

**Cambridge University Engineering Department**  
**Engineering Tripos Part IIA**  
**PROJECTS: Interim and Final Report Coversheet**

# IIA Projects

**TO BE COMPLETED BY THE STUDENT(S)**

Project:	SA1 - Aircraft Wing Analysis		
Title of report:	SA1 - Aircraft Wing Analysis Second Interim Report		
	Group Report / <del>Individual Report</del> (delete as appropriate)		
Name(s): (capitals)		crsID(s):	College(s):
SCHOEN CAO		zc282	Gonville & Caius
KAI TAN		ykt25	Wolfson
<u>Declaration</u> for: <del>Interim Report 1</del> / Interim Report 2 / <del>Final Report</del> (delete as appropriate)			
I/we confirm that, except where indicated, the work contained in this report is my/our own original work.			

**Instructions to markers of Part IIA project reports:**

**Grading scheme**

Grade	A / A*	B	C	D	E
Standard	Very Good / Excellent	Good	Acceptable	Minimum acceptable for Honours	Below Honours

Grade the reports on the scale A\* to D by marking the appropriate Overall Assessment box, and provide feedback against as many of the criteria as are applicable (or add your own). Feedback is particularly important for work graded C-E. Students should be aware that different projects and reports will require different characteristics.

*Penalties for lateness: Interim Reports: 3 marks per weekday; Final Reports: 0 marks awarded – late reports not accepted.*

Overall assessment (circle grade)	A*	A	B	C	D	E
Guideline standard	> 80%	70-80%	60-70%	50-60%	40-50%	< 40%

Marker:		Date:	
---------	--	-------	--

**Delete (1) or (2) as appropriate (for marking in hard copy – different arrangements apply for feedback on Moodle):**

- (1) Feedback from the marker is provided on the report itself.**
- (2) Feedback from the marker is provided on second page of cover sheet.**

# Project SA1 - Aircraft Wing Analysis Second Report

Schoen Cao (zc282), Kai Tan (ykt25)

May 23, 2019

## 1 Exercise 1

Listing 1: ueintbit.m

```
function f = ueintbit(xa,ua,xb,ub)

    %Calculate for variables
    ubar = (ua+ub)/2;
    du = ub-ua;
    dx = xb-xa;

    %Solve for integral
    f = (ubar^5 + 5/6*ubar^3*(du)^2 + 1/16*ubar*(du)^4)*dx;
end
```

Listing 2: Script for Exercise 1

```
clear
close all

%Input variables
ReL = 2500;
n = 101;
x = linspace(0,1,n);
ue = 1;

theta = zeros(1,n);    %initialising theta matrix

%Iterate for theta/L
for i=1:length(x);
    theta(i) = sqrt(.45/ReL*(ue)^-6*ueintbit(0,ue,x(i),ue));
end

hold on
plot(x,theta);    %Plot of Analytical Solution

thetab = 0.664/ReL^.5 .* (x).^-.5;    %Calculate Blasius Solution

plot(x,thetab); %Plot Blasius Solution

xlabel('Non-dimensional position , x/L');
ylabel('Non-dimensional momentum thickness , \theta/L');
title([ 'ReL=',num2str(ReL)]);
legend('Analytical','Blasius','location','Southeast');
hold off
```

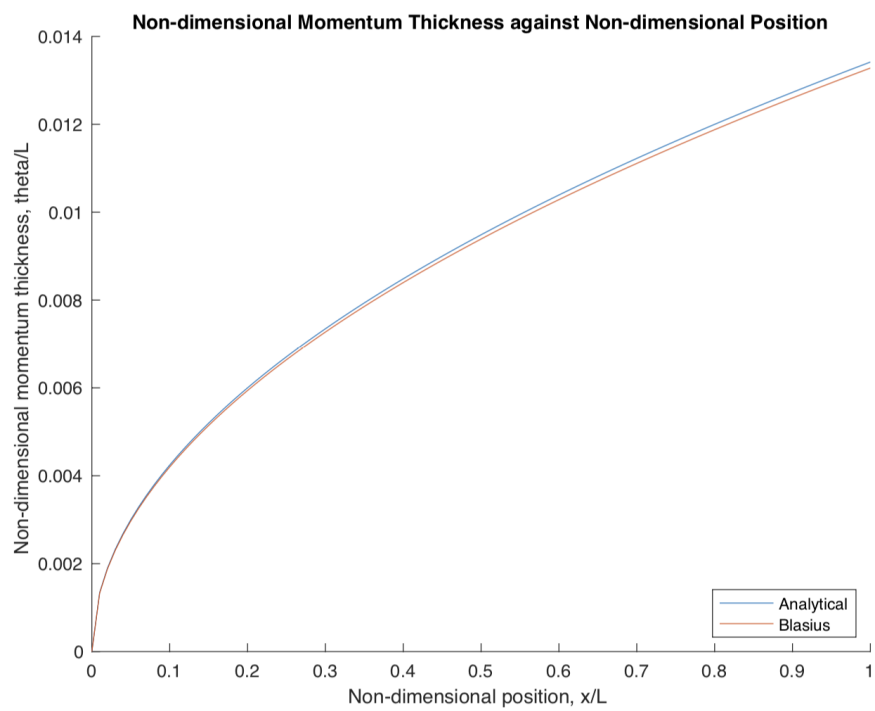


Figure 1: Plot of Non-Dimensional Momentum Thickness Against Non-Dimensional Position

