# Design, modeling and control of a Scara manipulator using PID controller

Marco Baltazar, Branco Palma, Yerack Paredes
Faculty of Engineering, School of Electronic Engineering
Universidad Peruana de Ciencias Aplicadas (UPC)

U201213835@upc.edu.pe U201417983@upc.edu.pe U201512654@upc.edu.pe

Abstract: this paper describes the design, modeling and control of a Scara manipulator using inverse dynamics following desired trajectories. To do this, the robot model is analyzed and the necessary equations are obtained to model the system and its corresponding control.

### I. INTRODUCTION

The main objective of this paper is to design a position controller for a robot based on the analysis of its behavior in space. A Scara manipulator is a robot that has four degrees of freedom, that is, four points of movement. The first three are of the rotary type and the fourth is displacement type.

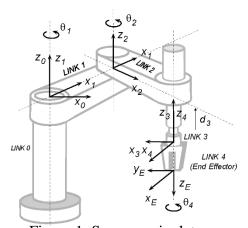


Figure 1: Scara manipulator

In order to achieve the design of a controller, it is necessary to have the basic concepts of geometry, transformation matrices, dynamic model and controller design. The controller will be in charge of governing the manipulator so that the desired position is achieved in a certain time from an input and the error measurement.

The modeling of the Scara manipulator in the coordinate axes is useful to understand not only how the behavior of the robot would be affected with respect to the input and output variables, but also, to understand the behavior of the system inside, analyze the internal variables and how these can provide more information about a real robot.

During the process, it will be necessary to consider possible non-linear behaviors of the system. This can cause unwanted behavior in the system and creating singularities which cannot be solved by the designed driver.

The behavior and design process can be represented through figure 2, which shows how the desired value goes through several stages until the robot is positioned where it is desired. In order to calculate each of the stages described in said figure, complex mathematical equations must be used, which must be detailed in detail in this paper.

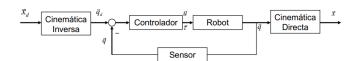


Figure 2: robot control system

The Denavit-Hartenberg (D-H) method is used to be able to determine the position of the end point of the robot with respect to the axis that is defined in the first joint of the manipulator robot. This method is also useful to establish the rotation and translation matrices of the system.

Starting from the D-H, the coordinate transformation matrix for each of the joints can be determined. This results from the multiplication of the corresponding matrices for each of the joints.

$$A_{i} = \begin{bmatrix} \cos\theta_{i} & -\cos\alpha_{i}\sin\theta_{i} & \sin\alpha_{i}\sin\theta_{i} & a_{i}\cos\theta_{i} \\ \sin\theta_{i} & \cos\alpha_{i}\cos\theta_{i} & -\sin\alpha_{i}\cos\theta_{i} & a_{i}\sin\theta_{i} \\ 0 & \sin\alpha_{i} & \cos\alpha_{i} & d_{i} \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

Figure 3: general matrix for each of the joints.

In order to design the robot controller, it is first necessary to know the equation of its dynamics, which will have the following general form:

$$M_{(q)}\ddot{q} + C_{(q,\dot{q})}\dot{q} + F_{(\dot{q})} + G_{(q)} + \tau_d = \tau$$

# Where:

- o M is the inertia matrix
- C is the matrix of centripetal force and Coriolis
- o F is the friction vector
- o G is the gravity vector
- $\circ$   $\tau_d$  represents the disturbances
- $\circ$   $\tau$  is the control input vector and contains the torques required at each joint

However, friction and disturbances can be neglected for mathematical analysis.

In order to control the Scara manipulator, it is necessary to use a PID controller, since it is probable that there is an error in steady state and a PD controller would not be sufficient in such a condition. An integrator is added in each of the joints. That is why touch control is represented by the equation:

$$\tau = M \left( \ddot{q}_d + k_v \dot{e} + k_p e + k_i \varepsilon \right) + N$$

Where:

- o  $k_v$  is the derivative gain
- $\circ$   $k_p$  is the proportional gain
- o  $k_i$  is the integral gain
- o e is the error
- $\circ$   $\dot{\varepsilon} = e$
- $\circ \quad N = \hat{C}\dot{q} + \hat{G}$

This controller adds stability to the PD type as long as the constant Ki is adequate.

#### II. WORK DEVELOPMENT AND ANALYSIS

In order to start the development process, it is first necessary to analyze the robot architecture from figure 4.

