



Chapter 1: Transformations and rigid movements

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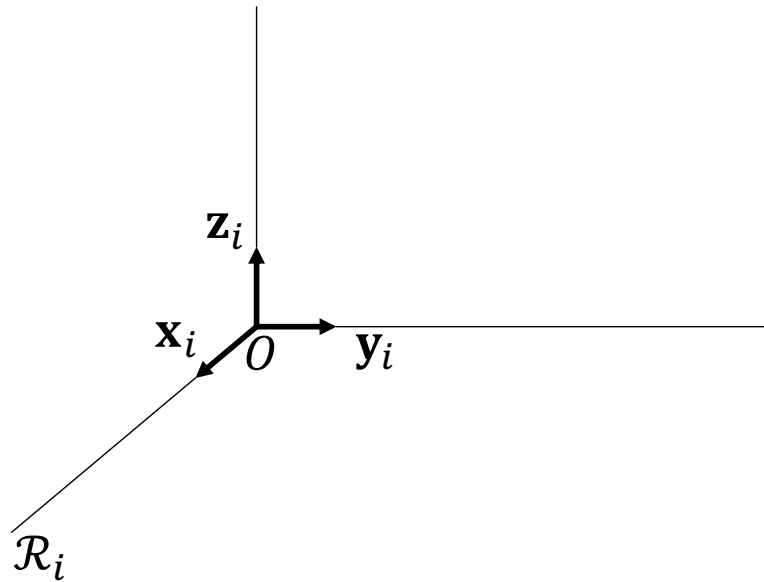
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NOTATIONS AND DEFINITIONS

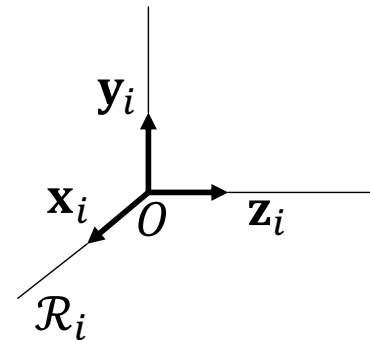
- ❖ Point : lowercase italic letter
- ❖ Vector : lowercase bold letter
- ❖ Null vector: **0**
- ❖ Matrix: bold capital letter
- ❖ Norm of a vector : $\| \ \|$
- ❖ Scalar product : $\mathbf{a} \cdot \mathbf{b}$
- ❖ Cross product: $\mathbf{a} \wedge \mathbf{b}$
- ❖ Pose = position + orientation

NOTATIONS AND DEFINITIONS

Let $\mathcal{R}_i = (O, \mathbf{x}_i, \mathbf{y}_i, \mathbf{z}_i)$ be a direct orthonormal frame

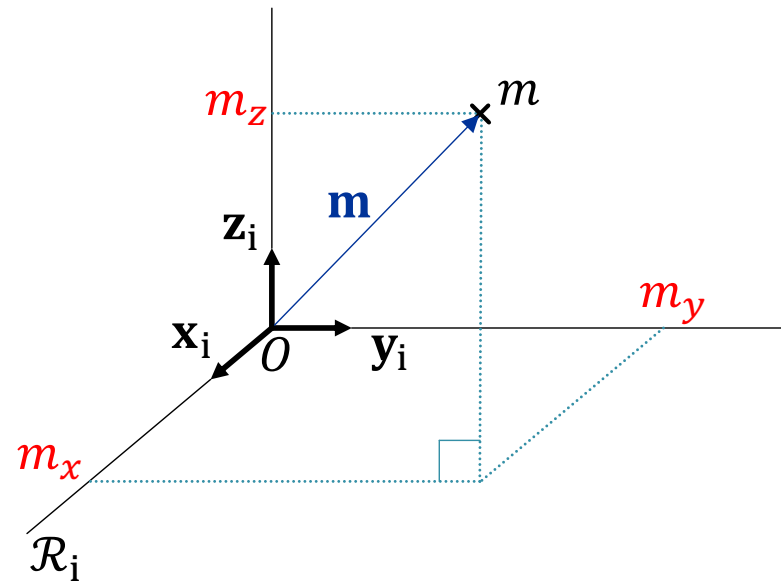


- ❖ Orthogonal frame: $\mathbf{x}_i \perp \mathbf{y}_i \perp \mathbf{z}_i$
- ❖ Normalized frame : $\|\mathbf{x}_i\| = \|\mathbf{y}_i\| = \|\mathbf{z}_i\| = 1$



Indirect frame

POINT REPRESENTATION



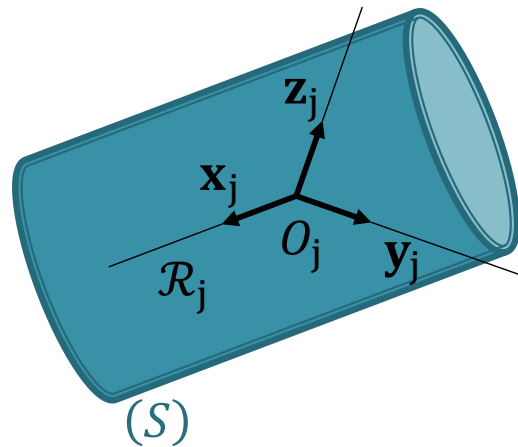
❖ The **coordinates** of a point m are represented by the vector:

$${}^i\mathbf{m} = \begin{pmatrix} m_x \\ m_y \\ m_z \end{pmatrix} \quad \text{to say that the vector } \mathbf{m} \text{ is expressed in the frame } \mathcal{R}_i$$

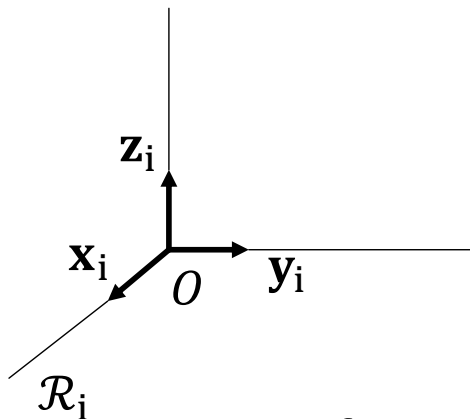
Or:

$${}^i\mathbf{m} = m_x \mathbf{x}_i + m_y \mathbf{y}_i + m_z \mathbf{z}_i$$

SOLID POSE REPRESENTATION



\mathcal{R}_j : frame attached
the solid (S)



\mathcal{R}_i : reference frame

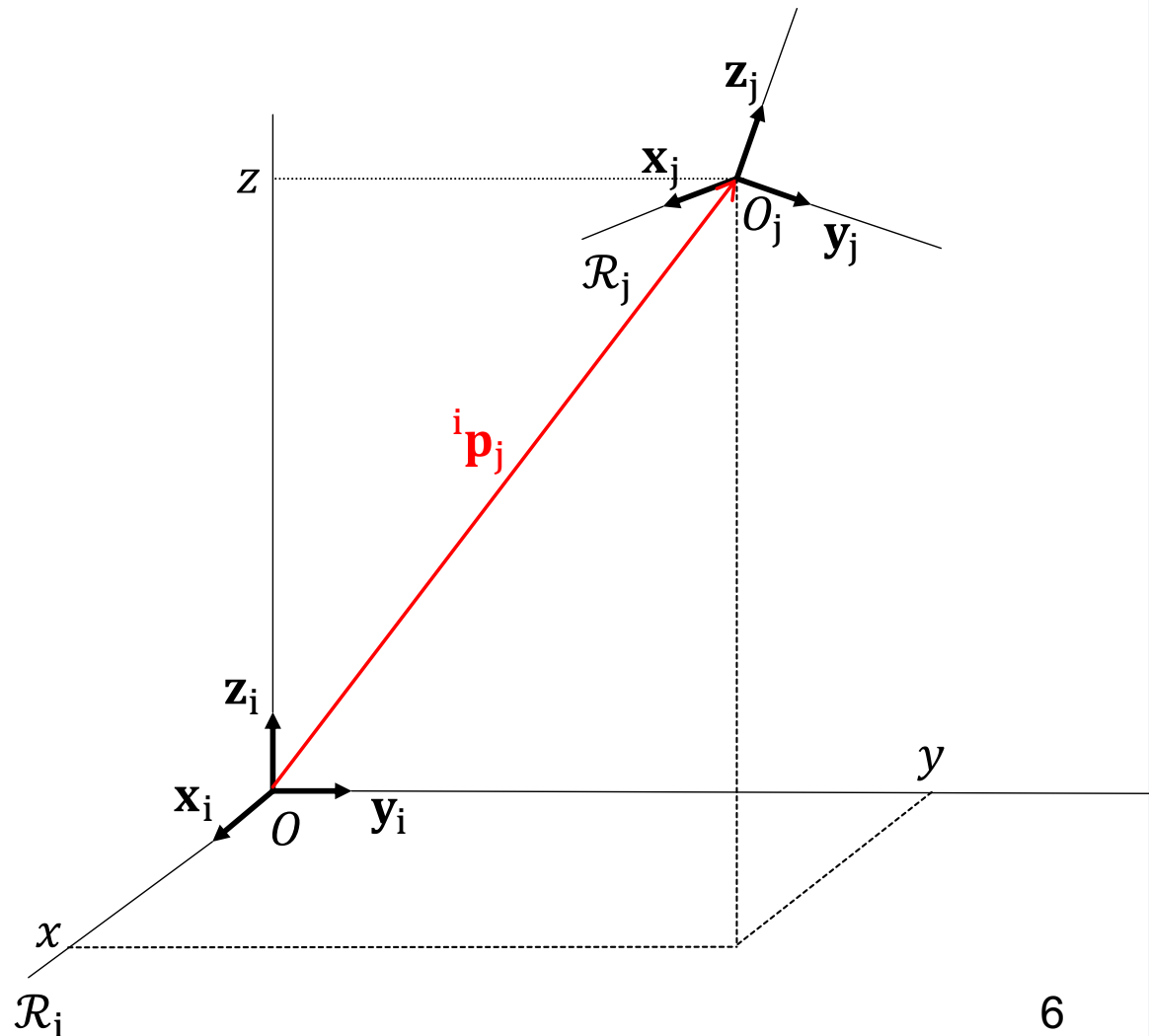
Q: What is the pose
(position + orientation) of
the solid with respect to
the frame \mathcal{R}_i ?

