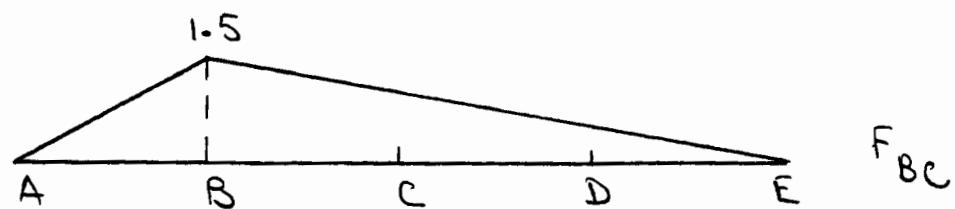
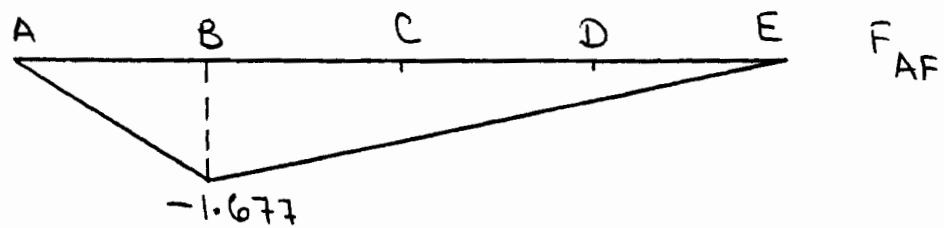


$$F_{GH}$$

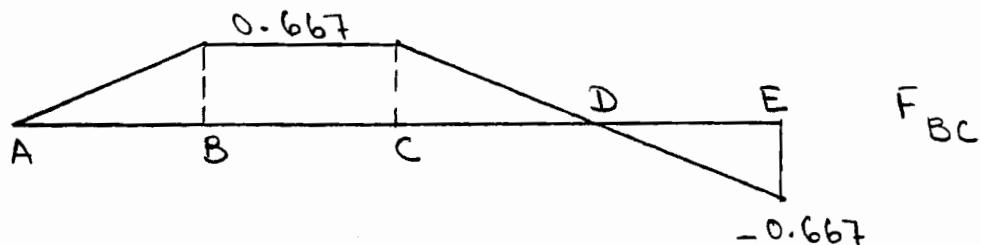
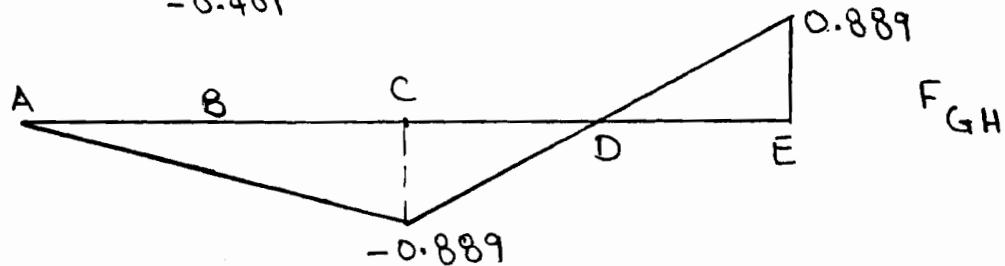
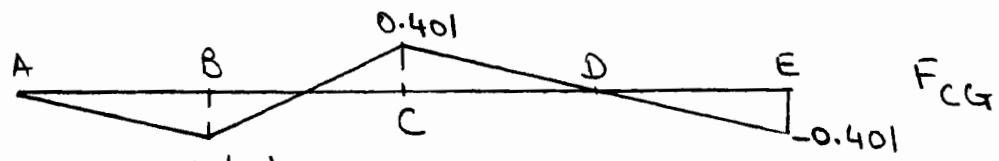
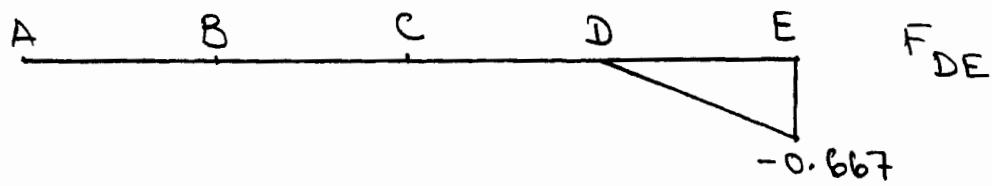


$$F_{CH}$$

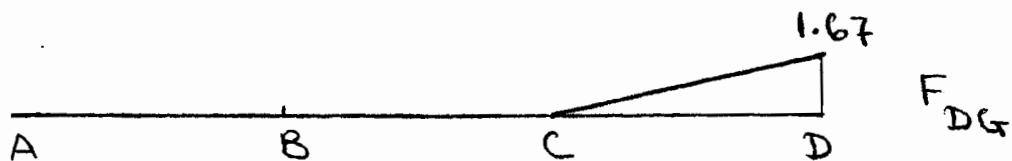
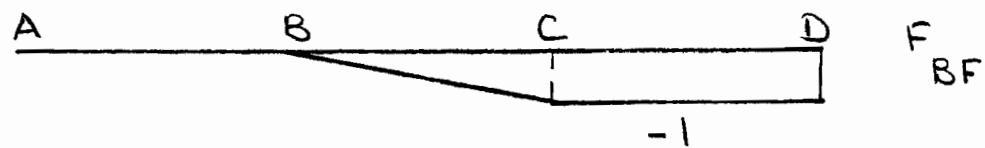
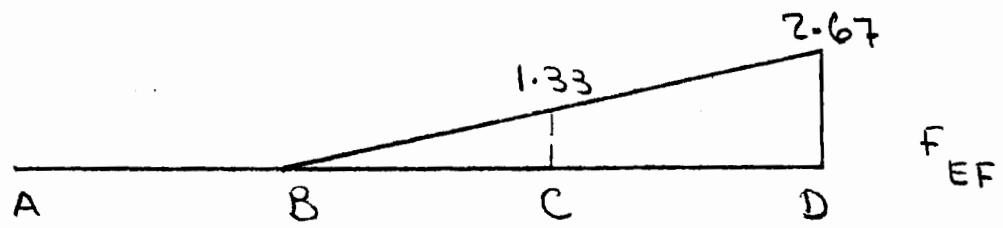
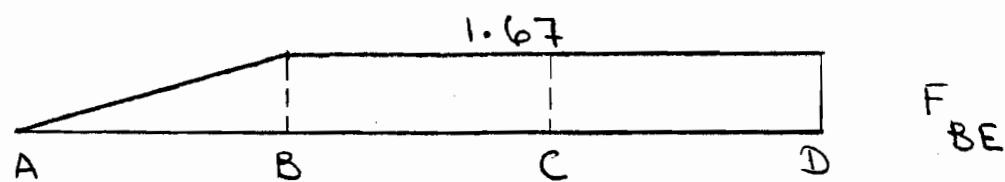
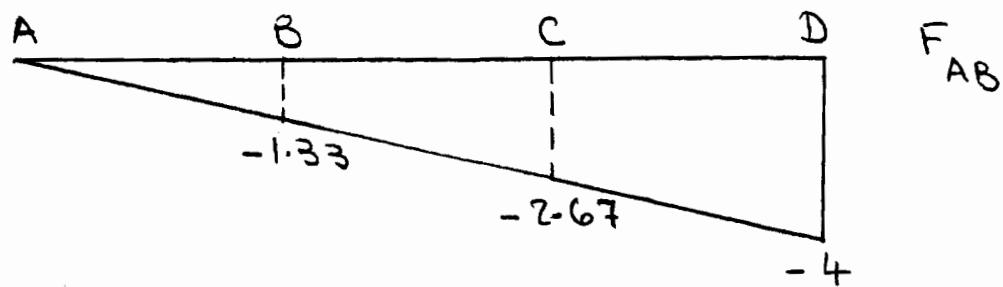
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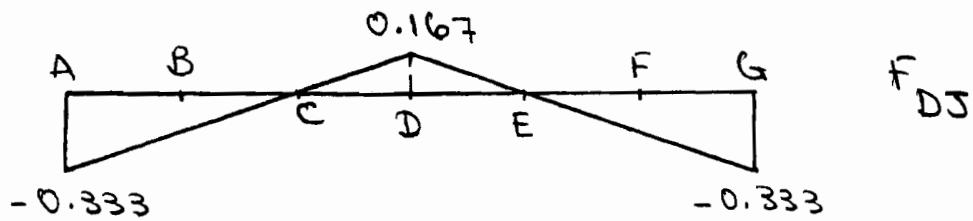
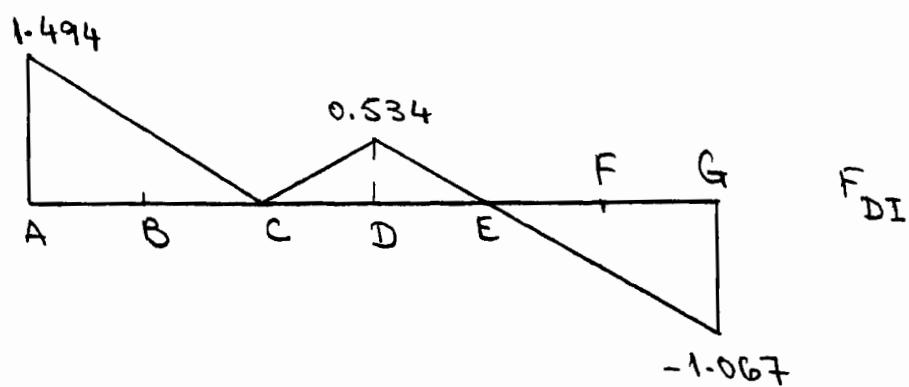
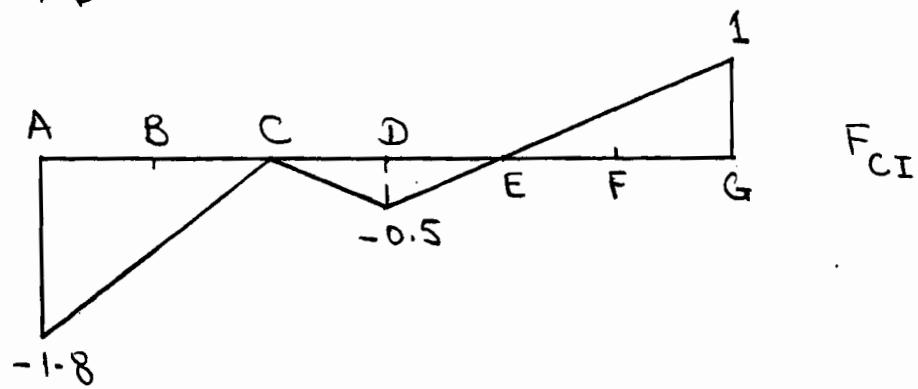
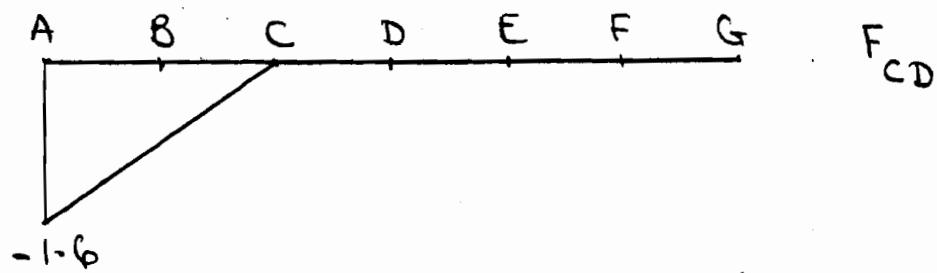
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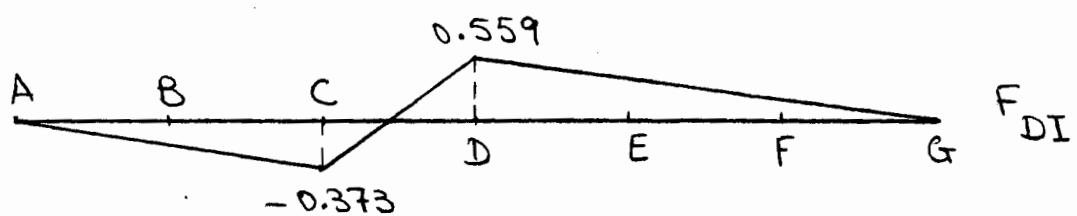
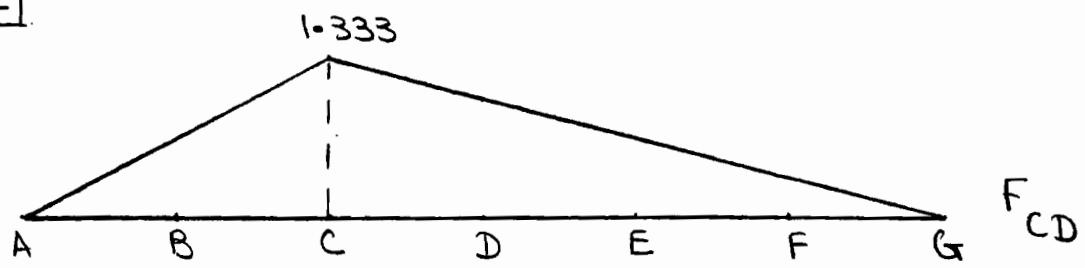
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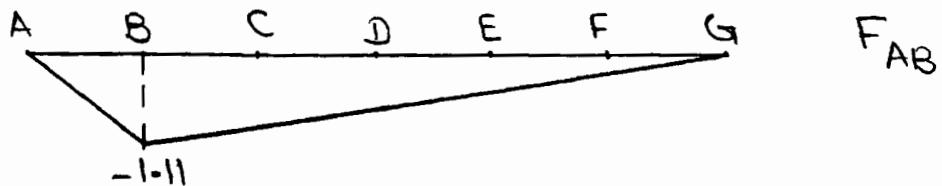
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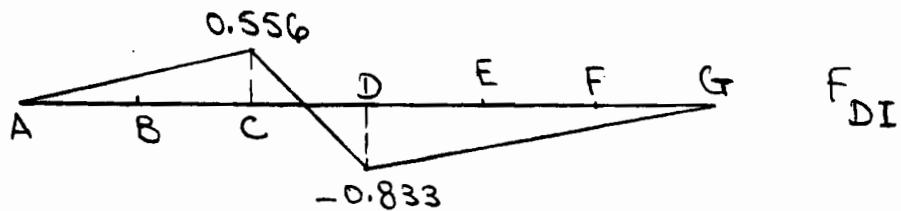
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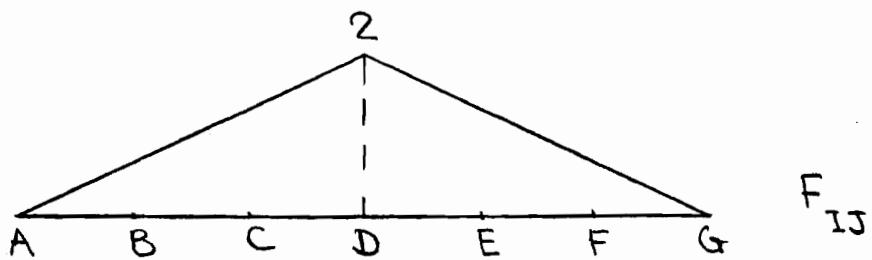
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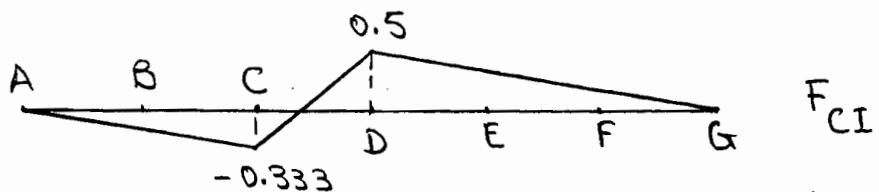
$$F_{AB}$$



$$F_{DI}$$

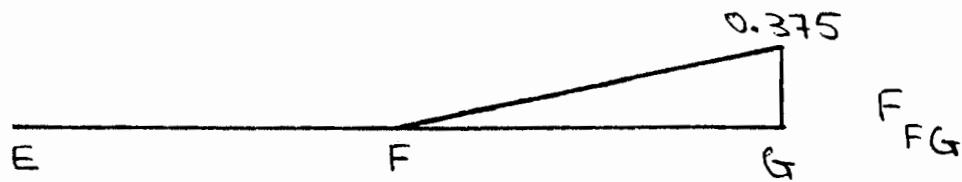
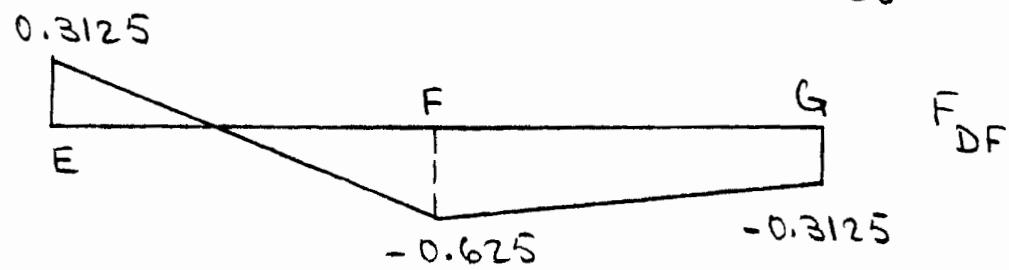
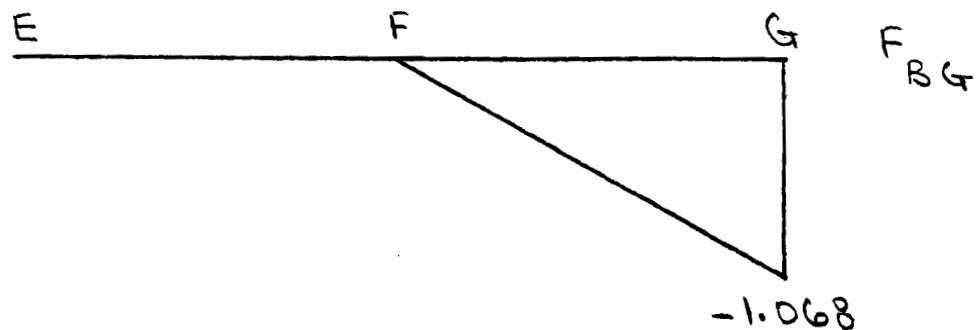
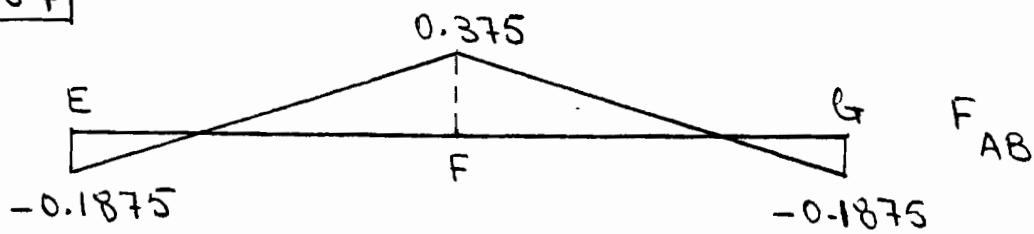


$$F_{IJ}$$

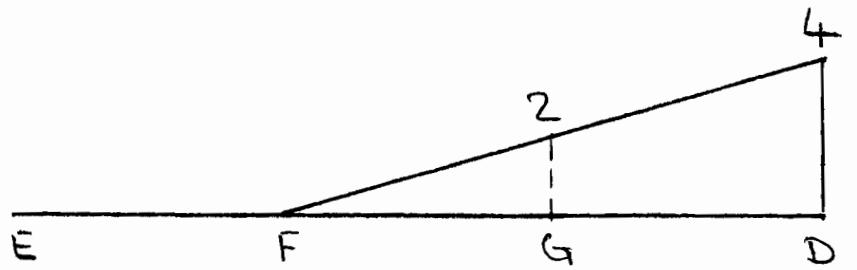
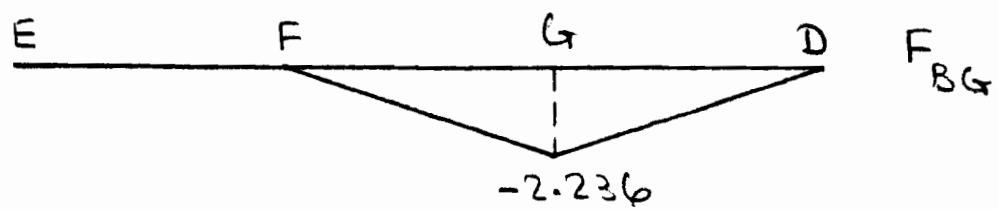
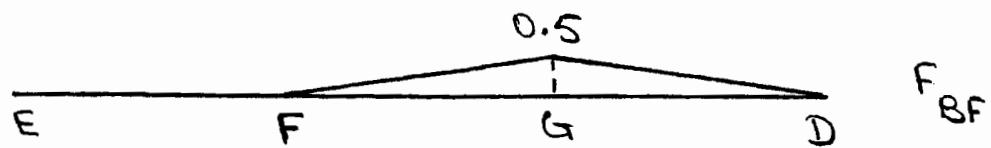
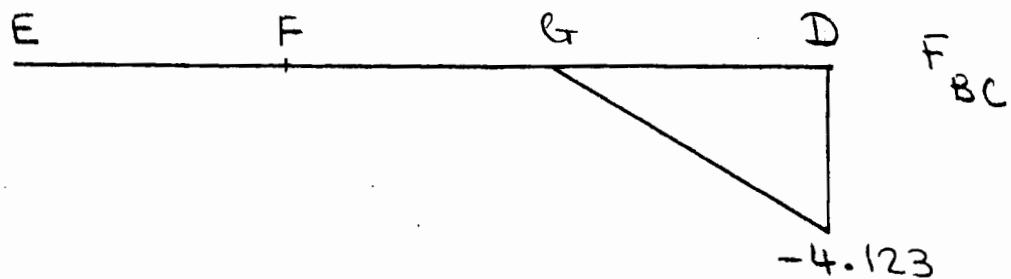


$$F_{CI}$$

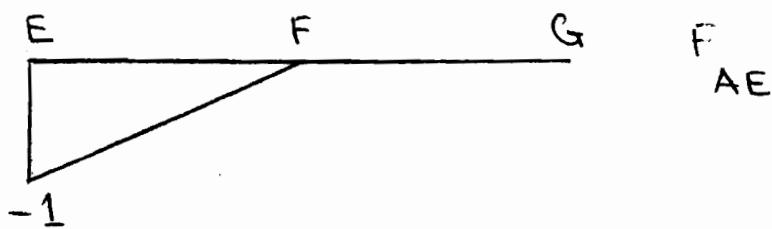
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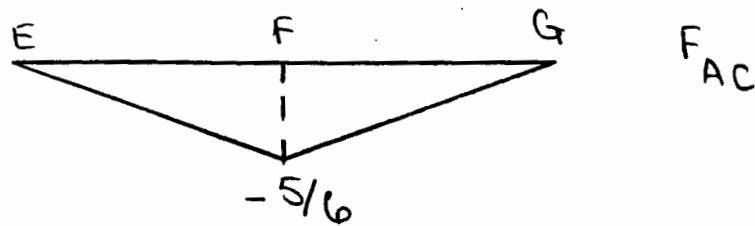
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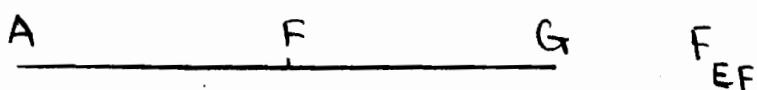
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F_{AE}

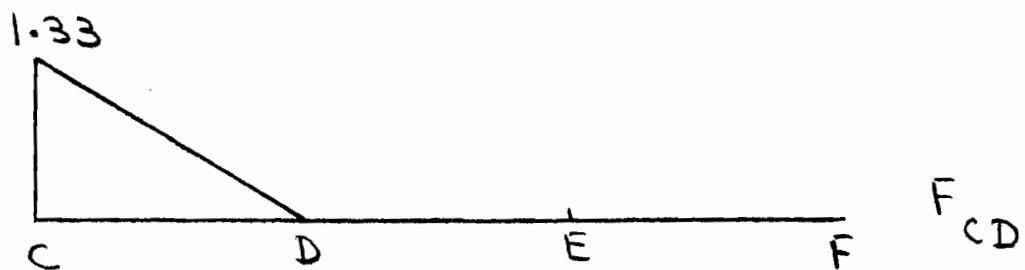
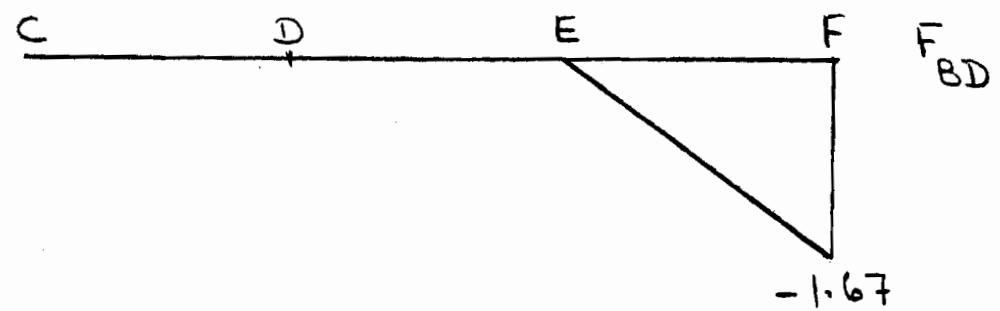
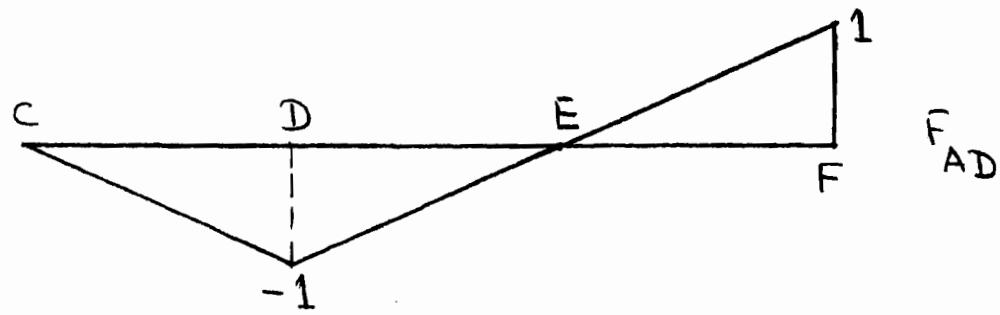


F_{AC}

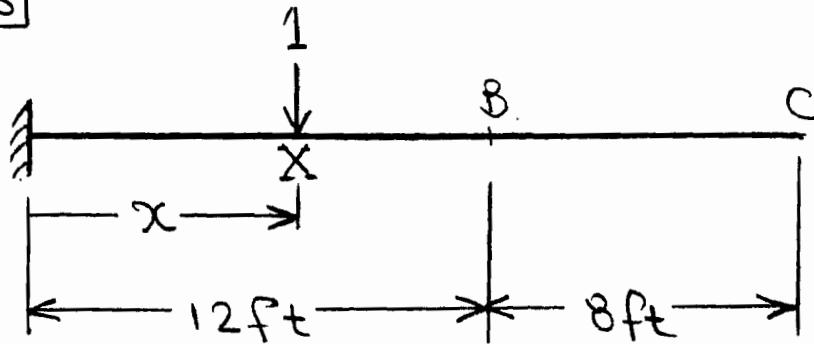


F_{EF}

8.57

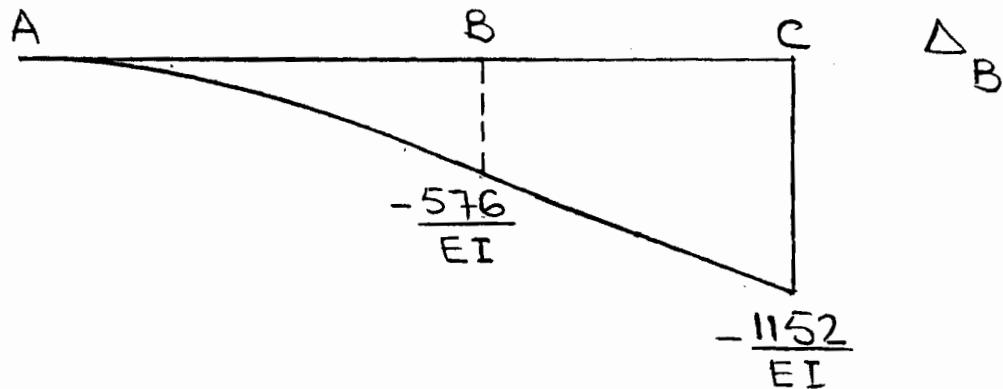


B.58

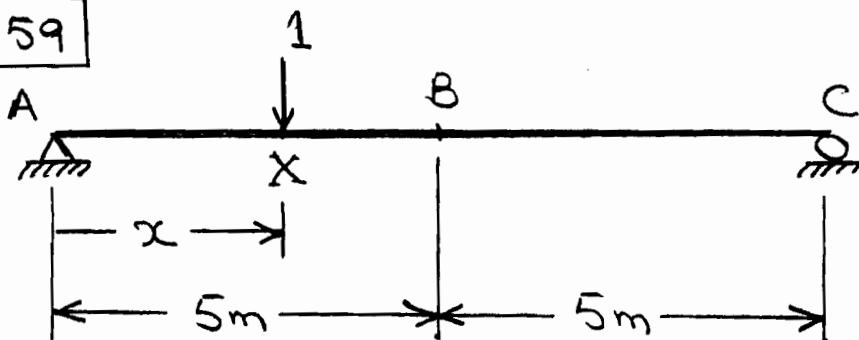


$$0 \leq x \leq 12': f_{Bx} = \frac{1}{6EI} (x^3 - 36x^2)$$

$$12' \leq x \leq 20': f_{Bx} = \frac{72}{EI} (4 - x)$$

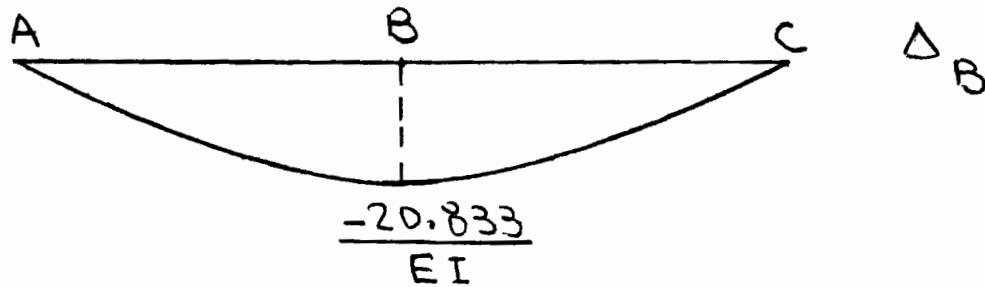


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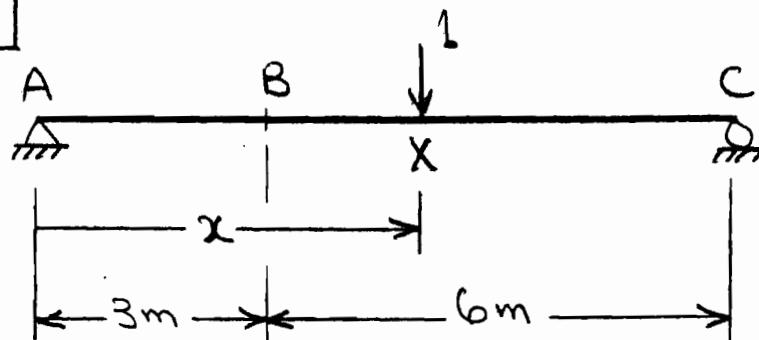


$$0 \leq x \leq 5m: f_{Bx} = \frac{1}{12EI} (x^3 - 75x)$$

$$5m \leq x \leq 10m: f_{Bx} = \frac{1}{12EI} [(10-x)^3 - 75(10-x)]$$

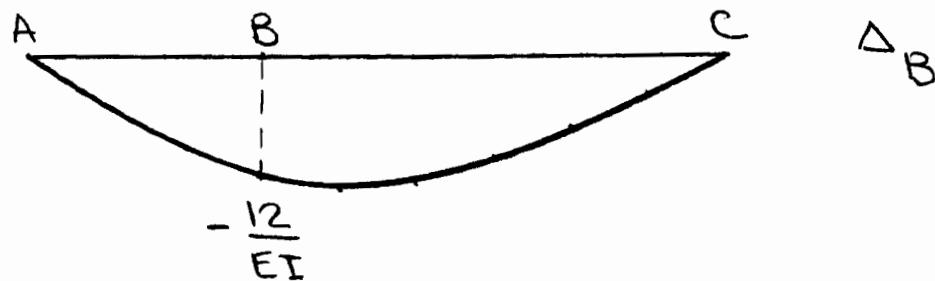


8.60

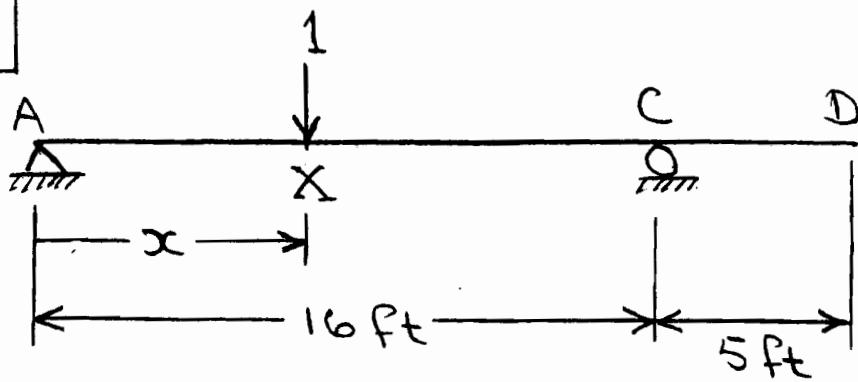


$$0 \leq x \leq 3m \quad f_{Bx} = \frac{1}{9EI} (x^3 - 45x)$$

$$3m \leq x \leq 9m \quad f_{Bx} = \frac{1}{18EI} (-x^3 + 27x^2 - 171x + 81)$$

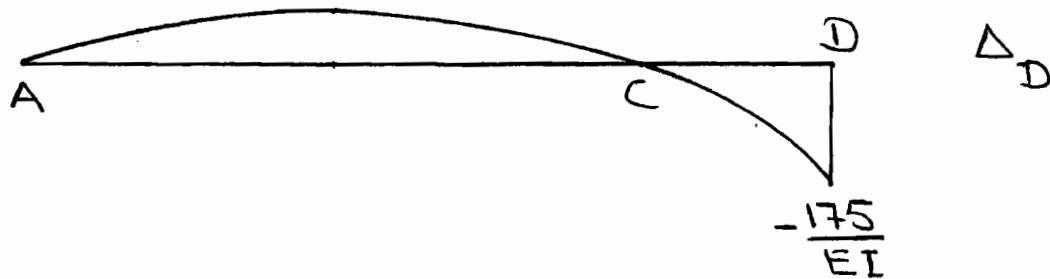


8.61



$$0 \leq x \leq 16': f_{DX} = \frac{5x}{96EI} (256 - x^2)$$

$$16' \leq x \leq 21': f_{DX} = \frac{1}{6EI} (-5376 + 1088x - 63x^2 + x^3)$$

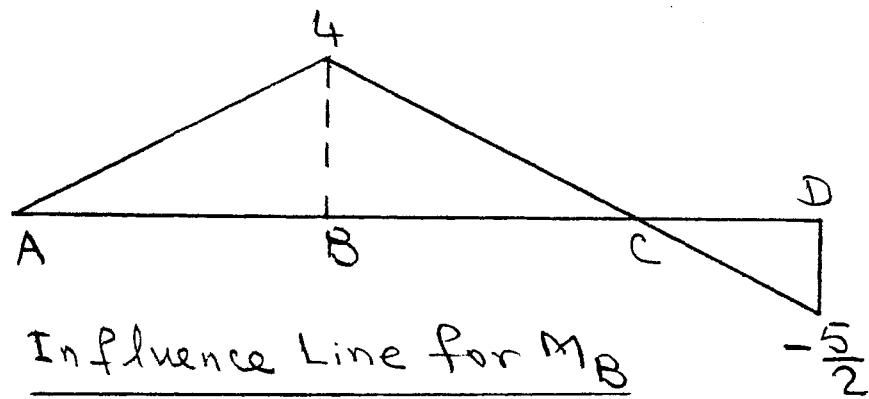


Chapter Nine

Application of Influence Lines

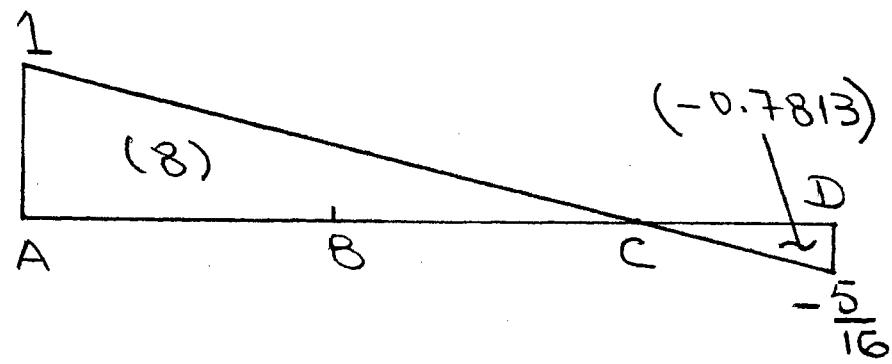
CHAPTER 9

9.1



$$\begin{aligned} \text{Maximum Negative } M_B &= 15 \left(-\frac{5}{2}\right) \\ &= \underline{-37.5 \text{ k-ft}} \end{aligned}$$

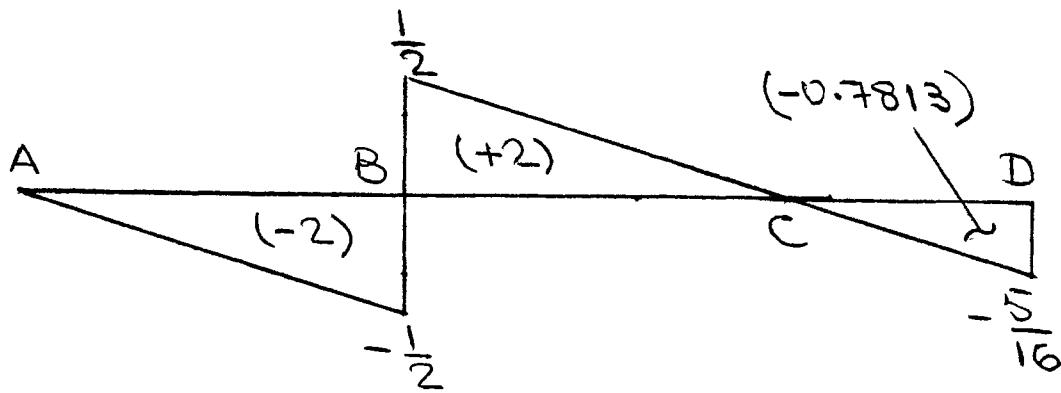
9.2



Influence Line for A_y

$$\text{Maximum Upward } A_y = 3(8) = \underline{24 \text{ k} \uparrow}$$

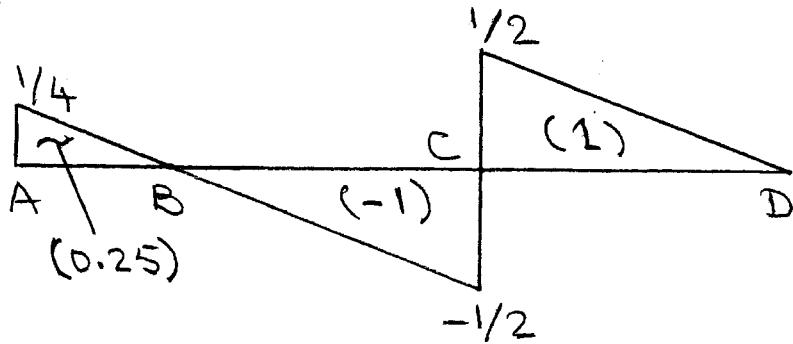
9.3



Influence Line for S_B

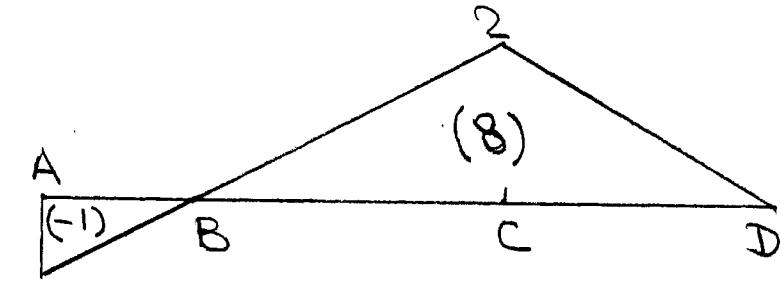
$$\begin{aligned}\text{Maximum Negative } S_B &= 3(-2 - 0.7813) \\ &= \underline{-8.344 \text{ k}}\end{aligned}$$

9.4

Influence Line for S_C

$$\begin{aligned} \text{Maximum Positive } S_C &= 100\left(\frac{1}{2}\right) + 50(0.25+1) \\ &\quad + 20(0.25-1+1) \\ &= \underline{\underline{117.5 \text{ kN}}} \end{aligned}$$

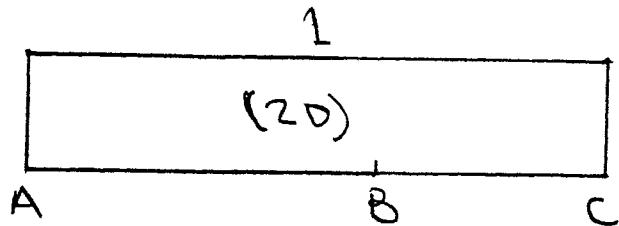
$$\begin{aligned} \text{Maximum Negative } S_C &= 100\left(-\frac{1}{2}\right) + 50(-1) \\ &\quad + 20(0.25-1+1) \\ &= \underline{\underline{-95 \text{ kN}}} \end{aligned}$$

Influence Line for M_C

$$\begin{aligned} \text{Maximum Positive } M_C &= 100(2) + 50(8) + 20(7) \\ &= \underline{\underline{740 \text{ kN.m}}} \end{aligned}$$

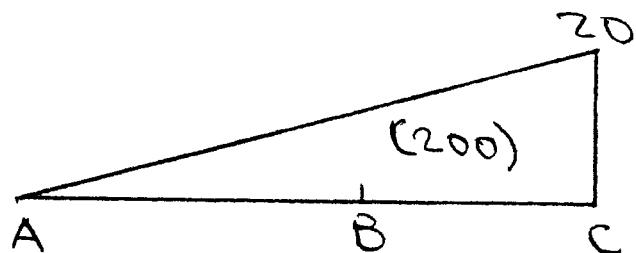
$$\begin{aligned} \text{Maximum Negative } M_C &= 100(-1) + 50(-1) + 20(7) \\ &= \underline{\underline{-10 \text{ kN.m}}} \end{aligned}$$

9.5



Influence Line for A_y

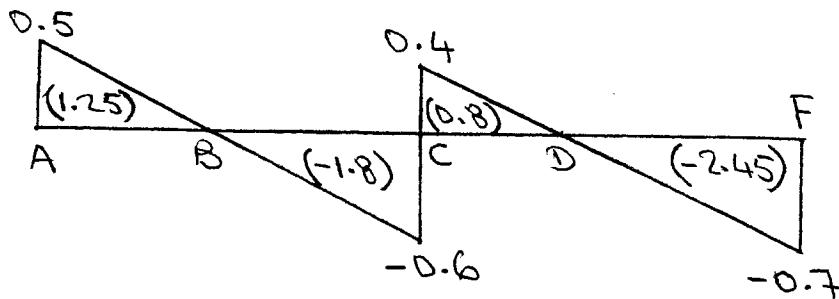
$$\begin{aligned}\text{Maximum Upward } A_y &= 25(1) + 2(20) + 0.5(20) \\ &= \underline{75 \text{ k} \uparrow}\end{aligned}$$



Influence Line for $M_A (+\zeta)$

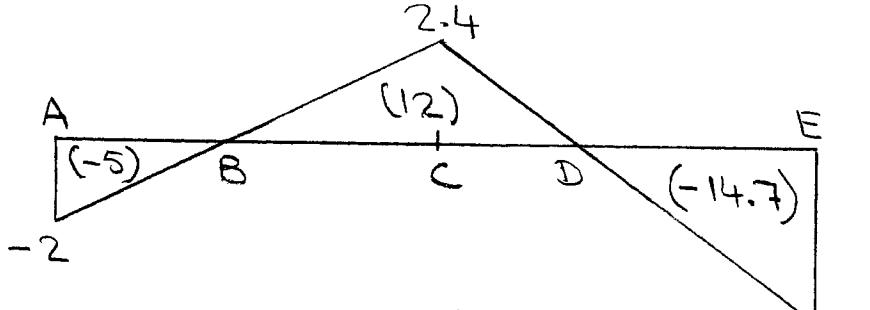
$$\begin{aligned}\text{Maximum Counterclockwise } M_A &= 25(20) + 2(200) + 0.5(200) \\ &= \underline{1000 \text{ k-ft} \leftarrow}\end{aligned}$$

9.6

Influence Line for S_c

$$\begin{aligned} \text{Maximum Positive } S_c &= 150(0.5) + 50(1.25 + 0.8) \\ &\quad + 25(1.25 - 1.8 + 0.8 - 2.45) \\ &= \underline{122.5 \text{ kN}} \end{aligned}$$

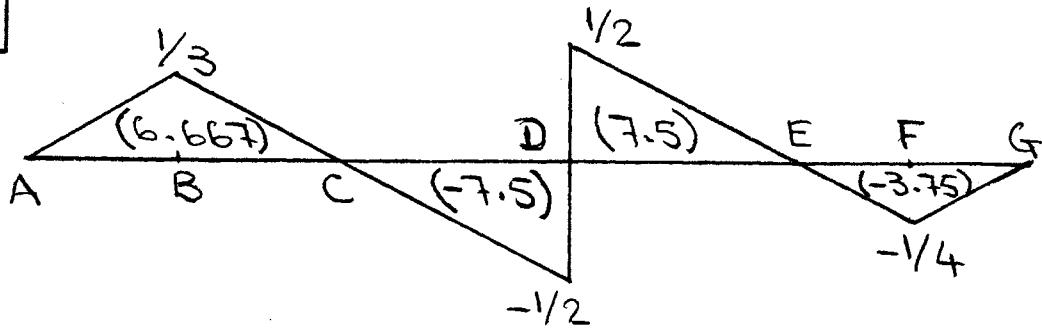
$$\begin{aligned} \text{Maximum Negative } S_c &= 150(-0.7) + 50(-1.8 - 2.45) \\ &\quad + 25(1.25 - 1.8 + 0.8 - 2.45) \\ &= \underline{-372.5 \text{ kN}} \end{aligned}$$

Influence Line for M_c

$$\begin{aligned} \text{Maximum Positive } M_c &= 150(2.4) + 50(12) \\ &\quad + 25(-5 + 12 - 14.7) \\ &= \underline{767.5 \text{ kN.m}} \end{aligned}$$

$$\begin{aligned} \text{Maximum Negative } M_c &= 150(-4.2) + 50(-5 - 14.7) \\ &\quad + 25(-5 + 12 - 14.7) \\ &= \underline{-1807.5 \text{ kN.m}} \end{aligned}$$

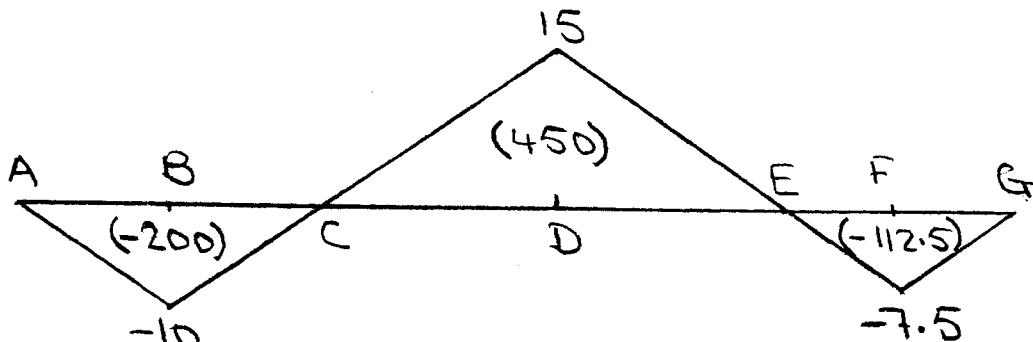
9.7



Influence Line for S_D

$$\begin{aligned} \text{Maximum Positive } S_D &= 30\left(\frac{1}{2}\right) + 3(6.667 + 7.5) \\ &\quad + 1(6.667 - 7.5 + 7.5 - 3.75) \\ &= \underline{60.417 \text{ k}} \end{aligned}$$

$$\begin{aligned} \text{Maximum Negative } S_D &= 30\left(-\frac{1}{2}\right) + 3(-7.5 - 3.75) \\ &\quad + 1(6.667 - 7.5 + 7.5 - 3.75) \\ &= \underline{-45.833 \text{ k}} \end{aligned}$$

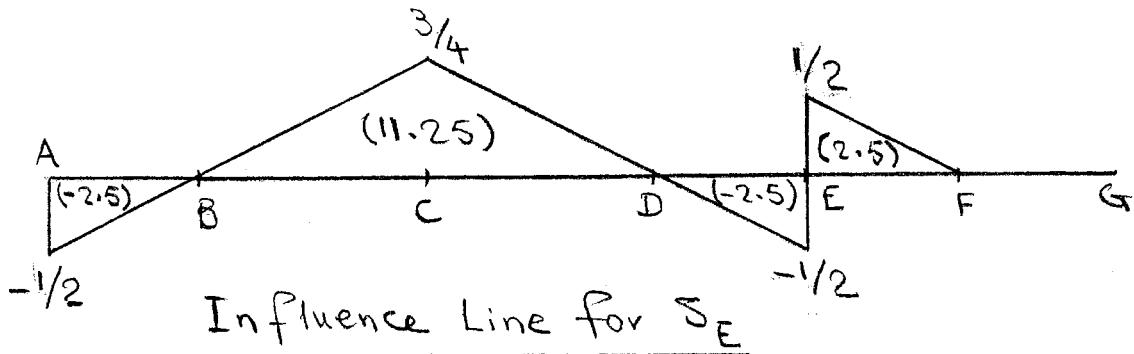


Influence Line for M_D

$$\begin{aligned} \text{Maximum Positive } M_D &= 30(15) + 3(450) \\ &\quad + 1(-200 + 450 - 112.5) \\ &= \underline{1937.5 \text{ k-ft}} \end{aligned}$$

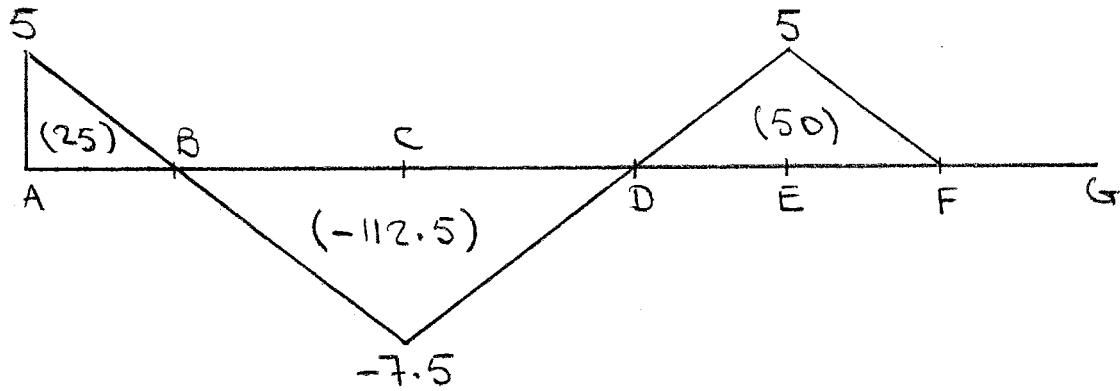
$$\begin{aligned} \text{Maximum Negative } M_D &= 30(-10) + 3(-200 - 112.5) \\ &\quad + 1(-200 + 450 - 112.5) \\ &= \underline{-1100 \text{ k-ft}} \end{aligned}$$

9.8



$$\begin{aligned} \text{Maximum Positive } S_E &= 40\left(\frac{3}{4}\right) + 2(11.25 + 2.5) + 1(-2.5 + 11.25) \\ &= \underline{66.25 \text{ k}} \end{aligned}$$

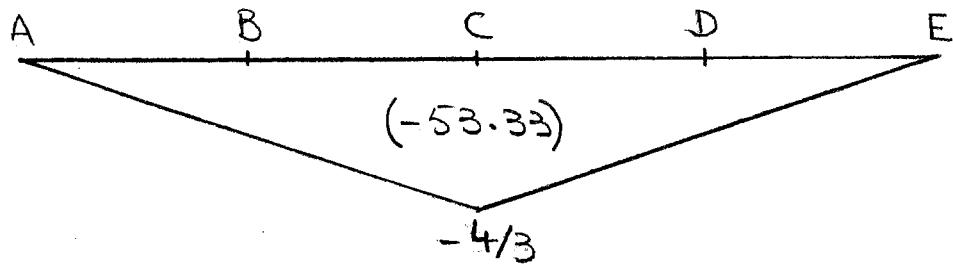
$$\begin{aligned} \text{Maximum Negative } S_E &= 40\left(-\frac{1}{2}\right) + 2(-2.5 - 2.5) + 1(-2.5 + 11.25) \\ &= \underline{-21.25 \text{ k}} \end{aligned}$$



$$\begin{aligned} \text{Maximum Positive } M_E &= 40(5) + 2(25 + 50) + 1(25 - 112.5 + 50) \\ &= \underline{312.5 \text{ k-ft}} \end{aligned}$$

$$\begin{aligned} \text{Maximum Negative } M_E &= 40(-7.5) + 2(-112.5) + 1(25 - 112.5 + 50) \\ &= \underline{-562.5 \text{ k-ft}} \end{aligned}$$

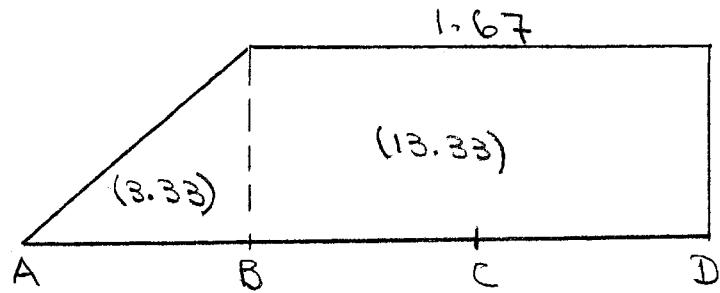
9.9



Influence Line for F_{GH}

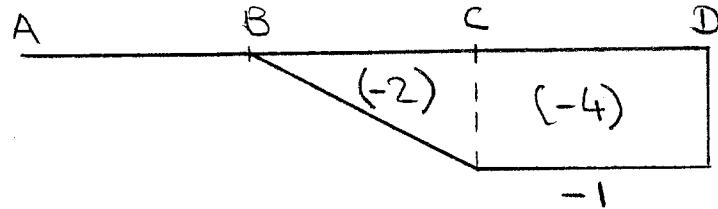
$$\begin{aligned} \text{Maximum Compressive } F_{GH} &= 30\left(-\frac{4}{3}\right) + 3(-53.33) \\ &= -200k = \underline{\underline{200k(C)}} \end{aligned}$$

9.10



Influence Line for F_{BE}

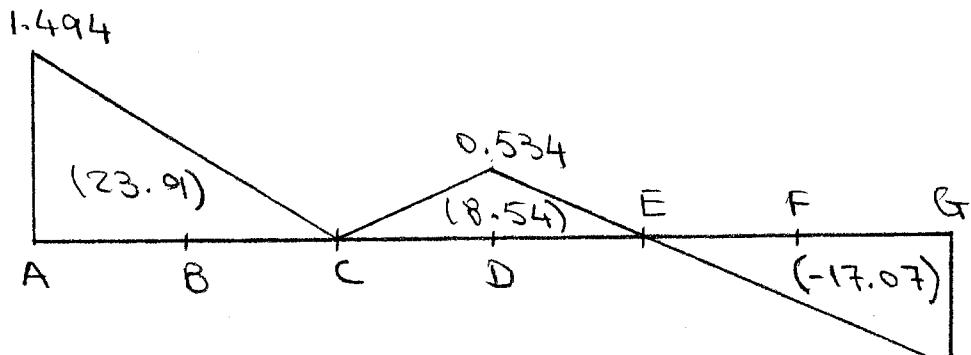
$$\begin{aligned} \text{Maximum Tensile } F_{BE} &= 120(1.67) + 60(3.33 + 13.33) \\ &= \underline{1200 \text{ kN (T)}} \end{aligned}$$



Influence Line for F_{BF}

$$\begin{aligned} \text{Maximum Compressive } F_{BF} &= 120(-1) + 60(-2-4) \\ &= -480 \text{ kN} \\ &= \underline{480 \text{ kN (C)}} \end{aligned}$$

9.11

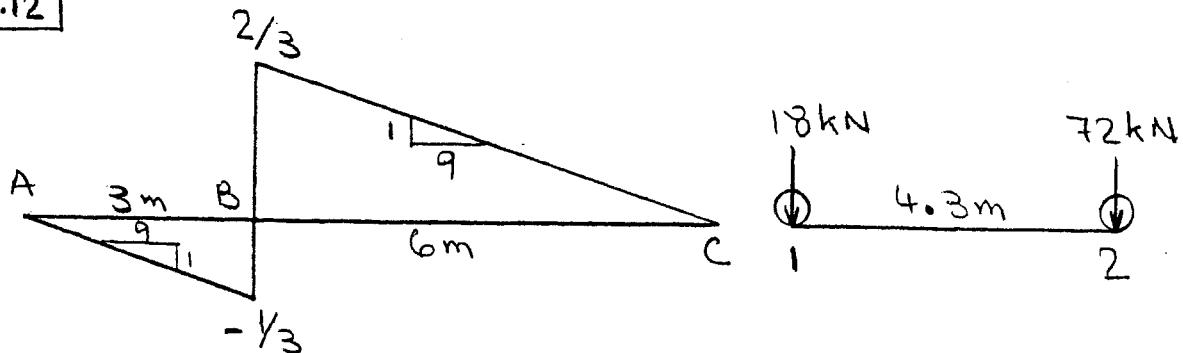


Influence Line for F_{DI}

$$\begin{aligned} \text{Maximum Tensile } F_{DI} &= 40(1.494) + 4(23.9 + 8.54) \\ &\quad + 2(23.9 + 8.54 - 17.07) = \underline{\underline{220.3 \text{ k (T)}}} \end{aligned}$$

$$\begin{aligned} \text{Maximum Compressive } F_{DI} &= 40(-1.067) + 4(-17.07) \\ &\quad + 2(23.9 + 8.54 - 17.07) = -80.2 \text{ k} \\ &= \underline{\underline{80.2 \text{ k (C)}}} \end{aligned}$$

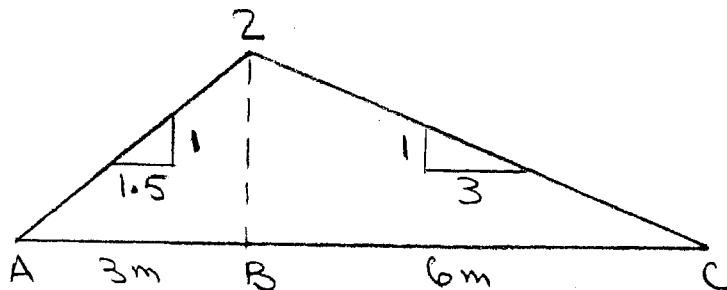
9.12

Influence Line for S_B Loading position 1:

$$S_B = 18 \left(\frac{2}{3}\right) + 72 \left(\frac{1.7}{9}\right) = 25.6 \text{ kN}$$

Loading position 2:

$$S_B = 72 \left(\frac{2}{3}\right) = 48 \text{ kN}$$

Max. Positive $S_B = \underline{48 \text{ kN}}$ Influence Line for M_B Loading position 1:

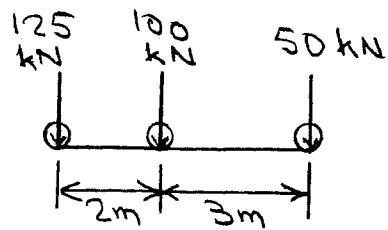
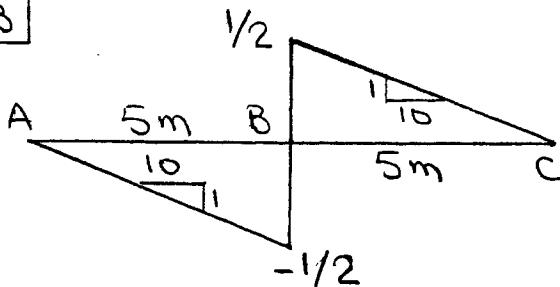
$$M_B = 18(2) + 72 \left(\frac{1.7}{3}\right) = 76.8 \text{ kN.m}$$

Loading position 2:

$$M_B = 72(2) = 144 \text{ kN.m}$$

Max. Positive $M_B = \underline{144 \text{ kN.m}}$

9.13

Influence Line for S_B Loading position 1:

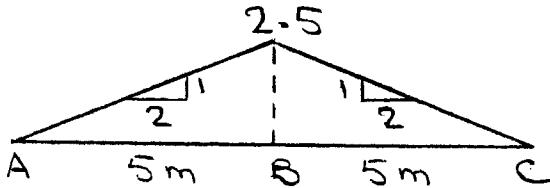
$$S_B = 125\left(\frac{1}{2}\right) + 100\left(\frac{3}{10}\right) = 92.5 \text{ kN}$$

Loading position 2:

$$S_B = 125\left(-\frac{3}{10}\right) + 100\left(\frac{1}{2}\right) + 50\left(\frac{2}{10}\right) = 22.5 \text{ kN}$$

Loading position 3:

$$S_B = 100\left(-\frac{2}{10}\right) + 50\left(\frac{1}{2}\right) = 5 \text{ kN}$$

Max. Positive $S_B = \underline{92.5 \text{ kN}}$ Influence Line for M_B Loading position 1:

$$M_B = 125(2.5) + 100(1.5) = 462.5 \text{ kN.m}$$

Loading position 2:

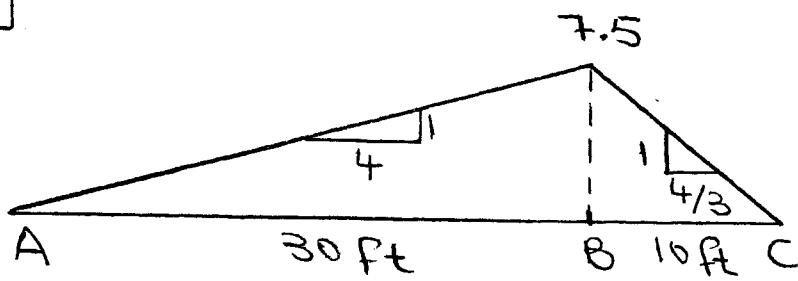
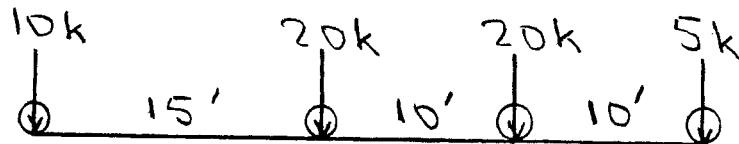
$$M_B = 125(1.5) + 100(2.5) + 50(1) = 487.5 \text{ kN.m}$$

Loading position 3:

$$M_B = 100(1) + 50(2.5) = 225 \text{ kN.m}$$

Max. Positive $M_B = \underline{487.5 \text{ kN.m}}$

9.14

Influence Line for M_B Loading position 1:

$$M_B = 10(7.5) = 75 \text{ k-ft}$$

Loading position 2:

$$M_B = 10\left(\frac{15}{4}\right) + 20(7.5) = 187.5 \text{ k-ft}$$

Loading position 3:

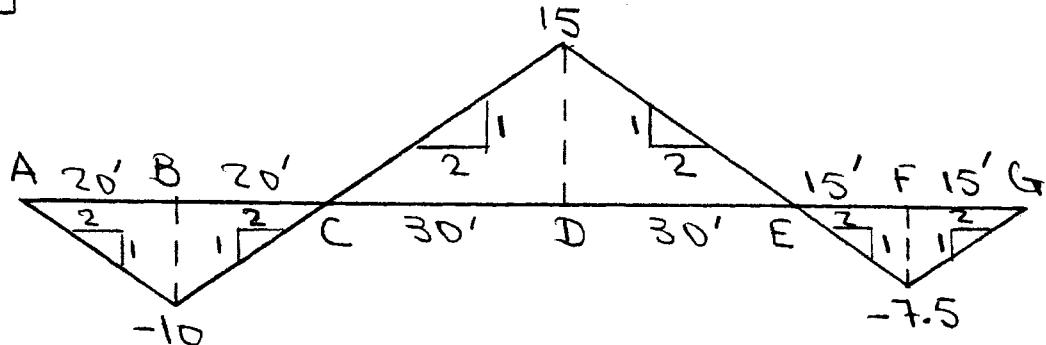
$$M_B = 10\left(\frac{5}{4}\right) + 20(5) + 20(7.5) = 262.5 \text{ k-ft}$$

Loading position 4:

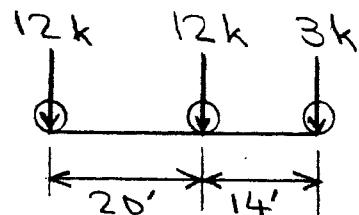
$$M_B = 20\left(\frac{10}{4}\right) + 20(5) + 5(7.5) = 187.5 \text{ k-ft}$$

Max. Positive $M_B = 262.5 \text{ k-ft}$

9.15



Influence Line for M_D



Loading position 1:

$$M_D = 12(15) + 12(5) + 3\left(-\frac{4}{2}\right) = 234 \text{ k-ft}$$

Loading position 2:

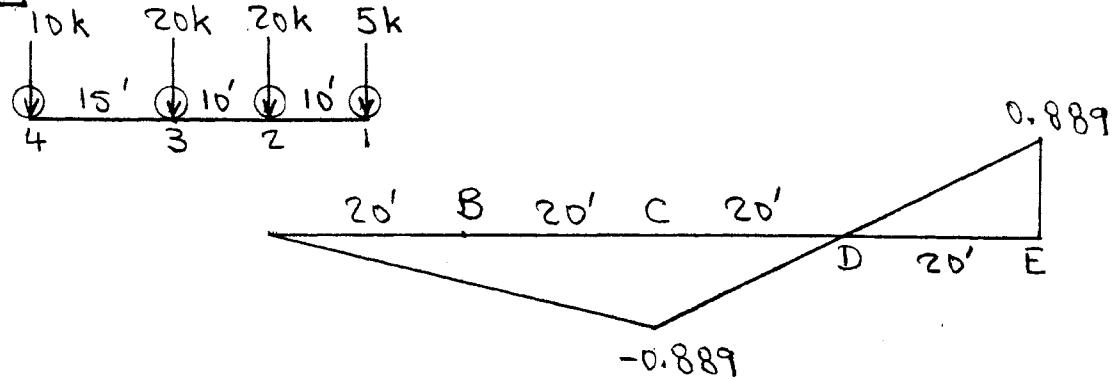
$$M_D = 12(5) + 12(15) + 3\left(\frac{15}{2}\right) = 264 \text{ k-ft}$$

Loading position 3:

$$M_D = 12\left(-\frac{4}{2}\right) + 12\left(\frac{16}{2}\right) + 3(15) = 117 \text{ k-ft}$$

Max. Positive $M_D = 264 \text{ k-ft}$

9.16

Influence Line for F_{GH} Loading position 1:

$$F_{GH} = [5(40) + 20(30+20) + 10(5)] \left(-\frac{0.889}{40}\right) = -27.78 \text{ k}$$

Loading position 2:

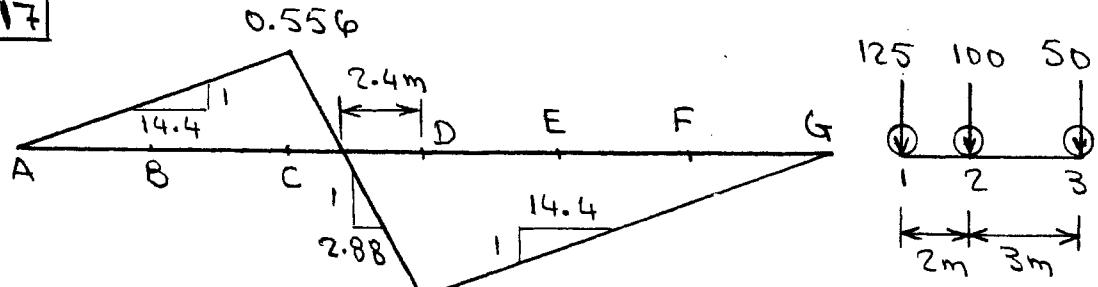
$$F_{GH} = 5(10) \left(-\frac{0.889}{20}\right) + [20(40+30) + 10(15)] \left(-\frac{0.889}{40}\right) \\ = -36.67 \text{ k}$$

Loading position 3:

$$F_{GH} = [5(0) + 20(10)] \left(-\frac{0.889}{20}\right) + [20(40) + 10(25)] \left(-\frac{0.889}{40}\right) \\ = -32.23 \text{ k}$$

Maximum compressive $F_{GH} = -36.67 \text{ k} = \underline{36.67 \text{ k (C)}}$

9.17



Influence Line for F_{DI}

Loading position 1:

$$F_{DI} = 125(0.556) - 100\left(\frac{0.4}{2.88}\right) - 50\left(\frac{11}{14.4}\right) = 17.42 \text{ kN}$$

Loading position 2:

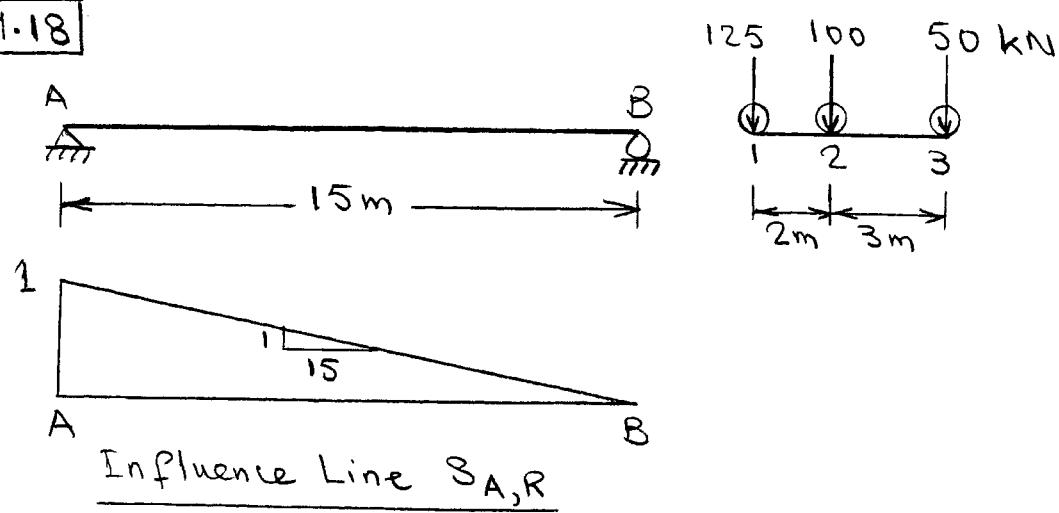
$$F_{DI} = 125\left(\frac{6}{14.4}\right) + 100(0.556) - 50\left(\frac{1.4}{2.88}\right) = 83.38 \text{ kN}$$

Loading position 3:

$$F_{DI} = 125\left(\frac{3}{14.4}\right) + 100\left(\frac{5}{14.4}\right) + 50(0.556) = 88.56 \text{ kN}$$

Maximum Tensile $F_{DI} = \underline{88.56 \text{ kN (T)}}$

9.18



Loading position 1:

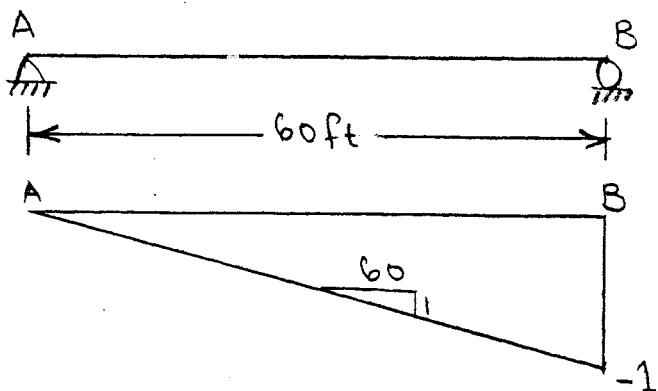
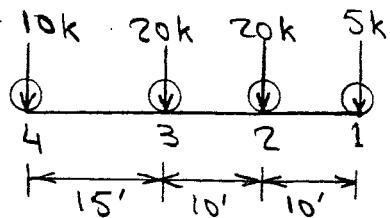
$$S_{A,R} = [125(15) + 100(13) + 50(10)] \frac{1}{15} = 245 \text{ kN}$$

Loading position 2:

$$S_{A,R} = [100(15) + 50(12)] \frac{1}{15} = 140 \text{ kN}$$

Absolute maximum shear = 245 kN

9.19



Influence Line for $S_{B,L}$

Loading position 1:

$$S_{B,L} = [5(60) + 20(50+40) + 10(25)] \left(-\frac{1}{60}\right) = -39.17 \text{ k}$$

Loading position 2:

$$S_{B,L} = [20(60+50) + 10(35)] \left(-\frac{1}{60}\right) = -42.5 \text{ k}$$

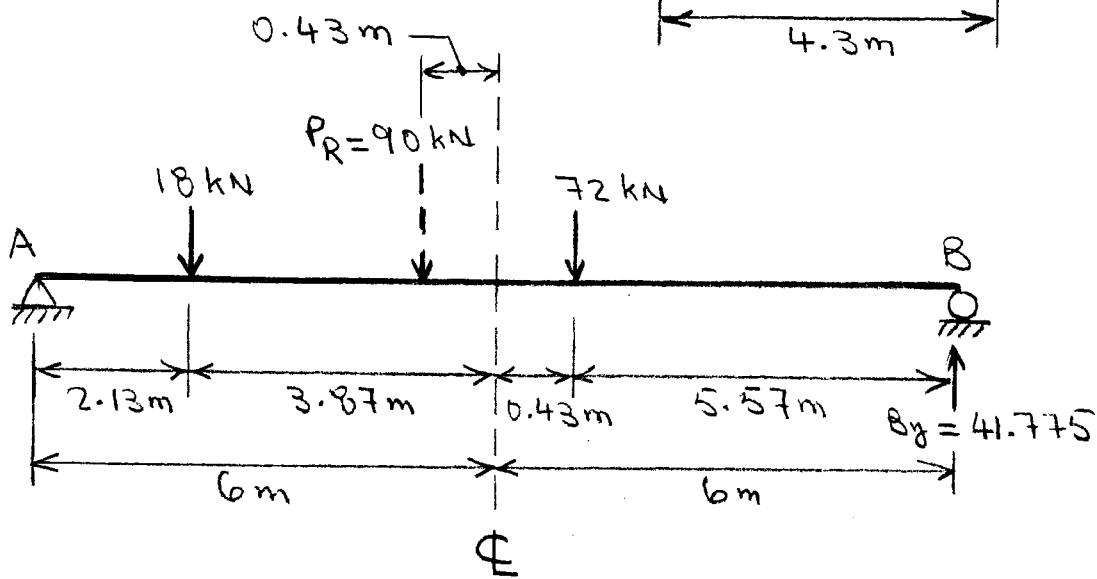
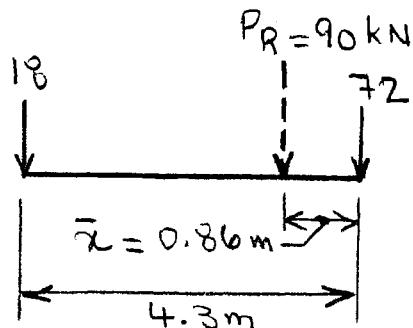
Loading position 3:

$$S_{B,L} = [20(60) + 10(45)] \left(-\frac{1}{60}\right) = -27.5 \text{ k}$$

Absolute maximum shear = 42.5 k

9.20

$$\bar{x} = \frac{18(4.3)}{90} = 0.86 \text{ m}$$

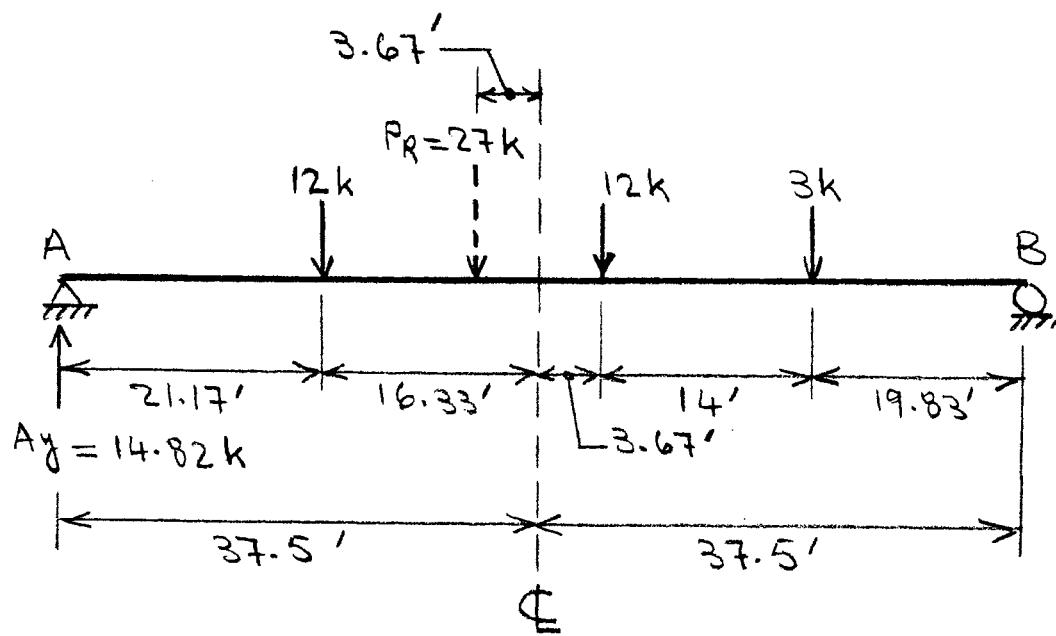
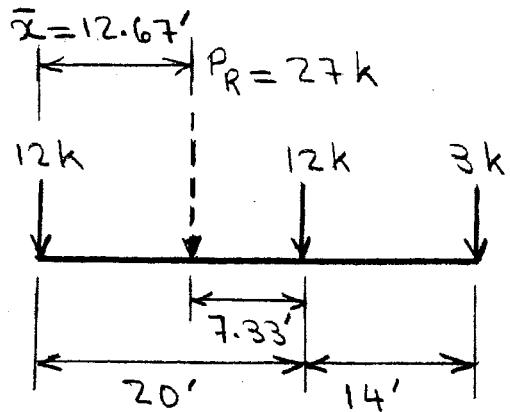


$$B_y = \frac{90(6 - 0.43)}{12} = 41.775 \text{ kN}$$

$$M_{\max} = 41.775 (5.57) = \underline{232.7 \text{ kN.m}}$$

9.21

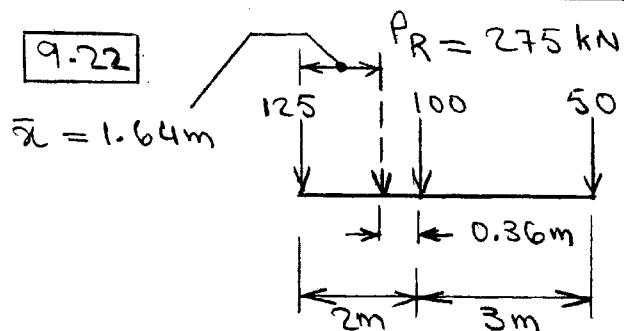
$$\bar{x} = \frac{12(20) + 3(34)}{27} = 12.67 \text{ ft}$$



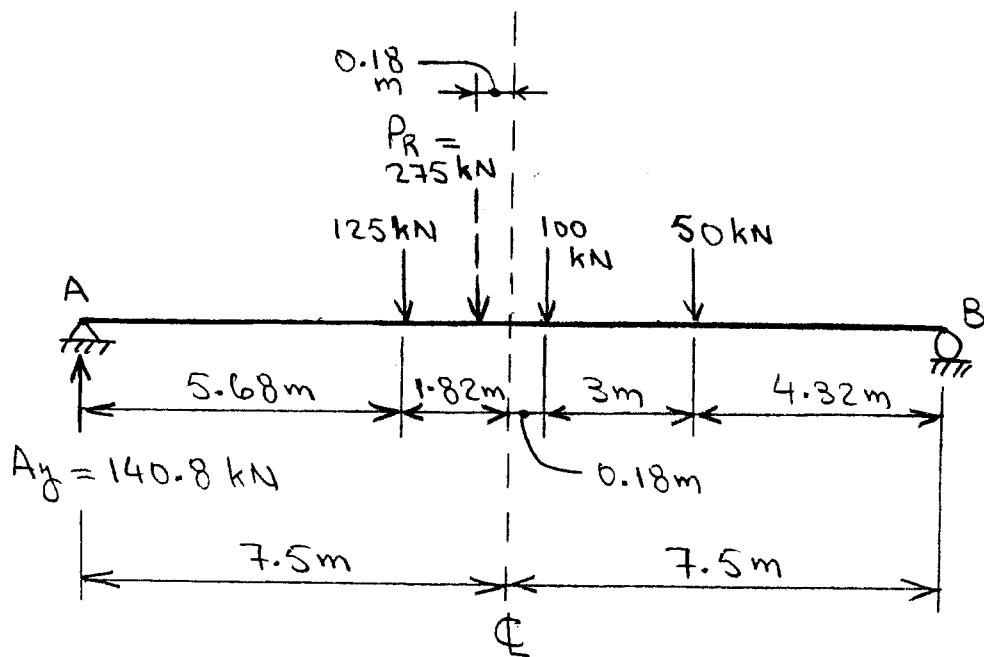
$$A_y = \frac{27(37.5 + 3.67)}{75} = 14.82 \text{ k}$$

$$M_{\max} = 14.82(37.5 + 3.67) - 12(20) = \underline{\underline{370.1 \text{ k-ft}}}$$

9.22



$$\bar{x} = \frac{100(2) + 50(5)}{275} = 1.64 \text{ m}$$

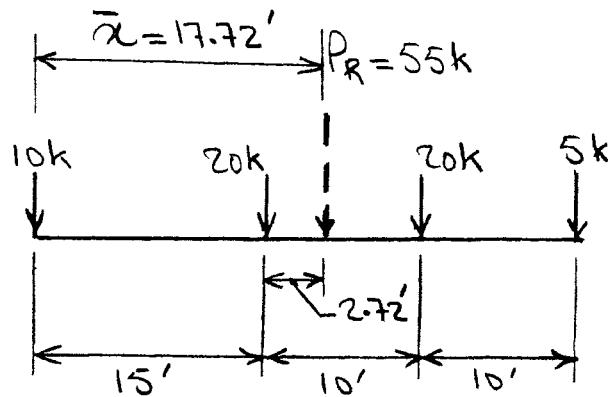


$$A_y = 140.8 \text{ kN}$$

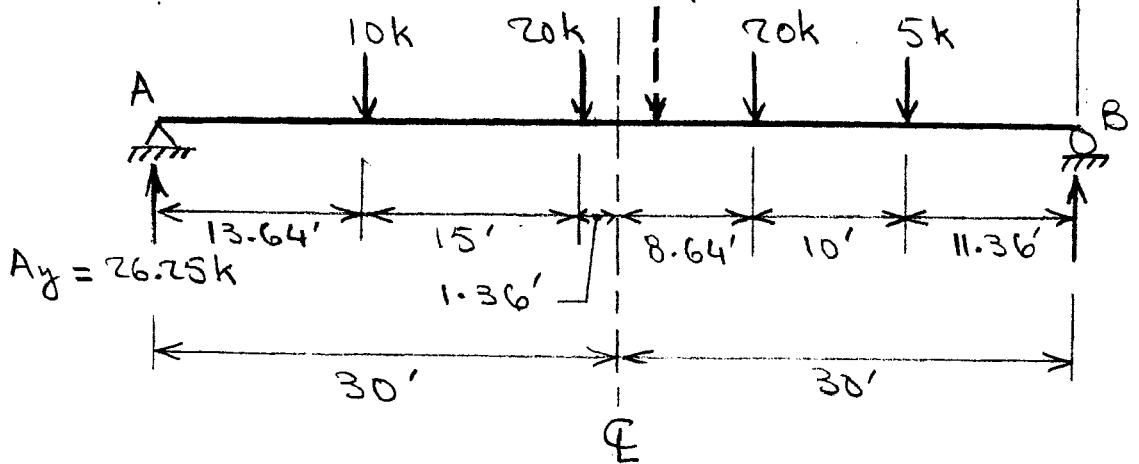
$$A_y = 275 \left(\frac{0.18 + 7.5}{15} \right) = 140.8 \text{ kN}$$

$$M_{max} = 140.8(7.5 + 0.18) - 125(2) = \underline{831.3 \text{ kN}\cdot\text{m}}$$

9.23



$$\bar{x} = \frac{20(15+25) + 5(35)}{55} \\ = 17.72'$$



$$A_y = 55(28.64/60) = 26.25k$$

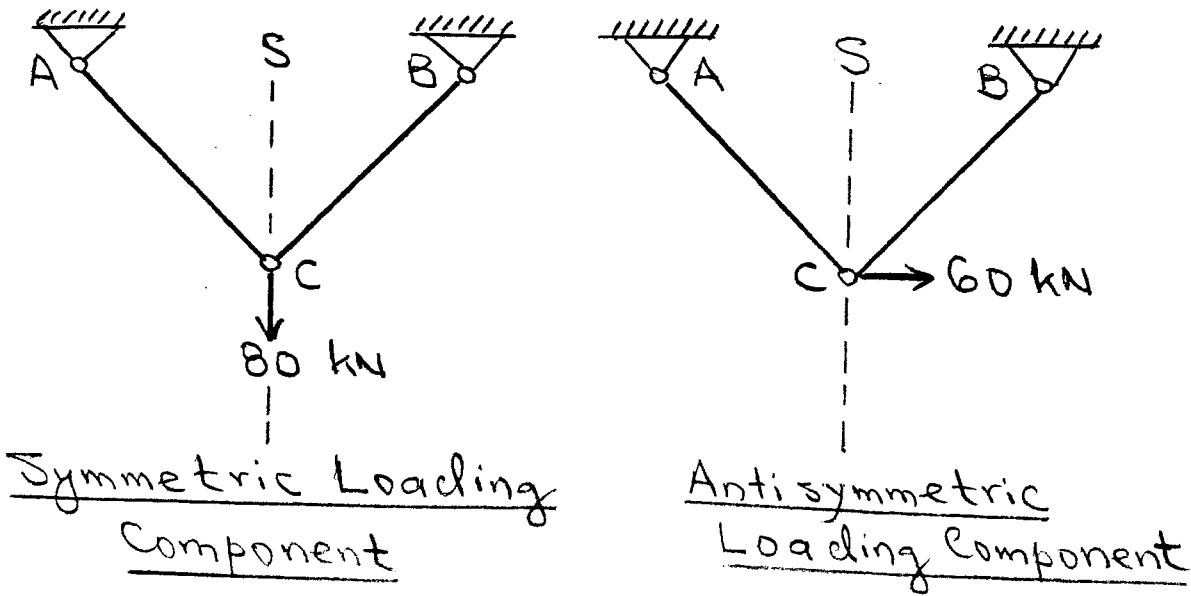
$$M_{max} = 26.25(28.64) - 10(15) = \underline{601.8 \text{ k-ft}}$$

Chapter Ten

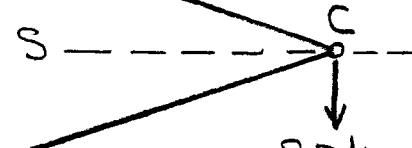
Analysis of Symmetric Structures

CHAPTER 10

10.1



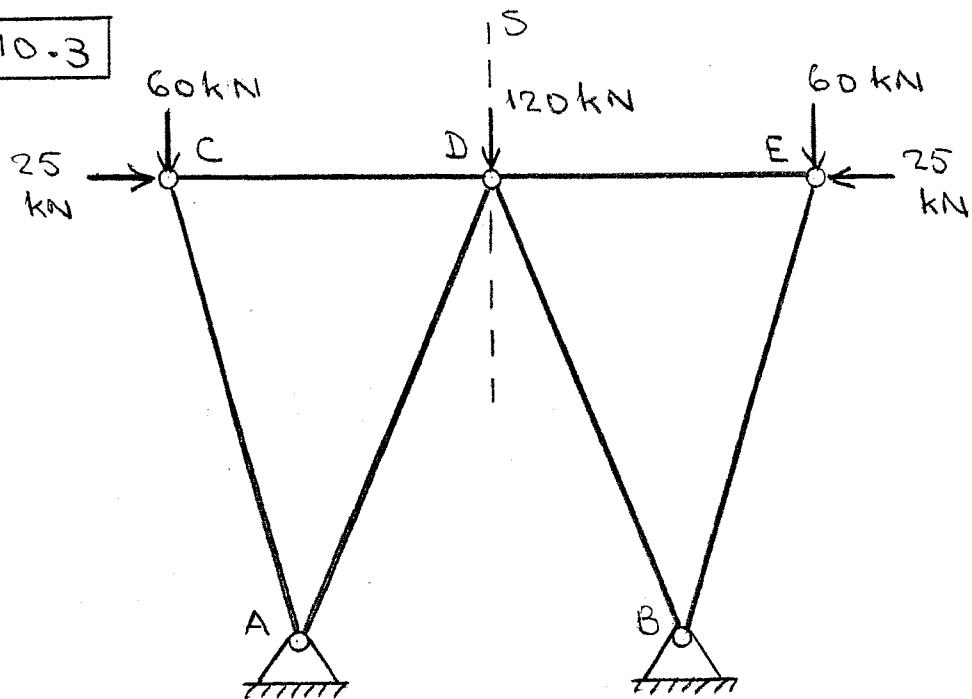
10.2



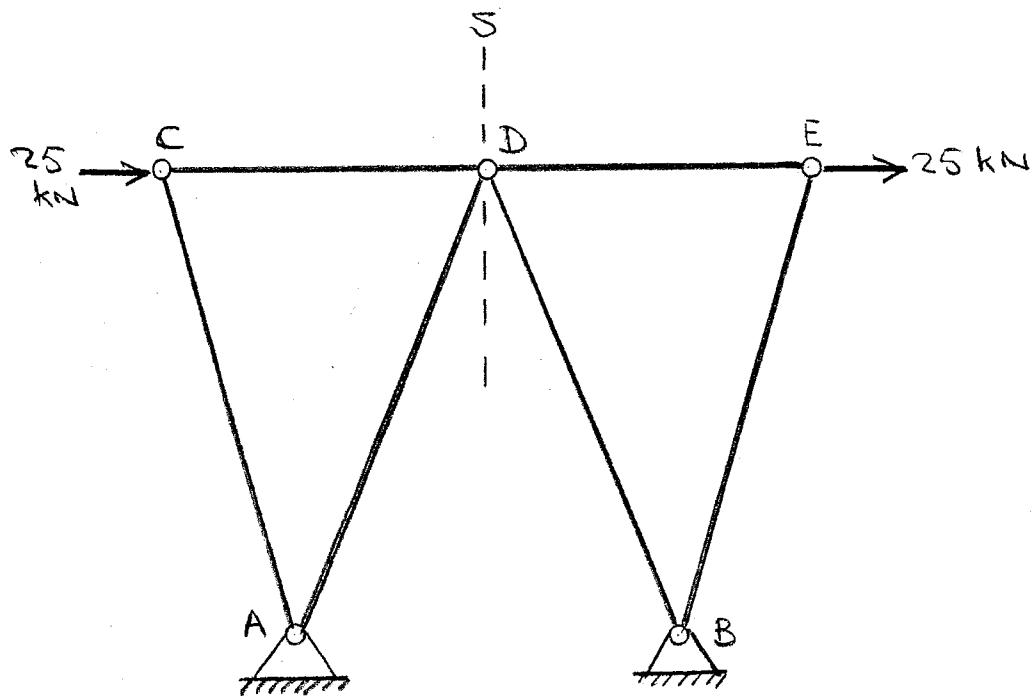
Symmetric Loading
Component

Antisymmetric Loading
Component

10.3

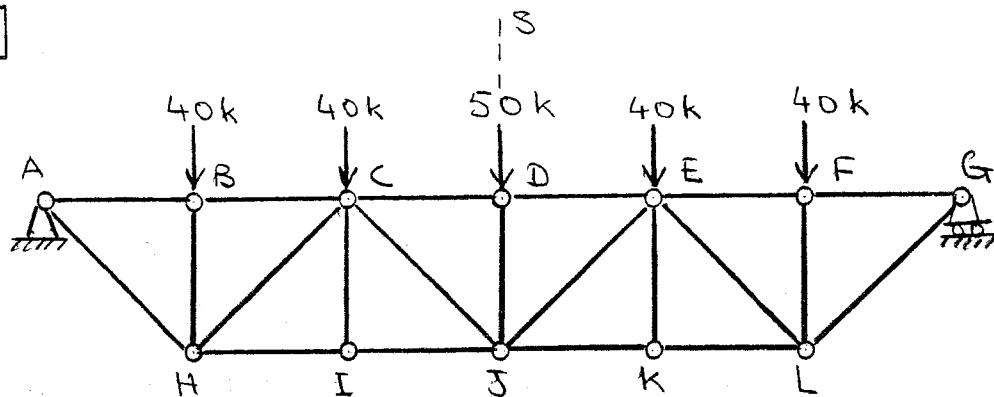


Symmetric Loading Component

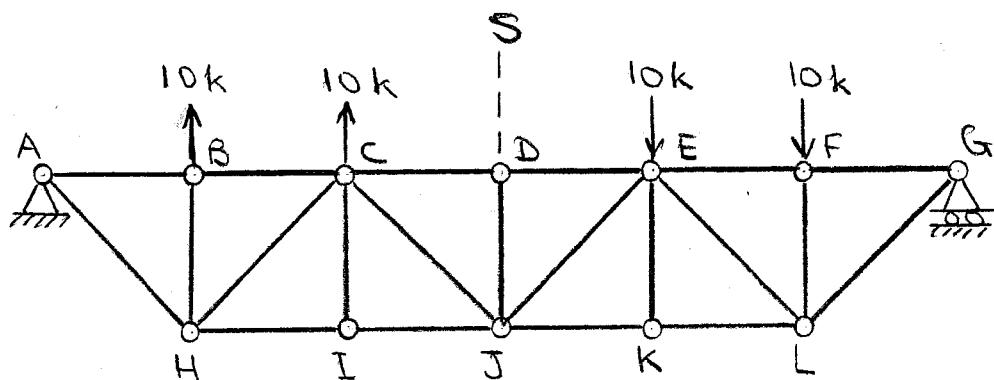


Anti-symmetric Loading Component

10.4

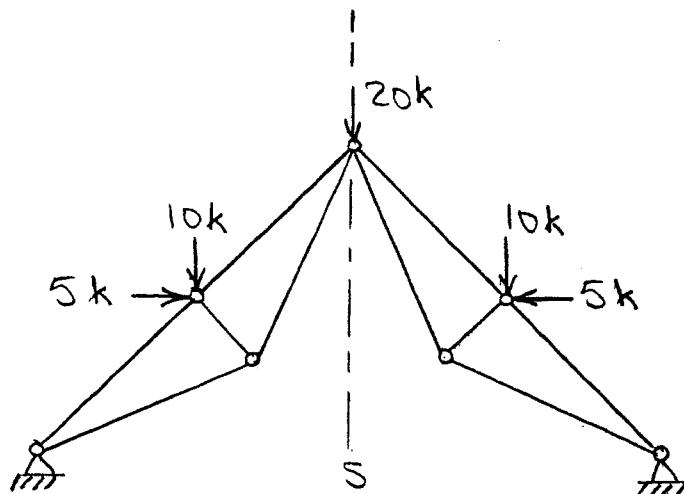


Symmetric Loading Component

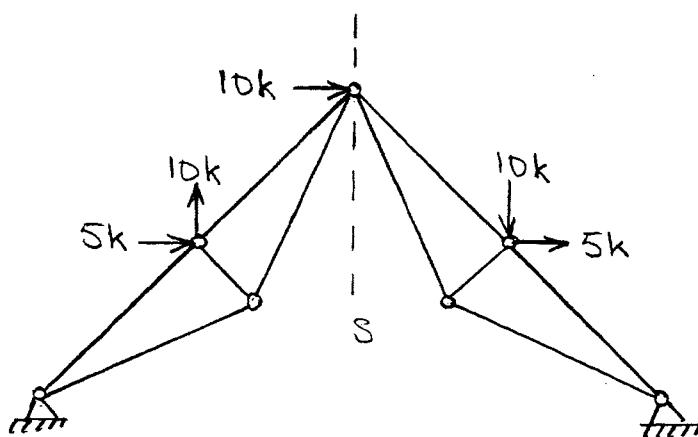


Antisymmetric Loading Component

10.5

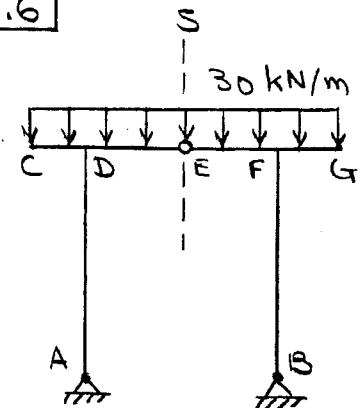


Symmetric Loading Component

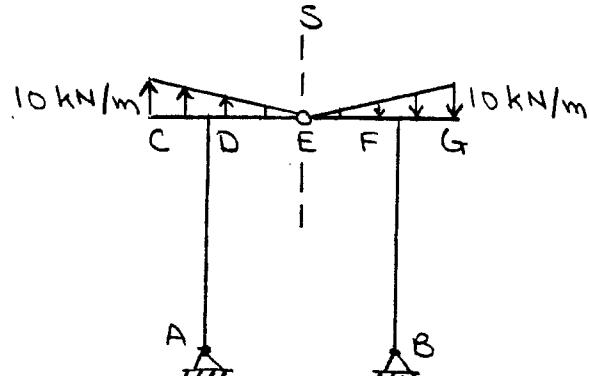


Antisymmetric Loading Component

10.6

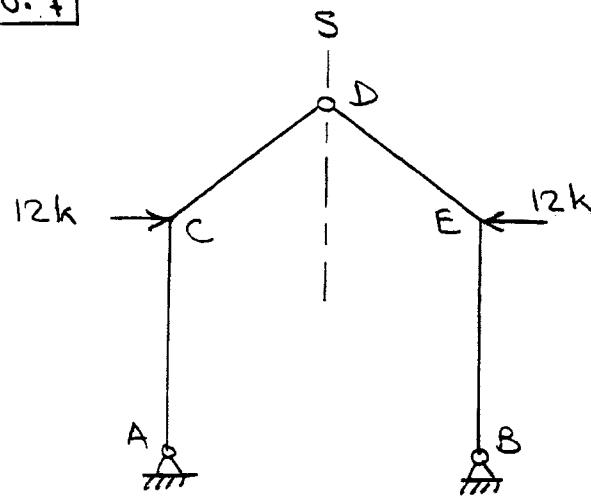


Symmetric Loading Component

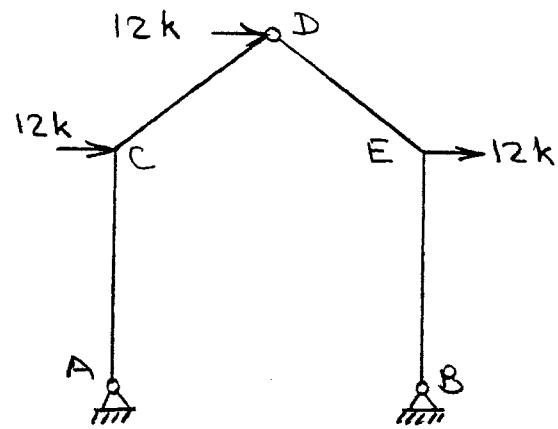


Antisymmetric Loading Component

10.7

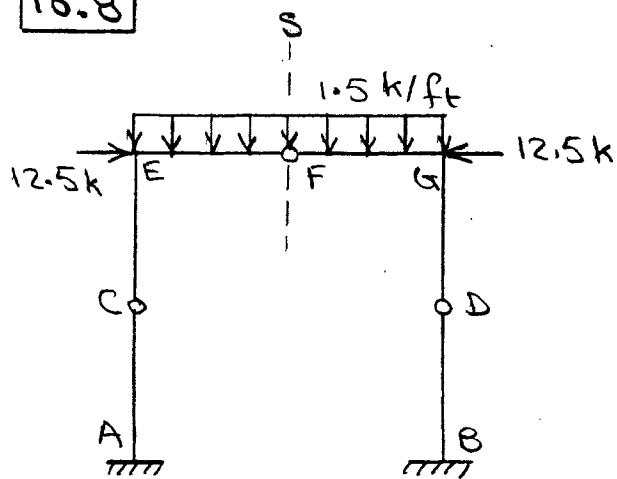


Symmetric Loading Component

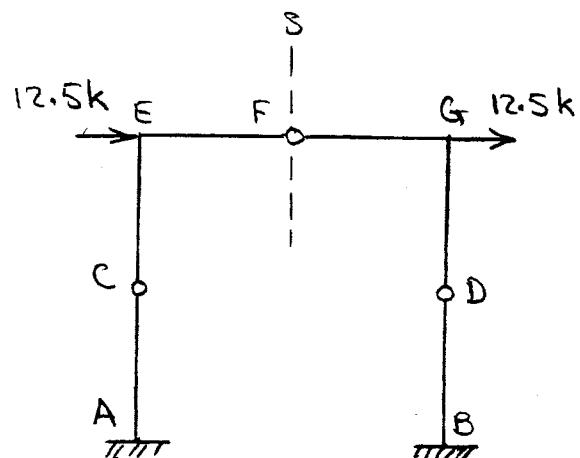


Antisymmetric Loading Component

10.8

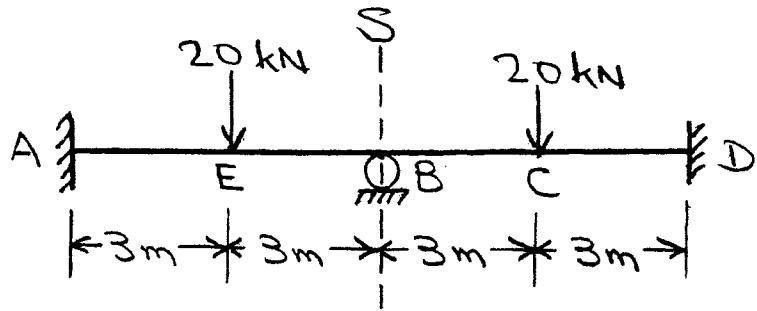


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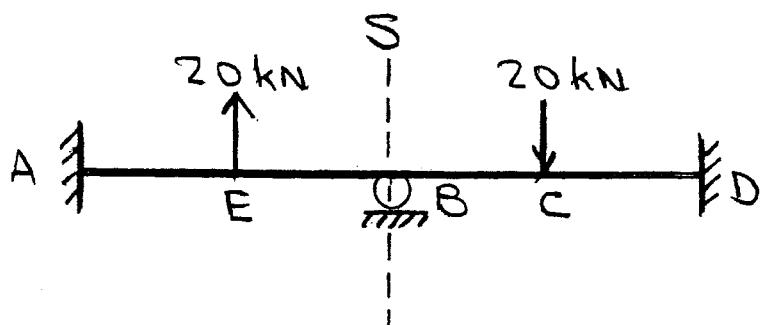