## 2 Exercise 2

Listing 3: Script for Exercise 2

```

clear
close all

%Input Conditions, change according to required by Exercise
Re = 1e6;
duedx = -.2;

%Initial Conditions and Discretisation steps
n = 101;
laminar = true;
x = linspace(0,1,n);
ue0 = 1;

%Initialise transition and separation indicators
int = 0;
ils = 0;

%initialising matrices
theta = zeros(1,n);
He = zeros(1,n);

%Generate a matrix of ue values
for i = 1:n;
    ue(i) = duedx*x(i)+ue0;
end

%To initialise loop
i = 1;
while laminar && i < n; %laminar loop
    i = i+1; %Increase iteration counter

    %Solve for theta, Rethet, m, H, He
    theta(i) = sqrt(.45/Re*(ue(i))^-6*ueintbit(0,ue(1),x(i),ue(i)));
    Rethet = theta(i)*Re*(ue(i));
    m = -Re*(theta(i))^2*duedx;
    H = thwaites_lookup(m);
    He(i) = laminar_He(H);

    %Check for transition
    if log(Rethet) >= 18.4*He(i) - 21.74;
        laminar = false; %Flow no longer laminar
        int = i; %Save iteration when transition occurs
        display([x(i), Rethet/1000]);
    end
end
end

```

$d(u_e/U)/d(x/L)$	$Re_L (10^6)$	$x/L$	$Re_\theta (10^3)$
-0.2	1	0.58	0.5666
-0.2	10	0.24	1.0802
-0.2	100	0.04	1.3498
0	10	0.37	1.2903
0	100	0.04	1.3416
0.2	100	0.04	1.3337

Table 1: Transition Locations and Corresponding  $Re_\theta$  for Various Conditions

### 3 Exercise 3

Listing 4: Script for Exercise 3

```
clear
close all

%Input Conditions, change according to required by Exercise
Re = 1e6;
duedx = -.5;

%Initial Conditions and Discretisation steps
n = 101;
laminar = true;
x = linspace(0,1,n);
ue0 = 1;

%Initialise transition and separation indicators
int = 0;
ils = 0;

%initialising matrices
ue = zeros(1,n);
theta = zeros(1,n);
He = zeros(1,n);

%Generate a matrix of ue values
for i = 1:n;
    ue(i) = duedx*x(i)+ue0;
end

%To initialise loop
i=1;
while laminar && i < n; %laminar loop
    i = i+1; %Increase iteration counter

    %Solve for theta, Rethet, m, H, He
    theta(i) = sqrt(.45/Re*(ue(i))^-6*ueintbit(0,ue(1),x(i),ue(i)));
    Rethet = theta(i)*Re*(ue(i));
    m = -Re*(theta(i))^2*duedx;
    H = thwaites_lookup(m);
    He(i) = laminar_He(H);

    %Check for transition
    if log(Rethet) >= 18.4*He(i) - 21.74; %Transition condition
        laminar = false; %Flow no longer laminar
        int = i; %Save iteration where transition occurs

    %Check for separation
    elseif m >= 0.09; %Separation Condition
        laminar = false; %Flow no longer laminar
        ils = i; %Save iteration where transition occurs
    end
end

if int ~= 0;
    disp(['Natural transition at ' num2str(x(int)) ' with Rethet ' num2str(Rethet)]);
elseif ils ~= 0;
    disp(['Laminar separation at ' num2str(x(ils)) ' with Rethet ' num2str(Rethet)]);
end
```

$Re_L$	Separation Location (x/L)
$10^4$	0.25
$10^5$	0.25
$10^6$	0.25

Table 2: Separation Location for Various Flow Conditions

Listing 5: Script for Exercise 3 to Find  $Re_L$  at which Transition Supplants Laminar Separation

```

clear
close all

%Input Conditions, change according to required by Exercise
Re = linspace(1e6,1e7,9001);
duedx = -.5;

%Initial Conditions and Discretisation steps
n = 101;
x = linspace(0,1,n);
ue0 = 1;

%Generate a matrix of ue values
for i = 1:n;
    ue(i) = duedx*x(i)+ue0;
end

%To initialise loop
for k = 1:length(Re);
    i=1;

    %Initialise indicators at the start of every loop
    int = 0;
    ils = 0;

    %Reset laminar flag at the start of every loop
    laminar = true;

    %Laminar loop
    while laminar && i < n;
        i = i+1;    %Increase interation counter

        %Solve for theta, Rethet, m, H, He
        theta(i) = sqrt(.45/Re(k)*(ue(i))^-6*ueintbit(0,ue(1),x(i),ue(i)));
        Rethet = theta(i)*Re(k)*(ue(i));
        m = -Re(k)*(theta(i))^2*duedx;
        H = thwaites_lookup(m);
        He(i) = laminar_He(H);

        %Check for transition
        if log(Rethet) >= 18.4*He(i) - 21.74;    %Transition condition
            laminar = false;    %Flow no longer laminar
            int = i;    %Save iteration where transition occurs

        %Check for separation
        elseif m >= 0.09;    %Separation Condition
            laminar = false;    %Flow no longer laminar
            ils = i;    %Save iteration where transition occurs
        end
    end

    %Display Re at which transition will supplant laminar separation
    if int ~= 0; %If transition occurs before separation, ils = 0
        disp(['Re_L at which transition supplants laminar transion is '...
            ,num2str(Re(k))])
        break %break loop once required Re is found
    end
end
end

