

Can you balance a cup on top of an inverted pendulum?

Laboratory Courseware
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Figure 1: Experimental lab setup, Quanser 2DOF Inverted Pendulum, [3]

1 Forewords

The inverted pendulum is a special, up-side-down positioned (regular) pendulum, see Fig. 1. In most of the case a rigid rod is mounted in a small cart/manipulator via a pivot. Since, the inverted pendulum is an unstable physical system, the task with such system is to balance (move the manipulator) in order to keep the rod in its upright position. The laboratory device is a 2-degree of freedom pendulum where the rod can turn in 2 orthogonal directions. Instead of a cart, this system has two robot arm manipulators, to move the pivot position and keep the rod in its upright position.

This courseware and lab experiment contributes to the understanding of stabilizing systems under uncertainty and implement robust and optimal control theory. Did you know rocket launching and landing or segways are inverted pendulum like problems, see Fig. 2?



Figure 2: Rocket and segway. Source: Wikipedia

This syllabus is supporting students who are taking the course *Robust and nonlinear control (ESS076)* at Chalmers University of Technology, Gothenburg, Sweden and are selecting the 2-DOF Inverted Pendulum as an option for their lab session. The information contained within this paper has two main parts; (i) (mandatory a-priori) **preparation**, (ii) real time **experiment**. Students are supposed to read this paper carefully as well as perform all the exercises before attending the session.

The layout of the labware is the following; on the basis of first principal formalism, a linear time invariant LTI model is derived and analyzed. The nominal model will be topped up by uncertainties that originates from balancing a cup. This LTI model will be used to develop robust and optimal output feedback \mathcal{H}_∞ controller (1 and 2 degree of freedom robust controller structures).

Special thanks to former MPSYS students Yalcin Kalafat and Peixi Gong to create the manual under the supervision of Balazs Kulcsar (Automatic Control Group, Division of SYSCON, Department E2, Chalmers University of Technology).

Aim:

1. Answer the title question
2. Turn theory to practice
3. Implement real-time linear \mathcal{H}_∞ controller design to control unstable and uncertain system.

2 Preparation

The equipment used is a 2-DOF Inverted Pendulum (Quanser) with four-links and two robot manipulators. Part of this labmanual is taken from [3].

As you can see from Figure 3, the servo x (left hand side servo) controls the pendulum in direction x while the servo y (right hand side servo) controls the pendulum in direction y .

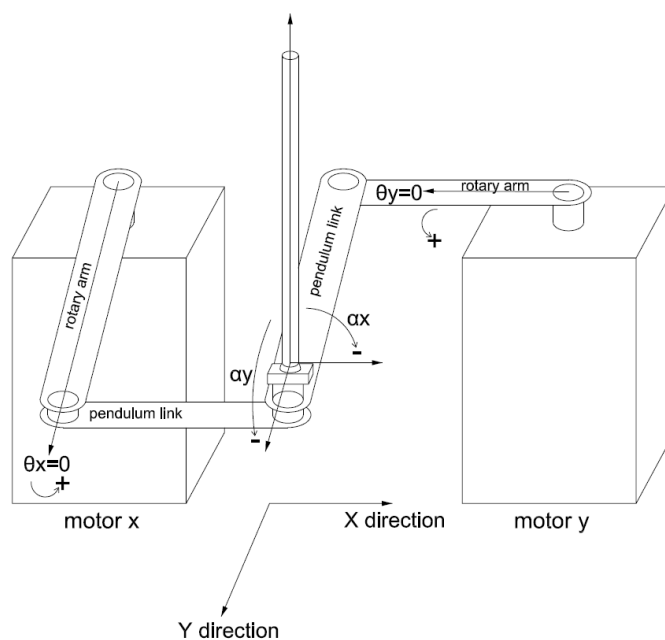


Figure 3: 2-DOF Inverted Pendulum [3].

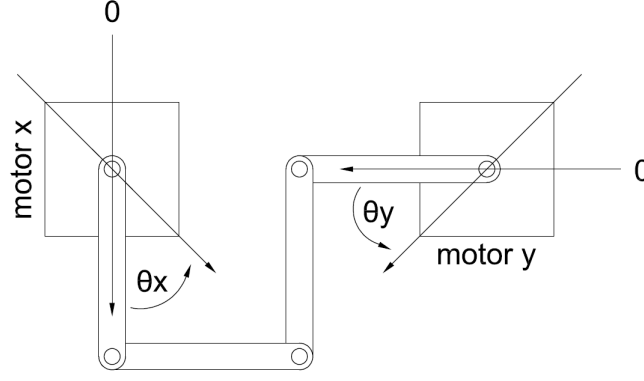


Figure 4: 2-DOF Robot HOME Position [3].

