

# Solver Validation with the TUDa compressor

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# Solver introduction

## AeroX:

Multi-block structured mesh on Cartesian coordinate system  
Spalart-Allmaras turbulence model

Spatial discretization:

- Convective fluxes: JST scheme with the scaled numerical dissipation
- Diffusive fluxes: evaluate the gradient of the velocity, temperature and turbulence quantity at cell center using the Gauss's theorem, then use central scheme to get face fluxes

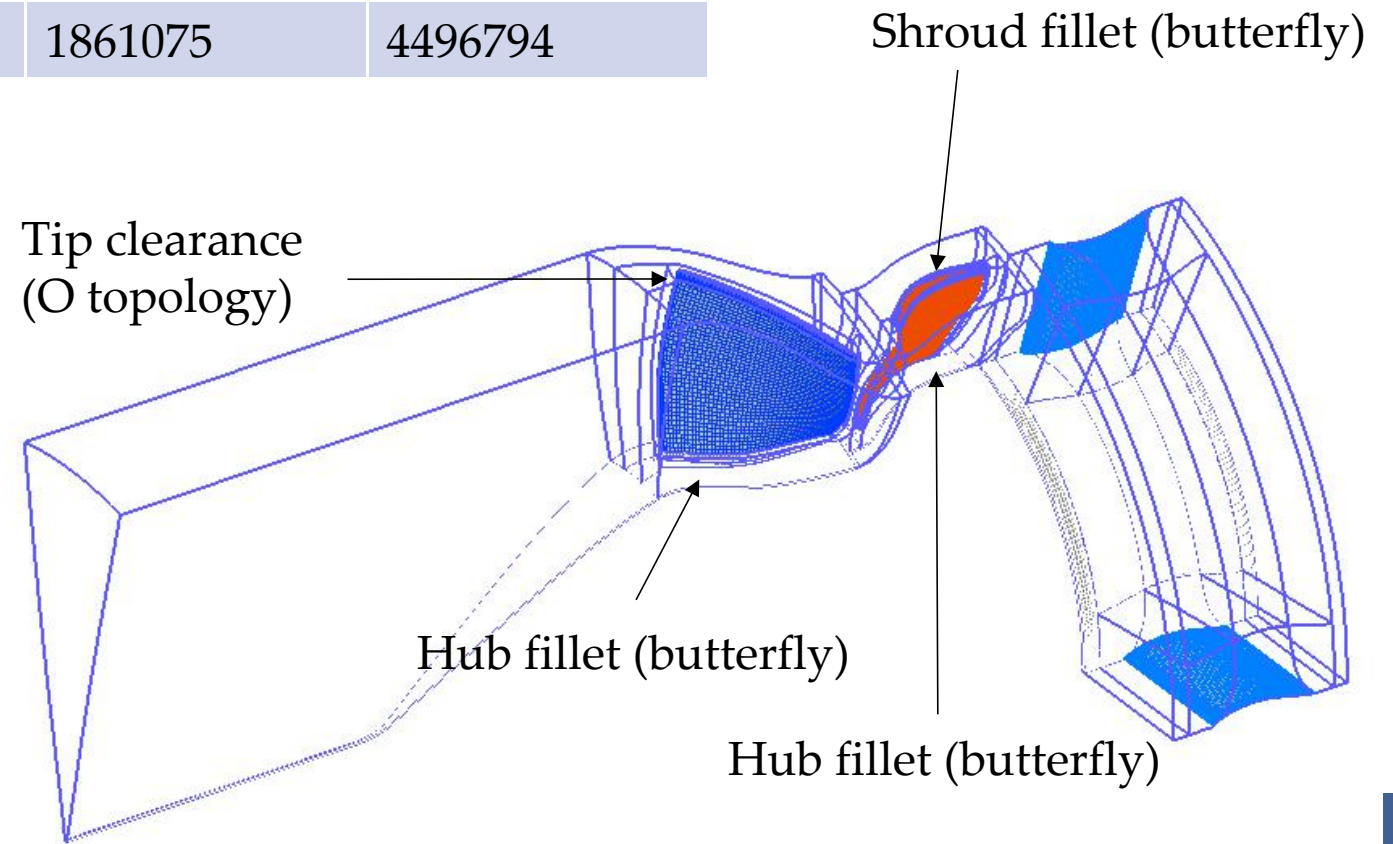
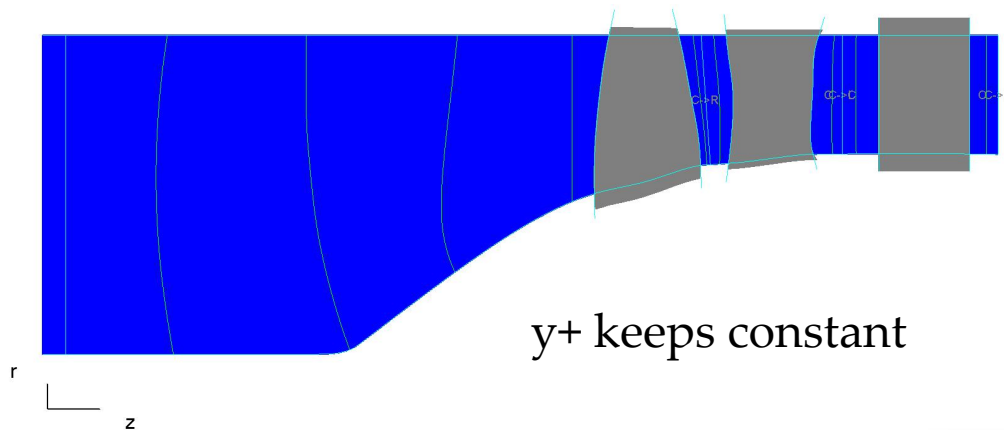
Time integration in pseudo time:

