



cFS Overview



Version 1.10 December 2023



Audience & Prerequisites



Objectives

Introduce flight software (FSW) concepts and core Flight System (cFS) architectural features

Intended audience

Spacecraft technical managers, systems engineers, discipline engineers, and software engineers

Prerequisites

- An interest in learning about FSW and the cFS
- No programming skills required



Outline



- 1. Flight Software Introduction
- 2. Core Flight System(cFS) Introduction

Appendix A: Acronyms

Appendix B: cFS Architectural Goals

Appendix C: Online Resources

cFS Overview





Flight Software Introduction

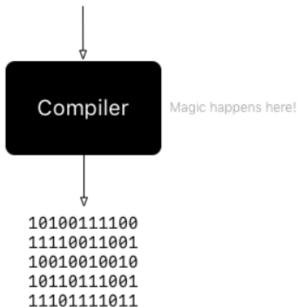


What is Software?



- Microcode are instructions and data processed by a microprocessor
- Programming languages are formal languages (i.e. grammatical rules) that are translated into microcode by a compiler

```
func greet() = {
    Console.println("Hello, World!")
}
```



 Interpreted languages such as Python, Ruby, and JavaScript run on a virtual machines (VMs) and the VM is ported to different processors



What is an Embedded Computer?



- An Embedded Computer is a computer that is integrated in a product
- Embedded Computers do not usually have a keyboard, mouse, or monitor interface



Embedded Software is software that that runs on an Embedded Computer in order to make a device or product work

"Embedded software is the opposing thumb of hardware"



Embedded Software Characteristics



Microcontroller

- Typically, limited resources: low power, small memory, and slower clock speeds
- Typically, no operating system
- For example, the processor embedded with reaction wheel assembly

Embedded Operating System

- Real-time preemptive multi-tasking
- Typically, much smaller than Windows or OS X
- Examples include VxWorks, RTEMS, and FreeRTOS

Embedded System

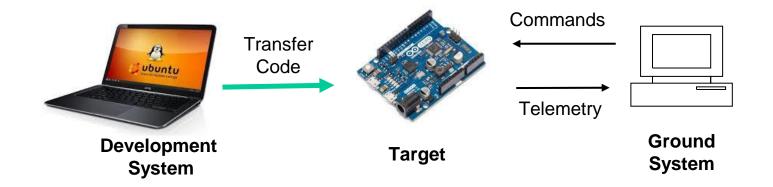
The combination of an Embedded Computer and Embedded Software



How do we Develop Embedded Software?



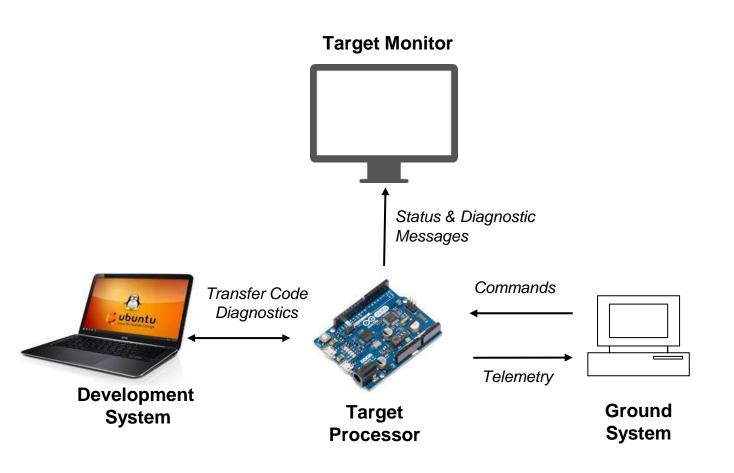
- Use a Windows, Mac, or Linux computer to write the code
 - Programming Languages used
 - For Flight Software: C, C++, Assembly Language, sometimes Ada
 - For Ground and test software: C, C++, Python, Java, Ruby, etc.
- Write code, cross-compile for target processor, transfer object code to embedded computer, control embedded system with a ground system
- This is known as Cross Development





Cross Development Environment





Target Processor

A processor that runs the cFS target image

Development System

- Used to build and transfer the cFS target image to the target processor
- Requires a 'cross compiler' if the target process is different than the development system
- May include runtime diagnostic tools

Target Monitor

- A common diagnostic tool used to help verify the embedded system is operating correctly
- Often a monitor connected over a serial port

Ground System

- An application that sends command messages to the target and receives telemetry messages from the target
- The command & telemetry communications link may vary between test configurations and operations

cFS Introduction



Spacecraft Definitions



A few terms:

