
HW1 Problem 1: Cartesian Coordinates to Keplerian Orbital Elements

```
fprintf('\n');
clearvars -except function_list pub_opt
close all

r = [-2436.45; -2436.45; 6891.037]; % km
v = [5.088611; -5.088611; 0.0]; % km/s
state = [r;v];
oe = cart2oe(state);
fprintf('a = %f km\n', oe(1))
fprintf('e = %f\n', oe(2))
fprintf('i = %f degrees\n', oe(3)*180/pi)
fprintf('RAAN = %f degrees\n', oe(4)*180/pi)
fprintf('Arg of Periapse = %f degrees\n', oe(5)*180/pi)
fprintf('True Anomaly = %f degrees\n', oe(6)*180/pi)

a = 7712.194677 km
e = 0.001001
i = 63.434003 degrees
RAAN = 135.000000 degrees
Arg of Periapse = 90.000000 degrees
True Anomaly = 0.000000 degrees
```

Published with MATLAB® R2013b