

Couette Flow with slip boundary condition

Ganesh Meshram

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1 Couette Flow with Slip Boundary Condition:

The velocity profile for an incompressible Couette flows with slip can be obtained by considering a two dimensional channel extending between $\mathbf{x} = \mathbf{0}$ to $\mathbf{x} = \mathbf{L}$ and $\mathbf{y} = \mathbf{0}$ to $\mathbf{y} = \mathbf{H}$, with top surface moving with a prescribed velocity \mathbf{U}_∞ .

Assumptions

- Steady, incompressible flow, and no mass transport.
- No end effects, $\frac{\partial}{\partial \mathbf{x}} () = \mathbf{0}$.
- No edge effects, $\frac{\partial}{\partial \mathbf{z}} () = \mathbf{0}$.
- Only gravity force.

Continuity equation:

$$\frac{\partial \mathbf{u}}{\partial \mathbf{x}} + \frac{\partial \mathbf{v}}{\partial \mathbf{y}} + \frac{\partial \mathbf{w}}{\partial \mathbf{z}} = \mathbf{0} \quad (1)$$

By neglecting edge effects and end effects, the above continuity equation(1) can be reduced to

$$\frac{\partial \mathbf{v}}{\partial \mathbf{y}} = \mathbf{0} \quad (2)$$

X-momentum equation: At $\mathbf{y} = \mathbf{0}$, $\mathbf{v} = \mathbf{0}$, $\mathbf{w} = \mathbf{0}$ everywhere

$$\frac{\partial \mathbf{u}}{\partial \mathbf{t}} + \mathbf{u} \frac{\partial \mathbf{u}}{\partial \mathbf{x}} + \mathbf{v} \frac{\partial \mathbf{u}}{\partial \mathbf{y}} + \mathbf{w} \frac{\partial \mathbf{u}}{\partial \mathbf{z}} = -\frac{1}{\rho} \frac{\partial \mathbf{P}}{\partial \mathbf{x}} + \nu \left[\frac{\partial^2 \mathbf{u}}{\partial \mathbf{x}^2} + \frac{\partial^2 \mathbf{u}}{\partial \mathbf{y}^2} + \frac{\partial^2 \mathbf{u}}{\partial \mathbf{z}^2} \right] + \mathbf{F}_\mathbf{x} \quad (3)$$

After considering steady state, neglecting edge effects and end effects, and other forces the above x-momentum equation(3) can be reduced to

$$\mathbf{0} = \nu \left[\frac{\partial^2 \mathbf{u}}{\partial \mathbf{y}^2} \right] \quad (4)$$

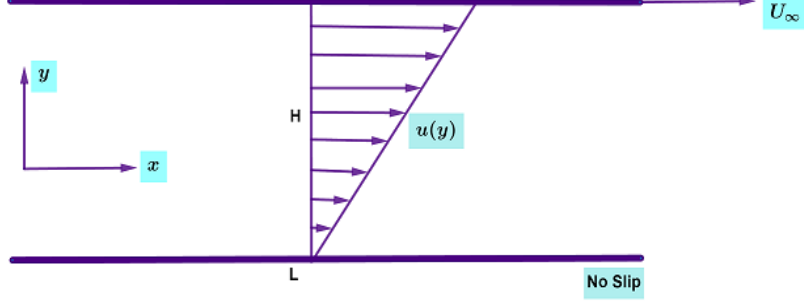


Figure 1: Analysis of Couette Flow with slip boundary condition

or

$$\left[\frac{\partial^2 \mathbf{u}}{\partial y^2} \right] = 0 \quad (5)$$

Now non- dimensionalizing the above equation

$$\mathbf{u}^* = \frac{\mathbf{u}}{u_\infty}, y^* = \frac{y}{h} \quad (6)$$

$$\mathbf{u} = u_\infty \mathbf{u}^*, y = h y^* \quad (7)$$

$$\frac{d^2(u_\infty \mathbf{u}^*)}{d(h y^*)^2} = 0 \quad (8)$$

$$\frac{u_\infty}{h^2} \frac{d^2 \mathbf{u}^*}{d y^{*2}} = 0 \quad (9)$$

$$\frac{d^2 \mathbf{u}^*}{d y^{*2}} = 0 \quad (10)$$

Now integrating the above equation (Eq:10), then we get

$$\mathbf{u}^* = C_1 y^* + C_2 \quad (11)$$

Slip boundary condition

$$\mathbf{u}_s - \mathbf{u}_w = \left(\frac{2 - \sigma_v}{\sigma_v} \right) \frac{1}{\rho} \sqrt{\frac{\pi}{2RT_\infty}} \tau_s + \frac{3}{4} \frac{\text{Pr}}{\rho RT_w} \left(\frac{\gamma - 1}{\gamma} \right) (-\mathbf{q}_s) \quad (12)$$

