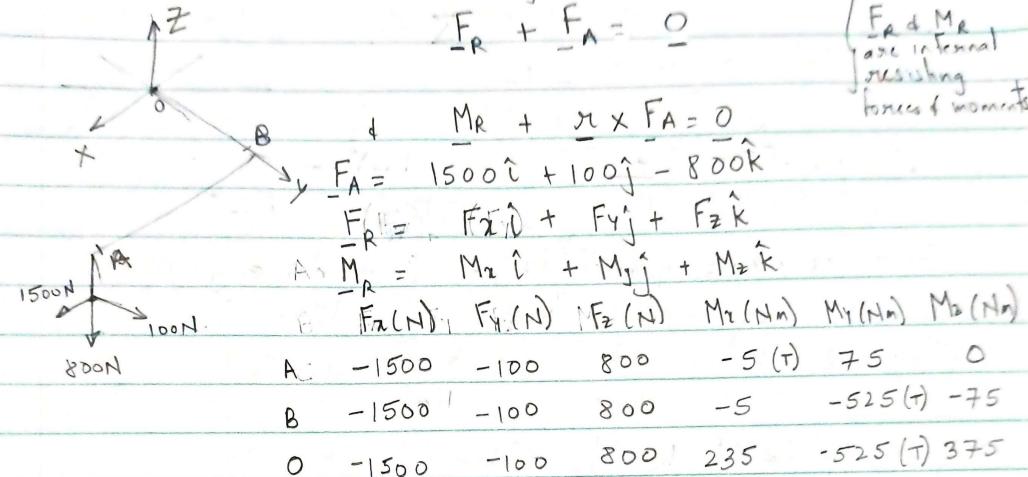


## Problem 1

## ME 423 - Quiz 1

1a. The internal resisting forces and moments at different locations are calculated using:

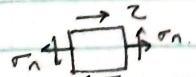


From the table it is seen that the most critical section is at pt O.

We will ignore the effect of direct shear due to  $F_x$  &  $F_z$ .  $M_y$  represents the torque  $\otimes$  y-axis.  $M_x$  represents bending moment  $\otimes$  x-axis &  $M_z$  represents bending  $\otimes$  z axis. The resultant bending moment is  $\tilde{M} = \sqrt{M_y^2 + M_x^2}$ .

$$\tau_{bmax} = \frac{32 \tilde{M}}{\pi d^3} M_y \quad \tau_n = \frac{4F_y}{\pi d^2}$$

$$\tau_n = \tau_{bmax} + \tau_{naxial} = \frac{32 \tilde{M}}{\pi d^3} + \frac{4F_y}{\pi d^2}$$



$$\tau_{naxial} = \frac{16 M_y}{\pi d^3}$$

$$\tau_{vm} = (\tau_n^2 + 3\tau_{max}^2)^{1/2}$$

To find the minimum required diameter

$$\tau_{vm} = \frac{\tau_y}{FoS} \quad \text{or} \quad \tau_{vm}^2 = \frac{\tau_y^2}{FoS^2}$$

$$\therefore \tau_n^2 + 3\tau_{max}^2 = \frac{\tau_y^2}{FoS^2}$$

$$\therefore \left( \frac{32 \tilde{M}}{\pi d^3} + \frac{4P}{\pi d^2} \right)^2 + 3 \left( \frac{16 M_y}{\pi d^3} \right)^2 = \frac{\tau_y^2}{FoS^2} \quad (A)$$

Solve the above equation (A) for d. It is not easy to solve unless you have the "right" calculator. To simplify, we assume that  $\frac{4P}{\pi d^2} \ll \frac{32 \tilde{M}}{\pi d^3}$ . We will verify the assumption at the end. Then we get.

$$\text{(can be solved using any calculator)} \quad \left( \frac{32 \tilde{M}}{\pi d^3} \right)^2 + 3 \left( \frac{16 M_y}{\pi d^3} \right)^2 = \frac{\tau_y^2}{FoS^2} \quad (B)$$

$$\text{Solving B : } d = 33.3012 \text{ mm.} \approx 33.3 \text{ mm.}$$

$$\text{Solving A : } d_2 = 33.3063 \text{ mm.} \approx 33.3 \text{ mm.}$$

We see that the difference in ansus is about 0.02%.

