

# A380X Procedures

For simulation purposes only



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# PRELIMINARY COCKPIT PREPARATION



# **AIRCRAFT POWER-UP ENGINES** ENG START selector...... NORM WIPERS BOTH WIPER selector ...... OFF **BATTERIES** ALL BAT pb-sw (BAT 1, ESS, 2, APU BAT)...... ON **EXT POWER** If the AVAIL light is on: EXT pb sw (2, 3, 1, 4)...... ON The AVAIL light goes off. The EXT power units should be connected, in the following order: - EXT 2 EXT 3 EXT 1

This order ensures the shortest start time of the display units.

At least two EXT power units are necessary to supply the entire electrical network.

If the flight crew anticipates high electrical demands (e.g. APU start or doors actuation):

- A third or a fourth EXT power unit is necessary to supply the entire electrical network, or
- The flight crew must temporarily reduce the electrical load when only two EXT power units are available, by setting the:
  - o CAB FANS pb-sw to OFF, and
  - o COOLG pb-sw to OFF.

EXT 4

Note: These temporary actions ensure that enough electrical power is available during a high electrical demand (e.g. to start the APU or to operate the doors). When the flight crew sets to OFF the CAB FANS pb-sw and the COOLG pb-sw, no degradation of the galley cooling system and of the air conditioning is expected for a few minutes, i.e. the time necessary to start the APU or to operate the doors.

As soon as it is possible, the flight crew should set to ON the CAB FANS pb-sw and the COOLG pb-sw.

When the APU is available, no reduction of load is necessary



## **COCKPIT LIGHTS**

COCKPIT LIGHTS AS RQ
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On the overhead panel, set the STBY COMPASS sw as required, and set to BRT or DIM the ANN LT as required.

On the pedestal, set the PEDESTAL FLOOD LT knob, the AMBIENT LT knob, and INTEG LT knob as required.

Adjust the brightness of the Display Units: PFD, ND, MFD, EWD, and SD.



# **OIS INITIALIZATION LAPTOP START** CAPT & FO LAPTOPS...... ON If the maintenance personnel did not set to ON the CAPT and FO laptops, on each lateral console, set the CAPT and FO laptops to ON. To turn on the laptops, slide the laptops ON/OFF sw to the right not more than 2 s. Set the laptop to ON sufficiently early to give enough time for laptop power on. Time for power on is approximately 5 min. BACKUP LAPTOP......ON The backup laptop rack is located in the FO lateral console. If the maintenance personnel did not set to ON the backup laptop, set the backup laptop to ON. To turn on the laptop, slide the laptop ON/OFF sw to the right for not more than 2 s. The backup laptop must be running to perform automatic data loading. **KEYBOARD AND CURSOR CONTROL UNIT (KCCU)** CCD sw and KBD sw 1 and 2...... ON **OIS APPLICATION START** OIT knob...... ON Turn the OIT knob in order to start the OIT and adjust brightness. OIT SLIDE sw...... NSS AVNCS LOGIN AS PILOT...... PERFORM The connection to NSS AVNCS takes a few minutes. During that time, the flight crew should launch the FLT OPS applications. OIT SLIDE sw...... FLT OPS LOGIN AS PILOT...... PERFORM NAV CHARTS......START OPS LIBRARY...... START **COMPANY COM(ACARS)INITIALIZATION** OIT SIDE sw...... NSS AVNCS INIT DATA...... SEND Send a request to the airline ground station via the INIT DATA button of COMPANY COM application. When the request is delivered on ground (I.e. an acknowledgement from ground is received onboard), the request message appears in the Sent items. Note: Send Initialization after check input data and display data. Note: Check Uplink message: CAT Recency, Full Thrust Recency And SOW Index Range.

Note: SOW Index Range isincluded for a month only if it is changed.



## **FLT OPS STATUS**

OIT SIDE sw		FLT OPS
In accordance w	vith Operators policy or if required by opera	ational regulation:
FLT OPS STS pag	ge	CHECK OIS VERSION
If required, the j	flight crew performs this check unless a spe	cific procedure is established as per
Operators policy	y to ensure that the correct version is on bo	ard.
	STS page, check OIS version number and conce by the Operator (e.g. on the company fli	•
NAV CHARTS		CHECK VALIDITY DATE
Check the validi	ity date of the NAV CHARTS on the FLT OPS	STS page.



ECAM/LOGBOOK CHECK	
ECAM RCL	
RCL pb	
LOGBOOK CHECK	
OIT SIDE sw	
<ul> <li>Check the deferred items list</li> <li>Crosscheck with ECAM recall</li> <li>Note the missing parts of the CDL deferred items</li> <li>Check the maintenance release status</li> <li>Enter the pre-flight parameters, in accordance with Operators policy.</li> </ul>	
MEC/CDL ITEMS CHECK	
OIT SIDE sw FLT OPS  MEL/CDL ITEMS (as appropriate) CHECK DISPATCH CONDITIONS	
Access the MEL and CDL items via the LOGBOOK STATUS panel of OPS LIBRARY.  MEL items are automatically actived in performance applications.  CDL items are automatically activated in performance applications, but without missing parts.	
AIRCRAFT ACCEPTANCE	
OIT SIDE sw	

<u>Note:</u> The aircraft acceptance can be performed later, but must be completed at the end of the Cockpit Preparation



## **APU AND FIRE TEST/APU START**

<u>1P</u>	RM
RMP 1 and 2ON	
FO switches on both RMPs.	
STBY RAD NAV key OFF	
COMMUNICATION FREQUENCIES	

#### Use:

- VHF will normally be selected for the active ATC communications
- VHF will normally be selected for the emergency (121.5 MHz)

<u>Note:</u> When A/C is parked at the gate, VHF2 can be used for ATIS or company frequencies to ensure that VHF3 is devoted to ACARS.

• VHF3 will normally be selected for ACARS (Also used for ATIS, and company radio in flight)

<u>Note:</u> When VHF3 is selected for voice mode, up/down links of all ACARS won't work, so it must be switched to data mode immediately after having finished using the voice.

Note: The speaker volume should be turned on for monitor during the flight.

The flight crew usually uses VHF3 for ACARS.

Turn the INT RECEPTION knob to adjust the volume and to enable contact with the ground crew.

### **APU AND ENG FIRE TEST**

Note:

The flight crew should apply the APU Start on Batteries supplementary procedure to start the APU on batteries. For more information, Refer to PRO-SUP-49 APU Start on Batteries. The flight crew should perform the APU and ENG fire tests when the APU is available.

APU FIRE pb-sw	CHECK IN and GUARDED
APU AGENT light	OFF
ENG 1(2)(3)(4) FIRE pb-sw	CHECK IN and GUARDED
ENG 1(2)(3)(4) AGENT 1 and 2 light	OFF
FIRE TEST pb	
•	

The flight crew should maintain the TEST pb pressed during the time of the test.

## TEST RESULT

Check that all ENG FIRE and APU FIRE detection and extinguishing systems are operative:

- The continuous repetitive chime sounds
- The MASTER WARN light flashes
- The ECAM displays the ENG 1(2)(3)(4) FIRE, APU FIRE and MLG BAY FIRE alerts
- All ENG FIRE pb-sw and the APU FIRE pb-sw come on in red
- All SQUIB lights of the ENG and APU AGENT pb come on
- All DISCH lights of the ENG and AUPU AGENT pb come on
- All FIRE lights on the ENG MASTER panel come on.



<u>PU START</u>	
APU MASTER SW pb-sw O	Ν
When the APU flap is fully open (I.e. FLAP OPEN indication appears on the APU SD page):	
APU START pb C	N
On the APU SD page, check that FLAP OPEN appears before setting to ON the APU START pb.	
<u>LEC</u>	
EXT PWR (if ON) AS RQF	₹D
The flight crew should keep ON the external power units to reduce the APU load, particularly in hot weather conditions.	



#### **OIS PREPARATION**

### **FMS/OIS FLT OPS STS**

The flight crew initializes the OIS with the planned city pair (FROM/TO) and flight number (FLT NBR). The OIS can be initialized automatically from the FMS or manually. Initialization of the FMS enables automatic initialization of the OIS Flight Ops applications (Performance and Navigation charts), the AOC and the Logbook. At aircraft power-up, the DATA/STATUS page is the default FMS page that appears. Display the FMS ACTIVE/INIT page by pressing the INIT key on the KCCU, or by selecting INIT in the ACTIVE menu on the General Menu Bar on the MFD.

If Company Flight Plan is received via ACARS: ACFT STATUS on the FMS DATA/STATUS page...... CHECK Check: The engine and aircraft types The active database validity period The pilot stored elements. Check if any waypoint, NAVAID, route, and runway appear. Review and delete them, if appropriate. RECEIVED CPNY F-PLN...... INSERT The Company Flight Plan is automatically loaded in the FMS. If the flight crew wants to insert the received flight plan in SEC 1(2)(3), they should clear the secondary flight plan by using the Delete button, before inserting the received flight plan. The Company Flight Plan can be received via ACARS even if the flight crew did not previously select CPNY F-PLN REQUEST. FLT NBR and FROM/TO on FMS ACTIVE/INIT page...... CHECK OIT SIDE sw...... FLT OPS FLT OPS STS page...... SELECT FLT OPS STS page...... CHECK UPDATED Check that FLT NBR and CITY PAIR in the OIS are in accordance with the FMS. If Company Flight Plan is not received via ACARS: OIT SLIDE sw...... FLT OPS FLT IOS STS page...... SELECT FLT NBR and FROM/TO...... INSERT Enter the FLT NBR and CITY PAIR on the FLT OPS STS page.

FLT OPS STS page...... CHECK UPDATED



### PRELIMINARY PERFORMANCE DETERMINATION

Compute the preliminary performance data in accordance with the technical condition of the aircraft and/or any other criteria that may impact the performance data (e.g. NOTAM, runway condition, aircraft configuration).

T.O PERF	START
Start the T.O PERF application.	
AIRFIELD DATA	OBTAIN
Obtain airfield data that are used for preliminary takeoff performance computation.	

If dispatch under MEL or CDL and in accordance with the logbook:

In the RUNWAY SELECTION panel, enter the runway characteristics

The flight crew should consider any NOTAM that affects the airport data.

Note: In anticipation of a possible runway change, select multiple runways, as applicable in the MULTIPLE RWY panel.

- o In the OUTSIDE CONDITIONS panel, enter the outside conditions
- o In the MEL/CDL panel, check selected items, if any
- In the AIRCRAFT CONFIGURATION panel, enter the aircraft configuration (e.g. aerodynamic configuration, air flow, thrust option)

Check the takeoff weight.

- Launch the computation and check the results.



# **COCKPIT PREPARATION**



## **OVERHEAD PANEL**

## WHITE LIGHTS ON THE OVERHEAD PANEL

	ng the scan sequence of the overhead panel:  'E LIGHTSOFF
	eral rule to turn off all white lights for all the systems during the scan sequence (Bottom to top,
<u>Note:</u> - -	Depending on Operators policy, the GND CONNECTION pb-sw on the MAINT panel and the REMOTE C/B CTL pb-sw on the maintenance ELEC panel may be kept ON during the cockpit preparation, if the maintenance personnel uses the Portable Multipurpose Access Terminal (PMAT) for maintenance purposes  During the scan of the overhead panel, check also that the only amber lights are the GEN FAULT lights.
RCDR RCDR GN	D CTL pb-swON
•	PT & PURS sw CAPT  APT/CAPT & PURS sw on the EVAC panel as per Company policy.
PROBE &	INDOW HEAT  WINDOW HEAT pb-sw AUTO  BE & WINDOW HEAT pb-sw should not be set to ON except in cold weather operations.
Many aire	ODE selector
<ul><li>This is a second of the contract of th</li></ul>	a complete alignment, if: is the first flight of the day e was a change in flight crew GPS is not available, and if the flight crew expects long segments in airspaces with poor radio AID coverage. flights, perform a fast alignment. wring the refuel operations or the Auto Ground Transfer (AGT), at least one IR should remain

<u>ELT</u>

ELT...... ARMED

operative. If the flight crew needs to align the IRs during the refuel operations or the AGT, they should switch off two IRs, then set them to NAV and wait for 40 s before switching off the third IR.



RESET BUTTONS (Left side)
EXTERIOR LIGHTS
STROBE sw
BEACON sw OFF
NAV swON
Set to ON the NAV sw, as required, to turn on the navigation and obstruction lights.
REMAINING EXTERIOR LIGHTS AS RQRD
<u>SIGNS</u>
SEAT BELTS swON
Set to ON the SEAT BELTS sw when refueling is complete.
NO SMOKING sw AUTO
EMER EXIT LT sw ARM
ENG START
ENG START selector NORM
<u>AIR</u>
APU BLEED pb-swON
Do not use the APU BLEED, if the maintenance personnel confirms that an HP ground air unit is
connected to the aircraft.
The flight crew should also check on the BLEED SD page, if there is pressure in the bleed air system, to
determine if an HP ground unit is connected.
XBLEED selector
AIR FLOW selector
As long as the number of passenger is not entered in the FMS, airflow is automatically adjusted for the
maximum number of passengers.
As soon as the number of passenger is entered, during FMS initialization, airflow is automactically
adjusted to the actual number of passengers.
CKPT selector
The temperature ranges between 18 ° C and 30 °C. The recommended temperature selection is
approximately 21.5 °C (about 10 o'clock). CABIN selector
Cabin temperature is controlled from the Flight Attendant Panel (FAP) only.
cash temperature is controlled from the ringht Attendant railer (LAF) only.



