Contents

| 1 | Aer | roshell | | | | 1 |
|---|------------------------|------------------------------|--------------------|---------------|--------------|------|
| | 1.1 | | INITIAL | IZE | | . 1 |
| | | setup | SETUP | | | |
| | | compute | | TE | | |
| | 1.4 compute partials | | COMPUTE_PARTIALS 3 | | | . 3 |
| 2 | $\mathbf{W}\mathbf{h}$ | eels | | | | 4 |
| | 2.1 | initialize | INITIAL | IZE | | . 4 |
| | 2.2 | setup | SETUP | | | |
| | 2.3 | 1 | | TE | | |
| | | 2.3.1 Circumferential Stress | | | | . 8 |
| 1 | A | eroshell | | | | |
| _ | | | Unita | Variable Name | | |
| | | Mass Quantity | Units | Mass | ; | |
| | | Area. | | Area | | |
| | | Lift Coefficient | | LiftCoeffici | ent. | |
| | | Drag Coefficient | | DragCoeffici | | |
| | | O . | | O | | |
| | - | penmdao.api import Explici | tCompone | ent | | |
| | - | numpy as np | | | | |
| cl | | AeroShell(ExplicitComponen | t): | | | |
| | | "Aeroshell of the pod""" | | | | |
| | | AeroShell_initialize>> | | | | |
| < <aeroshell_setup>> <<aeroshell_compute>> <<aeroshell_compute_partials>></aeroshell_compute_partials></aeroshell_compute></aeroshell_setup> | | | | | | |
| | | | | | | |
| | \\1 | rerobherr_compute_partrais | | | | |
| 1. | 1 i | nitialize | | | INITIA | LIZE |
| de | f in: | itialize(self): | | | | |
| | """I | Declare options""" | | | | |
| # Need to convert this to an input at some point | | | | | | |
| | | lf.options.declare('Mass', | | | | |
| | | default | =1., | | | |

types=np.ScalarType,

```
desc='Mass of the HyperJackets Pod')
    # The following properties need to be made more accurate ASAP.
    # These should not exist like this
    self.options.declare('Area'
                         default=1.
                         types=np.ScalarType,
                         desc='Pod Lift Area')
    self.options.declare('LiftCoefficient')
                         default=1.
                         types=np.ScalarType,
                         desc='Lift Coefficient of Pod')
    self.options.declare('DragCoefficient')
                         default=1.
                         types=np.ScalarType,
                         desc='Drag Coefficient of Pod')
1.2 setup
                                                          SETUP
def setup(self):
   """Declare inputs and outputs"""
    # Inputs
    self.add_input('AirDensity',
                   desc="Air Density around the aeroshell")
    self.add_input('Velocity',
                   desc="Speed of the aeroshell (and of the pod)")
    # Outputs
    self.add_output('Lift', 0.0,
                    units="kg*m/s^2",
                    desc="Lifting force due to the aeroshell")
    self.add_output('Drag', 0.0,
                    units="kg*m/s^2",
                    desc="Drag force due to the aeroshell")
```

```
# Independence
    self.declare_partials('Lift',
                          'DragCoefficient',
                          dependent=False)
    self.declare_partials('Drag',
                          'LiftCoefficient',
                          dependent=False)
1.3
     compute
                                                      COMPUTE
def compute(self, inputs, outputs):
    """Compute outputs"""
    # Properties of the tube
    rho = inputs['AirDensity']
    # Aerodynamic Coefficients
    c_l = self.options['LiftCoefficient']
    c_d = self.options['DragCoefficient']
    # Properties of the pod
    area = self.options['Area']
    vel = inputs['Velocity']
    lift = - 0.5*rho*c_l*area*vel*vel
    drag = 0.5*rho*c_d*area*vel*vel
    outputs['Lift'] = lift
    outputs['Drag'] = drag
1.4 compute partials
                                          COMPUTE PARTIALS
def compute_partials(self, inputs, partials):
    """ Computation of partial derivatives."""
    c_l = self.options["LiftCoefficient"]
    c_d = self.options["DragCoefficient"]
    area = self.options["Area"]
    vel = inputs["Velocity"]
```

```
partials['Lift', 'AirDensity'] = -0.5*c_l*area*vel*vel
    partials['Lift', 'Velocity'] = - c_l*area*rho*vel
   Wheels
from openmdao.api import ExplicitComponent
<<Wheels_wheelStress>>
class Wheels(ExplicitComponent):
    """ Wheel Material """
    <<Wheels_initialize>>
    <<Wheels_setup>>
    <<Wheels_compute>>
2.1 initialize
                                                      INITIALIZE
def initialize(self):
   """Declare options"""
    # Material Properties
    self.options.declare('Density',
                         default=1.,
                         types=np.ScalarType,
                         desc='Density of the wheel material')
    self.options.declare('PoissonsRatio',
                         default=1.
                         types=np.ScalarType,
                         desc="Poisson's Ratio for the wheel material")
    self.options.declare('FrictionCoefficient',
                         default=1,
                         types=np.ScalarType,
                         desc="Friction Coefficient of the wheel material")
    self.options.declare('YieldCircumferentialStress',
                          default=1,
                          types=np.ScalarType,
                          desc="Max Circumferential Stress")
    self.options.declare('YieldRadialStress',
```