The rod can tilt in direction x and y too. Links that are directly connected to the servos are called rotary arms. The *home position* of the 2-DOF Inverted Pendulum is represented by the schematic draw illustrated in Figure 4. This position is defined as $\theta_x = \theta_y = 0$, and rod angle $\alpha_x = \alpha_y = 0$. Angles increase positively in counter-clockwise (CCW), the servos (and thus the arm) turn CCW when the servo control voltage is positive [3]. The inverted pendulum angle α_x and α_y is zero when it is perfectly in the upright position and increases positively CCW [3]. The home position is selected and used as an operating point for the rest of the studies.

3 First principal pendulum model

If the rod does not deviate much from its home position, the coupling in the dynamics of two directions is negligible. Hence, the 2-DOF Inverted Pendulum can be modelled as the combination of two identical, but 1-DOF Inverted Pendulum (block diagonal). One pendulum moves in x direction while the other in y direction. In the sequel we will use this simplification.

First principal pendulum model can be obtained by finding the equations of motion for robot manipulators with multiple joints and rod displacement.

3.1 Nonlinear Equations of Motion and nominal LTI model

The nonlinear equations of motion for 1-DOF Rotary Pendulum are given

$$\ddot{\theta}(t) = f_3(t) = \frac{-\frac{1}{2}((4M_p L_p^2 \alpha(t) \dot{\theta}(t) \dot{\alpha}(t) - 8C_o V_m(t) + 8D_r \dot{\theta}(t))J_p + M_p^2 L_p^4 \alpha(t) \dot{\theta}(t) \dot{\alpha}(t))}{((4J_r + 4M_p L_r^2)J_p + M_p L_p^2 J_r)} - \frac{-\frac{1}{2}((M_p^2 L_p^3 L_r \dot{\theta}^2(t) + 20M_p^2 L_p^2 L_r g)\alpha(t) - 2M_p L_p^2 D_r \dot{\theta}(t) + 2M_p L_p^2 C_o V_m(t))}{((4J_r + 4M_p L_r^2)J_p + M_p L_p^2 J_r)} \quad (1)$$

$$\ddot{\alpha}(t) = f_4(t) = \frac{((M_p^2 L_p^2 L_r^2 + M_p L_p^2 J_r) \dot{\theta}^2(t) + 20J_r M_p L_p g + 20M_p^2 L_r^2 L_p g)\alpha(t)}{((4J_r + 4M_p L_r^2)J_p + M_p L_p^2 J_r)} + \frac{2M_p L_r L_p C_o V_m(t) - 2M_p L_r L_p D_r \dot{\theta}(t) - M_p^2 L_p^3 L_r \alpha(t) \dot{\theta}(t) \dot{\alpha}(t)}{((4J_r + 4M_p L_r^2)J_p + M_p L_p^2 J_r)} \quad (2)$$

Parameters that are used in the equations are listed in Table 1.

Parameters	Description	Value
M_p	Pendulum mass with T-fitting	0.1270 <i>kg</i>
L_r	Length of rotary arm	0.1270 <i>m</i>
L_p	Length of the pendulum(w/T-fitting)	0.3111 <i>m</i>
J_r	Equivalent inertia with the 4-bar linkage	0.0083 <i>kg.m²</i>
J_p	Pendulum inertia around CoG	0.0012 <i>kg.m²</i>
D_r	Arm viscous damping coefficient	0.0690 <i>N.m.s/rad</i>
C_o	Voltage convert coefficient	0.1285 <i>N.m.s/rad</i>
g	Gravitational constant	0.981 <i>kg.s²</i>

Table 1: Nominal parameter values. Boldfaced parameters are subjected to change and will be considered uncertain.

$V_m(t)$ is the control input signal (voltage to the servo motor). As about the model outputs, there are four encoders that separately measure $\theta_x(t)$, $\theta_y(t)$ and $\alpha_x(t)$, $\alpha_y(t)$. The above nonlinear equation can be linearized around the upright, home position. That position is an operating point for the linearization. Note, all variable are zero at the home position.

