

Plasma Spray Coating Research for Pratt and Whitney by Max Plomer

Summary: The following is a preliminary research of the Plasma Spray Coating process; we will review basic heat transfer material and then build a simple analytical/numerical estimate of the location of a particle when it has reached melting temperature and then when it has fully melted. There are 3 main problems that are combined into the solution:

- (1) Lumped capacitance model is used for low Biot number spheres being heated to melting temperature
- (2) Once particle reaches melting temperature this becomes a Stefan Problem where we calculate the particle's solid fraction using its latent heat of melting
- (3) Linear and quadratic drag coefficients are used to calculate the location of the particle when melted.

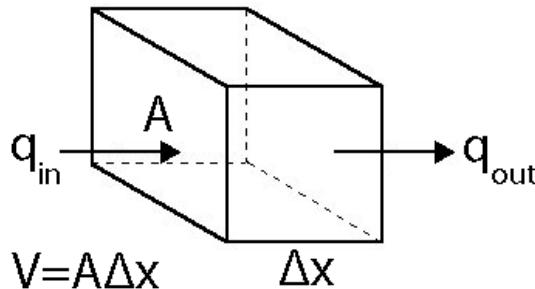
Sources for the following material is:

1. MIT material processing lecture notes part of the MIT Open Courseware initiative
2. Classical Mechanics. John R. Taylor
3. Fundamentals of Heat and Mass Transfer. Incropera, DeWitt, Bergman and Lavine
4. "Scaling Analysis and Prediction of Thermal Aspects of the Plasma Spraying Process Using a Discrete Particle Approach" Jinho Lee and Theodore L. Bergman

Fourier's Law

$$\vec{q} = -k\vec{\nabla}T$$
$$\vec{q} = \text{heat flux } \left[\frac{J}{m^2 s} = \frac{W}{m^2} \right] \quad k = \text{thermal conductivity } \left[\frac{W}{m K} \right]$$

Heat Balance in Small Element



$$\text{heat in} - \text{heat out} + \text{heat generation (chemical reaction)} = \text{heat accumulation}$$

$$Aq_{in} - Aq_{out} = V \frac{\partial H}{\partial t} \quad H = \text{enthalpy per unit volume}$$

$$[q]_x - [q]_{x+\Delta x} = \Delta x \frac{\partial H}{\partial t}$$

$$\frac{\left(\left[-k \frac{\partial T}{\partial t} \right]_x - \left[-k \frac{\partial T}{\partial t} \right]_{x+\Delta x} \right)}{\Delta x} = \frac{\partial H}{\partial t}$$

Limit as $\Delta x \rightarrow 0$

$$\frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) = \frac{\partial H}{\partial t}$$

If k is constant

$$k \frac{\partial^2 T}{\partial x^2} = \frac{\partial H}{\partial t}$$

From definition of heat capacity at constant pressure

$$\rho c_p = \frac{\partial H}{\partial T}$$

Divide by ∂t

$$\frac{\partial H}{\partial t} = \rho c_p \frac{\partial T}{\partial t} = k \frac{\partial^2 T}{\partial x^2}$$

Heat Conduction Equation

$$\frac{\partial T}{\partial t} = \alpha \frac{\partial^2 T}{\partial x^2}$$

$$\alpha = \text{thermal diffusivity} = \frac{k \left[\frac{W}{m * K} \right]}{\rho \left[\frac{kg}{m^3} \right] c_p \left[\frac{J}{K * kg} \right]} = \left[\frac{m^2}{s} \right]$$

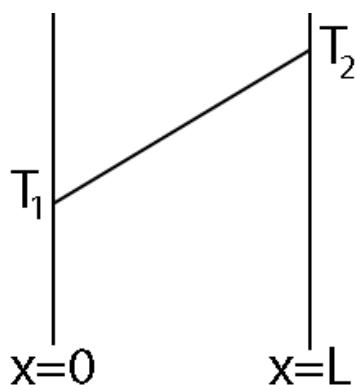
Assumes $k \neq f(x)$

$$\rho c_p \frac{\partial T}{\partial t} = \vec{\nabla} \cdot k \vec{\nabla} T \quad \text{went to} \quad \frac{\partial T}{\partial t} = \alpha \vec{\nabla}^2 T$$