$$d = 33.3063 \text{ mm.}$$

$$\text{For: } \tau_{naxial} \approx 0.11 \text{ MPa.}$$

$$\tau_b = 122.00 \text{ MPa.}$$

$$z = 72.37 \text{ MPa.}$$

$$\therefore \tau_{naxial} \ll \tau_b$$

$$\tau_{naxial} \ll z.$$

Hence our assumption of neglecting  $\tau_a$  is justified. This is true in almost all problems of combined loading.

## Problem 2

$$\textcircled{1} \quad (\tau_{\max})_1 = 340 \text{ MPa}, \quad (\tau_{\min})_1 = 160 \text{ MPa}, \quad n_1 = 8 \times 10^4.$$

$$\textcircled{2} \quad (\tau_{\max})_2 = 320 \text{ MPa}, \quad (\tau_{\min})_2 = -200 \text{ MPa}, \quad n_2 = ?$$

$$\sigma_y = 350 \text{ MPa}, \quad \sigma_{ult} = 420 \text{ MPa}, \quad f = 0.9 \quad \bar{\tau}_e = 175 \text{ MPa}.$$

$$\text{Miners rule} \quad \frac{n_1}{N_1} + \frac{n_2}{N_2} = 1$$

$N_i \rightarrow$  number of cycles for failure at fully reversed (ie  $\tau_m = 0$ ) cycle with amplitude  $\bar{\tau}_i$

Equivalent fully reversed stress amplitude for \textcircled{1}.

$$\left( \bar{\tau}_{a1}' \right) = \frac{\bar{\tau}_{a1}}{1 - \frac{\tau_{m1}}{\sigma_{ult}}} \quad \bar{\tau}_{a1} = 90 \text{ MPa} \quad \tau_{m1} = 250 \text{ MPa}$$

$$= 222.35 \text{ MPa}$$

$$\left( \bar{\tau}_{a2}' \right) = \frac{\bar{\tau}_{a2}}{1 - \frac{\tau_{m2}}{\sigma_{ult}}} \quad \bar{\tau}_{a2} = 260 \text{ MPa} \quad \tau_{m2} = 60 \text{ MPa}$$

$$= 303.33 \text{ MPa}$$

Need to find  $N_1$  &  $N_2$  corresponding to  $\bar{\tau}_{a1}'$  &  $\bar{\tau}_{a2}'$  from the Basquin eqn.

$$\bar{\tau}_a = \sigma_f (2N)^b$$

$$b = T \frac{\ln(\bar{\tau}_{a1}' / \bar{\tau}_e)}{\ln(10^3 / 10^6)} = \frac{\ln(f_{ult} / \bar{\tau}_e)}{\ln(10^3 / 10^6)} = -0.11148$$

$$\sigma_f = \frac{\bar{\tau}_e}{(2 \times 10^6)^b} = 882.07 \text{ MPa}$$

$$\therefore N_1 = \frac{1}{2} \left( \frac{\bar{\tau}_{a1}'}{\sigma_f} \right)^{\frac{1}{b}} = 116705.75$$

$$N_2 = \frac{1}{2} \left( \frac{\bar{\tau}_{a2}'}{\sigma_f} \right)^{\frac{1}{b}} = 7198.8558$$

$$\text{Now } n_2 = N_2 \left( 1 - \frac{N_1}{N_2} \right) \approx 2264 \text{ cycles.}$$

Note that

Note that if you had used  $(\tau_{\max})_1 = 360 \text{ MPa}$ , then  $n_1 > 1$ .