<u>ELEC</u>	
ELEC DC SD PAGE DISPLAY	
ALL BAT pb-sw (BAT 1, ESS, 2, and APU) OFF then ON	
<ul> <li>10 s after setting all BAT pb-sw to ON</li> </ul>	
On the ELEC DC SD page, check that the current charge of each battery is below 60 A, and is	
decreasing.	
<u>Fuel</u>	
TRIM TK FEED sw AUTO	
<u>MAINTENANCE</u>	
ALL LIGHTS OFF	
<ul> <li>Depending on Operators policy, the GND CONNECTION pb-sw on the MAINT panel and the REMOTE</li> </ul>	
C/B CTL pb-sw on the maintenance ELEC panel may be kept ON during the cockpit preparation, if the	2
maintenance personnel uses the Portable Multipurpose Access Terminal (PMAT) for maintenance	
purposes	
<ul> <li>Check that all lights are off. If not, select the corresponding pushbutton to turn off the light.</li> </ul>	
CARGO AIR COND	
CARGO AIR COND selectors AS RQRD	
<u>RMP 3</u>	
RMP 3 ON	
STBY RAD NAV key OFF	
<u>CVR</u>	
CVR TEST pb PRESS	
Press and release the pushbutton. If the test fails, the RECORDER CVR FAULT ECAM alert triggers 5 s	
after the flight crew presses the CVR TEST pb.	
<u>RESET</u>	
RESET BUTTONS (Right side) CHECK	
Check that all reset buttons are pushed.	



## **CENTER INTRUMENT PANEL**

<u>SWITCHING</u>	
SWITCHING	G selectorsNORM
Check that	the ATT HDG selector, the AIR DATA selector, and the FMS selector are set to NORM.
<u>ISIS</u>	
ISIS	CHECK
– Adjust	the brightness, as appropriate
– Check	
0	The airspeed
0	The barometric settings
0	The altitude indication
0	The attitude indications
0	The heading.
L/G GRVTY	
	swOFF
2, 0 0	
<u>CLOCK</u>	
	CHECK and SET AS NECESSARY
Check the	time and adjust, if necessary.
	ne and chronometer should be set to zero.
•	
A-SKID	
A CI/ID	ON



PEDESTAL
PARK BRK
PARK BRKON
If brakes are hot and chocks are in place:
PARK BRK OFF
Release the PARK BRK handle to improve brake cooling.
On the triple pressure indicator, check that the left and right brake pressure drops to zero.
BODY ACCU PRESSURE
BODY ACCU Pressure CHECK/REINFLATE
Check for normal indications.
The BODY ACCY pressure indication must be in the green band.
If required, press the ACCU REINFLATE pb to reinflate the BLG brake accumulators.
<u>ENG</u>
THRUST LEVERS
THRUST REVERSER LEVERS STOWED
ENG MASTER 1, 2, 3 AND 4 OFF
CKPT DOOR
CKPT DOOR swNORM



## MFD ATC COM On the MFD ATC COM/MSG RECORD page: MSG RECORD..... ERASE ALL Click on the ERASE ALL button in order to erase the record of all CPDLC messages of the previous flight before starting a new flight. This avoids confusing CPDLC messages from the previous flight with the ones of the current flight. • On the MFD ATC COM/CONNECT/CONNECTION STATUS page: If ADS services are expected: ADS...... CHECK ARMED **ATC CLEARANCE** ATC CLEARANCE...... THE MOST PROBABLE/OBTAIN The flight crew should obtain ATC clearance, or use the probable clearance. NAV CHARTS CLIPBOARD...... PREPARE At ATC clearance, or at any convenient time, the FO prepares the NAV CHARTS clipboard. **MFD SURV** On the MFD SURV/CONTROLS page:

SURV DEFAULT SETTINGS...... SELECT

- XPRD is set to AUTO. The SQWK code remains as previously selected
- TCAS is set to TA/RA and NORM
- WXR is set to AUTO
  - ELEVN/TILT is set to AUTO
  - MODE is set to WX
  - TURB is set to AUTO
  - GAIN is set to AUTO
  - WX on VD is set to ON
  - PRED W/S is set to AUTO
- All TAWS modes are set to ON.



MFD FMS INITIALIZATION				
FLIGHT PLA	IN INITIALIZATION			
FLIGHT I	PLAN INITIALIZATION COMPLETE, AS RQR			
Complet	e the FMS ACTIVE/INIT page, as required.			
<u>Note:</u>	For ATC needs, the crew should enter exactly the entire Flight number, as shown on the			
	ICAO flight plan, without inserting any space, on the MFD INIT page			
<u>Note:</u>	Enter the Cost Index will be decided in consideration with the economical efficiency by the			
	company.			
WINDS				
	ENTER AS APPROPRIAT			
	between the trip wind or the forecasted wind for climb, cruise, or descent, as appropriate.			
	average wind or forcasted wind from the computerized F-PLN.			
IRS ALIGNI	ΛFNT			
	ne GPS is available:			
_	and 3 CHECK NAV or ALIGN			
	MS POSITION/IRS page, check that the IRS are either aligned or in alignment.			
	ne GPS is not available, or is failed:			
	and 3 ALIGN			
, _,				
DEPARTUR	E SELECTION			
	URE SELECT/CHEC			
Select th	e runway, SID and TRANS, as appropriate, in the FMS DEPARTURE page. If the company route			
already	has the departure procedure, check this departure.			
NAVAIDS				
	S CHEC			
	MS POSITION/NAVAIDS page, check the VOR, ILS, and ADF, tuned by the FMS.			
	them, if required, and check that the ND and PFD (for the ILS) display the correct ident.			
_	DESELECTION AS RQRI			
	e list of deselected navaids on the FMS POSITION/NAVAIDS page. If the NOTAMs warn that			
	vaids are not available, deselect them.			



<b>FUEL AND LOA</b>	<u>D</u>
ZFW/ZFWC0	GINSERT
If the ZFWC	G and ZFW are not available, the flight crew can enter the expected values to enable:
– FMS pe	erformance predictions
– Optimu	m fuel distribution, if refueling is not complete.
BLOCK FUEL	INSERT
CAUTION	Some of the characteristic speeds that are displayed on the PFD (green dot, F, S, VLS) are based on the ZFW and ZFWCG that the flight crew enters in the FMS. In addition, when the first engine starts, the pitch trim is automatically set to the takeoff target, based on the entered ZFWCG and block fuel. It is therefore the Captain's responsibility to carefully check all of this data.
The fuel is a	utomatically distributed, in accordance with the entered ZFWCG and takeoff CG target.
If refuel is no	ot yet completed, the flight crew can enter the expected Fuel On Board (FOB).
_	ight crew enters the ZFWCG, ZFW and BLOCK FUEL, the FMS provides all the predictions, as EXTRA fuel, if any.
RTE RSV/FIN	IAL CHECK/INSERT AS APPROPRIATE
	T DEST CHECK/INSERT AS APPROPRIATE
Check that t	he minimum fuel at destination is in accordance with the required minimum fuel.
<b>TAKEOFF PERF</b>	<u>ORMANCE</u>
PRELIMINAF	RY TAKEOFF PERF DATAINSERT
Insert the pr	reliminary takeoff performance data on the T.O panel of the FMS ACTIVE/PERF page.
V1, VR and \	V2INSERT
TOGA/FLEX/	DERATED SELECT/INSERT
FLAPS	SELECT
ANTI-ICE	INSERT
THR RED/AC	CCEL ALTITUDE SET OR CHECK
NOISE PROC	CEDURE AS APPROPRIATE
TRANS ALTI	TUDE AS APPROPRIATE
EO ACCEL A	LTITUDE SET OR CHECK
CLIMB PERFOR	MANCE
	_B INSERT AS APPROPRIATE
	E or insert the selected derated climb on the CLB page of the FMS ACTIVE/PERF page, as
appropriate	
SPEED PRESELE	<u>ECTION</u>
	and CRZ panels of the FMS ACTIVE/PERF page:
	EDS AS RQRD



#### **ACTIVE F-PLN CHECK**

- 1. Lateral revision at departure airport
- 2. Lateral revision at waypoint for route modification, as appropriate
- 3. Vertical revision for climb speed limit/constraints in accordance with ATC clearance. Enter step altitudes as appropriate
- 4. Lateral revision for arrival.

Check the EOSID on the ND PLAN mode, if applicable

Check the F-PLN and the ND PLAN mode vs. The computerized flight plan or the navigation charts. Check the distance to destination along the F-PLN. Compare it with the toal distance computed by the computerized flight plan.

### SEC F-PLN

SECONDARY FLIGHT PLANS...... AS APPROPRIATE

Create the SEC 1, SEC 2, and SEC 3 F-PLN, as appropriate, using the FMS SEC INDEX page. The flight crew can use SEC 1 and SEC 2 to copy the active F-PLN, and to anticipate a runway change, an immediate return, or a diversion immediately after takeoff.

For an anticipated runway change, they must compute the takeoff performance accordingly. The flight crew can load an ATC flight plan in the SEC 3 only.

Note:

Before the flight crew creates a SEC 1(2)(3) flight plan, they should erase associated SEC 1(2) (3) data from the previous flight via the DELETE button on the SEC 1(2)(3) panel of the SEC INDEX page.

#### **ROUTE SUMMARY CHECK**

ROUTE SUMMARY...... CHECK versus ATC F-PLN Check the route summary via the FMS DATA/ROUTE page.

### **FMS DATA CROSSCHECK**

After the FO initialized the FMS, the CAPT crosschecks:

- The airfield data
- The IRS alignment
- The fuel and load data
- The takeoff performance data with the OIS takeoff performance data
- The F-PLN

### **NAV CHARTS CLIPBOAD**

NAV CHARTS CLIPBOARD.....IMPORT

The CAPT imports the NAV CHARTS clipboard that the FO prepared.



## **GLARESHIELD**

<u>LIGHTING</u>
INTEGRAL LIGHTS AS RQRD
LOUDSPEAKER
LOUDSPEAKER knob
Set the LOUDSPEAKER knob to approximately 1 o'clock position.
BAROMETRIC REFERENCE
BAROMETRIC REFERENCE SET
Set the QNH on the EFIS CP and on the ISIS.
Check the barometric reference and altitude indications on the PFD and on ISIS.
The maximum difference is:
<ul> <li>+/- 20 ft between both PFDs</li> </ul>
<ul> <li>+/- 30 ft between ISIS and PFD</li> </ul>
<ul> <li>+/- 25 ft between each PFD and airport elevation.</li> </ul>
Note: If the barometric unit is mercury inches, the flight crew may notice a discrepancy of 0.01 inHg
between the QNH values on the PFDs and the QNH values selected on the EFIS CP. This
discrepancy does not impact the altitude computation.
reaches of the control of the contro
EFIS CONTROL PANEL (EFIS CP)
ND MODE AND RANGE AS RQRD
MODE: Display the ARC mode on the ND, if the takeoff direction is in approximately the same direction as the departure clearance.
Display the NAV mode, if the change in direction will be more than 70°, after takeoff.
This enables the ND to display the area behind the aircraft.
RANGE: Set the minimum range to display the first waypoint after departure, or as required for weather purposes.
WX pb on EFIS CP OFF
Before pressing the WX pb on any EFIS CP, the flight crew should ensure that:
<ul> <li>No maintenance personnel is facing the aircraft within a distance less that 20 ft within an arc of 135° on either side of the aircraft centerline.</li> </ul>
<ul> <li>The aircraft is not directed toward any large metallic obstacle, such as a hangar, within 20 ft in an arc of 90° on either side of the aircraft centerline.</li> </ul>
The weather radar starts to emit as soon as the first engine is started if WX is selected on the EFIS CP to
display weather information.
OTHER EFIS OPTIONS AS RQRD

Select other EFIS options (e.g. constraints, waypoints, navaids, etc.) as required.



## **AFS CONTROL PANEL (AFS CP)**

The FO sets the AFS CP:	_
)ON	FD.
ORTH REF MAG	NO
neck that TRUE does not appear neither on the HDG/TRK window of the AFS CP, no on the PFD.	Che
PD/MACH window DASHED	SPD
DG/TRK window DASHED	HDO
_T window INITIAL EXPECTED CLEARANCE ALTITUDE	ALT
/S / FPA window DASHED	V/S
The CAPT crosschecks the AFS CP:	•
FS CP CROSSCHECK	AFS



#### **LATERAL CONSOLES**

### **OXYGEN MASK TEST**

Test the oxygen masks of each occupied station. During the test, check that the oxygen correctly reaches the oxygen mask by verifying that the oxygen mask blinker becomes yellow, and the flow of oxygen flow can be heard via the loudspeakers.

#### On the RMP:

INT/RAD sw...... INT

- On the mask stewage box:

OXYGEN MASK TEST..... PERFORM

## On the DOOR SD page:

The flight crew must perform this check after checking the masks of each occupied station, to ensure that the oxygen supply valve is open. Due to residual pressure between the oxygen supply valve and the oxygen mask, it may not be possible to detect that a supply valve is failed closed, during the oxygen mask test.

### **SLIDING WINDOWS**

#### **TAKEOFF BRIEFING**

TAKEOFF BRIEFING...... PERFORM

The purpose of the takeoff briefing is for the PF to inform the PM of the planned course of action for both normal and abnormal situations during takeoff, and to identify other operational risks.



