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NASA Technical Paper 1757
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Modular Instrumentation System

for Real-Time Measurements and Control

on Reciprocating Engines

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this modular instrumentation system are the real-time computation of parameters and talender a large number of consecutive cycles in

EP and MFBR, other instruments have been developed to provide the following functions angle to sweep the oscilloscope horizontal axis and generate pressure-volume diagrams

100- or

n a computation of the mean and the standard deviation of the sample.

e crankshaft angle where an event such as ignition, injection,

or peak pressure point occurs.

the peak-per-cycle value of a parameter such as cylinder or injection pressure.

Generate the average curve of a parameter of consecutive cycles, such as average pressure or average MFBR curves.

mands of controlled timing and duration to control ignition, injection,

or

ne operation. A modular plug-in concept was adopted to increase the flexibility oped for reciprocating engines. Among the parameters measured are the indicated mean erformed from measured cylinder pressure and crankshaft angle and are available in reagine. Data are displayed in bargraph form and the mean and the standard deviation are

### Introduction

e research program is to develop a computer model for the Otto cycle that will predict the ed primarily to provide input data to generate empirical equations

nd to verify its results. The instrumentation system provides

a real-time measurement

on of key combustion parameters and displays the results in

e (MFBR). IMEP is the work per unit of displacement volume performed during an engine njection) and combustion and the combustion time

een recognized as important indicators of combustion. They have typically been determi

of

hotographs. The reduction of photographed data has not been automated and does no addition, high-speed recording with postrun computer reduction is also being used. Neither

I the parameters are computed and displayed as

ed by the variations in engine operating speed, both steady state and transient. Time-dor different engine operating speeds. Angular position analysis eliminates engine speed vari

a

an absolute position optical shaft encoder directly coupled to the crankshaft. The encode 0.35"

riation in a parameter from one cycle to another under equal engine operating condition in a parameter from one cycle to another under equal engine operating condition module rack contains major power supply components (i.e., transformers, rectifiers, and parameter variations, the digital display is of the continuous-averaging type with a time ry engine cycle a serial digital output is produced by each module for use by the statistic

1

#### **Function Generator**

and the instrumentation systems. This module performs several other operations, as il

2).

This code is desirable in that synchronization between the bits is easily obtained but is not onverts the Gray code to a natural binary code. An extra bit is added to convert the erate an analog signal proportional to the crackshaft angle. This analog signal is used a

8

beed and speed fluctuations. The display is actually 2048 discrete dots, a dot for each in geometry and the crankshaft position. The cylinder volume, as a function of atile read-only memory. In the function generator this memory is used as al, which is used to generate pressure-volume diagram displays. A switch can select when

er unit of time. Reference marks are generated every lo", with larger

e function generator are displaying engine speed and generating reference marks. Eng

2 7200

Bit

ry craokshafl-angle data to-anlog Linear sweep 4 converter nder selection memory Digital- to-analog converter Volume signal

Angle decoder Timing marks

al Engine time Counter base speed display Figure

2. - Block diagram of function generator.

marks at 90" and once per 720" to generate 8

erence points for the oscilloscope displays. The

and 90" marks are derived directly from the binary

ory used to store the volume signal. The sweep, volume signal, and reference marks

om a power supply in the function generator. The function generation module is shown

4

Indicated Mean Effective Pressure cated mean effective pressure can be used as a measure of combustion efficiency. IMEP is a

ton face during a single thermodynamic cycle, normalized by the cylinder displacement.

a

d parameter and has not generally been available as

be readily determined from a pressure-volume (P-V) diagram, as shown in figure

5. The area enclosed by the

ent of the work done by the engine. Area A1 represents negative work, or

rk done on the engine. The algebraic difference

ternatively be computed from an analysis of pressure-crankshaft-angle data. This has

er of consecutive cycles that are not readily available from

P-V diagram analysis. In this instrument (refs.

P is defined as the closed integral of the pressure-volume relationship in the cylinder of

$$= - + Pdv A1$$

VD Crankshaft angle, dq

Figure 3. - Function generator outouts.