Anti-symmetric Loading Component

10.9

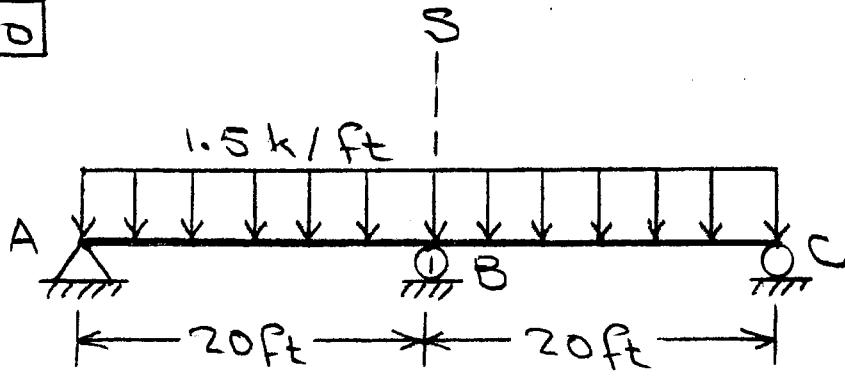


Symmetric Loading Component

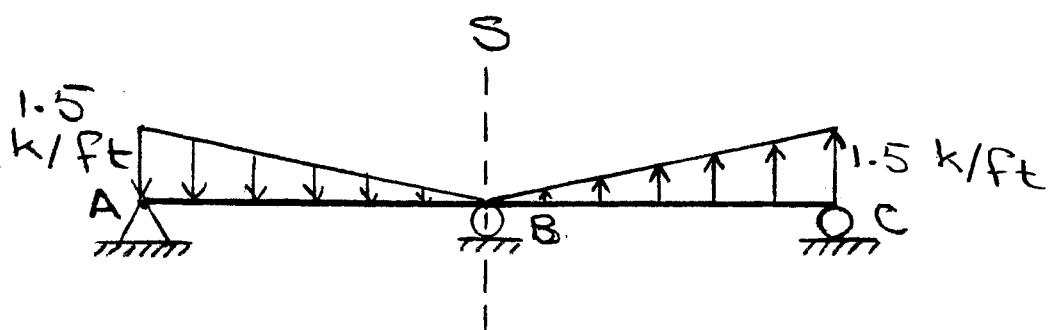


Antisymmetric Loading Component

10.10

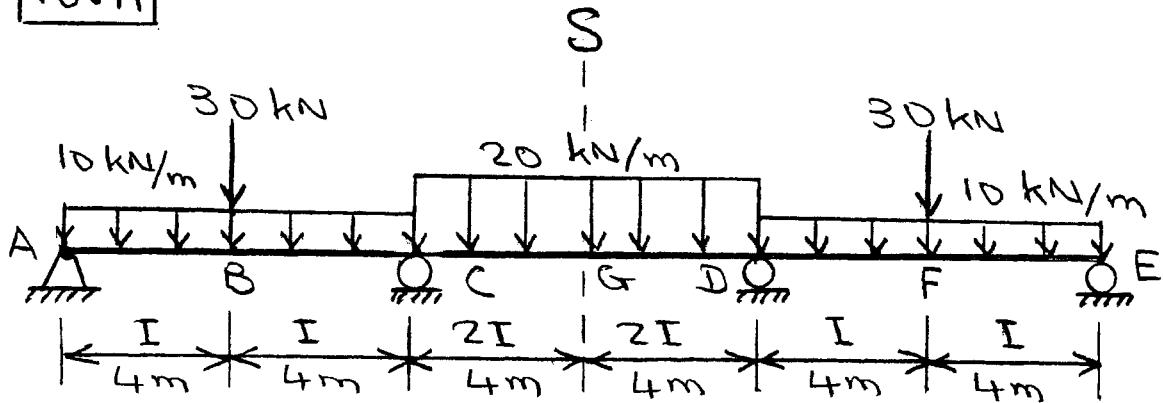


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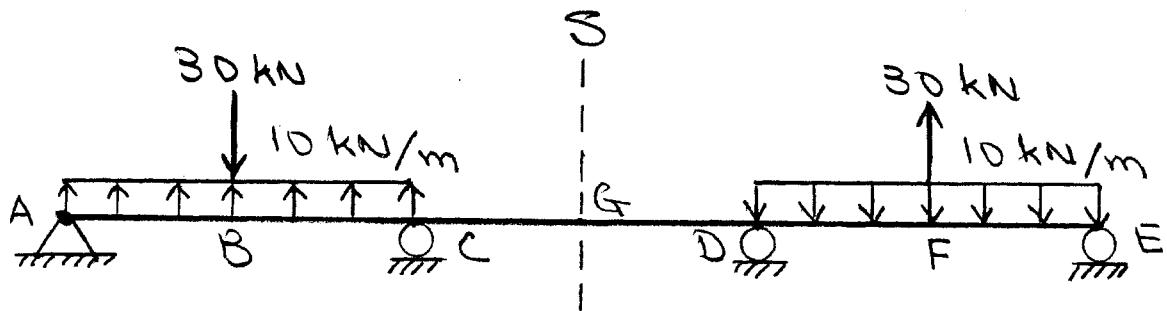


Anti-symmetric Loading Component

10.11

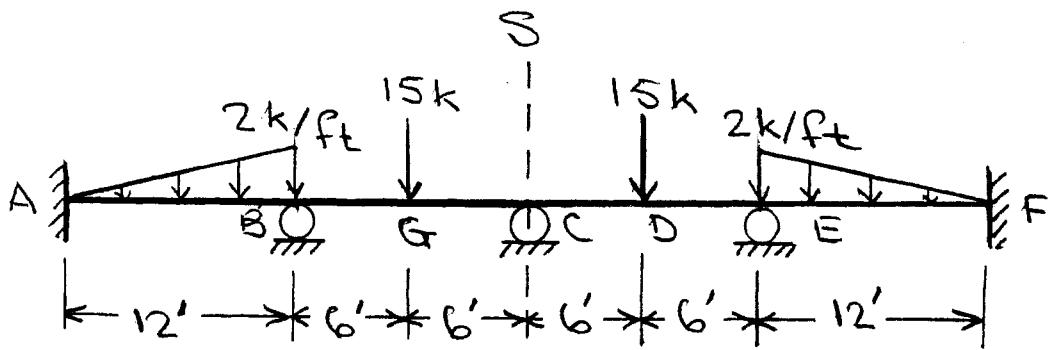


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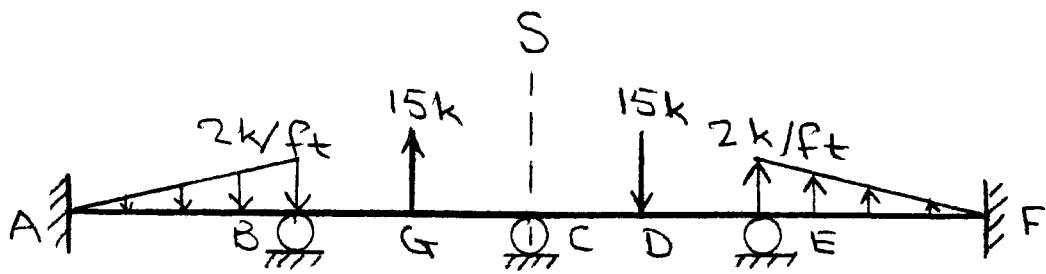


Antisymmetric Loading Component

10.12

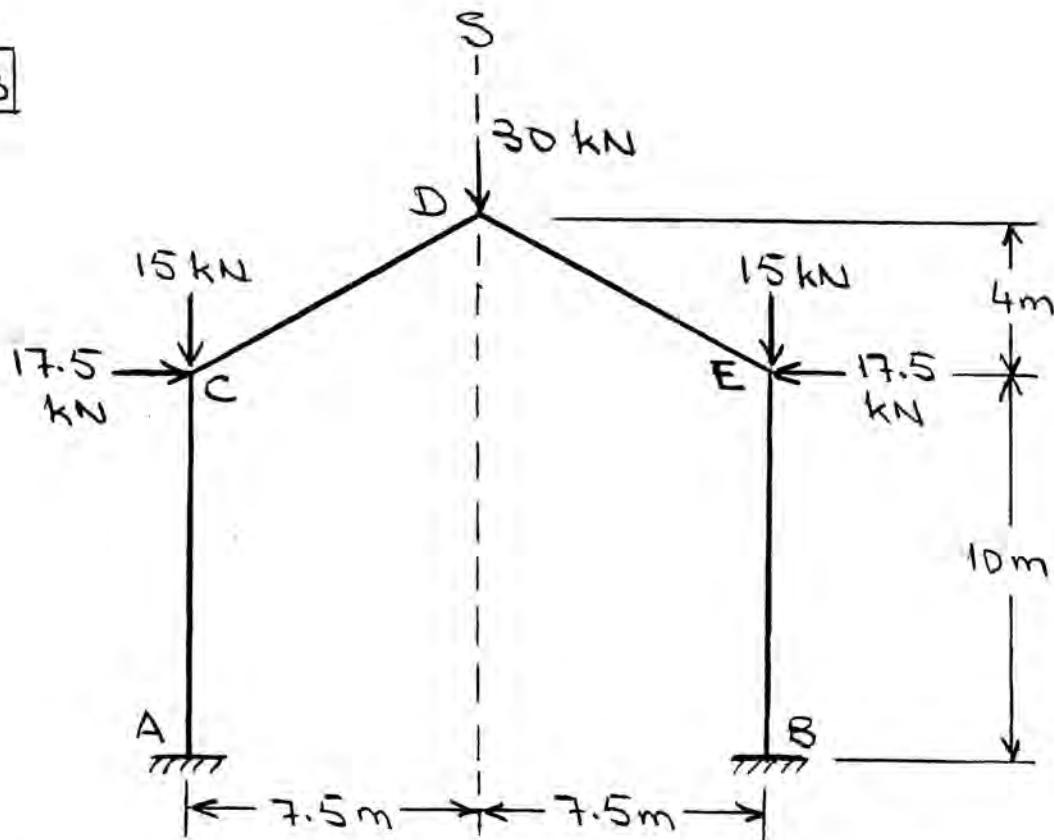


Symmetric Loading Component

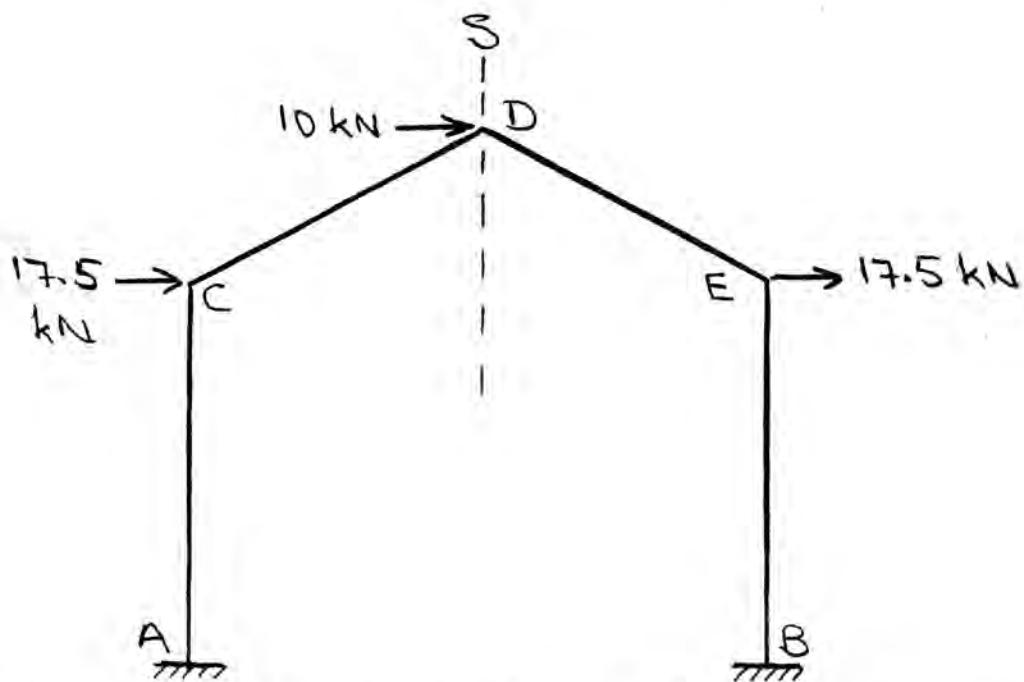


Antisymmetric Loading Component

10-13

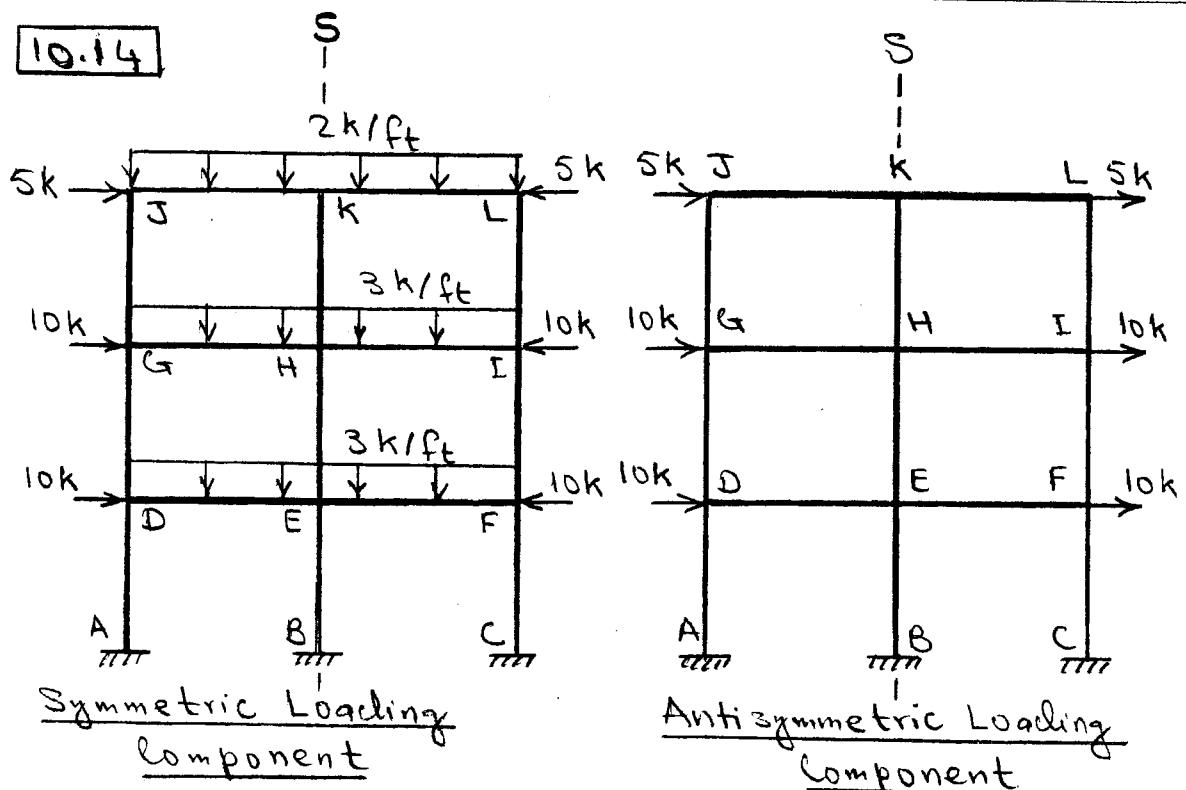


Symmetric Loading Component

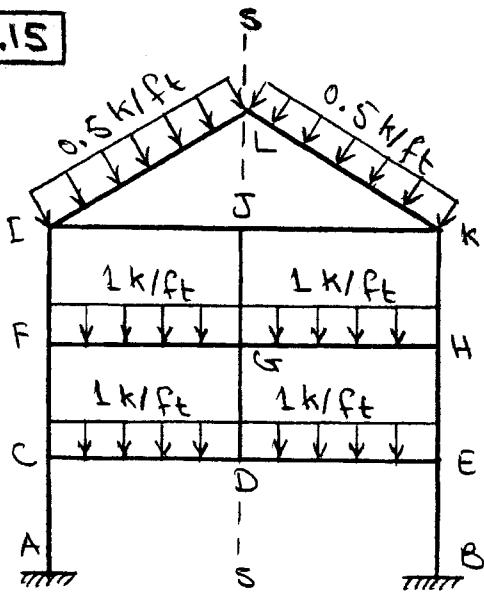


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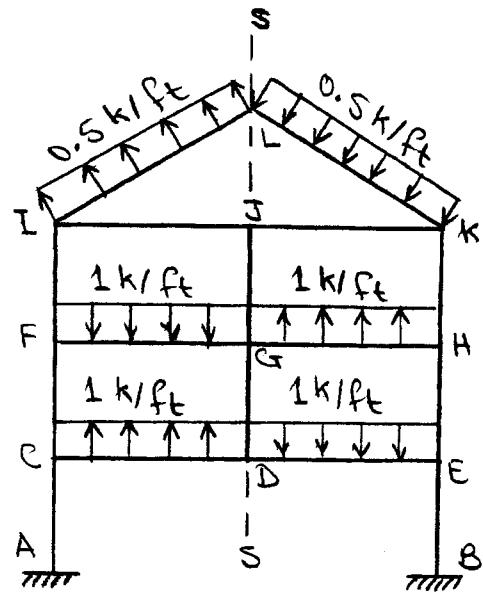
10.14



10.15

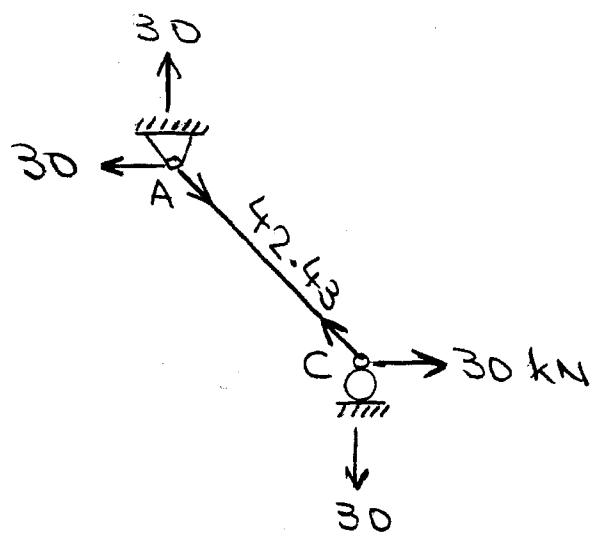
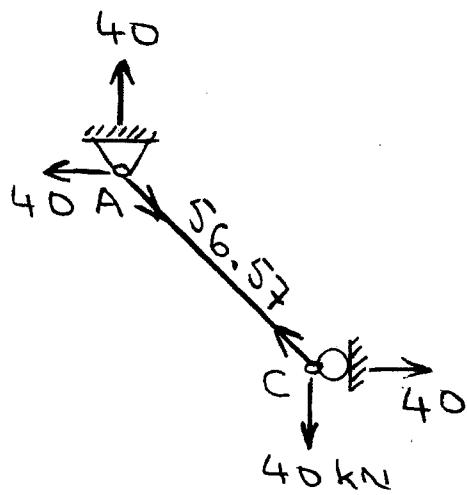


Symmetric Loading Component

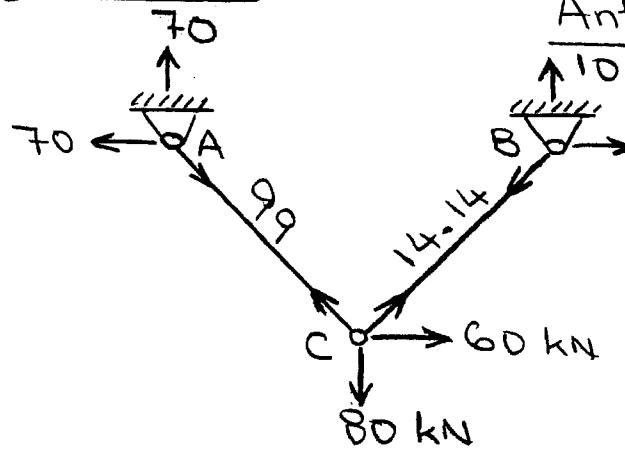


Anti-symmetric Loading Component

10.16



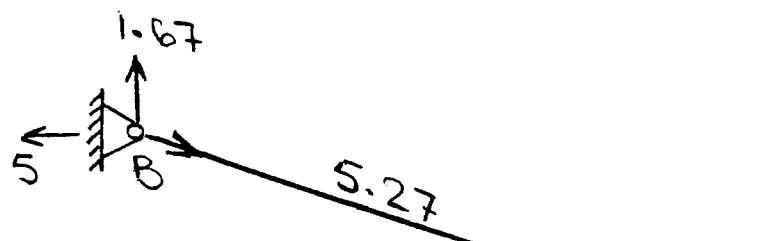
Symmetric



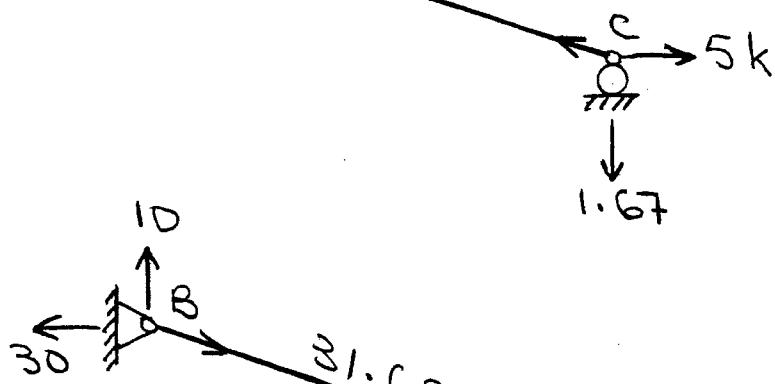
Anti-symmetric

Member Forces

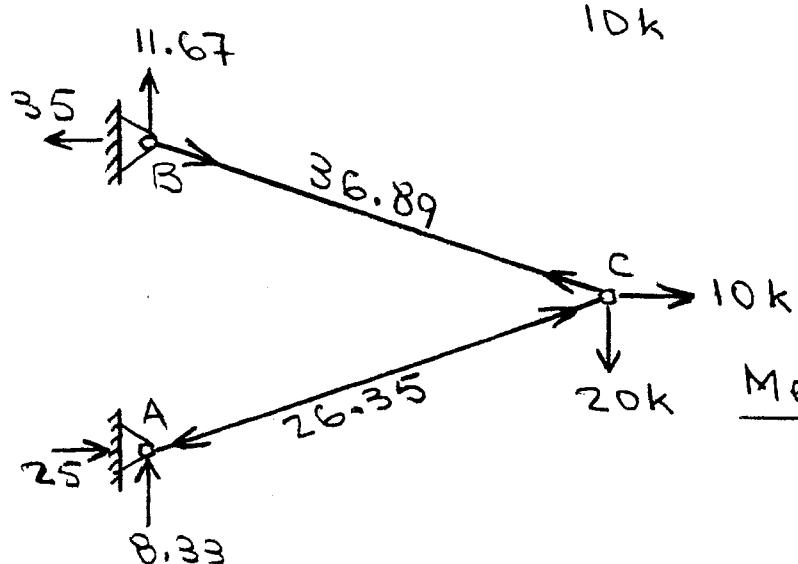
10.17



Symmetric



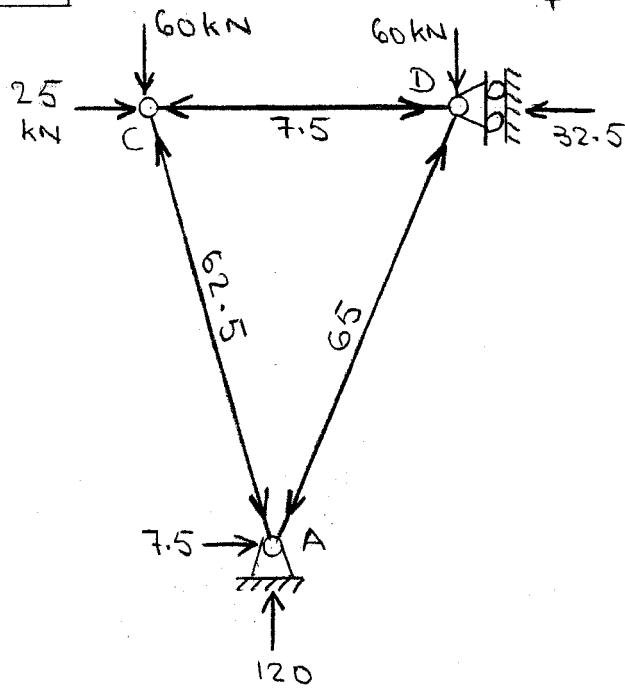
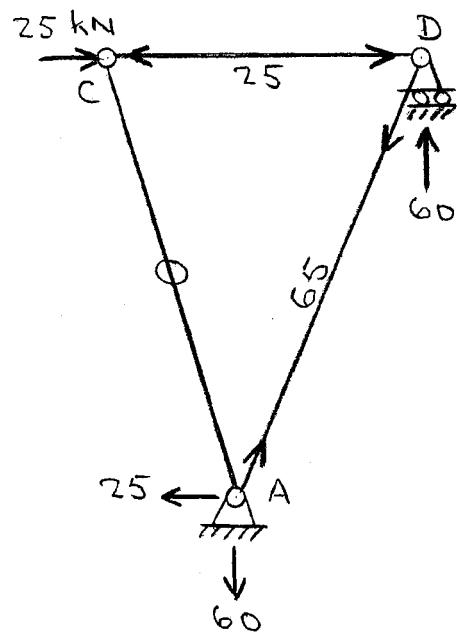
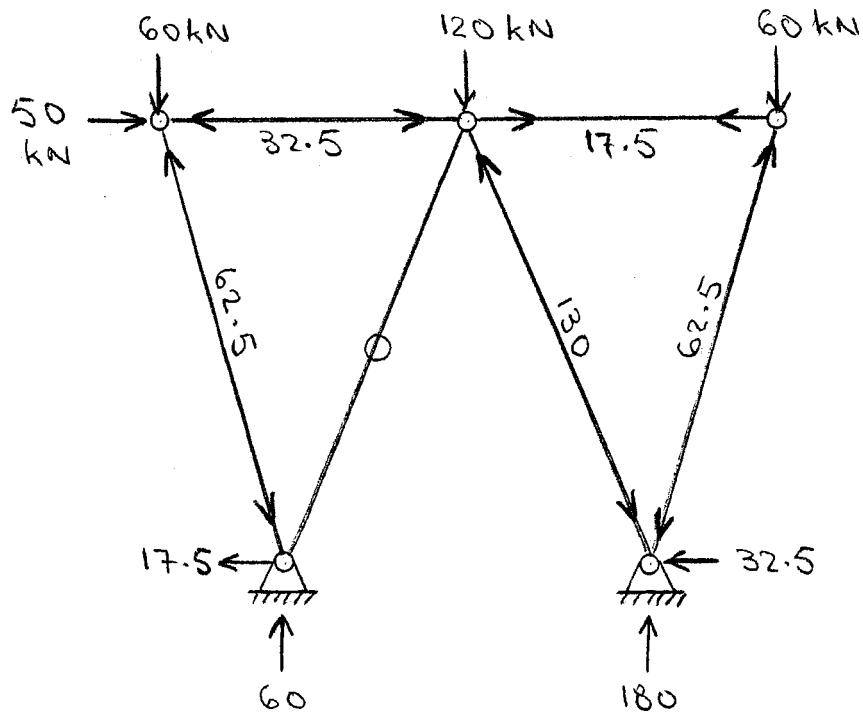
Antisymmetric



Member Forces

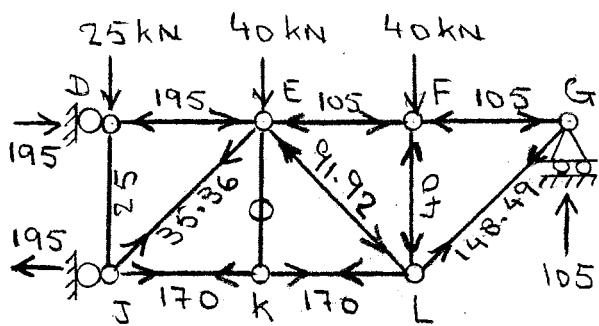
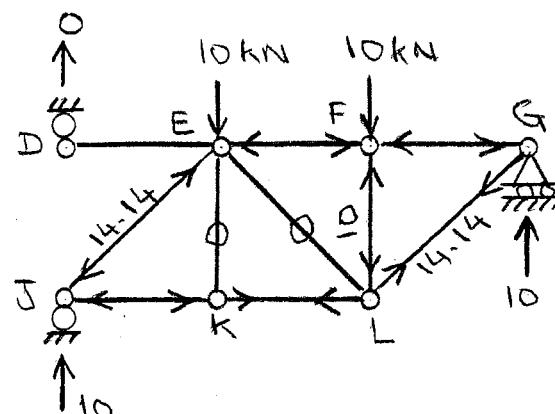
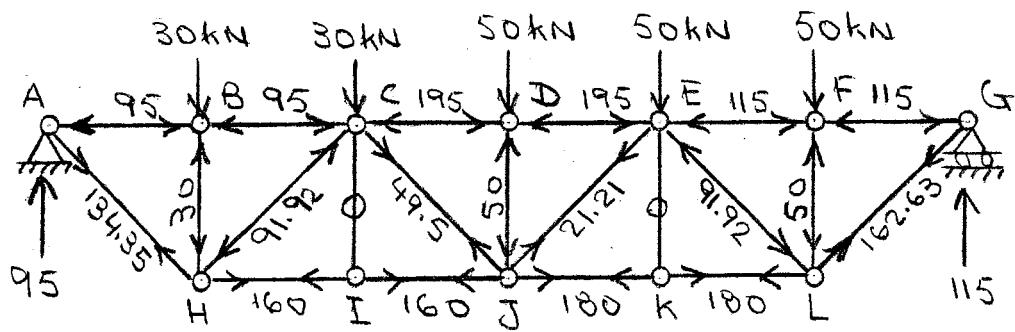
10.18

See Solution of Problem 10.3

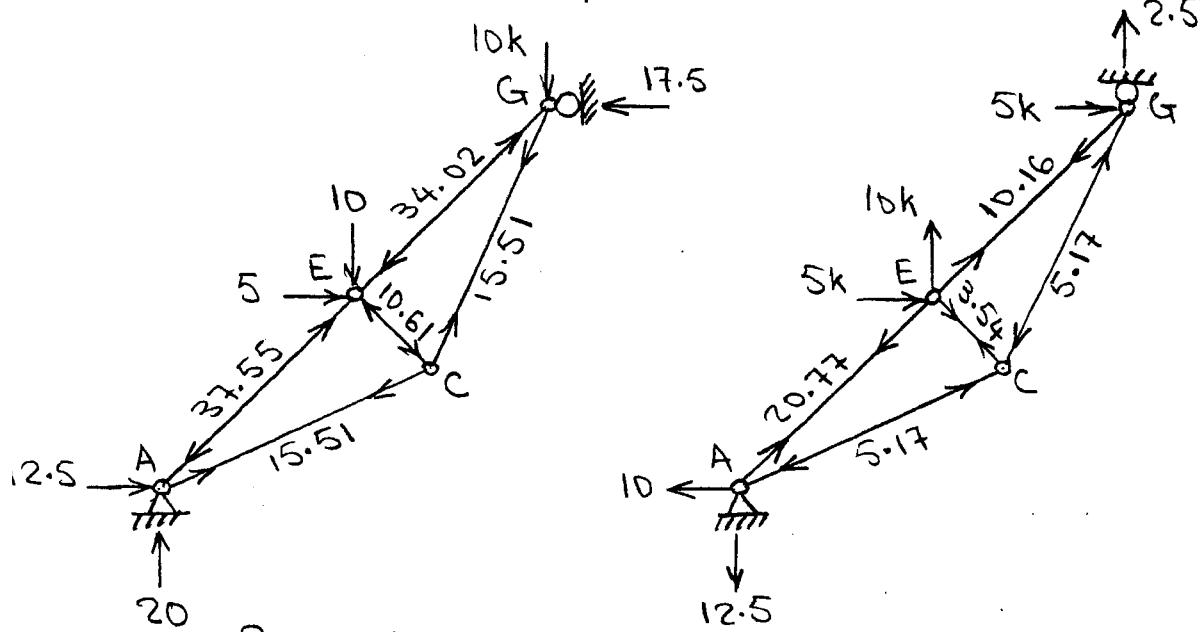
SymmetricAntisymmetricMember Forces

10-19

See solution of Problem 10.4.

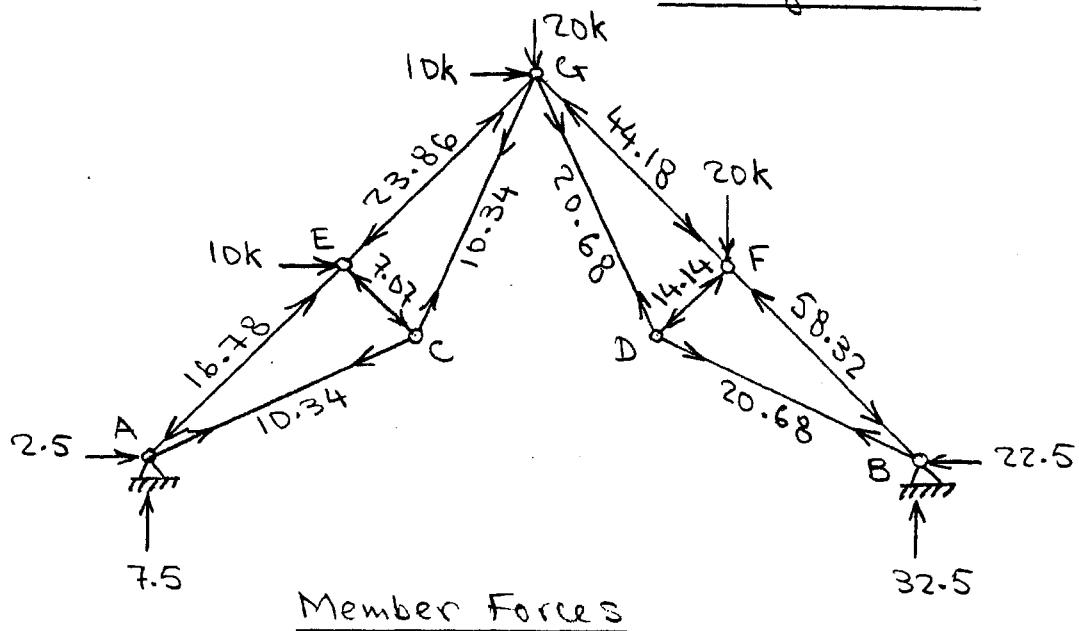
SymmetricAnti-SymmetricMember Forces

10.20 See solution of Problem 10.5.



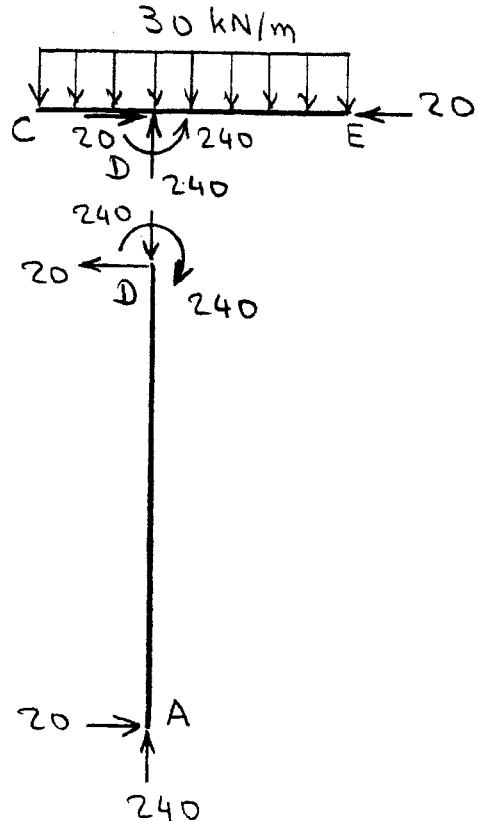
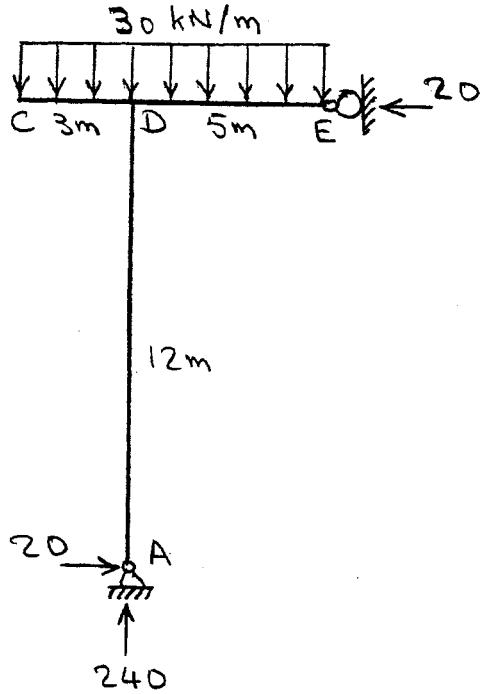
Symmetric

Antisymmetric

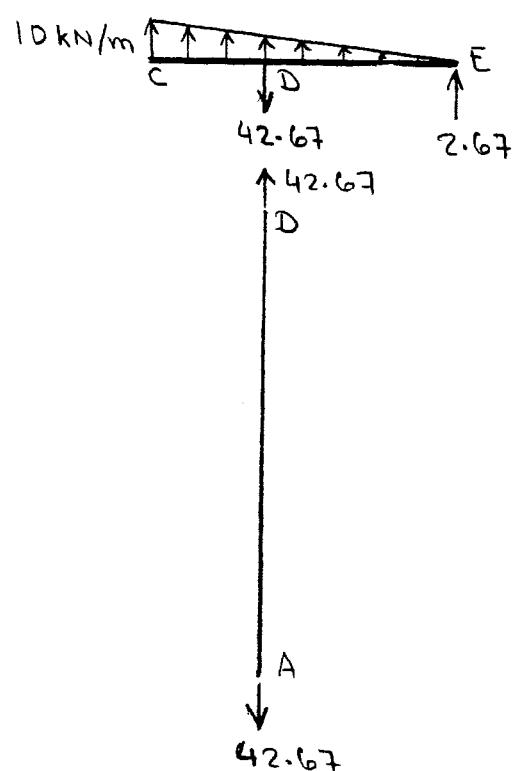
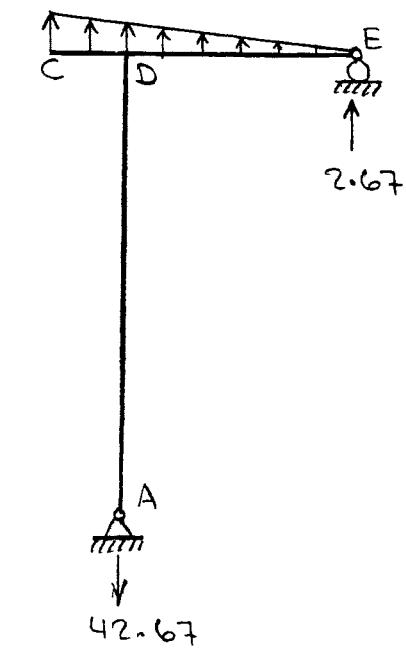


Member Forces

10.21 See solution of Problem 10.6.

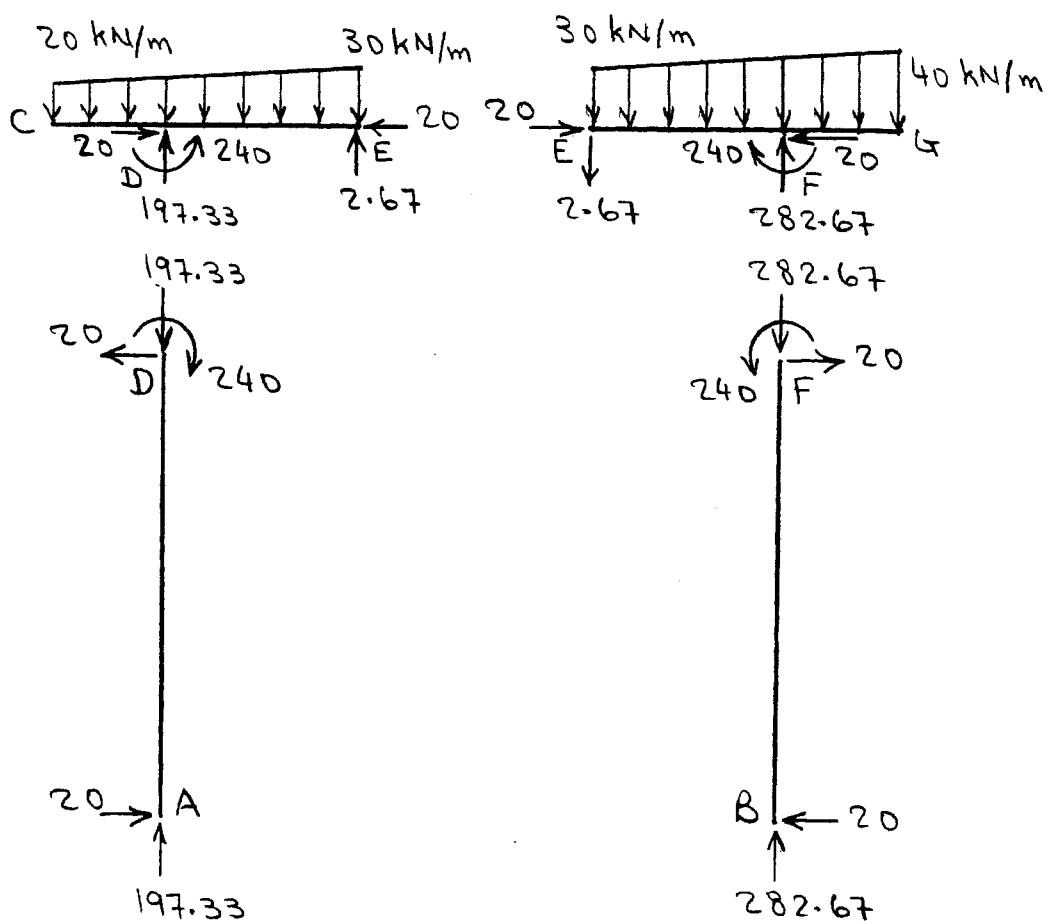


Symmetric



Anti-symmetric

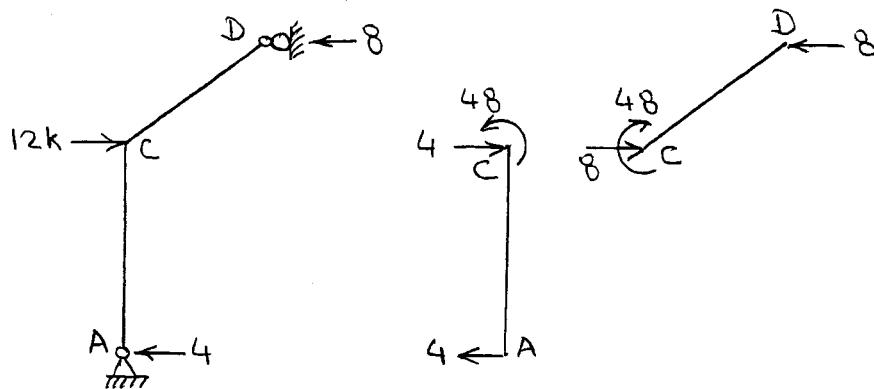
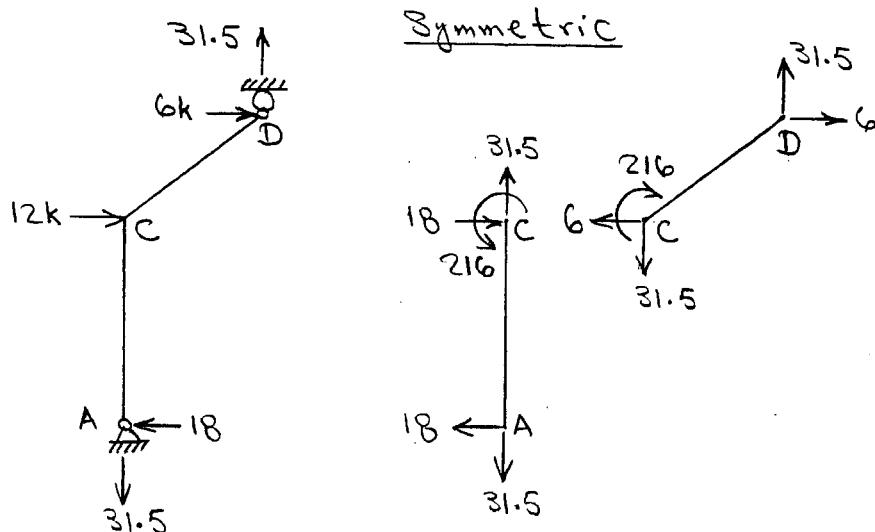
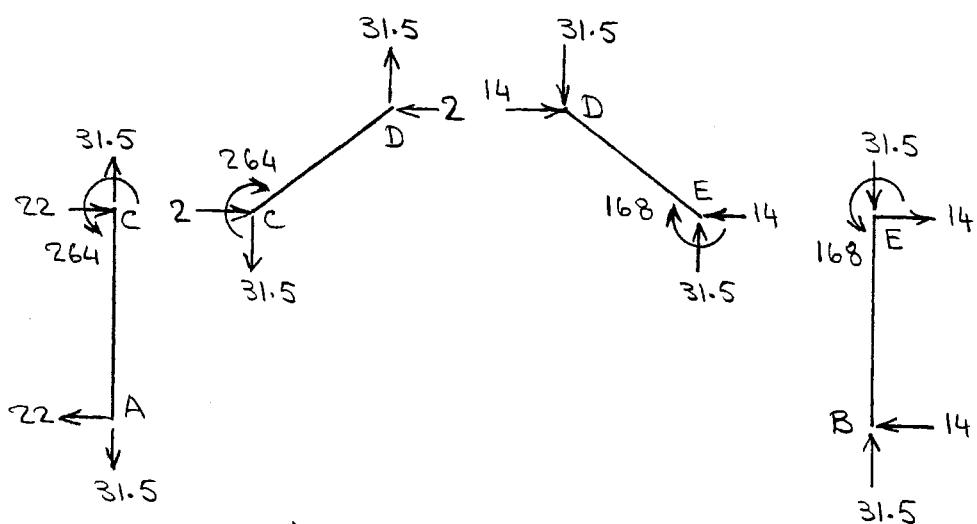
10.21 (Contd.)



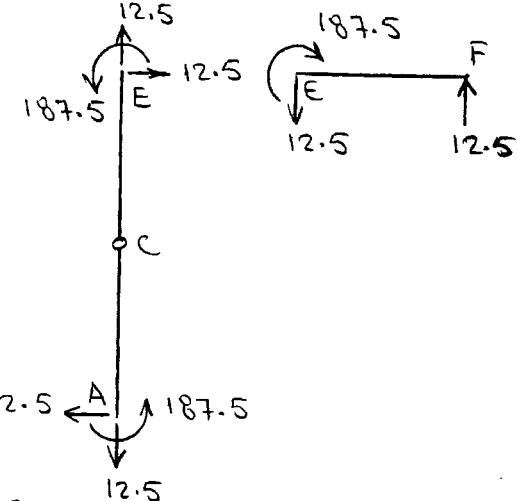
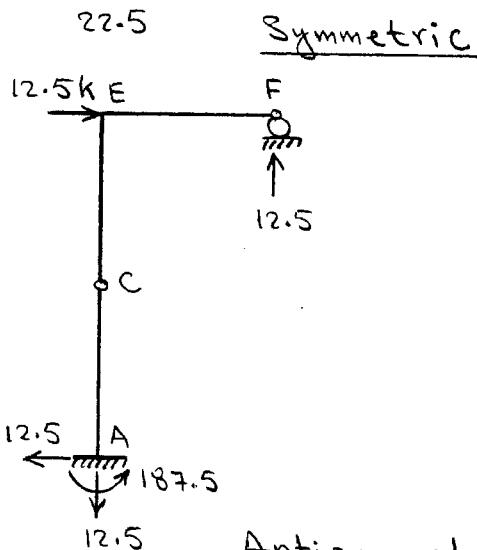
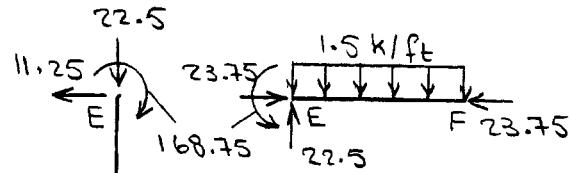
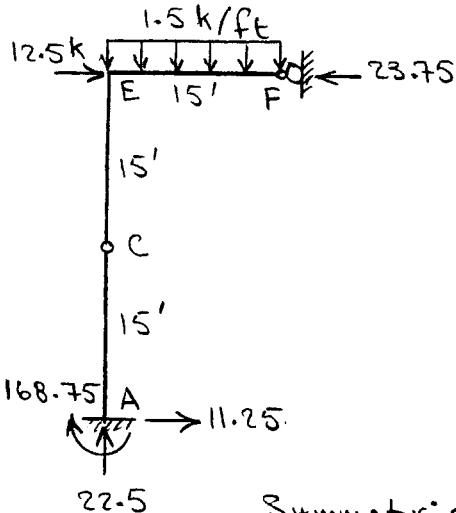
Member End Forces

10.22

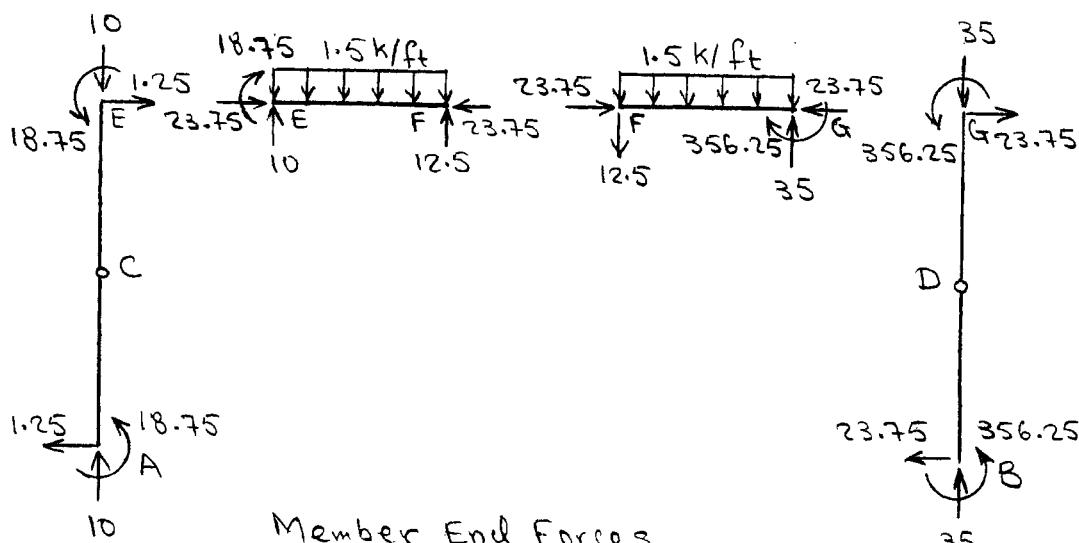
See solution of Problem 10.7.

SymmetricAntisymmetricMember End Forces

10.23 See solution of Problem 10.8.



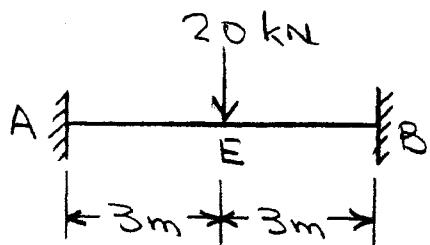
Antisymmetric



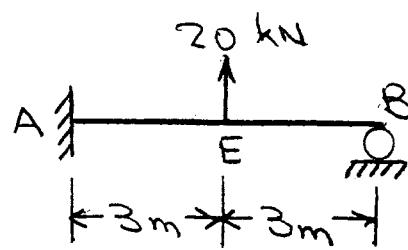
Member End Forces

10.24

See solution of Problem 10.9.



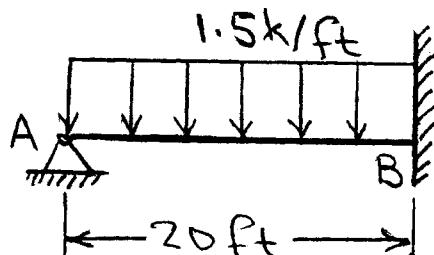
Symmetric



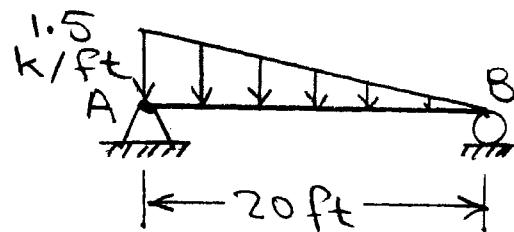
Antisymmetric

10.25

See solution of Problem 10.10.



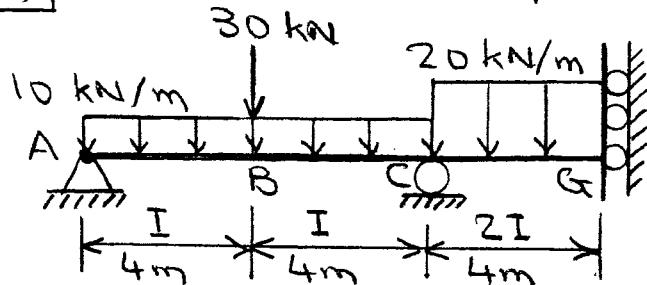
Symmetric



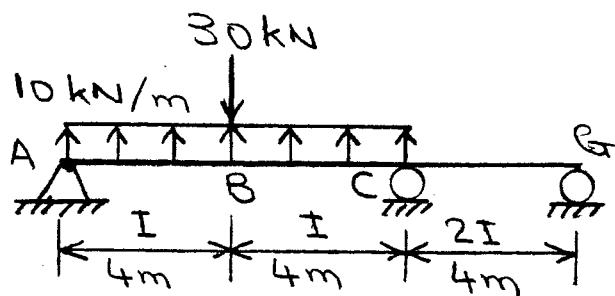
Antisymmetric

10.26

See solution of Problem 10.11



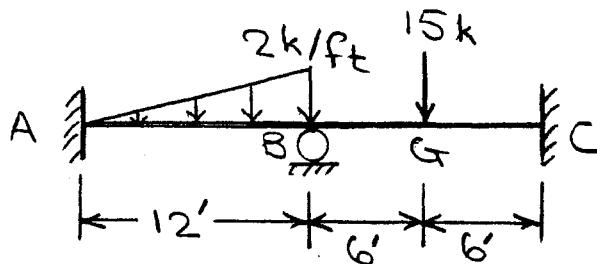
Symmetric



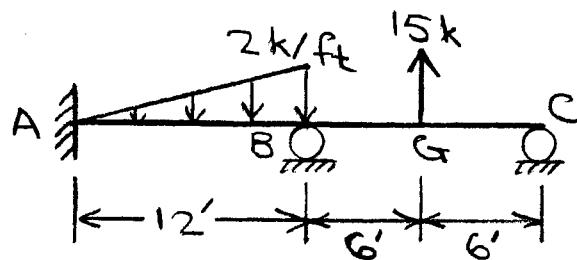
Antisymmetric

10.27

See solution of Problem 10.2.



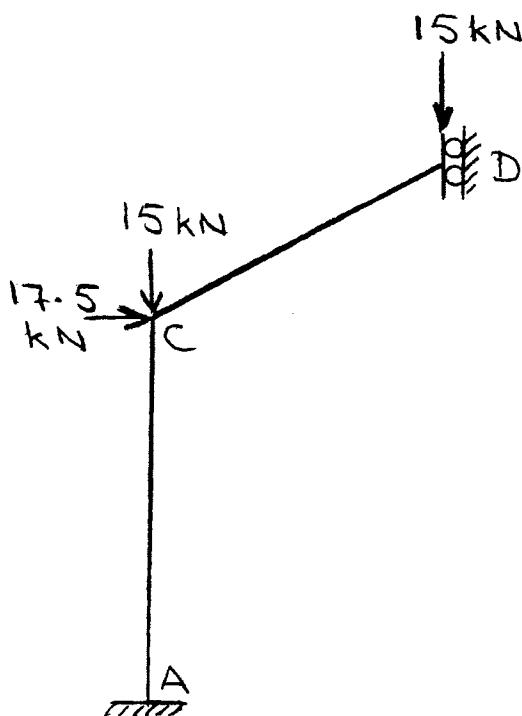
Symmetric



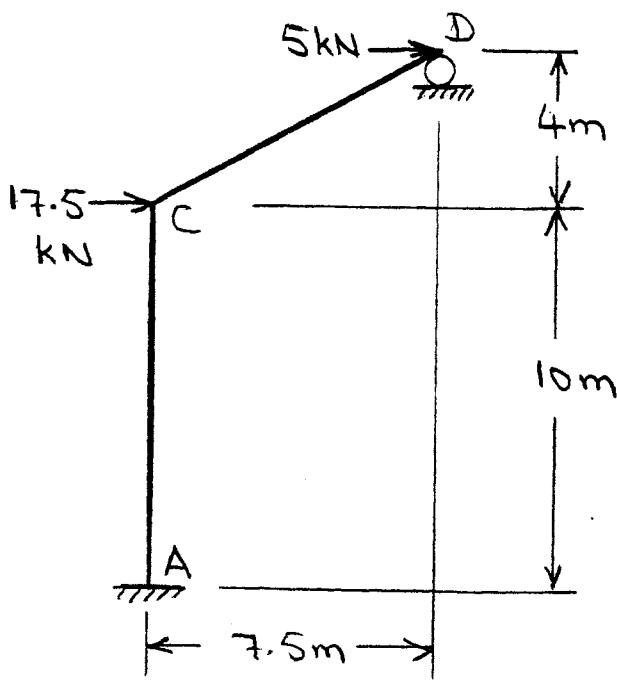
Antisymmetric

10.28

See solution of Problem 10.13.

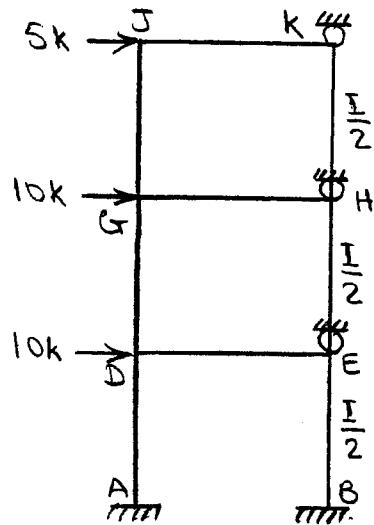
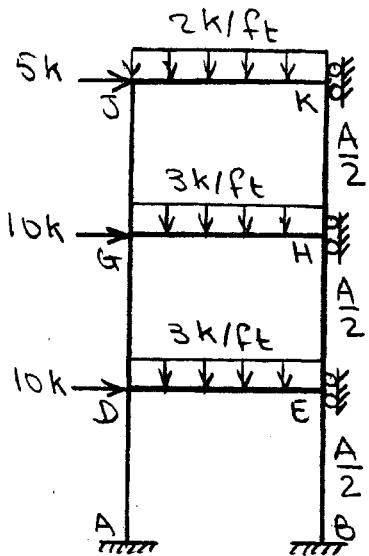


Symmetric

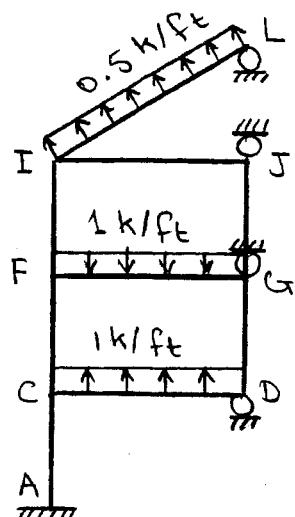
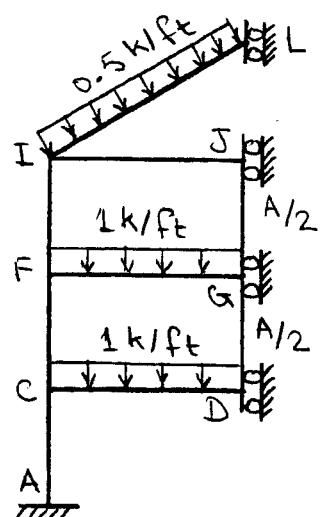


Antisymmetric

10.29 See solution of Problem 10.14.



10.30 See solution of Problem 10.15.

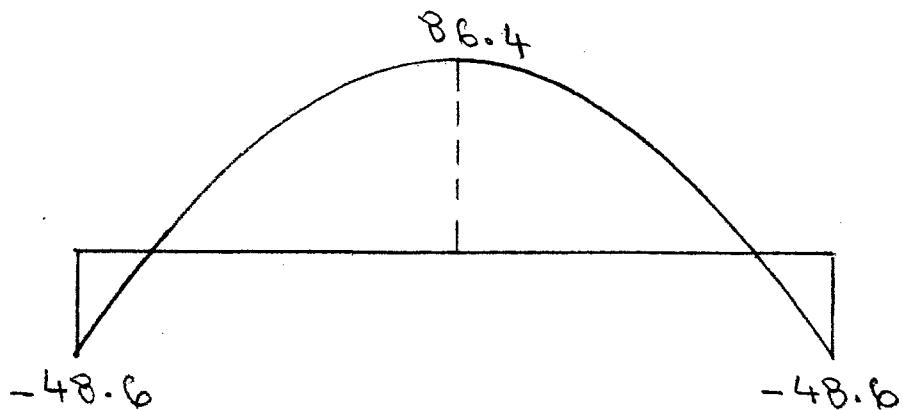
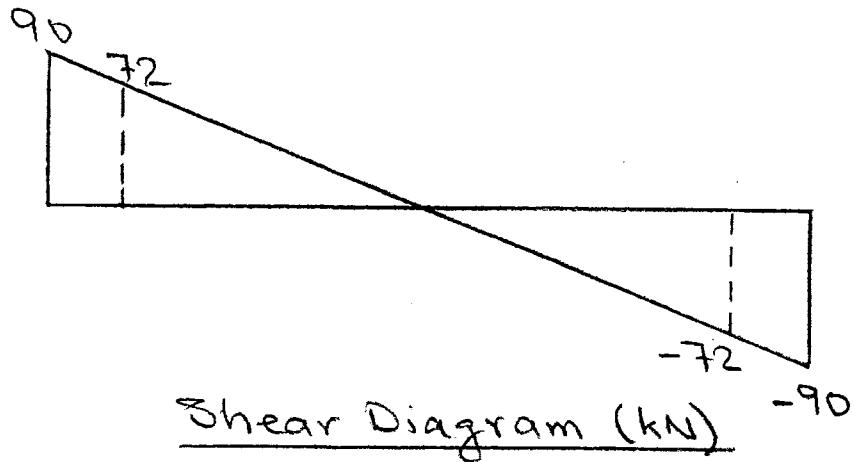
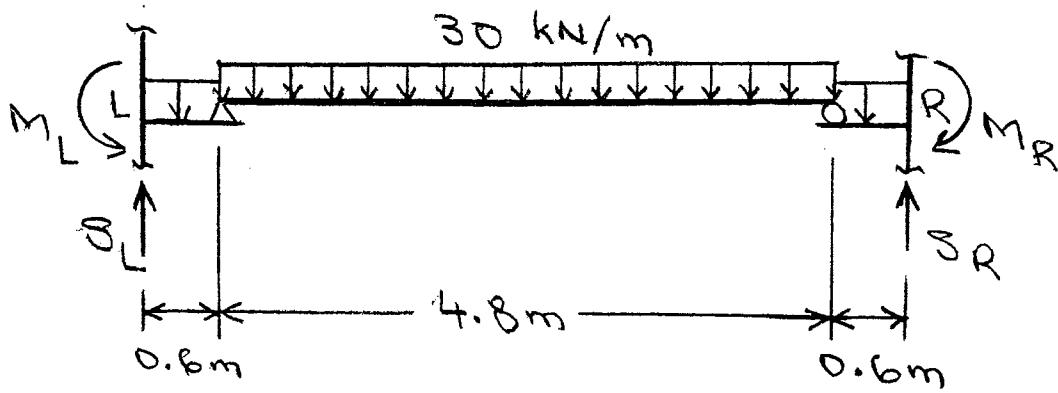


Chapter Twelve

Approximate Analysis of Rectangular Building Frames

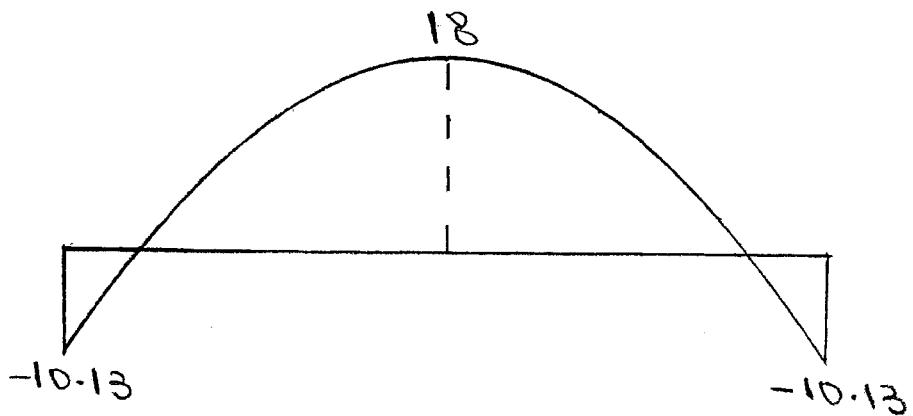
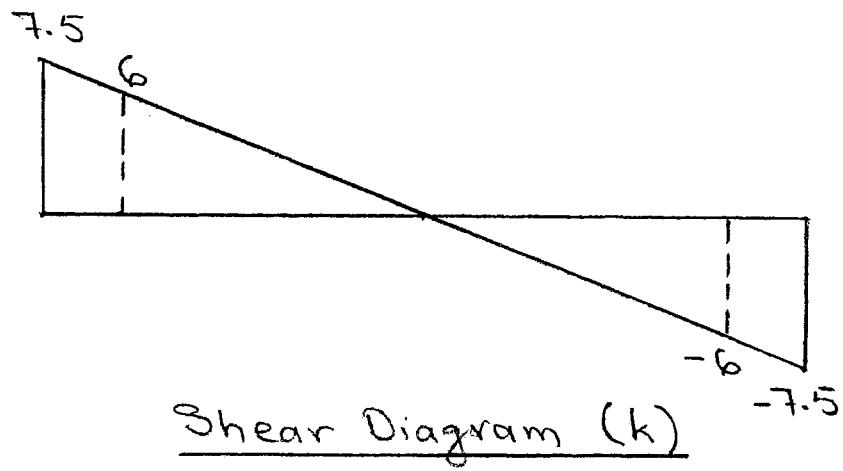
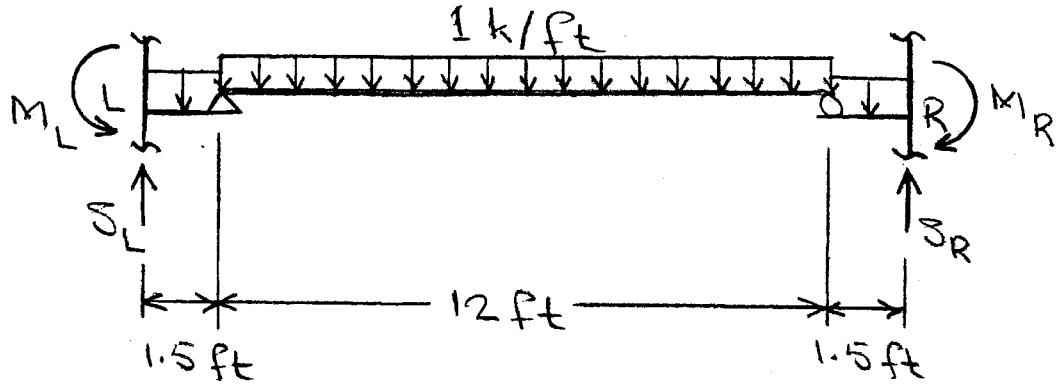
CHAPTER 12

12-1

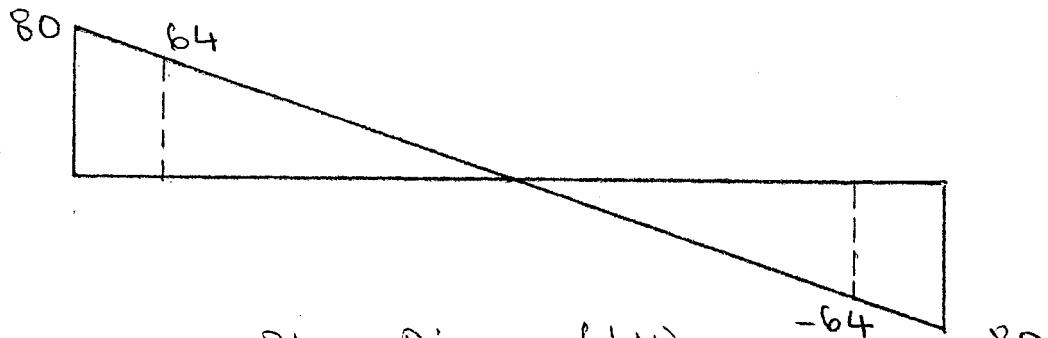
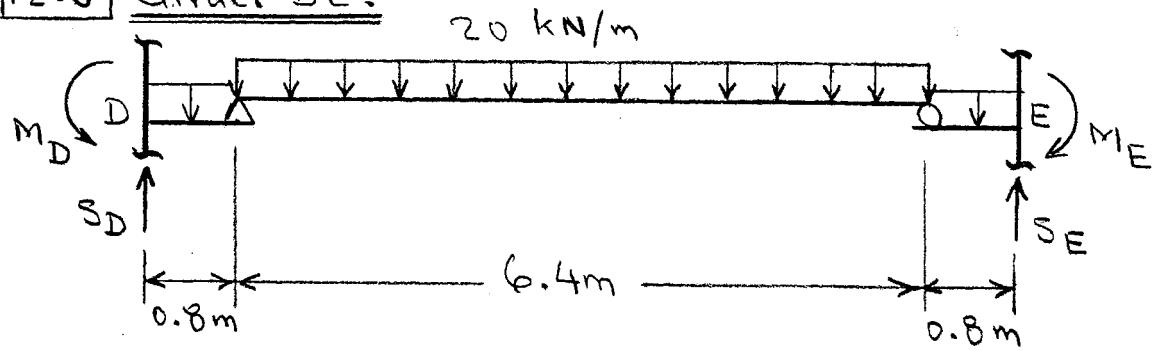


Bending Moment Diagram (kN.m)

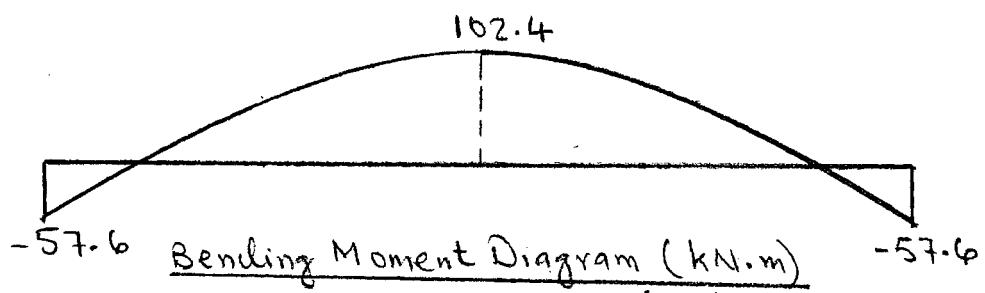
12.2



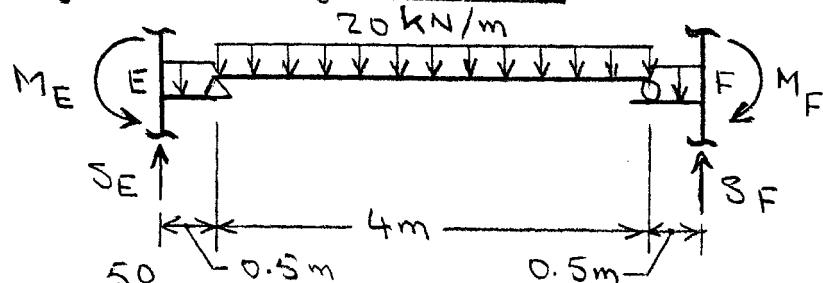
12.3 Girder DE:



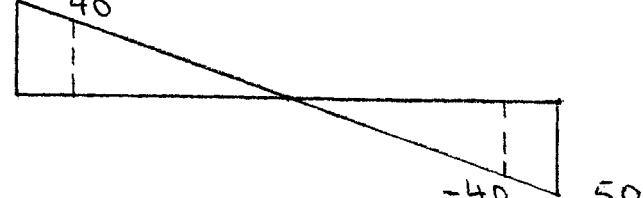
Shear Diagram (kN)



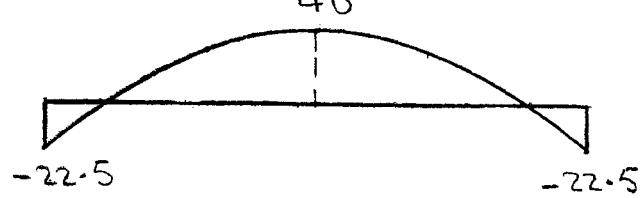
Girder EF:



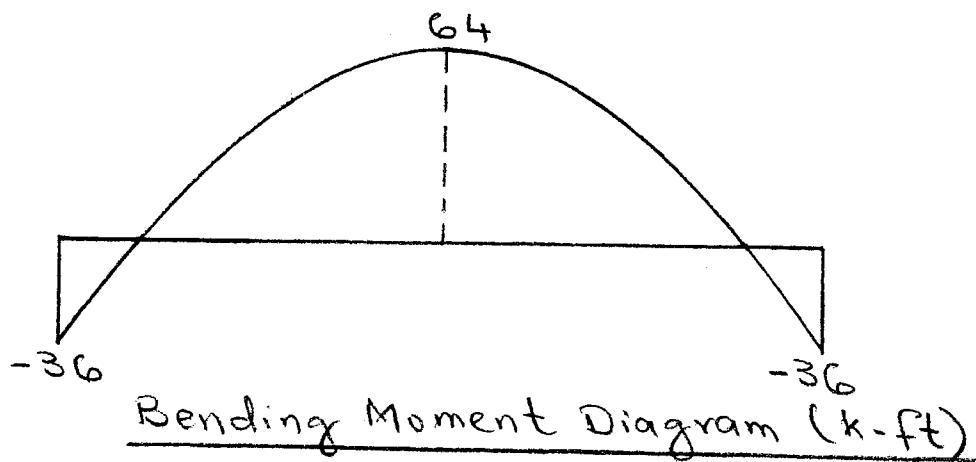
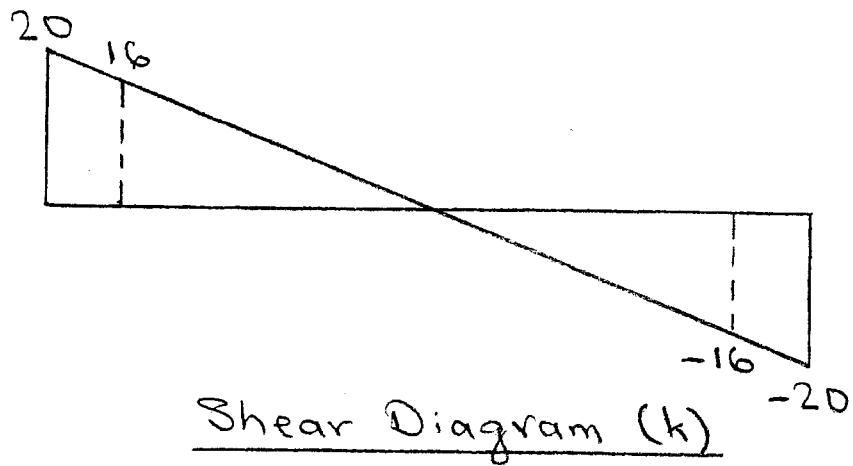
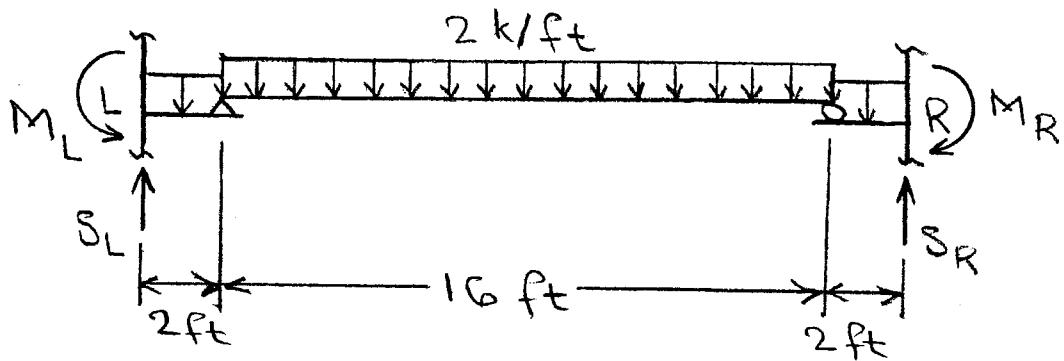
Shear Diagram (kN)



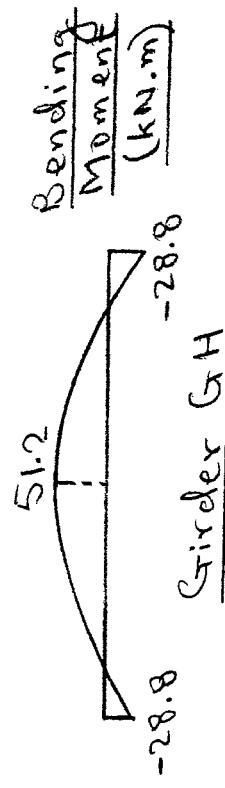
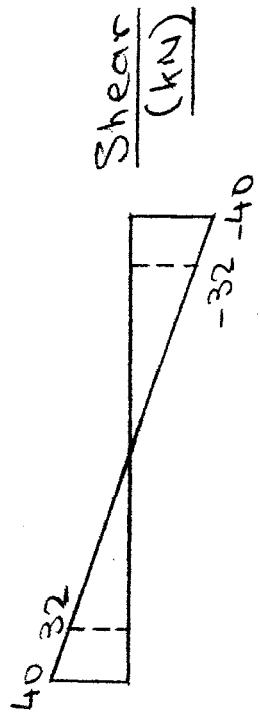
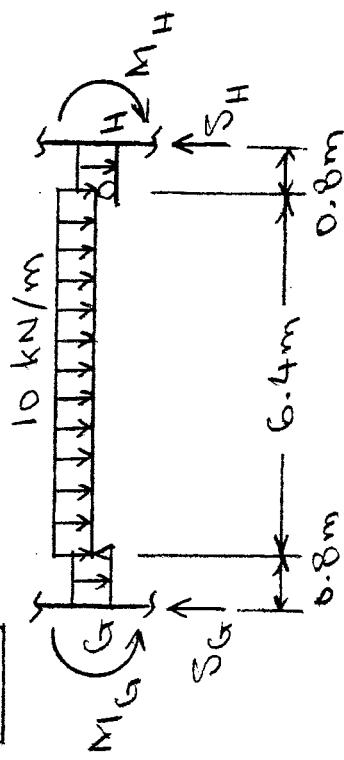
Bending Moment Diagram (kN.m)



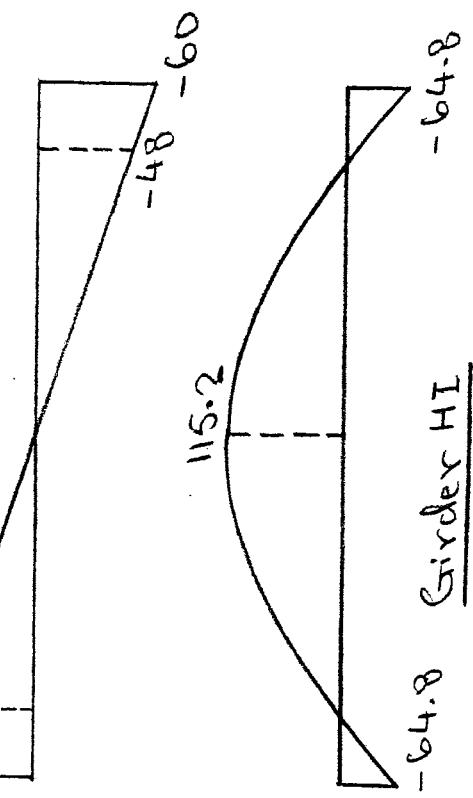
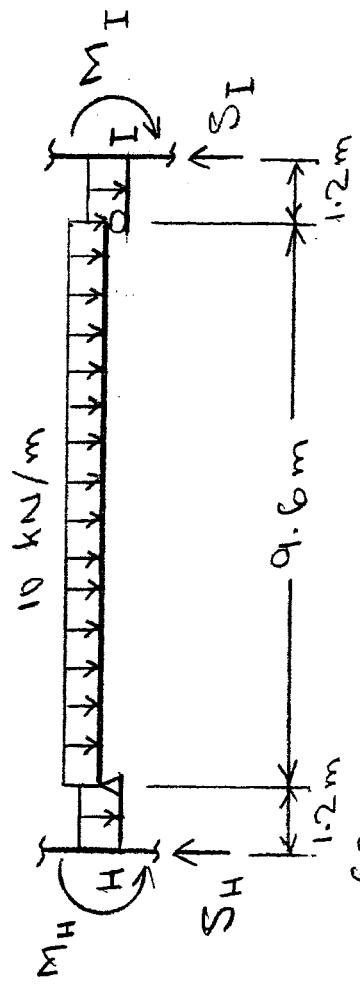
12-4



12-5



Girder GH

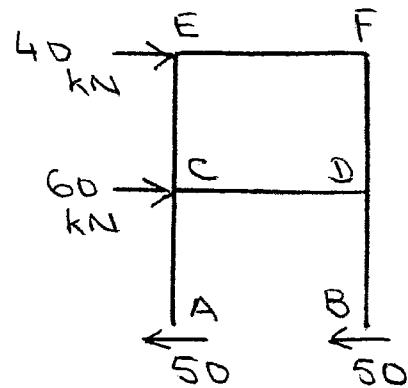
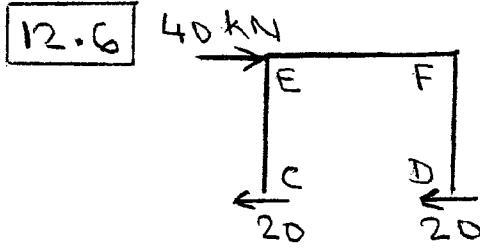
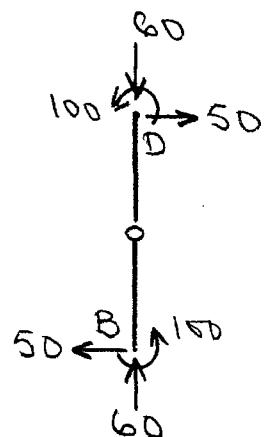
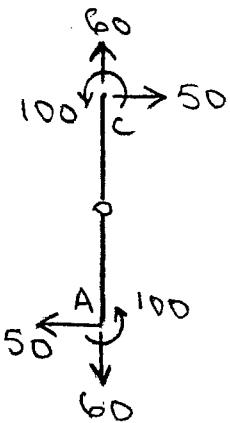
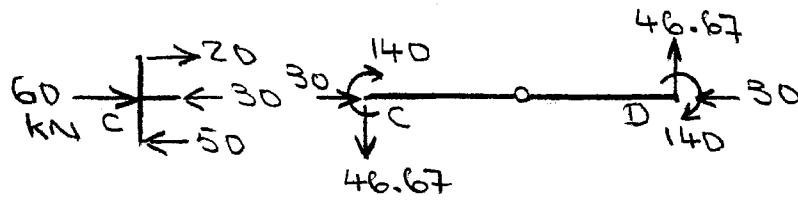
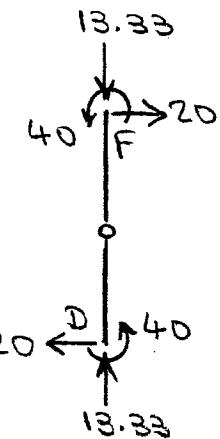
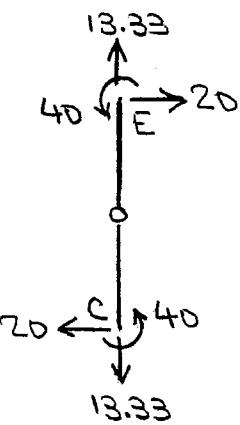
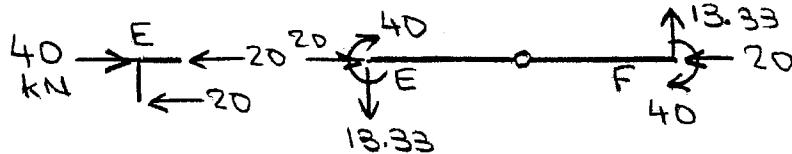


Girder HI

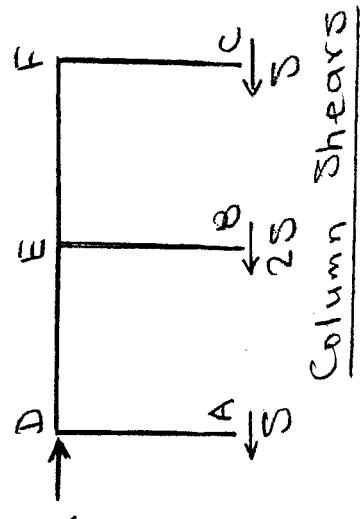
Shear in Girder DE = 2 x Shear in Girder GH.
Moment in Girder DE = 2 x Moment in Girder GH.

Shear in Girder EF = 2 x Shear in Girder HI.
Moment in Girder EF = 2 x Moment in Girder HI.

12.6

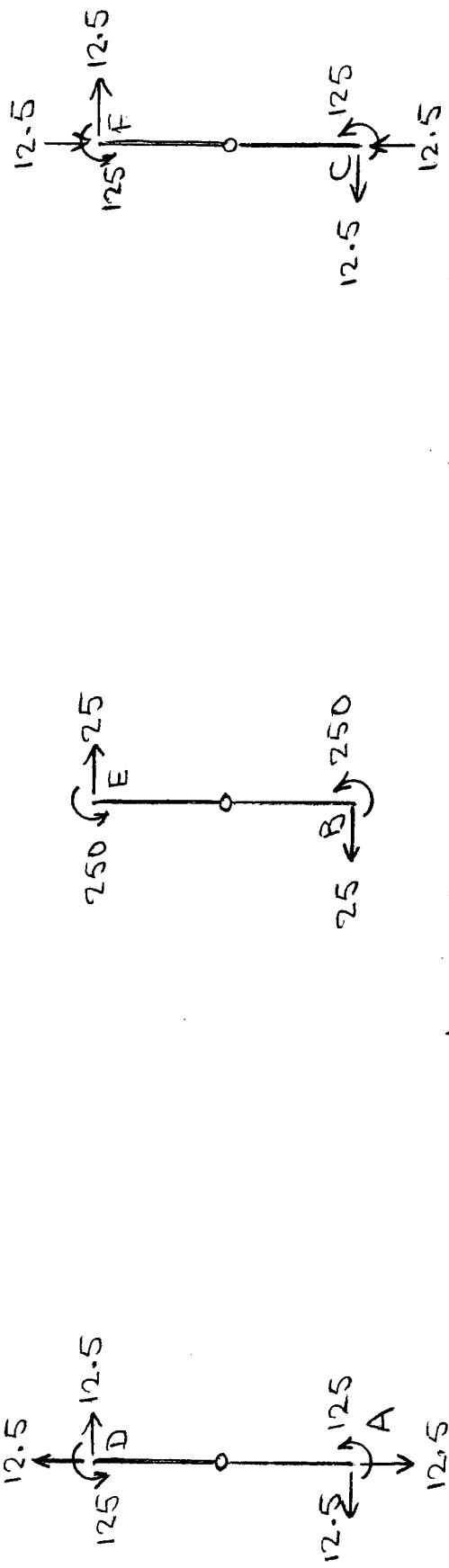
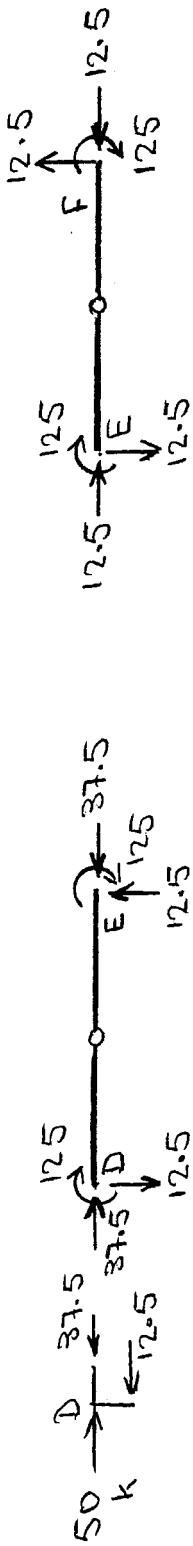
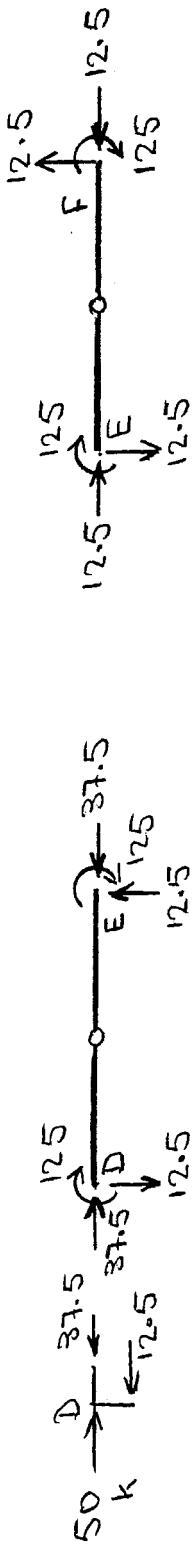
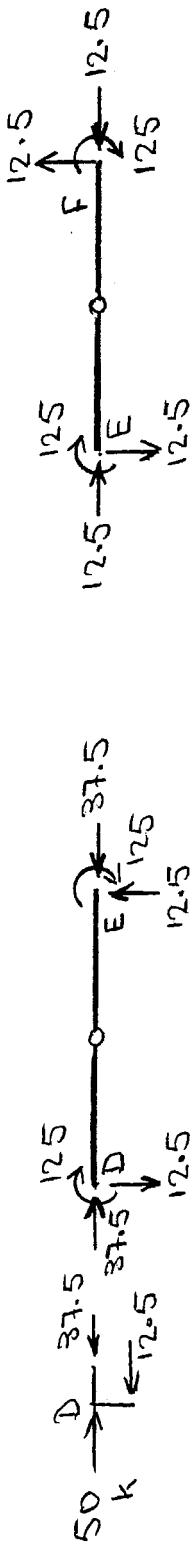
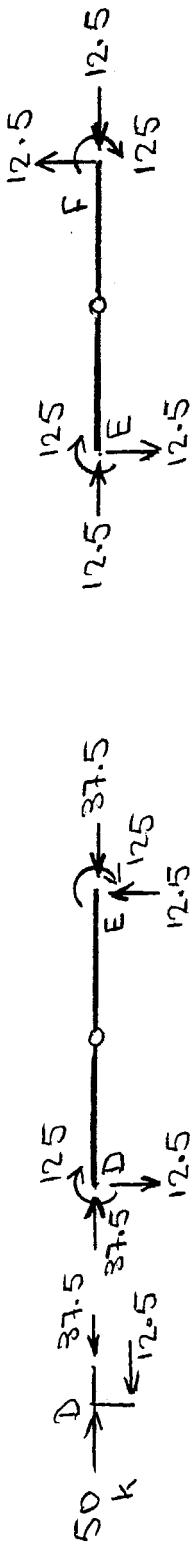
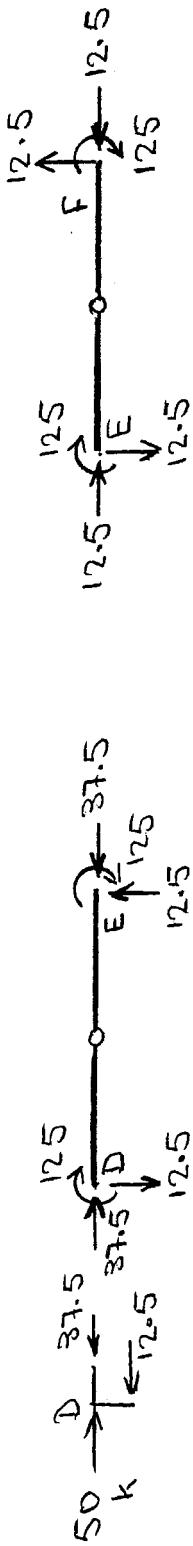
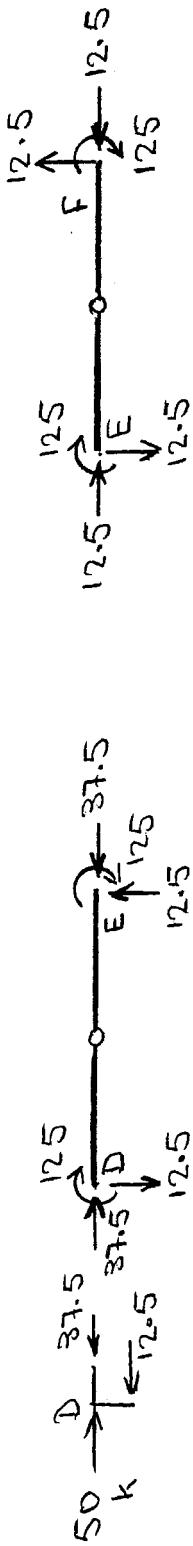
Column ShearsMember Axial Forces, Shears, and Moments

12.7
12.1



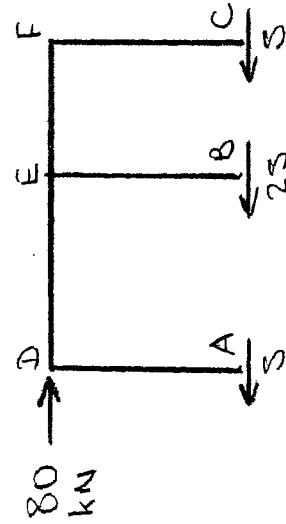
$$4S = 50 \quad S = 12.5 \text{ k}$$

Column Shears



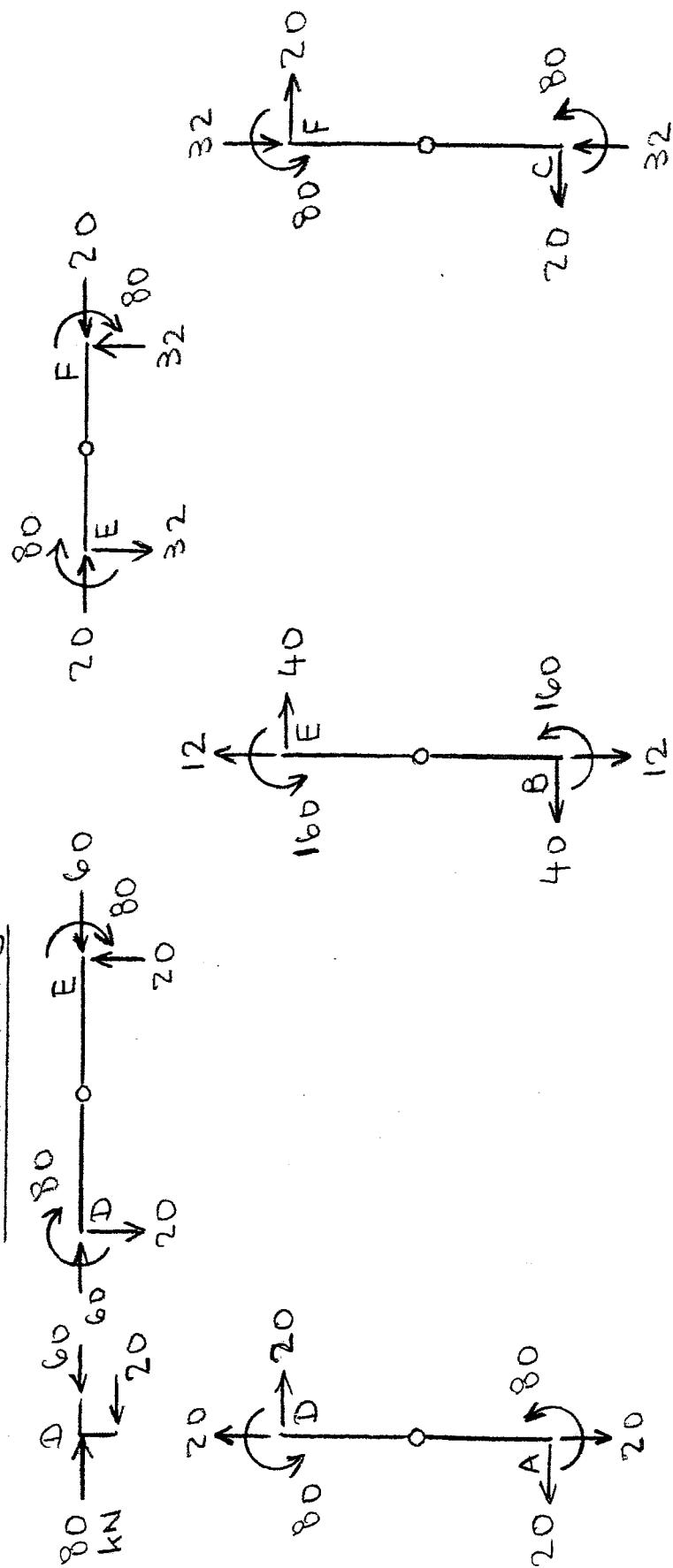
Member Axial Forces, Shears, and Moments

12.8

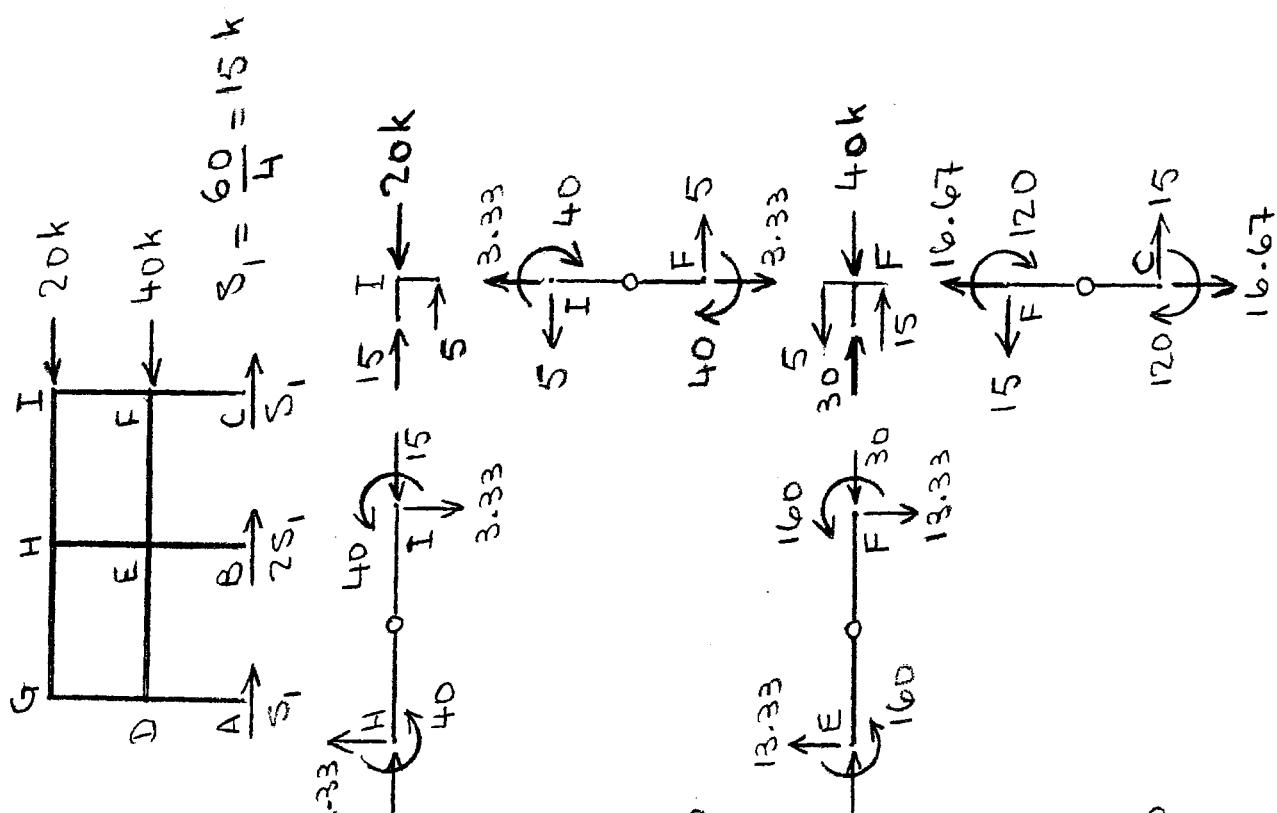
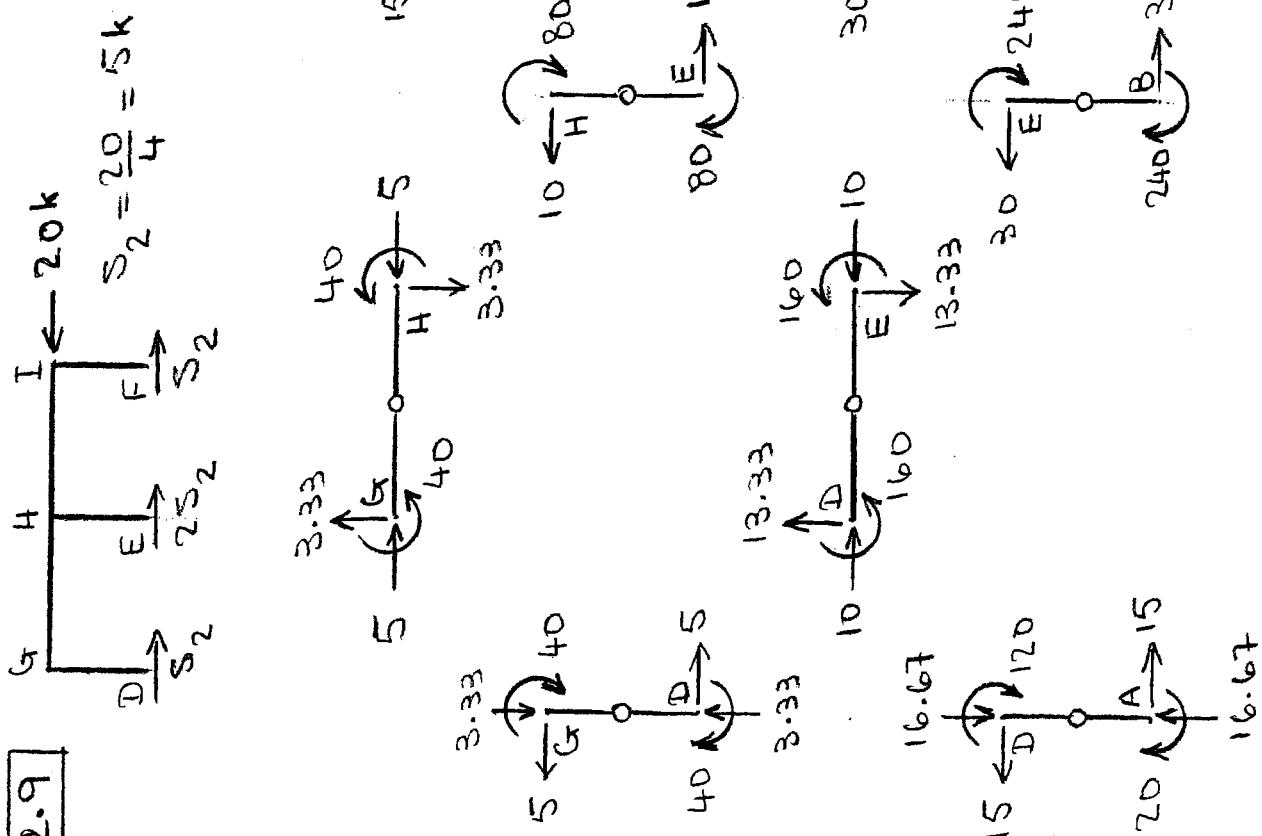


