M6 - Dynamic 4-Bar Mechanism

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${\bf Contents}$

1	Introduction	2
2	Exercises	2
3	Dynamics 3.1 Theory 3.2 Constraints	3 3
4	Numerical Integration	4
5	Plots	6
6	6.4 Parameters	17 18 18 19 19



1 Introduction

A 4 bar mechanism is shown on figure 1 below.

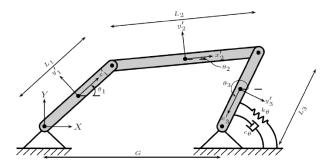


Figure 1: A 4 bar mechanism with a rotational spring damper element

The mechanis consists of an interconnected chain of 3 bars that are ttached to ground at the 2 'free' ends. The model is tied together by 4 ervolute join constraints. The mass of the left most lnk is negligible, which means it can be replaced by an absolute distance constraint. A rotational spring-damper element is placed between ground and the right most body. The aim of the following set of exercises is to make a parametrix model of the mechanism in Matlab, which means that the model should be able to analyze mechanism with arbitraty bar lengths, and spring-damper properties, k_{θ} and c_{theta} respectively.

2 Exercises

- a. Calculate the constraint jacobian Φ_q
- b. Set up the acceleration equation γ
- c. Set up a valid initial configuration and displace it to a configuration that gives substantial forces in the spring. Assume zero initial velocities for all links and run a simulation forward in time until the mechanism reaches rest.
- d. plot position, velocities and acceleration vs time for all generalized coordinates
- e. plot the spring and damper forces vs time
- f. Plot the reaction forces in the revolute join between link 2 and link 3 $\,$



3 Dynamics

3.1 Theory

Looking at the dynamics of the pendulum it is wished to solve for Newton's equation of motion for a differential mass which is described as

$$\ddot{r}^P \cdot dm(P) = F(P) \cdot dm(P) + \left[\int_m f(P,R) dm(R) \right] dm(P) \tag{3.1}$$

The equation is difficult to use as it explicitly involves internal forces which are unknown. The equation is also applied for every differential element of the mass which create too many equation of motion. Therefore to simplify the equation is to use the variations or virtual work approach. By using both theories the variational form of the equations of planar motion can be written as

$$\delta q^T [M\ddot{q} - Q] = 0 \tag{3.2}$$

where

$$M = diag(m, m, J') = \begin{bmatrix} m_1 & 0 & 0 & 0 & 0 & 0 \\ 0 & m_1 & 0 & 0 & 0 & 0 \\ 0 & 0 & J'_1 & 0 & 0 & 0 \\ 0 & 0 & 0 & m_2 & 0 & 0 \\ 0 & 0 & 0 & 0 & m_2 & 0 \\ 0 & 0 & 0 & 0 & 0 & J'_2 \end{bmatrix}$$

$$(3.3)$$

where J' is the polar mass moment of inertia. Q is only the applied force as the internal force is not known and removed by saying that the condition holds for any virtual displacement that is consistent with the set of constraint in the system. The condition is true if

$$\Phi_q \delta q = 0 \tag{3.4}$$

The last issue is to change the virtual work which can be done by applying Lagrange Multiplier Theorem which shows a unique linear correlation between A and b if $x^Tb = 0$ as soon as Ax = 0. b is a linear combination of the rows of a and can be written as $b = -A^T\lambda$ The equation of motion can now be written as

$$M\ddot{q} + \Phi_q^T \lambda = Q_A \tag{3.5}$$

By combining it with the one of the kinematic constraints

$$\begin{split} &\Phi(q,t)=0\\ &\Phi_{q}\cdot\dot{q}=\nu\\ &\Phi\ddot{q}=\gamma \end{split} \tag{3.6}$$

a linear system can be made and solved by using a numerical integrator such as Runge-Kutta

$$\begin{bmatrix} M & \Phi_q^T \\ \Phi_q & 0 \end{bmatrix} \begin{bmatrix} \ddot{q} \\ \lambda \end{bmatrix} = \begin{bmatrix} Q^A \\ \gamma \end{bmatrix} \tag{3.7}$$

3.2 Constraints

The 4 bar mechanism seen on figure 1, is one body having 6 generalized cartesian coordinates. By applying absolute constraints on the x- and y-axis to the ground, the system have 1 DOF

By applying absolute constraint on body 3 in the x- and y-axis to ground with constant distance two DOF are constrained Applying an revolute joint between body 2 and 3 constrain 2 DOF. Furthermore to have 1 DOF, a absolute constraint between the origin of the global coordinate and body 2 is made.

