

# The Laplace Transformation

## Introduction

### Time domain approach

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t), \quad x(0) = x_0 \in \mathbb{R}^n \\ y(t) = Cx(t) \end{cases}$$

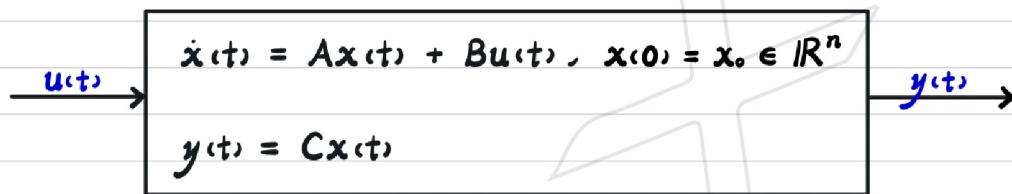
$$\rightarrow \text{let } x = x_h + x_p = e^{At}x_0 + x_p \rightarrow \dot{x} = Ae^{At}x_0 + \dot{x}_p = A(e^{At}x_0 + x_p) + Bu$$

$$\rightarrow \dot{x}_p = Ax_p + Bu \rightarrow \text{let } x_p = \int_0^t e^{A(t-s)}Bu(s)ds$$

$$\frac{d}{dt} \int_0^t f(x, s) ds = \int_0^t \frac{\partial}{\partial t} f(x, s) ds + \frac{db}{dt} f(b, t) - \frac{db}{dt} f(a, t) \rightarrow \dot{x}_p = \int_0^t Ae^{A(t-s)}Bu(s)ds + Bu(t) = Ax_p + Bu$$

$$\rightarrow y = Ce^{At}x_0 + \int_0^t Ce^{A(t-s)}Bu(s)ds$$

### Frequency domain approach



### Laplace transforms

$$F(s) = L\{f(t)\} = \int_0^\infty e^{-st} f(t) dt \rightarrow \text{care only about } t \geq 0 \text{ and } \text{Re}(s) > a \text{ (ROC)}$$

$$\text{linear } L\{af(t) + bg(t)\} = aF(s) + bG(s)$$

$$\text{derivatives } L\{f^{(n)}(t)\} = s^n F(s) - \sum_{k=1}^n s^{n-k} f^{(k-1)}(0) = s^n F(s) \text{ for } f^{(k)}(0) = 0, k \in \mathbb{N}$$

$$\text{derivative } L\{t^n f(t)\} = (-1)^n \frac{d^n F(s)}{ds^n}$$

$$s \text{ shift theorem } L\{e^{at}f(t)\} = F(s-a)$$

$$t \text{ shift theorem } L\{f(t-a)H(t-a)\} = e^{-as} F(s)$$

$$\text{scale theorem } L\{f(at)\} = \frac{1}{a} F\left(\frac{s}{a}\right)$$

$$\text{convolution theorem } L\left\{\int_0^t f(u)g(t-u)du\right\} = F(s)G(s)$$

# Transfer Function

## The transfer function

second-order system  $\ddot{y}(t) + 2\xi\omega_n\dot{y}(t) + \omega_n^2 y(t) = u(t), y(0) = \dot{y}(0) = 0$

$$\xrightarrow{\text{L}} s^2 Y(s) + 2\xi\omega_n s Y(s) + \omega_n^2 Y(s) = U(s), s \in \mathbb{C}$$

$$\rightarrow G(s) = \frac{Y(s)}{U(s)} = \frac{1}{s^2 + 2\xi\omega_n s + \omega_n^2}$$

general control system  $\dot{x}(t) = Ax(t) + Bu(t), y(t) = Cx(t) + Du(t), x(0) = x_0$

$$\xrightarrow{\text{L}} sX(s) = AX(s) + BU(s) \rightarrow X(s) = (sI - A)^{-1}BU(s)$$

$$\rightarrow G(s) = \frac{Y(s)}{U(s)} = C(sI - A)^{-1}B + D$$

$$\rightarrow \begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\omega_n^2 & -2\xi\omega_n \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u(t), y(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix}$$

## The frequency response

$$u = re^{j\theta} \Rightarrow y = L^{-1}\{G\}u = gre^{j\theta} = Ae^{j\phi}re^{j\theta} = Are^{j(\theta+\phi)}$$

$$\xrightarrow{\text{Re}} u = \cos(\omega t) \Rightarrow y = A \cos(\omega t + \phi)$$

