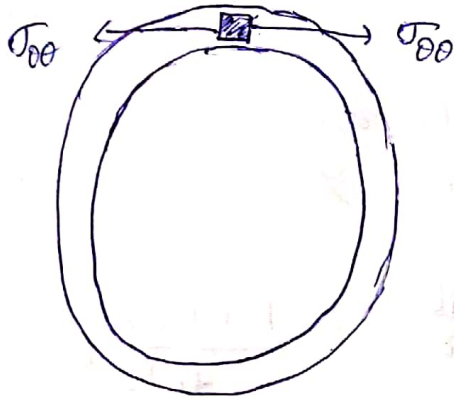
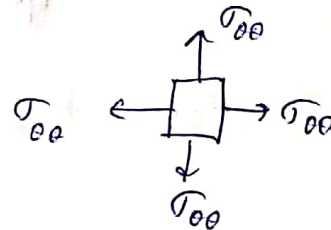


- ① cut the sphere diametrically using a plane

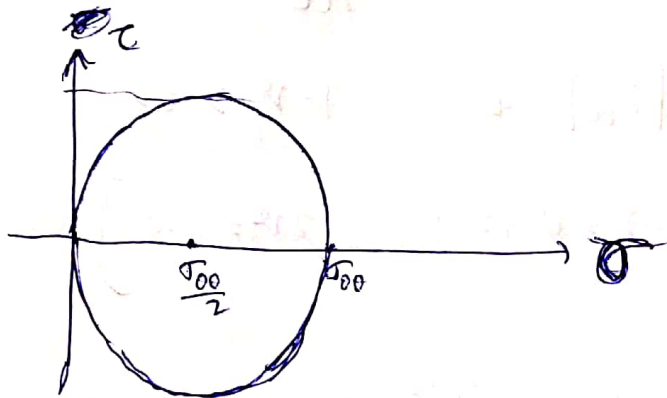


Top view:-



there is no  $r-r$  component of stress

$$\sigma_{00} \times 2\pi r t = \pi r^2 \times p \Rightarrow \sigma_{00} = \frac{Pr}{2t}$$



Max Tensile stress =  $\sigma_{00}$

Max shear stress =  $\frac{\sigma_{00}}{2}$   
 ↳ In  $45^\circ$  plane.

Given Yield values :- In tension = 140,000 psi  
 In shear = 65,000 psi

~~If it yields first :-~~

- a) Shear yield stress is less than half of Tensile yield stress. But ~~the~~ Max. Tensile stress is twice the Max shear stress.

⇒ Shear yield will happen first if the failure happens by yield.

If it happens by yield, :-

$$\frac{P_r}{4t} = \left( \frac{\text{shear yield stress}}{\text{FOS}} \right)$$

$$\Rightarrow P_1 = \frac{65000}{2.75} \times \frac{4t}{\gamma} = \frac{130000}{11} t$$

If the normal strain limit reaches first

Strain in xx :-  $\frac{-P_r \times 2\nu}{2tE}$

Strain in yy :-  $\frac{P_r(1-\nu)}{2tE}$

$$|\epsilon_{xx}| < |\epsilon_{yy}| \text{ as } 1-\nu > 2\nu$$

$$\left[ \because 1-\nu = 0.72 ; \quad 2\nu = 0.56 \right]$$

⇒ First brittle failure if it happens, happens through yy - direction

$$\Rightarrow \frac{P_r(1-\nu)}{2tE} = 1000 \times 10^{-6}$$

$$\Rightarrow P = \frac{1000 \times 10^{-6} \times 2 \times E \times t}{(1-\nu) \times \gamma}$$

$$\Rightarrow P_2 = \frac{10^{-3} \times 2 \times 30 \times 10^6}{0.72 \times 8} t = \frac{250000}{3} t$$

$$= \frac{31250}{3} t$$

$$\cancel{P_2 > P_1} \quad P_2 < P_1$$

⇒ Failure will happen because of normal strain limit i.e.; Brittle failure.

b) Min. permissible thickness:-

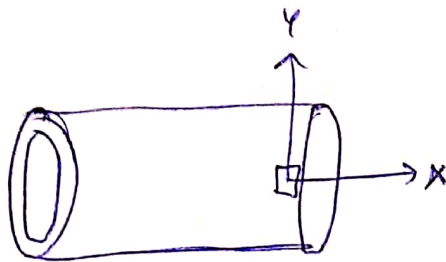
$$P_2 = \frac{31250}{3} t \quad \Rightarrow \quad \cancel{t = \frac{3 P_2 \times 31250}{3}}$$

$$t = \frac{3 P_2}{31250} \quad \Rightarrow \quad t = \frac{3 \times 3000}{31250} \text{ in}$$

$$\Rightarrow t = 0.288 \text{ in}$$

2

1) Failure will occur likely at either A or B.