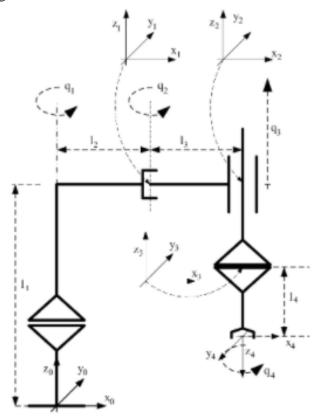


Figure 4: Scara robot architecture

Where four joint values are identified. In addition, you can also see the displacements to achieve that the point on the x4, y4 and z4 axis is located with reference to x0, y0 and z0.

Starting from a graphic analysis, it is possible to determine the D-H of the system.

	θ	d	а	α
1	q1	l1	12	0
2	q2	0	13	0
3	0	q3	0	0
4	q4	-14	0	π

Figure 5: D-H Scara robot

This table explains the behavior in the joints and how the coordinates Xx, Yx, Zx are with respect to a fixed point or a mobile one as appropriate.

Starting from D-H it is possible to determine the Ai matrices for each of the joints using the general matrix.

$$A_4 = \begin{array}{cccc} \cos(q4) & sen(q4) & 0 & 0 \\ sen(q4) & -\cos(q4) & 0 & 0 \\ 0 & 0 & -1 & -l4 \\ 0 & 0 & 0 & 1 \end{array}$$

Later it is necessary to carry out the multiplication of the matrices in order to determine the transformation matrix.

$$T = A_1 \times A_2 \times A_3 \times A_4$$

$$T \\ = \begin{matrix} C_{1+2+4} & S_{1+2+4} & 0 & l3 \times C_{1+2} + l2 \times C_1 \\ S_{1+2+4} & -C_{1+2+4} & 0 & l3 \times S_{1+2} + l2 \times S_1 \\ 0 & 0 & -1 & l1 - l4 + q3 \\ 0 & 0 & 0 & 1 \end{matrix}$$

We start from this point to determine the dynamics of the system, for which it is necessary to find the value of each of the variables within the following equation.

$$M_{(q)}\ddot{q} + C_{(q,\dot{q})}\dot{q} + G_{(q)} = \tau$$

From the matrix T the equations for Xi, Yi and Zi can be obtained. These equations will be derived and squared in order to determine the velocity.

$$dx_i = \frac{dx_i}{dq_1} \times dq_1 + \frac{dx_i}{dq_2} \times dq_2 + \frac{dx_i}{dq_3} \times dq_1$$

$$dy_i = \frac{dy_i}{dq_1} \times dq_1 + \frac{dy_i}{dq_2} \times dq_2 + \frac{dy_i}{dq_3} \times dq_1$$

$$dz_i = \frac{dz_i}{dq_1} \times dq_1 + \frac{dz_i}{dq_2} \times dq_2 + \frac{dz_i}{dq_3} \times dq_1$$

$$v^2 = (dx)^2 + (dy)^2 + (dz)^2$$

By finding the velocity it is possible to determine the kinetic energy with respect to each more in the robot.

$$K_i = \frac{1}{2} \times m_i \times v_i^2$$

$$K = \sum_{n=i}^{N} K_n$$

$$N = \text{degrees of freedom}$$

To calculate the inertia matrix, the equation of kinetic energy with respect to qi is derived twice.

$$M_{11} = \frac{\left(m2 \times (2 \times l2^{2} + 4 \times C_{2} \times l2 \times l3 + 2 \times l3^{2})\right)}{2} + \frac{\left(m3 \times (2 \times l2^{2} + 4 \times C_{2} \times l2 \times l3 + 2 \times l3^{2})\right)}{2} + \left(m4 \times \frac{\left(2 \times l2^{2} + 4 \times C_{2} \times l2 \times l3 + 2 \times l3^{2}\right)\right)}{2} + l2^{2} \times m1$$

$$M_{21} = l3 \times (l3 + l2 \times C_{2}) \times (m2 + m3 + m4)$$

$$M_{31} = 0$$

$$M_{41} = 0$$

$$M_{12} = l3 \times (l3 + l2 \times C_{2}) \times (m2 + m3 + m4)$$

$$M_{22} = l3^{2} \times (m2 + m3 + m4)$$

$$M_{32} = 0$$

$$M_{42} = 0$$

$$M_{13} = 0 \qquad M_{23} = 0$$

$$M_{33} = \frac{m3}{4} + \frac{m4}{4} \qquad M_{43} = 0$$

$$M_{14} = 0 \qquad M_{24} = 0$$

$$M_{34} = 0 \qquad M_{44} = 0$$

Similarly, using partial derivatives on each of the elements of the inertia matrix, it is possible to obtain the matrix of Centripetal Force and Coreolis.