# BEFORE PUSHBACK OR START



#### **BEFORE START CLEARANCE**

# **LOADSHEET** FINAL LOADSHEET...... CHECK Carefully check the final loadsheet, particularly for significant errors. Make sure that the loadsheet data is correct (e.g. correct flight number, aircraft, dry operating weight/CG, configuration, fuel on board). FUEL ON BOARD...... CHECK On the permanent data of the SD, check that the Fuel On Board (FOB) corresponds to the F-PLN and to the Loadsheet. ZFW/ZFWCG in FMS...... CHECK/REVISE The PF compares the ZFW/ZFCG of the loadsheet with the ZFW/ZFWCG entered in the FMS ACTIVE/FUEL & LOAD page, and revises them if necessary. ZFW/ZFWCG in FMS...... CROSSCHECK The CAPT crosschecks the ZFW / ZFWCG entered in the FMS ACTIVE/FUEL& LOAD page. LOADSHEET TOCG AND ECAM GWCG...... CROSSCHECK Check the loadsheet TOCG vs. the ECAM GWCG: If the difference is less than 1 %, no further action is required The ECAM GWCG is reliable. If there is more than 1 % difference, check that the ZFW and the ZFCG are correctly entered in the

- FMS.
- If not, revise the ZFW / ZFWCG in the FMS and compare the ECAM GWCG and the loadsheet TOCG
- If the difference is confirmed, compute again the loadsheet using the trim tank fuel quantity displayed on the FUEL SD page For more information on how to compute the loadsheet using trim tank fuel quantity, Refer to PER-LOD-FDC Trim Tank Adjustment.
- If the difference between the loadsheet TOCG and the ECAM GWCG is still greater than 1 %, suspect an abnormal fuel distribution. Confirm the fuel distribution with load control and compute again the loadsheet accordingly.

ECAM GWCG	CHECK WITHIN OPERATIONAL LIMITS
Check that the Gross Weight CG displayed	on the permanent data of the SD is within the operational
limits using the OIS Loadsheet application	or referring to the loadsheet.



THS FPR_ in FMS ACTIVE/PERF page INSERT T.O CG (in %)  Use the TOCG indicated on the loadsheet as the reference for insertion of the T.O CG in the FMS.
Use the TOCG indicated on the loadsheet as the reference for insertion of the T.O CG in the FMS.
THS FOR_ in FMS ACTIVE/PERF page CROSSCHECK
FINAL LOADSHEET SIGN and EXPORT
If the loadsheet is modified, or if required by the authorities or by the airline policym the Captain sends
the loadsheet to the ground via EXPORT function in the LOADSHEET application.
<ul> <li>If the ECAM GWCG is not within the operational limits:</li> </ul>
AUTO GND XFR pb-swON
Ground transfer is activated to automatically obtain the ground CG target in accordance with the final
ZFW / ZFWCG values entered in the FMS .
<ul> <li>If time permits, continue the Automatic Ground Transfer (AGT) until the FUEL AUTO GND XFR</li> </ul>
COMPLETED alert triggers
$-\hspace{0.1cm}$ If departure is imminent, monitor the ECAM GWCG during the AGT . The flight crew can manually
stop the AGT when the ECAM CG is within the operational limits.
Note: - The flight crew can obtain the amount of fuel that should be transferred from/to the trim
tank to be inside the operational envelope, via the Loadsheet application - The flight crew
should not launch the automatic ground transfer if the aircraft is moving (e.g. during
towing.) The AGT is inhibited as soon as at least two engines are running.
AUTO GND XFR MONITOR
THS FOR _ in FMS ACTIVE/PERF page INSERT T.O CG (in %)
Use the TOCG indicated on the loadsheet as the reference for the insertion of the T.O CG in the FMS.
THS FOR _ in FMS ACTIVE/PERF page CROSSCHECK
FINAL LOADSHEET SIGN and EXPORT
If the loadsheet is modified, or if required by the authorities or by the airline policym the Captain sends
the loadsheet to the ground via EXPORT function in the LOADSHEET application.
TAKEOFF DATA
OIS FINAL TAKEOFF PERF CONFIRM or RECOMPUTE
<ul> <li>If takeoff conditions did not change, verify and confirm that the preliminary takeoff data are still</li> </ul>
valid
<ul> <li>If takeoff conditions changed, calculate the final takeoff performance, using the T.O PERF</li> </ul>
application on the OIS.
FMS TAKEOFF DATA CHECK/REVISE, AS REQUIRED
The FO checks or revises the takeoff data in the T.O. panel of the FMS ACTIVE/PERF page.
REVISED FMS TAKEOFF DATA CROSSCHECK
The CAPT crosschecks the takeoff speeds and the flexible temperature. The CAPT crosschecks the FMS
entries made by the FO, using the XCHECK WITH AVNCS function in the T.O PERF application.
The flight crew should pay particular attention in determining the takeoff configuration.
<ul> <li>In accordance with Airlines policy or if required by operational regulation:</li> </ul>
FINAL TAKEOFF PERFORMANCE STORE IN CURRENT EFF



## **SEATING POSITION**

SEATS, SEAT BEL	LTS, HARNESSES, RUDDER PEDALS, ARMRESTS	ADJUST
	ectly adjusted when the pilot's eyes are in line with the red ball and the who	ite light. For
<u>HUD</u>		
HUD		DEPLOY
HUD knob		ADJUST
	O and adjust the brightness according to conditions	AS RORD
Select the declut	tter mode as required. The flight crew should select the crosswind mode or not within the display area of the HUD.	
CAUTION	In order to avoid head contact with the HUD, the flight crew should:	
	Tight the seat belts and the fifth strap for takeoff and landing	
	Maintain the seat belts and the fifth strap fasten, in all other flig	ght phases
<u> </u>		
<u>MFD</u>		
MFD	IN TAKEOFF CO	NFIGURATION
<ul> <li>ACTIVE/F-PI</li> </ul>	LN page on the PM side	
• T.O panel o	f the ACTIVE/PERF page on the PF side.	
EXTERNAL POWER		
EXT PWR		CHECK AVAIL
Check that the E	EXT PWR pb-sw are set to AVAIL before the request of the external power o	lisconnection.
EXT PWR DISCO	NNECTION	REQUEST
LP GROUND CART		
LP GROUND CAF	RTS CHECK D	SCONNECTED
If LP ground card	ts are used for air conditioning, request disconnection before engine start.	If LP ground
	cted during the engine start, air pressure in the mixer unit may be detected	l out of range,
and the associat	ted alert COND MIXER PRESS REGUL FAULT may trigger.	
BEFORE START CHE	ECKLIST DOWN TO THE LINE	
BEFORE START C	C/L down to the line	COMPLETE



	AT START CLEARANCE
PUSHBACK/ST	ART UP CLEARANCE
PUSHBACK/ST/	ART UP CLEARANCE OBTAIN
Obtain ATC pus	shback/start up clearance and ground crew clearance.
<u>IDEO</u>	
TAXI VIDEO	AS RQRD
As required, se	t the TAXI video on the PFD or on the SD to see the aircraft vicinity.
/INDOWS AND [	DOORS .
WINDOWS AN	D DOORS CHECK CLOSEI
<ul><li>On the DO</li></ul>	OR SD page, check that all doors and windows are closed
<ul><li>When requ</li></ul>	uired by local authorities, check that the cockpit door is closed and locked (I.e. OPEN doe
-	r on the CKPT DOOR light).
• •	
<u>GHTS</u>	
	ON
When cleared	for start or pushback, set to ON the BEACON sw.
,	,
HRUST LEVERS	
HRUST LEVERS	IDI
THR LEVERS	
	The engines start regardless of the thrust lever position. If the thrust levers are not
THR LEVERS	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position,
THR LEVERS	The engines start regardless of the thrust lever position. If the thrust levers are not
THR LEVERS	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.
THR LEVERS  CAUTION  ARKING BRAKE	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbac	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Sk is not required:
THR LEVERS  CAUTION  ARKING BRAKE A  1. If pushbac  PARK BRK	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  ik is not required:  ON
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbac  PARK BRK  On the triple pi	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  ON  Treessure indicator, the left and right BRK pressure indications should be above 3 500 PSI.
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbace PARK BRK  On the triple pide  If pushbace	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  ik is not required:  ON  ressure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  ik is required:
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbac  PARK BRK  On the triple properties of the park BRK	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  ON  Tressure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Ek is required:  OF
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbace PARK BRK  On the triple picture of the pushbace PARK BRK  CAUTION	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Is is not required:  ON  Tressure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Is is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbac  PARK BRK  On the triple pi  If pushbac  PARK BRK  PARK BRK  CAUTION  N/W STEER DIS	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  Cressure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Ek is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.  SC MEMO.  CHECK DISPLAYED
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbac  PARK BRK  On the triple pi  If pushbac  PARK BRK  CAUTION  N/W STEER DIS  In the case of a	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  ON  Tessure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Ek is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.  SC MEMO
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbac  PARK BRK  On the triple properties  If pushbac  PARK BRK  CAUTION  N/W STEER DIST  In the case of a towing position	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  Oressure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Ek is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.  SC MEMO.  CHECK DISPLAYED pushback (towbarless or conventional), the ground crew should set the tow pin to the
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbace PARK BRK On the triple properties of the case of a towing position.	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Is is not required:  Or ressure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Is is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.  IS C MEMO
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbac  PARK BRK  On the triple properties  If pushbac  PARK BRK  CAUTION  N/W STEER DIST  In the case of a towing position	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  ON  Tessure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Ek is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.  EC MEMO
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbace PARK BRK On the triple properties of the case of a towing position.	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  ON  Tessure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Ek is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.  EC MEMO
THR LEVERS  CAUTION  ARKING BRAKE  1. If pushbace PARK BRK On the triple properties of the case of a towing position.	The engines start regardless of the thrust lever position. If the thrust levers are not set to IDLE, then thrust rapidly increases to the corresponding thrust lever position, causing a hazardous situation.  AND NOSEWHEEL STEERING  Ek is not required:  Orcessure indicator, the left and right BRK pressure indications should be above 3 500 PSI.  Ek is required:  OF  Do not use the brakes during pushback, unless required due to an emergency.  EC MEMO

When the pushback is completed, set to ON the PARK BRK handle, and ask the ground crew for towbar disconnection.



## BEFORE START CHECKLIST BELOW THE LINE



# **ENGINE START**



#### **AUTOMATIC ENGINE START**

ENG START selector...... IGN START The ENG SD page appears. WAIT FOR 10 s The flight crew should wait approximately for 10 s before the ENG MASTER lever of the engines 2 and 3 are set to ON. A 10 s delay permits to detect the loss of a thrust reverser function. If no ENG 2(3) REVERSER FAULT ECAM alert appears: START ENGINES 1 and 2...... ANNOUNCE For any operational reason, any engine can be started first. Note: ENG MASTER 1, then 2...... ON Set the ENG MASTER levers to ON when all amber crosses and messages no longer appear on the engine parameters of the EWD and of the ENG SD page. Note: Parameter callouts are not mandatory. Note: The level of N2 vibration may exceed 5 units and the indication may pulse for a short period of time during the engine start sequence. Depending on the time between the engine is shutdown and the next engine start, the HP rotor can bow due to the thermal stabilization of the engine. As a result, during the engine start sequence, the N2 vibration indication may exceed 5 units. In order to minimize the rotor bow effect during the engine automatic start sequence, the FADEC cranks the engine for 20 s after the N2 reaches 20 %. • When the engine reaches idle (I.e. AVAIL appears on the EWD): ENG IDLE PARAMETERS...... CHECK NORMAL At ISA, sea level, with bleed off, check that: • THR is approximately 3 % • EGT is approximately 455 °C • N1 is approximately 20 % • N2 is approximately 66 % FF is approximately 1 400 lbs/h. START ENGINES 3 and 4...... ANNOUNCE ENG MASTER 3 then 5...... ON Apply the same procedure as indicated for engines 1 and 2.



<ul> <li>If ENG 2(3) REVERSER FAULT ECAM alert appears:</li> </ul>	
ENG START selector	NORM
<ul> <li>When XX appears on ENG parameters:</li> </ul>	
ENG START selector	IGN START
The action to set the ENG START selector to NORM then	to IGN START enables to reset the
electrical power of the FADEC and ETRAC, and to clea	r thrust reverser failure.
<ul> <li>If ENG 2(3) REVERSER FAULT ECAM alert no longer a</li> </ul>	ppears after 10 s :
ENGINES 1 and 2	START
Apply the same procedure as previously indica	ted under condition "If no ENG 2(3) REVERSER
FAULT ECAM alert appears".	
ENGINES 3 and 4	START
Apply the same procedure as previously indicated under	condition "If no ENG 2(3) REVERSER
FAULT ECAM alert appears".	



# **AFTER START**



#### **AFTER START**

FΝ	G	<b>STA</b>	RT	SFI	<b>FCT</b>	OR
LIN	v	217		JLL		$\mathbf{v}$

ENG START selector...... NORM

- Setting to NORM the ENG START selector indicates the end of the start sequence if the ENG START selector remains in the IGN START position, this prevents a continuous ignition on ground. The flight crew should set the ENG START selector to NORM then to IGN START to activate continuous ignition on ground. The flight crew should then perform the AFTER START actions.
- The WHEEL SD page appears instead of the ENG SD page

<u>Note:</u> If the flight crew does not set the ENG START selector to NORM, the WHEEL SD page automatically appears instead of the ENG SD page 30 s after the last engine start. To prevent thermal shock, the engine should run at idle or taxi thrust for a minimum of 3 min (5 min recommended) prior to takeoff.