frequency response  $\omega \mapsto G(i\omega)$

$$\text{gain } A = |G(i\omega)|$$

$$\text{phase } \phi = \angle G(i\omega)$$

## Bode diagram

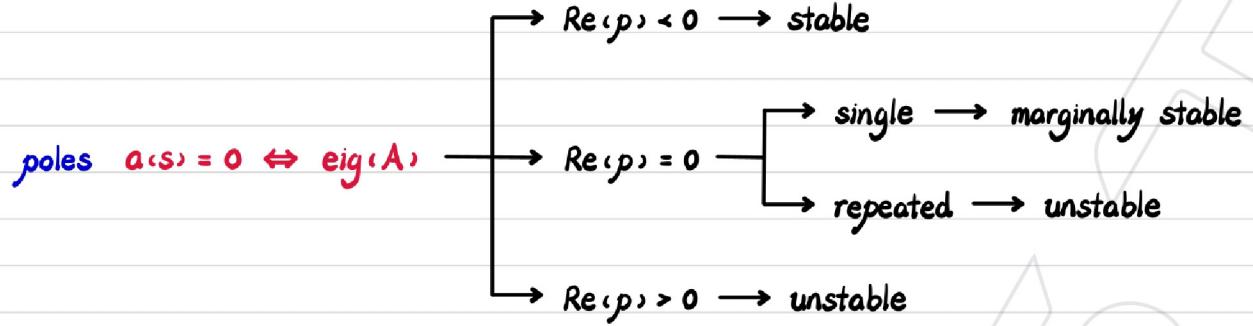


$$L(s) = K(s)G(s) \quad 20 \log_{10}|L(i\omega)| = 20 \log_{10}|K(i\omega)| + 20 \log_{10}|G(i\omega)| \text{ and } \angle L(i\omega) = \angle K(i\omega) + \angle G(i\omega)$$

## Poles, Zeros and Performance

### Poles and zeros

transfer function of a SISO system  $G(s) = \frac{b(s)}{a(s)} = \frac{b_0 + b_1 s + \dots + b_m s^m}{a_0 + a_1 s + \dots + a_n s^n}, m \leq n$



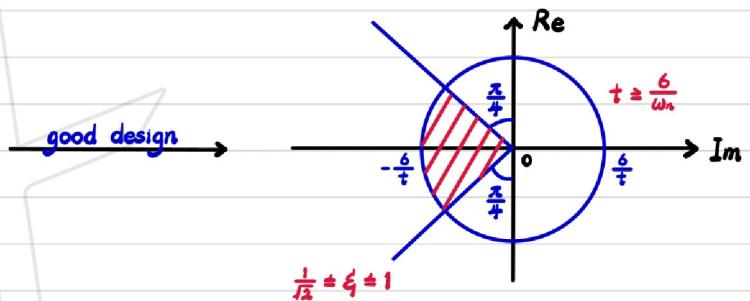
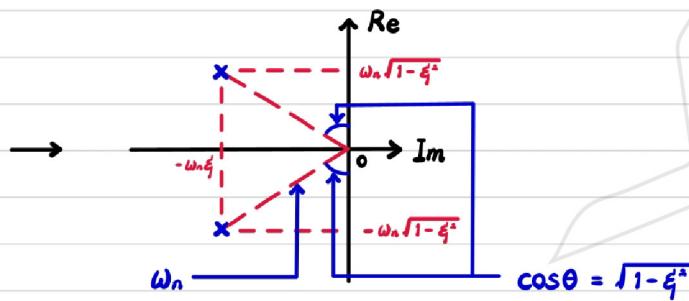
zeros  $b(s) = 0 \rightarrow \text{system unresponsive to certain inputs}$

### Poles of a second-order system

$$G(s) = \frac{Y(s)}{U(s)} = \frac{1}{s^2 + 2\xi\omega_n s + \omega_n^2}$$

$$\rightarrow s^2 + 2\xi\omega_n s + \omega_n^2 = 0$$

lightly damped ( $0 < \xi < 1$ )  $\rightarrow s = \omega_n(-\xi \pm i\sqrt{1-\xi^2})$



### Routh-Hurwitz stability criterion

2nd order system  $a(s) = s^2 + a_1 s + a_0$  is stable if  $a_1, a_0 > 0$

3rd order system  $a(s) = s^3 + a_2 s^2 + a_1 s + a_0$  is stable if  $a_2 a_1 > a_0$  and  $a_3, a_2, a_1, a_0 > 0$

4th order system

$a(s) = s^4 + a_3 s^3 + a_2 s^2 + a_1 s + a_0$  is stable if  $a_3 a_2 a_1 > a_3^2 a_0 + a_1^2 - a_3 a_1 > a_1$  and  $a_3, a_2, a_1, a_0 > 0$

## PID Control Using Transfer Function

### The Final Value Theorem

$\lim_{t \rightarrow \infty} y(t) = \lim_{s \rightarrow 0} s Y(s)$  only correct if system is stable