$$C_{11} = -l2 \times l3 \times q2p \times S_2 \times (m2 + m3 + m4)$$

$$C_{21} = -l2 \times l3 \times q1p \times S_2 \times (m2 + m3 + m4)$$

$$C_{11} = 0 \quad C_{11} = 0$$

$$C_{12} = -l2 \times l3 \times S_2 \times (\dot{q}1 + \dot{q}2) \times (m2 + m3 + m4)$$

$$C_{13} = 0 \quad C_{14} = 0$$

$$C_{22} = 0 \quad C_{23} = 0 \quad C_{24} = 0$$

$$C_{32} = 0 \quad C_{33} = 0 \quad C_{34} = 0$$

$$C_{42} = 0 \quad C_{43} = 0 \quad C_{44} = 0$$

To determine the gravity vector, it is required to use the power energy formula, which depends only on the position in zi.

$$E_{Pi} = m \times g \times z_{i}$$

$$E_{p} = \sum_{n=i}^{N} E_{Pi}$$

$$G = \underbrace{g \times (m3 + m4)}_{2}$$

For the design of the controller it is necessary to take into account the following equation:

$$\tau = M \left( \ddot{q}_d + k_v \dot{e} + k_p e + k_i \varepsilon \right) + N$$

Where N has been calculated based on the equation:

$$N = C \times \dot{q2}$$

$$\dot{q3}$$

Finally, the error obtained is:

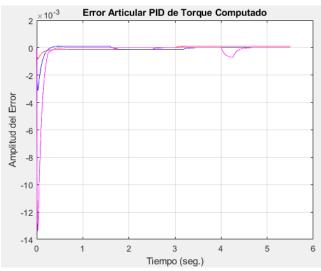


Figure 6: Joint torque error

In addition, the behavior of the position, velocity and corresponding acceleration for a given time can be determined.

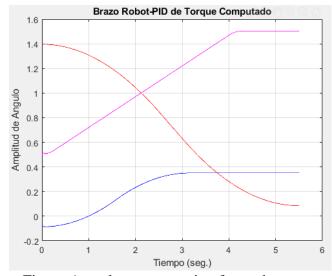


Figure 6: angle representation for each rotary joint

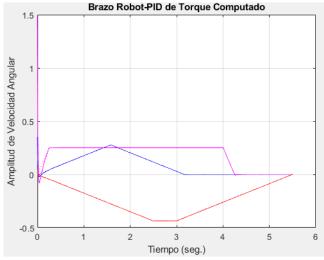


Figure 7: angular velocity for each rotary joint

# III. Conclusions

By using the PID controller to manipulate the final position of the Scara robotic arm, it is possible to find its position with a low margin of error.

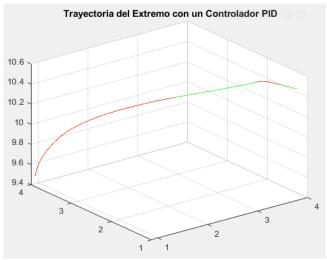


Figure 8: Position of the final point with PID controller

The result is more optimal when using an integrator, thus achieving a lower steady-state error.

# IV. BIBLIOGRAPHY

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- [2] Antonio Barrientos, Fundamentos de Robótica 2da ed., 2017
- [3] Enrique Arnaez Braschi; Enfoque práctico de la teoría de robots: con aplicaciones de Matlab. Lima: Editorial UPC, 2015.