SOLID POSE REPRESENTATION

Position of a solid (S)

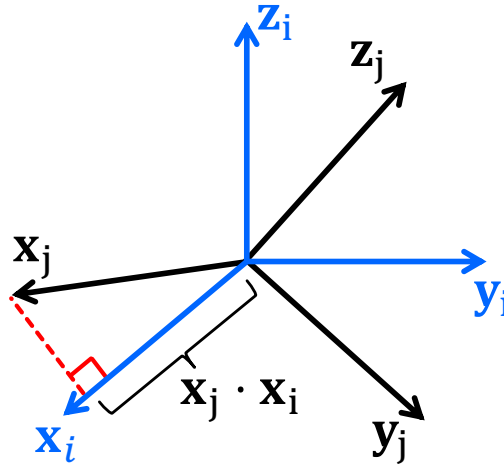
❖ The position \mathcal{R}_j with respect to \mathcal{R}_i is represented by the vector:

$${}^i\mathbf{p}_j = \overrightarrow{OO_j} = \begin{pmatrix} x \\ y \\ z \end{pmatrix}$$



SOLID POSE REPRESENTATION

Orientation of the solid (S)



The vector of the base \mathcal{R}_j can be expressed in the frame \mathcal{R}_i as:

$${}^i\mathbf{x}_j = \begin{pmatrix} \mathbf{x}_j \cdot \mathbf{x}_i \\ \mathbf{x}_j \cdot \mathbf{y}_i \\ \mathbf{x}_j \cdot \mathbf{z}_i \end{pmatrix};$$

$${}^i\mathbf{y}_j = \begin{pmatrix} \mathbf{y}_j \cdot \mathbf{x}_i \\ \mathbf{y}_j \cdot \mathbf{y}_i \\ \mathbf{y}_j \cdot \mathbf{z}_i \end{pmatrix};$$

$${}^i\mathbf{z}_j = \begin{pmatrix} \mathbf{z}_j \cdot \mathbf{x}_i \\ \mathbf{z}_j \cdot \mathbf{y}_i \\ \mathbf{z}_j \cdot \mathbf{z}_i \end{pmatrix}$$

SOLID POSE REPRESENTATION

The orientation of the frame \mathcal{R}_j with respect to \mathcal{R}_i is represented by:

$${}^i\mathbf{R}_j = \begin{pmatrix} {}^i\mathbf{x}_j & {}^i\mathbf{y}_j & {}^i\mathbf{z}_j \end{pmatrix}$$

${}^i\mathbf{R}_j$: 3×3 rotation matrix, expressing the transition from the frame \mathcal{R}_i towards the frame \mathcal{R}_j

Properties:

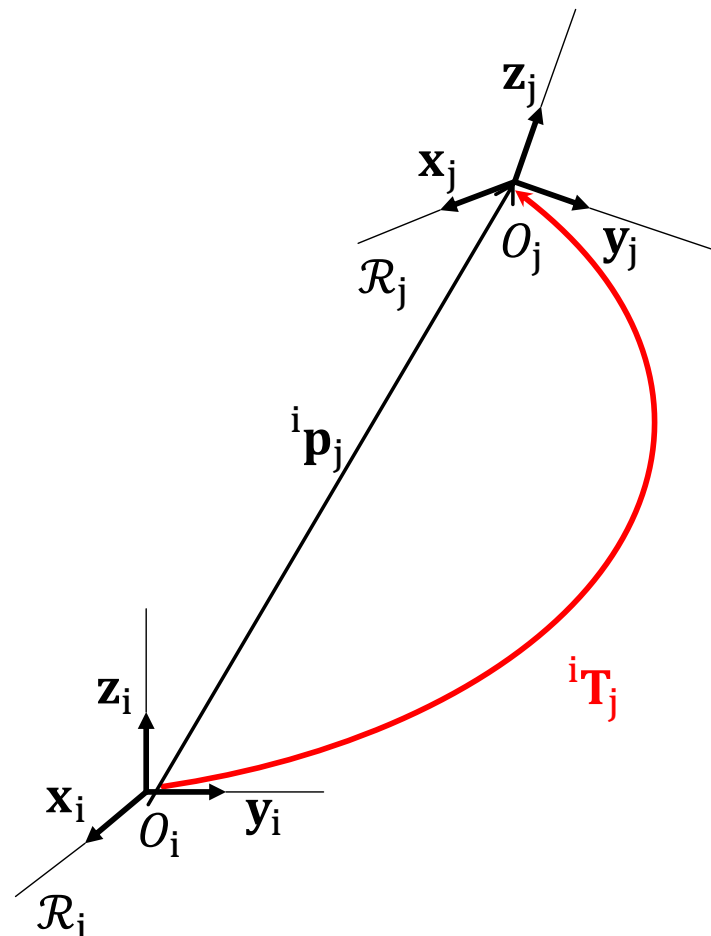
- ❖ $\| {}^i\mathbf{x}_j \| = \| {}^i\mathbf{y}_j \| = \| {}^i\mathbf{z}_j \| = 1$
- ❖ ${}^i\mathbf{x}_j \cdot {}^i\mathbf{y}_j = {}^i\mathbf{x}_j \cdot {}^i\mathbf{z}_j = {}^i\mathbf{y}_j \cdot {}^i\mathbf{z}_j = 0$

The pose of the solid (S), represented by the vector ${}^i\mathbf{p}_j$, and the matrix ${}^i\mathbf{R}_j$ are gathered inside a matrix ${}^i\mathbf{T}_j$, known as homogeneous transformation matrix

HOMOGENEOUS TRANSFORMATION MATRIX

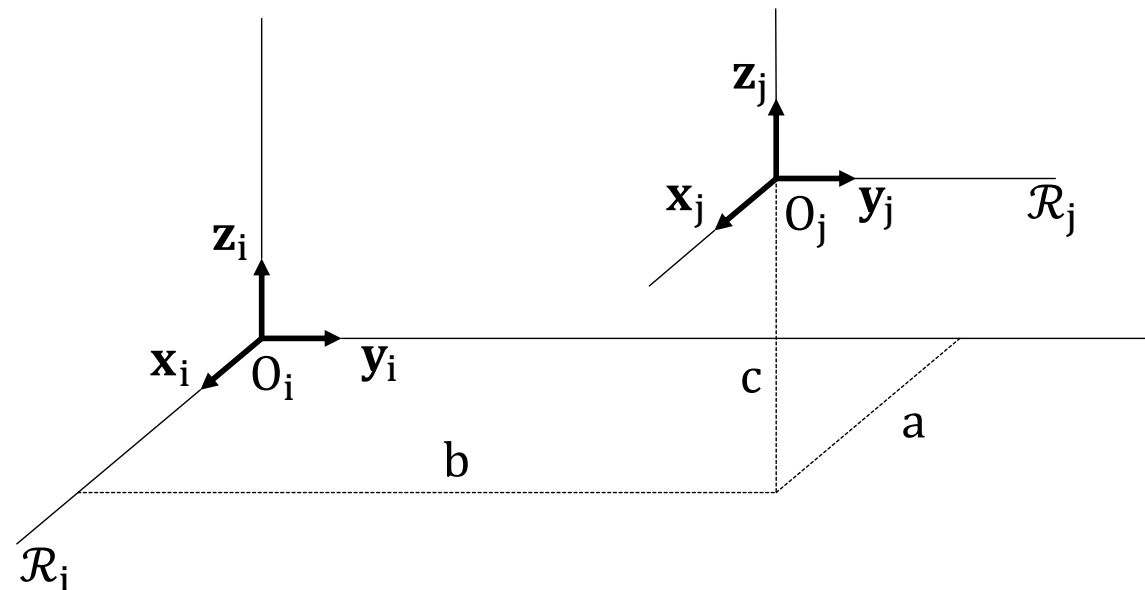
${}^i\mathbf{T}_j$: transition matrix from the frame \mathcal{R}_i towards the frame \mathcal{R}_j

$${}^i\mathbf{T}_j = \begin{pmatrix} & {}^i\mathbf{R}_j & & \\ 0 & 0 & 0 & 1 \end{pmatrix}$$



HOMOGENEOUS TRANSFORMATION MATRIX

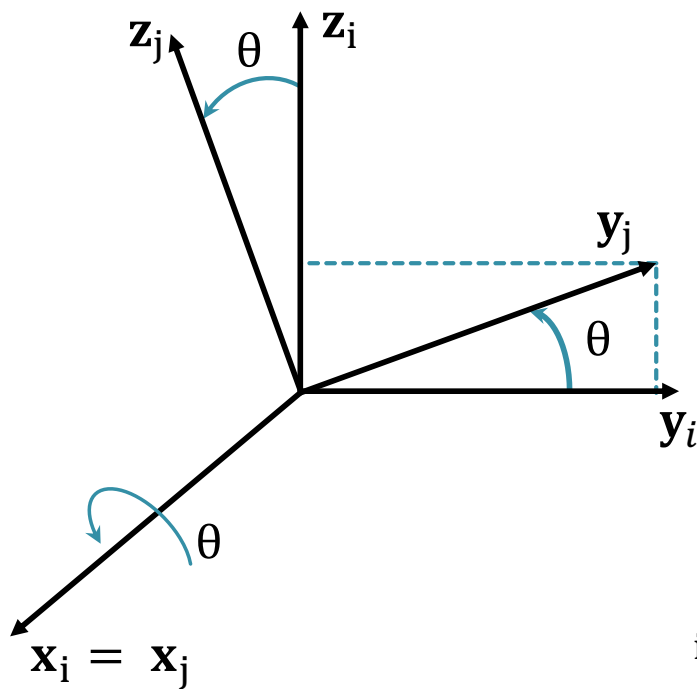
Pure translation



$${}^i\mathbf{T}_j = \mathbf{Trans}(a, b, c) = \begin{pmatrix} 1 & 0 & 0 & a \\ 0 & 1 & 0 & b \\ 0 & 0 & 1 & c \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

