

Instability of Flow In Magnetic Nozzle

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Outline of Presentation

- 1 Introduction
- 2 Spectral Method
- 3 Singular Perturbation
- 4 Future Work
- 5 Appendix: Numerical Experiments

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Instability of Plasma Flow

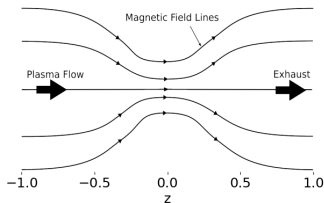
- The instability of plasma flow refers to the tendency of a plasma system to deviate from a stable, equilibrium state and exhibit perturbations or fluctuations in its behavior. [1]

To investigate instability, we assume oscillating perturbed quantities ,
 $\tilde{n}, \tilde{v} \sim \exp(-i\omega t)$.

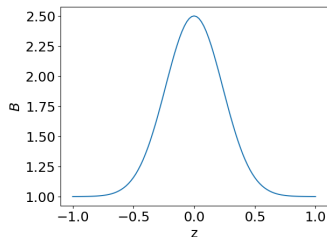
- ① If $\text{Im}(\omega) > 0$, then it is unstable flow since the perturbations grow exponential in time, $\exp(\text{Im}(\omega)t)$.
- ② If $\text{Im}(\omega) \leq 0$, then it is stable flow since the perturbations decay/unchanged in time.

Magnetic Nozzle

- A magnetic nozzle is a device that uses a magnetic field to shape and control the flow of charged particles in a plasma propulsion system.
- Instabilities may affect magnetic nozzle operation and the resulting thrust. [4]



(a) Simplified representation of magnetic nozzle.



(b) A simplified magnetic field of magnetic nozzle.

Figure 1: Simplified representation of magnetic nozzle. Length is normalized.

Governing Equations

The nondimensionalized governing equations for the plasma flow in magnetic nozzle are

$$\text{Cons. of Den.} \quad \frac{\partial n}{\partial t} + n \frac{\partial v}{\partial z} + v \frac{\partial n}{\partial z} - nv \frac{\partial_z B}{B} = 0 \quad (1)$$

$$\text{Cons. of Mom.} \quad n \frac{\partial v}{\partial t} + nv \frac{\partial v}{\partial z} = - \frac{\partial n}{\partial z} \quad (2)$$

where n, v are density and velocity, respectively.

The equilibrium quantities n_0, v_0 must satisfy the condition,

$$\frac{\partial}{\partial z} \left(\frac{n_0 v_0}{B} \right) = 0 \quad (3)$$

$$v_0 \frac{\partial v_0}{\partial z} = - \frac{1}{n_0} \frac{\partial n_0}{\partial z} \quad (4)$$

Equilibrium Velocity Profiles

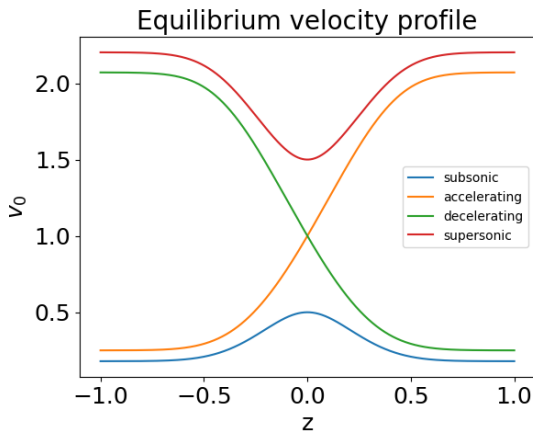


Figure 2: Velocity if normalized to sound speed. There are 4 different cases for velocity profile, subsonic, supersonic, accelerating and decelerating case.

Polynomial Eigenvalue Problem

By linearizing the governing equations, and assume oscillating perturbed quantities, $\tilde{n}, \tilde{v} \sim \exp(-i\omega t)$. We can derive the following equation,

$$\begin{aligned} & \omega^2 \tilde{v} \\ & + 2i\omega \left(v_0 \frac{\partial}{\partial z} + \frac{\partial v_0}{\partial z} \right) \tilde{v} \\ & + \left[(1 - v_0^2) \frac{\partial^2}{\partial z^2} - \left(3v_0 + \frac{1}{v_0} \right) \frac{\partial v_0}{\partial z} \frac{\partial}{\partial z} \right. \\ & \quad \left. - \left(1 - \frac{1}{v_0^2} \right) \left(\frac{\partial v_0}{\partial z} \right)^2 - \left(v_0 + \frac{1}{v_0} \right) \frac{\partial^2 v_0}{\partial z^2} \right] \tilde{v} = 0 \end{aligned} \tag{5}$$

It is a polynomial eigenvalue problem.