### **APU BLEED**

APU BLEED pb-sw...... OFF

- The APU bleed valve closes
- All engine bleed valve opens
- All crossbleed valves close.

### **ENG ANTI-ICE**

All ENG ANTI-ICE pb-sw...... AS RQRD

- Set to ON the ENG ANTI-ICE pb-sw , if:
  - o Icing conditions exist or are anticipated, or
  - $\circ$  Standing water, slush, ice, or snow is on the taxiways or runway when the OAT is less than 10 °C .
- When taxiing in icing conditions, if temperature is less than + 3 °C:
  - With freezing fog conditions: the flight crew must apply the ice shedding procedure
  - Without freezing fog conditions: In order to shed any possible ice on the fan, the flight crew must increase the thrust to 60 % N1 momentarily on two symmetric engines at a time (i.e. 1 and 4, 2 and 3) with parking brake on (or brake with pedals). The engines run-up must be performed at least every 30 min of total taxi time (including taxi-in and taxi-out time) and when lined-up before applying the takeoff thrust.

CAUTION

Pay particular attention to the aircraft movement during this procedure. If during thrust increase, the aircraft starts to move, immediately retard the thrust levers to IDLE.

If the aircraft moves, apply the thrust on only one engine at a time.

### APU

– If the APU is not required:

APU MASTER SW pb-sw...... OFF



GROUND SPOILERS
GND SPLRS ARM
RUDDER TRIM
RUDDER TRIM ZERO
If the rudder trim position is not zero, press the RESET pb on the RUDDER TRIM panel.
<u>FLAPS</u>
FLAPS SET
<ul> <li>Set the FLAPS for takeoff</li> </ul>
<ul> <li>Check the slats and flaps position on the slats/flaps display of the PFD</li> </ul>
<ul> <li>If taxiing in icing conditions with rain, slush, or snow, maintain the slats/flaps retracted, until the aircraft reaches the holding point of the takeoff runway, in order to prevent contamination of the slats/flaps mechanism.</li> </ul>
PITCH TRIM
PITCH TRIM CHECK
Check that the pitch trim is set to the takeoff target on the pitch trim display of the PFD.
ECAM STATUS
ECAM STATUS CHECK
<ul> <li>The F/O checks and the CAPT crosschecks that there is no status reminder (STS) on the EWD</li> </ul>
<ul> <li>If STS appears on the EWD, press the STS pb on the ECP to display the STATUS page.</li> </ul>
GROUND CREW
CLEAR TO DISCONNECT ANNOUNCE
The ground crew will:
Remove the chocks
Remove the tow pin
Disconnect the interphone
Make the hand signal on the left or right side.
N/W STEER DISC MEMO

N/W STEER DISC MEMO...... CHECK NOT DISPLAYED



#### **FLIGHT CONTROLS**

FLIGHT CONTROLS...... CHECK
The flight crew should perform the flight control check when the slats/flaps are set to the takeoff
configuration.

The CAPT remains silent, and applies full longitudinal and lateral sidestick deflection. On the F/CTL SD page, the FO checks full travel of all elevators and all ailerons, and the correct deflection and retraction of all spoilers. The FO announces "full up", "full down", "neutral", "full left", "full right", "neutral", when each applicable full travel/neutral position is reached. The CAPT remains silent, and checks that the FO calls are in accordance with the sidestick order.

<u>Note:</u> To reach full travel, maintain full sidestick for a sufficient period of time.

- The CAPT remains silent, presses the PEDAL DISC pb to disconnect the NWS and applies full left rudder, full right rudder, and neutral. The FO follows on the rudder pedals and announces "full left", "full right", "neutral" as each full travel/neutral position is reached. Note: To reach full travel, apply full rudder for a sufficient period of time.
- The FO applies full longitudinal and lateral sidestick deflection, remains silent, and checks full travel and the correct sense of all elevators and all ailerons, and the correct deflection and retraction of all spoilers, on the F/CTL SD page.
- <u>Note:</u> The F/CTL SD page automatically appears during 20 s after any flight crew action on the sidestick or on the rudder pedals
  - After the F/CTL checks, the rudder trim returns to a position between  $\pm 3$  °. As a result, the rudder trim check at neutral in the AFTER START C/L may appear as not completed (I.e. in blue).

#### **AFTER START CHECKLIST**

AFTER START C/L...... COMPLETE



# TAXI



#### TAXI

The Captain will turn ON then OFF the Landing Light/Taxi Light/Turnoff Light in response to the Ramp Coordinator's signal.

#### **CLEARANCE**

TAXI CLEARANCE...... OBTAIN

#### **EXTERNAL AND TAXI AID CAMERA SYSTEM (ETACS)**

TAXI pb on EFIS CP...... AS RQRD

#### **AIRPORT NAVIGATION**

Direct visual obvservation out of the cockpit windows remains the primary means of taxiing. If there is a conflict between the outside and the OANS display, the reference must be the outside view.

- If necessary, set the ND RANGE selector to ZOOM to activate the Onboard Airport Navigation System (OANS). Then, select ARC, ROSE, or PLAN mode, as appropriate
- If the DATABASE CYCLE NOT VALID message appears on the STATUS panel, check the active database cycle validity period, and swap the active and second database.

#### **EXTERIOR LIGHTS**

RWY TURN OFF & CAMERA sw..... AS RQRD

#### **PARK BRK**

PARK BRK..... OFF Check that the left and right brake pressure is zero on the triple pressure indicator.

#### **THRUST LEVERS**

THRUST LEVERS...... AS RQRD

- Little, if any, power above idle thrust can be necessary to move the aircraft (e.g. approximately up to 10 % THR at heavy weight, with a uphill taxi slope). Excessive thrust application on engines can result in exhaust-blast damage (e.g. on airport signs). The flight crew should use symmetrical thrust. When the aircraft begins to move, less thrust is necessary.
- If the inner engines are located over unconsolidated or unprepared ground during taxi, avoid high thrust settings on the inner engines (i.e. engine ingestion (FOD) risk increases). If additional thrust is necessary, the flight crew should preferably use the outer engines.
- The use of engine anti-ice increases the ground idle thrust
- If the aircraft was parked for a long time (i.e. more than 6 h) in high tire temperature conditions and with a high aircraft weight, then the flight crew can feel a "square wheel effect" when the aircraft begins to move.



#### **BRAKES**

BRAKES CHECK		
CAUTION	If the aircraft was parked in wet conditions for a long time, the first brake application	
	at low speed is less effective	

- When the aircraft begins to move, the PF smoothly presses the brake pedals to check the braking efficiency of the normal braking system. The FO checks that the left and right BRK pressure on the triple pressure indicator remains at zero.
- The maximum taxi speed should be 30 kt in a straight line on long taxiways, and 8 to 10 kt on a sharp turn. It is difficult to assess the ground speed. Therefore, monitor the ground speed on the ND or on the ETACS video. Avoid continuous brake application. When the aircraft exceeds 30 kt with idle thrust, smoothly brake, and decelerate to 10 kt. Release the brakes, to enable the aircraft to accelerate again.

#### **NOSEWHEEL STEERING**

NOSEWHEEL STEERING...... AS RQRD

- Use smooth and progressive handwheel inputs.
- The nosewheel steering angle is limited to 70 °.

#### **ATC CLEARANCE**

#### **TAKEOFF DATA**

If the takeoff data changed, or in the case of a runway change, prepare updated takeoff data and recompute takeoff performance, as appropriate.

#### **TAKEOFF DATA COMPUTATION**

• If multiple runway selection was used in anticipation of the runway change:

T.O PERF	ACCESS
SELECTED RWY AND INPUT DATA	. CHECK
Check that the new runway was previously selected and check that the conditions used for t	akeoff
performance computation are still valid.	
TAKEOFF DATA	CHECK
Check that the takeoff data for the selected runway.	

If multiple runway selection was not used, or if the takeoff conditions changed:

T.O PERF	ACCESS
NEW RUNWAY/NEW CONDITIONS	
TAKEOFF PERF	-
Launch the computation for the new selected runway or the new takeoff conditions.	651111 6112
TAKEOFF PERFORMANCE RESULTS	CROSSCHECK



# IN THE FMS ACTIVE/PERF PAGE

• In the case of ATC clearance or takeoff change:

• In the case of ATC clearance or takeoff change:	
TAKEOFF PERF DATA	
Update the FMS in accordance with the new results of the takeoff pe	erformance computation: I.e.
update the T.O panel of the FMS ACTIVE/PERF page, and the FMS ACTIVE	CTIVE F-PLN page.
If the flight crew anticipated the runway change in a secondary fligh	t plan, activate this SEC F-PLN.
V1, VR AND V2	UPDATE
FLX T.O TEMPERATURE	UPDATE
FLAPS	UPDATE
FMS UPDATES	CROSSCHECK
The CAPT crosschecks the new FMS inputs.	
FLAPS lever	AS APPROPRIATE
Set the FLAPS for takeoff	
AFS/FLIGHT INSTRUMENTS	
<ul> <li>In the case of ATC clearance or takeoff runway change:</li> </ul>	
F-PLN (SID, TRANS)	REVISE or CHECK
Carefully confirm that the ATC clearance agrees with the FMS, if NA	V mode is to be used.
INITIAL CLIMB SPEED AND SPEED LIMIT	REVISE or CHECK
Revise the SPD panel of the FMS ACTIVE/VERT REV page associated	with the departure, or the CLB
waypoint.	
CLEARED ALTITUDE ON AFS CP	SET
On the PFD, crosscheck and confirm the first cleared altitude.	
HDG ON AFS CP	PRESET, AS RQRD
<ul> <li>If the ATC requires a heading after takeoff, or in the case of</li> </ul>	a radar vector departure, preset
the heading on the AFS CP . NAV disarms.	
<ul> <li>RWY TRK mode maintains the aircraft on the runway track up</li> </ul>	ntil HDG engages
FD	CHECK SELECTED ON
PFD/ND	CHECK
Check that the PFD and the ND indications (e.g. airspeeds, initial tar	get altitude, heading, slats/flaps
configuration, FMA modes) are in accordance with the departure.	
MFD	. IN TAKEOFF CONFIGURATION
<ul> <li>The PF displays the T.O panel of the FMS ACTIVE/PERF page</li> </ul>	
<ul> <li>The PM displays the FMS ACTIVE/F-PLN page.</li> </ul>	

## **TAKEOFF BRIEFING**

TAKEOFF BRIEFING...... CONFIRM The takeoff briefing should usually be only a brief confirmation of the entire takeoff briefing that is done at the gate. The flight crew should address any clearance changes at this time.



## **AUTO BRK**

RTO pb...... ARM

- The RTO ARM light comes on
- BRK RTO appears on the FMA
- The flight crew can arm the AUTO BRK in the RTO mode, even if the PARK BRK is set to ON
- The RTO mode improves the safety in the case of an aborted takeoff. If the takeoff is aborted, the AUTO BRK system applies maximum braking as soon as the thrust levers are set to idle, and if the speed is above 72 kt (i.e. corresponding to the order of ground spoiler extension). Below 72 kt, the AUTO BRK does not activate, because the ground spoilers do not automatically extend.

#### **ATC**

ATC CODE...... CONFIRM/SET

## **FINAL CHECK**

TO CONFIG pb TEST
Press the TO CONFIG pb on the ECP, and check that the EWD displays the T.O CONFIG NORMAL memo.
T.O MEMO NO BLUE LINE
Check that the EWD does not display any blue line in the T.O memo section.
CABIN READY
Check CABIN READY on the EWD, or obtain the report from the chief: "Cabin ready for takeoff"

## **BEFORE TAKEOFF CHECKLIST DOWN TO THE LINE**



# **BEFORE TAKEOFF**



# BEFORE TAKEOFF

TAKEOFF OR LINE UP CLEARANCE	
TAKEOFF OR LINE UP CLEARANCEOBTA	λIN
<u>CABIN CREW</u>	
CABIN CREW ADVI About 3 minutes prior to the estimated takeoff time, captain call "TAKEOFF SIGNAL"	SE
<u>PACKS</u>	
<ul> <li>PACKS 1 and 2</li></ul>	
EXTERIOR LIGHTS	
EXTERIOR LIGHTS	ET
<u>ETACS</u>	
<ul> <li>If the ETACS was used during the taxi:</li> <li>TAXI pb on EFIS CP</li></ul>	
EFIS CP	
ND RANGE selector	RD
EFIS OPTIONS (WX pb or TERR pb)	
TRAF pb	Νכ



# **BEFORE ENTERING THE RUNWAY**

APF	ROACH PATH CHECK CLEAR OF TRAFFIC
STR	OBE sw ON
Set	the STROBE sw to ON to cross or enter a runway.
TAk	EOFF RUNWAY CONFIRM
Cor - - -	firm that the line up is performed on the intended runway. Useful aids are: The runway markings The runway lights Be careful that in low visibility, edge lights could be mixed up with the center line lights. The ILS signal If the runway is ILS equipped, the flight crew can press the LS pb: The LOC deviation should be centered after line up. The runway symbol on the ND.
<u>SLIDIN</u>	G TABLE
SLII	ING TABLE STOWE
<u>TCAS</u>	
TA	b TA ONLY or TA/RA
The Ope	RA is the default mode of the TCAS. flight crew may use the TA ONLY mode in specific airports, and for specific procedures (identified by rators) that may provide resolution advisories that are neither wanted nor appropriate (e.g. closely- ced parallel or converging runways).
<u>BEFOR</u>	E TAKEOFF CHECKLIST BELOW THE LINE
BEF	ORE TAKEOFF C/L below the line COMPLETE

Read the checklist below the line, when the flight crew obtains the line up clearance.



