# CH<sub>8</sub>

### **CH\_8**

8.1 Spherical Pressure Vessels

Definition

Stresses at the Outer Surface

Stresses at the Inner Surface

8.2 Cylindrical Pressure Vessels

**Circumferential Stress** 

**Longitudinal Stress** 

Stresses at the Outer Surface

in-plane shear stresses

out-of-plane shear stresses

Stresses at the Inner Surface

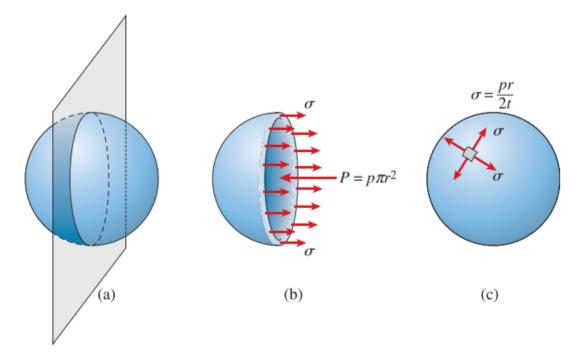
8.3 Maximum Stresses in Beams

8.4 Combined Loadings

# **8.1 Spherical Pressure Vessels**

### **Definition**

pressure vessels are closed structures containing liquids or gases under pressure



according to the above figure

$$p\pi r^2 = \sigma(2\pi r t)$$
  $\sigma = rac{pr}{2t}$ 

### Stresses at the Outer Surface

every plane of the vessel is a principle plane and every direction is a principle direction

$$\sigma_1=\sigma_2=rac{pr}{2t}$$

and the maximum shear stresses is made by 45 degree rotation

$$au_{max} = rac{\sigma}{2} = rac{pr}{4t}$$

## Stresses at the Inner Surface

the principle stresses are the same

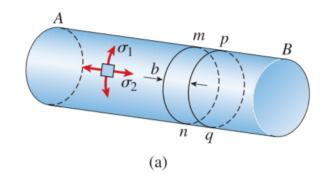
$$\sigma_1=\sigma_2=rac{pr}{2t}$$

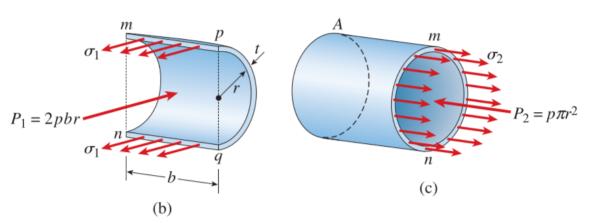
the in-plane shear stresses are zero, but the maximum out-of-plane shear stress is

$$au_{max} = rac{\sigma + p}{2} = rac{pr}{4t} + rac{p}{2}$$

# **8.2 Cylindrical Pressure Vessels**

## **Circumferential Stress**





$$\sigma_1(2bt) - 2pbr = 0$$

$$\sigma_1 = \frac{pr}{t}$$

# **Longitudinal Stress**

$$\sigma_2(2\pi rt)-p\pi r^2=0$$
  $\sigma_2=rac{pr}{2t}$ 

### Stresses at the Outer Surface

$$au_{max} = rac{\sigma_1}{2} = rac{pr}{2t}$$

#### in-plane shear stresses

$$( au_{max})_z=rac{\sigma_1-\sigma_2}{2}=rac{\sigma_1}{4}=rac{pr}{4t}$$

### out-of-plane shear stresses

$$( au_{max})_x = rac{\sigma_1}{2} = rac{pr}{2t} \quad ( au_{max})_y = rac{\sigma_2}{2} = rac{pr}{4t}$$

### Stresses at the Inner Surface

$$\sigma_1 = \frac{pr}{t} \quad \sigma_2 = \frac{pr}{2t} \quad \sigma_3 = -p$$

$$(\tau_{max})_x = \frac{\sigma_1 - \sigma_3}{2} = \frac{pr}{2t} + \frac{p}{2}$$

$$(\tau_{max})_y = \frac{\sigma_2 - \sigma_3}{2} = \frac{pr}{4t} + \frac{p}{2}$$

$$(\tau_{max})_z = \frac{\sigma_1 - \sigma_2}{2} = \frac{pr}{4t}$$

### 8.3 Maximum Stresses in Beams

$$\sigma = -rac{My}{I} \qquad au = rac{VQ}{Ib}$$

# 8.4 Combined Loadings

- 1. Select a point in the structure where the stresses and strains are to be determined. (The point is usually selected at a cross section where the stresses are large, such as at **a cross section where the bending moment has its maximum value**.)
- 2. For each load on the structure, determine the stress resultants at the cross section containing the selected point. (The possible stress resultants are an axial force, a twisting moment, a bending moment, and a shear force.)
- 3. Calculate the normal and shear stresses at the selected point due to each of the stress resultants. Also, if the structure is a pressure vessel, determine the stresses due to the internal pressure.  $\sigma = P/A$   $\tau = T\rho/I_P$   $\sigma = My/I$   $\tau = VQ/Ib$   $\sigma = pr/t$
- 4. Combine the individual stresses to obtain the resultant stresses at the selected point. In other words, obtain the stresses  $\sigma_x$ ,  $\sigma_y$ , and  $\tau_{xy}$  acting on a stress element at the point. (Note that in this chapter we are dealing only with elements in plane stress.)
- 5. Determine the principal stresses and maximum shear stresses at the selected point, using either the stress-transformation equations or Mohr's circle. If required, determine the stresses acting on other inclined planes.
- 6. Determine the strains at the point with the aid of Hooke's law for plane stress.
- 7. Select additional points and repeat the process. Continue until enough stress and strain information is available to satisfy the purposes of the analysis.