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Spectral Method

Eq.(5) can be reformulate as

$$\begin{bmatrix} 0 & 1 \\ \hat{M} & \hat{N} \end{bmatrix} \begin{bmatrix} \tilde{v} \\ \omega \tilde{v} \end{bmatrix} = \omega \begin{bmatrix} \tilde{v} \\ \omega \tilde{v} \end{bmatrix} \quad (6)$$

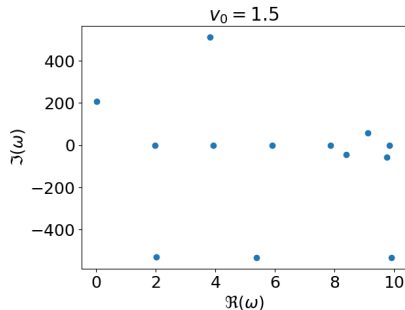
where the operators \hat{M} and \hat{N} are defined as

$$\begin{aligned} \hat{M} = & - \left[(1 - v_0^2) \frac{\partial^2}{\partial z^2} - \left(3v_0 + \frac{1}{v_0} \right) \frac{\partial v_0}{\partial z} \frac{\partial}{\partial z} \right. \\ & \left. - \left(1 - \frac{1}{v_0^2} \right) \left(\frac{\partial v_0}{\partial z} \right)^2 - \left(v_0 + \frac{1}{v_0} \right) \frac{\partial^2 v_0}{\partial z^2} \right] \\ \hat{N} = & -2i \left(v_0 \frac{\partial}{\partial z} + \frac{\partial v_0}{\partial z} \right) \end{aligned}$$

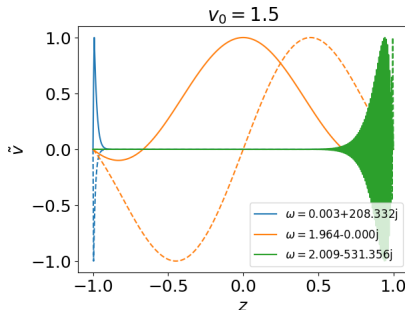
Then by discretizing operators \hat{M}, \hat{N} , this becomes an algebraic eigenvalue problem.

Spectral Pollution

- All modes of Eq.(5) with $v_0 = \text{const}$ are stable.
- Yet all discretization show unstable modes.



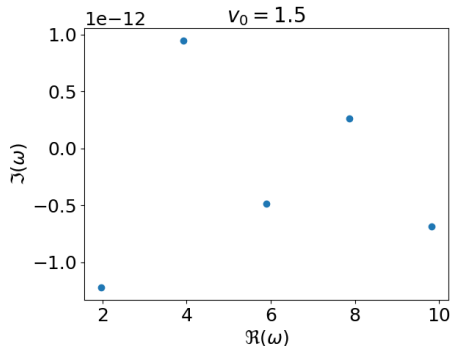
(a) Unfiltered eigenvalues.



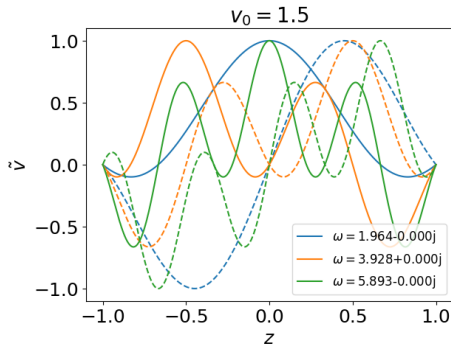
(b) First few unfiltered eigenfunctions.

Figure 3: Finite difference discretization was used. Spurious modes occurs regardless of the resolution.

Filtering Spurious Modes



(a) Filtered eigenvalues.



(b) First few filtered eigenfunctions.

Figure 4: The spurious modes are changing under different resolution. We can filter them by convergence test.

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Existence of Singularity

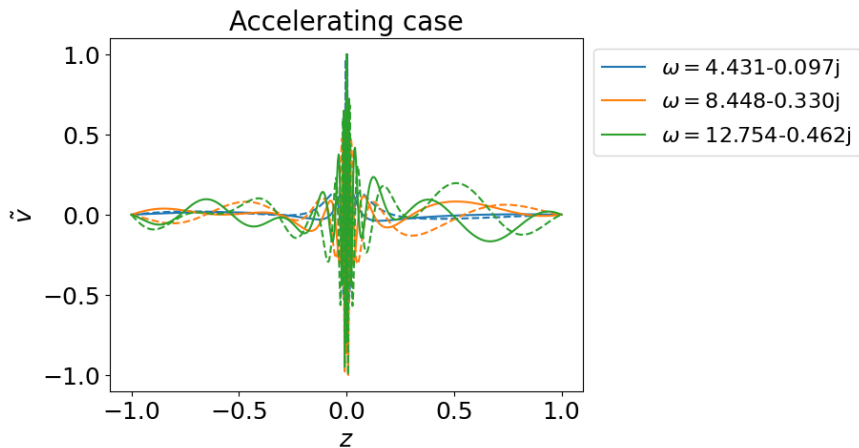


Figure 5: Dirichlet boundary conditions are set at the two ends, all eigenfunctions are squeezed to the singular point.