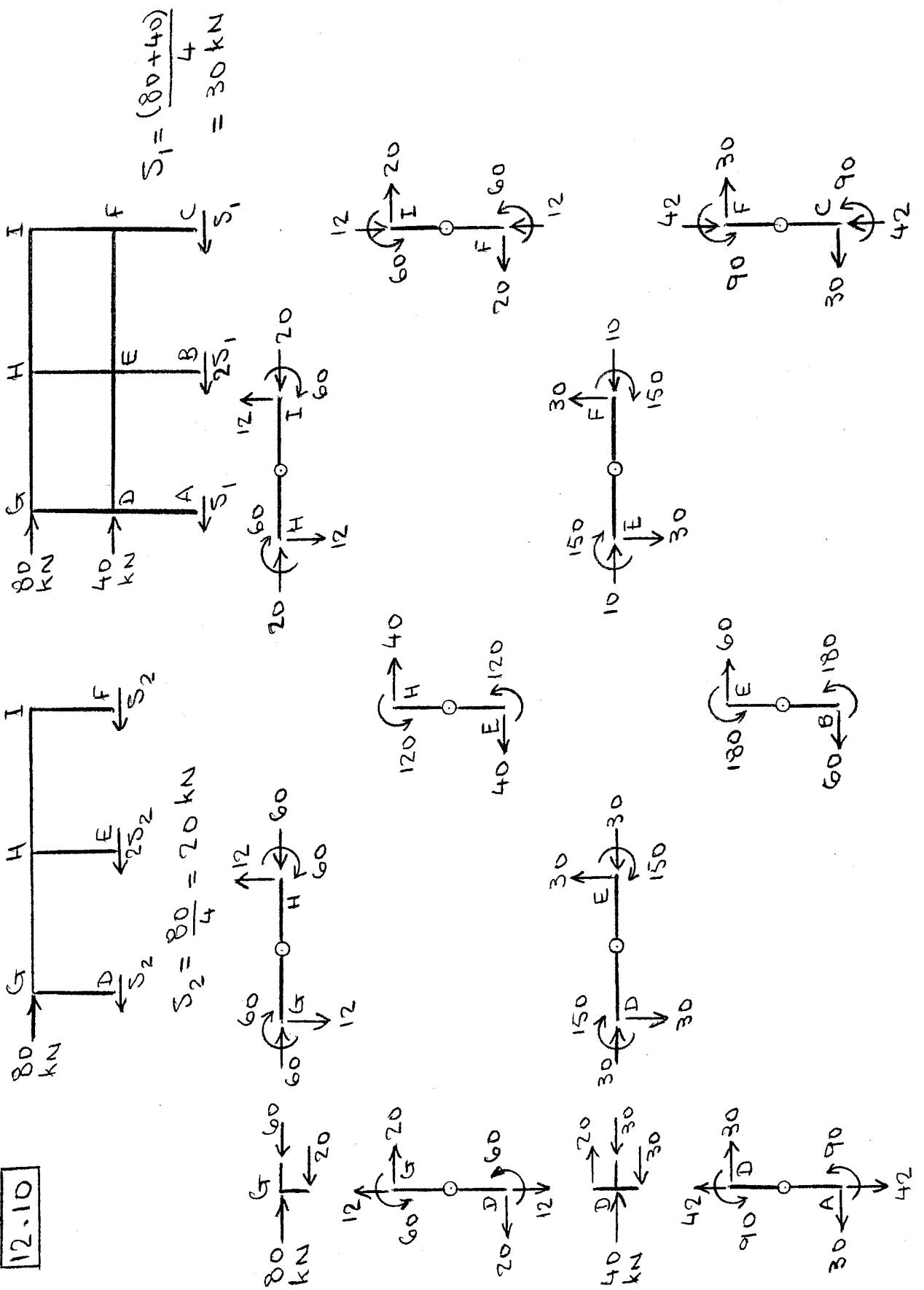
Column Shears

$$4S = 80 \quad S = 20 \text{ kN}$$

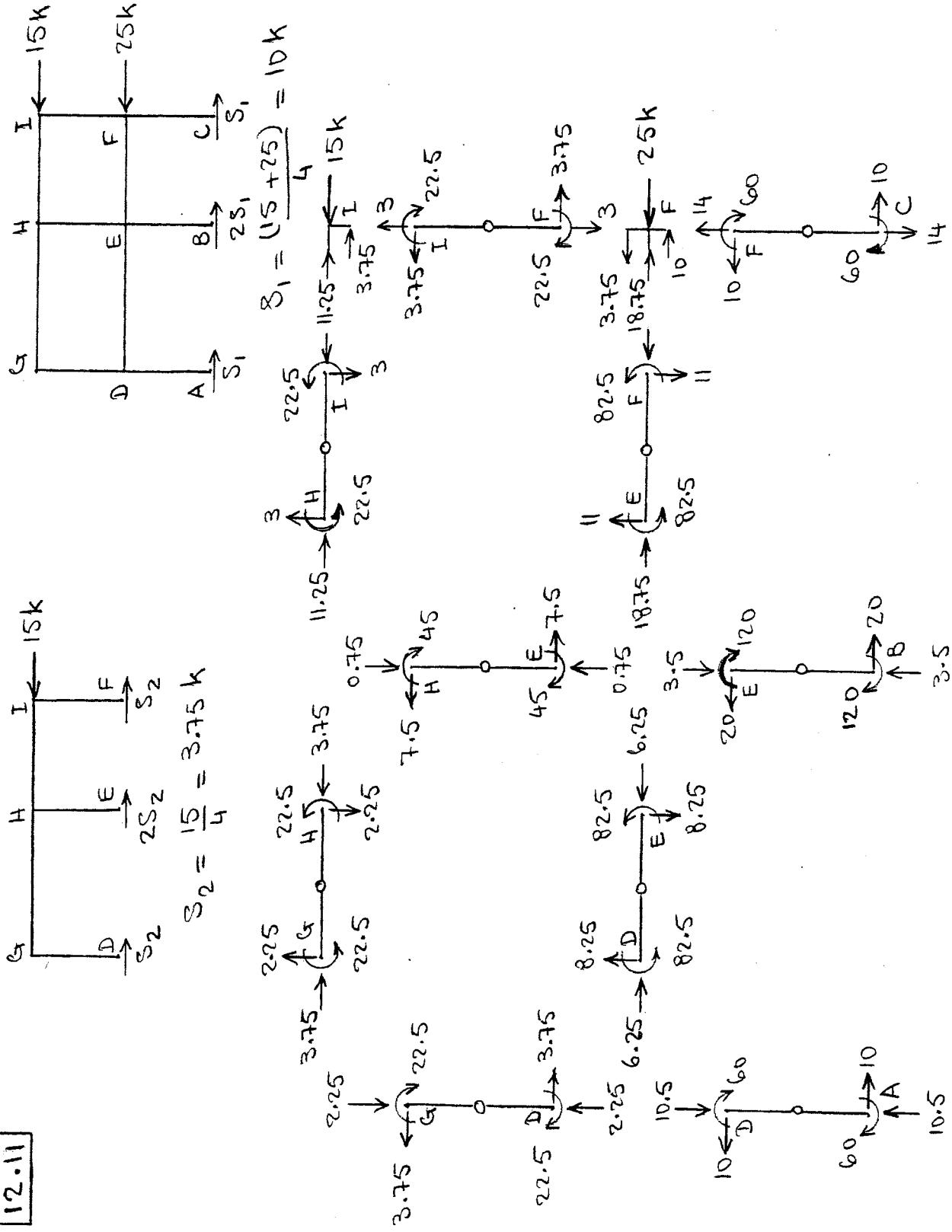


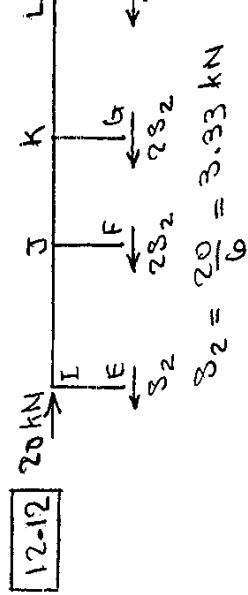
12.9



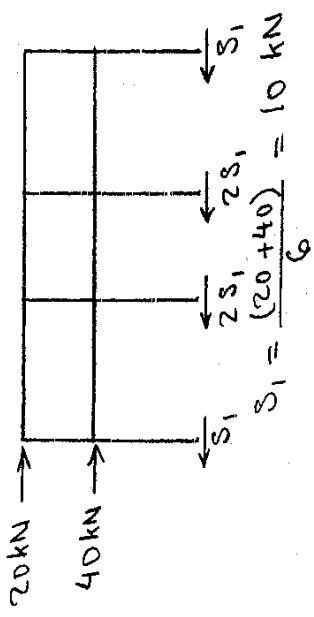


12.11

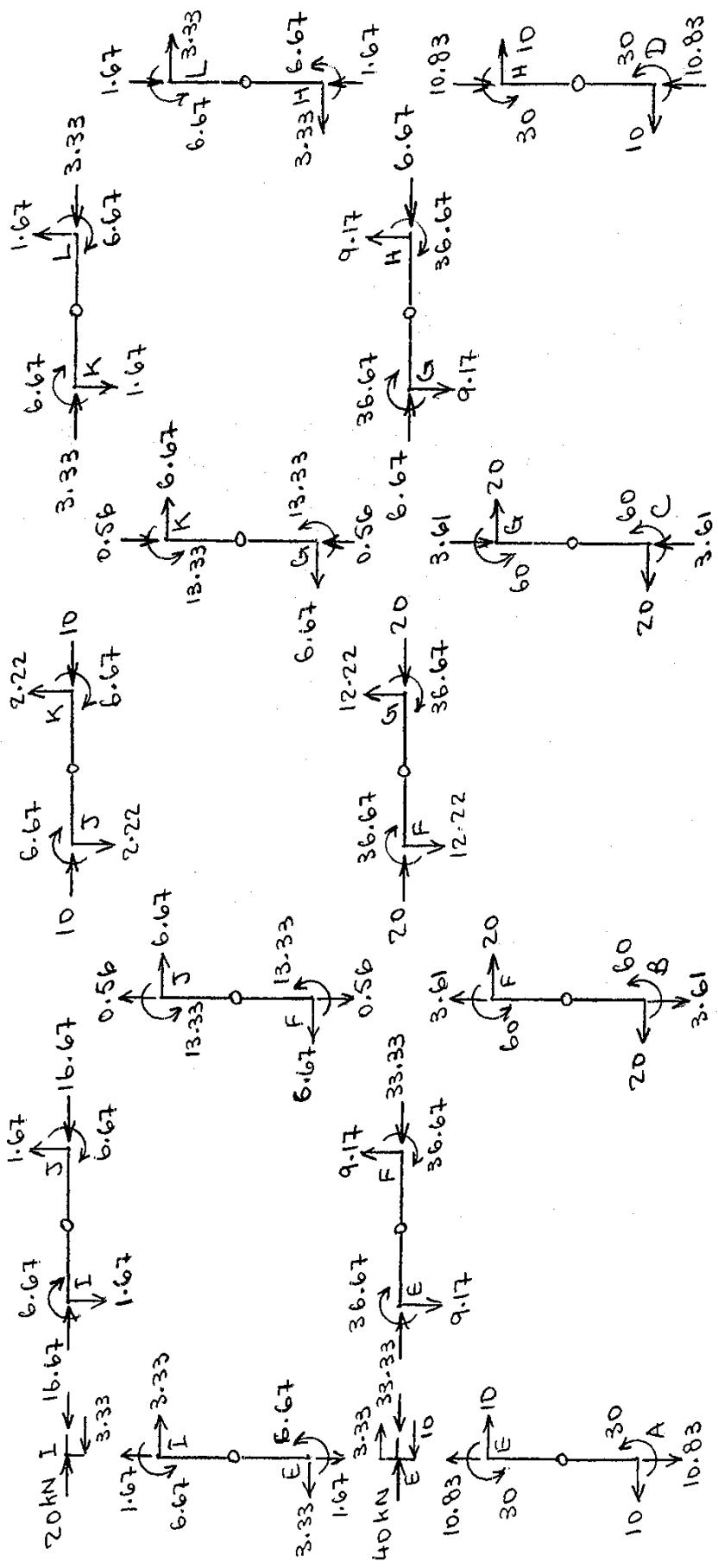




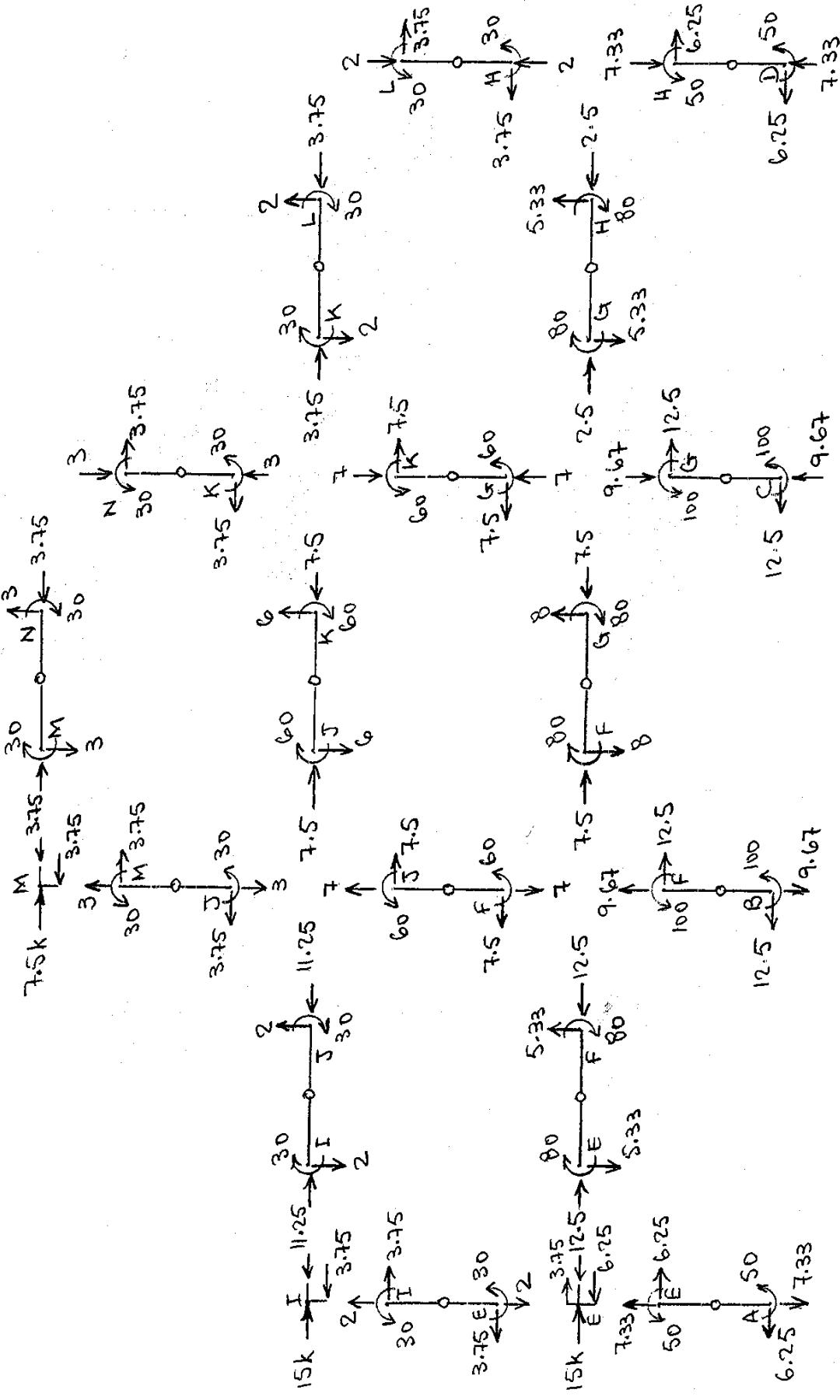
$$S_2 = \frac{20}{6} = 3.33 \text{ kN}$$



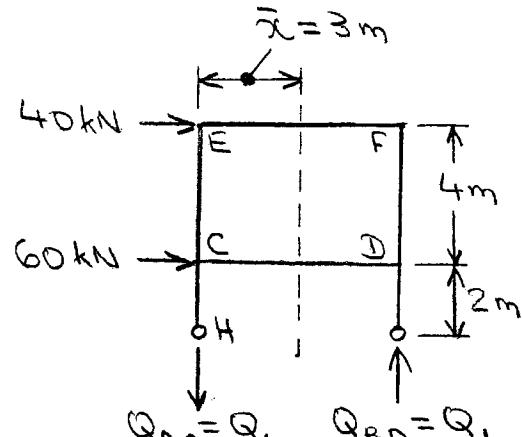
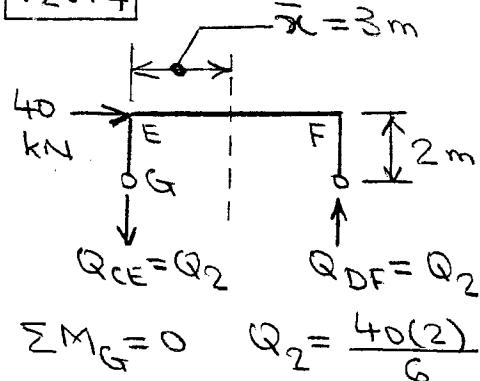
$$S_1 = \frac{(20+40)}{6} = 10 \text{ kN}$$



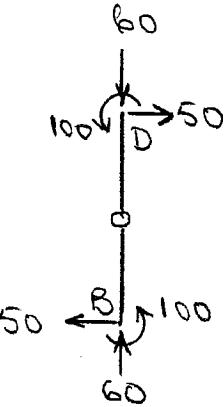
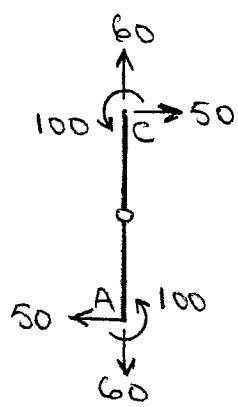
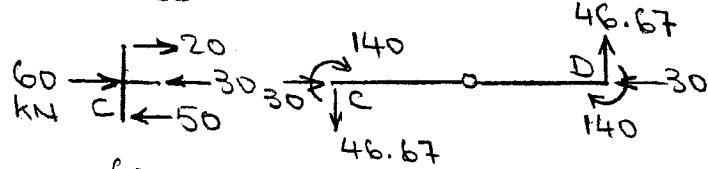
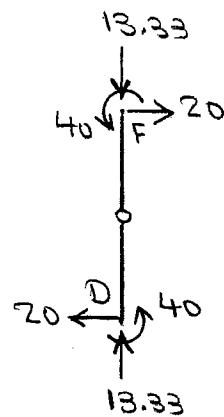
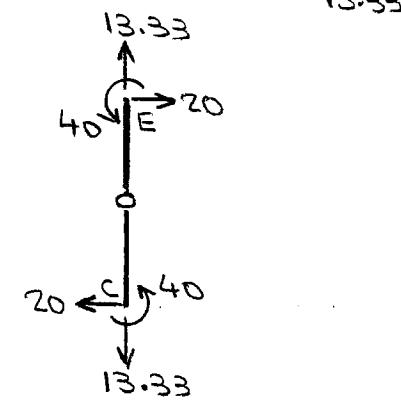
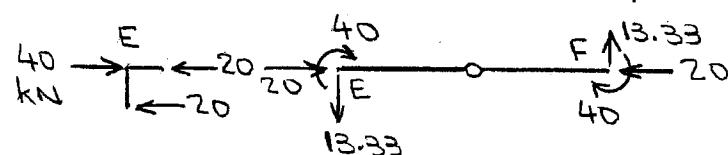
$$12.13 \quad S_3 = \frac{7.5}{2} = 3.75 \text{ k}; \quad S_2 = \frac{(7.5+15)}{6} = 3.75 \text{ k}; \quad S_1 = \frac{(7.5+15+15)}{6} = 6.25 \text{ k}$$



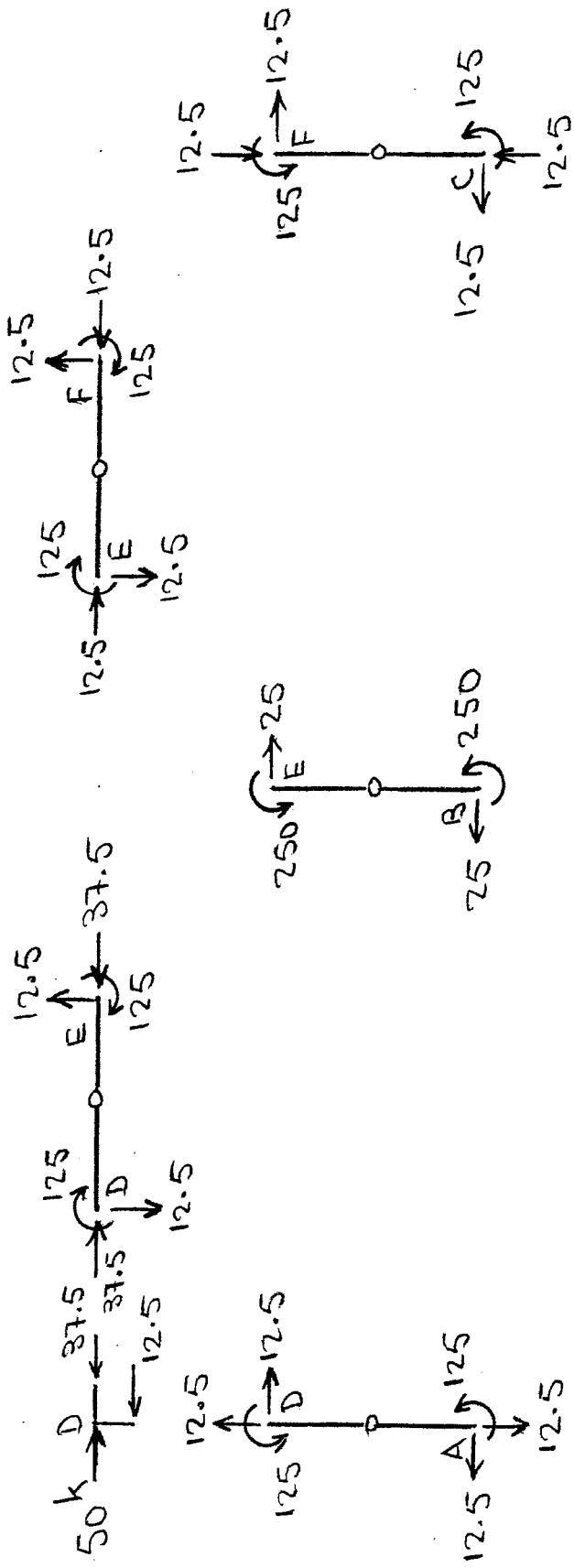
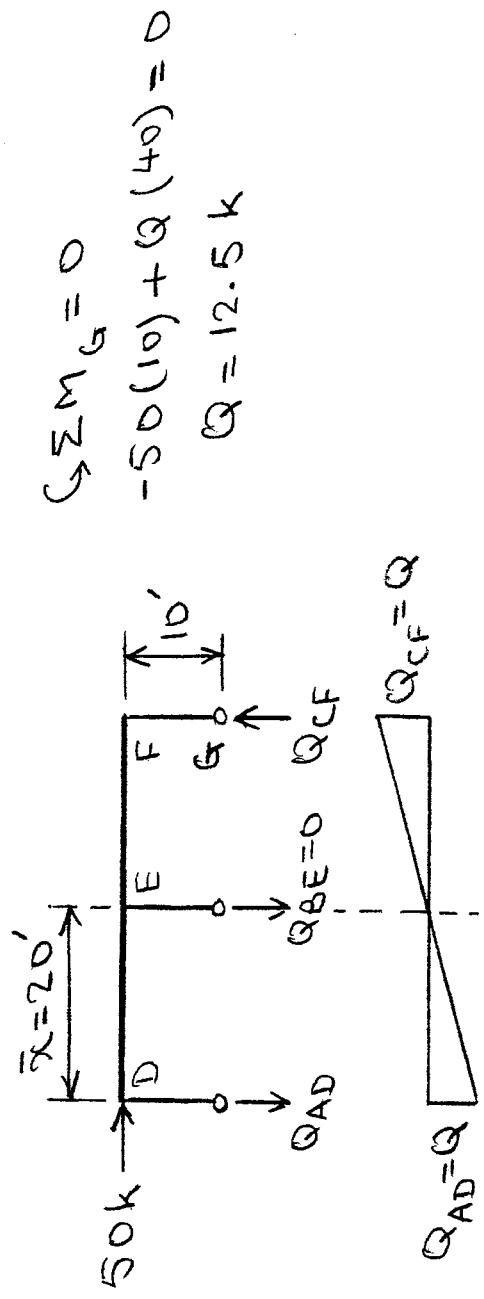
12.14

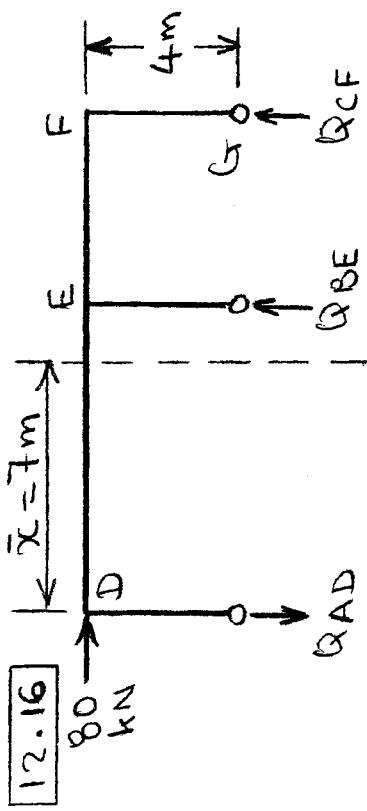


$$Q_1 = \frac{40(6) + 60(2)}{6} = 60 \text{ kN}$$



12.15





$$D = \sum_{j=1}^n \sum_{i=1}^{m_j}$$

$$-80(4) + Q(13) - \frac{1}{4}Q(5) = 0$$

$$G = 26.05 \text{ kN}$$

$$Q_{AD} = 26.05 \text{ kN}, Q_{BE} = 3.72 \text{ kN}$$

$$\sigma_{EF} = 22.33 \text{ kN}$$

$$QCF = \frac{Q}{T}$$

四
二

$$\alpha_{BE} = \frac{1}{7}$$

26.05

22.33
55.87

A free body diagram of a horizontal beam. At the left end, there is a reaction force of 80 kN pointing downwards. At the right end, there is a reaction force of 22.33 pointing downwards. There are two vertical loads: one at the left end of 26.05 pointing upwards, and another at the center of the beam of 104.2 pointing upwards. There are two horizontal loads: one at the left end of 76.05 pointing to the left, and another at the center of the beam of 13.96 pointing to the right. There are two clockwise moments: one at the left end of 53.96 and another at the center of the beam of 53.96 . There is one counter-clockwise moment at the center of the beam of 13.96 .

$$160 \rightarrow E \rightarrow 40 \quad | \quad 55.82 \quad | \quad F \rightarrow 13.96$$

20

10

—

2

33.96 → 55.83

22.33

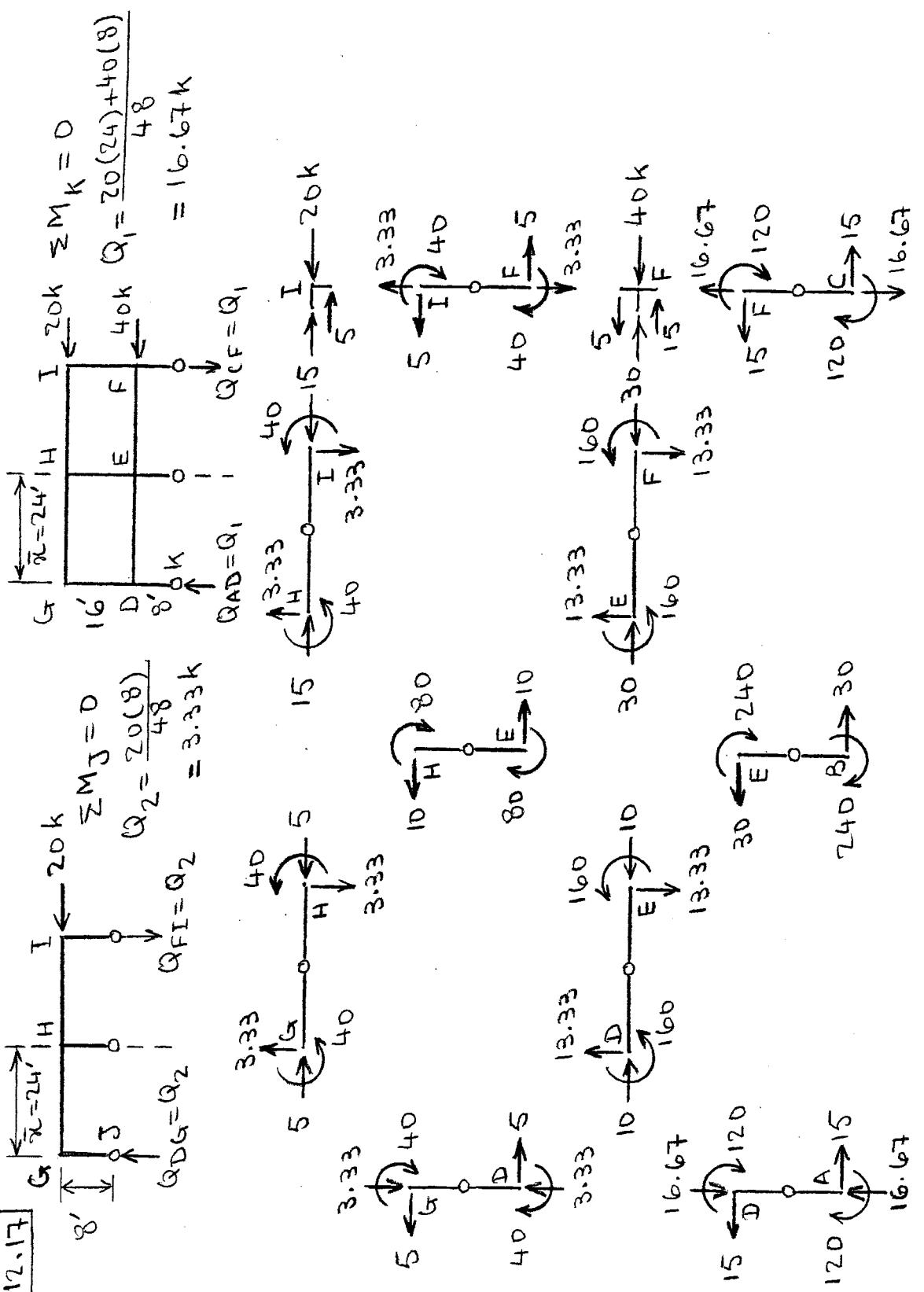
- 60 -

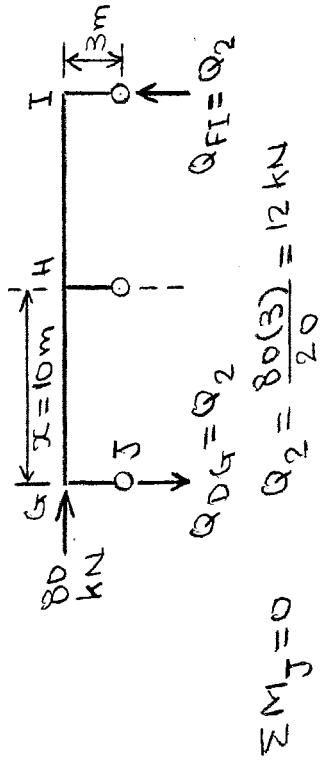
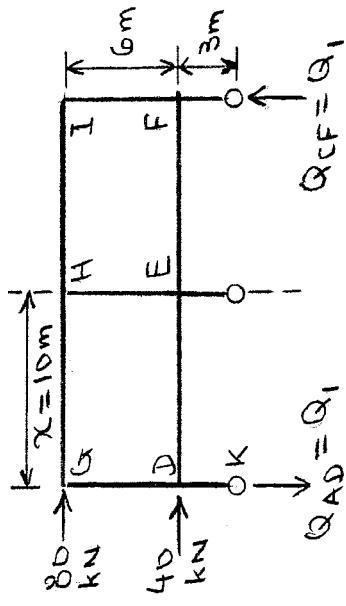
26

$$26.05 \rightarrow A \downarrow 104.2$$

29.05

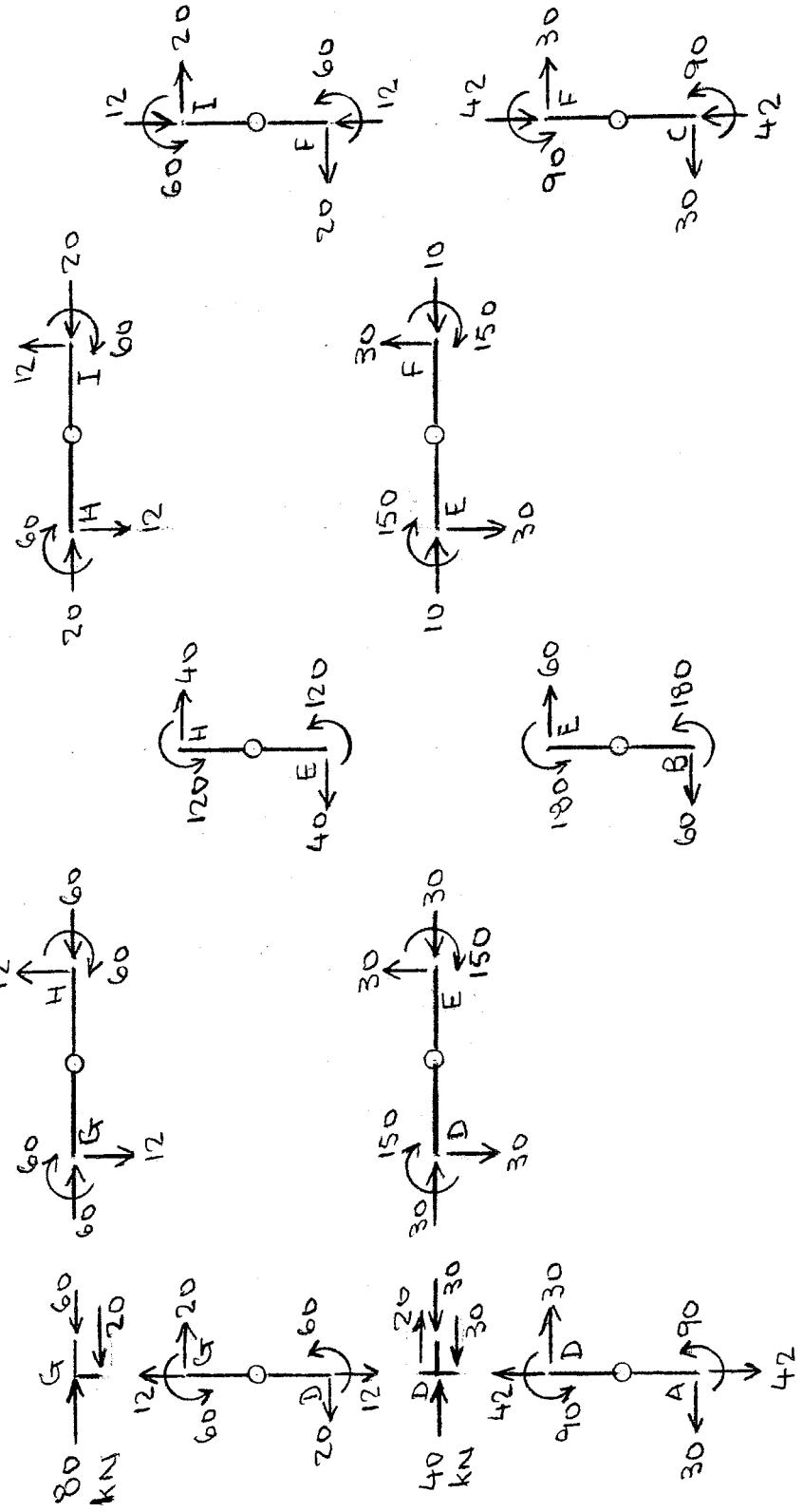
12.17

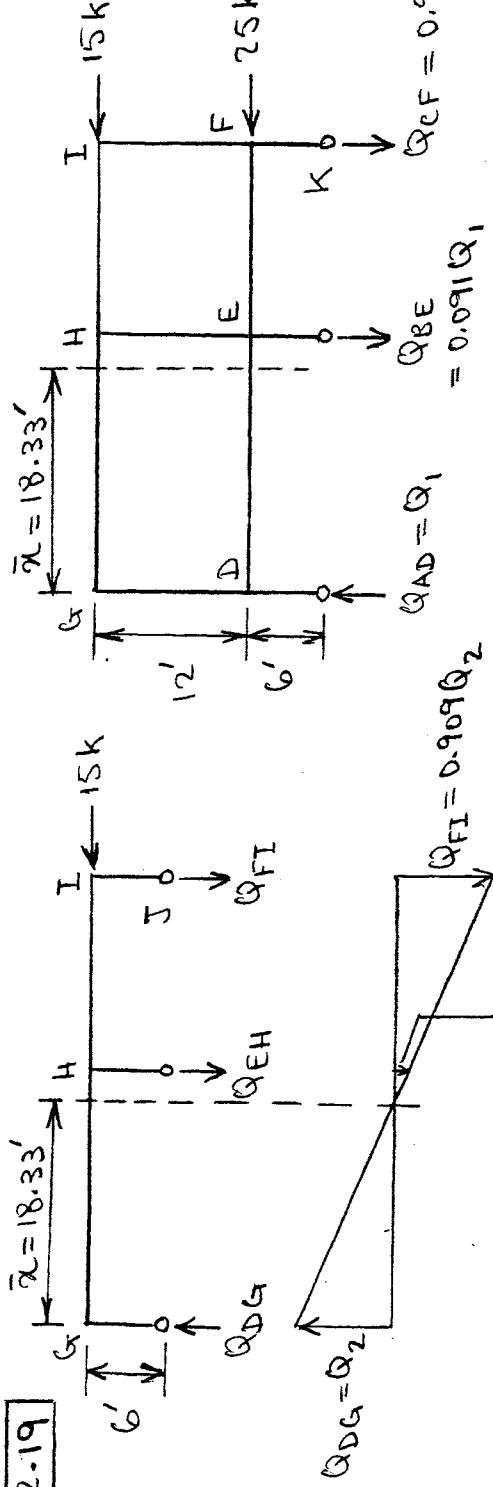




12.18

$$\Sigma M_K = 0 \quad Q_1 = \frac{80(9) + 40(3)}{20} = 42 \text{ kN}$$





$$\begin{aligned}
 & Q_{BC} = 3.41k, \quad Q_{CE} = 0.947k \\
 & Q_2 = 2.71k \\
 & 15(6) - 4Q_2(35) + 0.091Q_2(20) = 0 \\
 & Q_2 = 2.71k \\
 & Q_{AD} = 12.66k, \quad Q_{BE} = 1.15k \\
 & 15(18) + 2Q_1(6) - Q_1(35) + 0.091Q_1(20) = 0 \\
 & Q_1 = 12.66k \\
 & Q_{CF} = 11.51k
 \end{aligned}$$

$$Q_{FI} = 2.46 \text{ k}$$

$$Q_{\text{E}} = 0.31 \text{ k}, Q_{\text{EH}} = 0.24 \text{ k}$$

$$Q_{CF} = 11.51 \text{ k}$$

$$Q_{AD} = 12.66K, \quad Q_{BE} = 1.15K$$

$$Q_1 = 12.66 \text{ k}$$

$$15(6) - \Phi_2(35) + 0.091\Phi_2(20) = 0$$

$$+ C \sum_{\sigma} S^{\sigma} = 0$$

$$Q_{EH} = 0.091 Q_2$$

$$Q_{FI} = 0.909 Q_2$$

$$Q_{CF} = 0.909 Q_1$$

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十一

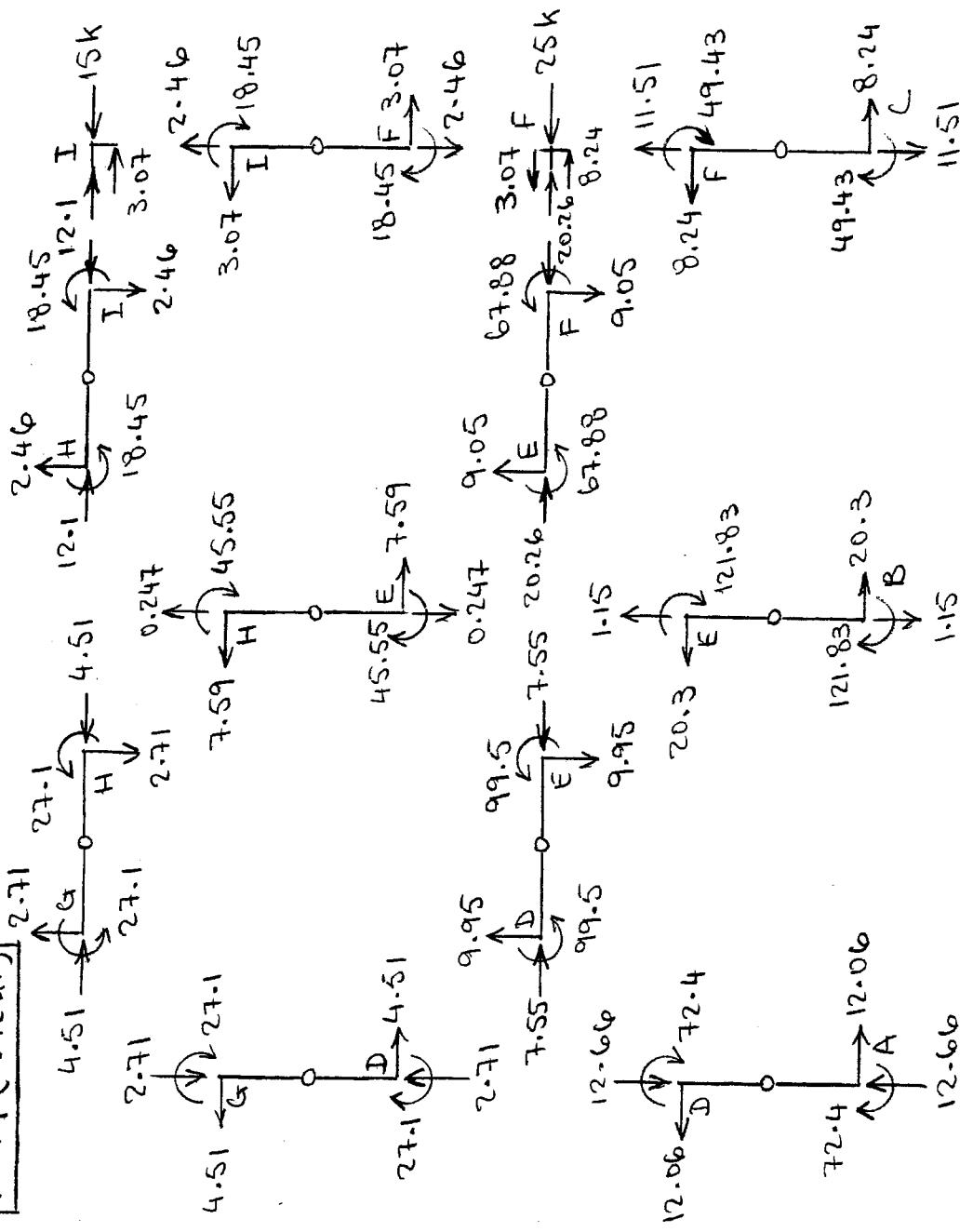
I
5

\rightarrow E

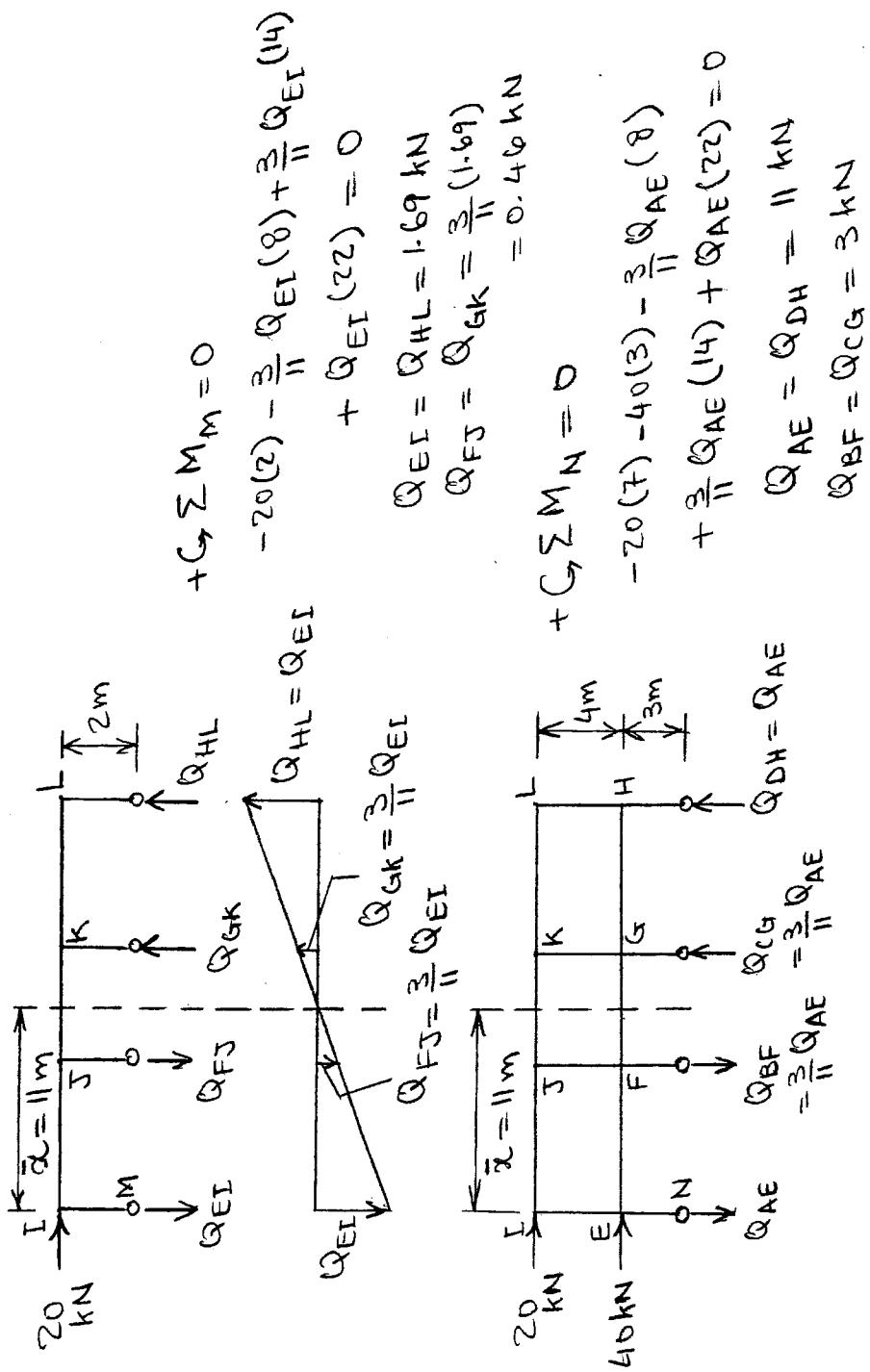
5
6

12.19

12.19 (cont'd.)

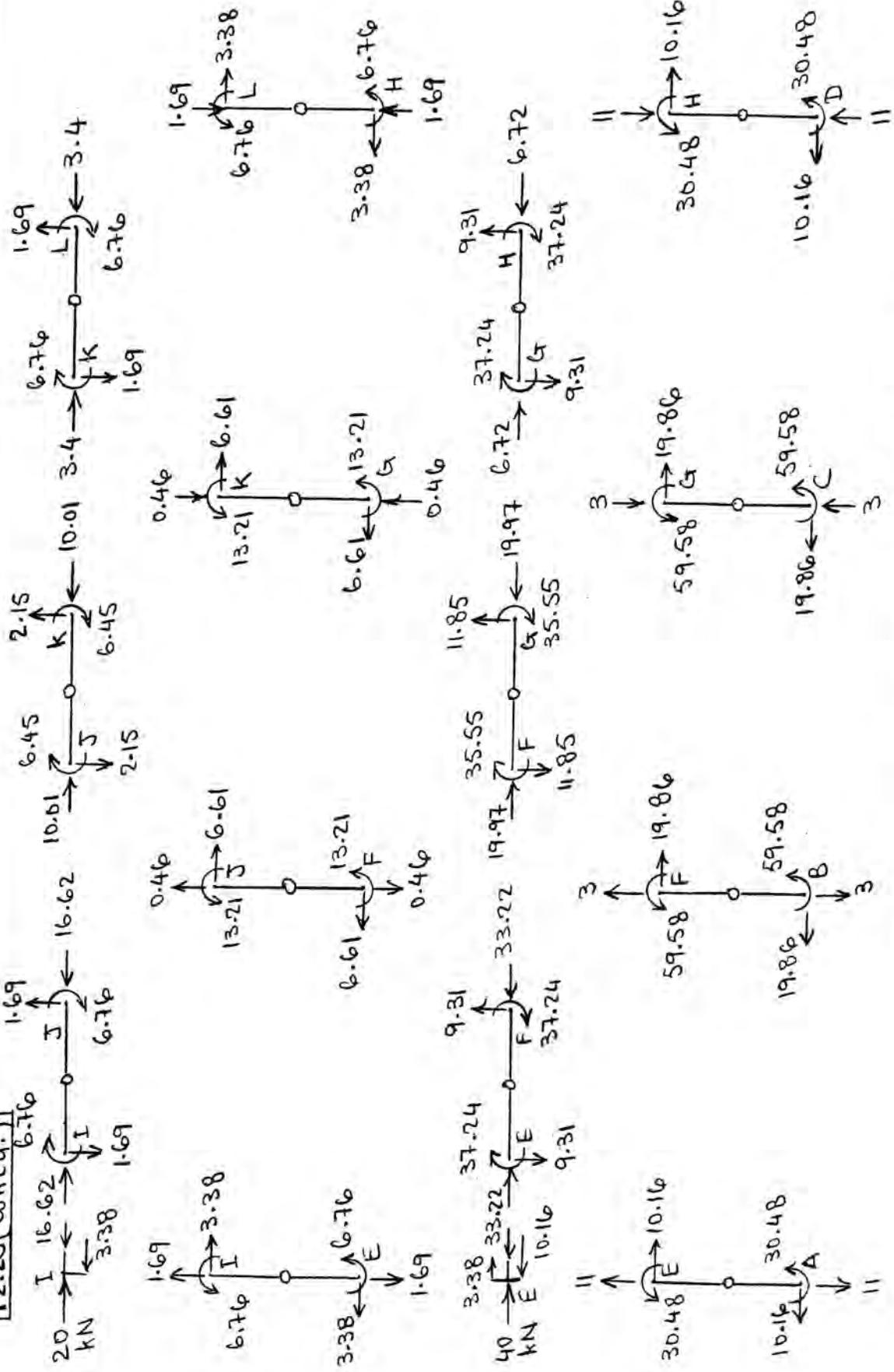


12.20



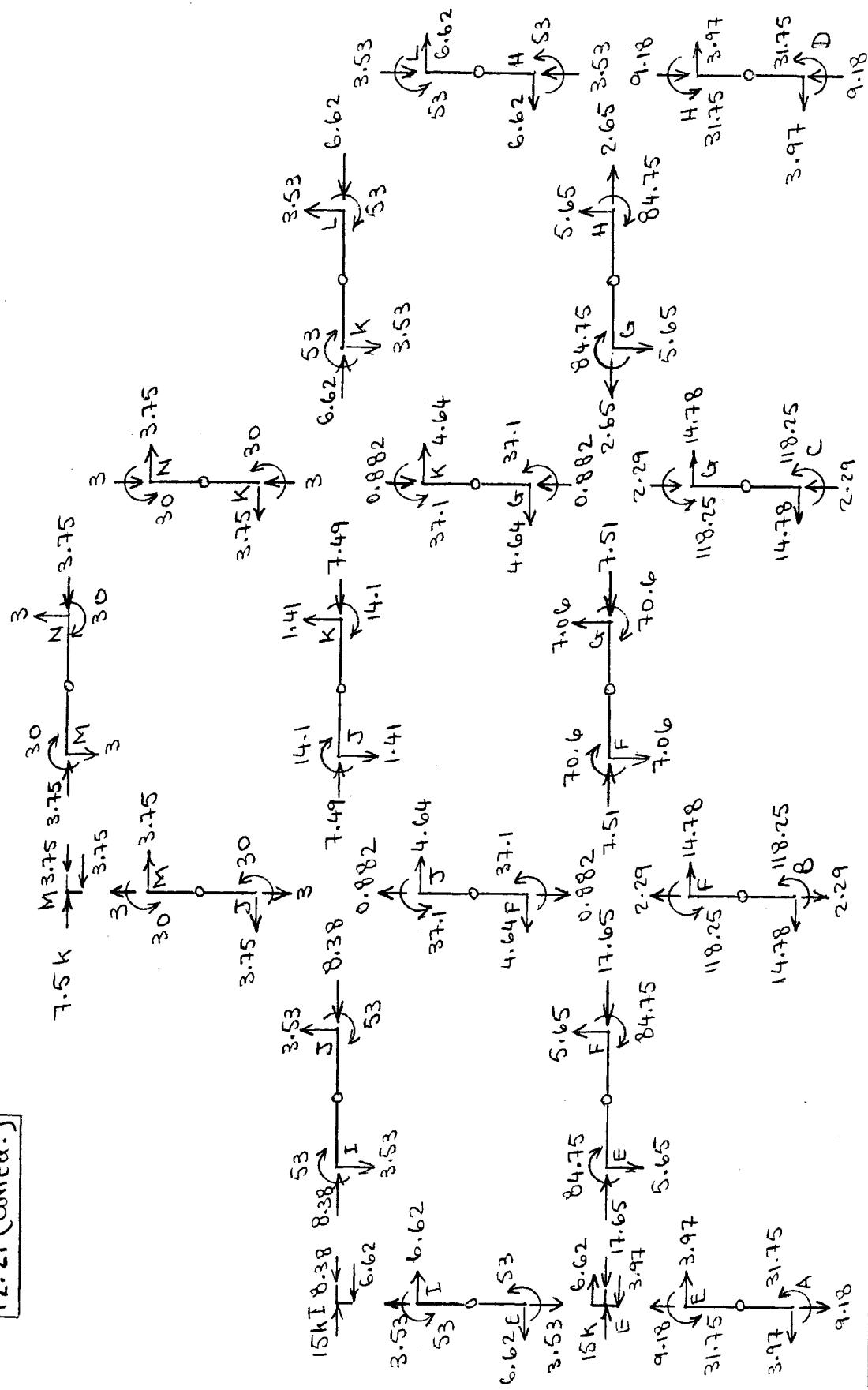
Column Axial Forces

12.20 (cont'd.)



Member Axial Forces, Shears, and Moments

12.21 (contd.)

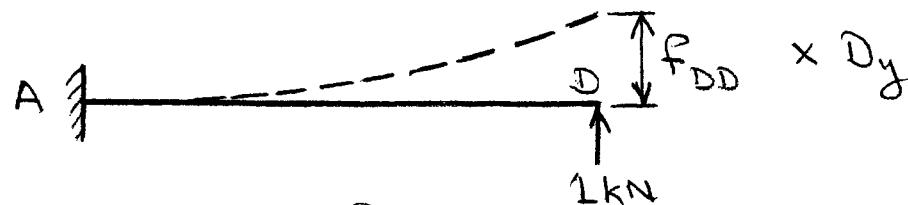
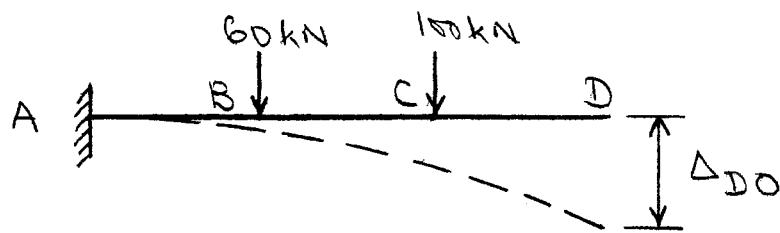


Chapter Thirteen

Method of Consistent Deformations – Force Method

CHAPTER 13

13.1

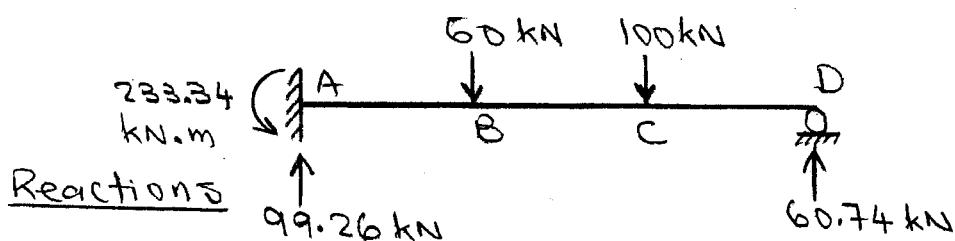


Using beam deflection formulas:

$$\Delta_{DD} = -\frac{14760 \text{ kN.m}^3}{EI} \quad F_{DD} = \frac{243 \text{ kN.m}^3/\text{kN}}{EI}$$

Compatibility Equation:

$$\Delta_{DD} + F_{DD} D_y = 0 \quad D_y = \frac{14760}{243} = \underline{60.74 \text{ kN} \uparrow}$$

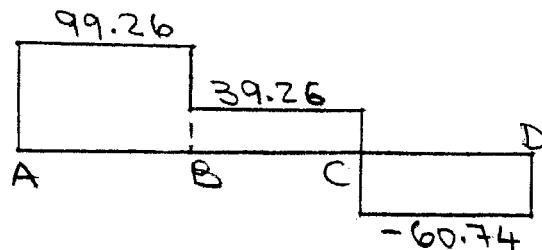


Reactions

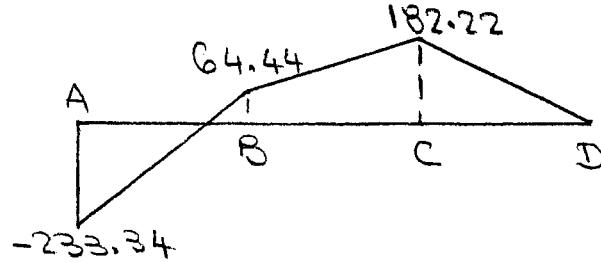
99.26 kN

60.74 kN

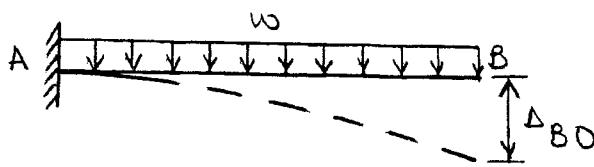
Shear Diagram (kN)



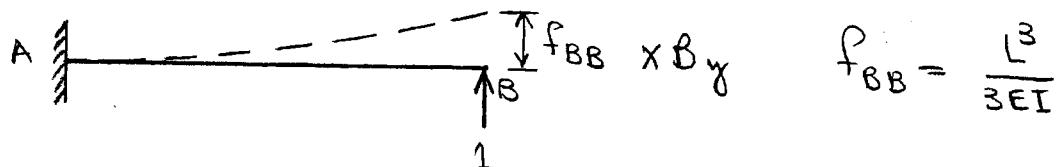
Bending Moment Diagram (kN.m)



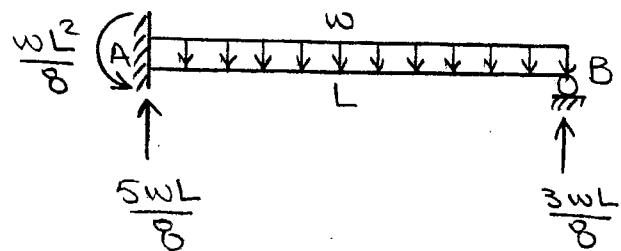
13.2



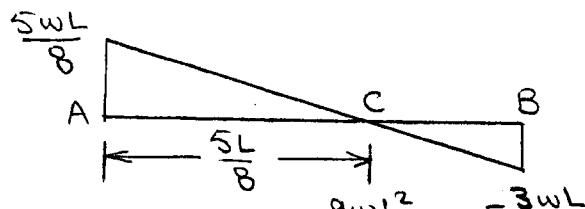
$$\Delta_{B0} = -\frac{wL^4}{8EI}$$



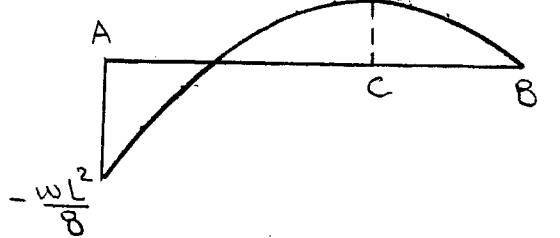
$$\Delta_{B0} + P_{BB} B_y = 0 \quad B_y = \left(\frac{wL^4}{8EI}\right) \frac{3EI}{L^3} = \frac{3wL}{8} \uparrow$$



Reactions

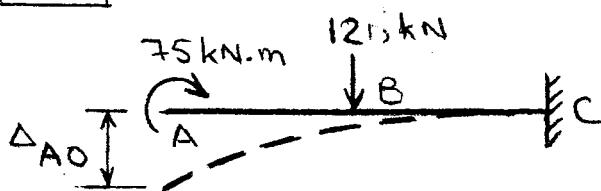


Shear Diagram



Bending Moment Diagram

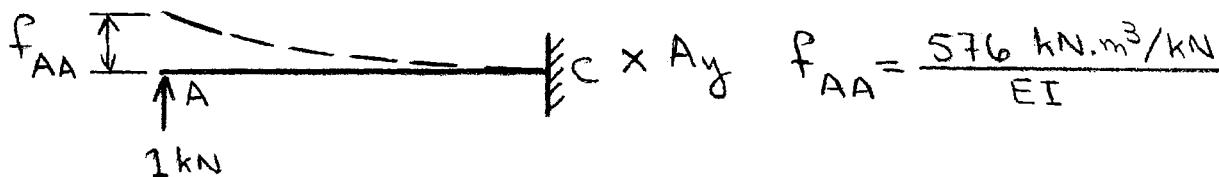
13.3



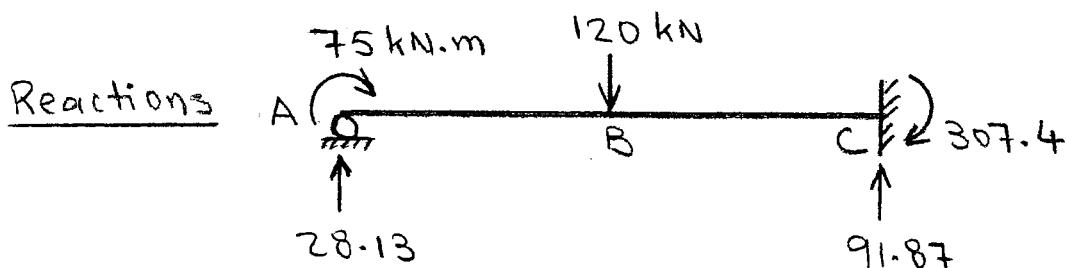
Using beam deflection

formulas:

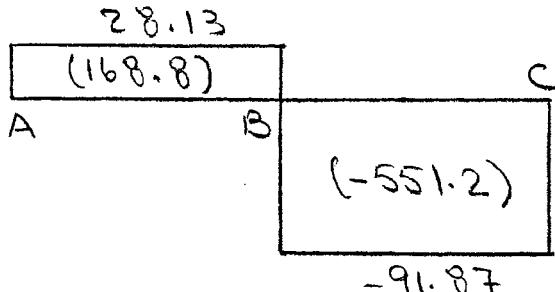
$$\Delta_{AO} = \frac{5400}{EI} - \frac{21600}{E\sum} \\ = - \frac{16200 \text{ kN.m}^3}{EI}$$



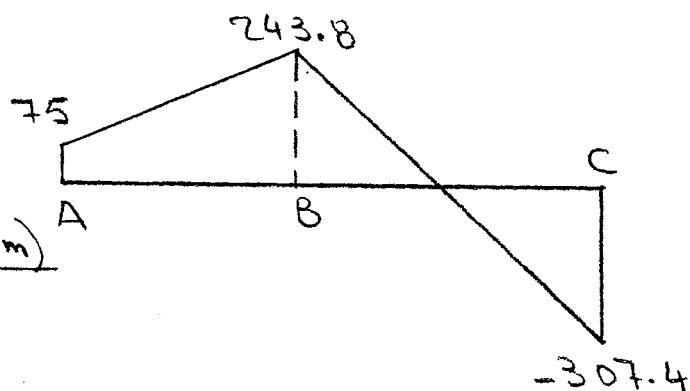
$$\Delta_{AO} + f_{AA} A_y = 0 \quad A_y = \frac{16200}{576} = 28.13 \text{ kN} \uparrow$$



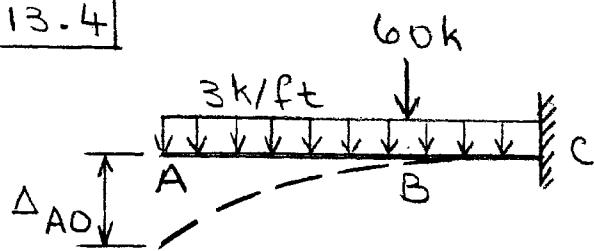
Shear Diagram (kN)



Bending Moment Diagram (kN.m)



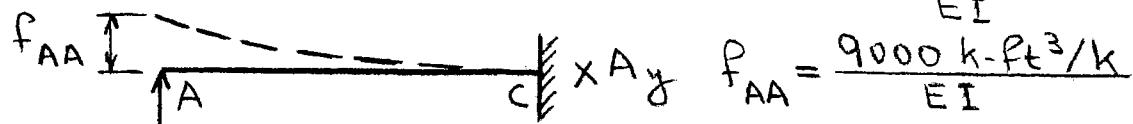
13.4



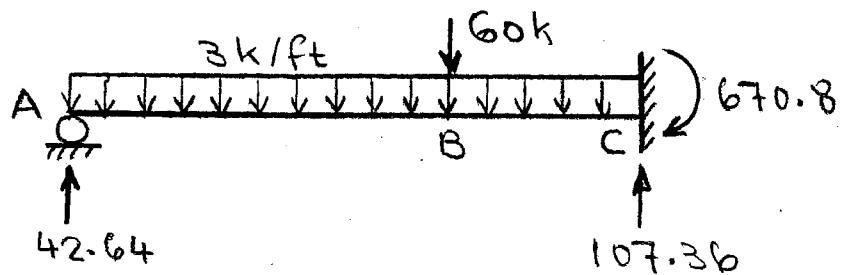
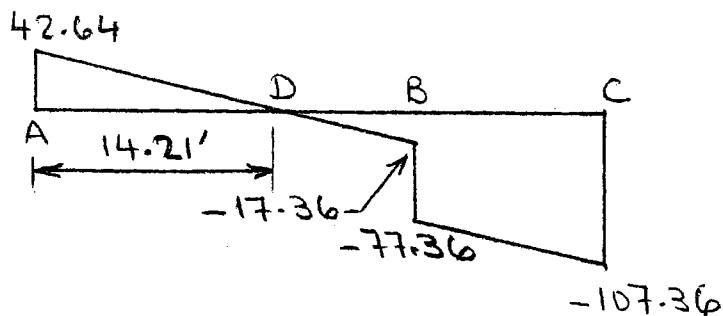
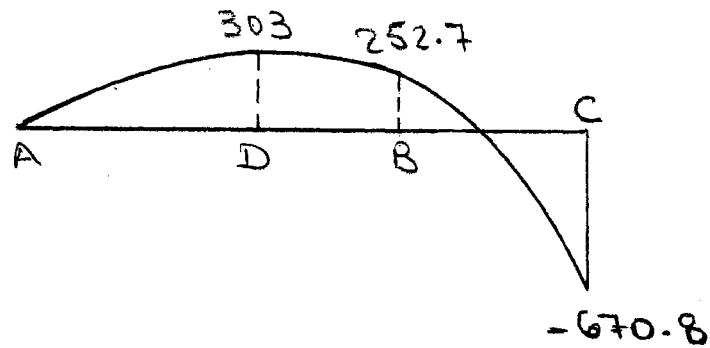
Using beam deflection formulas:

$$\Delta_{AO} = -\frac{303750}{EI} - \frac{80000}{EI}$$

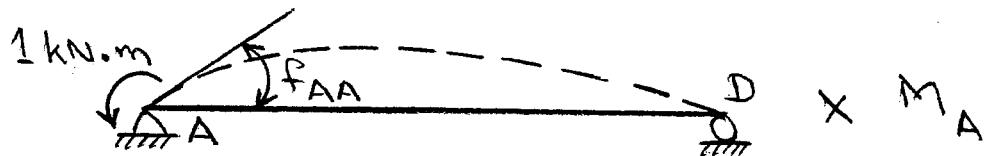
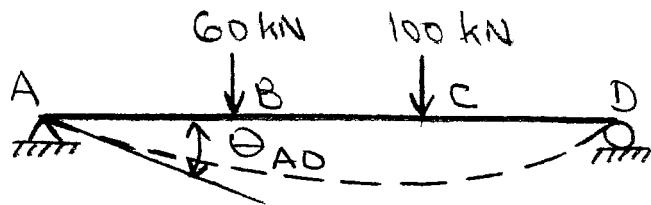
$$= -\frac{383750}{EI} \text{ k-ft}^3$$



$$\Delta_{AO} + f_{AA} A_y = 0 \quad A_y = \frac{383750}{9000} = 42.64 \text{ k} \uparrow$$

ReactionsShear Diagram (k)Bending Moment Diagram (k-ft)

13.5



Using beam deflection formulas:

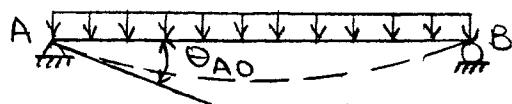
$$\theta_{AD} = -\frac{700 \text{ kN}\cdot\text{m}^2}{EI} \quad f_{AA} = \frac{3 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI}$$

Compatibility Equations:

$$\theta_{AD} + f_{AA} M_A = 0 \quad M_A = \frac{700}{3} = 233.33 \text{ kN}\cdot\text{m}$$

For reactions, and shear and moment diagrams, see solution of Problem 13.1.

13.6



$$\theta_{AO} = -\frac{wL^3}{24EI}$$

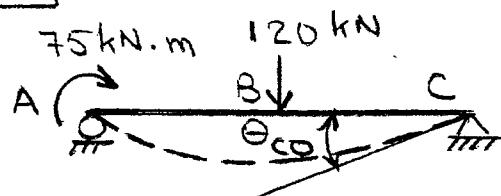


$$f_{AA} = \frac{L}{3EI}$$

$$\theta_{AO} + f_{AA} M_A = 0 \quad M_A = \frac{wL^3}{24EI} \left(\frac{3EI}{L} \right) = \frac{wL^2}{8} \rightarrow$$

For reactions, and shear and moment diagrams,
see solution of Problem 13.2.

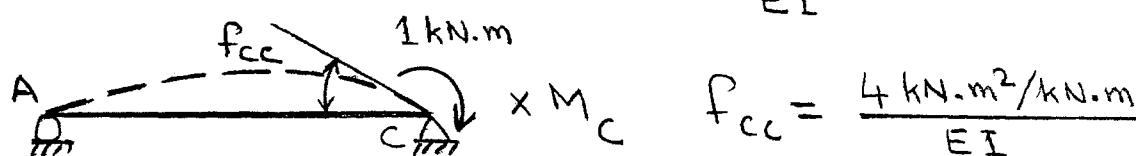
13-7



Using beam deflection formulas:

$$\Theta_{CO} = -\frac{150}{EI} - \frac{1080}{EI}$$

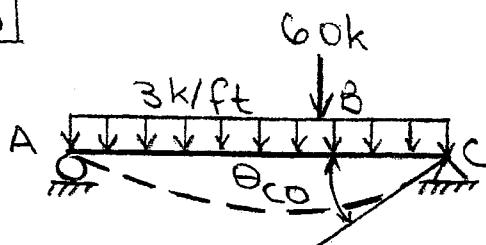
$$= -\frac{1230 \text{ kN}\cdot\text{m}^2}{EI}$$



$$\Theta_{CO} + f_{CC}M_C = 0 \quad M_C = \frac{1230}{4} = \underline{\underline{307.5 \text{ kN}\cdot\text{m}}}$$

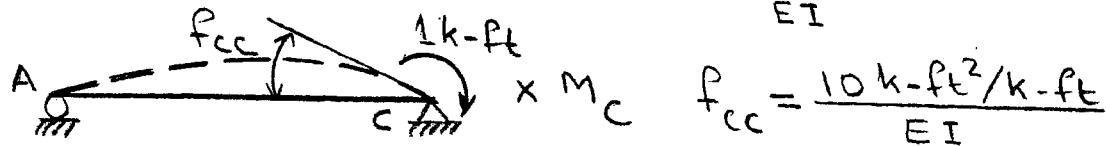
For reactions, and shear and moment diagrams,
see solution of Problem 13-3.

13.8



Using beam deflection formula:

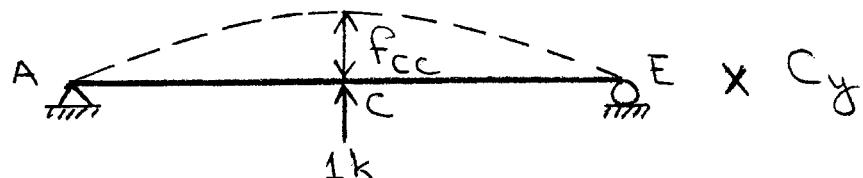
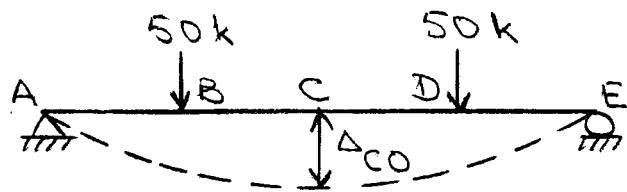
$$\theta_{co} = -\frac{3375}{EI} - \frac{3333}{EI}$$
$$= -\frac{6708 \text{ k-ft}^2}{EI}$$



$$\theta_{co} + f_{cc}M_c = 0 \quad M_c = \frac{6708}{10} = 670.8 \text{ k-ft}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.4.

13.9

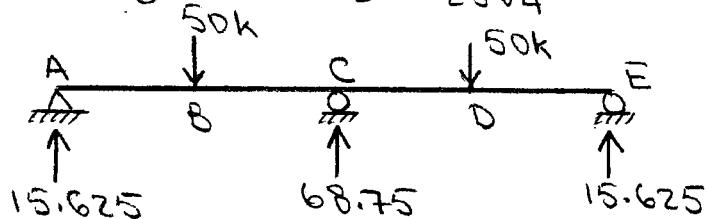


Using beam deflection formulas:

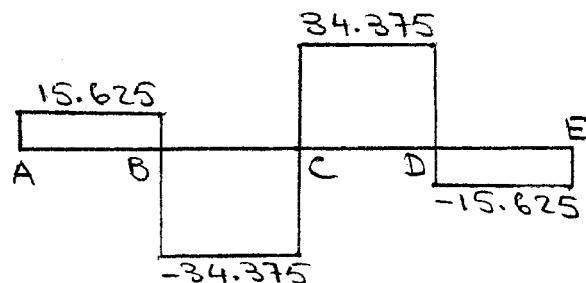
$$\Delta_{CO} = -\frac{158400 k \cdot ft^3}{EI} \quad f_{CC} = \frac{2304 k \cdot ft^3 / k}{EI}$$

Compatibility Equation:

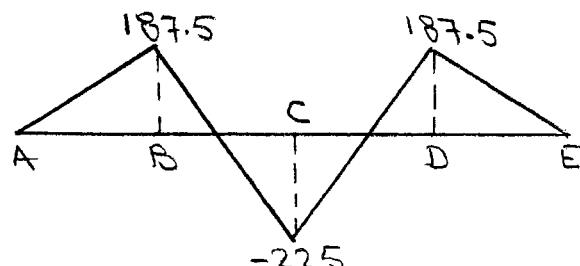
$$\Delta_{CO} + f_{CC} C_y = 0 \quad C_y = \frac{158400}{2304} = 68.75 k \uparrow$$



Reactions

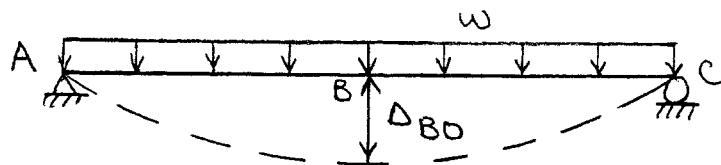


Shear Diagram (k)

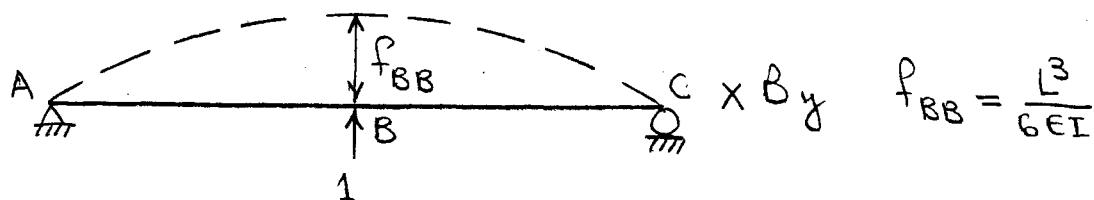


Bending Moment Diagram (k-ft)

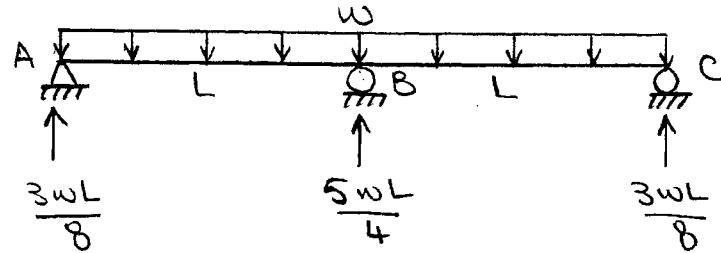
13.10



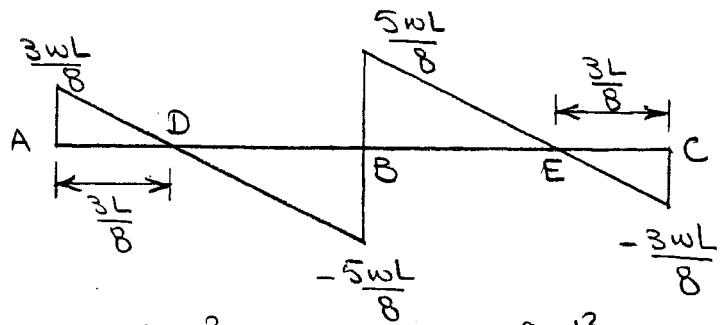
$$\Delta_{BO} = -\frac{5wL^4}{24EI}$$



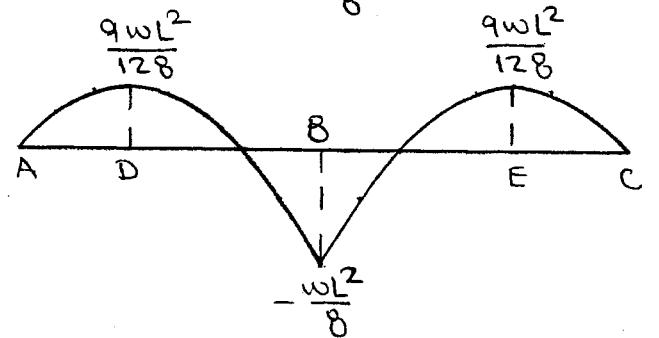
$$\Delta_{BO} + f_{BB} B_y = 0 \quad B_y = \frac{5wL^4}{24EI} \left(\frac{6EI}{L^3} \right) = \frac{5wL}{4} \uparrow$$



Reactions

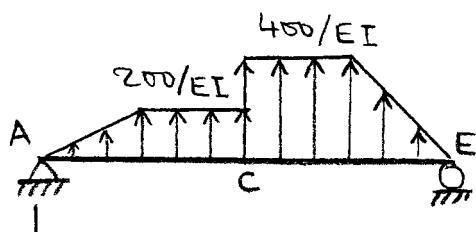
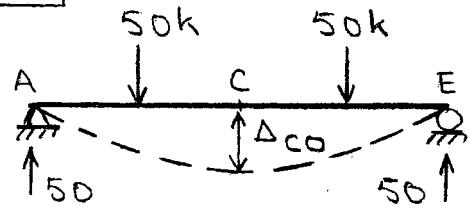


Shear Diagram



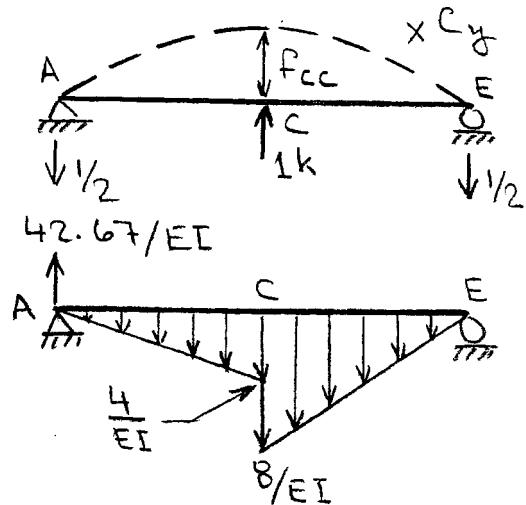
Bending Moment
Diagram

13.11



$$\underline{3133.33}$$

EI
Conjugate Beam for
External Loading



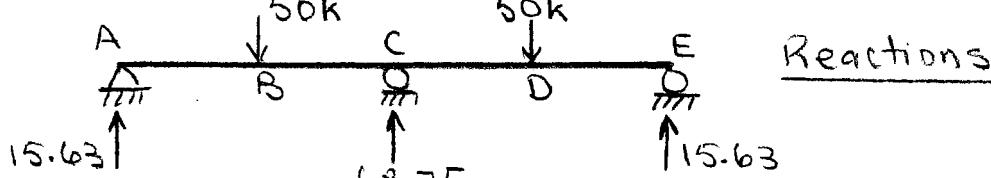
Conjugate Beam for
Unit Value of
Redundant C_y

$$\Delta_{c_0} = -\frac{3133.33}{EI} (16) + \frac{1}{2} (8) \left(\frac{200}{EI} \right) \left(\frac{8}{3} + 8 \right) + \left(\frac{200}{EI} \right) 8 (4)$$

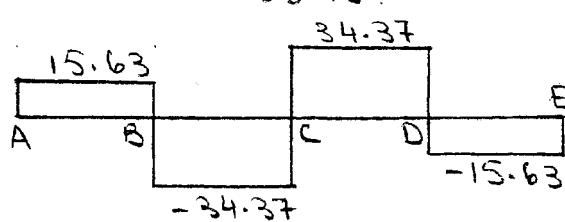
$$= -\frac{35200}{EI}$$

$$P_{cc} = \frac{42.67}{EI} (16) - \frac{1}{2} (16) \left(\frac{4}{EI} \right) \left(\frac{16}{3} \right) = \frac{512}{EI}$$

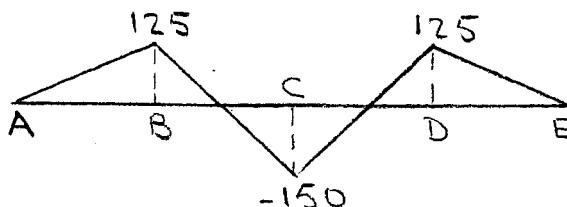
$$\Delta_{c_0} + P_{cc} C_y = 0 \quad C_y = \frac{35200}{512} = 68.75 k \uparrow$$



Reactions

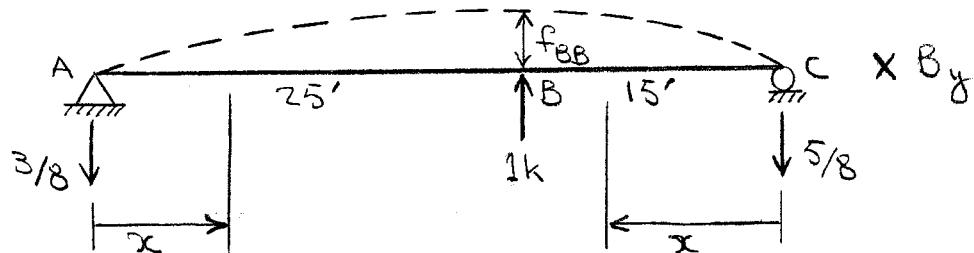
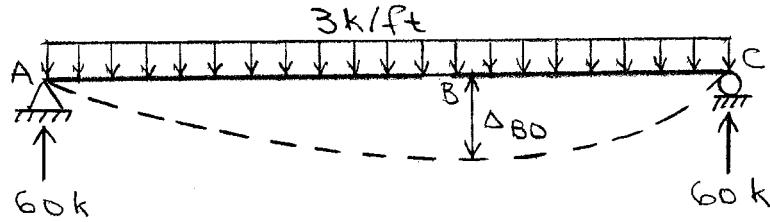


Shear Diagram (k)



Bending Moment
Diagram (k-ft)

13.12



Using the virtual work method:

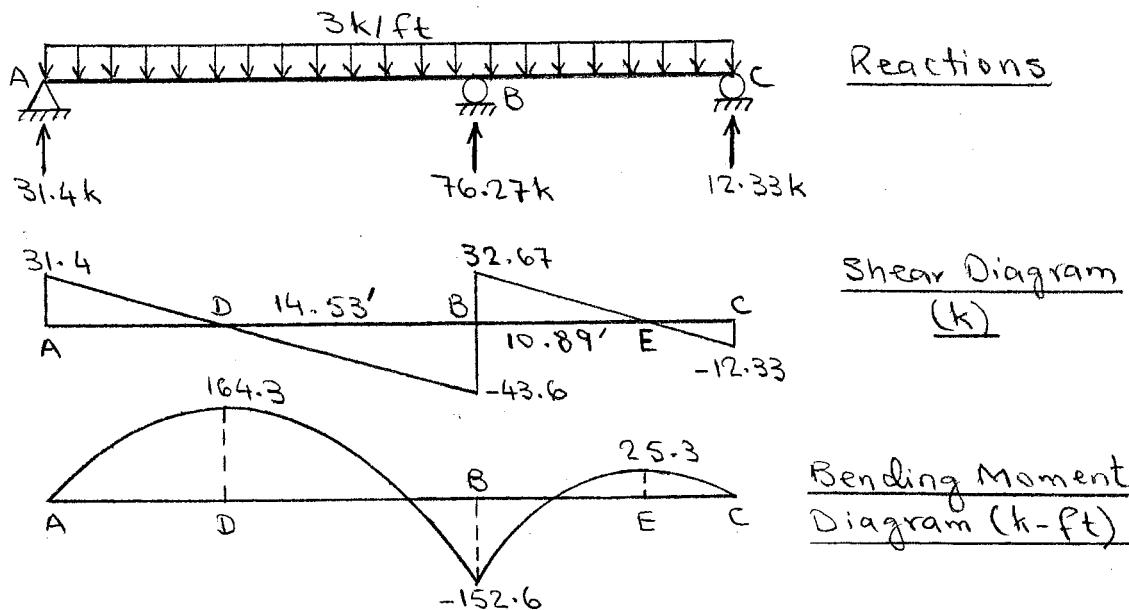
$$\Delta_{B0} = \frac{1}{EI} \left[\frac{1}{2} \int_0^{25} (60x - \frac{3x^2}{2}) (-\frac{3x}{8}) dx + \int_0^{15} (60x - \frac{3x^2}{2}) (-\frac{5x}{8}) dx \right]$$

$$= -\frac{61450.2 \text{ k-ft}^3}{EI}$$

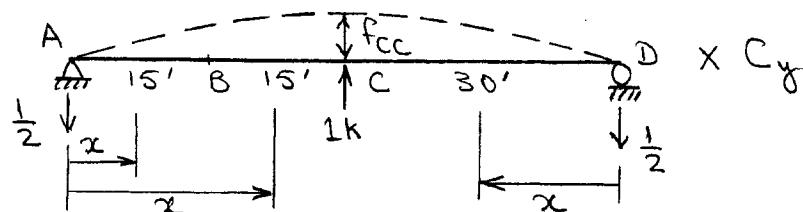
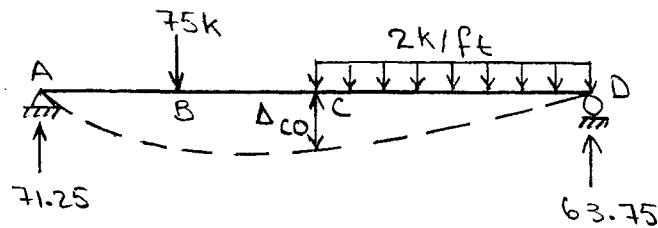
$$f_{BB} = \frac{1}{EI} \left[\frac{1}{2} \int_0^{25} (-\frac{3x}{8})^2 dx + \int_0^{15} (-\frac{5x}{8})^2 dx \right] = \frac{805.66 \text{ k-ft}^3/\text{k}}{EI}$$

Compatibility Equation:

$$\Delta_{B0} + f_{BB} B_y = 0 \quad B_y = \frac{61450.2}{805.66} = 76.27 \text{ k} \uparrow$$



13.13



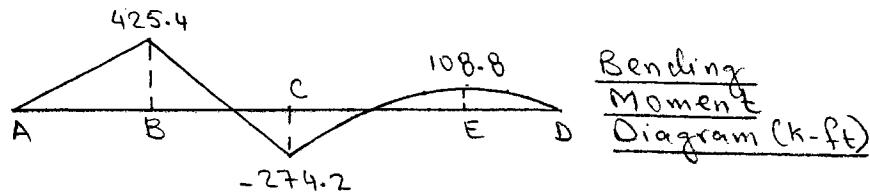
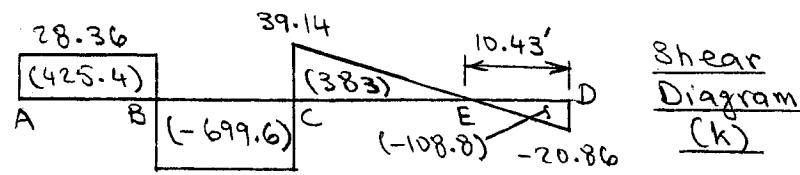
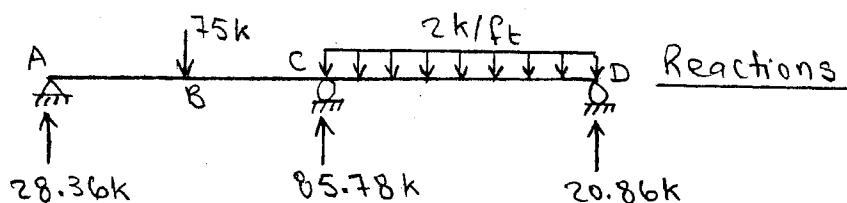
Using the method of virtual work:

$$\begin{aligned}\Delta_{CO} &= \frac{1}{EI} \left[\frac{1}{3} \int_0^{15} (71.25x) \left(-\frac{x}{2}\right) dx \right. \\ &\quad + \frac{1}{3} \int_{15}^{30} \{71.25x - 75(x-15)\} \left(-\frac{x}{2}\right) dx \\ &\quad \left. + \int_0^{30} (63.75x - x^2) \left(-\frac{x}{2}\right) dx \right] \\ &= -\frac{257343.75}{3000} \text{ k-ft}^3\end{aligned}$$

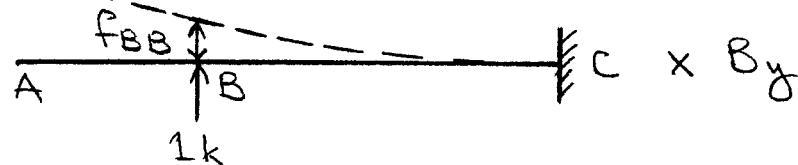
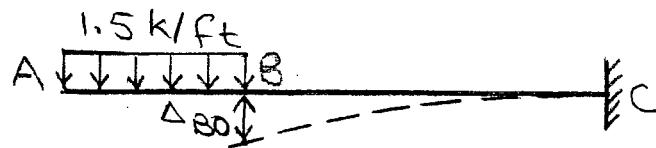
$$f_{CC} = \frac{1}{EI} \left[\frac{1}{3} \int_0^{30} \left(-\frac{x}{2}\right)^2 dx + \int_0^{30} \left(-\frac{x}{2}\right)^2 dx \right] = \frac{3000 \text{ k-ft}^3/k}{EI}$$

Compatibility Equation: $\Delta_{CO} + f_{CC} C_y = 0$

$$C_y = \frac{257343.75}{3000} = 85.78 \text{ k } \uparrow$$



13.14

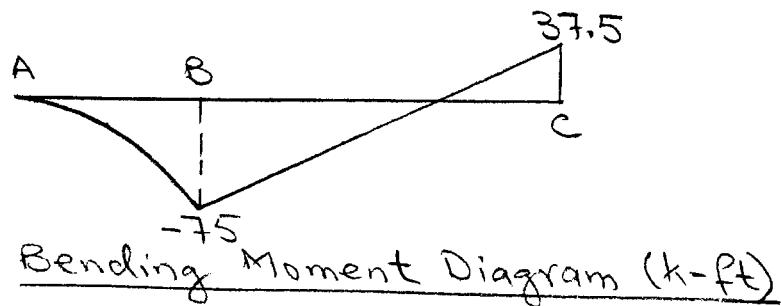
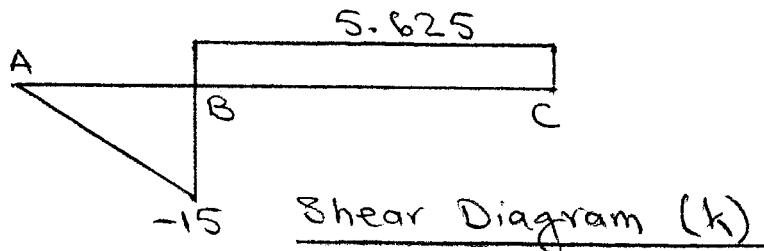
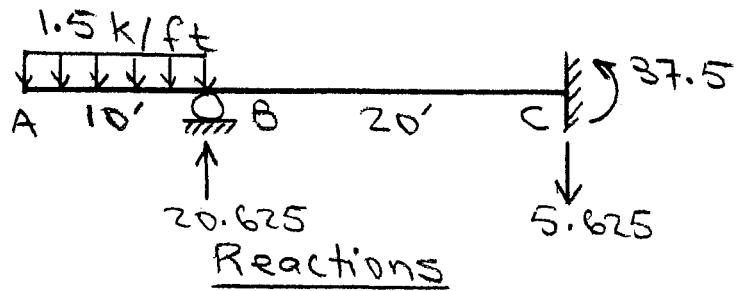


Using beam deflection formulas:

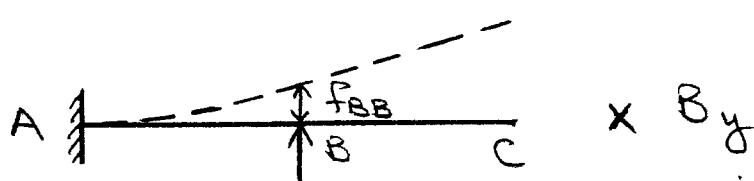
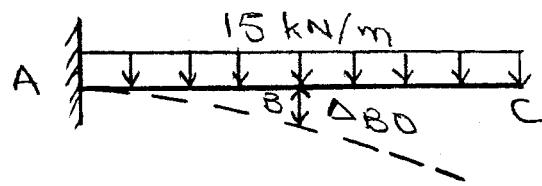
$$\Delta_{B0} = -\frac{55000 \text{ k-ft}^3}{EI} \quad f_{BB} = \frac{2666.67 \text{ k-ft}^3/\text{k}}{EI}$$

Compatibility Equation:

$$\Delta_{B0} + f_{BB} B_y = 0 \quad B_y = \frac{55000}{2666.67} = 20.625 \text{ k} \uparrow$$



13.15

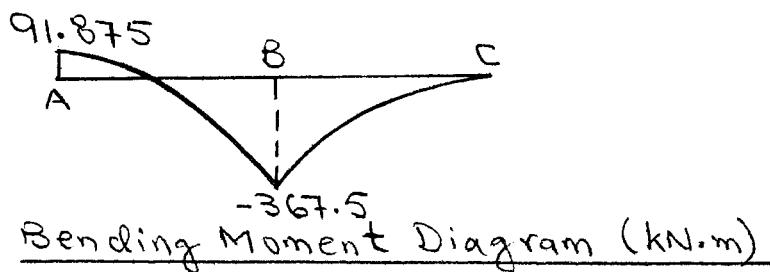
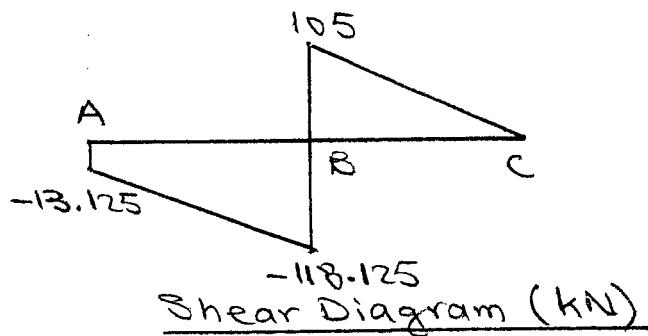
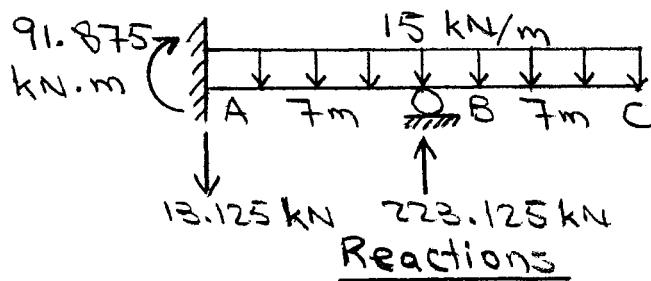


Using beam deflection formulas:

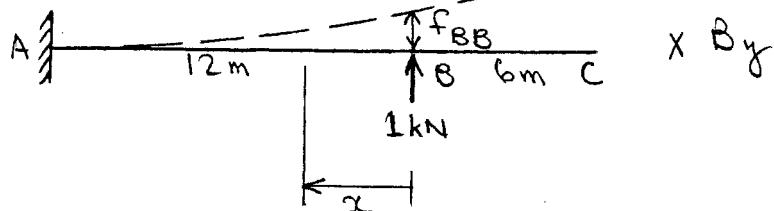
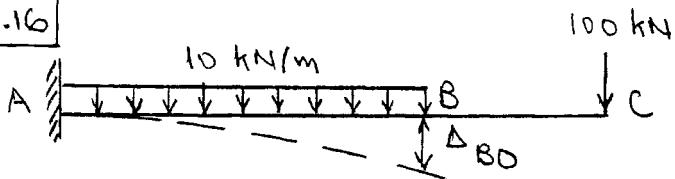
$$\Delta_{B0} = -\frac{25510.625 \text{ kN}\cdot\text{m}^3}{EI} \quad f_{BB} = \frac{114.333 \text{ kN}\cdot\text{m}^3/\text{kN}}{EI}$$

Compatibility Equation: $\Delta_{B0} + f_{BB} B_y = 0$

$$B_y = \frac{25510.625}{114.333} = 223.125 \text{ kN} \uparrow$$



B.16



Using the virtual work method:

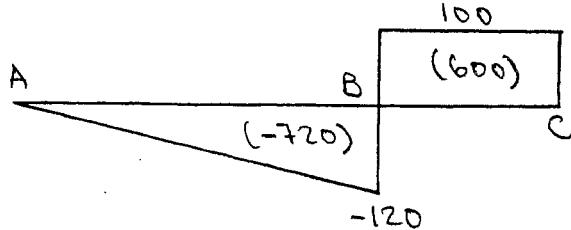
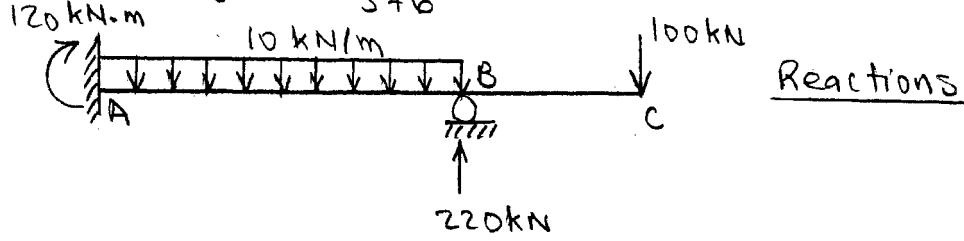
$$\Delta_{BD} = \frac{1}{EI} \left[\int_0^{12} \left\{ -100(6+x) - 5x^2 \right\} x \, dx \right]$$

$$= -\frac{126720}{EI} \text{ kN-m}^3$$

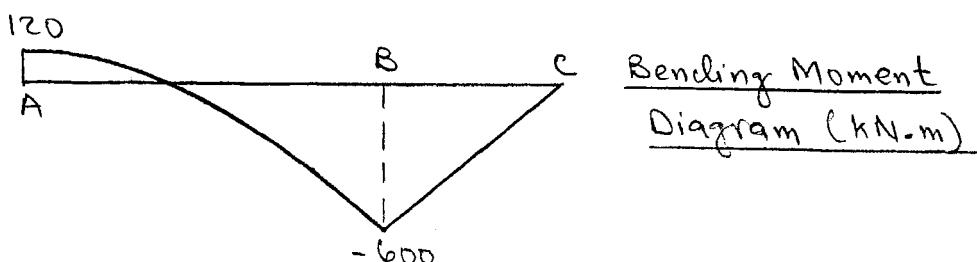
$$f_{BB} = \frac{1}{EI} \left[\int_0^{12} x^2 \, dx \right] = \frac{576}{EI} \text{ kN-m}^3/\text{kN}$$

Compatibility Equation: $\Delta_{BD} + f_{BB} B_y = 0$

$$B_y = \frac{126720}{576} = 220 \text{ kN } \uparrow$$

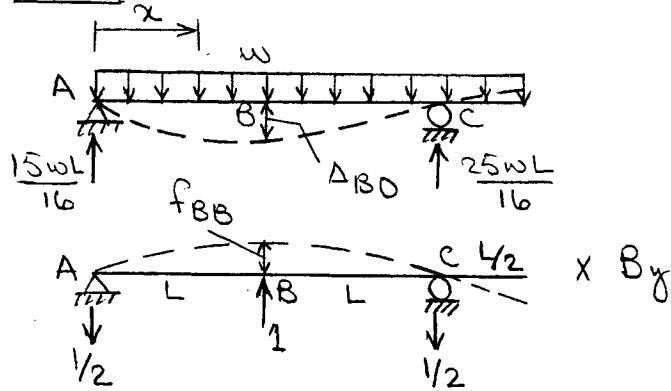


Shear Diagram
(kN)



Bending Moment
Diagram (kN-m)

13-17



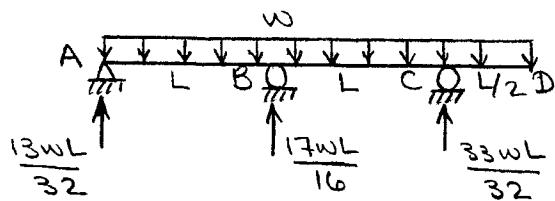
Using the virtual work method:

$$\begin{aligned}\Delta_{BD} &= \frac{1}{EI} \left[\int_0^L \left(\frac{15wLx}{16} - \frac{wx^2}{2} \right) \left(-\frac{x}{2} \right) dx \right. \\ &\quad \left. + \int_L^{2L} \left(\frac{15wLx}{16} - \frac{wx^2}{2} \right) \left\{ -\frac{x}{2} + 1(x-L) \right\} dx \right] \\ &= -\frac{17wL^4}{96EI}\end{aligned}$$

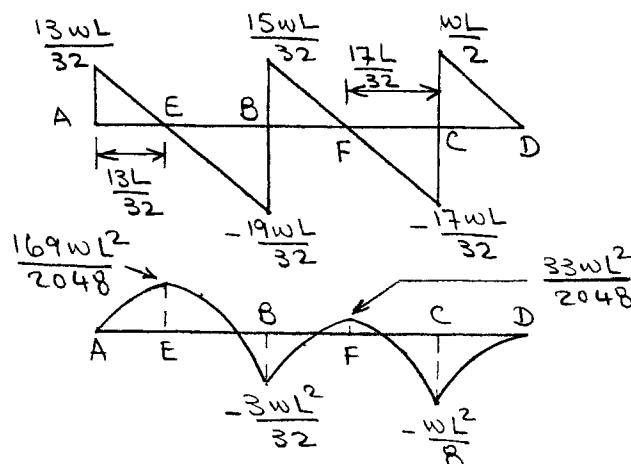
$$f_{BB} = \frac{2}{EI} \int_0^L \left(-\frac{x}{2} \right)^2 dx = \frac{L^3}{6EI}$$

Compatibility Equation: $\Delta_{BD} + f_{BB} B_y = 0$

$$B_y = \frac{17wL^4}{96EI} \left(\frac{6EI}{L^3} \right) = \frac{17wL}{16} \uparrow$$



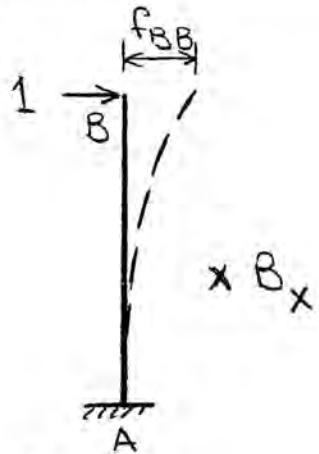
Reactions



Shear Diagram

Bending Moment Diagram

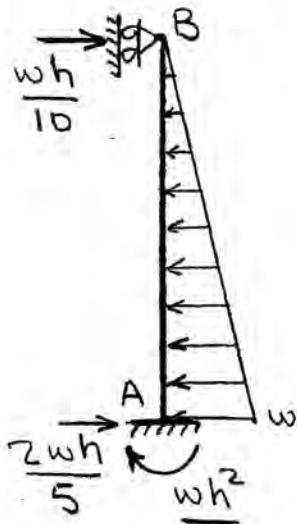
13-18



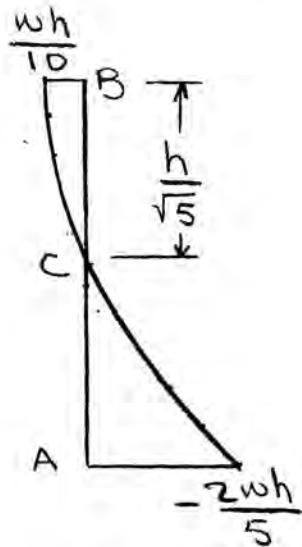
Using beam deflection formulas:

$$\Delta_{BD} = -\frac{wh^4}{30EI}; \quad f_{BB} = \frac{h^3}{3EI}$$

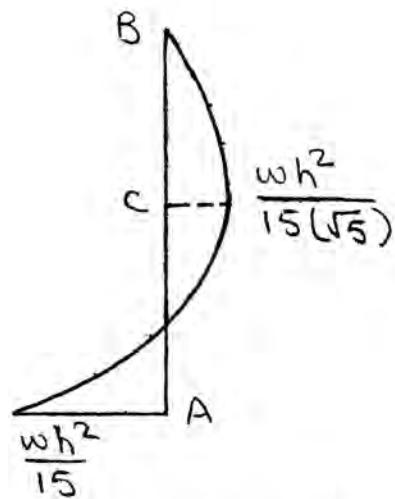
$$\Delta_{BD} + f_{BB} B_x = 0 \quad B_x = \left(\frac{wh^4}{30EI}\right) \left(\frac{3EI}{h^3}\right) = \frac{wh}{10} \rightarrow$$



Reactions

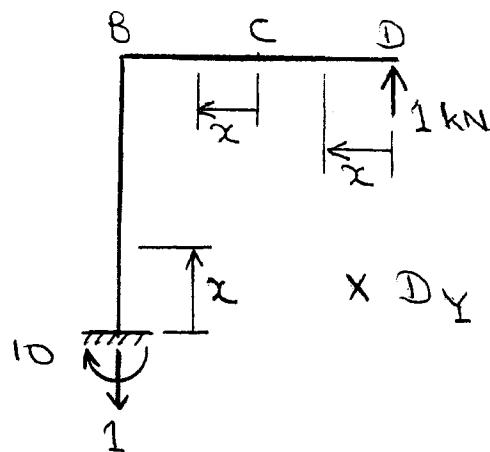
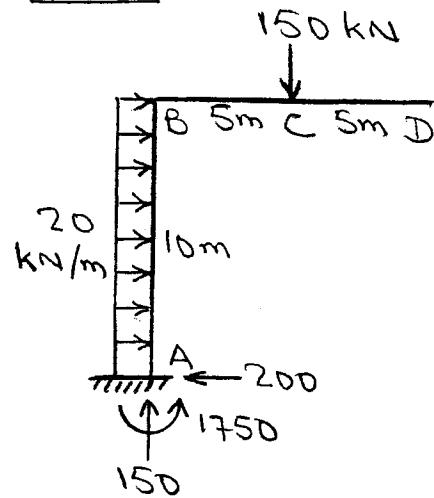


Shear Diagram



Bending Moment Diagram

13.19



Using the virtual work method:

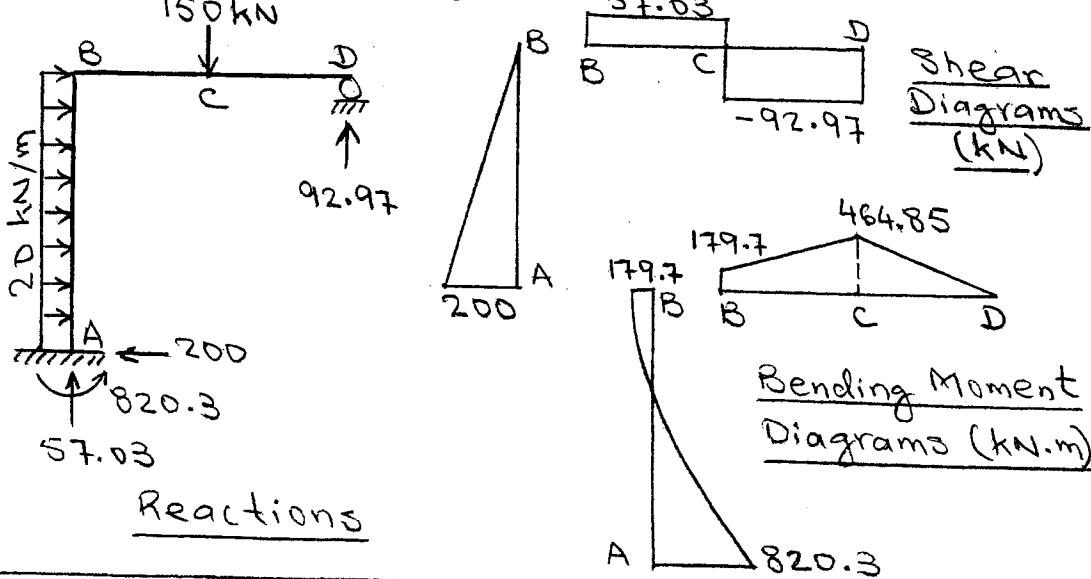
$$\Delta_{DD} = \frac{1}{EI} \left[\int_0^{10} (-1750 + 200x - 10x^2)(10) dx + \int_0^5 (-150x) 1(x+5) dx \right]$$

$$= -\frac{123958.33}{EI} \text{ kN.m}^3$$

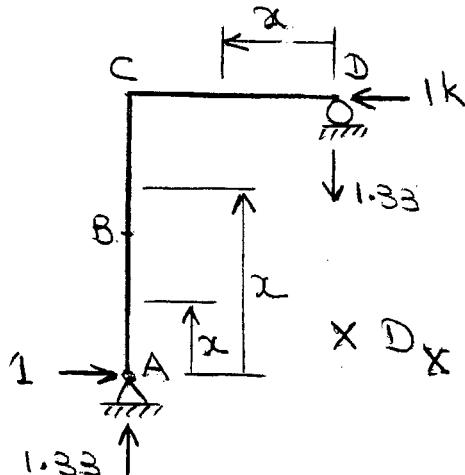
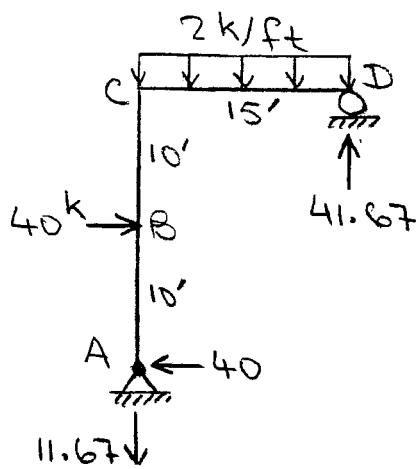
$$f_{DD} = \frac{1}{EI} \left[\int_0^{10} (10)^2 dx + \int_0^5 (x)^2 dx \right] = \frac{1333.33}{EI} \text{ kN.m}^3/\text{kN}$$

Compatibility Equation: $\Delta_{DD} + f_{DD} D_y = 0$

$$D_y = \frac{123958.33}{1333.33} = \frac{92.97}{57.03} \text{ kN} \uparrow$$



13.20



Using the virtual work method:

$$\Delta_{DD} = \frac{1}{EI} \left[\int_0^{10} (40x)(-x) dx + \int_{10}^{20} (400)(-x) dx + \int_0^{15} (41.67x - x^2)(-1.33x) dx \right]$$

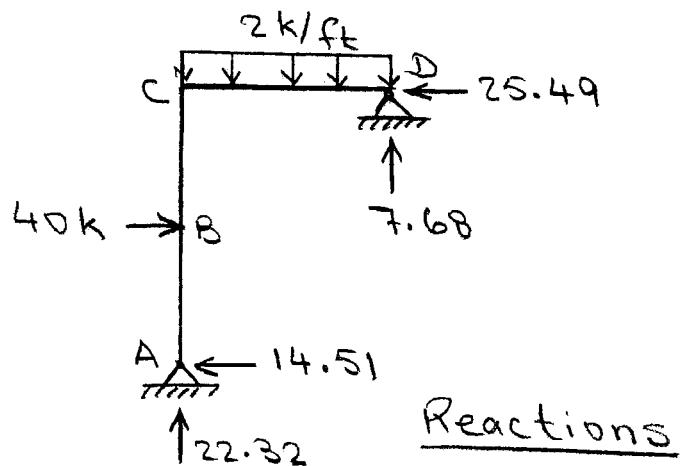
$$= -\frac{118958.33 \text{ k-ft}^3}{EI}$$

$$f_{DD} = \frac{1}{EI} \left[\int_0^{20} (-x)^2 dx + \int_0^{15} (-1.33x)^2 dx \right]$$

$$= \frac{4666.67 \text{ k-ft}^3/k}{EI}$$

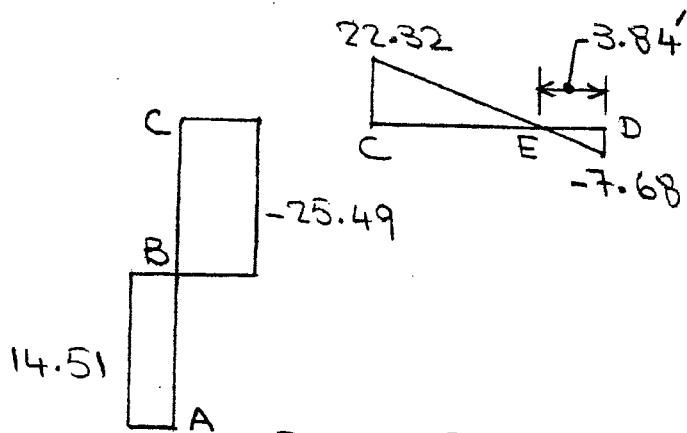
Compatibility Equation: $\Delta_{DD} + f_{DD} D_x = 0$

$$D_x = \frac{118958.33}{4666.67} = 25.49 \text{ k} \leftarrow$$

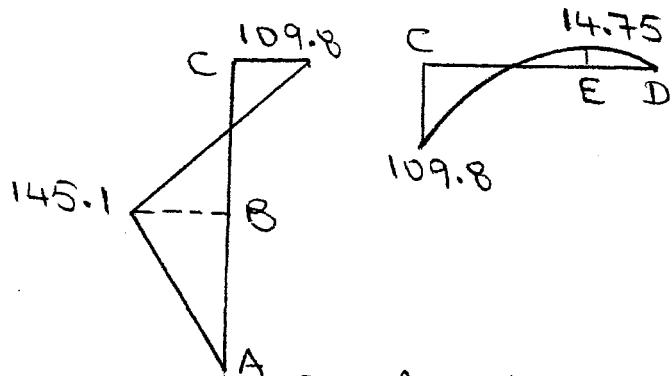


Reactions

13.20 (cont'd.)

