#### **Table of Contents**

Data Fetch	2
Local gravitational acceleration	
Accelerometers	
Rate gyros	2
Magnetometers	
Remove bias	2
Initial Pitch, Roll, and Yaw	3
Final Pitch, Roll, Yaw	4
RK4 Integration	5

### **PROBLEM 4**

```
clear variables; close all; clc
```

#### **Data Fetch**

```
data_table_acc = readtable('Accelerometer_Att_est.csv');
data_table_gyro = readtable('Gyroscope_Att_est.csv');
data_table_mag = readtable('Magnetometer_Att_est.csv');
% CORRECT VALUES FOR THESE BIASES AND VARIANCES AS COMPUTED IN
EXPERIMENT 1
bias_acc = [1.948244;
              1.926003;
              -3.76083]; % biases in accelerometer x,y,z
bias_mag = [-12.11214937;
              -19.67616054;
              22.73696197]; % biases in magnetometer x,y,z
bias_gyro = [0.00001194560806;
              -0.00000912316961;
              -0.00000169621783]; % biases in gyro x,y,z
var_acc = [4.94; 5.23; 13.5]*10^-5;
var_gyro = [2.1; 2.6; 4.9]*10^-6;
          = [0.467657; 0.744017; 0.463155];
*----
Warning: Column headers from the file were
modified to make them valid MATLAB
identifiers before creating variable names
for the table. The original column headers
are saved in the VariableDescriptions
property.
Set 'PreserveVariableNames' to true to use
```

```
the original column headers as table
variable names.
Warning: Column headers from the file were
modified to make them valid MATLAB
identifiers before creating variable names
for the table. The original column headers
are saved in the VariableDescriptions
property.
Set 'PreserveVariableNames' to true to use
the original column headers as table
variable names.
Warning: Column headers from the file were
modified to make them valid MATLAB
identifiers before creating variable names
for the table. The original column headers
are saved in the VariableDescriptions
property.
Set 'PreserveVariableNames' to true to use
the original column headers as table
variable names.
```

# Local gravitational acceleration

```
g = 9.80333; % m/s/s
```

### **Accelerometers**

```
time_stamps_acc = data_table_acc{:, 1};
acc_x = data_table_acc{:, 2};
acc_y = data_table_acc{:, 3};
acc_z = data_table_acc{:, 4};
```

# Rate gyros

```
time_stamps_gyro = data_table_gyro{:, 1};
gyro_x = data_table_gyro{:, 2};
gyro_y = data_table_gyro{:, 3};
gyro_z = data_table_gyro{:, 4};
```

# **Magnetometers**

```
time_stamps_mag = data_table_mag{:, 1};
mag_xb = data_table_mag{:, 2};
mag_yb = data_table_mag{:, 3};
mag_zb = data_table_mag{:, 4};
```

### Remove bias

```
acc_x_wo_bias = acc_x - bias_acc(1);
acc_y_wo_bias = acc_y - bias_acc(2);
acc_z_wo_bias = acc_z - bias_acc(3);
```

```
gyro_x_wo_bias= gyro_x - bias_gyro(1);
gyro_y_wo_bias= gyro_y - bias_gyro(2);
gyro_z_wo_bias= gyro_z - bias_gyro(3);
mag_xb_wo_bias = mag_xb - bias_mag(1);
mag_yb_wo_bias = mag_yb - bias_mag(2);
mag_zb_wo_bias = mag_zb - bias_mag(3);
```

# Initial Pitch, Roll, and Yaw

```
initial_few_pts = 5;
roll_data = atan( acc_y_wo_bias(1:initial_few_pts) ./
 acc_z_wo_bias(1:initial_few_pts) );
pitch_data = asin( acc_x_wo_bias(1:initial_few_pts) / g );
initial_roll = mean( roll_data )
disp('rad')
initial_pitch = mean( pitch_data )
disp('rad')
tmp1 = [...
 cos(initial_pitch) sin(initial_pitch)*sin(initial_roll)
 sin(initial_pitch)*cos(initial_roll); ...
 0 cos(initial_roll) -sin(initial_roll); ...
 -sin(initial_pitch) cos(initial_pitch)*sin(initial_roll)
 cos(initial_pitch)*cos(initial_roll)] * ...
 [mag_xb_wo_bias(1:initial_few_pts)'; ...
 mag_yb_wo_bias(1:initial_few_pts)';
 mag_zb_wo_bias(1:initial_few_pts)'];
mag_x_wo_bias = tmp1(1,:)';
mag_y_wo_bias = tmp1(2,:)';
magnetic_heading_data = -atan2( mag_y_wo_bias, mag_x_wo_bias );
              = -14.07*pi/180;
declination
% Declination for Worcester, MA found using World Magnetic Model
% https://www.ngdc.noaa.gov/geomag/calculators/
magcalc.shtml#declination
true_heading_data = declination + magnetic_heading_data;
initial_yaw = mean(true_heading_data)
disp('rad')
initial roll =
    0.0248
rad
initial_pitch =
```

```
-0.1978

rad

initial_yaw =
-1.2761

rad
```