Interesting Connection to Black Hole

- The sonic horizon is an exact sonic analogue of black hole horizon. [5]
- A quasi-1D fluid flow is ruled by [2]

$$\frac{\partial}{\partial t}(\rho A) + \frac{\partial}{\partial x}(\rho A v) = 0 \quad (7)$$

$$\frac{\partial}{\partial t}(\rho A v) + \frac{\partial}{\partial x}[(\rho v^2 + p)A] = 0 \quad (8)$$

$$\frac{\partial}{\partial t} \left(\frac{\rho v^2}{2} - \frac{p}{1-\gamma} A \right) + \frac{\partial}{\partial x} \left[\left(\frac{\rho v^2}{2} - \frac{\gamma}{1-\gamma} A \right) A v \right] = 0 \quad (9)$$

In [2, 3], the acoustic analogue of tortoise coordinate is used to transform the above to Schrödinger-type equation,

$$x^* = c_{s0} \int [c_s(x)(1 - M(x)^2)]^{-1} dx$$

where c_{s0} denotes the stagnation speed of sound, $c_s = dp/d\rho$ is the local speed of sound and $M(x) = v(x)/c_s(x)$ is the Mach number.

Shooting Method

- Shooting method is employed.
- Initial values are obtained by expanding solution at the singularity.

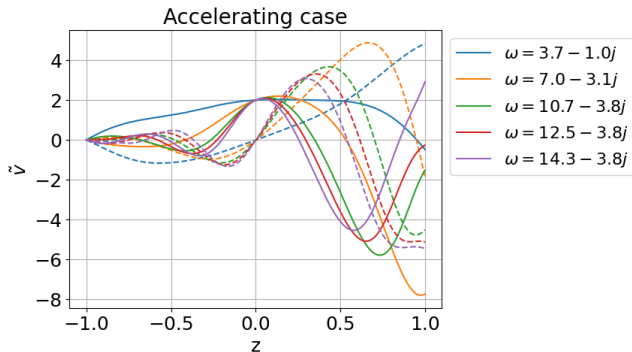


Figure 6: The solutions cross the singular point smoothly. All modes are stable.

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Future Work

- Investigate and interpret the instability of an accelerating flow with non-zero left boundary. See Fig.7
- Compare results to analytically solvable problems with similar configuration.

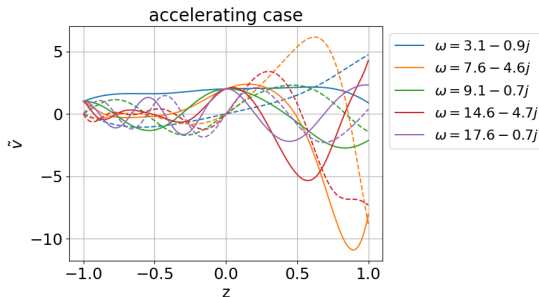


Figure 7: What is the physical interpretation of "non-zero" boundary value? How do we interpret these eigenvalues?



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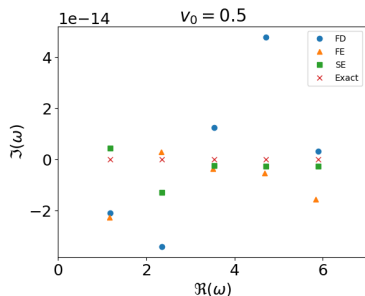
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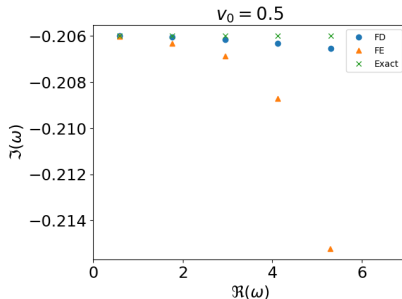
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Constant Velocity Case - Subsonic



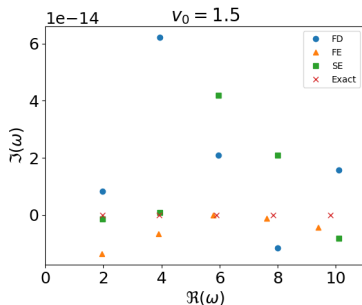
(a) Dirichlet boundary, all modes are stable.