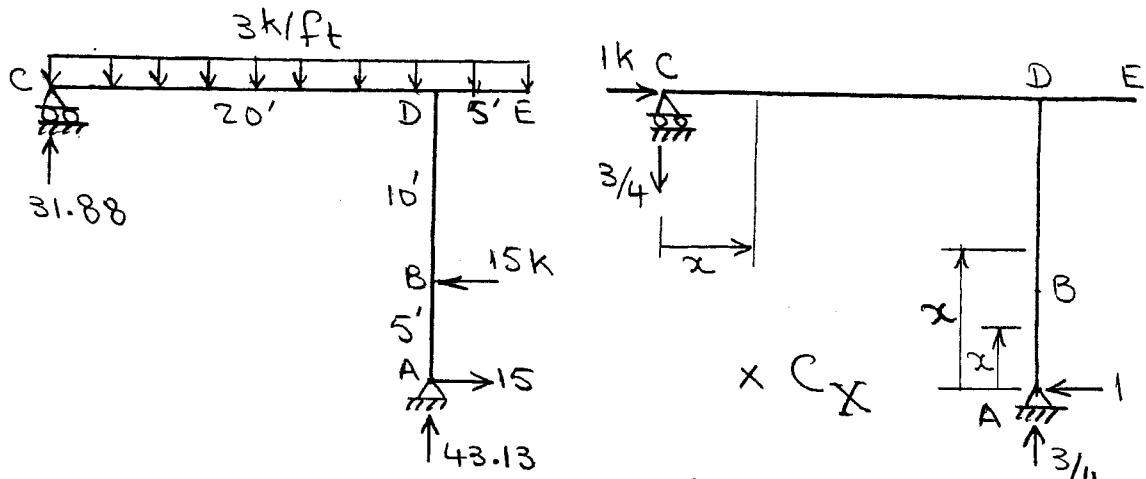


Shear Diagrams (k)



Bending Moment Diagrams (k-ft)

13.21



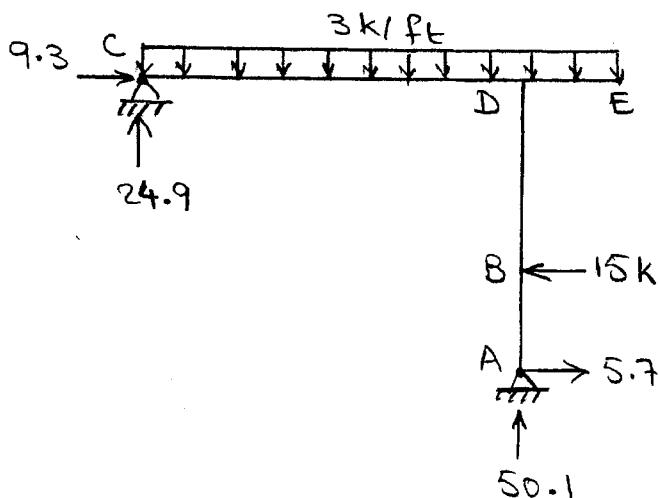
Using the virtual work method:

$$\Delta_{co} = \frac{1}{EI} \left[\frac{1}{2} \int_0^{20} (31.88x - 1.5x^2) (-\frac{3}{4}x) dx + \int_0^5 (-15x)x dx + \int_5^{15} \{-15x + 15(x-5)\} x dx \right] = -\frac{17505 \text{ k-ft}^3}{EI}$$

$$f_{cc} = \frac{1}{EI} \left[\frac{1}{2} \int_0^{20} (-\frac{3}{4}x)^2 dx + \int_0^{15} (x)^2 dx \right] = \frac{1875 \text{ k-ft}^3/k}{EI}$$

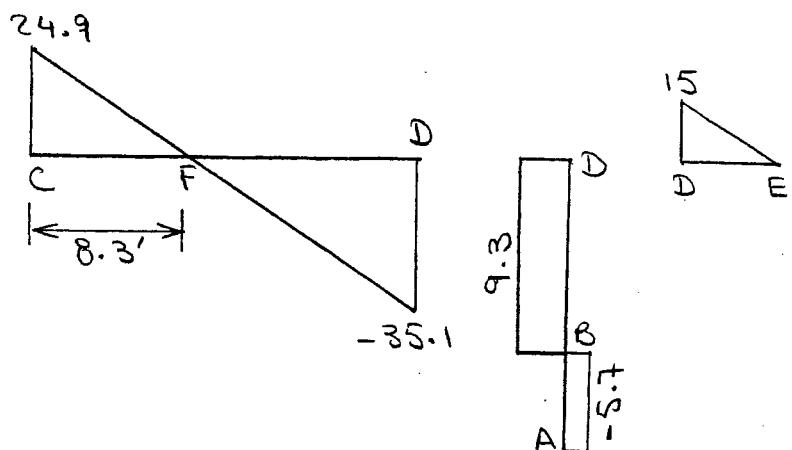
Compatibility Equation: $\Delta_{co} + f_{cc} C_x = 0$

$$C_x = \frac{17505}{1875} = \underline{9.3 \text{ k}} \rightarrow$$

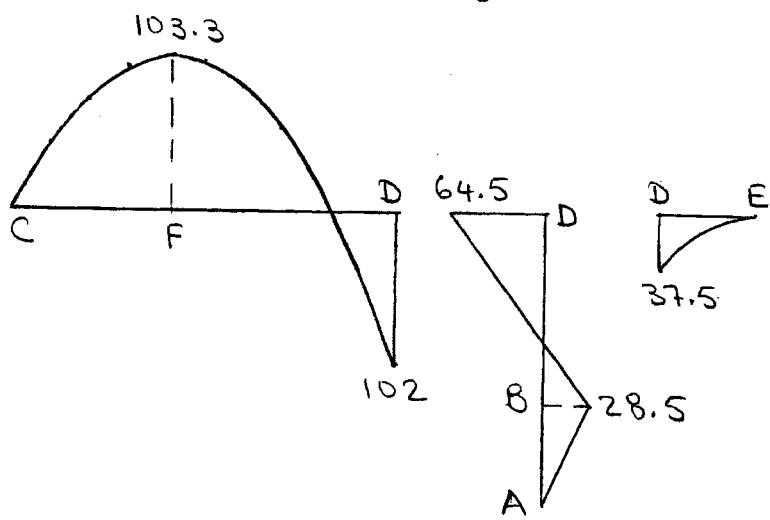


Reactions

13.21 (contd.)

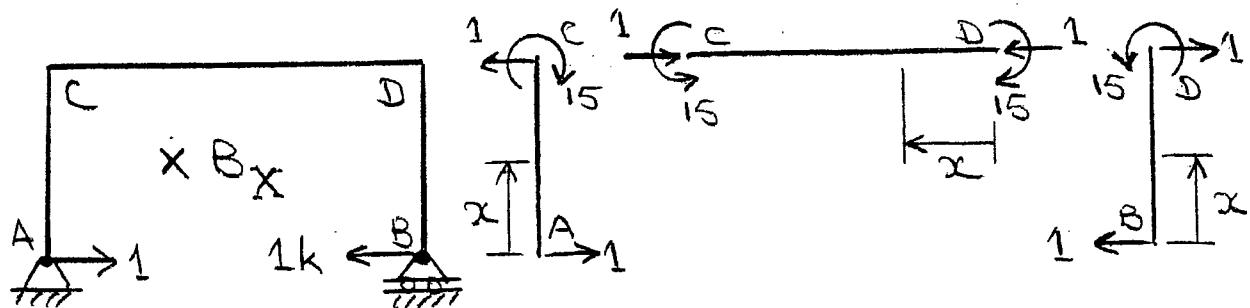
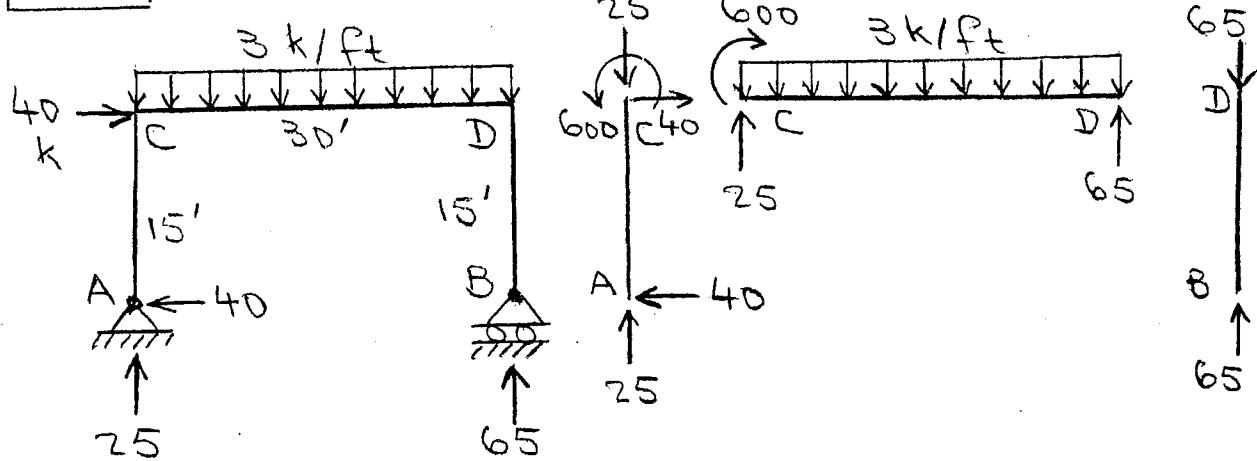


Shear Diagrams (k)



Bending Moment Diagrams (k-ft)

13.22



Using the virtual work method:

$$\Delta_{BD} = \frac{1}{EI} \left[\int_0^{15} (40x)(-x) dx + \int_0^{30} (65x - 1.5x^2)(-15) dx \right]$$

$$= -\frac{281250 \text{ k-ft}^3}{EI}$$

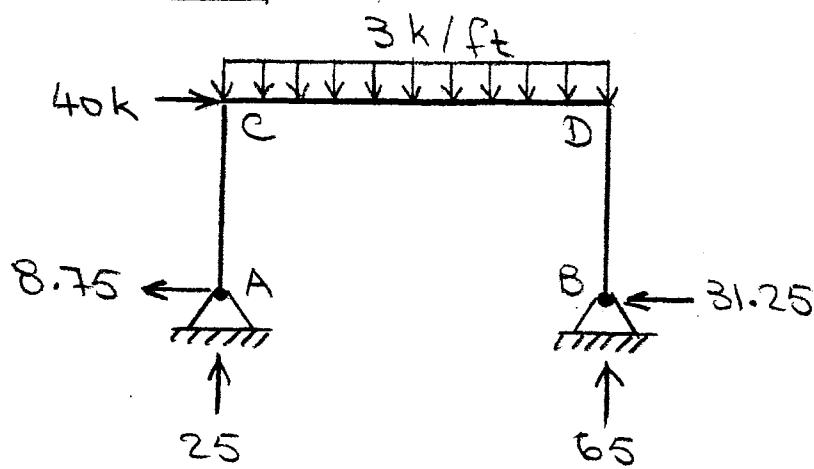
$$f_{BB} = \frac{1}{EI} \left[\int_0^{15} (-x)^2 dx + \int_0^{30} (-15)^2 dx + \int_0^{15} x^2 dx \right]$$

$$= \frac{9000 \text{ k-ft}^3 / k}{EI}$$

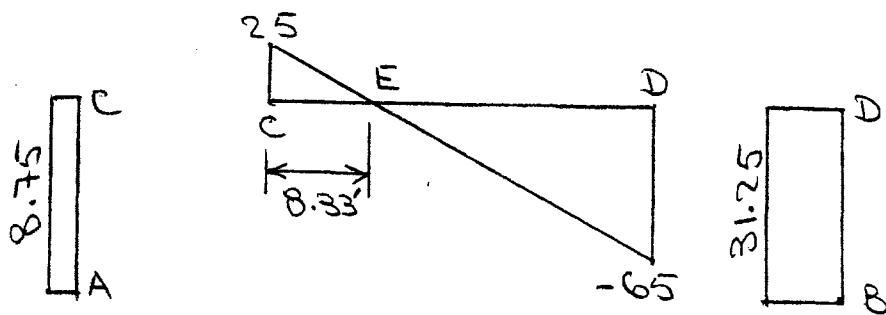
Compatibility Equation: $\Delta_{BD} + f_{BB} B_x = 0$

$$B_x = \frac{281250}{9000} = \underline{\underline{31.25 \text{ k}}} \leftarrow$$

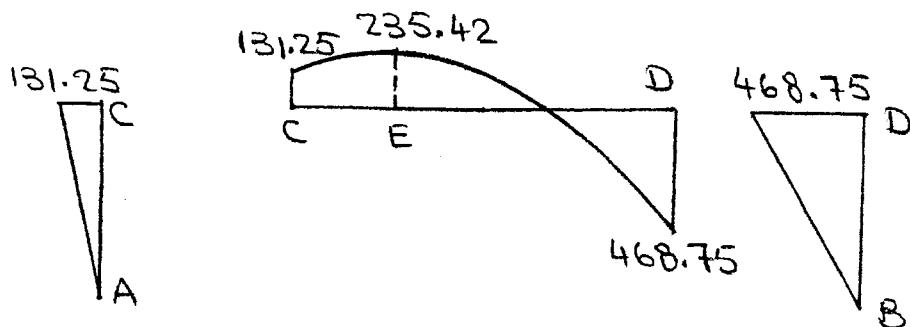
13.22 (Cont'd.)



Reactions

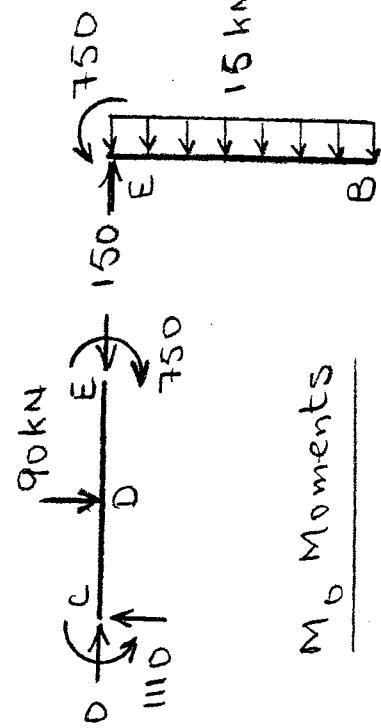
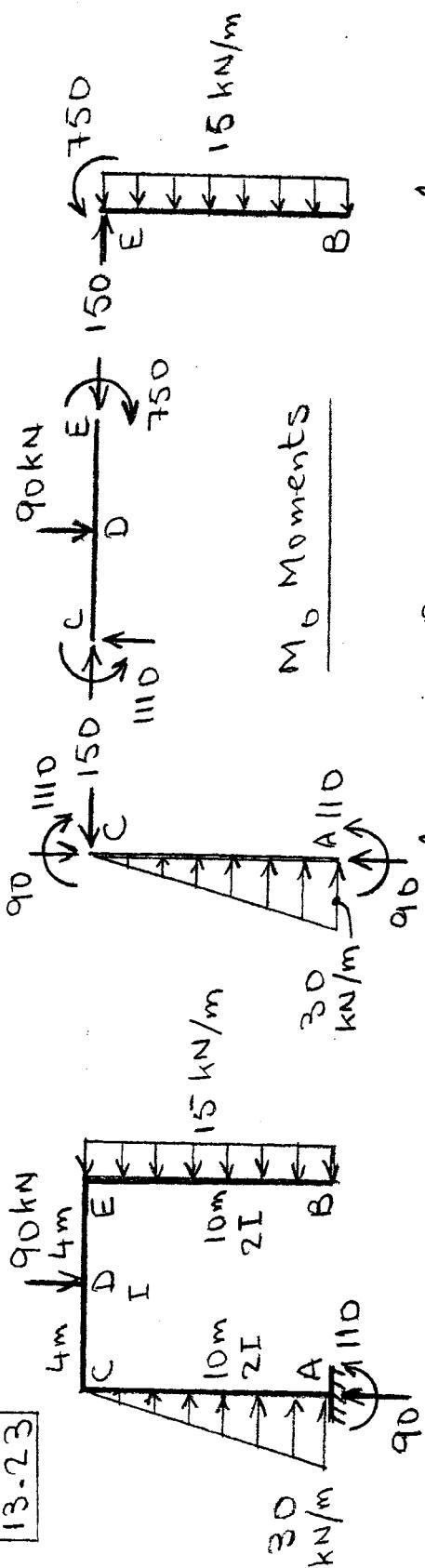


Shear Diagrams (k)

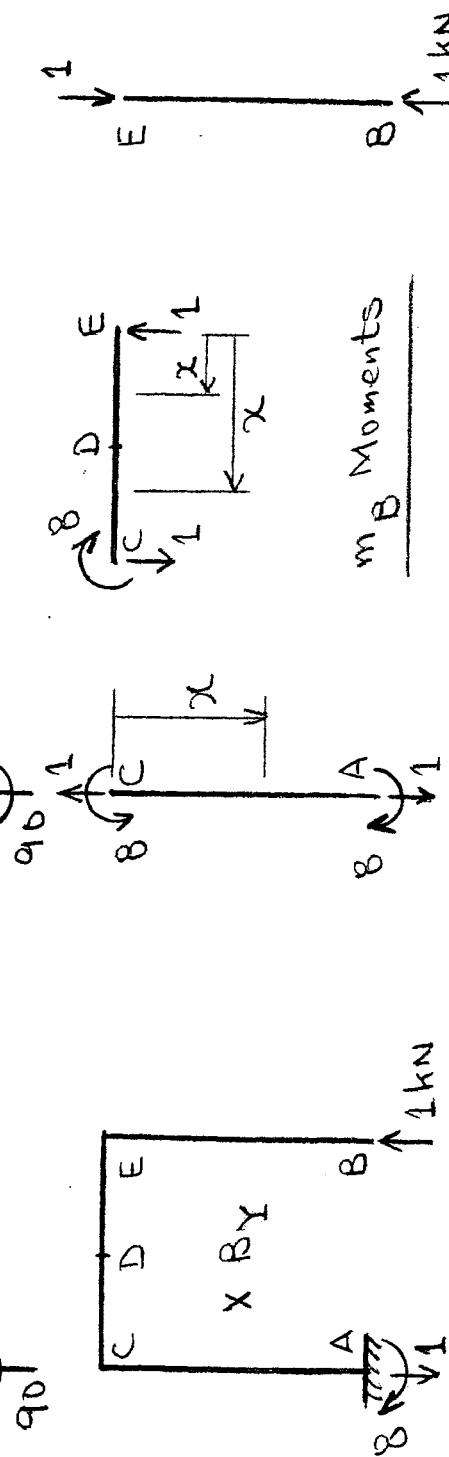


Bending Moment Diagrams (k-ft)

13.23



M_0 Moments



Segment	x coordinate Origin	Limits (m)	M_0 (kN.m)	m_B (kN.m/kN)
CA	C	0-10	$150x - 1110 - \frac{1}{2}(x)3x(\frac{1}{3})$	8
ED	E	0-4	-750	1x
DC	E	4-8	$-750 - 90(x-4)$	1x

13.23 (contd.)

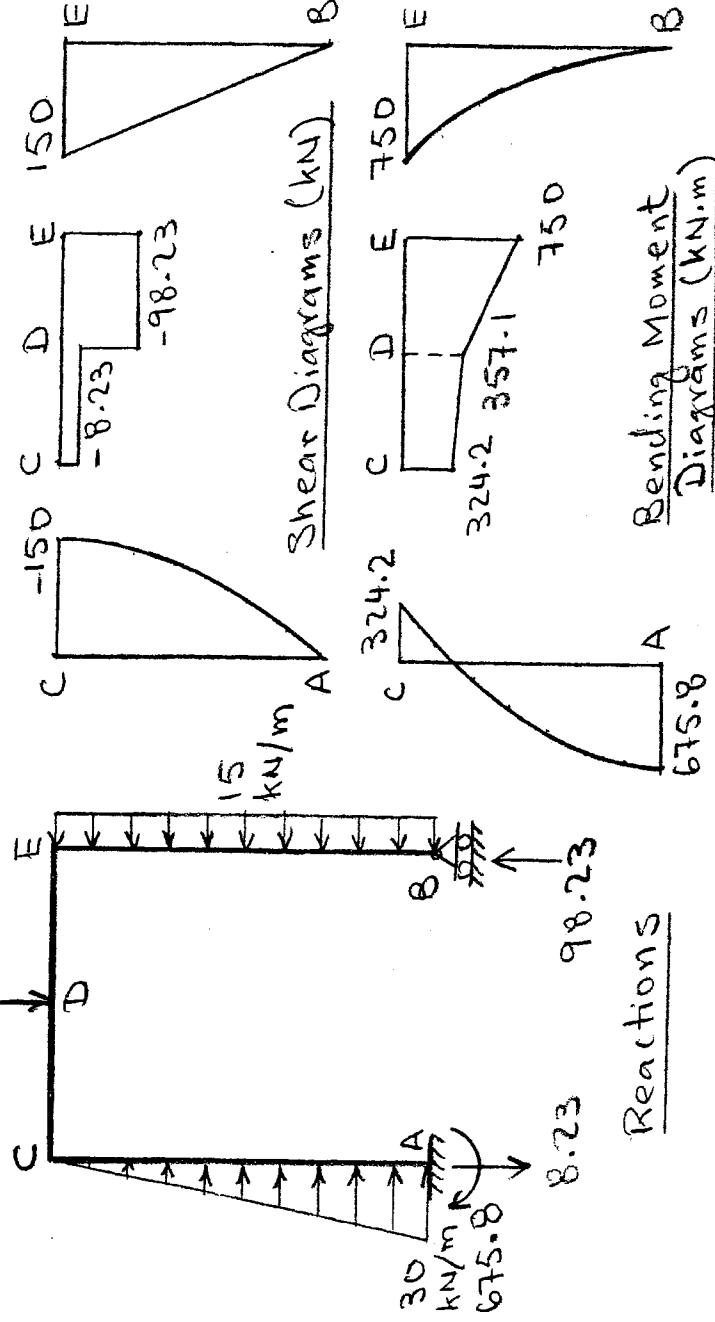
Using the virtual work method:

$$\Delta_{B_0} = \sum \int \frac{M_{0B}}{EI} dx = \frac{1}{EI} \left[\frac{1}{2} \int_0^{10} (150x - 1110 - \frac{x^3}{2})(8) dx + \int_0^4 (-750)x dx \right]$$

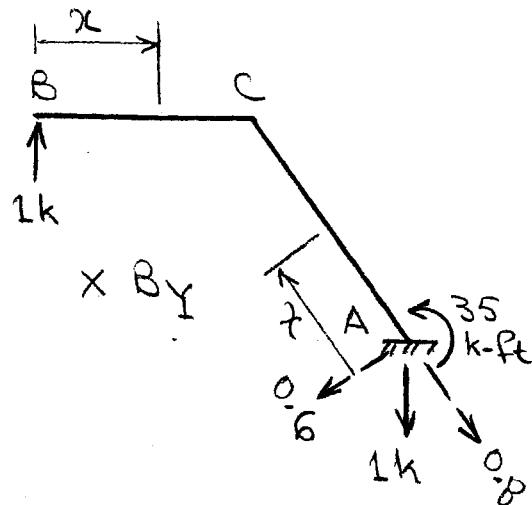
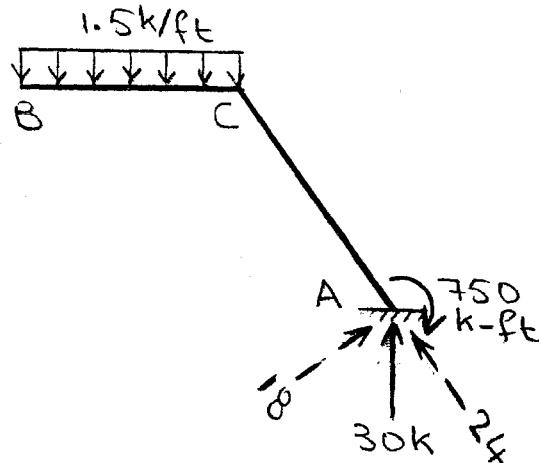
$$+ \int_4^8 (-390 - 90x)x dx = -\frac{48250}{EI} \text{ kN}\cdot\text{m}^3$$

$$f_{B_B} = \sum \int \frac{m_B}{EI} dx = \frac{1}{EI} \left[\frac{1}{2} \int_0^{10} 64 dx + \int_0^8 x^2 dx \right] = \frac{490.67}{EI} \text{ kN}\cdot\text{m}^3/\text{kN}$$

Compatibility Equation: $\Delta_{B_0} + f_{B_B} \beta_Y = 0$ $\beta_Y = \frac{48250}{490.67} = \frac{98.23}{1} \text{ kN} \uparrow$



13.24



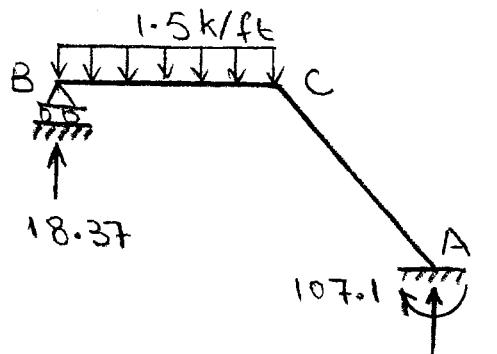
Using the virtual work method:

$$\Delta_{B_0} = \frac{1}{EI} \left[\int_0^{20} \left(-\frac{1.5x^2}{2} \right) (1x) dx + \int_0^{25} (18x - 750)(-0.6x + 35) dx \right] \\ = -\frac{405000 \text{ k-ft}^3}{EI}$$

$$f_{BB} = \frac{1}{EI} \left[\int_0^{20} (1x)^2 dx + \int_0^{25} (-0.6x + 35)^2 dx \right] \\ = \frac{22041.667 \text{ k-ft}^3/k}{EI}$$

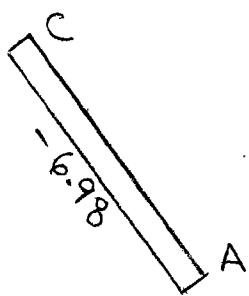
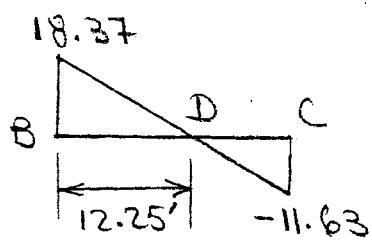
Compatibility Equation: $\Delta_{B_0} + f_{BB} B_Y = 0$

$$B_Y = \frac{405000}{22041.667} = 18.37 \text{ k } \uparrow$$

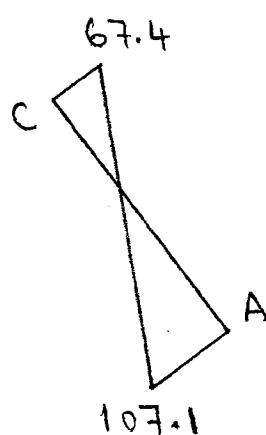
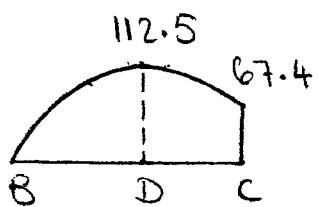


Reactions

13.24 (contd.)

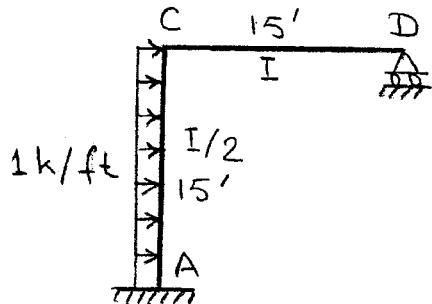


Shear Diagrams (k)

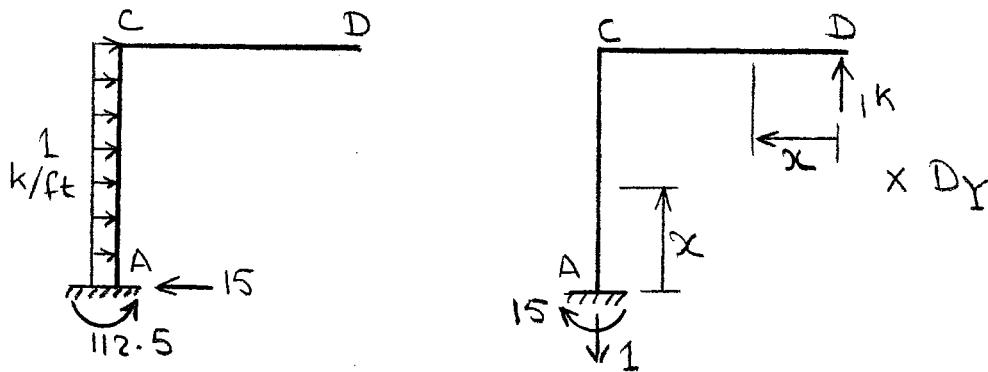


Bending Moment Diagrams (k-ft)

13-25 As the frame is symmetric subjected to antisymmetric loading, only a half of the structure needs to be analyzed.



Substructure for Analysis



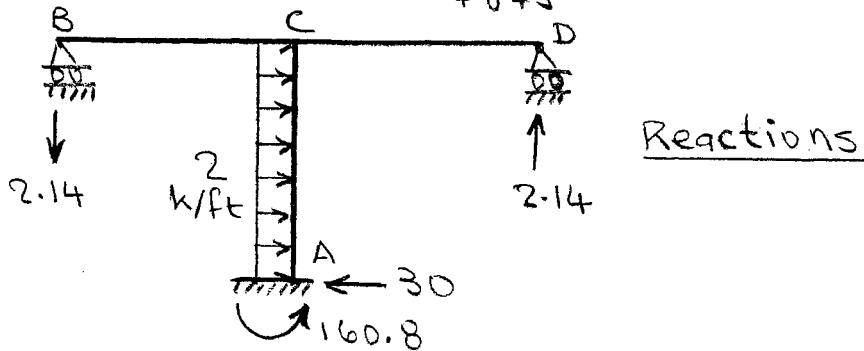
Using the virtual work method:

$$\Delta_{D_0} = \frac{1}{EI} \left[2 \int_0^{15} (-112.5 + 15x - \frac{x^2}{2})(15) dx \right] = -\frac{16875}{EI}$$

$$F_{DD} = \frac{1}{EI} \left[2 \int_0^{15} (15)^2 dx + \int_0^{15} (1x)^2 dx \right] = \frac{7875}{EI}$$

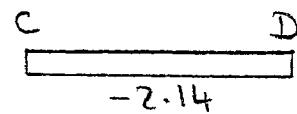
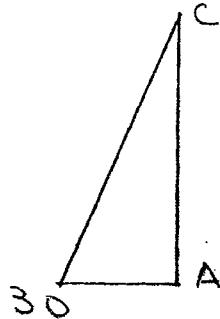
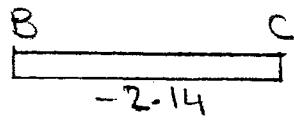
Compatibility Equation:

$$\Delta_{D_0} + F_{DD} D_Y = 0 \quad D_Y = \frac{16875}{7875} = \underline{2.14 \text{ k} \uparrow}$$

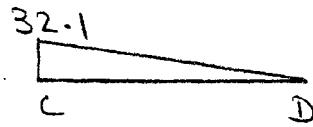
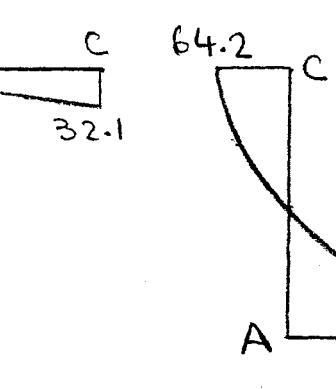
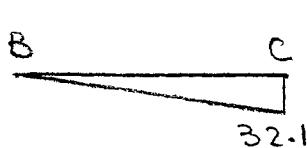


Reactions

13.25 (contd.)



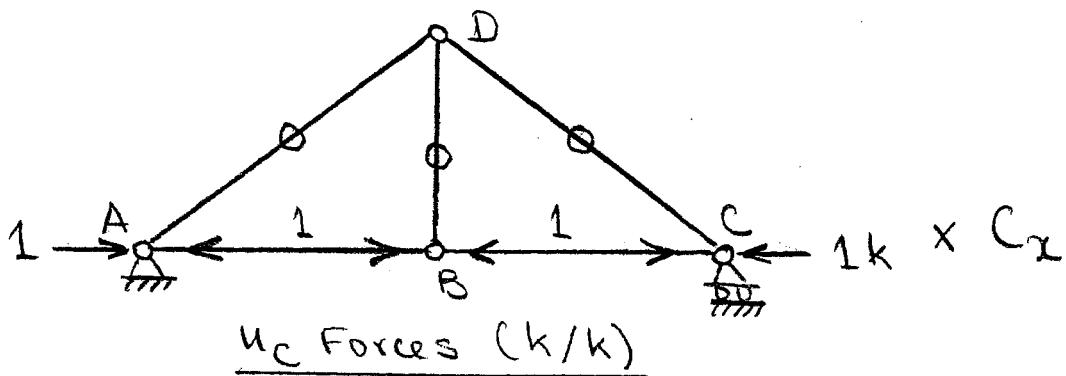
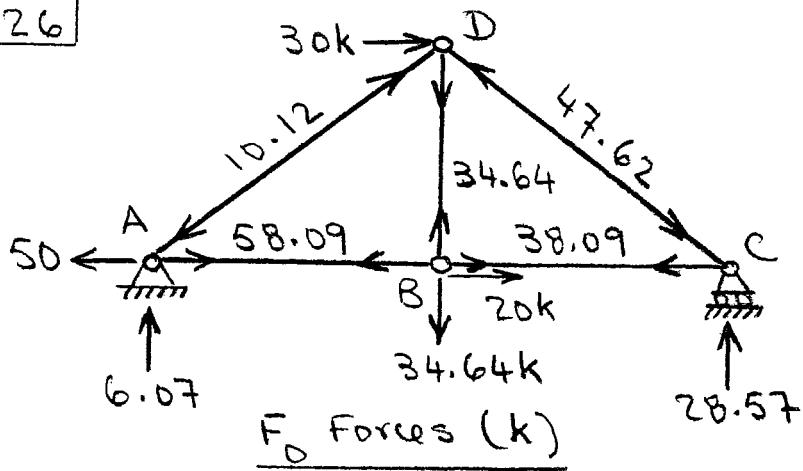
Shear Diagrams (k)



160.8

Bending Moment Diagrams (k-ft)

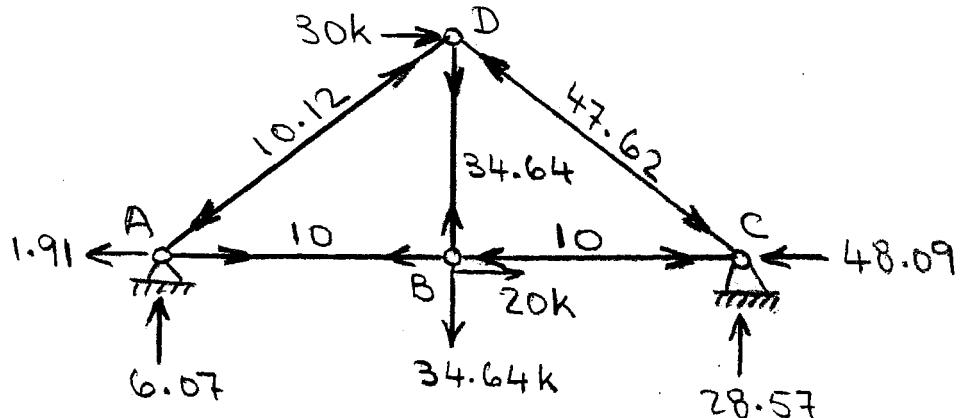
13.26

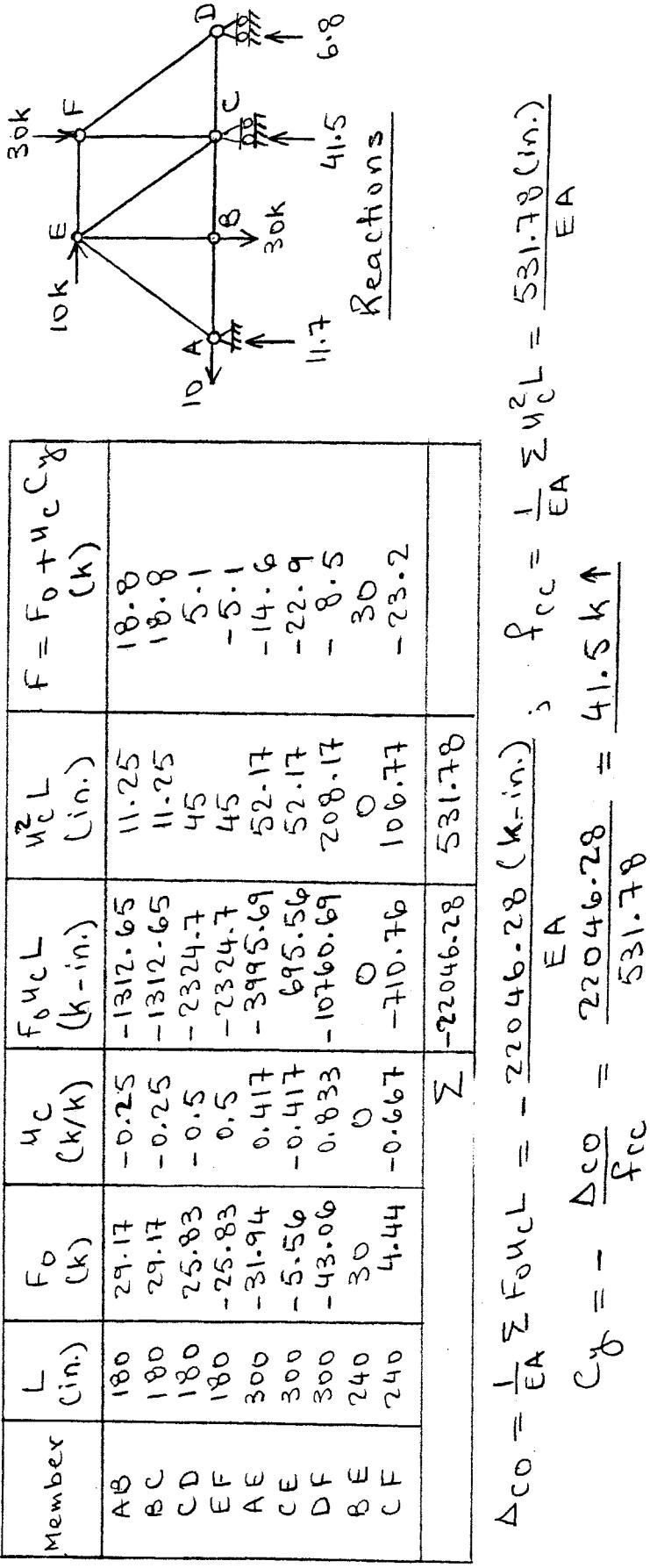
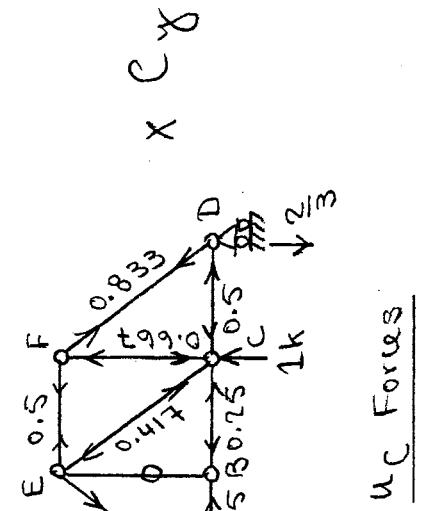
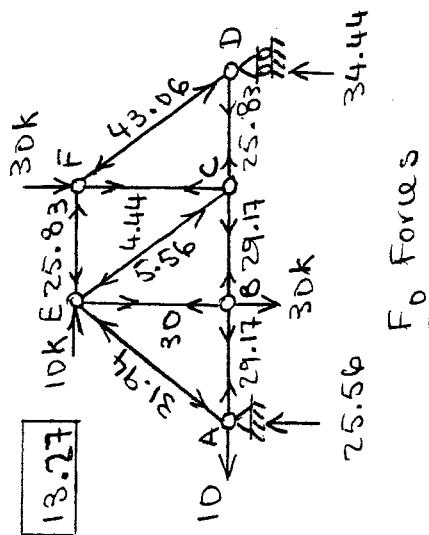


$$\Delta_{CO} = \frac{1}{EA} \sum F_0 u_C L = \frac{1}{EA} [58.09(-1)8 + 38.09(-1)8] \\ = -\frac{769.44 \text{ k-ft}}{EA}$$

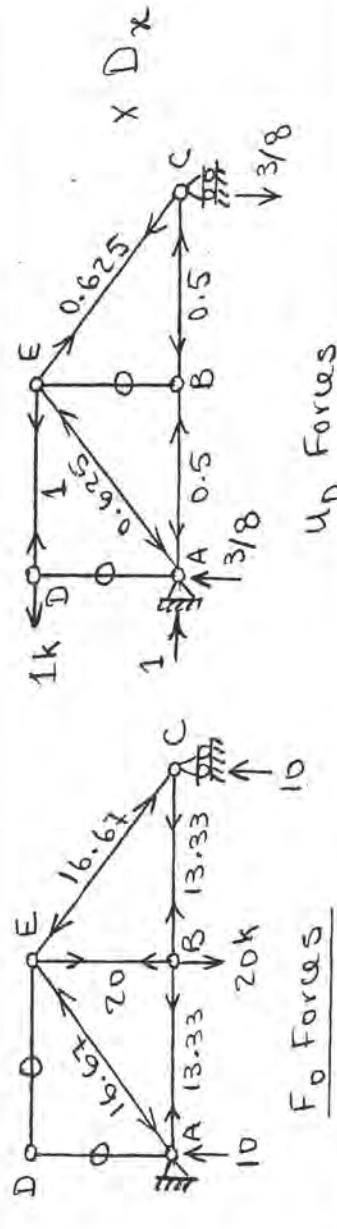
$$f_{CC} = \frac{1}{EA} \sum u_C^2 L = \frac{1}{EA} [2(-1)^2 8] = \frac{16 \text{ ft}}{EA}$$

$$C_2 = -\frac{\Delta_{CO}}{f_{CC}} = \frac{769.44}{16} = \underline{48.09 \text{ k} \leftarrow}$$

Member Forces and Reactions



13.78



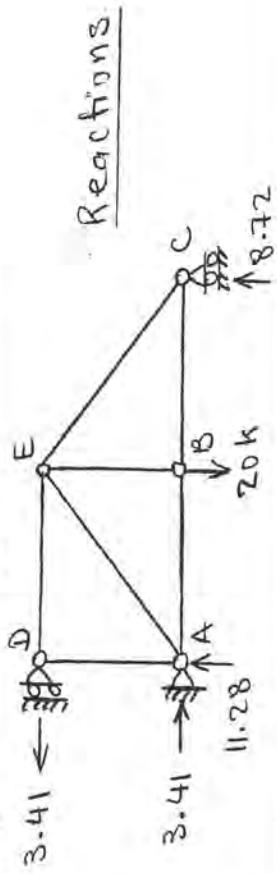
F₀ Forces

U_D Forces

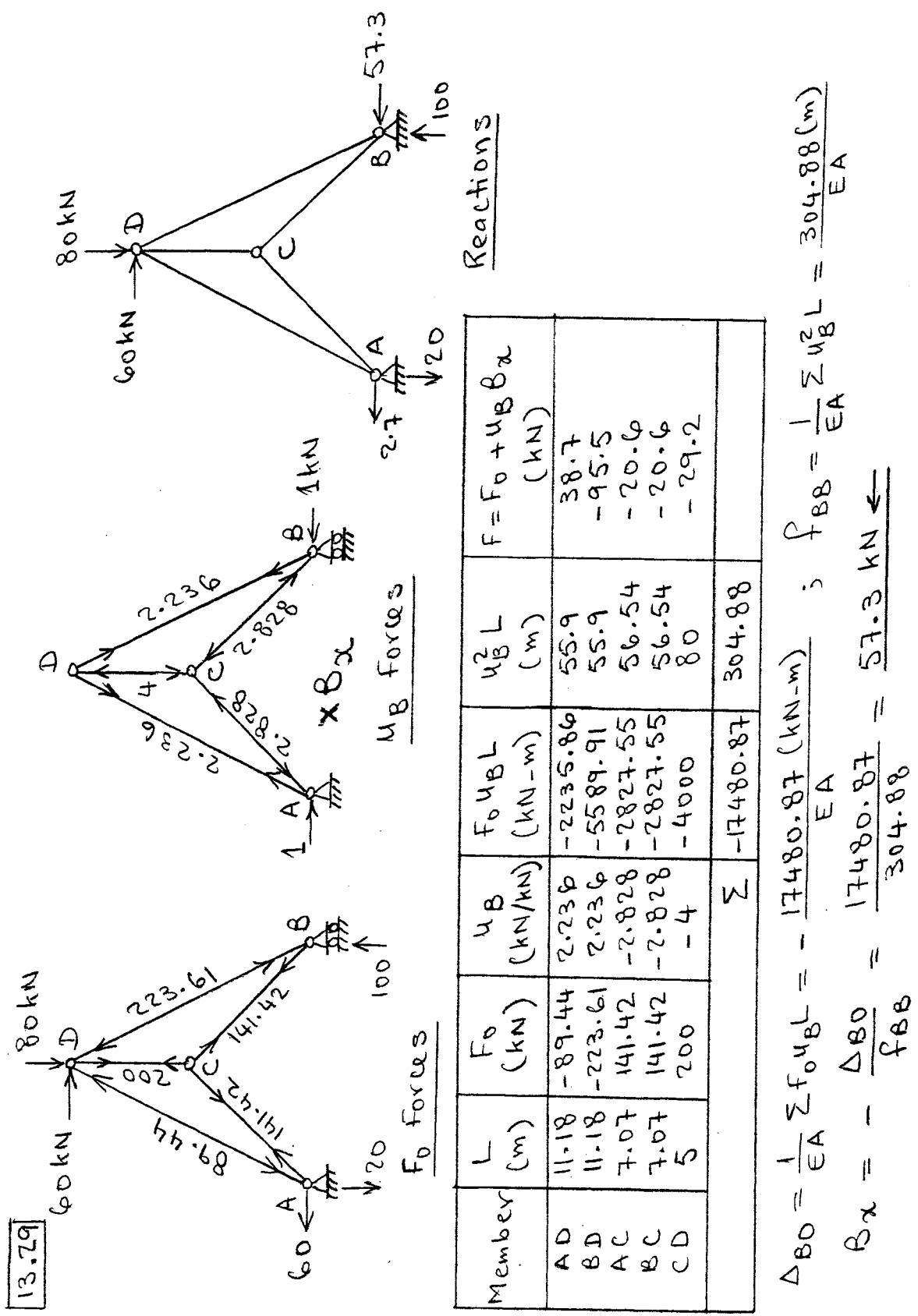
Member	L (in.)	A (in ²)	F ₀ (k)	U _D (k/in.)	F ₀ U _D L/A (k/in.)	U _D ² L/A (1/in.)	F = F ₀ + U _D Dx (k)
A B	192	20	13.33	-0.5	-159.96	6	11.63
B C	192	20	13.33	-0.5	-159.96	6	11.63
D E	192	0	0	1	0	24	3.41
A E	240	20	-16.67	-0.625	416.75	15.63	-18.8
C E	240	20	-16.67	0.625	-312.56	11.72	-14.54
A D	144	0	0	0	0	0	0
B E	144	20	0	0	0	0	20
Σ					-215.73	63.35	

$$\Delta_{D0} = \frac{1}{E} \sum \frac{F_0 U_{DL}}{A} = -\frac{215.73}{E} \text{ (k/in.)} ; \quad f_{DD} = \frac{1}{E} \sum \frac{U_{DL}^2}{A} = \frac{63.35}{E} \text{ (1/in.)}$$

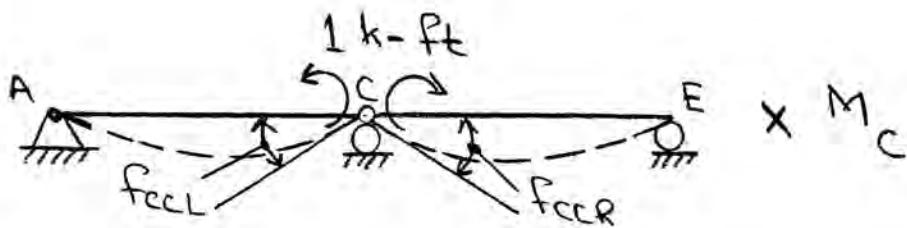
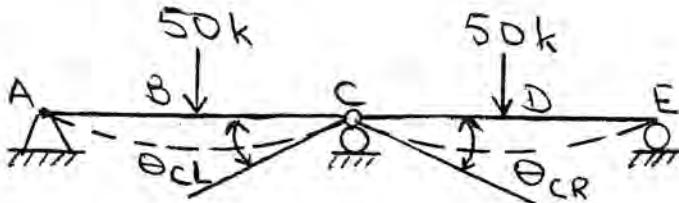
$$D_x = -\frac{\Delta_{D0}}{f_{DD}} = \frac{215.73}{63.35} = 3.41 \text{ k} \leftarrow$$



Reactions



13.30



Using beam deflection formulas:

$$\theta_{CL} = \theta_{CR} = \frac{1800 \text{ k-ft}^2}{EI}$$

$$\theta_{corel.} = \theta_{CL} + \theta_{CR} = \frac{3600 \text{ k-ft}^2}{EI}$$

$$f_{CCL} = f_{CCR} = \frac{8 \text{ ft}}{EI}$$

$$f_{corel.} = f_{CCL} + f_{CCR} = \frac{16 \text{ ft}}{EI}$$

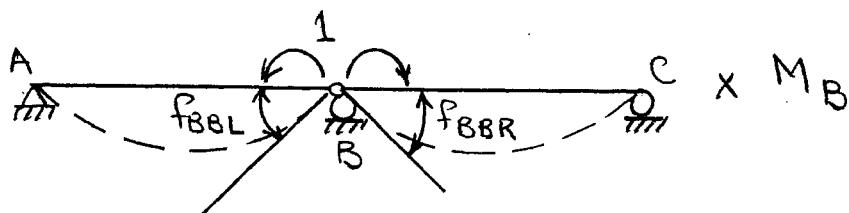
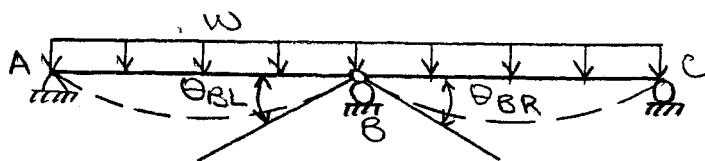
Compatibility Equation:

$$\theta_{corel.} + f_{corel.} M_C = 0$$

$$M_C = -\frac{3600}{16} = -225 \text{ k-ft}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.9.

13.31



Using beam deflection formulas:

$$\theta_{BL} = \theta_{BR} = \frac{wL^3}{24EI}$$

$$\theta_{B\text{rel.}} = \theta_{BL} + \theta_{BR} = \frac{wL^3}{12EI}$$

$$f_{BBL} = f_{BBR} = \frac{L}{3EI}$$

$$f_{B\text{rel.}} = f_{BBL} + f_{BBR} = \frac{2L}{3EI}$$

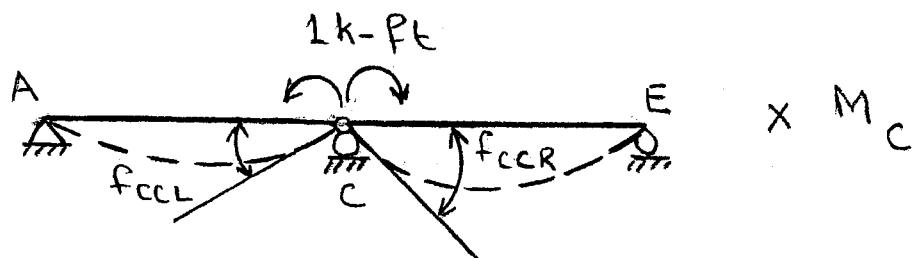
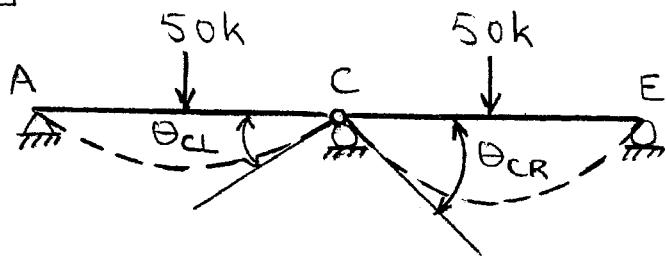
Compatibility Equation:

$$\theta_{B\text{rel.}} + f_{B\text{rel.}} M_B = 0$$

$$M_B = -\frac{wL^3}{12EI} \left(\frac{3EI}{2L} \right) = -\frac{wL^2}{8}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.10.

13.32



Using beam deflection formulas:

$$\theta_{CL} = \frac{400}{EI} \quad ; \quad \theta_{CR} = \frac{800}{EI}$$

$$\theta_{\text{Correl.}} = \theta_{CL} + \theta_{CR} = \frac{1200 \text{ k-ft}^2}{EI}$$

$$f_{CCL} = \frac{2.67}{EI} \quad ; \quad f_{CCR} = \frac{5.33}{EI}$$

$$f_{\text{Correl.}} = f_{CCL} + f_{CCR} = \frac{8 \text{ ft}}{EI}$$

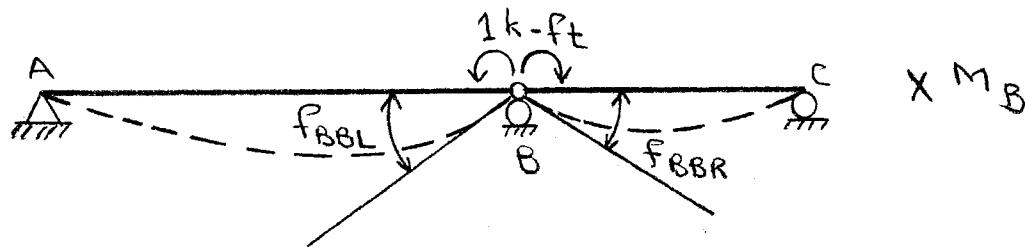
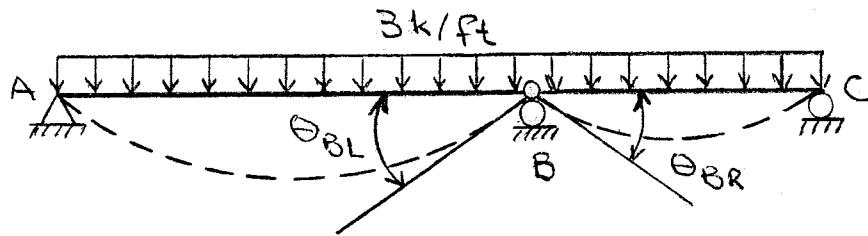
Compatibility Equation:

$$\theta_{\text{Correl.}} + f_{\text{Correl.}} M_C = 0$$

$$M_C = -\frac{1200}{8} = -150 \text{ k-ft}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.11.

13.33



Using beam deflection formulas:

$$\theta_{BL} = \frac{976.56 k-ft^2}{EI}; \quad \theta_{BR} = \frac{421.88 k-ft^2}{EI}$$

$$\theta_{B\text{rel.}} = \theta_{BL} + \theta_{BR} = \frac{1398.44 k-ft^2}{EI}$$

$$f_{BBL} = \frac{4.17 ft}{EI}; \quad f_{BBR} = \frac{5 ft}{EI}$$

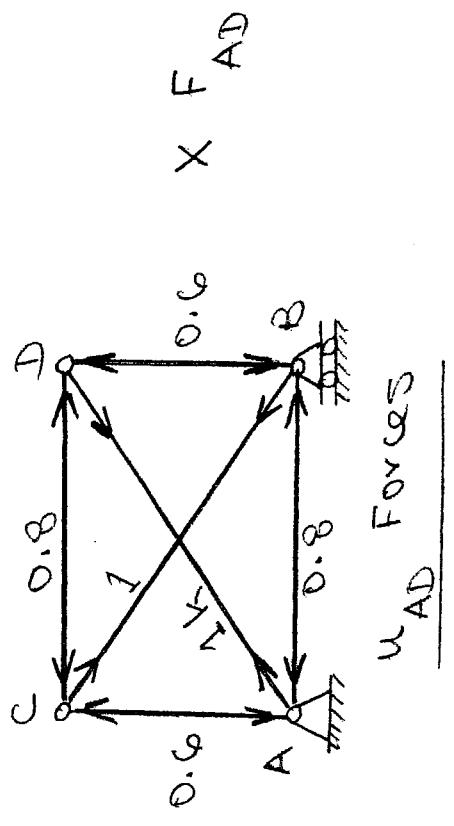
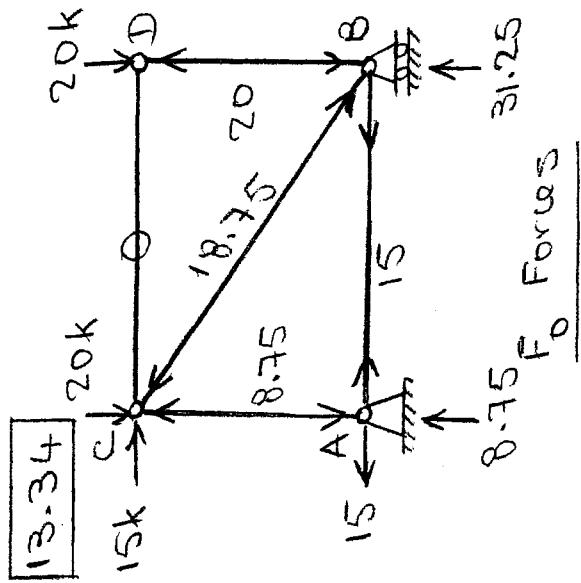
$$f_{B\text{rel.}} = f_{BBL} + f_{BBR} = \frac{9.17 ft}{EI}$$

Compatibility Equation:

$$\theta_{B\text{rel.}} + f_{B\text{rel.}} M_B = 0$$

$$M_B = - \frac{1398.44}{9.17} = -152.6 k-ft$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.12.

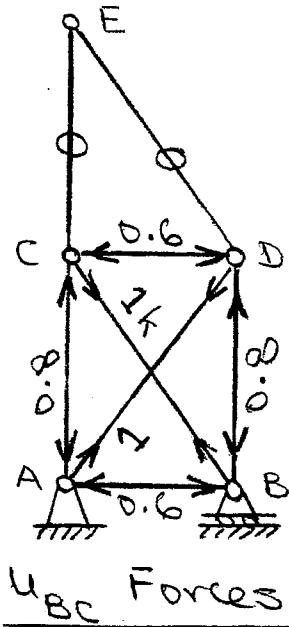
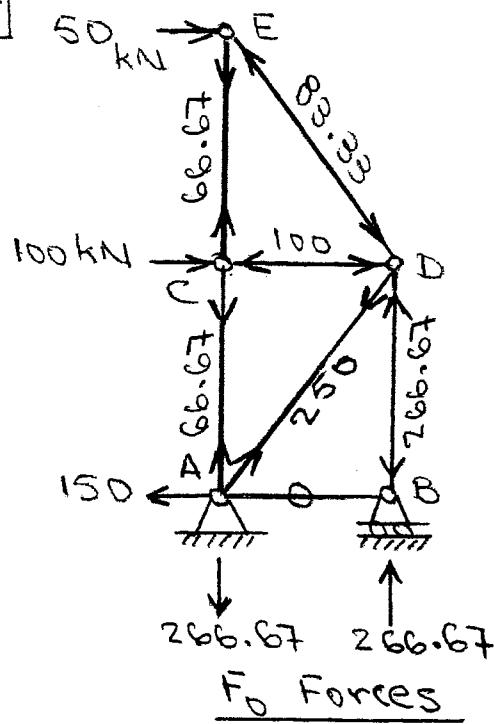


Member	L (in.)	A (in. ²)	F_0 (k)	u_{AD} (k/k)	u_{AD}^2/A (k/in.)	$u_{AD}^2 L/A$ (1/in.)	$F = F_0 + u_{AD} F_{AD}$ (k)
AB	192	6	-15	-0.8	-384	20.48	11.4
CD	192	6	0	0	0	20.48	-3.6
AC	144	6	-8.75	-0.6	126	8.64	-11.6
BC	144	6	-20	-0.6	288	8.64	-22.4
BD	240	20	-18.75	-1	-562.5	30	-14.2
AD	240	20	0	0	0	30	4.5
Σ					-532.5	118.24	

$$\Delta_{ADD} = \frac{1}{E} \sum \frac{F_{AD} u_{ADL}}{A} = -\frac{532.5 \text{ (k-in.)}}{E} ; F_{AD,AD} = \frac{1}{E} \sum \frac{u_{AD}^2 L}{A} = \frac{118.24 \text{ (1/in.)}}{E}$$

$$F_{AD} = -\frac{\Delta_{ADD}}{F_{AD,AD}} = \frac{532.5}{118.24} = \frac{4.5 \text{ k (T)}}{}$$

13.35



Member	L (m)	F_0 (kN)	u_{BC} (kN/kN)	$F_0 u_{BC} L$ (kN.m)	$u_{BC}^2 L$ (m)	$F = F_0 + u_{BC} F_{BC}$ (kN)
AB	3	0	-0.6	0	1.08	71.9
CD	3	-100	-0.6	180	1.08	-28.1
AC	4	66.67	-0.8	-213.33	2.56	162.5
BD	4	-266.67	-0.8	853.33	2.56	-170.8
CE	4	66.67	0	0	0	66.7
AD	5	250	1	1250	5	130.2
DE	5	-83.33	0	0	5	-83.3
BC	5	0	1	0	5	-119.8
		Σ	2070	17.28		

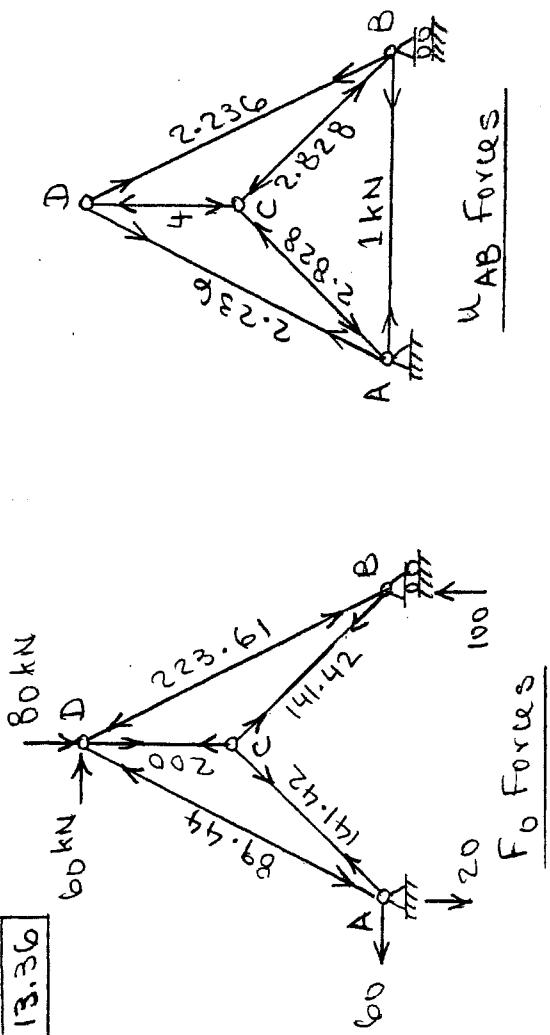
$$\Delta_{BCD} = \frac{1}{EA} \sum F_0 u_{BC} L = \frac{2070 \text{ (kN.m)}}{EA}$$

$$f_{BC,BC} = \frac{1}{EA} \sum u_{BC}^2 L = \frac{17.28 \text{ (m)}}{EA}$$

$$F_{BC} = -\frac{\Delta_{BCD}}{f_{BC,BC}} = -\frac{2070}{17.28} = -119.8 \text{ kN}$$

$$= \underline{119.8 \text{ kN (C)}}$$

13.36



u_{AB} Forces

F_0 Forces

Member	L (m)	F_0 (kN)	u_{AB} (kN/m)	$F_0 u_{AB} L$ (kN-m)	$u_{AB}^2 L$ (m)	$F = F_0 + u_{AB} F_{AB}$
AD	11.18	-89.44	2.236	-2235.86	55.9	34.7
BD	11.18	-223.61	2.236	-5589.91	55.9	-99.5
AC	7.07	141.42	-2.828	-2827.55	56.54	-15.6
BC	7.07	141.42	-2.828	-2827.55	56.54	-15.6
CD	5	200	-4	-4000	80	-22
AB	10	0	1	0	10	55.5
				$\Sigma -17480.87$	314.88	

$$\Delta_{ABD} = \frac{1}{EA} \sum F_0 u_{AB} L = -\frac{17480.87}{EA} \quad ; \quad f_{AB,AB} = \frac{1}{EA} \sum u_{AB}^2 L = \frac{314.88}{EA}$$

$$F_{AB} = -\frac{\Delta_{ABD}}{f_{AB,AB}} = \frac{17480.87}{314.88} = \underline{55.5 \text{ kN} (\tau)}$$

13.37 Using beam deflection formulas:

$$\Delta_{B0} = -\frac{17wL^4}{384EI} = -\frac{17(25)(16)^4}{384EI} = -\frac{72533.33}{EI} \text{ kN.m}^3$$

$$\Delta_{C0} = -\frac{wL^4}{8EI} = -\frac{25(16)^4}{8EI} = -\frac{204800}{EI} \text{ kN.m}^3$$

$$f_{BB} = \frac{L^3}{24EI} = \frac{(16)^3}{24EI} = \frac{170.67}{EI} \text{ kN.m}^3/\text{kN}$$

$$f_{BC} = f_{CB} = \frac{5L^3}{48EI} = \frac{5(16)^3}{48EI} = \frac{426.67}{EI} \text{ kN.m}^3/\text{kN}$$

$$f_{CC} = \frac{L^3}{3EI} = \frac{(16)^3}{3EI} = \frac{1365.33}{EI} \text{ kN.m}^3/\text{kN}$$

Compatibility Equations:

$$\Delta_{B0} + f_{BB} B_y + f_{BC} C_y = 0$$

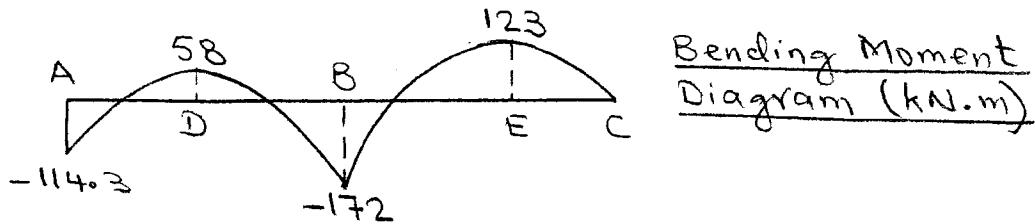
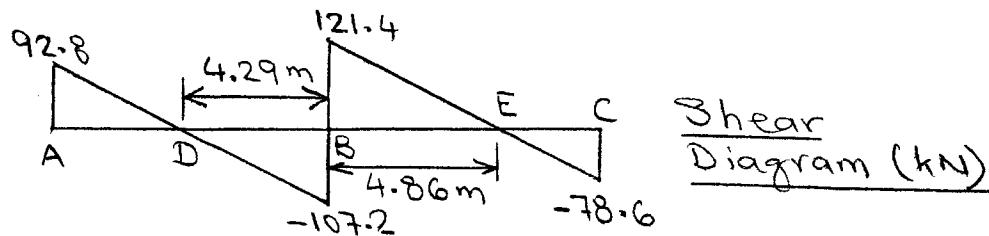
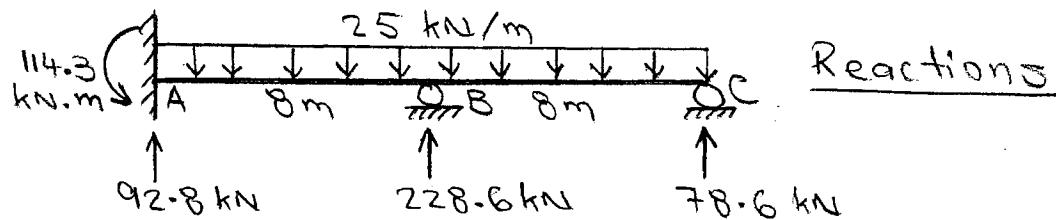
$$\Delta_{C0} + f_{CB} B_y + f_{CC} C_y = 0$$

$$-72533.33 + (170.67)B_y + (426.67)C_y = 0$$

$$-204800 + (426.67)B_y + (1365.33)C_y = 0$$

Solving these equations, we obtain

$$B_y = 228.6 \text{ kN} \uparrow \quad C_y = 78.6 \text{ kN} \uparrow$$



13.38 Using beam deflection formulas:

$$\Delta_{B0} = -\frac{819166.67 k \cdot ft^3}{EI}; \quad \Delta_{D0} = -\frac{280000 k \cdot ft^3}{EI}$$

$$f_{BB} = \frac{21333.33 k \cdot ft^3/k}{EI}; \quad f_{BD} = f_{DB} = \frac{6666.67 k \cdot ft^2/k}{EI}$$

$$f_{DD} = \frac{2666.67 k \cdot ft^3/k}{EI}$$

Compatibility Equations:

$$\Delta_{B0} + f_{BB} B_y + f_{BD} D_y = 0$$

$$\Delta_{D0} + f_{DB} B_y + f_{DD} D_y = 0$$

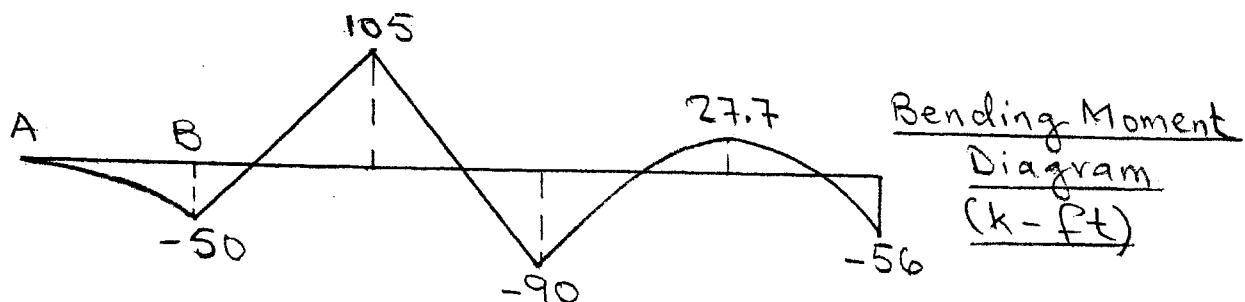
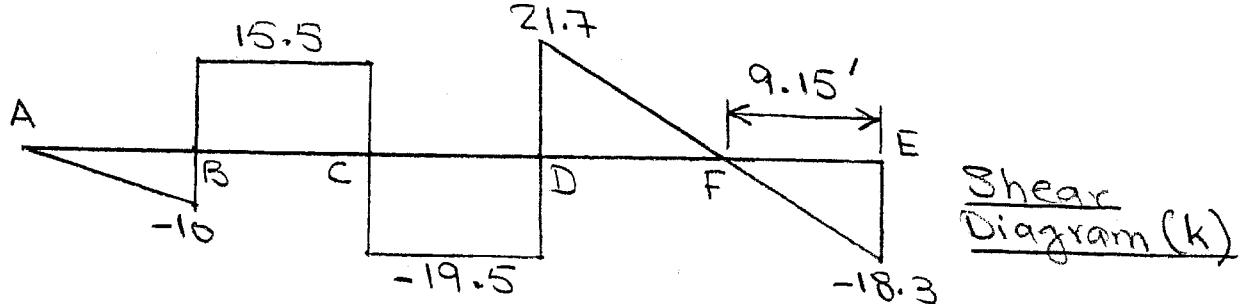
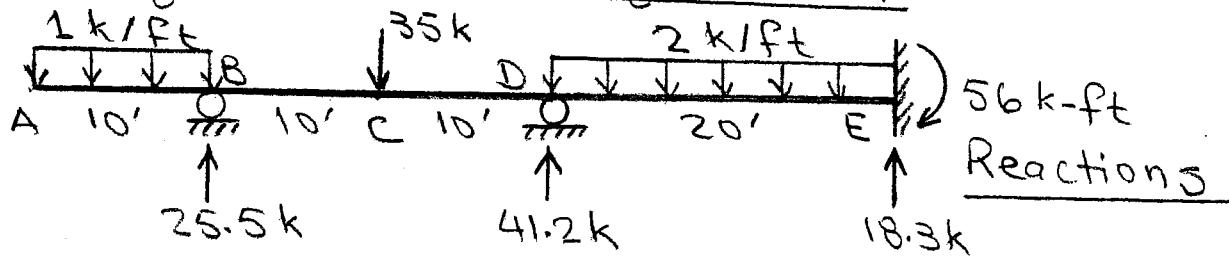
$$-819166.67 + (21333.33) B_y + (6666.67) D_y = 0$$

$$-280000 + (6666.67) B_y + (2666.67) D_y = 0$$

Solving these equations, we obtain

$$B_y = 25.5 k \uparrow$$

$$D_y = 41.2 k \uparrow$$



13.39 The bending moments at C and E are selected as the redundants. Using beam deflection formulas, we obtain:

$$\Theta_{C\text{rel.}} = \Theta_{CL} + \Theta_{CR} = \frac{1}{EI} (768 + 336) = \frac{1104 \text{ kN}\cdot\text{m}^2}{EI}$$

$$\Theta_{E\text{rel.}} = \Theta_{EL} + \Theta_{ER} = \frac{1}{EI} (384 + 600) = \frac{984 \text{ kN}\cdot\text{m}^2}{EI}$$

$$f_{CC\text{rel.}} = f_{CL} + f_{CR} = \frac{1}{EI} \left(\frac{10}{3} + \frac{10}{6}\right) = \frac{5 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI}$$

$$f_{CE} = f_{EC} = \frac{10}{12EI} = \frac{0.833 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI}$$

$$f_{EE\text{rel.}} = f_{EL} + f_{ER} = \frac{1}{EI} \left(\frac{10}{6} + \frac{8}{3}\right) = \frac{4.33 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI}$$

Compatibility Equations: $\Theta_{\text{corel.}} + f_{CC\text{rel.}} M_C + f_{CE} M_E = 0$
 $\Theta_{E\text{rel.}} + f_{EC} M_C + f_{EE\text{rel.}} M_E = 0$

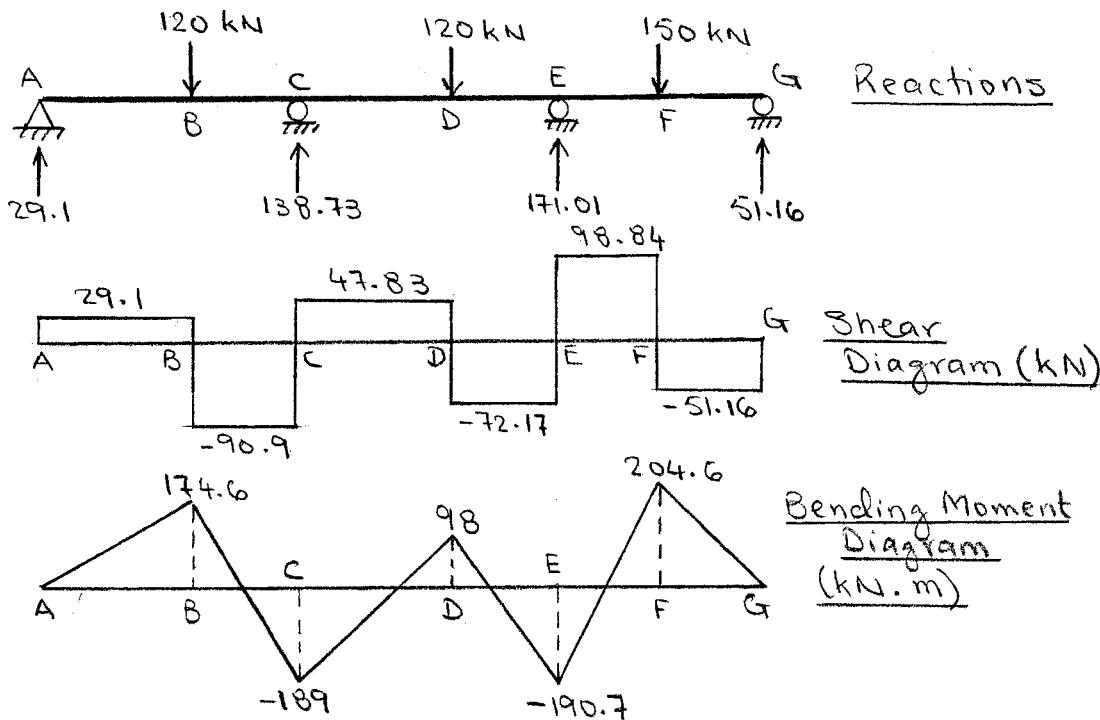
$$1104 + 5M_C + 0.833M_E = 0$$

$$984 + 0.833M_C + 4.33M_E = 0$$

Solving these equations, we obtain

$$\underline{M_C = -189 \text{ kN}\cdot\text{m}}$$

$$\underline{M_E = -190.7 \text{ kN}\cdot\text{m}}$$



13.40 The bending moments at B and C are selected as the redundants. Using beam deflection formulas, we obtain:

$$\Theta_{B\text{rel.}} = \Theta_{BL} + \Theta_{BR} = \frac{1}{EI} (1440 + 720) = \frac{2160 \text{ k-ft}^2}{EI}$$

$$\Theta_{C\text{rel.}} = \Theta_{CL} + \Theta_{CR} = \frac{1}{EI} (720 + 1440) = \frac{2160 \text{ k-ft}^2}{EI}$$

$$f_{BB\text{rel.}} = f_{BBL} + f_{BRR} = \frac{1}{EI} (8 + 4) = \frac{12 \text{ k-ft}^2/\text{k-ft}}{EI}$$

$$f_{CC\text{rel.}} = f_{CL} + f_{CR} = \frac{1}{EI} (4 + 8) = \frac{12 \text{ k-ft}^2/\text{k-ft}}{EI}$$

$$f_{BC} = f_{CB} = \frac{2 \text{ k-ft}^2/\text{k-ft}}{EI}$$

Compatibility Equations:

$$\Theta_{B\text{rel.}} + f_{BB\text{rel.}} M_B + f_{BC} M_C = 0$$

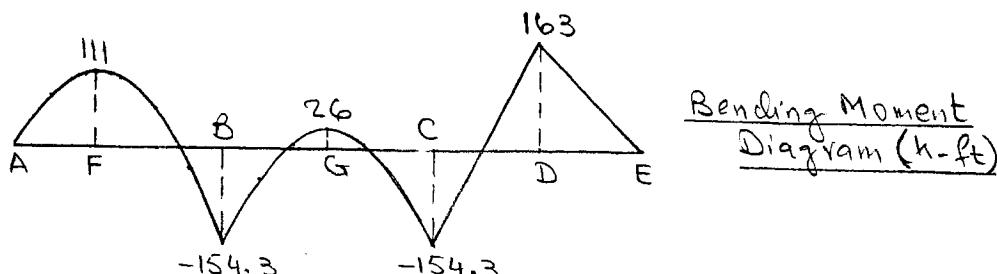
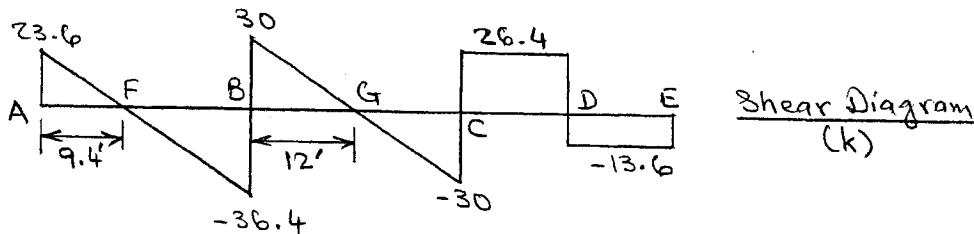
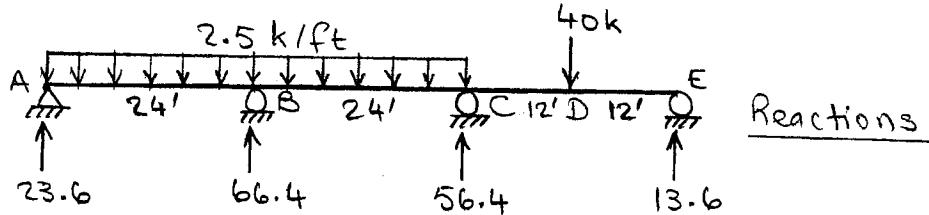
$$\Theta_{C\text{rel.}} + f_{CB} M_B + f_{CC\text{rel.}} M_C = 0$$

$$2160 + 12 M_B + 2 M_C = 0$$

$$2160 + 2 M_B + 12 M_C = 0$$

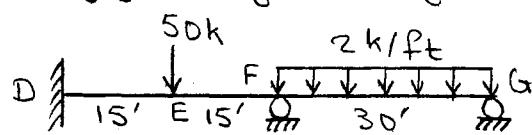
Solving these equations, we obtain

$$M_B = M_C = -154.3 \text{ k-ft}$$



Bending Moment Diagram (k-ft)

13.41 As the beam and the loading are symmetric, we will analyze only the right half DG of the beam.



Using beam deflection formulas:

$$\Delta_{FO} = -\frac{1085625 \text{ k-ft}^3}{EI}; \quad \Delta_{GO} = -\frac{3076875 \text{ k-ft}^3}{EI}$$

$$P_{FF} = \frac{9000 \text{ k-ft}^3/k}{EI}; \quad P_{FG} = F_{GF} = \frac{22500 \text{ k-ft}^3/k}{EI}$$

$$F_{GG} = \frac{72000 \text{ k-ft}^3/k}{EI}$$

Compatibility Equations:

$$\Delta_{FO} + P_{FF} F_y + P_{FG} G_y = 0$$

$$\Delta_{GO} + P_{GF} F_y + F_{GG} G_y = 0$$

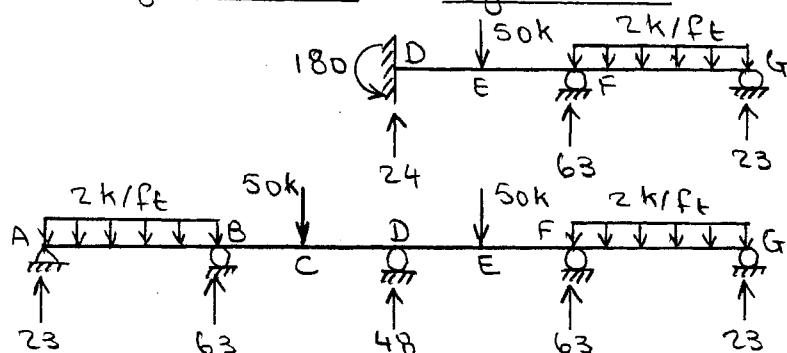
$$-1085625 + 9000 F_y + 22500 G_y = 0$$

$$-3076875 + 22500 F_y + 72000 G_y = 0$$

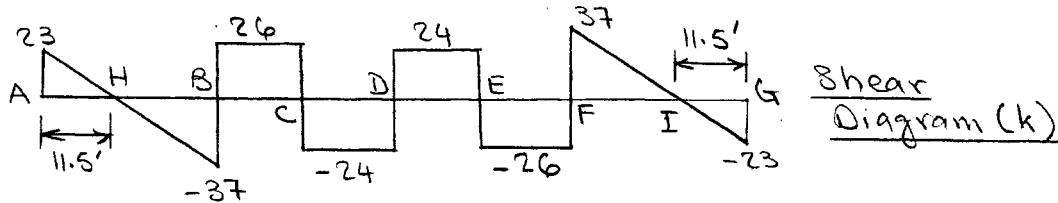
Solving these equations, we obtain

$$F_y = 63 \text{ k}\uparrow$$

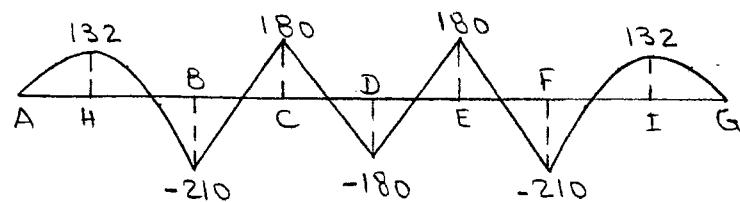
$$G_y = 23 \text{ k}\uparrow$$



Reactions

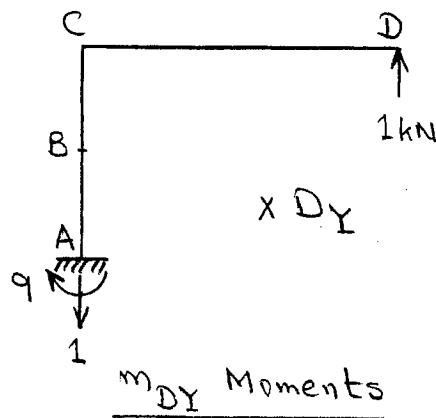
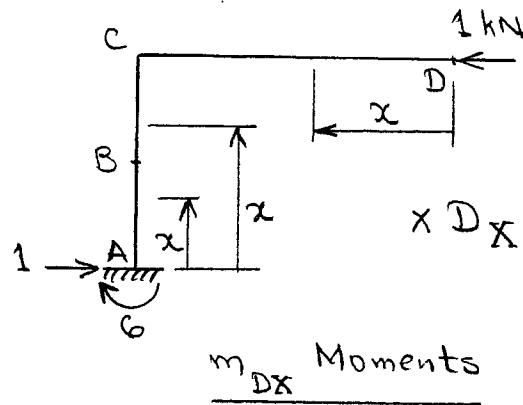
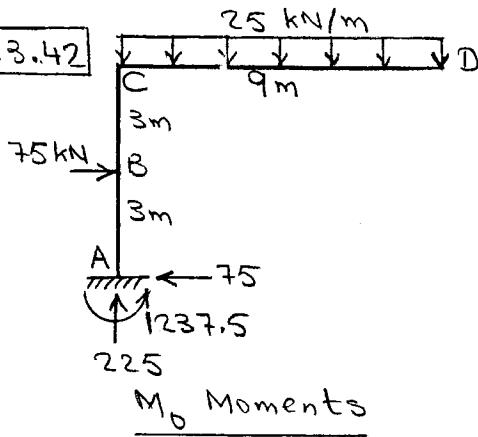


Shear Diagram (k)



Bending Moment Diagram (k-ft)

13.42



Segment	z Coordinate		M_0 (kN.m)	m_{DX} (kN.m/kN)	m_{DY} (kN.m/kN)
	Origin	Limits (m)			
AB	A	0-3	-1237.5 + 75x	6-x	9
BC	A	3-6	-1012.5	6-x	9
DC	D	0-9	-12.5x ²	0	x

$$\Delta_{DX0} = \sum \int \frac{M_0 m_{DX}}{EI} dx = - \frac{19912.5 \text{ kN-m}^3}{EI}$$

$$\Delta_{DY0} = \sum \int \frac{M_0 m_{DY}}{EI} dx = - \frac{78215.625 \text{ kN-m}^3}{EI}$$

$$f_{DX,DY} = \sum \int \frac{m_{DX}^2}{EI} dx = \frac{72 \text{ kN-m}^3/\text{kN}}{EI}$$

$$f_{DY,DY} = \sum \int \frac{m_{DY}^2}{EI} dx = \frac{729 \text{ kN-m}^3/\text{kN}}{EI}$$

$$f_{DX,DY} = f_{DY,DX} = \sum \int \frac{m_{DX} m_{DY}}{EI} dx = \frac{162 \text{ kN-m}^3/\text{kN}}{EI}$$

13.42 (contd.) Compatibility Equations:

$$\Delta_{DX0} + f_{DX,DX} D_X + f_{DX,DY} D_Y = 0$$

$$\Delta_{DY0} + f_{DY,DX} D_X + f_{DY,DY} D_Y = 0$$

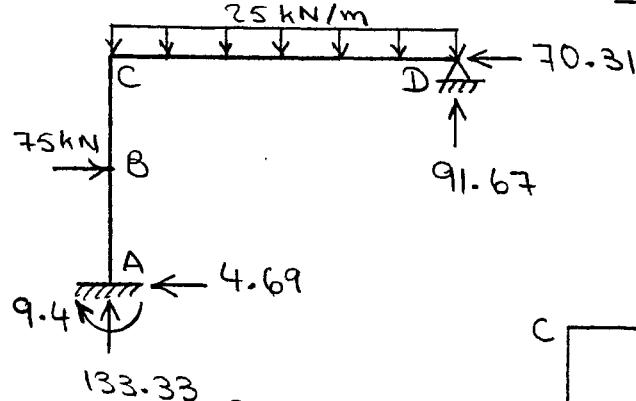
$$-19912.5 + 72 D_X + 162 D_Y = 0$$

$$-78215.625 + 162 D_X + 729 D_Y = 0$$

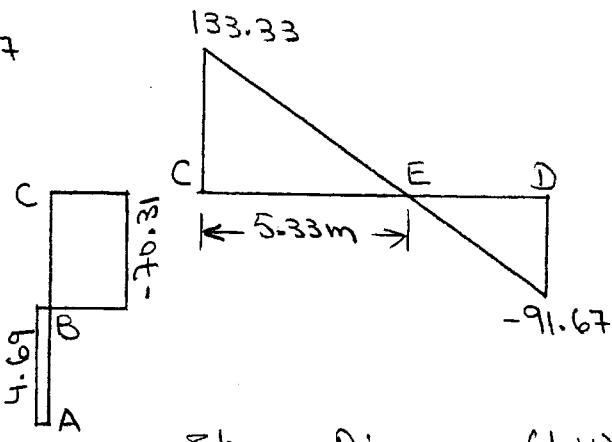
Solving these equations, we obtain

$$D_X = 70.3 \text{ kN} \leftarrow$$

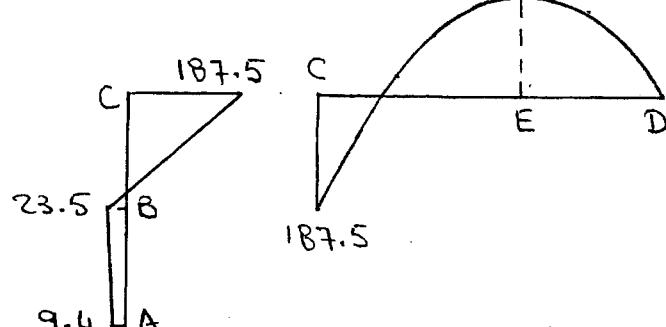
$$D_Y = 91.7 \text{ kN} \uparrow$$



Reactions

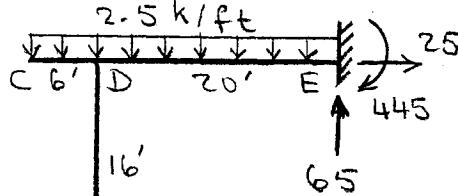


Shear Diagrams (kN)



Bending Moment Diagrams (kN.m)

13.43 The reactions A_x and A_y are selected as the redundants.

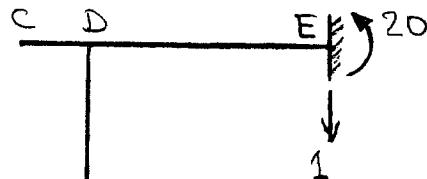


B $\leftarrow 25 \text{ k}$

8'

A

M_0 Moments



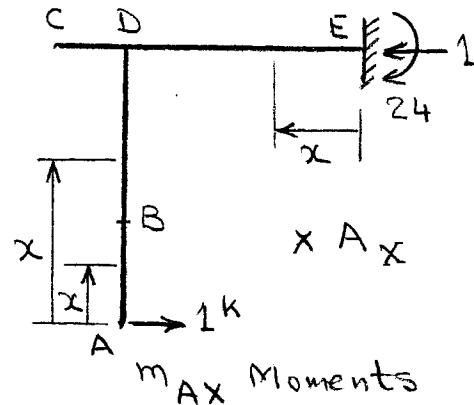
B

A

$x A_Y$

↑ 1 k

m_{AY} Moments



m_{AX} Moments

Segment	x Coordinate		M_0 (k-ft)	m_{AX} (k-ft/k)	m_{AY} (k-ft/k)
	Origin	Limits (ft)			
AB	A	0-8	0	-1x	0
BD	A	8-24	25(x-8)	-1x	0
ED	E	0-20	-445 + 65x - 1.25x ²	-24	20-x

$$\Delta_{AX0} = \sum \int \frac{M_0 m_{AX}}{EI} dx = - \frac{78133.33 \text{ k-ft}^3}{EI}$$

$$\Delta_{AY0} = \sum \int \frac{M_0 m_{AY}}{EI} dx = - \frac{19000 \text{ k-ft}^3}{EI}$$

$$f_{AX,AX} = \sum \int \frac{m_{AX}^2}{EI} dx = \frac{16128 \text{ k-ft}^3/k}{EI}$$

$$f_{AY,AY} = \sum \int \frac{m_{AY}^2}{EI} dx = \frac{2666.67 \text{ k-ft}^3/k}{EI}$$

$$f_{AX,AY} = f_{AY,AX} = \sum \int \frac{m_{AX} m_{AY}}{EI} dx = - \frac{4800 \text{ k-ft}^3/k}{EI}$$