HOMOGENEOUS TRANSFORMATION MATRIX

Pure rotation (1)



$${}^i\mathbf{x}_j = \begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix};$$

$${}^i\mathbf{y}_j = \begin{pmatrix} 0 \\ \cos \theta \\ \sin \theta \end{pmatrix};$$

$${}^i\mathbf{z}_j = \begin{pmatrix} 0 \\ -\sin \theta \\ \cos \theta \end{pmatrix}$$

$${}^i\mathbf{T}_j = \mathbf{Rot}(\mathbf{x}, \theta) = \begin{pmatrix} 1 & 0 & 0 & 0 \\ 0 & \cos \theta & -\sin \theta & 0 \\ 0 & \sin \theta & \cos \theta & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

HOMOGENEOUS TRANSFORMATION MATRIX

Pure rotation (2)

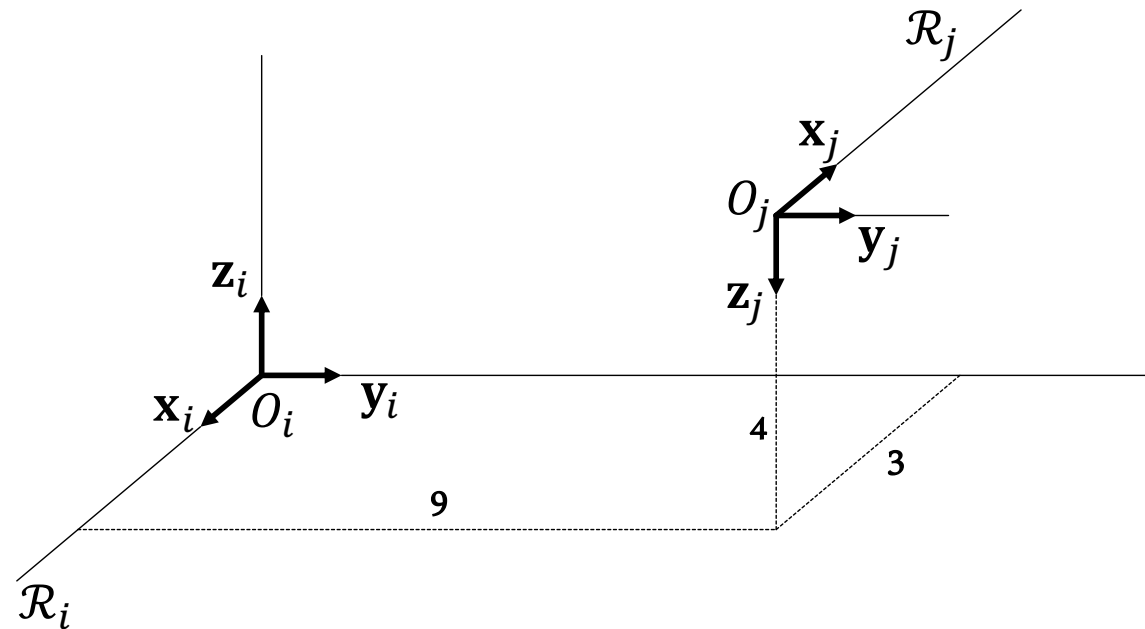
$$\mathbf{Rot}(\mathbf{x}, \theta) = \begin{pmatrix} 1 & 0 & 0 & 0 \\ 0 & \cos \theta & -\sin \theta & 0 \\ 0 & \sin \theta & \cos \theta & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

$$\mathbf{Rot}(\mathbf{y}, \theta) = \begin{pmatrix} \cos \theta & 0 & \sin \theta & 0 \\ 0 & 1 & 0 & 0 \\ -\sin \theta & 0 & \cos \theta & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

$$\mathbf{Rot}(\mathbf{z}, \theta) = \begin{pmatrix} \cos \theta & -\sin \theta & 0 & 0 \\ \sin \theta & \cos \theta & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

HOMOGENEOUS TRANSFORMATION MATRIX

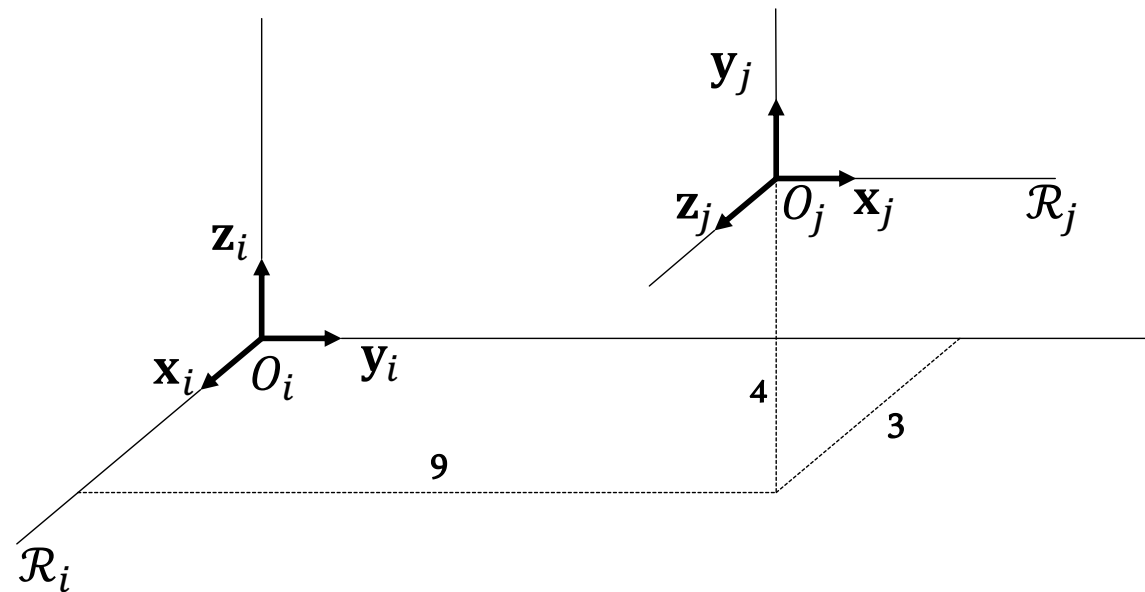
Example 1: compute the homogeneous transformation matrix ${}^i T_j$



Response:

HOMOGENEOUS TRANSFORMATION MATRIX

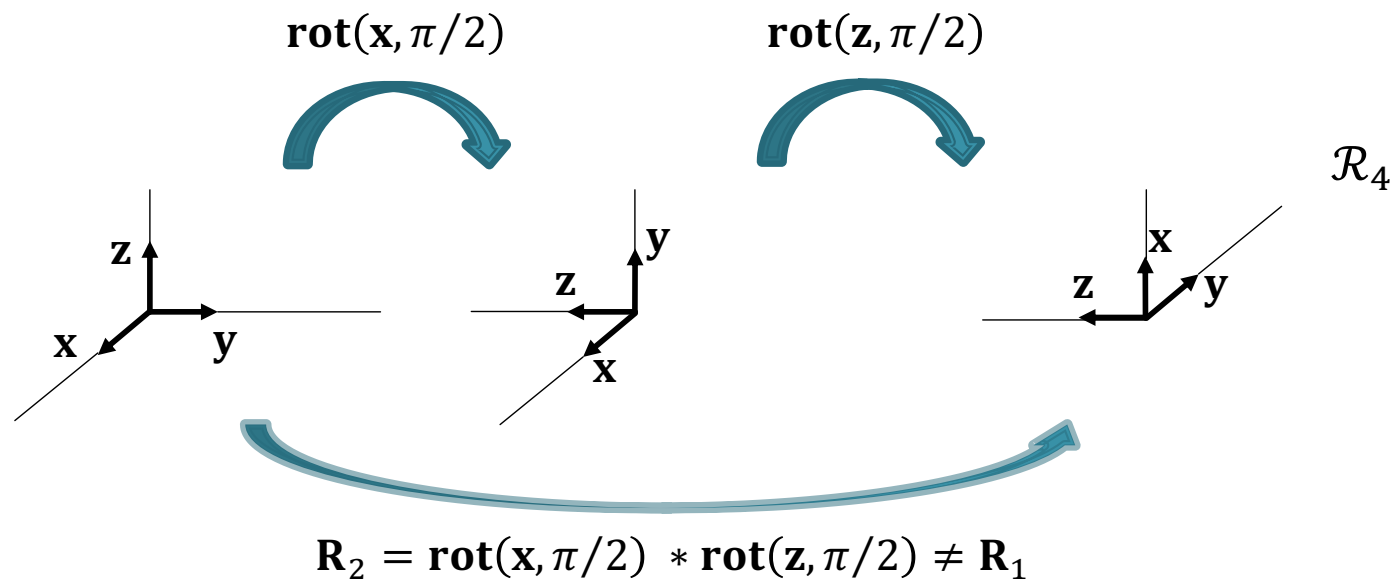
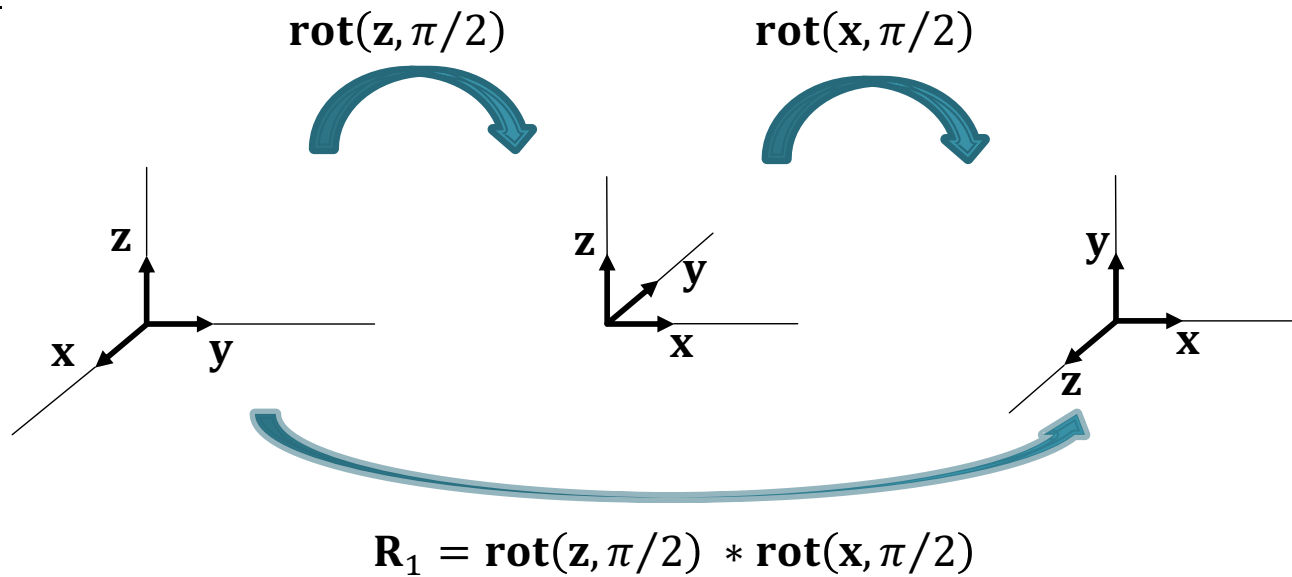
Example 2: compute the homogeneous transformation matrix ${}^i T_j$