- Local time stepping
- Hybrid method combining explicit five stage Runge-Kutta method and implicit LU-SGS method

# Mesh generation

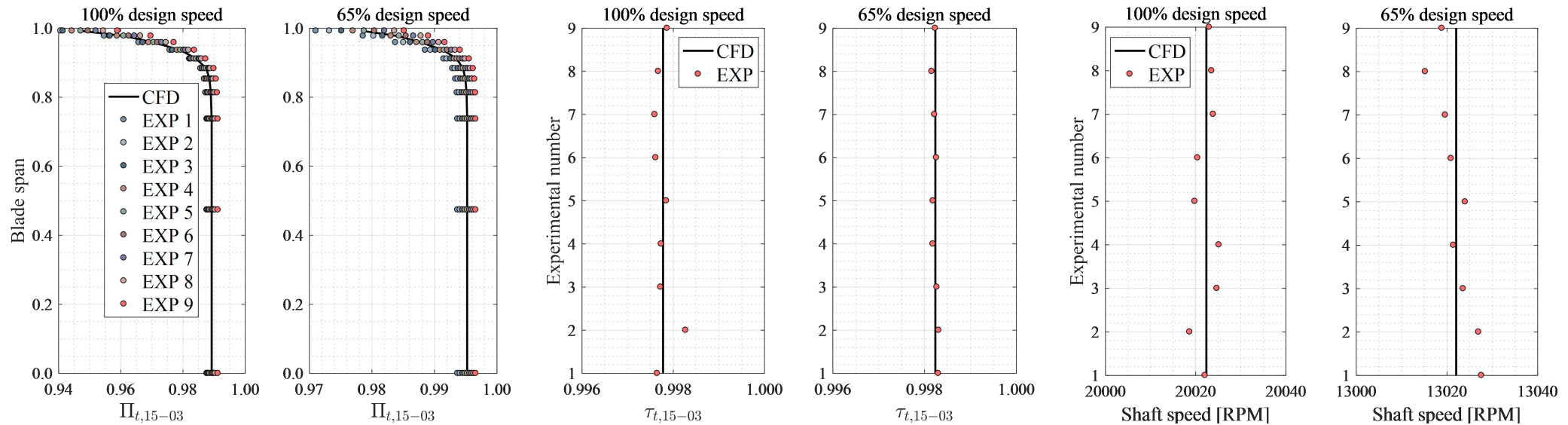
| Mesh   | Row1    | Row2   | Row3    | Total   |
|--------|---------|--------|---------|---------|
| Coarse | 444875  | 260227 | 484071  | 1189173 |
| Medium | 874607  | 576715 | 899795  | 2351117 |
| Fine   | 1646583 | 989139 | 1861075 | 4496794 |