13.43 (contd.) Compatibility Equations:

$$\Delta_{AX0} + f_{AX,AX} A_x + f_{AX,AY} A_y = 0$$

$$\Delta_{AY0} + f_{AY,AX} A_x + f_{AY,AY} A_y = 0$$

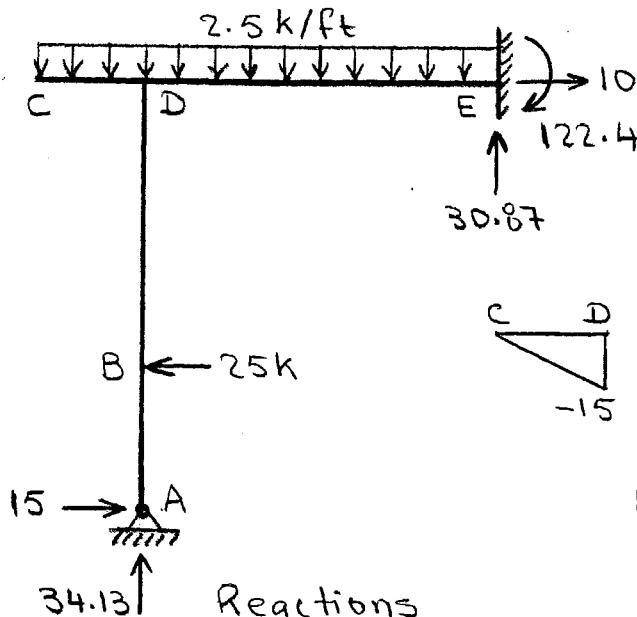
$$-78133.33 + 16128 A_x - 4800 A_y = 0$$

$$-19000 - 4800 A_x + 2666.67 A_y = 0$$

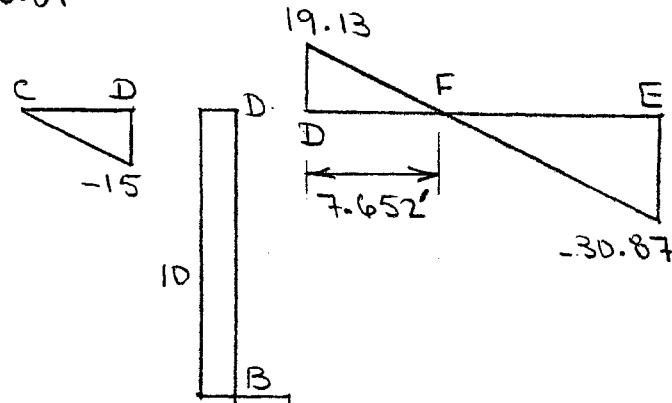
Solving these equations, we obtain:

$$A_x = 15 \text{ k} \rightarrow$$

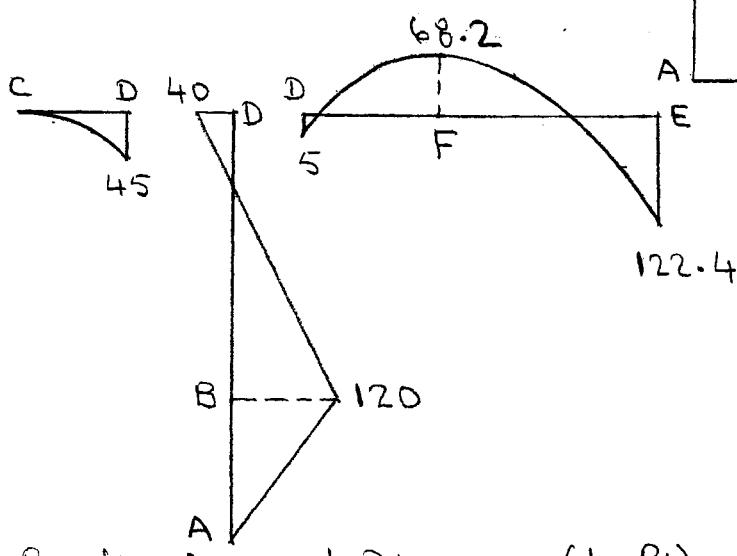
$$A_y = 34.13 \text{ k} \uparrow$$



Reactions

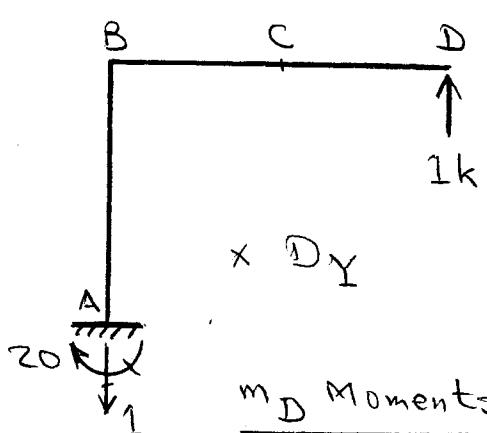
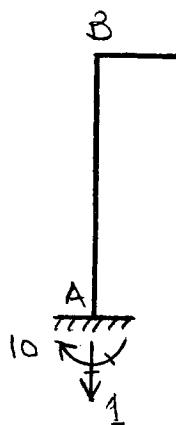
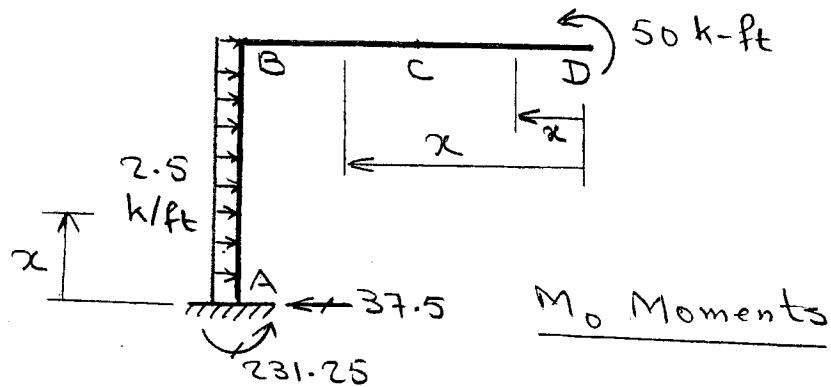


Shear Diagrams (k)



Bending Moment Diagrams (k-ft)

13.44



Member	x coordinate		M_o (k-ft)	m_c	m_d
	Origin	Limits (ft)			
AB	A	0 - 15	$-231.25 + 37.5x - 1.25x^2$	10	20
CB	D	10 - 20	50	$1(x-10)$	$1x$
DC	D	0 - 10	50	0	$1x$

$$\Delta_{co} = \sum \int \frac{M_o m_c}{EI} dx = - \frac{4062.5 \text{ k-ft}^3}{EI}$$

$$\Delta_{do} = \sum \int \frac{M_o m_d}{EI} dx = - \frac{3125 \text{ k-ft}^3}{EI}$$

$$f_{cc} = \sum \int \frac{m_c^2}{EI} dx = \frac{1833.33 \text{ k-ft}^3/k}{EI}$$

$$f_{dd} = \sum \int \frac{m_d^2}{EI} dx = \frac{8666.67 \text{ k-ft}^3/k}{EI}$$

$$f_{cd} = f_{dc} = \sum \int \frac{m_c m_d}{EI} dx = \frac{3833.33 \text{ k-ft}^3/k}{EI}$$

13.44 (contd.) Compatibility Equations:

$$\Delta_{CO} + f_{OC} C_Y + f_{CD} D_Y = 0$$

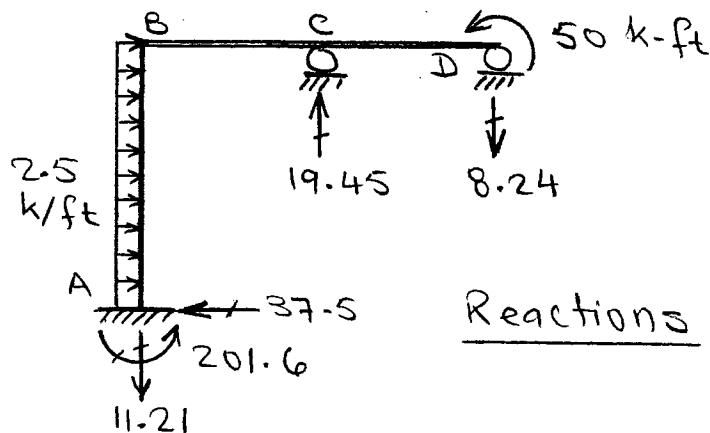
$$\Delta_{DO} + f_{DC} C_Y + f_{DD} D_Y = 0$$

$$-4062.5 + 1833.33 C_Y + 3833.33 D_Y = 0$$

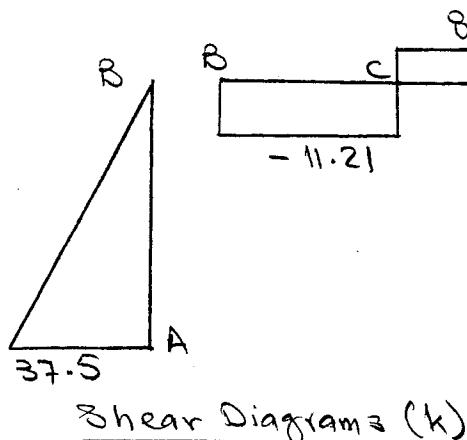
$$-3125 + 3833.33 C_Y + 8666.67 D_Y = 0$$

Solving these equations, we obtain:

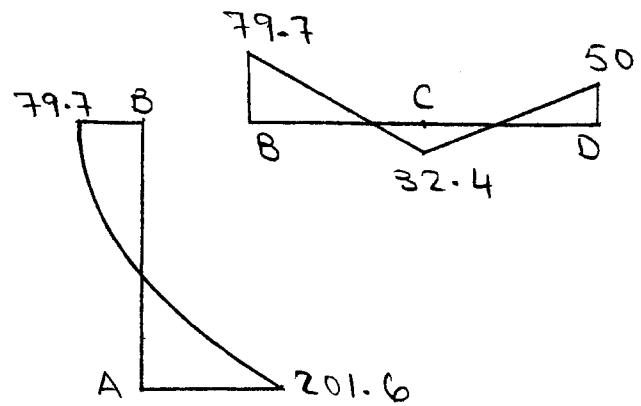
$$C_Y = 19.45 \text{ k} \uparrow \quad D_Y = -8.24 \text{ k} = 8.24 \text{ k} \downarrow$$



Reactions

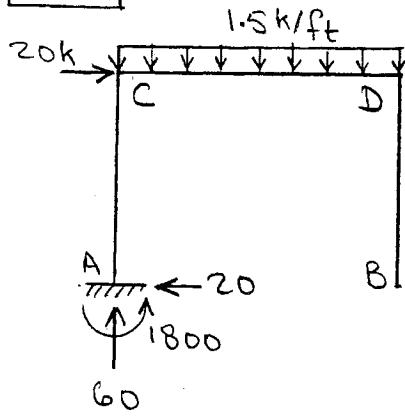
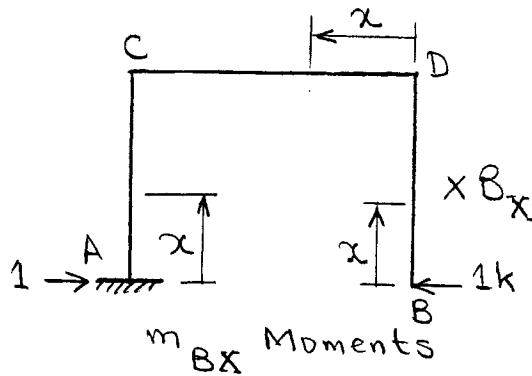
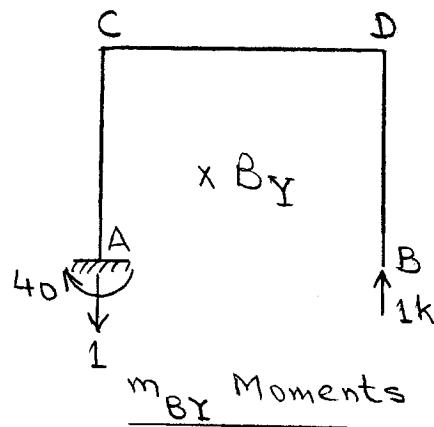
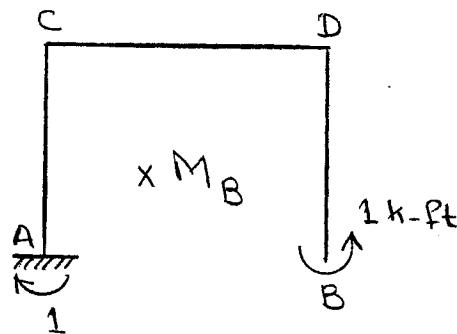


Shear Diagrams (k)



Bending Moment Diagrams (k-ft)

13.45

 M_0 Moments m_{BX} Moments m_{BY} Moments m_{MB} Moments

Member	x Coordinate		Moment of Inertia	M_0 (k-ft)	m_{BX}	m_{BY}	m_{MB}
	Origin	Limits (ft)					
AC	A	0-30	I	$-1800 + 20x$	-x	40	1
DC	D	0-40	2I	$-0.75x^2$	-30	x	1
BD	B	0-30	I	0	x	0	-1

$$\Delta_{BXO} = \sum \int \frac{M_0 m_{DX}}{EI} dx = \frac{870000 k \cdot ft^3}{EI}$$

$$\Delta_{BYO} = \sum \int \frac{M_0 m_{DY}}{EI} dx = -\frac{2040000 k \cdot ft^3}{EI}$$

$$\Theta_{BO} = \sum \int \frac{M_0 m_{MB}}{EI} dx = -\frac{53000 k \cdot ft^2}{EI}$$

$$f_{BX,BX} = \sum \int \frac{m_{BX}^2}{EI} dx = \frac{36000 k \cdot ft^3 / k}{EI}$$

$$B.45 \text{ (Cont'd.)} \quad f_{BY,BY} = \sum \int \frac{m_{BY}^2}{EI} dx = \frac{58666.67}{EI}$$

$$f_{MB,MB} = \sum \int \frac{m_{MB}^2}{EI} dx = \frac{80}{EI}$$

$$f_{BX,BY} = f_{BY,BX} = \sum \int \frac{m_{BX} m_{BY}}{EI} dx = -\frac{30000}{EI}$$

$$f_{BX,MB} = f_{MB,BX} = \sum \int \frac{m_{BX} m_{MB}}{EI} dx = -\frac{1500}{EI}$$

$$f_{BY,MB} = f_{MB,BY} = \sum \int \frac{m_{BY} m_{MB}}{EI} dx = \frac{1600}{EI}$$

Compatibility Equations:

$$\Delta_{BX0} + f_{BX,BX} B_X + f_{BX,BY} B_Y + f_{BX,MB} M_B = 0$$

$$\Delta_{BY0} + f_{BY,BX} B_X + f_{BY,BY} B_Y + f_{BY,MB} M_B = 0$$

$$\Theta_{B0} + f_{MB,BX} B_X + f_{MB,BY} B_Y + f_{MB,MB} M_B = 0$$

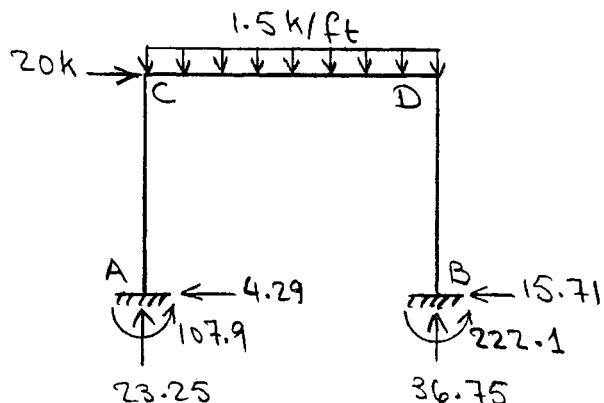
$$870000 + 36000 B_X - 30000 B_Y - 1500 M_B = 0$$

$$-2040000 - 30000 B_X + 58666.67 B_Y + 1600 M_B = 0$$

$$-53000 - 1500 B_X + 1600 B_Y + 80 M_B = 0$$

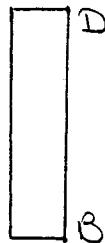
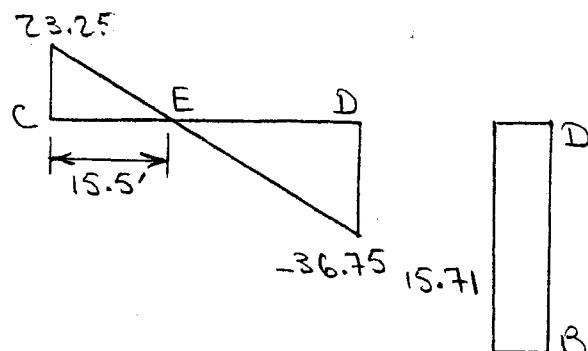
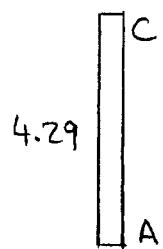
Solving these equations, we obtain:

$$\underline{B_X = 15.71 \text{ k} \leftarrow} \quad \underline{B_Y = 36.75 \text{ k} \uparrow} \quad \underline{M_B = 222.1 \text{ k-ft} \leftarrow}$$

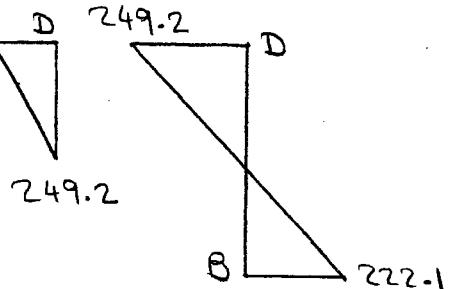
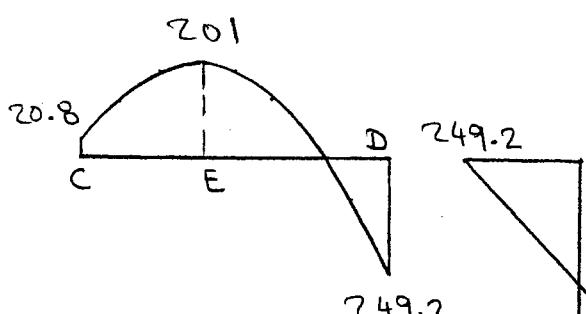
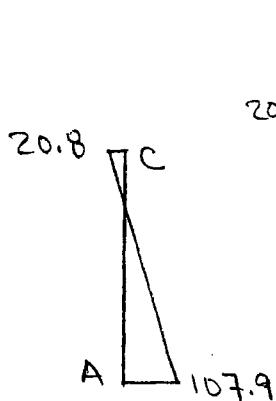


Reactions

13.45 (contd.)

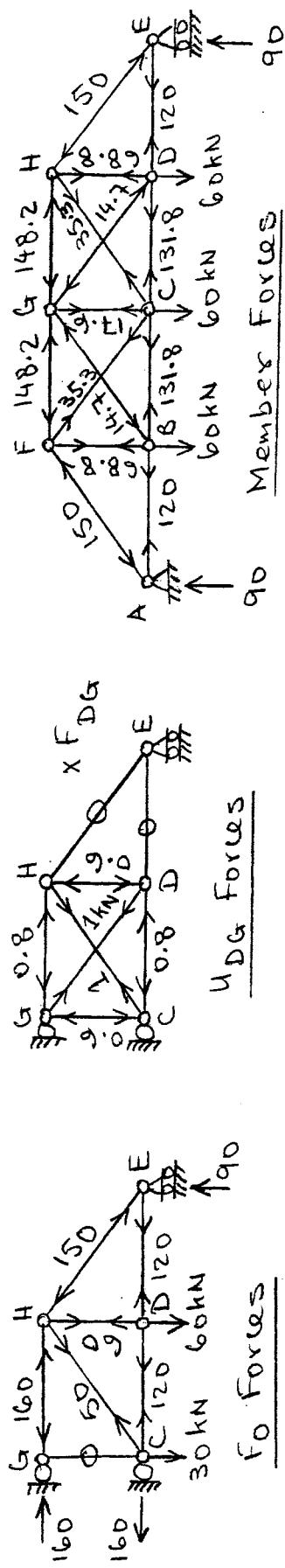


Shear Diagrams (k)



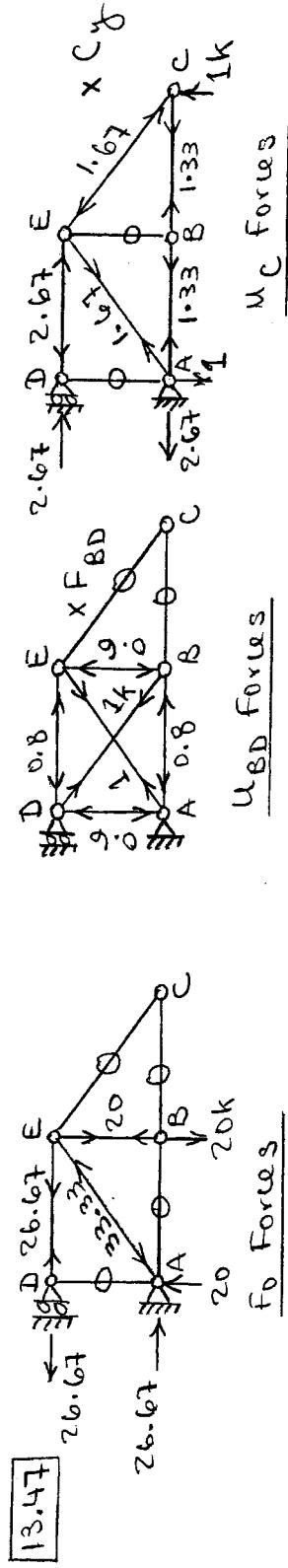
Bending Moment Diagrams (k-ft)

13.46 As the truss and the loading are symmetric, we will analyze only the right half CEGH of the truss.



Member	L (m)	A	F_0 (kN)	u_{DG} (kN/kN)	$\frac{F_0 u_{DG} L}{A}$	$\frac{u_{DG}^2 L}{A}$	$F = F_0 + u_{DG} f_{DG}$ (kN)
C D	8	1	120	-0.8	- 768	5.12	131.8
D E	8	1	120	0	0	0	120
G H	8	1/2	-160	-0.8	1024	5.12	-148.2
C G	6	1	0	0	0	4.32	8.8
D H	6	1	60	-0.6	-216	2.16	68.8
C H	10	1	50	1	500	10	35.3
E H	10	1	-150	0	0	0	-150
D G	10	1	0	1	0	10	-14.7
Σ			540			36.72	

$$F_{D6} = -\frac{540}{36 \cdot t_2} = -14.7 \text{ kN} = \underline{14.7 \text{ kN (c)}}$$



Member (in.)	A (in ²)	f ₀ (k)	U _{BD} (k/k)	U _C (k/k)	F ₀ U _{BD} A	F ₀ U _C A	U _{BD} A	U _C A	F = F ₀ + U _{BD} F _{BD} + U _C C _y
A B	192	8	0	-0.8	1.33	0	15.36	42.45	-25.54
B C	192	8	0	0	1.33	0	0	42.45	-1.44
D E	192	8	26.67	-0.8	-2.67	-512.04	-1709.01	171.09	8.29
A D	144	9	0	-0.6	0	0	15.36	0	0.31
B E	144	9	20	-0.6	-2.88	0	8.64	0	-7.3
A E	240	9	-33.33	1	1.67	-1333.2	-2276.44	40	12.7
C E	240	9	0	0	-1.67	0	0	0	-10.4
B D	240	9	0	1	0	0	0	0	12.16
\sum									
					-2133.26	-3935.45	12.8	451.22	92.52

$$\Delta_{BDO} = \frac{-2133.26}{E}; \quad \Delta_{CO} = -\frac{3935.45}{E}; \quad f_{BDO, BD} = \frac{12.8}{E}; \quad f_{CC} = \frac{451.22}{E}$$

$$f_{BDO, C} = f_{C, BD} = \frac{92.52}{E}$$

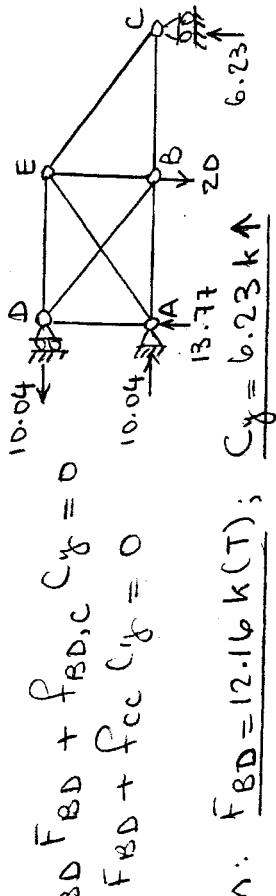
Compatibility Equations:

$$\Delta_{BDO} + f_{BD, BD} F_{BD} + f_{BD, C} C_y = 0$$

$$\Delta_{CO} + f_{CO, BD} F_{BD} + f_{CC, C} C_y = 0$$

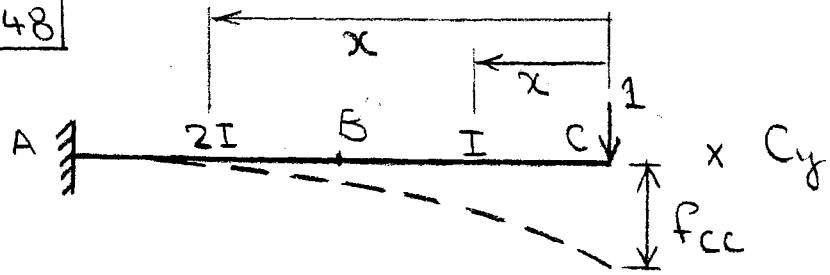
$$-2133.26 + 12.8 F_{BD} + 92.52 C_y = 0$$

$$-3935.45 + 92.52 F_{BD} + 451.22 C_y = 0$$



Solving these equations, we obtain: $F_{BD} = 12.16 \text{ k(N)}$; $C_y = 6.23 \text{ k(N)}$

13.48



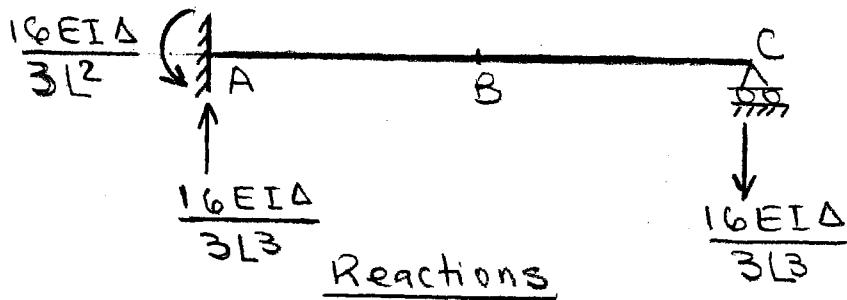
Using the virtual work method:

$$F_{cc} = \frac{1}{EI} \left[\int_0^{L/2} (-1x)^2 dx + \frac{1}{2} \int_{L/2}^L (-1x)^2 dx \right]$$

$$= \frac{3L^3}{16EI}$$

Compatibility Equation: $F_{cc} c_y = \Delta$

$$c_y = \frac{16EI\Delta}{3L^3} \downarrow$$



13.49 From the solution of Problem 13-1

$$\Delta_{DD} = -\frac{14760 \text{ kN.m}^3}{EI} = -\frac{14760}{200(3250)} = -0.0227 \text{ m}$$

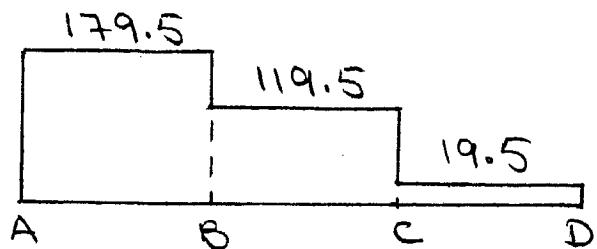
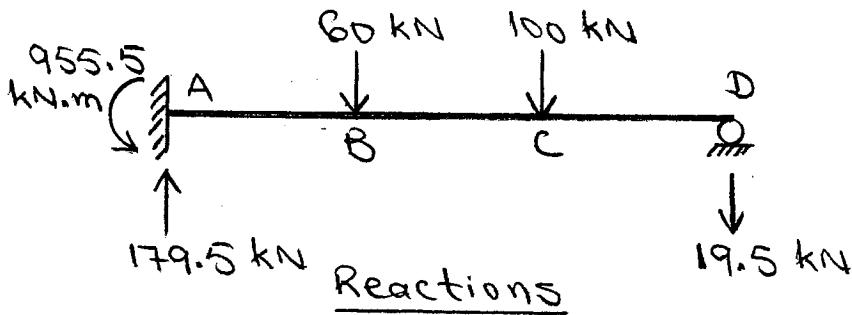
$$f_{DD} = \frac{243 \text{ kN.m}^3/\text{kN}}{EI} = \frac{243}{200(3250)} = 0.000374 \text{ m/kN}$$

Compatibility Equation:

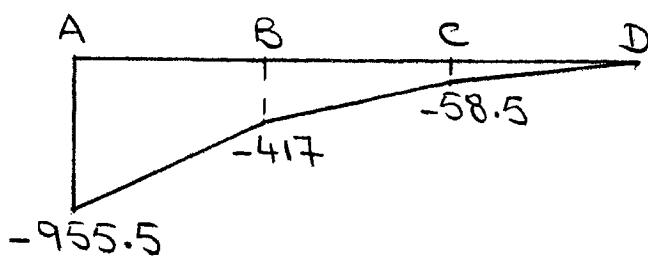
$$\Delta_{DD} + f_{DD} D_y = \Delta_D$$

$$-0.0227 + (0.000374) D_y = -0.03$$

$$D_y = -\frac{0.0073}{0.000374} = -19.5 \text{ kN} = \underline{\underline{19.5 \text{ kN} \downarrow}}$$



Shear Diagram (kN)



Bending Moment Diagram (kN.m)

13.50 From the solution of Problem 13.9:

$$\Delta_{CO} = -\frac{158400 \text{ k-ft}^3}{EI} = -\frac{158400 (12)^3}{29000 (1500)} = -6.292 \text{ in.}$$

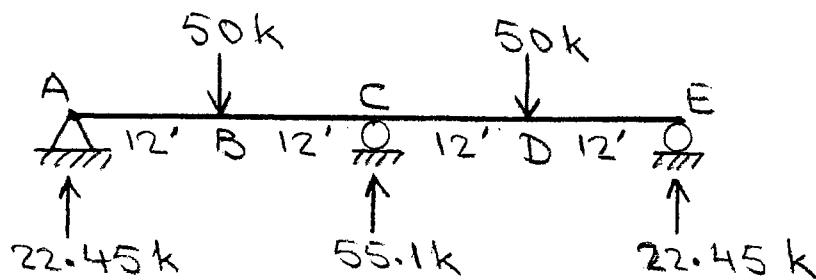
$$f_{CC} = \frac{2304 \text{ k-ft}^3/\text{k}}{EI} = \frac{2304 (12)^3}{29000 (1500)} = 0.0915 \text{ in./k}$$

Compatibility Equation:

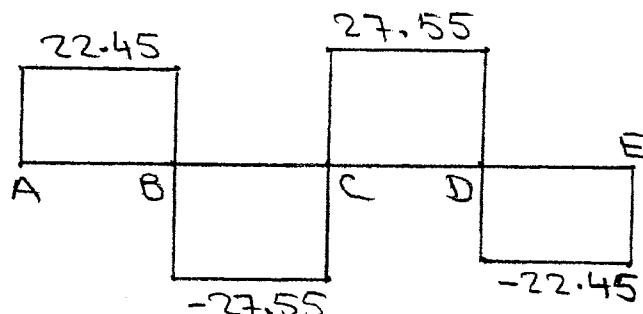
$$\Delta_{CO} + f_{CC} C_y = \Delta_C$$

$$-6.292 + (0.0915) C_y = -1.25$$

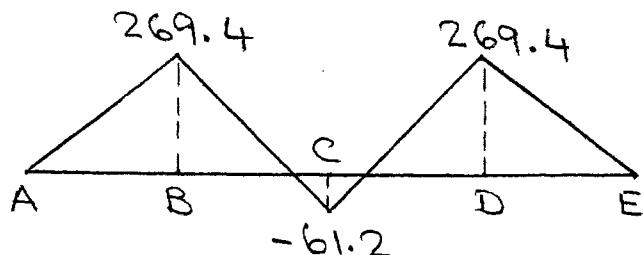
$$C_y = \frac{5.042}{0.0915} = \underline{\underline{55.1 \text{ k}\uparrow}}$$



Reactions



Shear Diagram (k)

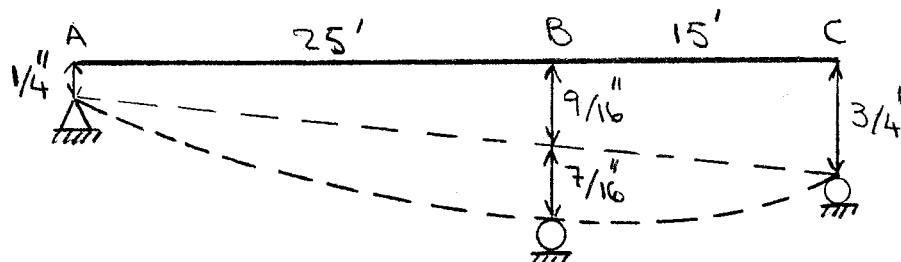


Bending Moment Diagram (k-ft)

13.51 From the solution of Problem 13-12:

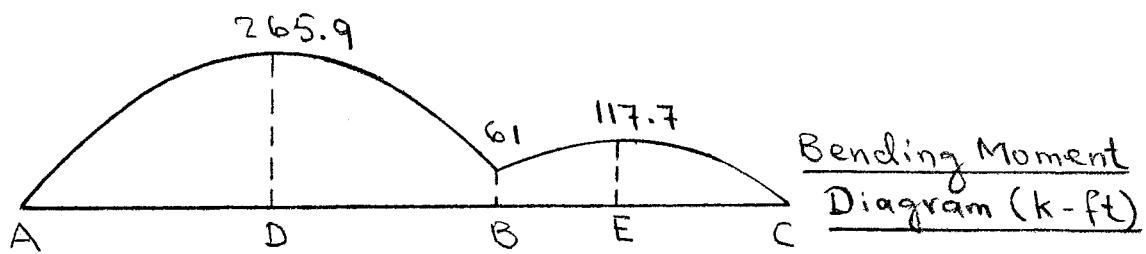
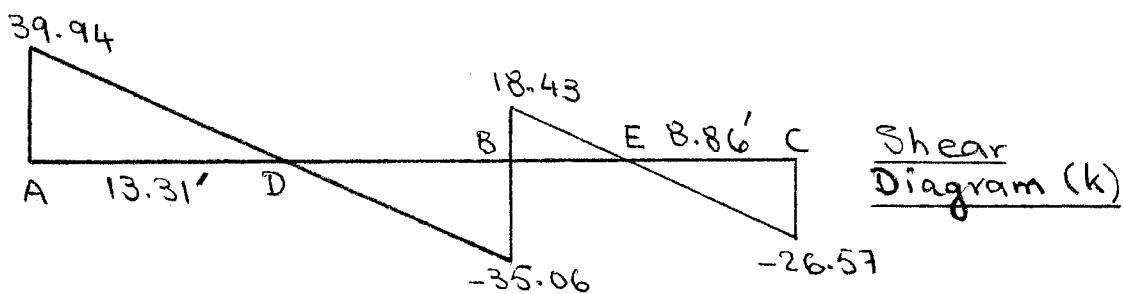
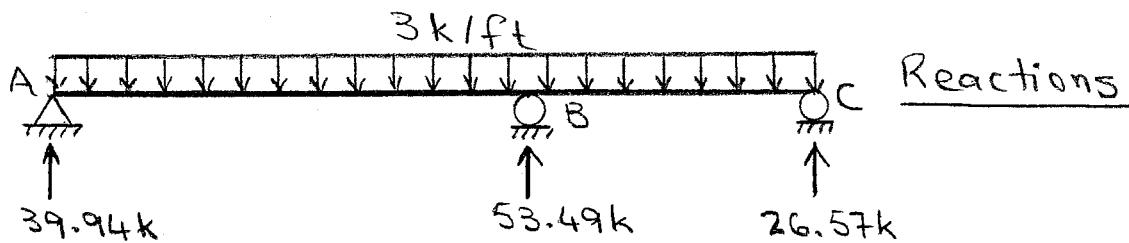
$$\Delta_{B0} = -\frac{61450.2 \text{ k-ft}^3}{EI} = -\frac{61450.2 (12)^3}{29000 (2500)} = -1.465 \text{ in.}$$

$$f_{BB} = \frac{805.66 \text{ k-ft}^3/\text{k}}{EI} = \frac{805.66 (12)^3}{29000 (2500)} = 0.0192 \text{ in./k}$$



Compatibility Equation:

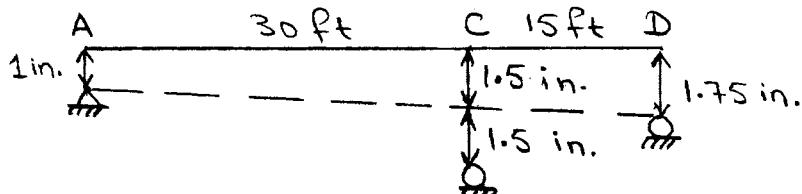
$$-1.465 + 0.0192 B_y = -\frac{7}{16} \quad B_y = 53.49 \text{ k} \uparrow$$



13.52 From the solution of Problem 13.27:

$$\Delta_{CO} = -\frac{22046.28 \text{ (k-in)}}{EA} = -\frac{22046.28}{29000(6)} = -0.127 \text{ in.}$$

$$f_{CC} = \frac{531.78 \text{ (in.)}}{EA} = \frac{531.78}{29000(6)} = 0.00306 \text{ in./k}$$

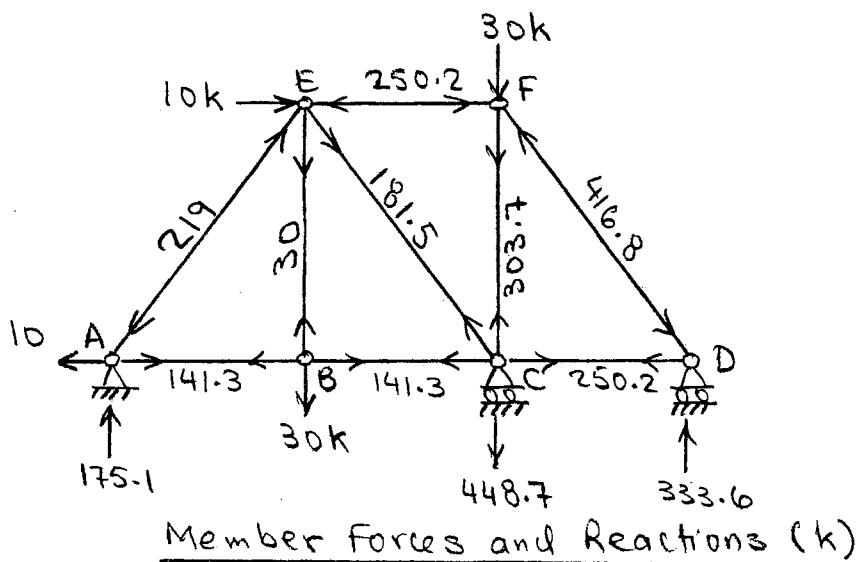


Compatibility Equation:

$$-0.127 + (0.00306) C_y = -1.5$$

$$C_y = -448.7 \text{ k} = \underline{448.7 \text{ k} \downarrow}$$

By using the F_o and M_C forces computed in the solution of Problem 13.17, we obtain the following member forces.



13.53 From the solution of Problem 13.37:

$$\Delta_{BD} = -\frac{72533.33 \text{ kN}\cdot\text{m}^3}{EI} = -\frac{72533.33}{70(1300)} = -0.797 \text{ m}$$

$$\Delta_{CD} = -\frac{204800 \text{ kN}\cdot\text{m}^3}{EI} = -\frac{204800}{70(1300)} = -2.251 \text{ m}$$

$$f_{BB} = \frac{170.67 \text{ kN}\cdot\text{m}^3/\text{kN}}{EI} = \frac{170.67}{70(1300)} = 0.00188 \text{ m/kN}$$

$$f_{BC} = f_{CB} = \frac{426.67 \text{ kN}\cdot\text{m}^3/\text{kN}}{EI} = \frac{426.67}{70(1300)} = 0.00469 \text{ m/kN}$$

$$f_{CC} = \frac{1365.33 \text{ kN}\cdot\text{m}^3/\text{kN}}{EI} = \frac{1365.33}{70(1300)} = 0.015 \text{ m/kN}$$

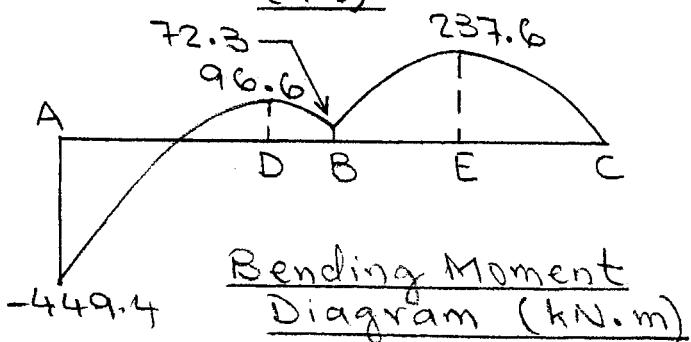
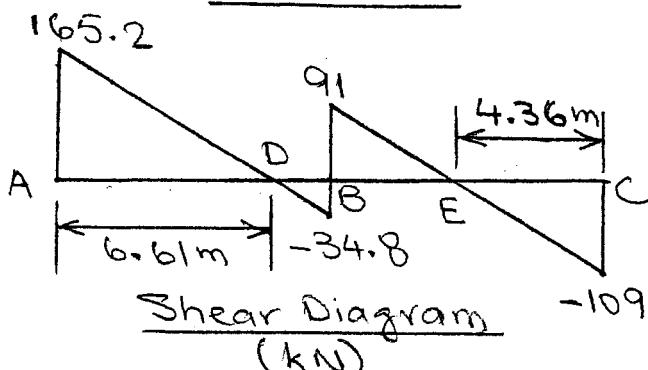
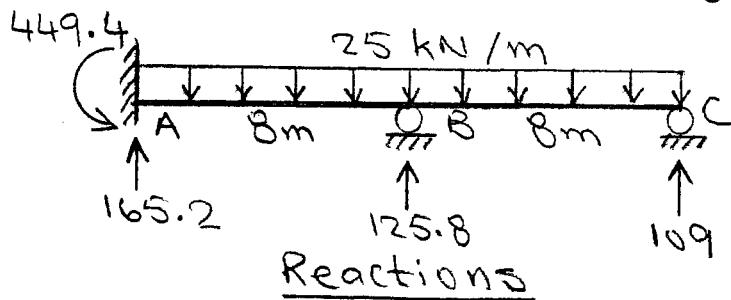
Compatibility Equations:

$$-0.797 + (0.00188)By + (0.00469)Cy = -0.05$$

$$-2.251 + (0.00469)By + (0.015)Cy = -0.025$$

Solving these equations, we obtain

$$By = 125.8 \text{ kN} \uparrow \quad Cy = 109 \text{ kN} \uparrow$$



13.54 From the solution of Problem 13.39:

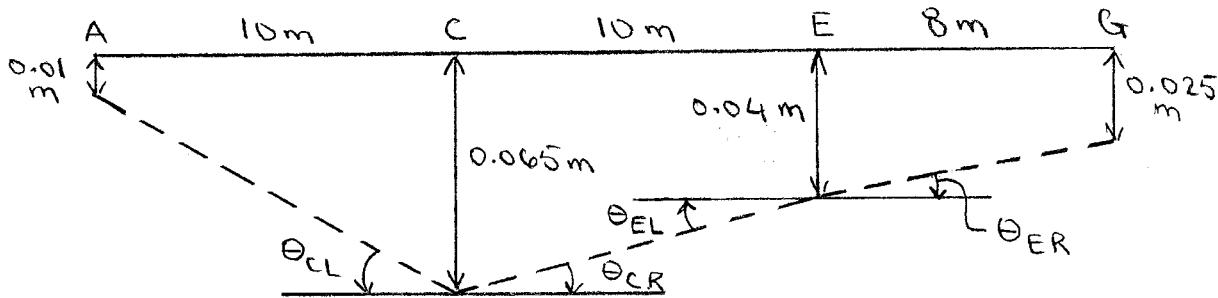
$$\theta_{C\text{rel.}} = \frac{1104 \text{ kN}\cdot\text{m}^2}{EI} = \frac{1104}{200(500)} = 0.01104 \text{ rad.}$$

$$\theta_{E\text{rel.}} = \frac{984 \text{ kN}\cdot\text{m}^2}{EI} = 0.00984 \text{ rad.}$$

$$f_{C\text{rel.}} = \frac{5 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI} = 0.00005 \text{ rad./kN}\cdot\text{m}$$

$$f_{E\text{rel.}} = \frac{4.33 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI} = 0.0000433 \text{ rad./kN}\cdot\text{m}$$

$$f_{CE} = f_{EC} = \frac{0.833 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI} = 0.00000833 \text{ rad./kN}\cdot\text{m}$$



$$\theta_{CL} = \frac{0.065 - 0.01}{10} = 0.0055 \text{ rad.}$$

$$\theta_{CR} = \frac{0.065 - 0.04}{10} = 0.0025 \text{ rad.}$$

$$\theta_C = \theta_{CL} + \theta_{CR} = 0.008 \text{ rad.}$$

$$\theta_{EL} = \frac{0.04 - 0.065}{10} = -0.0025 \text{ rad.}$$

$$\theta_{ER} = \frac{0.04 - 0.025}{8} = 0.001875 \text{ rad.}$$

$$\theta_E = \theta_{EL} + \theta_{ER} = -0.000625 \text{ rad.}$$

Compatibility Equations:

$$\theta_{C\text{rel.}} + f_{C\text{rel.}} M_C + f_{CE} M_E = \theta_C$$

$$\theta_{E\text{rel.}} + f_{EC} M_C + f_{EE\text{rel.}} M_E = \theta_E$$

$$1104 + 5M_C + 0.833M_E = 800$$

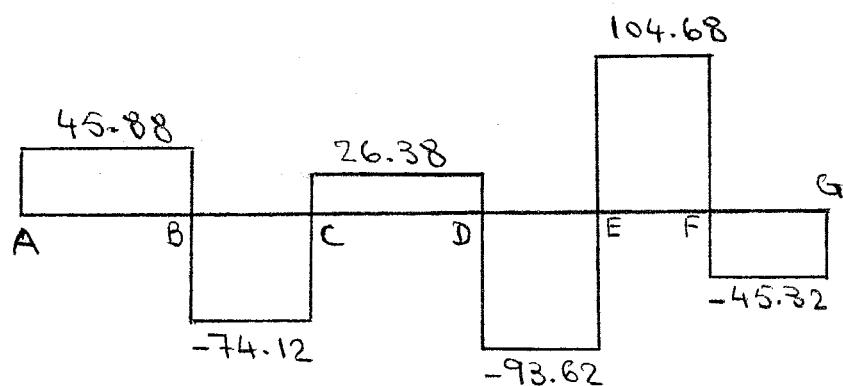
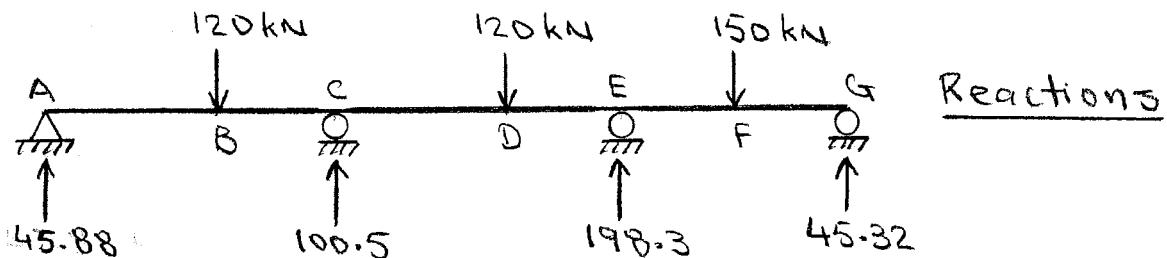
$$984 + 0.833M_C + 4.33M_E = -62.5$$

13-54 (contd.)

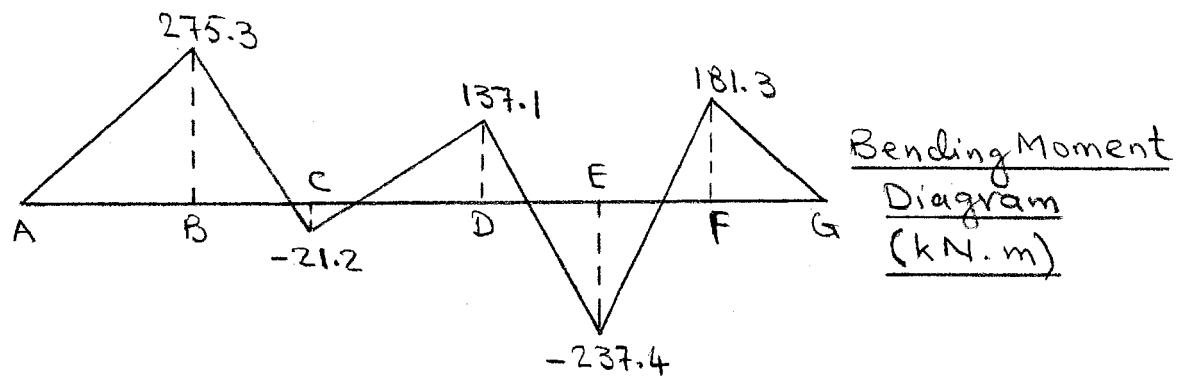
Solving these equations, we obtain

$$M_C = -21.29 \text{ kN.m}$$

$$M_E = -237.42 \text{ kN.m}$$

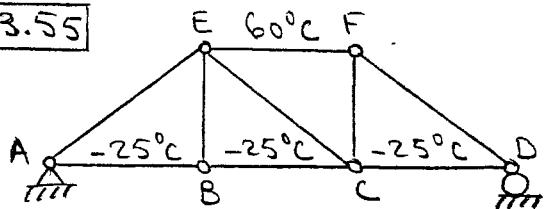


Shear
Diagram
(kN)

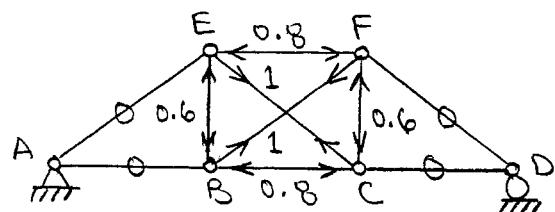


Bending Moment
Diagram
(kN.m)

13.55



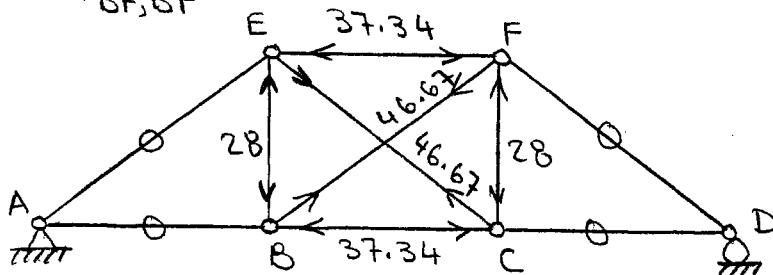
Temperature Changes

 Δ_{BF} Forces (kN)

$$\Delta_{BFD} = \alpha \sum (\Delta T) L u_{BF} = 1.2 \times 10^5 (8) [-25(-0.8) + 60(-0.8)] = -0.002688 \text{ m} = -2.688 \text{ mm}$$

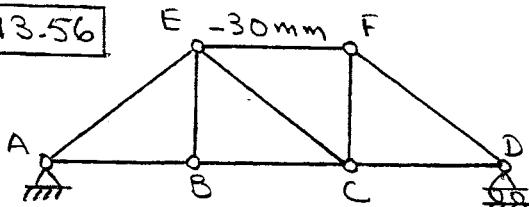
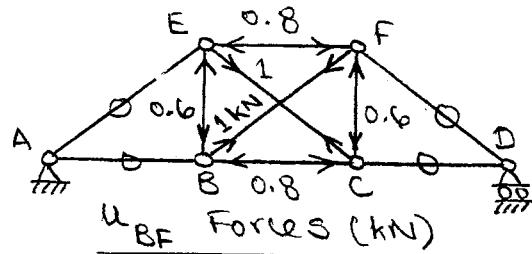
$$\begin{aligned} f_{BF, BF} &= \frac{1}{EA} \sum u_{BF}^2 L \\ &= \frac{1}{200(10^6)(0.003)} [2(1)^2(10) + 2(-0.8)^2(8) + 2(-0.6)^2(6)] = 57.6 (10^{-6}) \text{ m/kN} \\ &= 0.0576 \text{ mm/kN} \end{aligned}$$

$$F_{BF} = -\frac{\Delta_{BFD}}{f_{BF, BF}} = \frac{46.67 \text{ kN } (\tau)}{57.6 (10^{-6})} = 807.6 \text{ kN } (\tau)$$



Member Forces (kN)

13.56

Fabrication Errors

$$\Delta_{BFO} = \sum \delta u_{BF} = -30(-0.8) = 24 \text{ mm}$$

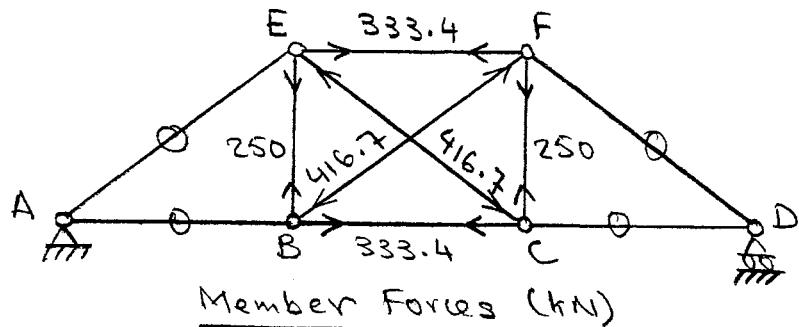
$$f_{BF,BF} = \frac{1}{EA} \sum u_{BF}^2 L$$

$$= \frac{1}{200(10^6)(0.003)} [2(1)^2 10 + 2(-0.8)^2 8 + 2(-0.6)^2 6]$$

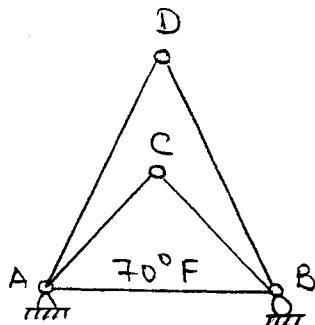
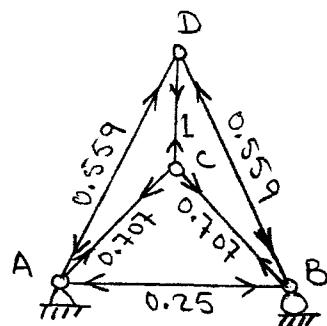
$$= 57.6 (10^{-6}) \text{ m/kN} = 0.0576 \text{ mm/kN}$$

$$F_{BF} = - \frac{\Delta_{BFO}}{f_{BF,BF}} = - \frac{24}{0.0576} = -416.7 \text{ kN}$$

$$= \underline{416.7 \text{ kN (C)}}$$

Member Forces (kN)

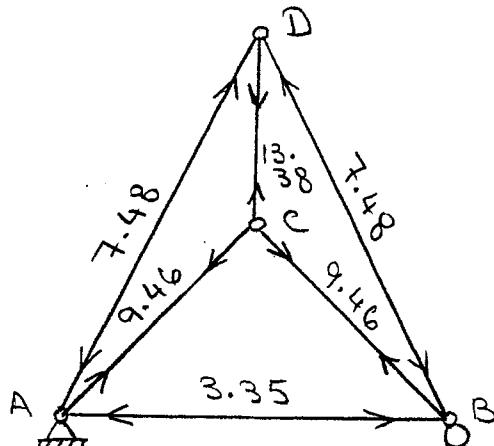
13.57

Temperature Change u_{CD} Forces (k)

$$\Delta_{CDO} = \sum \alpha(\Delta T) L u_{CD} = 6.5 \times 10^{-6} (70)(20)(-0.25) \\ = -0.002275 \text{ ft} = -0.0273 \text{ in.}$$

$$f_{CD,CD} = \frac{1}{EA} \sum u_{CD}^2 L \\ = \frac{1}{29000(8)} [(1)^2(10) + (-0.25)^2(20) + 2(-0.559)^2 \times \\ (22.36) + 2(0.707)^2(14.14)] = 0.00017 \text{ ft/k} \\ = 0.00204 \text{ in./k.}$$

$$F_{CD} = -\frac{\Delta_{CDO}}{f_{CD,CD}} = \underline{13.38 \text{ k}(\tau)}$$

Member Forces (k)

Chapter Fourteen

Three Moment Equation and the Method of Least Work

CHAPTER 14

14.1

Three-Moment Equation at Joint C:

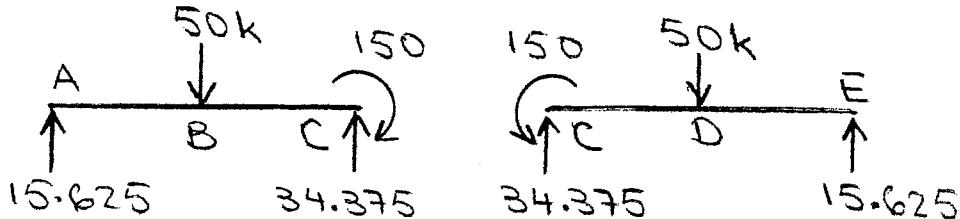
$$M_A = M_E = 0$$

$$2M_C \left(\frac{16}{2I} + \frac{16}{I} \right) = -\frac{50(16)^2(0.5)(1-0.25)}{2I}$$

$$-\frac{50(16)^2(0.5)(1-0.25)}{I}$$

$$48M_C = -7200$$

$$\underline{M_C = -150 \text{ k-ft}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.11.

14.2

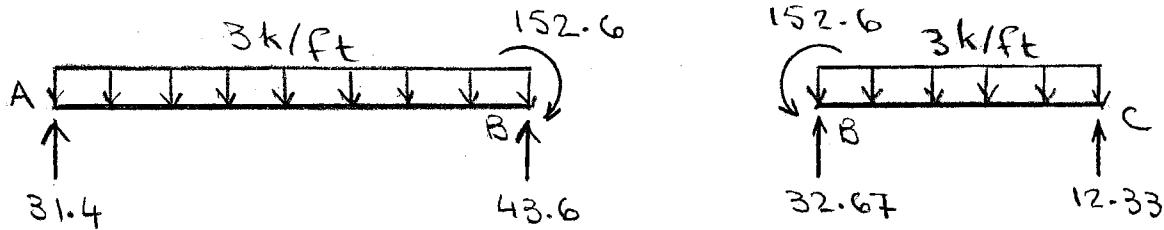
Three - Moment Equation at Joint B:

$$M_A = M_C = 0$$

$$2M_B \left(\frac{25}{2I} + \frac{15}{I} \right) = -\frac{3(25)^3}{4(2I)} - \frac{3(15)^3}{4I}$$

$$55M_B = -8390.625$$

$$\underline{M_B = -152.6 \text{ k-ft}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.12.

14.3

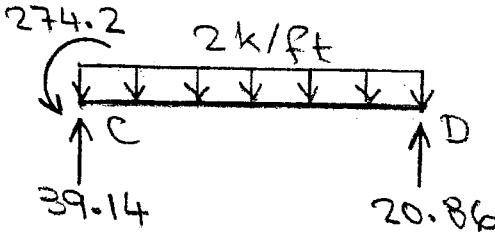
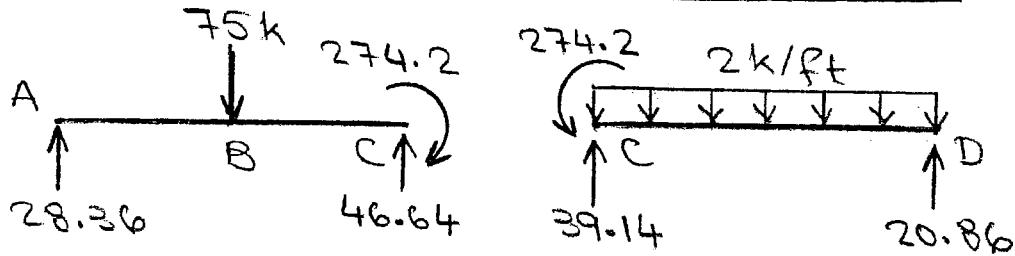
Three-Moment Equation at Joint C:

$$M_A = M_D = 0$$

$$2M_C \left(\frac{30}{3I} + \frac{30}{I} \right) = -\frac{75(30)^2(0.5)(1-0.25)}{3I} - \frac{2(30)^3}{4I}$$

$$80M_C = -21937.5$$

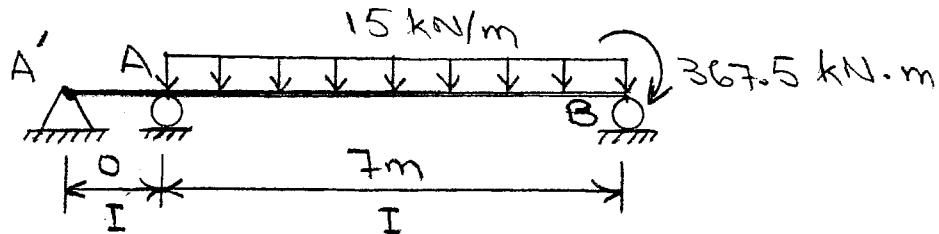
$$\underline{M_C = -274.2 \text{ k-ft}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.13.

14.4



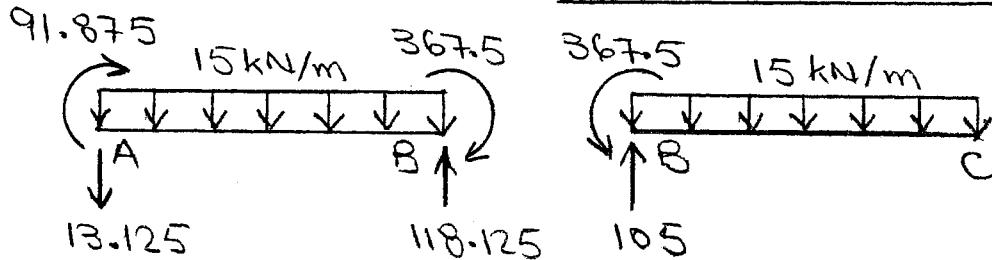
Three-Moment Equation at Joint A:

$$M_{A'} = 0, \quad M_B = -367.5 \text{ kN.m}$$

$$2M_A(0+7) - 367.5(7) = -\frac{1}{4}(15)(7)^3$$

$$14M_A = 1286.25$$

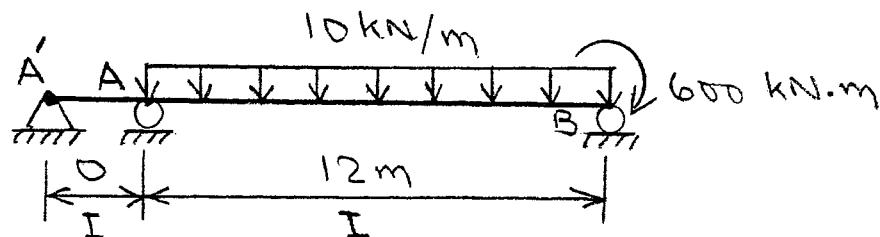
$$\underline{M_A = 91.875 \text{ kN.m}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.15.

14.5



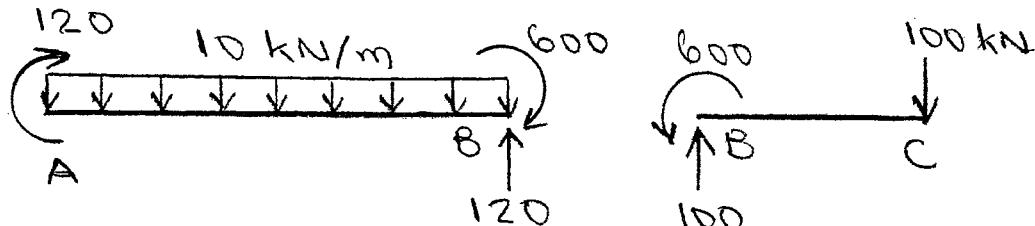
Three-Moment Equation at Joint A:

$$M_{A'} = 0, \quad M_B = -600 \text{ kN.m}$$

$$2M_A(0+12) - 600(12) = -\frac{1}{4}(10)(12)^3$$

$$24M_A = 2880$$

$$\underline{M_A = 120 \text{ kN.m}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.16.

14.6

Three-Moment Equation at Joint C:

$$M_A = 0$$

$$2M_C \left(\frac{10}{I} + \frac{10}{2I} \right) + M_E \left(\frac{10}{2I} \right) = - \frac{120(10)^2(0.6)(1-0.36)}{I}$$

$$- \frac{120(10)^2(0.4)(1-0.16)}{2I}$$

$$30M_C + 5M_E = -6624 \quad (1)$$

Three-Moment Equation at Joint E:

$$M_G = 0$$

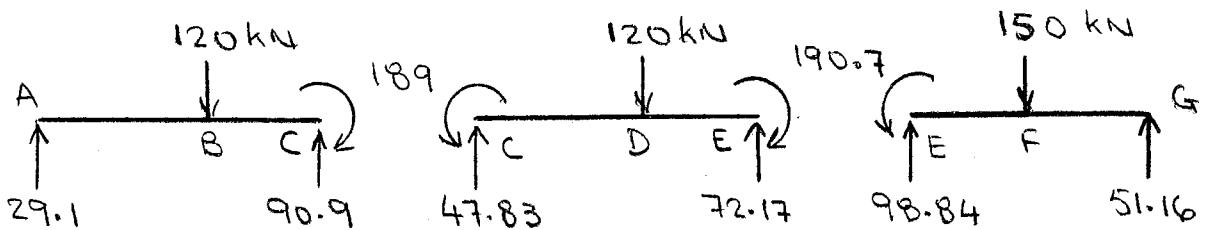
$$M_C \left(\frac{10}{2I} \right) + 2M_E \left(\frac{10}{2I} + \frac{8}{I} \right) = - \frac{120(10)^2(0.6)(1-0.36)}{2I}$$

$$- \frac{150(8)^2(0.5)(1-0.25)}{I}$$

$$5M_C + 26M_E = -5904 \quad (2)$$

Solving Eqs. (1) and (2) simultaneously, we obtain:

$$\underline{M_C = -189 \text{ kN.m}} \quad \underline{M_E = -190.7 \text{ kN.m}}$$

Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.39.

14.7

Three-Moment Equation at Joint B:

$$M_A = 0$$

$$2M_B \left(\frac{24}{I} + \frac{24}{2I} \right) + M_C \left(\frac{24}{2I} \right) = -\frac{2.5(24)^3}{4I} - \frac{2.5(24)^3}{4(2I)}$$

$$72M_B + 12M_C = -12960 \quad (1)$$

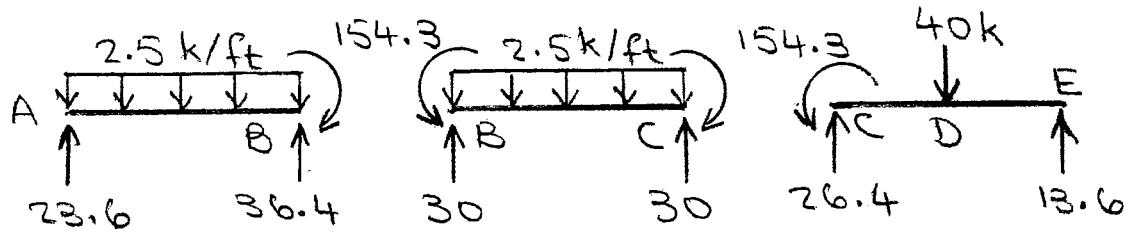
Three-Moment Equation at Joint C: $M_E = 0$

$$M_B \left(\frac{24}{2I} \right) + 2 \left(\frac{24}{2I} + \frac{24}{I} \right) = -\frac{2.5(24)^3}{4(2I)} - \frac{40(24)^2(0.5)(1-0.25)}{I}$$

$$12M_B + 72M_C = -12960 \quad (2)$$

Solving Eqs. (1) and (2) simultaneously, we obtain:

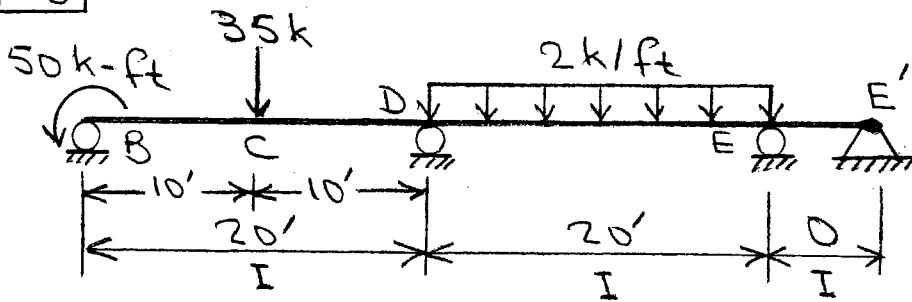
$$\underline{M_B = M_C = -154.3 \text{ k-ft}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.40.

14.8



Three-Moment Equation at Joint D:

$$M_B = -50 \text{ k-ft}$$

$$-50(20) + 2M_D(20+20) + M_E(20) = -35(20)^2(0.5)(0.75)$$

$$- \frac{1}{4}(2)(20)^3$$

$$80M_D + 20M_E = -8250 \quad (1)$$

Three-Moment Equation at Joint E:

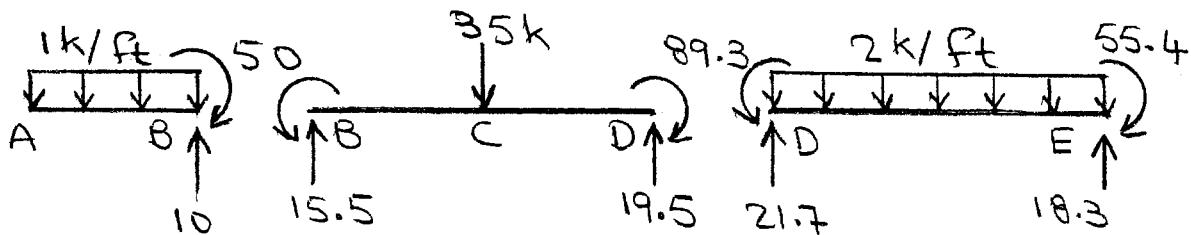
$$M_{E'} = 0$$

$$M_D(20) + 2M_E(20+0) = -\frac{1}{4}(2)(20)^3$$

$$20M_D + 40M_E = -4000 \quad (2)$$

Solving Eqs. (1) and (2) simultaneously, we obtain:

$$\underline{M_D = -89.3 \text{ k-ft}} \quad \underline{M_E = -55.4 \text{ k-ft}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of problem 13.38.

14.9

Three-Moment Equation at Joint B:

$$M_A = M_C = 0$$

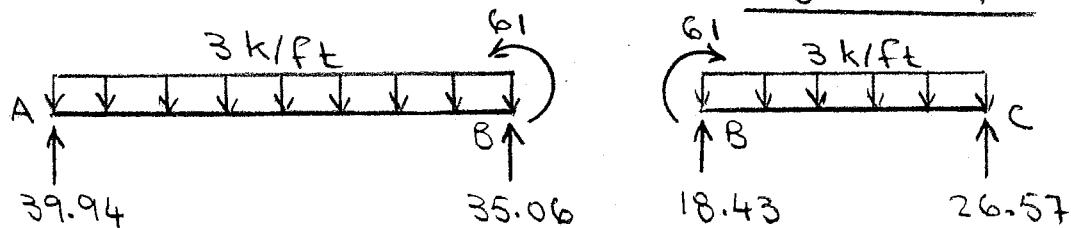
$$2M_B \left(\frac{25}{2I} + \frac{15}{I} \right) = -\frac{3(25)^3}{4(2I)} - \frac{3(15)^3}{4I} - 6E \left[\frac{0.25-1}{25(12)} + \frac{0.75-1}{15(12)} \right]$$

$$55M_B = -8390.625 + 0.0233EI$$

Substituting $EI = \frac{29000(2500)}{(12)^2} = 503472.2 \text{ k-ft}^2$, we obtain:

$$55M_B = 3357$$

$$\underline{M_B = 61 \text{ k-ft}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of problem 13.51.

14.10

Three-Moment Equation at Joint C: $M_A = 0$

$$2M_C \left(\frac{10}{I} + \frac{10}{2I}\right) + M_E \left(\frac{10}{2I}\right) = -\frac{120(10)^2(0.6)(1-0.36)}{I}$$

$$-\frac{120(10)^2(0.4)(1-0.16)}{2I} - 6E \left[\frac{0.01-0.065}{10} + \frac{0.04-0.065}{10} \right]$$

Substituting $EI = 200(500) = 100000 \text{ kN.m}^2$, we obtain:

$$30M_C + 5M_E = -1824 \quad (1)$$

Three-Moment Equation at Joint E: $M_G = 0$

$$M_C \left(\frac{10}{2I}\right) + 2M_E \left(\frac{10}{2I} + \frac{8}{I}\right) = -\frac{120(10)^2(0.6)(1-0.36)}{2I}$$

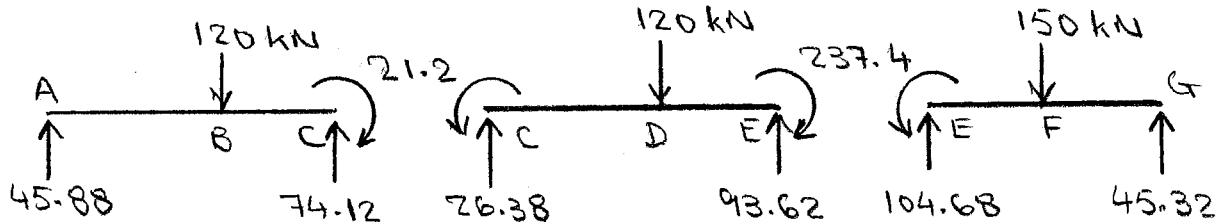
$$-\frac{150(8)^2(0.5)(1-0.25)}{I} - 6E \left[\frac{0.065-0.04}{10} + \frac{0.025-0.04}{8} \right]$$

$$5M_C + 26M_E = -6279 \quad (2)$$

Solving Eqs. (1) and (2) simultaneously, we obtain

$$\underline{M_C = -21.2 \text{ kN.m}}$$

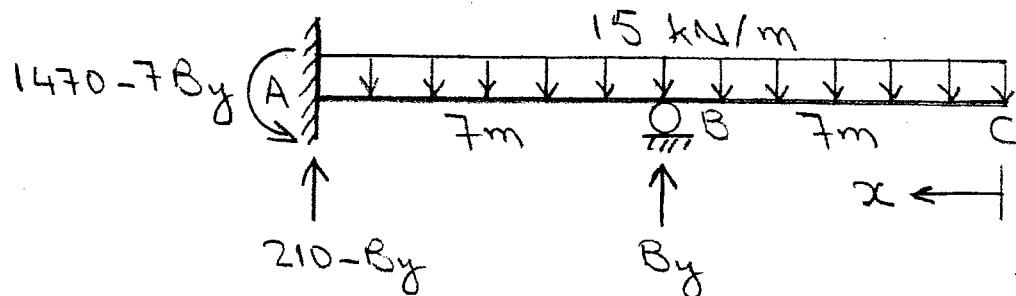
$$\underline{M_E = -237.4 \text{ kN.m}}$$



Span End Moments and Shears

For reactions, and shear and bending moment diagrams, see solution of Problem 13.54.

14.11



Segment	x Coordinate		M	$\frac{\partial M}{\partial B_y}$
	Origin	Limits (m)		
CB	C	0 - 7	$-7.5x^2$	0
BA	C	7 - 14	$-7.5x^2 + B_y(x-7)$	$x-7$

$$\frac{\partial U}{\partial B_y} = \int \left(\frac{\partial M}{\partial B_y} \right) \frac{M}{EI} dx = 0$$

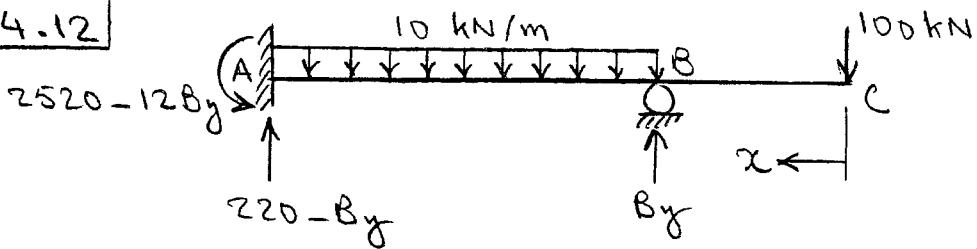
$$\frac{1}{EI} \int_7^{14} (x-7) [-7.5x^2 + B_y(x-7)] dx = 0$$

$$-25510.625 + (114.33) B_y = 0$$

$$\underline{B_y = 223.125 \text{ kN} \uparrow}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.15.

14.12



Segment	x coordinate		M	$\frac{\partial M}{\partial B_y}$
	Origin	Limits (m)		
CB	C	0 - 6	-100x	0
BA	C	6 - 18	-100x - 5(x-6)^2 + By(x-6)	x - 6

$$\frac{\partial U}{\partial B_y} = \int \left(\frac{\partial M}{\partial B_y} \right) \frac{M}{EI} dx = 0$$

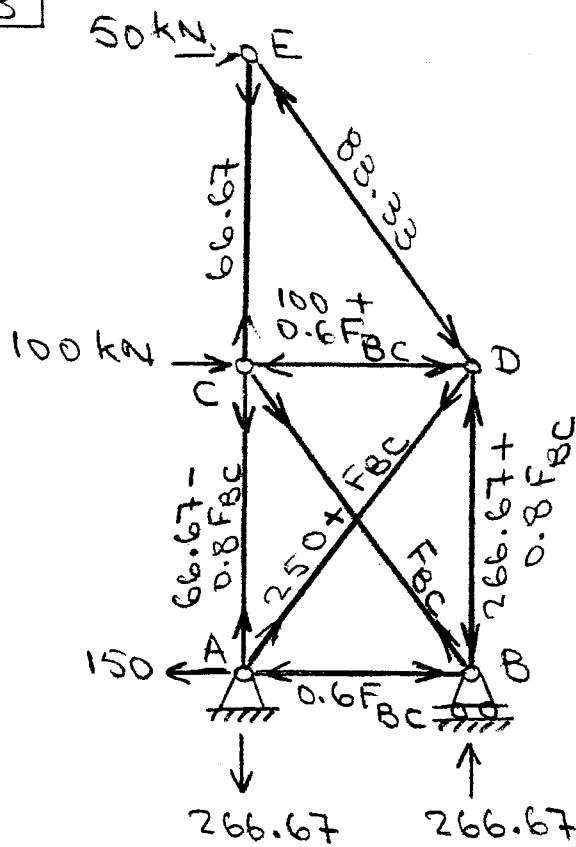
$$\frac{1}{EI} \int_6^{18} (x-6) [-100x - 5(x-6)^2 + By(x-6)] dx = 0$$

$$-126720 + 576By = 0$$

$$By = \underline{220 \text{ kN } \uparrow}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.16.

14.13



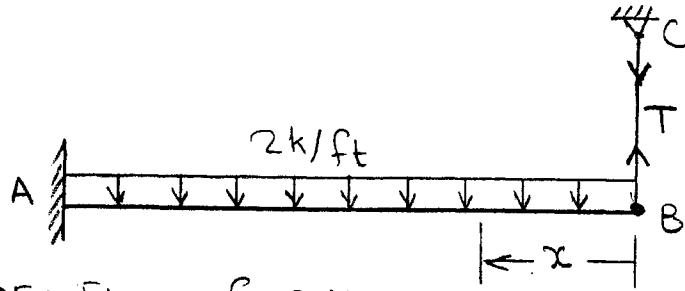
Member	L (m)	F	$\frac{\partial F}{\partial F_{BC}}$	$\left(\frac{\partial F}{\partial F_{BC}}\right) FL$	F (kN)
AB	3	$-0.6 F_{BC}$	-0.6	$1.08 F_{BC}$	71.9
CD	3	$-100 - 0.6 F_{BC}$	-0.6	$180 + 1.08 F_{BC}$	-28.1
AC	4	$66.67 - 0.8 F_{BC}$	-0.8	$-213.33 + 2.56 F_{BC}$	162.5
BD	4	$-266.67 - 0.8 F_{BC}$	-0.8	$853.33 + 2.56 F_{BC}$	-170.8
CE	4	66.67	0	0	66.7
AD	5	$250 + F_{BC}$	-1	$1250 + 5 F_{BC}$	130.2
DE	5	-83.33	0	0	-83.3
BC	5	F_{BC}	1	$5 F_{BC}$	-119.8
		\sum		$2070 + 17.28 F_{BC}$	

$$\frac{\partial \Delta}{\partial F_{BC}} = \frac{1}{EA} \sum \left(\frac{\partial F}{\partial F_{BC}} \right) FL = 0$$

$$\frac{1}{EA} (2070 + 17.28 F_{BC}) = 0$$

$$F_{BC} = -119.8 \text{ kN} = \underline{\underline{119.8 \text{ kN (C)}}}$$

14-14



$$\frac{\partial V}{\partial T} = \sum \left(\frac{\partial F}{\partial T} \right) \frac{FL}{AE} + \int \left(\frac{\partial M}{\partial T} \right) \frac{M}{EI} dx = 0$$

Cable BC: $F = T$; $\frac{\partial F}{\partial T} = 1$

Beam AB: $M = Tx - x^2$; $\frac{\partial M}{\partial T} = x$

$$\begin{aligned}\frac{\partial V}{\partial T} &= \frac{1}{E} \left[\frac{T(5)(12)^2}{0.5} + \frac{(12)^4}{500} \int_0^{15} x(Tx - x^2) dx \right] \\ &= 48096T - 524880 = 0\end{aligned}$$

$$\underline{T = 10.9 \text{ k (T)}}$$

Chapter Fifteen

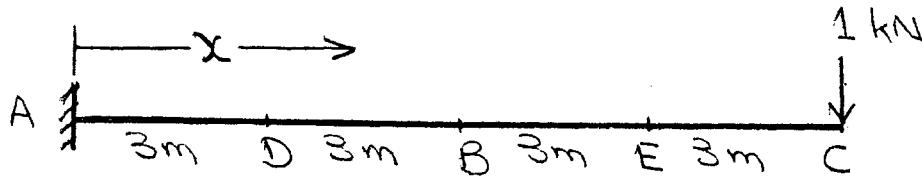
Influence Lines for

Statically Indeterminate

Structures

CHAPTER 15

15-1 Select $C_y (\uparrow +)$ as the redundant.



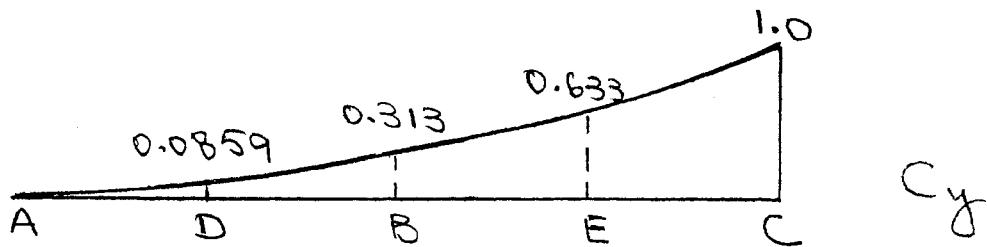
$$C_y = - \frac{f_{xc}}{\bar{f}_{cc}} = - \frac{f_{xc}}{f_{cc}}$$

Using beam deflection formulas:

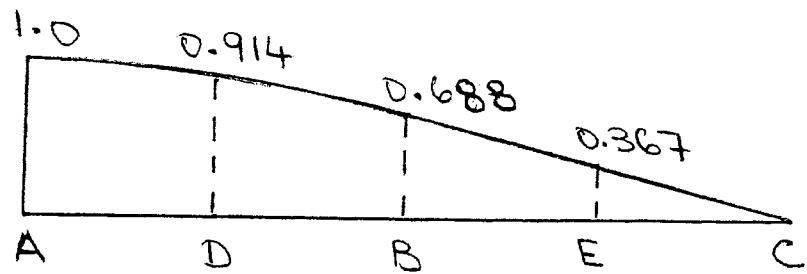
$$f_{xc} = - \frac{1}{6EI} (x^3 - 36x^2)$$

$$\bar{f}_{cc} = - f_{cc} = + \frac{576 \text{ kN.m}^3/\text{kN}}{EI}$$

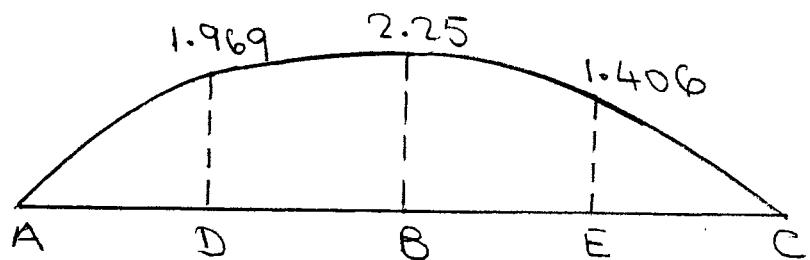
x (m)	EIf_{xc}	C_y (kN/kN)	Ay (kN/kN)	M_A (kN.m/kN)	S_B (kN/kN)	M_B (kN.m/kN)
0	0	0	1	0	0	0
3	-49.5	0.0859	0.914	1.989	-0.0859	0.516
6	-180	0.313	0.688	2.25	-0.313(L) 0.687(R)	1.875
9	-364.5	0.633	0.367	1.406	0.367	0.797
12	-576	1	0	0	0	0



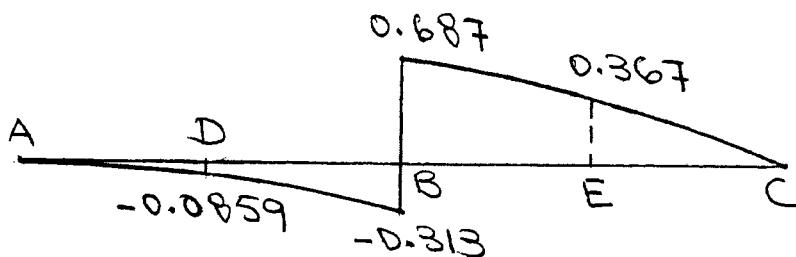
15.1 (Contd.)



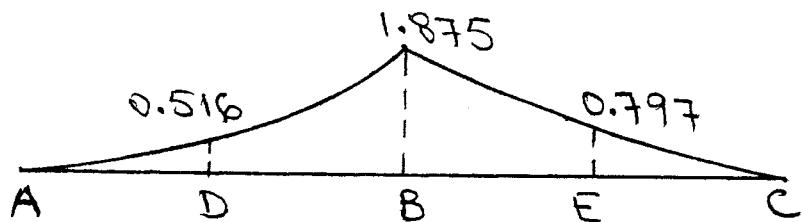
A_y



$M_A(G+)$

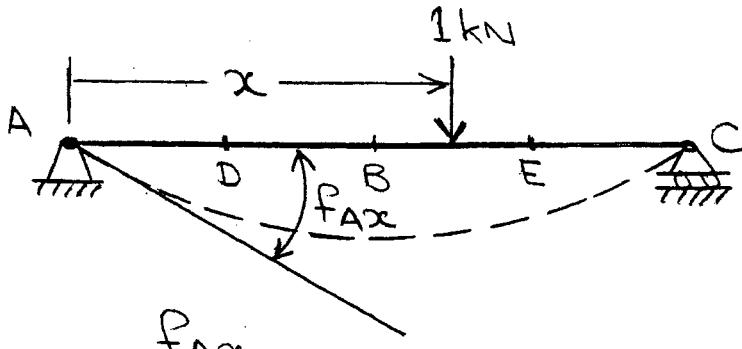


S_B



M_B

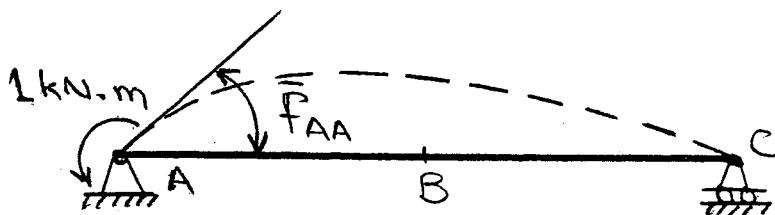
15.2 Select M_A ($G+$) as the redundant.



$$M_A = -\frac{F_{Ax}x}{f_{AA}}$$

Using beam deflection formulas:

$$f_{Ax} = -\frac{1}{72EI} (x^3 - 36x^2 + 288x)$$

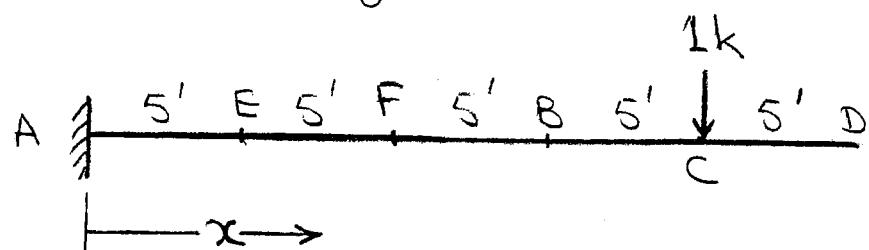


$$f_{AA} = + \frac{4 \text{ kN}\cdot\text{m}^2/\text{kN}\cdot\text{m}}{EI}$$

x (m)	$EI f_{Ax}$	M_A
0	0	0
3	-7.875	1.969
6	-9	2.25
9	-5.625	1.406
12	0	0

For influence lines, see solution of Problem 15.1.

15.3 Select C_y ($\uparrow +$) as the redundant.



$$C_y = -\frac{f_{xcx}}{\bar{f}_{cc}} = -\frac{f_{xc}}{\bar{f}_{cc}}$$

Using beam deflection formulas:

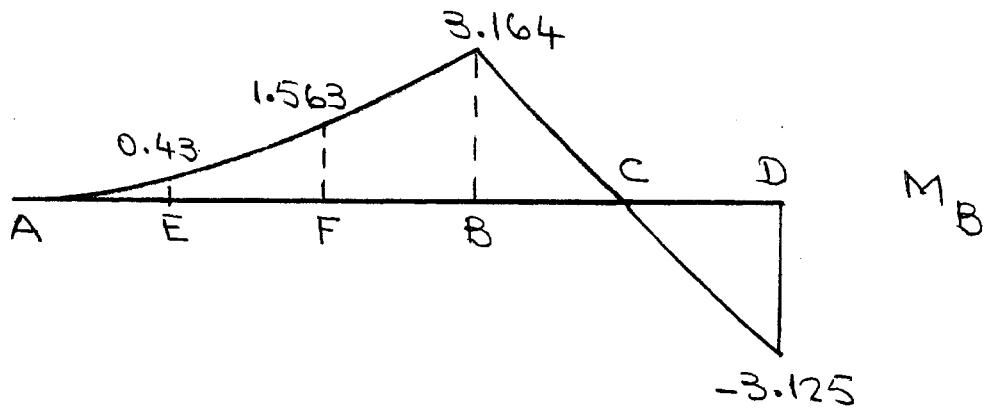
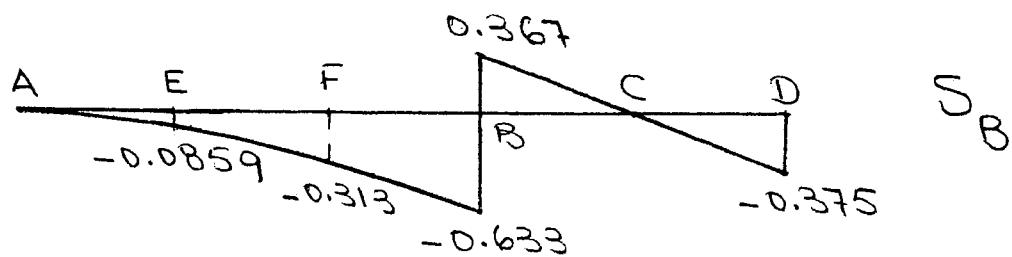
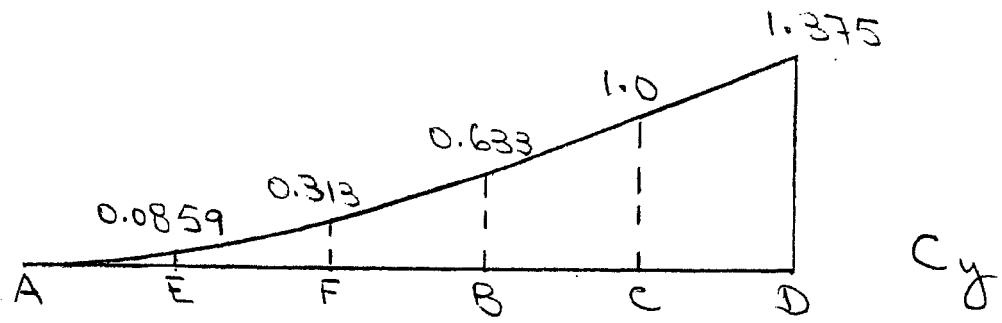
$$\text{for } 0 \leq x \leq 20' \quad f_{xc} = \frac{1}{6EI} (x^3 - 60x^2)$$

$$\text{for } 20' \leq x \leq 25' \quad f_{xc} = \frac{200}{3EI} (20 - 3x)$$

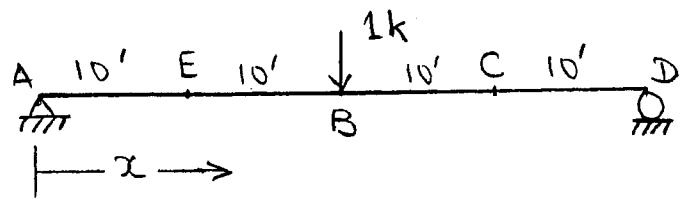
$$\bar{f}_{cc} = -f_{cc} = +\frac{2666.67 k \cdot ft^3 / k}{EI}$$

x (ft)	$EI f_{xc}$	C_y (k/k)	S_B (k/k)	M_B (k-ft/k)
0	0	0	0	0
5	-229.17	0.0859	-0.0859	0.43
10	-833.33	0.313	-0.313	1.563
15	-1687.5	0.633	-0.633 (L) 0.367 (R)	3.164
20	-2666.67	1	0	0
25	-3666.67	1.375	-0.375	-3.125

15.3 (Cont'd.)



15.4 Select B_y ($\uparrow +$) as the redundant.



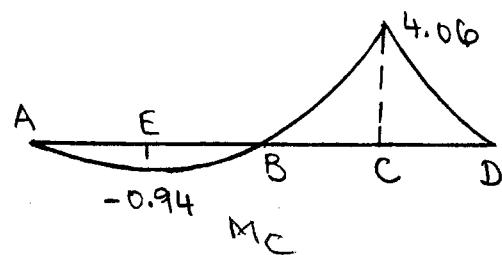
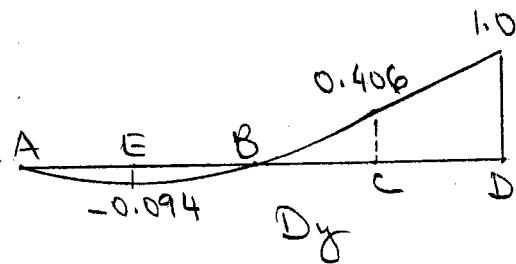
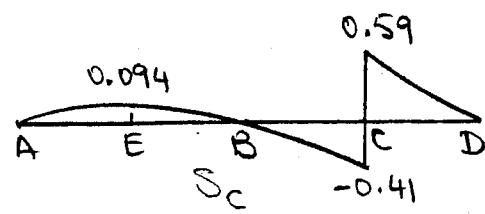
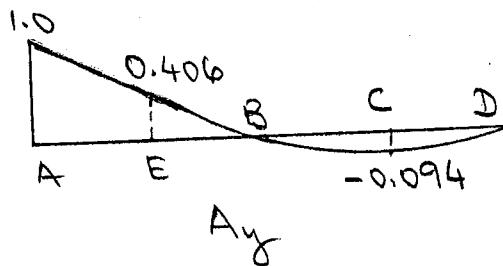
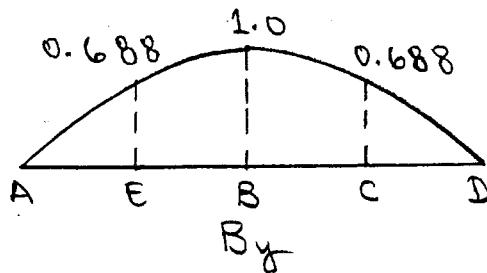
$$B_y = -\frac{f_{Bx}}{\bar{f}_{BB}} = -\frac{f_{xB}}{\bar{f}_{BB}}$$

Using beam-deflection formulas:

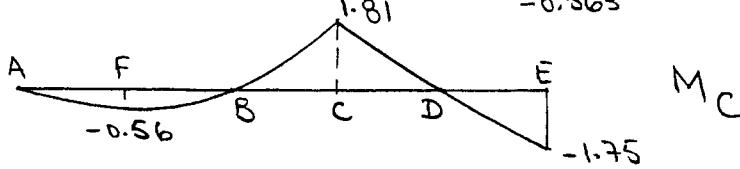
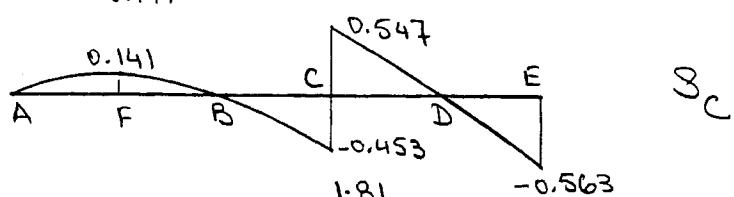
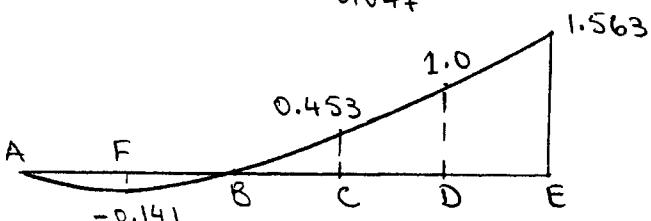
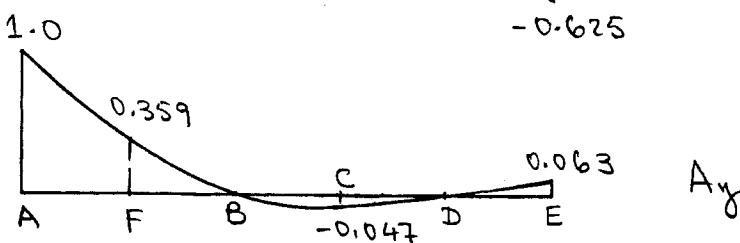
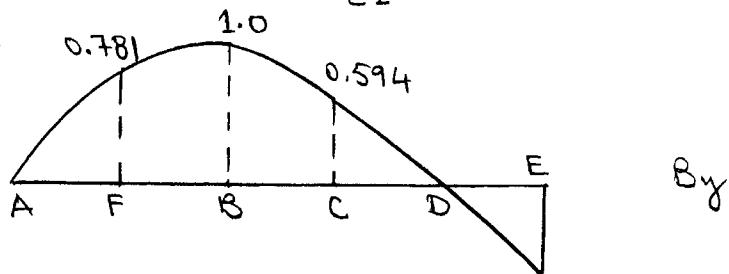
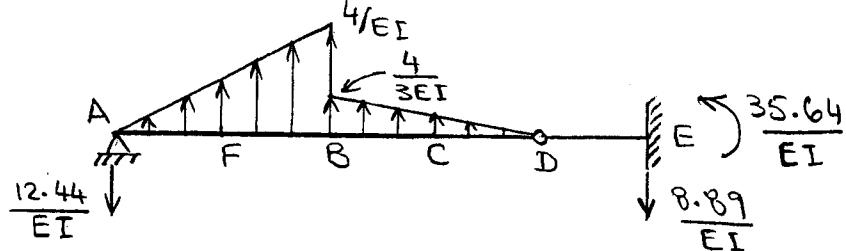
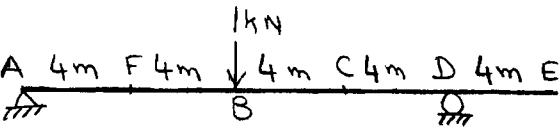
$$f_{EB} = f_{CB} = -\frac{916.7 k \cdot ft^3/k}{EI}$$

$$\bar{f}_{BB} = -\frac{1333.3 k \cdot ft^3/k}{EI}$$

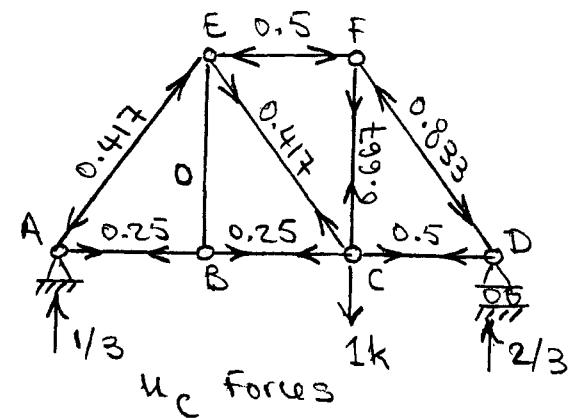
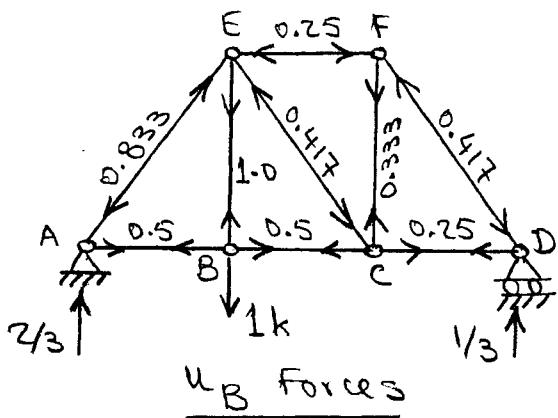
$$\bar{f}_{BB} = -f_{BB} = +\frac{1333.3 k \cdot ft^3/k}{EI}$$



15.5 Select B_y ($\uparrow +$) as the redundant.

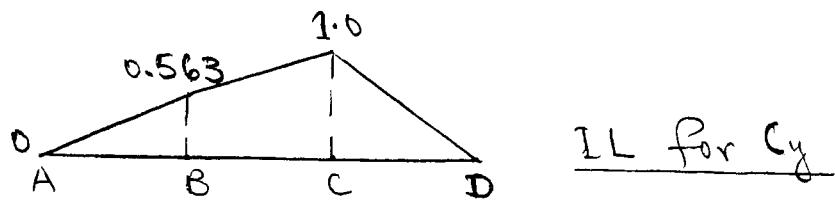


15-6 Select C_y ($\uparrow +$) as the redundant.



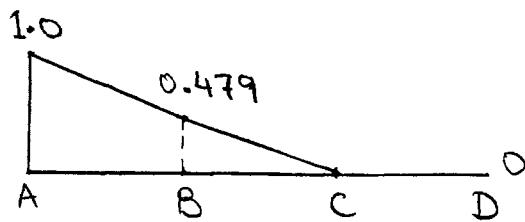
Member	L (ft)	u_B	u_C	$u_B u_C L$	$u_C^2 L$
AB	15	0.5	0.25	1.875	0.938
BC	15	0.5	0.25	1.875	0.938
CD	15	0.25	0.5	1.875	3.75
EF	15	-0.25	-0.5	1.875	3.75
BE	20	1.0	0	0	0
CF	20	0.333	0.667	4.442	8.898
AE	25	-0.833	-0.417	8.684	4.347
CE	25	-0.417	0.417	-4.347	4.347
DF	25	-0.417	-0.833	8.684	17.347
		\sum	24.963	44.315	

Unit load at B: $C_y = -\frac{24.963}{44.315} = -0.563 \text{ k}$ or $0.563 \text{ k} \uparrow$

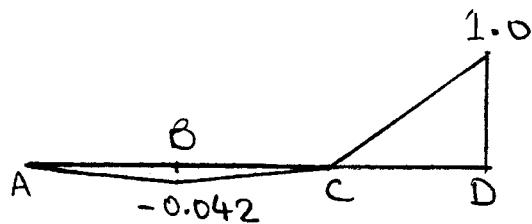


IL for C_y

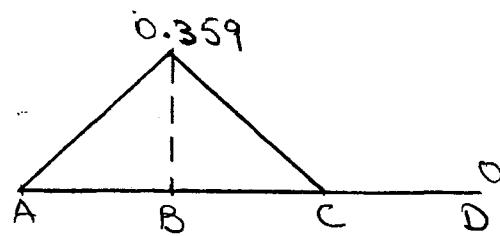
15.6 (cont'd.)