```

From this code, it was found that the minimum  $Re_L$  for transition to supplant laminar separation is  $1.792 \times 10^6$

## 4 Exercise 4

Listing 6: thickdash.m

```
function dthickdx = thickdash(xmx0,thick);
    %import global variables
    global Re ue0 duedx

    %find theta and deltaE as defined
    theta = thick(1);
    deltaE = thick(2);

    %calculate He
    He = deltaE/theta;

    %calculate H
    if He >= 1.46;
        H = (11*He+15) / (48*He-59);
    else
        H = 2.803;
    end

    %calculate ue
    ue = ue0 + duedx * xmx0;

    %calculate Re_theta
    Rethet = Re * ue * theta;

    %calculate cf
    cf = 0.091448*((H-1)*Rethet)^(-.232)*exp(-1.260*H);

    %calculate cdiss
    cdiss = 0.010019*((H-1)*Rethet)^(-1/6);

    %calculate f
    dthickdx = zeros(2,1);
    dthickdx(1) = cf/2 - (H+2) / ue * duedx * theta;
    dthickdx(2) = cdiss - 3/ue * duedx * deltaE;
end
```

Listing 7: Script for Exercise 4

```
clear
close all

%Defining global variables value
global Re ue0 duedx

Re = 1e7;
ue0 = 1;
duedx = 0;

%Initial values of theta and delta-E
x0 = 0.01;
thick0(1) = 0.037*x0*(Re*x0)^(-1/5);
thick0(2) = 1.80 * thick0(1);

[delx thickhist] = ode45(@thickdash,[0 0.99],thick0);

for i = 1:length(delx);

    theta_7(i) = 0.037 * (x0+delx(i)) * (Re * (x0+delx(i)))^(-1/5);
    theta_9(i) = 0.023 * (x0+delx(i)) * (Re * (x0+delx(i)))^(-1/6);
end
```



```

hold on
plot(deltax+x0, thickhist(:,1));
plot(deltax+x0, theta_7);
plot(deltax+x0, theta_9);
xlabel('non-dimensional position x/L');
ylabel('non-dimensional momentum thickness \theta/L');
legend('Differential equation', '1/7^{th} Power Law Estimate', '1/9^{th} Power Law ...
Estimate', 'location', 'Southeast');
hold off

```

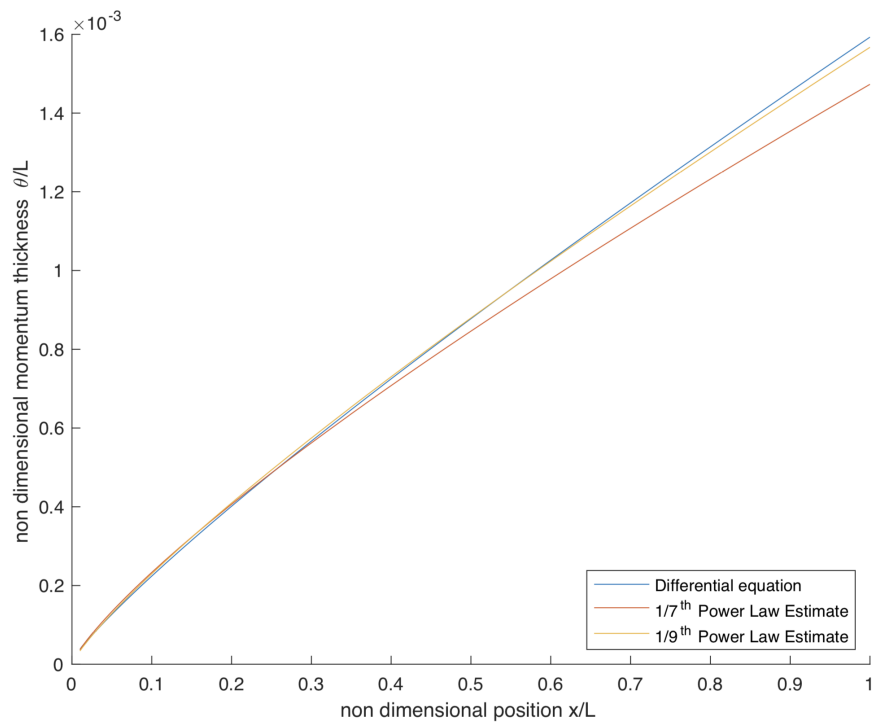


Figure 2: Non-Dimensional Momentum Thickness against Non-Dimensional Position for Various Solutions

