

# User Instructions and Manual for the SDC

## System Simulation Spreadsheet

Version 1.0, March 2024

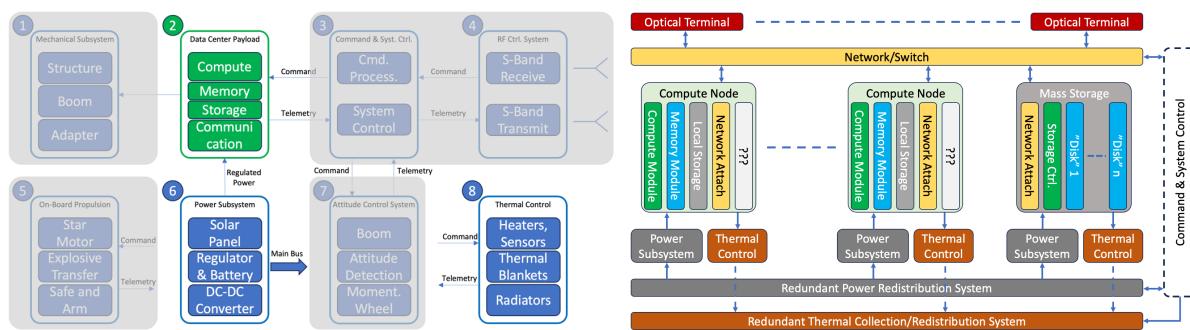
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# Introduction

The SDC (Space Data Center) System Simulator is an Excel, visual basic (VBA) enhanced system performance assessment tool for edge-computing scenarios in space. It has an underlying model of different system components of a satellite with a compute- and communication payload. The System-components which are modeled with various degrees of details are shown in below figure left (2), (6) and (8). Other system components (greyed-out) are accounted for on an aggregated level. The compute payload is then further refined into compute- and storage nodes, including thermal and power infrastructure (figure below on the right).



The results are computed by combining values of many different/relevant technology roadmaps, together with user-requirements into a few specific figures of merit (FOMs), such as **Cost per Compute**, **Power per Compute** or **Cost per Power**. Additional values are also provided, either numerically or a selection thereof can be plotted against each other.

It is imperative to **Enable Macros** when opening the spreadsheet for it to work properly!

To improve robustness, values are not referred to with their cell-coordinates, but rather by definition of Named Variables (see -> **Formulas -> Name Manager** for the full list).

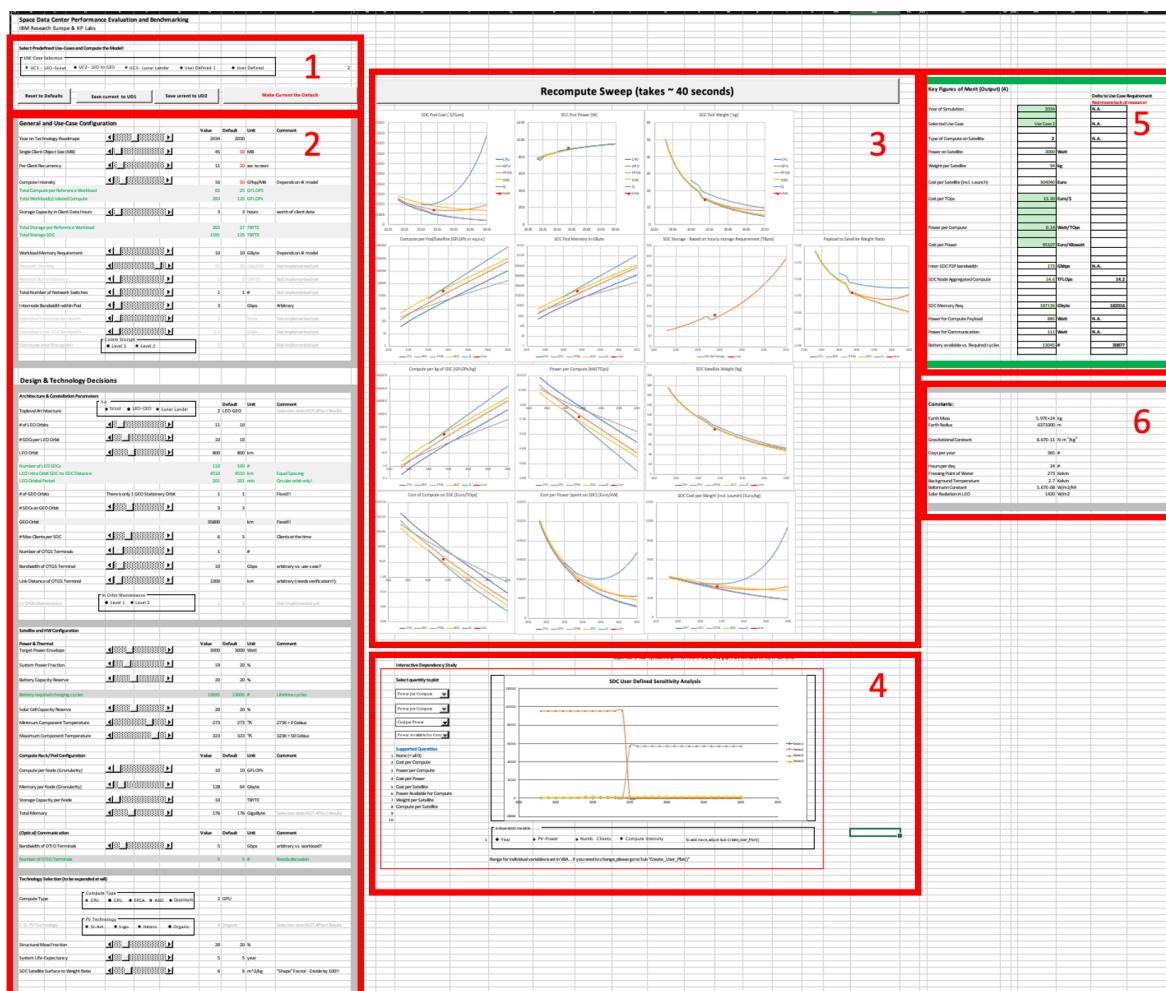
By using the “Name Manager” and named ranges extensively, the entire spreadsheet could be made relatively robust against shifting cells in sheets. However, it is highly discouraged to change the names of the individual sheets, as it was necessary to hard code their access with their respective names all over in the VBA code.