- Spacecraft Bus usually refers to the fundamental systems of a spacecraft, i.e.
 - Mechanical Structure
 - Electrical System
 - Power System
 - Command and Data Handling System (C&DH)
 - Attitude Control System/ Propulsion System
 - RF System
 - Thermal System
- **Payloads** refers to the instruments on board, i.e.
 - Cameras, Telescopes, Radars, etc
- Observatory Usually refers to the entire system, i.e. the combination of the Spacecraft Bus and the Payloads



FSW Challenges – Limited Resources



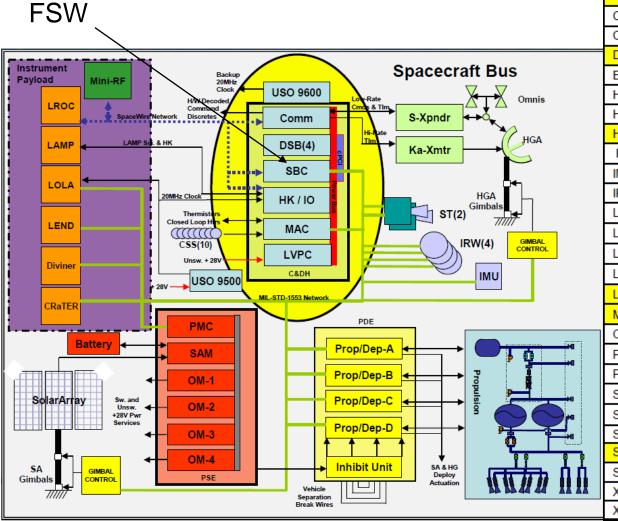
- Flight processor clock speeds are in the MHz range while our laptops are in the GHz range
 - NICER (6/12/17) 83Mhz Broad Reach 440 Power PC
- Non-volatile memory
 - GPM (2/27/14) 2MB banks of EEPROM for each FSW image
 - Compressed operating system and apps
- Flight RAM in Megabyte range while our laptops are in the Gigabyte range
 - GPM (2/27/14) 36MB RAM



FSW Challenges – Large number of Custom Interfaces



More than 20 Interfaces controlled by FSW on this single string example



C&DH	Command & Data Handling			
Comm	Communication			
CRaTER	Cosmic Ray Telescope for the Effects of Radiation			
CSS	Coarse Sun Sensor			
DSB	Data Storage Board			
EVD	Engine Valve Driver			
HGA	High-Gain Antenna			
H/W	Hardware			
HK	Housekeeping			
Ю	Input/Output			
IMU	Inertial Measurement Unit			
IRW	Integrated Reaction Wheel			
LAMP	Lyman-Alpha Mapping Project			
LEND	Lunar Exploration Neutron Detector			
LOLA	Lunar Orbiter Laser Altimeter			
LROC	Lunar Reconnaissance Orbiter Camera			
LVPC	Low Voltage Power Converter			
MAC	Multi-Function Analog Card			
OM	Output Module			
PMC	Power Management Controller			
PSE	Power System Electronics			
SA	Solar Array			
SAM	Solar Array Module			
ST	Star Tracker			
SBC	Single Board Computer			
SSR	Solid State Recorder			
Xmtr	Transmitter			
Xpndr	Xpndr Transponder			



FSW Challenges – Robustness & Remote Diagnostics



Curiosity Probe's Computer Reset (2/25/19)

 "Last week, its computer rebooted without warning. Now, NASA engineers are trying to figure out what caused the unprompted restart."



Israeli Lunar Lander Suffers Glitch on Way to the Moon (2/27/19)



- "During the pre-maneuver phase the spacecraft computer reset unexpectedly, causing the maneuver to be automatically cancelled,"
- "The engineering teams ... are examining the data and analyzing the situation. At this time, the spacecraft's systems are working well, except for the known problem in the star tracker.
- https://www.space.com/israel-moon-landersuffers-glitch.html



FSW Challenges – Manage & Protect Hardware



- Detect flight hardware anomalies and failures
 - Sensors, Actuators, Clocks, CPU, Memory, Power, Thermal, Communications
- Respond to onboard anomalies/failures pre-planned actions, e.g.,
 - Capture Diagnostic Data
 - Use "numerically safe" data for current control law cycle
 - Reconfigure onboard hardware and/or software minimize the problem
 - 'Safe' all Flight Hardware (Science Instruments, Spacecraft Hardware)
 - Ensure Sun is on the Solar Array Panels -- Maintain Power Positive State
 - Notify Ground of problems
- FSW can often compensate for hardware problems found during spacecraft I&T or post-launch



FSW Challenges – Serving Multiple Customers



Principle Investigator

"I want all science data in a timely manner"

Other spacecraft subsystems

"It would be really nice if I could see data X, Y, and Z at rate N"

FSW Test Team

"We want to see all of the data all of the time"

Spacecraft Integration and Test team

- "We want to see or at least log all of the data all of the time"
- "We also need to test as we will fly"

Operational Managers and Spacecraft Operators

"Make my life simple and cheap"

FSW On Orbit Maintenance Team

"Can we access/view data/code X and change it to Y?"