Response:

HOMOGENEOUS TRANSFORMATION MATRIX

Remarks:



HOMOGENEOUS TRANSFORMATION MATRIX

Remarks:

$$\text{Let } \mathbf{T}_1 = \begin{pmatrix} \mathbf{R}_1 & \mathbf{p}_1 \\ 0 & 1 \end{pmatrix} \text{ and } \mathbf{T}_2 = \begin{pmatrix} \mathbf{R}_2 & \mathbf{p}_2 \\ 0 & 1 \end{pmatrix}.$$

$$\mathbf{T}_1 \mathbf{T}_2 = \begin{pmatrix} \mathbf{R}_1 \mathbf{R}_2 & \mathbf{R}_1 \mathbf{p}_2 + \mathbf{p}_1 \\ 0 & 1 \end{pmatrix}$$

$$\mathbf{T}_2 \mathbf{T}_1 = \begin{pmatrix} \mathbf{R}_2 \mathbf{R}_1 & \mathbf{R}_2 \mathbf{p}_1 + \mathbf{p}_2 \\ 0 & 1 \end{pmatrix}$$

$$\Rightarrow \mathbf{T}_1 \mathbf{T}_2 \neq \mathbf{T}_2 \mathbf{T}_1$$

HOMOGENEOUS TRANSFORMATION MATRIX

Properties:

$$1) \quad \mathbf{R}^{-1} = \mathbf{R}^T$$

$$2) \quad \mathbf{R}^T \mathbf{R} = \mathbf{I}_3$$

$$3) \quad \det(\mathbf{R}) = 1$$

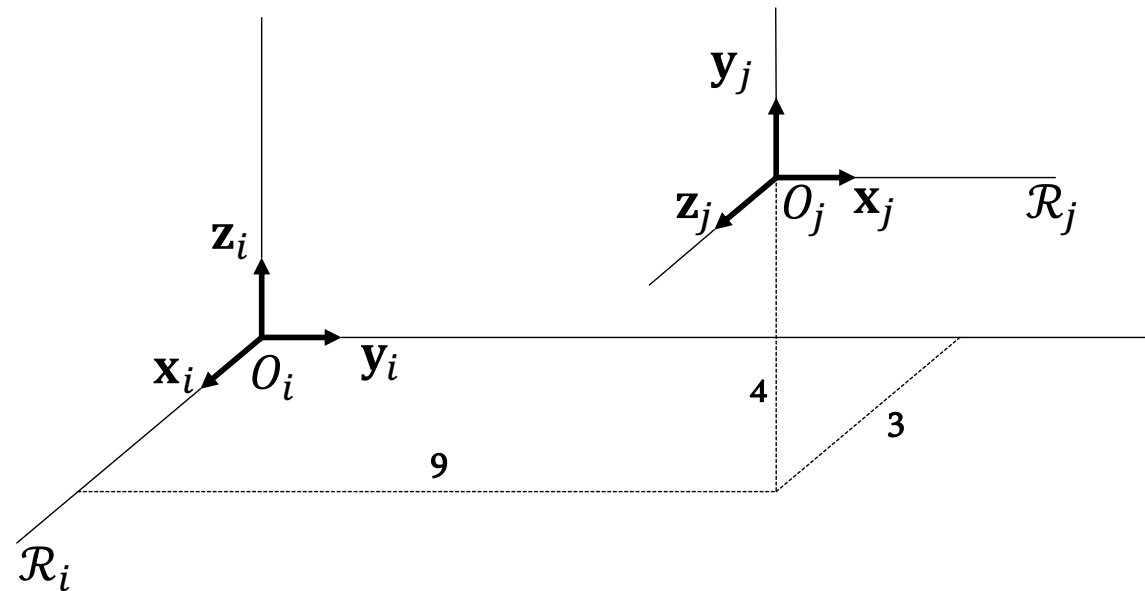
$$4) \quad {}^i\mathbf{T}_j^{-1} = {}^j\mathbf{T}_i$$

$$5) \quad \mathbf{T} = \begin{pmatrix} \mathbf{R} & \mathbf{p} \\ 0 & 1 \end{pmatrix} \Rightarrow \mathbf{T}^{-1} = \begin{pmatrix} \mathbf{R}^T & -\mathbf{R}^T \mathbf{p} \\ 0 & 1 \end{pmatrix}$$

$$6) \quad {}^0\mathbf{T}_k = {}^0\mathbf{T}_1 {}^1\mathbf{T}_2 {}^2\mathbf{T}_3 \dots {}^{k-1}\mathbf{T}_k$$

MATRICES DE TRANSFORMATION HOMOGÈNES

Example 3: from the example 2, compute jT_i using the properties 4 and 5



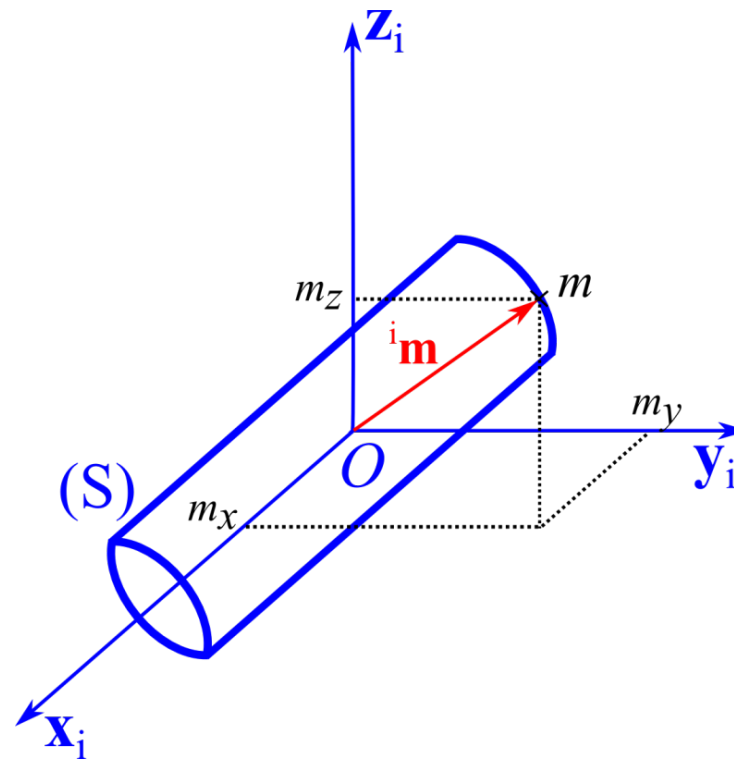
Response:

$${}^jT_i = \begin{pmatrix} 0 & 1 & 0 & -9 \\ 0 & 0 & 1 & -4 \\ 1 & 0 & 0 & -3 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