Steady State 1D Conduction

$$\vec{\nabla}^2 T = 0 \quad \text{Laplace Equation}$$

$$1D \quad \frac{\partial^2 T}{\partial x^2} = 0 \quad (\text{a line})$$



$$\int \left[\frac{d}{dx} \left(\frac{dT}{dx} \right) = 0 \right] dx$$

$$\int \left[\frac{dT}{dx} = A \right] dx$$

$$T = Ax + B$$

At $x=0, T=T_1$

$$T_1 = B$$

At $x=L, T=T_2$

$$T_2 = AL + T_1 \quad A = \frac{T_2 - T_1}{L}$$

$$T = \left(\frac{T_2 - T_1}{L} \right) x + T_1$$

$$\frac{T - T_1}{T_2 - T_1} = \frac{x}{L}$$

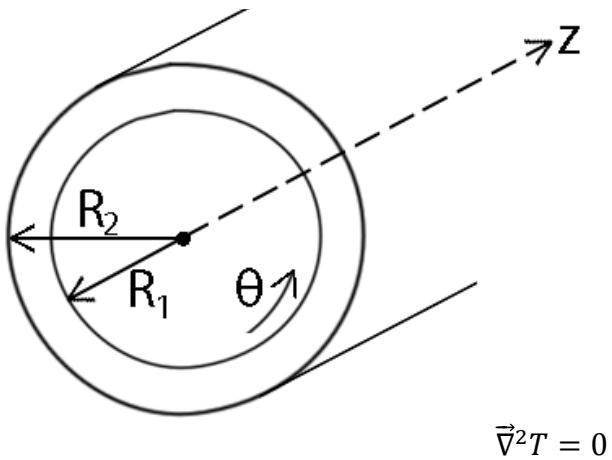
Dimensionless variables T^* and x^*

$$T^* = \frac{T - T_1}{T_2 - T_1} \quad x^* = \frac{x}{L} \quad T^* = x^*$$

Constant heat flux

$$q = -k \frac{dT}{dx} \quad \frac{dT}{dx} = \frac{T_2 - T_1}{L}$$

Steady State in Cylinder



$$\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial T}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 T}{\partial \theta^2} + \frac{\partial^2 T}{\partial z^2} = 0$$

T constant in z and θ directions

$$\frac{1}{r} \frac{d}{dr} \left(r \frac{dT}{dr} \right) = 0$$

$$\int \left[\frac{d}{dr} \left(r \frac{dT}{dr} \right) = 0 \right] dr$$

$$r \frac{dT}{dr} = A$$

$$\int \left[\frac{dT}{dr} = \frac{A}{r} \right] dr$$

Note: $\int dx/x = \ln(x) + C$

$$T = A \ln(r) + B$$

At $r=R_1, T=T_1$

$$T_1 = A \ln(R_1) + B \quad eq1$$

At $r=R_2, T=T_2$

$$T_2 = A \ln(R_2) + B \quad eq1$$

From eq1

$$B = T_1 - A \ln(R_1)$$

From eq2

$$B = T_2 - A \ln(R_2)$$

Substitute eq2 \rightarrow eq1

$$T_1 - A \ln(R_1) = T_2 - A \ln(R_2)$$

$$T_1 - T_2 = A \ln(R_1) - A \ln(R_2) = A \ln\left(\frac{R_1}{R_2}\right)$$

$$A = \frac{T_1 - T_2}{\ln\left(\frac{R_1}{R_2}\right)} \quad B = T_1 - \frac{T_1 - T_2}{\ln\left(\frac{R_1}{R_2}\right)} \ln(R_1)$$

$$T = \frac{T_1 - T_2}{\ln\left(\frac{R_1}{R_2}\right)} \ln(r) + T_1 - \frac{T_1 - T_2}{\ln\left(\frac{R_1}{R_2}\right)} \ln(R_1)$$

Subtract T_1 , divide by $(T_1 - T_2)$

$$\frac{T - T_1}{T_1 - T_2} = \frac{\ln\left(\frac{r}{R_1}\right)}{\ln\left(\frac{R_1}{R_2}\right)}$$