# **TAKEOFF**



## TAKEOFF

TAKEOFF
THRUST SETTING
Rolling takeoff is permitted
TAKEOFF ANNOUNCE
THR
Apply 25 % of THR on all four engines with the brakes set to on. However, the flight crew can release the brakes to perform a rolling takeoff.
<ul> <li>If the crosswind is at, or below 23 kt, and there is no tailwind:</li> </ul>
BRAKES
THRUST LEVERS
<u>Note:</u> The ECAM displays the ENG SD page, instead of the WHEEL SD page.
<ul> <li>If the crosswind is above 23 kt, and/or in the case of tailwind:</li> </ul>
BRAKES
THRUST LEVERS 50 %
At 20 kt ground speed:
THRUST LEVERS FLX or TOGA
Note: When FO is PF, After calling FMA "MAN FLEX", the Captain's hand must be on the
thrust levers until V1, and FO's hand must be removed from the thrust levers.
CHRONO START
DIRECTIONAL CONTROL
PFD/ND SCAN  – Check the FMA on the PFD:
MAN FLX or MAN TOGA
Note: If an ILS is tuned associated with the departure runway, RWY mode appears. In all other cases, no lateral mode appears until the aircraft lifts off.
<ol> <li>If GPS PRIMARY LOST, check on the ND the FMS position: I.e that the aircraft is on the runway centerline. The FMS updates the aircraft position at takeoff, only if GPS PRIMARY LOST.</li> </ol>
TAKEOFF THRUST CHECK

Before the aircraft reaches 80 kt, check the thrust setting, i.e. check that the actual THR of each engine

reached the thrust rating limit and the thrust target indicated by the blue dot.



1111/031 3E1	ANNOUNCE
PFD and ENG indications	SCAN
Scan the airspeed and the THR throughout the takeoff.	
<u>AT 100 KT</u>	
ONE HUNDRED KNOTS	ANNOUNCE
<ul> <li>Below 100 kt, the PF can decide to abort the takeoff, depending on the</li> <li>Above 100 kt, rejecting the takeoff is a more serious matter.</li> </ul>	ne circumstances
<u>AT V1</u>	
V1	MONITOR or ANNOUNCE
<u>AT VR</u>	
ROTATION	ORDEF
ROTATION	PERFORM
<ul> <li>If the PF uses the HUD:</li> <li>At VR, initiate the rotation, and bring the inverted T toward the horiz</li> <li>After liftoff, fly the velocity vector and follow the SRS using the flight put of the PF does not use the HUD:</li> <li>At VR, initiate the rotation toward a pitch attitude of 12.5° (10°, if a lange of the liftoff, follow the SRS pitch command bar.</li> <li>Note: If a tail strike occurs, the ECAM triggers the TAIL STRIKE alert.</li> <li>At 30 ft, if NAV is armed (i.e. the flight crew did not preset any HDG of engages</li> <li>If NAV is not armed, RWY TRK engages at 50 ft and remains displayed another lateral mode.</li> </ul> WHEN POSITIVE CLIMB	oath director.  one engine is failed)  r TRK), NAV automatically
POSITIVE CLIMB  Announce positive climb, when the vertical speed indication is positive and increased	
L/G UP	
L/GAP	
POSITIVE CLIMB Announce positive climb, when the vertical speed indication is positive and increased L/G UPL/G	d the radio altitude has

When the aircraft reaches 1 500 ft, or at the thrust reduction altitude, whichever occurs the

first, the CRUISE SD page appears instead of the ENG SD page.

Note:



# **ABOVE THE ACCELERATION ALTITUDE**

At F speed:

Note: For takeoffs in CONFIG 1+F, F speed does not appear.	
FLAPS 1	ORDER
FLAPS 1	SET
<ul><li>At S speed:</li></ul>	
FLAPS ZERO	ORDER
FLAPS ZERO	SET
EXTERIOR LIGHTS	SET
<ul> <li>Set to OFF the NOSE sw and the RWY TURN OFF &amp; CAMERA sw</li> </ul>	
GND SPLRS	DISARM



# AFTER TAKEOFF



AFTER TAKEOFF	
<u>APU</u>	
<ul> <li>If the APU was used to supply the air conditioning during takeoff:</li> </ul>	
APU BLEED pb-sw	OFF
APU MASTER SW pb-sw	OFF
<u>TCAS</u>	
<ul> <li>If the takeoff was performed with TA ONLY:</li> </ul>	
TA pb	TA/RA
ANTI-ICE PROTECTION	
ANTI-ICE pb-sw	AS RQRD
The flight crew should set to ON the ENG ANTI-ICE pb-sw, when icing condition	ns are expected with a TAT
at, or below 10℃.	
SLIDING TABLE	
SLIDING TABLE	RELEASE, AS NECESSARY
AFTER TAKEOFF/CLIMB CHECKLIST DOWN TO THE LINE	
AFTER TAKEOFF/CLIMB C/L down to the line	COMPLETE



# CLIMB



# **CLIMB INITIAL CLIMB** CRZ FL...... SET AS RORD If the ATC clears the aircraft to the scheduled CRZ FL or above, it is not necessary to modify the CRZ FL that was inserted on the ACTIVE/INIT page during the cockpit preparation. The flight guidance automatically takes into account any altitude that the flight crew selects on the AFS CP above the CRZ FL. If the ATC limits the CRZ FL to a lower level than the one entered on the ACTIVE/INIT page (or than the one that appears on the PERF page) insert this lower CRZ FL on the PERF page. If the flight crew does not enter this lower CRZ FL on the PERF page, there is no transition to the CRZ phase. Therefore, the managed speed/Mach targets are not modified and A/THR SOFT mode is not available. In that case, the FMA displays ALT instead of ALT CRZ in the second column. CLIMB SPEED MODIFICATIONS...... AS RQRD If ATC requests, or operational considerations lead to change the speed: Select the new speed with the SPD/MACH knob on the AFS CP and pull The new speed target is now activated Press the SPD/MACH knob to return to the managed speed profile. The speed target becomes managed. The best speed (and rate of climb) for long-term situations is between Green Dot (GD) and Note: the ECON speed. At high altitude, an acceleration from GD to ECON speed can take a long time At high altitude and depending on the Mach (selected or computed by the FMS), the airspeed can be below GD. AFTER TAKEOFF/CLIMB CHECKLIST BELOW THE LINE AFTER TAKEOFF/CLIMB C/L below the line...... COMPLETE **ANTI-ICE PROTECTION** ANTI-ICE pb-sw...... AS RQRD The flight crew shoud set to ON the ENG ANTI-ICE pb-sw, when the aircraft encounters icing conditions, unless the SAT is below -40°C. AT 10 000 FT EFIS OPTIONS...... AS RQRD ECAM MEMO...... REVIEW On PF request, or at least with PF approval: Clear the manually tuned NAVAIDS in the TUNED FOR DISPLAY panel of the POSITION/NAVAIDS page. OPT/MAX ALT...... CHECK

Check the optimum and maximum altitude capability.



## **AT THE TRANSITION ALTITUDE**

BAROMETRIC REFERENCE...... SET STD/CROSSCHECK

At the transition altitude, the barometric setting flashes on the PFD. The flight crew should set STD on the EFIS CP and on the ISIS SFD - Crosscheck the barometric settings and the altitude indications For additional training-oriented information on the VD in relation to the barometric reference setting, Refer to FCTM/NO-140 Vertical Display. - For additional information on the associated callouts, Refer to PRO-NOR-SCO Summary for Each Flight Phase



# CRUISE



CRUISE
ALT CRZ
ALT CRZ on FMA CHECK
Check that the aircraft flies at the cruise flight level that was entered on the CRZ panel of the FMS ACTIVE/PERF page. This enables the aircraft to:
<ul> <li>Fly at the ECON cruise Mach/speed The cruise Mach/speed is targeted and cruise fuel consumption is optimized.</li> </ul>
<ul> <li>Benefit from the A/THR SOFT mode</li> </ul>
<ul> <li>Have accurate predictions.</li> </ul>
Note: If the selected AFS CP altitude is below the FMS CRZ FL, then change the CRZ to the current altitude.
<u>ECAM</u>
ECAM MEMO REVIEW
SD PAGES REVIEW
Periodically review the SD pages, and particularly monitor the ENG oil pressure.
<u>Note:</u> During the cruise, the oil quantity variation is not linear. The flight crew can notice a rapid
decrease, particularly at the beginning of the flight. This rapid decrease is due to an oil
temperature decrease that leads to a longer oil transit time in the sumps (i.e. more oil is retained in the sumps).
Periodically review the following SD pages:
<ul> <li>BLEED: For bleed parameters</li> </ul>
— ELEC: For electrical parameters and GEN load
FLIGHT PROGRESS
FLIGHT PROGRESS
STEP FLIGHT LEVEL
STEP FLIGHT LEVEL AS APPROPRIATE



# **DESCENT PREPARATION**



#### **DESCENT PREPARATION**

#### **LANDING INFORMATION**

#### **BAROMETRIC REFERENCE**

#### **ECAM**

STATUS PAGE/STATUS MORE PAGE...... CHECK

- Check the STATUS page before completing the approach briefing. Review the active DEFERRED PROCs (i.e. ALL PHASES, APPR and LDG) and LIMITATIONS, and take particular note of any degradation in landing capability, or any other aspect affecting the approach and landing
- The STATUS page also automatically appears, if not empty, when the flight crew sets the barometric reference, or when the slats are extended
- Check the ALERTS IMPACTING LDG PERF on the STATUS MORE page, in order to verify if any alert triggered during the flight has an impact on the landing performance.

## **LANDING PERFORMANCE**

In the LDG PERF application, modify the selections in accordance with the estimated arrival conditions:

- In the AIRPORT/RUNWAY part, select the applicable runway
- In the CONDITIONS part, enter the estimated landing conditions
- In the AIRCRAFT STATUS part, check the selected items, if any
- Launch the computation and check the results versus Airline policy or applicable regulations



#### **FMS**

FMS ACTIVE/F-PLN/ARRIVAL page..... INSERT/CHECK

- Insert APPR, STAR, TRANS, and APPR VIA, if applicable
- If the NO FLS FOR THIS APPR message appears on the FMS message area, the flight crew will fly the NPA without FLS function.

DES panel of the FMS ACTIVE/PERF page...... INSERT/CHECK

- Before the descent, check the MANAGED MACH/SPD
- If a speed different from the ECON Mach/speed is required, insert that Mach/speed in the ECON entry fields.

The managed speed profile of the descent has a default speed limit of 250 kt below 10 000 ft . The flight crew can delete or modify this speed limit, if necessary, on the SPD panel of the VERT REV page.

APPR panel of the FMS ACTIVE/PERD page...... INSERT/CHECK Insert and check the approach data on the APPR panel of the FMS ACTIVE/PERF page.

- Enter the destination QNH, temperature, and wind. The entered wind should be the average wind value provided by the ATC or ATIS. Do not enter the gust values (e.g. if the wind is 150/20-25, insert the lower speed 150/20. In the managed speed mode, the ground speed mini function takes into account the gusts).
- Insert the MINIMUM (i.e. BARO or RADIO), as applicable. If the BARO/RADIO MINIMUM value is not a multiple of ten, round up the MINIMUM (BARO or RADIO) to the nearest ten feet. For example, if the MINIMUM is equal to 91 ft, insert '100 ft' on the APPR panel.
- <u>Note:</u> Any change of the RWY or the type of arrival (e.g. VOR , ILS ) automatically deletes the previous MINIMUM
  - To avoid undershooting the published MDA ( MDH ) during a go-around due to the aircraft inertia during the pull-up, some Authorities may require Operators to add a specific number of feet to the published MDA ( MDH ).
- Check or modify the landing configuration.

Always select the landing configuration on the APPR panel of the PERF page:

- The flight crew can choose FLAPS 3 rather than FLAPS FULL for landing, depending on the available runway length or go-around performance, or if they expect windshear or severe turbulence during the approach
- In the case of a system failure, the ECAM can require a specific landing configuration. Select the correct landing configuration on the APPR panel.
  - Select CONF 3 on the APPR panel for landing in CONF 3
  - Select FULL for all other slats/flaps configuration.

As a general rule, the flight crew can use the managed speed if the landing configuration and the configuration selected on the APPR panel are the same (if they are not the same, the managed speed will not drop down to the approachspeed).

— In the case of an in-flight failure that affects the landing performance and increases the VAPP, use the LDG PERF application for VAPP computation. Insert the computed VAPP in the APPR panel of the FMS ACTIVE/PERF page. L2 The flight crew can modify VAPP. The ground speed mini function takes into account this new value.



