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Aerospace Mechanics 1 (Orbits)

3613 – 102

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Homework #1

1. Completed Reading
2. Scored 10/10 on syllabus quiz
3. A spherical planet, “Planet X” has an equatorial radius R_x of $5R_{\oplus}$ and a density ρ_x of $5\rho_{\oplus}$. The necessary equations are the scalar form of Newton’s Universal Law of Gravitation, $F_G = G \frac{m_1 m_2}{r^2}$, Where m is mass, r is the distance between the COM (center of mass) of the two bodies, F_G is the force of gravity, and G is the universal gravitational constant, $G \approx 6.6743 \cdot 10^{-11} \frac{m^3}{kg \cdot s^2}$, and the scalar form of Newton’s Second Law, $F = m \cdot a$, where a is acceleration.
 - a. When the Falcon is 1 AU away from the center of the planet the gravitational force F_G is given to be 2.5 N. Find the mass of the Falcon in metric tons and US tons.
 - i. To determine m_{falcon} , units are converted to be consistent. 1 AU is approximately $1.496 \cdot 10^{11}$ meters, so $r \approx 1.496 \cdot 10^{11}$ meters. The remaining values are already in consistent units.
 - ii. Newton’s Universal Law of Gravitation is rewritten using non-generic variables, $F_G = G \frac{m_x m_{falcon}}{r^2}$, and is rearranged to solve for m_{falcon} :
$$\frac{r^2 \cdot F_G}{G \cdot m_x} = m_{falcon}.$$
 - iii. m_x (mass of Planet X) must be found to calculate m_{falcon} .
 - iv. $m_x = \rho_x \cdot V_x$ where V_x is the volume of planet X and $\rho_x = 5\rho_{\oplus}$. Since Planet X is a perfect sphere, $V_x = \frac{4}{3}\pi R_x^3$, where $R_x = 5R_{\oplus}$. By substituting these values for ρ_x and R_x , the new equation is as follows:
$$m_x = 5\rho_{\oplus} \cdot V_x = \frac{4}{3}\pi(5R_{\oplus})^3.$$
 The equatorial radius of planet Earth, R_{\oplus} , is

approximately $6.371 \cdot 10^6$ meters, and the density of planet Mercury ρ_{Mercury} is approximately $5429 \frac{\text{kg}}{\text{m}^3}$. Using these values, $m_x \approx 3.676 \cdot 10^{27}$ kg.

- v. Since all values other than m_{falcon} have been defined, the rearranged gravitational equation can now be solved to get $m_{\text{falcon}} \approx 2.281 \cdot 10^5$ kg.
- vi. To find the mass of the Falcon in metric tons and US tons the conversion factors $1 \text{ metric ton} = 1000 \text{ kg}$ and $1 \text{ metric ton} \approx 1.102 \text{ US tons}$ are used to get that **the Millennium Falcon is approximately**

228.08 metric tons, or approximately 251.41 US tons.

- b. The Falcon's thrusters fail, and the craft begins to drift towards the planet, at one point recording a gravitational acceleration, a_G , of $0.1g$; where g is the acceleration due to gravity at sea level on planet Earth. Find the distance of the Falcon to the center of Planet X in kilometers (assume the Falcon is a point mass).
 - i. To find the distance between the COMs of Planet X and the Falcon, both Newton's Universal Law of Gravitation and Newtons Second Law must be used.
 - ii. Since a_G is given to be $0.1g$, and g is known to be approximately $9.807 \frac{\text{m}}{\text{s}^2}$, the force F in Newton's Second Law is the force due to gravity, F_G , and the equation can be written as $F_G = \frac{g \cdot m_{\text{falcon}}}{10}$. This is the same force found in the gravitational equation, so for this instance the second law equation can be substituted in to the gravitational equation as follows:

$$\frac{g \cdot m_{\text{falcon}}}{10} = G \frac{m_x m_{\text{falcon}}}{r^2}.$$
 - iii. The equation is now rearranged to solve for the distance between the COMs of the two bodies: $r = \sqrt{10 \cdot \frac{G \cdot m_{\text{falcon}} \cdot m_x}{m_{\text{falcon}} \cdot g}}$, and further m_{falcon} divides out to get a simplified equation: $r = \sqrt{10 \cdot \frac{G \cdot m_x}{g}}$.
 - iv. All values other than r are known, so the equation can be solved to get $r \approx 5.0014 \cdot 10^8$ meters, which is simply $5.0014 \cdot 10^5$ kilometers, meaning that **the distance between the Millennium Falcon and the center of Planet X is approximately $5.0014 \cdot 10^5$ kilometers.**

