II, Iz, I3 : Invariants of Stress

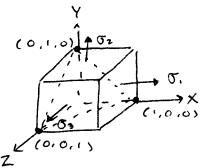
Three roots: 5, 7 52 7 53

In the principal axes, the state of stress

$$\begin{bmatrix}
5. & 0 & 0 \\
0 & 5_1 & 0 \\
0 & 0 & 5_3
\end{bmatrix}$$

=> 
$$\begin{cases} I_1 = \sigma_1 + \sigma_2 + \sigma_3 \\ I_2 = \sigma_1 \sigma_2 + \sigma_2 \sigma_3 + \sigma_3 \sigma_1 \\ I_3 = \sigma_1 \sigma_2 \sigma_3 \end{cases}$$

2.4.4 Octahedral Stress



XYZ: the principal directions

Consider the Family of planes whose und normal vectors satisfy: L2 = M2 = 12 = 1/3

The normal stress:

$$\int_{\text{out}}^{2} = \int_{-\infty}^{2} \int_{-\infty}^{\infty} + m^{2} \int_{yy}^{2} + n^{2} \int_{zz}^{2} + 2 \int_{\text{more}}^{\infty} + 2 \int_{yz}^{\infty} + 2 \int_{yz}^{\infty} + 2 \int_{zz}^{\infty} +$$

2.4.5 Mean and Deviator Stresses

The mean (normal) Stress:

Der:ne

[Tm]: hydro static Stress

[Ta] : pure shear stress

Example: The state of stress at a point is given by 
$$[T] = \begin{bmatrix} -19.0 & -4.70 & 6.45 \\ -4.70 & 4.60 & 11.8 \end{bmatrix}$$
 MPa  $\begin{bmatrix} 6.45 & 11.8 & -8.30 \end{bmatrix}$ 

Determine the principal stress and their orientation wir. t.

the original coordinate system.

Solution: 
$$I_1 = Gxx + Gyy + Gzz$$

$$= -19.0 + 4.60 - 8.30 = -22.7 \text{ MPa}$$

$$\text{necessary}$$

$$I_2 = Gxx Gyy + Gyy Gzz + Gxx Gzz - Gxy^2 - Gyz^2 - Gzz^2$$

$$= -170.8125 \text{ MPa}^2$$

$$I_3 = |T| = 2647.522 \text{ MPa}^3$$

The cubic eq.n:  

$$0^3 - I_1 0^2 + I_2 0 - I_3 = 0$$
  
=>  $0^3 + 22.70^2 - 170.8250 - 2647.523 = 0$   
 $0^3 = 25 \cos \alpha + 1/3 I_1$   
 $0^3 = 25 \cos (\alpha + 120^\circ) + 1/3 I_2$   
 $0^3 = 25 \cos (\alpha + 240^\circ) + 1/3 I_3$   
Here  $0^3 = \sqrt{1/2} - 1/3 I_2$ 

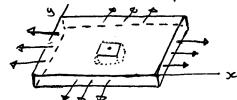
$$Q = \frac{3}{3}(-22.7)(-170.8125) - (2647.522) - \frac{2}{27}(-22.7)$$
= -488.590

$$\begin{array}{l} \therefore \ \, \text{$\mathbb{K}$} = 1/3 \ \cos^4\left(-\frac{-488.690}{2\times 1220.26}\right) \\ = 0.466412 \ \text{rad} = 26.151^{\circ} \\ \therefore \ \, \text{$\mathbb{T}$} = 2 \left(10.6861\right) \cos 26.161^{\circ} + \left(-\frac{32.7}{3}\right) = 11.6179 \\ \text{$\mathbb{T}$} = 2 \left(10.6861\right) \cos \left(26.151^{\circ} + 190^{\circ}\right) + \left(-\frac{72.7}{3}\right) = -25.3163 \\ \text{$\mathbb{C}$} = 2 \left(10.6861\right) \cos \left(26.151^{\circ} + 240^{\circ}\right) + \left(-\frac{72.7}{3}\right) = -2.63163 \\ \text{$\mathbb{C}$} = 2 \left(10.6861\right) \cos \left(26.151^{\circ} + 240^{\circ}\right) + \left(-\frac{72.7}{3}\right) = -9.0015 \\ \text{$\mathbb{C}$} = -9.0015 \ \text{$\mathbb{MP}$} \\ \text{$\mathbb{C}$} = -9.0015 \ \text{$\mathbb{C}$} \\ \text{$\mathbb{C}$} = -9.$$