GA panel of the FMS ACTIVE/PERF page...... CHECK

· · · · · · · · · · · · · · · · · · ·
Check the THR RED and ACCEL altitudes, and modify them if necessary
FMS POSITION/NAVAIDS pageCHECK
<ul> <li>Set the NAVAIDS as required, and check the VOR/ADF idents on the ND s and the ILS (GLS) ident on the PFD s</li> </ul>
<ul> <li>For an ILS (GLS) approach, check the frequency (channel) and the course of the selected ILS (GLS)</li> </ul>
<ul> <li>If a VOR/DME exists close to the airfield, select it and enter the associated ident in the BRG/DIST O</li> </ul>
entry fields of the POSITION/MONITOR page, for NAV ACCURACY monitoring during descent.
SEC pages AS RQRD
The flight crew should prepare a SEC F-PLN before the top of descent, either to an alternative runway for
destination, or to the landing runway in the case of circling. In all cases, the routing to the alternate
should be available. If there is a last-minute runway change, then the flight crew only needs to activate
the appropriate SEC F-PLN , without forgetting to set the new MINIMUM and NAVAIDS.
FMSCROSSCHECK
OANS

OANS DLINIV

- BTV locates the dry and wet lines according to the runway threshold and to ensure that the flight crew will select an achievable runway exit The dry and wet lines computation takes into account the current weight of the aircraft (i.e. not the predicted landing weight).
- ROW/ROP needs the real position of the runway end to ensure the aircraft protection against runway excursion.



#### **BTV**

The use of the BTV autobrake mode is recommended in the case of dry or wet runway conditions.

CAUTION	Do not use BTV in the case of:
	<ul> <li>Contaminated runway</li> </ul>

- Contaminated runway
- Any reverse inoperative, or in the case of any aircraft failure affecting landing performance.

ND MODE sel	ector	PLAN
ND RANGE se	lector	ZOOM
_	Γ	
NONVAL LAII		JLLLCI

Select the exit in accordance with the runway conditions, i.e. wet or dry. It is recommended to select an exit beyond the WET line, even on a dry runway, in order to:

- Anticipate the last minute change of the runway condition from dry to damp or wet
- Provide a smooth deceleration for passenger comfort.

The flight crew chooses the exit by taking into account

- The destination airport gate
- The airport ground circulation
- The runway exit configuration (i.e. high speed turn off on dry runway)
- The predictive turn around times displayed on the ND.

#### **AUTOBRAKE**

AUTO BRK/BTV....... SELECT, AS RQRD Set the AUTO BRK Mode selector to the appropriate autobrake mode, depending on the condition and the length of the runway. To use BTV, set the AUTO BRK Mode selector to BTV before removing the Airport Navigation display from the ND (i.e. before the selection of a ND range different from ZOOM). If the flight crew sets another ND range before arming the BTV autobrake mode, the BTV preparation is lost.

On contaminated runways, use autobrake mode 3. Do not use BTV.

If BTV is not available, the HI mode produces a smooth but high level of deceleration on dry runways, that can be useful in short runway conditions.

#### • If the crew selected BTV:

OANS RUNWAY LENGTH Vs CHARTS RUNWAY LENGHT...... CROSSCHECK

The flight crew crosschecks the runway length displayed on the Airport Navigation display with the LDA published in the charts, when BTV is armed. The difference between the runway length displayed on the Airport Navigation display and the LDA published in the charts must not exceed 115 ft.

#### **LANDING BRIEFING**

LANDING BRIEFING...... PERFORM

The main objective of the landing briefing is for the PF to inform the PM of the planned course of action for the approach. It is recommended to use the FMS pages as a guide for the descent and approach briefing.



## **DESCENT CLEARANCE**

DESCENT CLEARANCE OBTAIN	
CLEARED ALTITUDE ON AFS CP	
When the flight crew obtains the ATC clearance, they should set the cleared altitude(FL) on the AFS CP	
taking into account the safe altitudes. If the lowest safe altitude is above the cleared altitude, check with	
the ATC to determine if this constraint applies. If the ATC confirms that this constrain applies, set the	
safe altitude on the AFS CP until it is safe to reach the cleared altitude.	

## **ANTI-ICE PROTECTION**



# DESCENT



#### **DESCENT**

#### **DESCENT INITIATION**

DESCENT...... INITIATE

The standard method to initiate a descent is to engage the DES mode at the Top of Descent (T/D) computed by the FMS.

The FMS computes the T/D based on the following assumptions:

- The aircraft will beging the descent in DES mode with managed speed
- The FMS will guide the aircraft along the descent profile computed with all the vertical F-PLN data (i.e. ALT CSTR, MANAGED MACH/SPD, SPD CSTR, SPD LIMIT) to reach VAPP at 1 000 ft AGL.

The T/D appears on the ACTIVE/F-PLN page and on the ND.

**Note:** The ND does not display the Top of Descent when HDG (Or Track) mode is engaged.

When the aircraft reaches the T/D

ALT knob...... PUSH

• If the ATC requires an early descent

Use the DES mode.

The DES mode guides the aircraft down to a lower vertical speed, in order to converge with the required descent path. (The flight crew may use a V/S of 1 000 ft/min ).

• If the ATC delays the descent

Beyond the T/D, the T/D REACHED message appears on the PFD and MFD. This suggests that the flight crew should reduce the speed torward green dot speed with ATC permission. When cleared the descent, engage DES mode with the managed speed active.

#### **DESCENT MONITORING**

DESCENT...... MONITOR

#### **DESCENT ADJUSTMENT**

RATE of DESCENT...... ADJUST, AS RQRD

To increase the rate of descent, increase the descent speed by using the selected speed, if comfort and the ATC permit. It is more cost-effective (Time/Fuel) than using the speed brakes:

- Maintain a high speed as long as possible, provided that the ATC permits to fly above the speed limits
- If the aircraft is high and at high speed, it is more efficient to keep the high speed until ALT\* then to decelerate at level-off, instead of combining descent and deceleration.

If the aircraft goes below the desired profile, use the selected speed and the V/S mode to adjust the rate of descent.

SPEED BRAKES...... AS RQRD

In the OP DES mode, the flight crew can use the speed brakes to increase the rate of descent. The flight crew can also use the speed brakes to maintain the required rate of descent, when engine anti-ice is used. In the DES mode, if the aircraft is on, or below the flight path and the ATC requires a higher rate of descent, do not use the speed brakes because the rate of descent is imposed by the planned flight path. Therefore, the A/THR may increase thrust to compensate for the increase in drag. In this case, use the OP DES mode with speed brakes.



AT 20 000	<u>FT</u>	
Cabin C	Crew	ADVICE
<u>Note:</u>	The Captain Use Seat Belt Selector 3 chimes then Auto or On and make a PA Prepare For Landing"	"Cabin Crew,
TAWS AND		
	b on EFIS CP	
WX pb o	on EFIS CP	AS RQRD
AT 10 000	<u>FT</u>	
LAND sv	W	ON
	tht crew may set to ON the landing lights, depending on the airline policy, or follo mendations.	wing regulatory
SEAT BE	ELTS sw	ON
CSTR pb	b on EFIS CP	ON
•	he LS pb, if one of the following approaches is planned:	
– ILS/		
•	G/S out, LOC only, or LOC B/C	
	PA with the FLS function	
The PFD ARRIVA	D displays the landing system data, in accordance with the approaches selected on the selected of the landing system data, in accordance with the approaches selected of the land of the l	
the FMS	S ARRIVAL page, the FMA displays the FLS capability.	
• For	r NPA flown with the FLS function:	
FLS CAPAB	BILITY	CHECK
The FLS cap	pability does not appear, if one of the following conditions occured:	
_	The NO FLS FOR THIS APPR has appeared on the FMS message area	
_	The required conditions to display the FLS capability are not met.	
FLS DA	NTA	CHECK
NAVAID	OS	AS RQRD/CHECK
HOLDING		
	NG PATTERN	AS RORD
	tht crew can insert an holding pattern in the FMS if required during the descent.	
	H CHECKLIST	
APPRO <i>A</i>	ACH C/L	COMPLETE



# PRECISION APPROACH



#### **INITIAL APPROACH**

#### **INITIAL APPROACH**

APPROACH PHASE...... CHECK/ACTIVATE

- If the aircraft overflies the DECEL pseudo waypoint in NAV mode, the APPR phase activates automatically
- If the aircraft is in HDG or TRACK mode approximately 15 nm from touchdown, activate and confirm the APPR phase on the FMS ACTIVE/PERF page.

POSITIONING...... CHECK/ACTIVATE

- In NAV mode, use the VERT DEV on the PFD and on the DES panel on the ACTIVE/PERF page
- In HDG or TRACK mode, use the energy circle on the ND that shows the required distance to land.

MANAGED SPEED...... CHECK Check that the managed speed is active and monitor the target speed.

<u>Note</u>: The aircraft decelerates automatically at the DECEL pseudo waypoing when managed speed is active and NAV, LOC\* or LOC mode is engaged.

During the approach, the autothrust maintains the maneuvering speed of the current configuration (O, S, F, VAPP).

If ATC requires a particular speed, use selected speed. Adjust the aircraft configuration accordingly. When the ATC speed constraint (e.g. « maintain 170 kt to the outer marker ») no longer applies, return to managed speed.

If ATC orders successive step descents down to the final approach flight path, use V/S or FPA mode and monitor the VERT DEV.

SPEED BRAKES...... AS RQRD

#### **NAVIGATION ACCURACY**

#### If GPS PRIMARY LOST:

NAVIGATION ACCURACY...... MONTIOR

On the POSITION/MONITOR page, check that the required navigation accuracy is appropriate to the phase of flight.

Monitor the navigation accuracy and be prepared to change the approach strategy. If the NAV ACCUR DOWNGRADED message appears on the FMS message area and on the ND, use raw data to check the navigation accuracy.

The navigation accuravy determines the autopilot modes that the flight crew should use for the approach, the type of ND displays, and the use of TAWS.



## INTERMEDIATE/FINAL APPROACH

#### **INTERMEDIATE/FINAL APPROACH**

#### General

The preferred technique for flying an ILS(GLS) approach is to fly a decelerated approach using the AP / FD s, the LOC and G/S modes, the autothrust is SPEED mode, and the managed speed target.

The objective is to stabilize the aircraft on the final descent path at VAPP in the landing configuration, at 1 000 ft above the airfield elevation in instrument conditions, or at 500 ft above airfield elevation in visual conditions, after continuous deceleration on the glide scope.

To be stabilized, all of the following conditions must be achieved prior to, or upon reaching this stabilization height:

- The aircraft is on the correct lateral and vertical flight plan,
- The aircraft is in the desired landing configuration,
- The thrust is stabilized, usually above idle, to maintain the target approach speed along the desired final approach path,
- There is no excessive flight parameter deviation.

If the aircraft is not stabilized on the approach path in landing configuration at 1 000 ft above airfield elevation in instrument conditions, or at 500 ft above airfield elevation in visual conditions, or as restricted by Operator policy/regulations, a go-around must be initiated unless the crew estimates that only small corrections are necessary to rectify minor deviations from stabilized conditions due, amongst others, to external perturbations.

#### Decelerated Approach

The decelerated approach technique descends the aircraft to 1 000 ft, at VAPP. In most cases, the interception of the final descent path is achieved with CONF 1 at S speed.

The advantages are:

- Decrease in fuel consumption
- Decrease in the noise level
- Time saving
- Flexibility and ability to vary speed, to be in accordance with ATC requests.

Note: For ILS approach, the ICAO defines the envelope in which the quality of the G/S signal ensures a normal capture. This envelope is within 10 nm, +- 8 degress from the centerline of the ILS glide path, and up to 1.75 teta (teta, being the nominal glide path angle).

If the approach is armed when the aircraft is far outside from the standard glideslope (G/S) capture envelope, spurious G/S \* engagement may occur, due to an incorrect G/S deviation signal. This spurious G/S capture will order a pitch up, if the aircraft is below the glide beam, and a pitch down, if the aircraft is above the glide beam.



Each time that the flight crew notices pitch movement, or a spurious G/S \*, or a trajectory deviation, they must immediately disconnect the AP, if engaged, in order to re-establish a normal attitude and disengage APPR mode. It is then recommended to arm/rearm the APP (ILS) mode within the normal capture zone.

#### APPR MODE ACTIVATION

•	When the ATC clears the aircraft for the approach :	
	APPR ph on AFS CP	PRFSS

 The flight crew should press the APPR pb only after the ATC clears the aircraft for the approach.

When pressed, the LOC and G/S modes arm provided that the flight crew:

- Selected ILS(GLS) approach on the FMS ARRIVAL page, or
- Manually-tuned an ILS(GLS) in the POSITION/NAVAIDS page and did not select an approach or only selected a runway on the FMS ARRIVAL page, or
- Set the RAD NAV key to STBY and selected the LS on RMP 1 or 2.
- LOC and/or G/S modes engaged not sooner than 3 s after being armed.

Note: If the flight crew selects a non-precision approach in the active flight plan, and if they manually tune an ILS(GLS) on the POSITION/NAVAIDS page, the MFD and PFD display CHECK APPR SEL. This message is a reminder to the flight crew that, although an ILS(GLS) is tuned in the POSITION/NAVAIDS page, the available approach guidance modes are F-LOC and F-G/S modes (for NPA flown with the FLS function) when the APPR pb is pressed on the AFS CP.

#### • If GPS PRIMARY LOST:

The pre-capture zone of the LOC beam is not available. Therefore, the LOC capture performance may be degraded, and overshoot of the LOC beam may occur.

The following graph illustrates the angle of interception vs. the distance, that ensures a capture with a single overshoot. This overshoot is less than 1.3 dots.



#### LOC Capture Domain

The LOC capture point is the point at which the aircraft track projection intercepts the LOC centerline.