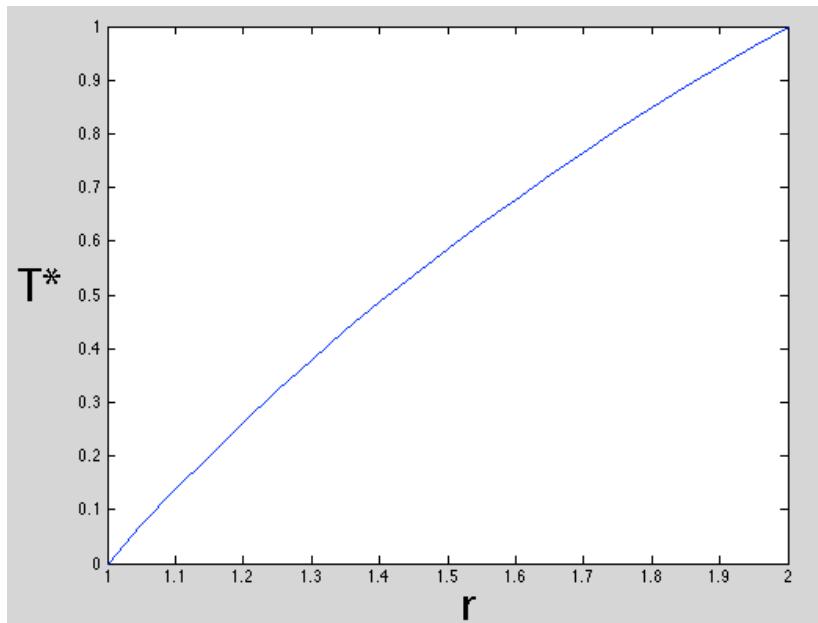
Dimensionless variable T^*

$$T^* = \frac{T - T_1}{T_2 - T_1} = \frac{\ln\left(\frac{r}{R_1}\right)}{\ln\left(\frac{R_2}{R_1}\right)}$$

Constant total heat flux, but flux not constant

$$q = -k \frac{dT}{dr} \neq \text{constant}$$

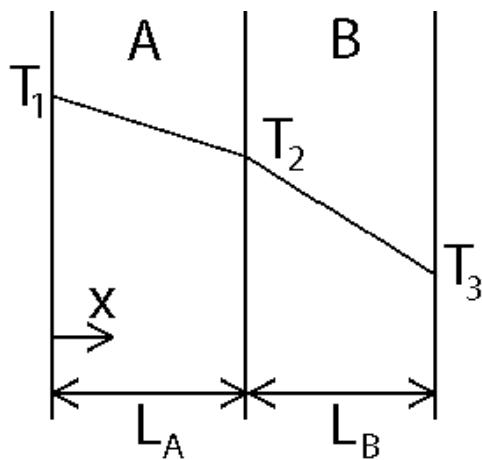
$$A = 2\pi r \quad qA = \text{constant}$$



At $r=1$ $T=T_1$

At $r=2$ $T=T_2$

Steady State in Composite Wall



T_2 is unknown

$$\frac{\partial^2 T}{\partial x^2} = 0 \quad \text{in material A and B}$$

At $x=L_A$, $T=T_2$, $q_{\text{in}}=q_{\text{out}}$

$$q_{\text{in}} = -k_A \left[\frac{dT}{dx} \right]_{L_A^-} = -k_B \left[\frac{dT}{dx} \right]_{L_A^+} = q_{\text{out}}$$

$$k_A \frac{T_1 - T_2}{L_A} = k_B \frac{T_2 - T_3}{L_B}$$

$$T_1 \frac{k_A}{L_A} - T_2 \frac{k_A}{L_A} = T_2 \frac{k_B}{L_B} - T_3 \frac{k_B}{L_B}$$

$$T_2 = \frac{T_1 \frac{k_A}{L_A} + T_3 \frac{k_B}{L_B}}{\frac{k_A}{L_A} + \frac{k_B}{L_B}}$$