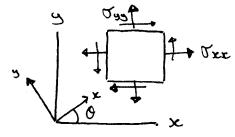
For  $\sigma_2$ : (-0.6209, 0.3802, -0.6856) For  $\sigma_3$ : (-0.7834, -0.3306, 0.5862)

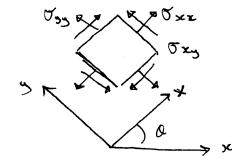
## 2.4.6 Plane Stress

Consider a thin plate



non zero stress: Txx, Tyy, Tzz } plane
others: Tzx = Tzy = Tzz = & Stress





$$\begin{array}{lll}
\overline{N}_{i} &= l_{i}\overline{i}^{2} + m_{i}\overline{j}^{2} + l_{i}\overline{k}^{2} \\
l_{i} &= Cosd; m_{i} &= Sind; n_{i} &= \emptyset \\
\overline{N}_{a} &= l_{z}\overline{i}^{2} + m_{z}\overline{j}^{2} + l_{z}\overline{k}^{2} \\
l_{z} &= -Sind; m_{z} &= Cosd; l_{z} &= \emptyset
\end{array}$$

$$\frac{\sigma_{\text{avg}} = \sigma_{\text{xr}} + \sigma_{\text{yy}}}{2} = \frac{I}{2}$$

$$(*) \mathcal{O}_{XY} = -\frac{\mathcal{O}_{XX} - \mathcal{O}_{YY}}{2} \leq n 20 + \mathcal{O}_{XY} \cos 20$$

max or min. normal stress occurs when 
$$\sigma_{xy} = \sigma$$
  $\sigma_{xy} = \sigma_{xy}$   $\sigma_{xx} = \sigma_{xy}$   $\sigma_{xx} = \sigma_{xy}$   $\sigma_{xx} = \sigma_{xy}$   $\sigma_{xx} = \sigma_{xy}$  the principal stresses:

$$\mathcal{T}_{1} = \mathcal{T}_{avg} + R$$
,  $\mathcal{T}_{2} = \mathcal{T}_{avg} - R$ 
Here,  $\mathcal{R} = \sqrt{\left(\frac{\mathcal{T}_{xx} - \mathcal{T}_{yy}}{2}\right)^{2} + \mathcal{T}_{xy}^{2}}$ 

Find the principal stresses

Solution: Txx = 350 Tyy = -200 Tru = 500

$$\int avg = \frac{0xx + 0yy}{2} = \frac{350 - 200}{2} = 75$$

$$R = \sqrt{\left(\frac{0xx - 0yy}{2}\right)^2 + 0xy^2}$$

$$= > \sqrt{\left(\frac{350 + 200}{2}\right)^2 + 500^2} = > 570.64$$

$$tan 20p = 20xy = 2(500) = 1.82$$
 $0xx - 0yy = 350 - 200$ 
 $0p = 30.59^{\circ}, 120.59^{\circ}$ 

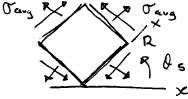
Let \$ = 30,59°

Max: n-plane shear stress

$$\frac{d\sigma_{xy}}{d\sigma} = 0 = \frac{-\sigma_{xx} - \sigma_{yy}}{2} (2\cos 2\theta) - \sigma_{xy}(2\sin 2\theta) = 0$$

$$= \frac{2}{2} \tan 2\theta_s = -\sigma_{xx} - \sigma_{yy}$$
and  $\sigma_{xy}$ , max = R

the normal stress is Vavy



READING

2.5 Differential equations of motion of a deformable body 1° 10 stress analysis - axial member

$$AG(x)$$
 $X$ 
 $X+dx$ 
 $+dG(x)+dG(x)$ 

$$\begin{aligned}
&\mathcal{E} F_{x} = \emptyset : \\
&T(x) d_{x} + (\sigma(x) + d\sigma) A - A\sigma(x) = \emptyset \\
&\Rightarrow A d\theta + T(x) dx = \emptyset \\
&\Rightarrow \int_{0}^{\infty} dx + T(x) A = \emptyset
\end{aligned}$$

Define: 
$$Bx = T(x)$$
 (force/volume)
$$\frac{d\sigma}{dx} + Bx = 0$$

$$\mathcal{E}Fx = \emptyset:$$

$$- 0xxdy + (0xx + \frac{\partial 0xx}{\partial x} dx)dy ...$$
... - 0xydx + (0xy + \frac{\partial 0xy}{\partial y} dy)dxe ...