FSW Challenges – Complex Functionality



- FSW systems often have a large number of functional requirements as shown on the next slide
- The functions need to operate concurrently and at different rates
- The functions have different levels of criticality and lower critical functions must not impact mission critical functions



Hardware & Software

Memory Integrity

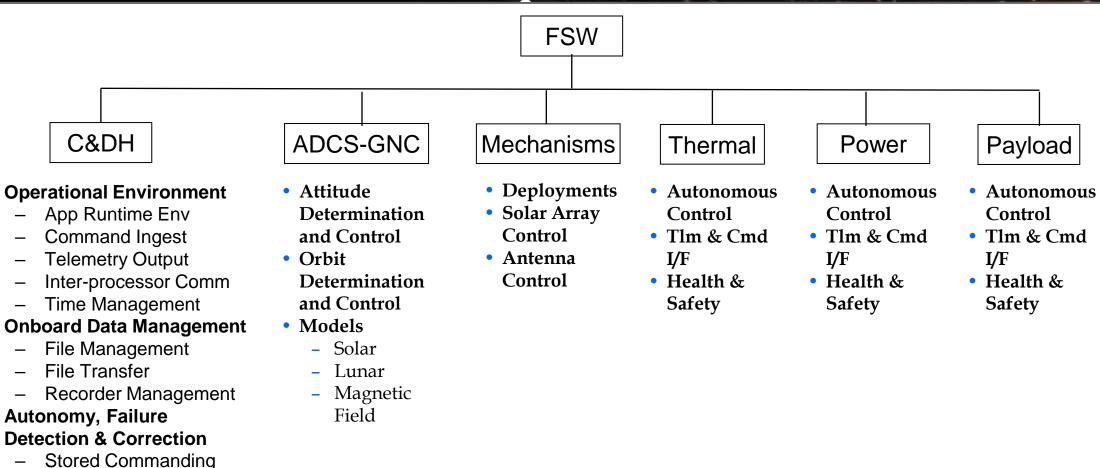
Support & Checks

Monitoring

Typical Spacecraft Flight Software Requirements



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ADCS: Attitude Determination & Control System

C&DH: Command & Data Handling **GNC**: Guidance Navigation and Control

cFS Introduction





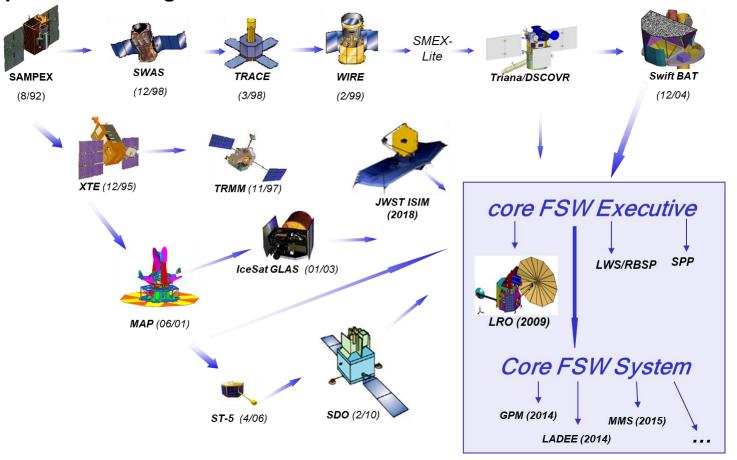
Core Flight System Introduction



Core Flight System Motivation



For decades the NASA Goddard Space Flight Center has been designing, building, and operating spacecraft customized for specific science goals



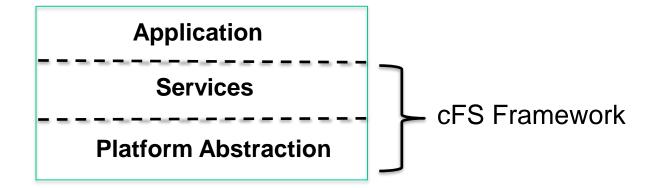
- Flight software (FSW) design patterns evolved that leveraged commonality and parametrized variability
- The core Flight System (cFS) was developed as project-independent FSW product line



