## EEE334 Solutions 2014

1 a

In surveillance radar we want an antenna with a narrow beam to give good angular resolution. However, a narrow beam means we have to scan more positions in the sky which results in a longer scan time. Hence there is a trade-off. This trade-off is significant for mechanically scanned antennas but is less critical for modern electronically scanned phased arrays.

(2)

b

Hemisphere of sky contains  $2\pi$  steradians Antenna beamwidth  $\Delta\Omega = \Delta\theta\Delta\phi$ 

Number of beam positions  $N_B = \frac{2\pi}{\Delta\Omega} = \frac{2\pi}{\Delta\theta\Delta\phi}$ 

Using gain  $G = \frac{4\pi}{\Delta\theta\Delta\phi}$  gives  $N_B = \frac{G}{2}$ 

G = 40dB = 10,000 which gives  $N_B = 5,000$ 

G = 30dB = 1000 which gives  $N_B = 500$ 

Total scan time = 6s, therefore dwell time =  $\frac{6s}{500}$  = 12ms

(4)

С

wavelength 
$$\lambda = \frac{c}{f} = \frac{3*10^8}{560*10^8} = 53.6cm$$

PRT = pulse-width/duty-cycle = 
$$\frac{1.3 \mu S}{8.3 \times 10^{-4}}$$
 = 1.57mS

PRF=1/PRT = 638.5Hz

Average Power = Peak Power \* Duty cycle =  $279kW \times 8.3 \times 10^{-4} = 231.6W$ 

Number of hits 
$$n = \frac{\Delta\theta \times PRF}{6 \times RPM}$$
 so  $\Delta\theta = \frac{6 \times RPM \times n}{PRF} = \frac{6 \times 16 \times 9.9}{638.5} = 1.49^{\circ}$ 

Gain  $G = \frac{4\pi}{\Delta\theta\Delta\phi}$  with beamwidths in radians

So for beamwidths in degrees,  $G = \frac{4\pi}{\Delta\theta\Delta\phi} \times \left(\frac{180}{\pi}\right)^2 = \frac{4\times(180)^2}{1.49\times4\times\pi} = 6920 \text{ or } 38\text{dB}$ 

Max range = 
$$\frac{c}{2 \times PRF} = \frac{3 \times 10^8}{2 \times 638.5} = 3.35 \text{km}$$

Range resolution = 
$$\frac{c \times \tau}{2} = \frac{3 \times 10^8 \times 1.3 \times 10^{-6}}{2} = 195m$$

ċ

Performance of radar system is proportional to the transmit power multiplied by the transmit and receive antenna gains, So we can write

$$PG^2 = K$$

Where P is transmit power, G is the antenna gain (Tx = Rx) and K is a constant.

Now we are told that

$$C = C_P + C_A = PC_k + AC_{sm}$$

Substituting for  $P = \frac{K}{A^2}$  gives

$$C = \frac{KC_{kW}}{A^2} + AC_{sm}$$

Differentiating wrt A and setting to zero gives

$$C_{sm} = \frac{2KC_{kW}}{A^3}$$

But 
$$P = \frac{K}{A^2}$$
 so

$$AC_{sm} = 2PC_{kW}$$

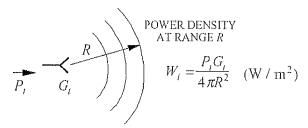
or

$$C_A = 2C_P$$

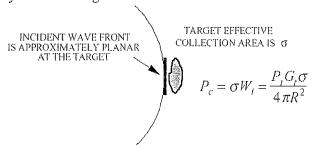
(6)

2a

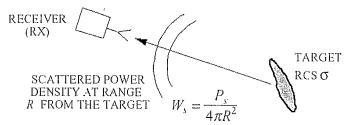
Power density incident on the target



Power collected by the radar target



The RCS gives the fraction of incident power that is scattered back toward the radar. Therefore,  $P_s = P_c$  and the scattered power density at the radar is obtained by dividing by  $4\pi R^2$ .