POINT BELONGING TO A SOLID IN PURE ROTATION

- ❖ Let m be a point belonging to a solid (S). The coordinates of m , expressed in the frame $\mathcal{R}_i(O, \mathbf{x}_i, \mathbf{y}_i, \mathbf{z}_i)$, are represented by a vector:

$${}^i\mathbf{m} = (m_x \quad m_y \quad m_z)^T$$



POINT BELONGING TO A SOLID IN PURE ROTATION

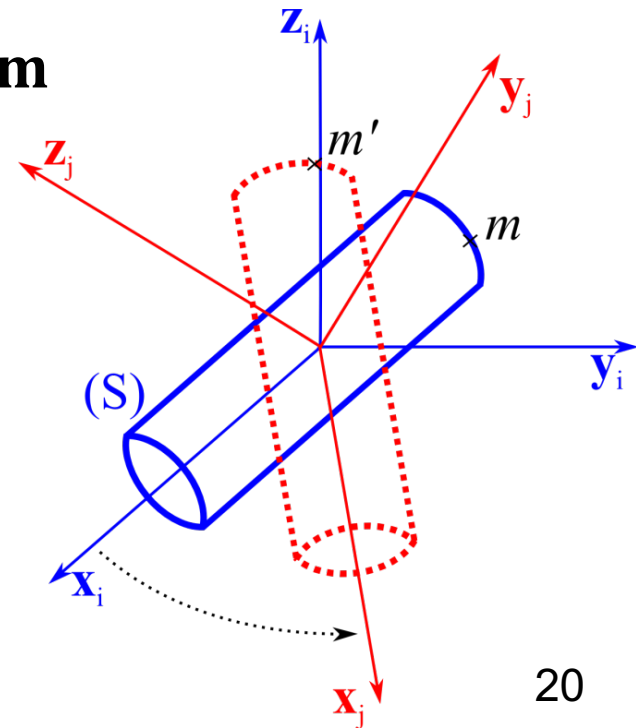
- ❖ After a pure rotation of the solid around the origin O , the coordinates of m' , expressed in the frame $\mathcal{R}_j(O, \mathbf{x}_j, \mathbf{y}_j, \mathbf{z}_j)$, are:

$${}^j\mathbf{m}' = (m_x \quad m_y \quad m_z)^T = {}^i\mathbf{m}$$

- ❖ The coordinates of m' , expressed in the frame $\mathcal{R}_i(O, \mathbf{x}_i, \mathbf{y}_i, \mathbf{z}_i)$, are:

$${}^i\mathbf{m}' = {}^i\mathbf{R}_j {}^j\mathbf{m}' = {}^i\mathbf{R}_j {}^i\mathbf{m}$$

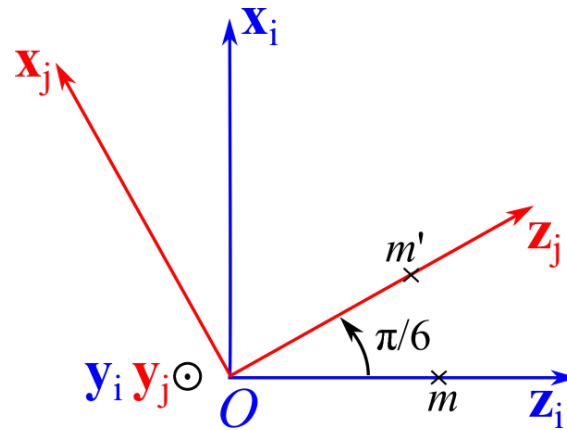
with ${}^i\mathbf{R}_j$ being the rotation matrix $\mathcal{R}_i \rightarrow \mathcal{R}_j$



POINT BELONGING TO A SOLID IN PURE ROTATION

Example 1:

Let m be a point with the coordinates $(0 \ 1 \ \sqrt{3})^T$, expressed in the frame $\mathcal{R}_i(O, \mathbf{x}_i, \mathbf{y}_i, \mathbf{z}_i)$. Compute the coordinates of this transformed point (now point m') after a rotation of the frame \mathcal{R}_i by $\pi/6$ around the axis \mathbf{y}_i .



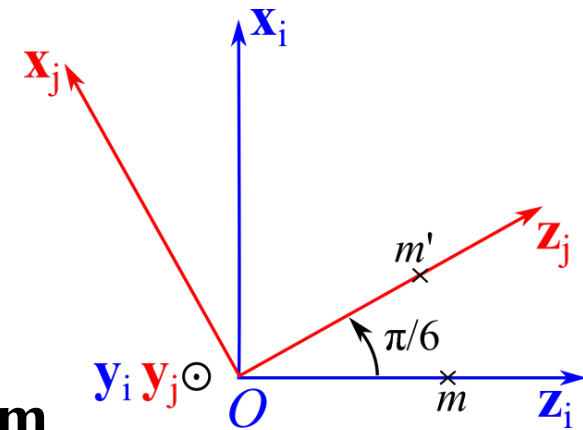
Response:

POINT BELONGING TO A SOLID IN PURE ROTATION

❖ Solution :

$${}^i\mathbf{m} = \begin{pmatrix} 0 \\ 1 \\ \sqrt{3} \end{pmatrix} = {}^j\mathbf{m}'$$

$${}^i\mathbf{m}' = {}^i\mathbf{R}_j {}^j\mathbf{m}' = {}^i\mathbf{R}_j {}^i\mathbf{m}$$



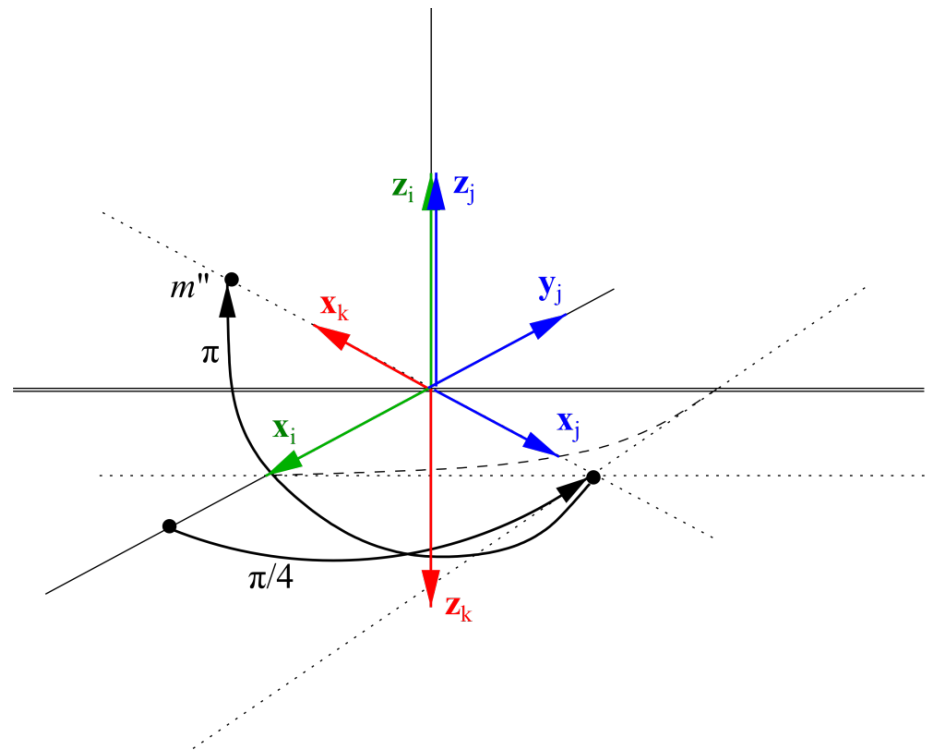
$${}^i\mathbf{R}_j = \begin{pmatrix} \cos \theta & 0 & \sin \theta \\ 0 & 1 & 0 \\ -\sin \theta & 0 & \cos \theta \end{pmatrix} = \begin{pmatrix} \frac{\sqrt{3}}{2} & 0 & \frac{1}{2} \\ 0 & 1 & 0 \\ -\frac{1}{2} & 0 & \frac{\sqrt{3}}{2} \end{pmatrix}$$

$${}^i\mathbf{m}' = {}^i\mathbf{R}_j {}^j\mathbf{m}' = {}^i\mathbf{R}_j {}^i\mathbf{m} = \begin{pmatrix} \frac{\sqrt{3}}{2} & 0 & \frac{1}{2} \\ 0 & 1 & 0 \\ -\frac{1}{2} & 0 & \frac{\sqrt{3}}{2} \end{pmatrix} \begin{pmatrix} 0 \\ 1 \\ \sqrt{3} \end{pmatrix} = \begin{pmatrix} \frac{\sqrt{3}}{2} \\ 1 \\ \frac{3}{2} \end{pmatrix}$$

POINT BELONGING TO A SOLID IN PURE ROTATION

Exemple 2:

Let m'' be a point of coordinates $(\sqrt{2} \ 0 \ 0)^T$, expressed in the frame $\mathcal{R}_k(O, \mathbf{x}_k, \mathbf{y}_k, \mathbf{z}_k)$. Determine the coordinates of the same point in the frame \mathcal{R}_i

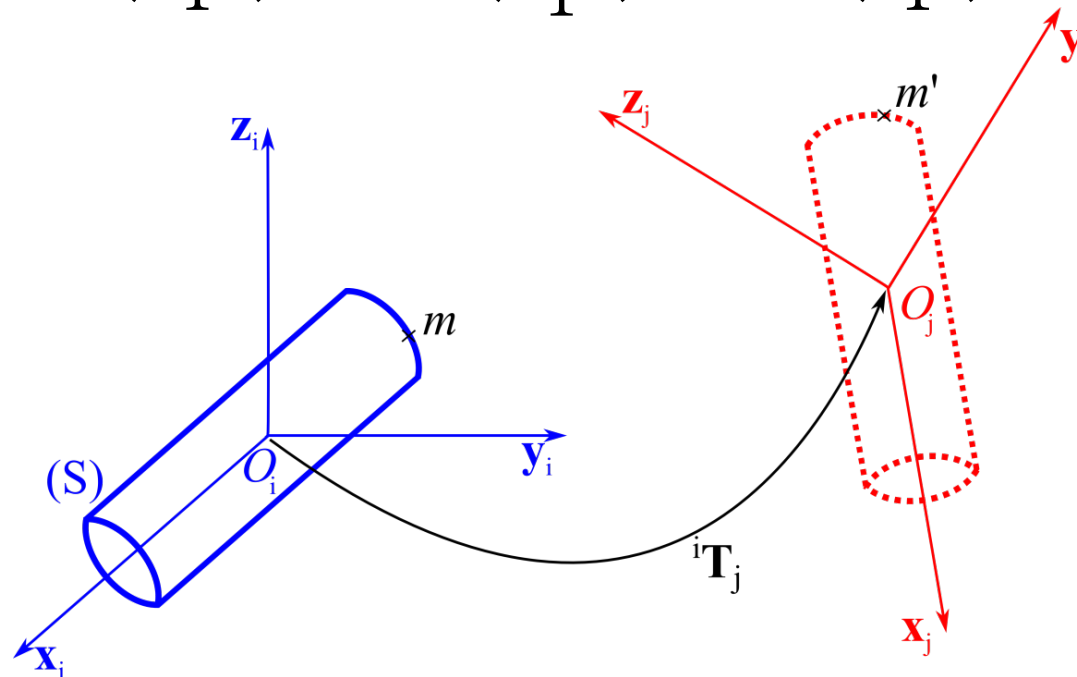


Réponse :