**Please Note:** For the spreadsheet to work in its current way, many technology performance metrics had to be simplified and were mapped to linear or exponential growth curves, sometimes simply by creating fractions of current performance, power and weight indications. **The relevance and value of the results of this “simulator” has thus to be seen in the trends and comparison between different quantities/trends it provides, and NOT in absolute values!**

## User Interface (UI)

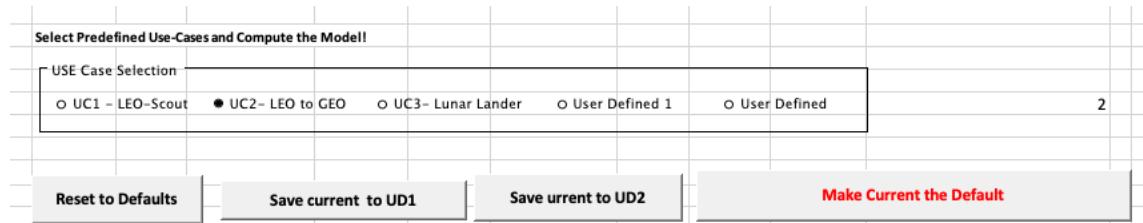
The Simulation Spreadsheet consists of various tabs. However, the main sheet to interact with the tool, is the ***SDC Summary*** sheet, as shown below. For best experience, a rather large screen is recommended.

The main interface consists of 5 user-relevant sections (1-5).



## UI Section 1

This section allows to select and preload existing and user-defined scenarios. The values for the different scenarios are fetched from or saved to a large table in on the sheet **Use-Case Definitions**. To adjust a pre-set use-case, values on the latter table must be changed, while interactive, user-defined use-cases can be saved with a button-click to one of the two User-Defined configurations.



## UI Section 2

This section allows to manually adjust individual values and to experiment with the results. This is possible either by using the sliders (despite that they are configured for manual sliding and page “jumps”, mostly single up/down increments work smoothly, since every change triggers a recompute of the entire model and update of the charts in **Section 3**) or by entering values directly into the “Value” column, which will be reflected in the slider position as well.

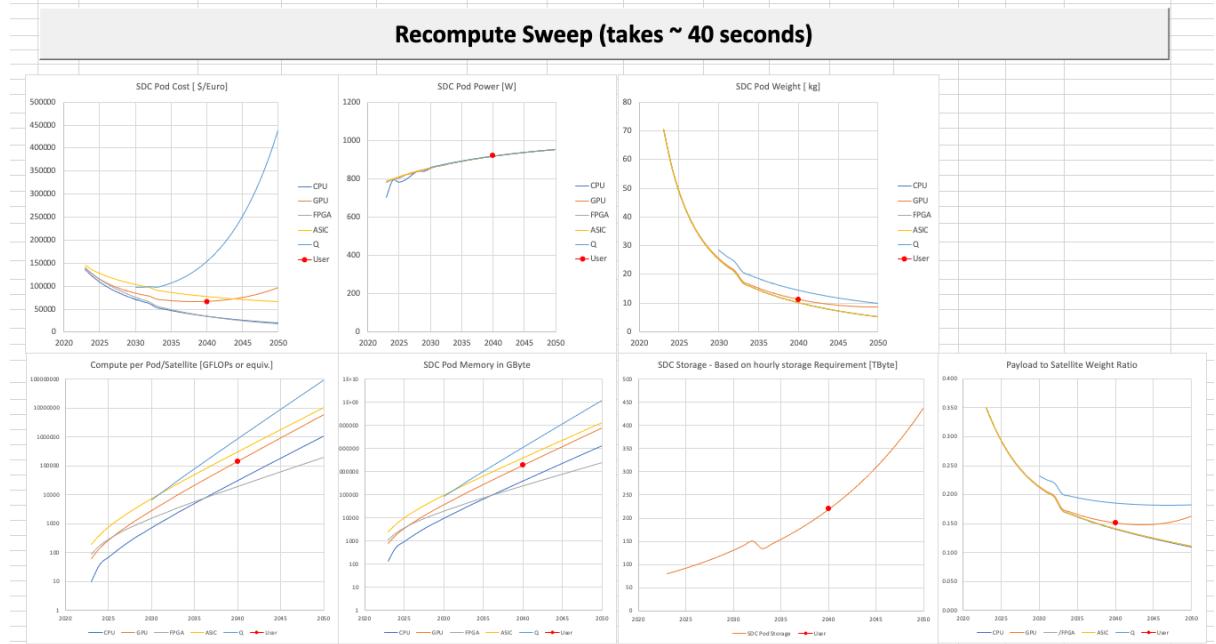
Below, a screenshot of one of the many subsections of **Section 2** is shown. Additionally, to numeric values, also radio-buttons are used to facilitate selecting e.g. architectural or technological choices. Changes made in this section will immediately be reflected by the **red dot** in the figures in **Section 3**.

If numbers are displayed in **green**, they are computed in the cell, based on other values and should not be touched/changed (see also the coloring convention).

Architecture & Constellation Parameters		Default	Unit	Comment
Toplevel Architecture	<input checked="" type="radio"/> Scout <input checked="" type="radio"/> LEO-GEO <input type="radio"/> Lunar Lander	LEO-GEO	LEO-GEO	Selection does NOT Affect Results
# of LEO Orbit	<input type="button" value="◀"/> <input type="button" value="▶"/>	11	10	
# SDCs per LEO Orbit	<input type="button" value="◀"/> <input type="button" value="▶"/>	10	10	
LEO Orbit	<input type="button" value="◀"/> <input type="button" value="▶"/>	800	800 km	
Number of LEO SDCs		110	100 #	
LEO Intra Orbit SDC-to-SDC Distance		4510	4510 km	
LEO Orbital Period		201	201 min	Equal Spacing Circular orbit only!
# of GEO Orbit	There is only 1 GEO Stationary Orbit	0	1	Fixed!!

## UI Section 3

This is the main reporting section, where various performance relevant quantities are plotted over time.



To refresh the plots, hit the “**Recompute Sweep**” button. This will trigger a recompute of the entire underlying data, for every year and for every type of compute (CPU, GPU, … , Quantum), with the current requirements set in **Section 2**. The data for the plots is written to the **temp\_data** sheet. During the recompute process, you will be able to see the **Technology\_Roadmap** sheet, where you can observe continuous changes in the value-column. On the very top the year will count up to 2050, which gives you an indication how close to completion the process is. After completion, you will be automatically redirected to the **SDC Summary** sheet again, for inspecting the new graphs and continue your evaluation of other requirements.