Figure 4. - Function generator.

e displacement volume. This integral can be performed in the

8 domain by a change of variables:

v, .H=O

the instrument this integral is approximated by a

summation of finite difference products: where

A0 =720"/n and n= 1024. The term AV/AB is

a function only of engine geometry and 0: m L

In

m a

.&

Intake m L

In

m a

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Intake

П

**TDC BDC** 

Volume

sure -volume diagram. IMEP = Net WorklDisplaced volume -

(A2 - A1)IVp

where

٧

Α

R

r

е

vc volume cylinder area rod length

crankshaft postion (measured from bottom dead center

(BDC))

clearance volume

AV -=A[ -r sin e A0

red in a memory as is the function-generator volume signal. The computation of pressure by the stored function and summing the products. The computation is perform

1024 times during each engine cycle or at

approximately 0.7" increments of 8.

ument is shown in cycle the accumulator value, as an analog signal, is figure and the to attain the high speed required. The pressure signal accumulator is reset for toterval. The during the computation are shown in figure 7. Figure divided with 7(a) is the absolute value, as an analog signal, of an adder-accumulator. At the engine AV/A8. These traces are shown for one 720" engine

Pressure transducer 1 Pressure I,

transducer

lemory - Multiplier - Analog- converter encoder - IMEP accumulator Adder-

AA

Control and timing -

Figure 6. - Block diagram of IMEP instrument.

20 0 360 720 Crankshaft angle, deg (a) Absolute value

b) Absolute value of P(AVVIA9) versus crankshaft angle.

versus crankshafl angle.,

0 350 720 Crankshaft angle,

dq

'VIA9) Ld) Bargraph of 100 consecutive versus crankshaft angle. IMEP values.

10 psildivision. Figure 7.

- Signals during computation.

nter (TDC) of the combustion stroke in the center. The actual value of g the second and fourth quadrants of the trace. The absolute value of

e would be negative during the second and fourth cycles. Figure 7(c) is the running sum reacting on the other side of the piston. Including crankcase pressure would effectivel

7(b) and (c) to indicate negative work during the intake stroke. Figure 7(d) is a bargrap

MEP module but is formatted by the statistical signal processor module. The amplitude

nis sample was taken near the lean limit (equivalence ratio, considerable variation in IMEP: The standard deviation was

22

ne mean. The IMEP data have been verified by extensive

iagram analysis. (Agreement of the data has been

ding at 20 percent of full scale.) The module can be adapted to

emory (a plug-in component) and the calibration setting. The IMEP module is shown i

8.

indication of combustion efficiency, it gives little insight into the combustion process. Eve P(AV/AB)

FBR) will indicate several characteristics of the combustion. The MFBR curve displays

crankshaft angle. The MFBR curve is shown in figure

ition delay is the interval between ignition and the crankshaft angle where

The burn interval is defined as the interval between crankshaft angles where

engine cycle. Although some variation in the total mass burned is expected because o

6 ..~. C-77-583

Figure 8. - IMEP module. Pressure

50 psildivision - I

Crankshaft angle Figure

- Pressure and mass-fraction burn rate curves.

he IMEP instrument and also by using only pressure and crankshaft-angle information.

nced by Rasweiler and Withrow (ref.4). Basically, the pressure rise, or decrease, in

mponents. Piston motion will compress, or expand, the volume and cause

an additional pressure rise if any of the fuel mass was burned during the interval. The

@'@p+APb (7) where

ange during interval APp pressure change due to compression or

ige due to mass burning Let Pi be the actual pressure at the end i-1 be the pressure at the beginning of the interval. We can define ressure Pi' that would be the pressure at the end change was due to expansion or compression. Then it follows that expansion ming a quasi-static adiabatic process, PV? =Constant where

y polytropic gas constant, Cp/C,

Then by backsubstitution or The mass burned during the interval proportional to the preous cylinder volume at the end of the interval. Therefore L\M= k Vi(APb) (14) where k is the constant of proportionality, or

the last term will be abbreviated to A(PV7) in the following discussion.) The mass fraction of the instrument (block diagram of figure ressure duqng each 0 increment (0.35") is multiplied by

ous interval. Any deviation is due to mass burning. The A(PV7) terms are multiplied by

V/Vr ve. The computation is only valid when all valves are closed and is therefore performed

interval about top dead center. The term is stored in a memory and multiplied by P in a multiplying digital-to-analog converter. A

a -45" to 135"

ator (ref.5) determines A(PV7). This is digitally multiplied by nory, and summed. The MFBR curve is then normalized to determine the

d 90-percent points. The delay between ignition and percent mass burned and the interval between

10 and 90

nt mass burned are determined. The MFBR curve

e are not directly accounted for in the analysis but can be compensated for by changes in instrument. Results obtained by using the mass fraction burn rate instrument compar

of

kshaft-angle data. The MFBR instrument is shown in figure

11.