## 5 Exercise 5

Listing 8: Script for Exercise 5

```
clear
close all

%Defining global variables value
global Re ue0 duedx

Re = 1e7;
ue0 = 1;
duedx = -.3;

%Boundary values of theta and delta_E
x0 = 0.01;
thick0(1) = 0.037*x0*(Re*x0)^(-1/5);
thick0(2) = 1.80 * thick0(1);

[delx thickhist] = ode45(@thickdash,[0 0.99],thick0);

%Calculate He
for i = 1:length(delx);
    He(i) = thickhist(i,2)/thickhist(i,1);
end

%Plotting duedx = -.6
clear He
duedx = -.6;
[delx thickhist] = ode45(@thickdash,[0 0.99],thick0);
for i = 1:length(delx);
    He(i) = thickhist(i,2)/thickhist(i,1);
end

%Plot theta and deltaE
figure(1);
hold on
plot(delx+x0,thickhist(:,1));
plot(delx+x0,thickhist(:,2));
xlabel('non_dimensional_position_x/L');
ylabel('non_dimensional_thickness');
legend('\theta','\delta_E','location','Northwest');
title(['Re_L=',num2str(Re),' d(u_e/U)/d(x/L)=',num2str(duedx)]);
hold off

%He plot to be done command window and by changing the input variables
```

Listing 9: Command Window Scripts Used to Plot Manually

```
%Plot He
plot([0 1],[1.46 1.46], '—'); %Plot reference value
hold on
plot(delx+x0,He); %Plot He, repeat for every run with different condition
xlabel('non_dimensional_position_x/L');
ylabel('Energy_shape_factor_H_E');
title(['Re_L=',num2str(Re)]);
plot(delx+x0,He);
xlabel('non_dimensional_position_x/L');
ylabel('Energy_shape_factor_H_E');
title(['Re_L=',num2str(Re)]);
legend('H_E=1.46 Separation condition','d(u_e/U)/d(x/L)=-0.3',...
'd(u_e/U)/d(x/L)=-0.6','d(u_e/U)/d(x/L)=-0.9','location','Southwest');
```

By plotting the graphs for  $H_E$  again  $x/L$  using the command window, the turbulent separation for the various flow conditions can be found. This is done by running the code at various conditions, and plotting the  $H_E$  for each condition. The list of commands are shown in listing 9. The results are tabulated in Table 3 and Table 4. It should be noted that these values are read off the graph and are approximated values of the actual separation location.

$d(u_e/U)/d(x/L)$	Turbulent Separation Location $x/L$
-0.3	NA
-0.6	0.81
-0.9	0.54

Table 3: Turbulent Separation Location for Various Velocity Gradient at  $Re_L = 10^7$

$Re_L$	Turbulent Separation Location $x/L$
$10^6$	0.73
$10^7$	0.81
$10^8$	0.90

Table 4: Turbulent Separation Location for Various  $Re_L$  at  $d(u_e/U)/d(x/L) = -0.6$

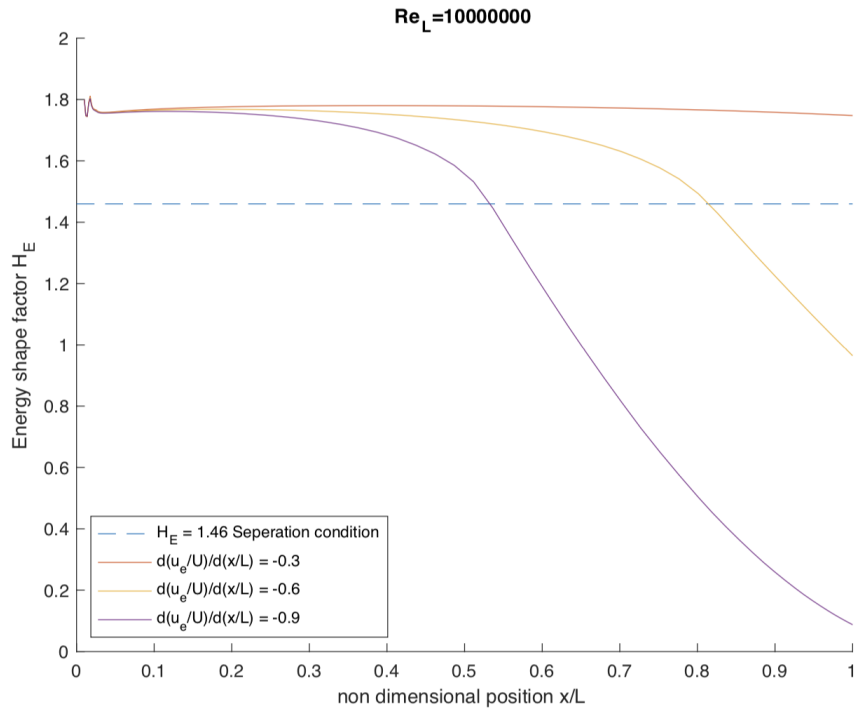


Figure 3: Plot of Turbulent Separation for Various Velocity Gradient at  $Re_L = 10^7$