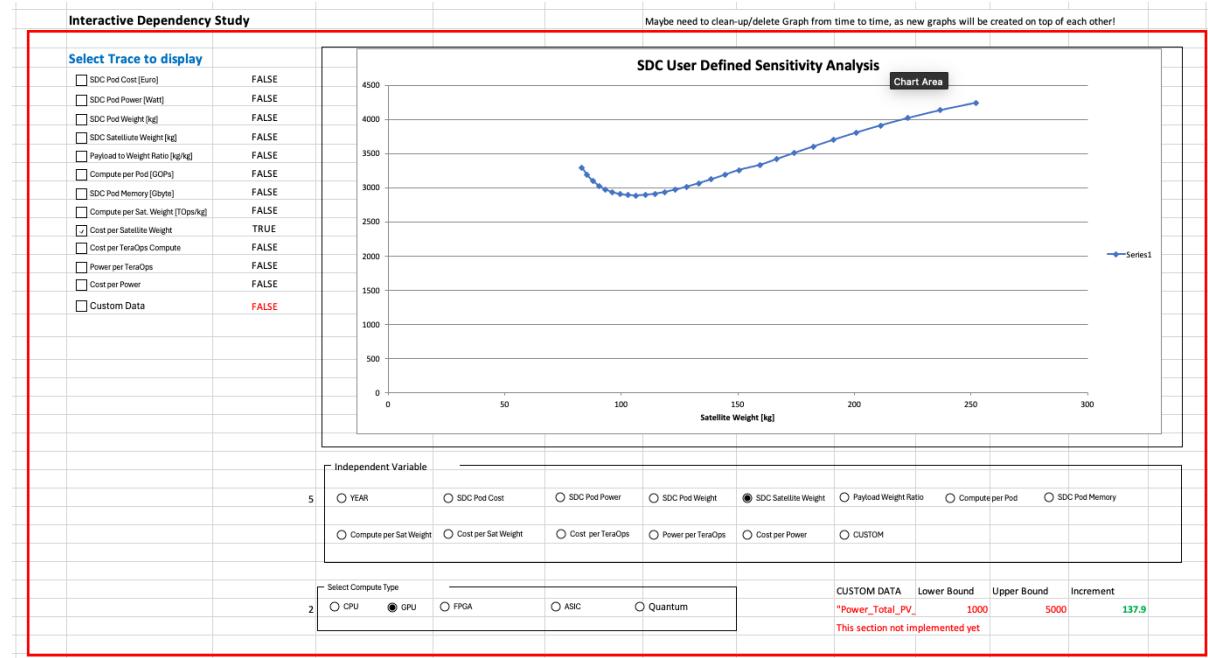
## UI Section 4

In this section, you can choose to plot all computed FOMs against each other or against time. Every time you select a different x-axis, the graph will be recreated, with all checked data from the left. As the figure will be regenerated on top of the existing chart, you may want to clean-up, i.e. delete underlying charts from time to time.

All data is based on the results from applying the “Recompute Button” and is NOT affected by temporary changes of the requirements. If such need to be accounted for, please **press “Recompute Sweep”** before.

Please note, that all data here is fetched from the **temp\_data** sheet, thus is not fully independent, but has rather always time as an implicit parameter.

To be able to select any of the requirement variables from **Section 2** as a sweep parameter, is not implemented yet, but may yield interesting dependencies in the future. Curves in this section are always only given for one of the compute types available, which must be selected (a change thereof also triggers a regeneration of the chart).



## UI Section 5

This section provides numerical values for the key figures of merit (FOMs) as defined in the statement of work, including some references to the “conditions” they were evaluated.

The right column evaluates “reserves” with respect to specific use-case requirements, where applicable.

Please note: The key figures of merit displayed here, are not the result of an optimization scheme, but rather derived from the given power-envelope, i.e. the compute available is the maximum compute performance possible within the available power of the solar panels and the given compute technology.

Key Figures of Merit		Delta to Use Case Requirement <i>Red means lack of resource!</i>
Year of Simulation	2036	N.A.
Selected Use Case	Use Case 2	N.A.
Type of Compute on Satellite	2	N.A.
Power on Satellite	5000 Watt	
Weight per Satellite	141 kg	
Cost per Satellite (incl. Launch)	499007 Euro	
Cost per TOps	2.20 Euro/\$	
Power per Compute	0.02 Watt/TOps	
Cost per Power	99801 Euro/Kilowatt	
Inter-SDC P2P bandwidth	173 Gbitps	N.A.
SDC Node Aggregated Compute	227.1 TFLOps	226.7
SDC Memory Req.	2906880 Gbyte	2901760
Power for Compute Payload	1496 Watt	N.A.
Power for Communication	97 Watt	N.A.
Battery available vs. Required cycles	13045 #	34030

## UI Section 6

This section is reserved for physical and other constants required during model evaluation.

## Coloring Convention

The coloring convention for values in the spreadsheet is generally as follows:

- Black: Values are defined/entered there
- Blue: Values are defined somewhere else and only shown for reference
- Green: Values are computed in the cell (should not be changed)
- Red/Yellow: Issue/uncertainty with/of the value or not implemented yet. Special attention should be given to these values
- Grey: Values/quantities for future use/implementation

Color of frames/outlines:

- No consistent scheme applied here.

# Simulation Data Source

The simulation model relies on 2 principal sources of data. 1) the Use-Case Definitions (sheet), which are also reflected and modified on the UI value column and 2) the Technology\_Roadmaps (sheet) of a vast numbers of quantities, of which always the values in the “Value” column will be used to evaluate the current model.

## Use-Case Definitions Sheet

On the **Use-Case Definitions** Sheet, different predefined and user-defined requirements, specific to a given use-case can be entered.

To simplify the underlying VBA code, all data is on the very same rows as the user-adjustable requirements from the UI on the **SDC Summary** sheet. Using the radio-buttons on the UI sheet, simply copies the relevant column from Use-Case Definitions to the “value” column of the SDC Summary sheet, which is used by the model(s) to compute the figures of merit and the “red point” on the graphs, or the graphs themselves during the sweeps.

If the user has a set of requirements between which he/she wants to switch, 2 user-defined columns are available to store them. By means of button clicks on the UI sheet (Section 1), values can be stored to or by selecting the radio buttons, retrieved from these sections. The same applies to the default column.

## Technology Roadmap Sheet

This sheet constitutes the major source of data that is used in computing the models. It consists of over 50 individual “roadmaps” of technologies involved or required in the building of a space data center satellite constellation. Most of them are assumed to be of exponential nature with the time in years where performance doubling is expected in column “F”. Current performance metric (as of 2023) is in the “Default” column and growth rate are often educated guesses from literature studies, but absolutely open to discussion and improvement. The latter on Column “F”.

For some values a linear growth is assumed, which is easily visible by the yellow font of the growth rate.

For practical reasons, performance roadmaps for the different compute types (CPU, GPU, FPGA, ASIC and Quantum), is separated from this list. Values are copied “on-demand” into the upper section (row 4-6) depending on the selection of the compute-type, either by user or VBA scripts.

Also in this section, extensive use of named ranges has been made, by naming the first entries per row and using their row-index in the software code. This will allow to add more rows in the future without the need to rewrite and correct the VBA code.