In fact  $(\tau_{\max})_1 > \sigma_y$  and the answer would

3. For a cantilever beam.  $\delta = \frac{PL^3}{3EI}$  For a rectangular c/s  $I = \frac{bh^3}{12}$

$$\therefore \delta = \frac{4PL^3}{Ebh^3}$$

$$\text{or } P = \frac{Eb h^3 \delta}{4L^3}$$

$$P_{\max} = \frac{Eb h^3 \delta_{\max}}{4L^3}, \quad P_{\min} = \frac{Eb h^3 \delta_{\min}}{4L^3}$$

$$\tau_{\max} = \frac{M_{\max} h/2}{I} = \frac{(P_{\max} L) h/2}{I} = \frac{3 Eh \delta_{\max}}{2 L^2}$$

$$\text{Similarly, } \tau_{\min} = \frac{3 Eh \delta_{\min}}{2 L^2}$$

$$\tau_a = \frac{\tau_{\max} - \tau_{\min}}{2} = \frac{3 Eh}{4 L^2} (\delta_{\max} - \delta_{\min})$$

$$\tau_m = \frac{3 Eh}{4 L^2} (\delta_{\max} + \delta_{\min})$$

Substituting  $E = 200 \text{ GPa}$ ,  $h = 4 \text{ mm}$ ,  $L = 300 \text{ mm}$ ,

$$\delta_{\max} = 20 \text{ mm}, \quad \delta_{\min} = 10 \text{ mm}$$

$$\tau_a = 66.66 \text{ MPa}, \quad \tau_m = 200 \text{ MPa}$$

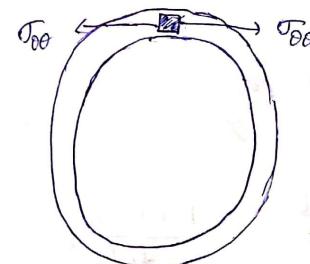
Goodman criterion for infinite life.

$$\frac{\tau_a}{\tau_c} + \frac{\tau_m}{\tau_{ult}} = \frac{1}{n}$$

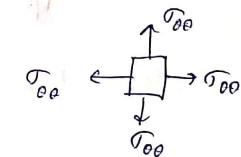
Substituting the above values with  $\tau_c = 0.5 \tau_{ult}$ ,  $n = 1.4 > 1$ . Hence the factor of safety based on infinite life as per Goodman's criterion is 1.4.

Note: No correction factor is assumed for the endurance limit but not was mentioned/given. In practice, one does need to use the corrections factors to modify the uncorrected endurance strength

① Cut the sphere diametrically using a plane

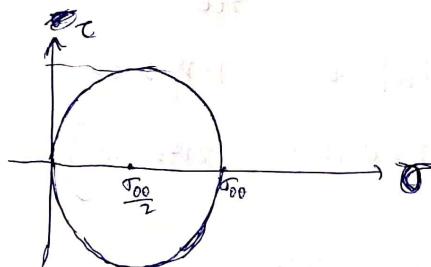


Top view:



There is no  $\sigma-\sigma$  component of stress

$$\sigma_{xx} \times 2\pi rt = \pi r^2 \times p \Rightarrow \sigma_{xx} = \frac{Pr}{2t}$$



Max Tensile Stress =  $\sigma_{xx}$

Max shear stress =  $\frac{\sigma_{xx}}{2}$   
In  $45^\circ$  plane.

Given Yield values are:  
In tension = 140,000 psi  
In shear = 65,000 Psi

If it yields first:

- a) Shear yield stress is less than half of Tensile yield stress.  
But Max. Tensile stress is twice the shear stress.

⇒ Shear yield will happen first if the failure happens by yield.

If it happens by yield:-

$$\frac{P_r}{t} = \left( \frac{\text{Shear yield stress}}{\text{FOS}} \right)$$

$$P_1 = \frac{65000}{2.75} \times \frac{1}{\gamma} = \frac{130000}{11} t$$

If the normal strain limit reaches first :-

- Strain in  $\sigma\sigma$  :-  $\frac{-P_r}{2tE} \times 2\gamma$
- Strain in  $\theta\theta$  :-  $\frac{P_r}{2tE} (1-\gamma)$

$$|\varepsilon_{rr}| < |\varepsilon_{\theta\theta}| \text{ as } 1-\gamma > 2\gamma$$