AP 1+2	ON
When the APPR mode is selected, the flight crew should engage both AP s.	
FMA C	HECK
Check that the FMA displays the approach capability (CAT2, CAT3 SINGLE, CAT3 DUAL, or AUTO LAND) the intended ILS(GLS) approach.	) for

#### **APPROACHING GREEN DOT SPEED**

FLAPS 1	ORDER
FLAPS 1	SET

- The flight crew should not set FLAPS 1 later than 3 nm from the Final Approach Fix (FAF)
- Check the aircraft deceleration toward S speed
- The aircraft must reach, or be established on, the glideslope with FLAPS 1 and S speed at or above
   2 500 ft AGL
- If the aircraft speed is significantly above S on the glideslope, or if the aircraft does not decelerate
  on the glideslope, extend the landing gear in order to decelerate the aircraft.
   The flight crew can also use the speed brakes. However, the flight crwe should be more aware of
  the VLS increse, and the limited effect of the speed brakes at low speeds.

LOC CAPTURE....... MONITOR

The flight crw must always monitor the capture of a LOC beam. During the capture phase, when the LOC deviation is within the LOC scale, the LOC deviation on the PFD and the ND should move towards the center of the scale.

If GPS PRIMARY is lost, the LOC capture performance may be degraded, and overshoot of the LOC beam may occur.

<u>Note:</u> For ILS capture, the ICAO requires the LOC beam to ensure a normal capture within 10 nm, and +-35 degrees from the centerline of the course. Some current ILS systems just meet the requirement and are the subject to an erroneous capture outside these limits.



G/S CAPTURE	MONITOR
Note: The glideslope may be capture independently of the LOC beam cap	ture, depending on the PRIM
pin program.	
If above the glideslope :	
V/S MODE	AS RORD
Do not exceed 2000 ft/min	
AFS CP ALTITUDE	SET ABOVE A/C ALTITUDE
Select an altitude above the aircraft altitude, in order to prevent	•
<ul><li>When G/S Capture ( G/S* ) :</li></ul>	
GO-AROUND ALTITUDE	SET
Note:  - This procedure prevents from an undesired level off - If the aircraft intercepts the ILS(GLS) above the valid  (i.e. when the radio altitude indication is not yet a appears on the FMA. Check that the FMA displays intended approach when the airraft is below 5 00	ity range of the radio altimeter vailable on the PFD ), CAT 1 the correct capability for the
BELOW VFE NEXT, AT 2500 FT AGL MINIMUM	
FLAPS 2	
FLAPS 2	SET
<ul> <li>Check the aircraft decelerated toward F speed</li> </ul>	
<ul> <li>If the aircraft intercepts the ILS(GLS) glideslope below 2 500 ft AGL below the glideslope</li> </ul>	, select FLAPS 2 at one dot
<ul> <li>If the aircraft speed is significantly above F on the glideslope, or if on the glideslope, extend the landing gear in order to decelerate t brakes is not recommended.</li> </ul>	
AFTER FLAPS 2 SELECTION, AT 2000 FT AGL MINIMUM	
L/G DOWN	ORDER
L/G	DOWN
AUTO BRK	CONFIRM
If the runway conditions deteriorated since the approach briefing, consider performance assessment and if there is a need for another braking model.	•
When BTV is selected, if the aircraft lands on the runway different from t	
settings, the autobrake reverts automatically toautobrake mode Hi in sh	•
GND SPLRS	ARM
EXTERIOR LIGHTS	SET



#### AFTER LANDING GEAR DOWN DOWN SELECTION

Below VFE NEXT :	
FLAPS 3	ORDER
FLAPS 3	SET
WHEEL SD PAGE	CHECK
<ul> <li>The WHEEL SD page appears below 800 ft, or at landing gear extension</li> </ul>	
<ul> <li>Check the five landing gear green indications. At least one green triangle gear strut is sufficient to indicate that this landing gear strut is down and</li> </ul>	_
Below VFE NEXT :	
FLAPS FULL	ORDER
FLAPS FULL	SET
Check that the aircraft decelerates to VAPP.	
A/THR CHECK IN SPEED	) MODE or OFF
For GLS CAT I with autoland, if the A/THR is available, activate the A/THR	
SLIDING TABLE	STOWED
LDG MEMO	IO BLUE LINE
CABIN	
Check CABIN READY on the EWD, or obtain the report from the chief purser : « Cabin	
LANDING C/L	
FLIGHT PARAMETERS	CHECK
The PF announces any modification to the FMA	
The PM announces any flight parameter deviation.	
At 500 ft AGL (or RA) and below and if the flight crew selected BTV autobra	ke mode :
The PM may take advantage to adjust the ND RANGE selector to display the	dry and wet lines
on the Airport Navigation display, when the landing is positively considered.	
• At 350 ft AGL (or RA) :	
LAND ON FMA	ANNOUNCE
ILS(GLS) COURSE	CHECK
Check the ILS(GLS) course on the PFD.	



#### AT MINIMUM+100 FT

ONE HUNDRED ABOVE...... MONITOR or ANNOUNCE

#### **AT MINIMUM**

MINIMUM...... MONITOR or ANNOUNCE

LANDING or GO-AROUND...... ANNOUNCE

Maintain the aircraft on a stabilized flight path up to the flare.

At 50 ft, one dot below the glide slope means that the aircraft is 7 ft below the glide slope.

Do not duck under the glide slope.

ALL ILS APPROACHES (CAT I) Procedure and Call out

#### **AT MINIMUM**

The following approach pattern is based on the assumption that managed speed is used : (inline image not included)

If in selecte speed, select:

- S speed after FLAPS 1 selection
- F speed after FLAPS 2 selection
- VAPP after landing FLAPS selection.

Note: If earlier stabilization at VAPP is required, start the deeleration at a higher altitude.



## **NON-PRECISION APPROACH**



INITIAL APPROACH	
INTERMEDIATE/FINAL APPROACH	
INTERMEDIATE/TINALATTROACH	



## VISUAL APPROACH



#### **INITIAL APPROACH**

#### **INITIAL/INTERMEDIATE APPROACH**

The flight plan selected on the FMS should include the selection of the landing runway. The downwind leg may also be part of the flight plan. This may produce a useful indication of the aircraft position in the circuit, on the ND. However, the flight crew must use the appropriate visual references.

#### AT THE BEGINNING OF THE DOWNWIND LEG

APPROACH PHASE	ACTIVATE
On the APPR panel of the ACTIVE/PERF page, use the ACTIVATE AF	PPR button to activate the
approach phase	
FD OFF	ORDER
FD	OFF
FLYING REFERENCE	TRK-FPA
A/THR ACTIVE	CHECK

#### ON THE DOWNWIND LEG

Abeam threshold, extend the downwind leg to 45 s (+/- wind correction). Turn into the base leg with a maximum of 30° of bank. Descend with the appropriate FPA.

#### • Below VFE NEXT:

FLAPS 2 OI	RDER
FLAPS 2	. SET
Check the aircraft deceleration toward F speed.	

#### **FINAL APPROACH**

- The speed trend arrow and FPV help the flight crew to make timely and correct thrust settings (if in manual thrust) and approach path corrections. Avoid descent through the correct approach path with idle thrust. (Late recognition of this situation without a prompt thrust increase may lead to considerable speed decay and altitude loss).
- Have the aircraft stabilized by 500 ft AGL, on the correct approach path at VAPP (or ground speed mini) with the appropriate thrust applied. If the aircraft is not stabilized on the approach path in landing configuration at 500 ft above airfield elevation, or as restricted by Operator policy/regulations, a go-around must be initiated unless the crew estimates that only small corrections are necessary to rectify minor deviations from stabilized conditions due, amongst others, to external pertubations.
- Avoid any tendency to duck under in the late stages of the approach.
- Avoid destabilizing the approach in the last 100 ft to have the best chance of making a good touchdown at the desired position.



# LANDING



LANDING	
FOR MANUAL LANDING	
AP	OFF
The flight crew should disengage the AP whatever the type of appro	pach is. A/THR SPEED mode remains
engaged.	
AT AROUND 40 FT RA	
From a stabilized approach, the flate height is approximately 40 ft.	
FLARE	INITIATE
ATTITUDE	MONITOR
THRUST LEVERS	IDLE
Move the thrust levers to idle, and begin a gentle progressive flare to down without a prolonged float. If the autothrust is engaged, it the thrust levers are set to idle detent. At 20 ft, an automatic "reminder	t automatically disconnects when all 'RETARD" callout will trigger, as a
<u>Note:</u> Ground spoilers extension is inhibited if two or more the detent.	rust levers remain above the IDLE
FOR AUTOMATIC LANDING	
BETWEEN 50 FT AND 40 FT RA	
FMA	CHECK FLARE
FLARE	MONITOR
AT APPROXIMATELY 30 FT RA	
FMA	CHECK THR IDLE
The autothrust begins to decrease thrust to idle.	
AT 10 FT RA	
An automatic "RETARD" callout triggers.	
THRUST LEVERS	IDLE
The autothrust disconnects.	
LATERAL GUIDANCE	MONITOR
Monitor the lateral guidance by using external references.	
AT TOUCHDOWN	CHECK BOLL OUT
FMA"  • If AUTO ROLL OUT:	CHECK ROLL OUT
	LINTH END OF BOLL OUT
AP KEEP ENGAGED,	ONTIL LIND OF ROLL OUT
<ul> <li>DEROTATION</li> <li>As soon as the main landing gear touches down:</li> <li>DEROTATION</li> </ul>	INITIATE





#### **LANDING ROLL**

REVERSER LEVERS...... PULL Select MAX REV immediately after the main landing gear touches down If the airport regulations limit the use of the thrust reversers and if landing performance permits, select and maintain IDLE REV until the aircraft reaches the taxi speed In the case of a failure of one reverser, it is possible to use the opposite reverser If required for performance reasons, braking may begin before the nosewheel touchdown. However, if passenger comfort is the priority, the flight crew should delay braking on dry runways only, until the nosewheel touches down During rollout, avoid sidestick inputs (either lateral or longitudinal). If the flight crew encounters directional control problems, they should reduce the thrust to reverse idle until directional control is satisfactory After the flight crew selects reverse thrust, they should perform a full stop landing. GND SPLRS EXTENDED...... ANNOUNCE Check the slats/flaps display on the lower part of the PFD, to ensure that the ground spoilers are extended. If no ground spoilers are extended: Check that all thrust levers are set to IDLE detent Set both thrust reverser levers to MAX REV, and fully press the brake pedals. If ground spoilers are not armed, ground spoilers will extend at reverser thrust selection. Check that the EWD displays the expected reverser deployment (i.e. REV). DIRECTIONAL CONTROL...... MONITOR/ENSURE Do not use the nosewheel steering control handle before reaching taxi speed. If autobrake selected: AUTO BRK...... CHECK/ANNOUNCE Check and announce BTV, BRK LO, BRK 2, BRK 3, or BRK HI on the FMA. Note: If no ground spoilers are extended, the autobrake is not activated. • If no autobrake: BRAKES"..... AS RQRD The flight crew feels the deceleration. They confirm the deceleration by using speed trend on the PFD. If AUTO ROLLOUT, before 20 kt: **AT 80 KT** EIGHTY KNOTS...... ANNOUNCE REVERSER LEVERS...... IDLE **CAUTION** Avoid using high levels of reverse thrust at low speed, unless required due to an emergency or if "KEEP MAX REVERSE" sounds. The distortion of the airflow caused by

gases that re-enter the compressor can cause engine stalls that may result in excessive

EGT.



#### FOR CAT II & CAT III OPERATIONS WITH BTV

• If the aircraft ground speed is more than 10 kt when the aircraft enters the last 1 000 ft of the runway (i.e. the centerline runway lights are continuously red):

The runway center line is color coded. Continuous red lights mark the last 1 000 ft of a runway designed for CATII and CATIII operations. In low visibility condition, the flight crew must select an exit 1 000 ft before the runway end. In normal BTV operation, BTV deactivates when the aircraft reaches 10 kt ground speed. If the aircraft ground speed is more than 10 kt when the aircraft enters the last 1 000 ft of the runway, the flight crew must override BTV and apply manual braking as required.

BTV AUTOBRAKE MODE...... OVERRIDE
MANUAL BRAKING....... APPLY AS RQRD

#### **AT TAXI SPEED**

REVERSER LEVERS...... STOW

When the aircraft reaches the taxi speed, and before it leaves the runway, stow the reversers.

CAUTION	Except in an emergency, do not use reverse thrust to control the aircraft speed while on		
	taxiways.		

On taxiways, the use of reversers, even restricted to idle thrust, may have the following effects:

- The engines may ingest fine sand and debris that may be detrimental to the engines and airframe systems.
- On snow-covered areas, snow will recirculate into the air intel, and may cause an engine flameout or rollback.

AUTO BRK...... DISARM

- The autobrake may be disarmed at pilot's discretion.
- The flight crew should use one of the A/THR Instinctive Disconnect pb to disarm the autobrake.
- If BTV is active and the flight crew did not disarm manually the autobrake, the autobrake disarms automatically at 10 kt.