rho = inputs["AirDensity"]

```
default=1,
                          types=np.ScalarType,
                          desc="Max Radial Stress")
    # Engineering Properties
    self.options.declare('FactorOfSafety',
                          default=1,
                          types=np.ScalarType,
                          desc="Factor of Safety for the wheels")
    # Wheel Properties
    self.options.declare('InnerRadius',
                         default=1.,
                         types=np.ScalarType,
                         desc="Inner Radius of the wheel")
    self.options.declare('OuterRadius',
                         default=1.,
                         types=np.ScalarType,
                         desc="Outer Radius of the wheel")
    self.options.declare("Multiplicity",
                         default=4.,
                         types=np.ScalarType,
                         desc="Number of wheels used on pod")
2.2 setup
                                                          SETUP
def setup(self):
    """Declare inputs and outputs"""
    # Inputs
    self.add_input('NormalForce',
                   0.5,
                   desc="Normal Force applied on wheels due to weight of the pod")
    self.add_input('Velocity',
                   0.5,
                   desc="Velocity of the pod")
    # Outputs
    self.add_output('RevolutionsPerMinute',
                    0.5,
```

```
desc="Revolutions per minute of the wheel")
    self.add_output('FrictionForce',
                    0.5,
                    desc="FrictionForce applied to the wheels of the car")
    # Output Stresses experienced by the wheel
    self.add_output('MaximumCircumferentialStress',
                   desc="Circumferential Stress experienced due to rotation")
    self.add_output('MaximumRadialStress',
                   0.5,
                   desc="Radial Stress experienced due to rotation")
    # Independence
    self.declare_partials('CircumferentialStress',
                          'NormalForce',
                          dependent=False)
    self.declare_partials('RadialStress',
                          'NormalForce',
                          dependent=False)
2.3
    compute
                                                      COMPUTE
def compute(self, inputs, outputs):
    """Compute outputs"""
    # Material Properties of the wheel
    density = self.options["Density"]
    m = self.options["PoissonsRatio"]
    c_f = self.options["FrictionCoefficient"]
    circumferential_max = self.options["YieldCircumferentialStress"]
    radial_max = self.options["YieldRadialStress"]
    # Wheel Properties
    r1 = self.options["InnerRadius"]
    r2 = self.options["OuterRadius"]
    multiplicity = self.options["Multiplicity"]
    # Pod Properties
    vel = inputs["Velocity"]
```

```
normal_force = inputs["NormalForce"]
# Engineering
# Derived Properties
omega = vel/r2
# Circumferential & Radial Stresses
 if r1 is 0:
     # We're dealing with a solid disc
     # Both stress are max at r = 0
     # Circumferential Stress & Radial Stress
     c_stress, r_stress = wheelRotationalStress(radius = 0,
                                                 innerRadius = r1,
                                                 outerRadius = r2,
                                                 omega = omega,
                                                poissonsRatio = m,
                                                 density = density):
else:
     # We're dealing with a hollow disc
     # Both stress are max at r = (r1 * r2) ** (0.5)
     # Circumferential Stress & Radial Stress
     c_stress, r_stress = wheelRotationalStress(radius = (r1*r2)**(0.5),
                                                 innerRadius = r1,
                                                 outerRadius = r2,
                                                 omega = omega,
                                                poissonsRatio = m,
                                                 density = density):
if c_stress > circumferential_max:
     failure = "Due to stress above yield circumferential stress"
     raise WheelFailure(failure)
elif c_stress > circumferential_max / factor_of_safety:
     failure = "Circumferential stress is past allowable "
     raise WheelFailure(failure)
else:
     outputs["CircumferentialStress"] = c_stress
 if r_stress > radial_max:
```

```
raise WheelFailure("Your wheels have ripped apart due to radial stress")
elif r_stress > radial_max / factor_of_safety:
    raise WheelFailure("Your radial stress is past the allowable stress")
else:
    outputs["RadialStress"] = r_stress

# RevolutionsPerMinute
# FrictionForce
```

2.3.1 Circumferential Stress

```
# Desired Radius
def wheelRotationalStress(radius = 0,
                            innerRadius = 0,  # Inner Radius of the wheel
outerRadius = 1,  # Outer Radius of the wheel
omega = 1,  # Rotational Velocity of the wheel
                            poissonsRatio = 1, # Poisson's Ratio. Denoted by 1/m
                                                 # Density of the material
                            density = 1):
    if innerRadius is 0:
        # We're dealing with a solid disc
        C_1 = (3 + poissonsRatio)*(1/4)
        C_1 *= (density*(omega**2)*(outerRadius**2))
        C_2 = 0
    else:
        # We're dealing with a hollow disc
        C_1 = (3 + poissonsRatio)*(1/4)
        C_1 *= (density*(omega**2)*(innerRadius**2 + outerRadius**2))
        C_2 = (3 + poissonsRatio)*(1/8)
        C_2 *= (density*(omega**2)*(innerRadius**2)*(outerRadius**2))
    sigma_radial = C_1/2 + C_2/(radius**2)
    sigma_radial -= (3 + poissonsRatio)*(1/8)*(density*(omega**2)*(radius**2))
    sigma_circum = C_1/2 - C_2/(radius**2)
    sigma_circum -= (1 + 3*poissonsRatio)*(1/8)*(density*(omega**2)*(radius**2))
```

return sigma_radial, sigma_circum