- c. Plot the magnitude of the gravitational force between Planet X and the Millennium Falcon as a function of distance ranging from the initial position (1 AU) to the planet's surface (R_x meters from the planet's center).
- Maximum distance is given as 1 AU, or $r_{max} \approx 1.496 \cdot 10^{11}$ meters, and minimum distance is given as the planet's surface, or $r_{min} \approx 3.186 \cdot 10^7$ meters.
 - $r_{min} = R_x$, $R_x = 5R_{\oplus}$, $R_{\oplus} \approx 6.371 \cdot 10^6$ meters, so $R_x \approx 3.186 \cdot 10^7$ meters.
 - r is converted from meters to kilometers to get $r_{max} \approx 1.496 \cdot 10^8$ kilometers, and $r_{min} \approx 3.186 \cdot 10^4$ kilometers.
 - r in Newton's Universal Law of Gravitation equation is now used as an independent variable ranging from r_{max} to r_{min} : $F_G = G \frac{m_x m_{falcon}}{r^2}$, and MatLab is used to plot the result. (figures 1 and 2)

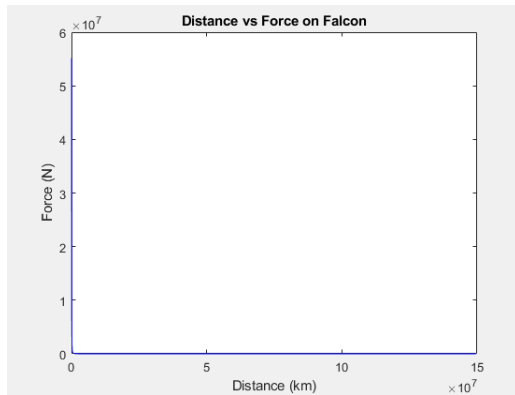


Figure 1 (Basic viewing window)

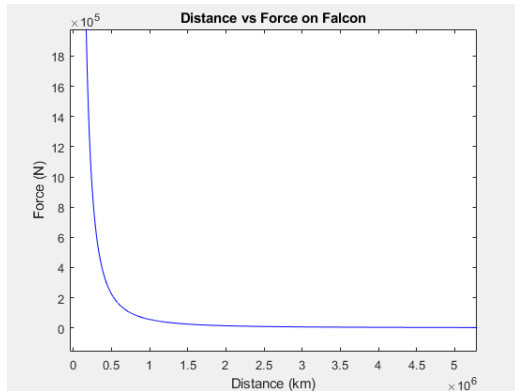


Figure 2 (Zoomed viewing window)

- d. To plot the magnitude of the gravitational force between Planet X and the Millennium Falcon an initial value problem must be set up.
 - i. Distance, r , must be written as a function of time by rewriting Newton's Second Law as $a_G = \frac{F_G}{m_{falcon}}$.
 - ii. Both acceleration, a , and force, F_G , must be taken as functions of time to get $a_G(t) = \frac{F_G(t)}{m_{falcon}}$.
 - iii. Newton's Universal Law of Gravitation is also written with respect to time: $F_G(t) = G \frac{m_x m_{falcon}}{(r(t))^2}$
 - iv. Solve the initial value problem for $r(t)$ using the initial conditions $r(0) = 1 \text{ AU}$ and assume $v(0) = 0 \frac{m}{s}$, and $a(0) = 0 \frac{m}{s^2}$.
 - v. Replace r in the gravitational equation with the equation $r(t)$ and plot the resulting equation, $F_G(t)$, against time.
4. Emailed professor
5. SpaceX's "monopolization" of the space industry and its effects on other launch vehicle companies domestic and abroad.
 - a. SpaceX has arguably dominated the space industry since NASA began contracting most of its missions to them in 2008. This meant that the government primarily supported SpaceX and was very biased in outsourcing their projects directly to them over other companies. While this may have led to SpaceX having something of a monopoly in the space industry, it seems to be for the best considering their numbers and accomplishments to date, although personally I think most of SpaceX's numbers and accomplishments are grossly embellished or could be just as easily achieved by any other semi-decent contractor.
 - b. The Artemis program aims to achieve regular manned missions to the moon as well as to serve as a sort of precursor to the Mars project. Personally, I see no point in these missions other than to say that we as a species are regularly travelling to other celestial bodies. It seems very premature for us to attempt such expensive and challenging missions with such little justification. If I were to change something about the Artemis missions it would be to do away with them altogether and focus on more relevant and cost-effective missions in their stead.
 - c. Considering the complications encountered by Boeing's Starliner I would understand the astronauts' concern with using it as their primary vehicle. If NASA and Boeing conclude that it's perfectly safe to use the vehicle to return to Earth, I would value their opinions over those of the two astronauts, especially with how expensive it would be to send up another craft just so they could feel a bit more comfortable on their journey back. Again, that is assuming NASA and Boeing found the craft to be perfectly safe for the astronauts return, if the craft is less than

