

ROBUST CONTROL VIA INTEGRAL ACTION

It is a known fact, also from basic control courses, that if the loop function has a pole in the origin then the regulated error converges to zero robustly.

As we shall see here this is true also for (a class of) nonlinear systems

STANDING ASSUMPTION: The error $\tilde{y}_r = y_r - y_r^*$ between the regulated output y_r and its target value y_r^* is measured

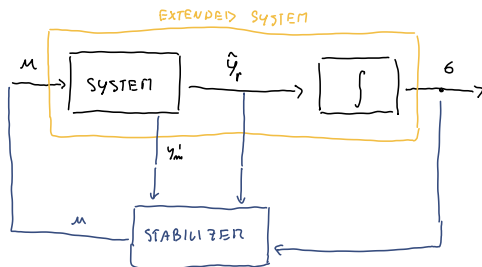
$$\hookrightarrow y_m = \begin{pmatrix} \tilde{y}_r \\ y_m' \end{pmatrix} \quad \text{for some } y_m'$$

Moreover, the number of inputs equals that of regulated outputs: $n_u = n_r$

GENERAL PARADIGM: we add an integrator (POLE IN THE ORIGIN) in series to the controlled system processing the error \tilde{y}_r (in the multi-variable case it's one integrator for each component of \tilde{y}_r).

This produces an "EXTENDED SYSTEM".

Then, we stabilize the extended system



BIBLIOGRAPHY: H. KHALIL, NONLINEAR SYSTEMS, Chap. 12.3

PRELIMINARY MATERIAL

Let us start considering the following SERIES interconnection:



QUESTION: when is this series stabilizable from u ?

The series satisfies:

$$\begin{pmatrix} \dot{x} \\ \dot{z} \end{pmatrix} = \begin{pmatrix} A & 0 \\ C & 0 \end{pmatrix} \begin{pmatrix} x \\ z \end{pmatrix} + \begin{pmatrix} B \\ 0 \end{pmatrix} u \quad (x \in \mathbb{R}^n, u \in \mathbb{R}^m, z \in \mathbb{R}^p)$$

RESULT. Let $A \in \mathbb{R}^{n \times n}$, $B \in \mathbb{R}^{n \times m}$, $C \in \mathbb{R}^{p \times n}$ be such that (A, B) is stabilizable. Then

$$\begin{pmatrix} A & 0 \\ C & 0 \end{pmatrix}, \begin{pmatrix} B \\ 0 \end{pmatrix}$$

is stabilizable if

$$\text{rank} \begin{pmatrix} A & B \\ C & 0 \end{pmatrix} = \# \text{ rows} = n + p \quad (\text{NON-RESONANCE CONDITION})$$

(proof from Module 1)

RESULT. Let $K = [K_1 \ K_2]$ ($K_1 \in \mathbb{R}^{m \times n}$, $K_2 \in \mathbb{R}^{m \times p}$) be such that

$$\begin{pmatrix} A & 0 \\ C & 0 \end{pmatrix} + \begin{pmatrix} B \\ 0 \end{pmatrix} K = \begin{pmatrix} A+BK_1 & BK_2 \\ C & 0 \end{pmatrix}$$

is Hurwitz. Then, K_2 is invertible.

Proof.

Suppose, by contradiction that K_2 is singular. Then

$$\exists w \in \mathbb{R}^p \setminus \{0\}, K_2 w = 0.$$

But this implies

$$\begin{bmatrix} A+BK_1 & BK_2 \\ C & 0 \end{bmatrix} \cdot \begin{bmatrix} 0 \\ w \end{bmatrix} = \begin{bmatrix} BK_2 w \\ 0 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} = 0 \cdot \begin{bmatrix} 0 \\ w \end{bmatrix}$$

$\Rightarrow 0$ is an eigenvalue of $\begin{pmatrix} A+BK_1 & BK_2 \\ C & 0 \end{pmatrix}$, a contradiction.