$$DOF = nb - nh = 6 - 5 = 1 (3.8)$$

The constraint is defined as

$$\Phi_{k} = \begin{bmatrix}
\left(x_{2} - \frac{L_{2}\cos(\theta_{2})}{2}\right)^{2} - L_{1}^{2} + \left(y_{2} - \frac{L_{2}\sin(\theta_{2})}{2}\right)^{2} \\
x_{2} - x_{3} + \frac{L_{2}\cos(\theta_{2})}{2} + \frac{L_{3}\cos(\theta_{3})}{2} \\
y_{2} - y_{3} + \frac{L_{2}\sin(\theta_{2})}{2} + \frac{L_{3}\sin(\theta_{3})}{2} \\
x_{3} - G + \frac{L_{3}\cos(\theta_{3})}{2} \\
y_{3} + \frac{L_{3}\sin(\theta_{3})}{2}
\end{bmatrix}$$
(3.9)



In this case Φ_k is the same as the vector in the absolute frame pointing to CG of the body in the local frame. To find the Jacobian matrix, the following equation and the generalized coordinates can be used

$$\Phi_{q} = \frac{\partial \Phi_{i}(\mathbf{q})}{\partial q_{j}} = \begin{bmatrix}
2x_{2} - L_{2}\cos(\theta_{2}) & 2y_{2} - L_{2}\sin(\theta_{2}) & L_{2}\sin(\theta_{2}) & \left(x_{2} - \frac{L_{2}\cos(\theta_{2})}{2}\right) - L_{2}\cos(\theta_{2}) & \left(y_{2} - \frac{L_{2}\sin(\theta_{2})}{2}\right) & 0 & 0 & 0 \\
1 & 0 & \frac{L_{2}\cos(\theta_{2})}{2} & -1 & 0 & \frac{L_{3}\sin(\theta_{3})}{2} & 0 & -1 & 0 \\
0 & 1 & \frac{L_{2}\cos(\theta_{2})}{2} & 0 & -1 & \frac{L_{3}\cos(\theta_{3})}{2} & 0 & 0 \\
0 & 0 & 0 & 0 & 1 & 0 & -\frac{L_{3}\sin(\theta_{3})}{2} & 0 & 0 & 0 \\
0 & 0 & 0 & 0 & 0 & 1 & \frac{L_{3}\cos(\theta_{3})}{2} & 0 & 0 & 0 & 0
\end{pmatrix}$$
(3.10)

As there is no driving constraint, ν is a zero matrix. γ can be found by applying the chain rule on ν the equation will be derived as

$$\gamma = \Phi_{q} \cdot \ddot{q} = -\left(\Phi_{q} \cdot \dot{q}\right)_{q} \cdot \dot{q} + \Phi_{qt} \cdot \dot{q} + \Phi_{tq} \cdot \dot{q} - \Phi_{tt} = -\left(\Phi_{q} \cdot \dot{q}\right)_{q} \cdot \dot{q} - \Phi$$

The kinematic constraint are now defined.

To find the mass matrix, equation (3.3) can be applied for one body. As the body is a thin rectangular rod where it is constrained in the end, the polar mass moment of inertia can be defined as $J' = \frac{m \cdot l^2}{12}$. The applied force in the equation (3.7) is gravity in the y-axis and a rotational spring-damper force in the angle as no other force is applied.

$$Q^{A} = \begin{bmatrix} 0 \\ -m_{1} \cdot g \\ 0 \\ 0 \\ -m_{2} \cdot g \\ -n \end{bmatrix}$$
 (3.12)

where $n = K_0 \cdot (q(6) - q(6)_{start}) + C0 \cdot \dot{q}(6)$

To solve for the acceleration in the equation of motion it can be rewritten as

$$\begin{bmatrix} \ddot{q}_{6x1} \\ \lambda_{5x1} \end{bmatrix} = \begin{bmatrix} M_{6x6} & \Phi_{q(6x5)}^T \\ \Phi_{q(5x6)} & 0_{5x5} \end{bmatrix}_{11x11}^{-1} \begin{bmatrix} Q^A \\ \gamma \end{bmatrix}_{11x1}$$
(3.13)

Furthermore as the constraint forces using the Lagrange multiplier theorem is neglected, they can be found by following equation

$$F_{constraint(6x1)} = -\Phi_{q(6x5)}^T \cdot \lambda_{(5x1)}$$

$$\tag{3.14}$$

Equation (3.14) can be used for the constraint forces for the generalized coordinated but if wanting to look at the constraint forces in a connection, the following equation can be written as

$$F_i^{nk} = -C_i^T A_i^T \Phi_{ri}^{k T} \lambda^k \tag{3.15}$$

It should be noted at some situation such as this, the C- and A-matrix will be the identity matrix.