Core Flight System Architecture



- The cFS was intentionally designed to meet specific cFS architectural goals and quality attributes that are summarized in Appendix C
- The cFS uses a three-tiered software architecture that provides a portable and extendable framework with a product line deployment model
 - Platform Abstraction Layer supports portability
 - The Service Layer provides an Application Programming Interface (API) enabling application reuse across missions
 - Compile-time configuration parameters and run-time command/table defined parameters support adaptability and scalability





cFS Benefits

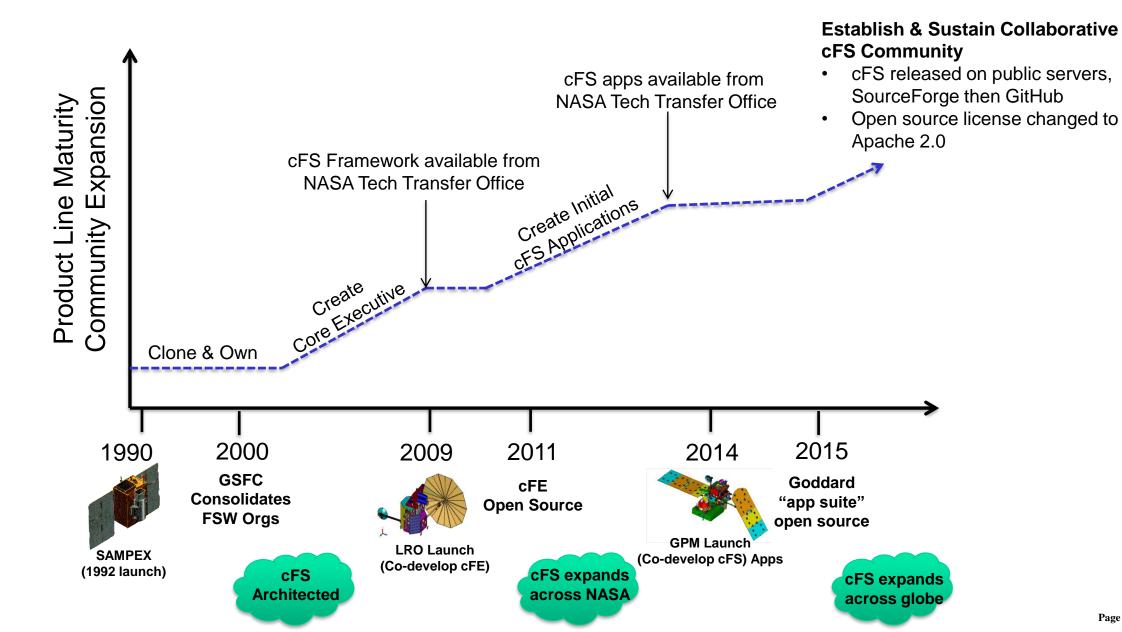


- The cFS Framework and reusable app suite designs are based on lessons learned over decades of experience
 - These lessons include the entire spacecraft lifecycle that includes development, operations, and maintenance
 - This perspective is very important because each phase may impose design requirements and/or constraints that are not always complimentary
- The cFS component designs and parameterization address many of the FSW challenges previously listed, for example...
 - The cFS framework provides services to preserve data across process resets
 - Mission functionality is distributed across apps that each have their own execution thread and priority
 - Message filter tables allow telemetry downlink content and rates to be tuned for specific testing and operational use cases
- FSW engineering is a systems engineering endeavor because the FSW is integrating multiple subsystems into a single system that must operate as a single system. In order to design the best cFS system, or any FSW system, a FSW expert should be involved from early formulation stages to...
 - Participate in trades: ground/flight, hardware/software, build/buy, budget options, etc.
 - Champion requirements, design, and interface features with knowledge of all of the customers and spacecraft lifecycle needs