POINT BELONGING TO A SOLID IN A FREE MOVEMENT

- ❖ ${}^i\mathbf{m} = (m_x \ m_y \ m_z)^T$, coordinate of m expressed in \mathcal{R}_i
- ❖ ${}^j\mathbf{m}' = (m'_x \ m'_y \ m'_z)^T$, coordinates of m' expressed in \mathcal{R}_j
- ❖ The coordinates of m' , expressed in the frame \mathcal{R}_i are:

$$\begin{pmatrix} {}^i\mathbf{m}' \\ 1 \end{pmatrix} = {}^i\mathbf{T}_j \begin{pmatrix} {}^j\mathbf{m}' \\ 1 \end{pmatrix} = {}^i\mathbf{T}_j \begin{pmatrix} {}^i\mathbf{m} \\ 1 \end{pmatrix}$$

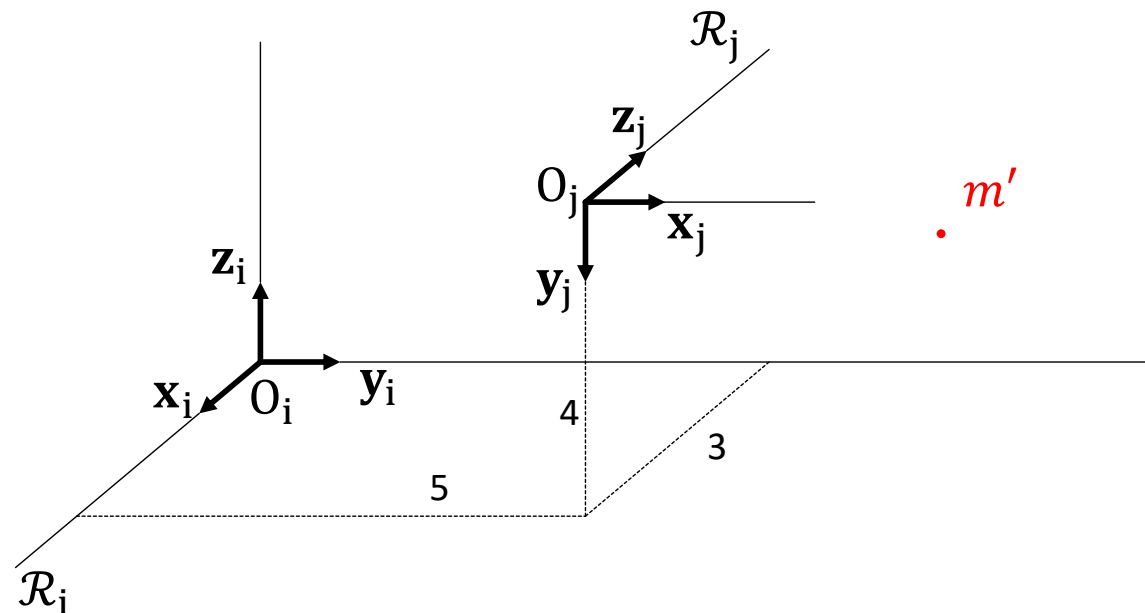


POINT BELONGING TO A SOLID IN A FREE MOVEMENT

Example 3:

Coordinates of m' in \mathcal{R}_j : ${}^j\mathbf{m}' = (\sqrt{3} \quad 4 \quad 5)^T$

Compute the coordinates of m' in \mathcal{R}_i



Response:

POINT BELONGING TO A SOLID IN A FREE MOVEMENT

❖ Solution:

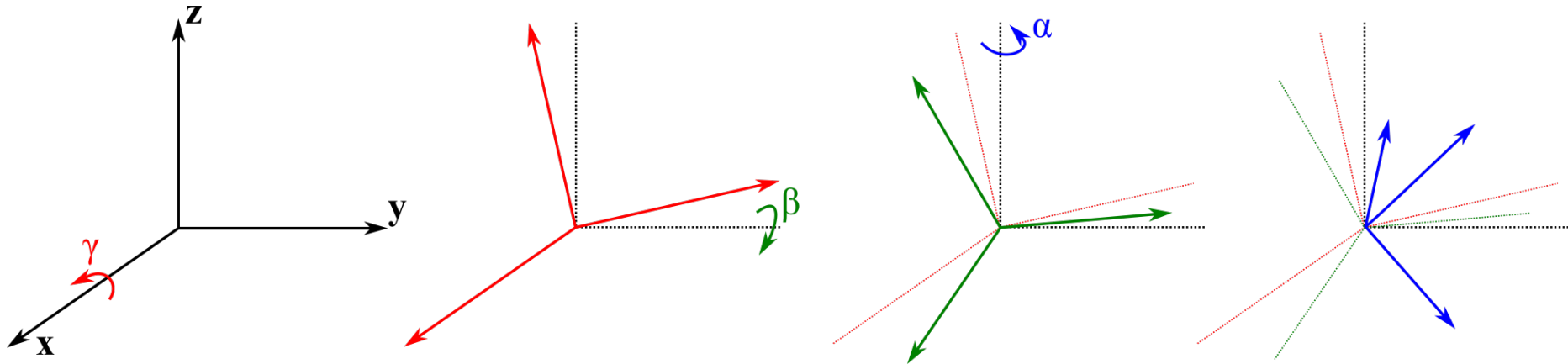
$${}^j\mathbf{m}' = \begin{pmatrix} \sqrt{3} \\ 4 \\ 5 \end{pmatrix}$$
$${}^i\mathbf{m}' = {}^i\mathbf{T}_j {}^j\mathbf{m}'$$

$${}^i\mathbf{T}_j = \begin{pmatrix} 0 & 0 & -1 & 3 \\ 1 & 0 & 0 & 5 \\ 0 & -1 & 0 & 4 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

$${}^i\mathbf{T}_j {}^j\mathbf{m}' = \begin{pmatrix} 0 & 0 & -1 & 3 \\ 1 & 0 & 0 & 5 \\ 0 & -1 & 0 & 4 \\ 0 & 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} \sqrt{3} \\ 4 \\ 5 \\ 1 \end{pmatrix} = \begin{pmatrix} -2 \\ \sqrt{3} + 5 \\ 0 \\ 1 \end{pmatrix}$$

ORIENTATIONS REPRESENTATION

Knowing the Roll (γ) Pitch (β) Yaw (α) angles of a solid, the associated rotation matrix can be expressed as:



$$\mathbf{R} = \mathbf{rot}(\mathbf{z}, \alpha) \mathbf{rot}(\mathbf{y}, \beta) \mathbf{rot}(\mathbf{x}, \gamma)$$

$$= \begin{pmatrix} \cos \alpha & -\sin \alpha & 0 \\ \sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} \cos \beta & 0 & \sin \beta \\ 0 & 1 & 0 \\ -\sin \beta & 0 & \cos \beta \end{pmatrix} \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos \gamma & -\sin \gamma \\ 0 & \sin \gamma & \cos \gamma \end{pmatrix}$$

$$= \begin{pmatrix} \cos \alpha \cos \beta & -\sin \alpha \cos \gamma + \cos \alpha \sin \beta \sin \gamma & \sin \alpha \sin \gamma + \cos \alpha \sin \beta \cos \gamma \\ \sin \alpha \cos \beta & \cos \alpha \cos \gamma + \sin \alpha \sin \beta \sin \gamma & -\cos \alpha \sin \gamma + \sin \alpha \sin \beta \cos \gamma \\ -\sin \beta & \cos \beta \sin \gamma & \cos \beta \cos \gamma \end{pmatrix}$$

ORIENTATIONS REPRESENTATION

Inversely, to determine the Roll Pitch Yaw angles from a given rotation

matrix $\mathbf{R} = \begin{pmatrix} x_x & y_x & z_x \\ x_y & y_y & z_y \\ x_z & y_z & z_z \end{pmatrix}$, one can proceed as follow:

❖ if $x_z \neq \pm 1$

$$\begin{aligned}\alpha &= \text{atan2}(x_y, x_x) \\ \beta &= \text{atan2}\left(-x_z, \sqrt{x_x^2 + x_y^2}\right) \\ \gamma &= \text{atan2}(y_z, z_z)\end{aligned}$$