	A	B	C	D	E	F	G	H	I	J	K	L	M	N
1	Reference Technology Values for Sim - Values used in Scenario!													
2	Value	Default	Unit	YEAR										
3	YEAR	2036	2023	Performance Doubling Every XY Years red = exponential, yellow = linear										
4	Power per Compute:	0.006	0.1 W/GFLOPS	Assume Start in 2030	1.5	Quantum								
5	Cost per Compute:	0.006	0.10 \$/GFLOPS		1.5	Quantum								
6	Weight per Compute:	0.031	0.50 kg/TOps		1.5	Quantum								
7														
8														
9	Cost per Memory	0.04	4.00 \$/GBYTE		2.0		4.00	2.83	2.00	1.41	1.00	0.71	0.50	
10	Memory per Module	80.63	4.00 GBYTE		3.0		4.00	5.04	6.35	8.00	10.08	12.70	16.00	
11														
12	Mass (Storage_)													
13	Terabyte per module	285.41	30.00 TBYTE		4.0		30.00	35.68	42.43	50.45	60.00	71.35	84.85	
14	Cost per Capacity	0.09	8.00 \$/TBYTE		2.0		8.00	5.66	4.00	2.83	2.00	1.41	1.00	
15	Weight per Capacity	0.11	10.00 g/TBYTE		2.0		10.00	7.07	5.00	3.54	2.50	1.77	1.25	
16	Power per Capacity	0.05	1.00 W/TBYTE		3.0		1.00	0.79	0.63	0.50	0.40	0.31	0.25	
17														
18	Network/Switch (Network_)													
19	Power per Capacity per Port	0.03	0.20 W/Gbps/Port		5.0		0.20	0.17	0.15	0.13	0.11	0.10	0.09	
20	Cost per Capacity per Port	1.65	10.00 \$/Gbps/Port		5.0		10.00	8.71	7.58	6.60	5.74	5.00	4.35	
21	Weight per Port	13	20 g/Port		20.0		20	19	19	18	17	17	16	
22														
23	Optical Terminals													
24	In-Orbit (OTIO_)													
25	Max Range of optical Terminal	17236	7000 km	SCOT80	10.0		7000	7502	8041	8618	9237	9899	10610	
26	Maximum Datarate	24.6	10.0 Gbps		10.0		10.0	10.7	11.5	12.3	13.2	14.1	15.2	
27	Max. Simultaneous Connections	7	3 #	Allow fraction of connecti	10.0		3	3	3	4	4	4	5	

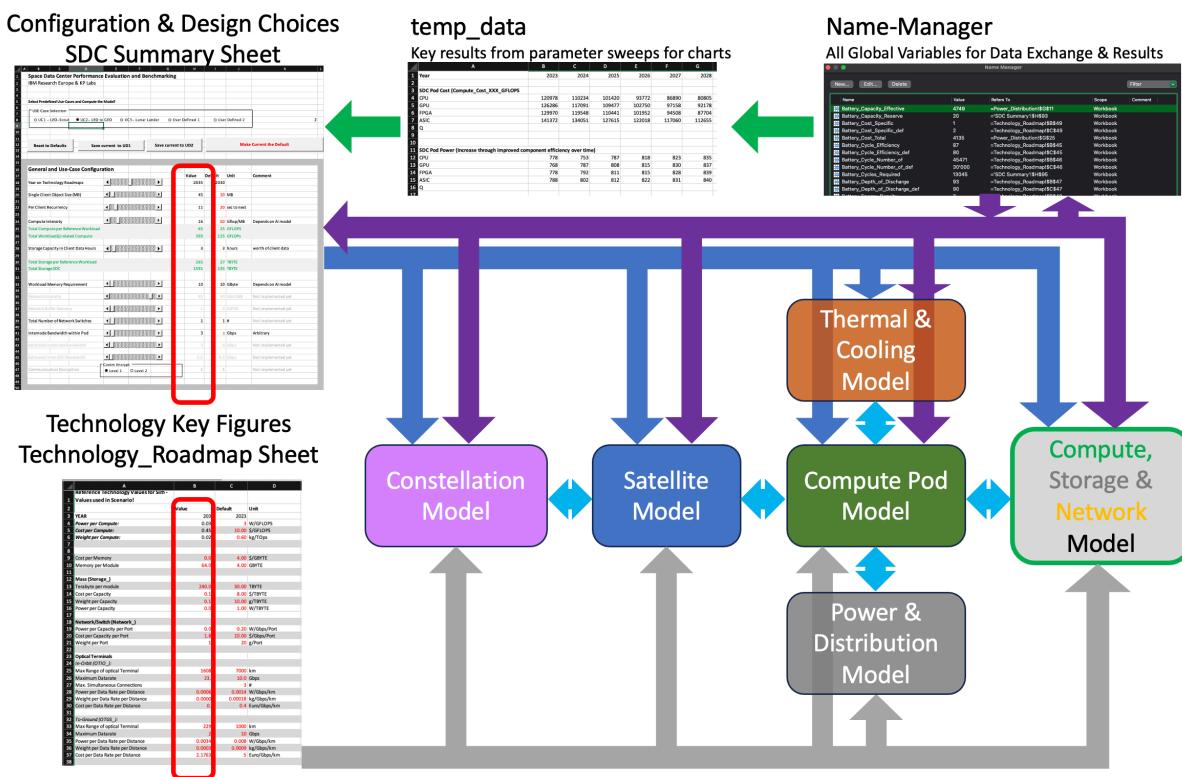
Below row 151 there are additional technology roadmaps which maybe useful for future refinements of the simulator and are provided/left as a reference.

The “Reference Benchmark Section” (Row 75 and following), serves to create a static plot of the SDC Summary sheet on edge-computing cost vs. raw-data-download cost. These numbers are not used anywhere else in the model.

# Simulation Model

The simulation model is broken down into 6 different blocks, as outlined in below figure. Starting from the constellation model, requirements are derived or forwarded (the latter by means of “Name-Manager” variables), to the satellite model, which in turn consists of the Power- and Power-Distribution Model, the Thermal and Cooling Model and the Compute Pod Model. The latter then consists of the Compute, Storage and Network model. The latter three are all computed on the “Compute\_Storage\_Network” sheet.

As depicted in the UI-section of this document, key result sweeps over time can be computed by the press of a button. All results of these sweeps are stored on the temp\_data sheet for visualization on the UI. By sweeping over years, most significant “dynamics” of variables are expected to be captured. Arbitrary variables cannot be swept independently yet.

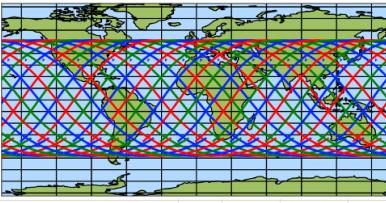
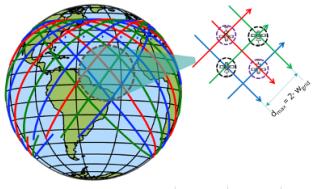


## Important:

For such a model to work consistently, a decision had to be made to either start at the compute pod model and then derive power-requirements, which need to propagate through all other model blocks, or vice-versa. The latter was adopted, to maintain our envisioned top-down approach. An overall power- and thermal envelope was defined (by means of the variable **Power\_Total\_PV\_Capacity** (SDC Summary, Cell H89), total available power of the solar panels and indirectly an approximation of the overall satellite size) and a usable fraction thereof of made available to supply the Compute Pod. As a result, total available compute power (variable **SDC\_Pod\_Compute\_Power\_Limit**) is one of the results of the model.