$$\left[ \because 1-\gamma = 0.72 ; \quad 2\gamma = 0.56 \right]$$

⇒ First brittle failure if it happens, happens through  $\theta\theta$ -direction

$$\frac{P_r}{2tE} (1-\gamma) = 1000 \times 10^{-6}$$

$$P = \frac{1000 \times 10^{-6} \times 2tE \times t}{(1-\gamma) \times \gamma}$$

$$\Rightarrow P_2 = \frac{10^{-3} \times 2 \times 30 \times 10^6}{0.72 \times 8} t = \frac{250000}{3} t = \frac{31250}{3} t$$

$$\cancel{P_1 > P_2} \quad P_2 < P_1$$

⇒ Failure will happen because of normal strain limit i.e.; Brittle failure.

b) min. permissible thickness:-

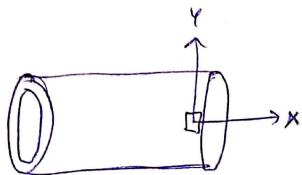
$$P_2 = \frac{31250}{3} t \Rightarrow t = \frac{3P_2}{31250}$$

$$t = \frac{3P_2}{31250} \Rightarrow t = \frac{3 \times 3000}{31250} \text{ in}$$

$$\Rightarrow t = 0.288 \text{ in}$$

(2)

i) Failure will occur likely at either A or B.



$y \rightarrow$  tangential direction

$x \rightarrow$  cylinder's Axis  
direction,

~~Given  $P = 3.5 \text{ MPa}$~~  But required ~~resists the operating~~  
~~pressure~~

~~Point A~~

$$d_2 = 250 \text{ mm}; P = 3.5 \text{ MPa}; t_{\text{out}} = 125 \text{ mm}.$$

$$I = \pi r^3 t$$

$$\sigma_{in} = 125 \text{ mm} - 6 \text{ mm} = 119 \text{ mm}.$$

Due to pressure:  
and bending,



$$\sigma_x = \frac{M r}{I} + \frac{P r_i}{2t}$$

$$= \frac{(45 \times 1000)(750 \times 10^{-3}) \times 125 \text{ mm}}{(6.846 \times 10^7 \text{ mm}^4)} + \frac{(3.5 \text{ MPa} \times 119 \text{ mm})}{2 \times 6 \text{ mm}}$$

$$= 158 \text{ MPa.}$$

~~Failure on~~ ~~due to~~ ~~stress~~ ~~MPa.~~

$$\sigma_y = \frac{P r_i}{t} = \frac{(3.5 \text{ MPa}) \times (119 \text{ mm})}{6 \text{ mm}}$$

$$= 69.4 \text{ MPa.}$$

$$\tau_{xy} = \frac{T r_2}{J} = \frac{45000 \times 125}{6.846 \times 10^7 \text{ mm}^4} = \frac{(45000 \text{ N} \times 125 \text{ mm}) \times (125 \text{ mm})}{6.846 \times 10^7 \text{ mm}^4} = 10.27 \text{ MPa.}$$

At B there is no Bending as it is on neutral axis.

At A, there will be Maximum Bending load.

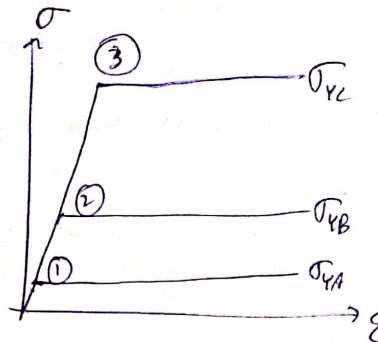
Every other stress is same between A and B

$\Rightarrow$  Stress at A  $>$  Stress at B.

$\Rightarrow$  Failure is likely to occur at A.