❖ if $x_z = \pm 1$

$$\alpha - \text{sign}(\beta)\gamma = \text{atan2}(z_y, z_x)$$

α et γ are indeterminate

ORIENTATIONS REPRESENTATION

$$\sin \frac{\pi}{4} = \frac{\sqrt{2}}{2}$$

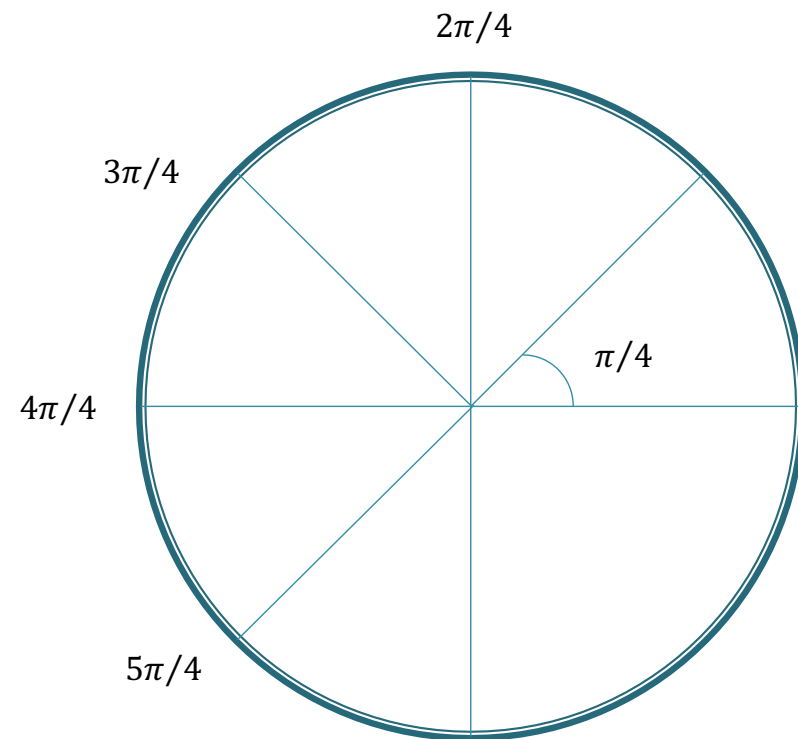
$$\Rightarrow \tan \frac{\pi}{4} = 1$$

$$\cos \frac{\pi}{4} = \frac{\sqrt{2}}{2}$$

$$\sin \frac{5\pi}{4} = -\frac{\sqrt{2}}{2}$$

$$\Rightarrow \tan \frac{5\pi}{4} = 1$$

$$\cos \frac{5\pi}{4} = -\frac{\sqrt{2}}{2}$$



$$\tan^{-1}(1) = \text{atan}(1) = ???$$

ORIENTATIONS REPRESENTATION

$$\text{atan2}(\sin \theta, \cos \theta)$$

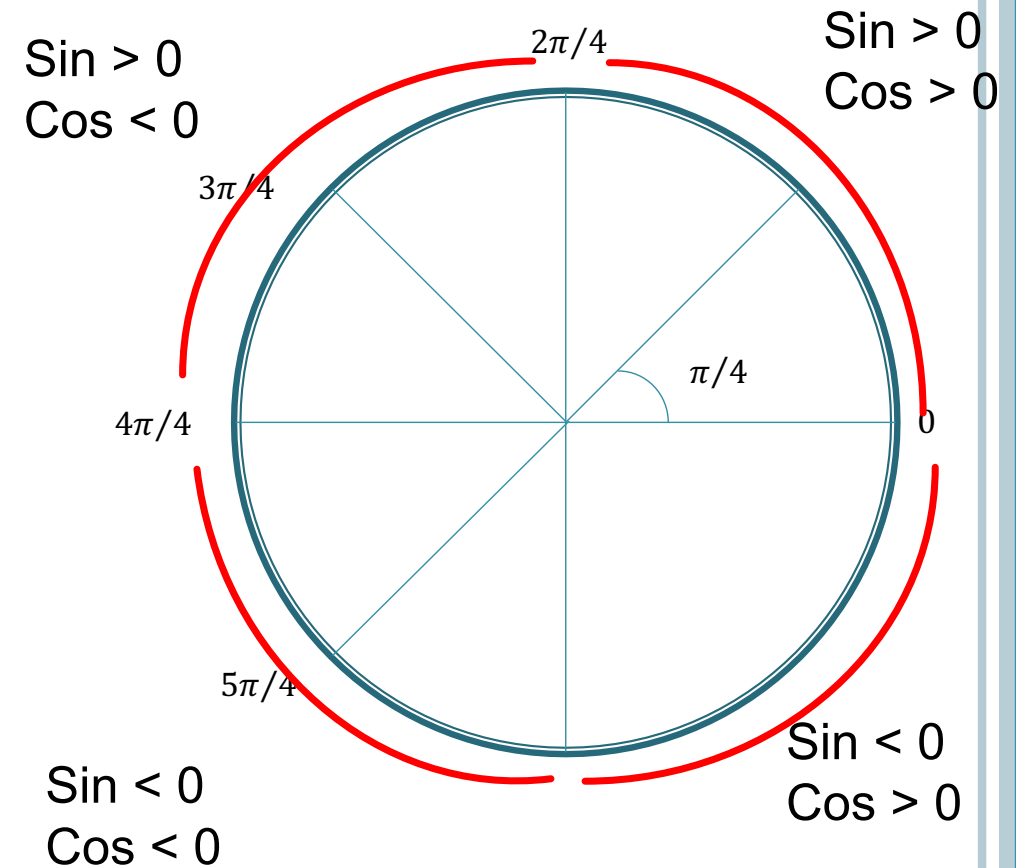
Example :

$$\text{atan2}\left(\frac{\sqrt{2}}{2}, \frac{\sqrt{2}}{2}\right) = \frac{\pi}{4}$$

$$\text{atan2}\left(-\frac{\sqrt{2}}{2}, \frac{\sqrt{2}}{2}\right) = \frac{7\pi}{4}$$

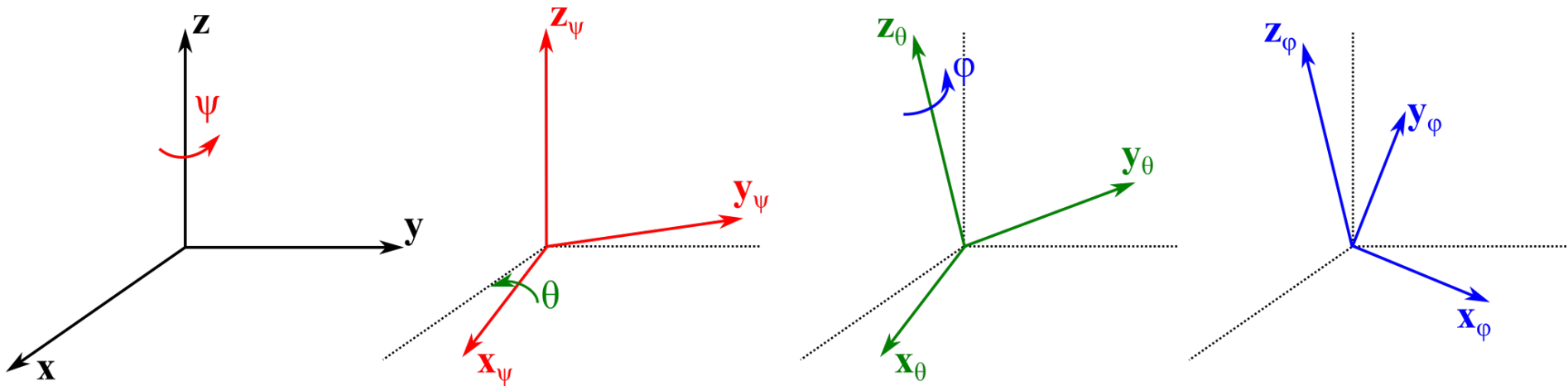
$$\text{atan2}\left(-\frac{\sqrt{2}}{2}, -\frac{\sqrt{2}}{2}\right) = \frac{5\pi}{4}$$

$$\text{atan2}\left(\frac{\sqrt{2}}{2}, -\frac{\sqrt{2}}{2}\right) = \frac{3\pi}{4}$$



ORIENTATIONS REPRESENTATION

Knowing the Euler angles (ψ, θ, φ) angles of a solid, the associated rotation matrix can be expressed as:



$$\mathbf{R} = \text{rot}(\mathbf{z}, \psi) \text{rot}(\mathbf{x}_\psi, \theta) \text{rot}(\mathbf{z}_\theta, \varphi)$$

$$= \begin{pmatrix} \cos \psi & -\sin \psi & 0 \\ \sin \psi & \cos \psi & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos \theta & -\sin \theta \\ 0 & \sin \theta & \cos \theta \end{pmatrix} \begin{pmatrix} \cos \varphi & -\sin \varphi & 0 \\ \sin \varphi & \cos \varphi & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

$$= \begin{pmatrix} \cos \psi \cos \varphi - \sin \psi \cos \theta \sin \varphi & -\cos \psi \sin \varphi - \sin \psi \cos \theta \cos \varphi & \sin \psi \sin \theta \\ \sin \psi \cos \varphi + \cos \psi \cos \theta \sin \varphi & -\sin \psi \sin \varphi + \cos \psi \cos \theta \cos \varphi & -\cos \psi \sin \theta \\ \sin \theta \sin \varphi & \sin \theta \cos \varphi & \cos \theta \end{pmatrix}$$

ORIENTATIONS REPRESENTATION

Inversely, to determine the Euler angles from a given rotation matrix

$$\mathbf{R} = \begin{pmatrix} x_x & y_x & z_x \\ x_y & y_y & z_y \\ x_z & y_z & z_z \end{pmatrix}, \text{ we can proceed as follow:}$$

❖ if $z_z \neq \pm 1$

$$\psi = \text{atan2}(z_x, -z_y)$$

$$\theta = \text{acos}(z_z)$$

$$\varphi = \text{atan2}(x_z, y_z)$$

❖ If $z_z = \pm 1$

$$\theta = \pi(1 - z_z)/2$$

$$\psi + z_z \varphi = \text{atan2}(y_x, x_x)$$

ψ et φ are indeterminate

ORIENTATIONS REPRESENTATION

Example :