Useful conclusions

$$\frac{\Delta T_A}{\Delta T_B} = \frac{\frac{L_A}{k_A}}{\frac{L_B}{k_B}}$$

$$q = -k \frac{\Delta T}{L} \rightarrow \Delta T \propto \frac{L}{k}$$

Thermal circuit

$$R_{cond,A} = \frac{L_A}{k_A} \quad R_{cond,B} = \frac{L_B}{k_B} \quad q = \frac{T_1 - T_3}{\frac{L_A}{k_A} + \frac{L_B}{k_B}}$$

Note: since $q = -k \frac{\partial T}{dx}$, we can use the q from the thermal circuit to go directly to the temperature gradient

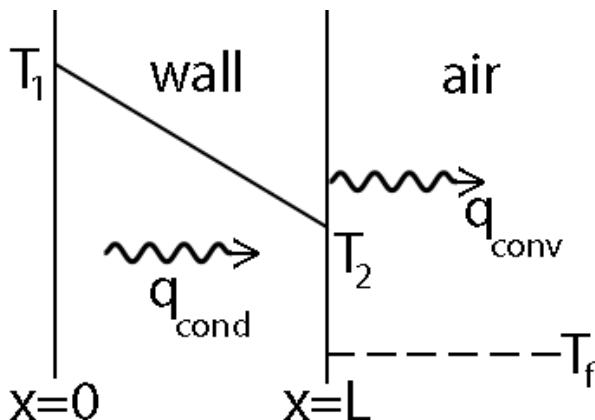
$$\frac{\partial T}{dx} = \frac{-q}{k}$$

Comparison of Steel vs. Mullite

$$k_{Steel} = 30 \frac{W}{m K} \quad k_{Mullite} = 3 \frac{W}{m K}$$

$$\text{if } L_{Steel} = L_{Mullite}, \quad \frac{\Delta T_{Mullite}}{\Delta T_{Steel}} = 10$$

Steady State Convection on Wall



T_2 is unknown

At $x=L$

$$q = h(T_2 - T_f)$$

$$h = \text{convection heat transfer coefficient } \left[\frac{W}{m^2 K} \right]$$

$$q = \text{heat flux } \left[\frac{W}{m^2} \right]$$

$$\frac{\partial^2 T}{\partial x^2} = 0 \quad \text{in wall} \quad T = T_1 + \frac{T_2 - T_1}{L} x$$

At $x=L$, $q_{\text{cond}}=q_{\text{conv}}$

$$-k \frac{dT}{dx} = h(T_2 - T_f) \quad \frac{dT}{dx} = \frac{T_2 - T_1}{L}$$

$$-\frac{k}{L}T_2 + \frac{k}{L}T_1 = hT_2 - hT_f \quad \frac{k}{L}T_1 + hT_f = T_2 \left(h + \frac{k}{L} \right)$$

$$T_2 = \frac{\frac{k}{L}T_1 + hT_f}{h + \frac{k}{L}}$$

$$T = T_1 + \frac{x}{L} \left[\frac{\frac{k}{L}T_1 + hT_f}{h + \frac{k}{L}} - T_1 \right]$$

$$T - T_1 = \frac{x}{L} \left[\frac{h(T_f - T_1)}{h + \frac{k}{L}} \right]$$

$$\frac{T - T_1}{T_f - T_1} = \frac{x}{L} \left[\frac{1}{1 + \frac{k}{hL}} \right]$$

Dimensionless variables T^* and x^*

$$T^* = \frac{T - T_1}{T_f - T_1} \quad x^* = \frac{x}{L} \quad T^* = x^* \left[\frac{1}{1 + \frac{k}{hL}} \right]$$

Thermal circuit

$$R_{cond} = \frac{L}{k} \quad R_{conv} = \frac{1}{h} \quad q = \frac{T_1 - T_f}{\frac{L}{k} + \frac{1}{h}}$$

Biot Number = $B_i = \frac{hL}{k}$ = ratio of conductive resistance to convective

