Harrison M. Leece

Westchester, CA 90045 https://github.com/HarrisonLeece

harryleecemail@gmail.com (669)258-1729

May 2020 graduate with experience in the design, operation and analysis of launch vehicles. Strong technical, communication and leadership skills developed through internships and academic projects. Seeking a role at an innovative launch provider in the design, test and analysis of aerospace fluid, thermodynamic and propulsion systems.

Education

Loyola Marymount University, Los Angeles CA

BSE, Mechanical Engineering

Expected: May 2020 GPA: 3.84

Work Experience

The Aerospace Corporation, El Segundo, CA, Propulsion Systems Intern

March 2019-August 2019

- Designed and developed the backend and user interface for a utility that enables users to rapidly download cross-discipline launch vehicle telemetry for use in launch risk assessment
- Supported the Air Force launch rehearsal and day of launch activities by developing a Falcon 9 stage propulsion handbook for use during day of launch support and as a new employee quick-start to the Falcon vehicles
- Reverse engineered, and converted Aerospace's Atlas V solid rocket boosters analysis code from MATLAB to Python
- Supported four Falcon 9 launches by analyzing telemetry for anomalous behavior, operating software on Linux machines, and reporting findings to Air Force engineering

Moog Aerospace, Torrance, CA, Engineering Intern

May 2018-August 2018

- Performed various engineering tasks, including processing ECNs and ECPs, and executed solid model changes, drawing changes, tolerance stack up analysis and tool designs
- Investigated failures to find how parts at risk of being scrapped could potentially comply with customer specifications after modification. Investigated tolerance stack up issues to prevent future failures

LMU Campus Event Operations, Los Angeles CA, Operations Assistant

September 2016-Current

Engineering Projects

Loyola Marymount Aerospace Research Society (LMARS), VP/Lead Engineer

Jan 2018-Present

- Designed and modeled an injector and oxidizer feed system capable of delivering desired mass flow of oxidizer to combustion chambers of subscale and full-scale hybrid rocket motors
- Modeled test stand and modular oxidizer feed system capable of measuring thrust curve of subscale motor used for hybrid motor research at LMU
- Led team transition from solid and hybrid technologies to liquid propulsion, including preliminary research, university relations and recruitment of rocketry experts as mentors
- Initial designer and programmer for LMU Rocket Trajectory Simulation code
- Analyzed combustion properties to determine liquid engine requirements
- Designed suitable annular orifice for pintle injector to prevent flow oscillation in rocket engine
- Responsible for engineering management tasks and team leadership, including system architecture and integration, driving task completion and educating new team members

Skills/Other

CAD: SolidWorks, NX; **Code:** MATLAB (Simulink), Java, Python (Data science, Numerical Methods), Excel, Git, NASA CEA, Linux; First aid; **Interim secret clearance**