

GPS and curved space

1) The GPS system is made of

- 6x4 satellites at $R_s = 26600 \text{ km}$, i.e. 20200 km from earth surface ($R_\oplus = 6400 \text{ km}$, $g = 9.8 \text{ m/s}^2$)

- 4 satellites on each of the 6 planes

1 plane splits the angles into 2 pieces

2 planes

4

3 planes

6

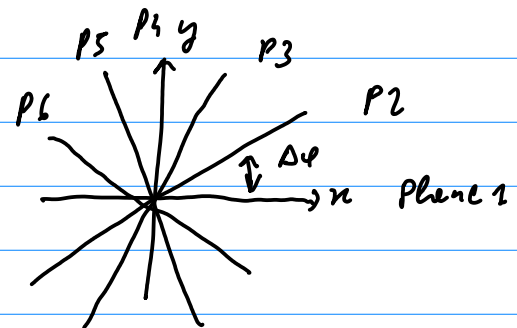
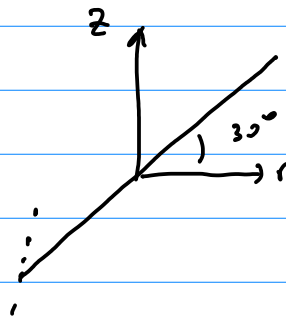
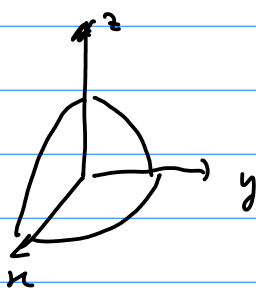
6

12

\Rightarrow

$$\Delta\varphi = \frac{360^\circ}{12} = 30^\circ$$

- each plane is at 30° wrt the previous (this does not mean that all planes are obtained from one plane by a rotation along the same axis)



- almost always at least 4 satellites are visible

2) Why 4?

We need to solve the eq.

$$(*) \quad |\vec{r}_{\text{GPS}} - \vec{r}_s| = c (t_{\text{GPS}} - t_s) \quad s=1, \dots, 4$$

where (ct_s, \vec{r}_s) are the coords of satellites
and $(ct_{\text{GPS}}, \vec{r}_{\text{GPS}})$ are the coords of receiver
 \Rightarrow 4 unknowns!

Precision

$$\text{If we want } |\Delta r_{\text{GPS}}| \sim 1 \text{ m} \Rightarrow |c \Delta t_s| \sim 1 \text{ m} \\ \Rightarrow |\Delta t_s| \sim 3 \cdot 10^{-9} \text{ s}$$

so time on satellites must be precise with such an error
and they must be synchronized with such an error, i.e.
 $|t_r - t_s| \leq 3 \cdot 10^{-9} \text{ s}$

This must happen also after $1 \text{ d} = 86400 \text{ s}$ hence the error
must be $\left| \frac{\Delta t}{dt} \right| \sim 3.47 \cdot 10^{-14}$

3) Fact

The synchronized GPS clocks have frequency

$\nu_{\text{earth}} = 10.23 \text{ MHz}$ on earth measured with earth proper time

$\nu_{\text{satellite}} = 10.22999999543 \text{ MHz}$ on satellites in their proper time

$$\text{so } \left| \frac{\Delta \nu}{\nu_{\oplus}} \right| = \frac{\nu_{\oplus} - \nu_s}{\nu_{\oplus}} = 4.47 \cdot 10^{-10}$$

$$\text{or } \left| \frac{\Delta P}{P_{\oplus}} \right| = \frac{P_s - P_{\oplus}}{P_{\oplus}} = \frac{\nu_{\oplus} - \nu_s}{\nu_s} \sim \frac{\nu_{\oplus} - \nu_s}{\nu_{\oplus}} \sim 4.47 \cdot 10^{-10}$$

N.B. \oplus Earth ; \odot sun ; ♀ Venus ; ♂ Mars

In other words.

If we use the maximum of \vec{E} in an e.m. wave to measure time intervals, an e.m. wave emitted with frequency $\nu_{\text{satellite}}$ when measured on satellite with its proper time is measured with frequency ν_{\oplus} on earth in earth proper time.