To find the torsion the equation can be written as

$$T_i^{nk} = (s_i^{'PT} B_i^T \Phi_{ri}^{kT} - \Phi_{\phi i}^{kT}) \lambda^k$$
 (3.16)

4 Numerical Integration

To solve the equation of motion, a numerical integration solver has to be used. A option is Runge-Kutta which finds an approximation of non linear equations. By applying an initial problem such as:

$$\frac{dy}{dt} = f(t,y) \tag{4.1}$$

using an initial start guesses $y(t_0) = y_0$ where y_0 represent all unknown variables. The integrator uses Euler's method 4 times and weight averaging the guesses.

$$y_{i+1} = y_i + \frac{1}{6}h(k_1 + 2k_2 + 2k_3 + k_4)$$
(4.2)



where h is the step which can be defined by δt . Each k guess is defined as:

$$k_{1} = f(t_{i}, y_{i})$$

$$k_{2} = f(t_{i} + \frac{h}{2}, y_{i} + h\frac{k_{1}}{2}$$

$$k_{3} = f(t_{i} + \frac{h}{2}, y_{i} + h\frac{k_{2}}{2}$$

$$k_{4} = f(t_{i} + h, y_{i} + hk_{3})$$

$$(4.3)$$



5 Plots

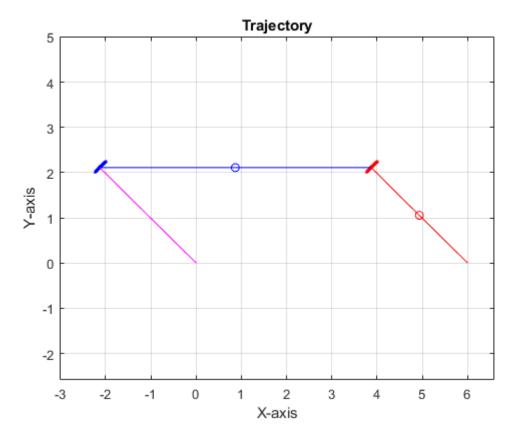


Figure 2: Trajectory



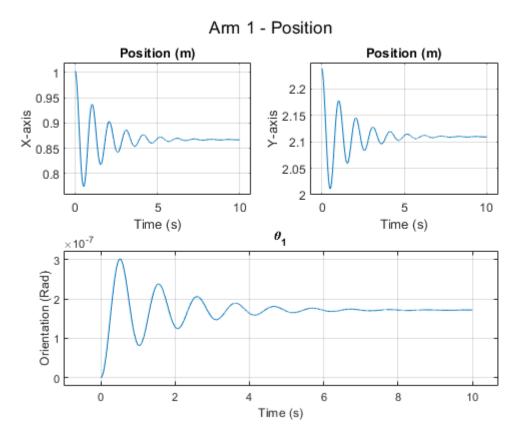


Figure 3: Position of arm 1

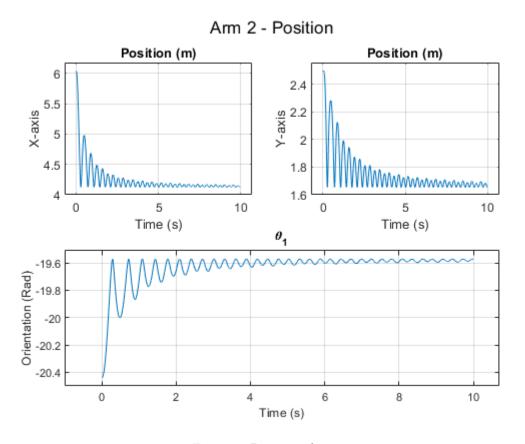


Figure 4: Position of arm 2



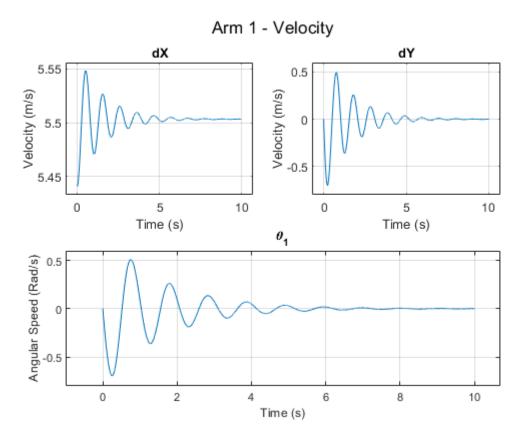


Figure 5: Velocity of arm 1

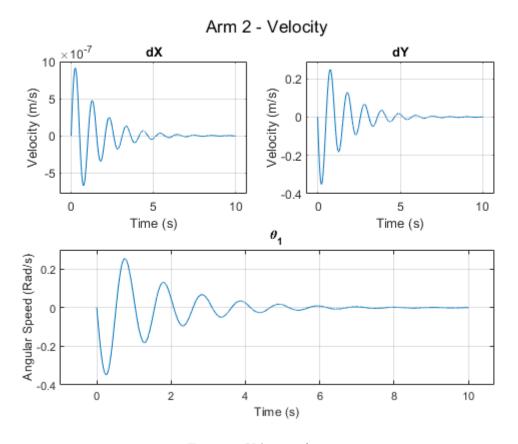


Figure 6: Velocity of arm 2



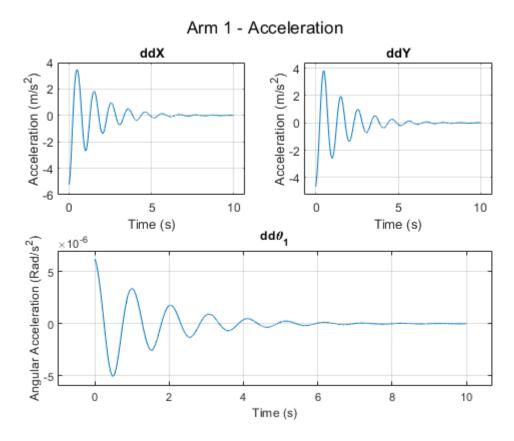


Figure 7: Acceleration of arm 1

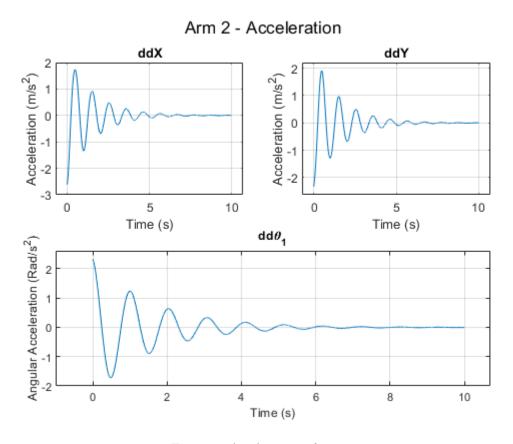


Figure 8: Acceleration of arm 2



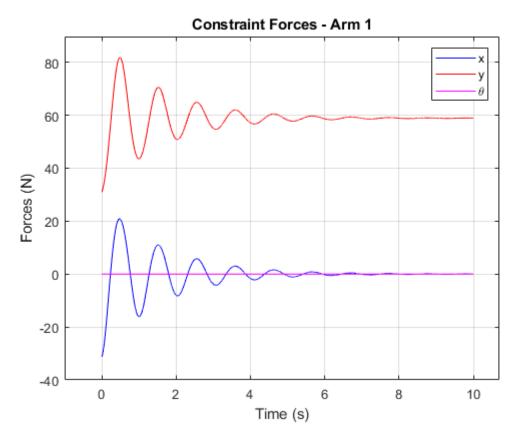


Figure 9: Constraint forces of arm 1

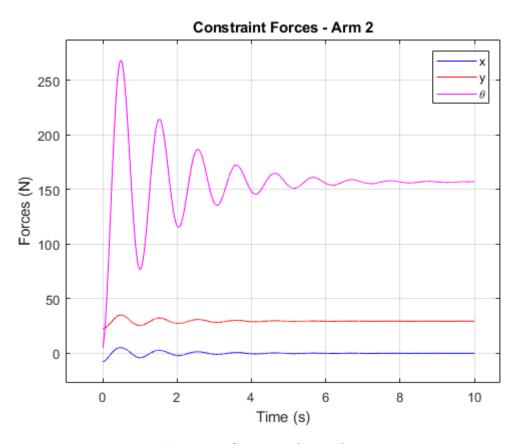


Figure 10: Constraint forces of arm 2



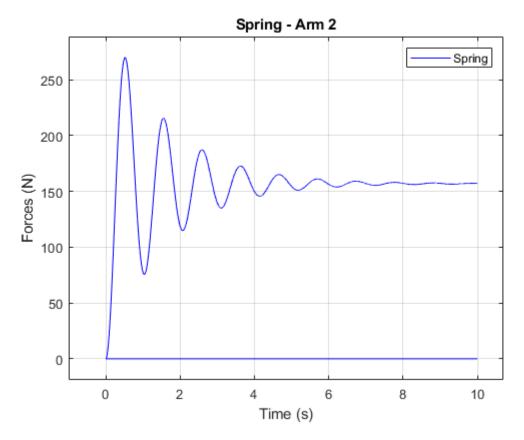


Figure 11: Spring Forces

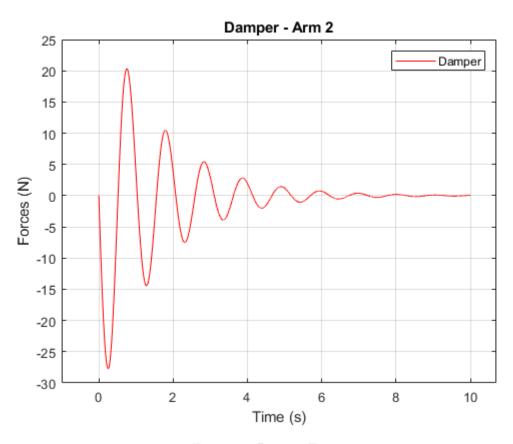


Figure 12: Damper Forces



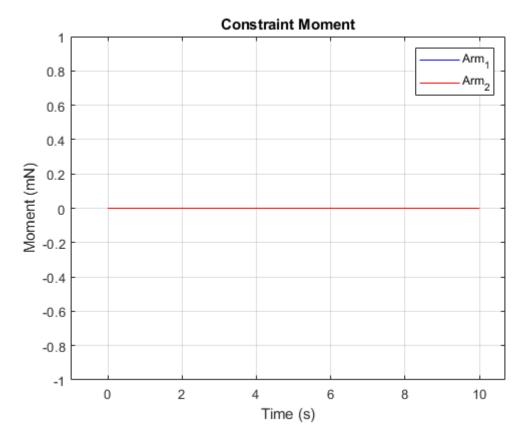


Figure 13: Moment



6 Appendix

6.1 Main Code

```
1 close all; clear all;
2 tspan = 0:0.001:10;
3 params = params;
4 L_2 = params.L2;
5 L_3 = params.L3;
6 y0 = zeros(1,17);
   q_guess = [-2; 0.5; pi/4; 1; 1; pi/10];
9 grashof = [params.L1, params.L2, params.L3, params.G];
10 if max(grashof)+min(grashof) < sum(grashof)-max(grashof)+min(grashof) % See if Grashof ...</pre>