cFS Historical Timeline



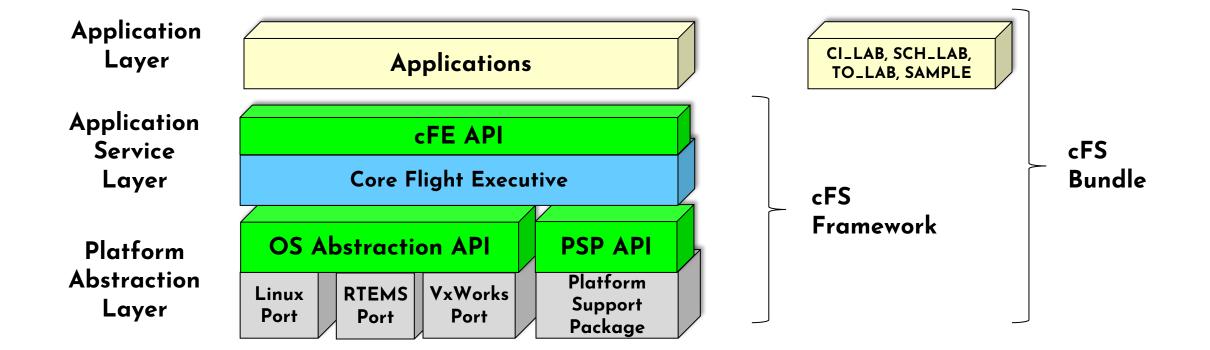






cFS Layered Architecture





TBD Notes

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Application Layer Benefits



- Write once run anywhere the cFS framework has been deployed
- Framework has been ported to many popular hardware platform/operating system platforms including MUSTANG and SpaceCube
- Goddard has released 15 applications that provide common command and data handling functionality such as
 - Stored command management and execution
 - Onboard data storage file management
- Reduce project cost and schedule risks
 - High quality flight heritage applications
 - Focus resources on mission-specific functionality
- Framework provides seamless application transition from technology efforts to flight projects



Key Definitions



Framework

- The set of individual services, applications, tools, and infrastructure managed by the open source community Configuration Control Board (CCB).

Bundle

- An executable version of the framework configured for a nominal Linux system. Links compatible versions of the framework elements as a recommended starting point for new cFS-based systems.

Component

An individual application, service, or tool that can be used in a cFS-based system

Distribution

 A set of custom components packaged together with the framework; generally created and provided by a cFS user (individual or group) with specific needs (e.g. a NASA center, the GSFC SmallSat Project Office)

cFE vs cFS

 cFE is the Core Flight Executive services and API – cFS is a general collective term for the framework and the growing set of components



Current State of cFS Open Source Program



- cFS Framework funding primarily comes from projects
 - In 2023 the Artemis Gateway program is the primary funding (cFS is in their interoperability spec)
 - cFS strategic decisions are driven by NASA projects which may or may not align with the open source community's priorities
- A NASA multi-center Configuration Control Board (CCB) manages releases of the open source cFS
 Framework
 - The CCB works with projects to determine which issues and changes are in each release
- Cumbersome cFS Framework release process (discussed later)
- Applications are maintained by the organization that created the apps
 - Currently NASA Ames, Goddard, and Johnson maintain individual app repositories
 - The organization is responsible for funding and for maintaining compatibility with different cFS Framework versions
 - Other organizations have created apps that are included in a cFS Distribution (discussed later)



cFS Open Source Community Challenges



Hard for organizations to adopt the cFS

- Steep learning curve with some online training material but no courses offered
- Requires highly motivated teams to train themselves
- On small teams, there may only be one cFS expert so they become critical to the project's success

Successfully using the cFS on a mission requires disciplined systems engineering

- Most of the available online resources do not address cFS systems engineering
- The cFS repo list available components and distributions but it's up to the user to figure which components they
 may need and how to use them
- Apps often work together to achieve goals and implement functional requirements
 - The reusable apps are documented as separate components so it is often unclear how the apps can be used collectively

Github discussions are the primary technical support mechanism

They're helpful but are responses are not always timely

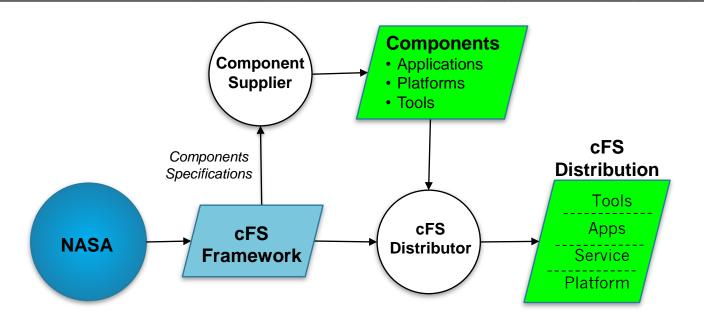
Some artifacts were developed for Goddard-specific ground systems and have not been redesigned

- Table tools Generate binary table files and display binary table files
- Testing scripts



Conceptual cFS Product Model





- Organizations, not just NASA, supply components (applications, platforms and tools) and cFS Distributions
 - What would motivate an organization to be a supplier or distributor?
 - Would organizations spend resources on assets given the current NASA cFS Framework strategic path is driven by NASA projects?
- One component supplier value proposition
 - As the number of supported platforms increases then apps become more valuable
 - As the number of apps increases then supporting a cFS platform become more valuable

cFS Introduction



What Components and Distributions have been produced?



