W/B / Performance calculation Cessna 150M OH-CKY OM App 5B 10.8.2011

Deter	WEIGI	HT & BAL	ANCE	
Date://		MASS	ARM	MOM.
	BASIC MASS (BW)	579	ж.	457,4
PIC: Student:	FRONT SEATS		1	
	BAGGAGE AREA 1 (max 54 kg)		1,63	
Doute	BAGGAGE AREA 2 (max 18 kg)		2,13	
Route:	ZERO FUEL WEIGHT			
	FUEL (max 92,5 kg)		1,07	
Prepared:	RAMP MASS			
	TAXI FUEL	- 2	1,07	- 2,1
Aroramayadı	T/O MASS		_	
Approved:	TRIP FUEL (-)		1,07	
	LDG WEIGHT AT DEST.		-	

	BASIC INFO								
A/C	MTOW - kg	BW - kg	Usable fuel	Max. fuel	MOM - kgm				
ОН-СКҮ	798	579	128,5 I / 92,5 kg	144 I / 103 kg	457,4				

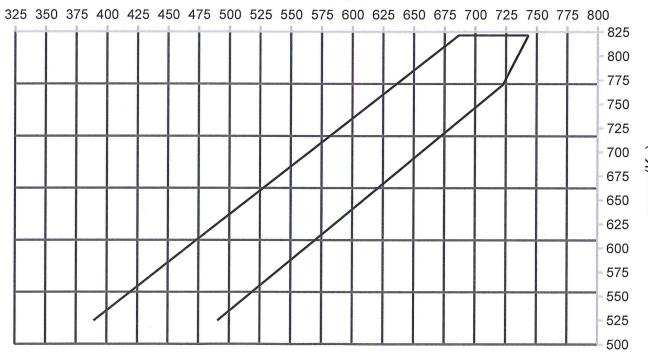
SEATS 1 & 2						
MASS	MOM					
kg	kgm					
50	50					
55	54					
60	59					
65	64					
70	69					
75	74					
80	79					
85	84					
90	89					
95	94					

BAGGAGE (max 54 kg								
MASS	MOM							
kg	Fwd	Aft						
2	3	4						
4	7	9						
6	10	13						
8	13	17						
10	16	21						
18	29	38						
30	49	-						
40	65	-						
50	82	-						
54	88	-						

FUEL							
Qty	Mass	Manage kana					
litres	kg	Moment - kgm					
5	3,6	3,8					
10	7,1	7,6					
20	14,2	15,2					
30	21,3	22,1					
40	28,4	28,4					
50	35,5	37,6					
60	42,6	42,6					
70	49,7	53,2					
80	56,8	60,8					
90	63,9	68,4					
100	72,0	77,0					
110	79,2	84,7					
120	86,4	92,5					
130	93,6	100,5					
140	100,8	107,9					
144	103,7	110,9					

Consumption = 38 l/h

moment (Kgm)



mass (Kg

T/O DISTANCE / SHORT FIELD

CONDITIONS:

Flaps 10° / Full throttle prior brake release / Paved, level, dry runway / Zero wind

NOTES:

- 1. Short field technique as specified on section IV.
- 2. Prior to takeoff from fields above 3000 ft elevation, the mixture should be leaned to give maximum RPM in a full throttle, static runup.
- 3. Decrease distances 10 % for each 9 kts headwind. For operation with tailwinds up to 10 kts, increase distances 10 % for each 2 kts.

4. For operation on dry grass runway, increase distances by 15 % of ground roll figure.

FLIGHT-	SPEEI	D/KIAS	PRESS	0°	C	10	°C	20	°C	30	°C	40	°C
MASS kg	ROTATE	AT15m	ALTITUDE ft	GRND ROLL m	TOTAL TO CLEAR 15m OBST								
757	50	54	0	195	363	212	393	230	424	247	456	267	489
1			1000	215	399	233	433	251	466	271	501	293	539
			2000	236	440	256	477	277	515	299	555	322	597
1			3000	261	488	282	527	305	570	329	616	355	666
			4000	287	541	311	585	335	634	363	686	392	744
1			5000	317	600	343	652	370	707	401	770	344	838
			6000	349	671	379	730	410	796	443	870	479	953
			7000	387	753	419	824	454	902	492	992	532	1094
	2		8000	428	853	465	936	504	1035	547	1148	591	1279

LANDING DISTANCE / SHORT FIELD

CONDITIONS:

Flaps 30° / Power idle / Maximum braking / Paved, level, dry runway / Zero wind

NOTES:

- Short field technique as specified on section IV
- Decrease distance 10 % for each 9 kts headwind. For operation with tailwinds up to 10 kts, increase distance 10 % each 2 kts.
- 3. For operation on dry grass runway, increase 45 % on ground roll figure.

4. If landing with no flaps becomes necessary, increase approach speed by 7 KIAS and increase landing distance 35 %

				3 /							9	
FLIGHT-	AIRSPEED	PRESSURE	0°	,C	10	°C	20)° C	30	°C	4	0°C
MASS	AT 15m	ALTITUDE ft	GRND ROLL	TOTAL TO CLEAR								
kg	KIAS		m	15m OBST								
757	50	0	137	354	142	362	148	370	152	378	157	386
	54	1000	142	361	148	370	152	378	158	387	163	316
		2000	148	370	152	378	158	387	163	396	169	405
		3000	152	378	158	389	165	398	171	407	175	415
		4000	158	389	165	398	171	407	177	418	183	427
		5000	165	398	171	407	177	418	183	427	189	437
		6000	171	408	177	418	184	430	191	439	197	450
		7000	178	419	184	430	191	439	198	451	204	462
		8000	184	430	192	442	198	451	206	463	212	474

TAKE-OFF RUN AVAILABLE	LANDING RUN AVAILABLE	
TAKE-OFF RUN REQUIRED	LANDING RUN REQUIRED	2
TAKE-OFF DISTANCE TO 15 m / 50 FT	LANDING DISTANCE FROM15 ft / 50 FT	-
CROSSWIND COMPONENT	HEADWIND COMPONENT	

RUNWAY AVAILABLE

EFHF 18/36 TORA = TODA = 1280 m, LDA 18 = 1080 m, LDA 36 = 1160 m

EFHF 09/27 TORA = TODA = 1024 m, LDA 09 = 1024 m, LDA 27 = 985 m

EFTP 06/24 TORA = TODA = LDA = 2700 m

EFTU 08/26 TORA = TODA = LDA = 2500 m

MAKSIMUM CROSSWIND-COMPONENT = 15 KTS

10.8.2012