## SDC\_Constellation Sheet

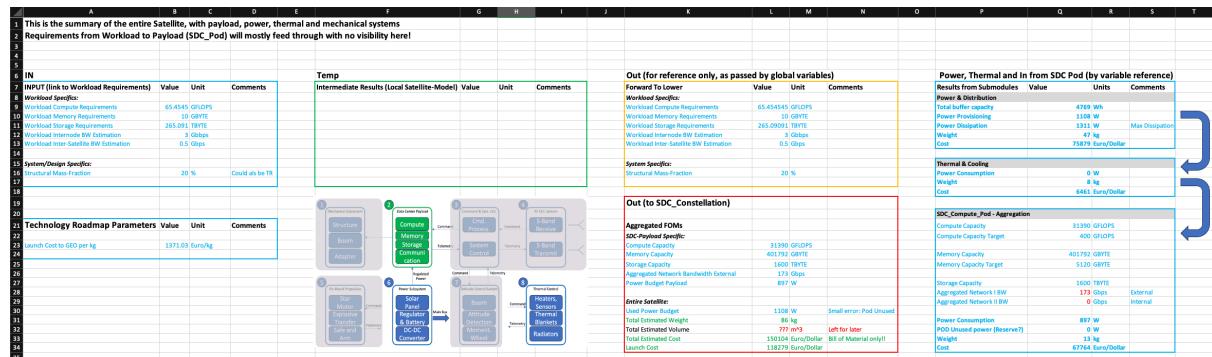
This is the sheet that implements the highest level of the overall simulation model, the “constellation”. It is mainly concerned with orbits, their number of and aggregation values. Most of the latter are not reflected in the primary output of the simulation tool but serve as a reference and are for future use.

This is the summary of the entire Constellation Requirements from Workloads for individual Satellites ... later also distributed workload requirements			
IN			
<b>Workload Specifics:</b>			
Workload Compute Requirements		65.45455 GFLOPS	
Workload Memory Requirements		10 GBYTE	
Workload Storage Requirements		265.0909 TBYTE	
Workload Internode BW Estimation		3 Gbps	
Workload Inter-Satellite BW Estimation		0.5 Gbps	
<b>System/Design Specifics:</b>			
Structural Mass-Fraction		20 %	Could als be TR
Serves also as kind of an overview...not all values are used here			
Technology Roadmap Parameters			
Launch Cost to GEO per kg	1371.031	Euro/kg	x
Spacing between Satellites	4510 km	o	
Inter Satellite Bandwidth	5 Gbps	o	
Ground Station Link Length	1000 km	o	
			
			
Temp			
Intermediate Results (Local Satellite-Model)			
<b>Cost Overview:</b>		Value	Unit
Aggregated Cost per Satellite		306'065	Euro/Dollar
Launch Cost		192'942	Euro/Dollar
Number of Satellites		110 #	
NRE (Design, Tools etc.)		2'030'631	Euro/Dollar
Constellation Deployment Cost		56'921'349	Euro/Dollar
<b>Performance Overview:</b>			
Aggregated Compute		44'000	GFLOPS
Aggregated Memory		563'200	GBYTE
Aggregated Storage		176'000	TERABYTE
Aggregated Inter-SDC Bandwidth		2'200	Gbps
Aggregated Bandwidth to SDC from Clients		3'300	Gbps
Assuming 1-4 SDC connections			
Out (To SDC Satellite)			
Forward To Lower			
<b>Workload Specifics:</b>		Value	Unit
Workload Compute Requirements		65.45455	GFLOPS
Workload Memory Requirements		10	GBYTE
Workload Storage Requirements		265.0909	TBYTE
Workload Internode BW Estimation		3	Gbps
Workload Inter-Satellite BW Estimation		0.5	Gbps
<b>System Specifics:</b>			
Structural Mass-Fraction		20 %	
Out (to Overview & (time)Series)			
Aggregated FOMs			
<i>Per Spacecraft (SDC Satellite):</i>			
Compute Capacity	400	GFLOPS	
Memory Capacity	5120	GBYTE	
Storage Capacity	1600	TBYTE	
Aggregated Network Bandwidth External	173	Gbps	
Estimated Power Budget	1847	W	From PV (sunny side)
Total Estimated Weight	141	kg	
Total Estimated Volume	???	m^3	
Total Estimated BOM Cost	306065	Euro/Dollar	Bill of Material only!!
Launch Cost	192942	Euro/Dollar	
<b>Constellation Specific:</b>			
Total Number of Spacecrafts	110 #		
Non Recurring Cost for Constellation	2'030'631	Euro/Dollar	NOT Ground-Segment
Overall Compute Capability	44000	GFLOPS	
Total Memory Available	563200	GBYTE	
Total Storage Available	176000	TBYTE	
Inter SDC Bandwidth	2200	Gbps	
To SDC Upload Bandwidth	3300	Gbps	
Constellation Deployment Cost Estimate	56921349.4	Euro/Dollar	
In from SDC Satellite (by global reference)			
Results from Submodules			
<b>Aggregated FOMs</b>		Value	Units
<i>SDC-Payload Specific:</i>			
Compute Capacity	400	GFLOPS	
Memory Capacity	5120	GBYTE	
Storage Capacity	1600	TBYTE	
Aggregated Network Bandwidth External	173	Gbps	
Power Budget Payload	1496	W	
<i>Entire Satellite:</i>			
Estimated Power Budget	1847	W	From PV (sunny side)
Total Estimated Weight	141	kg	
Total Estimated Volume	???	m^3	
Total Estimated BOM Cost	306065	Euro/Dollar	Bill of Material only!!
Launch Cost	192942	Euro/Dollar	

It is thus clear that use-case requirements are “just forwarded” to the model of the individual satellite, as currently considered workload do NOT require distributed computing as required by very large workloads, and exploit constellation properties only for “communication/downlink coverage”, which at this stage is not explicitly modeled or reflected in the FOMs.