4) Why?

Can we understand this fact using special relativity?

1st try

Satellites move "fast"

$$v_s = 3.87 \text{ km s}^{-1} \quad \beta = 1.23 \cdot 10^{-5}$$

$$\gamma - 1 = \frac{1}{\sqrt{1-\beta^2}} - 1 \sim \frac{1}{2} \beta^2 \sim 0.83 \cdot 10^{-10}$$

Easy to derive

$$g = \frac{GM_{\oplus}}{R_{\oplus}^2}$$

$$\frac{v^2}{R_s} = \frac{GM_{\oplus}}{R_s^2}$$

$$\Rightarrow \frac{v^2}{R_s g} = \frac{R_{\oplus}^2}{R_s^2} \Rightarrow v^2 = g \frac{R_{\oplus}^2}{R_s}$$

$$v^2 \sim 10 \frac{\text{m}}{\text{s}^2} \frac{(6 \cdot 10^6 \text{ m})^2}{27 \cdot 10^6 \text{ m}} = \frac{1}{2.7} \cdot 36 \cdot 10^6 \frac{\text{m}^2}{\text{s}^2} \sim 12 \cdot 10^6 \frac{\text{m}^2}{\text{s}^2}$$

It follows that

$P_s = \gamma P_{\oplus}$ for a signal emitted from earth since $\Delta x_g = 0$

$$\text{hence} \quad \frac{\Delta P}{P} = \frac{P_s - P_{\oplus}}{P_{\oplus}} = \gamma - 1 = +0.83 \cdot 10^{-10}$$

or

$P_{\oplus} = \gamma P_s$ for a signal emitted from satellite since $\Delta x_s = 0$
and measured on earth

$$\text{hence} \quad \frac{\Delta P}{P} = \frac{P_s - P_{\oplus}}{P_{\oplus}} = \frac{1-\gamma}{\gamma} \sim +1-\gamma \sim -0.83 \cdot 10^{-10}$$

Which is correct?

- We must ask what we want measure in which system.
we measure 1) a time interval
2) in earth system

⇒ The 2nd is right

- Here we are actually comparing frequencies as measured in satellite proper time with those measured by an observer which is static wrt the \oplus observer and synchronized with the \oplus obs

• 2nd try

Suppose light feels gravitation, or better energy feels gravitation and not only mass.

Hence write

$$V = - \frac{G M_{\oplus} m}{r} \Rightarrow - \frac{G M_{\oplus} E}{c^2 r}$$

Then given photon of frequency $\nu \Rightarrow E = h\nu$
and we can compute what happens when it falls from the satellite.

$$E_S + V_S = \left(1 - \frac{G M_{\oplus}}{c^2 R_S}\right) E_S = E_{\oplus} + V_{\oplus} = \left(1 - \frac{G M_{\oplus}}{c^2 R_{\oplus}}\right) E_{\oplus}$$

$$\Rightarrow E_{\oplus} \sim E_S \left(1 + \frac{G M_{\oplus}}{c^2} \left(\frac{1}{R_{\oplus}} - \frac{1}{R_S}\right)\right) > E_S$$

i.e. photons acquire energy when falling

Equivalently

$$\begin{aligned} \left| \frac{\Delta \nu}{\nu_{\oplus}} \right| &= \frac{E_{\oplus} - E_S}{E_S} = \frac{G M_{\oplus}}{c^2} \left(\frac{1}{R_{\oplus}} - \frac{1}{R_S} \right) = \\ &= \frac{g R_{\oplus}^2}{c^2} \left(\frac{1}{R_{\oplus}} - \frac{1}{R_S} \right) \sim 5.3 \cdot 10^{-10} \end{aligned}$$

Combining 1st and 2nd we get

$$\frac{\Delta \nu}{\nu_{\oplus}} \sim 5.3 \cdot 10^{-10} - 0.83 \cdot 10^{-10} \sim 4.47 \cdot 10^{-10}$$

which is the right result

How do we interpret this?