STATE - FEEDBACK CASE :

$$y_m' = x$$

RESULT. Given y_r^* , let (x^*, u^*) be a solution to the SOLVABILITY Eqs

$$\begin{cases} 0 = f(x^*, u^*) \\ y_r^* = h_r(x^*, u^*) \end{cases}$$

and let $A = \frac{\partial f}{\partial x}(x^*, u^*)$, $B = \frac{\partial f}{\partial u}(x^*, u^*)$, $C_r = \frac{\partial h_r}{\partial x}(x^*)$. Suppose that

(A, B) is stabilizable and

$$\text{rank} \begin{pmatrix} A & B \\ C_r & 0 \end{pmatrix} = n_x + n_r$$

and let $K = [K_1 \ K_2]$ with $K_1 \in \mathbb{R}^{n_m \times n_x}$, $K_2 \in \mathbb{R}^{n_m \times n_r}$ be such that

$$\begin{pmatrix} A & 0 \\ C_r & 0 \end{pmatrix} + \begin{pmatrix} B \\ 0 \end{pmatrix} K = \begin{pmatrix} A + BK_1 & BK_2 \\ C_r & 0 \end{pmatrix}$$

is Hurwitz (always possible to find such K due to the NON-RESONANCE condition).

Then, for every estimate \hat{u}^* and \hat{x}^* of u^* and x^* (even $\hat{u}^* = 0$, $\hat{x}^* = 0$)

the controller

$$\begin{cases} \dot{\hat{\sigma}} = \tilde{y}_r \\ \mu = \hat{u}^* + K_1(x - \hat{x}^*) + K_2 \hat{\sigma} \end{cases} \quad \left\{ \begin{array}{l} \text{has the form} \\ \dot{\xi} = g(\xi, y_m) \\ \mu = \delta(\xi, y_m) \end{array} \right.$$

is such that there exists $\hat{\sigma}^* \in \mathbb{R}^{n_r}$ s.t. the equilibrium $(x^*, \hat{\sigma}^*)$ is LAS for the closed-loop system.

PROOF.

We can write (see previous classes)

$$\dot{\tilde{x}} = A \tilde{x} + B \tilde{u} + f_{\text{hor}}(x, u) \quad (\tilde{x} = x - x^*, \tilde{u} = u - u^*)$$

$$\tilde{y}_r = C_r \tilde{x} + h_{r, \text{hor}}(x, u)$$

Taking

$$\begin{cases} \mu = \hat{u}^* + K_1(x - \hat{x}^*) + K_2 \hat{\sigma} \\ \dot{\hat{\sigma}} = \tilde{y}_r = C_r \tilde{x} + h_{r, \text{hor}}(x, u) \end{cases}$$

leads to

$$\dot{\hat{x}} = A \hat{x} + B \left(\hat{u}^* + K_1 (x - \hat{x}^*) + K_2 \delta - u^* \right) + f_{\text{hor}}(x, u)$$

$$\downarrow$$

$$= A \hat{x} + B K_1 (x - \hat{x}^* - \hat{x}^*) + B K_2 \delta + B (\hat{u}^* - u^*) + f_{\text{hor}}(x, u)$$

$$= (A + B K_1) \tilde{x} + B K_2 \delta + B \left[K_1 (x^* - \hat{x}^*) + \hat{u}^* - u^* \right] + f_{\text{hor}}(x, u)$$

$$\dot{\delta} = C_r \tilde{x} + h_{r, \text{hor}}(x, u)$$

Define

$$\delta^* = K_2^{-1} \cdot \left[u^* - \hat{u}^* - K_1 (x^* - \hat{x}^*) \right] \quad (K_2 \text{ is invertible, see above})$$

and change coordinates

$$\delta \mapsto \tilde{\delta} = \delta - \delta^* \quad (\delta = \tilde{\delta} + \delta^*)$$

Then, we obtain:

$$\dot{\tilde{x}} = (A + B K_1) \tilde{x} + B K_2 \tilde{\delta} + B K_2 \delta^* + B \underbrace{(K_1 (x^* - \hat{x}^*) + \hat{u}^* - u^*)}_{= -K_2 \delta^*} + f_{\text{hor}}(x, u)$$

$$\downarrow$$

$$= (A + B K_1) \tilde{x} + B K_2 \tilde{\delta} + f_{\text{hor}}(x, u)$$

$$\dot{\tilde{\delta}} = \dot{\delta} - \underbrace{\dot{\delta}^*}_0 = \dot{\delta} = C_r \tilde{x} + h_{r, \text{hor}}(x, u)$$