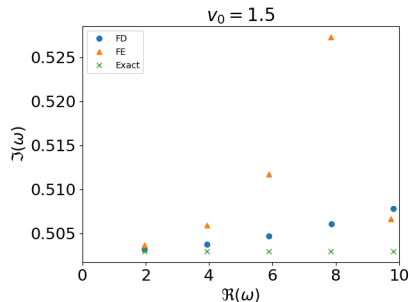
(b) Fixed-open boundary, all modes are stable.

Figure 8: Showing the first 5 eigenvalues. In the Dirichlet boundary case, all methods are close to the exact eigenvalues. Meanwhile, finite-difference method has higher accuracy than finite-element method in fixed-open case.

Constant Velocity Case - Supersonic



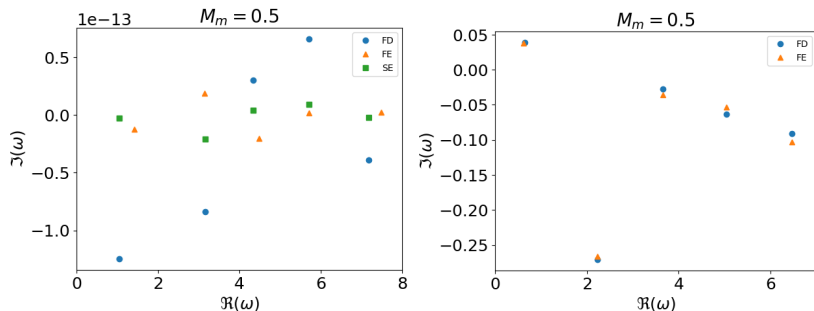
(a) Dirichlet boundary, filtered modes are stable.



(b) Fixed-open boundary, all modes are unstable.

Figure 9: Showing the first 5 eigenvalues. In the Dirichlet boundary case, all methods are close to the exact eigenvalues. Meanwhile, finite-difference method has higher accuracy than finite-element method in fixed-open case.

Subsonic Case

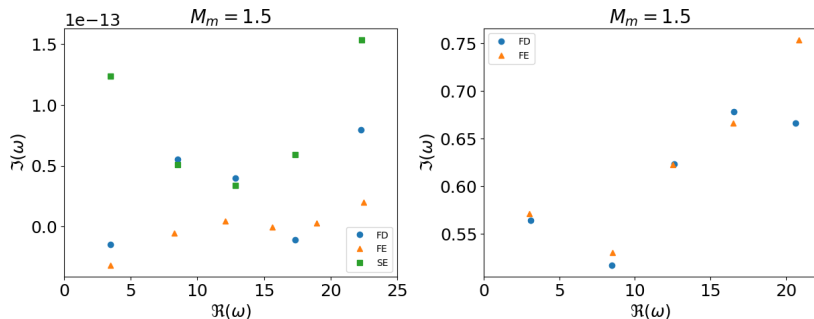


(a) Dirichlet boundary, all modes are stable.

(b) The ground mode is unstable, other modes are stable.

Figure 10: Showing the first 5 modes. It suggests that the subsonic flow in magnetic nozzle is stable.

Supersonic Case



(a) Dirichlet boundary, filtered modes are stable.

(b) Fixed-open boundary, all modes are unstable.

Figure 11: This suggests that the supersonic flow is stable if the boundary is Dirichlet and unstable if the boundary is left-fixed-right-open.

Accelerating Case

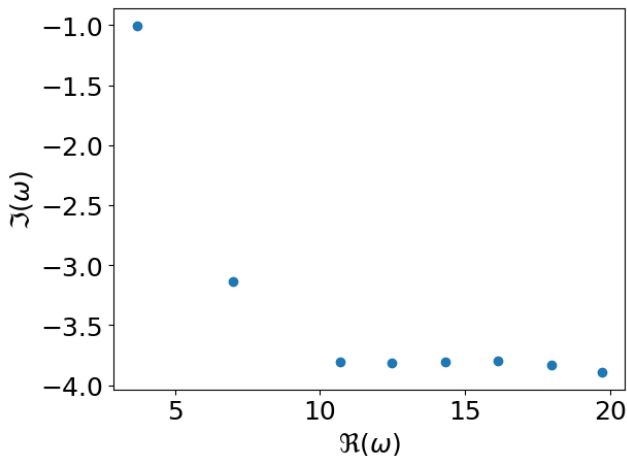


Figure 12: All modes are stable.