#### V. Anexo 1

% Cinemática Robot

```
clc;close all;clear all
syms q1 q2 q3 q4 11 12 13 14 15
% a d al th
%DH = [12 11 0 q1]
       13 0 0 q2
        0 q3 0 0
        0 -14 pi q4];
   DH = [q1 11 12 0]
       q2 0 13 0
       0 q3 0 0
       q4 -14 0 pi];
A1 =
matra(DH(1,1),DH(1,2),DH(1,3),DH(1,4))
matra(DH(2,1),DH(2,2),DH(2,3),DH(2,4))
matra(DH(3,1),DH(3,2),DH(3,3),DH(3,4))
matra(DH(4,1),DH(4,2),DH(4,3),DH(4,4))
T1 = A1
T2 = T1*A2
T3 = T2*A3
T4 = simplify(T3*A4)
% Calculando el Modelo Dinámico.
clear all; close all; clc
syms q1 q2 q3 q4 11 12 13 14 15 q1p q2p
q3p q4p pi
% a d al th
DH = [q1 11 12 0]
       q2 0 13 0
       0 q3 0 0
       q4 -14 0 pi];
% masa 1
% a d al th
matra(DH(1,1),DH(1,2)/2,DH(1,3),DH(1,4))
T1 = A1m;
x1 = T1(1,4)
y1 = T1(2,4)
z1 = T1(3,4)
% Derivamos con respecto al tiempo
x1p =
diff(x1, 'q1')*q1p+diff(x1, 'q2')*q2p+diff
(x1, 'q3')*q3p + diff(x1, 'q4')*q4p;
y1p =
diff(y1,'q1')*q1p+diff(y1,'q2')*q2p+diff
(y1, 'q3')*q3p + diff(y1, 'q4')*q4p;
```

```
z1p =
diff(z1, 'q1') *q1p+diff(z1, 'q2') *q2p+diff
(z1, 'q3')*q3p + diff(z1, 'q4')*q4p;
% Elevamos al cuadrado
x1p2 = x1p^2; y1p2 = y1p^2; z1p2 =
z1p^2;
% Vel. cuadrado v^2 = xp^2 + yp^2 + zp^2
v12 = simplify(x1p2 + y1p2 + z1p2)
% masa 2
A1 =
matra(DH(1,1),DH(1,2),DH(1,3),DH(1,4));
{\tt matra}\,({\tt DH}\,(2,1)\,,{\tt DH}\,(2,2)\,,{\tt DH}\,(2,3)\,,{\tt DH}\,(2,4)\,)\,;
T2 = A1*A2;
x2 = T2(1,4);
y2 = T2(2,4);
z2 = T2(3,4);
% Derivamos con respecto al tiempo
= \alpha 2x
diff(x2, 'q1')*q1p+diff(x2, 'q2')*q2p+diff
(x2, 'q3')*q3p + diff(x2, 'q4')*q4p;
y2p =
diff(y2,'q1')*q1p+diff(y2,'q2')*q2p+diff
(y2, 'q3')*q3p + diff(y2, 'q4')*q4p;
z2p =
diff(z2,'q1')*q1p+diff(z2,'q2')*q2p+diff
(z2, 'q3')*q3p + diff(z2, 'q4')*q4p;
% Elevamos al cuadrado
x2p2 = x2p^2; y2p2 = y2p^2; z2p2 =
z2p^2;
% Vel. cuadrado v^2 = xp^2 + yp^2 + zp^2
v22 = simplify(x2p2 + y2p2 + z2p2);
% masa 3
A1 =
matra(DH(1,1),DH(1,2),DH(1,3),DH(1,4));
matra(DH(2,1),DH(2,2),DH(2,3),DH(2,4));
A3 =
matra(DH(3,1),DH(3,2)/2,DH(3,3),DH(3,4))
;
T3 = A1*A2*A3;
x3 = T3(1,4);
y3 = T3(2,4);
z3 = T3(3,4);
% Derivamos con respecto al tiempo
x3p =
diff(x3, 'q1')*q1p+diff(x3, 'q2')*q2p+diff
(x3, 'q3')*q3p + diff(x3, 'q4')*q4p;
y3p =
diff(y3, 'q1')*q1p+diff(y3, 'q2')*q2p+diff
(y3, 'q3')*q3p + diff(y3, 'q4')*q4p;
z3p =
diff(z3, 'q1')*q1p+diff(z3, 'q2')*q2p+diff
(z3, 'q3')*q3p + diff(z3, 'q4')*q4p;
% Elevamos al cuadrado
```

```
x3p2 = x3p^2; y3p2 = y3p^2; z3p2 =
                                               m21 = m12;
z3p^2;
                                               m22 =
% Vel. cuadrado v^2 = xp^2 + yp^2 + zp^2
                                               simplify(diff(diff(K,'q2p'),'q2p'));
v32 = simplify(x3p2 + y3p2 + z3p2);
                                               m23 =
                                               simplify(diff(diff(K,'q2p'),'q3p'));
                                               m24 =
                                               simplify(diff(diff(K,'q2p'),'q4p'));
                                               m31 = m13;
                                               m32 = m23;
% masa 4
                                               m33 =
A1 =
                                               simplify(diff(diff(K, 'q3p'), 'q3p'));
matra(DH(1,1),DH(1,2),DH(1,3),DH(1,4));
                                               simplify(diff(diff(K, 'q3p'), 'q4p'));
matra(DH(2,1),DH(2,2),DH(2,3),DH(2,4));
                                               m41 = m14;
                                               m42 = m24;
matra(DH(3,1),DH(3,2)/2,DH(3,3),DH(3,4))
                                               m43 = m34;
                                               m44 =
A4 =
                                               simplify (diff (diff (K, 'q4p'), 'q4p'));
matra (DH(4,1), DH(4,2)/2, DH(4,3), DH(4,4))
T4 = A1*A2*A3*A4;
x4 = T4(1,4);
y4 = T4(2,4);
                                               M = [ m11 m12 m13 m14 ]
z4 = T4(3,4);
                                               m21 m22 m23 m24
% Derivamos con respecto al tiempo
                                               m31 m32 m33 m34
diff(x4, 'q1')*q1p+diff(x4, 'q2')*q2p+diff
                                               m41 m42 m43 m44]
(x4, 'q3')*q3p + diff(x4, 'q4')*q4p;
y4p =
diff(y4,'q1')*q1p+diff(y4,'q2')*q2p+diff
(y4, 'q3')*q3p + diff(y4, 'q4')*q4p;
                                               % Matriz de Fuerzas Centrípetas y de
z4p =
                                               Coriolis
diff(z4, 'q1')*q1p+diff(z4, 'q2')*q2p+diff
                                               % Empleamos los términos de Christoffel
(z4, 'q3')*q3p + diff(z4, 'q4')*q4p;
                                               % c11
% Elevamos al cuadrado
                                               c11 =
x4p2 = x4p^2; y4p2 = y4p^2; z4p2 =
                                               1/2*(diff(m11, 'q1')+diff(m11, 'q1')-
z4p^2;
                                               diff(m11, 'q1')) *q1p;