The number of grid points is twice as much as the coarser mesh



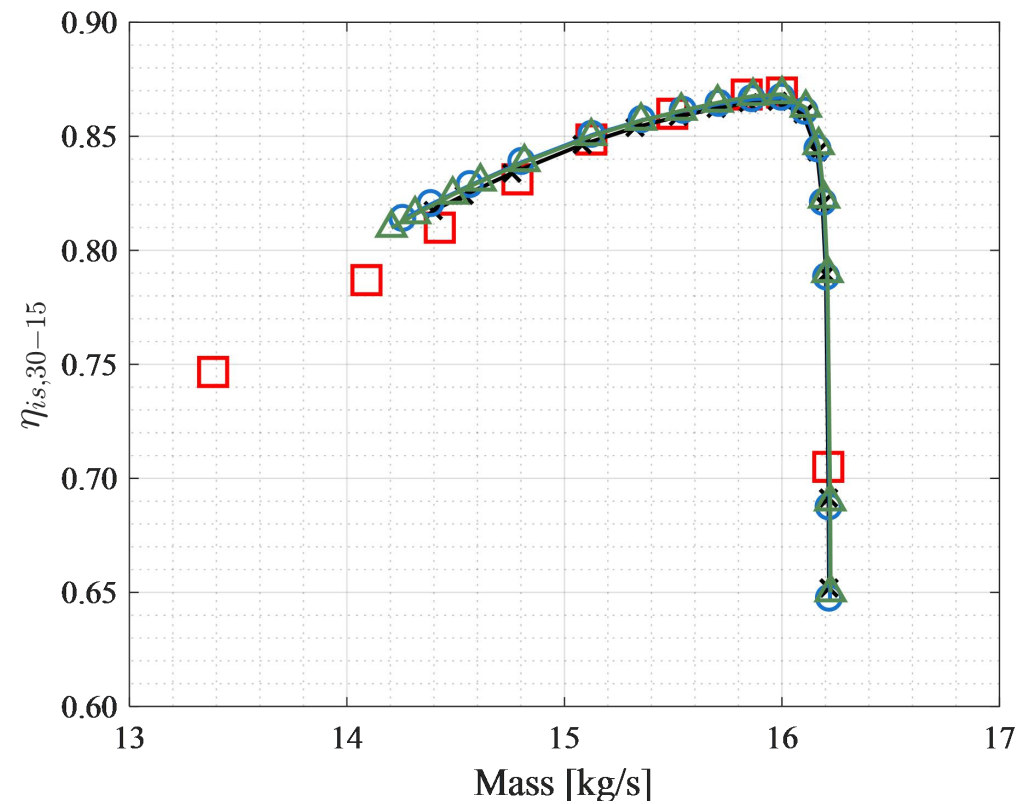
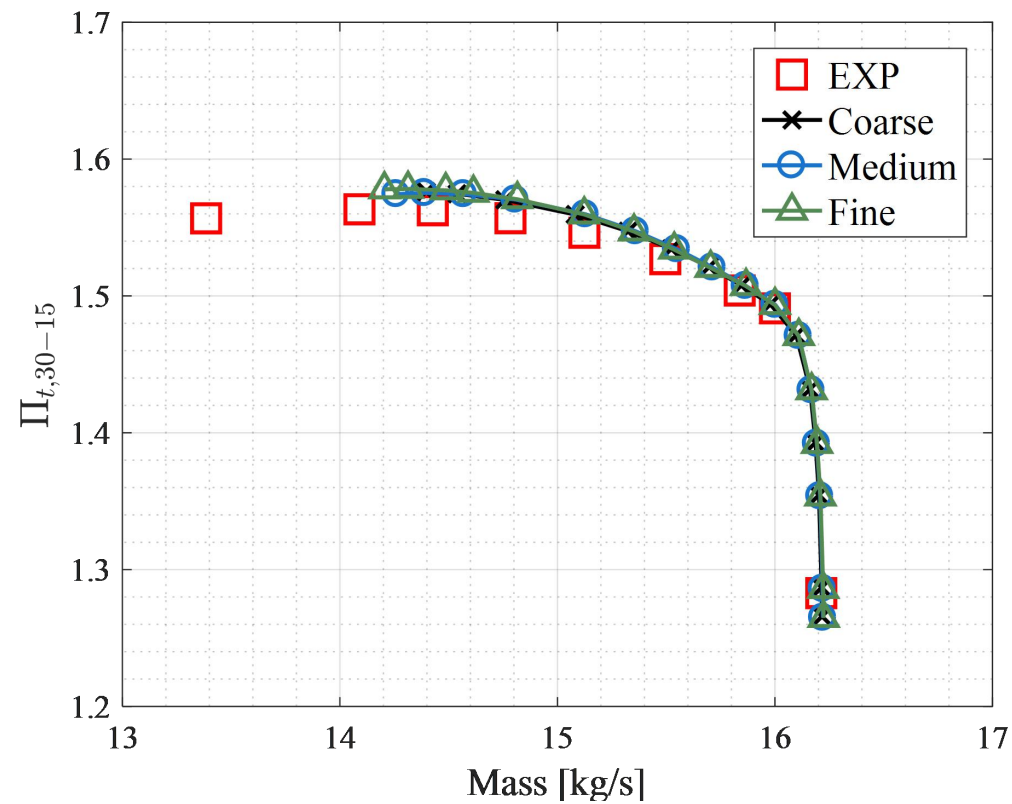
# Inlet boundary settings