safe in any way but NASA decides it's better to save money by using Starliner, the astronauts' lives are certainly more important than saving a bit of money.

Citations:

- Encyclopedia Britannica, "Gravitational Constant," Encyclopedia Britannica. [Online]. Available: <https://www.britannica.com/science/gravitational-constant>. [Accessed: 23-Aug-2024].
- **NASA Jet Propulsion Laboratory (JPL)**, "Astronomical Unit (AU)," CNEOS. [Online]. Available: <https://cneos.jpl.nasa.gov/glossary/au.html>. [Accessed: 23-Aug-2024].
- **NASA's National Space Science Data Center (NSSDC)**, "Earth Fact Sheet," NASA. [Online]. Available: <https://nssdc.gsfc.nasa.gov/planetary/factsheet/earthfact.html>. [Accessed: 23-Aug-2024].
- **NASA's National Space Science Data Center (NSSDC)**, "Mercury Fact Sheet," NASA. [Online]. Available: <https://nssdc.gsfc.nasa.gov/planetary/factsheet/mercuryfact.html>. [Accessed: 23-Aug-2024].
- **Metric Conversions**, "Metric Tons to Short Tons," [Online]. Available: <https://www.metric-conversions.org/weight/metric-tons-to-short-tons.htm>. [Accessed: 23-Aug-2024].

MATLAB output

```
>> hwk01_prob3
```

Problem 3a

Mass of Millennium Falcon: 228.08 metric tons

Mass of Millennium Falcon: 251.41 tons

Problem 3b

Distance of Millennium Falcon from center of planet: 500138.65 km

Problem 3c

```
>>
```

MATLAB Code

```
%-----  
  
disp('Problem 3a');  
  
R = 5*(6.371e+06); % Planet radius  
row = 5*5429; % Planet Density  
r = 1.496e+11; % Distance of falcon to planet mass center  
G = 6.67430e-11; % Universal gravitational constant  
F = 2.5;  
  
% Density to mass  
m_planet = (row)*(4/3)*pi*(R^3);  
  
% Mass of falcon  
m_falcon = (F*(r^2))/(G*m_planet); % kg  
m_falcon_mtons = m_falcon / 1e3; % metric tons  
m_falcon_tons = m_falcon_mtons*1.10231; % tons  
fprintf('Mass of Millennium Falcon: %.2f metric tons \n', m_falcon_mtons);  
fprintf('Mass of Millennium Falcon: %.2f tons \n', m_falcon_tons);  
  
%-----  
  
disp('Problem 3b');  
  
g = 9.807; % m/(s^2)  
a = 0.1*g;
```

```
dist = sqrt((G*m_planet)/a); % Meters
```

```
dist_km = dist/1000; % km
```

```
fprintf('Distance of Millennium Falcon from center of planet: %.2f km \n', dist_km);
```

```
%-----
```

```
disp('Problem 3c');
```

```
num_points = 1e6; % Number of points to plot
```

```
distance = linspace(r, R, num_points); % Meters
```

```
distance_km = distance / 1000; % km
```

```
force = (G * m_planet * m_falcon) ./ (distance.^ 2); % Newtons
```

```
figure(1);
```

```
plot(distance_km, force, 'b');
```

```
xlabel('Distance (km)');
```

```
ylabel('Force (N)');
```

```
title('Distance vs Force on Falcon');
```