### Final Pitch, Roll, Yaw

```
Final_few_pts = (1185:1194);
roll_data_f = atan( acc_y_wo_bias(Final_few_pts) ./
 acc_z_wo_bias(Final_few_pts) );
pitch_data_f = asin( acc_x_wo_bias(Final_few_pts) / g );
final_roll = mean( roll_data_f )
disp('rad')
final_pitch = mean( pitch_data_f )
disp('rad')
tmp2 = [...
 cos(final_pitch) sin(final_pitch)*sin(final_roll)
 sin(final_pitch)*cos(final_roll); ...
 0 cos(final_roll) -sin(final_roll); ...
 -sin(final_pitch) cos(final_pitch)*sin(final_roll)
 cos(final_pitch)*cos(final_roll)] * ...
 [mag_xb_wo_bias(Final_few_pts)'; ...
 mag_yb_wo_bias(Final_few_pts)'; mag_zb_wo_bias(Final_few_pts)'];
mag_x_wo_bias_f = tmp2(1,:)';
mag_y_wo_bias_f = tmp2(2,:)';
magnetic_heading_data_f = -atan2( mag_y_wo_bias_f, mag_x_wo_bias_f );
               = -14.07*pi/180;
declination
% Declination for Worcester, MA found using World Magnetic Model
% https://www.ngdc.noaa.gov/geomag/calculators/
magcalc.shtml#declination
true_heading_data_f = declination + magnetic_heading_data_f;
final_yaw = mean(true_heading_data_f)
disp('rad')
final roll =
   -0.1675
rad
final_pitch =
```

```
-0.2021

rad

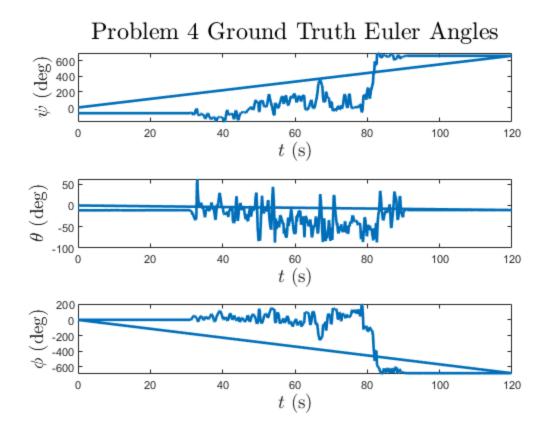
final_yaw =
-1.6215

rad
```

# **RK4 Integration**

```
dt = 0.001;
t = max([time_stamps_acc(1), time_stamps_mag(1),
time_stamps_gyro(1)]);
x_t = [initial_yaw; initial_pitch; initial_roll];
tfinal = min([time_stamps_acc(end), time_stamps_mag(end),
time_stamps_gyro(end)]);
n time pts = round( tfinal /dt );
t_ground_truth_store = zeros(1, n_time_pts);
x_ground_truth_store = zeros(3, n_time_pts);
t_ground_truth_store(1, 1) = t;
x_ground_truth_store(:, 1) = x_t;
column number = 1;
while (t < tfinal)</pre>
 col_gyro = find(time_stamps_gyro <= t, 1, 'last');</pre>
% col_acc = find(time_stamp_acc <= t, 1, 'last');</pre>
% col_mag = find(time_stamp_mag <= t, 1, 'last');</pre>
 u_t = [gyro_x_wo_bias(col_gyro); gyro_y_wo_bias(col_gyro);
 gyro_z_wo_bias(col_gyro)];
 k1 = dt*attitude_kinematics_asg3(x_t, u_t);
 k2 = dt*attitude_kinematics_asg3((x_t + 0.5*k1), u_t);
 k3 = dt*attitude_kinematics_asg3((x_t + 0.5*k2), u_t);
 k4 = dt*attitude_kinematics_asg3((x_t + k3), u_t);
 x_{t} = x_{t} + (1/6)*k1 + (1/3)*k2 + (1/3)*k3 + (1/6)*k4;
 column_number = column_number + 1;
 t = t + dt; % New time
 x_t = x_tplusdt; % x_t is the state at time t
 t_ground_truth_store(1, column_number) = t;
 x_ground_truth_store(:, column_number) = x_t;
end
figure;
```

```
subplot(311)
plot(t ground truth store(1, :), x ground truth store(1, :)*180/
pi, 'LineWidth', 2);
ylabel('$\psi$ (deg)', 'Interpreter', 'latex', 'FontSize', 14)
xlabel('$t$ (s)', 'Interpreter', 'latex', 'FontSize', 14)
title('Problem 4 Ground Truth Euler
Angles', 'Interpreter', 'latex', 'FontSize', 18)
subplot(312)
plot(t_ground_truth_store(1, :), x_ground_truth_store(2, :)*180/
pi, 'LineWidth', 2);
ylabel('$\theta$ (deg)', 'Interpreter', 'latex', 'FontSize', 14)
xlabel('$t$ (s)', 'Interpreter', 'latex', 'FontSize', 14)
subplot(313)
plot(t_ground_truth_store(1, :), x_ground_truth_store(3, :)*180/
pi, 'LineWidth', 2);
ylabel('$\phi$ (deg)', 'Interpreter', 'latex', 'FontSize', 14)
xlabel('$t$ (s)', 'Interpreter', 'latex', 'FontSize', 14)
disp('The result gathered is a valid result as it matches the devices
 orientation. It is')
disp('consistent with results gathered in step 4 with the only
 inconsistancy being the large')
disp('angle measurments due to full rotations during movement')
function x_dot = attitude_kinematics_asg3(x_, u_)
 theta_ = x_{(2)};
phi_{-} = x_{-}(3);
 x_{dot} = [-\sin(theta_{id}) \ 0 \ 1; \dots]
    sin(phi_)*cos(theta_) cos(phi_) 0; ...
    cos(phi_)*cos(theta_) -sin(phi_) 0] \setminus u_;
end
The result gathered is a valid result as it matches the devices
 orientation. It is
consistent with results gathered in step 4 with the only inconsistancy
being the large
angle measurments due to full rotations during movement
```



Published with MATLAB® R2020a