3

All the tie-rods have same  $E$  (Young's Modulus)



Order of yielding:-

A, B, C

Deflection will be same in all the rods.

$$\delta_A = \delta_B = \delta_C \Rightarrow \varepsilon_A = \varepsilon_B = \varepsilon_C \quad [As L is same]$$

$$F_A + F_B + F_C = F$$

$$\sigma_A \times A_A + \sigma_B \times A_B + \sigma_C \times A_C = F$$

$$a) \text{ when } A \text{ yields, } \sigma_A = \sigma_{yA}$$

$$\text{until yielding, } \sigma_A = E\varepsilon_A, \sigma_B = E\varepsilon_B, \sigma_C = E\varepsilon_C$$

$$\text{But, } \varepsilon_A = \varepsilon_B = \varepsilon_C$$

$$\Rightarrow \sigma_A = \sigma_B = \sigma_C = E\varepsilon$$

$$\Rightarrow \varepsilon = \frac{\sigma_{yA}}{E} = \frac{50 \times 10^6 \text{ Pa}}{100 \times 10^9 \text{ Pa}} = 5 \times 10^{-4}$$

$$\Rightarrow F = 50 \times 10^6 \times [5+3+1] \times 10^{-4} \\ = 50 \times 9 \times 10^2 = 45 \times 10^3 \text{ N.}$$

$$\text{Deflection } \delta = \frac{E \times l}{5 \times 10^4 \times \frac{1}{10} \text{ m}} = 0.05 \text{ mm}$$

b) when B yields,

$$\text{Here, } \sigma_A = \sigma_{yA}, \sigma_B = \sigma_{yB}, \sigma_C = \sigma_{yC} \\ [\text{Hasn't yielded yet}]$$

$$\Rightarrow \varepsilon_B = \varepsilon_C = \frac{\sigma_{yB}}{E} = \varepsilon_A$$

$$F = \sigma_A A_A + \sigma_B A_B + \sigma_C A_C$$

$$\Rightarrow F = \sigma_{yA} \times A_A + (\sigma_{yB})(A_B + A_C)$$

$$\Rightarrow \varepsilon = \frac{\sigma_{yB}}{E} = \frac{100 \times 10^6}{100 \times 10^9} = 10^{-3} \Rightarrow \delta = 10^{-3} \times \frac{1}{10} = 0.1 \text{ mm}$$

$$\Rightarrow F = 50 \times 10^6 \times 5 \times 10^{-4} + 100 \times 10^6 (3+1) \times 10^{-4}$$

$$\Rightarrow F = 65000 \text{ N.} = 65 \times 10^3 \text{ N.}$$

c) when C yields,

$$\sigma_A = \sigma_{yA}, \sigma_B = \sigma_{yB}, \sigma_C = \sigma_{yC}$$

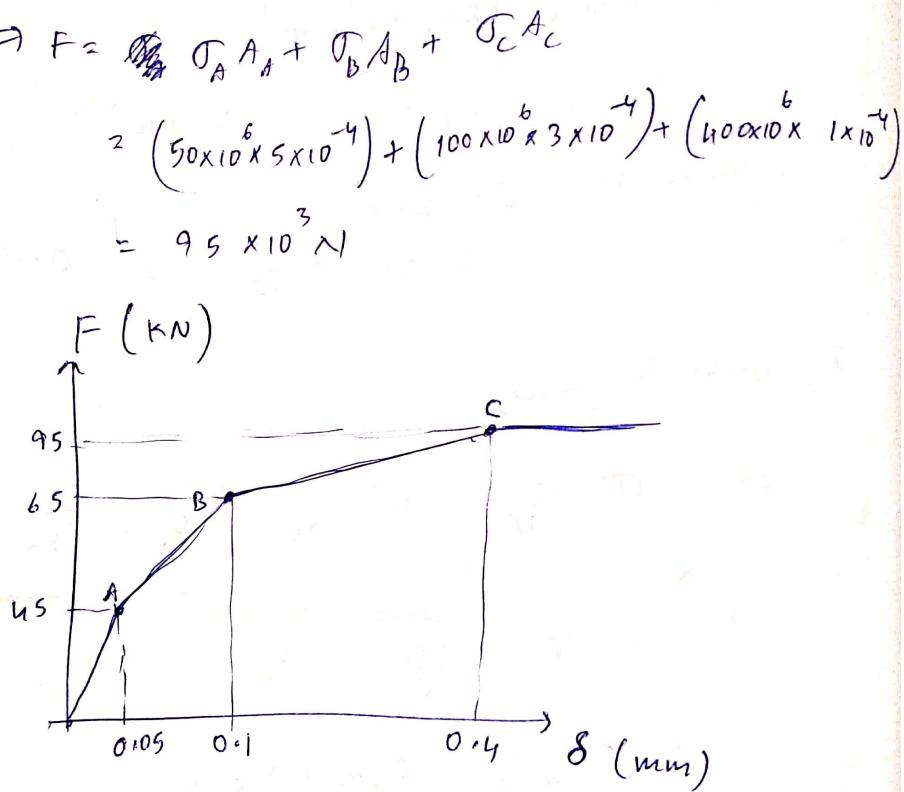
$$\Rightarrow \varepsilon_C = \frac{\sigma_{yC}}{E} = \frac{400 \times 10^6}{100 \times 10^9} = 4 \times 10^{-3}$$