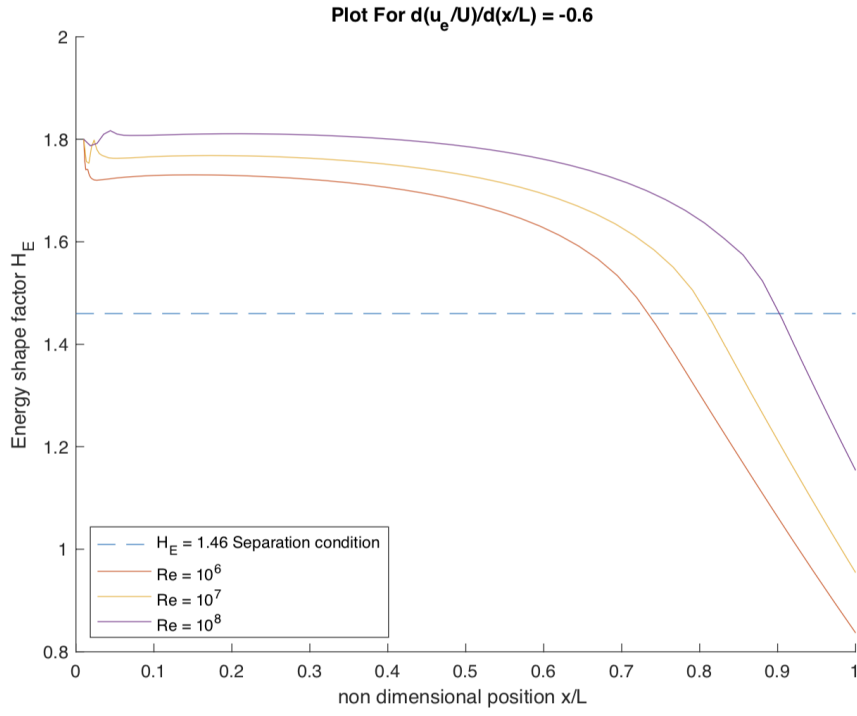


Figure 4: Plot of Turbulent Separation Location for Various  $Re_L$  at  $d(u_e/U)/d(x/L) = -0.6$

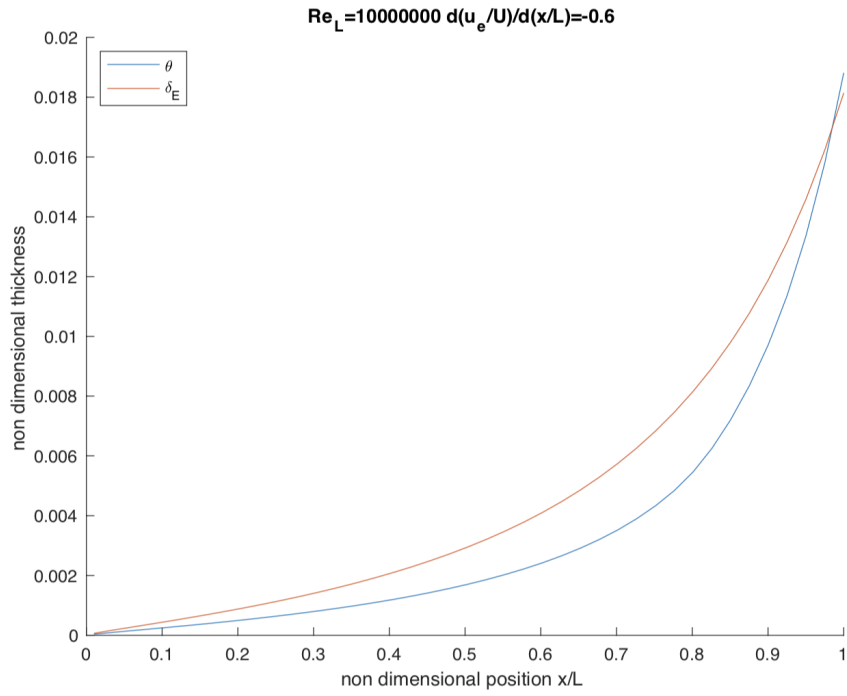


Figure 5: Plot of  $\frac{\delta_E}{L}$  and  $\frac{\theta}{L}$  Against Non-Dimensional Position