        is satisfied
        %q_quess = NewtonRaphston(tspan(1),q_quess)
11
12 else
       error("Grashof is not satisfied")
13
14 end
15 q_guess = [1.0036; 2.2393; 0; 5.0018; 1.1197; 5.4405];
y0 = [q_guess; zeros(11,1)];
17
   % Runge Kutta with correction
opts = odeset; opts=odeset(opts,'RelTol',1e-13,'AbsTol',1e-14);
19 [t y] = ode45(@(t,y) EOM(t,y,params),tspan,y0',opts);
20
   % Position, velocity, acceleration
q = y(:, 1:6);
22 q_dot = y(:,6:12);
23 B = @(theta) [-sin(theta) -sin(theta); cos(theta) -sin(theta)];
24 % Acceeleration, Constraint Forces, Constraint Torso
dy = zeros(length(tspan), 17);
   lambda = zeros(5,length(tspan));
   for i = 1:length(tspan)
27
28
       dy(i,:) = EOM(tspan(i),y(i,:)',params);
29
       lambda(:,i) = dy(i,13:17);
       CF(i,:) = -Phi_q(params, q(i,:)).'*lambda(:,i);
30
       phik = Phi_q(params, q(i,:));
31
       CT(i,1) = ([L_2/2;0].'*B(q(i,3)).'*phik(2:3,1:2).'-phik(2:3,3).')*lambda(2:3,i);
32
        \texttt{CT(i,2)} \ = \ ([-\texttt{L}\_3/2;0].'*\texttt{B(q(i,6))}.'*\texttt{phik(2:3,4:5)}.'-\texttt{phik(2:3,6)}.')*\texttt{lambda(2:3,i)}; 
33
34 end
35 q_ddot = dy(:,7:12);
36
37 % Constraint forces - Arm 1
38 figure (14)
39
   plot(t,CT(:,1),'b',t,CT(:,2),'r')
40 hold on
41 axis padded
   grid on
43 legend('Arm_1','Arm_2')
44 title('Constraint Moment')
45 ylabel('Moment (mN)')
46 xlabel('Time (s)')
47 hold off
49
50 % Spring-Damper Forces
51 K0 = params.K0;
52 C0 = params.C0;
53 Spring_Force = zeros(length(tspan));
   Damper_Force_Force = zeros(length(tspan));
54
   for i = 1:length(tspan)
55
       Spring_Force(i) = K0*(q(i,6)-y0(6));
       Damper_Force(i) = C0*q_dot(i,6);
57
   end
58
59
60
  % Getting points
61
   A = @(theta) [cos(theta) -sin(theta); sin(theta) cos(theta)];
   for i = 1:length(tspan)
62
       P1(i,:) = [q(i,1);q(i,2)] + A(q(i,3))*[-L_2/2;0]; % Second arm left point
63
       P2(i,:) = [q(i,1);q(i,2)] + A(q(i,3))*[L_2/2;0]; % Second arm right point
64
       P3(i,:) = [q(i,4);q(i,5)] + A(q(i,6))*[-L_3/2;0]; % Third arm left point
65
       P4(i,:) = [q(i,4);q(i,5)] + A(q(i,6))*[L_3/2;0]; % Third arm right point
