

Partitioned Nonlinear Structural Analysis of Wind Turbines using BeamDyn



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Background

- Previous beam model used in FAST (v7.0 and ElastoDyn module of v8.0)
 - Euler-Bernoulli beam model with shortening effect
 - Two degree-of-freedoms (No torsion, shear, and axial DoFs)
 - Assumed-mode method (Two bending modes in one direction)
- Beam models used in other wind turbine tools
 - Multibody-formulation
 - · Linear beam models
 - Constraints introduced between linear beams to describe large deflections and rotations
 - Finite element method
- Partitioned Analysis
 - FAST modularization framework
 - Tight coupling
 - Loose coupling

Objective

- Objective: create efficient high-fidelity beam models for wind turbine blade analysis
 - Based on Geometrically Exact Beam Theory (GEBT)
 - Advanced numerical techniques for implementation
 - Achieve the speed of computational design without significant loss of accuracy comparing to the ultimate accuracy obtained by 3D nonlinear FEA
 - Compatible with the FAST modularization framework
 - Verification and Validation





NWTC static and dynamic blade tests showing typical large, elastic deflections.

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 - Mixed implementation (Yu and Blair, 2012; Wang et al., 2013)
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 - Successfully applied to simulation of fluid dynamics, geophysics, elastodynamics
 - Limited usage in structural dynamics
 - FAST Modularization Framework (Jonkman, 2013; Sprague et al., 2015)
 - Loose coupling
 - Predictor-correction
- BeamDyn, an alternative of ElastoDyn in FAST for blades

Governing Equation

$$\frac{\dot{\underline{h}} - \underline{F'} = \underline{f}}{\dot{\underline{g}} + \dot{\overline{u}}\underline{h} - \underline{M'} - (\widetilde{x}'_0 + \widetilde{u}')\underline{F} = \underline{m}}$$

Constitutive Equation

$$\left\{ \frac{\underline{h}}{\underline{g}} \right\} = \underline{\mathcal{M}} \left\{ \frac{\underline{\dot{u}}}{\underline{\omega}} \right\} \\
 \left\{ \underline{\underline{F}}_{\underline{M}} \right\} = \underline{\underline{\mathcal{C}}} \left\{ \underline{\underline{\epsilon}}_{\underline{K}} \right\}$$

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Strain Measures

$$\left\{ \underline{\underline{\epsilon}} \right\} = \left\{ \underline{\underline{x}}_0' + \underline{\underline{u}}' - (\underline{\underline{R}} \ \underline{\underline{R}}_0) \overline{\iota}_1 \right\}$$

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- Timoshenko-like beam model

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- Geometrically exact: deformed beam geometry is represented exactly

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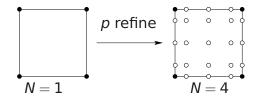
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- $\underline{\underline{M}}$ and $\underline{\underline{C}}$ are 6 × 6 sectional mass and stiffness matrices, respectively
- Elastic couplings are captured
- Timoshenko-like beam model
- Geometrically exact: deformed beam geometry is represented exactly
- Small strains

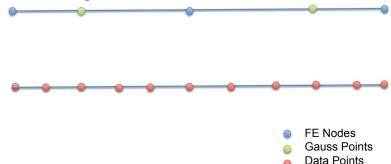
Legendre Spectral Finite Elements

- LSFE methods combine the geometric flexibility of the FE method with the accuracy of global spectral methods.
 - Solution improved through increased basis polynomial order (p-refinement)
 - LSFEs employ Lagrangian interpolant shape functions with nodes at Gauss-Lobatto-Legendre (GLL) points
 - Exponential convergence rates for sufficiently smooth solutions



Numerical Integration

▶ FE Node, Integration Point, and Data Point

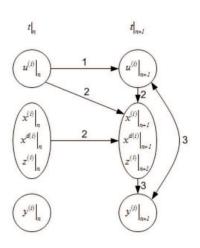


- Gauss Integration
- Trapezoidal-Rule Integration

$$\int_{a}^{b} f(x)dx \approx \frac{1}{2} \sum_{k=1}^{N} (x_{k+1} - x_k) [f(x_{k+1}) + f(x_k)]$$

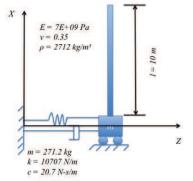
Partitioned Analysis

- Module Coupling Algorithm
 - Prediction by extrapolating based on known values.
 - Update states.
 - Global input-output solve. Range from a simple transfer of information to a full nonlinear-system solve.
 - 4. Correction. Either accept the current solutions or repeat 2 and 3.