% Vel. cuadrado v^2 = xp^2 + yp^2 + zp^2
                                               c11 = c11 +
v42 = simplify(x4p2 + y4p2 + z4p2);
                                               1/2*(diff(m11,'q2')+diff(m12,'q1')-
                                               diff(m21, 'q1')) *q2p;
                                               c11 = c11 +
                                               1/2*(diff(m11, 'q3')+diff(m13, 'q1')-
                                               diff(m31, 'q1')) *q3p;
                                               c11 = c11 +
                                               1/2*(diff(m11,'q4')+diff(m14,'q1')-
                                               diff(m41, 'q1')) *q4p;
syms m1 m2 m3 m4% Masas
                                               c11 = simplify(c11);
% Energía Cinética del Sistema
                                               % c12
% K = K1 + K2 ---> Ki = 1/2*mi*vi^2
                                               c12 =
K1 = 1/2*m1*v12; K2 = 1/2*m2*v22;
                                               1/2*(diff(m12, 'q1')+diff(m11, 'q2')-
K3 = 1/2*m3*v32; K4 = 1/2*m4*v42;
                                               diff(m12, 'q1')) *q1p;
K = K1+K2+K3+K4;
                                               c12 = c12 +
% Matriz de Inercias
                                               1/2*(diff(m12, 'q2')+diff(m12, 'q2')-
m11 =
                                               diff(m22, 'q1')) *q2p;
simplify(diff(diff(K,'q1p'),'q1p'));
                                               c12 = c12 +
m12 =
                                               1/2*(diff(m12, 'q3') + diff(m13, 'q2') -
simplify (diff (diff (K, 'q1p'), 'q2p'));
                                               diff(m32, 'q1')) *q3p;
m13 =
                                               c12 = c12 +
simplify(diff(diff(K, 'q1p'), 'q3p'));
                                               1/2*(diff(m12,'q4')+diff(m14,'q2')-
m14 =
                                               diff(m42, 'q1')) *q4p;
simplify(diff(diff(K,'q1p'),'q4p'));
                                               c12 = simplify(c12);
```

```
% c13
                                              c22 = simplify(c22);
                                              % c23
1/2*(diff(m13,'q1')+diff(m11,'q3')-
                                              c23 =
diff(m13, 'q1'))*q1p;
                                             1/2*(diff(m23,'q1')+diff(m21,'q3')-
c13 = c13 +
                                             diff(m13,'q2'))*q1p;
1/2*(diff(m13,'q2')+diff(m12,'q3')-
                                            c23 = c23 +
                                             1/2*(diff(m23,'q2')+diff(m22,'q3')-
diff(m23, 'q1')) *q2p;
c13 = c13 +
                                            diff(m23, 'q2')) *q2p;
1/2*(diff(m13, 'q3')+diff(m13, 'q3')-
                                            c23 = c23 +
diff(m33,'q1'))*q3p;
                                             1/2*(diff(m23, 'q3')+diff(m23, 'q3')-
                                             diff(m33, 'q2'))*q3p;
c13 = c13 +
1/2*(diff(m13,'q4')+diff(m14,'q3')-
                                              c23 = c23 +
diff(m43, 'q1')) *q4p;
                                             1/2*(diff(m23,'q4')+diff(m24,'q3')-
c13 = simplify(c13);
                                              diff(m43,'q2'))*q4p;
% c14
                                             c23 = simplify(c23);
c14 =
1/2*(diff(m14, 'q1') + diff(m11, 'q4') -
                                             % c24
diff(m14, 'q1')) *q1p;
                                              c24 =
c14 = c14 +
                                              1/2*(diff(m24,'q1')+diff(m21,'q4')-
1/2*(diff(m14,'q2')+diff(m12,'q4')-
                                            diff(m14,'q2'))*q1p;
diff(m24,'q1'))*q2p;
                                             c24 = c24 +
c14 = c14 +
                                             1/2*(diff(m24,'q2')+diff(m22,'q4')-
1/2*(diff(m14, 'q3') + diff(m13, 'q4') -
                                            diff(m24, 'q2')) *q2p;
diff(m34, 'q1')) *q3p;
                                             c24 = c24 +
c14 = c14 +
                                             1/2*(diff(m24,'q3')+diff(m23,'q4')-
1/2*(diff(m14,'q4')+diff(m14,'q4')-
                                            diff(m34,'q2'))*q3p;
diff(m44, 'q1')) *q4p;
                                             c24 = c24 +
c14 = simplify(c14);
                                             1/2*(diff(m24,'q4')+diff(m24,'q4')-
                                              diff(m44, 'q2'))*q4p;
                                              c24 = simplify(c24);
% c21
c21 =
1/2*(diff(m21, 'q1')+diff(m21, 'q1')-
diff(m11, 'q2')) *q1p;
c21 = c21 +
                                             % c31
1/2*(diff(m21,'q2')+diff(m22,'q1')-
                                              c31 =