Grouping the equations yields:

$$\begin{pmatrix} \dot{\tilde{x}} \\ \dot{\tilde{\delta}} \end{pmatrix} = \underbrace{\begin{pmatrix} A + B K_1 & B K_2 \\ C_r & 0 \end{pmatrix}}_{\text{Hurwitz linear part}} \underbrace{\begin{pmatrix} \tilde{x} \\ \tilde{\delta} \end{pmatrix}}_{\text{H.O.T}} + \underbrace{\begin{pmatrix} f_{\text{hor}}(x, u) \\ h_{r, \text{hor}}(x, u) \end{pmatrix}}_{\text{H.O.T}}$$

Hurwitz linear part

H.O.T

\Downarrow

The result follows from Lyapunov's indirect method.

REMARKS

1) The control law works for every \hat{u}^*, \hat{x}^* (also $\hat{u}^* = 0$ and $\hat{x}^* = 0$)

2) The equilibrium \hat{e}^* of \hat{e} is

$$\hat{e}^* = K_2^{-1} \cdot \left[u^* - \hat{u}^* - K_1 (x^* - \hat{x}^*) \right]$$

Thus, repeating the computations of the proof, we get


$$u = \hat{u}^* + K_1 (x - \hat{x}^*) + K_2 \hat{e} \quad (\text{by definition})$$

$$= u^* + K_1 \hat{x}^* + K_2 \hat{e} \quad (\text{using } \hat{e}^*) \quad \leftarrow \text{control law we had implemented if we knew } x^* \text{ and } u^*$$

$\Rightarrow \hat{e}(t)$ converges to the value \hat{e}^* making up for our mistakes in estimating x^* and u^*

\hookrightarrow related to ITERATIVE LEARNING when implemented in discrete time

3) The closed-loop system is

$$\begin{pmatrix} \dot{\hat{x}} \\ \dot{\hat{e}} \end{pmatrix} = \begin{pmatrix} A+BK_1 & BK_1 \\ C_r & 0 \end{pmatrix} \begin{pmatrix} \hat{x} \\ \hat{e} \end{pmatrix} + \begin{pmatrix} f_{\text{hor}}(x, u^* + K_1 \hat{x}^* + K_2 \hat{e}) \\ h_{r,\text{hor}}(x) \end{pmatrix}$$


Compared to the state-feedback law $u = u^* + K_1 \hat{x}$ with perfect knowledge of u^* and x^* , here we have an additional term $K_2 \hat{e}$

\downarrow
Transients linked to \hat{e} (poor knowledge of u^* and x^*) may reduce the domain of attraction

4) The gains K_1 and K_2 may still depend on x^* and u^* as they are tuned on A, B, C_r

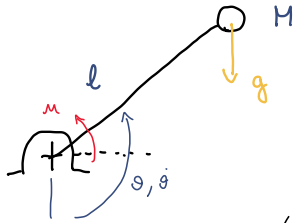
\hookrightarrow Less critical as K_1 and K_2 must only ensure closed-loop stability and:

1) Stability is robust: small uncertainties in A and B are tolerated

2) Often the robustness margins are high

\uparrow
 \nwarrow see examples

EXAMPLE: ACTUATED PENDULUM



GOAL: stabilize the pendulum to a given position $y_r^* = \theta^*$

pendulum equations: $(x_1 = \theta, x_2 = \dot{\theta})$

$$\begin{cases} \dot{x}_1 = x_2 \\ \dot{x}_2 = -\frac{g}{l} \sin x_1 - \frac{b}{\pi l^2} x_2 + \frac{1}{\pi l^2} u \\ y_m = x_1 \\ y_r = x_1 \end{cases}$$