Step response

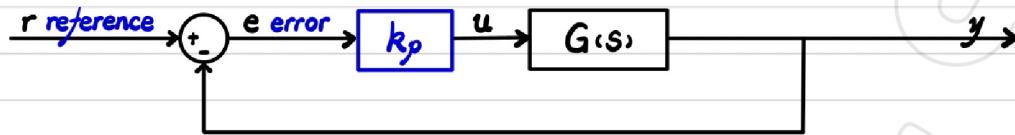
for a step input  $u(t) = \begin{cases} 1, & t \geq 0 \\ 0, & t < 0 \end{cases}$

$$\rightarrow U(s) = \frac{1}{s}, \operatorname{Re}(s) > 0$$

$$\rightarrow \lim_{t \rightarrow \infty} y(t) = \lim_{s \rightarrow 0} sY(s) = \lim_{s \rightarrow 0} sG(s)U(s) = \lim_{s \rightarrow 0} G(s) = \lim_{s \rightarrow 0} C(sI - A)^{-1}B + D$$

$$\rightarrow \lim_{t \rightarrow \infty} y(t) = -CA^{-1}B + D$$

## Proportional control



$$T_{\text{try}}(s) = \frac{k_p G(s)}{1 + k_p G(s)}$$

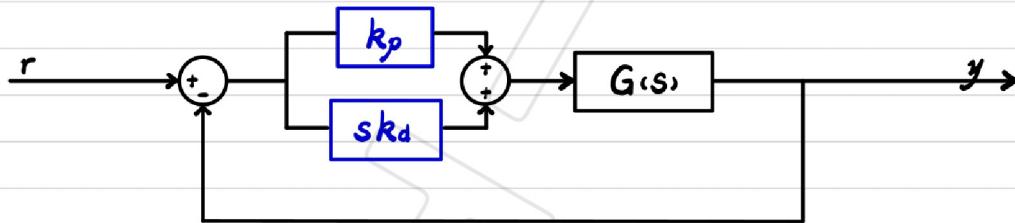
$$G(s) = \frac{1}{(1+s)^2} \quad T_{\text{try}}(s) = \frac{k_p}{(1+s)^2 + k_p} = \frac{k_p}{s^2 + 2s + (1+k_p)} \rightarrow \omega_n = \sqrt{k_p + 1} \text{ and } \xi = \frac{1}{\sqrt{k_p + 1}}$$

$$\text{track step} \quad \lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} [r(t) - y(t)] = \lim_{s \rightarrow 0} \left[ 1 - \frac{k_p}{s^2 + 2s + (1+k_p)} \right] = \frac{1}{1+k_p} \neq 0$$

→ can not eliminate steady-state error

→ increase  $k_p$  reduce steady-state error  $\Leftrightarrow$  increase oscillation

## Proportional-Derivative (PD) control



$$T_{\text{try}}(s) = \frac{(k_p + sk_d)G(s)}{1 + (k_p + sk_d)G(s)}$$

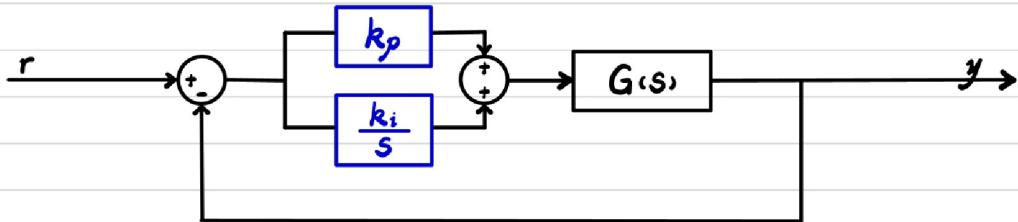
$$G(s) = \frac{1}{(1+s)^2} \quad T_{\text{try}}(s) = \frac{k_p + sk_d}{(1+s)^2 + (k_p + sk_d)} = \frac{k_p + sk_d}{s^2 + (2+k_d)s + (1+k_p)} \rightarrow \omega_n = \sqrt{k_p + 1} \text{ and } \xi = \frac{2+k_d}{2\sqrt{k_p + 1}}$$

$$\text{track step} \quad \lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} [r(t) - y(t)] = \lim_{s \rightarrow 0} \left[ 1 - \frac{k_p + sk_d}{s^2 + (2+k_d)s + (1+k_p)} \right] = \frac{1}{1+k_p} \neq 0$$

→ can not eliminate steady-state error

→ increase  $k_d$  reduce oscillation  $\Leftrightarrow$  more sensitive to noise

## Proportional-Integral (PI) control



$$T_{ry}(s) = \frac{(k_p + k_i/s)G(s)}{1 + (k_p + k_i/s)G(s)} = \frac{(sk_p + k_i)sG(s)}{s + (sk_p + k_i)G(s)}$$

track step  $\lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} [r(t) - y(t)] = \lim_{s \rightarrow 0} [1 - \frac{(sk_p + k_i)sG(s)}{s + (sk_p + k_i)G(s)}] = 1 - 1 = 0$

track ramp  $\lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} [r(t) - y(t)] = \lim_{s \rightarrow 0} [\frac{1}{s} - \frac{(sk_p + k_i)sG(s)}{s + (sk_p + k_i)G(s)} \frac{1}{s}] = \frac{1}{k_i G(0)} \neq 0$