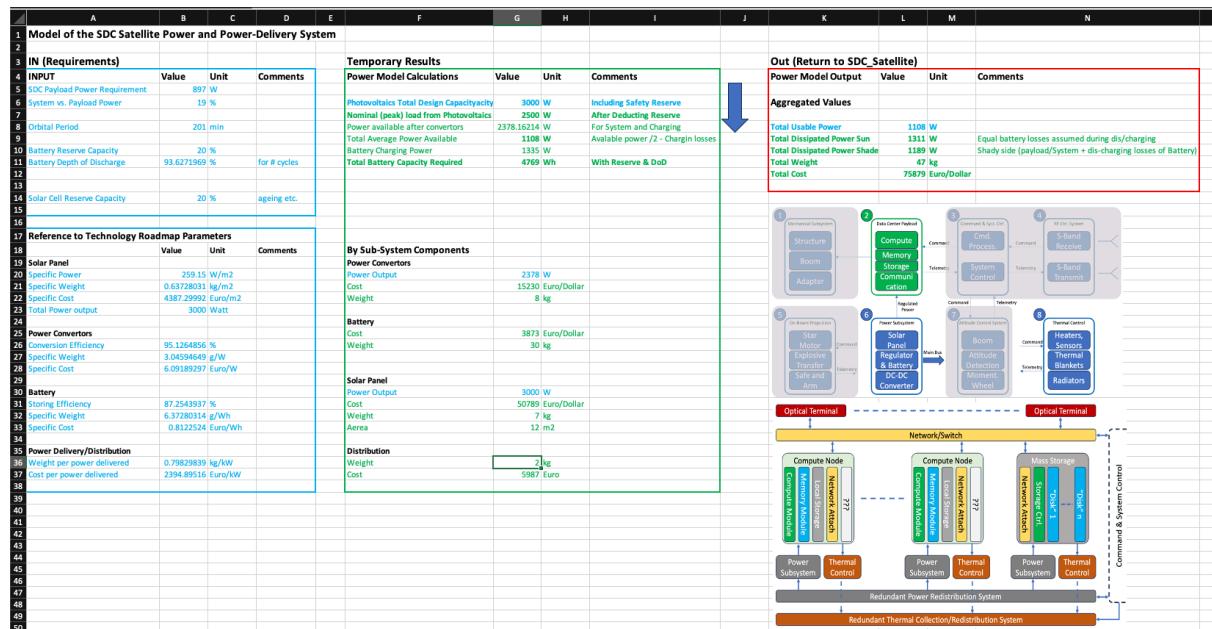
## SDC\_Satellite Sheet

This sheet mainly “forwards” requirements to underlying sub-models (which it does not actively, as all parameters are passed by named (global) variables). Below figure at the right already indicates, that the model first accounts for available power, of which losses in the power-train and buffering power for orbital-night-periods is deduced, then cooling requirements are computed (by underlying model) and finally the remainder of the available power is fed into the compute (SDC\_Pod/Compute\_Storage\_Network) model to get available compute capacity, weight and cost figures. Compute capacity is always “maxed-out” to the available power, with given compute technology and granularity.



## Power\_Distribution Sheet

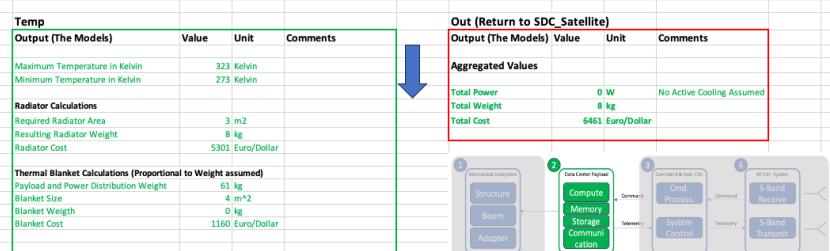
This model is based on assumptions of conversion efficiency, battery technology (e.g. depth of discharge [DoD]) and orbital parameters (duration of night -> battery capacity required). The sub-model key output is available power to the compute pod and cost and weight of the battery, solar panel, and power-conversion systems.



## Thermal\_and\_Cooling Sheet

The main contribution of this sub-model is the derivation of the cooling requirements, mostly weight, secondary also cost, to be able to operate at the given power-envelope. Only less than half of what the solar panels provide at peak, will be available for satellite system and payload operation, as the other half is needed to re-charge the batteries, such that the satellite is always operational, irrespective of day- or night-time. While this may be questioned, as the majority of observation-data is acquired during daytime, our assumption constitutes the most generic scenario, which with emerging hyper-spectral and SAR sensors will become more relevant. Also, any node in the constellation may be assigned tasks, such as compute and communication, irrespective of solar exposure, i.e. where they are w.r. to their orbital cycle.

A	B	C	D	E	F	G	H	I	J	K	L	M	N	O
<b>1 Model of the SDC Satellite Cooling and Thermal Management System</b>														
2 Simplified model where no Solsails- or Structural co-use for thermal management is assumed														
<b>3</b>														
<b>4 IN (Requirements)</b>														
<b>5 INPUT</b> Value Unit Comments														
6 Total Power Dissipated 1311 W														
7 Maximum Module Temperature 323 °K 323K + 50 C														
<b>10</b>														
<b>11 Reference to Technology Roadmap Parameters</b>														
<b>12</b> Value Unit Comments														
13 Radiator Efficiency 63.25 %														
14 Radiator Specific Weight 2.39489516 kg/m²														
<b>15</b>														
<b>16</b>														
<b>17</b>														
<b>18</b>														
<b>19</b>														
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<b>29</b>														
<b>30</b>														



## SDC\_Pod Sheet

In this sub-model, satellite requirements are broken down into node requirements (we assume here a single compute pod (i.e. the compute payload), with several/many nodes. For future scenarios with larger satellites, this approach can be extended towards several pods per satellite too.

As indicated with arrows, from workload/satellite requirements, the node-requirements are derived and provided to the “**Compute\_Storage\_Network**” model, where compute capacity, network and communication performance is computed according to available (power) resources.

The results from the latter are then aggregated to satellite level and returned to “**SDC\_Satellite**” (again by means of global variables).

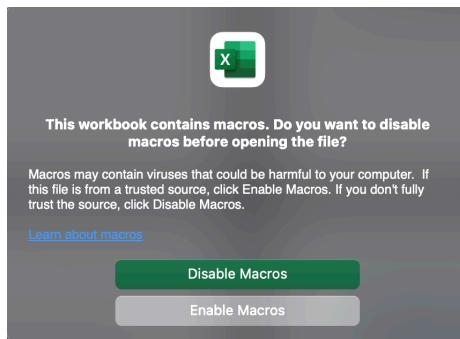
A	B	C	D	E	F	G	H	I	J	K	L	M	N	O	P	Q	R	S
This is the summary of the entire SDC Payload Section of the satellite																		
IN																		
INPUT (link directly from Requirements_Overview) Value Unit Comments																		
Compute																		
9	Compute per Workload	1000	GFLOPS															
10	Memory per Workload	100	GBYTE															
11	Total Memory Required per Satellite/Pod	10000	GBYTE															
12	Total Memory Required per Satellite/Pod	10000	GBYTE															
13	Number of Nodes Required	600	#															
14	Mass Storage																	
15	Capacity per workload	324	TBITE															
16	Number of Switches required	1	#															
17	Inter-Satellite Bandwidth	5	Gbps															
18	Intermediate Network Buffer Memory	1	Gbyte															
19	All to all switch																	
20	Network/Switch																	
21	Number of Switches required	1	#															
22	Inter-Satellite																	
23	Optical Terminal																	
24	Number of inter-Satellite Terminals	6	#															
25	Inter-Satellite Distance LEO-LEO	4510	km															
26	Inter-Satellite Bandwidth	5	Gbps															
27	Intermediate Network Station (IS) Terminals	1	#															
28	IS Line Distance	1000	km															
29	IS Bandwidth	10	Gbps															
30	Other																	
31	Max number of clients per SDC	6	#															
32	Technology Roadmap Parameters																	
33	Value	Unit	Comments															
34	Sub-System Aggregation Values																	
35	Power Envelope Perspective (lower limited FOMs)																	
36	Total Power available for Payload/Pod	1144	Watt															
37	Power Available for Compute and Storage	759	Watt															
38	Power Available for Memory	331	Watt															
39	Maximum Number of Nodes within Power Limit	3250	GFLOPS															
40	Resulting Available Compute capacity	3250	GFLOPS															
41	Resulting Available Memory capacity	41600	Gbyte															
42	Compute Node																	
43	Compute Node																	
44	Compute Node																	
45	Compute Node																	
46	Compute Node																	
47	Compute/Unit	10	GFLOPS	x														
48	Memory/Unit	128	GBYTE	x														
49	Total Memory Required per Satellite/Pod	10000	GBYTE															
50	Total Memory Required per Satellite/Pod	10000	GBYTE															
51	Number of Nodes Required	600	#															
52	Mass Storage Node																	
53	Mass Storage Node																	
54	Mass Storage Node																	
55	Mass Storage Node																	
56	Mass Storage Node																	
57	Mass Storage Node																	
58	Mass Storage Node																	
59	Mass Storage Node																	
60	Network/Switch																	
61	Network/Switch																	
62	Number of Switches required	1	#															
63	Inter-Satellite Bandwidth	5	Gbps															
64	Intermediate Network Buffer Memory	1	Gbyte															
65	All to all switch																	
66	Network/Switch																	
67	Optical Terminal																	
68	Number of inter-Satellite Terminals	7	#															
69	Inter-Satellite Distance LEO-LEO	21830	km															
70	Inter-Satellite Bandwidth	0.5	Gbps															
71	Intermediate Network Station (IS) Terminals	1	#															
72	IS Line Distance	20800	km															
73	IS Bandwidth	0.5	Gbps															
74	Other																	
75	Max number of clients per SDC	20	#															
76	Technology Roadmap Parameters																	
77	Value	Unit	Comments															
78	Sub-System Aggregation Values																	
79	Power Envelope Perspective (lower limited FOMs)																	
80	Total Power available for Payload/Pod	632	Watt															
81	Power Available for Compute and Storage	585	Watt															
82	Power Available for Memory	45	Watt															
83	Maximum Number of Nodes within Power Limit	1512	GFLOPS															
84	Resulting Available Compute capacity	1512	GFLOPS															
85	Resulting Available Memory capacity	21500	Gbyte															
86	Compute Node																	
87	Compute Node																	
88	Compute Node																	
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119	Compute Node																	
120	Compute Node			</td														