Statistical Signal Processor Module tical signal processor module was designed to provide

a meaningful real-time display of

le in the system typically produces a measurement or computation for every cycle

g rpm, data are produced too rapidly for the operator to comprehend. For example,

3000 rpm,

neasurements or computations are performed by

7 P Pressure transducer

Shaft position encoder Control and timing Figure

10. - Block diagram of MFBR instrument.

u .,

F'

c-77-34

Figure 11 - MFBR instrument.

cond. The digital displays on each module do provide an indication

tical sense, and provide no information as to the cyclic variations in the parameters.

ta from the various modules connected to it. Upon initiation of a sample command, the

in the module, each divided into an upper and lower partition. Thus data can be collecte

the data are recalled from the memory and formatted into a bargraph display for each

t of each bar is proportional to the parameter value for a particular engine cycle. The

of

cycle on the left. These bargraphs are displayed on an oscilloscope.

A block diagram

of the statistical signal processor module is

shown in figure 12.

lso processed by analog circuitry to produce the statistical mean and the standard dev

for the 100-

cle sample. A low-pass filter in Data LOW > pass Mean

in + Memory Data

-b

High pass + True rm

S

Digital- converter Memory

4

address 4

r to-analog 4 output multiplier \* Analog Y-axis output

Α

Oscillator Control logic Figure

2. - Block diagram of statistical display module.

ter is used to compute the standard deviation. The mean and the standard deviation a

channel is displayed on the readouts. However, all the mean and standard deviation valu

of the module rack

the standard deviation the statistical signal processor module also measures the ma

n and the standard deviation by using the same channel selector switch. All the maximu

ary recording. The statistical signal processor module is shown in figure

13.

**Event Detector** 

ed primarily to determine the point at which ignition occurs. However, the design was made so that other events Standard deviation

## **Figure**

tector module has been developed to measure the angle

13. - Statistical signal processor.

parameters such as pressures cross a selectable value from either direction, or, in conjuctation or minimum value. The module accepts analog or digital inputs in the range of its provided on the front panel to set the trigger level. A switch is provided to select whether the first or last crossing is detected. Additionally, a window 180" wide

to

vs parameters such as ignition timing to be determined for any one of the cylinders. The Peak Detector

to determine the maximum or minimum value of an analog input signal. A switch ne peak or valley mode of operation. In the peak detect mode an internal track-hold and cycle. The amplifier is then placed in the track mode and will follow the input signal es into the hold mode and retains the peak value at its output. A status output signal is will again track the input until a new (higher) peak is reached. Operation in the valley most the input until a minimum level is detected. As with the event detector

rement to portions of the 720" cycle. An additional switch position is used to disable the entire cycle can be examined. The status output signal

peak value and the crackshaft-angle where the peak occurred can be measured. The

# **Curve-Averaging Module**

15.

of prime concern. However, because of the cyclic variations in combustion, it is difficulentative curve for a parameter. The curve-averaging module (ref.

2) was developed to provide a

ssive cycles and divides the output by 100 to obtain the average. The average curve is soint approximation to a 720" curve, or it can be used

our- channel mode, averaging two or four separate curves

Event detector module. Figure 15. - Peak detector module.

10 Figure 16. - Curve-averaging module.

le with 1024 or 512 points per curve. The input signal range is

f 10 volts. This module is shown in figure 16.

**Event Generator Module** 

various points in the engine cycle. Switches are provided on the front panel to set the ev

) on the compression stroke and the event duration

n each compression stroke of the engine. This allows for control of engines with up to Concluding Remarks

e the performance of internal combustion engines has been hindered by

6

oustion process. Data from the combustion process have been tedious to obtain of data reduction. This instrumentation system makes available the real-time calculation speed fluctuations. The modular concept allows flexibility in test system configurations.

25, 1980, 505-41.

### References

s' Coordination Meeting, Energy Research and Development Administration, Division of T Indicated Mean-Effective Pressure Instrument:

NASA Tech Brief B76-10542, 1977. 3. Rice, W.

: Indicated Mean-Effective Pressure Instrument.

Gerald M.; and Withrow. Lloyd: Motion Pictures of Engine Flames Correlated with Pressure

SAE

980. (Originally presented Jan. 14, 1938) 5. Birchenough. Arthur

, William J.: Fast Differential Analog-to Digital Conversion.

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## 15. Supplementary Notes

16. Abstract

been developed for reciprocating engines. Among the parameters measured nean effective pressure, or theoretical work per cycle, and the

er. These computations are performed from measured cylinder pressure and cranksha

real

ecutive-cycle sample is analyzed to reduce the effect of cyclic variations in the engine.

and the mean and the standard deviation are computed. Other instruments are also described.

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