→ might eliminate steady-state error depends on reference signal

### Proportional - Integral - Derivative (PID) control

$$K(s) = k_p + \frac{k_i}{s} + sk_d$$

→ both damping and steady-state error can be controlled

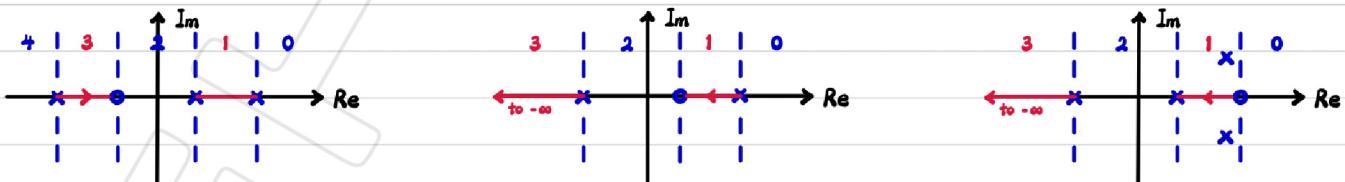
## Root Locus Design

### Root Locus Design

#### Plotting the root locus diagram

- ① for open-loop transfer function  $G(s)$
- P poles(s) :  $p_1, p_2, \dots, p_p \rightarrow$  plot as  $x$
  - Z zero(s) :  $z_1, z_2, \dots, z_z \rightarrow$  plot as  $o$

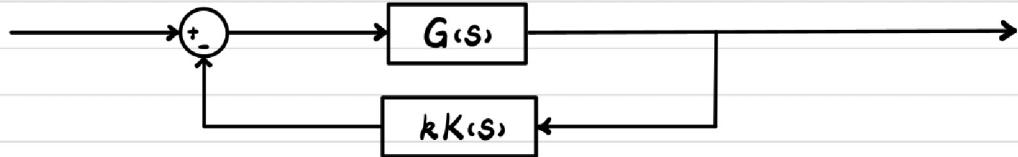
- ② real axis pole(s) converge to real axis zero(s) or conjugate poles or  $-\infty$  at 'odd regions'



- ③ find asymptotes with asymptote centre  $\frac{1}{P-Z}(\sum_{i=1}^P p_i - \sum_{j=1}^Z z_j)$  at evenly-spaced angles  $\frac{(2n+1)\pi}{P-Z}$

- ④ find break point(s) at  $G'(s) = 0$

### Feedback controller



$$G_{cl}(s) = \frac{G(s)}{1 - [-G(s)kK(s)]} = \frac{G(s)}{1 + kG(s)K(s)} = \frac{b_a/a_g}{1 + k(b_a/a_g)(b_k/a_k)} = \frac{a_gb_k}{a_ga_k + kb_ab_k}$$

→ closed-loop pole(s) at  $a_{GK}(s) + kb_{GK}(s) = 0 \xrightarrow{k=0}$  open-loop pole(s) at  $a_{GK}(s) = 0$

→ plotting the root locus for  $G(s)K(s)$  to show the movement of closed-loop poles as  $k > 0$  increases

## The Nyquist Criterion

### The Nyquist Diagram

#### Plotting the Nyquist diagram

$\omega \mapsto G(i\omega)$ ,  $-\infty < \omega < \infty$  run from  $-\infty$  up to  $\infty$  on the complex plane

①  $\overline{G(i\omega)} = \frac{\overline{b(i\omega)}}{\overline{a(i\omega)}} = \frac{\overline{b(-i\omega)}}{\overline{a(-i\omega)}} = G(-i\omega) \rightarrow$  symmetric about x-axis

② calculate asymptotic behaviour as  $\omega \rightarrow 0$  and  $\omega \rightarrow \infty$

→ find  $\angle G(\pm i\infty) = \mp \frac{2n}{\pi}$  at asymptotic values for nth order system

③ multiply transfer function by a factor  $k$  → scaling the entire Nyquist plot by a factor  $k$

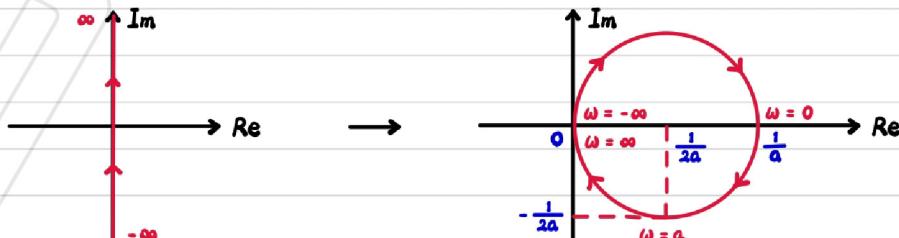
$$G(s) = -(2+s)$$



$$G(s) = \frac{1}{s+a} \quad \lim_{\omega \rightarrow \infty} G(i\omega) = 0 \rightarrow \lim_{\omega \rightarrow \infty} \angle G(i\omega) = -\frac{\pi}{2}$$