- In 2019 vendors starting to offer processor boards integrated with the cFS
 - Al Tech partnered with Embedded Flight Systems to offer the cFS integrated on the SP0-S Single Board Computer
 - Genesis Engineering developing an integrated GEN6000 (SpaceCube 2.0) cFS product
 - Genesis pursuing a Space Act Agreement (SAA) that would include the creation of a platform certification test suite
 - I am not aware of anything since 2019
- A list of operating system ports is maintained at https://github.com/cfs-tools/cfs-platform-list
- Apps have not been individually been released, but new apps have been produced as part of a distributions that are listed on the next slide
- In 2023 cFS Basecamp supports an "app store" model for cFS target integration
 - Developed for cFS-based educational courses and projects
 - Long term goal is to provide an app migration path from cFS Basecamp to flight missions
 - Relies on a NASA cFS Electronic Data Sheet toolchain that is not part of the official cFS Framework
 - Currently does not use any NASA apps



cFS Distributions



Name/Link	Provider	Status	Description
cFS Basecamp	Open STEMware	Use a cFS tech branch that is not current with the latest cFS Framework.	Basecamp provides a lightweight environment to help you learn NASA's core Flight System (cFS) and create app-based solutions for your projects. Basecamp's default cFS target runs on Linux and includes an app suite that provides a complete operational environment including support for onboard file management and transferring files between the ground and flight systems.
cFS Bundle	NASA GSFC	Current with latest cFS Framework	Contains the cFS Framework packaged with additional components to create a system that can easily be built, executed, and unit tested on a Linux platform. The bundle is not fully verified as an operational system, and is provided as a starting point vs an end product.
cFS Framework- 101	NASA MSFC	Functional but outdated. Last updates in 2019.	A training tool for individuals to learn how to develop software with NASA-developed Core Flight software (CFS) framework.
NASA Operational Simulator for Small Satellites (NOS3)	NASA IV&V	Not current with latest cFS Framework	NOS3 provides a complete cFS system designed to support satellite flight software development throughout the project life cycle. It includes •42 Spacecraft dynamics and visualization, NASA GSFC • cFS – core Flight System, NASA GSFC • COSMOS – Ball Aerospace • ITC Common – Loggers and developer tools, NASA IV&V ITC • NOS Engine – Middleware bus simulator, NASA IV&V ITC
OpenSatKit (OSK)	Open STEMware	Runs on Ubuntu 18.04 and earlier. No longer supported	OSK provides a complete cFS system to simplify the cFS learning curve, cFS deployment, and application development. The kit combines three open source tools to achieve these goals: • cFS – core Flight System, NASA GSFC • COSMOS – command and control platform for embedded systems, Ball Aerospace • 42 dynamic simulator, NASA GSFC

cFS Introduction





- Intended to help with many of the issues listed on "Consequences of cFS Open Source Program" slide
- Shortens path to productivity with built-in tutorials and hands on exercises
- > Simplifies the creation of hardware/software STEM educational projects
 - Download and integrate cFS libraries & apps with a few mouse clicks
- Online educational resources
- > [future] Automated transition to OpenC3 ground system

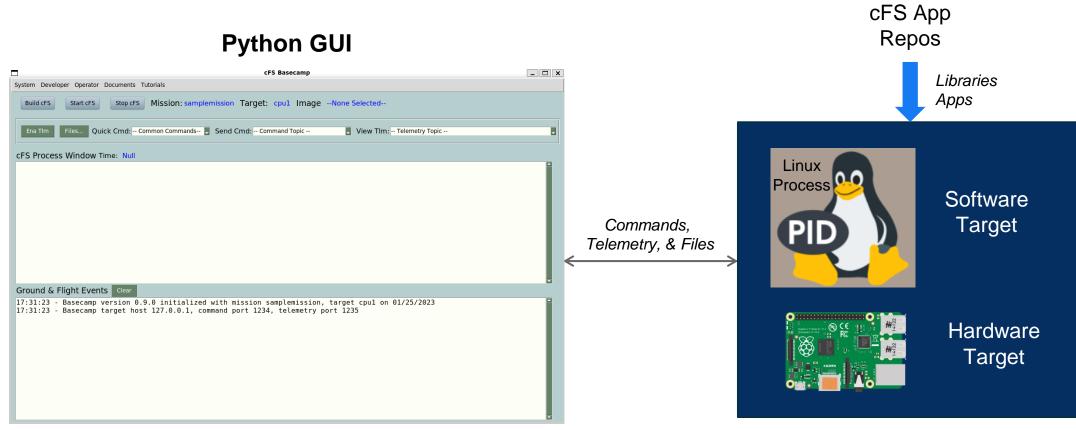


cFS Basecamp (2 of 2)



GitHub

- Send commands, display telemetry, and transfer files between the lightweight Python graphical user interface and a cFS Target
- Supports local and remote cFS targets







Concluding Remarks: Version Control



Release candidates (rc) are checkpoints where the development team has completed everything up to the submittal to software release processing folks for approval as an official release. Occasionally there's a week or to of finalizing artifacts and getting them all loaded after an rc is created, but it's usually fairly quick.