... + Bx dxdy = \O

=> \frac{\partial 0xx}{\partial x} + \frac{\partial 0xy}{\partial y} + Bx = \O

\text{EFy:}

$$\frac{\partial \delta xy}{\partial x} + \frac{\partial \delta yy}{\partial y} + \beta y = \emptyset$$

$$\frac{\partial \nabla xx}{\partial x} + \frac{\partial \nabla xy}{\partial y} + \frac{\partial \nabla xz}{\partial z} + Bx = 0$$

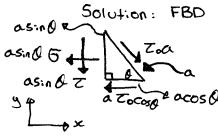
$$\frac{\partial \nabla xy}{\partial x} + \frac{\partial \nabla yy}{\partial y} + \frac{\partial \nabla zz}{\partial z} + By = 0$$

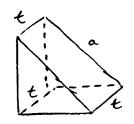
$$\frac{\partial \nabla xz}{\partial x} + \frac{\partial \nabla yz}{\partial y} + \frac{\partial \nabla zz}{\partial z} + Bz = 0$$

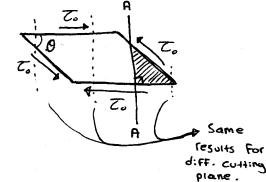


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Example: A thin plate is subjected to uniform shear stress Find the stress vector on the cutting plane A-A



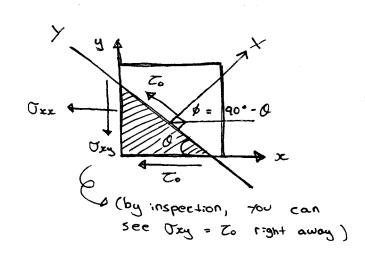




=> 
$$G = -2 T_0 \frac{\cos \theta}{\sin \theta} = -2 T_0 \cot \theta$$
 (normal stress)  
£Fy = 0:

(Putting Small triangle into differential element)

State of Stress



Stress transformation

$$\mathcal{T}_{XY} = \mathcal{T}_0 = -\left(\frac{\mathcal{T}_{XX} - \mathcal{T}_{39}}{2}\right) \leq :n 2(90^{\circ} - 0) + \mathcal{T}_{X9} \cos 2(90^{\circ} - 0)$$

$$T_0 = -\frac{\sigma_{xx}}{2} \sin 2\theta - \sigma_{xy} \cos 2\theta$$
 2

$$\underbrace{\text{Example}}_{[T]} : \begin{bmatrix} 5 & 0 & 0 \\ 0 & -6 & -12 \\ 0 & -12 & 1 \end{bmatrix} \quad (MPa)$$

Determine

b) the principal directions

Solution: 
$$Oxx = 5$$
  $Oxy = Oxz = 0$ 

$$Oyx = 0 Oyy = -6 Oyz = -12$$

$$Ozx = 0 Ozy = -12 Ozz = 1$$

and I, = 
$$Gxx + Gyy + Gzz = 5-6+1=0$$
  
 $I_2 = GxxGyy + GyyGzz + GzzGxx - Gxy^2 - Gyz^2 - Gzx^2$   
 $\Rightarrow -175$ 

$$I_{3} = \begin{vmatrix} 5 & 0 & 0 \\ 0 & -6 & -12 \\ 0 & -12 & 1 \end{vmatrix} = -750$$

=> 
$$0^3 - I_10^2 + I_20 - I_3$$
  
 $0^3 - 1950 + 950 = 0$ 

$$0 = 5$$
 is a root (by inspection)

$$0 = 10$$
 and  $0 = -15$  are the rest of the roots

$$\frac{0^3 - 1750 + 750}{0-5} = 0^2 + 50 - 150 = 0$$