$$G(0) = \frac{1}{a} \rightarrow \angle G(0) = 0$$

$$G(ia) = \frac{1}{2a} - i\frac{1}{2a} \rightarrow \angle G(ia) = -\frac{\pi}{4}$$



$$G(s) = \frac{1}{s}$$

avoid singularity at  $\omega = 0 \rightarrow s = \begin{cases} i\omega, & -\infty < \omega < -\varepsilon \\ e^{i\theta}, & -\frac{\pi}{2} < \theta < \frac{\pi}{2} \\ i\omega, & \varepsilon < \omega < \infty \end{cases}$

$$\lim_{\omega \rightarrow \infty} G(i\omega) = 0 \rightarrow \lim_{\omega \rightarrow \infty} \angle G(i\omega) = \lim_{\omega \rightarrow \infty} \angle(1) - \angle(s) = 0 - \frac{\pi}{2} = -\frac{\pi}{2}$$

$$G(0) = \frac{1}{\varepsilon} \rightarrow \angle G(0) = 0$$

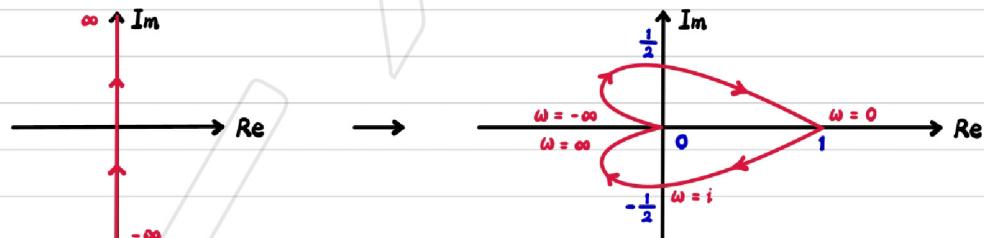
$$G(i\varepsilon) = -i\frac{1}{\varepsilon} \rightarrow \angle G(i\varepsilon) = -\frac{\pi}{4}$$



$$G(s) = \frac{1}{s^2 + 2s + 1} \quad \lim_{\omega \rightarrow \infty} G(i\omega) = 0 \rightarrow \lim_{\omega \rightarrow \infty} \angle G(i\omega) = \lim_{\omega \rightarrow \infty} \angle(1) - 2\angle(s+1) = -\pi$$

$$G(0) = 1 \rightarrow \angle G(0) = 0$$

$$G(i) = -\frac{1}{2}i \rightarrow \angle G(i) = -\frac{\pi}{2}$$



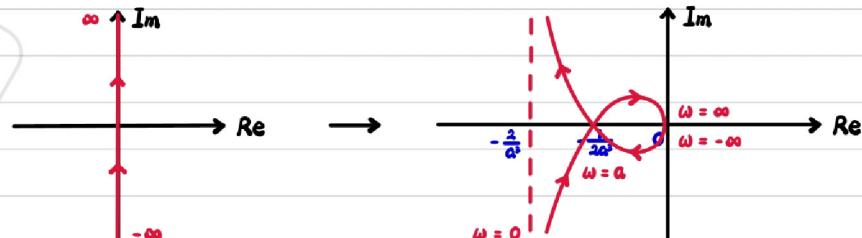
$$G(s) = \frac{1}{s(s+a)^2}$$

$$G(i\omega) = \frac{1}{i\omega(a^2 - \omega^2 + 2i\omega)} = \frac{1}{-2\omega^2 + i(a^2\omega - \omega^3)} = -\frac{2a}{(a^2 + \omega^2)^2} - i\frac{a^2 - \omega^2}{\omega(a^2 + \omega^2)^2}$$

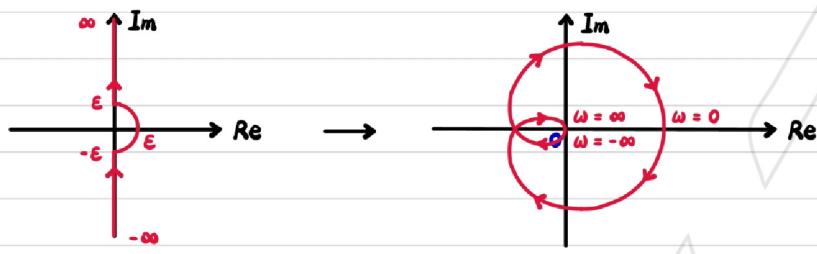
$$\lim_{\omega \rightarrow \infty} G(i\omega) = 0 \rightarrow \lim_{\omega \rightarrow \infty} \angle G(i\omega) = \lim_{\omega \rightarrow \infty} \angle(1) - [\angle(s) + 2\angle(s+a)] = -\frac{3}{2}\pi$$

$$G(0) = -\frac{2}{a^3} - i\infty \rightarrow \angle G(0) = -\frac{3}{2}\pi$$

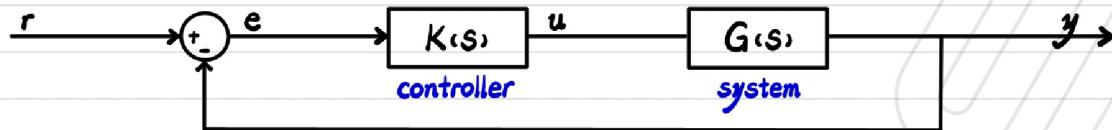
$$G(ia) = -\frac{1}{2a^3} \rightarrow \angle G(ia) = \pi$$