We compare a static satellite in orbit with a static obs on earth so we miss time dilation for the satellite and Doppler for the \oplus obs
 \Rightarrow we added time dilation to gravity effects

Earth observer sees in her system and with her proper time
$$\overset{\text{proper time}}{\nu_{\oplus}} = 23 \text{ MHz} = \overset{\text{blue shifted}}{\nu_s} > \overset{\text{proper time}}{\nu_s}$$

Satellite observer sees

$$\overset{\text{red shifted}}{\nu_{\oplus}} = \overset{\text{proper time}}{\nu_s} < \overset{\text{proper time}}{\nu_{\oplus}}$$

5) Principle of equivalence

All matter behaves in the same way in gravitational fields (up to spin dependent coupling)

The previous is an application of it.

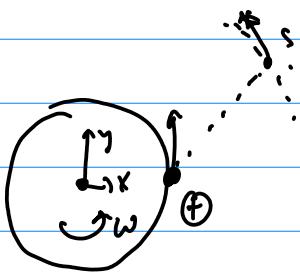
NOTICE

1) Eq (*) is strictly speaking valid in ECI system (Earth Centered Inertial system) since on earth light geodesic is not a straight line because of earth rotation

2) We have to consider that receiver is moving so
$$-ds^2 = c^2 dt_{GPS}^2 = c^2 \gamma_{rec}^2 dt_{receiver}^2$$

3) Because receiver is moving signals from satellites are subject to Doppler effect (both transverse which we have already considered and radial which we consider now which is bigger!)

$$\frac{v_{rec}}{v_{GPS}} = \frac{\sqrt{1-\beta^2}}{1+\beta_r} \sim 1 - \frac{v_r}{c} + \frac{1}{2} \left(\frac{v_r}{c} \right)^2 - \frac{1}{2} \left(\frac{v}{c} \right)^2 + O(c^{-3})$$



Now $v_r \sim O(100 \text{ m/s})$

as an obs on equator is moving

$$\text{at } v_{eq} = \frac{4 \cdot 10^4 \text{ km}}{1d} = 462 \frac{\text{m}}{\text{s}}$$

4) To keep satellites synchronized they must be updated frequently.

Clocks on earth must be synchronized but earth is rotating and with different speeds at different latitudes

\Rightarrow Sagnac effect

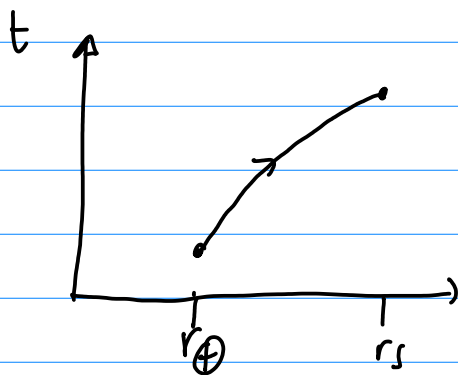
not possible to synchronize all clocks

Schild's argument

Can we explain blue/red shift in flat space?

Suppose space is Minkowski.

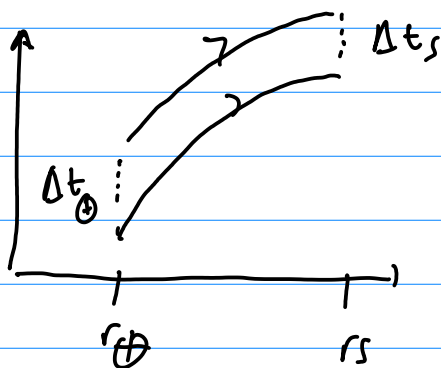
Forget entire rotation.



A γ climbing out the gravitational well.

The path followed may be at non constant speed

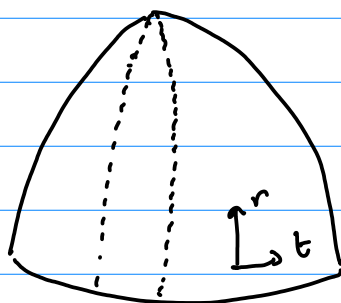
Consider a 2nd photon. It must follow an equal path since the system is static.



$$\text{Hence } \Delta t_{\oplus} = \Delta t_s$$

Hence the space cannot be flat

Ex.



Sphere