The control law found before (see previous notes) was

$$u(t) = \underbrace{M g l \sin \theta^*}_{u^* \text{ (uncertain)}} + K \left[x(t) - \underbrace{\begin{pmatrix} \theta^* \\ 0 \end{pmatrix}}_{x^*} \right]$$

The robustified controller is (pick $\hat{u}^* = 0$ and $\hat{x}^* = x^*$ as we know θ^*)

$$\begin{cases} \dot{\hat{x}} = x_1 - \theta^* \\ u = K_1 (x - \hat{x}^*) + K_2 \hat{x} \end{cases} \quad (\text{NO EXPLICIT DEPENDENCY FROM } u^*)$$

in which K_1 and K_2 must be such that

$$\begin{pmatrix} A + BK_1 & BK_2 \\ C & 0 \end{pmatrix}, \quad A = \begin{pmatrix} 0 & 1 \\ -\frac{g}{l} \cos \theta^* & -\frac{b}{\pi l^2} \end{pmatrix}, \quad B = \begin{pmatrix} 0 \\ \frac{1}{\pi l^2} \end{pmatrix}, \quad C = (1 \quad 0)$$

is Hurwitz

Let see what values of K_1 and K_2 work:

$$A_{cl} = \begin{pmatrix} A+BK_1 & BK_2 \\ C & 0 \end{pmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ -\frac{g}{\ell} \cos \theta^* + \frac{K_1}{m\ell^2} & -\frac{g}{m\ell^2} + \frac{K_{12}}{m\ell^2} & \frac{K_2}{m\ell^2} \\ 1 & 0 & 0 \end{bmatrix}$$

Let us change bases using

$$T = \begin{bmatrix} 0 & 0 & 1 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \quad T^{-1} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 1 & 0 & 0 \end{bmatrix} \rightarrow \begin{cases} T \begin{pmatrix} x \\ \delta \end{pmatrix} = \begin{pmatrix} \delta \\ x \end{pmatrix} \\ \downarrow \\ \text{we just swapped } \delta \\ \text{and } x \end{cases}$$

Then

$$\begin{aligned} T A_{cl} T^{-1} &= \begin{pmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ \frac{K_2}{m\ell^2} & -\frac{g}{\ell} \cos \theta^* + \frac{K_{11}}{m\ell^2} & \frac{K_{12}-g}{m\ell^2} \end{pmatrix} \\ &= \begin{pmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -\alpha_0 & -\alpha_1 & -\alpha_2 \end{pmatrix} \quad \begin{cases} \alpha_0 = -\frac{K_2}{m\ell^2} \\ \alpha_1 = \frac{g}{\ell} \cos \theta^* - \frac{K_{11}}{m\ell^2} \\ \alpha_2 = \frac{g-K_{12}}{m\ell^2} \end{cases} \end{aligned}$$

Recall from Module 1 that

$$\varphi(A_{cl}) = \alpha_0 + \alpha_1 \lambda + \alpha_2 \lambda^2 + \lambda^3$$

Let's find conditions on K_1 and K_2 ensuring that A_{cl} is Hurwitz in a robust way:

RESULT (ROUTH - HURWITZ CRITERION) the eigenvalues of A_c have all negative real part iff $\alpha_0, \alpha_1, \alpha_2 > 0$ and $\alpha_1 \alpha_2 > \alpha_0$

↳ In our case, these conditions are equivalent to

$$\begin{cases} K_2 < 0 & (I) \\ K_{11} < M g \cos \theta^* & (II) \\ K_{12} < \beta & \leftarrow \text{always true if } K_{12} < 0 \quad (III) \\ \left(\frac{g}{e} \cos \theta^* - \frac{K_{11}}{M e^2} \right) \left(\frac{\beta - K_{12}}{M e^2} \right) > - \frac{K_2}{M e^2} & (IV) \end{cases}$$

RESULT Fix $K_2 < 0$ and $K_{12} < 0$, and then choose $K_{11} < 0$ so that

$$|K_{11}| > M g e + \frac{|K_2|}{|K_{12}|} M e^2 \quad (\text{we do not know } M g e \text{ but we can easily find an upper bound!})$$

Then, (I) - (IV) are satisfied

↓

If $c > 0$ is such that $c > M g e$ then the condition can be substituted by $|K_{11}| > c + \frac{|K_2|}{|K_{12}|}$

PROOF.