Inlet boundary (ME15) conditions: converted to ideal gas model: 101325Pa/288.15K



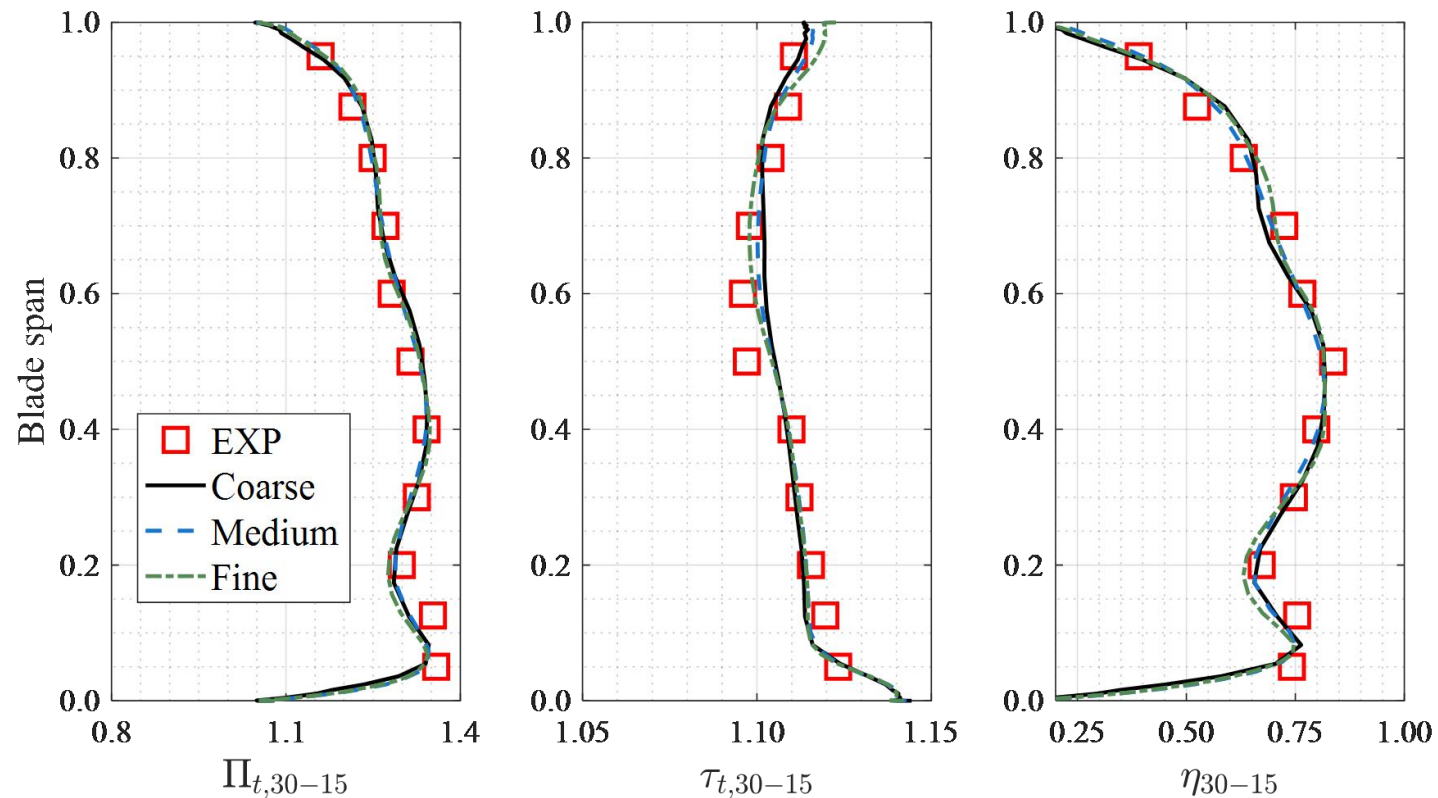
Performance map is obtained by increasing the back pressure

# Performance map – 100% design speed



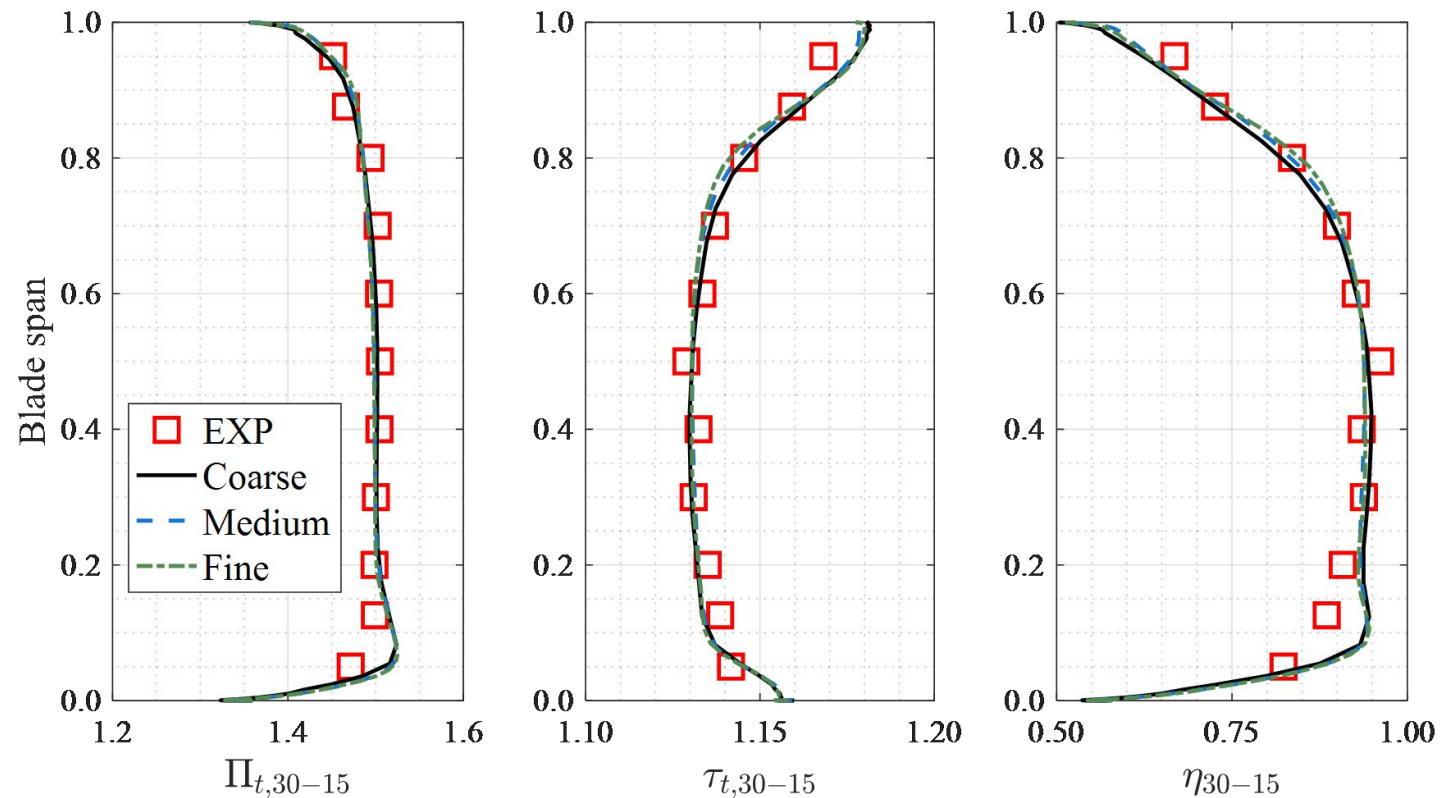
# Radial profile (ME30 stator exit 100% speed)