$$\mathbf{u}_s - \mathbf{u}_w = \left(\frac{2 - \sigma_v}{\sigma_v} \right) \frac{\mu}{\rho} \sqrt{\frac{\pi}{2RT_\infty}} \left(\frac{\partial \mathbf{u}_s}{\partial n} \right)_{\text{wall}} + \frac{3}{4} \frac{\text{Pr}}{\rho RT_w} \left(\frac{\gamma - 1}{\gamma} \right) \left(\mathbf{k} \frac{\partial \mathbf{T}}{\partial \mathbf{x}} \right)_{\text{wall}} \quad (13)$$

Where, $\tau_s = \mu \left(\frac{\partial \mathbf{u}_s}{\partial y} \right)_{\text{wall}}$ and $-\mathbf{q}_s = (\mathbf{k} \frac{\partial \mathbf{T}}{\partial \mathbf{x}})_{\text{wall}}$
 After non-dimensionalizing the above slip B.C:

$$\mathbf{u}_s^* - \mathbf{u}_w^* = \left(\frac{2 - \sigma_v}{\sigma_v} \right) \text{Kn} \left(\frac{\partial \mathbf{u}_s^*}{\partial \mathbf{n}^*} \right)_{\text{wall}} + \frac{3}{2\pi} \frac{\text{Kn}^2 \text{Re}}{\text{Ec}} \left(\frac{\gamma - 1}{\gamma} \right) \left(\frac{\partial \mathbf{T}^*}{\partial \mathbf{x}^*} \right)_{\text{wall}} \quad (14)$$

or

$$\mathbf{u}_s^* - \mathbf{u}_w^* = \alpha \left(\frac{\partial \mathbf{u}_s^*}{\partial \mathbf{n}^*} \right)_{\text{wall}} + \beta \quad (15)$$

Where,

$$\alpha = \left(\frac{2 - \sigma_v}{\sigma_v} \right) \text{Kn} \quad (16)$$

$$\beta = \frac{3}{2\pi} \frac{\text{Kn}^2 \text{Re}}{\text{Ec}} \left(\frac{\gamma - 1}{\gamma} \right) \quad (17)$$

Applying boundary conditions:

a) At lower wall :

$$\mathbf{y} = \mathbf{0}, \mathbf{y}^* = \frac{\mathbf{y}}{\mathbf{h}} \Rightarrow \frac{\mathbf{0}}{\mathbf{h}}, \Rightarrow \mathbf{y}^* = \mathbf{0}$$

$$\mathbf{u}^* = \mathbf{u}_s^*, \mathbf{u}_w = \mathbf{0} (\text{stationary wall})$$

$$\mathbf{u}_s^* = \mathbf{u}_w^* + \alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{n}^*} + \beta \quad (18)$$

$$\mathbf{u}_w^* = \frac{\mathbf{u}_w}{\mathbf{u}_\infty} \Rightarrow \frac{\mathbf{0}}{\mathbf{u}_\infty} = \mathbf{0}$$

Also,

$$\alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{n}^*} = \alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{y}^*} \quad (19)$$

From equation (11),

$$\mathbf{u}^* = \mathbf{C}_1 \mathbf{y}^* + \mathbf{C}_2 \quad (20)$$

$$\frac{\partial \mathbf{u}^*}{\partial \mathbf{y}^*} = \mathbf{C}_1 \quad (21)$$

Now, from equation(18)

$$\mathbf{u}_s^* = \mathbf{0} + \alpha \mathbf{C}_1 + \beta \quad (22)$$

$$\mathbf{u}_s^* = \alpha \mathbf{C}_1 + \beta \quad (23)$$

From equation(11),

$$\mathbf{u}_s^* = \mathbf{C}_1 \mathbf{y}^* + \mathbf{C}_2 \quad (24)$$

$$\alpha \mathbf{C}_1 + \beta = \mathbf{C}_1(0) + \mathbf{C}_2 \quad (25)$$

$$\alpha \mathbf{C}_1 + \beta = \mathbf{C}_2 \quad (26)$$

b) At upper wall:

$$\mathbf{u}_w = \mathbf{u}_\infty, \mathbf{u}^* = \mathbf{u}_s^*, \mathbf{u}_w^* = \frac{\mathbf{u}_w}{\mathbf{u}_\infty} = \frac{\mathbf{u}_\infty}{\mathbf{u}_\infty} = \mathbf{1}, \Rightarrow \mathbf{u}_w^* = \mathbf{1}$$