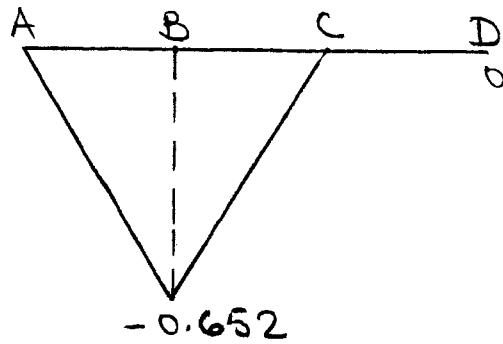
IL for A_y



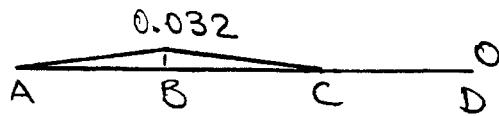
IL for D_y



IL for F_{BC}

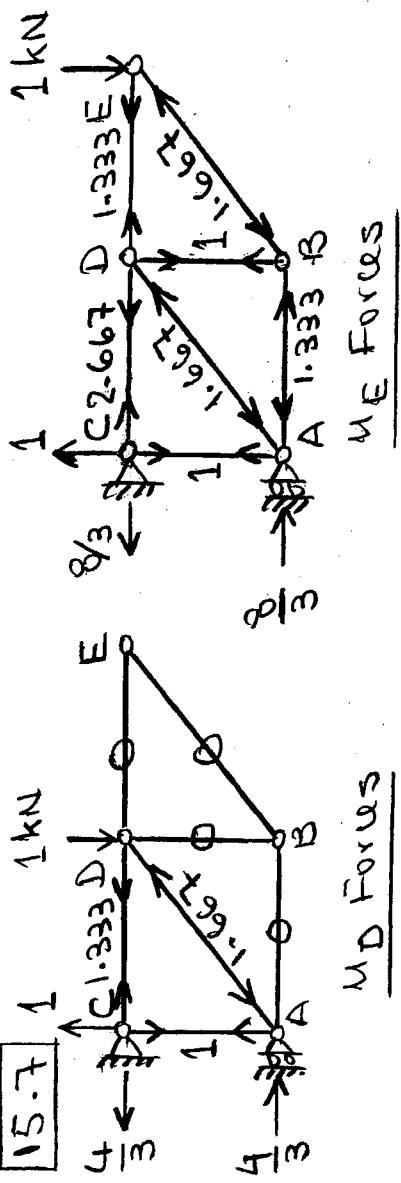


IL for F_{CE}

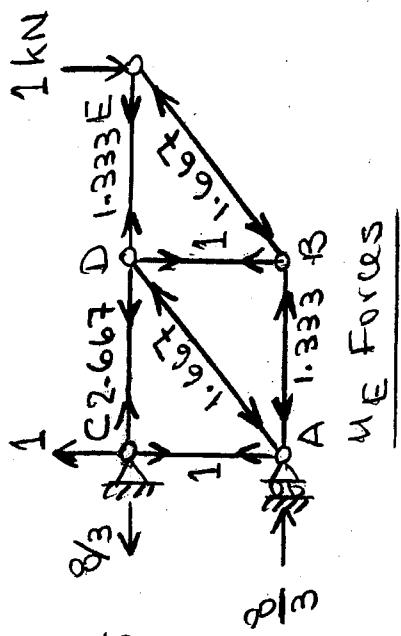


IL for F_{EF}

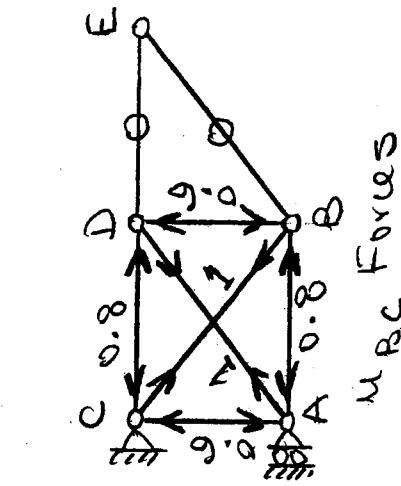
15.7



UD Forces



UE Forces



UEC Forces

Member	L (m)	u _D	u _E	u _{BC}	u _{DE}	u _{AC}	u _{BC} ²	u _{DE} ²
AB	8	0	-1.333	-0.8	0	0	8.533	5.12
CD	8	1.333	2.667	-0.8	-8.533	-17.067	5.12	0
DE	8	0	1.333	0	0	0	0	0
AC	6	1	1	-0.6	-3.6	-3.6	2.16	2.16
BD	6	0	1	-0.6	0	-3.6	-16.967	10
AD	10	-1.667	-1.667	1	1	0	0	0
BE	10	0	-1.667	0	0	0	0	0
BC	10	0	0	0	1	0	0	10
						Σ	-28.8	-32.4
								34.56

15.7 (contd.)

Unit load at D:

$$F_{BC,D} + F_{BC,E} = 0$$

$$F_{BC} = \frac{28.8}{34.56} = 0.833 \text{ kN/kN (T)}$$

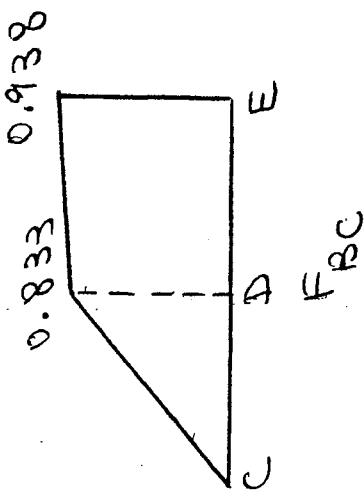
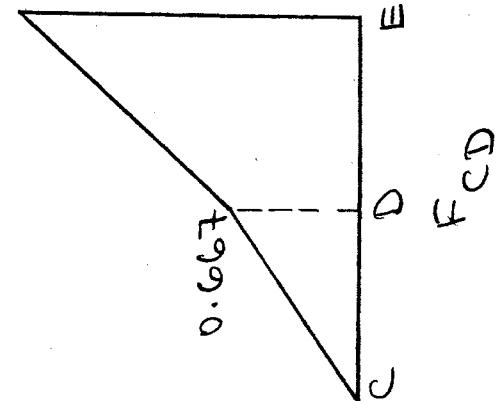
$$F_{CD} = u_D + u_E F_{BC} = 1.333 - 0.8 (0.833) \\ = 0.667 \text{ kN/kN (T)}$$

Unit load at E:

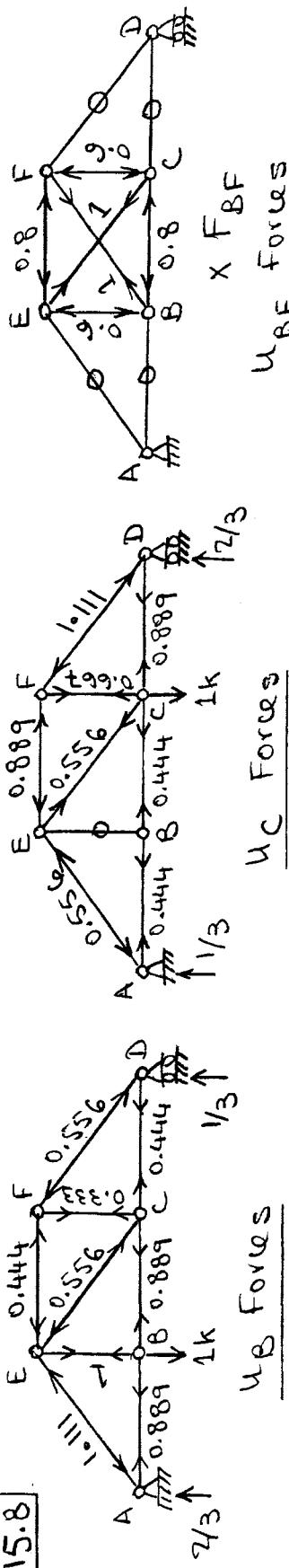
$$F_{BC,E} + F_{BC,EC} F_{BC} = 0 \\ F_{BC} = \frac{32.4}{34.56} = 0.938 \text{ kN/kN (T)}$$

$$F_{CD} = u_E + u_E F_{BC} = 2.667 - 0.8 (0.938) \\ = 1.917 \text{ kN/kN (T)}$$

1.917



15.8



Member	L (in.)	A	U_B	U_C	U_{BF}	$\frac{U_B U_{BFL}}{A}$	$\frac{U_C U_{BFL}}{A}$	$\frac{U_{BF}^2}{A}$
AB	192	8	0.889	0.444	0	0	-17.07	15.36
BC	192	8	0.889	0.444	-0.8	0	0	0
CD	192	8	0.444	0.889	0	0	8.52	15.36
EF	192	8	-0.444	-0.889	-0.8	8.52	17.07	0
BE	144	6	1	0	-0.6	-14.4	0	8.64
CF	144	6	0.333	0.667	-0.6	-4.8	-9.6	0
AE	240	8	-1.111	-0.556	0	0	0	0
DF	240	8	-0.556	-1.111	0	-22.24	22.24	40
CE	240	6	-0.556	0.556	1	0	0	40
BF	240	6	0	0	0	0	0	0
Σ						-49.99	21.19	128

15.8 (cont'd.)

Unit load at B:

$$f_{BF} = \frac{49.99}{12.8} = 0.39 \text{ k}$$

$$f_{BC} = 0.889 - 0.8(0.39) = 0.577 \text{ k}$$

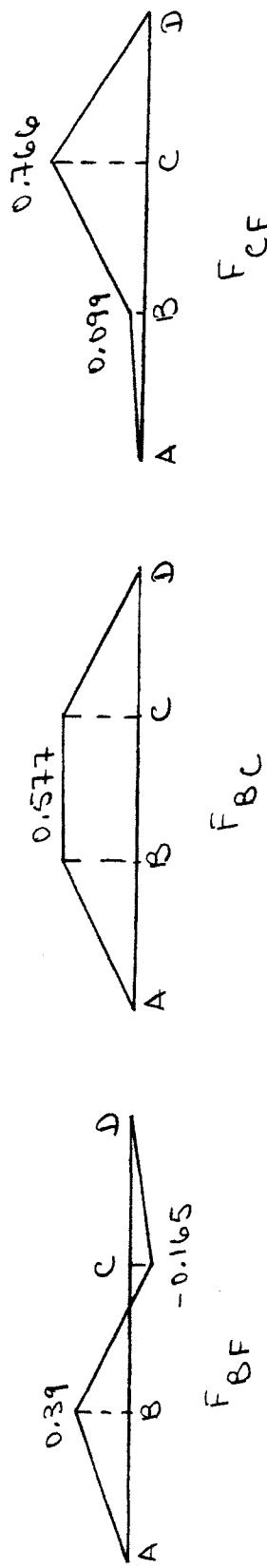
$$F_{CF} = 0.333 - 0.6(0.39) = 0.099 \text{ k}$$

Unit load at C:

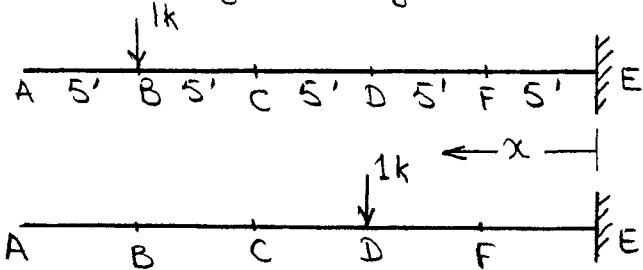
$$f_{BF} = -\frac{21.19}{12.8} = -0.165 \text{ k}$$

$$f_{BC} = 0.444 - 0.8(-0.165) = 0.577 \text{ k}$$

$$F_{CF} = 0.667 - 0.6(-0.165) = 0.766 \text{ k}$$



15.9 Select B_y and D_y as the redundants.



$$\text{Compatibility Eqs.: } f_{Bx} + \bar{f}_{BB} B_y + \bar{f}_{BD} D_y = 0$$

$$f_{Dx} + \bar{f}_{DB} B_y + \bar{f}_{DD} D_y = 0$$

Using beam deflection formulas:

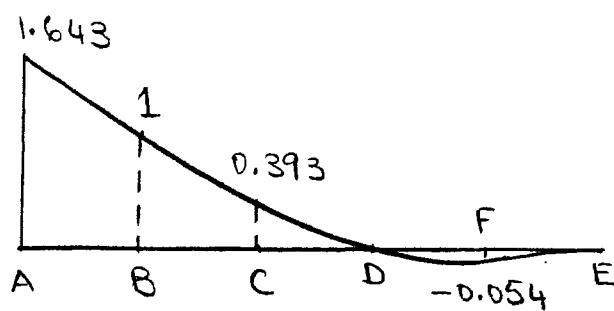
$$f_{Bx} = f_{x_B} = \begin{cases} \frac{1}{6EI} (x^3 - 60x^2) & \text{for } 0 \leq x \leq 20' \\ \frac{200}{3EI} (20 - 3x) & \text{for } 20' \leq x \leq 25' \end{cases}$$

$$f_{Dx} = f_{x_D} = \begin{cases} \frac{1}{6EI} (x^3 - 30x^2) & \text{for } 0 \leq x \leq 10' \\ \frac{50}{3EI} (10 - 3x) & \text{for } 10' \leq x \leq 25' \end{cases}$$

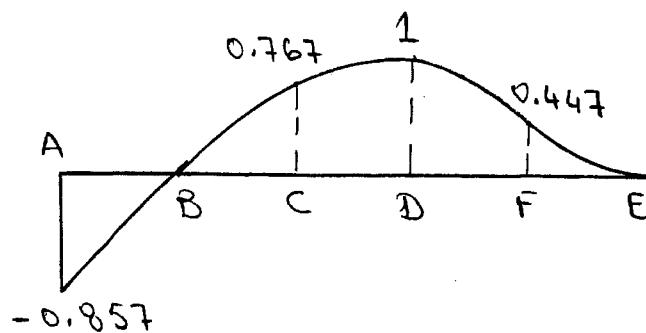
$$\bar{f}_{BB} = \frac{2666.7}{EI}; \quad \bar{f}_{BD} = \bar{f}_{DB} = \frac{833.3}{EI}; \quad \bar{f}_{DD} = \frac{333.3}{EI}$$

x (ft.)	$EI f_{Bx}$	$EI f_{Dx}$	B_y	D_y	δ_C	M_C
0	0	0	0	0	0	0
5	-229.2	-104.2	-0.054	0.447	-0.054	-0.27
10	-833.3	-333.3	0	1	0	0
15	-1687.5	-583.3	0.393	0.767	-0.607(L) 0.393(R)	1.97
20	-2666.7	-833.3	1	0	0	0
25	-3666.7	-1083.3	1.643	-0.857	0.643	-1.79

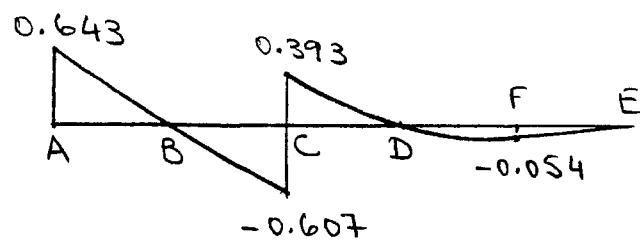
15.9 (Contd.)



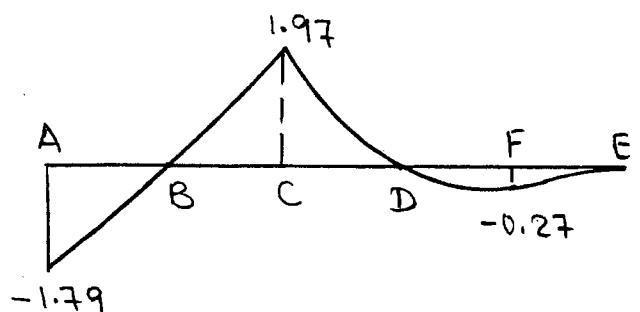
B_y



D_y

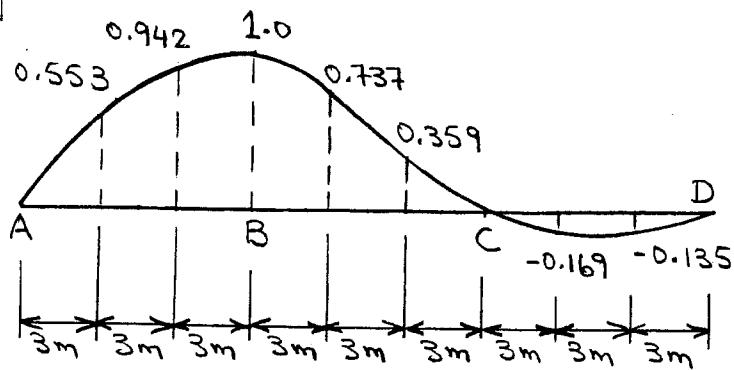
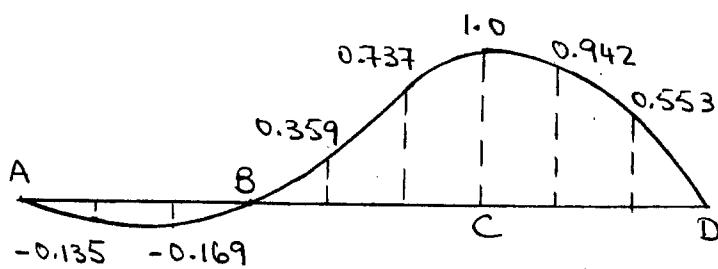
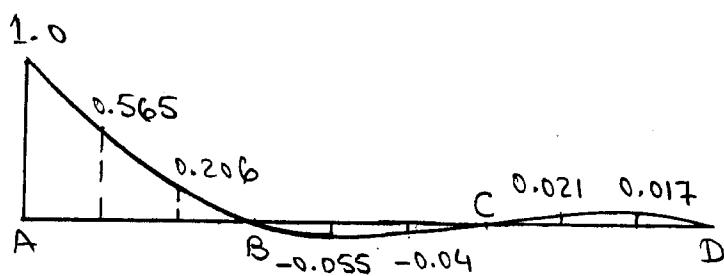
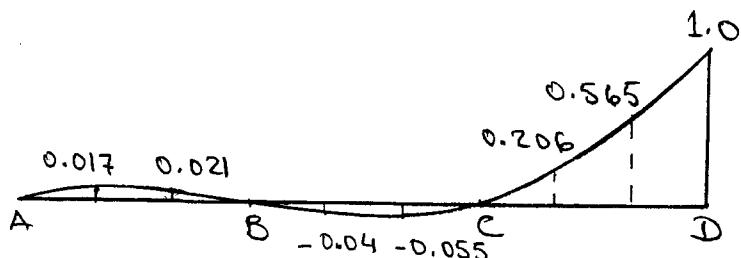


S_c

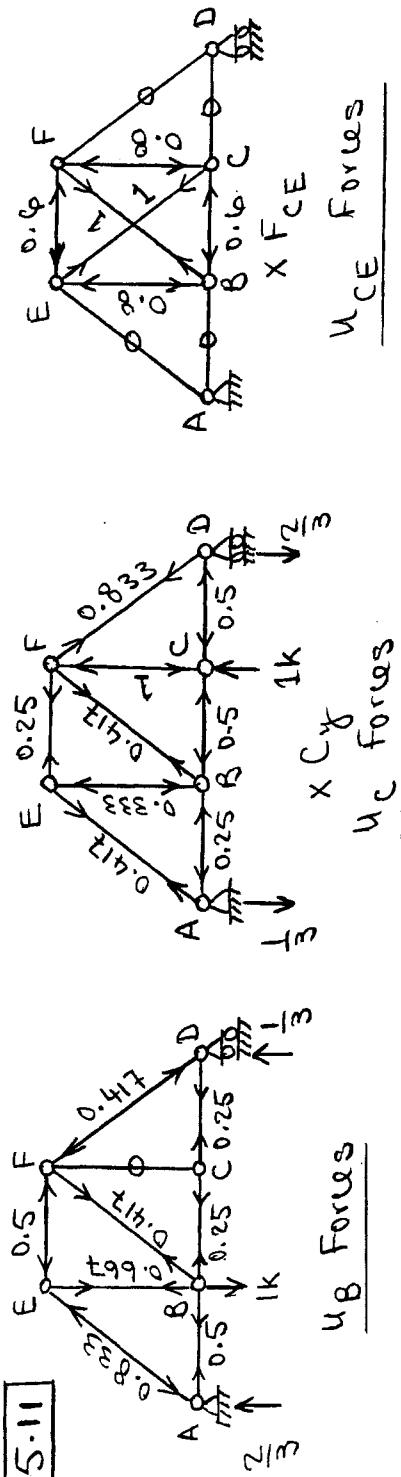


M_c

15.10

 B_y  C_y  A_y  D_y

15.11



u_B Forces

u_C Forces

u_E Forces

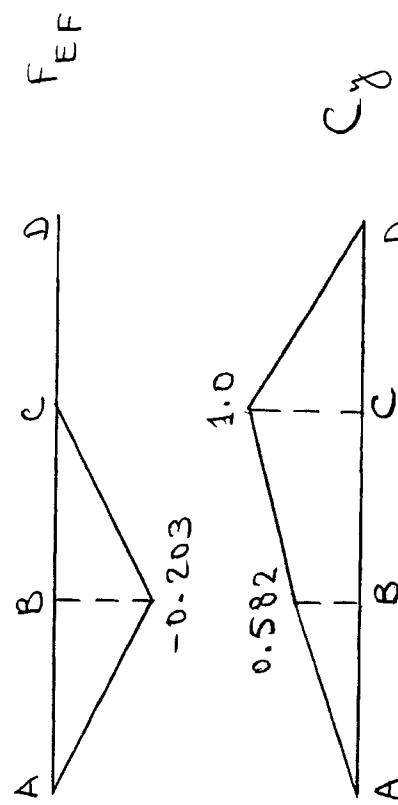
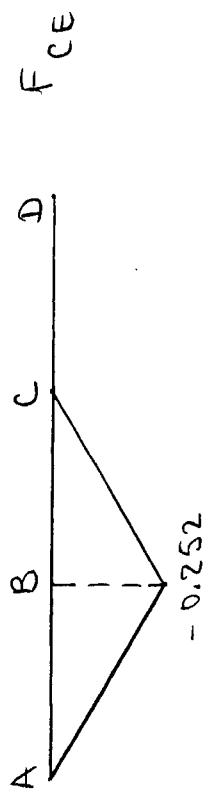
u_F Forces

Member	L (ft)	u_B	u_C	u_E	u_F	u_{BCE}	u_{BCD}	u_{CDE}	u_{CEF}
A-B	15	0.5	-0.25	0	0	-1.875	0	0.938	0
B-C	15	0.25	-0.5	-0.6	-0.6	-1.875	3.75	5.4	4.5
C-D	15	0.25	0.5	0	0	-1.875	3.75	0	0
E-F	15	-0.5	0.25	-0.6	-0.6	-1.875	4.5	0.938	-2.25
B-E	20	0.667	-0.333	-0.8	-0.8	-4.442	-10.672	2.218	5.320
C-F	20	0	-1	-0.8	-0.8	0	0	12.8	16
A-E	25	-0.833	0.417	0	0	-3.684	4.347	0	0
F-D	25	-0.417	0.833	0	0	-8.984	17.347	0	0
B-F	25	0.417	0.417	1	1	4.347	4.347	25	10.425
C-E	25	0	0	0	0	0	0	0	0
Σ		-24.963	2.003	57.635	86.4	34.003			

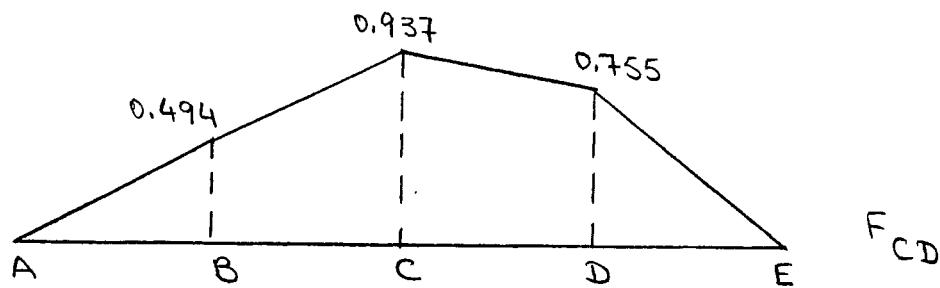
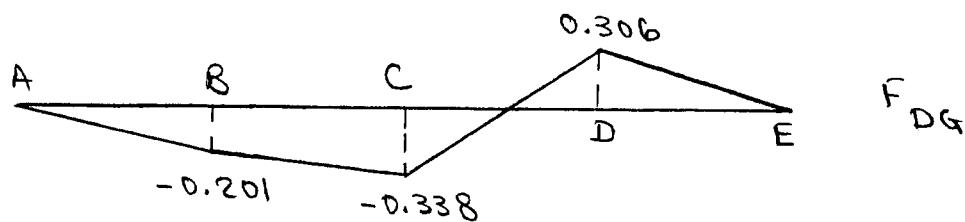
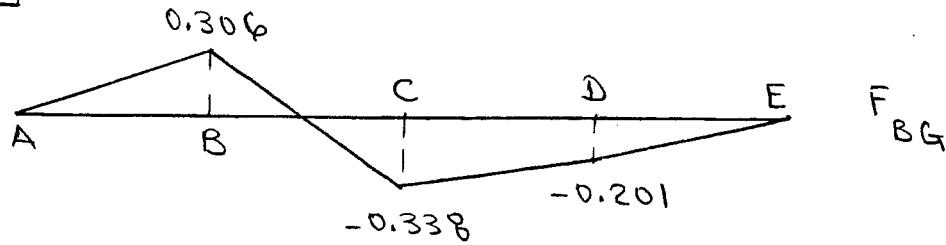
$$\begin{aligned} \text{Unit load at B: } & -24.963 + 57.635 C_F + 34.003 F_{CE} = 0 \\ & 2.003 + 34.003 C_F + 86.4 F_{CE} = 0 \end{aligned}$$

Solving these equations, we obtain: $C_F = 0.582$ $F_{CE} = -0.252$ k

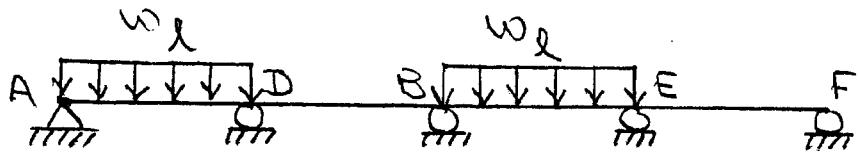
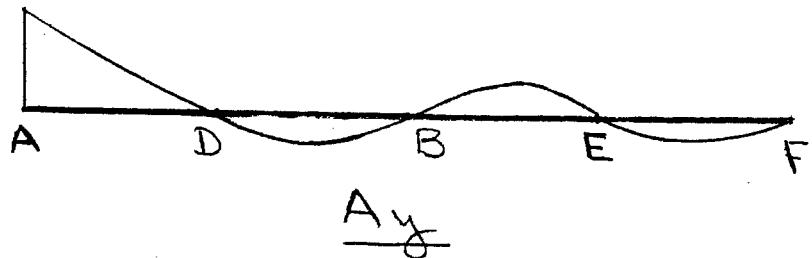
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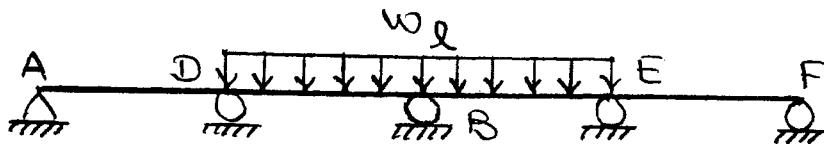
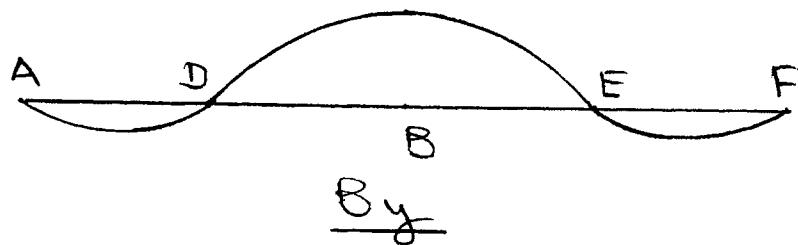
15.12



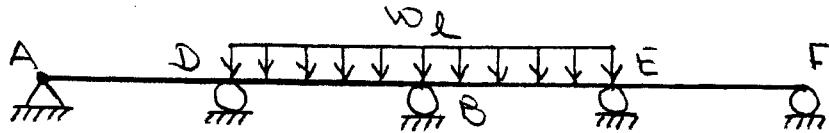
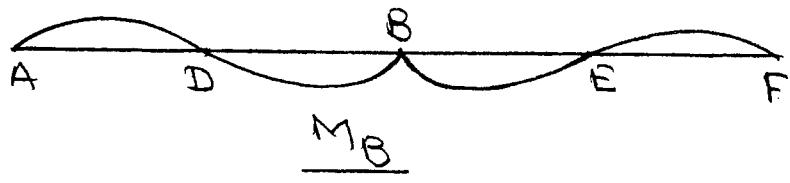
15-13



Live Load Arrangement for Maximum Positive A_y

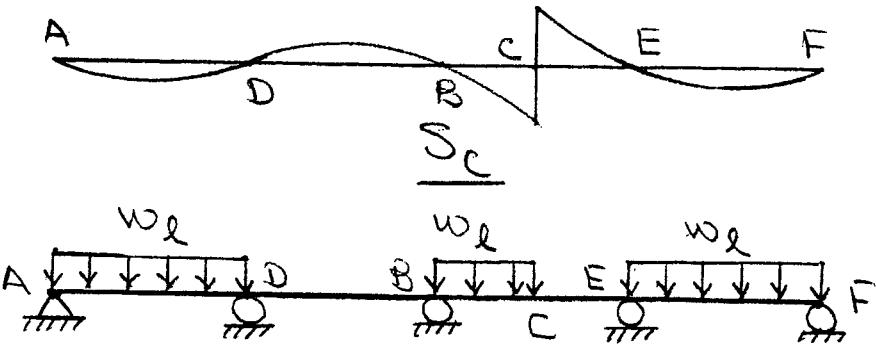


Live Load Arrangement for Maximum Positive B_y

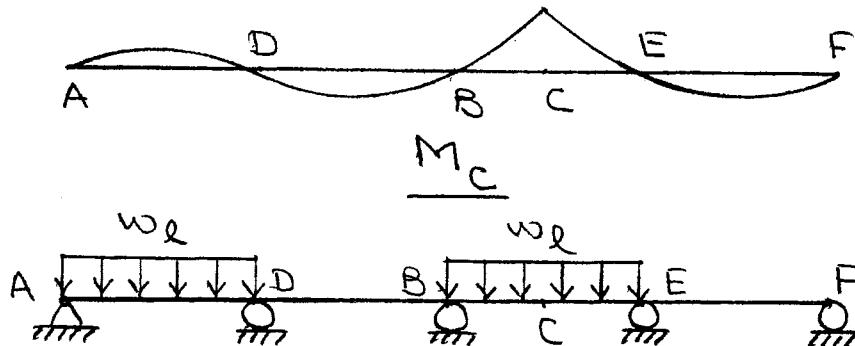


Live Load Arrangement for Maximum Negative M_B

15.13 (Cont'd.)

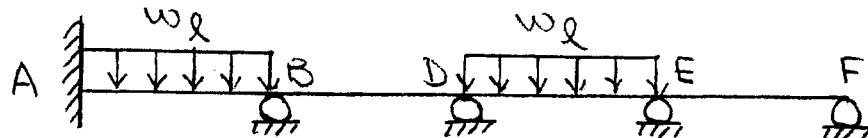
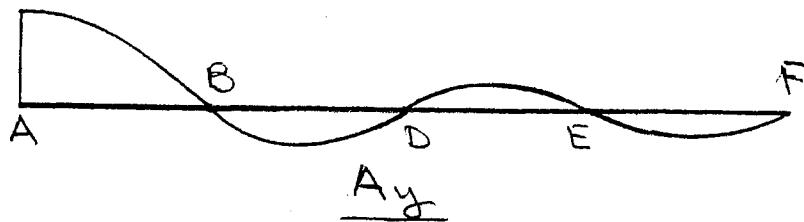


Live Load Arrangement for Maximum Negative S_C

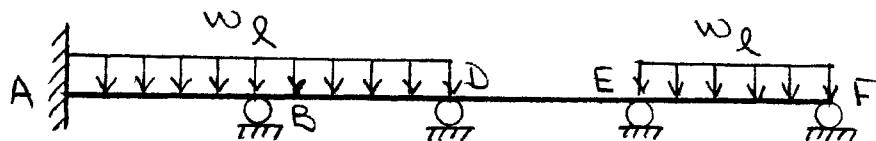
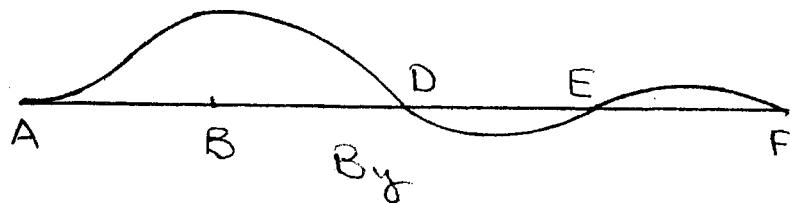


Live Load Arrangement for Maximum Positive M_C

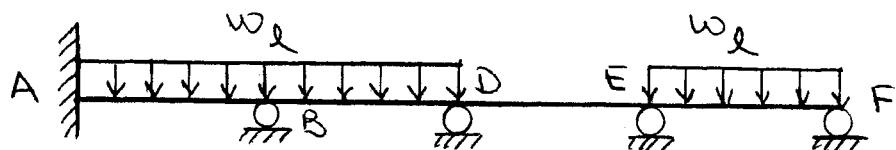
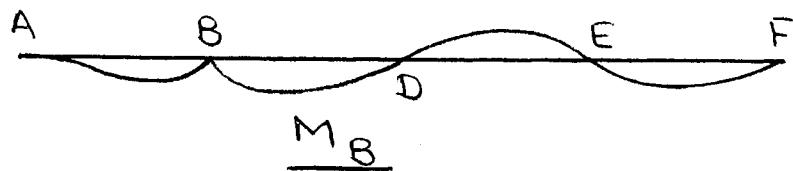
15.14



Live Load Arrangement for Maximum Positive A_y

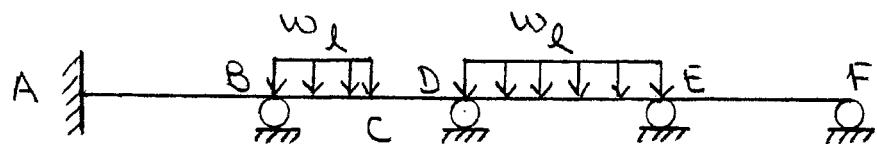
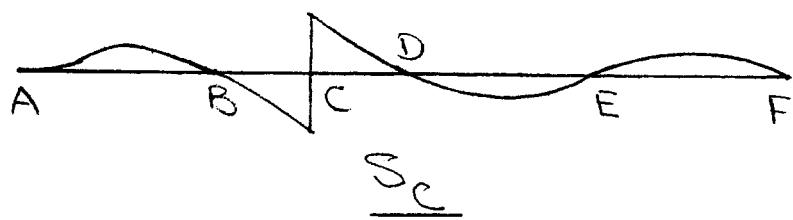


Live Load Arrangement for Maximum Positive B_y

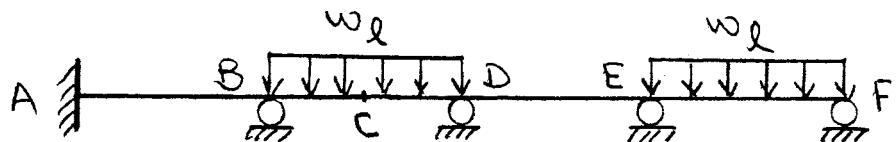
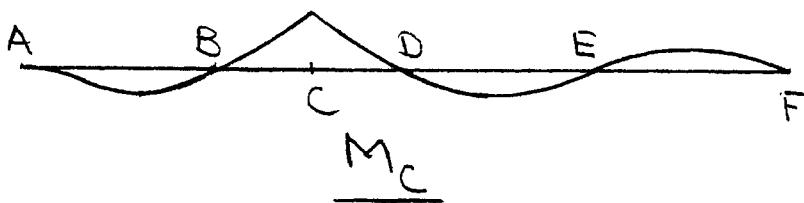


Live Load Arrangement for Maximum Negative M_B

15.14 (contd.)

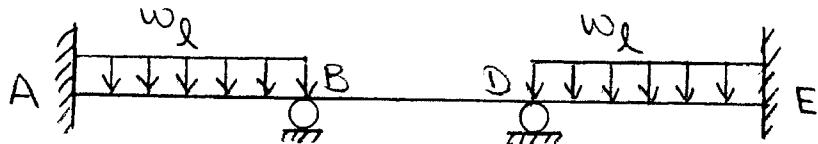
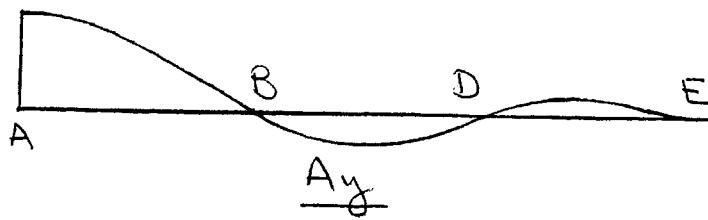


Live Load Arrangement for Maximum Negative S_c

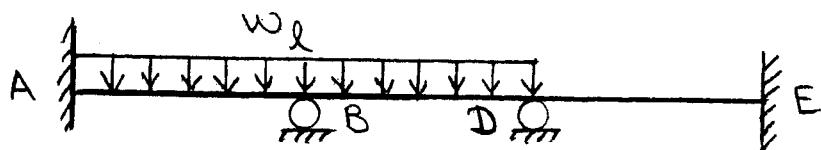
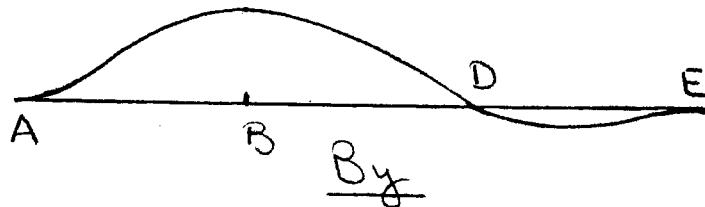


Live Load Arrangement for Maximum positive M_c

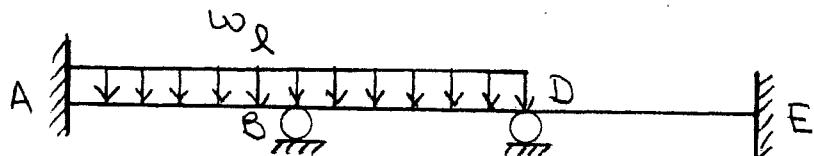
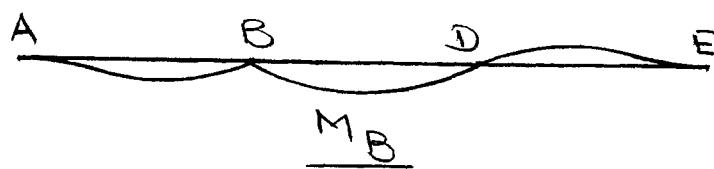
15.15



Live Load Arrangement for Maximum Positive A_y

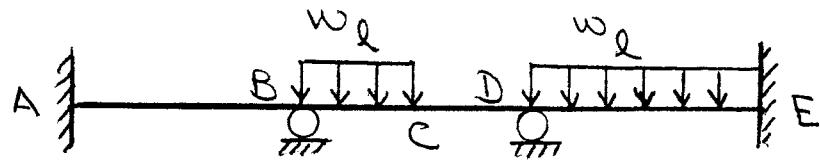
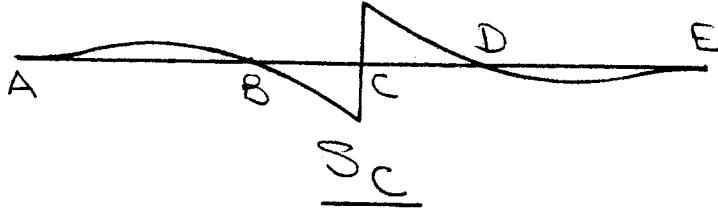


Live Load Arrangement for Maximum Positive B_y

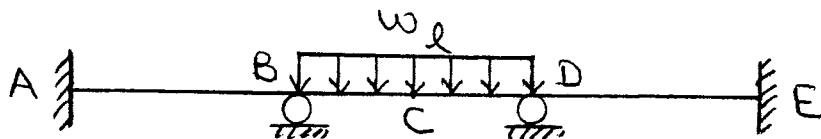
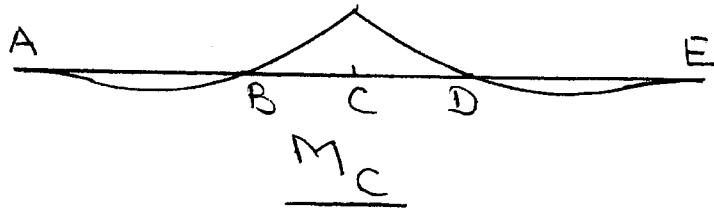


Live Load Arrangement for Maximum Negative M_B

15.15 (Contd.)

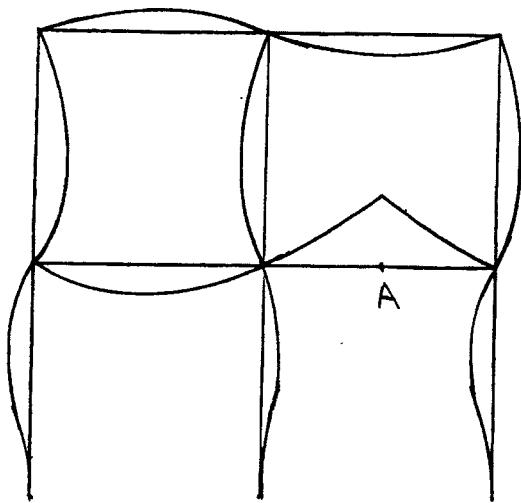


Live Load Arrangement for Maximum Negative S_C

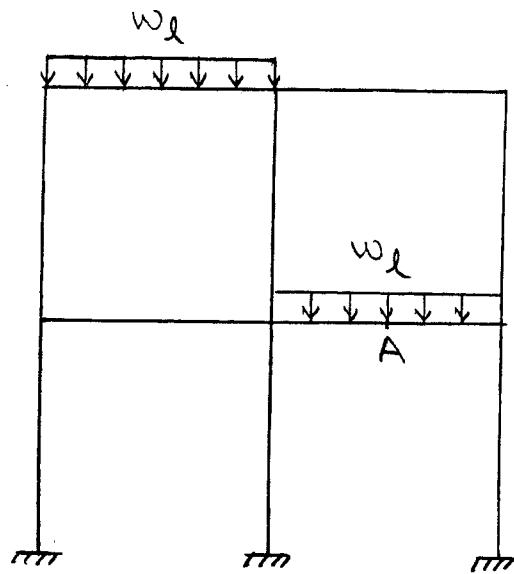


Live Load Arrangement for Maximum Positive M_C

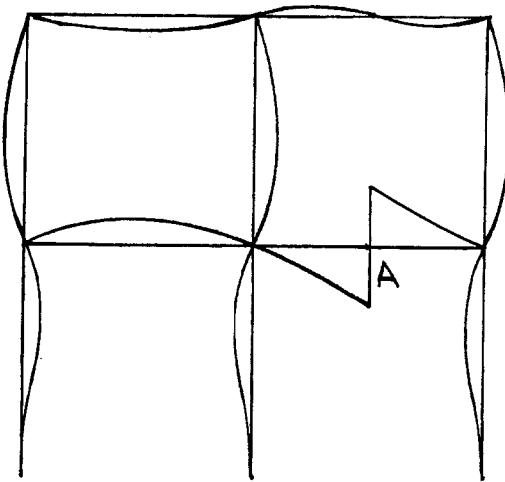
15.16



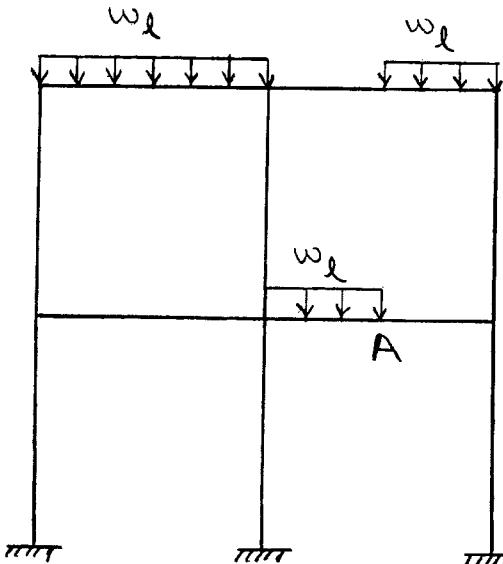
M_A



Live Load Arrangement
for Max. Positive M_A

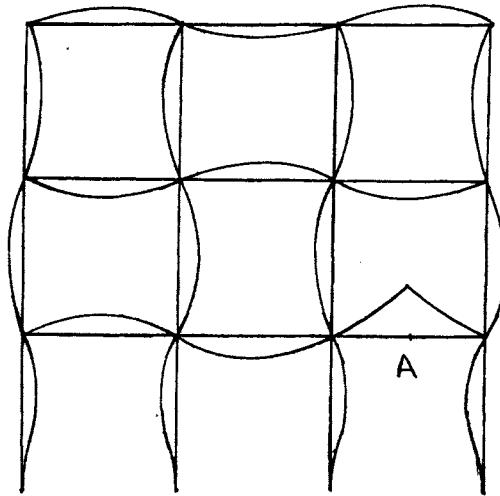


S_A

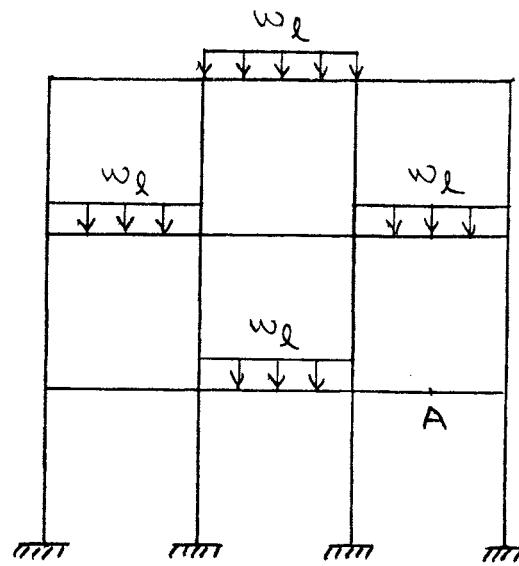


Live Load Arrangement
for Max. Negative S_A

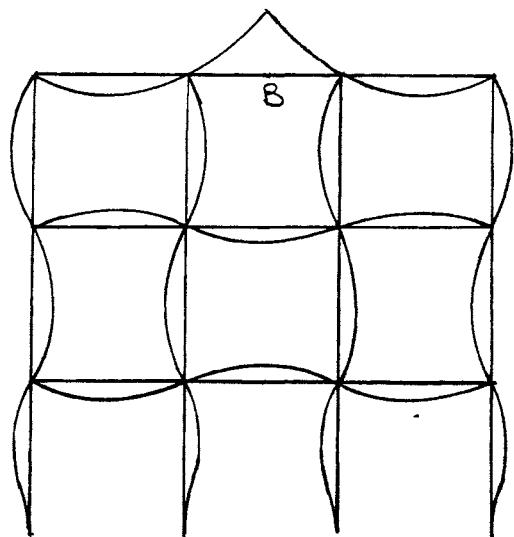
15.17



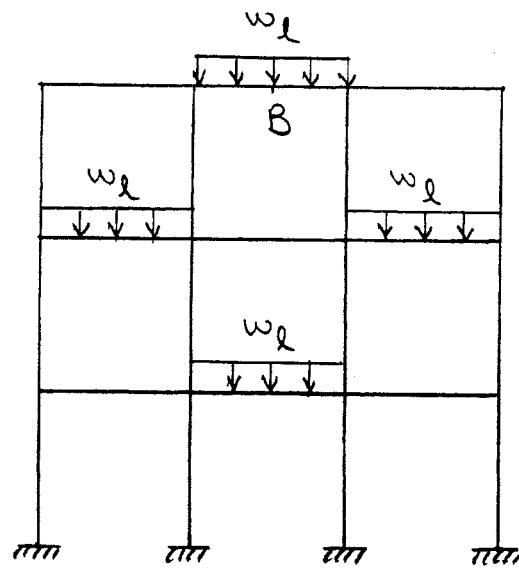
Qualitative I.L. for M_A



Live Load Arrangement
for Max. Negative M_A



Qualitative I.L. for M_B



Live Load Arrangement
for Max. Positive M_B

Chapter Sixteen

Slope-Deflection Method

CHAPTER 16

16-1 Fixed end moments:

$$FEM_{AC} = +40 \text{ k-ft}; \quad FEM_{CA} = -80 \text{ k-ft}$$

$$FEM_{CE} = +37.5 \text{ k-ft}; \quad FEM_{EC} = -37.5 \text{ k-ft}$$

Slope-deflection equations:

$$M_{AC} = 0.0667 EI \theta_C + 40; \quad M_{CA} = 0.133 EI \theta_C - 80$$

$$M_{CE} = 0.133 EI \theta_C + 37.5; \quad M_{EC} = 0.0667 EI \theta_C - 37.5$$

Equilibrium equation: $M_{CA} + M_{CE} = 0$

$$(0.133 EI \theta_C - 80) + (0.133 EI \theta_C + 37.5) = 0$$

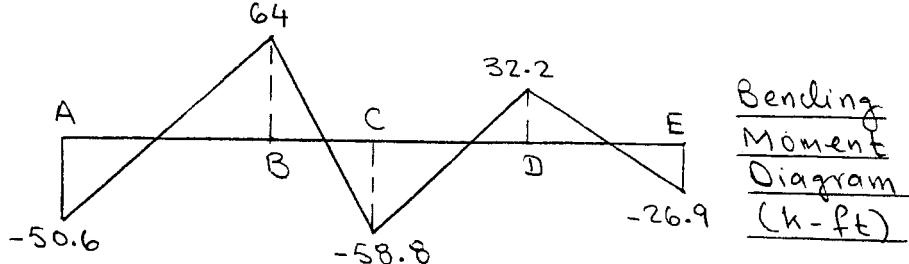
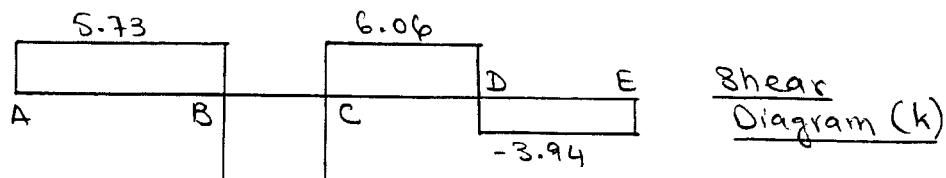
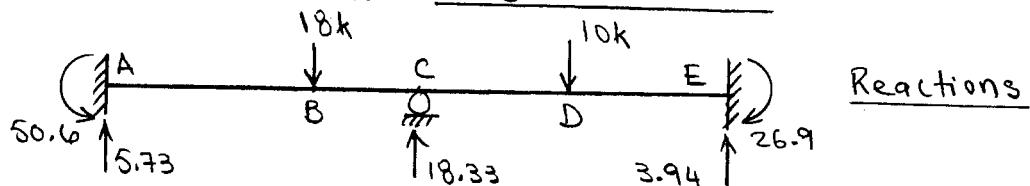
$$0.267 EI \theta_C - 42.5 = 0$$

$$EI \theta_C = 159.2 \text{ k-ft}^2$$

Member end moments. Substituting the numerical value of $EI \theta_C$ into the slope-deflection equations, we obtain:

$$\underline{M_{AC} = 50.6 \text{ k-ft}}; \quad \underline{M_{CA} = -58.8 \text{ k-ft}}$$

$$\underline{M_{CE} = 58.8 \text{ k-ft}}; \quad \underline{M_{EC} = -26.9 \text{ k-ft}}$$



16.2 Fixed-end Moments:

$$FEM_{AB} = \frac{1.5(30)^2}{12} + \frac{20(30)}{8} = 187.5 \text{ k-ft}$$

$$FEM_{BA} = -187.5 \text{ k-ft}; FEM_{BC} = \frac{3(20)^2}{12} = 100 \text{ k-ft}$$

$$FEM_{CB} = -100 \text{ k-ft.}$$

Slope-deflection equations:

$$M_{AB} = 0.0667 EI \theta_B + 187.5; M_{BA} = 0.133 EI \theta_B - 187.5$$

$$M_{BC} = 0.2 EI \theta_B + 100; M_{CB} = 0.1 EI \theta_B - 100$$

Equilibrium equation: $M_{BA} + M_{BC} = 0$

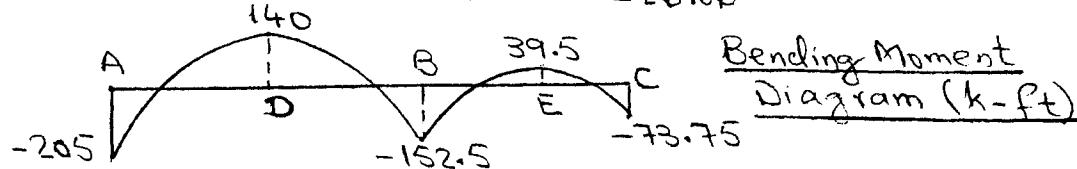
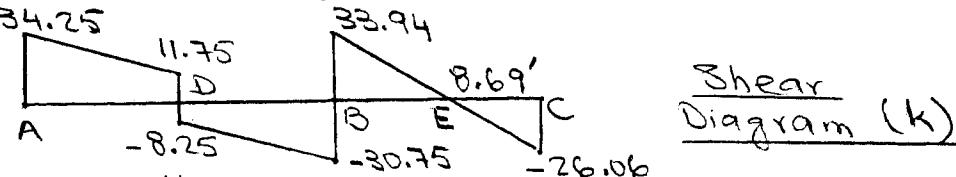
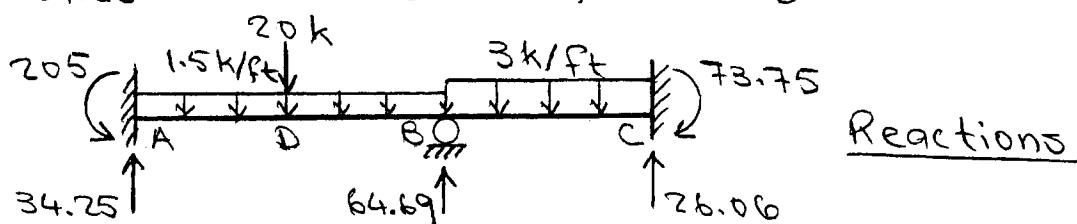
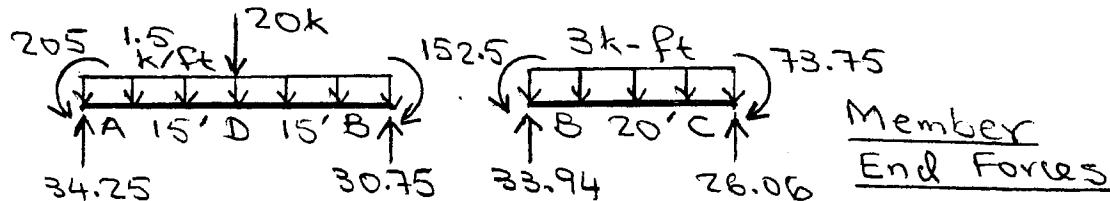
$$(0.133 EI \theta_B - 187.5) + (0.2 EI \theta_B + 100) = 0$$

$$EI \theta_B = 262.5 \text{ k-ft}^2$$

Member end moments: Substituting the numerical value of $EI \theta_B$ into the slope-deflection equations, we obtain:

$$M_{AB} = 205 \text{ k-ft}; M_{BA} = -152.5 \text{ k-ft}$$

$$M_{BC} = 152.5 \text{ k-ft}; M_{CB} = -73.75 \text{ k-ft}$$



16.3 Fixed end moments:

$$FEM_{BE} = +400 \text{ kN.m}; \quad FEM_{EB} = -400 \text{ kN.m}$$

Slope-deflection equations:

$$M_{AB} = 0.222 EI\theta_B; \quad M_{BA} = 0.444 EI\theta_B$$

$$M_{BE} = 0.444 EI\theta_B + 400; \quad M_{EB} = 0.222 EI\theta_B - 400$$

$$\text{Equilibrium equation: } M_{BA} + M_{BE} = 0$$

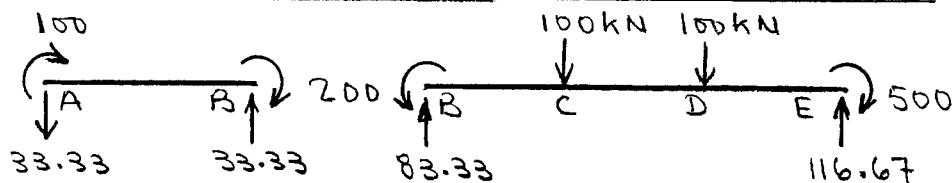
$$0.444 EI\theta_B + (0.444 EI\theta_B + 400) = 0$$

$$EI\theta_B = -450 \text{ kN.m}^2$$

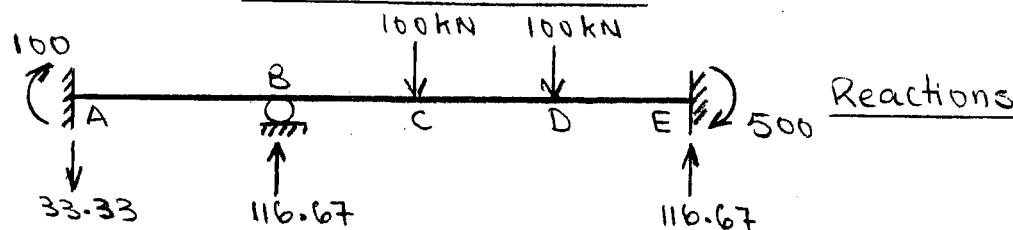
Member end moments: Substituting the numerical value of $EI\theta_c$ into the slope-deflection equations, we obtain

$$\underline{M_{AB} = -100 \text{ kN.m}}; \quad \underline{M_{BA} = -200 \text{ kN.m}}$$

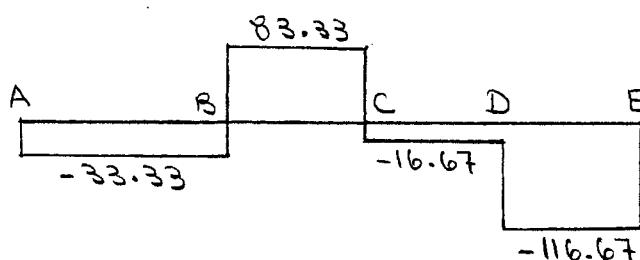
$$\underline{M_{BE} = 200 \text{ kN.m}}; \quad \underline{M_{EB} = -500 \text{ kN.m}}$$



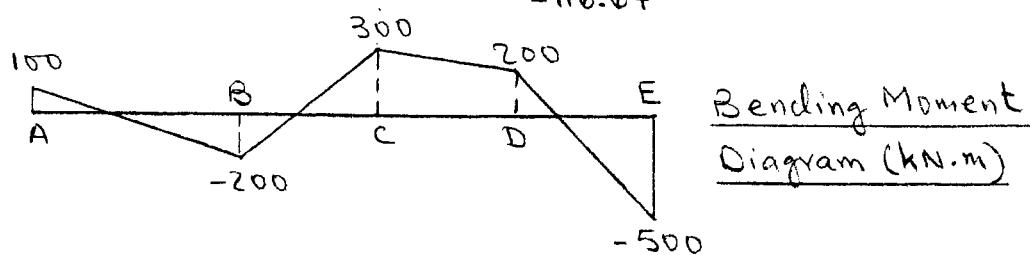
Member End Forces



Reactions



Shear Diagram (kN)



Bending Moment Diagram (kN.m)

16.4 Fixed-end moments:

$$FEM_{AB} = \frac{25(8)^2}{12} = 133.33 \text{ kN.m}; FEM_{BA} = -133.33 \text{ kN.m}$$

$$FEM_{BC} = 133.33 \text{ kN.m}; FEM_{CB} = -133.33 \text{ kN.m}$$

Slope-deflection equations: $M_{CB} = 0$

$$M_{AB} = 0.25 EI\theta_B + 133.33; M_{BA} = 0.5 EI\theta_B - 133.33$$

$$M_{BC} = 0.375 EI\theta_B + 200$$

Equilibrium equation: $M_{BA} + M_{BC} = 0$

$$(0.5 EI\theta_B - 133.33) + (0.375 EI\theta_B + 200) = 0$$

$$EI\theta_B = -76.19 \text{ kN.m}^2$$

Member end moments: Substituting the numerical value of $EI\theta_B$ into the slope-deflection equations, we obtain:

$$\underline{M_{AB} = 114.3 \text{ kN.m}}; \underline{M_{BA} = -171.4 \text{ kN.m}}$$

$$\underline{M_{BC} = 171.4 \text{ kN.m}}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.37.

16.5

Fixed-end moments:

$$FEM_{AB} = \frac{3(25)^2}{12} = 156.25 \text{ k-ft}; \quad FEM_{BA} = -156.25 \text{ k-ft}$$

$$FEM_{BC} = \frac{3(15)^2}{12} = 56.25 \text{ k-ft}; \quad FEM_{CB} = -56.25 \text{ k-ft}$$

Slope-deflection equations: $M_{AB} = M_{CB} = 0$

$$M_{BA} = \frac{3EI(2I)}{25} \theta_B - 156.25 - \frac{156.25}{2} = 0.24EI\theta_B - 234.38$$

$$M_{BC} = \frac{3EI}{15} \theta_B + 56.25 + \frac{56.25}{2} = 0.2EI\theta_B + 84.38$$

Equilibrium equation: $M_{BA} + M_{BC} = 0$

$$(0.24EI\theta_B - 234.38) + (0.2EI\theta_B + 84.38) = 0$$

$$EI\theta_B = 340.91 \text{ k-ft}^2$$

Member end moments. Substituting the numerical value of $EI\theta_B$ into the slope-deflection equations, we obtain

$$M_{BA} = -152.6 \text{ k-ft}; \quad M_{BC} = 152.6 \text{ k-ft}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.12.

16.6 Fixed-end moments;

$$FEM_{AB} = 187.5 \text{ k-ft}; \quad FEM_{BA} = -187.5 \text{ k-ft}$$

$$FEM_{BC} = 100 \text{ k-ft}; \quad FEM_{CB} = -100 \text{ k-ft}$$

Chord rotations: A $\xrightarrow[360'']{\Psi_{AB}} 0.5''$ B $\xrightarrow[240'']{\Psi_{BC}}$ C

$$\Psi_{AB} = -\frac{0.5}{360} = -0.00139$$

$$\Psi_{BC} = +\frac{0.5}{240} = +0.00208$$

Slope-deflection equations:

Using $EI = \frac{29000(1650)}{(12)^2} \text{ k-ft}^2$, we write

$$M_{AB} = 0.0667 EI \theta_B + 92.3 + 187.5 = 0.0667 EI \theta_B + 279.8$$

$$M_{BA} = 0.133 EI \theta_B + 92.3 - 187.5 = 0.133 EI \theta_B - 95.2$$

$$M_{BC} = 0.2 EI \theta_B - 207.7 + 100 = 0.2 EI \theta_B - 107.7$$

$$M_{CB} = 0.1 EI \theta_B - 207.7 - 100 = 0.1 EI \theta_B - 307.7$$

Equilibrium equation: $M_{BA} + M_{BC} = 0$

$$(0.133 EI \theta_B - 95.2) + (0.2 EI \theta_B - 107.7) = 0$$

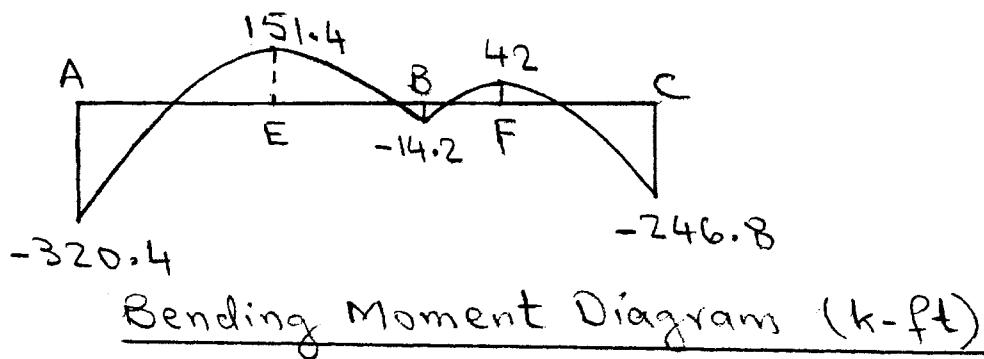
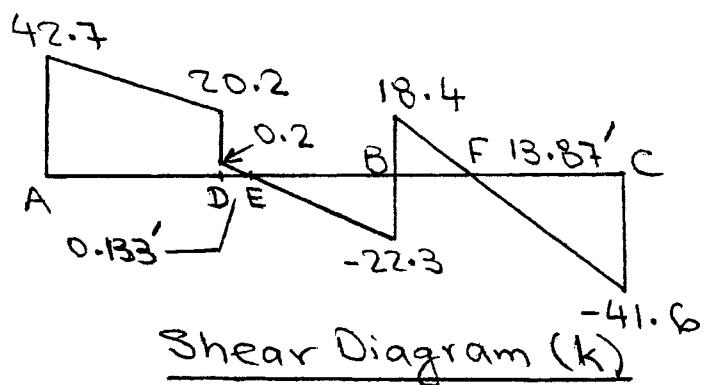
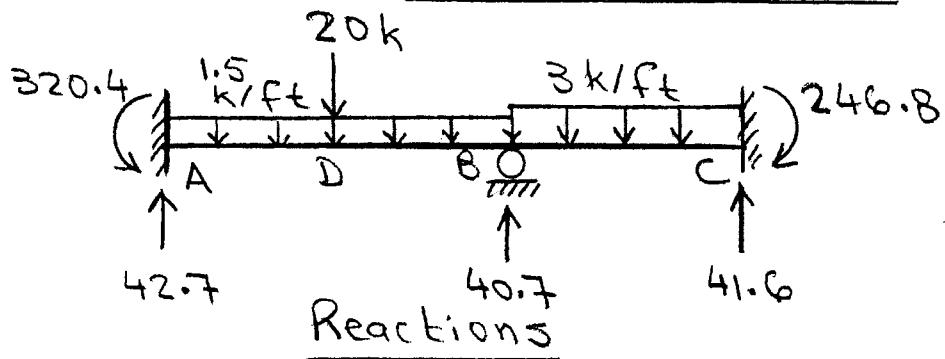
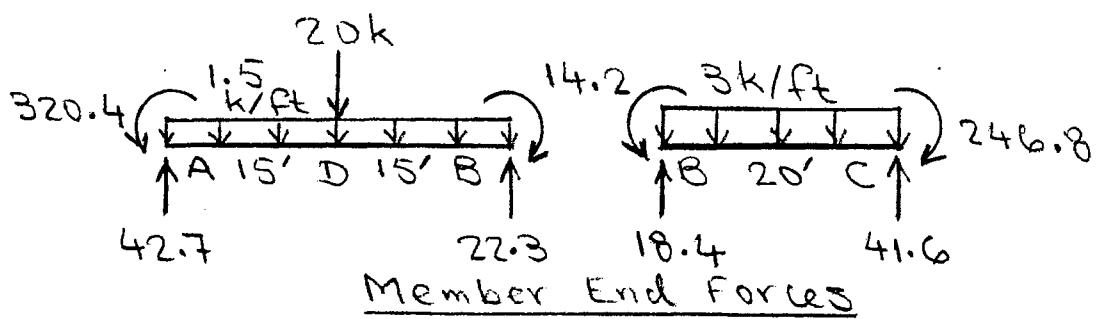
$$EI \theta_B = 609.31 \text{ k-ft}^2$$

Member end moments: Substituting the numerical value of $EI \theta_B$ into the slope-deflection equations, we obtain:

$$M_{AB} = 320.4 \text{ k-ft}; \quad M_{BA} = -14.2 \text{ k-ft}$$

$$M_{BC} = 14.2 \text{ k-ft}; \quad M_{CB} = -246.8 \text{ k-ft}$$

16.6 (contd.)

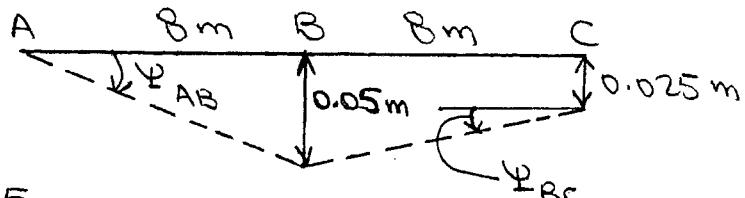


16.7 Fixed-end moments:

$$FEM_{AB} = 133.33 \text{ kN.m}; FEM_{BA} = -133.33 \text{ kN.m}$$

$$FEM_{BC} = 133.33 \text{ kN.m}; FEM_{CB} = -133.33 \text{ kN.m}$$

Chord rotations:



$$\Psi_{AB} = -\frac{0.05}{8} = -0.00625$$

$$\Psi_{BC} = +\frac{0.025}{8} = +0.003125$$

Slope-deflection equations: $M_{CB} = 0$

Using $EI = 70(1300) = 91000 \text{ kN.m}^2$, we write

$$M_{AB} = 0.25EI\theta_B + 426.56 + 133.33 = 0.25EI\theta_B + 559.89$$

$$M_{BA} = 0.5EI\theta_B + 426.56 - 133.33 = 0.5EI\theta_B + 293.23$$

$$\begin{aligned} M_{BC} &= 0.375EI\theta_B - 106.64 + 133.33 + \frac{133.33}{2} \\ &= 0.375EI\theta_B + 93.36 \end{aligned}$$

Equilibrium equation: $M_{BA} + M_{BC} = 0$

$$(0.5EI\theta_B + 293.23) + (0.375EI\theta_B + 93.36) = 0$$

$$EI\theta_B = -441.82 \text{ kN.m}^2$$

Member end moments: Substituting the numerical value of $EI\theta_B$ into the slope-deflection equations, we obtain:

$$\underline{M_{AB} = 449.4 \text{ kN.m}}; \underline{M_{BA} = 72.3 \text{ kN.m}}$$

$$\underline{M_{BC} = -72.3 \text{ kN.m}}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.53.

16.8 Fixed-end moments:

$$FEM_{AB} = \frac{1.5(25)^2}{12} = 78.125 \text{ k-ft}; FEM_{BA} = -78.125 \text{ k-ft}$$

$$FEM_{BC} = \frac{1.5(20)^2}{12} = 50 \text{ k-ft}; FEM_{CB} = -50 \text{ k-ft.}$$

$$FEM_{CD} = 78.125 \text{ k-ft}; FEM_{DC} = -78.125 \text{ k-ft.}$$

Slope-deflection equations:

$$M_{AB} = 0.08EI\theta_B + 78.125; M_{BA} = 0.16EI\theta_B - 78.125$$

$$M_{BC} = 0.2EI\theta_B + 0.1EI\theta_C + 50; M_{CB} = 0.1EI\theta_B + 0.2EI\theta_C - 50$$

$$M_{CD} = 0.16EI\theta_C + 78.125; M_{DC} = 0.08EI\theta_C - 78.125$$

Equilibrium equations: $M_{BA} + M_{BC} = 0$

$$M_{CB} + M_{CD} = 0$$

$$0.36EI\theta_B + 0.1EI\theta_C = 28.125$$

$$0.1EI\theta_B + 0.36EI\theta_C = -28.125$$

By solving these equations, we obtain:

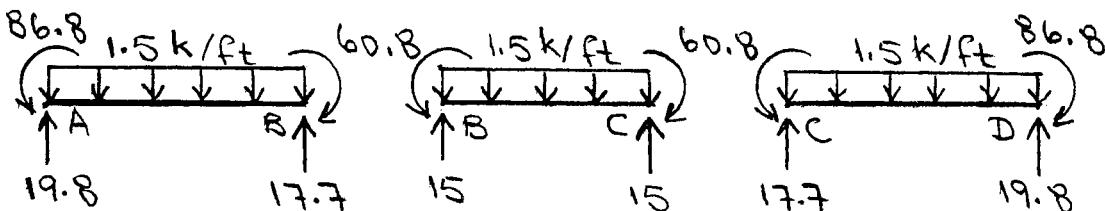
$$EI\theta_B = 108.173 \text{ k-ft}^2; EI\theta_C = -108.173 \text{ k-ft}^2$$

Member end moments: Substituting the numerical values of $EI\theta_B$ and $EI\theta_C$ into the slope-deflection equations, we obtain:

$$\underline{M_{AB} = 86.8 \text{ k-ft}}; \underline{M_{BA} = -60.8 \text{ k-ft}}$$

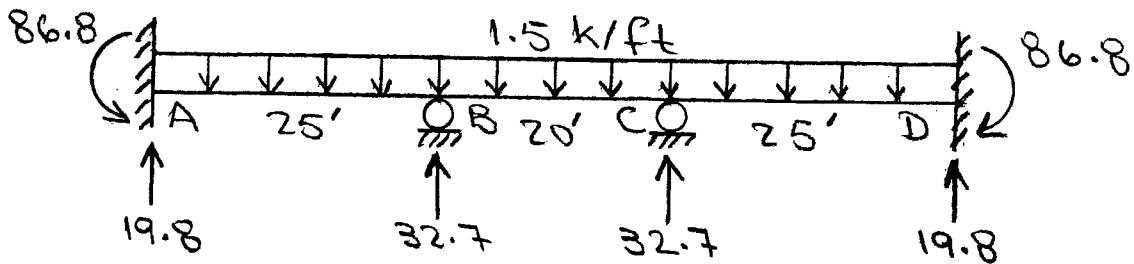
$$\underline{M_{BC} = 60.8 \text{ k-ft}}; \underline{M_{CB} = -60.8 \text{ k-ft}}$$

$$\underline{M_{CD} = 60.8 \text{ k-ft}}; \underline{M_{DC} = -86.8 \text{ k-ft}}$$

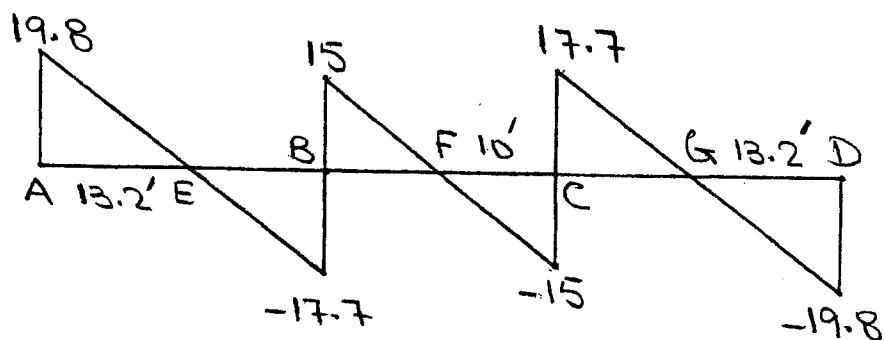


Member End Moments and Shears

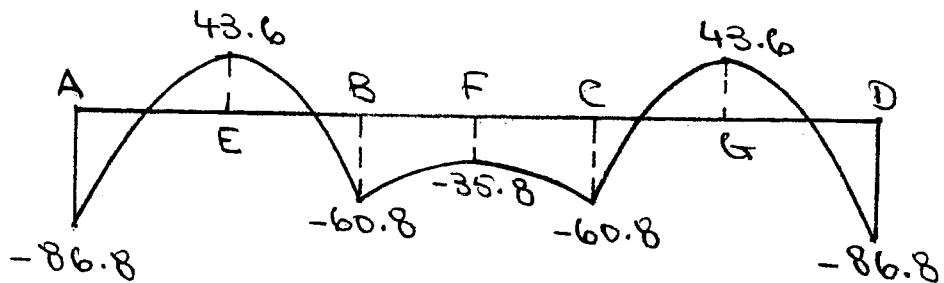
16.8 (Contd.)



Reactions



Shear Diagram (k)



Bending Moment Diagram (k-ft)

16.9 Fixed end moments:

$$FEM_{AB} = \frac{20(8)^2}{12} = 106.7 \text{ kN.m}; FEM_{BA} = -106.7 \text{ kN.m}$$

$$FEM_{BC} = 106.7 \text{ kN.m}; FEM_{CB} = -106.7 \text{ kN.m}$$

$$FEM_{CE} = \frac{60(8)}{8} = 60 \text{ kN.m}; FEM_{EC} = -60 \text{ kN.m}$$

Slope-deflection equations:

$$M_{AB} = 0.25EI\theta_B + 106.7; M_{BA} = 0.5EI\theta_B - 106.7$$

$$M_{BC} = 0.5EI\theta_B + 0.25EI\theta_C + 106.7$$

$$M_{CB} = 0.25EI\theta_B + 0.5EI\theta_C - 106.7$$

$$M_{CE} = 0.5EI\theta_C + 60; M_{EC} = 0.25EI\theta_C - 60$$

Equilibrium equations: $M_{BA} + M_{BC} = 0$

$$M_{CB} + M_{CE} = 0$$

$$EI\theta_B + 0.25EI\theta_C = 0$$

$$0.25EI\theta_B + EI\theta_C = 46.7$$

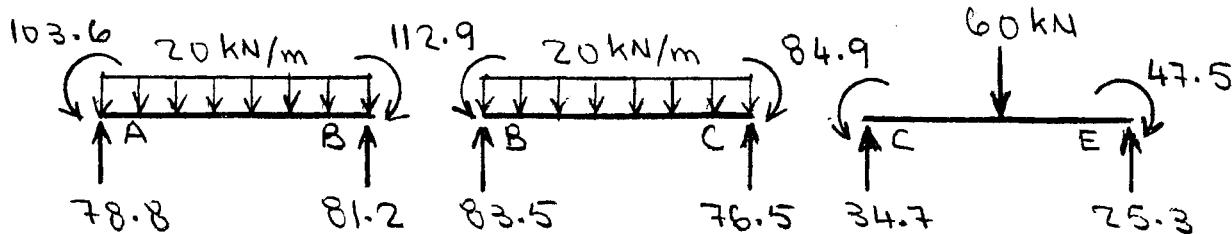
By solving these equations, we obtain

$$EI\theta_B = -12.45 \text{ kN.m}^2; EI\theta_C = 49.81 \text{ kN.m}^2$$

Member end moments: Substituting the numerical values of $EI\theta_B$ and $EI\theta_C$ into the slope-deflection equations, we obtain

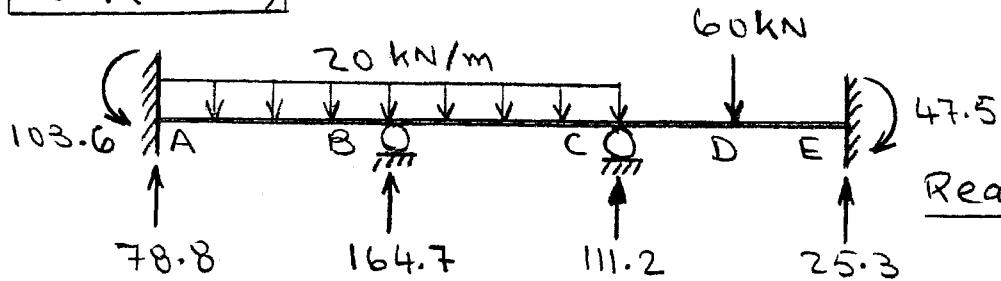
$$M_{AB} = 103.6 \text{ kN.m}; M_{BA} = -112.9 \text{ kN.m}; M_{BC} = 112.9 \text{ kN.m}$$

$$M_{CB} = -84.9 \text{ kN.m}; M_{CE} = 84.9 \text{ kN.m}; M_{EC} = -47.5 \text{ kN.m}$$

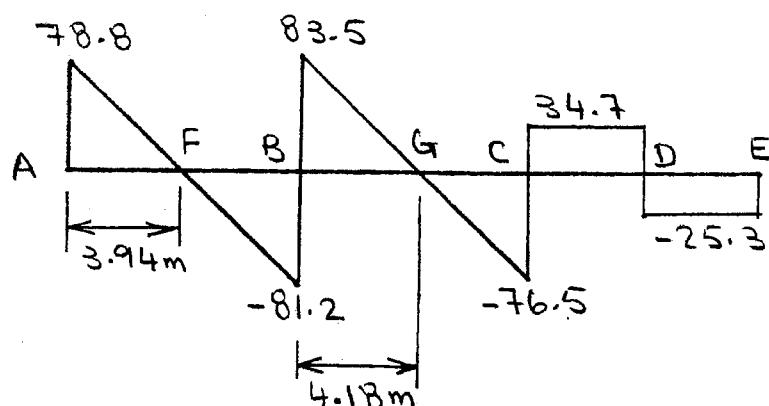


Member End Moments and Shears

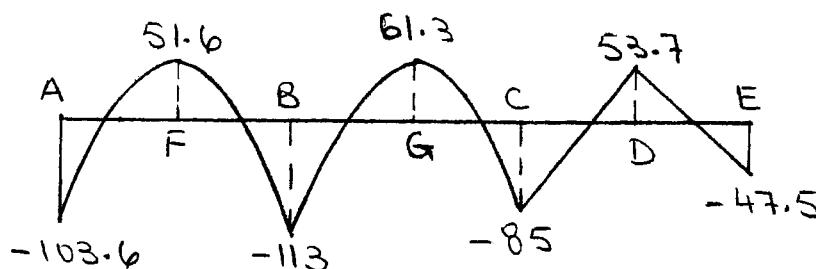
16-9 (contd.)



Reactions



Shear Diagram (kN)



Bending Moment Diagram (kN.m)

16.10 Fixed-end moments:

$$FEM_{AB} = \frac{2(15)^2}{12} = 37.5 \text{ k-ft}; FEM_{BA} = -37.5 \text{ k-ft}$$

$$FEM_{BC} = \frac{3(10)^2}{30} = 10 \text{ k-ft}; FEM_{CB} = -\frac{3(10)^2}{20} = -15 \text{ k-ft}$$

$$FEM_{CD} = 15 \text{ k-ft}; FEM_{DC} = -10 \text{ k-ft}$$

Slope-deflection equations:

$$M_{AB} = 0.133 EI \theta_B + 37.5; M_{BA} = 0.267 EI \theta_B - 37.5$$

$$M_{BC} = 0.4 EI \theta_B + 0.2 EI \theta_C + 10$$

$$M_{CB} = 0.2 EI \theta_B + 0.4 EI \theta_C - 15$$

$$M_{CD} = 0.4 EI \theta_C + 15; M_{DC} = 0.2 EI \theta_C - 10$$

Equilibrium equations: $M_{BA} + M_{BC} = 0$

$$M_{CB} + M_{CD} = 0$$

$$0.667 EI \theta_B + 0.2 EI \theta_C = 27.5$$

$$0.2 EI \theta_B + 0.8 EI \theta_C = 0$$

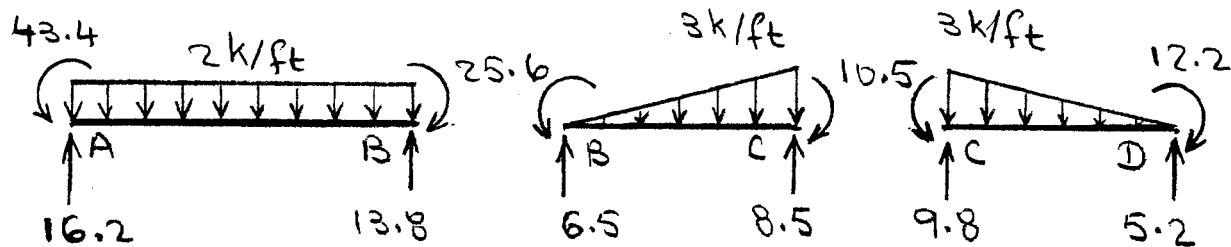
By solving these equations, we obtain

$$EI \theta_B = 44.59 \text{ k-ft}^2; EI \theta_C = -11.15 \text{ k-ft}^2$$

Member end moments: Substituting the numerical values of $EI \theta_B$ and $EI \theta_C$ into the slope-deflection equations, we obtain

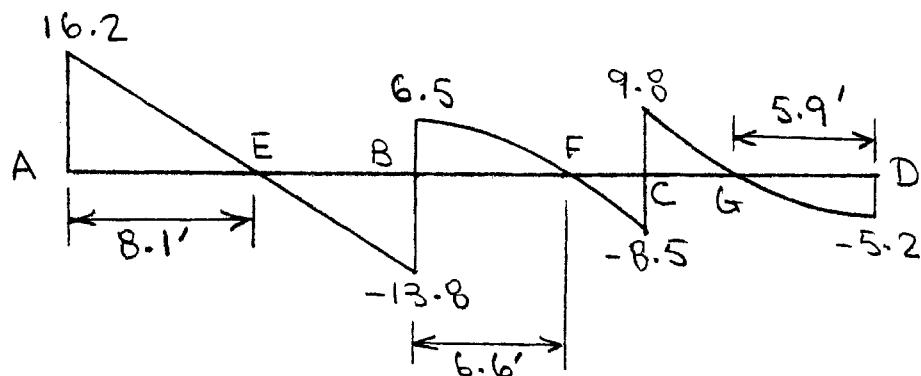
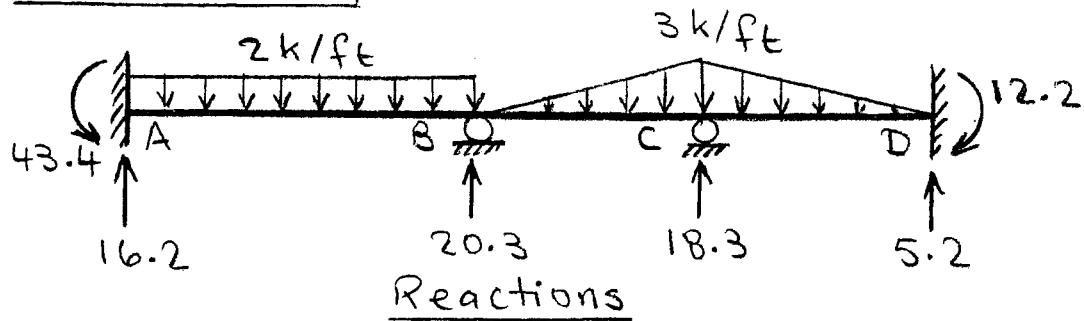
$$M_{AB} = 43.4 \text{ k-ft}; M_{BA} = -25.6 \text{ k-ft}; M_{BC} = 25.6 \text{ k-ft}$$

$$M_{CB} = -10.5 \text{ k-ft}; M_{CD} = 10.5 \text{ k-ft}; M_{DC} = -12.2 \text{ k-ft}$$

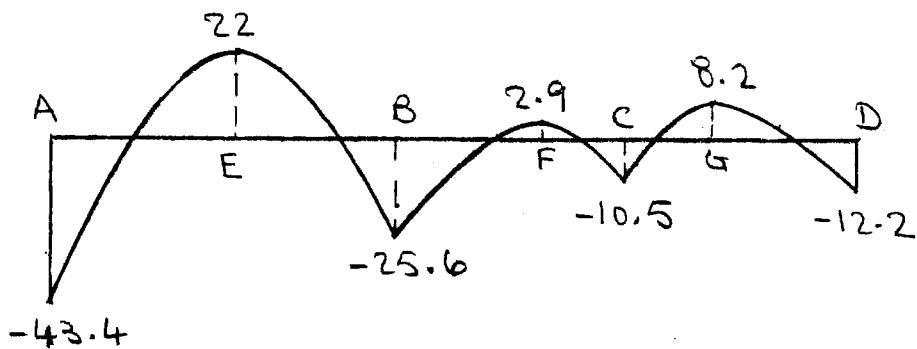


Member End Moments and Shears

16.10 (contd.)

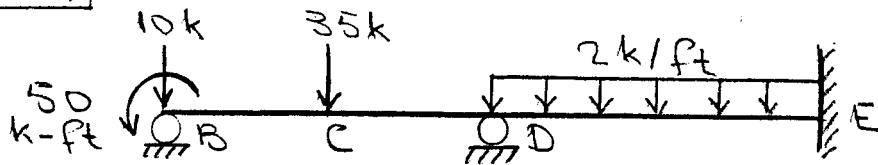


Shear Diagram (k)



Bending Moment Diagram (k-ft)

16.11

Fixed-end moments:

$$FEM_{BD} = \frac{35(20)}{8} = 87.5 \text{ k-ft}; \quad FEM_{DB} = -87.5 \text{ k-ft}$$

$$FEM_{DE} = \frac{2(20)^2}{12} = 66.67 \text{ k-ft}; \quad FEM_{ED} = -66.67 \text{ k-ft}$$

Slope-deflection equations:

$$M_{BD} = 0.2EI\theta_B + 0.1EI\theta_D + 87.5$$

$$M_{DB} = 0.1EI\theta_B + 0.2EI\theta_D - 87.5$$

$$M_{DE} = 0.2EI\theta_D + 66.67; \quad M_{ED} = 0.1EI\theta_D - 66.67$$

$$\underline{\text{Equilibrium equations:}} \quad M_{BD} - 50 = 0$$

$$M_{DB} + M_{DE} = 0$$

$$0.2EI\theta_B + 0.1EI\theta_D = -37.5$$

$$0.1EI\theta_B + 0.4EI\theta_D = 20.83$$

By solving these equations, we obtain:

$$EI\theta_B = -244.05 \text{ k-ft}^2; \quad EI\theta_D = 113.1 \text{ k-ft}^2$$

Member end moments: Substituting the numerical values of $EI\theta_B$ and $EI\theta_D$ into the slope-deflection equations, we obtain:

$$\underline{M_{BD} = 50 \text{ k-ft}; \quad M_{DB} = -89.3 \text{ k-ft}}$$

$$\underline{M_{DE} = 89.3 \text{ k-ft}; \quad M_{ED} = -55.4 \text{ k-ft}}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 18.38.

16.12 Fixed-end moments:

$$FEM_{AC} = FEM_{CE} = \frac{120(6)(4)^2}{(10)^2} = 115.2 \text{ kN.m};$$

$$FEM_{CA} = FEM_{EC} = -\frac{120(4)(6)^2}{(10)^2} = -172.8 \text{ kN.m}$$

$$FEM_{EG} = \frac{150(8)}{8} = 150 \text{ kN.m}; \quad FEM_{GE} = -150 \text{ kN.m}$$

Slope-deflection equations: $M_{AC} = M_{GE} = 0$

$$M_{CA} = \frac{3EI}{10}\theta_C - 172.8 - \frac{115.2}{2} = 0.3EI\theta_C - 230.4$$

$$M_{CE} = \frac{2EI(2L)}{10}(2\theta_C + \theta_E) + 115.2 = 0.4EI(2\theta_C + \theta_E) + 115.2$$

$$M_{EC} = 0.4EI(\theta_C + 2\theta_E) - 172.8$$

$$M_{EG} = \frac{3EI}{8}\theta_E + 150 + \frac{150}{2} = 0.375EI\theta_E + 225$$

Equilibrium equations: $M_{CA} + M_{CE} = 0$

$$M_{EC} + M_{EG} = 0$$

$$1.1EI\theta_C + 0.4EI\theta_E = 115.2$$

$$0.4EI\theta_C + 1.175EI\theta_E = -52.2$$

By solving these equations, we obtain

$$EI\theta_C = 137.96 \text{ kN.m}^2; \quad EI\theta_E = -91.39 \text{ kN.m}^2$$

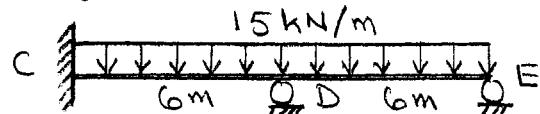
Member end moments. Substituting the numerical values of $EI\theta_C$ and $EI\theta_E$ into the slope-deflection equations, we obtain

$$\underline{M_{CA} = -189 \text{ kN.m}}; \quad \underline{M_{CE} = 189 \text{ kN.m}}$$

$$\underline{M_{EC} = -190.7 \text{ kN.m}}; \quad \underline{M_{EG} = 190.7 \text{ kN.m}}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 13.39.

16.13 As the beam and the loading are symmetric, we will analyze only the right half CE of the beam.



Fixed-end moments: $FEM_{CD} = 45 \text{ kN.m}$; $FEM_{DC} = -45 \text{ kN.m}$

$FEM_{DE} = 45 \text{ kN.m}$; $FEM_{ED} = -45 \text{ kN.m}$

Slope-deflection equations: $M_{ED} = 0$

$$M_{CD} = 0.333 EI\theta_D + 45; M_{DC} = 0.667 EI\theta_D - 45$$

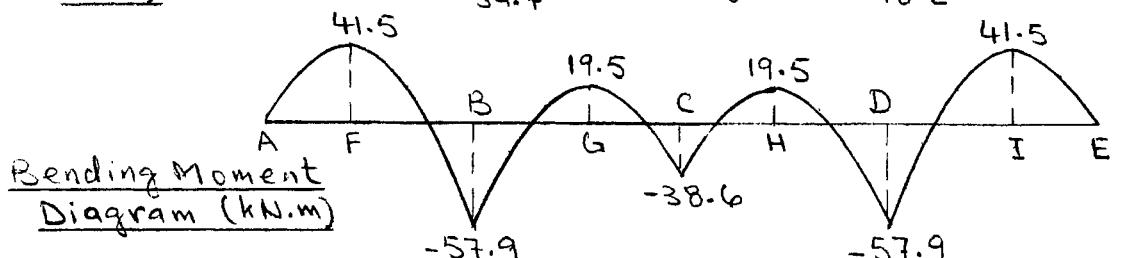
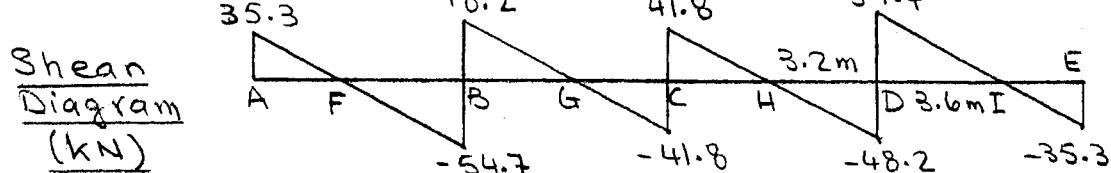
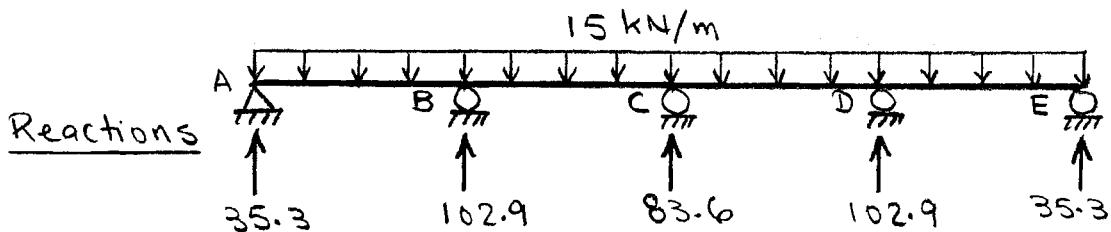
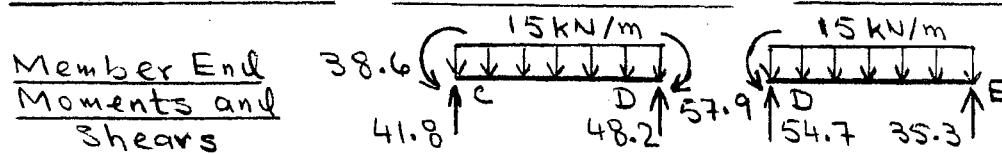
$$M_{DE} = \frac{3EI}{6}(\theta_D) + 45 + \frac{45}{2} = 0.5EI\theta_D + 67.5$$

Equilibrium equations: $M_{DC} + M_{DE} = 0$

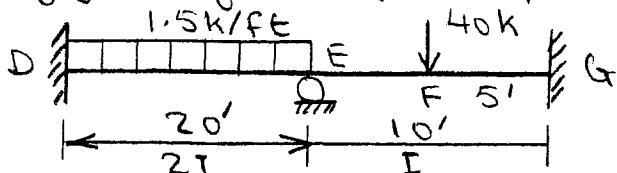
$$1.167 EI\theta_D + 22.5 = 0; EI\theta_D = -19.28 \text{ kN.m}^2$$

Member end moments: Substituting the numerical value of $EI\theta_D$ into the slope-deflection equations, we obtain

$$M_{CD} = 38.6 \text{ kN.m}; M_{DC} = -57.9 \text{ kN.m}; M_{DE} = 57.9 \text{ kN.m}$$



16-14 As the beam and loading are symmetric, we will analyze only the right half DG of the beam.



Fixed end moments: $FEM_{DE} = 50 \text{ k-ft}$, $FEM_{ED} = -50 \text{ k-ft}$

$FEM_{EG} = 100 \text{ k-ft}$; $FEM_{GE} = -100 \text{ k-ft}$

Slope deflection equations:

$$M_{DE} = 0.2EI\theta_E + 50; M_{ED} = 0.4EI\theta_E - 50$$

$$M_{EG} = 0.2EI\theta_E + 100; M_{GE} = 0.1EI\theta_E - 100$$

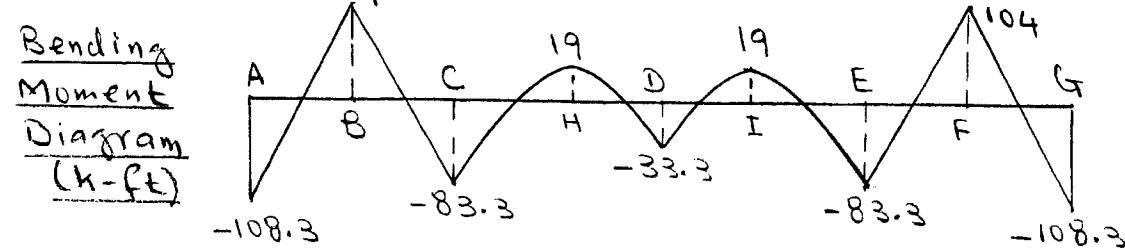
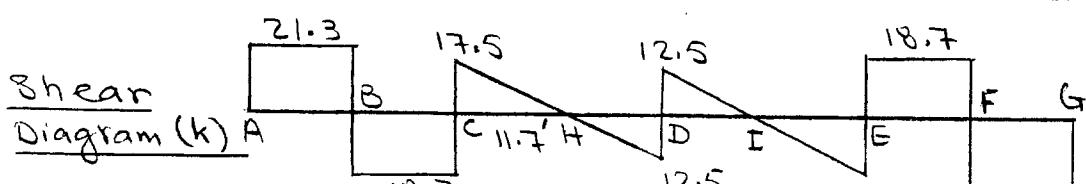
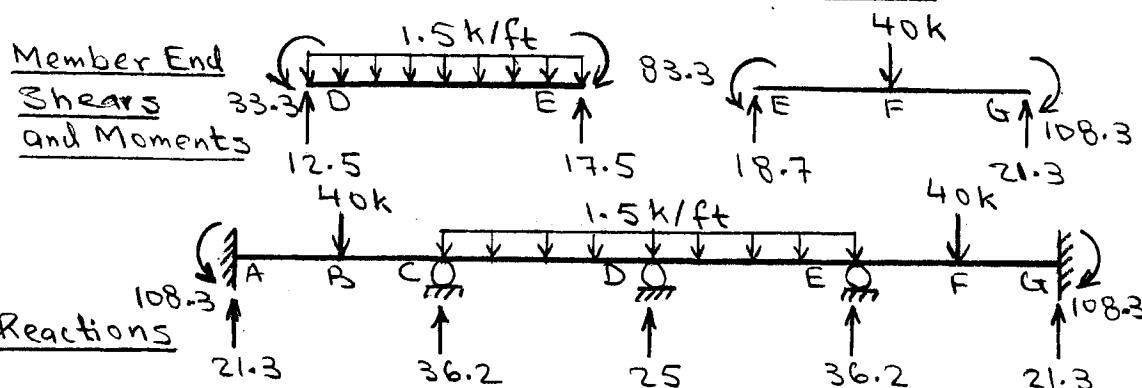
Equilibrium equation: $M_{ED} + M_{EG} = 0$

$$0.6EI\theta_E + 50 = 0 \quad EI\theta_E = -83.33 \text{ k-ft}^2$$

Member end moments: Substituting the numerical value of $EI\theta_E$ into the slope-deflection equations, we obtain

$$M_{DE} = 33.3 \text{ k-ft}; M_{ED} = -83.3 \text{ k-ft}$$

$$M_{EG} = 83.3 \text{ k-ft}; M_{GE} = -108.3 \text{ k-ft}$$



16.15 Fixed end moments:

$$FEM_{AB} = 106.7 \text{ kN.m};$$

$$FEM_{BA} = -106.7 \text{ kN.m}$$

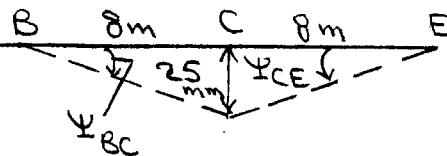
$$FEM_{BC} = 106.7 \text{ kN.m};$$

$$FEM_{CB} = -106.7 \text{ kN.m}$$

$$FEM_{CE} = 60 \text{ kN.m};$$

$$FEM_{EC} = -60 \text{ kN.m}$$

Chord rotations: A



$$\Psi_{BC} = -\frac{0.025}{8} = -0.003125$$

$$\Psi_{CE} = +\frac{0.025}{8} = +0.003125$$

Slope-deflection equations:

$$M_{AB} = 0.25EI\theta_B + 106.7; M_{BA} = 0.5EI\theta_B - 106.7$$

$$M_{BC} = 0.5EI\theta_B + 0.25EI\theta_C + 23.8$$

$$M_{CB} = 0.25EI\theta_B + 0.5EI\theta_C + 24.6$$

$$M_{CE} = 0.5EI\theta_C - 71.3; M_{EC} = 0.25EI\theta_C - 191.3$$

Equilibrium equations: $M_{BA} + M_{BC} = 0$

$$M_{CB} + M_{CE} = 0$$

$$EI\theta_B + 0.25EI\theta_C = -131.3$$

$$0.25EI\theta_B + EI\theta_C = 46.7$$

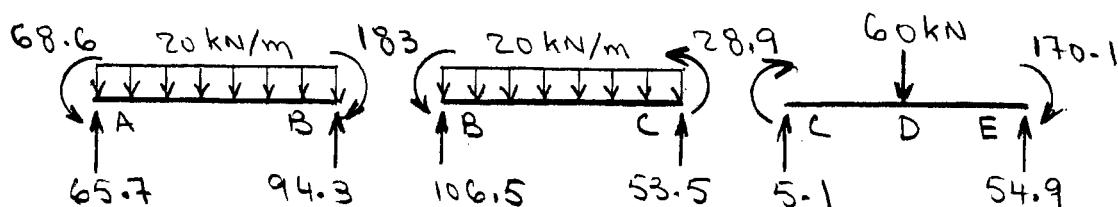
By solving these equations, we obtain

$$EI\theta_B = -152.51 \text{ kN.m}^2; EI\theta_C = 84.83 \text{ kN.m}^2$$

Member end moments: Substituting the numerical values of $EI\theta_B$ and $EI\theta_C$ into the slope-deflection equations, we obtain

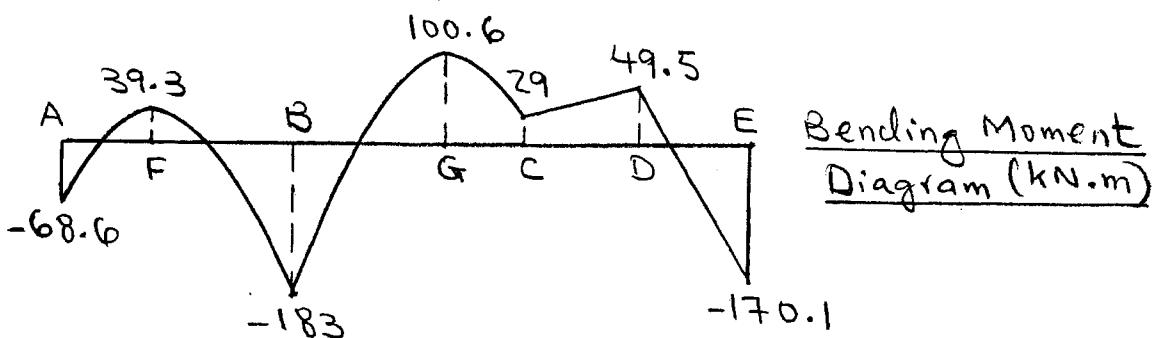
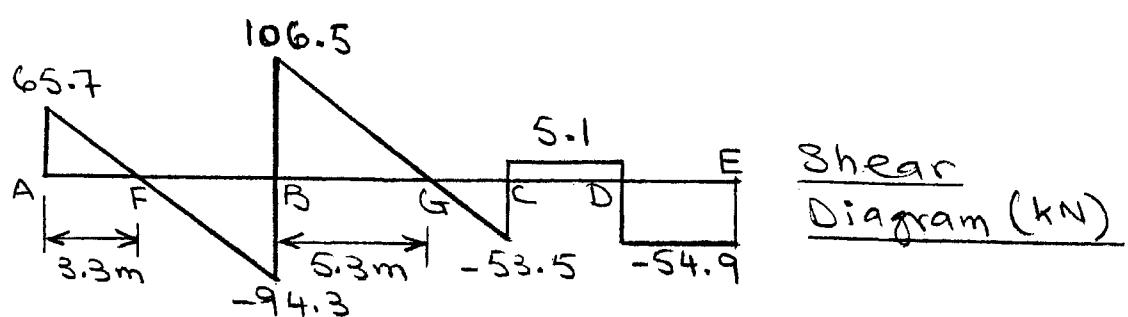
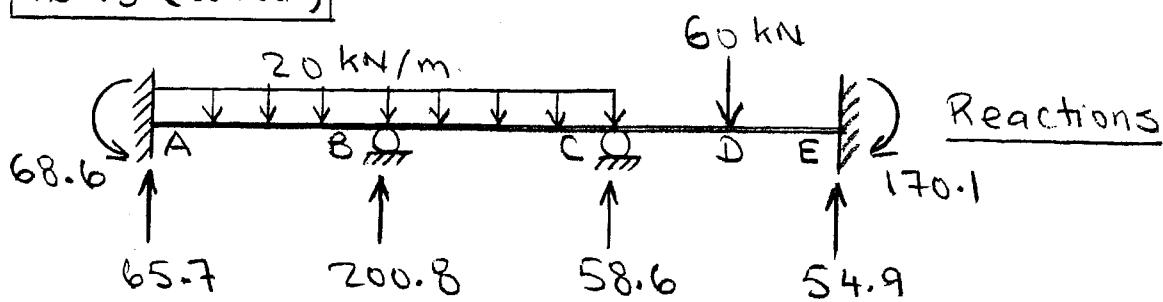
$$M_{AB} = 68.6 \text{ kN.m}; M_{BA} = -183 \text{ kN.m}; M_{BC} = 183 \text{ kN.m}$$

$$M_{CB} = 28.9 \text{ kN.m}; M_{CE} = -28.9 \text{ kN.m}; M_{EC} = -170.1 \text{ kN.m}$$



Member End Moments and Shears

16.15 (contd.)



