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function [Pos, Vel, P_f, refrence_time] = sequential_rv_filter(times, measurements, R_obs, mu, Epoch)
%-----
% Extended Kalman Filter for orbit estimation from ECI position and
% velocity
%-----
% Inputs:
% times [Nx1] times in seconds of observations (assumes
% ascending order
%
% measurements [Nx4] [r, v] in [m] and [v/s]
%
% R_obs [4x4] covariance of each measurement (constant diagonal)
%
% mu [1] Gravitational Parameter [m^3/s^2 ]
%
% Epoch datetime The Epoch of all the measurements

% Outputs:
% Pos [3x1] Estimated final position in ECI [m]
%
% Vel [3x1] Estimated final velcoity in ECI [m]
%
% P_f [6x6] covariance of coodinate system [m^2 and (m/s)^2]
%
% refrence_time The time [s] of the final calcualted estimated trajectory
%
% Assumptions:
% - initial state is in the ECI frame
% - Keplerian two-body motion (no perturbations)

N = length(times);

% Initialization
X_0_star = measurements(1,:); % Refrence trajectory is first measurments
X_k_star = X_0_star;
P_k = R_obs(:,:,1);

% Initialize for History
X_k_hist = zeros(6, N);
X_k_hist(:, 1) = X_k_star;
P_k_hist = zeros(6, 6, N);
P_k_hist(:,:1) = P_k;

% Initialize Times
t_prev = times(1);
Epoch_prev = Epoch;
dt_max = 5*60; % 15 minutes in seconds largest propogation interval

for k = 2:N

    % Propagation Step
    dt_total = times(k) - t_prev;

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% Splitting if dt is to large:
if dt_total<=dt_max
    %Single step propogation
    Phi = Kepler_STM_UV(dt_total, X_k_star, mu, 'seconds');
    X_pred = Phi * X_k_star;
    P_k = Phi* P_k *Phi';
else
    N_steps = ceil(dt_total / dt_max);
    dt_step = dt_total / N_steps;
    X_temp = X_k_star;
    P_temp = P_k;
    for i = 1:N_steps
        Phi_step = Kepler_STM_UV(dt_step, X_temp, mu, 'seconds');
        X_temp = Phi_step * X_temp;
        P_temp = Phi_step * P_temp * Phi_step';
    end
    X_pred = X_temp;
    P_k = P_temp;
end

% Measurements Reading
Y_k = measurements(k,:)';
G = eye(length(measurements(1,:))); % Measruements in r and v already
y_k = Y_k - G*X_pred;
H_k_tilda = eye(6);

% Kalman gain
if (isscalar(R_obs(1,1,:)))
    R = R_obs;
else
    R = R_obs(:,:,k);
end
S = H_k_tilda * P_k * H_k_tilda' + R;
K = P_k * H_k_tilda' / S;

% Update
x_k_hat = K * y_k;
X_k_star = X_pred + x_k_hat;
I = eye(6);
P_k = (I - K*H_k_tilda) * P_k;
Epoch_prev = Epoch_prev+seconds(dt_total);
t_prev = times(k);

% Save
X_k_hist(:,k) = X_k_star;
P_k_hist(:,:,k) = P_k;
end

X_f_star = X_k_hist(:, end);
Pos = X_f_star(1:3);
Vel = X_f_star(4:6);
P_f = P_k_hist(:,:,:end);
refrence_time = t_prev;
end

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