## 6 Exercise 6

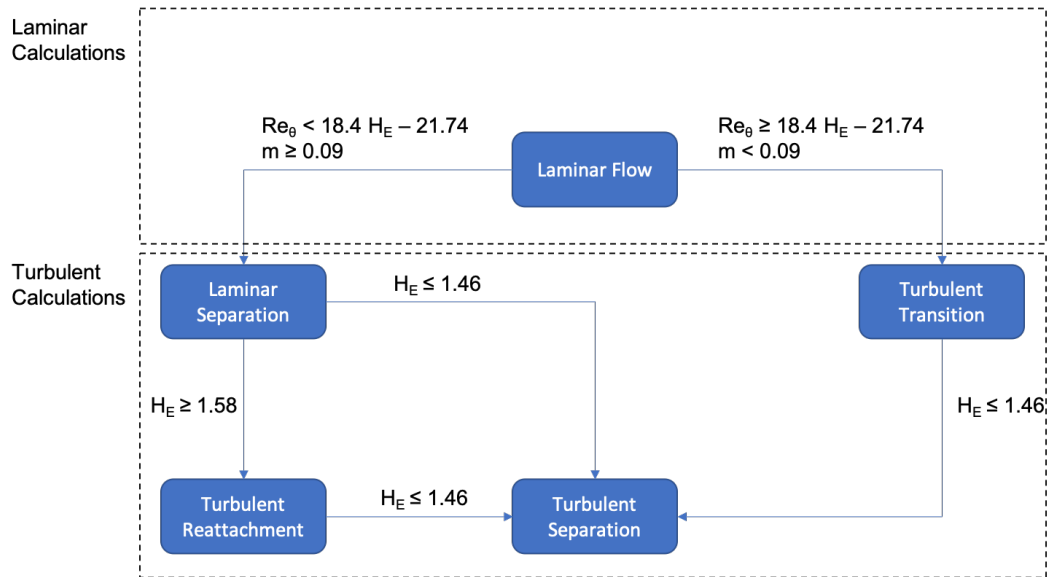


Figure 6: Flow Chart of Development of Flow

Listing 10: Script for Exercise 6

```

clear
close all

%Defining global variables value
global Re ue0 duedx

%Define simulation conditions
Re = 1e5;
duedx = -0.25;
ue0 = 1;

%Iteration setting & initial conditions
n = 101;
laminar = true;
x = linspace(0,1,n);

%%initialising indicators
int = 0; %natural transition
ils = 0; %laminar separation
itr = 0; %turbulent reattachment
its = 0; %turbulent separation

%generating ue matrix
for i = 1:n;
    ue(i) = duedx*x(i)+ue0;
end

%initialising i
i = 1;
while laminar && i < n; %laminar loop
    i = i+1; %Iteration counter
    theta(i) = sqrt(.45/Re*(ue(i))^-6*ueintbit(0,ue(1),x(i),ue(i)));
    Rethet = theta(i)*Re*(ue(i));
    m = -Re*(theta(i))^2*duedx;
    H = thwaites_lookup(m);
    He(i) = laminar_He(H);

    %Check for Transition
    
```

```

    if log(Rethet) >= 18.4*He(i) - 21.74; %laminar check
        laminar = false; %laminar flag & end loop
        int = i; %set pointer
        disp(['Turbulent Transition at x/L= ' num2str(x(int)) ' at Re_L ...
            num2str(Re)]);

    %Check for Laminar Separation
    elseif m >= 0.09;
        laminar = false; %laminar flag & end loop
        ils = i;
        He(i) = 1.51509; %set He to separated value
        disp(['Laminar Separation at x/L= ' num2str(x(ils)) ' at Re_L ...
            num2str(Re)]);
    end
end

%Value for He for Laminar Flow
He(1) = 1.57258;

%Calculate deltaE matrix
deltaE = He.*theta;

%Turbulent Loop after separation or transition
while its == 0 && i < n;
    thick0(1) = theta(i); %y matrix, value at elemental plate's start
    thick0(2) = deltaE(i);
    i = i+1;
    [delx thickhist] = ode45(@thickdash,[0,x(i)-x(i-1)],thick0);
    theta(i) = thickhist(length(delx),1);
    deltaE(i) = thickhist(length(delx),2);
    He(i) = deltaE(i)/theta(i);

    %Check for turbulent reattachment
    if ils > 0 && He(i) >= 1.58 && itr == 0;
        itr = i;
        disp(['Turbulent Reattachmemt at x/L= ' num2str(x(itr)) ...
            ' at Re_L ' num2str(Re)]);
    end

    %Check for turbulent separation
    if He(i) <= 1.46; %turbulent seperation check
        its = i;
        H=2.803; %H at seperation
        disp(['Turbulent Separation at x/L= ' num2str(x(its)) ...
            ' at Re_L ' num2str(Re)]);
    end
end

while i < n; %final loop
    theta(i+1) = theta(i)*(ue(i)/ue(i+1))^(H+2); %theta for cf=0
    i = i+1;
    He(i) = He (its); %H assumed to remain constant since He is constant
end

%Plot graph using command window as with Ex 5.

```

As mentioned in Exercise 5, the graphs for Exercise 6 is done is plotted manually using the command window using similar commands.

```

clear
close all

%Defining global variables value
global Re ue0 duedx

%Define simulation conditions
Re = 1e5;
duedxtest = linspace(-0.55,-0.25,31); %Create an array for test gradient
ue0 = 1;

%Iteration setting & initial conditions
n = 101;
laminar = true;
x = linspace(0,1,n);

for k = 1:length(duedxtest); %loop for various velocity gradient conditions
    duedx = duedxtest(k);

    %generating ue matrix
    for i = 1:n;
        ue(i) = duedx*x(i)+ue0;
    end

    %initialising i
    i = 1;
    laminar = true;
    int = 0;    %natural transition
    ils = 0;    %laminar seperation
    itr = 0;    %turbulent reattachment
    its = 0;    %turbulent seperation

    while laminar && i < n; %laminar loop
        i = i+1;    %increase i counter
        theta(i) = sqrt(.45/Re*(ue(i))^-6*ueintbit(0,ue(1),x(i),ue(i)));
        Rethet = theta(i)*Re*(ue(i));
        m = -Re*(theta(i))^2*duedx;
        H = thwaites_lookup(m);
        He(i) = laminar_He(H);

        %Check for Transition
        if log(Rethet) >= 18.4*He(i) - 21.74;
            laminar = false;    %laminar flag & end loop
            int = i;    %set pointer

        %Check for Laminar Separation
        elseif m >= 0.09;
            laminar = false;    %also end laminar loop & turbulent formula
            ils = i;
            He(i) = 1.51509;    %set He to separated value
        end
    end

    %Value for He for Laminar Flow
    He(1) = 1.57258;

    %Calculate deltaE matrix
    deltaE = He.*theta;

    %Turbulent Loop
    while its == 0 && i < n;
        thick0(1) = theta(i);    %y matrix, value at elemental plate's start
    end
end