diff(m21, 'q2')) *q2p;
                                             1/2*(diff(m31,'q1')+diff(m31,'q1')-
c21 = c21 +
                                            diff(m11,'q3'))*q1p;
1/2*(diff(m21, 'q3') + diff(m23, 'q1') -
                                            c31 = c31 +
diff(m31, 'q2')) *q3p;
                                             1/2*(diff(m31, 'q2') + diff(m32, 'q1') -
c21 = c21 +
                                            diff(m21, 'q3')) *q2p;
1/2*(diff(m21,'q4')+diff(m24,'q1')-
                                             c31 = c31 +
diff(m41, 'q2')) *q4p;
                                             1/2*(diff(m31, 'q3') + diff(m33, 'q1') -
c21 = simplify(c21);
                                              diff(m31, 'q3')) *q3p;
% c22
                                              c31 = c31 +
c22 =
                                              1/2*(diff(m31, 'q4') + diff(m34, 'q1') -
1/2*(diff(m22, 'q1')+diff(m21, 'q2')-
                                              diff(m41, 'q3')) *q4p;
diff(m12, 'q2')) *q1p;
                                              c31 = simplify(c31);
c22 = c22 +
1/2*(diff(m22, 'q2')+diff(m22, 'q2')-
                                              % c32
diff(m22, 'q2')) *q2p;
                                             c32 =
c22 = c22 +
                                             1/2*(diff(m32,'q1')+diff(m31,'q2')-
1/2*(diff(m22,'q3')+diff(m23,'q2')-
                                             diff(m12, 'q3')) *q1p;
diff(m32, 'q2')) *q3p;
                                              c32 = c32 +
c22 = c22 +
1/2*(diff(m22,'q4')+diff(m24,'q2')-
                                              1/2*(diff(m32, 'q2')+diff(m32, 'q2')-
                                             diff(m22, 'q3')) *q2p;
diff(m42, 'q2')) *q4p;
                                              c32 = c32 +
                                              1/2*(diff(m32, 'q3') + diff(m33, 'q2') -
                                              diff(m32, 'q3')) *q3p;
```

```
c32 = c32 +
                                             c42 = c42 +
1/2*(diff(m32,'q4')+diff(m34,'q2')-
                                            1/2*(diff(m42,'q4')+diff(m44,'q2')-
diff(m42,'q3'))*q4p;
                                            diff(m42,'q4'))*q4p;
c32 = simplify(c32);
                                             c42 = simplify(c42);
% c33
                                             % c43
c33 =
                                             c43 =
1/2*(diff(m33, 'q1') + diff(m31, 'q3') -
                                            1/2*(diff(m43,'q1')+diff(m41,'q3')-
diff(m13, 'q3')) *q1p;
                                            diff(m13,'q4'))*q1p;
c33 = c33 +
                                            c43 = c43 +
1/2*(diff(m33, 'q2') + diff(m32, 'q3') -
                                            1/2*(diff(m43,'q2')+diff(m42,'q3')-
diff(m23, 'q3')) *q2p;
                                            diff(m23, 'q4'))*q2p;
c33 = c33 +
                                             c43 = c43 +
1/2*(diff(m33, 'q3')+diff(m33, 'q3')-
                                             1/2*(diff(m43, 'q3') + diff(m43, 'q3') -
diff(m33, 'q3')) *q3p;
                                            diff(m33, 'q4'))*q3p;
c33 = c33 +
                                             c43 = c43 +
1/2*(diff(m33,'q4')+diff(m34,'q3')-
                                             1/2*(diff(m43,'q4')+diff(m44,'q3')-
diff(m43,'q3'))*q4p;
                                             diff(m43,'q4'))*q4p;
c33 = simplify(c33);
                                             c43 = simplify(c43);
% c34
                                             % c44
c34 =
                                             c44 =
1/2*(diff(m34,'q1')+diff(m31,'q4')-
                                             1/2*(diff(m44,'q1')+diff(m41,'q3')-
diff(m14,'q3'))*q1p;
                                            diff(m14, 'q4'))*q1p;
c34 = c34 +
                                             c44 = c44 +
1/2*(diff(m34,'q2')+diff(m32,'q4')-
                                            1/2*(diff(m44,'q2')+diff(m42,'q3')-
diff(m24, 'q3')) *q2p;
                                            diff(m24,'q4'))*q2p;
c34 = c34 +
                                             c44 = c44 +
1/2*(diff(m34,'q3')+diff(m33,'q4')-
                                            1/2*(diff(m44,'q3')+diff(m43,'q3')-
diff(m34, 'q3')) *q3p;
                                             diff(m34, 'q4'))*q3p;
c34 = c34 +
                                             c44 = c44 +
1/2*(diff(m34,'q3')+diff(m34,'q4')-
                                            1/2*(diff(m44,'q4')+diff(m44,'q2')-
                                            diff(m44,'q4'))*q4p;
diff(m44, 'q3')) *q4p;
c34 = simplify(c34);
                                             c44 = simplify(c44);
% c41
c41 =
1/2*(diff(m41,'q1')+diff(m41,'q1')-
diff(m11, 'q4')) *q1p;
c41 = c41 +