| Method | Mass    | $\Pi_{t,30-15}$ | $\eta_{is,30-15}$ |
|--------|---------|-----------------|-------------------|
| EXP    | 16.2130 | 1.2828          | 0.7051            |
| Coarse | 16.2143 | 1.2870          | 0.6914            |
| Medium | 16.2191 | 1.2861          | 0.6873            |
| Fine   | 16.2219 | 1.2857          | 0.6899            |



# Radial profile (ME30 stator exit 100% speed)

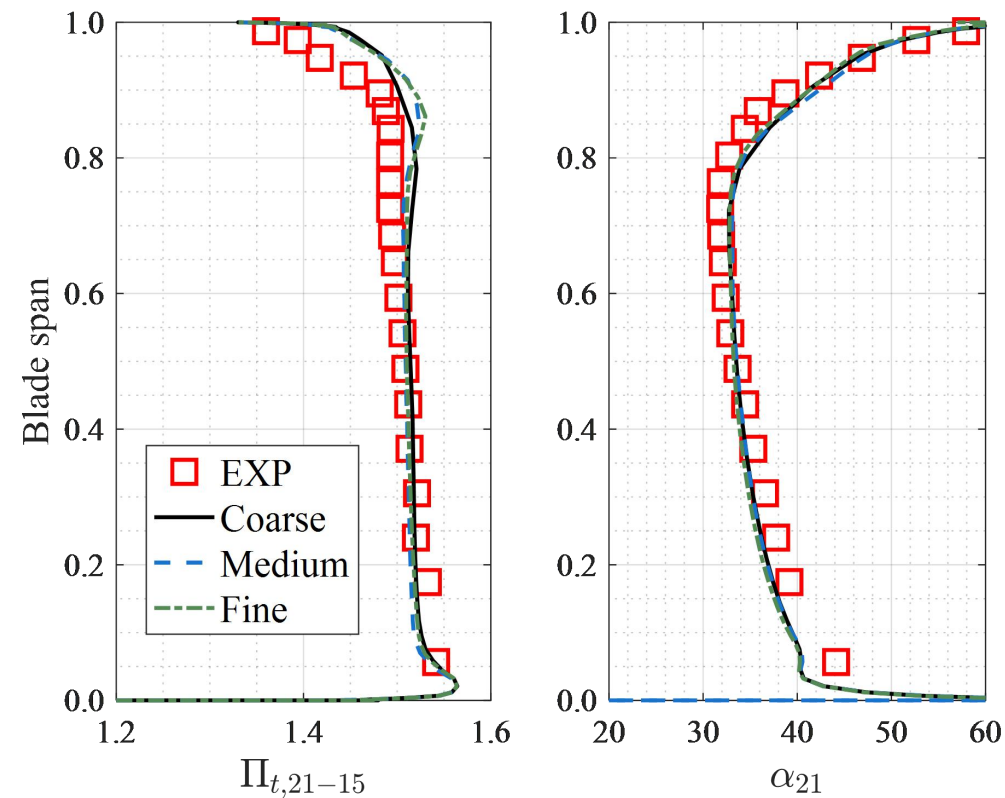
| Method | Mass  | $\Pi_{t,30-15}$ | $\eta_{is,30-15}$ |
|--------|-------|-----------------|-------------------|
| EXP    | 16.00 | 1.4949          | 0.8691            |
| Coarse | 15.98 | 1.4939          | 0.8653            |
| Medium | 16.00 | 1.4936          | 0.8669            |
| Fine   | 16.00 | 1.4933          | 0.8684            |