Note not every rc gets submitted to software release folks, but it's a version that has all the artifacts/processes/testing applied such that it could. Historically software release processing has taken up to 18 months, and could be up to 2 years from what I'm told, which is why we separate the two activities if projects do want to use the rc. It's a project choice if this software release processing (basically paperwork and "official release" designation) is value added.

I typically just use the latest RC (and occasionally advance if I need a bug fix or change that got merged).

RC's and dev cycles are more for coordinating across open source components and project needs for a checkpoint, version numbers are for impact.

In theory "breaking" changes are indicated by major version number updates (6.9.1 to 7.0.0), and there's a best-effort attempt to keep changes within a major version number backwards compatible. Note sometimes changes sneak in where using open source cFS components across minor versions or rc's does break (using Draco-rc1 osal with Draco-rc4 cfe might not actually work), but the impacts to external apps should be at most minimal. Occasionally custom config a project may be using can break if they do something unanticipated or uncovered by open source CI, but again best effort is made to keep things backwards compatible until there's a major version number update.

Feature Deprecation Process

- Mark feature as deprecated on any release
- Provide tools/process that will warning applications when a feature is marked as deprecated
- Only deprecate on major versions

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Concluding Remarks: User Responsibilities



User Contributions

Community Contribution process and Contributor License Agreement (CLA)

The cFS Framework has a NASA NPR-7150.2C Class E classification

- Class states Design Concept, Research, Technology and General-Purpose Software
- The cFS Framework provides artifacts to support Class B missions and a subset of artifacts to support Class A missions
- End-users are responsible for classifying the software system that uses the cFS Framework
- This is consistent with the Apache license's "Disclaimer of Warranty" that states, "...provides the Work...on an "AS IS" BASIS, WITHOUT WARRANTIES OR CONDITIONS OF ANY KIND, either express or implied, including, without limitation, any warranties or conditions of ... FITNESS FOR A PARTICULAR PURPOSE."

End-users are responsible for complying with International Traffic in arms Regulations (ITAR)





Appendix A

Acronyms





API Application Programmer Interface

APL Applied Physics Lab

ASIST Advanced Spacecraft Integration and System Testing

ATS Absolute Time Sequence

BC Bus Controller
BT Build Test

bps bits-per seconds
Bps Bytes-per seconds

BSP Board Support Package

C&DH Command and Data Handling

CCSDS Consultative Committee for Space Data Systems

CDS Critical Data Store

CESE Center for Experimental Software Engineering

CFDP CCSDS File Delivery Protocol

cFE Core Flight Executive

cFS Core Flight Software SystemCM Configuration Management

CMD Command

COTS Commercial Off The Shelf

cPCI Compact PCI

CRC Cyclic Redundancy Check

CS Checksum

DMA Direct Memory Access

DS Data Storage

EEPROM Electrically Erasable Programmable Read-Only Memory

EOF End of File

ES Executive Services
EVS Event Services

FDC Failure Detection and Correction

FDIR Failure Detection, Isolation, and Recovery
FM File Management, Fault Management

FSW Flight Software





GNC Guidance Navigation and Control

GSFC Goddard Space Flight Center GOTS Government Off The Shelf

GPM Global Precipitation Measurement

GPS Global Positioning System
Hi-Fi High-Fidelity Simulation

HK Housekeeping HS Health & Safety

HW Hardware

Hz Hertz

I&T Integration and Test

ICD Interface Control Document

IPP Innovative Partnership Program Office IRAD Internal Research and Development

ITAR International Traffic in Arms Regulations

ISR Interrupt Service Routine

ITOS Integration Test and Operations System IV&V Independent Verification and Validation

JHU Johns Hopkins University

KORI Korean Aerospace Research Institute

LADEE Lunar Atmosphere and Dust Environment Explorer

LC Limit Checker

LDS Local Data Storage

LRO Lunar Reconnaissance Orbiter

Mbps Megabits-per seconds

MD Memory Dwell

MET Mission Elapsed Timer

MM Memory Manager
MS Memory Scrub

NACK Negative-acknowledgement

NASA National Aeronautics Space Agency





NESC NASA Engineering and Safety Center

NOOP No Operation
OS Operating System

OSAL Operating System Abstraction Layer
PCI Peripheral Component Interconnect

PSP Platform Support Package RAM Random-Access Memory

RM Recorder Manager ROM Read-Only Memory RT Remote Terminal

R/T Real-time

RTOS Real-Time Operating System
RTS Relative Time Sequence

SARB Software Architecture Review Board

S/C Spacecraft
SB Software Bus

SBC Single-Board Computer

SC Stored Command

SCH Scheduler

S-COMM S-Band Communication Card SDO Solar Dynamic Observatory SDR Spacecraft Data Recorder SIL Simulink Interface Layer