```



```
67 end
68
69 % figure(1)% Live Trajectory
70 % hold on
71 % grid on
72 % title('Live Trajectory')
73 % xlabel('X-axis')
74 % ylabel('Y-axis')
   % xlim([-4 8])
76 % ylim([-4 5])
77 % tic
   % for p = 1:length(tspan)
78
          \label{eq:body1} \mbox{Body1} = \mbox{line([P1(p,1) \ P2(p,1)], \ [P1(p,2) \ P2(p,2)]);}
79 %
   응
          Body1.Color = 'r';
          Body2 = line([P3(p,1) P4(p,1)], [P3(p,2) P4(p,2)]);

Body2.Color = 'b';
81
82
   2
   용
          Body3 = line([0 P1(p,1)], [0 P1(p,2)]);
          Body3.Color = 'm';
84
85
          pause(0.000001)
          delete(Body1);
86
87 응
          delete(Body2);
   오
          delete(Body3);
88
89 % end
90 % toc
91 % plot(q(p,1),q(p,2),'ro-');
92 % plot(q(p,4),q(p,5),'bo-');
93 % Body1 = line([P1(end,1) P2(end,1)], [P1(end,2) P2(end,2)]);
   % Body1.Color = 'r';
94
95 % Body2 = line([P3(end, 1) P4(end, 1)], [P3(end, 2) P4(end, 2)]);
96 % Body2.Color = 'b';
97 % Body3 = line([0 P1(end,1)], [0 P1(end,2)]);
98 % Body3.Color = 'm';
99 % hold off
100 %
101 % Trajectory
102 % figure(2)
103 % plot(P1(:,1),P1(:,2),'.','Color','b');
104 % hold on
105 % plot(P2(:,1),P2(:,2),'b');
106 % plot(P3(:,1),P3(:,2),'.','Color','r');
107 % plot(P4(:,1),P4(:,2),'r');
108 % plot(q(end,1),q(end,2),'bo-');
109 % plot(q(end, 4), q(end, 5), 'ro-');
110 % axis equal
111 % grid on
112 % axis padded
113 % title('Trajectory')
114 % xlabel('X-axis')
115 % ylabel('Y-axis')
116 % Body3 = line([0 P1(end,1)], [0 P1(end,2)]);
117 % Body3.Color = 'm';
118 % Body1 = line([P1(end,1) P2(end,1)], [P1(end,2) P2(end,2)]);
119 % Body1.Color = 'b';
120 % Body2 = Line([P3(end, 1) P4(end, 1)], [P3(end, 2) P4(end, 2)]);
121 % Body2.Color = 'r';
122 % hold off
123
124 % Constraint forces - Arm 1
125 % figure (9)
126 % plot(t, CF(:,1), 'b',t, CF(:,2), 'r',t, CF(:,3), 'm')
127 % hold on
128 % axis padded
129 % grid on
130 % legend('x','y','\theta')
131 % title('Constraint Forces - Arm 1')
132 % ylabel('Forces (N)')
133 % xlabel('Time (s)')
134 % hold off
135 %
136
   % % Constraint forces - Arm 2
137 % figure(10)
138 % plot(t,CF(:,4),'b',t,CF(:,5),'r',t,CF(:,6),'m')