$Bi \ll 1$

-Slow convection, no temperature gradients in solid

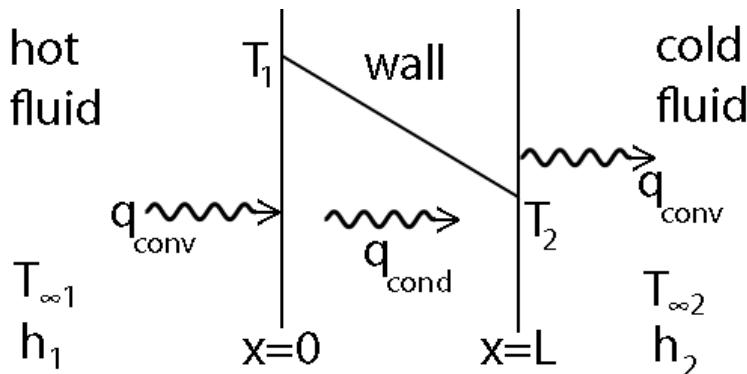
$Bi \approx 1$

-Conduction and convection equally important, transient solution

$Bi \gg 1$

- $T_2 = T_f$, rapid convection

1D Plane Wall Conduction with Convection on Both Sides



T_1 and T_2 are unknown, assuming k is constant otherwise $\frac{d}{dx} \left(k \frac{dT}{dx} \right) = 0$

$$\frac{\partial^2 T}{\partial x^2} = 0 \quad \text{in wall} \quad T = Ax + B$$

At $x=0$, $q_{cond} = q_{conv}$

At $x=L$, $q_{cond}=q_{conv}$

$$-k \left[\frac{dT}{dx} \right]_{x=0+} = h_1(T_{\infty 1} - T_1)$$

$$-k \left[\frac{dT}{dx} \right]_{x=L-} = h_2(T_2 - T_{\infty 2})$$

$$\frac{dT}{dx} = A \quad T_1 = B \quad T_2 = AL + B$$

$$eq1: -kA = h_1(T_{\infty 1} - B)$$

$$eq2: -kA = h_2(AL + B - T_{\infty 2})$$

from eq1: $B = \frac{kA}{h_1} + T_{\infty 1}$ substitute into eq2

$$-kA = h_2 \left(AL + \frac{kA}{h_1} + T_{\infty 1} - T_{\infty 2} \right)$$

Solve for A

$$-kA = Ah_2 \left(L + \frac{k}{h_1} \right) + h_2(T_{\infty 1} - T_{\infty 2})$$

$$A \left(-k - h_2 \left(L + \frac{k}{h_1} \right) \right) = h_2(T_{\infty 1} - T_{\infty 2})$$

$$A = \frac{h_2(T_{\infty 1} - T_{\infty 2})}{-k - h_2 \left(L + \frac{k}{h_1} \right)} = \frac{-(T_{\infty 1} - T_{\infty 2})}{k \left(\frac{1}{h_1} + \frac{1}{h_2} + \frac{L}{k} \right)}$$

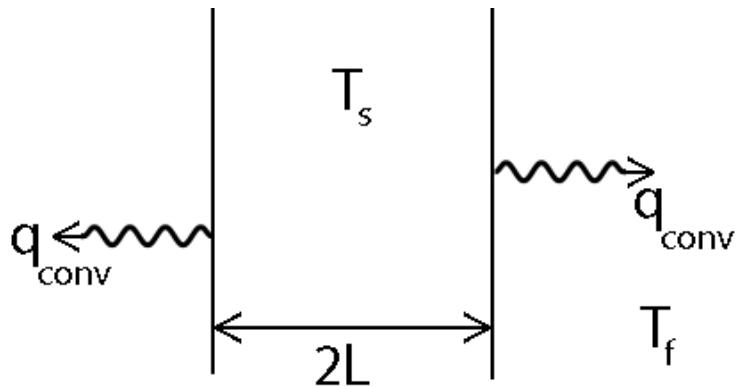
$$B = \frac{kA}{h_1} + T_{\infty 1}$$

$$q = -k \frac{dT}{dx} = -kA$$

Thermal circuit

$$R_{cond} = \frac{L}{k} \quad R_{conv1} = \frac{1}{h_1} \quad R_{conv2} = \frac{1}{h_2} \quad q = \frac{T_{\infty 1} - T_{\infty 2}}{\frac{L}{k} + \frac{1}{h_1} + \frac{1}{h_2}}$$