```

```

thick0(2) = deltaE(i);
i = i+1;
[delx thickhist] = ode45(@thickdash,[0,x(i)-x(i-1)],thick0);
theta(i) = thickhist(length(delx),1);
deltaE(i) = thickhist(length(delx),2);
He(i) = deltaE(i)/theta(i);

%Check for turbulent reattachment
if ils > 0 && He(i) >= 1.58 && itr == 0;
    itr = i;

end

%Check for turbulent separation
if He(i) <= 1.46; %turbulent seperation check
    its = i;
    H=2.803; %H at seperation
end
if i==101 && its ~=0
    disp(['Critical_Velocity_Gradient_is_' num2str(duedx)]);
    break %break loop if turbulent separation occurs at x/L = 1.
end

end
end
end

```

Critical Velocity Gradient is found to be -0.51 (2 significant figures)

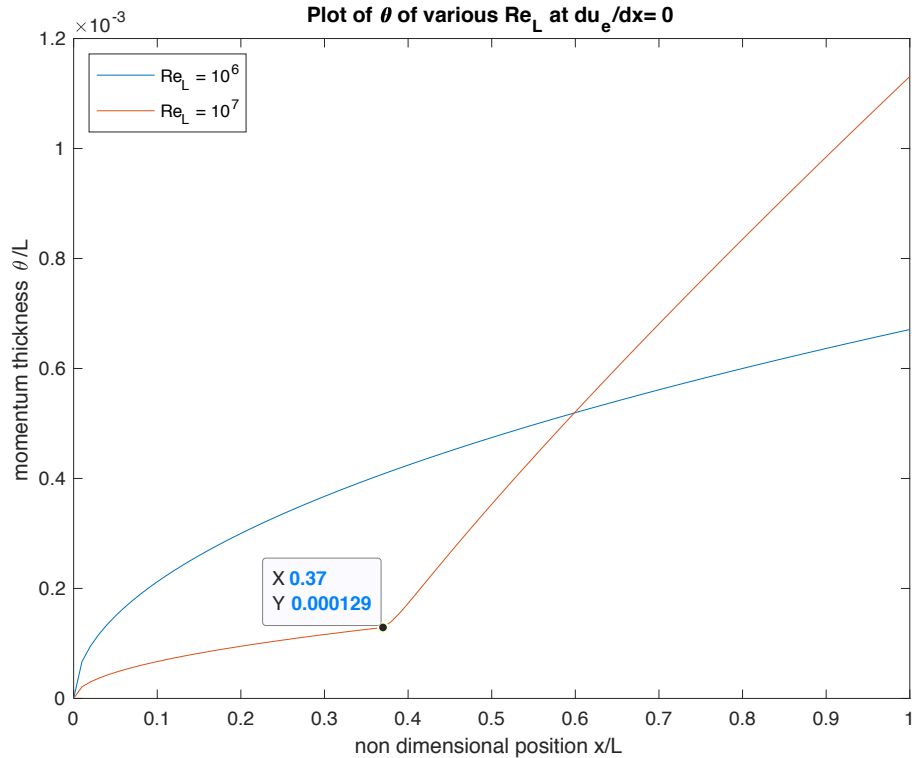


Figure 7: Plot of  $\frac{\theta}{L}$  Against  $\frac{x}{L}$  for Various  $Re$  and  $d(u_e/U)/d(x/L) = 0$



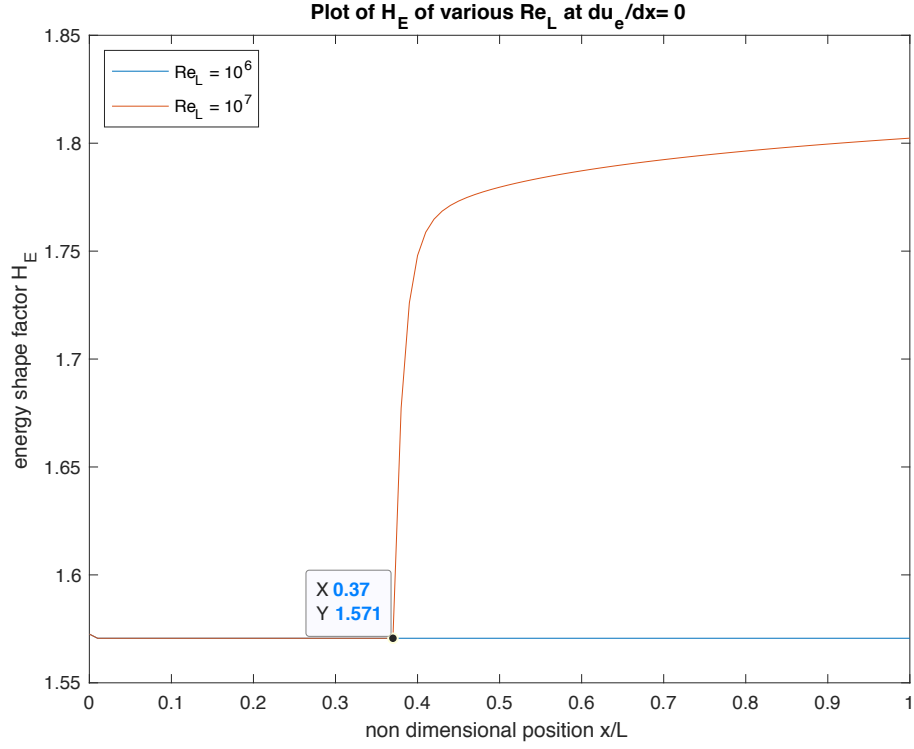


Figure 8: Plot of  $H_E$  Against  $\frac{x}{L}$  for Various  $Re$  and  $d(ue/U)/d(x/L) = 0$