Example 1: Partitioned Analysis

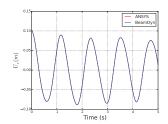
Beam-Spring-Damper-Mass System



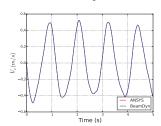
- Time integrator for SDM: Adams-Bashforth-Moulton (ABM4)
- Time integrator for beam: Generalized- α
- Initial condition solution from $U_Z = 0.1$ m.
- Benchmark: ANSYS 60 BEAM188 elements; 10⁻⁵s time increment

Example 1: Partitioned Analysis

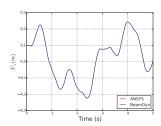
Root Displacement



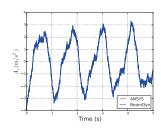
Root Velocity



► Tip Displacement

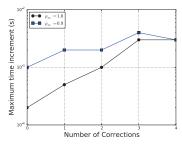


Root Acceleration

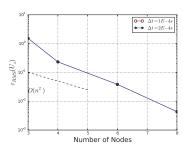


Example 1: Partitioned Analysis

Stability



Accuracy



- Corrections and numerical damping help maximum allowable time step size
- ► RMS Error

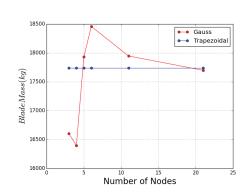
$$arepsilon_{RMS} = \sqrt{rac{\sum_{k=0}^{n_{max}} [U_{Z}^{k} - U_{b}(t^{k})]^{2}}{\sum_{k=0}^{n_{max}} [U_{b}(t^{k})]^{2}}}$$

- NREL 5-MW Wind Turbine
 - 61.5m long with initial twist
 - 49 cross-sectional stations
 - Blade modeled with a SINGLE element

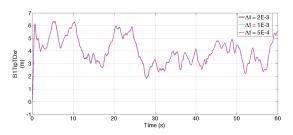
Tip Displacement

-6 -7 -7 -9 -10 -11 -10 -15 20 25 Number of Nodes

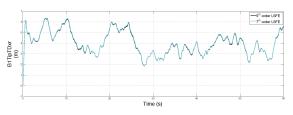
► Blade Mass



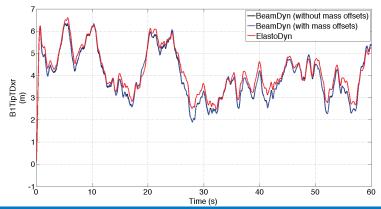
▶ Time Discretization



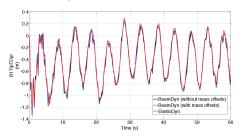
Space Discretization



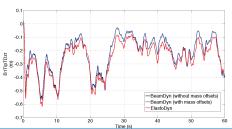
- Aero-Servo-Elastic Coupled Analysis
- ▶ Mean Wind Speed 12 m/s² with Turbulence
- ▶ Test Case 26 in FAST Archive
- ► Tip Displacement *U_{flap}*



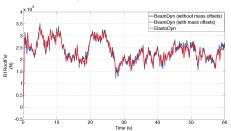
► Tip Displacement *U_{edge}*



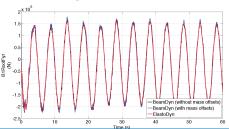
► Tip Displacement *U*_{axial}



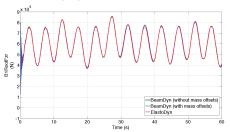
Root Reaction Force F_{flap}



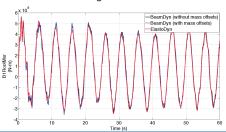
Root Reaction Force F_{edge}



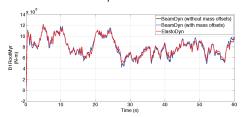
► Root Reaction Force F_{axial}



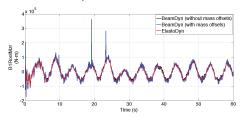
▶ Root Reaction Moment M_{edge}



Root Reaction Moment M_{flap}



► Root Reaction Torque M_{pitch}



Summary

Conclusion

- Based on geometrically exact beam theory, BeamDyn is capable
 of dealing with geometric nonlinear beam problems with arbitrary
 magnitude of displacements and rotations for both static and
 dynamic analyses
- Along with a preprocessor like PreComp or VABS, BeamDyn takes full elastic coupling effects into account
- Spectral Finite Elements and new Trapezoidal-Rule quadrature have been adopted in the implementation of BeamDyn. These new techniques further improved the efficiency and accuracy.
- BeamDyn is implemented following the programming requirements (data structures and interfaces) of the FAST modularization framework. The module coupling algorithm has been implemented and verified through numerical examples.
- BeamDyn optionally replaces previous blade model of FAST when high-fidelity modeling of aero-elastically tailored blades is required.
- More verification and validation results in "FAST v8 Verification and Validation for a MW-Scale Wind Turbine with Aeroelastically Tailored Blades", Guntur, S. et al.

Questions?

Acknowledgments

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