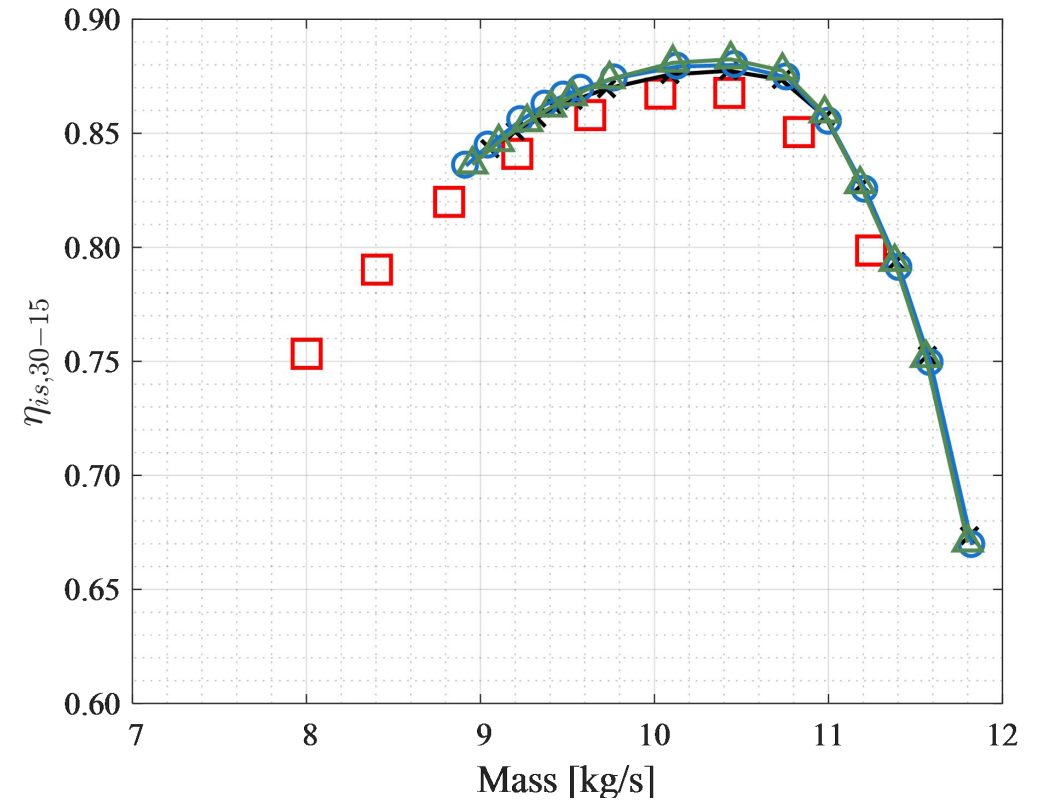
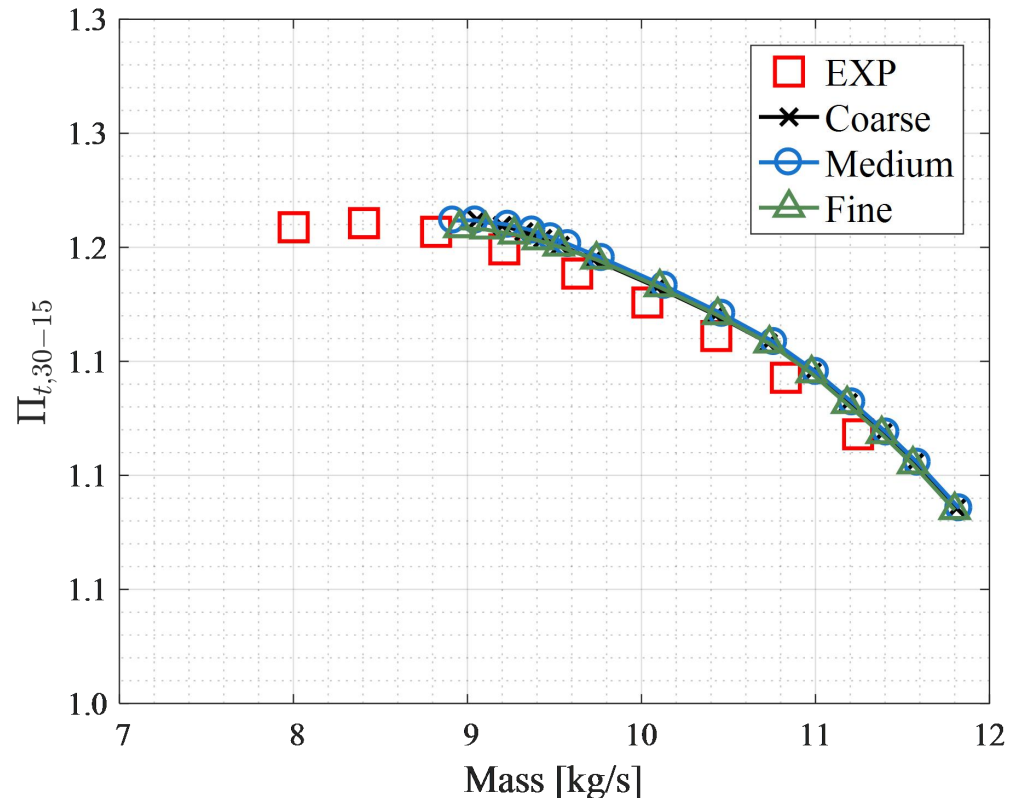
# Radial profile (ME21 rotor exit 100% speed)