1/2*(diff(m41, 'q2') + diff(m42, 'q1') -
diff(m21, 'q4')) *q2p;
c41 = c41 +
1/2*(diff(m41, 'q3') + diff(m43, 'q1') -
diff(m31, 'q4'))*q3p;
c41 = c41 +
1/2*(diff(m41,'q4')+diff(m44,'q1')-
diff(m41, 'q4')) *q4p;
                                            C = [c11 c12 c13 c14]
                                             c21 c22 c23 c24
c41 = simplify(c41);
                                             c31 c32 c33 c34
                                             c41 c42 c43 c44]
% c42
c42 =
1/2*(diff(m42,'q1')+diff(m41,'q2')-
diff(m12,'q4'))*q1p;
                                             % Cálculo de la Energía Potencial
c42 = c42 +
                                             % P = P1 + P2 ---> Pi = mi*q*zi
1/2*(diff(m42,'q2')+diff(m42,'q2')-
                                            syms g
diff(m22, 'q4')) *q2p;
                                            P1 = m1*g*z1;
c42 = c42 +
                                             P2 = m2*g*z2;
1/2*(diff(m42, 'q3') + diff(m43, 'q2') -
                                            P3 = m3*g*z3;
diff(m32, 'q4')) *q3p;
                                             P4 = m4*g*z4;
```

```
= x(6);
P = P1 + P2 + P3 + P4;
                                             X7(k,1) = x(7); X8(k,1) = x(8); X9(k,1)
% Determinación del Vector de Gravedad
                                             = x(9);
g1 = simplify(diff(P,'q1'));
                                             q1 = x(1); q2 = x(2); q3 = x(3);
g2 = simplify(diff(P, 'q2'));
                                             q1p = x(4); q2p = x(5); q3p = x(6);
g3 = simplify(diff(P, 'q3'));
g4 = simplify(diff(P, 'q4'));
G = conj([g1 g2 g3 g4]')
                                             % Errores de Seguimiento
                                             e = qd(:,k) - x(1:3); ep = qdp(:,k) -
                                             x(4:6);
                                             E(k,1:3) = e'; EP(k,1:3) = ep';
% Robot Plotter con PDT
clear all; close all; clc
% Condiciones Iniciales
ff = pi/180; % Factor de Conv. Sex a Rad
amax(1) = 10*ff; % rad/s^2
                                             % Variable Intermedia para reducir las
vmax(1) = 25*ff; % rad/s
                                             dimesiones
amax(2) = 10*ff; % rad/s^2
                                             % Cálculo de las Matrices del
vmax(2) = 18*ff; % rad/s
                                             Manipulador
amax(3) = 1.00; % mts/s^2
                                             % Matriz de Inercias
vmax(3) = 0.25; % mts/s
amax(4) = 10*ff; % rad/s^2
                                             M = [(m2*(2*12^2 + 4*cos(q2)*12*13 +
vmax(4) = 25*ff; % rad/s
                                             2*13^2))/2 + (m3*(2*12^2 +
                                             4*\cos(q2)*12*13 + 2*13^2)/2 +
amax = amax';
                                             (m4*(2*12^2 + 4*cos(q2)*12*13 +
vmax = vmax';
                                             2*13^2)/2 + 12^2*m1, 13*(13 +
% Constantes de la Trayectoria
                                             12*\cos(q2))*(m2 + m3 + m4),
p0 = [80*ff -5*ff 0.5]';
pf = [5*ff 20*ff 1.5]';
                                             13*(13 + 12*\cos(q2))*(m2 + m3 + m4),
                                             13^2* (m2 + m3 + m4),
t0 = 0;
dt = 0.005; % Pequeño para simular
                                                                                   0,
                                             Ο,
continuidad
                                             m3/4 + m4/4];
x = [p0'pf']';
k = 1;
[R1, V1, A1] =
                                            % Inversa de M(q)
trayrobot2(amax, vmax, p0, pf, dt);
                                            MI = inv(M);
QD = [R1];
                                             % Matriz de Fuerzas Centrípetas y de
QDP = [V1];
                                             Coriolis
QDPP = [A1];
T = 0: (max(size(QD))-1);
T = T'*dt;
                                             C = [-12*13*q2p*sin(q2)*(m2 + m3 + m4),
qd = QD(:,1:3)';
qdp = QDP(:, 1:3)';
                                             -12*13*sin(q2)*(q1p + q2p)*(m2 + m3 +
qdpp = QDPP(:, 1:3)';
                                             m4), 0
                                               12*13*q1p*sin(q2)*(m2 + m3 + m4),
                                             0,0
                                                                               0,
                                             0, 0];
m1 = 20; m2 = 10; 11 = 4; 12 = 2; 13 = 2;
14=-5; % masas y longitudes
m3 = 20; m4 = 15;
g = 9.81; % Gravedad (m/s^2)
% Parámetros del Controlador
Kp = 1000 * eye(3); Kv = 100 * eye(3);
Ki=1*eye(3);
x = [x(1:3); x(4:6); [0 0 0]'];
for t=0:dt:(max(T))
                                            N = C*[q1p q2p q3p]';
X1(k,1) = x(1); X2(k,1) = x(2); X3(k,1)
                                             % Vector de Gravedad
= x(3);
```

