Satellite Geodesy Homework 1

1. Given a satellite's position and velocity at epoch t_0 in the Conventional Inertial System (CIS) as follows:

x = -7856125.132570323 m y = -3153615.476193391 z = -8815677.901873229 $\dot{x} = 2296.291456475990 ms^{-1}$ $\dot{y} = 3944.771956823101$ $\dot{z} = -3449.641080085869$

Please use numerical integration to compute this satellite's positions and velocities over 24 hours, starting from $t_0 = 0^h$ UT on January 1 (initial epoch), using the following GRS80 constants:

$$GM = 3.986005 \times 10^{14} \, m^3 s^{-2}$$
, $a_e = 6378137 m$.

Output your computational results at 1-hr interval in the format of $(t, x, y, z, \dot{x}, \dot{y}, \dot{z})$.

- 2. Refer to Sec. 3.2.2.3, equations (3.129)-(3.131) for the computation of acceleration vector $(\ddot{x}, \ddot{y}, \ddot{z})$ with the central (Keplerian) term and the zonal geopotential coefficient J_2 .
- 3. Refer to Sec. 3.3.2.2 for numerical orbit integration and determination of the *satellite state* $(x, y, z, \dot{x}, \dot{y}, \dot{z})$.
- 4. You can use any *language* and *integration method* of your choice, although *Matlab* and the *Runge Kutta method* are recommended.
- 5. To get the correct answer, you will need to re-calculate the acceleration vector at every integration step (with a step-size h).
- 6. The goals of this homework are (1) to compute the satellite's orbits, and (2) to evaluate the effect of J_2 (Earth's oblateness) on this particular satellite.
- 7. Grading is based on the correctness (partial correct values in next page) and completeness of your report, so please be thorough.

Output example (correct values) in m and m/s

0 h	-7856125. 132570	2296. 291456
	-3153615. 476193	3944. 771957
	-8815677. 901873	-3449. 641080
1 1-	E7EE001 000044	2200 720706
1 h	5755031.696944	3399. 730706
	8754204. 804534	1044. 813888
	-6377968. 551538	4456. 737814
2 h	6750033. 719893	-2966. 160665
	1390833. 072660	-4129. 525473
	10210899. 040342	2522. 921916
3 h	-7005107. 829396	-2819. 934307
	-9017018. 941616	-239. 596487
	4482654. 791563	-4949. 617196
4 h	-5263070. 117739	3559, 071884
	448913. 238516	4208. 814790
	-11020866.890861	-1532.591175

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