3.2 Linear Time-invariant State-space Model

The standard form for a dynamic model, suitable for simulation and controller design, is the state-space form

$$\dot{x}(t) = A_\delta x(t) + B_\delta u(t) \quad (3)$$

and

$$y(t) = C_\delta x(t) + D_\delta u(t), \quad (4)$$

where $x(t)$ is the state, $u(t)$ is the control input, A_δ , B_δ , C_δ , and D_δ are state-space matrices. For 1-DOF Rotary Pendulum, the linear states are defined

$$x(t) = [\delta\theta(t) \ \delta\alpha(t) \ \delta\dot{\theta}(t) \ \delta\dot{\alpha}(t)]^T, \quad (5)$$

where $\delta\cdot$ refers to the fact that small variables change around the known operating point. Only angles can be sensed by encoders, hence, the states $x_1(t)$ and $x_2(t)$ are measured and the outputs of 1-DOF Rotary Pendulum are obtained as

$$y(t) = [\delta\theta(t) \ \delta\alpha(t)]^T = [x_1(t) \ x_2(t)]^T. \quad (6)$$

The control input becomes

$$u(t) = \delta V_m(t). \quad (7)$$

According to eqs.(1) – (2), the system state equations are given by

$$\frac{d\theta(t)}{dt} = \dot{\theta}(t) = f_1(t) \quad (8)$$

$$\frac{d\alpha(t)}{dt} = \dot{\alpha}(t) = f_2(t) \quad (9)$$

$$\frac{d\dot{\theta}(t)}{dt} = f_3(t) \quad (10)$$

$$\frac{d\dot{\alpha}(t)}{dt} = f_4(t) \quad (11)$$

After determining the operating point, linearization results in matrices A_δ and B_δ as

$$A_\delta = \left[\begin{array}{ccc} \frac{\partial f_1}{\partial x_1} & \cdots & \frac{\partial f_1}{\partial x_n} \\ \vdots & \ddots & \vdots \\ \frac{\partial f_n}{\partial x_1} & \cdots & \frac{\partial f_n}{\partial x_n} \end{array} \right] \bigg|_{(\theta=\alpha=\dot{\theta}=\dot{\alpha}=0)}$$

$$B_\delta = \left[\begin{array}{ccc} \frac{\partial f_1}{\partial u_1} & \cdots & \frac{\partial f_1}{\partial u_n} \\ \vdots & \ddots & \vdots \\ \frac{\partial f_n}{\partial u_1} & \cdots & \frac{\partial f_n}{\partial u_n} \end{array} \right] \bigg|_{(\theta=\alpha=\dot{\theta}=\dot{\alpha}=0)}$$

For C_δ and D_δ , the measured states are $\delta\theta(t)$ and $\delta\alpha(t)$, meanwhile, the control input does not influence the measurement, so

$$C_\delta = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix}, \quad D_\delta = \begin{bmatrix} 0 \\ 0 \end{bmatrix}. \quad (12)$$

Exercise 2 Linearize the nonlinear model with the nominal parameter values depicted in Table 1 around the operating point. Find the numerical values for the matrices A_δ and B_δ . Is the LTI system model stable and minimal?

Answer

3.3 Uncertain pendulum models

Create a block diagonal nominal LTI system model as

$$A = \begin{bmatrix} A_\delta & 0 \\ 0 & A_\delta \end{bmatrix}, \quad B = \begin{bmatrix} B_\delta & 0 \\ 0 & B_\delta \end{bmatrix}, \quad C = \begin{bmatrix} C_\delta & 0 \\ 0 & C_\delta \end{bmatrix}, \quad D = \begin{bmatrix} D_\delta & 0 \\ 0 & D_\delta \end{bmatrix}.$$

The inverted pendulum in Figure 1 will be modified with a small platform (cup-holder) and a cup will be added on top too. Therefore, the nominal system model $P_{nom} = (A, B, C, D)$ will be changed

