CH₅

CH 5

5.1 Pure Bending and Nonuniform Bending

Pure Bending

Nonuniform Bending

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Maximum Stresses at a Cross Section

Doubly Symmetric Shapes

Rectangular Cross Section

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5.8 Shear Stresses in Rectangular Beams

Shear Formula

5.1 Pure Bending and Nonuniform Bending

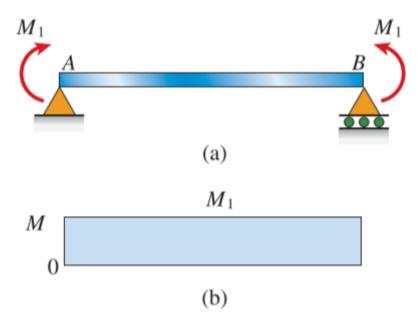
Pure Bending

the flexure of a beam under a constant bending moment

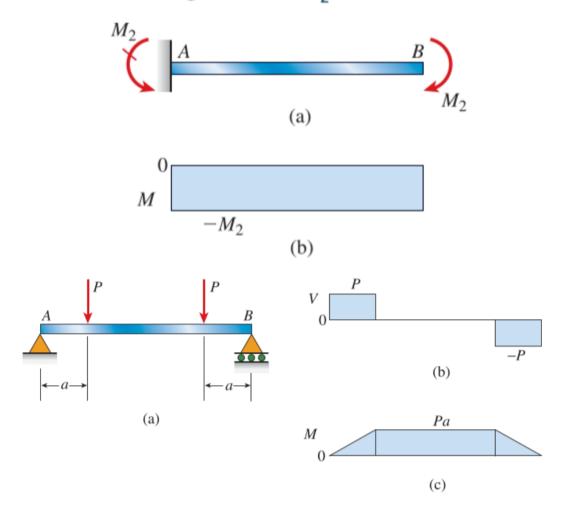
Nonuniform Bending

the flexure in the presence of shear forces

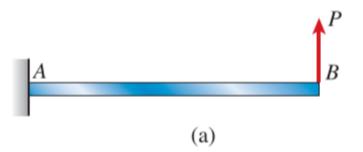
Simple beam in pure bending $(M = M_1)$

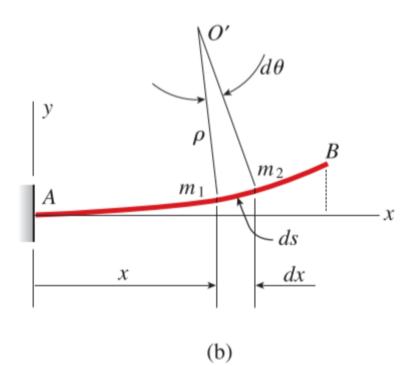


Cantilever beam in pure bending $(M = -M_2)$



5.2 Curvature of a Beam



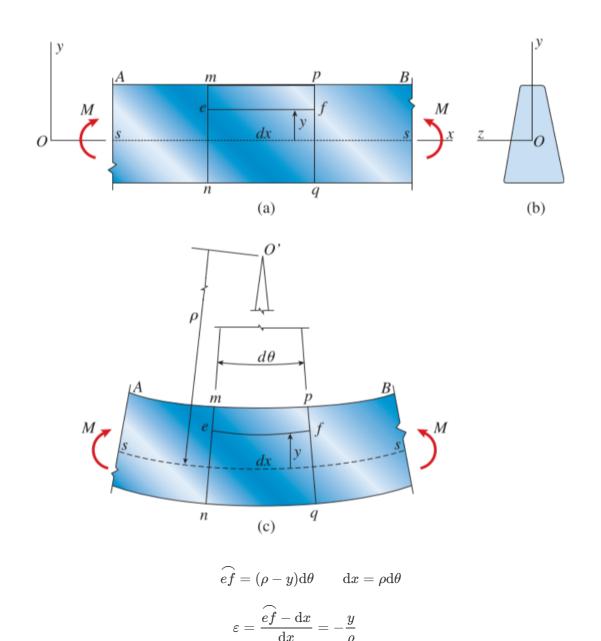


$$\rho d\theta = ds$$

$$\kappa = \frac{1}{\rho} = \frac{d\theta}{ds}$$

$$= \frac{d\theta}{dx}$$

5.3 Longitudinal Strains in Beams



5.4 Normal Stress in Beams

$$\sigma_x = E arepsilon_x = rac{Ey}{
ho} = - E \kappa y$$

Location of Neutral Axis

$$\int_{A}\sigma_{x}\mathrm{d}A=-\int_{A}E\kappa y\mathrm{d}A=0$$
 $\int_{A}y\mathrm{d}A=0$

Moment Curvature Relationship

$$\mathrm{d}M = -\sigma_x y \mathrm{d}A$$

$$M = -\int_A \sigma_x y \mathrm{d}A$$

$$= \int_A \kappa E y^2 \mathrm{d}A = \kappa E \int_A y^2 \mathrm{d}A$$

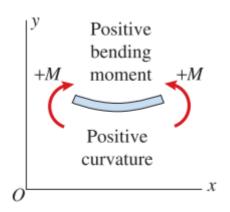
$$= \kappa E I$$

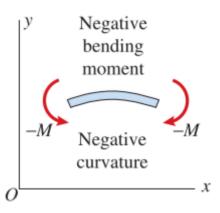
$$\kappa = \frac{1}{\rho} = \frac{M}{EI}$$

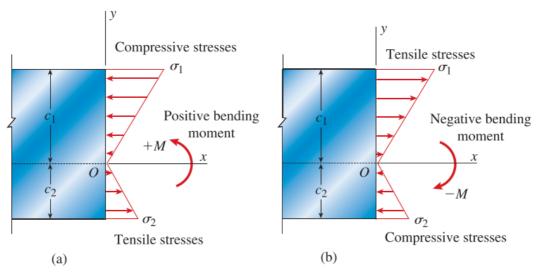
Flexure Formula

bending stresses (flexure stresses)

$$\sigma_x = -rac{My}{I}$$







For positive moment:
$$\sigma_c=-rac{Mar{y}}{I}~\sigma_t=-rac{M(ar{y}-h)}{I}$$
 For negative moment: $\sigma_c=-rac{M(ar{y}-h)}{I}~\sigma_t=-rac{Mar{y}}{I}$

Maximum Stresses at a Cross Section

the *maximum tensile* and *compressive bending stresses* acting at any given cross section occur at points **located farthest from the neutral axis**

$$\sigma_{max} = -rac{Mc}{I} = -rac{M}{S}$$
 $S = rac{I}{c}$

where S is known as the **section moduli** of the cross-sectional area

Doubly Symmetric Shapes

Rectangular Cross Section

$$I = \frac{bh^3}{12} \qquad S = \frac{bh^2}{6}$$

Circular Cross Section

$$I = \frac{\pi d^4}{64} \qquad S = \frac{\pi d^3}{32}$$

5.8 Shear Stresses in Rectangular Beams

Shear Formula

$$au = rac{VQ}{I_z b}$$

- V: internal resultant shear force
- I: moment of inertia of entire x-sectional area computed about the neutral axis
- ullet b: width of the member's x-sectional area, measured at the pt where au is to be determined
- Q: $rac{1}{2}b(rac{h^2}{4}-y^2)$, where y is measured from neutral axis

$$au=rac{6V}{hh^3}(rac{h^2}{4}-y^2)$$