Y → tangential direction  
X → Cylinder's Axis direction.

~~Given  $P = 3.5 \text{ MPa}$  (But requires  $1.5$  times the operating pressure)~~

$$I = \pi r^3 t$$

~~$P = 3.5 \text{ MPa}$~~

POINT A:

$$d_z = 250 \text{ mm}; \quad P = 3.5 \text{ MPa}; \quad r_{\text{out}} = 125 \text{ mm}.$$

$$r_{\text{in}} = 125 \text{ mm} - 6 \text{ mm} = 119 \text{ mm}.$$

Due to pressure:-  
and Bending,

$$\sigma_x = \frac{Mr_{\text{out}}}{I} + \frac{Pr_i}{2t}$$

$$= \frac{(45 \times 1000)(750 \times 10^{-3}) \times 125 \text{ mm}}{(6.846 \times 10^7 \text{ mm}^4)} + \frac{(3.5 \text{ MPa} \times 119 \text{ mm})}{2 \times 6 \text{ mm}}$$

$$= 158 \text{ MPa}.$$

~~$\sigma_y = 69.4 \text{ MPa}$~~

$$\sigma_y = \frac{Pr_i}{t} = \frac{(3.5 \text{ MPa}) \times (119 \text{ mm})}{6 \text{ mm}}$$

$$= 69.4 \text{ MPa}.$$

$$Z_{xy} = \frac{Tr_z}{J} = \frac{\cancel{45000} \times 125}{\frac{(45000 \text{ N} \times 125 \text{ mm}) \times (125 \text{ mm})}{6.846 \times 10^7 \text{ mm}^4}} = 10.27 \text{ MPa.}$$

At B there is no Bending as it is on neutral axis.

At A, there will be Maximum Bending load.

Every other stress is same between A and B

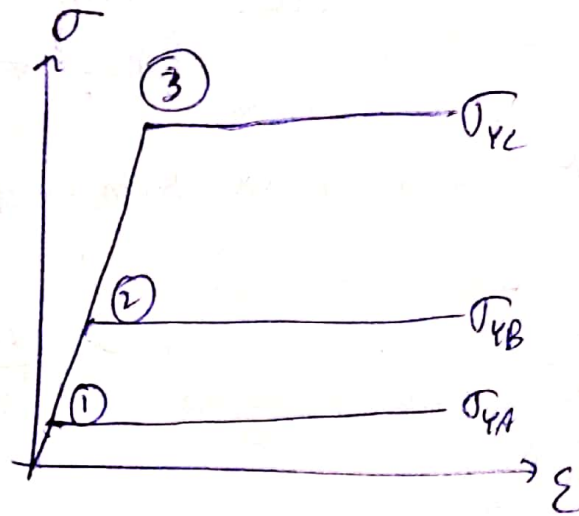
⇒ Stress at A > Stress at B.

⇒ Failure is likely to occur at A.



3

All the tie-rods have same  $E$  [Young's Modulus]



Order of yielding:-

A, B, C

Deflection will be same in all the rods.

$$\delta_A = \delta_B = \delta_C \Rightarrow \epsilon_A = \epsilon_B = \epsilon_C \quad \left[ \begin{array}{l} \text{As } L \text{ is} \\ \text{same} \end{array} \right]$$

$$F_A + F_B + F_C = F$$

$$\sigma_A \times A_A + \sigma_B \times A_B + \sigma_C \times A_C = F$$

a) when A yields,  $\sigma_A = \sigma_{YA}$

until yielding,  $\sigma_A = E\epsilon_A$ ,  $\sigma_B = E\epsilon_B$ ,  $\sigma_C = E\epsilon_C$

$$\text{But, } \epsilon_A = \epsilon_B = \epsilon_C$$

$$\Rightarrow \sigma_A = \sigma_B = \sigma_C = E\epsilon$$

$$\Rightarrow \epsilon = \frac{\sigma_{YA}}{E} = \frac{50 \times 10^6 \text{ Pa}}{100 \times 10^9 \text{ Pa}} = 5 \times 10^{-4}$$