$\Leftrightarrow$  consider singularity at  $\omega = 0$



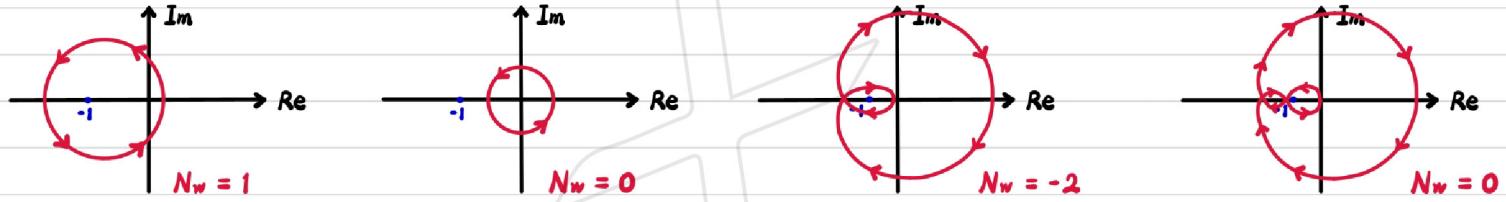
## Feedback stability



$$\text{loop transfer function } L(s) = G(s)K(s) \rightarrow T_{Ny}(s) = \frac{L(s)}{1 + L(s)} = \frac{a_L(s)}{a_L(s) + b_L(s)}$$

## The Nyquist stability criterion

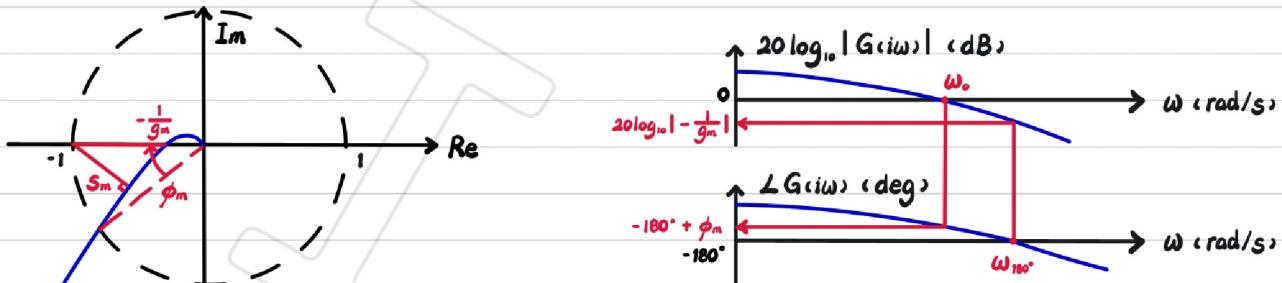
$N_w$  number of times  $-1$  is encircled in the **anticlockwise** direction by the Nyquist contour



→ for stable open-loop system, closed-loop system is stable if  $N_w = 0$

→ for unstable open-loop system, closed-loop system is stable if  $N_w = \text{number of unstable poles}$

## Stability margins



gain margin  $g_m$

Bode stable region if  $\angle L(i\omega)$  cross  $180^\circ$  from above only once at  $\omega_{180}$ . system is stable if  $|L(i\omega_{180})| < 1$

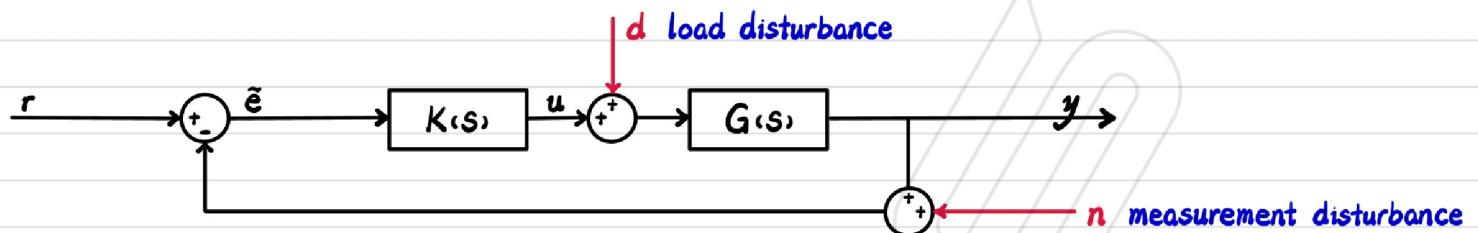
phase margin  $\phi_m$

stability margin  $s_m$

$G(s)$  is an approximation of true system and  $K(s)$  cannot be implemented directly  $\rightarrow L(i\omega)$  in reality might be  $gL(i\omega)e^{j\phi}$