Let the rotation matrix be $\mathbf{R} = \begin{pmatrix} \frac{\sqrt{3}}{2} & \frac{-1}{2} & 0 \\ 0 & 0 & -1 \\ \frac{1}{2} & \frac{\sqrt{3}}{2} & 0 \end{pmatrix}$

Compute either the Euler angles and/or the Roll-Pitch-Yaw angles associated to this rotation matrix

$$\alpha = \text{atan2} \left(0, \frac{\sqrt{3}}{2} \right) \text{ impossible}$$

$$\beta = \text{atan2} \left(-1/2, \frac{\sqrt{3}}{2} \right) \text{ impossible}$$

$$\gamma = \text{atan2} \left(\frac{\sqrt{3}}{2}, 0 \right) \text{ impossible}$$

ORIENTATIONS REPRESENTATION

Example :

Let the rotation matrix be $\mathbf{R} = \begin{pmatrix} \frac{\sqrt{3}}{2} & \frac{-1}{2} & 0 \\ 0 & 0 & -1 \\ \frac{1}{2} & \frac{\sqrt{3}}{2} & 0 \end{pmatrix}$

Compute either the Euler angles and/or the Roll-Pitch-Yaw angles associated to this rotation matrix

$$\psi = \text{atan2}(0, 1) = 0$$

$$\theta = \text{acos}(0) = \pm \frac{\pi}{2}$$

$$\varphi = \text{atan2}\left(\frac{1}{2}, \frac{\sqrt{3}}{2}\right) = \frac{\pi}{6}$$

ORIENTATIONS REPRESENTATION

Example :

Let the rotation matrix be $\mathbf{R} = \begin{pmatrix} 0.7071 & -0.3536 & -0.6124 \\ 0.7071 & 0.3536 & 0.6124 \\ 0 & -0.8660 & 0.5 \end{pmatrix}$

Compute either the Euler angles and/or the Roll-Pitch-Yaw angles associated to this rotation matrix

$$\alpha = \text{atan2}(0.7071, 0.7071) = \frac{\pi}{4}$$

$$\beta = \text{atan2}(0, 1) = 0$$

$$\gamma = \text{atan2}\left(-\frac{\sqrt{3}}{2}, \frac{1}{2}\right) = -\frac{\pi}{3}$$

ORIENTATIONS REPRESENTATION

Example :

Let the rotation matrix be $\mathbf{R} = \begin{pmatrix} 0.3536 & -0.3536 & 0.8660 \\ 0.6124 & -0.6124 & -0.5 \\ 0.7071 & 0.7071 & 0 \end{pmatrix}$

Compute either the Euler angles and/or the Roll-Pitch-Yaw angles associated to this rotation matrix

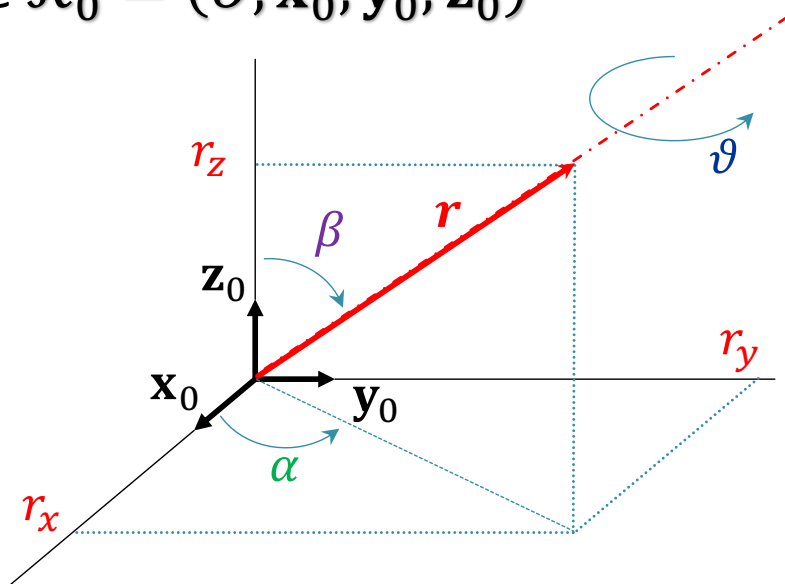
$$\psi = \text{atan2}(0.8660, 0.5) = \frac{\pi}{3}$$

$$\theta = \text{acos}(0) = \frac{\pi}{2}$$

$$\varphi = \text{atan2}(0.7071, 0.7071) = \frac{\pi}{4}$$

ORIENTATION REPRESENTATION: ANGLE AND AXIS

- ❖ Let $\mathbf{r} = [r_x \ r_y \ r_z]^T$ be a unit vector of a rotation axis with respect to the reference frame $\mathcal{R}_0 = (O, \mathbf{x}_0, \mathbf{y}_0, \mathbf{z}_0)$



$$\text{rot}(\mathbf{r}, \vartheta) = \text{rot}(\mathbf{z}, \alpha) \text{rot}(\mathbf{y}, \beta) \text{rot}(\mathbf{z}, \vartheta) \text{rot}(\mathbf{y}, -\beta) \text{rot}(\mathbf{z}, -\alpha)$$

$$\sin \alpha = \frac{r_y}{\sqrt{r_x^2 + r_y^2}}; \quad \cos \alpha = \frac{r_x}{\sqrt{r_x^2 + r_y^2}};$$

$$\sin \beta = \sqrt{r_x^2 + r_y^2}; \quad \cos \beta = r_z$$

ORIENTATION REPRESENTATION: ANGLE AND AXIS

$$\mathbf{rot}(\mathbf{r}, \vartheta) = \begin{pmatrix} r_x^2(1 - C_\vartheta) + C_\vartheta & r_x r_y(1 - C_\vartheta) - r_z S_\vartheta & r_x r_z(1 - C_\vartheta) + r_y S_\vartheta \\ r_x r_y(1 - C_\vartheta) + r_z S_\vartheta & r_y^2(1 - C_\vartheta) + C_\vartheta & r_y r_z(1 - C_\vartheta) - r_x S_\vartheta \\ r_x r_z(1 - C_\vartheta) - r_y S_\vartheta & r_y r_z(1 - C_\vartheta) + r_x S_\vartheta & r_z^2(1 - C_\vartheta) + C_\vartheta \end{pmatrix}$$

The vector \mathbf{r} is constrained by the following relationship:

$$r_x^2 + r_y^2 + r_z^2 = 1$$

ORIENTATION REPRESENTATION: ANGLE AND AXIS

To solve the inverse problem and determine the angle ϑ and the unit vector \mathbf{r} from a given rotation matrix $\mathbf{R} = \begin{pmatrix} x_x & y_x & z_x \\ x_y & y_y & z_y \\ x_z & y_z & z_z \end{pmatrix}$, we can proceed as follow:

$$\vartheta = \cos^{-1} \left(\frac{x_x + y_y + z_z - 1}{2} \right)$$

$$\mathbf{r} = \frac{1}{2 \sin \vartheta} \begin{pmatrix} y_z - z_y \\ z_x - x_z \\ x_y - y_x \end{pmatrix}$$

Important:

If $\sin \vartheta = 0$, the expression of \mathbf{r} becomes meaningless. To solve this problem, it is necessary to refer to particular expressions for $\vartheta = 0$ and $\vartheta = \pi$ (not given in this course).

ORIENTATION REPRESENTATION: UNIT QUATERNION

- ❖ The drawbacks of the axis/angle representation can be overcome by a different four-parameter representation; namely the *unit quaternion*. In other words Euler parameters, defined as $\mathcal{Q} = \{\eta, \epsilon\}$ where :

$$\eta = \cos \frac{\vartheta}{2}$$
$$\epsilon = \sin \frac{\vartheta}{2} \mathbf{r}$$

η is called the scalar part of the quaternion and $\epsilon = [\epsilon_x \quad \epsilon_y \quad \epsilon_z]^T$ is called the vector part of the quaternion. They are constrained by the condition:

$$\eta^2 + \epsilon_x^2 + \epsilon_y^2 + \epsilon_z^2 = 1$$

Hence the name of unit quaternion

ORIENTATION REPRESENTATION: UNIT QUATERNION

❖ The associated rotation matrix is given by:

$$\mathbf{rot}(\eta, \boldsymbol{\epsilon}) = \begin{pmatrix} 2(\eta^2 + \epsilon_x^2) - 1 & 2(\epsilon_x\epsilon_y - \eta\epsilon_z) & 2(\epsilon_x\epsilon_z + \eta\epsilon_y) \\ 2(\epsilon_x\epsilon_y + \eta\epsilon_z) & 2(\eta^2 + \epsilon_y^2) - 1 & 2(\epsilon_y\epsilon_z - \eta\epsilon_x) \\ 2(\epsilon_x\epsilon_z - \eta\epsilon_y) & 2(\epsilon_y\epsilon_z + \eta\epsilon_x) & 2(\eta^2 + \epsilon_z^2) - 1 \end{pmatrix}$$

ORIENTATION REPRESENTATION: UNIT QUATERNION

- ❖ To solve the inverse problem and compute the quaternion from a given rotation matrix $\mathbf{R} = \begin{pmatrix} x_x & y_x & z_x \\ x_y & y_y & z_y \\ x_z & y_z & z_z \end{pmatrix}$, one can proceed as follow:

$$\eta = \frac{1}{2} \sqrt{x_x + y_y + z_z + 1}$$

$$\epsilon = \frac{1}{2} \begin{pmatrix} \text{sign}(y_z - z_y) \sqrt{x_x - y_y - z_z + 1} \\ \text{sign}(z_x - x_z) \sqrt{y_y - z_z - x_x + 1} \\ \text{sign}(x_y - y_x) \sqrt{z_z - x_x - y_y + 1} \end{pmatrix}$$

End of chapter 1