| Method | Mass  | $\Pi_{t,30-15}$ | $\eta_{is,30-15}$ |
|--------|-------|-----------------|-------------------|
| EXP    | 16.00 | 1.4949          | 0.8691            |
| Coarse | 15.98 | 1.4939          | 0.8653            |
| Medium | 16.00 | 1.4936          | 0.8669            |
| Fine   | 16.00 | 1.4933          | 0.8684            |



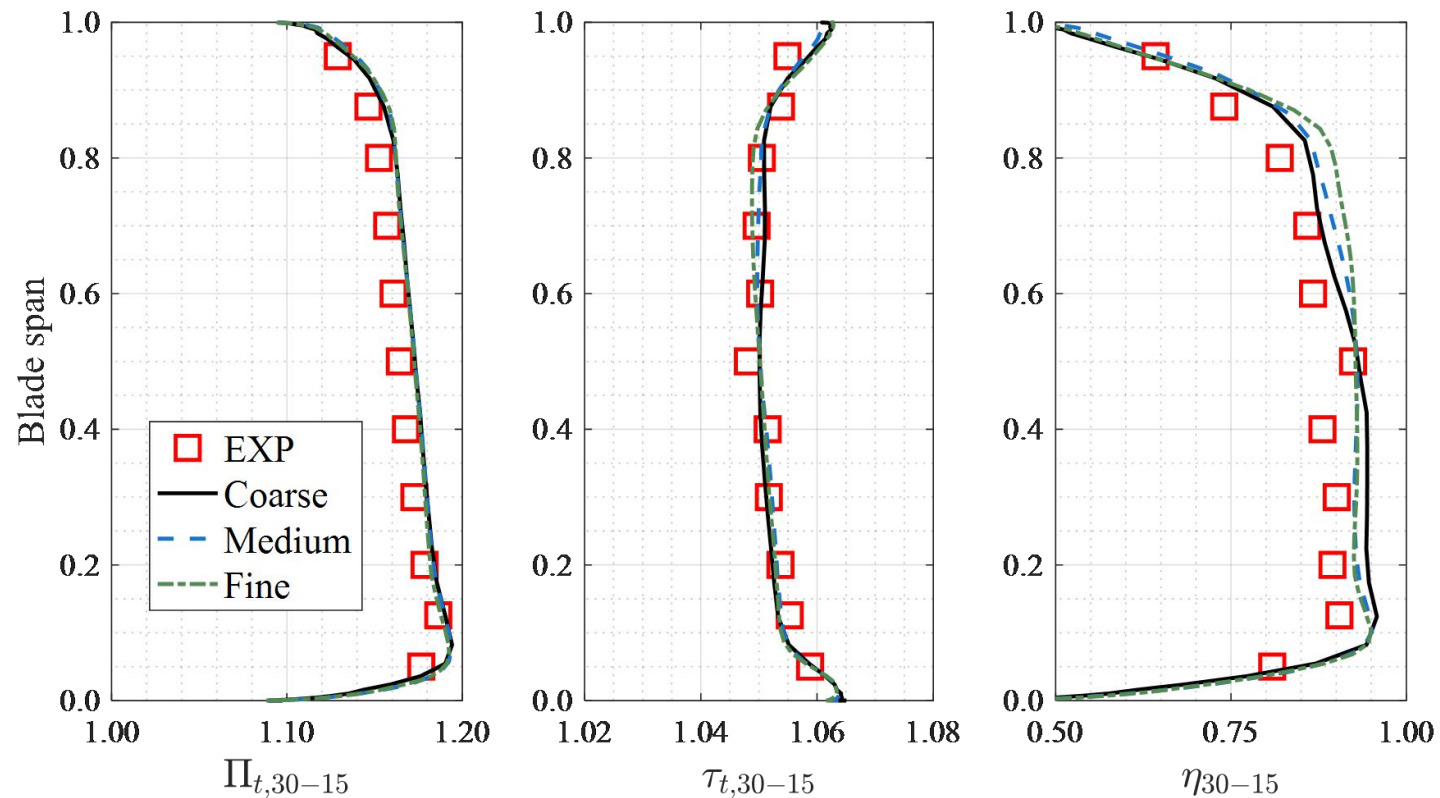


# Performance map – 65% design speed



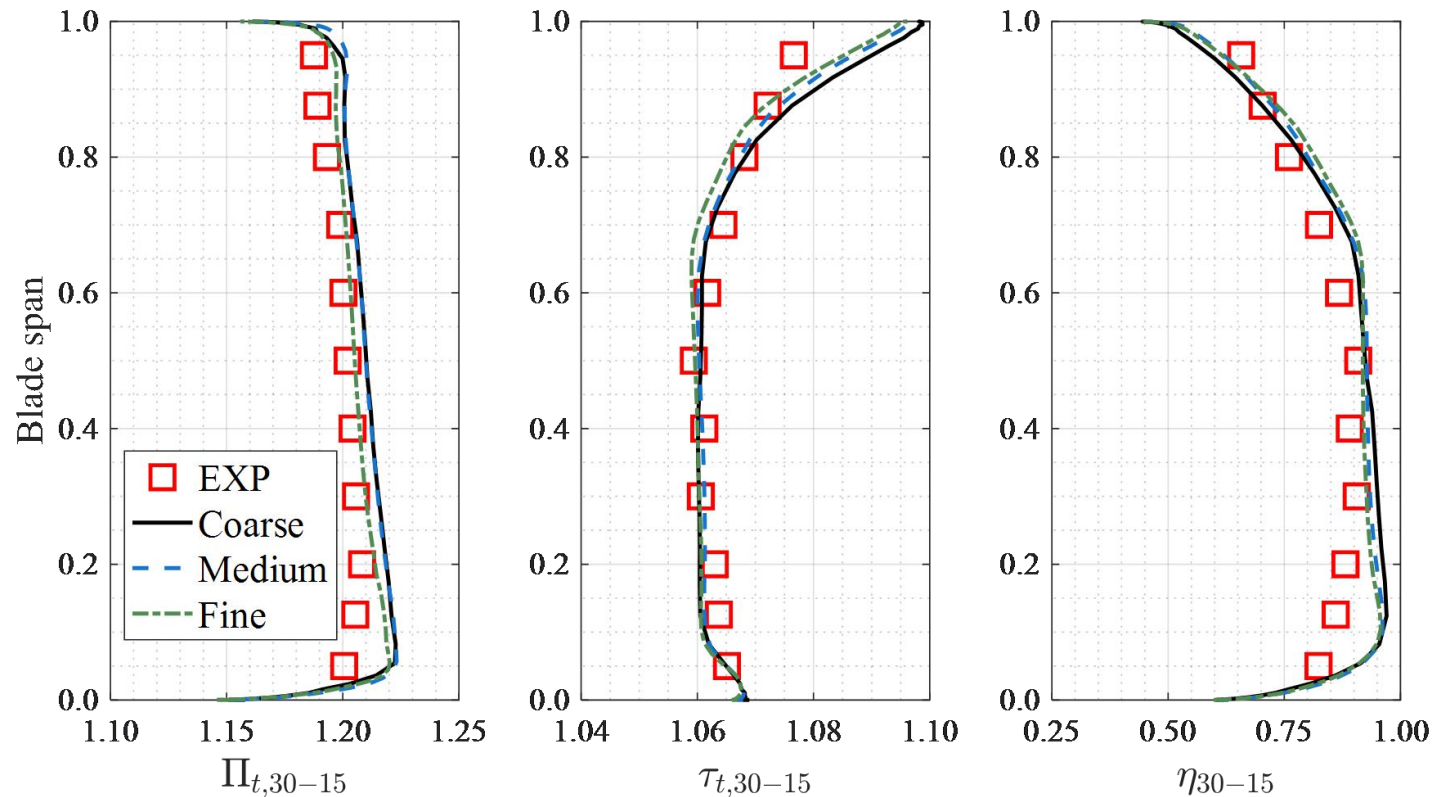
# Radial profile (ME30 stator exit 65% speed)

| Method | Mass  | $\Pi_{t,30-15}$ | $\eta_{is,30-15}$ |
|--------|-------|-----------------|-------------------|
| EXP    | 10.43 | 1.1611          | 0.8677            |
| Coarse | 10.42 | 1.1705          | 0.8774            |
| Medium | 10.46 | 1.1709          | 0.8800            |
| Fine   | 10.44 | 1.1702          | 0.8824            |



# Radial profile (ME30 stator exit 65% speed)

| Method | Mass  | $\Pi_{t,30-15}$ | $\eta_{is,30-15}$ |
|--------|-------|-----------------|-------------------|
| EXP    | 9.212 | 1.1991          | 0.8410            |
| Coarse | 9.200 | 1.2096          | 0.8508            |
| Medium | 9.234 | 1.2099          | 0.8558            |
| Fine   | 9.268 | 1.2056          | 0.8549            |



# Conclusion

## □ TUDa compressor

- The solutions by the three in-house meshes show a very low grid dependency for both the performance map and the radial profiles, at both the 100% speed and 65% speed
- The solution predicted by AeroX has a good quantitative agreement with the experimental data in general

THANKS FOR YOUR ATTENTION