$$\Rightarrow F = 50 \times 10^6 \times (5+3+1) \times 10^{-4}$$

$$= 50 \times 9 \times 10^2 = 45 \times 10^3 \text{ N}$$

Deflection  $\delta = \epsilon \times l$

$$= 5 \times 10^{-4} \times \frac{1}{10} \text{ m} = 0.05 \text{ mm}$$

b) when B yields,

Here,  $\sigma_A = \sigma_{YA}$ ,  $\sigma_B = \sigma_{YB}$ ,  $\sigma_C = \sigma_{YB}$   
[Hasn't yielded yet]

$$\Rightarrow \epsilon_B = \epsilon_C = \frac{\sigma_{YB}}{E} = \epsilon_A$$

$$F = \sigma_A A_A + \sigma_B A_B + \sigma_C A_C$$

$$\Rightarrow F = \sigma_{YA} \times A_A + (\sigma_{YB})(A_B + A_C)$$

$$\Rightarrow \epsilon = \frac{\sigma_{YB}}{E} = \frac{100 \times 10^6}{100 \times 10^9} = 10^{-3} \Rightarrow \delta = 10^{-3} \times \frac{1}{10} = 0.1 \text{ mm}$$

$$\Rightarrow F = 50 \times 10^6 \times 5 \times 10^{-4} + 100 \times 10^6 (3+1) \times 10^{-4}$$

$$\Rightarrow F = 65000 \text{ N} = 65 \times 10^3 \text{ N}$$

c) when C yields,

$\sigma_A = \sigma_{YA}$ ,  $\sigma_B = \sigma_{YB}$ ,  $\sigma_C = \sigma_{YC}$

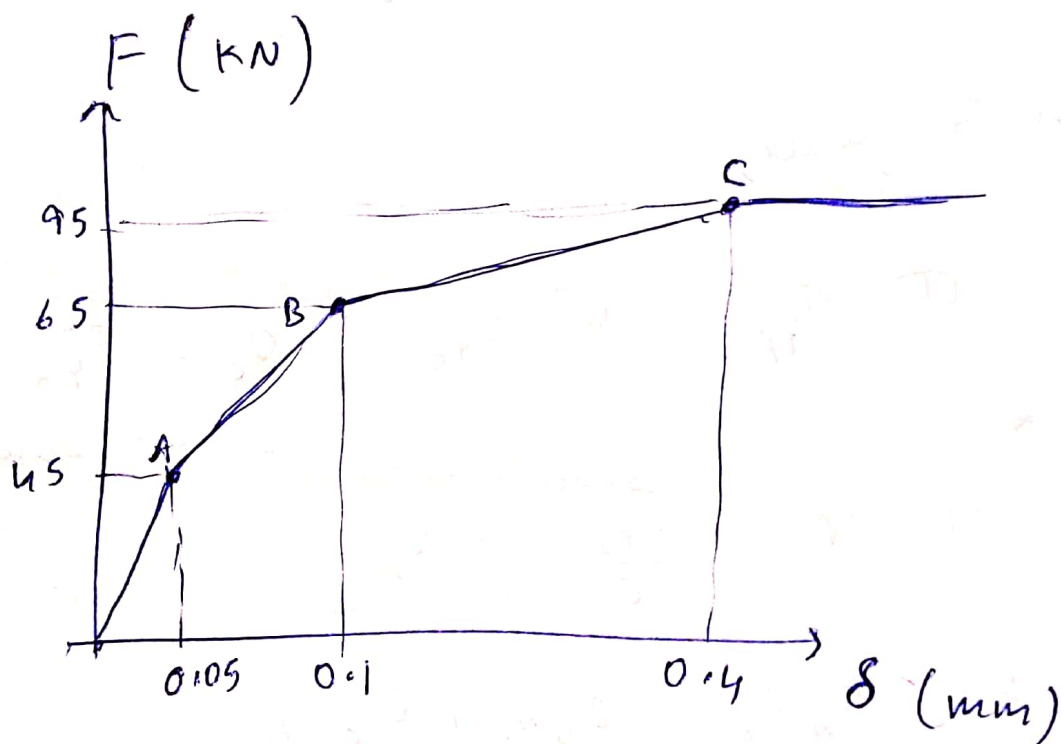
$$\Rightarrow \epsilon_C = \frac{\sigma_{YC}}{E} = \frac{400 \times 10^6}{100 \times 10^9} = 4 \times 10^{-3}$$

$$\Rightarrow \delta = 4 \times 10^{-3} \times \frac{1}{10} = 0.4 \text{ mm}$$

$$\Rightarrow F = \sigma_A A_A + \sigma_B A_B + \sigma_C A_C$$

$$= (50 \times 10^6 \times 5 \times 10^{-4}) + (100 \times 10^6 \times 3 \times 10^{-4}) + (400 \times 10^6 \times 1 \times 10^{-4})$$

$$= 95 \times 10^3 \text{ N}$$





4

Given is the slender rod [legs of table]