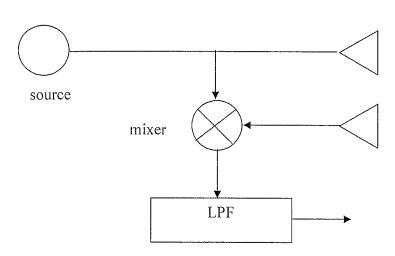


The target scattered power collected by the receiving antenna is  $W_s A_{er}$ . Thus the maximum target scattered power that is available to the radar is

$$P_r = \frac{P_t G_t \sigma A_{er}}{(4\pi R^2)^2} = \frac{P_t G_t G_r \sigma \lambda^2}{(4\pi)^3 R^4}$$

(6)

2b



Tx signal =  $\cos(\omega_0 t)$ 

Rx signal =  $B\cos(\omega_d t)$ 

At mixer Rx is multiplied by a signal with same frequency as Tx signal

Output from mixer is  $S = B\cos(\omega_d t)\cos(\omega_0 t)$ 

Expand to give 
$$S = \frac{B}{2} \Big[ \cos \Big[ (\omega_d - \omega_0) t \Big] + \cos \Big[ (\omega_d + \omega_0) t \Big] \Big]$$

Low-pass filtering leaves only difference term i.e.  $\frac{B}{2}\cos(\omega_d - \omega_0)$  where Doppler

frequency  $\Delta \omega = \omega_d - \omega_0$ 

(4)

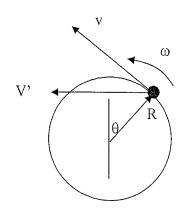
2c

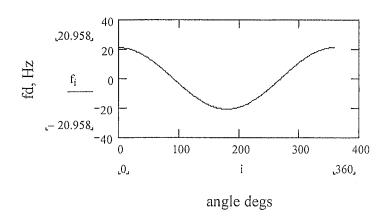
Assuming that  $d\gg R$  we can assume that the target is illuminated by a plane-wave and  $\alpha=0$ 

We have  $v = R\omega$  and  $v' = v\cos\theta$  where v' is the component of the targets velocity in the direction of the illuminating beam of the radar.

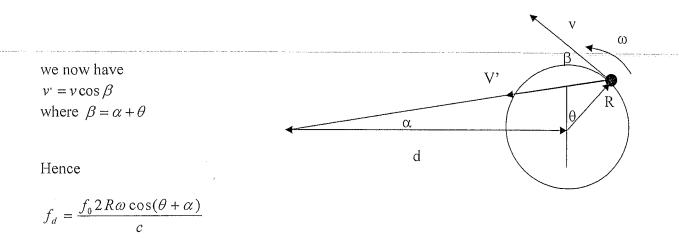
Hence Doppler shift is given by  $f_d = \frac{f_0 2R\omega\cos\theta}{c}$  where  $f_0$  is the radar frequency

Use numerical values and plot. Note that 30RPM = pi rads/s

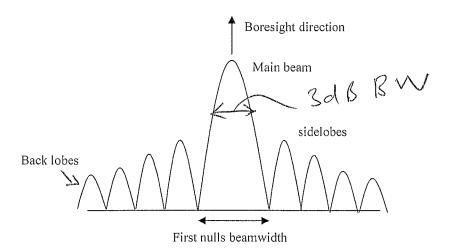




For the second part of the problem we must take into account the angle  $\,lpha$ 



3a



note change in a note reduction Solution is based on Friis transmission equation

$$Pr := Pt \cdot Gt \cdot Gr \left(\frac{\lambda}{4 \cdot \pi \cdot R}\right)^2$$

First calculate some additional parameters from given information

$$\lambda := \frac{3 \cdot 10^8}{10.8 \cdot 10^9} \qquad \lambda = 0.028$$

$$\lambda = 0.028$$

metres

$$Pr := 500 \cdot 10^{-9}$$

Watts

Range 
$$\mathbb{R}:=150\cdot10^3$$

metres

Dt := 2.1 
$$\eta t := 0.75$$

TX efficiency

Diamiter of RX dish

$$Dr := 1.8 \quad \eta r := 0.65$$

RX efficiency

At := 
$$\left(\frac{Dt}{2}\right)^2 \cdot \pi \cdot \eta t$$
 At = 2.598

$$Gt := \frac{4 \cdot \pi \cdot At}{\lambda^2}$$

$$Gt = 4.231 \times 10^4$$

$$Gt = 4.231 \times 10^4$$

$$Ar := \left(\frac{Dr}{2}\right)^2 \cdot \pi \cdot \eta r \qquad Ar = 1.654$$

$$Ar = 1.654$$

$$Gr := \frac{4 \cdot \pi \cdot Ar}{\lambda^2}$$

$$Gr = 2.694 \times 10^4$$

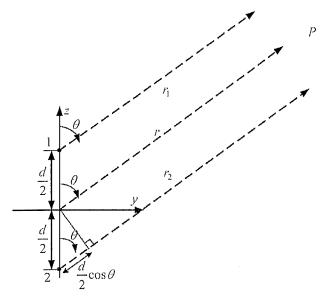
$$Gr = 2.694 \times 10^4$$

Therefore TX power is

$$Pt := \frac{Pr}{\left[Gt \cdot Gr \cdot \left(\frac{\lambda}{4 \cdot \pi \cdot R}\right)^{2}\right]}$$