Low Biot Number Solutions



Global heat balance, $T = f(t)$

$$Ah(T - T_f) = -\rho c_p \frac{dT}{dt} V$$

Separation of variable

$$\int \frac{-Ah}{\rho c_p V} dt = \int \frac{1}{T - T_f} dT$$

Note: Integrate RHS using u-substitution,

$$\begin{aligned} u &= T - T_f, & du &= dT, & \int \frac{1}{T - T_f} dT &= \ln(u) + C \\ && \frac{-Ah}{\rho c_p V} t + C &= \ln(T - T_f) \end{aligned}$$

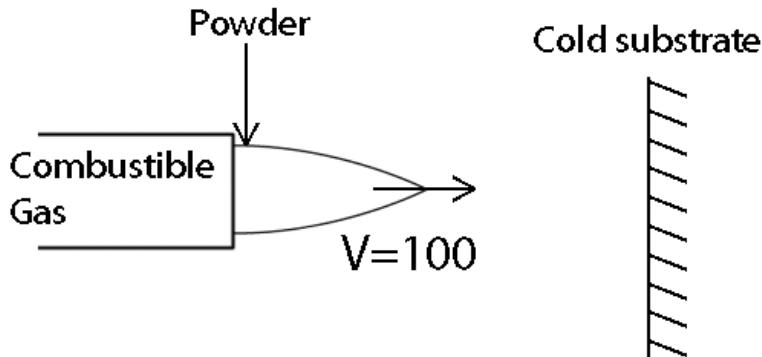
At $t=0, T=T_s$

$$C = \ln(T_s - T_f)$$

$$\frac{-Ah}{\rho c_p V} t = \ln(T - T_f) - \ln(T_s - T_f) = \ln\left(\frac{T - T_f}{T_s - T_f}\right)$$

$$\frac{T - T_f}{T_s - T_f} = \exp\left(\frac{-Ah}{\rho c_p V} t\right)$$

Thermal Spray Coatings / Plasma Spray



Spray is quasi-steady state because it is assumed that neither the plasma torch nor substrate moves with respect to the other, and that the particle is heated isothermally. This assumption is justified with $Bi \ll 1$.

oxyacetylene torch: $T = 3000K$

powder: Ni alloy MAR-M200 $R = 2 - 50 \mu m$ ($\mu m = 10^{-6}m$)

Problem: need to find time that spherical particle reaches melting temperature T_m

At $r=0$, symmetry $\frac{\partial T}{\partial r} = 0$

At $r=R$ convection into ambient air, $q = h(T - T_f)$ $T_f = 3000K$ $h \approx 500 \frac{W}{m^2 K}$

Governing equations

$$Bi = \frac{hL_c}{k}$$

L_c is the character length defined as volume/area ratio, $R/3$ for sphere, L for wall of thickness $2L$, $R/2$ for cylinder

$$h = 500 \frac{W}{m^2 K} \quad R = 50 * 10^{-6}m \quad k = 16 \frac{W}{m K}$$

$$Bi = \frac{hR}{k3} = 5.2 * 10^{-4} \ll 1 \quad \text{Can use lumped capacitance model}$$

$$\frac{T - T_f}{T_0 - T_f} = \exp\left(\frac{-Ah}{\rho c_p V} t\right)$$

$$\frac{V}{A} = \frac{\frac{4}{3}\pi r^3}{4\pi r^2} = \frac{R}{3}$$

Time to reach melting temperature

$$t_1 = \ln\left(\frac{T_m - T_f}{T_0 - T_f}\right) \frac{-\rho c_p R}{h 3} = 0.1035 s$$

$$T_f = 3000K \quad T_0 = 300K \quad h = 500 \frac{W}{m^2 K}$$

$$\rho = 8500 \frac{kg}{m^3} \quad c_p = 500 \frac{J}{kg K} \quad T_{m,Ni} = 1700K$$

Once particle gets to melting temperature, still need to transfer energy to melt it