   % hold on
```



```
140 % axis padded
141 % grid on
142 % legend('x','y','\theta')
143 % title('Constraint Forces - Arm 2')
144 % ylabel('Forces (N)')
145 % xlabel('Time (s)')
146 % hold off
147
148 % Damper Forces
149 figure(15)
plot(t,Damper_Force,'r')
151 hold on
152 axis padded
153 grid on
154 legend('Damper')
155 title('Damper - Arm 2')
156 ylabel('Forces (N)')
157 xlabel('Time (s)')
158 hold off
159
160 % Spring Force
161 % figure(11)
162 % plot(t,Spring_Force,'b')
163 % hold on
164 % axis padded
165 % grid on
166 % legend('Spring')
167 % title('Spring - Arm 2')
168 % ylabel('Forces (N)')
169 % xlabel('Time (s)')
170 % hold off
171
172 % % Position - Arm 1
173 % figure(3)
174 % sgtitle('Arm 1 - Position')
175 % subplot (2,2,1,'Color','r')
176 % plot(t,q(:,1))
177 % grid on
178 % axis padded
179 % title('Position (m)')
180 % xlabel('Time (s)')
181 % ylabel('X-axis')
182 % subplot (2,2,2)
183 % plot(t,q(:,2))
184 % grid on
185 % axis padded
186 % title('Position (m)')
187 % xlabel('Time (s)')
188 % ylabel('Y-axis')
189 % subplot(2,2,[3 4],'Color','b')
190 % plot(t,q(:,3))
191 % grid on
192 % axis padded
193 % title('\theta_1')
194 % xlabel('Time (s)')
195 % ylabel('Orientation (Rad)')
196 %
197 % % Velocity - Arm 1
198 % figure (4)
199 % sgtitle('Arm 1 - Velocity')
200 % grid on
201 % axis padded
202 % subplot (2,2,1)
203 % plot(t,q_dot(:,1))
204 % grid on
205 % axis padded
206 % title('dX')
207 % xlabel('Time (s)')
208 % ylabel('Velocity (m/s)')
209 % subplot (2,2,2)
210 % plot(t,q_dot(:,2))
211 % grid on
212 % axis padded
```



```
213 % title('dY')
214 % xlabel('Time (s)')
215 % ylabel('Velocity (m/s)')
216 % subplot(2,2,[3 4])
217 % plot(t,q_dot(:,3))
218 % grid on
219 % axis padded
220 % title('\theta_1')
221 % xlabel('Time (s)')
222 % ylabel('Angular Speed (Rad/s)')
223 %
224 % % Acceleration - Arm 1
225 % figure(5)
226 % sgtitle('Arm 1 - Acceleration')
227 % grid on
228 % axis padded
229 % subplot (2,2,1)
230 % plot(t,q_ddot(:,1))
231 % grid on
232 % axis padded
233 % title('ddX')
234 % xlabel('Time (s)')
235 % ylabel('Acceleration (m/s^2)')