$$Pt = 2.02$$

Watts



Field at P given by

$$E = \frac{e^{-jkr_1}}{r_1} + \frac{e^{-jkr_2}}{r_2}$$

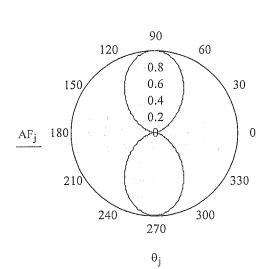
In far-field can assume that  $r_1 = r_2$  for amplitude variations

For phase variations (due to difference in path length)

$$r_1 = r - \frac{d}{2}\cos\theta$$
 and  $r_2 = r + \frac{d}{2}\cos\theta$ 

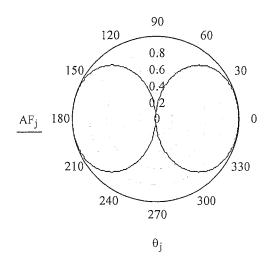
Field now given by  $E = \frac{1}{r} \left[ e^{-jk(r-d_2'\cos\theta)} + e^{-jk(r+d_2'\cos\theta)} \right]$ 

After some further manipulation this gives



$$E = \frac{e^{-jkr}}{r} 2\cos\left[\frac{kd}{2}\cos\theta\right] \text{ and normalised}$$
 array factor given by  $AF = \cos\left[\frac{kd}{2}\cos\theta\right]$ 

If the elements are driven in anti-phase the main beam of radiation is in the endfire direction



(8)

4a

Wireless links more efficient in terms of power when the communication link is over a long distance. In wireless systems the power drops off as  $1/R^2$  and good cable system may have losses of 5dB per km. So, for example, if a system has a 100dB of loss at 20km doubling the distance would produce 200dB of loss in a cable system but only 106dB in a wireless system.  $[1/R^2$  to  $1/(2R)^2$  gives a reduction of 1/4 or 6dB]

(4)

46 (1)

Repart locus of E voctor

AR = a/b

I < MR < 00

Lircular linear policy

9 b/(11) Polarisation diversity. Tel TEn TEn XXI Tx2 Ey Ey 7 XX2 Tx1 and txx vertically polarised Tx2 and Rx2 Horizontally " Signal from Tx, necessed by Rx1 But not received by RXL Signal for Tx2 necessal by Rx2 but not by Rx, Hence can transmit 2 signals using same frequency and dribbe capacity. (an also me LHC + RHC pol Circula pod can he generated corrected from 2 orthogonal depoles driven with a 40° phane difference with a 90° phane difference (c) +90° ...

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4c

The first step in this problem is to work out the directivity (or gain as the antenna is lossless)

Pattern maximum  $U_{\text{max}} = 1$ 

Total radiated power 
$$P_{rad} = \int_{0}^{2\pi} \int_{0}^{\pi} U(\theta, \phi) \sin \theta d\theta d\phi = P_{rad} = 2\pi \int_{0}^{\pi} \sin^4 \theta d\theta$$

Now using the standard integral with  $x = \theta$  and a = 1 and evaluating gives

$$P_{rad} = \frac{3\pi^2}{4}$$

Directivity given by 
$$D = 4\pi \frac{U_{\text{max}}}{P_{ad}} = \frac{16}{3\pi} = 1.7 \text{ or } 2.3 \text{dB}$$

Effective area of antenna given by 
$$A_e = \frac{\lambda^2}{4\pi}D = \frac{30^2}{4\pi} \times 1.7 = 122m^2$$

Power accepted by antenna  $P_r = A_e W_r = 122 \times 5 \times 10^{-6} = 6.1 \times 10^{-4} W$ 

The antenna is not matched to the transmission line so some power will be reflected. To calculated how much we work out the refection coefficient

$$\rho = \frac{Z_L - Z_0}{Z_L + Z_0} = \frac{73 - 50}{73 + 50} = 0.187$$

power transferred to coax cable  $P_a = P_r \left(1 - \rho^2\right) = 6.1 \times 10^{-4} \times (1 - 0.187^2) = 5.89 \times 10^{-4} W$  or  $589 \mu W$ 

(8)