Workshop SciTech2024 Adaptive grid convergence study on the Joukovski airfoil

M. Visonneau, E. Guilmineau & J. Wackers



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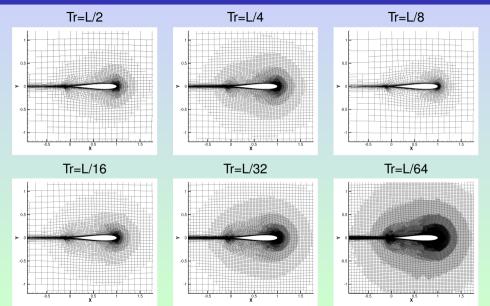
Introduction

- ► ISIS-CFD, our in-house flow solver (aka FINE/Marine, Cadence Design System) is used for this 2D verification study around the Joukovski airfoil:
 - Spalart-Allmaras model with R rotation correction and QCR2000 non-linear addition (SA-R-QCR2000) with SA-neg formulation and ICCFD7 2012 modifications,
 - Anisotropic adaptive grid refinement based on the Flux-Component Hessian using a series of 11 grids, starting from an initial built with Hexpress,
- Systematic grid convergence based on a linear reduction of the threshold (Tr ranging from L/2 to L/64),
- Estimate of discretisation error based on Eca & Hoekstra procedure.

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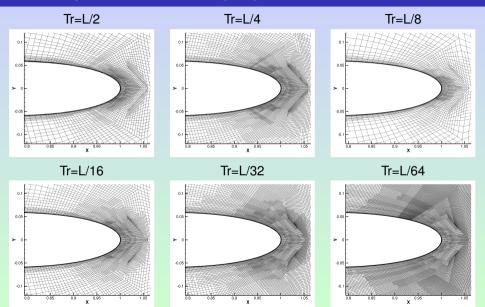
Series of adapted meshes - Global view



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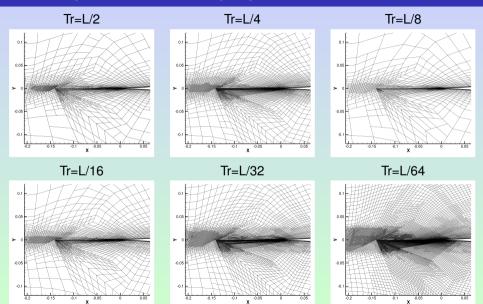
Series of adapted meshes - Leading edge



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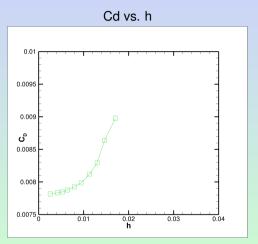
Series of adapted meshes - Trailing edge

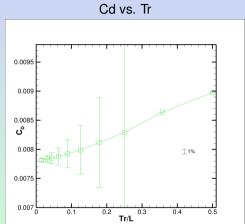


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Grid convergence of the drag





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Comparison with converged values mentioned in the paper

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Cd	Cd	Cdv	Cdv	Cdp	Cdp
(SANS)	(ISIS-CFD)	(SANS)	(ISIS-CFD)	(SANS)	(ISIS-CFD)
0.007872	0.007816	0.006627	0.006652	0.001244	0.001164

- First numerical study around the Joukovski airfoil with an initial grid built with Hexpress:
- Regular convergence wrt. the threshold,
- Converged values are not in perfect agreement with the ones obtained with SANS,
- Estimated order of accuracy: 1.3 (not a rigourous estimate with a series of adapted grids),
- Next step 1: perform the same study with the structured grids proposed in this collaborative action,
- Next step 2: Start from a more regular grid e.g. a coarse structured grid taken from this series.

Thank you for your attention!

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