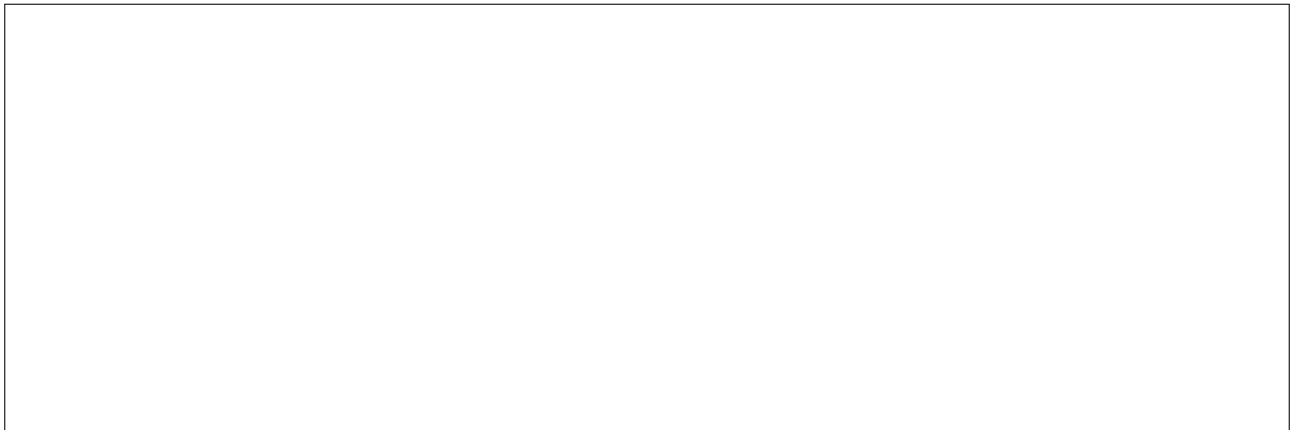
Exercise 3 Expand the nominal LTI model with real parametric uncertainties in $S = (M_p, L_p, J_p, C_o)$ with $(\pm 100\%, \pm 50\%, \pm 100\%, \pm 10\%)$ respectively. These are the parameters that we expect to change once a extra glass is added. Generate 100 samples from and plot their singular values. Which parameter influences low/high frequency behaviour the most?

Answer



Exercise 3 Create an equivalent unstructured uncertainty representation to the previous real parametric uncertainties. We aim at finding a 4th order equivalent input multiplicative uncertainty and related overbound weighting W_{iM} (hint: find a scalar overbound and use it as w_{iM} , $W_{iM} = \begin{bmatrix} w_{iM} & 0.1w_{iM} \\ 0.1w_{iM} & w_{iM} \end{bmatrix}$). Draw the block diagram of the equivalent unstructured uncertainty representation.

Answer



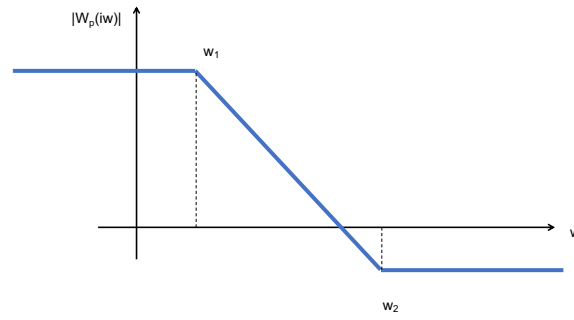


Figure 5: Shape for tracking performance weight W_p in frequency domain.

Exercise 4

Design an 1 Degree of Freedom \mathcal{H}_∞ controller that can balance the extra load (glass). Draw the augmented block diagram first. Use the following performance metrics:

- Input multiplicative weights W_{iM}
- Penalized the input by W_u
- Angle tracking $(y - T_{id}r)$ performance weighted by W_p

and the input weightings as

- W_{id} as the weight to shape the reference signal
- Sensor noise weighting W_n

Answer

Exercise 5

Find the weightings for the above terms with the following conditions.

- The input signal is hard bounded at $\pm 10V$. Try to account for an input voltage normalization to include the previous information.
- In order to weight tracking error in α we use the following shape for the channel-wise weights (see Figure 5). The performance weight should have a magnitude at most $34dB$ in steady state value and $-135dB$ for high frequencies. It will then change within the range $[\omega_1 \ \omega_2] = [1E + 1 \ 9E + 6]$ with slope 20 dB/decade .