Shear Diagram (kN)

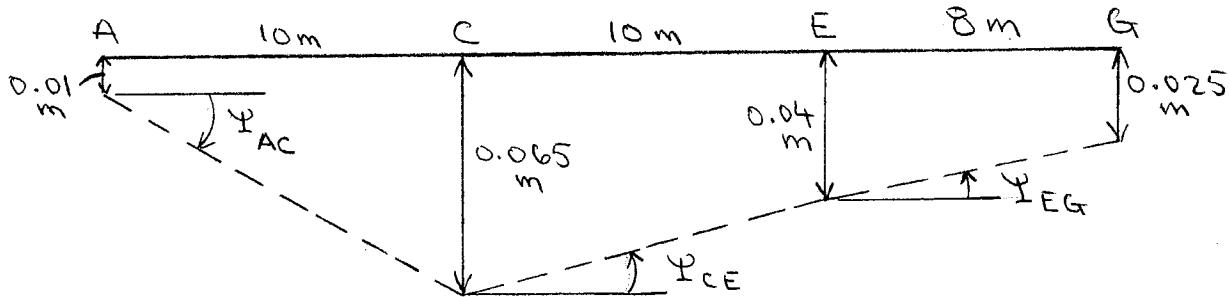
Bending Moment Diagram (kN.m)

16.16 Fixed-end moments: $FEM_{AC} = FEM_{CE} = 115.2 \text{ kN}\cdot\text{m}$

$FEM_{CA} = FEM_{EC} = -172.8 \text{ kN}\cdot\text{m}; FEM_{EG} = 150 \text{ kN}\cdot\text{m}$

$FEM_{GE} = -150 \text{ kN}\cdot\text{m}$

Chord rotations:



$$\Psi_{AC} = -\frac{0.055}{10} = -0.0055; \quad \Psi_{CE} = \frac{0.025}{10} = 0.0025$$

$$\Psi_{EG} = \frac{0.015}{8} = 0.001875$$

Slope-deflection equations: $M_{AC} = M_{GE} = 0$

Using $EI = 200(500) = 100000 \text{ kN}\cdot\text{m}^2$, we write:

$$M_{CA} = 0.3EI\theta_C + 165 - 230.4 = 0.3EI\theta_C - 65.4$$

$$M_{CE} = 0.4EI(2\theta_C + \theta_E) - 300 + 115.2 = 0.4EI(2\theta_C + \theta_E) - 184.8$$

$$M_{EC} = 0.4EI(\theta_C + 2\theta_E) - 300 - 172.8 = 0.4EI(\theta_C + 2\theta_E) - 472.8$$

$$M_{EG} = 0.375EI\theta_E - 70.31 + 225 = 0.375EI\theta_E + 154.69$$

Equilibrium equations:

$$M_{CA} + M_{CE} = 0 \Rightarrow 1.1EI\theta_C + 0.4EI\theta_E = 250.2$$

$$M_{EC} + M_{EG} = 0 \Rightarrow 0.4EI\theta_C + 1.175EI\theta_E = 318.11$$

By solving these equations, we obtain

$$EI\theta_C = 147.23 \text{ kN}\cdot\text{m}^2; \quad EI\theta_E = 220.61 \text{ kN}\cdot\text{m}^2$$

Member end moments:

$$\underline{M_{CA} = -21.2 \text{ kN}\cdot\text{m}; \quad M_{CE} = 21.2 \text{ kN}\cdot\text{m}}$$

$$\underline{M_{EC} = -237.4 \text{ kN}\cdot\text{m}; \quad M_{EG} = 237.4 \text{ kN}\cdot\text{m}}$$

For reactions, and shear and bending moment

diagrams, see solution of Problem 13.54.

16.17 Fixed-end moments:

$$FEM_{AC} = \frac{75(6)}{8} = 56.25 \text{ kN.m}; FEM_{CA} = -56.25 \text{ kN.m}$$

$$FEM_{CD} = \frac{25(9)^2}{12} = 168.75 \text{ kN.m}; FEM_{DC} = -168.75 \text{ kN.m}$$

Slope-deflection equations: $M_{DC} = 0$

$$M_{AC} = 0.333 EI\theta_C + 56.25; FEM_{CA} = 0.667 EI\theta_C - 56.25$$

$$M_{CD} = \frac{3EI}{9}(\theta_C) + 168.75 + \frac{168.75}{2} = 0.333 EI\theta_C + 253.13$$

Equilibrium equation: $M_{CA} + M_{CD} = 0$

$$EI\theta_C = -196.88 \text{ kN.m}^2$$

Member end moments. Substituting the numerical value of $EI\theta_C$ into the slope-deflection equations, we obtain:

$$\underline{M_{AC} = -9.4 \text{ kN.m}}; \underline{M_{CA} = -187.5 \text{ kN.m}}$$

$$\underline{M_{CD} = 187.5 \text{ kN.m}}$$

For reactions, see solution of Problem 13.42.

16-18 Fixed-end moments:

$$FEM_{AC} = FEM_{CA} = FEM_{CD} = FEM_{DC} = 0$$

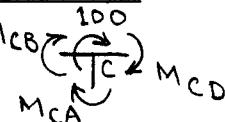
$$FEM_{BC} = \frac{2(15)^2}{12} = 37.5 \text{ k-ft}; FEM_{CB} = -37.5 \text{ k-ft}$$

Slope-deflection equations:

$$M_{AC} = 0.133EI\theta_C; M_{CA} = 0.267EI\theta_C$$

$$M_{BC} = 0.133EI\theta_C + 37.5; M_{CB} = 0.267EI\theta_C - 37.5$$

$$M_{CD} = 0.2EI\theta_C; M_{DC} = 0$$

Equilibrium equation: 

$$M_{CA} + M_{CB} + M_{CD} + 100 = 0$$

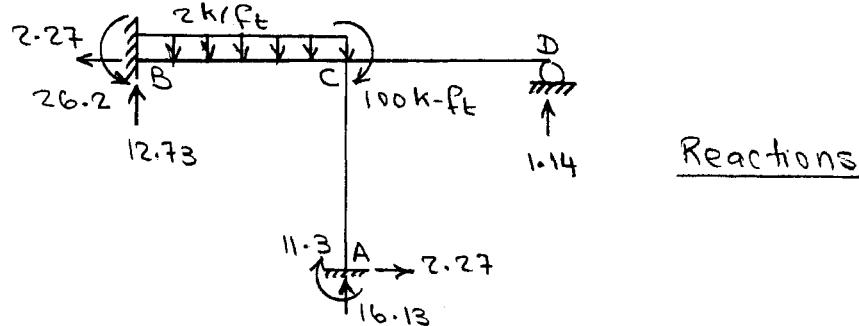
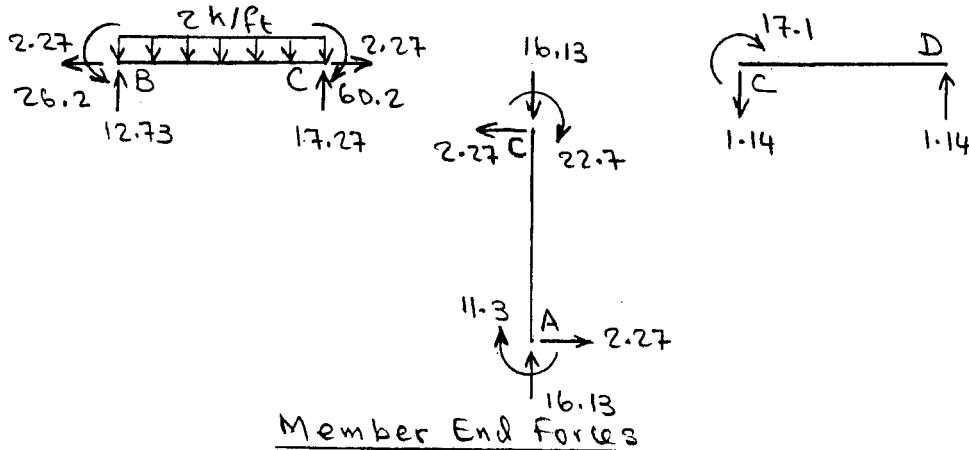
$$(0.267EI\theta_C) + (0.267EI\theta_C - 37.5) + (0.2EI\theta_C) + 100 = 0$$

$$0.734EI\theta_C = -62.5 \quad EI\theta_C = -85.15 \text{ k-ft}^2$$

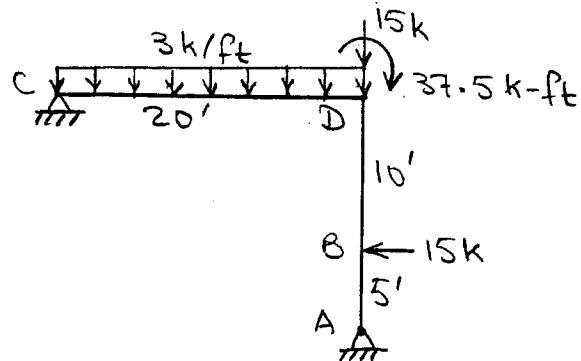
Member end moments: $M_{AC} = -11.3 \text{ k-ft};$

$$M_{CA} = -22.7 \text{ k-ft}; M_{BC} = 26.2 \text{ k-ft}; M_{CB} = -60.2 \text{ k-ft};$$

$$M_{CD} = -17.1 \text{ k-ft}.$$



16.19

Fixed-end moments:

$$FEM_{AD} = -\frac{15(5)(10)^2}{(15)^2} = -33.33 \text{ k-ft}$$

$$FEM_{DA} = \frac{15(10)(5)^2}{(15)^2} = 16.67 \text{ k-ft}$$

$$FEM_{CD} = \frac{3(20)^2}{12} = 100 \text{ k-ft}; \quad FEM_{DC} = -100 \text{ k-ft.}$$

Slope-deflection equations: $M_{AD} = M_{CD} = 0$

$$M_{DA} = \frac{3EI}{15}(\theta_D) + 16.67 + \frac{-33.33}{2} = 0.2EI\theta_D + 33.34$$

$$M_{DC} = \frac{3EI}{20}(\theta_D) - 100 - \frac{100}{2} = 0.3EI\theta_D - 150$$

Equilibrium equation: $M_{DA} + M_{DC} + 37.5 = 0$

$$(0.2EI\theta_D + 33.34) + (0.3EI\theta_D - 150) + 37.5 = 0$$

$$0.5EI\theta_D = 79.16; \quad EI\theta_D = 158.32 \text{ k-ft}^2$$

Member end moments:

$$M_{DA} = 65 \text{ k-ft}; \quad M_{DC} = -102.5 \text{ k-ft}$$

For reactions, see solution of Problem 13.21.

16.20 Fixed-end moments: $FEM_{AC} = FEM_{CA} = 0$

$$FEM_{BD} = FEM_{DB} = 0$$

$$FEM_{CD} = \frac{30(10)^2}{12} = 250 \text{ kN.m}; FEM_{DC} = -250 \text{ kN.m}$$

Slope-deflection equations:

$$M_{AC} = 0.25 EI \theta_C; M_{CA} = 0.5 EI \theta_C$$

$$M_{BD} = 0.25 EI \theta_D; M_{DB} = 0.5 EI \theta_D$$

$$M_{CD} = 0.4 EI \theta_C + 0.2 EI \theta_D + 250$$

$$M_{DC} = 0.2 EI \theta_C + 0.4 EI \theta_D - 250$$

Equilibrium equations: $M_{CA} + M_{CD} = 0$

$$M_{DB} + M_{DC} = 0$$

$$0.9 EI \theta_C + 0.2 EI \theta_D = -250$$

$$0.2 EI \theta_C + 0.9 EI \theta_D = 250$$

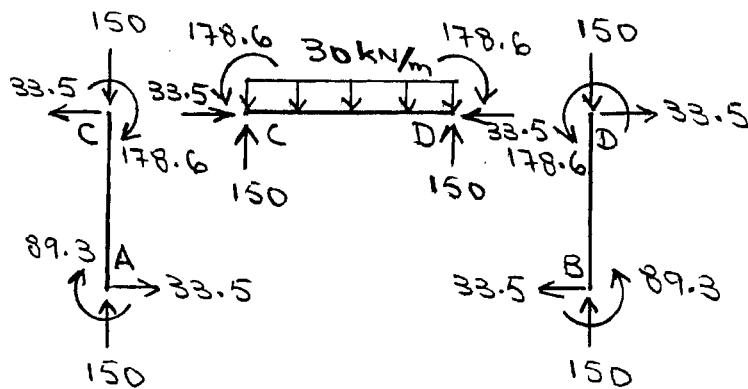
By solving these equations, we obtain:

$$EI \theta_C = -357.14 \text{ kN.m}^2; EI \theta_D = 357.14 \text{ kN.m}^2$$

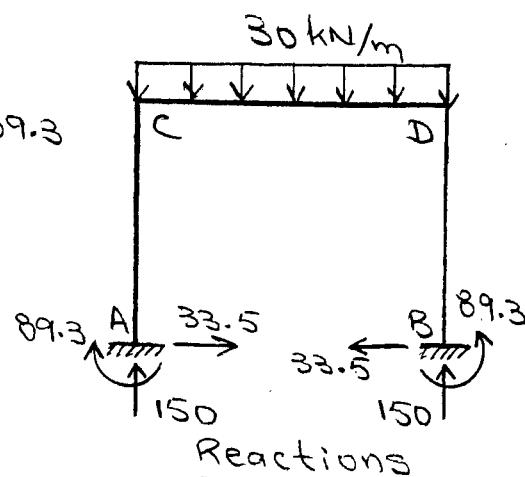
Member end moments: $M_{AC} = -89.3 \text{ kN.m};$

$$M_{CA} = -178.6 \text{ kN.m}; M_{BD} = 89.3 \text{ kN.m}; M_{DB} = 178.6 \text{ kN.m};$$

$$M_{CD} = 178.6 \text{ kN.m}; M_{DC} = -178.6 \text{ kN.m}.$$



Member End Forces

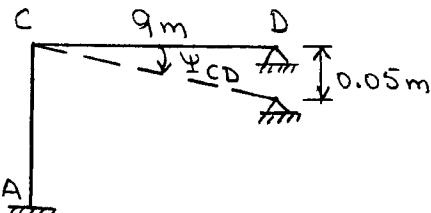


Reactions

16.21 Fixed-end moments:

$$FEM_{AC} = 56.25 \text{ kN.m}; FEM_{CA} = -56.25 \text{ kN.m}$$

$$FEM_{CD} = 168.75 \text{ kN.m}; FEM_{DC} = -168.75 \text{ kN.m}$$



Chord rotation:

$$\Psi_{CD} = -\frac{0.05}{9} = -0.00556$$

Slope-deflection equations: $M_{DC} = 0$

$$M_{AC} = 0.333 EI \theta_C + 56.25; M_{CA} = 0.667 EI \theta_C - 56.25$$

Using $EI = 200(400) \text{ kN.m}^2$, we write

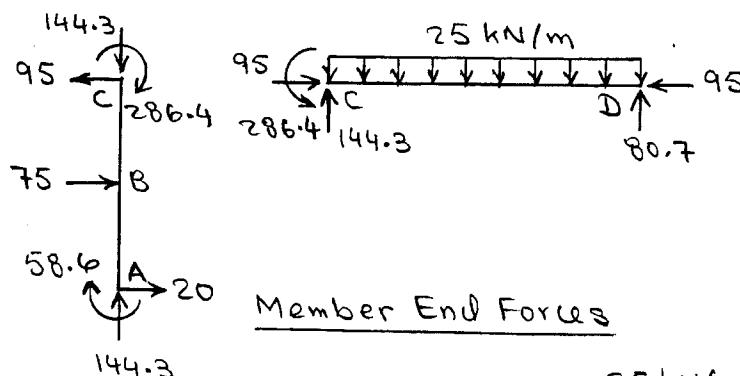
$$M_{CD} = 0.333 EI \theta_C + 148.27 + \frac{168.75}{2} \\ = 0.333 EI \theta_C + 401.4$$

Equilibrium equation: $M_{CA} + M_{CD} = 0$

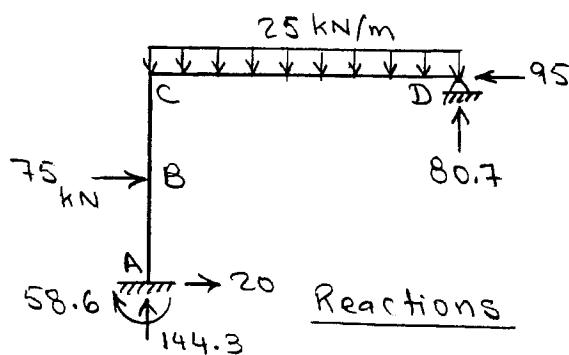
$$EI \theta_C = -345.15 \text{ kN.m}^2$$

Member end moments: $M_{AC} = -58.6 \text{ kN.m}$

$$M_{CA} = -286.4 \text{ kN.m}; M_{CD} = 286.4 \text{ kN.m}$$



Member End Forces



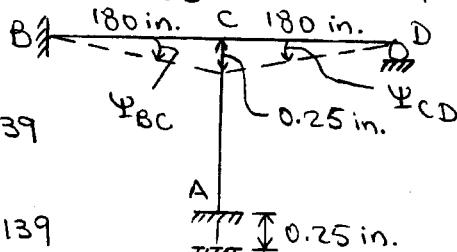
Reactions

16.22 Fixed-end moments:

$$FEM_{AC} = FEM_{CA} = FEM_{CD} = FEM_{DC} = 0$$

$$FEM_{BC} = 37.5 \text{ k-ft}; \quad FEM_{CB} = -37.5 \text{ k-ft}.$$

Chord rotations..



$$\Psi_{BC} = -\frac{0.25}{180} = -0.00139$$

$$\Psi_{CD} = +\frac{0.25}{180} = +0.00139$$

Slope-deflection equations. $M_{DC} = 0$

Using $EI = \frac{29000(3500)}{(12)^2}$ k-ft², we write

$$M_{AC} = 0.133 EI \theta_c; \quad M_{CA} = 0.267 EI \theta_c$$

$$M_{BC} = 0.133 EI \theta_c + 391.9 + 37.5 = 0.133 EI \theta_c + 429.4$$

$$M_{CB} = 0.267 EI \theta_c + 391.9 - 37.5 = 0.267 EI \theta_c + 354.4$$

$$M_{CD} = 0.2 EI \theta_c - 196$$

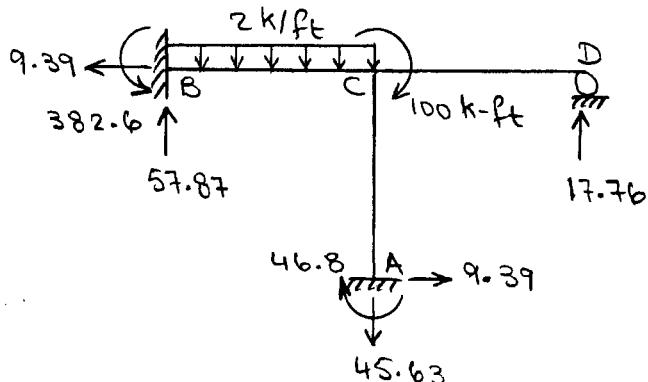
Equilibrium equation: $M_{CA} + M_{CB} + M_{CD} + 100 = 0$

$$0.734 EI \theta_c = -258.4 \quad EI \theta_c = -352 \text{ k-ft}^2$$

Member end moments: $M_{AC} = -46.8 \text{ k-ft};$

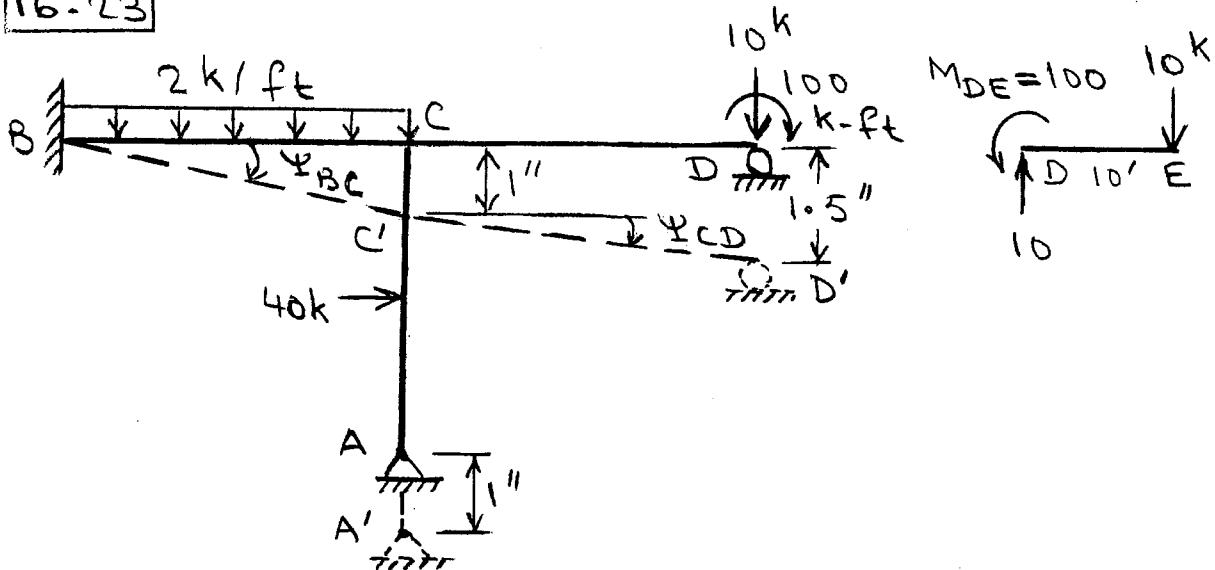
$$M_{CA} = -94 \text{ k-ft}; \quad M_{BC} = 382.6 \text{ k-ft};$$

$$M_{CB} = 260.4 \text{ k-ft}; \quad M_{CD} = -266.4 \text{ k-ft}.$$



Reactions

16-23



Fixed-end moments: $FEM_{CD} = FEM_{DC} = 0$

$$FEM_{BC} = 150 \text{ k-ft}; \quad FEM_{CB} = -150 \text{ k-ft}$$

$$FEM_{AC} = 100 \text{ k-ft}; \quad FEM_{CA} = -100 \text{ k-ft}$$

Chord rotations:

$$\Psi_{BC} = -\frac{1}{30(12)} = -0.00728; \quad \Psi_{CD} = -\frac{0.5}{30(12)} = -0.00139$$

Slope-deflection equations: $M_{AC} = 0; \quad M_{DE} = 100 \text{ k-ft}$

$$M_{CA} = 0.15EI\theta_C - 150; \quad M_{BC} = 0.0667EI\theta_C + 265.7$$

$$M_{CB} = 0.133EI\theta_C - 34.3$$

$$M_{CD} = 0.133EI\theta_C + 0.0667EI\theta_D + 57.9$$

$$M_{DC} = 0.0667EI\theta_C + 0.133EI\theta_D + 57.9$$

Equilibrium equations: $M_{CA} + M_{CB} + M_{CD} = 0$

$$M_{DC} + 100 = 0$$

$$0.417EI\theta_C + 0.0667EI\theta_D = 126.4$$

$$0.0667EI\theta_C + 0.133EI\theta_D = -157.9$$

By solving these equations, we obtain:

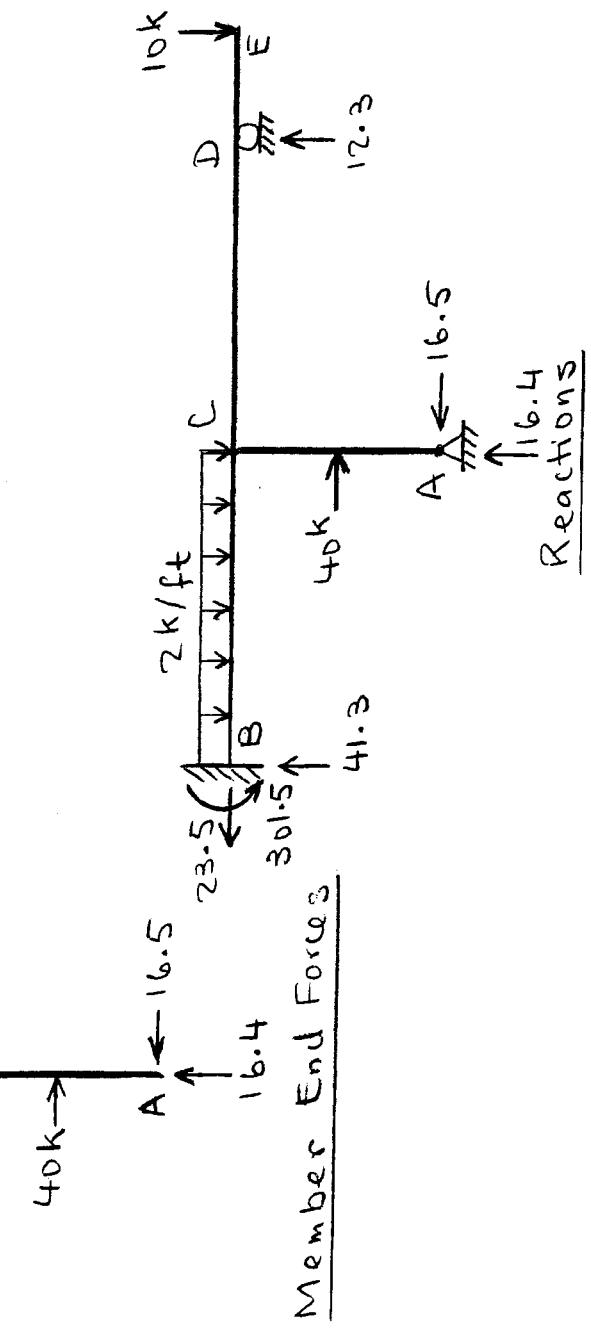
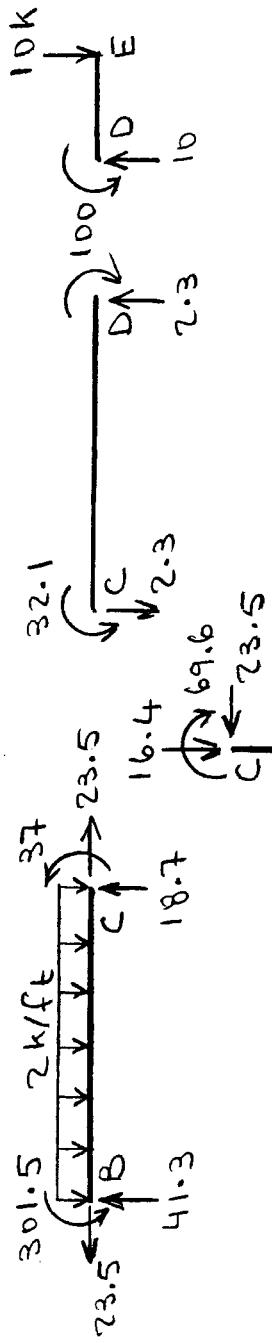
$$EI\theta_C = 536 \text{ k-ft}^2; \quad EI\theta_D = -1456 \text{ k-ft}^2$$

16.23 (cont'd.)

Member end moments: Substituting the numerical values of $EI\theta_C$ and $EI\theta_D$ into the slope-deflection equations, we obtain

$$M_{CA} = -69.6 \text{ k-ft}; M_{BC} = 301.5 \text{ k-ft}; M_{CB} = 37 \text{ k-ft}; M_{CD} = 32.1 \text{ k-ft}$$

$$M_{DC} = -100 \text{ k-ft.}$$



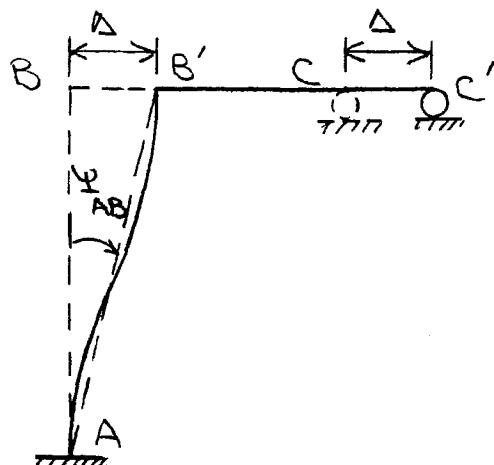
16.24 Fixed-end moments: $FEM_{AB} = FEM_{BA} = 0$

$$FEM_{BC} = \frac{2(15)^2}{12} = 37.5 \text{ k-ft}; \quad FEM_{CB} = -37.5 \text{ k-ft}$$

Chord rotations:

$$\Psi_{AB} = -\frac{\Delta}{20}$$

$$\Psi_{BC} = 0$$



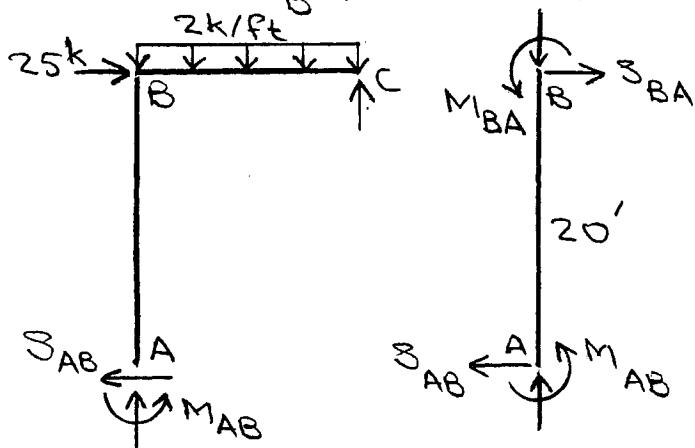
Slope-deflection equations:

$$M_{AB} = 0.1EI\theta_B + 0.015EI\Delta; \quad M_{BA} = 0.2EI\theta_B + 0.015EI\Delta$$

$$M_{BC} = 0.2EI\theta_B + 56.25; \quad M_{CB} = 0$$

$$\text{Equilibrium equations: } M_{BA} + M_{BC} = 0$$

$$0.4EI\theta_B + 0.015EI\Delta = -56.25 \quad (1)$$



$$\sum F_x = 0 \quad S_{AB} = 25 \quad \frac{M_{AB} + M_{BA}}{20} = 25$$

$$0.3EI\theta_B + 0.03EI\Delta = 500 \quad (2)$$

By solving equations (1) and (2) simultaneously, we obtain:

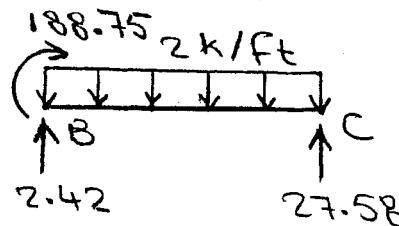
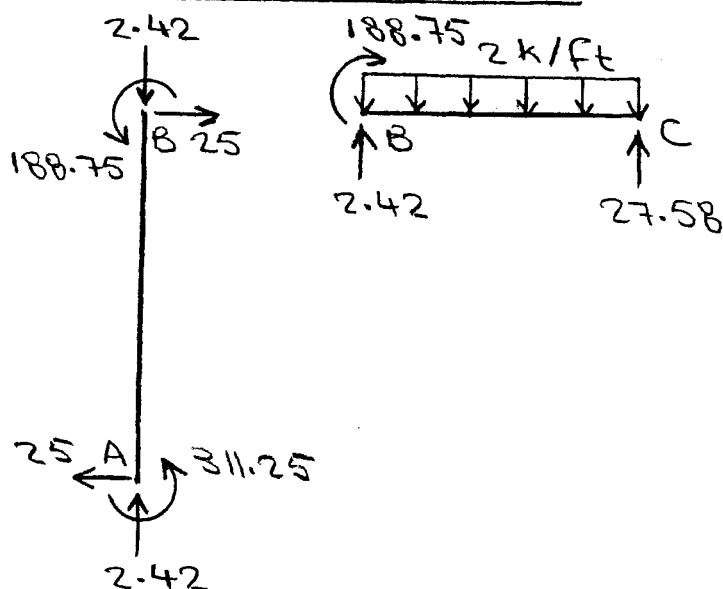
$$EI\theta_B = -1225 \text{ k-ft}^2; \quad EI\Delta = 28916.67 \text{ k-ft}^3.$$

16.24 (contd.)

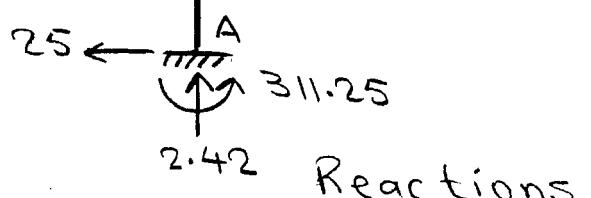
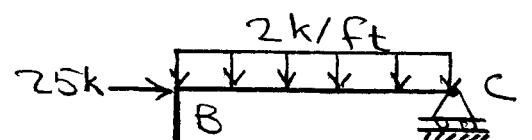
Member end moments: Substituting the numerical values of $EI\theta_B$ and $EI\Delta$ into the slope-deflection equations, we obtain:

$$M_{AB} = 311.25 \text{ k-ft}; \quad M_{BA} = 188.75 \text{ k-ft};$$

$$M_{BC} = -188.75 \text{ k-ft}.$$



Member End Forces



Reactions

16.25 Fixed-end moments. The non-zero fixed-end moments are: $FEM_{CD} = \frac{1.5(40)^2}{12} = 200 \text{ k-ft}$; and $FEM_{DC} = -200 \text{ k-ft}$.

Chord rotations:

$$\Psi_{AC} = \Psi_{BD} = -\frac{\Delta}{30}$$

$$\Psi_{CD} = 0$$

Slope-deflection equations:

$$M_{AC} = 0.0667EI(\theta_C + 0.1\Delta); \quad M_{CA} = 0.0667EI(2\theta_C + 0.1\Delta)$$

$$M_{BD} = 0.0667EI(\theta_D + 0.1\Delta); \quad M_{DB} = 0.0667EI(2\theta_D + 0.1\Delta)$$

$$M_{CD} = 0.1EI(2\theta_C + \theta_D) + 200$$

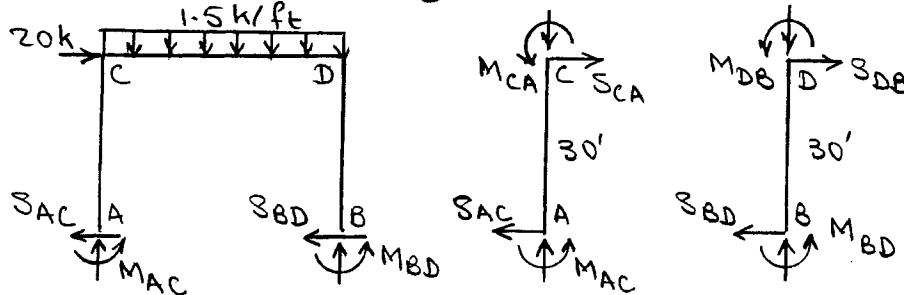
$$M_{DC} = 0.1EI(\theta_C + 2\theta_D) - 200$$

Equilibrium equations: $M_{CA} + M_{CD} = 0$

$$0.333EI\theta_C + 0.1EI\theta_D + 0.0667EI\Delta = -200 \quad (1)$$

$$M_{DC} + M_{DB} = 0$$

$$0.1EI\theta_C + 0.333EI\theta_D + 0.0667EI\Delta = 200 \quad (2)$$



$$\sum F_x = 0 \quad S_{AC} + S_{BD} = 20$$

$$\frac{M_{AC} + M_{CA}}{30} + \frac{M_{BD} + M_{DB}}{30} = 20$$

$$0.00667EI\theta_C + 0.00667EI\theta_D + 0.000889EI\Delta = 20 \quad (3)$$

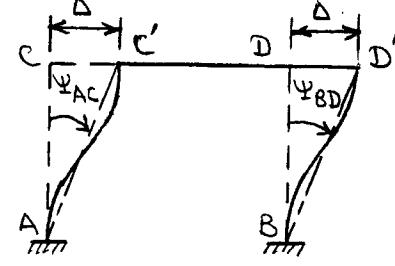
Solving Eqs. (1) thru (3) simultaneously, we obtain:

$$EI\theta_C = -1308 \text{ k-ft}^2; \quad EI\theta_D = 408 \text{ k-ft}^2; \quad EI\Delta = 29242 \text{ k-ft}^3$$

Member end moments: $M_{AC} = 107.8 \text{ k-ft}$; $M_{CA} = 20.8 \text{ k-ft}$

$$M_{BD} = 222.3 \text{ k-ft}; \quad M_{DB} = 249.2 \text{ k-ft}; \quad M_{CD} = -20.8 \text{ k-ft}$$

M_{DC} = -249.2 k-ft. For reactions, see solution of Problem 13.45.



16.26 Fixed-end moments: $M_{AC} = M_{CA} = 0$

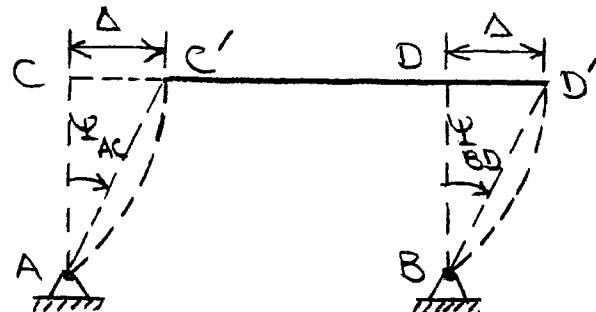
$$M_{BD} = M_{DB} = 0; M_{eD} = \frac{3(30)^2}{12} = 225 \text{ k-ft};$$

$$M_{DC} = -225 \text{ k-ft.}$$

Chord rotations:

$$\Psi_{AC} = \Psi_{BD} = -\frac{\Delta}{15}$$

$$\Psi_{CD} = 0$$



Slope-deflection equations: $M_{AC} = M_{BD} = 0$

$$M_{CA} = 0.2 EI \theta_C + 0.0133 EI \Delta$$

$$M_{DB} = 0.2 EI \theta_D + 0.0133 EI \Delta$$

$$M_{CD} = 0.133 EI \theta_C + 0.0667 EI \theta_D + 225$$

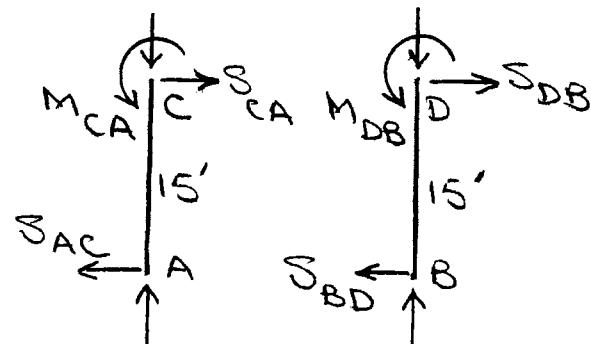
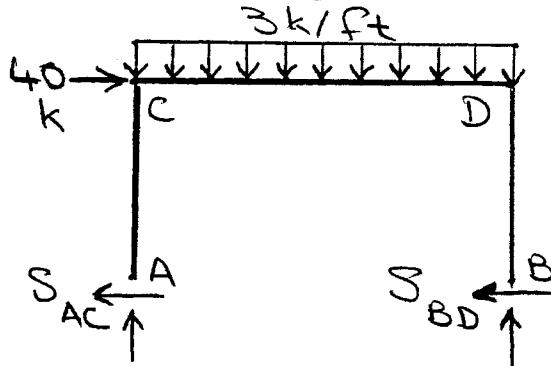
$$M_{DC} = 0.0667 EI \theta_C + 0.133 EI \theta_D - 225$$

Equilibrium equations: $M_{CA} + M_{eD} = 0$

$$0.333 EI \theta_C + 0.0667 EI \theta_D + 0.0133 EI \Delta = -225 \quad (1)$$

$$M_{DB} + M_{DC} = 0$$

$$0.0667 EI \theta_C + 0.333 EI \theta_D + 0.0133 EI \Delta = 225 \quad (2)$$



$$\sum F_x = 0 \quad S_{AC} + S_{BD} = 40 \quad \frac{M_{CA}}{15} + \frac{M_{DB}}{15} = 40$$

$$0.0133 EI \theta_C + 0.0133 EI \theta_D + 0.00178 \Delta = 40 \quad (3)$$

16.26 (Contd.)

By solving Eqs. (1) thru (3) simultaneously, we obtain:

$$EI\theta_c = -2343.75 \text{ k-ft}^2; \quad EI\theta_D = -656.25 \text{ k-ft}^2;$$

$$EI\Delta = 45000 \text{ k-ft}^3.$$

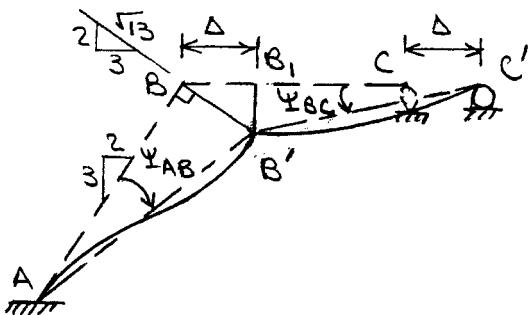
Member end moments:

$$\underline{M_{CA} = 131.25 \text{ k-ft}}; \quad \underline{M_{DB} = 468.75 \text{ k-ft}},$$

$$\underline{M_{CD} = -131.25 \text{ k-ft}}; \quad \underline{M_{DC} = -468.75 \text{ k-ft}}.$$

For reactions, see solution of Problem 13.22.

15.27 Chord rotations:



$$\Psi_{AB} = -\frac{BB'}{L_{AB}} = -\frac{(\sqrt{13}/3)\Delta}{14.42} = -0.0833 \Delta$$

$$\Psi_{BC} = \frac{B_1B'}{L_{BC}} = \frac{(2/3)\Delta}{12} = 0.0556 \Delta$$

Slope-deflection equations: $M_{CB} = 0$

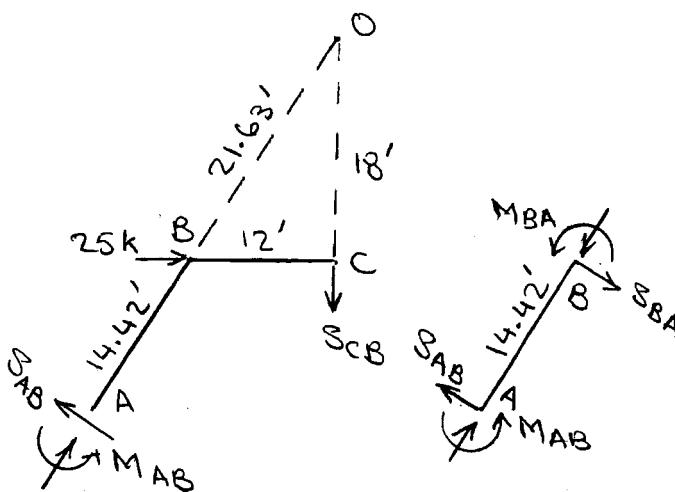
$$M_{AB} = 0.139 EI \theta_B + 0.0347 EI \Delta$$

$$M_{BA} = 0.277 EI \theta_B + 0.0347 EI \Delta$$

$$M_{BC} = 0.25 EI \theta_B - 0.0139 EI \Delta$$

Equilibrium equations: $M_{BA} + M_{BC} = 0$

$$0.527 EI \theta_B + 0.0208 EI \Delta = 0 \quad (1)$$



$$+\sum M_O = 0 \quad M_{AB} - S_{AB}(36.05) + 25(18) = 0$$

$$M_{AB} - \left(\frac{M_{AB} + M_{BA}}{14.42} \right)(36.05) + 450 = 0$$

$$0.901 EI \theta_B + 0.139 EI \Delta = 450 \quad (2)$$

16.27 (Contd.)

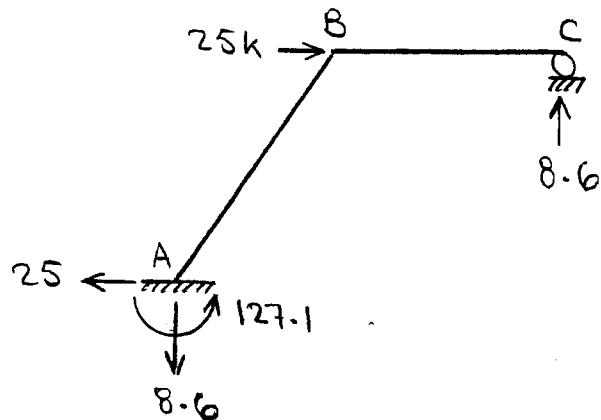
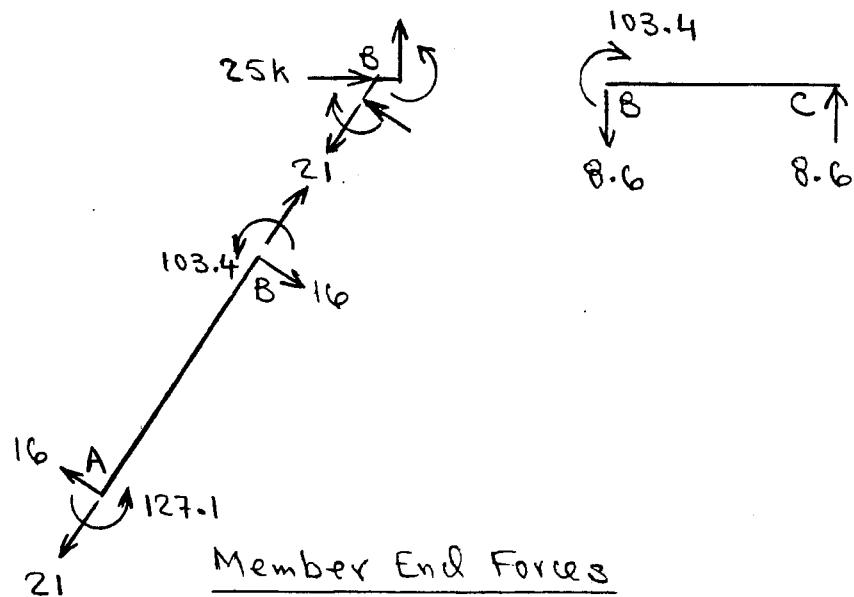
By solving Eqs. (1) and (2), we obtain:

$$EI\theta_B = -171.7 \text{ k-ft}^2; EI\Delta = 4350 \text{ k-ft}^3$$

Member end moments. Substituting the numerical values of $EI\theta_B$ and $EI\Delta$ into the slope-deflection equations, we obtain:

$$M_{AB} = 127.1 \text{ k-ft}; M_{BA} = 103.4 \text{ k-ft};$$

$$M_{BC} = -103.4 \text{ k-ft}.$$

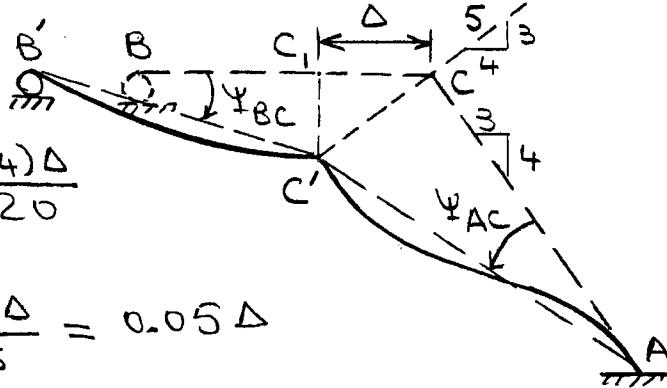


Reactions

16.28 Fixed end moments:

$$FEM_{BC} = \frac{1.5(20)^2}{12} = 50 \text{ k-ft}; FEM_{CB} = -50 \text{ k-ft}$$

Chord rotations:



$$\Psi_{BC} = -\frac{c_c c'_c}{L_{BC}} = -\frac{(3/4)\Delta}{20} = -0.0375\Delta$$

$$\Psi_{AC} = \frac{c_c c'_c}{L_{AC}} = \frac{(5/4)\Delta}{25} = 0.05\Delta$$

Slope-deflection equations: $M_{BC} = 0$

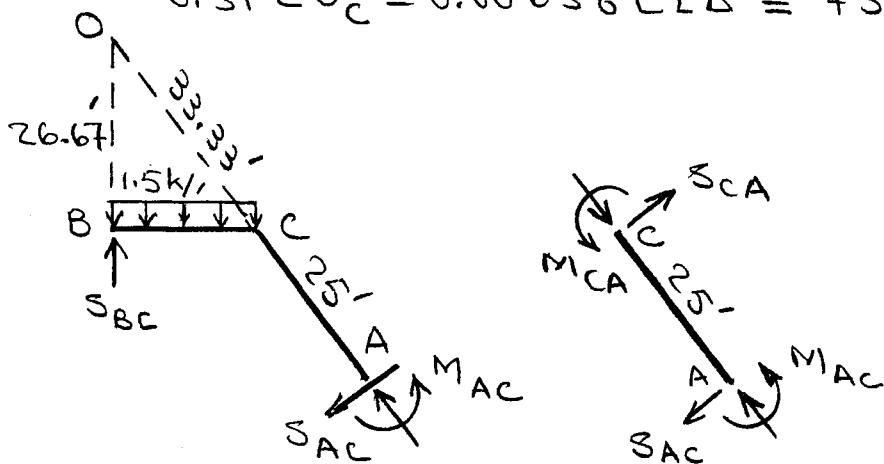
$$M_{CB} = 0.15EI(\theta_c + 0.0375\Delta) = 75$$

$$M_{CA} = 0.08EI(2\theta_c - 0.15\Delta)$$

$$M_{AC} = 0.08EI(\theta_c - 0.15\Delta)$$

Equilibrium equations: $M_{CB} + M_{CA} = 0$

$$0.31EI\theta_c - 0.00638EI\Delta = 75 \quad (1)$$



$$+\sum M_O = 0 \quad M_{AC} - S_{AC}(58.33) - 1.5(20)10 = 0$$

$$M_{AC} - \left(\frac{M_{AC} + M_{CA}}{25}\right)(58.33) - 300 = 0$$

$$-0.48EI\theta_c + 0.044EI\Delta = 300 \quad (2)$$

By solving Eqs. (1) and (2), we obtain

$$EI\theta_c = 493 \text{ k-ft}^2; EI\Delta = 12196 \text{ k-ft}^3.$$

16.28 (contd.) Member end moments: Substituting the numerical values of $EI\theta_c$ and $EI\Delta$ into the slope-deflection equations, we obtain

$$\underline{M_{CB} = 67.5 \text{ k-ft}} ; \underline{M_{CA} = -67.5 \text{ k-ft}}$$
$$\underline{M_{AC} = -106.9 \text{ k-ft}}$$

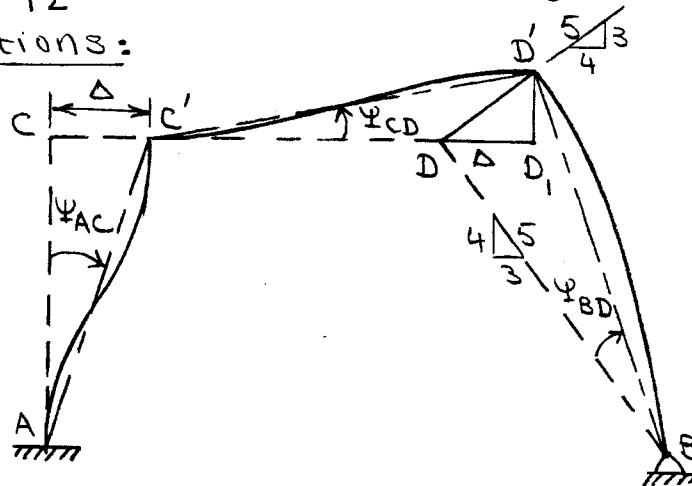
For reactions, see solution of Problem 13.24.

16-29 Fixed-end moments: $FEM_{BD} = FEM_{DB} = 0$

$$FEM_{AC} = \frac{50(4)}{8} = 25 \text{ kN.m}; FEM_{CA} = -25 \text{ kN.m}$$

$$FEM_{CD} = \frac{18(5)^2}{12} = 37.5 \text{ kN.m}; FEM_{DC} = -37.5 \text{ kN.m}$$

Chord rotations:



$$\Psi_{AC} = -\frac{CC'}{L_{AC}} = -\frac{\Delta}{4} = -0.25\Delta$$

$$\Psi_{BD} = -\frac{DD'}{L_{BD}} = -\frac{(5/4)\Delta}{5} = -0.25\Delta$$

$$\Psi_{CD} = \frac{D_1D'}{L_{CD}} = \frac{(3/4)\Delta}{5} = 0.15\Delta$$

Slope-deflection equations: $M_{BD} = 0$

$$M_{AC} = 0.5EI(\theta_C + 0.75\Delta) + 25$$

$$M_{CA} = 0.5EI(2\theta_C + 0.75\Delta) - 25$$

$$M_{CD} = 0.4EI(2\theta_C + \theta_D - 0.45\Delta) + 37.5$$

$$M_{DC} = 0.4EI(\theta_C + 2\theta_D - 0.45\Delta) - 37.5$$

$$M_{DB} = 0.6EI(\theta_D + 0.25\Delta)$$

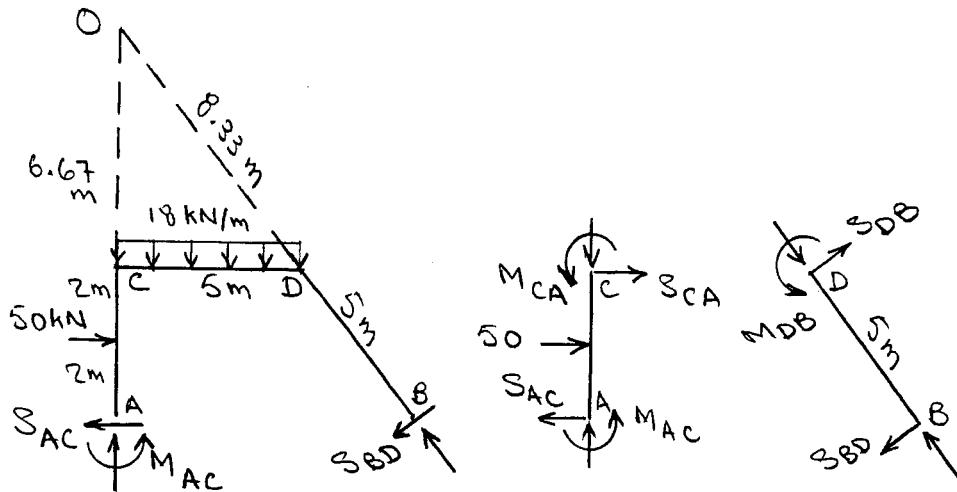
Equilibrium equations: $M_{CA} + M_{CD} = 0$

$$1.8EI\theta_C + 0.4EI\theta_D + 0.195EI\Delta = -12.5 \quad (1)$$

$$M_{DC} + M_{DB} = 0$$

$$0.4EI\theta_C + 1.4EI\theta_D - 0.03EI\Delta = 37.5 \quad (2)$$

16.29 (contd.)



$$+G \sum M_O = 0$$

$$M_{AC} - S_{AC}(10.67) - S_{BD}(13.33) + 50(8.67) - 18(5)2.5 = 0$$

$$M_{AC} - \left(25 + \frac{M_{AC} + M_{CA}}{4}\right)10.67 - \left(\frac{M_{DB}}{5}\right)13.33 + 208.5 = 0$$

$$3.5EI\theta_C + 1.6EI\theta_D + 2.025EI\Delta = -33.25 \quad (3)$$

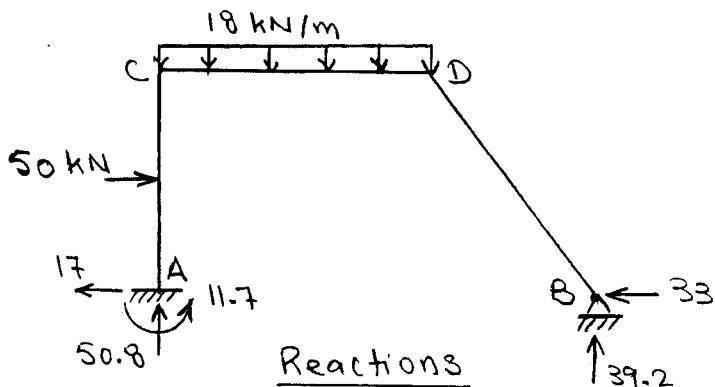
By solving Eqs. (1) thru (3) simultaneously, we obtain:
 $EI\theta_C = -11.33 \text{ kN}\cdot\text{m}^2$; $EI\theta_D = 29.59 \text{ kN}\cdot\text{m}^2$

$$EI\Delta = -20.22 \text{ kN}\cdot\text{m}^3$$

Member end moments. Substituting the numerical values of $EI\theta_C$, $EI\theta_D$ and $EI\Delta$ into the slope-deflection equations, we obtain:

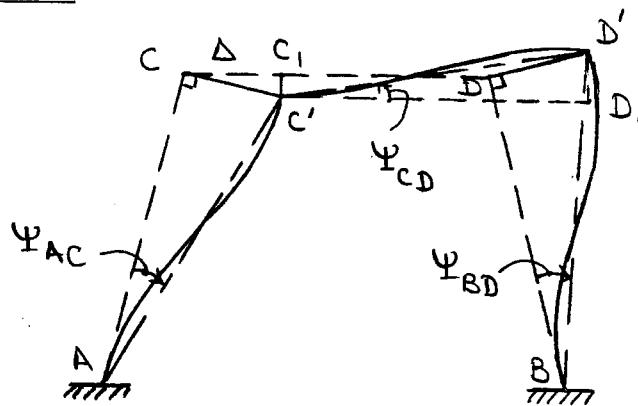
$$M_{AC} = 11.7 \text{ kN}\cdot\text{m}; \quad M_{CA} = -43.9 \text{ kN}\cdot\text{m};$$

$$M_{CD} = 43.9 \text{ kN}\cdot\text{m}; \quad M_{DC} = -14.7 \text{ kN}\cdot\text{m}; \quad M_{DB} = 14.7 \text{ kN}\cdot\text{m}$$



16.30 Fixed-end moments: The non-zero fixed-end moments are: $FEM_{CD} = \frac{3(16)^2}{12} = 64 \text{ k-ft}$ and $FEM_{DC} = -64 \text{ k-ft}$.

Chord rotations:



$$\Psi_{AC} = -\frac{CC'}{L_{AC}} = -\frac{(\sqrt{17}/4)\Delta}{16.49} = -0.0625\Delta$$

$$\Psi_{BD} = -\frac{DD'}{L_{BD}} = -\frac{(\sqrt{17}/4)\Delta}{16.49} = -0.0625\Delta$$

$$\Psi_{CD} = \frac{D_1D'}{L_{CD}} = \frac{2(1/4)\Delta}{16} = 0.03125\Delta$$

Slope-deflection equations:

$$M_{AC} = 0.121EI(\theta_C + 0.188\Delta); M_{CA} = 0.121EI(2\theta_C + 0.188\Delta)$$

$$M_{BD} = 0.121EI(\theta_D + 0.188\Delta); M_{DB} = 0.121EI(2\theta_D + 0.188\Delta)$$

$$M_{CD} = 0.125EI(2\theta_C + \theta_D - 0.0938\Delta) + 64$$

$$M_{DC} = 0.125EI(\theta_C + 2\theta_D - 0.0938\Delta) - 64$$

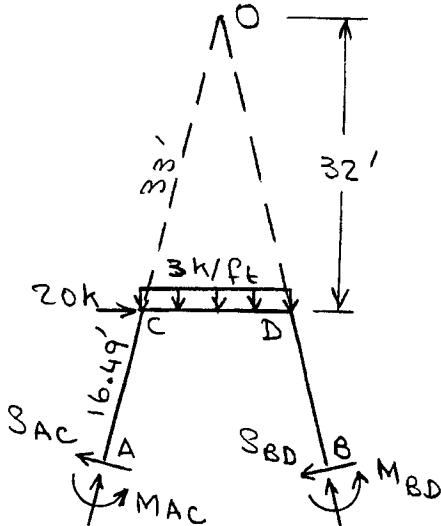
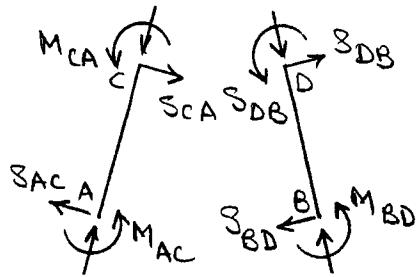
$$\text{Equilibrium equations: } M_{CA} + M_{CD} = 0$$

$$0.492EI\theta_C + 0.125EI\theta_D + 0.011EI\Delta = -64 \quad (1)$$

$$M_{DC} + M_{DB} = 0$$

$$-0.125EI\theta_C + 0.492EI\theta_D + 0.011EI\Delta = 64 \quad (2)$$

16.30 (Contd.)



$$+C \sum M_O = 0$$

$$M_{AC} + M_{BD} - (S_{AC} + S_{BD})49.49 + 20(32) = 0$$

$$M_{AC} + M_{BD} - \frac{49.49}{16.49}(M_{AC} + M_{CA} + M_{BD} + M_{DB}) + 640 = 0$$

$$0.968 EI\theta_C + 0.968 EI\theta_D + 0.227 EI\Delta = 640 \quad (3)$$

By solving Eqs. (1) thru (3), we obtain:

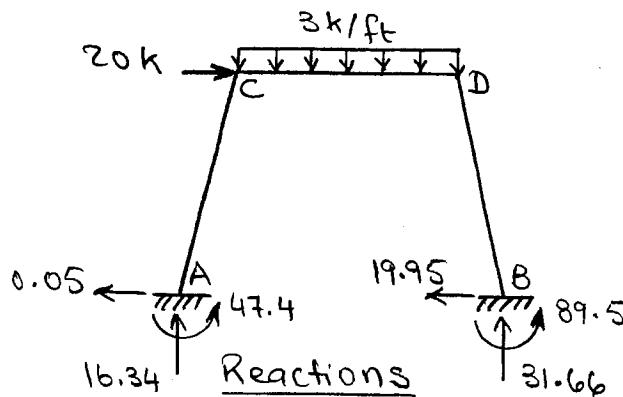
$$EI\theta_C = -233.7 \text{ k-ft}^2, EI\theta_D = 115.1 \text{ k-ft}^2;$$

$$EI\Delta = 3325 \text{ k-ft}^3$$

Member end moments. Substituting the numerical values of $EI\theta_C$, $EI\theta_D$ and $EI\Delta$ into the slope-deflection equations, we obtain: $M_{AC} = 47.4 \text{ k-ft}$;

$$M_{CA} = 19 \text{ k-ft}; \quad M_{BD} = 89.5 \text{ k-ft}; \quad M_{DB} = 103.5 \text{ k-ft};$$

$$M_{CD} = -19 \text{ k-ft}; \quad M_{DC} = -103.5 \text{ k-ft}.$$

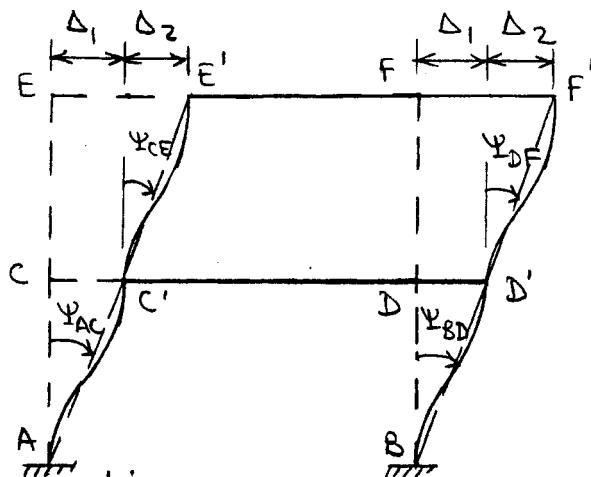


16.31 Chord rotations:

$$\Psi_{AC} = \Psi_{BD} = -\frac{\Delta_1}{15}$$

$$\Psi_{CE} = \Psi_{DF} = -\frac{\Delta_2}{15}$$

$$\Psi_{CD} = \Psi_{EF} = 0$$



Slope-deflection equations:

$$M_{AC} = 0.133EI(\theta_C + 0.2\Delta_1); M_{CA} = 0.133EI(2\theta_C + 0.2\Delta_1)$$

$$M_{BD} = 0.133EI(\theta_D + 0.2\Delta_1); M_{DB} = 0.133EI(2\theta_D + 0.2\Delta_1)$$

$$M_{CE} = 0.133EI(2\theta_C + \theta_E + 0.2\Delta_2)$$

$$M_{EC} = 0.133EI(\theta_C + 2\theta_E + 0.2\Delta_2)$$

$$M_{DF} = 0.133EI(2\theta_D + \theta_F + 0.2\Delta_2)$$

$$M_{FD} = 0.133EI(\theta_D + 2\theta_F + 0.2\Delta_2)$$

$$M_{CD} = 0.133EI(2\theta_C + \theta_D); M_{DC} = 0.133EI(\theta_C + 2\theta_D)$$

$$M_{EF} = 0.133EI(2\theta_E + \theta_F); M_{FE} = 0.133EI(\theta_E + 2\theta_F)$$

$$\text{Equilibrium equations: } M_{CA} + M_{CD} + M_{CE} = 0$$

$$EI(6\theta_C + \theta_D + \theta_E + 0.2\Delta_1 + 0.2\Delta_2) = 0 \quad (1)$$

$$M_{DB} + M_{DC} + M_{DF} = 0$$

$$EI(\theta_C + 6\theta_D + \theta_F + 0.2\Delta_1 + 0.2\Delta_2) = 0 \quad (2)$$

$$M_{EC} + M_{EF} = 0$$

$$EI(\theta_C + 4\theta_E + \theta_F + 0.2\Delta_2) = 0 \quad (3)$$

$$M_{FD} + M_{FE} = 0$$

$$EI(\theta_D + \theta_E + 4\theta_F + 0.2\Delta_2) = 0 \quad (4)$$

$$\theta_{CE} + \theta_{DF} = 9$$

$$\frac{1}{15} [(M_{CE} + M_{EC}) + (M_{DF} + M_{FD})] = 9$$

16.31 (Contd.)

$$EI(3\theta_C + 3\theta_D + 3\theta_E + 3\theta_F + 0.8\Delta_2) = 1015 \quad (5)$$

$$S_{AC} + S_{BD} = 27$$

$$\frac{1}{15} [(M_{AC} + M_{CA}) + (M_{BD} + M_{DB})] = 27$$

$$EI(3\theta_C + 3\theta_D + 0.8\Delta_1) = 3045 \quad (6)$$

By solving Eqs. (1) thru (6) simultaneously, we

obtain: $EI\theta_C = EI\theta_D = -267.6 \text{ k-ft}^2$;

$$EI\theta_E = EI\theta_F = -110.7 \text{ k-ft}^2; \quad EI\Delta_1 = 5813 \text{ k-ft}^3;$$

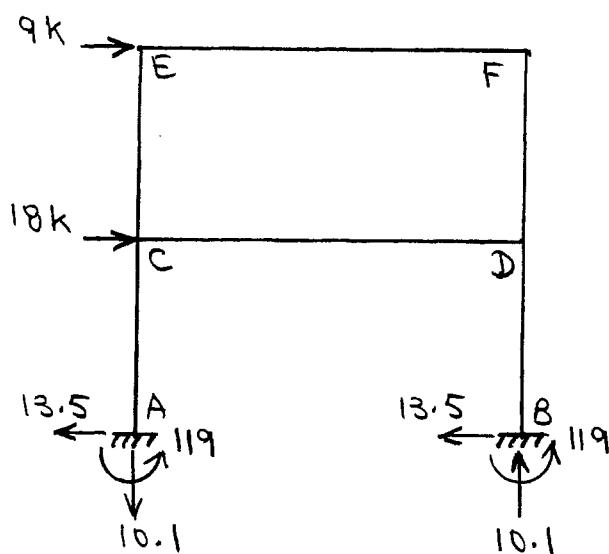
$$EI\Delta_2 = 4106 \text{ k-ft}^3.$$

Member end moments: $M_{AC} = M_{BD} = 119 \text{ k-ft}$;

$$M_{CA} = M_{DB} = 83.5 \text{ k-ft}; \quad M_{CE} = M_{DF} = 23.3 \text{ k-ft};$$

$$M_{EC} = M_{FD} = 44.2 \text{ k-ft}; \quad M_{CD} = M_{DC} = -106.8 \text{ k-ft};$$

$$M_{EF} = M_{FE} = -44.2 \text{ k-ft}.$$



Reactions

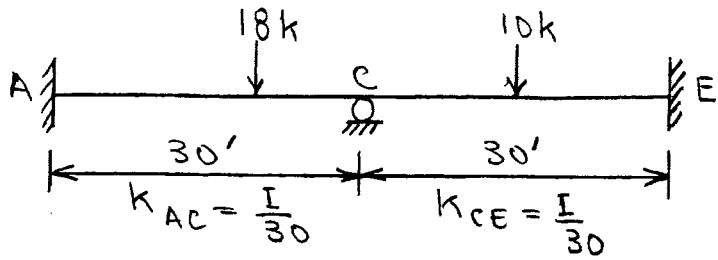
Chapter Seventeen

Moment-Distribution

Method

CHAPTER 17

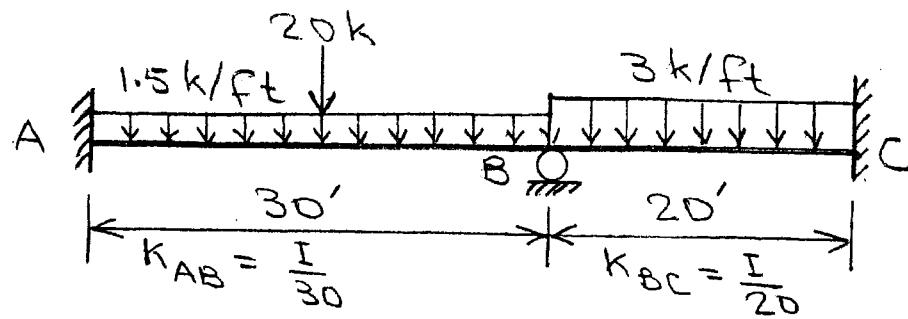
17-1



DF	0.5	0.5	
FEM	40	-80	37.5 -37.5
	21.3	21.3	
	10.6 ←	10.6 →	
Final Moments	50.6	-58.7	58.8 -26.9

For reactions, and shear and bending moment diagrams, see solution of Problem 16-1.

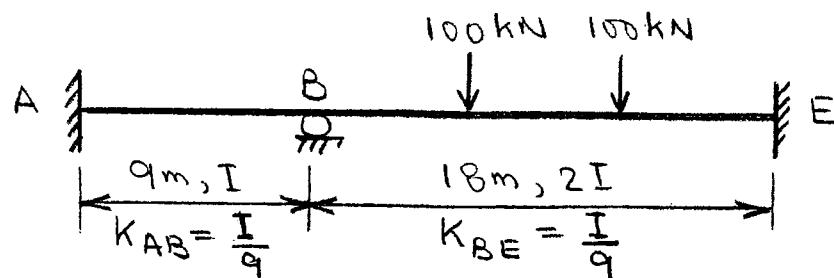
17.2

DF
FEM

	0.4	0.6		
187.5	-187.5	100	-100	
	35	52.5		
17.5			26.25	
Final Moments	205	-152.5	152.5	-73.75

For reactions, and shear and bending moment diagrams, see solution of Problem 16.2.

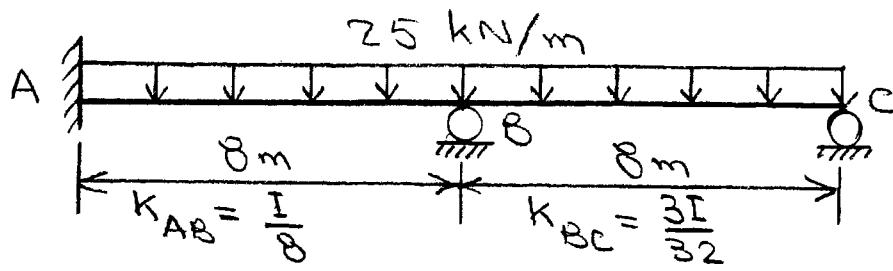
17.3



DF FEM	0.5	0.5	
0	0	400	-400
	-200	-200	
-100			-100
Final Moments	-100	-200	-500

For reactions, and shear and bending moment diagrams, see solution of Problem 16.3.

17.4

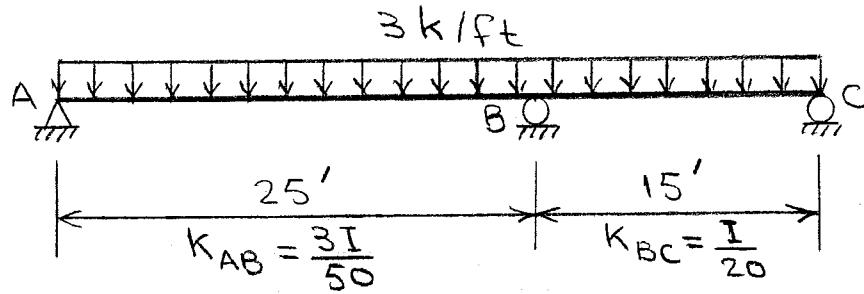


DF
FEM

	0.571	0.429	
133.3	-133.3	133.3	-133.3
		66.7	133.3
	-38.1	-28.6	
-19			
Final Moments	114.3	-171.4	171.4
			0

For reactions, and shear and bending moment diagrams, see solution of Problem 13.37.

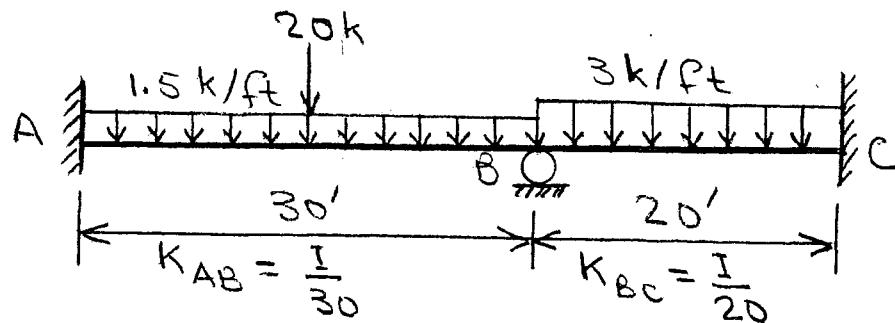
17-5



DF		0.545	0.455	
FEM	156.3	-156.3	56.3	-56.3
	-156.3	54.5	45.5	56.3
		-78.1	28.1	
		27.3	22.7	
Final Moments	0	-152.6	152.6	0

For reactions, and shear and bending moment diagrams, see solution of Problem 13.12.

17.6



$$\begin{aligned} FEM_{AB} &= \frac{1.5(30)^2}{12} + \frac{20(30)}{8} + \frac{6(29000)(1650)(0.5)}{(30)^2(12)^3} \\ &= 112.5 + 75 + 92.3 = 279.8 \text{ k-ft} \\ FEM_{BA} &= -112.5 - 75 + 92.3 = -95.2 \text{ k-ft} \\ FEM_{BC} &= \frac{3(20)^2}{12} - \frac{6(29000)(1650)(0.5)}{(20)^2(12)^3} \\ &= 100 - 207.7 = -107.7 \text{ k-ft} \\ FEM_{CB} &= -100 - 207.7 = -307.7 \text{ k-ft} \end{aligned}$$

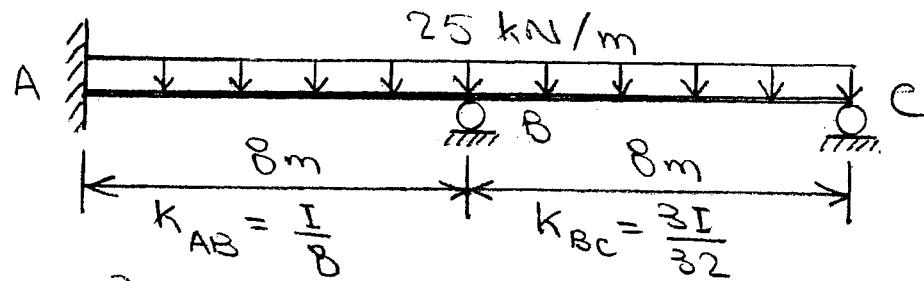
DF FEM

	0.4	0.6		
279.8	-95.2	-107.7	-307.7	
	81.2	121.7		
40.6			60.9	
320.4	-14	14	-246.8	

Final Moments

For reactions, and shear and bending moment diagrams, see solution of Problem 16.6.

17.7



$$\text{FEM}_{AB} = \frac{25(8)^2}{12} + \frac{6(70)(1300)(0.05)}{(8)^2}$$

$$= 133.3 + 426.6 = 559.9 \text{ kN.m}$$

$$\text{FEM}_{BA} = -133.3 + 426.6 = 293.3 \text{ kN.m}$$

$$\text{FEM}_{BC} = 133.3 - \frac{6(70)(1300)(0.025)}{(8)^2}$$

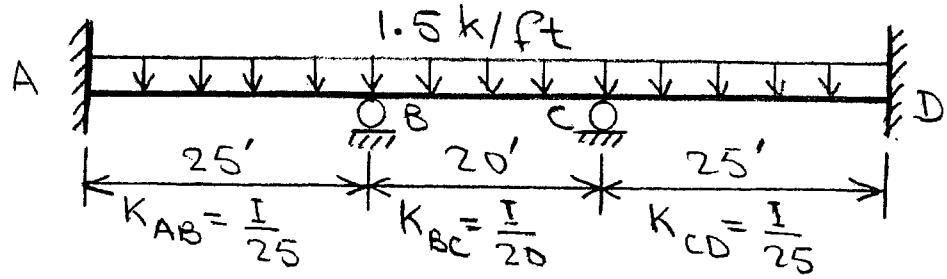
$$= 133.3 - 213.3 = -80 \text{ kN.m}$$

$$\text{FEM}_{CB} = -133.3 - 213.3 = -346.6 \text{ kN.m}$$

DF	0.571	0.429	
FEM	559.9	293.3	-80
	-121.8	-91.5	346.6
-60.9	173.3	-74.3	
-49.5	-99		
Final Moments	449.5	72.5	-72.5
			0

For reactions, and shear and bending moment diagrams, see solution of Problem 13.53.

17.8

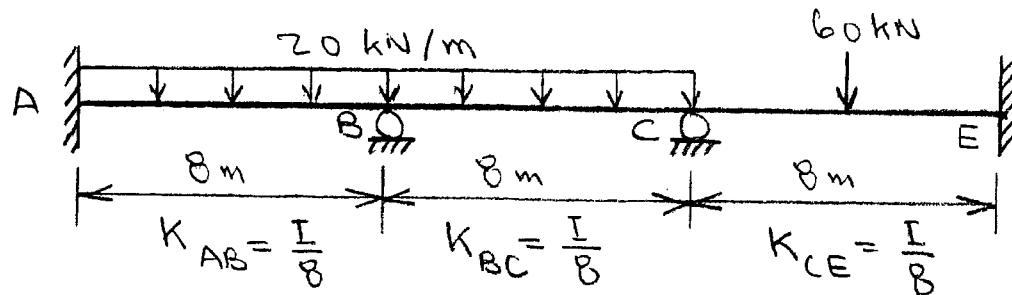
DF
FEM

	$4/9$	$5/9$	$5/9$	$4/9$	
78.1	-78.1	50	-50	78.1	-78.1
12.5	15.6	-15.6	-15.6	-12.5	
6.2		-7.8	7.8		-6.2
3.5	4.3	-4.3	-4.3	-3.5	
1.7		-2.2	2.2		-1.7
1	1.2	-1.2	-1.2	-1	
0.5		-0.6	0.6		-0.5
0.3	0.3	-0.3	-0.3	-0.3	
0.1		-0.2	0.2		-0.1
0.1	0.1	-0.1	-0.1	-0.1	
86.6	-60.7	60.7	-60.7	60.7	-86.6

Final Moments

For reactions, and shear and bending moment diagrams, see solution of Problem 16.8.

17.9

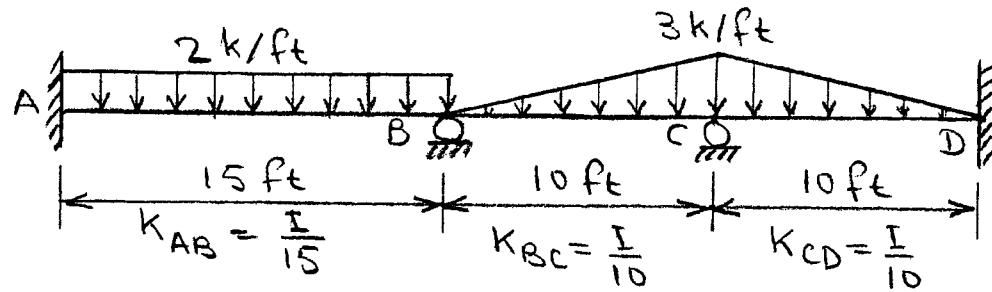
DF
FEM

	0.5	0.5	0.5	0.5		
106.7	-106.7	106.7	-106.7	60	-60	
			23.4	23.4		
	-5.9	11.7			11.7	
-3			-3			
			1.5	1.5		
	0.8				0.8	
-0.4			-0.4			
-0.2			-0.2			
			0.1	0.1		
103.5	-113	112.9	-84.9	85	-47.5	

Final Moments

For reactions, and shear and bending moment diagrams, see solution of Problem 16.9.

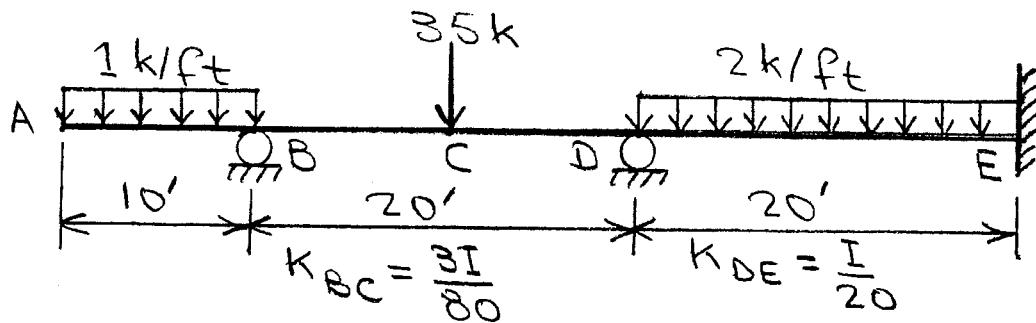
17.10



DF FEM	0.4	0.6	0.5	0.5		
37.5	-37.5	10	-15	15	-10	
5.5	11	16.5	8.3			
			-4.2	-4.2	-2.1	
		-2.1				
0.4	0.8	1.3	0.7			
			-0.3	-0.3	-0.2	
		-0.2				
Final Moments	43.4	-25.6	25.6	-10.5	10.5	-12.3

For reactions, and shear and bending moment diagrams, see solution of Problem 16.10.

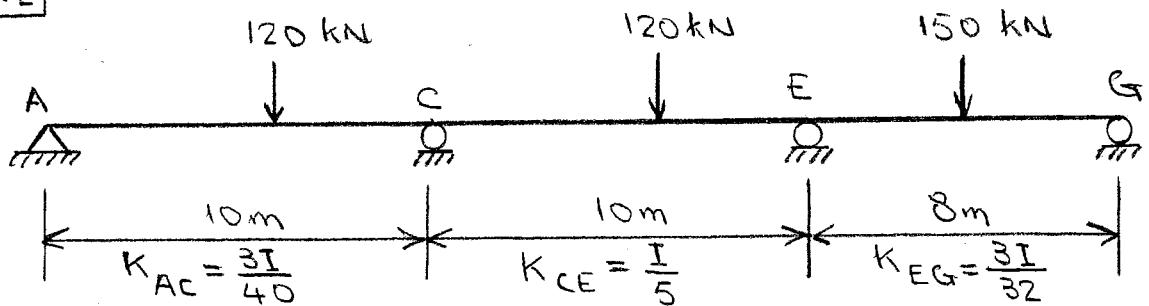
17.11



DF FEM	1	0.429	0.571	
-50	87.5	-87.5	66.7	-66.7
	-37.5	8.9	11.9	
		-18.8		5.9
		8.1	10.7	
				5.4
Final Moments	-50	50	-89.3	89.3
				-55.4

For reactions, and shear and bending moment diagrams, see solution of Problem 13.38.

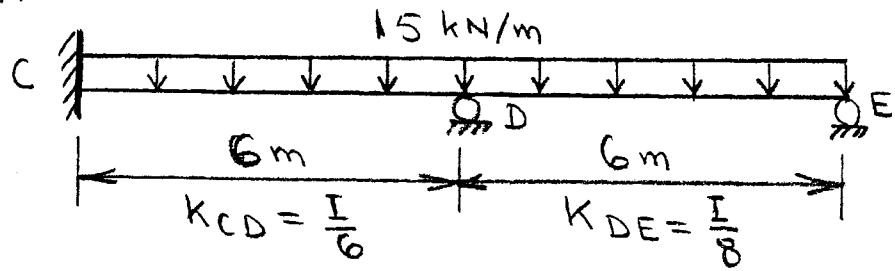
17.12



DF	0.273	0.727	0.681	0.319	
FEM	115.2	-172.8	115.2	-172.8	150
	-115.2	15.7	41.9	15.5	7.3
		→ -57.6	7.8 ←	→ 20.9	75 ←
		13.6	36.2	-65.3	-30.6
			-32.7	18.1	
		8.9	23.8	-12.3	-5.8
			-6.2	11.9	
		1.7	4.5	-8.1	-3.8
			-4.1	2.3	
		1.1	3	-1.6	-0.7
			-0.8	1.5	
		0.2	0.6	-1	-0.5
		0.1	0.4	-0.2	-0.1
Final Moments	0	-189.1	189.1	-190.8	190.8
					0

For reactions, and shear and bending moment diagrams, see solution of Problem 13.39.

17.13 As the beam and loading are symmetric, we will analyze only the right half CE of the beam.



DF	0.571	0.429	1
FEM	45	-45	45
		22.5	-45
	-12.8	-9.7	45
-6.4			
Final Moments	38.6	-57.8	57.8
			0

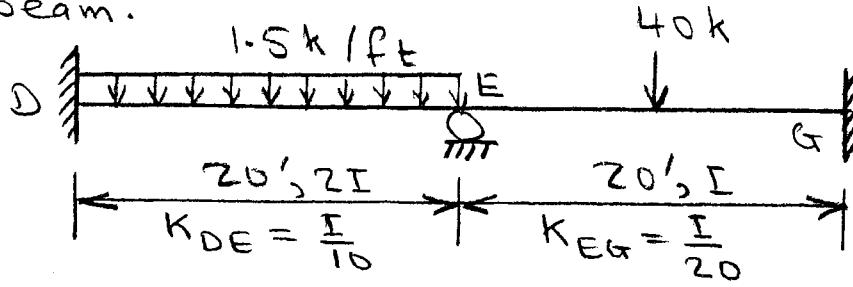
Because of symmetry, the member end moments for the left half of the beam are:

$$M_{AB} = 0; M_{BA} = -57.8 \text{ kN.m}$$

$$M_{BC} = 57.8 \text{ kN.m}; M_{CB} = -38.6 \text{ kN.m}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 16.13.

17-14 As the beam and loading are symmetric, we will analyze only the right half DG of the beam.



DF
FEM

	$\frac{2}{3}$	$\frac{1}{3}$	
50	-50	100	-150
	-33.3	-16.7	
-16.7			-8.3
Final Moments	33.3	-83.3	83.3
			-108.3

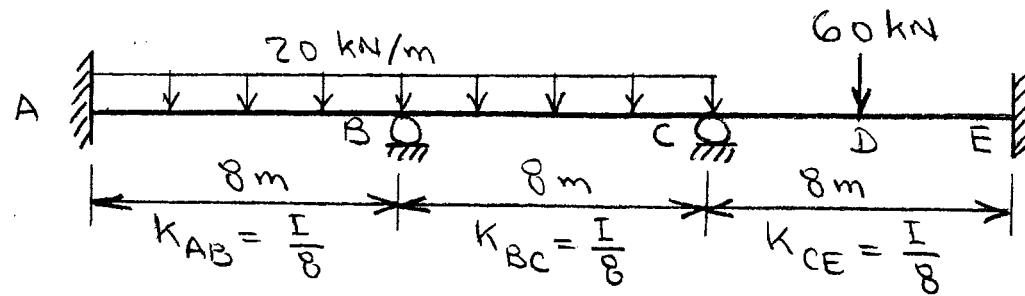
Because of symmetry, the member end moments for the left half of the beam are:

$$M_{AC} = 108.3 \text{ k-ft}; \quad M_{CA} = -83.3 \text{ k-ft}$$

$$M_{CD} = 83.3 \text{ k-ft}; \quad M_{DC} = -33.3 \text{ k-ft}$$

For reactions, and shear and bending moment diagrams, see solution of Problem 16-14.

17.15



$$FEM_{AB} = \frac{20(8)^2}{12} = 106.7 \text{ kN.m}; FEM_{BA} = -106.7 \text{ kN.m}$$

$$FEM_{BC} = 106.7 + \frac{6(70)(800)(0.025)}{(8)^2} = 106.7 + 131.3 \\ = 238 \text{ kN.m}$$

$$FEM_{CB} = -106.7 + 131.3 = 24.6 \text{ kN.m}$$

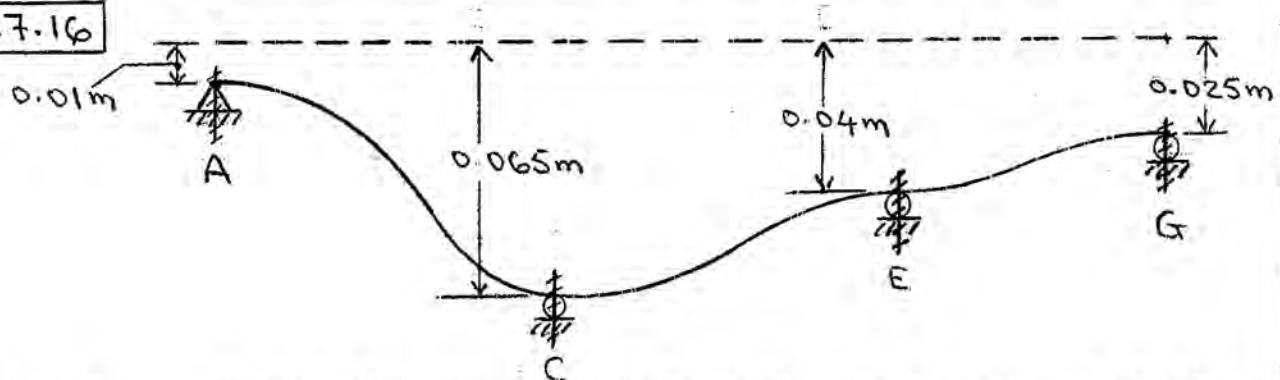
$$FEM_{CE} = \frac{60(8)}{8} - 131.3 = 60 - 131.3 = -71.3 \text{ kN.m}$$

$$FEM_{EC} = -60 - 131.3 = -191.3 \text{ kN.m}$$

DF	0.5	0.5	0.5	0.5		
FEM	106.7	-106.7	238	24.6	-71.3	-191.3
	-65.7	-65.7	23.4	23.4		
	-32.8	11.7	-32.8		11.7	
	-5.8	-5.8	16.4	16.4		
	-2.9	8.2	-2.9		8.2	
	-4.1	-4.1	1.5	1.5		
	-2.1	0.7	-2.1		0.7	
	-0.4	-0.4	1	1		
	-0.2	0.5	-0.2		0.5	
	-0.2	-0.2	0.1	0.1		
Final Moments	68.7	-182.9	182.9	29	-28.9	-170.2

For reactions, and shear and bending moment diagrams, see solution of Problem 16.15.

17.16



$$FEM_{AC} = \frac{120(6)(4)^2}{(10)^2} + \frac{6(250)(500)(0.055)}{(10)^2} = 115.2 + 330 = 445.2 \text{ kN.m}$$

$$FEM_{CA} = -\frac{120(6)^2(4)}{(10)^2} + 330 = -172.8 + 330 = 157.2 \text{ kN.m}$$

$$FEM_{CE} = 115.2 - \frac{6(250)(1000)(0.025)}{(10)^2} = 115.2 - 300 = -184.8 \text{ kN.m}$$

$$FEM_{EC} = -172.8 - 300 = -472.8 \text{ kN.m}$$

$$FEM_{EG} = \frac{150(8)}{8} - \frac{6(250)(500)(0.015)}{(8)^2} = 150 - 140.6 = 9.4 \text{ kN.m}$$

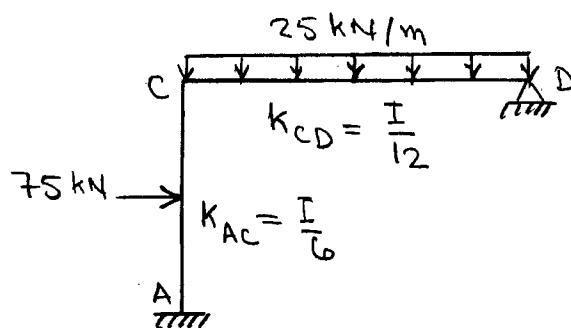
$$FEM_{GE} = -150 - 140.6 = -290.6 \text{ kN.m}$$

DF	0.273	0.727	0.681	0.319	
FEM	445.2	157.2	-184.8	-472.8	
-445.2	7.5	20.1	315.6	147.8	
-222.6	157.8	10	145.3	290.6	
17.7	47.1	-105.8	-49.5		
-52.9	23.6				
14.4	38.5	-16.1	-7.5		
-8	19.2				
2.2	5.8	-13.1	-6.1		
-6.5	2.9				
1.8	4.7	-2	-0.9		
-1	2.4				
0.3	0.7	-1.6	-0.8		
-0.8	0.4				
0.2	0.6	-0.3	-0.1		
0	-21.3	21.3	-237.6	237.6	0

Final Moments

For reactions, and shear and bending moment diagrams,
see solution of Problem 13.54.

17.17

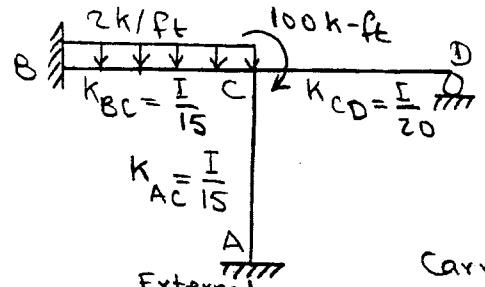


DF FEM	AC	CA	CD	DC
	2/3	1/3		
56.3	-56.3	168.8	-168.8	
	-75	-37.5	168.8	
-37.5		84.4		
	-56.3	-28.1		
-28.2				
	-9.4	-187.6	187.6	0

Final Moments

for reactions, see solution of problem 18.42

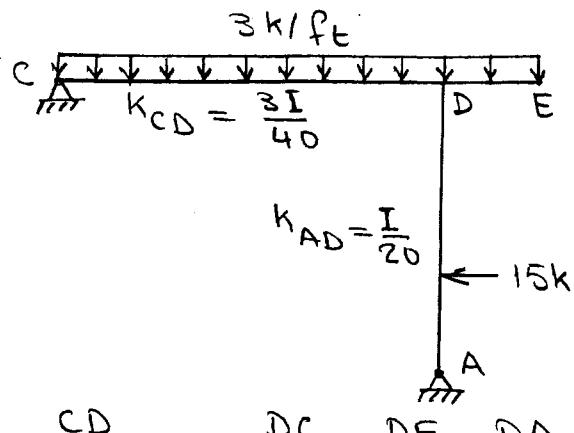
17-18



	BC	CB	External Moment at Joint C	CA	CD	DC	AC
DF	0.364	-	0.364	0.272			
FEM	37.5	-37.5	100	0	0	0	0
	-22.7	-	-22.7	-17.1			
Final Moments	26.2	-60.2	100	-22.7	-17.1	0	-11.3

For reactions, see solution of Problem 16.18.

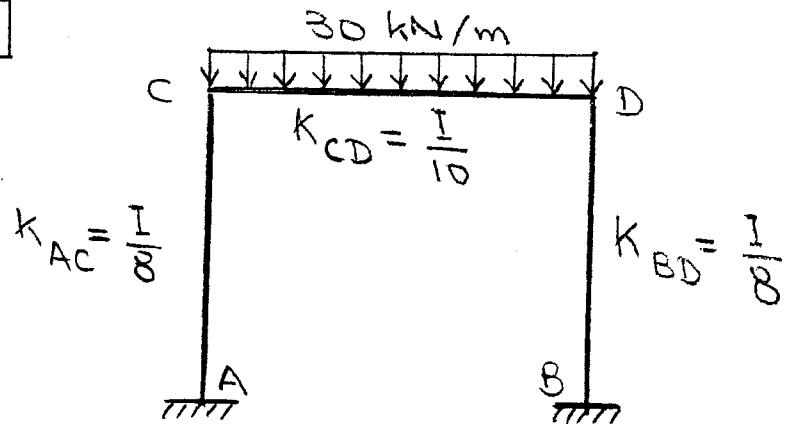
17.19



	CD	DC	DE	DA	AD
DF	0.6	-	0.4		
FEM	100	-100	37.5	16.7	-33.3
	-100	27.5		18.3	33.3
		-50		16.7	
		20		13.3	
Final Moments	0	-102.5	37.5	65	0

For reactions, see solution of Problem 13.21.

17.20

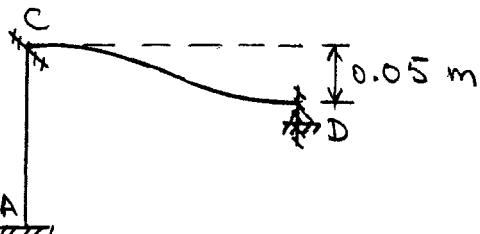


	AC	CA	CD	DC	DB	BD
DF FEM		$\frac{5}{9}$	$\frac{4}{9}$	$\frac{4}{9}$	$\frac{5}{9}$	
0	0	250	-250	0	0	0
-69.4	-138.9	-111.1	111.1	138.9	69.4	
-15.4	-30.9	-24.7	24.7	30.9	15.4	
-3.4	-6.8	-5.5	5.5	6.8	3.4	
-0.8	-1.5	-1.2	1.2	1.5	0.8	
-0.2	-0.3	0.6	-0.6	0.3	0.2	
	-89.2	-178.4	178.5	-178.5	178.4	89.2

Final Moments

For reactions, see solution of problem 16.20.

17.21



$$FEM_{AC} = \frac{75(6)}{8} = 56.3 \text{ k-ft}; FEM_{CA} = -56.3 \text{ k-ft}$$

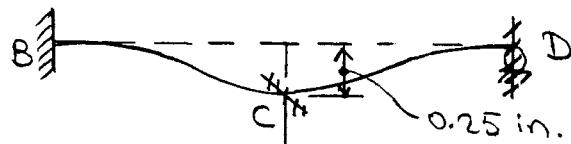
$$FEM_{CD} = \frac{25(9)^2}{12} + \frac{6(200)400(0.05)}{(9)^2} = 168.8 + 296.3 \\ = 465.1 \text{ k-ft}$$

$$FEM_{DC} = -168.8 + 296.3 = 128.2 \text{ k-ft}$$

	AC	CA	CD	DC
DF	$\frac{2}{3}$	$\frac{1}{3}$		
FEM	56.3	-56.3	465.1	128.2
	-272.5		-136.3	-128.2
	-136.3		-64.1	
		42.7	21.4	
	21.4			
Final Moments	-58.6	-286.1	286.1	0

For reactions, see solution of Problem 16-21

17.22



$$FEM_{AC} = FEM_{CA} = 0$$



$$FEM_{BC} = \frac{2(15)^2}{12} + \frac{6(29000)(3500)(0.25)}{(15)^2(12)^3} = 37.5 + 391.6 \\ = 429.1 \text{ k-ft}$$

$$FEM_{CB} = -37.5 + 391.6 = 354.1 \text{ k-ft}$$

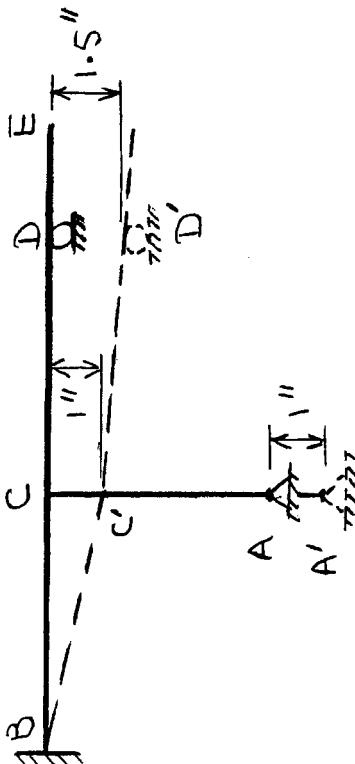
$$FEM_{CD} = FEM_{DC} = -391.6 \text{ k-ft}$$

Carry Over

	BC	CB	External Moment at Joint C	CA	CD	DC	AC
DF FEM	0.364	-	0.364	0.272			
429.1	354.1	100	0	-391.6	-391.6		0
-22.7			-22.7	-17.1	391.6		
-11.3				195.8			-11.3
	-71.3		-71.3	-53.2			
-35.7							-35.7
Final Moments	382.1	260.1	100	-94	-266.1	0	-47

For reactions, see solution of Problem 16.22.

17.23



$$FEM_{AC} = \frac{40(20)}{8} = 100 \text{ k-ft}; FEM_{CA} = -100 \text{ k-ft}$$

$$FEM_{BC} = \frac{2(30)^2}{12} + \frac{6(10000)(3000)(1)}{(30)^2 (12)^3} = 150 + 115.7 = 265.7 \text{ k-ft}$$

$$FEM_{CB} = -150 + 115.7 = -34.3 \text{ k-ft}; FEM_{CD} = \frac{6(10000)(3000)(0.5)}{(30)^2 (12)^3} = 57.9 \text{ k-ft}$$

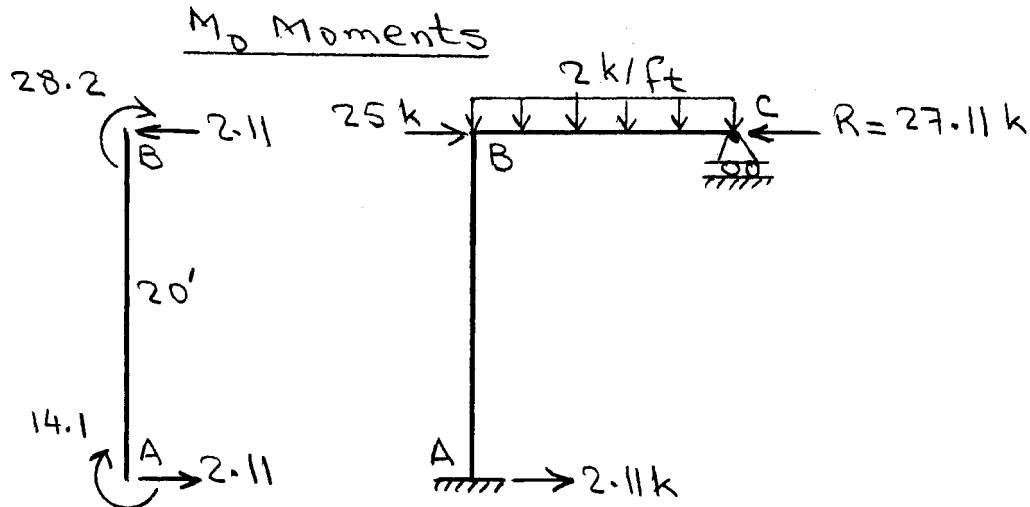
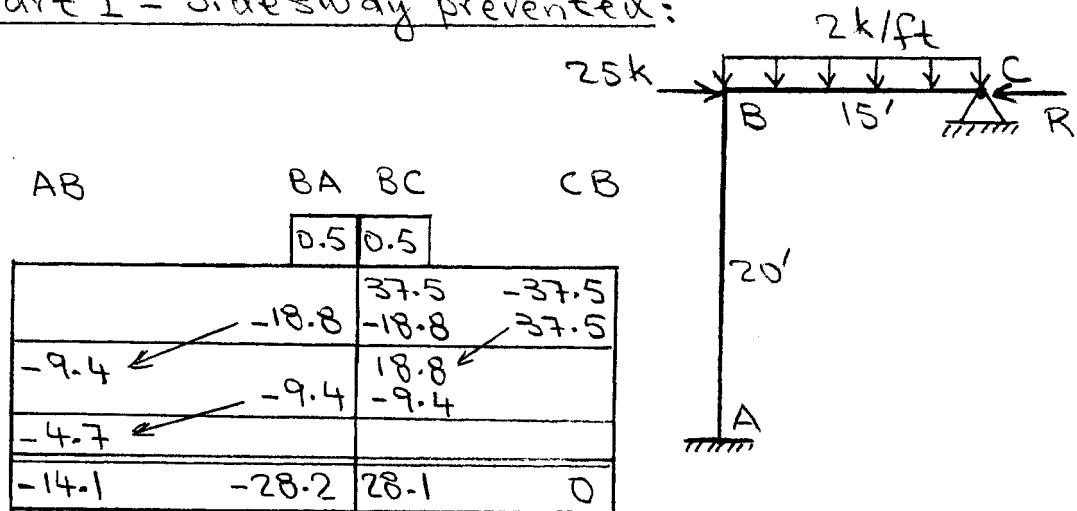
$$FEM_{DC} = 57.9 \text{ k-ft}; FEM_{DE} = 10(10) = 100 \text{ k-ft}.$$

	BC	CB	CA	CD	DC	DE	AC	
265.7	0.348	0.391	0.261		1	-	1	DF
-34.3	-34.3	-100	57.9	19.9	57.9	100	100	FEM
13.3	26.6	29.9	19.9	-7.9	-157.9	-100	-100	
22.4	44.9	50.4	33.7					
301.4	37.2	-69.7	32.5	-100	100	0	0	Final Moments

For reactions, see solution of problem 16.23.

$$17.24 \quad K_{AB} = \frac{I}{20}, \quad K_{BC} = \frac{I}{4} \left(\frac{I}{15} \right) = \frac{I}{20}$$

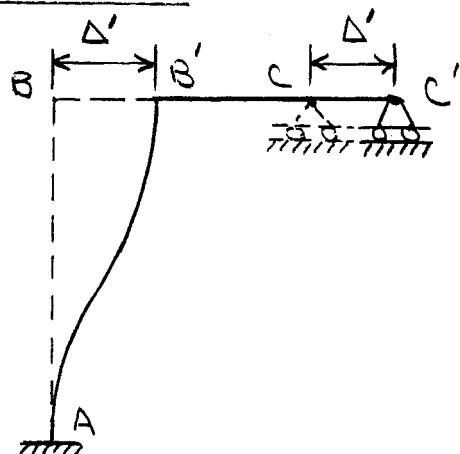
Part I - Sidesway prevented:



Part II - Sidesway permitted:

$$\begin{aligned} FEM_{AB} &= FEM_{BA} \\ &= \frac{6EI\Delta'}{(20)^2} \end{aligned}$$

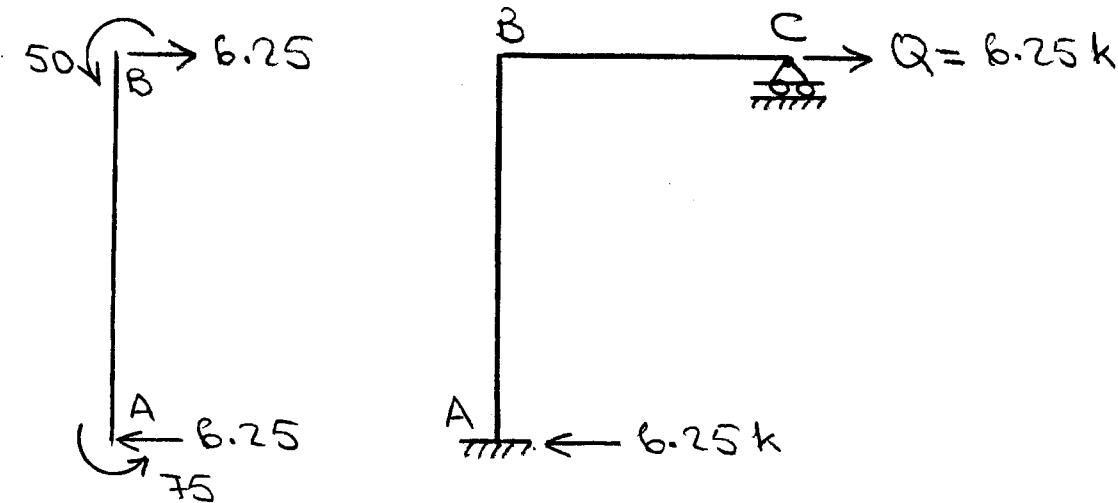
$$\begin{aligned} \text{Let } FEM_{AB} &= FEM_{BA} \\ &= 100 \text{ k-ft.} \end{aligned}$$



17.24 (Contd.)

