HW7 Problem 1

```
fprintf('\n');
clearvars -except function_list hw_pub toolsPath
close all
CelestialConstants; % import useful constants
X0 = [5492.000; %km]
 3984.001 ;%km
 2.955 ;%km
 -3.931 ;%km/sec
 5.498 ;%km/sec
 3.665 ];%km/sec
% Anon fcn to calculate specific energy. It shouldn't change!
spec\_energy = @(X) norm(X(4:6))^2/2 - Earth.mu/norm(X(1:3));
% Classical orbit elements
[a,e,i,RAAN,w,f] = cart2OE(XO(1:3),XO(4:6),Earth.mu);
% Get the stuff that's propagated
n = sqrt(Earth.mu/a/a/a);
M0 = E2M(f2E(f,e),e);
for t = [100 1e6]; %s
    % Final mean anom is easy...
    Mf = M0 + n*t;
    % Unwinde the mean anom
    while Mf > 2*pi
        Mf = Mf - 2*pi;
    end
    % Final true anom
    ff = E2f(M2E(Mf,e),e);
    % Back to ECI!
    [r f, v f] = OE2cart(a,e,i,RAAN,w,ff,Earth.mu);
    fprintf('r_f(t=%d):\n',t)
    disp(r f);
    fprintf('delta Energy(t=%d):\n',t)
    disp(spec_energy([r_f;v_f]) - spec_energy(X0));
end
% Anonymous function to calculate 2-body accel
two_body = @(t,X) [X(4);X(5);X(6);...
    -Earth.mu*X(1)/norm(X(1:3))^3;...
    -Earth.mu*X(2)/norm(X(1:3))^3;...
    -Earth.mu*X(3)/norm(X(1:3))^3];
% Anon fcn to calculateposition difference.
calc_dr = @(X_exp, r_f) \ sqrt((X_exp(1)-r_f(1))^2 ...
        +(X \exp(2)-r f(2))^2...
        +(X_exp(3)-r_f(3))^2;
```

```
options=odeset('RelTol',tol,'AbsTol',[tol tol tol tol tol tol]);
for t = [100 1e6]
    [t array, X array] = ode45(two body, [0 t], X0, options);
    fprintf('r_f(t=%d)(integrated):\n',t_array(end))
   disp(X array(end,1:3)');
    fprintf('delta Energy(t=%d)(integrated):\n',t)
   disp(spec energy(X array(end,1:6)) - spec energy(X0));
end
for tol = [1e-12 1e-10 1e-8 1e-6 1e-4]
    options=odeset('RelTol',tol,'AbsTol',[tol tol tol tol tol tol));
    [t array, X array] = ode45(two body, [0 1e6], X0, options);
    fprintf('Position Diff @ tol = %e:\n',tol)
   disp(calc dr(X array(end,1:3)',r f));
    fprintf('delta Energy @ tol = %f:\n',tol)
   disp(spec_energy(X_array(end,1:6)) - spec_energy(X0));
end
        r_f(t=100):
           1.0e+03 *
           5.064753117135560
           4.507257422053243
           0.368658029818315
        delta Energy(t=100):
             3.552713678800501e-15
        r_f(t=1000000):
           1.0e+03 *
           1.407037337984632
           6.270084686694161
           2.306266072702379
        delta Energy(t=1000000):
             1.421085471520200e-14
        r_f(t=100)(integrated):
           1.0e+03 *
           5.064753168465420
           4.507257366803862
           0.368657989226936
        delta Energy(t=100)(integrated):
             4.973799150320701e-14
        r_f(t=1000000)(integrated):
           1.0e+03 *
           1.407037859051284
```

tol=1e-12;

- 6.270084634401733 2.306265884003925
- delta Energy(t=1000000)(integrated):
 2.198543569420508e-09
- Position Diff @ tol = 1.000000e-12: 5.566435667007229e-04
- delta Energy @ tol = 0.000000: 2.198543569420508e-09
- Position Diff @ tol = 1.000000e-10: 0.043311061412548
- Position Diff @ tol = 1.000000e-08: 3.948961410452488
- Position Diff @ tol = 1.000000e-06: 31.540622301006042
- delta Energy @ tol = 0.000001: -1.331167062197380e-04
- Position Diff @ tol = 1.000000e-04: 1.204741406084945e+04
- delta Energy @ tol = 0.000100: -2.587559250330802

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