$$\Rightarrow \delta = 4 \times 10^{-3} \times \frac{1}{10} = 0.4 \text{ mm}$$

(4)

(Given is the slender rod [leg of table])

<u>Function:-</u>	Table leg [should bear compressive load].
<u>Objective:-</u>	* Minimizing mass * Improving slenderness (minimize $\frac{L}{r}$ )
<u>Constraint:-</u>	* $L$ is given. * Avoid Buckling
<u>Free variables:-</u>	Radius of leg, Material.



$$m = \pi r^2 L \times \rho$$

For Buckling, critical load will be

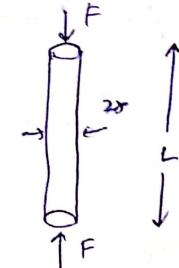
$$F_{\text{critical}} = \frac{\pi^2 EI}{L^2}$$

$$\text{where } I = \frac{\pi r^4}{4}$$

Given load  $F$  must be below  $F_{\text{critical}}$

$$\Rightarrow F < \frac{\pi^2 E}{L^2} \times \frac{\pi r^4}{4} = \frac{\pi^3 E r^4}{4 L^2}$$

$$\text{But, } r^2 = \frac{m}{\pi L \rho} \Rightarrow r^4 = \frac{m^2}{\pi^2 L^2 \rho^2}$$



$$\Rightarrow F < \frac{\pi^3 E}{4L^2} \times \frac{m^2}{M^2 L^2 \rho^2}$$

$$\Rightarrow F < \frac{\pi}{4L^4} \times \frac{E}{\rho^2} m^2$$

$$\Rightarrow m > \sqrt{\frac{4F}{\pi}} L^2 \times \left(\frac{\rho}{E}\right)$$

$$\Rightarrow m > \sqrt{\frac{4F \times L^2}{\pi} \times \frac{1}{M_1}} \quad \text{where } \boxed{M_1 = \frac{E \rho}{\ell}}$$

$\Rightarrow$  To minimize mass,  $M_1$  needs to be maximized

Also,  $F < \frac{\pi^3 E \rho^4}{4L^2} \Rightarrow \rho > \left(\frac{4FL^2}{\pi^3}\right)^{1/4} \times \left(\frac{1}{E}\right)^{1/4}$

$$\Rightarrow \rho > \left[\frac{4FL^2}{\pi^3}\right]^{1/4} \times \left[\frac{1}{M_2}\right]^{1/4} \quad \text{where } \boxed{M_2 = E}$$

$\Rightarrow$  To minimize radius,  $M_2$  needs to be maximized

From the material selection chart, plotting two material indices,

Choices of materials can be +



Material:	$M_1 = \frac{E \rho}{\ell} \left[ \frac{GPa}{kg/m^3} \right]$	$M_2 = \frac{E}{\rho} \left[ GPa \right]$	Comments:
Mg Alloys	0.115	44.8	Very low $M_1$ , -
Al Alloys	0.327	70	Very low $M_1$ ,
CFRP	6.201	.96	Best acc to Indices but very costly.
GFRP	2.211	21	Very low $M_2$ .
Wood.	4.234	93	Consider cost, wood is the last material after CFRP for $M_1$ .