During heating:

$$Ah(T - T_f) = -\rho c_p \frac{dT}{dt} V$$

During melting:

$$Ah(T_m - T_f) = \rho V h_{sf} \frac{df_s}{dt}$$

f_s = particle's solid fraction h_{sf} = particle's latent heat of melting

$$h_{sf,Ni} = 297 * 10^3 \frac{J}{kg}$$

$$Ah(T_m - T_f) = \rho V h_{sf} \frac{df_s}{dt}$$

Separation of variable

$$\int \frac{Ah(T_m - T_f)}{\rho V h_{sf}} dt = \int df_s$$

$$f_s = \frac{Ah(T_m - T_f)}{\rho V h_{sf}} t + C$$

At t=0, $f_s=1$, $C=1$

$$\frac{V}{A} = \frac{R}{3}$$

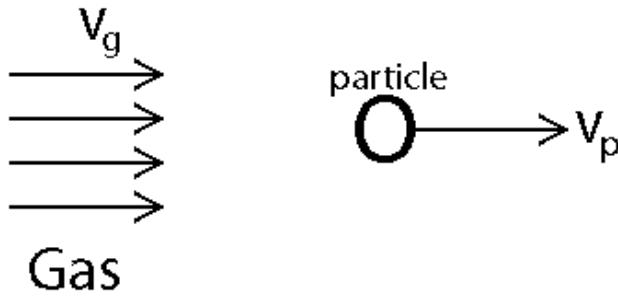
Want time when $f_s=0$

$$t_2 = \frac{-\rho V h_{sf}}{Ah(T_m - T_f)} = \frac{-\rho R h_{sf}}{3h(T_m - T_f)} = 0.0647 s$$

Add time to reach T_m and time to melt to get total time

$$t_{tot} = 0.1683 s$$

Calculating Acceleration and Velocity of Particle Using Air Resistance



For speeds not approaching the speed of sound

$$f_{drag} = f_{lin} + f_{quad} = bv + cv^2$$

For spherical projectile b and c have the form

$$b = \beta D \quad c = \gamma D^2$$

For air at standard temperature (0°C) and pressure (1 atm), which is inaccurate but they were the values that were available

$$\beta = 1.6 * 10^{-4} \frac{N s}{m^2} \quad \gamma = 0.25 \frac{N s^2}{m^4}$$

We can compare f_{lin} and f_{quad} to see if one can be neglected, using $v=100\text{m/s}$ and $D = 100 * 10^{-6}\text{m}$

$$\frac{f_{quad}}{f_{lin}} = \frac{\gamma D^2 v^2}{\beta D v} = \frac{\gamma D v}{\beta} = 15.625$$

Quadratic is dominant at $v=100\text{m/s}$ but its dominance diminishes as the particle gets up to speed, for the sake of completeness we will solve using both linear and quadratic drag.

From F=ma

$$\frac{dv}{dt} = \frac{b}{m}(v_g - v) + \frac{c}{m}(v_g - v) * abs(v_g - v)$$

$$m = V\rho = \frac{4}{3}\pi r^3 \rho \quad v_g = 100 \frac{m}{s} \quad \gamma = 0.25 \frac{N s^2}{m^4} \quad c = \gamma D^2$$

$$\begin{aligned} \beta &= 1.6 * 10^{-4} \frac{N s}{m^2} & b &= \beta D \\ D &= 100 * 10^{-6} m & R &= 50 * 10^{-6} m & \rho &= 8500 \frac{kg}{m^3} \end{aligned}$$

I chose to solve this numerically using Matlab

At $t = 0.1683\text{ s}$ distance traveled is 13.1060 m

Code used:

```
function velocity_using_quadandlinear_drag
roe=8500;
R=50e-6;
D=2*R;
gamma=0.25;
beta=1.6e-4;
c=gamma*D^2;
b=beta*D;
V=4/3*pi*R^3;
m=roe*V;
vg=100;

%initial conditions
v0=0;

%time span 0 to 10 sec
tspan = [0 .1683];

[t ,v] = ode15s(@quaddrag, tspan, v0);

%calculate distance
distance=trapz(t,v)

function dvdt = quaddrag(t, v)%time, mass fraction
    dvdt = b/m*(vg-v)+c/m*(vg-v)*abs(vg-v);
end
end
```