# Example System Analysis

In this section, a step-by-step use of the system simulator is executed to guide the user through all necessary steps. The application is run on an apple Macintosh (OSX Sonoma 14.3.1). As Microsoft Excel has a reduced functionality on OSX compared to Windows operating systems, we assume the user-experience on Windows should be equal or better.

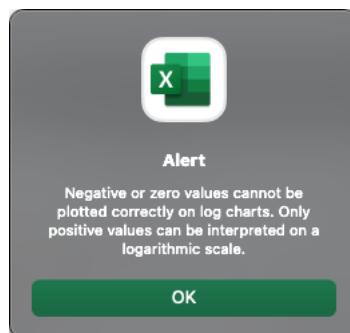
## 1. Open the Spreadsheet

Open the spreadsheet file and allow running macros, otherwise the simulator cannot function properly:



## 2. Select the Use Case

Open the **SDC Summary** sheet, if not already open. As many charts use log-scale, it is likely you will eventually be presented with Alerts. You can safely acknowledge them.



Now select a use-case by means of the radio-buttons.

Select Predefined Use-Cases and Compute the Model!				
USE Case Selection				
<input checked="" type="radio"/> UC1 - LEO-Scout	<input type="radio"/> UC2- LEO to GEO	<input type="radio"/> UC3- Lunar Lander	<input type="radio"/> User Defined 1	<input type="radio"/> User Defined 2

For this example we will stick with Use-Case 1. To be sure, press the radio button again. This loads all configuration parameters into the active value column **SDC Summary H**.

### 3. Verify and adjust individual settings

It's always good to quickly move the year 1 forward and 1 backward, this assures all values from the technology roadmap are properly copied into the active "value" column.

General and Use-Case Configuration				
Year on Technology Roadmaps	Value	Default	Unit	Comment
2030	2030	2030		

While the use-case properly assigns the top-level architecture in the value-column, you may want to select the corresponding top-level architecture also from the radio-button, as it was not possible to have it automatically update the button position. Of course, you can change the architecture if you want, but it only marginally affects the results (see comments in cells).

Design & Technology Decisions			
Architecture & Constellation Parameters			
Toplevel Architecture	<input type="radio"/> Scout <input type="radio"/> LEO-GEO <input checked="" type="radio"/> Lunar Lander	Scout	Default LEO-GEO <small>See SDC_Satellite L34, E_5_N G22-24 USE "Scout" for ALL LEO applications</small>

The same applies for the Radio-button of the compute type...just confirm the value in the "value" column or select another compute type. This will of course affect your results clearly!

### 4. Recompute the Sweeps

Now everything is ready to recompute all the parameter sweeps, so press the large button. As indicated, this may take a moment to complete. During the computation, you will be brought to the **Technology Roadmap** sheet. The entire "value" column will change all the time, in column "G" the compute type may vary occasionally, but mainly you will be able to monitor progress by looking at the "Year" in the value column, which iterates from 2023 to 2050. Upon completion you will be brought back to the **SDC Summary** sheet.

A	B	C	D	E	F	G	H	I	J
Reference Technology Values for Sim - Values used in Scenario!									
1	Value	Default	Unit			YEAR	2023	2024	20
2	2030	2023				red = exponential, yellow = linear			
3	YEAR								
4	Power per Compute:	0.233	3 W/GFLOPS			1.9 GPU	3.00000	2.08298	1.446
5	Cost per Compute:	1.647	10.00 \$/GFLOPS			2.7 GPU	10.00	7.73	5.
6	Weight per Compute:	0.099	0.60 kg/TOps			2.7 GPU	0.60	0.46	0.
7									
8									
9	Cost per Memory	0.35	4.00 \$/GBYTE			2.0	4.00	2.83	2.
10	Memory per Module	20.16	4.00 GBYTE			3.0	4.00	5.04	6.
11									
12	Mass (Storage_)								
13	Terabyte per module	100.91	30.00 TBYTE			4.0	30.00	35.68	42.
14	Cost per Capacity	0.71	8.00 \$/TBYTE			2.0	8.00	5.66	4.
15	Weight per Capacity	0.09	10.00 kg/TBYTE			2.0	10.00	7.07	6.

To update the "red dot" in the figures, cycle again one up/down on the YEAR slider. Now you can inspect all relevant quantities in the charts and in the green field.