Conditions (I) and (III) are obviously true since K_{12} and K_2 have been chosen negative.

To prove (II) note that:

$$|K_{11}| > M g e + \frac{|K_2|}{|K_{12}|} M e^2 \Rightarrow K_{11} = -|K_{11}| < -M g e \leq M g e \cos \theta^* \leftarrow \text{since } \cos \theta^* \geq -1$$

Finally, we have

$$-K_{11} = |K_{11}| > M g e + \frac{|K_2|}{|K_{12}|} M e^2 \geq -M g e \cos \theta^* + \frac{-K_2}{\beta - K_{12}} M e^2 \Rightarrow (IV)$$

↑
since $1 \geq -\cos \theta^*$

↑

Since $\beta - K_{12} \geq -K_{12}$

$\Rightarrow \frac{1}{-K_{12}} \geq \frac{1}{\beta - K_{12}}$

REMARKS

- ① In this case, closed-loop stability can be guaranteed ROBUSTLY with arbitrarily large margin

↳ Indeed it is enough choose $K_{12}, K_2, K_{11} < 0$ such that $|K_{11}|$ is sufficiently large

⇒ The integral control law is ROBUST

- ② The steady-state equilibrium of δ is

$$\delta^* = \frac{1}{K_2} \mu^* = \frac{1}{K_2} M g l \sin \vartheta^*$$

⇒ the integral action $K_2 \delta$ asymptotically gives the ideal feedforward control action needed to keep the pendulum in the right position

→ Such an action basically compensates the gravitational torque

↳ In robotics this relates to the "gravity-compensation" terms

- ③ By defining the "regulation error"

$$e(t) \doteq \vartheta(t) - \vartheta^* \quad (= x_1(t) - \vartheta^*)$$

and noting that $\dot{e}(t) = \dot{\vartheta}(t) = x_2(t)$, the control law reads ($e(t=0)=0$)

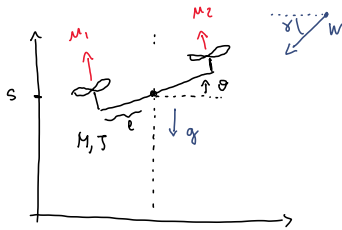
$$\begin{cases} \dot{e}(t) = x_1(t) - \vartheta^* = e(t) \\ \mu(t) = K_1 (x_1(t) - \vartheta^*) + K_2 \delta(t) = K_{11} e(t) + K_{12} \dot{e}(t) + K_2 \delta(t) \end{cases}$$

$$\Rightarrow \mu(t) = K_p e(t) + K_I \int_0^t e(s) ds + K_D \dot{e}(t) \quad \left(K_p \doteq K_{11}, K_I \doteq K_2, K_D \doteq K_{12} \right)$$

which is the equation of a PID controller

EXAMPLE: PLANAR DRONE

MODEL: (see previous notes)



$$\begin{cases} \dot{x}_1 = \frac{1}{M} \cos x_2 \cdot (m_1 + m_2) - g - \frac{W}{M} \sin \gamma - \frac{\mu}{M} x_1 \\ \dot{x}_2 = x_3 \\ \dot{x}_3 = -\frac{\beta}{J} x_3 + \frac{\ell}{J} (m_2 - m_1) \end{cases}$$

G-DAC: drive the vertical velocity $y_r = x_1$ to $y_r^* = 0$

Previous control law (with $x_2^* = 0$)

$$u(t) = u^* + K \left(x(t) - x^* \right) = \begin{pmatrix} \frac{Mg + W \sin \gamma}{2} \\ \frac{Mg + W \sin \gamma}{2} \end{pmatrix} + K x(t)$$