- For the θ tracking error performance, we follow the shape of the function (see Figure 5), with magnitudes for low ($44dB$) and high ($-85dB$) frequency, however over the transition between the extremal magnitude values is within the range $[\omega_1 \ \omega_2] = [1E - 2 \ 9E + 6]$ with slope 20 dB/decade.

and the input weightings as

- The ideal response to reference signals (for every channel) is a second order lag with real poles located both at -2 . For the second order low pass filters, their steady state value is always 1.
- Account for 5 degree equivalent noise in all of the measurement channels.

What is the steady state tracking error expected by W_p ?

Answer

Exercise 6

Create the system interconnection for the robust controller design in Matlab. Design a \mathcal{H}_∞ controller. What is the γ value found to be minimal? Is the robust performance condition fulfilled? Check robust stability condition, is that kept? Check the time-domain closed loop simulations with the help of the Simulink file *ExperimentRobustControl*. Is it stabilizing (nominal stability)? What is the settling time?

Answer

Save the controllers state-space matrices as variables ah, bh, ch, dh . **PLEASE BRING ALONG YOUR MATLAB SCRIPT AND SIMULINK FILE TO THE LABSESSION.**

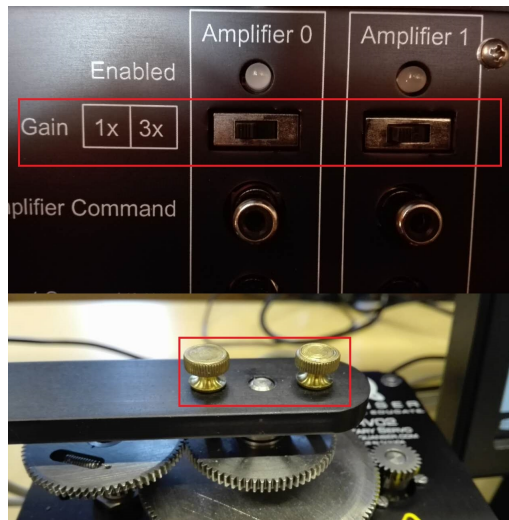


Figure 6: Voltage Amplifier Gain.

4 Experiment

DOS and DONTs

- Be aware, the inverted pendulum device is quite sensitive, work gently with it.
- Put on and take off cup from the top of the pendulum gentle.
- Before you run your code, ask permission to execute it from the teaching assistant!
- Before turning on the voltage amplifier device, ensure the voltage gains of both amplifier ports are set to $1\times$ (Figure 6), or the pendulum device will be damaged!
- If there comes error about '*quarc_comm*' when running the experiment with Simulink, go to *SRV02+2DIP-E* \rightarrow *HIL-Initialize*, set *Board type* to *q8_usb*. If the problem still persists, select QUARC|Set default options.
- After long time running, the 6 thumb screws (Figure 6) on the pendulum might be loose and have some bad impact, ensure the thumb screws are properly adjusted to be tense.

The **ExperimentRobustControl.mdl** in Simulink as shown in Figure 7 is used to balance the 2-DOF Inverted Pendulum to keep the rod in the upright position. The 2-DOF Inverted Pendulum contains **QUARC®** blocks that interface with the DC motor and the angular sensors of the 2-DOF Inverted Pendulum system.

As you can see from Figure 7, the direction x and y are separately controlled by two identical LQG controllers, due to the fact of link coupling around the home position.

Exercise 7

First we do an experiment with an LQG controller that has been designed in class SSY285.

1. Open the script **ESS076setup_experiment.m**. First turn the manual switch that selects the controller to LQG.
2. Load in your \mathcal{H}_∞ controller parameter matrices to the appropriate part in the code (load).
3. Run the script.