236 % subplot (2,2,2)
237 % plot(t,q_ddot(:,2))
238 % grid on
239 % axis padded
240 % title('ddY')
241 % xlabel('Time (s)')
242 % ylabel('Acceleration (m/s^2)')
243 % subplot (2,2,[3 4])
244 % plot(t,q_ddot(:,3))
245 % grid on
246 % axis padded
247 % title('dd\theta_1')
248 % xlabel('Time (s)')
249 % ylabel('Angular Acceleration (Rad/s^2)')
250 %
251 % % Position - Arm 2
252 % figure(6)
253 % sgtitle('Arm 2 - Position')
254 % subplot(2,2,1,'Color','r')
255 % plot(t,q(:,4))
256 % grid on
257 % axis padded
258 % title('Position (m)')
259 % xlabel('Time (s)')
260 % ylabel('X-axis')
261 % subplot(2,2,2)
262 % plot(t,q(:,5))
263 % grid on
264 % axis padded
265 % title('Position (m)')
266 % xlabel('Time (s)')
267 % ylabel('Y-axis')
268 % subplot(2,2,[3 4],'Color','b')
269 % plot(t,q(:,6))
270 % grid on
271 % axis padded
272 % title('\theta_1')
273 % xlabel('Time (s)')
274 % ylabel('Orientation (Rad)')
275 %
276 % % Velocity - Arm 2
277 % figure(7)
278 % sgtitle('Arm 2 - Velocity')
279 % grid on
280 % axis padded
281 % subplot (2,2,1)
282 % plot(t,q_dot(:,4))
283 % grid on
284 % axis padded
285 % title('dX')
```



```
286 % xlabel('Time (s)')
287 % ylabel('Velocity (m/s)')
288 % subplot (2,2,2)
289 % plot(t,q_dot(:,5))
290 % grid on
291 % axis padded
292 % title('dY')
293 % xlabel('Time (s)')
294 % ylabel('Velocity (m/s)')
295 % subplot (2,2,[3 4])
296 % plot(t,q_dot(:,6))
297 % grid on
298 % axis padded
299 % title('\theta_1')
300 % xlabel('Time (s)')
301 % ylabel('Angular Speed (Rad/s)')
303 % % Acceleration - Arm 2
304 % figure(8)
305 % sgtitle('Arm 2 - Acceleration')
306 % grid on
307 % axis padded
308 % subplot (2,2,1)
309 % plot(t,q_ddot(:,4))
310 % grid on
311 % axis padded
312 % title('ddX')
313 % xlabel('Time (s)')
314 % ylabel('Acceleration (m/s^2)')
315 % subplot (2,2,2)
316 % plot(t,q_ddot(:,5))
317 % grid on
318 % axis padded
319 % title('ddY')
320 % xlabel('Time (s)')
321 % ylabel('Acceleration (m/s^2)')
322 % subplot(2,2,[3 4])
323 % plot(t,q_ddot(:,6))
324 % grid on
325 % axis padded
326 % title('dd\theta_1')
327 % xlabel('Time (s)')
328 % ylabel('Angular Acceleration (Rad/s^2)')
```