X4(k,1) = x(4); X5(k,1) = x(5); X6(k,1)

```
G= [
               0
 (g*(m3 + m4))/2];
% Torque Computado (Señal de Control)
S = qdpp(:,k) + Kv*ep + Kp*e+ Ki*x(7:9);
tau = M*S+N+G;
% Ecuación de Estado
xp = [x(4:6)]
MI*(-N-G+tau)
el;
% Integración
x = x + xp*dt;
k = k + 1;
end
%Graficando
figure(1)
plot(T, E(:,1), 'r', T, E(:,2), 'b', T, E(:,3),
title('Error Articular PID de Torque
Computado')
xlabel('Tiempo (seg.)')
ylabel('Amplitud del Error')
grid on; zoom on
figure (2)
plot(T, X1, 'r-', T, X2, 'b-', T, X3, 'm-
',T,QD(:,1),'r:',T,QD(:,2),'b:',T,QD(:,3
), 'm:')
title('Brazo Robot-PID de Torque
Computado')
xlabel('Tiempo (seg.)');
ylabel('Amplitud de Angulo')
grid on; zoom on
figure(3)
plot(T, X4, 'r-', T, X5, 'b-', T, X6, 'm-
',T,QDP(:,1),'r:',T,QDP(:,2),'b:',T,QDP(
:,3),'m:')
title ('Brazo Robot-PID de Torque
Computado')
xlabel('Tiempo (seg.)')
ylabel('Amplitud de Velocidad Angular')
grid on; zoom on
% Trajectoria (PID) en espacio
cartesiano
x cart = 13*cos(X1 + X2) + 12*cos(X1);
y = 13*sin(X1 + X2) + 12*sin(X1);
z cart = 11 - 14 + X3;
xd cart = 13*cos(QD(:,1) + QD(:,2)) +
12*\cos(QD(:,1));
```

```
yd_cart = 13*sin(QD(:,1) + QD(:,2)) +
12*sin(QD(:,1));
zd_cart = 11 - 14 + QD(:,3);
figure(4)
plot3(x_cart,y_cart,z_cart,'r-
',xd_cart,yd_cart,zd_cart,'g-');
title(' Trayectoria del Extremo con un Controlador PID')
rotate3d on; grid on
```