HIGHLY UNCERTAIN

The integral control law is

$$\begin{cases} \dot{e} = x_1 \\ \dot{u} = K_1 x + K_2 e \end{cases}$$

In which K_1 and K_2 are such that the following matrix is Hurwitz

$$A_{cc} = \begin{pmatrix} A+BK_1 & BK_2 \\ C & 0 \end{pmatrix} = \begin{bmatrix} \frac{K_{11}^1 + K_{21}^1 - \mu}{M} & \frac{K_{12}^1 + K_{22}^1}{M} & \frac{K_{13}^1 + K_{23}^1}{M} & \frac{K_1^2 + K_2^2}{M} \\ 0 & 0 & 1 & 0 \\ \frac{\ell}{J} (K_{21}^1 - K_{11}^1) & \frac{\ell}{J} (K_{22}^1 - K_{12}^1) & \frac{\ell}{J} (K_{23}^1 - K_{13}^1) - \frac{\beta}{J} & \frac{\ell}{J} (K_2^2 - K_1^2) \\ 1 & 0 & 0 & 0 \end{bmatrix}$$

$$\text{where } K_1 = \begin{pmatrix} K_{11}^1 & K_{12}^1 & K_{13}^1 \\ K_{21}^1 & K_{22}^1 & K_{23}^1 \end{pmatrix} \quad K_2 = \begin{pmatrix} K_1^2 \\ K_2^2 \end{pmatrix}$$

An example of a robust choice for K_1 and K_2 is

$$K_1 = -\frac{1}{2} \begin{pmatrix} \gamma_1 & -\gamma_2 & -\gamma_3 \\ \gamma_1 & \gamma_2 & \gamma_3 \end{pmatrix}, \quad K_2 = -\frac{1}{2} \begin{pmatrix} \gamma_4 \\ \gamma_4 \end{pmatrix} \quad \text{for any } \gamma_1 > 0, \gamma_2 > 0, \gamma_3 > 0, \gamma_4 > 0$$

Indeed with this choice we have

$$A_{CL} = \begin{bmatrix} -\frac{\gamma_1 + \mu}{M} & 0 & 0 & -\frac{\gamma_4}{M} \\ 0 & 0 & 1 & 0 \\ 0 & -\gamma_2 \frac{\ell}{J} & -\frac{\beta + \gamma_3 \ell}{J} & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix}$$

Transforming A_{CL} with

$$T = \begin{bmatrix} 0 & 0 & 0 & 1 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \quad T^{-1} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 1 & 0 & 0 & 0 \end{bmatrix}$$

yields:

$$T A_{CL} T^{-1} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ -\frac{\gamma_1 + \mu}{M} & 0 & 0 & -\frac{\gamma_4}{M} \\ 0 & 0 & 1 & 0 \\ 0 & -\gamma_2 \frac{\ell}{J} & -\frac{\beta + \gamma_3 \ell}{J} & 0 \end{bmatrix} \quad T^{-1} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -\frac{\gamma_4}{M} & -\frac{\gamma_1 + \mu}{M} & 0 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & -\gamma_2 \frac{\ell}{J} & -\frac{\beta + \gamma_3 \ell}{J} \end{bmatrix}$$

$$\Rightarrow \varphi(A_{CL}) = \left(\lambda^2 + \left(\frac{\gamma_1 + \mu}{M} \right) \lambda + \frac{\gamma_4}{M} \right) \left(\lambda^2 + \left(\frac{\beta + \gamma_3 \ell}{J} \right) \lambda + \gamma_2 \frac{\ell}{J} \right)$$

(Routh-Hurwitz) all eigenvalues have negative real parts since

$$\frac{\gamma_1 + \mu}{M} > 0, \quad \frac{\gamma_4}{M} > 0, \quad \frac{\beta + \gamma_3 \ell}{J} > 0, \quad \gamma_2 \frac{\ell}{J} > 0$$

REMARK. Also in this case the integral control law is ROBUST since

- K_1 and K_2 can be chosen independently from the model's parameters
 - The integral action automatically compensates gravity and wind by producing u^*
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REMARK. An OUTPUT-FEEDBACK solution where

$$y_m = \begin{pmatrix} y_r \\ y_m' \end{pmatrix}$$

and $y_m' \neq x$ can be obtained via the separation principle as before by relying on a Luenberger observer.