# Feedback System Performance and Design

## Feedback system



$$\bar{e}(s) = \bar{r}(s) - Y(s)$$

$$= \bar{r}(s) - G(s)[U(s) + \bar{d}(s)]$$

$$= \bar{r}(s) - G(s)K(s)\bar{e}(s) + G(s)\bar{d}(s)$$

$$= \bar{r}(s) - G(s)K(s)[\bar{e}(s) - \bar{n}(s)] + G(s)\bar{d}(s)$$

$$\rightarrow \bar{e}(s) = \frac{1}{1+L(s)}\bar{r}(s) + \frac{G(s)}{1+L(s)}\bar{d}(s) - \frac{L(s)}{1+L(s)}\bar{n}(s)$$

$$\text{sensitivity function } S(s) = \frac{1}{1+L(s)}$$

$$\text{load sensitivity } S(s)G(s) = \frac{G(s)}{1+L(s)}$$

$$\text{complementary sensitivity } T(s) = \frac{L(s)}{1+L(s)}$$

$$\rightarrow S(s) + T(s) = 1 \rightarrow \text{trade-off}$$

- make  $S(i\omega)$  small at low  $\omega \rightarrow |L(i\omega)| \gg 1$
- make  $T(i\omega)$  small at high  $\omega \rightarrow |L(i\omega)| \ll 1$

## Sensitivity function

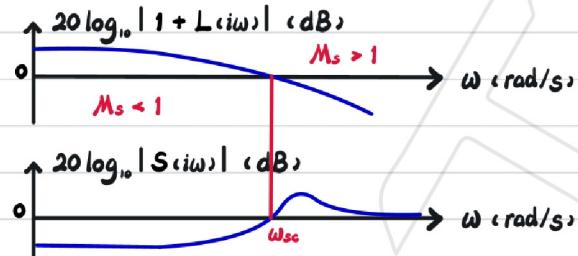
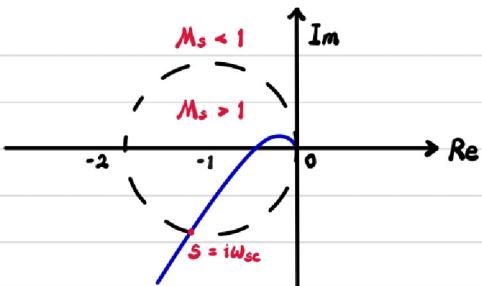
$$Y_{ol}(s) = L(s)\bar{r}(s) + G(s)\bar{d}(s)$$

$$Y_{cl}(s) = \frac{L(s)\bar{r}(s) + G(s)\bar{d}(s)}{1+L(s)}$$

$$\rightarrow \frac{Y_{cl}(s)}{Y_{ol}(s)} = S(s) = \frac{1}{1+L(s)}$$

$$\text{maximum sensitivity } M_s = \max_{\omega} |S(i\omega)| = \frac{1}{|1+L(i\omega)|} = \frac{1}{s_m}$$

$$\text{sensitivity crossover frequency } \omega_{sc} \rightarrow |S(i\omega_{sc})| = 1$$



## Bode's integral formula

supposed  $L(s)$  has no poles on the RHP and tends to 0 faster than  $\frac{1}{s}$  as  $s \rightarrow \infty$

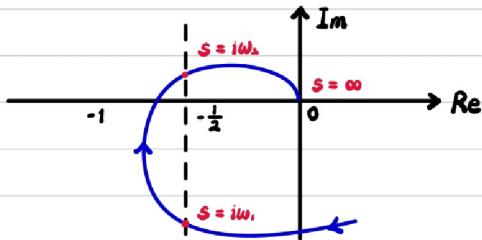
$$\rightarrow \int_0^\infty \ln |S(iw)| dw = 0$$

$\rightarrow$  Waterbed effect if sensitivity is pushed down for some  $w$ , it must increase at some other  $w$

## Complementary sensitivity

$$T(s) = \frac{L(s)}{1+L(s)}$$

at  $|T(iw)| = \left| \frac{L(iw)}{1+L(iw)} \right| = 1 \rightarrow |L(iw)| = |1+L(iw)| \rightarrow \operatorname{Re}\{L(iw)\} = -\frac{1}{2}$