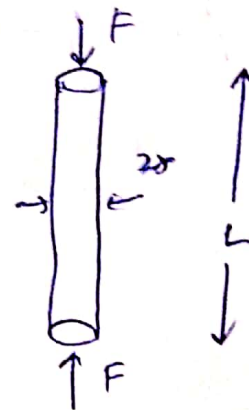
|                         |  |
|-------------------------|--|
| <u>Function:-</u>       | Table leg [should bear compressive load].  |
| <u>Objective:-</u>      | <ul style="list-style-type: none"> <li>* Minimizing mass</li> <li>* Improving slenderness (minimize <math>\frac{r}{L}</math>)</li> </ul> |
| <u>Constraints:-</u>    | <ul style="list-style-type: none"> <li>* <math>L</math> is given.</li> <li>* Avoid Buckling</li> </ul>                                   |
| <u>Free variables:-</u> | Radius of leg, Material.   |

$$m = \pi r^2 L \times \rho$$

For Buckling, critical load

will be :-

$$F_{\text{Critical}} = \frac{\pi^2 EI}{L^2}$$



where  $I = \frac{\pi r^4}{4}$

Given load  $F$  must be below  $F_{\text{critical}}$

$$\Rightarrow F < \frac{\pi^2 E}{L^2} \times \frac{\pi r^4}{4} = \frac{\pi^3 E r^4}{4 L^2}$$

But,  $r^2 = \frac{m}{\pi L \rho} \Rightarrow r^4 = \frac{m^2}{\pi^2 L^2 \rho^2}$

$$\Rightarrow F < \frac{\pi^3 E}{4L^2} \times \frac{m^2}{\pi^2 L^2 \rho^2}$$

$$\Rightarrow F < \frac{\pi}{4L^4} \times \frac{E}{\rho^2} m^2$$

$$\Rightarrow m > \sqrt{\frac{4F}{\pi}} L^2 \times \left( \frac{\rho}{E^{1/2}} \right)$$

$$\Rightarrow m > \sqrt{\frac{4F}{\pi}} \times L^2 \times \frac{1}{M_1} \quad \text{where } \boxed{M_1 = \frac{E^{1/2}}{\rho}}$$

$\Rightarrow$  To minimize mass,  $M_1$  needs to be maximized

$$\text{Also, } F < \frac{\pi^3 E \gamma^4}{4L^2} \Rightarrow \gamma > \left( \frac{4FL^2}{\pi^3} \right)^{1/4} \times \left( \frac{1}{E} \right)^{1/4}$$

$$\Rightarrow \gamma > \left[ \frac{4FL^2}{\pi^3} \right]^{1/4} \times \left[ \frac{1}{M_2} \right]^{1/4} \quad \text{where } \boxed{M_2 = E}$$

$\Rightarrow$  To minimize radius,  $M_2$  needs to be maximized

From the material selection chart, plotting two material indices,


Choices of materials can be:



| Material: | $M_1 = \frac{E^{1/2}}{\rho} \left[ \frac{(\text{GPa})^{1/2}}{\text{kg/m}^3} \right]$ | $M_2 = E \text{ [GPa]}$ | Comments:   |
|-----------|--|-------------------------|---|
| Al Alloys | 0.115  | 44.8                    | Very low $M_1$  |
| Al Alloys | 0.327  | 70                      | Very low $M_1$  |
| CFRP      | 6.201  | 96                      | Best acc. to indices but very costly.                         |
| GFRP      | 2.211  | 21                      | Very low $M_2$  |
| Wood.     | 4.234  | 9.3                     | Consider cost, wood is the best material after CFRP for $M_1$ |

Best choice without cost

Best choice with cost



With only Material Indices, CFRP is the best choice

With Material Indices and cost, ~~wood is the best choice.~~  
wood and GFRP are good choices depending on Material Index.