# GO AROUND



### **GO-AROUND GO-AROUND INITIATION** If TOGA thrust is not required, set the thrust levers to the TOGA detent then, retard the thrust levers as required. This enables to engage the GO-AROUND phase with the associated AP / FD modes. If the thrust levers are not briefly set to the TOGA detent, the FMS does not engage the GO-Notes: AROUND phase, and when the aircraft flies over, or close to the airport (less than 7 nm) the FMS will sequence the destination waypoint in the F-PLN. The GO-AROUND phase engages. The previously-flown approach is automatically strung back into the flight plan at the end of the missed approach procedure. Initiate the rotation toward a pitch attitude of 12.5 $^{\circ}$ ( 10 $^{\circ}$ , if one engine is failed) to obtain a positive rate of climb, then follow the SRS pitch command bar. GO-AROUND...... ANNOUNCE FLAPS...... RETRACT ONE STEP FMA...... CHECK/ANNOUNCE If the flight crew decides not to fly the flight plan after the go-around, they can use the HDG/TRK preset function. POSITIVE CLIMB...... ANNOUNCE If the pitch attitude goes above 20°, or below 10° up: The PM announces: "PITCH" If there is no climb rate: The PM announces: "SINK RATE" L/G UP......ORDER L/G.......UP NAV or HDG...... AS RQRD AT GO-AROUND THRUST REDUCTION ALTITUDE (LVR CLB FLASHING ON FMA) THRUST LEVERS...... CL AT GO-AROUND ACCELERATION ALTITUDE If the speed target does not increase to initial CLB speed: AFS CP ALTITUDE...... CHECK ALT pb...... PRESS FLAPS...... ORDER RETRACION ON SCHEDULE FLAPS...... RETRACT ON SCHEDULE EXTERIOR LIGHTS...... SET AFTER TAKEOFF/CLIMB C/L down to the line...... COMPLETE If necessary, at the transition altitude: BAROMETRIC REFERENCE...... SET STD/CROSSCHECK AFTER TAKEOFF/CLIMB C/L below the line...... COMPLETE



_	To pre	pare for a second approach:	
	APPRO	DACH PHASE	ACTIVATE
	Activa	te the APPR phase on the FMS ACTIVE/PERF page. If the APPR phase is not activa	te:
	0	Manage approach speed is not available	
	0	BARO/RADIO indications do not appear on the PFD.	
•	To div	ert to the alternate:	
FMS		UP	DATE
_		light crew prepared an alternate flight plan, they should use ENABLE ALTN in the revisior opoint on the ACTIVE/F-PLN page	n menu of the
	If the fl ACTIVE	light crew prepared a SEC/F-PLN to the diversion airfield, they should swap the SEC F-PLI F-PLN.	V to the
_	If the f	flight crew did not prepare any alternate, they should:	
	0	Initiate a selected climb	
	0	Perform a lateral revision to insert a new destination	
	0	When cleared to a waypoint:	
	DIREC	T TO PERF	ORM
The FM	IS auto	matically reverts to the CLB phase. The FMS automatically sets the CRZ FL at	the
-		ate CRZ FL (FL 220, or FL 310), and maintains the previous cost index. The ese targets if necessary	flight crew
FMS			HECK



## AFTER LANDING



AFTER LANDING	
GROUND SPOILERS	
GND SPLRS	DISARM
<u>FLAPS</u>	
FLAPS	RETRACT
<u>APU</u>	
APU MASTER SW pb-sw	ON
APU START may be delayed until just prior to engine shutdown.	
• If the APU flap is fully open (i.e. FLAP OPEN appears on the APU SD page):	_
APU START pb	
On the APU SD page, check that FLAP OPEN appears before setting to ON the APU S	TART pb.
ENG START	
ENG START selector	CHECK NORM
ANTI-ICE	
ANTI ICE	AS RQRE
If engine anti-ice is used, carefully control taxi speed, particularly on wet or slipped ground idle is increased. L1 When taxiing in freezing fog conditions, if temperature °F), ice may accumulate on the engines fans. In order to shed the ice, the flight constant shedding procedure at least every 30 min of total taxi time before takeoff. The tocumulative time of the taxi-in time of the previous flight and taxi-out time of the because it is necessary to determine the remaining taxi-out time that is allowed flight crew must record in the logbook the taxi-in time in freezing fog conditions.	re is less than + 3°C ( 3 rew must apply the ice stal taxi time is the next flight. Therefore,
EXTERIOR LIGHTS	
LAND sw	OFF
Turn off the LAND lights, if they are not necessary	
STROBE sw	AUTC
When leaving the runway, set the STROBE sw to AUTO.	
OTHER EXTERIOR LIGHTS	
Set the NAV sw to ON, as required, to turn on the navigation and obstruction ligh	
NOSE sw	TAX
Set NOSE sw to ON, when the aircraft leaves the runway.	
RWY TURN OFF & CAMERA sw	AS RQRE
Turn on the RWY TURN OFF & CAMERA sw at night for ETACS purpose.	
AIRPORT NAVIGATION	
ND RANGE selector Z	OOM, AS APPROPRIATI



#### **EFIS CONTROL PANEL (EFIS CP)**

TAXI pb	AS RQRD
When the TAXI pb is set to ON, the ETACS appears on the PFD.	
WX pb	CHECK OFF
The weather radar automatically switches off 60 s after landing	

#### **BRAKE TEMPERATURE**

BRAKE TEMPERATURE...... MONITOR

- Check the brake temperature for discrepancies and high temperature on the WHEEL SD page
- Maintenance action is due in the following cases:
  - The temperature difference between two brakes on a same gear is above 150°C, and the temperature of either brake is above or equal to 600°C, or
  - The temperature difference between two brakes on a same gear is above 150°C, and the temperature of either brake is below or equal to 60°C, or
  - The difference between the average temperature on the left and right brakes is above or equal to 200°C, or
  - o A fuse plug has melt, or
  - o THe brake temperature exceeds 800 °C.

#### **AFTER LANDING CHECKLIST**

AFTER LANDING C/L..... COMPLETE



## **PARKING**



PARKING
ANTI-ICE
ANTI-ICE OFF
APU BLEED
APU BLEED pbON
Set the APU BLEED pb to ON immediately before the engine shutdown to prevent engine exhaust fumes
from entering the air conditioning.
PARK BRK
Park BRK ON
<ul> <li>If the temperature of one brake is above 500 °C, avoid applying the parking brake, unless operationally necessary.</li> </ul>
<ul> <li>On the triple pressure indicator, check the left and right brake pressures.</li> </ul>
ENG MASTERS 1, 2, 3, and 4
ENG MASTERS 1, 2, 3, and 4
<ul> <li>After high thrust operations, such as the use of maximum reverse thrust at landing, operate the</li> </ul>
engine at idle for 3 min before shutting down the engine. This 3 min period thermally stabilize the
hot section of the engine. This 3 min period can include operational time at idle, such as taxiing.
Depending on the circumstances (e.g. urgent need to open a cargo door or to connect to staires),
the flight crew may decide to shut down the engine regardless of the time at idle.
— Check that engine parameters decrease
<ul> <li>The DOOR/OXY SD page appears</li> </ul>
<ul> <li>If the APU is not available, connect external power before shutting down the engines.</li> </ul>
<ul> <li>After the last engine shutdown, an automatic test of the fuel crossfeed valves, the LP valves, and the</li> </ul>
heat exchanger valves, begins. If a failure is detected, the associated ECAM alert triggers, and
should be entered in the logbook. This test lasts approximately 2 min.
ELAPSED TIME
ELAPSED TIME (If applicable)
SEAT BELTS
SEAT BELTS sw OFF
<u>SLIDES</u>
SLIDES DISARMED
Check that the slides are disarmed on the DOOR/OXY SD page. If any slide is not disarmed, warn the
cabin crew.



EXTERIOR LIGHTS	
BEACON sw	OFF
Turn off the BEACON lights when all engines spooled down.	
OTHER EXTERIOR LIGHTS A	S RQRD
GROUND CONTACT	
GROUND CONTACT ESTAE	BLISHED
<ul> <li>The flight crew should establish communication with ground crew</li> </ul>	
<ul> <li>Check that chocks are in place.</li> </ul>	
FUEL PUMPS	
FUEL PUMPS	OFF
HUD	
 HUD	STOW
FUEL QUANTITY	
FUEL QUANTITY	CHECK
Check that the sum of fuel on board and the fuel used is consistent with the fuel on board at deg	
If the flight crew detects a discrepancy that is not usual, maintenance action is due.	
PARKING CHECKLIST	
PARKING C/LCON	MPLETE
	MPLETE
PARKING C/LCOM	
PARKING C/LCOM	AS RQRD
PARKING C/L	AS RQRD Ike when
PARKING C/L	AS RQRD Ike when
PARKING C/L	AS RQRD ike when leave
PARKING C/L	AS RQRD ike when leave for an
PARKING C/L	AS RQRD ike when leave for an
PARK BRK  PARK BRK	AS RQRD ike when leave for an
PARK BRK  PARK BRK	AS RQRD ike when leave for an onally
PARK BRK  PARK BRK	AS RQRD  Ike when  Ieave  for an  onally
PARK BRK  PARK BRK  PARK BRK  PARK brace  If the ECAM displays the BRAKES BRAKE HOT alert, the flight crew should release the parking brace the wheel chocks are in position.  Notes: When winds including guts exceed 30 knots or, if slope of parking ramp is excessive, in parking brakes set.  Releasing the parking brake prevents critical structure from being exposed to high temperature of extended period of time. However, the flight crew may keep the parking brake applied if operation required (e.g. slippery tarmac).  ONBOARD INFORMATION SYSTEM (OIS) CLOSURE  OIT SLIDE sw	AS RQRD  Ike when  Ieave  for an  onally  FLT OPS  . CLOSE
PARK BRK  PARK BRK	AS RQRD  Ike when  Ieave  for an  onally  FLT OPS  . CLOSE



### **LOGBOOK**

OIT SIDE sw	<b>AVNCS</b>
Set the OIT SIDE sw to NSS AVNCS in order to access the logbook.	
FLIGHT CLOSURE PEI	RFORM
Before the electronic flight closure, check that VHF 3 DATA mode is active.	



## SECURING THE AIRCRAFT



SECURING THE AIRCRAFT
PARK BRK
PARK BRKON
Keep the parking brake on in order to reduce the rate of the hydraulic leak in the LEHGS accumulators.
OXYGEN CREW SUPPLY
OXYGEN CREW SUPPLY pb-sw OFF
ADIRS (1+2+3)
ADIRS (1+2+3)
The flight crew should not turn off the ADIRS during transits at latitudes above 70 °N, in order to avoid
excessive alignment time. After turning off the ADIRS, wait at least 10 s before turning off the electrical
supply, in order to ensure that the ADIRS will memorize the latest data.
EXTERIOR LIGHTS
EXTERIOR LIGHTS OFF
GROUND SERVICING  AS DODD
GND SERVICING pb-sw AS RQRD  If the ground crew or the servicing personnel requires electrical power, consider setting to ON the GND
SERVICING pb-sw (in the forward cabin, near the M1 door), before powering off the aircraft.
APU BLEED
APU BLEEED pb-sw OFF
EXT POWER
EXT pb-sw AS RQRD
At least two EXT power units are necessary to supply the entire electrical network.
If the electrical charge of at least one APU generator is above 50 %:
<ul> <li>At least three EXT power units are necessary to supply the aircraft with the APU OFF</li> </ul>
<ul> <li>If only two EXT power units are available, it is recommended to keep the APU ON or to reduce the</li> </ul>
load demand for a while (e.g. by setting to OFF the CAB FANS pb-sw and COOLG pb-sw ).
Note: These temporary actions ensure that enough electrical power is available during a high
electrical demand (e.g. to operate the cargo doors). When the flight crew sets to OFF the CAB
FANS pb-sw and the COOLG pb-sw, no degradation of the galley cooling system and of the
air conditioning is expected for a few minutes, i.e. the time necessary to operate the doors. As
soon as it is possible, the flight crew should set to ON the CAB FANS pb-sw and the COOLG pb-sw.
APU
APU MASTR SW pb-sw OFF
Turn off the APU after all passengers disembarked.



EMER LIGHTS AND SIGNS
EMER EXIT LT sw OFF
The flight crew must set to OFF the EMER EXIT LT sw to avoid battery discharge, when:
<ul> <li>Only batteries supply the aircraft, or</li> </ul>
<ul> <li>No electrical power supplies the aircraft (i.e. external power, APU and all batteries are set to OFF).</li> </ul>
When EMER EXIT LT sw is set to ARM or ON and the aircraft is not supplied, the Emergency lighting in cabin operates. In this case the HOT BUS via the BAT 1, the ESS BAT and the internal EPSU batteries supply the emergency lighting
NO SMOKING sw OFF
The flight crew must set to OFF the NO SMOKING sw to avoid battery discharge, when only the batteries supply the aircraft.
ONBOARD INFORMATION SYSTEM (OIS)
ALL LAPTOPS OFF
Switch off all laptops only if the aircraft is left unattended. To turn off the laptops, click on the SWITCH
OFF LAPTOP button on the LOGIN page of the FLT OPS Domain.
Note: The Network Server System (NSS) automatically shuts down at aircraft power down.
OIT knob OFF
The flight crew should switch off both OITs.
SECURING THE AIRCRAFT CHECKLIST
SECURING THE AIRCRAFT C/L
BAT 1, ESS, 2, AND APU BAT
ALL BAT pb-sw (BAT 1, ESS, 2, APU BAT) OFF
Wait until the APU flap is fully closed (i.e. around 2 min after the APU is set to OFF) before turning off the
APU battery. This ensures that the APU shutdown sequence is entirely completed.
COCKPIT WAY LIGHT
COCKPIT WAYLIGHT pb-sw ON, IF NECESSARY
If the aircraft is not electrically supplied, the flight crew can use the cockpit way light in order to leave
the aircraft. The cockpit way light goes off automatically after 60 s