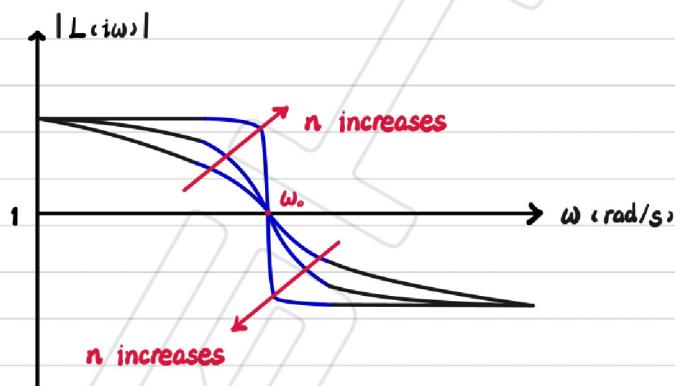


$$\begin{cases} |T(iw)| < 1 & -iw_1 < s < iw_1 \\ |T(iw)| > 1 & iw_1 < s < iw_2 \\ |T(iw)| < 1 & iw_2 < s < \infty \end{cases}$$

## Robustness

$$2 \leq g_m \leq 5, 30^\circ \leq \phi_m \leq 60^\circ \text{ and } 0.5 \leq S_m \leq 0.8$$

Bode's Gain-Phase relationship  $\angle L(iw) \approx 90^\circ \frac{d \ln |L(iw)|}{d \ln w}$



at cross-over region,  $|L(iw_c)| = \frac{1}{\omega^n}$

$$\angle L(iw_c) \approx 90^\circ \frac{d \ln \omega^n}{d \ln w} = 90^\circ \frac{(-n) d \ln \omega}{d \ln w} = -90^\circ n$$

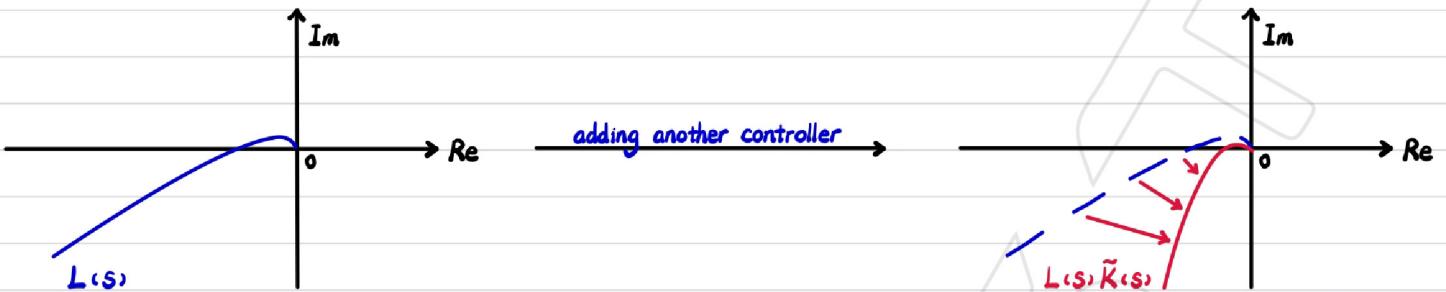
$$\rightarrow \angle L(iw_c) \approx -90^\circ n$$

$$\rightarrow -180^\circ + \phi_m \approx -90^\circ n \rightarrow \phi_m \approx 180^\circ - 90^\circ n$$

$$\rightarrow n \leq 2$$

$\rightarrow$  transition must happen gradually to ensure robustness

## Loop shaping



increasing phase margin if  $\omega_{new,0} = \omega_0$ ,  $\phi_{new,m} = \phi_m + \angle \tilde{K}(j\omega_0)$

attenuating load disturbances at low frequencies

choose PI controller  $\rightarrow \tilde{K}(s) = 1 + \frac{b}{s}$

$$\begin{cases} \lim_{s \rightarrow 0} |\tilde{L}(s)| = \lim_{s \rightarrow 0} |L(s)| \lim_{s \rightarrow 0} |K(s)| = \infty \rightarrow \lim_{s \rightarrow 0} |\tilde{S}(s)| = 0 \\ \lim_{s \rightarrow \pm\infty} |\tilde{L}(s)| = \lim_{s \rightarrow \pm\infty} |L(s)| \lim_{s \rightarrow \pm\infty} |K(s)| = 0 \rightarrow \lim_{s \rightarrow \pm\infty} |\tilde{S}(s)| = \infty \end{cases}$$

attenuating measurement disturbances at high frequencies

choose PD controller  $\rightarrow \tilde{K}(s) = 1 + as$

$$\begin{cases} \lim_{s \rightarrow \pm\infty} |\tilde{L}(s)| = \lim_{s \rightarrow \pm\infty} |L(s)| \lim_{s \rightarrow \pm\infty} |K(s)| = \infty \rightarrow \lim_{s \rightarrow \pm\infty} |\tilde{S}(s)| = 0 \\ \lim_{s \rightarrow 0} |\tilde{L}(s)| = \lim_{s \rightarrow 0} |L(s)| \lim_{s \rightarrow 0} |K(s)| = 0 \rightarrow \lim_{s \rightarrow 0} |\tilde{S}(s)| = \infty \end{cases}$$

$\rightarrow$  avoid infinite gain by use phase compensators  $\frac{s+a}{s+b}$

