# 1 Etestigngngnn XAMPLE TEX FILE

Here are some example aerodynamic equations used in the analysis of flight dynamics:

## 1.1 Lift Equation

The lift force generated by an object is given by:

$$L = \frac{1}{2}\rho v^2 C_L A \tag{1}$$

where:

- L is the lift force (N),
- $\rho$  is the air density (kg/m<sup>3</sup>),
- v is the velocity of the object relative to the air (m/s),
- $C_L$  is the lift coefficient (dimensionless),
- A is the reference area ( $m^2$ ).

## 1.2 Drag Equation

The drag force acting on an object is given by:

$$D = \frac{1}{2}\rho v^2 C_D A \tag{2}$$

where:

- D is the drag force (N),
- $C_D$  is the drag coefficient (dimensionless).

#### 1.3 Reynolds Number

The Reynolds number, a dimensionless quantity, is used to predict flow patterns in different fluid flow situations:

$$Re = \frac{\rho vL}{\mu} \tag{3}$$

where:

- Re is the Reynolds number,
- L is the characteristic length (m),
- $\mu$  is the dynamic viscosity of the fluid (Pa·s).

## 1.4 Bernoulli's Equation

Bernoulli's principle for incompressible flow is expressed as:

$$P + \frac{1}{2}\rho v^2 + \rho gh = \text{constant} \tag{4}$$

where:

- $\bullet$  P is the static pressure (Pa),
- g is the acceleration due to gravity (m/s<sup>2</sup>),
- h is the height above a reference point (m).

### 1.5 Equation of Continuity

The equation of continuity for incompressible flow is:

$$A_1 v_1 = A_2 v_2 (5)$$

where:

- $A_1$  and  $A_2$  are the cross-sectional areas (m<sup>2</sup>),
- $v_1$  and  $v_2$  are the flow velocities (m/s).