$\mathbf{y} = \mathbf{h}, \mathbf{y}^* = \frac{\mathbf{y}}{\mathbf{h}} = \frac{\mathbf{h}}{\mathbf{h}} = 1 \Rightarrow \mathbf{y}^* = 1$
Now,

$$\mathbf{u}_s^* = \mathbf{u}_w^* + \alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{n}^*} + \beta \quad (27)$$

Here,

$$\alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{n}^*} = -\alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{y}^*} = -\frac{\partial (\mathbf{C}_1 \mathbf{y}^* + \mathbf{C}_2)}{\partial \mathbf{y}^*} \quad (28)$$

$$\alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{n}^*} = -\alpha \frac{\partial \mathbf{u}_s^*}{\partial \mathbf{y}^*} = -\mathbf{C}_1 \quad (29)$$

$$\mathbf{u}_s^* = 1 + \alpha(-\mathbf{C}_1) + \beta \quad (30)$$

$$\mathbf{u}_s^* = 1 - \alpha \mathbf{C}_1 + \beta \quad (31)$$

Again,

$$\mathbf{u}_s^* = \mathbf{C}_1 \mathbf{y}^* + \mathbf{C}_2 \quad (32)$$

$$1 - \alpha \mathbf{C}_1 + \beta = \mathbf{C}_1(1) + \mathbf{C}_2 \quad (33)$$

$$1 - \alpha \mathbf{C}_1 + \beta = \mathbf{C}_1 + \mathbf{C}_2 \quad (34)$$

From equation (26), put the value of \mathbf{C}_2 in equation 33

$$1 - \alpha \mathbf{C}_1 + \beta = \mathbf{C}_1 + \alpha \mathbf{C}_1 + \beta \quad (35)$$

$$1 = \mathbf{C}_1 + \alpha \mathbf{C}_1 + \alpha \mathbf{C}_1 = \mathbf{C}_1 + 2\alpha \mathbf{C}_1 \quad (36)$$

$$1 = (1 + 2\alpha \mathbf{C}_1) \quad (37)$$

$$\mathbf{C}_1 = \frac{1}{(1 + 2\alpha)} \quad (38)$$

From equation 26,

$$\alpha \mathbf{C}_1 + \beta = \mathbf{C}_2 \quad (39)$$

$$\frac{\alpha}{1 + 2\alpha} + \beta = \mathbf{C}_2 \quad (40)$$

$$\mathbf{C}_2 = \left(\frac{\alpha}{1 + 2\alpha} \right) + \beta \quad (41)$$

$$\Rightarrow \mathbf{u}^* = \mathbf{C}_1 \mathbf{y}^* + \mathbf{C}_2 \quad (42)$$

$$\Rightarrow \mathbf{u}^* = \left(\frac{1}{1 + 2\alpha} \right) \mathbf{y}^* + \left(\frac{\alpha}{1 + 2\alpha} \right) + \beta \quad (43)$$

$$\Rightarrow \mathbf{u}^* = \left(\frac{\mathbf{y}^* + \alpha}{1 + 2\alpha} \right) + \beta \quad (44)$$

$$\Rightarrow \frac{\mathbf{u}}{\mathbf{u}_\infty} = \left(\frac{\frac{\mathbf{y}}{\mathbf{h}} + \alpha}{1 + 2\alpha} \right) + \beta \quad (45)$$

Now, replacing the values of α , and β , then the above equation (45) can be reduced to,

$$\Rightarrow \frac{u}{u_\infty} = \left[\frac{\frac{y}{h} + \left(\frac{2-\sigma_v}{\sigma_v} \right) Kn}{1 + 2 \left(\frac{2-\sigma_v}{\sigma_v} \right) Kn} + \frac{3}{2\pi} \frac{Kn^2 Re}{E_c} \left(\frac{\gamma-1}{\gamma} \right) \left(\frac{\partial T_s}{\partial x} \right) \right] \quad (46)$$

This is the expression of slip velocity for couette flow.