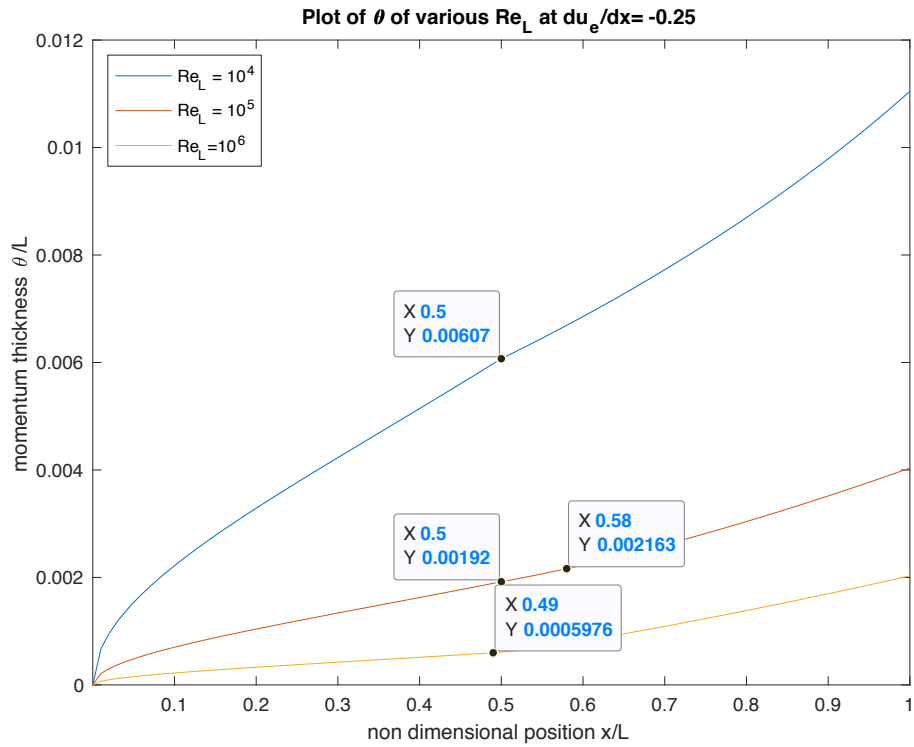


Figure 9: PPlot of  $\frac{\theta}{L}$  Against  $\frac{x}{L}$  for Various  $Re$  and  $d(ue/U)/d(x/L) = -0.25$

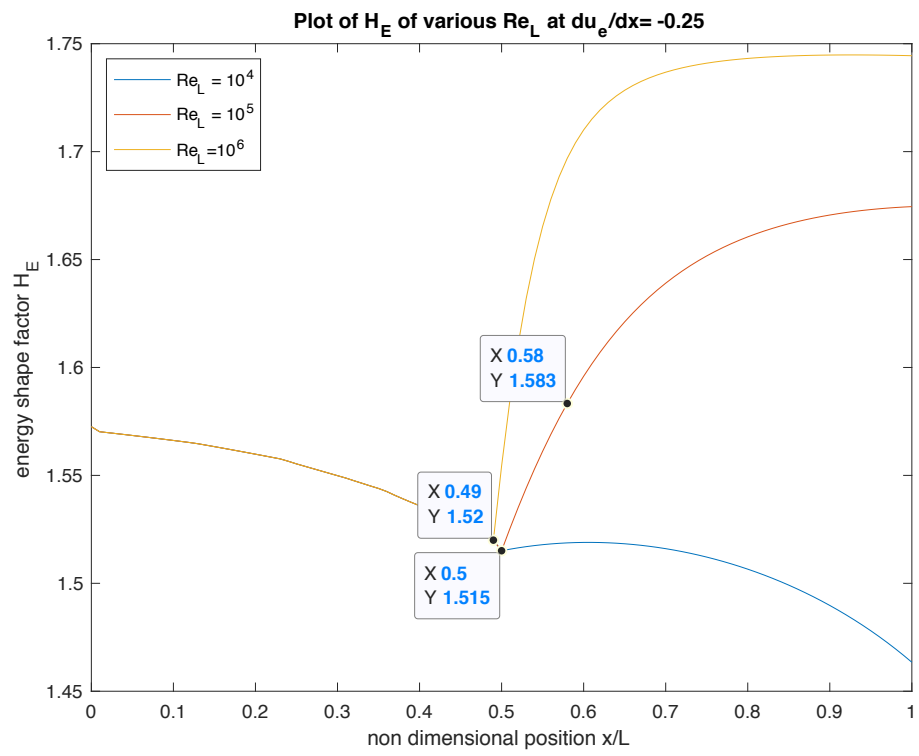


Figure 10: Plot of  $H_E$  Against  $\frac{x}{L}$  for Various  $Re$  and  $d(u_e/U)/d(x/L) = 0$