Key Figures of Merit		Delta to Use Case Requirement Red means lack of resource!	
Year of Simulation	2030	N.A.	
Selected Use Case	Use-Case 1	N.A.	
Type of Compute on Satellite	GPU	N.A.	
Power on Satellite	3000	Watt	
Weight per Satellite	96	kg	
Cost per Satellite (incl. Launch)	316'257	Euro	
Cost per TOps	99	Euro/\$	
Power per Compute	0.94	Watt/TOps	
Cost per Power	105'419	Euro/Kilowatt	
Inter-SDC P2P bandwidth	2352.6	Gbitps	N.A.
SDC Node Aggregated Compute (Fully exploiting available power)	3.20	TFLOps	0.9
SDC Memory Req.	10'240	Gbyte	2912
Power for Compute Payload	935	Watt	N.A.
Power for Communication	10	Watt	N.A.
Battery Required vs. Available cycles	13'045	#	25191

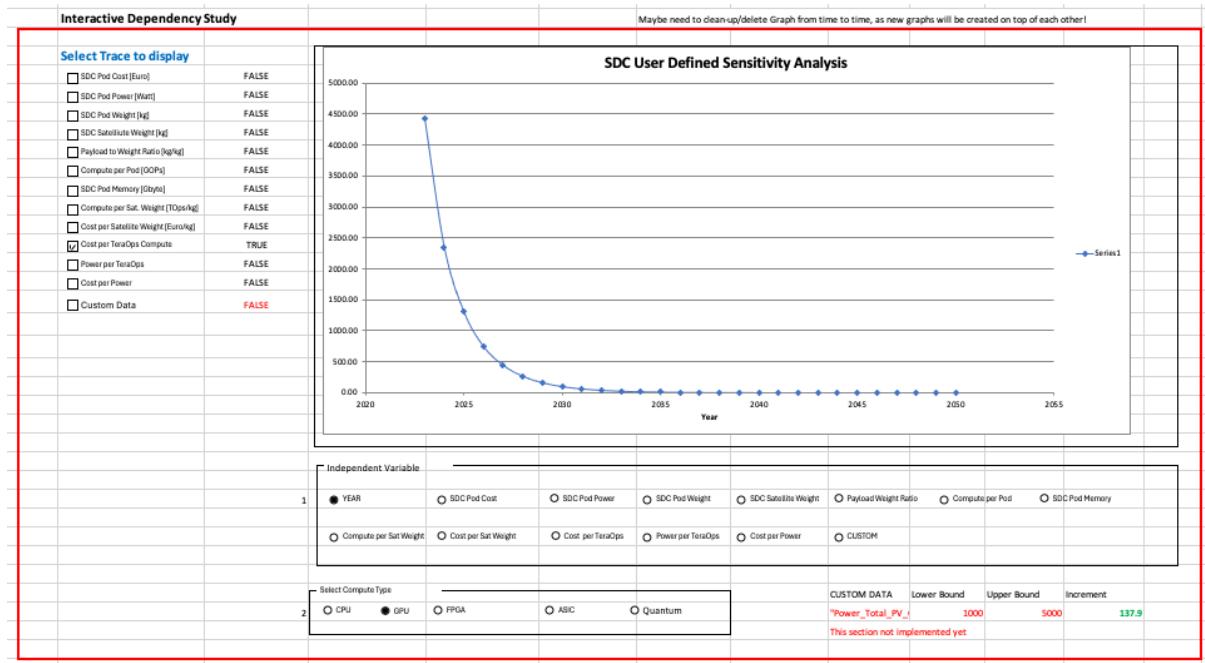
In case you get some red numbers in the green field, you can play with all values in the value column in the configuration section (either by slider action or by inserting values directly). This will recompute the model and the figures of merit as displayed above, for the values from the technology roadmap, corresponding to the year selected at the top of the configuration section.

If you have found some settings which seem promising, you should re-run the Recompute sweep to understand how the changes affect the key results over time.

## 5. Experiment with the Interactive dependency/sensitivity analysis

This window is kind of experimental. It allows you to plot many quantities that were computed during the parameter sweep against each other.

It is important to note, that time (i.e. year) is the curve parameter, meaning that they are not sweeps of two parameters against each other, everything else fixed, but always an implicit function of time. All values plotted origin from the **temp\_data** sheet and were computed during the main sweep.



# Solving Issues and known limitations and tips

As with every software tool, even more with research grade or early release tools, it is not possible to fully suppress all errors or misbehaviors. This section gives the user guidance on how to deal with potential issues and where to look for fixing them.

1. Problem: The actual value “red-dot” is NOT on the trace in the graph

Cause: The curves may have been computed/swept with different requirements values. This happens e.g. when in **Section 1** the Use-Case is changed.

Solution:

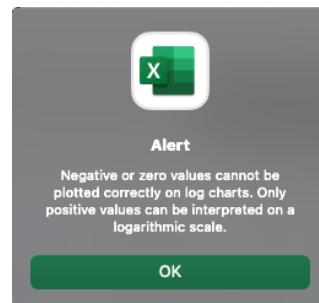
- a. Often a quick click on the year up/down solves the problem
- b. If the problem persists, the sweeps may have to be recomputed to represent the current combination of requirements and technology roadmaps
- c. It may be necessary to re-confirm the radio-button selections in Section 1,



2. Problem: Sometimes when applying changes, Excel complains about negative values for logarithmic charts.

Cause: On rare occasions, the system model may result in “0” values which cannot be properly plotted on logarithmic charts.

Solution: In our experience, these errors can at the current stage be ignored and “clicked-away”.

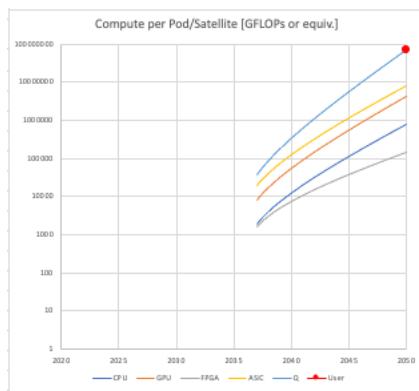


3. Problem: #REF (Reference Missing) in Section 5 (or on other pages)

Cause: This may be caused by some mismatch between functions for different use-cases that are not yet captured in the VBA code.

Solution: Check if any result is affected. Report the error (maybe save current configuration to one of the user-scenarios for sharing with the development team).

#### 4. Problem: Incomplete Graphs



Cause: Some of the numbers in **temp\_data** are negative, thus cannot be plotted on a log-plot. This can e.g. happen in the GEO case, when too much power is needed for communication and no power is left for compute, thus “negative” compute power would result (we tried to capture this with simple code though).

Solution: Either increase the power-envelope, reduce communication requirements or other power-consumers.

5. Problem: List of all named ranges/variables on the **Workbook\_Variables** sheet is no longer up to date.

Cause: You have added or removed variables.

Solution: Select cell A1 in **Workbook\_Variables**, Under “Formulas” -> Use in Formulas -> Paste Names -> (scroll to the bottom) Paste List ... and you will get an up to dated list of all named ranges/variables used in the workbook and in which cell they are defined.

6. Tip: Open multiple windows of the same workbook: Sometimes it’s helpful to see how changes in one sheet affect other sheets.

Cause: Too many sheets per workbook.

Solution: In the Window-Menu select New Window. This opens a new window of the same sheet. In the latter you can navigate to any sheet and changes in another sheet will be immediately visible to you.

# Administrative

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Contracting partner: KP Labs Sp. Z.o.o. (Poland)

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