AB	BA	BC	CB
	0.5	0.5	
100	100	-50	-50
-25 ←			
75	50	-50	0

M_Q Moments



Actual member end moments:

$$M_{AB} = -14.1 + \left(\frac{27.11}{6.25}\right)75 = \underline{\underline{311.2 \text{ k-ft}}}$$

$$M_{BA} = -28.2 + \left(\frac{27.11}{6.25}\right)50 = \underline{\underline{188.7 \text{ k-ft}}}$$

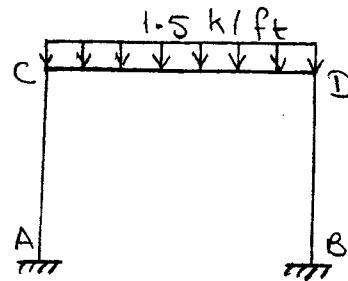
$$M_{BC} = 28.1 - \left(\frac{27.11}{6.25}\right)50 = \underline{\underline{-188.8 \text{ k-ft}}}$$

$$\underline{\underline{M_{CB} = 0}}$$

For reactions, see solution of Problem 16.24.

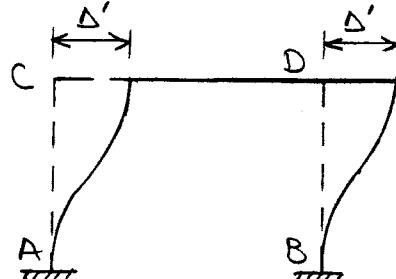
$$17.25 \quad K_{AC} = K_{BD} = \frac{I}{30}; \quad K_{CD} = \frac{2I}{40} = \frac{I}{20}$$

Part I - No Sidesway:



DF	AC	CA	CD	DC	DB	BD
FEM		0.4	0.6		0.6	0.4
0	0	200	-200	0	0	0
-80	-80	-120	120	80		
-40		60	-60		40	
-24		-36	36	24		
-12		18	-18		12	
-7.2		-10.8	10.8	7.2		
-3.6		5.4	-5.4		3.6	
-2.2		-3.2	3.2	2.2		
-1.1		1.6	-1.6		1.1	
-0.6		-1.0	1.0	0.6		
-0.3		0.5	-0.5		0.3	
-0.2		-0.3	0.3	0.2		
-0.1		0.2	-0.2		0.1	
	-0.1	-0.1	0.1	0.1		
M _o Moments	-57.1	-114.3	114.3	-114.3	114.3	57.1

Part II - Sidesway:



$$FEM_{CD} = FEM_{DC} = 0$$

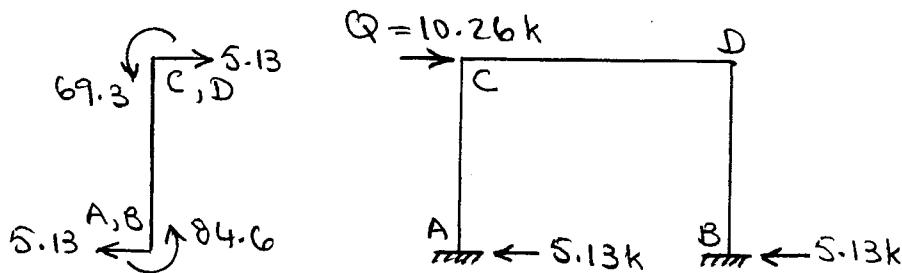
$$FEM_{AC} = FEM_{CA} = FEM_{BD} = FEM_{DB} = \frac{6EI\Delta'}{(30)^2}$$

$$\text{let } \frac{6EI\Delta'}{(30)^2} = 100 \text{ k-ft}$$

17.25 (contd.)

	AC	CA	CD	DC	DB	BD
DF	0.4	0.6		0.6	0.4	
FEM	100	100	0	0	100	100
-20		-60		-60		-20
12		18		18		12
6		9		9		6
-3.6		-5.4		-5.4		-3.6
-1.8		-2.7		-2.7		-1.8
1.1		1.6		1.6		1.1
0.6		0.8		0.8		0.6
-0.3		-0.5		-0.5		-0.3
-0.2		-0.3		-0.3		-0.2
0.1		0.2		0.2		0.1
84.6	69.3	-69.3	-69.3	69.3	84.6	

M_Q Moments



Actual member end moments: $M = M_0 + \left(\frac{20}{Q}\right) M_Q$

$$M_{AC} = 107.8 \text{ k-ft}; \quad M_{CA} = 20.8 \text{ k-ft};$$

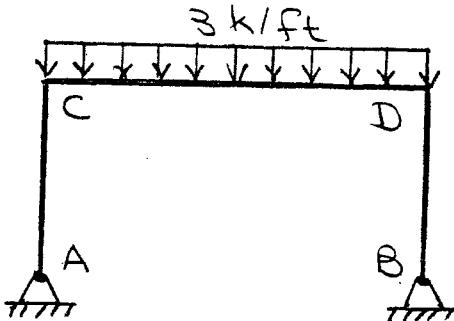
$$M_{CD} = -20.8 \text{ k-ft}; \quad M_{DC} = -249.4 \text{ k-ft};$$

$$M_{DB} = 249.4 \text{ k-ft}; \quad M_{BD} = 222 \text{ k-ft}$$

For reactions, see solution of Problem 13.45

$$17-26 \quad K_{AC} = K_{BD} = \frac{3}{4} \left(\frac{I}{15} \right) = \frac{I}{20}; \quad K_{CD} = \frac{I}{30}$$

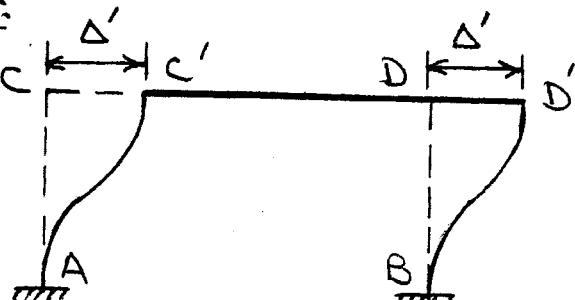
Part I - No Sidesway:



	AC	CA	CD	DC	DB	BD
DF FEM	0.6	0.4		0.4	0.6	
0	0	225	-225	0	135	0
-135	-135	-90	90	135	-135	
		45	-45			
-27	-27	-18	18	27		
		9	-9			
-5.4	-5.4	-3.6	3.6	5.4		
		1.8	-1.8			
-1.1	-1.1	-0.7	0.7	1.1		
		0.4	-0.4			
-0.2	-0.2	0.2	0.2	0.2		
M _o Moments	0	-168.7	168.7	-168.7	168.7	0

Part II - Sidesway:

$$FEM_{CD} = FEM_{DC} = 0$$

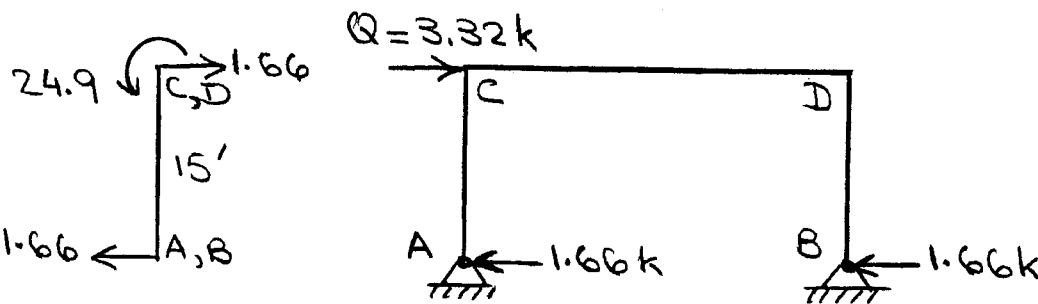


$$FEM_{AC} = FEM_{CA} = FEM_{BD} = FEM_{DB} = \frac{6EI\Delta'}{(15)^2}$$

$$\text{Let } \frac{6EI\Delta'}{(15)^2} = 100 \text{ k-ft.}$$

17.26 (Contd.)

	AC	CA	CD	DC	DB	BD
DF FEM	0.6	0.4		0.4	0.6	
100	100	0		0	100	100
-100	-60	-40		-40	-60	-100
	-50	-20		-20	-50	
	42	28		28	42	
	-8.4	14		14	-8.4	
		-5.6		-5.6		
	1.7	-2.8		-2.8	1.7	
		1.1		1.1		
	-0.4	0.6		0.6	-0.4	
		-0.2		-0.2		
M _Q Moments	0	24.9	-24.9	-24.9	24.9	0



Actual member end moments: $M = M_0 + \left(\frac{40}{Q}\right)M_Q$

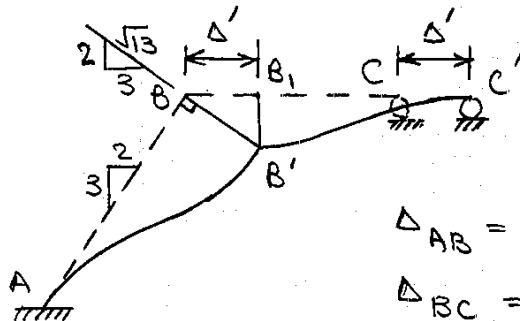
$$M_{AC} = M_{BD} = 0 ; \quad M_{CA} = 131.3 \text{ k-ft}$$

$$M_{CD} = -131.3 \text{ k-ft} ; \quad M_{DC} = -468.7 \text{ k-ft}$$

$$M_{DB} = 468.7 \text{ k-ft.}$$

For reactions, see solution of Problem 13.22.

$$17.27 \quad K_{AB} = \frac{I}{14.42}; \quad K_{BC} = \frac{3}{4} \left(\frac{I}{12} \right) = \frac{I}{16}$$



$$\Delta_{AB} = BB' = \frac{\sqrt{13}}{3} \Delta' = 1.202 \Delta'$$

$$\Delta_{BC} = B'C = \frac{2}{3} \Delta' = 0.667 \Delta'$$

$$FEM_{AB} = FEM_{BA} = \frac{6EI(1.202\Delta')}{(14.42)^2}$$

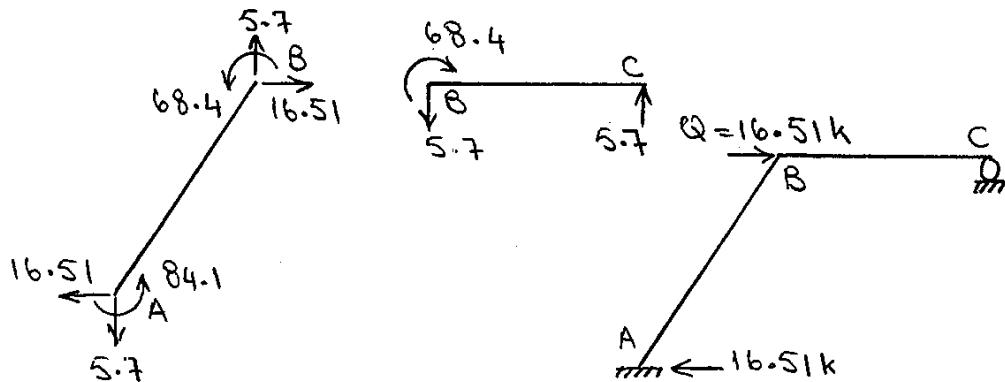
$$FEM_{BC} = FEM_{CB} = - \frac{6EI(0.667\Delta')}{(12)^2}$$

$$\text{Let } FEM_{AB} = FEM_{BA} = \frac{6EI(1.202\Delta')}{(14.42)^2} = 100 \text{ k-ft}$$

$$EI\Delta' = 2883.2$$

$$FEM_{BC} = FEM_{CB} = -80.1 \text{ k-ft}$$

	AB	BA	BC	CB
DF		0.526	0.474	
FEM	100	100	-80.1	-80.1
	-10.5	-9.4	40.1	80.1
-5.3		40.1		
	-21.1	-19		
-10.6				
M _Q Moments	84.1	68.4	-68.4	0



$$\text{Actual member end moments: } M = \frac{25}{16.51} M_Q$$

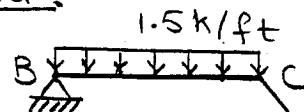
$$M_{AB} = 127.3 \text{ k-ft}; \quad M_{BA} = 103.6 \text{ k-ft};$$

$$M_{BC} = -103.6 \text{ k-ft}; \quad M_{CB} = 0$$

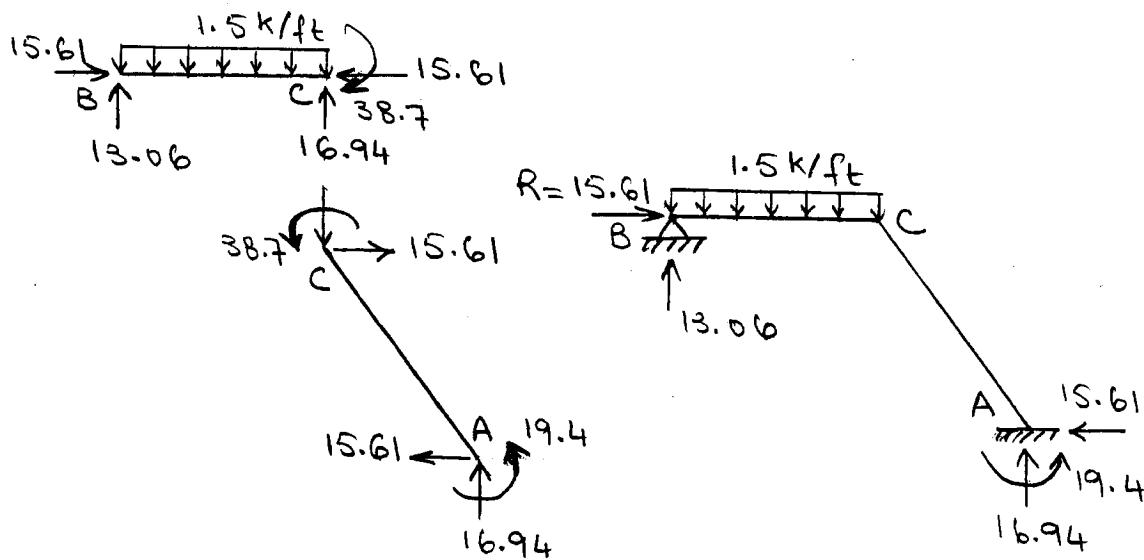
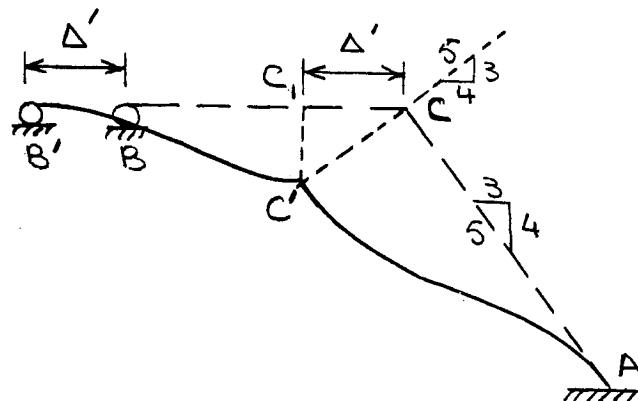
For reactions, see solution of Problem 16.27.

17.28

$$K_{CB} = \frac{3}{4} \left(\frac{I}{2} \right) = \frac{3I}{80}; \quad K_{CA} = \frac{I}{25}$$

Part I - Sidesway prevented:

	BC	CB	CA	AC
DF		0.484	0.516	
FEM	50 -50	-50 24.2	0 25.8	0
		-25 12.1	12.9	12.9
			6.5	6.5
M _o Moments	0	-38.7	38.7	19.4

Part II - Sidesway permitted:

$$\Delta_{BC} = \frac{3}{4} \Delta'$$

$$\Delta_{AC} = \frac{5}{4} \Delta'$$

17.28 (contd.)

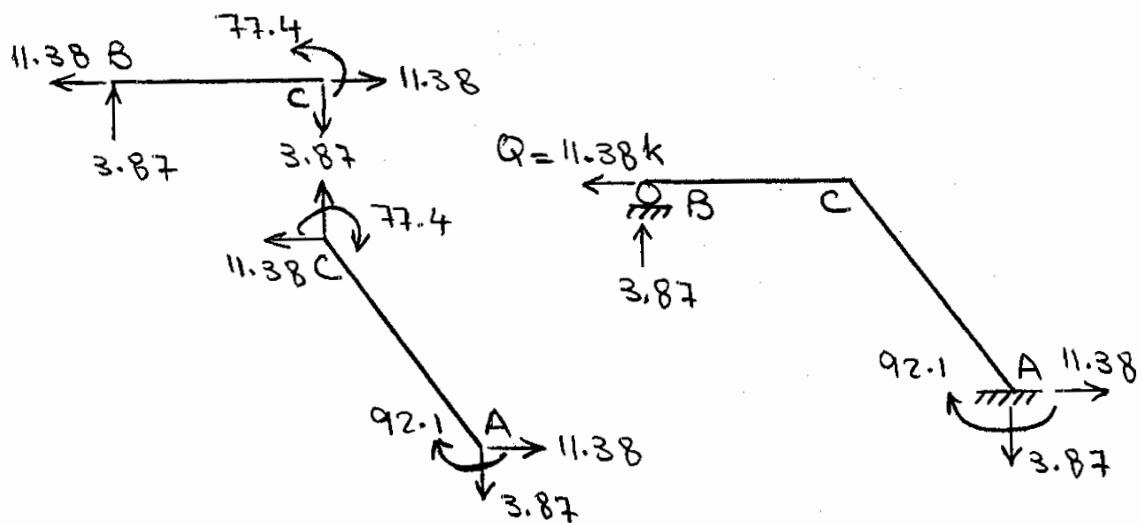
$$FEM_{BC} = FEM_{CB} = \frac{6EI(0.75\Delta')}{(20)^2}$$

Let $FEM_{BC} = FEM_{CB} = 100 \text{ k-ft}$

$$EI\Delta' = 8,888.89$$

$$FEM_{CA} = FEM_{AC} = -\frac{6EI(1.25\Delta')}{(25)^2} = -106.7 \text{ k-ft}$$

	BC	CB	CA	AC
DF		0.484	0.516	
FEM	100 -100	100 3.2	-106.7 3.5	-106.7
		-50 24.2	1.7 25.8	
				12.9
M _Q Moments	0	77.4	-77.4	-92.1

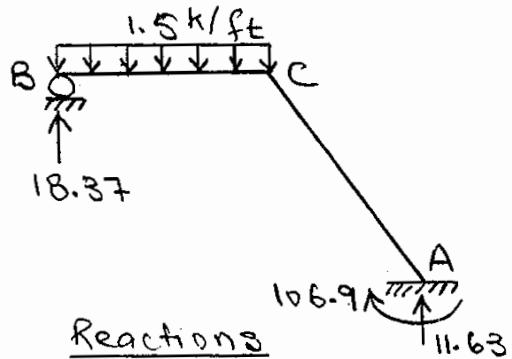


$$\text{Actual Moments: } M = M_0 + \left(\frac{R}{Q}\right) M_Q$$

$$M_{BC} = 0; \quad M_{CB} = 67.5 \text{ k-ft}$$

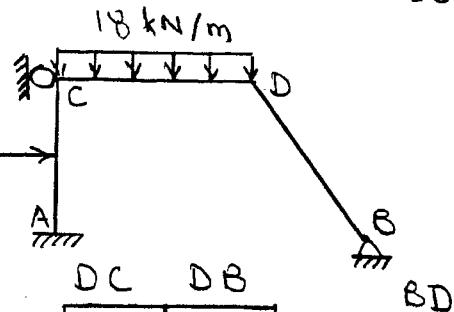
$$M_{CA} = -67.5 \text{ k-ft}$$

$$M_{AC} = -106.9 \text{ k-ft}$$

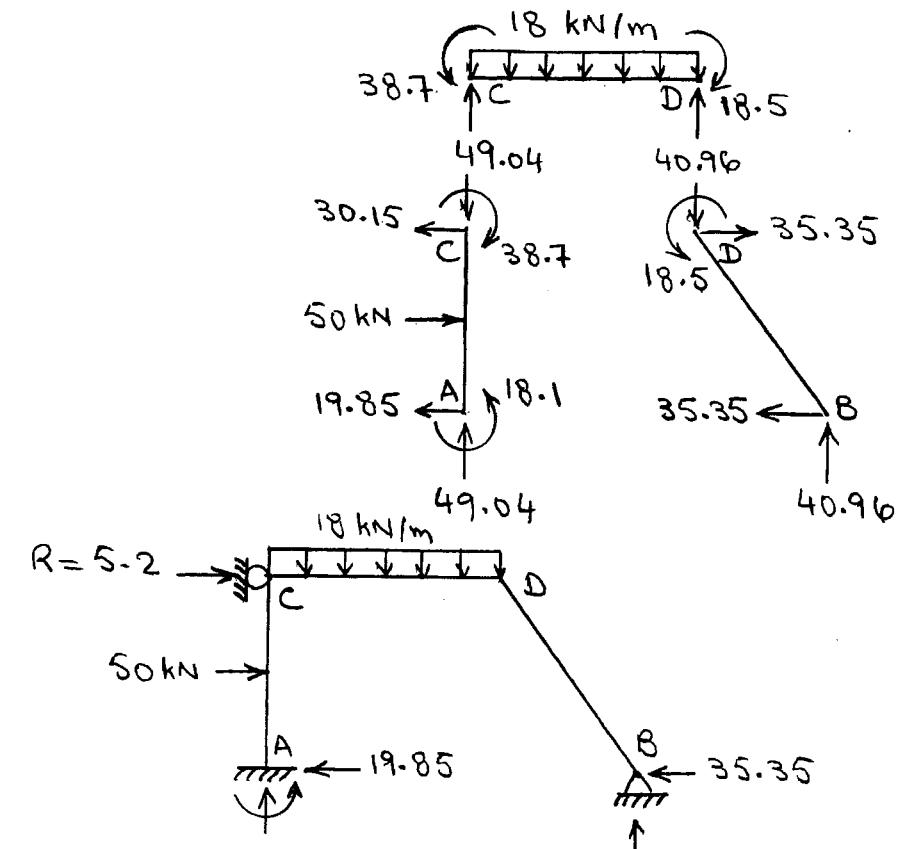


$$17.29 \quad k_{AC} = \frac{I}{4}; \quad k_{CD} = \frac{I}{5}; \quad k_{BD} = \frac{3}{4} \left(\frac{I}{5} \right) = \frac{3I}{20}$$

Part I - Sidesway prevented:

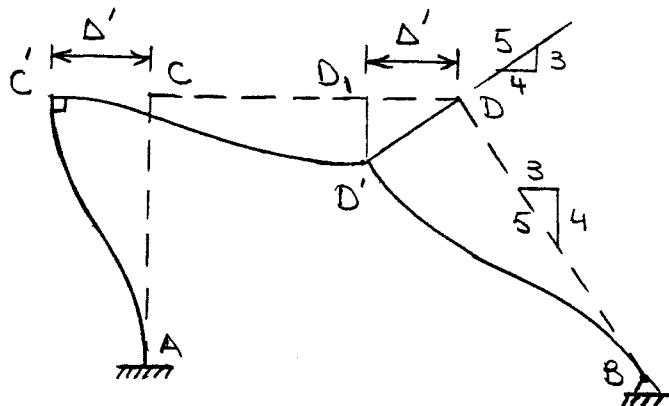


	AC	CA	CD	DC	DB	BD
DF FEM	0.556	0.444		0.571	0.429	
25	-2.5	37.5	-37.5	0	0	
	-7	-5.5	21.4	16.1		
-3.5		10.7	-2.8			
		-4.8	1.6	1.2		
-3	-5.9	0.8	-2.4			
	-0.4	-0.4	1.4	1.0		
-0.2		0.7	-0.2			
		-0.3	0.1	0.1		
-0.2	-0.4		-0.2			
			0.1	0.1		
M _o Moments	18.1	-38.7	38.7	-18.5	18.5	0



17.29 (contd.)

Part II - Sidesway permitted:



$$\Delta_{AC} = \Delta'$$

$$\Delta_{CD} = \frac{3}{4} \Delta' = 0.75 \Delta'$$

$$\Delta_{BD} = \frac{5}{4} \Delta' = 1.25 \Delta'$$

$$FEM_{AC} = FEM_{CA} = -\frac{6EI\Delta'}{(4)^2}$$

$$\text{let } FEM_{AC} = FEM_{CA} = -100 \text{ kN.m}$$

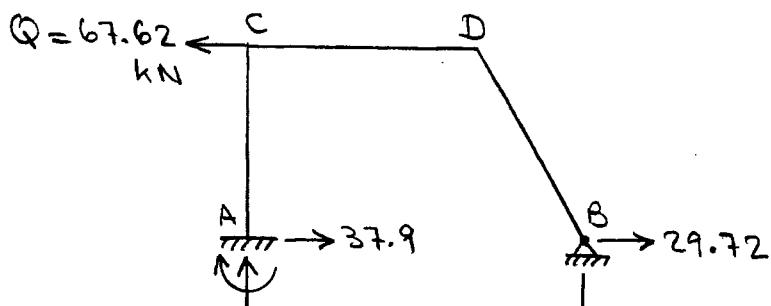
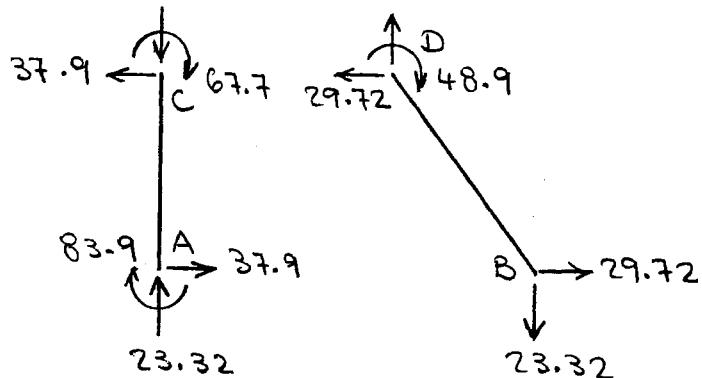
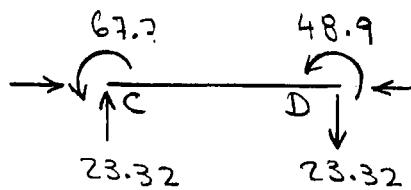
$$EI\Delta' = 266.7$$

$$FEM_{CD} = FEM_{DC} = \frac{6EI(0.75\Delta')}{(5)^2} = 48 \text{ kN.m}$$

$$FEM_{BD} = FEM_{DB} = -\frac{6EI(1.25\Delta')}{(5)^2} = -80 \text{ kN.m}$$

	AC	CA	CD	DC	DB	BD	
DF		0.556	0.444		0.571	0.429	
FEM	-100	-100	48	48	-80	-80	
		28.9	23.1	18.3	13.7	80	
	14.5		9.2	11.6	40		
		-5.1	-4.1	-29.5	-22.1		
	-2.6		-14.8	-2.1			
		8.2	6.6	1.2	0.9		
	4.1		0.6	3.3			
		-0.3	-0.3	-1.9	-1.4		
	-0.2		-1.0	-0.2			
		0.6	0.4	0.1	0.1		
	0.3			0.2			
				-0.1	-0.1		
M _Q Moments	-83.9	-67.7	67.7	48.9	-48.9	0	

17.29 (contd.)

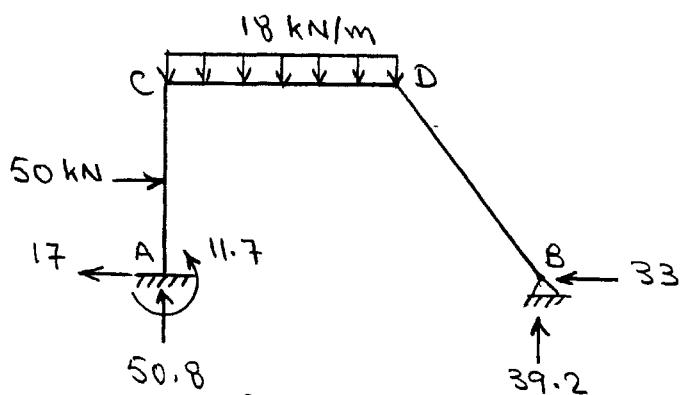


Actual member end moments: $M = M_0 + \left(\frac{R}{Q}\right)M_Q$

$$M_{AC} = 11.7 \text{ kN.m} ; M_{CA} = -43.9 \text{ kN.m}$$

$$M_{CD} = 43.9 \text{ kN.m} ; M_{DC} = -14.7 \text{ kN.m}$$

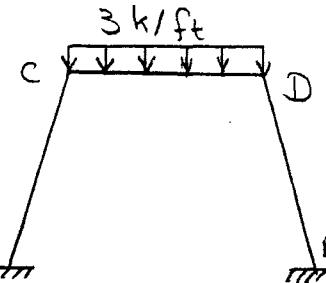
$$M_{DB} = 14.7 \text{ kN.m} ; M_{BD} = 0$$



Reactions

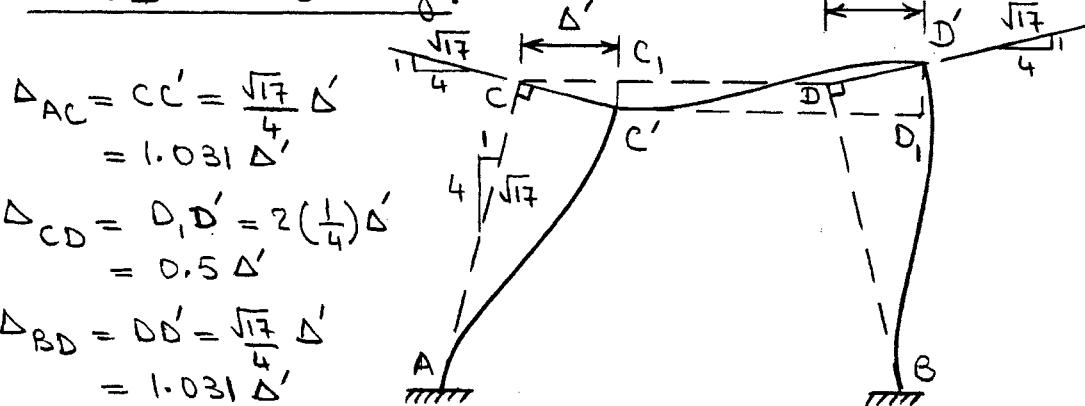
$$17.30 \quad K_{AC} = K_{BD} = \frac{I}{16.49} ; \quad K_{CD} = \frac{I}{16}$$

Part I - No Sidesway:



DF FEM	AC	CA	CD	DC	DB	BD
	0.492	0.508		0.508	0.492	
0	0	64		-64	0	0
-15.8	-31.5	-32.5		32.5	31.5	15.8
-4	-8	-8.3		8.3	8	4
-1.1	-2.1	-2.1		2.1	2.1	1.1
-0.3	-0.5	-0.6		0.6	0.5	0.3
	-0.1	-0.2		0.2	0.1	
M _o Moments	-21.2	-42.2	42.2	-42.2	42.2	21.2

Part II - Sidesway:



$$\Delta_{AC} = CC' = \frac{\sqrt{17}}{4} \Delta' \\ = 1.031 \Delta'$$

$$\Delta_{CD} = D_1 D'_1 = 2 \left(\frac{1}{4} \right) \Delta' \\ = 0.5 \Delta'$$

$$\Delta_{BD} = DD' = \frac{\sqrt{17}}{4} \Delta' \\ = 1.031 \Delta'$$

$$FEM_{AC} = FEM_{CA} = \frac{6EI(1.031\Delta')}{(16.49)^2}$$

$$\text{Let } FEM_{AC} = FEM_{CA} = 100 \text{ k-ft}$$

$$EI\Delta' = 4395.7$$

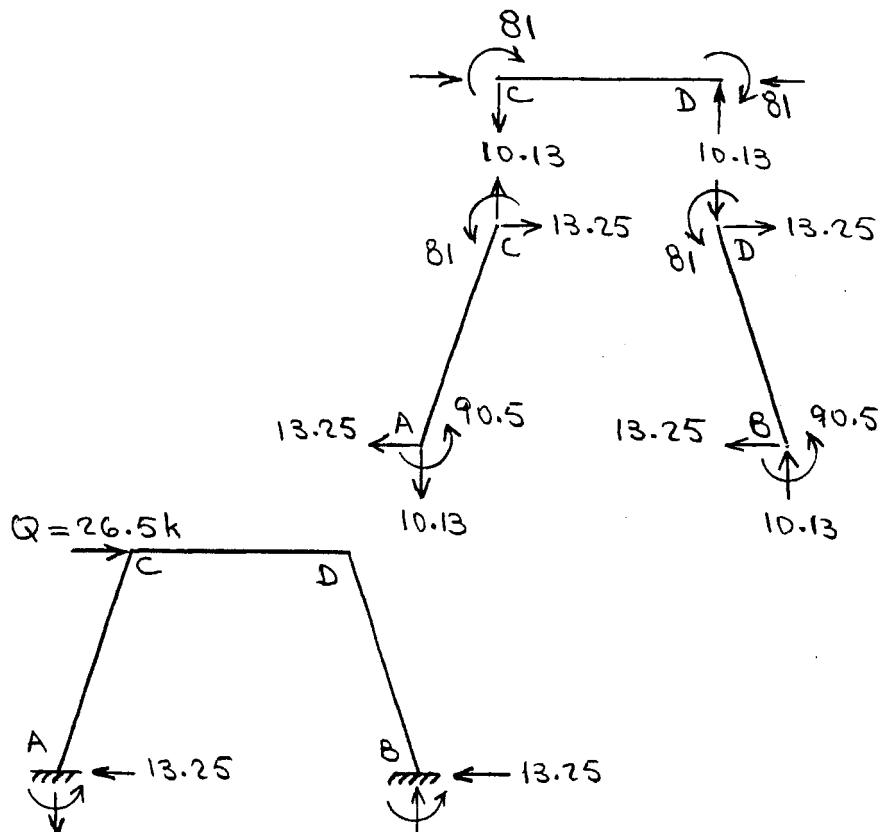
$$FEM_{BD} = FEM_{DB} = \frac{6EI(1.031\Delta')}{(16.49)^2} = 100 \text{ k-ft}$$

$$FEM_{CD} = FEM_{DC} = -\frac{6EI(0.5\Delta')}{(16)^2} = -51.5 \text{ k-ft.}$$

17.30 (contd.)

	AC	CA	CD	DC	DB	BD
DF FEM		0.492	0.508	0.508	0.492	
100	100	-51.5	-51.5	100	100	
-23.9	-24.6	-24.6	-24.6	-23.9	-12	
-12	6.1	6.2	6.2	6.1		
3.1		3.1	3.1		3.1	
-1.5		-1.6	-1.6	-1.5		
-0.8		-0.8	-0.8	-0.8		
0.4		0.4	0.4	0.4		
0.2		0.2	0.2	0.2		
-0.1		-0.1	-0.1	-0.1		
90.5	81	-81	-81	81	90.5	

M_Q Moments



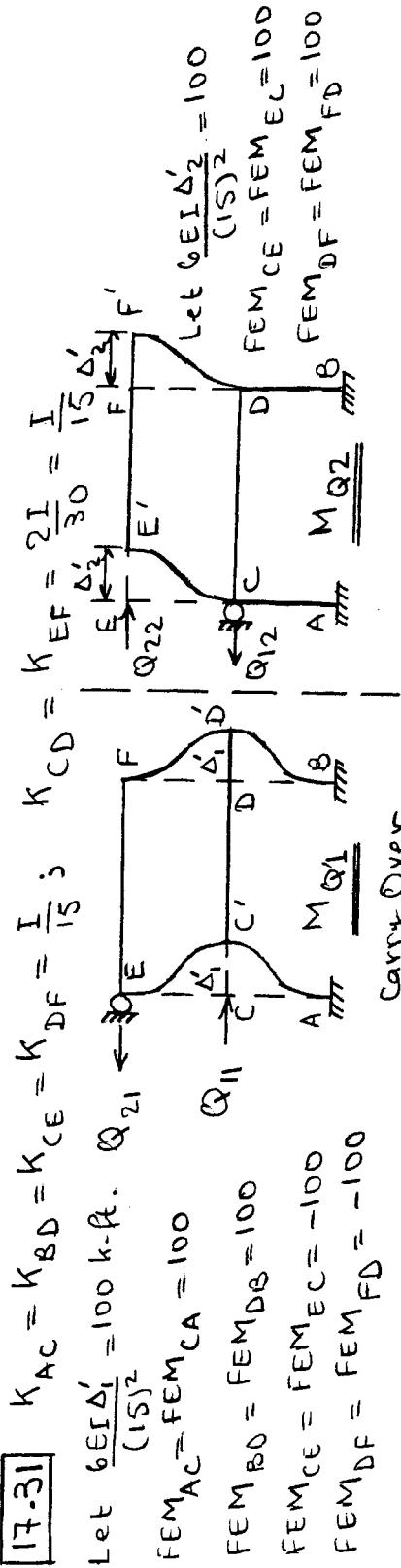
Actual member end moments: $M = M_0 + \left(\frac{20}{26.5}\right) M_Q$

$M_{AC} = 47.1 \text{ k-ft}$; $M_{CA} = 19 \text{ k-ft}$; $M_{CD} = -19 \text{ k-ft}$;

$M_{DC} = -103.3 \text{ k-ft}$; $M_{DB} = 103.3 \text{ k-ft}$; $M_{BD} = 89.5 \text{ k-ft}$.

For reactions, see solution of Problem 16.30.

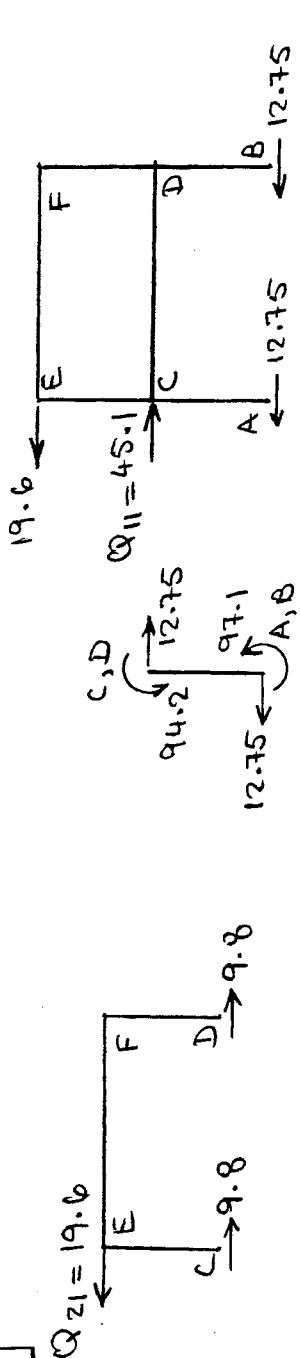
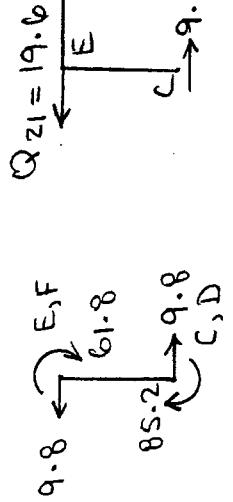
17.31



AC	CA			CD			CE			EF			FE			FD			BD		
	1/3	1/3	1/3	1/2	1/2	1/2	1/2	1/2	1/2	1/2	1/2	1/2	1/2	1/2	1/2	1/2	1/3	1/3	1/3		
100	100	0	-100	-100	0	0	-100	0	0	-100	0	-100	0	-100	0	0	100	100	100		
				50	50	50	50	50	50	50	50	50	50	50	50	50	50	50	50		
-8.4	-8.4	-8.3	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-12.5	-8.3	-8.3	-8.3			
-4.2	-4.2	-6.3	-4.2	-6.3	-6.3	-6.3	-6.3	-6.3	-6.3	-6.3	-6.3	-6.3	-6.3	-6.3	-6.3	-4.2	-4.2	-4.2			
3.5	3.5	3.5	5.3	5.3	5.3	5.3	5.3	5.3	5.3	5.3	5.3	5.3	5.3	5.3	5.3	3.5	3.5	3.5			
1.8	1.8	2.7	1.8	2.7	2.7	2.7	2.7	2.7	2.7	2.7	2.7	2.7	2.7	2.7	2.7	1.8	1.8	1.8			
-1.5	-1.5	-1.5	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-2.3	-1.5	-1.5	-1.5			
-0.8	-0.8	-1.2	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8	-0.8			
0.7	0.7	0.7	1	1	1	1	1	1	1	1	1	1	1	1	1	0.7	0.7	0.7			
0.4	0.4	0.5	0.4	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.4	0.4	0.4			
-0.3	-0.3	-0.3	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.5	-0.3	-0.3	-0.3			
-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2			
0.2	0.2	0.2	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.2	0.2	0.2			
0.1	0.1	0.2	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	-0.1	-0.1	-0.1		
-0.1	-0.1	-0.1	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2	0.1	0.1	0.1			
97.1	94.2	-8.8	-85.2	-61.8	61.7	61.7	-61.8	-61.8	-61.8	-61.8	-61.8	-61.8	-61.8	-61.8	-61.8	q7.1	q7.1	q7.1			

M_{Q1}
Moments

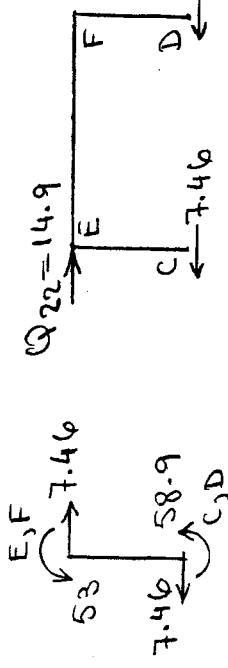
17-31 (contd.)



Carry Over

FEM	AC	CE			EF			FD			DB			BD
		1/3	1/3	1/2	1/2	1/2	1/2	1/3	1/3	1/3	1/3	1/3	1/3	
Q21	0	0	100	0	0	0	100	100	100	100	0	0	0	0
-33.3	-33.3	-33.3	-50	-50	-50	-50	-50	-50	-50	-50	-33.3	-33.3	-33.3	-33.3
-16.7	-16.7	-25	-16.7	-25	-25	-25	-16.7	-16.7	-16.7	-16.7	-16.7	-16.7	-16.7	-16.7
7	7	10.5	7	10.5	10.5	10.5	7	7	10.5	10.5	7	7	7	7
-5.8	-5.8	-5.8	-8.8	-8.8	-8.8	-8.8	-8.8	-8.8	-8.8	-8.8	-5.8	-5.8	-5.8	-5.8
-2.9	-2.9	-4.4	-2.9	-4.4	-4.4	-4.4	-2.9	-2.9	-4.4	-4.4	-2.9	-2.9	-2.9	-2.9
2.4	2.4	2.4	3.7	3.7	3.7	3.7	3.7	3.7	2.4	2.4	2.4	2.4	2.4	2.4
1.2	1.2	1.9	1.2	1.9	1.9	1.9	1.2	1.2	1.9	1.9	1.2	1.2	1.2	1.2
-1	-1	-1	-1.6	-1.6	-1.6	-1.6	-1.6	-1.6	-1	-1	-1	-1	-1	-1
-0.5	-0.5	-0.8	-0.5	-0.8	-0.8	-0.8	-0.5	-0.5	-0.8	-0.8	-0.5	-0.5	-0.5	-0.5
0.4	0.4	0.4	0.7	0.7	0.7	0.7	0.7	0.7	0.4	0.4	0.4	0.4	0.4	0.4
0.2	0.2	0.4	0.2	0.4	0.4	0.4	0.2	0.2	0.4	0.4	0.2	0.2	0.2	0.2
-0.2	-0.2	-0.2	-0.3	-0.3	-0.3	-0.3	-0.3	-0.3	-0.2	-0.2	-0.2	-0.2	-0.2	-0.2
-0.1	-0.1	-0.1	-0.1	-0.1	-0.1	-0.1	-0.1	-0.1	-0.2	-0.2	-0.1	-0.1	-0.1	-0.1
M _{Q2}	-11.8	-23.5	-35.4	58.9	53	-52.8	-52.8	53	58.9	-35.4	-23.5	-11.8	-11.8	-11.8
Moments														

17.31 (contd.)



By superimposing the horizontal forces at joints C and E, we write

$$45.1 c_1 - 19.6 c_2 = 18$$

$$-19.6 c_1 + 14.9 c_2 = 9$$

By solving these equations, we obtain $c_1 = 1.545$ and $c_2 = 2.636$.

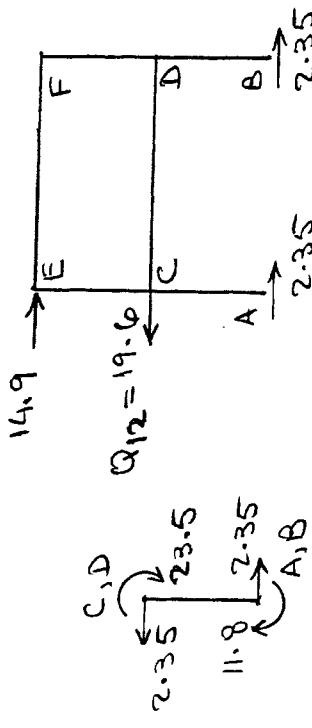
Member end moments: $M = c_1 M_{Q1} + c_2 M_{Q2}$

$$M_{AC} = M_{BD} = 119 \text{ k-ft}; \quad M_{CA} = M_{DB} = 83.5 \text{ k-ft}$$

$$M_{CD} = M_{DC} = -106.8 \text{ k-ft}; \quad M_{CE} = M_{DF} = 23.5 \text{ k-ft}$$

$$M_{EC} = M_{FD} = 44 \text{ k-ft}; \quad M_{EF} = M_{FE} = -44 \text{ k-ft}.$$

For reactions, see solution of Problem 16.31.



By superimposing the horizontal forces at joints C and E, we write

$$45.1 c_1 - 19.6 c_2 = 18$$

$$-19.6 c_1 + 14.9 c_2 = 9$$

$$M_{AC} = M_{BD} = 119 \text{ k-ft}; \quad M_{CA} = M_{DB} = 83.5 \text{ k-ft}$$

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$$M_{EC} = M_{FD} = 44 \text{ k-ft}; \quad M_{EF} = M_{FE} = -44 \text{ k-ft}.$$

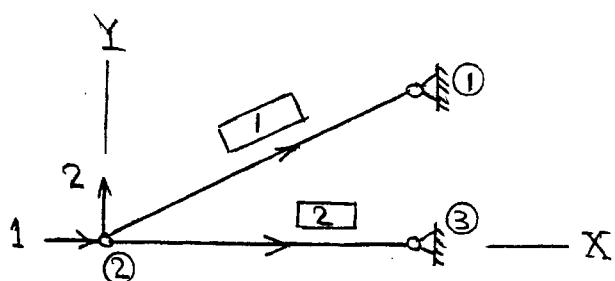
For reactions, see solution of Problem 16.31.

Chapter Eighteen

Introduction to Matrix Structural Analysis

CHAPTER 18

18.1



Member 1: $L = 11.18 \text{ ft}$; $\cos \theta = 0.894$; $\sin \theta = 0.447$

$$[K_1] = EA \begin{bmatrix} 1 & 2 & 0 & 0 \\ 0.0715 & 0.0357 & -0.0715 & -0.0357 \\ 0.0179 & -0.0357 & 0.0179 & -0.0179 \\ \text{sym.} & 0.0715 & 0.0357 & 0.0179 \\ & & & 0 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 0 \\ 0 \end{matrix}$$

Member 2: $L = 10 \text{ ft}$; $\cos \theta = 1$; $\sin \theta = 0$

$$[K_2] = EA \begin{bmatrix} 1 & 2 & 0 & 0 \\ 0.1 & 0 & -0.1 & 0 \\ 0 & 0 & 0 & 0 \\ \text{sym.} & 0.1 & 0 & 0 \\ & & 0 & 0 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 0 \\ 0 \end{matrix}$$

Structure stiffness matrix:

$$[S] = EA \begin{bmatrix} 1 & 2 \\ 0.1715 & 0.0357 \\ 0.0357 & 0.0179 \end{bmatrix} \begin{matrix} 1 \\ 2 \end{matrix}$$

Joint load vector: $\{P\} = \begin{bmatrix} 0 \\ -24 \end{bmatrix} \begin{matrix} 1 \\ k \end{matrix}$

Joint displacements: By solving the equations

$$\{P\} = [S]\{d\}, \text{ we obtain: } \{d\} = \frac{1}{EA} \begin{bmatrix} 477.26 \\ -2292.3 \end{bmatrix} \begin{matrix} 1 \\ 2 \end{matrix}$$

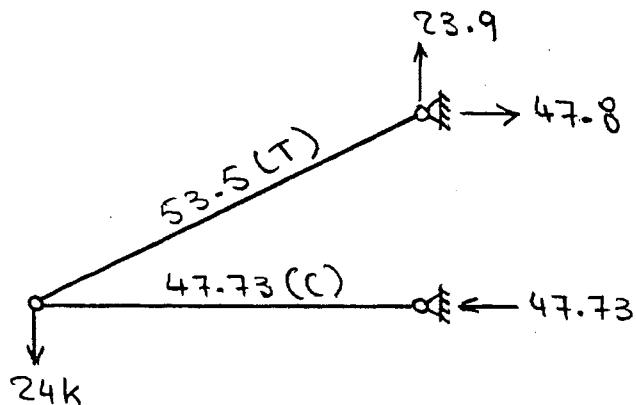
18.1 (Contd.) Member forces:

$$\{v_1\} = \frac{1}{EA} \begin{bmatrix} 477.26 \\ -2292.3 \\ 0 \\ 0 \end{bmatrix} \quad \{u_1\} = [T_1] \{v_1\} = \frac{1}{EA} \begin{bmatrix} -598 \\ 0 \end{bmatrix}$$

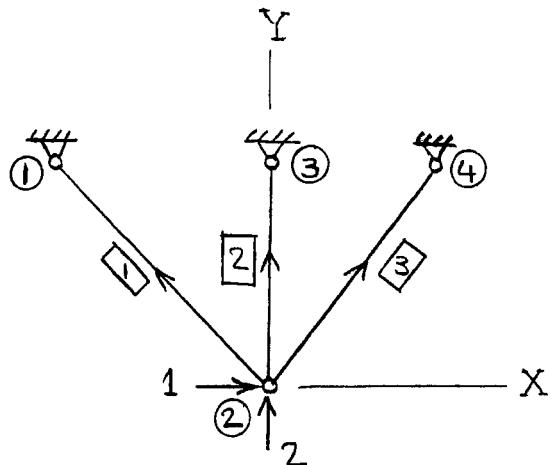
$$\{Q_1\} = [k_1] \{u_1\} = \begin{bmatrix} -53.5 \\ 53.5 \end{bmatrix} k ; \quad \{F_1\} = [T_1]^T \{Q_1\} = \begin{bmatrix} -47.8 \\ -23.9 \\ 47.8 \\ 23.9 \end{bmatrix} k$$

$$\{v_2\} = \frac{1}{EA} \begin{bmatrix} 477.26 \\ -2292.3 \\ 0 \\ 0 \end{bmatrix} \quad \{u_2\} = [T_2] \{v_2\} = \frac{1}{EA} \begin{bmatrix} 477.26 \\ 0 \end{bmatrix}$$

$$\{Q_2\} = [k_2] \{u_2\} = \begin{bmatrix} 47.73 \\ -47.73 \end{bmatrix} k ; \quad \{F_2\} = [T_2]^T \{Q_2\} = \begin{bmatrix} 47.73 \\ 0 \\ -47.73 \\ 0 \end{bmatrix} k$$



18.2.



Member 1: $L = 271.53 \text{ in.}$, $\cos\theta = -0.707$; $\sin\theta = 0.707$

$$[K_1] = 147.3$$

$$\begin{bmatrix} 1 & 2 & 0 & 0 \\ 1 & -1 & -1 & 1 \\ 1 & 1 & -1 & 2 \\ \text{sym.} & 1 & -1 & 0 \\ & 1 & 0 \end{bmatrix}$$

Member 2: $L = 192 \text{ in.}$; $\cos\theta = 0$; $\sin\theta = 1$.

$$[K_2] = 312.5$$

$$\begin{bmatrix} 1 & 2 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ 1 & 0 & -1 & 2 \\ 0 & 0 & 0 & 0 \\ & 1 & 0 \end{bmatrix}$$

Member 3: $L = 240 \text{ in.}$; $\cos\theta = 0.6$; $\sin\theta = 0.8$

$$[K_3] = 333.3$$

$$\begin{bmatrix} 1 & 2 & 0 & 0 \\ 0.36 & 0.48 & -0.36 & -0.48 \\ 0.64 & -0.48 & -0.64 & 0 \\ 0.36 & 0.48 & 0 & 0 \\ 0.64 & 0 & 0 & 0 \end{bmatrix}$$

Structure stiffness matrix:

$$[S] = \begin{bmatrix} 1 & 2 \\ 267.3 & 12.7 \\ 12.7 & 673.1 \end{bmatrix}$$

Joint load vector: $\{F\} = \begin{bmatrix} 20 \\ -50 \end{bmatrix} k$

18.2 (Contd.) Joint displacements: By solving the equations $\{P\} = [S]\{d\}$, we obtain:

$$\{d\} = \begin{bmatrix} 0.07842 \\ -0.07576 \end{bmatrix} \text{ in.}$$

Member forces:

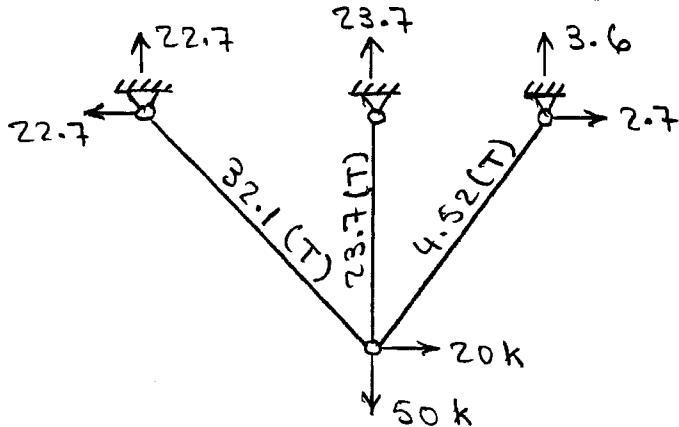
$$\{v_1\} = \{v_2\} = \{v_3\} = \begin{bmatrix} 0.07842 \\ -0.07576 \\ 0 \\ 0 \end{bmatrix} \text{ in.}$$

$$\{u_1\} = [T_1] \{v_1\} = \begin{bmatrix} -0.109 \\ 0 \end{bmatrix} \text{ in.}; \quad \{Q_1\} = [k_1] \{u_1\} = \begin{bmatrix} -32.1 \\ 32.1 \end{bmatrix} \text{ k}$$

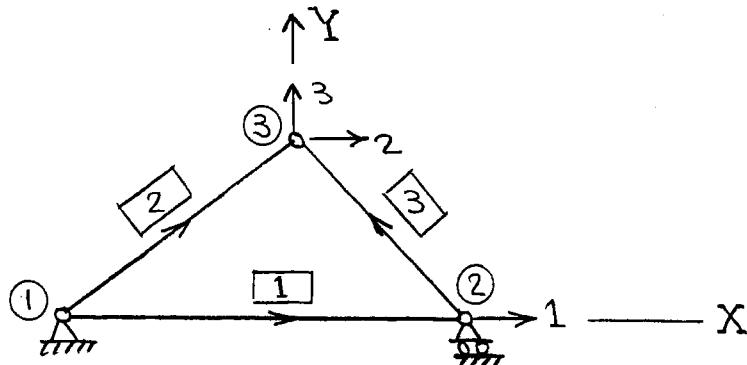
$$\{F_1\} = [T]^T \{Q_1\} = \begin{bmatrix} 22.7 \\ -22.7 \\ -22.7 \\ 22.7 \end{bmatrix} \text{ k} \quad \{u_2\} = \begin{bmatrix} 0 \\ -0.07576 \\ 0 \end{bmatrix} \text{ in.}$$

$$\{Q_2\} = \begin{bmatrix} -23.7 \\ 23.7 \end{bmatrix} \text{ k}; \quad \{F_2\} = \begin{bmatrix} 0 \\ -23.7 \\ 0 \\ 23.7 \end{bmatrix} \text{ k}$$

$$\{u_3\} = \begin{bmatrix} -0.01356 \\ 0 \end{bmatrix} \text{ in.}; \quad \{Q_3\} = \begin{bmatrix} -4.52 \\ 4.52 \end{bmatrix} \text{ k}; \quad \{F_3\} = \begin{bmatrix} -2.71 \\ -3.61 \\ 2.71 \\ 3.61 \end{bmatrix} \text{ k}$$



18.3



Member 1: $L = 14 \text{ m}$; $\cos \theta = 1$; $\sin \theta = 0$

$$[K_1] = \frac{EA}{14} \begin{bmatrix} 0 & 0 & 1 & 0 \\ 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1 \end{bmatrix}$$

Member 2: $L = 10 \text{ m}$; $\cos \theta = 0.8$; $\sin \theta = 0.6$

$$[K_2] = \frac{EA}{10} \begin{bmatrix} 0.64 & 0.48 & -0.64 & -0.48 \\ 0.48 & 0.36 & -0.48 & -0.36 \\ -0.64 & -0.48 & 0.64 & 0.48 \\ -0.48 & -0.36 & 0.48 & 0.36 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 2 \\ 3 \end{bmatrix}$$

Member 3: $L = 8.485 \text{ m}$; $\cos \theta = -0.707$; $\sin \theta = 0.707$

$$[K_3] = \frac{EA}{16.97} \begin{bmatrix} 1 & 0 & 2 & 3 \\ 1 & -1 & -1 & 1 \\ -1 & 1 & 1 & -1 \\ -1 & 1 & 1 & -1 \end{bmatrix} \begin{bmatrix} 1 \\ 1 \\ 2 \\ 3 \end{bmatrix}$$

Structure stiffness matrix:

$$[S] = EA \begin{bmatrix} 1 & 2 & 3 \\ 0.1304 & -0.05893 & 0.05893 \\ -0.05893 & 0.1229 & -0.01093 \\ 0.05893 & -0.01093 & 0.09493 \end{bmatrix} \begin{bmatrix} 1 \\ 2 \\ 3 \end{bmatrix}$$

Joint load vector:

$$\{\bar{P}\} = \begin{bmatrix} 0 \\ 80 \\ -120 \end{bmatrix} \begin{bmatrix} 1 \\ 2 \\ 3 \end{bmatrix} \text{ kN}$$

18.3 (contd.) Joint displacements: By solving the equations $\{P\} = [S]\{\delta\}$, we obtain:

$$\{\delta\} = \frac{1}{EA} \begin{bmatrix} 1439 \\ 1161 \\ -2024 \end{bmatrix}^T$$

Member forces:

$$\{v_1\} = \frac{1}{EA} \begin{bmatrix} 0 \\ 0 \\ 1439 \\ 0 \end{bmatrix}^T \quad \{u_1\} = [T_1] \{v_1\} = \frac{1}{EA} \begin{bmatrix} 0 \\ 1439 \end{bmatrix}^T$$

$$\{F_1\} = \begin{bmatrix} -102.8 \\ 0 \\ 102.8 \\ 0 \end{bmatrix}^T \text{ kN} ; \quad \{Q_1\} = [k_1] \{v_1\} = \begin{bmatrix} -102.8 \\ 102.8 \end{bmatrix}^T \text{ kN}$$

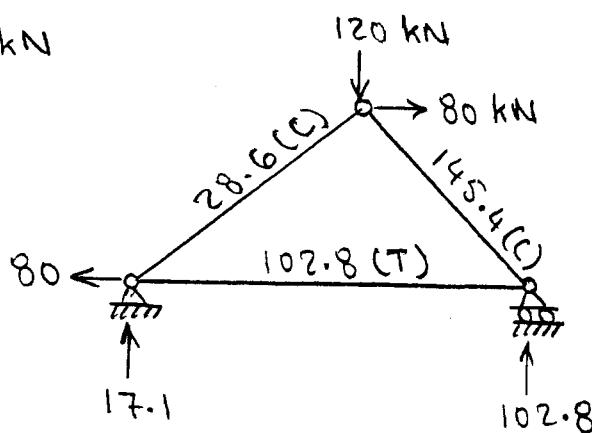
$$\{v_2\} = \frac{1}{EA} \begin{bmatrix} 0 \\ 0 \\ 1161 \\ -2024 \end{bmatrix}^T \quad \{u_2\} = \frac{1}{EA} \begin{bmatrix} 0 \\ 0 \\ 1161 \\ -2024 \end{bmatrix}^T$$

$$\{u_2\} = \frac{1}{EA} \begin{bmatrix} 0 \\ -285.6 \end{bmatrix}^T ; \quad \{Q_2\} = \begin{bmatrix} 28.56 \\ -28.56 \end{bmatrix}^T \text{ kN}$$

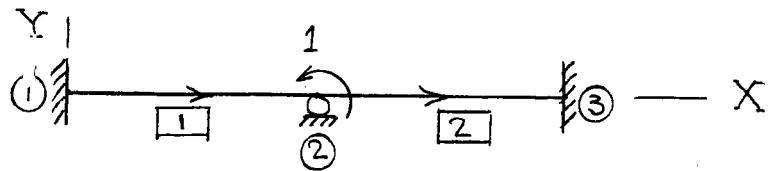
$$\{F_2\} = \begin{bmatrix} 22.85 \\ 17.14 \\ -22.85 \\ -17.14 \end{bmatrix}^T \text{ kN} ; \quad \{v_3\} = \frac{1}{EA} \begin{bmatrix} 1439 \\ 0 \\ 1161 \\ -2024 \end{bmatrix}^T$$

$$\{u_3\} = \frac{1}{EA} \begin{bmatrix} -1017.4 \\ -2251.8 \end{bmatrix}^T ; \quad \{Q_3\} = \begin{bmatrix} 145.4 \\ -145.4 \end{bmatrix}^T \text{ kN}$$

$$\{F_3\} = \begin{bmatrix} -102.8 \\ 102.8 \\ 102.8 \\ -102.8 \end{bmatrix}^T \text{ kN}$$



18.4

Member 2 \rightarrow 0 - 1 - 0 - 0Member 1 \rightarrow 0 0 0 1

$$[K_1] = [K_2] = \frac{EI}{L^3}$$

$$\begin{bmatrix} 12 & 180 & -12 & 180 \\ 3600 & -180 & 1800 & 0 \\ \text{sym.} & 12 & -180 & 0 \\ 3600 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 0 & 1 & 0 \\ 0 & 1 & 1 \\ 0 & 0 & 0 \\ 1 & 0 & 0 \end{bmatrix}$$

$$\{F_{f1}\} = \{Q_{f1}\} = \begin{bmatrix} 4.67 \\ 40 \\ 13.33 \\ -80 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1 \end{bmatrix}$$

$$\{F_{f2}\} = \{Q_{f2}\} = \begin{bmatrix} 5 \\ 37.5 \\ 5 \\ -37.5 \end{bmatrix} \begin{bmatrix} 0 \\ 1 \\ 0 \\ 0 \end{bmatrix}$$

$$[S] = \frac{EI}{L^3} [7200] \begin{bmatrix} 1 \end{bmatrix}; \quad \{P_f\} = [-42.5] \begin{bmatrix} 1 \end{bmatrix}$$

Joint displacements: $[S] \{d\} = -\{P_f\}$

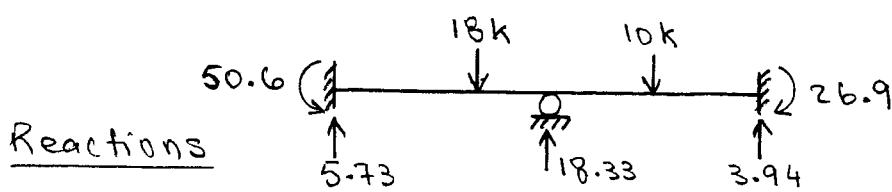
$$\{d\} = \frac{L^3}{EI} [0.005903]$$

Member forces:

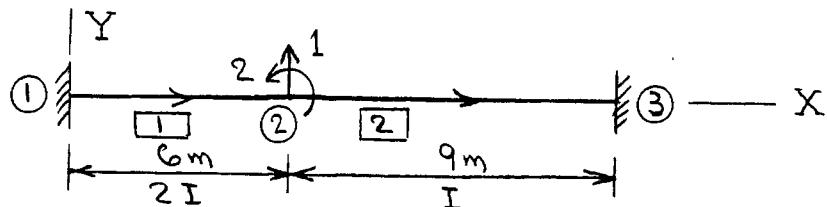
$$\{u_1\} = \{v_1\} = \frac{L^3}{EI} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0.005903 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1 \end{bmatrix}; \quad \{u_2\} = \{v_2\} = \frac{L^3}{EI} \begin{bmatrix} 0 \\ 0.005903 \\ 0 \\ 0 \end{bmatrix} \begin{bmatrix} 0 \\ 1 \\ 0 \\ 0 \end{bmatrix}$$

$$\{F_1\} = \{Q_1\} = [k_1] \{u_1\} + \{Q_{f1}\} = \begin{bmatrix} 5.73 \text{ k} \\ 50.6 \text{ k-ft} \\ 12.27 \text{ k} \\ -58.8 \text{ k-ft} \end{bmatrix}$$

$$\{F_2\} = \{Q_2\} = \begin{bmatrix} 6.06 \text{ k} \\ 58.8 \text{ k-ft} \\ 3.94 \text{ k} \\ -26.9 \text{ k-ft} \end{bmatrix}$$



18.5

Member 1:

$$[K_1] = EI$$

$$\begin{bmatrix} 0 & 0 & 1 & 2 \\ 0.111 & 0.333 & -0.111 & 0.333 \\ & 1.333 & -0.333 & 0.667 \\ & & 0.111 & -0.333 \\ & & & 1.333 \end{bmatrix}$$

sym.

Member 2:

$$[K_2] = EI$$

$$\begin{bmatrix} 1 & 2 & 0 & 0 \\ 0.0165 & 0.0741 & -0.0165 & 0.0741 \\ & 0.444 & -0.0741 & 0.222 \\ & & 0.0165 & -0.0741 \\ & & & 0.444 \end{bmatrix}$$

sym.

Structure stiffness matrix:

$$[S] = EI \begin{bmatrix} 1 & 2 \\ 0.1275 & -0.2589 \\ -0.2589 & 1.777 \end{bmatrix}$$

Joint load vector:

$$\{\bar{P}\} = \begin{bmatrix} -150 \\ 0 \end{bmatrix}$$

Joint displacements: By solving the equations

$$\{\bar{P}\} = [S] \{\bar{d}\}, \text{ we obtain: } \{\bar{d}\} = \frac{1}{EI} \begin{bmatrix} -1671 \\ -243.4 \end{bmatrix}$$

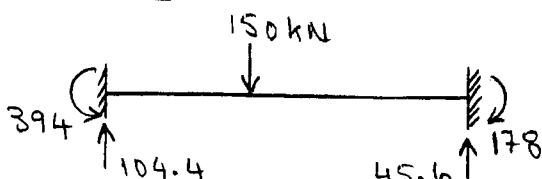
Member forces:

$$\{\bar{u}_1\} = \{\bar{v}_1\} = \frac{1}{EI} \begin{bmatrix} 0 \\ 0 \\ -1671 \\ -243.4 \end{bmatrix}$$

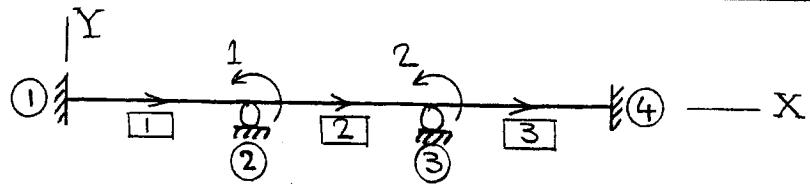
$$\{\bar{u}_2\} = \{\bar{v}_2\} = \frac{1}{EI} \begin{bmatrix} -1671 \\ -243.4 \\ 0 \\ 0 \end{bmatrix}$$

$$\{\bar{F}_1\} = \{\bar{Q}_1\} = [k_1] \{\bar{u}_1\} = \begin{bmatrix} 104.4 \text{ kN} \\ 394 \text{ kN.m} \\ -104.4 \text{ kN} \\ 232 \text{ kN.m} \end{bmatrix}$$

$$\{\bar{F}_2\} = \{\bar{Q}_2\} = \begin{bmatrix} -45.6 \text{ kN} \\ -232 \text{ kN.m} \\ 45.6 \text{ kN} \\ -178 \text{ kN.m} \end{bmatrix}$$

Reactions

18.6



Member 3 → 0 2 0 0

Member 2 → 0 1 0 2

Member 1 → 0 0 0 1

$$[K_1] = [K_2] = [K_3] = \frac{EI}{L^3} \begin{bmatrix} 12 & 108 & -12 & 108 \\ 1296 & -108 & 648 & 0 \\ & 12 & -108 & 0 \\ & 1296 & 0 & 1 \end{bmatrix} \begin{array}{l} \\ \\ \text{Sym.} \\ \end{array} \begin{array}{l} \\ \\ \\ \end{array}$$

$$\{F_{P1}\} = \{Q_{P1}\} = \begin{bmatrix} 27 \\ 81 \\ 27 \\ -81 \end{bmatrix} ; \quad \{F_{P3}\} = \{Q_{P3}\} = \begin{bmatrix} 13.5 \\ 40.5 \\ 13.5 \\ -40.5 \end{bmatrix}$$

Structure stiffness matrix:

$$[S] = \frac{EI}{L^3} \begin{bmatrix} 2592 & 648 \\ 648 & 2592 \end{bmatrix} \begin{array}{l} \\ \end{array} \quad \{P_f\} = \begin{bmatrix} -81 \\ 40.5 \end{bmatrix}$$

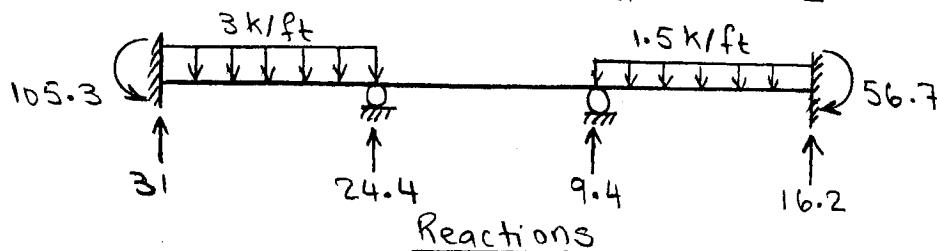
Joint displacements: By solving the equations

$$-\{P_f\} = [S]\{d\}, \text{ we obtain: } \{d\} = \frac{L^3}{EI} \begin{bmatrix} 0.0375 \\ -0.025 \end{bmatrix}$$

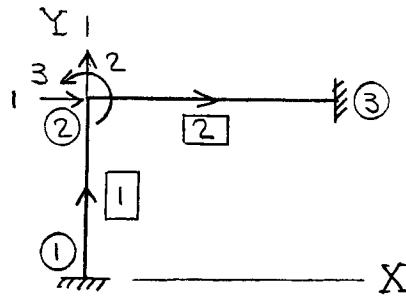
Member forces:

$$\{Q_1\} = \frac{L^3}{EI} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0.0375 \end{bmatrix} ; \quad \{Q_2\} = \frac{L^3}{EI} \begin{bmatrix} 0 \\ 0.0375 \\ 0 \\ -0.025 \end{bmatrix} ; \quad \{Q_3\} = \frac{L^3}{EI} \begin{bmatrix} 0 \\ -0.025 \\ 0 \\ 0 \end{bmatrix}$$

$$\{Q_1\} = \begin{bmatrix} 31k \\ 105.3 \text{ k-ft} \\ 23k \\ -32.4 \text{ k-ft} \end{bmatrix} ; \quad \{Q_2\} = \begin{bmatrix} 1.4k \\ 32.4 \text{ k-ft} \\ -1.4k \\ -8.1 \text{ k-ft} \end{bmatrix} ; \quad \{Q_3\} = \begin{bmatrix} 10.8k \\ 8.1 \text{ k-ft} \\ 16.2k \\ -56.7 \text{ k-ft} \end{bmatrix}$$



18.7



Member 1: $L = 15 \text{ ft}$; $\cos\theta = 0$; $\sin\theta = 1$

$$[k_1] = EI \times 10^3 \begin{bmatrix} 115.2 & 0 & 0 & -115.2 & 0 & 0 \\ 0 & 3.556 & 26.67 & 0 & -3.556 & 26.67 \\ 0 & 26.67 & 266.7 & 0 & -26.67 & 133.3 \\ \text{sym.} & & & 115.2 & 0 & 0 \\ & & & 0 & 3.556 & -26.67 \\ & & & & & 266.7 \end{bmatrix}$$

$$\{Q_{f1}\} = \begin{bmatrix} 0 \\ 15 \\ 37.5 \\ 0 \\ 15 \\ -37.5 \end{bmatrix}; [T_1] = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ -1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & -1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix}; \{F_{f1}\} = \begin{bmatrix} -15 \\ 0 \\ 0 \\ 37.5 \\ -15 \\ 1 \\ 0 \\ 2 \\ -37.5 \\ 3 \end{bmatrix}$$

$$[K_1] = [T_1]^T [k_1] [T_1] = EI \times 10^3 \begin{bmatrix} 0 & 0 & 0 & 1 & 2 & 3 \\ 3.556 & 0 & -26.67 & -3.556 & 0 & -26.67 \\ 0 & 115.2 & 0 & 0 & -115.2 & 0 \\ 266.7 & 26.67 & 266.7 & 0 & 0 & 133.3 \\ 0 & 3.556 & 0 & 26.67 & 1 \\ \text{sym.} & & & 115.2 & 0 & 2 \\ & & & & 266.7 & 3 \end{bmatrix}$$

Member 2: $L = 20 \text{ ft}$; $\cos\theta = 1$; $\sin\theta = 0$

$$[k_2] = [K_2] = EI \times 10^3 \begin{bmatrix} 1 & 2 & 3 & 0 & 0 & 0 \\ 86.4 & 0 & 0 & -86.4 & 0 & 0 \\ 1.5 & 15 & 0 & 0 & -1.5 & 15 \\ 200 & 0 & 0 & -15 & 100 & 0 \\ 86.4 & 0 & 0 & 0 & 0 & 0 \\ \text{sym.} & & & 1.5 & -15 & 0 \\ & & & & 200 & 0 \end{bmatrix}$$

$$\{F_{f2}\} = \{Q_{f2}\} = \begin{bmatrix} 0 \\ 10 \\ 50 \\ 0 \\ 10 \\ -50 \end{bmatrix}$$

18.7 (contd.) Structure matrices:

$$[S] = EI \times 10^{-3} \begin{bmatrix} 89.96 & 0 & 26.67 \\ 0 & 116.7 & 15 \\ 26.67 & 15 & 466.7 \end{bmatrix} ; \quad \{P_f\} = \begin{bmatrix} -15 \\ 10 \\ 12.5 \end{bmatrix}$$

Joint displacements: By solving the equations

$$-\{P_f\} = [S]\{d\}, \text{ we obtain:}$$

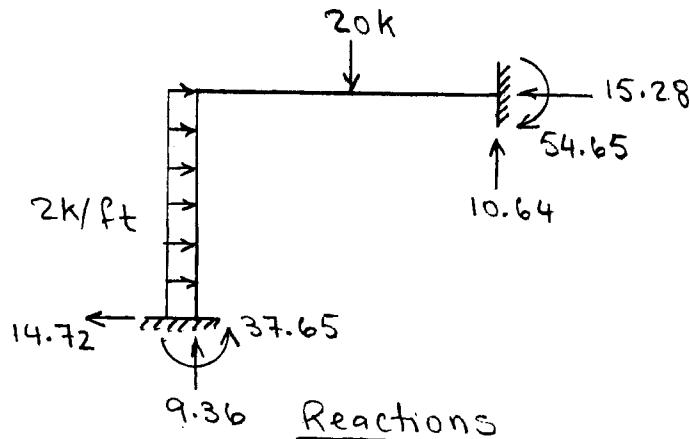
$$\{d\} = \frac{1}{EI \times 10^3} \begin{bmatrix} 0.1769 \\ -0.08129 \\ -0.03427 \end{bmatrix}$$

Member forces:

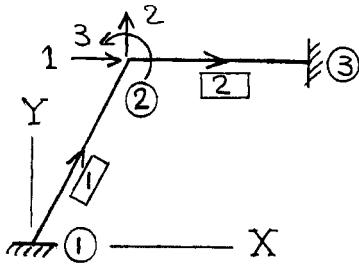
$$\{v_1\} = \frac{1}{EI \times 10^3} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0.1769 \\ -0.08129 \\ -0.03427 \end{bmatrix} ; \quad \{u_1\} = \frac{1}{EI \times 10^3} \begin{bmatrix} 0 \\ 0 \\ 0 \\ -0.08129 \\ -0.1769 \\ -0.03427 \end{bmatrix}$$

$$\{Q_1\} = \begin{bmatrix} 9.36 \text{ k} \\ 14.72 \text{ k} \\ 37.65 \text{ k-ft} \\ -9.36 \text{ k} \\ 15.29 \text{ k} \\ -41.92 \text{ k-ft} \end{bmatrix} ; \quad \{F_1\} = \begin{bmatrix} -14.72 \text{ k} \\ 9.36 \text{ k} \\ 37.65 \text{ k-ft} \\ -15.29 \text{ k} \\ -9.36 \text{ k} \\ -41.92 \text{ k-ft} \end{bmatrix}$$

$$\{u_2\} = \{v_2\} = \frac{1}{EI \times 10^3} \begin{bmatrix} 0.1769 \\ -0.08129 \\ -0.03427 \\ 0 \\ 0 \\ 0 \end{bmatrix} ; \quad \{F_2\} = \{Q_2\} = \begin{bmatrix} 15.28 \text{ k} \\ 9.36 \text{ k} \\ 41.92 \text{ k-ft} \\ -15.28 \text{ k} \\ 10.64 \text{ k} \\ -54.65 \text{ k-ft} \end{bmatrix}$$



18.8



Member 1: $L = 11.18 \text{ m}$; $\cos\theta = 0.447$; $\sin\theta = 0.894$

$$[k_1] = \begin{bmatrix} 71556 & 0 & 0 & -71556 & 0 & 0 \\ 0 & 687 & 3840 & 0 & -687 & 3840 \\ 0 & 28623 & 0 & 0 & -3840 & 14311 \\ 38 \text{ m.} & & 71556 & 0 & 0 & 0 \\ 0 & 687 & 0 & 687 & 0 & 0 \\ 0 & 28623 & 0 & 28623 & 0 & 0 \end{bmatrix}$$

$$[T_1] = \begin{bmatrix} 0.447 & 0.894 & 0 & 0 & 0 & 0 \\ -0.894 & 0.447 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0.447 & 0.894 & 0 \\ 0 & 0 & 0 & -0.894 & 0.447 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix}$$

$$[K_1] = [T_1]^T [k_1] [T_1] = \begin{bmatrix} 0 & 0 & 0 & 1 & 2 & 3 \\ 14847 & 28320 & -3433 & -14847 & -28320 & -3433 \\ 57327 & 1716 & -28320 & -57327 & 1716 & 0 \\ 28623 & 3433 & -1716 & 14311 & 0 & 0 \\ 14847 & 28320 & 3433 & 14847 & 28320 & 3433 \\ 57327 & -1716 & 14311 & 57327 & -1716 & 2 \\ 28623 & 0 & 0 & 28623 & 0 & 3 \end{bmatrix}$$

Member 2: $L = 10 \text{ m}$; $\cos\theta = 1$; $\sin\theta = 0$

$$[k_2] = [K_2] = \begin{bmatrix} 1 & 2 & 3 & 0 & 0 & 0 \\ 80000 & 0 & 0 & -80000 & 0 & 0 \\ 960 & 4800 & 0 & 0 & -960 & 4800 \\ 32000 & 0 & 0 & -4800 & 16000 & 0 \\ \text{Sym.} & 80000 & 0 & 0 & 960 & -4800 \\ & & & & 32000 & 0 \end{bmatrix}$$

Structure matrices:

$$[S] = \begin{bmatrix} 1 & 2 & 3 \\ 94847 & 28320 & 3433 \\ 28320 & 58287 & 3084 \\ 3433 & 3084 & 60623 \end{bmatrix}; \quad \{P\} = \begin{bmatrix} 1 \\ 2 \\ 3 \\ 0 \\ 0 \\ -150 \end{bmatrix}$$

18.8 (contd.) Joint displacements: By solving the equations $\{P\} = [S]\{d\}$, we obtain:

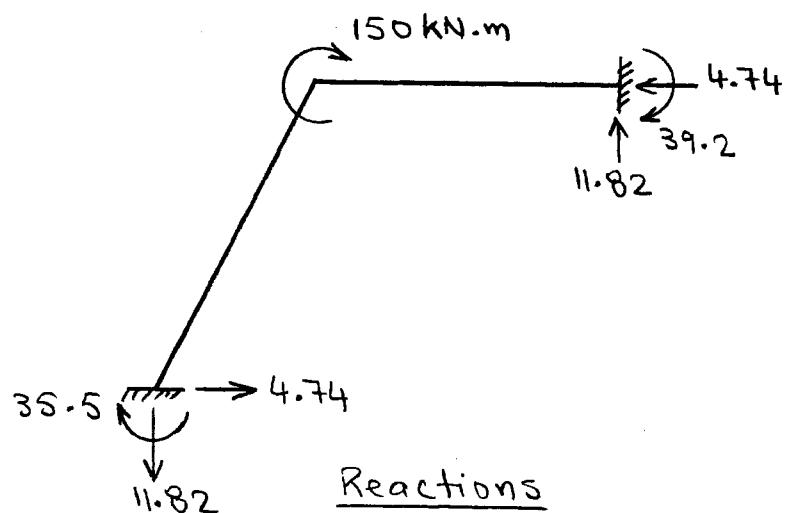
$$\{d\} = 10^{-4} \begin{bmatrix} 0.5924 \text{ m} \\ 1.0259 \text{ m} \\ -24.829 \text{ rad.} \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \end{matrix}$$

Member forces:

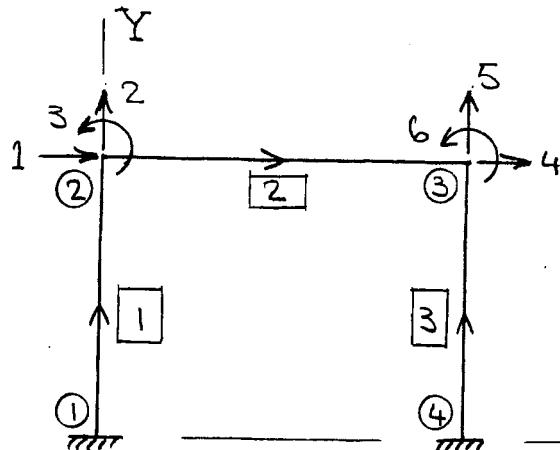
$$\{N_1\} = 10^{-4} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0.5924 \\ 1.0259 \\ -24.829 \end{bmatrix} ; \quad \{u_1\} = 10^{-4} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1.182 \\ -0.07103 \\ -24.829 \end{bmatrix}$$

$$\{Q_1\} = \begin{bmatrix} -8.46 \text{ kN} \\ -9.53 \text{ kN} \\ -35.5 \text{ kN.m} \\ 8.46 \text{ kN} \\ 9.53 \text{ kN} \\ -71 \text{ kN.m} \end{bmatrix} ; \quad \{F_1\} = \begin{bmatrix} 4.74 \text{ kN} \\ -11.82 \text{ kN} \\ -35.5 \text{ kN.m} \\ -4.74 \text{ kN} \\ 11.82 \text{ kN} \\ -71 \text{ kN.m} \end{bmatrix}$$

$$\{u_2\} = \{v_2\} = 10^{-4} \begin{bmatrix} 0.5924 \\ 1.0259 \\ -24.829 \\ 0 \\ 0 \\ 0 \end{bmatrix} ; \quad \{F_2\} = \{Q_2\} = \begin{bmatrix} 4.74 \text{ kN} \\ -11.82 \text{ kN} \\ -79 \text{ kN.m} \\ -4.74 \text{ kN} \\ 11.82 \text{ kN} \\ -39.2 \text{ kN.m} \end{bmatrix}$$



18.9



Members 1 and 3: $L = 30 \text{ ft}$; $\cos \theta = 0$; $\sin \theta = 1$

$$[k_1] = [k_3] = EI \times 10^{-3}$$

$$\begin{bmatrix} 80 & 0 & 0 & -80 & 0 & 0 \\ 0.444 & 6.67 & 0 & -0.444 & 6.67 & 0 \\ & 133.3 & 0 & -6.67 & 66.67 & \\ \text{Sym.} & & 80 & 0 & 0 & \\ & & & 0.444 & -6.67 & \\ & & & & 133.3 & \end{bmatrix}$$

$$[T_1] = [T_3] = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ -1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & -1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & -1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & -1 \end{bmatrix}$$

$$\begin{array}{c} \text{Member 3} \rightarrow \overline{0} \quad \overline{0} \quad \overline{0} \quad \overline{4} \quad \overline{5} \quad \overline{6} \\ \text{Member 1} \rightarrow \overline{0} \quad \overline{0} \quad \overline{0} \quad \overline{1} \quad \overline{2} \quad \overline{3} \end{array}$$

$$[K_1] = [K_3] = EI \times 10^{-3}$$

$$\begin{bmatrix} 0.444 & 0 & -6.67 & -0.444 & 0 & -6.67 & 0 & 0 \\ 80 & 0 & 0 & 80 & 0 & 0 & 0 & 0 \\ & 133.3 & 6.67 & 0 & 66.67 & 0 & 0 & 0 \\ \text{Sym.} & & 0.444 & 0 & 6.67 & 1 & 4 \\ & & & 80 & 0 & 2 & 5 \\ & & & & 133.3 & 3 & 6 \end{bmatrix}$$

Member 2: $L = 40 \text{ ft}$; $\cos \theta = 1$; $\sin \theta = 0$

$$[k_2] = [k_2] = EI \times 10^{-3}$$

$$\begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 \\ 120 & 0 & 0 & -120 & 0 & 0 \\ 0.375 & 7.5 & 0 & 0 & -0.375 & 7.5 \\ & 200 & 0 & -7.5 & 100 & 0 \\ \text{Sym.} & & 120 & 0 & 0 & 0 \\ & & & 0.375 & -7.5 & 200 \end{bmatrix} \begin{array}{l} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{array}$$

18.9 (contd.)

$$\{F_{f2}\} = \{Q_{f2}\} = \begin{bmatrix} 0 \\ 30 \\ 200 \\ 0 \\ 30 \\ -200 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{matrix}$$

Structure matrices:

$$[S] = EI \times 10^{-3} \begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 \\ 120.444 & 0 & 6.67 & -120 & 0 & 0 \\ 80.375 & 7.5 & 0 & -0.375 & 7.5 & 100 \\ 333.3 & 0 & -7.5 & 120.444 & 0 & 6.67 \\ & Sym. & & 80.375 & -7.5 & 333.33 \\ & & & & & 6 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{matrix}$$

$$\{P\} = \begin{bmatrix} 20 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}; \quad \{P_f\} = \{F_{f2}\}; \quad \{P\} - \{P_f\} = \begin{bmatrix} 20 \\ -30 \\ -200 \\ 0 \\ -30 \\ 200 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{matrix}$$

Joint displacements: By solving the equations

$\{P\} - \{P_f\} = [S]\{d\}$, we obtain:

$$\{d\} = \frac{1}{EI \times 10^{-3}} \begin{bmatrix} 29.4204 \\ -0.2906 \\ -1.314 \\ 29.289 \\ -0.4594 \\ 0.4044 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{matrix}$$

Member forces:

$$\{V_1\} = \frac{1}{EI \times 10^{-3}} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 29.4204 \\ -0.2906 \\ -1.314 \end{bmatrix} \begin{matrix} 0 \\ 0 \\ 0 \\ 1 \\ 2 \\ 3 \end{matrix}$$

$$\{V_4\} = \frac{1}{EI \times 10^{-3}} \begin{bmatrix} 0 \\ 0 \\ 0 \\ -0.2906 \\ -29.4204 \\ -1.314 \end{bmatrix}$$

18.9 (contd.)

$$\{Q_1\} = \begin{bmatrix} 23.26 \text{ k} \\ 4.3 \text{ k} \\ 108 \text{ k-ft} \\ -23.26 \text{ k} \\ -4.3 \text{ k} \\ 21 \text{ k-ft} \end{bmatrix}$$

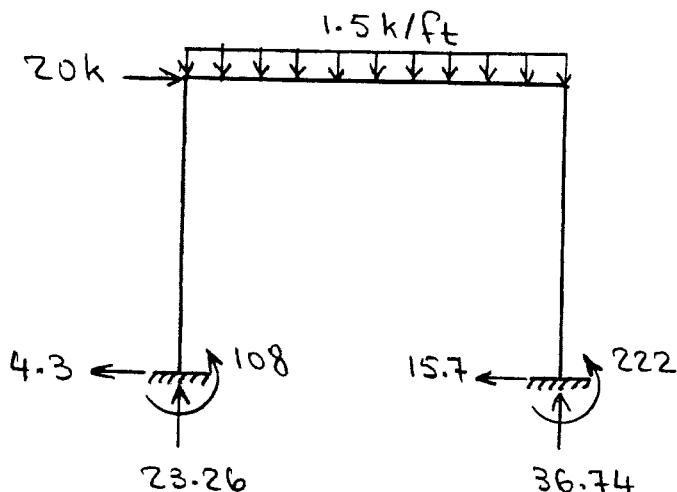
$$\{u_2\} = \{v_2\} = \{d\};$$

$$\{Q_2\} = \begin{bmatrix} 15.7 \text{ k} \\ 23.26 \text{ k} \\ -21 \text{ k-ft} \\ -15.7 \text{ k} \\ 36.74 \text{ k} \\ -249 \text{ k-ft} \end{bmatrix}$$

$$\{v_3\} = \frac{1}{EI \times 10^3} \begin{bmatrix} 0 \\ 0 \\ 0 \\ 29.289 \\ -0.4594 \\ 0.4044 \end{bmatrix} ;$$

$$\{u_3\} = \frac{1}{EI \times 10^3} \begin{bmatrix} 0 \\ 0 \\ 0 \\ -0.4594 \\ -29.289 \\ 0.4044 \end{bmatrix}$$

$$\{Q_3\} = \begin{bmatrix} 36.74 \text{ k} \\ 15.7 \text{ k} \\ 222 \text{ k-ft} \\ -36.74 \text{ k} \\ -15.7 \text{ k} \\ 249 \text{ k-ft} \end{bmatrix}$$



Reactions

Appendix B
Review of Matrix Algebra
&
Appendix C
Computer Software

APPENDIX B

B.1

$$[C] = [A] + 3[B] =$$

$$\begin{bmatrix} 18 & -11 & 18 \\ -11 & 19 & 28 \\ 18 & 28 & 4 \end{bmatrix}$$

B.2

$$[C] = 2[A] - [B] =$$

$$\begin{bmatrix} 7 & 8 \\ 11 & 7 \\ 1 & 0 \end{bmatrix}$$

B.3

$$[C] = \begin{bmatrix} -6 & 4 & -2 \end{bmatrix} \begin{bmatrix} 2 \\ -1 \\ 5 \end{bmatrix} = -12 - 4 - 10 = \underline{-26}$$

$$[D] = \begin{bmatrix} 2 \\ -1 \\ 5 \end{bmatrix} \begin{bmatrix} -6 & 4 & -2 \end{bmatrix} = \begin{bmatrix} -12 & 8 & -4 \\ 6 & -4 & 2 \\ -30 & 20 & -10 \end{bmatrix}$$

B.4

$$[C] = \begin{bmatrix} 2 & -5 \\ -5 & 3 \end{bmatrix} \begin{bmatrix} -3 & 4 \\ 4 & 1 \end{bmatrix} = \begin{bmatrix} -26 & 3 \\ 27 & -17 \end{bmatrix}$$

$$[D] = \begin{bmatrix} -3 & 4 \\ 4 & 1 \end{bmatrix} \begin{bmatrix} 2 & -5 \\ -5 & 3 \end{bmatrix} = \begin{bmatrix} -26 & 27 \\ 3 & -17 \end{bmatrix}$$

B.5

$$\begin{aligned}[A][B] &= \begin{bmatrix} 8 & -2 & 5 \\ 1 & -4 & 3 \\ 2 & 0 & 6 \end{bmatrix} \begin{bmatrix} 1 & -5 \\ 7 & 0 \\ 0 & -3 \end{bmatrix} \\ &= \begin{bmatrix} (8-14) & (-40-15) \\ (1-28) & (-5-9) \\ (2) & (-10-18) \end{bmatrix} = \begin{bmatrix} -6 & -55 \\ -27 & -14 \\ 2 & -28 \end{bmatrix} \\ [AB]^T &= \begin{bmatrix} -6 & -27 & 2 \\ -55 & -14 & -28 \end{bmatrix} \quad (1) \\ [B]^T[A]^T &= \begin{bmatrix} 1 & 7 & 0 \\ -5 & 0 & -3 \end{bmatrix} \begin{bmatrix} 8 & 1 & 2 \\ -2 & -4 & 0 \\ 5 & 3 & 6 \end{bmatrix} \\ &= \begin{bmatrix} (8-14) & (1-28) & (2) \\ (-40-15) & (-5-9) & (-10-18) \end{bmatrix} \\ &= \begin{bmatrix} -6 & -27 & 2 \\ -55 & -14 & -28 \end{bmatrix} \quad (2) \end{aligned}$$

From Eqs. (1) and (2) we can see that:

$$\underline{[AB]^T = [B]^T[A]^T}$$

B-6

$$\left[\begin{array}{ccc|c} 2 & 5 & -1 & 15 \\ 5 & -1 & 3 & 27 \\ -1 & 3 & 4 & 14 \end{array} \right]$$

$$\left[\begin{array}{ccc|c} 1 & 2.5 & -0.5 & 7.5 \\ 0 & -13.5 & 5.5 & -10.5 \\ 0 & 5.5 & 3.5 & 21.5 \end{array} \right]$$

$$\left[\begin{array}{ccc|c} 1 & 0 & 0.518 & 5.555 \\ 0 & 1 & -0.407 & 0.778 \\ 0 & 0 & 5.739 & 17.221 \end{array} \right]$$

$$\left[\begin{array}{ccc|c} 1 & 0 & 0 & 4 \\ 0 & 1 & 0 & 2 \\ 0 & 0 & 1 & 3 \end{array} \right]$$

Thus,

$$\underline{x_1 = 4 \quad x_2 = 2 \quad x_3 = 3}$$

B.7

$$\left[\begin{array}{ccc|c} -12 & -3 & 6 & 45 \\ 5 & 2 & -4 & -9 \\ 10 & 1 & -7 & -32 \end{array} \right]$$

$$\left[\begin{array}{ccc|c} 1 & 0.25 & -0.5 & -3.75 \\ 0 & 0.75 & -1.5 & 9.75 \\ 0 & -1.5 & -2 & 5.5 \end{array} \right]$$

$$\left[\begin{array}{ccc|c} 1 & 0 & 0 & -7 \\ 0 & 1 & -2 & 13 \\ 0 & 0 & -5 & 25 \end{array} \right]$$

$$\left[\begin{array}{ccc|c} 1 & 0 & 0 & -7 \\ 0 & 1 & 0 & 3 \\ 0 & 0 & 1 & -5 \end{array} \right]$$

Thus,

$$\underline{x_1 = -7 \quad x_2 = 3 \quad x_3 = -5}$$

B.8

$$\left[\begin{array}{ccccc} 5 & -2 & 6 & 0 & 0 \\ -2 & 4 & -1 & -1 & 18 \\ 6 & -1 & 6 & -1 & -29 \\ 0 & 3 & 8 & 7 & 11 \end{array} \right]$$

$$\left[\begin{array}{ccccc} 1 & -0.4 & 1.2 & 0 & 0 \\ 0 & 3.2 & 3.4 & 0 & 18 \\ 0 & 3.4 & -1.2 & 0 & -29 \\ 0 & 3 & 8 & 7 & 11 \end{array} \right]$$

$$\left[\begin{array}{ccccc} 1 & 0 & 1.625 & 0.375 & 2.25 \\ 0 & 1 & 1.063 & 0.938 & 5.625 \\ 0 & 0 & -4.814 & 4.811 & -48.125 \\ 0 & 0 & 4.811 & 4.186 & -5.875 \end{array} \right]$$

$$\left[\begin{array}{ccccc} 1 & 0 & 0 & 2 & -14 \\ 0 & 1 & 0 & -2 & -5 \\ 0 & 0 & 1 & 1 & 10 \\ 0 & 0 & 0 & 9 & -54 \end{array} \right]$$

$$\left[\begin{array}{ccccc} 1 & 0 & 0 & 0 & -2 \\ 0 & 1 & 0 & 0 & 7 \\ 0 & 0 & 1 & 0 & 4 \\ 0 & 0 & 0 & 1 & -6 \end{array} \right]$$

Thus,

$$x_1 = -2 \quad x_2 = 7 \quad x_3 = 4 \quad x_4 = -6$$

B.9

$$\left[\begin{array}{ccc|ccc} 4 & -3 & -1 & 1 & 0 & 0 \\ -2 & 5 & 1 & 0 & 1 & 0 \\ 6 & -4 & -5 & 0 & 0 & 1 \end{array} \right]$$

$$\left[\begin{array}{ccc|ccc} 1 & -0.75 & -0.25 & 0.25 & 0 & 0 \\ 0 & 3.5 & 0.5 & 0.5 & 1 & 0 \\ 0 & 0.5 & -3.5 & -1.5 & 0 & 1 \end{array} \right]$$

$$\left[\begin{array}{ccc|ccc} 1 & 0 & -0.143 & 0.357 & 0.215 & 0 \\ 0 & 1 & 0.143 & 0.143 & 0.286 & 0 \\ 0 & 0 & -3.572 & -1.572 & -0.143 & 1 \end{array} \right]$$

$$\left[\begin{array}{ccc|ccc} 1 & 0 & 0 & 0.42 & 0.22 & -0.04 \\ 0 & 1 & 0 & 0.08 & 0.28 & 0.04 \\ 0 & 0 & 1 & 0.44 & 0.04 & -0.28 \end{array} \right]$$

Thus,

$$[A]^{-1} = \begin{bmatrix} 0.42 & 0.22 & -0.04 \\ 0.08 & 0.28 & 0.04 \\ 0.44 & 0.04 & -0.28 \end{bmatrix}$$

B.10

$$\left[\begin{array}{cccc|cccc} 4 & 2 & 0 & -3 & 1 & 0 & 0 & 0 \\ 2 & 3 & -4 & 0 & 0 & 1 & 0 & 0 \\ 0 & -4 & 2 & -1 & 0 & 0 & 1 & 0 \\ -3 & 0 & -1 & 5 & 0 & 0 & 0 & 1 \end{array} \right]$$

$$\left[\begin{array}{cccc|cccc} 1 & 0.5 & 0 & -0.75 & 0.25 & 0 & 0 & 0 \\ 0 & 2 & -4 & 1.5 & -0.5 & 1 & 0 & 0 \\ 0 & -4 & 2 & -1 & 0 & 0 & 1 & 0 \\ 0 & 1.5 & -1 & 2.75 & 0.75 & 0 & 0 & 1 \end{array} \right]$$

$$\left[\begin{array}{cccc|cccc} 1 & 0 & 1 & -1.125 & 0.375 & -0.25 & 0 & 0 \\ 0 & 1 & -2 & 0.75 & -0.25 & 0.5 & 0 & 0 \\ 0 & 0 & -6 & 2 & -1 & 2 & 1 & 0 \\ 0 & 0 & 2 & 1.625 & 1.125 & -0.75 & 0 & 1 \end{array} \right]$$

$$\left[\begin{array}{cccc|cccc} 1 & 0 & 0 & -0.7917 & 0.2083 & 0.0833 & 0.1667 & 0 \\ 0 & 1 & 0 & 0.0833 & 0.0833 & -0.1667 & -0.3333 & 0 \\ 0 & 0 & 1 & -0.3333 & 0.1667 & -0.3333 & -0.1667 & 0 \\ 0 & 0 & 0 & 2.2917 & 0.7917 & -0.0833 & 0.3333 & 1 \end{array} \right]$$

$$\left[\begin{array}{cccc|cccc} 1 & 0 & 0 & 0 & 0.4818 & 0.0545 & 0.2818 & 0.3455 \\ 0 & 1 & 0 & 0 & 0.0545 & -0.1636 & -0.3455 & -0.0364 \\ 0 & 0 & 1 & 0 & 0.2818 & -0.3455 & -0.1182 & 0.1455 \\ 0 & 0 & 0 & 1 & 0.3455 & -0.0364 & 0.1455 & 0.4364 \end{array} \right]$$

Thus,

$$[A]^{-1} = \left[\begin{array}{cccc} 0.4818 & 0.0545 & 0.2818 & 0.3455 \\ 0.0545 & -0.1636 & -0.3455 & -0.0364 \\ 0.2818 & -0.3455 & -0.1182 & 0.1455 \\ 0.3455 & -0.0364 & 0.1455 & 0.4364 \end{array} \right]$$

APPENDIX C

C.1

$$\frac{L}{360} = \frac{80}{360} = 0.222 \text{ ft} = 2.67 \text{ in.}$$

(a) $A = 9.12 \text{ in}^2$

(b) $A = 6.33 \text{ in}^2$

(c) $A = 8.66 \text{ in}^2$

C.2

$$\frac{L}{360} = \frac{24}{360} = 0.0667 \text{ m}$$

(a) $A = 8470 \text{ mm}^2$

(b) $A = 7620 \text{ mm}^2$

(c) $A = 6860 \text{ mm}^2$

c.3

$$\underline{I = 1257 \text{ in}^4}$$