$$a = \begin{bmatrix} x \\ y \end{bmatrix}$$

$$R = \begin{bmatrix} 1 & 0 & 0 \\ 0 & G_{11}N_{2} & -S_{11}N_{2} \\ 0 & S_{11}N_{2} & G_{21}N_{2} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & -1 \\ 0 & 1 & 0 \end{bmatrix}$$

$$S(a) = \begin{bmatrix} 0 & -z & 0y \\ z & 0 & -x \\ -y & x & 0 \end{bmatrix}, \quad R^{T} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & -1 & 0 \end{bmatrix}$$

$$R^{T} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & -1 & 0 \end{bmatrix}$$

$$Ra = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & -1 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x \\ y \\ z \end{bmatrix} = \begin{bmatrix} x \\ -z \\ y \end{bmatrix}$$

$$\Rightarrow S(Ra) = \begin{bmatrix} 0 & -y - z \\ y & 0 & -x \\ +z & x & 0 \end{bmatrix}$$

Now

$$R S(a) R^{T} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & -1 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} 0 & -z & ty \\ z & 0 & -x \\ -y & x & 0 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & -1 \\ 0 & -1 & 0 \end{bmatrix}$$

$$= \begin{bmatrix} 0 & -y & -Z \\ +y & 0 & -x \\ Z & x & 0 \end{bmatrix}$$

For rotation of
$$8/1$$
 and 4700

$$P_1^0 = \begin{cases} c\theta_1 & -s\theta_1 & 0 & l_1c\theta_1 \\ s\theta_2 & c\theta_2 & 0 & l_1s\theta_1 \\ 0 & 0 & 0 & 1 \end{cases}$$

For rotation of
$$\sqrt{2}$$
 and translation by l_2

$$P_2 = \begin{bmatrix} CO_2 & -SO_2 & 0 & l_2 CO_2 \\ SO_1 & CO_2 & 0 & l_2 SO_2 \\ 0 & 0 & 1 & 0 \end{bmatrix}$$

Stanford type RRP 1 - Transformation from base to joint ?: To = Rotation (61) * translation (201) $= \begin{bmatrix} c0_1 & -30_1 & 0 & 0 \\ s0_1 & c0_1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \times \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$ T' = Rotation (x, N/2) * Rotation (Z, 02) * tran(x,d) $= \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & -1 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} * \begin{bmatrix} 0.02 & -502 & 0.007 \\ 502 & 0.02 & 0.007 \\ 0 & 0 & 0.017 \\ 0 & 0 & 0.017 \end{bmatrix}$ 73 = translation (x, d3)

Thus To = T, * T' * T'3

O since drone travels straight along z-axis

2) 30° rotation about x-axis

$$T_{2}^{1} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & \cos 30^{\circ} & -\sin 30^{\circ} & 0 \\ 0 & \sin 30^{\circ} & \cos 30^{\circ} & 0 \end{bmatrix}$$

3 60° votation about new z-axis

$$T_{3}^{2} = \begin{cases} Con 60^{\circ} - 8m60^{\circ} & 0 & 0 \\ Som 60^{\circ} & Con 60^{\circ} & 0 & 0 \\ 0 & 0 & 0 & 1 \end{cases}$$

 $T_3 = T_1^0 * T_2^1 * T_3^2$

$$T_3^0 = \begin{bmatrix} 0.86 & -0.5 & 0 & 0 \\ 0.43 & 0.75 & -0.5 & 0 \\ 0.25 & 0.43 & 0.86 & 10 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

Dobstacle's position (0,0,3,1) in base

$$P_{0} = T_{3}^{0} \begin{bmatrix} 0 \\ 0 \\ 3 \end{bmatrix} = \begin{bmatrix} -1.5 \\ 12.598 \end{bmatrix}$$

O Object. 20

of gear box are utilized: 1) Planetary searbox offers high torque trasmission and compact sixe, making it suitable where space is limited, Such as a sobotie arm, However it has some backlash error issuel. (2) harmonic drive gearbox offers high precision and minimal backlash. it is suitable in applications requiring accurate positioning. These gearbox are very expensive.

In drove application, a gearbox is typically not used with a motor due to emphasis on weight reduction. Gees add weight and complexity which has negative impact on drove performance.

2 Dels (x, y, z) Endpoint.

Z SCARA RRP

Y SCARA RRP

From the given diagram,

the coordinates of the end effector

from boxe frame orre:

1 coso, + l2 con (0, +02)

y - l, sino, + lz sin (0, +02)

Z = -d.

Now,

 $\frac{\partial x}{\partial \theta_1} = -l_1 \sin \theta_1 - l_2 \sin (\theta_1 + \theta_2)$

 $\frac{\partial x}{\partial \theta_2} = -l_2 \cdot Sm(\theta_1 + \theta_2)$

0× 0

2 24 = Ri Con 8, + l2 Con (0, + 02)

24 = 12 Cos (0, +02)

 $\frac{\partial y}{\partial z'} = 0$

$$\frac{\partial z}{\partial \theta_{1}} = 0$$

$$\frac{\partial z}{\partial \theta_{2}} = 1$$

$$\frac{\partial x}{\partial \theta_{1}} = 0$$

$$\frac{\partial x}{\partial \theta_{2}} = 0$$

$$\frac{\partial x}{\partial \theta_{1}} = 0$$

$$\frac{\partial x}{\partial \theta_{2}} = 0$$

$$\frac{\partial x}{\partial \theta_{1}} = 0$$

$$\frac{\partial x}{\partial \theta_{2}} = 0$$

$$\frac{\partial$$

$$X = l_1 co_1 + l_2 c(o_1 + o_2) + l_3 c(o_1 + o_2 + o_3)$$

$$Y = l_1 so_1 + l_2 s(o_1 + o_2) + l_3 s(o_1 + o_2 + o_3)$$

$$\frac{\partial x}{\partial \theta_{1}} = -\int_{0}^{1} S\theta_{1} - \int_{0}^{1} S(\theta_{1} + \theta_{2}) - \int_{0}^{1} S(\theta_{1} + \theta_{2} + \theta_{3})$$

$$\frac{\partial x}{\partial \theta_{2}} = -\int_{0}^{1} S(\theta_{1} + \theta_{2}) - \int_{0}^{1} S(\theta_{1} + \theta_{2} + \theta_{3})$$

$$\frac{\partial^{x}}{\partial \theta_{3}} = -l_{3}b(0, +0_{1}+0_{1})$$