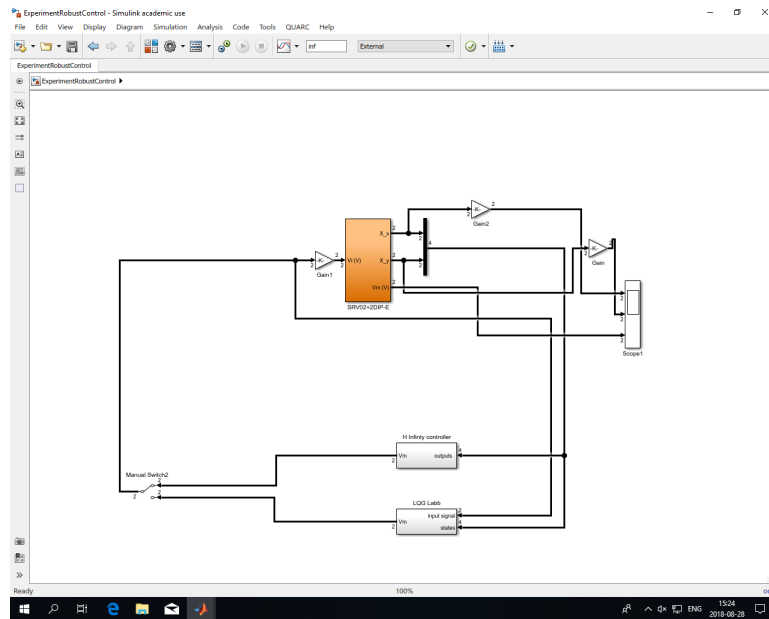


Figure 7: Simulink model used with **QUARC®** to run controller on the 2-DOF Inverted Pendulum.

4. Open Simulink file **ExperimentRobustControl.mdl**.
5. Open the scope.
6. In the Simulink diagram, run QUARC|Build.
7. Manually bring the pendulum to the upright vertical position and hold it by the uppermost tip (home position). Make sure that the pendulum is centered along both axes and is motionless.
8. To start the controller, click on the *Connect to target* button and then the *Start* on the Simulink toolbar (or select QUARC|Start from the menu). Gently release the pendulum once you feel the servo begin to stabilize the pendulum.
9. Inspect the movement of the pendulum, and check the scopes.

Try to gently apply a disturbance on the pendulum. Observe how it reacts. Take a glass located next to the pendulum and try to place on the platform located on the top of the pendulum. Be prepared to catch it in case the LQG controller fails to balance with.

Exercise 8

1. Open the script **ESS076setup_experiment.m**. First turn the manual switch that selects the controller to \mathcal{H}_∞ controller.
2. Load in your \mathcal{H}_∞ controller parameter matrices to the appropriate part in the code (load).
3. Run the script.
4. Open Simulink file **ExperimentRobustControl.mdl**.
5. Open the scope.

6. In the Simulink diagram, run QUARC|Build.
7. Manually bring the pendulum to the upright vertical position and hold it by the uppermost tip (home position). Make sure that the pendulum is centered along both axes and is motionless.
8. To start the controller, click on the *Connect to target* button and then the *Start* on the Simulink toolbar (or select QUARC|Start from the menu). Gently release the pendulum once you feel the servo begin to stabilize the pendulum.
9. Inspect the movement of the pendulum, and check the scopes.

Try to gently apply a disturbance on the pendulum. Observe how it reacts. Take a glass located next to the pendulum and try to place on the platform located on the top of the pendulum. Be prepared to catch it in case of power loss.

Disturbance reaction properties and tracking compensation of the \mathcal{H}_∞ controller differs (α, θ) .

Exercise 9

Change your code and modify the tracking performance requirement in your robust control design. Divide the entire W_p by 2. Redesign your \mathcal{H}_∞ controller. Try out the \mathcal{H}_∞ controller with the pendulum and a cup. What was your observation?

Change your code and modify the tracking performance requirement in your robust control design. Multiply the entire W_p by 2. Redesign your \mathcal{H}_∞ controller. Try out the \mathcal{H}_∞ controller with the pendulum and a cup. What was your observation?

What did we change with the change in the performance requirements?

Exercise 10 (optional)

Change your code and modify the tracking performance requirement in your robust control design. Change the frequency points of W_p shift upwards and downwards in ω for θ . Redesign your \mathcal{H}_∞ controller. Try out the \mathcal{H}_∞ controller with the pendulum and a cup. What was your observation?

What did we change with the change in the performance requirements?

References

- [1] O Wigström, *3-DOF Helicopter Laboratory Session*, laboratory courseware and manual, Chalmers University of Technology, Göteborg, Sweden, 2014.
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- [3] J Apkarian, H Lacheray, M Lévis, *LABORATORY GUIDE: 2 DOF Inverted Pendulum Experiment for MATLAB®/Simulink® Users*, Quanser, 2012.
- [4] B Kulcsar, *Lecture notes for Linear Control System Design, SSY285*, Chalmers University of Technology, Göteborg, 2013-16.