SpW Spacewire SRAM Static RAM

SSR Solid State Recorder

STCF Spacecraft Time Correlation Factor

SUROM Start-Up Read-Only Memory

SW Software, Spacewire

TAI International Atomic Time

TBD To be determined

cFS Introduction





TBL Table Services

TLM Telemetry

TDRS Tracking Data Relay Satellite

TM Time Manager
TO Telemetry Output

TRMM Tropical Rainfall Measuring System

UART Universal Asynchronous Receiver/Transmitter

UDP User Datagram Protocol UMD University of Maryland

UT Unit Test

UTC Coordinated Universal Time VCDU Virtual Channel Data Unit

XB External Bus

XBI Instrument 1553 External Bus XBS Spacecraft 1553 External Bus

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Appendix B

cFS Architectural Goals



Architecture Goals



- 1. Reduce time to deploy high quality flight software
- 2. Reduce project schedule and cost uncertainty
- 3. Directly facilitate formalized software reuse
- 4. Enable collaboration across organizations
- 5. Simplify sustaining engineering (On-Orbit FSW modifications) for long duration missions
- 6. Scale from small instruments to Hubble class missions
- 7. Build a platform for advanced concepts and prototyping
- 8. Create common standards and tools across the center



Quality Analysis - 1



Operability

The architecture must enable the flight system to operate in an efficient and understandable way

Reliability

 The architecture implementation must be known to behave correctly in nominal and expected off-nominal situations

Robustness

The architecture implementation must be predictable and safe in the presence of unexpected conditions

Performance

The architecture implementation must efficient in runtime resources given the targeted processing environments

Testability

The architecture implementation must be easily and comprehensively testable in situ in flight like scenarios

Maintainability

The architecture implementation must be maintainable in the operational environment



Quality Analysis - 2



Effective Reuse

The architecture must support an effective reuse approach. This includes the software and artifacts.
 Requirements, design, code, review presentations, test, operations guides, command and telemetry databases. The goal is to achieve 100% reuse of a software component with no code changes

Composability

- Properties established at the component level, such as interfaces, timeliness or testability, also hold at the system level. For an application or node to be composable the architecture and process must support:
 - Independent development of nodes
 - Integration of the node into a system should not invalidate services in the value and temporal domains
 - Integration of an additional node into a functioning system should not disturb the correct operation of the existing nodes
 - Replica determinism identical copies of nodes must produce identical results in an identical order, within a specified time interval

Predicable Development Schedule

Development estimates provided by the FSW team should be reliable



Quality Analysis - 3



Scalability

The FSW must scale with mission requirements. (Example: instruments or subsystem processor may only need a small amount of message buffer space. This should be configurable to avoid wasting memory resources)

Adaptability

The FSW must be capable of supporting a range of platforms and missions

Minimized Development Cost

Costs for mission functions should be as low as possible. The teams must consider the difference between NRE and costs for a given mission

Technology infusion

The FSW should support the infusion of new hardware and software technologies with minimal side effects





Appendix C

Online Resources



cFS Resources



- cFE Framework, http://github.com/nasa/cFE
 - Contains the OSALs, PSPs and cFE that are managed and released by the NASA CCB
- cFS Bundle (cFE Framework), http://github.com/nasa/cFS
 - Contains the cFE Framework packaged with additional components to create a system that can be built, run and unit tested on a Linux platform
 - Website also contains a list of apps, tools and distributions
 - Other components may exist
- cFS Apps, https://github.com/nasa/**
 - "**" is the app abbreviation such as FM for File Manager
- cFS External Code Interface (ECI) library, https://github.com/nasa/ECI



cFS Basecamp Resources



- cFS Basecamp, https://github.com/cfs-tools/cfs-basecamp
- cFS Basecamp App Repositories, https://github.com/cfs-apps
- cFE EDS Framework Toolchain, https://github.com/jphickey/cfe-eds-framework
- Open Mission Stack FSW Learning Resources, https://openmissionstack.com/
- YouTube Educational Videos, https://www.youtube.com/channel/UC2wfvAlkrrgyC4ITwL3zokg
- OpenSatKit, https://github.com/OpenSatKit/OpenSatKit/wiki
 - Combines cFS, COSMOS Ground System, https://cosmosrb.com/, and NASA 42 Simulator,
 https://github.com/ericstoneking/42
 - Nolonger supported