6.2 Equation of Motion

```
1 function dy = EOM(t,y,params)
2 m2 = params.m2;
3 m3 = params.m3;
4 J2 = params.J2;
5 J3 = params.J3;
6 g = params.g;
7 \text{ KO} = \text{params.KO};
8 C0 = params.C0;
9 persistent y0;
q = y(1:6);
11 q_{dot} = y(7:12);
12 lambda = y(13:17);
14
15 if t == 0
       y0 = q(6);
17 end
18 \quad n_1 = K0 * (q(6) - y0) + C0 * q_dot(6);
20 \, \text{M} = \text{diag([m2 m2 J2 m3 m3 J3]);}
                                                                             % Mass Matrix
Q_A = [0 -m2*g 0 0 -m3*g -n_1].';
                                                                                 % Applied Force
22 A = [M Phi_q(params,q).'; Phi_q(params,q) zeros(5)];
                                                                              % Combined Matrix
B = [Q_A; Gamma(params, q, q_dot)];
                                                                              % Combined Matrix
24 C = A \setminus B;
                                                                              % Acc, Lambda
```



6.3 Matlab - Symbolic

```
sympref('AbbreviateOutput', false);
_{\rm 2} syms x.2 y.2 theta.2 x.3 y.3 theta.3 L.1 L.2 L.3 G omega t;
3 syms x_dot_2 y_dot_2 theta_dot_2 x_dot_3 y_dot_3 theta_dot_3;
4 syms A(theta)
6 A(theta) = [\cos(\text{theta}) - \sin(\text{theta}); \sin(\text{theta}) \cos(\text{theta})];
  r_2 = [x_2; y_2];
s r_3 = [x_3; y_3];
10 sip_2 = [L_2/2;0];
11 \text{ sip}_3 = [L_3/2; 0];
13
  % Defining Arm 1
14 C = [0;0];
15 \quad C_3 = L_1;
rp2 = r_2 + A(theta_2) * - sip_2;
17
18 phi_k3 = r_3 + A(theta_3)*sip_3-[G;0]
                                                                             % Absolute ...
       constraint - L_3
   phi_k2 = r_2 + A(theta_2)*sip_2-r_3-A(theta_3)*-sip_3;
                                                                                      % Revolute ...
       constraint - L_3-L_2
phi_k1 = (rp2-C).'*(rp2-C)-C_3^2
                                                                             % Absolute ...
       constraint - L_2
21
22 % Constraints
phi_K = [phi_k1;phi_k2;phi_k3];
                                                                              % Kinematic ...
       constraint
q = [x_2 y_2 theta_2 x_3 y_3 theta_3].'
                                                                              % Generalized ...
       cartesian coordinates
25 phi = [phi_K]
                                                                              % Constraint matrix
26 phiq = jacobian(phi,q)
                                                                              % Jacobian Matrix
27 phit = -diff(phi,t)
                                                                              % Nu Matrix
28 q_dot = [x_dot_2 y_dot_2 theta_dot_2 x_dot_3 y_dot_3 theta_dot_3].';
                                                                              % Generealized ...
       cartesian velocity coordinates
29 gamma = -jacobian(phiq*q_dot,q)*q_dot-diff(phit,t)
                                                                              % Gamma matrix
30 %making functions of the symbolic
31 matlabFunction(phi,'file','Phi');
32 matlabFunction(phiq,'file','Phi_q');
33 matlabFunction(phit, 'file', 'Nu');
34 matlabFunction(gamma,'file','Gamma');
```

6.4 Parameters

```
1 function params = params
_2 params.m2 = 6;
3 \text{ params.m3} = 3;
4 params.L1 = 3;
5 params.L2 = 6;
6 params.L3 = 3;
7 params.J2 = params.m2/12*params.L2^2;
8 params.J3 = params.m3/12*params.L3^2;
9 params.g = 9.81;
10 params.G = 6;
11 params.K0 = 2500;
12 params.C0 = 80;
14 % params = params;
  % L_2 = params.L2;
16 % L_3 = params.L3;
17 \% x_2 = q(1);
```



```
18 % y_2 = q(2);
19 % theta_2 = q(3);
20 % x_3 = q(4);
21 % y_3 = q(5);
22 % theta_3 = q(6);
23 % x_dot_2 = q_dot(1);
24 % y_dot_2 = q_dot(2);
25 % theta_dot_2 = q_dot(3);
26 % x_dot_3 = q_dot(4);
27 % y_dot_3 = q_dot(5);
28 % theta_dot_3 = q_dot(6);
```

6.5 Φ_q

```
function phiq = Phi_q(params,q)
2
   %PHI_Q
3 %
        PHIQ = PHI_Q(L_2, L_3, THETA_2, THETA_3, X_2, Y_2)
4
5
        This function was generated by the Symbolic Math Toolbox version 8.7.
6 %
        08-May-2022 16:02:29
8 L_2 = params.L2;
9 L_3 = params.L3;
10 theta_2 = q(3);
11 theta_3 = q(6);
x_2 = q(1);
y_2 = q(2);
14
15
t2 = cos(theta_2);
17 	 t3 = cos(theta_3);
18 t4 = \sin(theta_2);
19 t5 = \sin(theta_3);
20 	 t6 = (L_2.*t2)./2.0;
t7 = (L_3.*t3)./2.0;
120 \text{ t8} = (L_2.*t4)./2.0;
23 	 t9 = (L_3.*t5)./2.0;
24 	 t10 = -t8;
25 t11 = -t9;
26 phiq = ...
       reshape([x_2.*2.0-L_2.*t2,1.0,0.0,0.0,0.0,y_2.*2.0-L_2.*t4,0.0,1.0,0.0,0.0,-L_2.*t4.*(t6-x_2)+L_2.*t2.
```

6.6 γ

```
1 function gamma = Gamma(params,q,q-dot)
2 %GAMMA
3 %
        GAMMA = GAMMA(L_2,L_3,THETA_2,THETA_3,THETA_DOT_2,THETA_DOT_3,X_2,X_DOT_2,Y_2,Y_DOT_2)
5 %
        This function was generated by the Symbolic Math Toolbox version 8.7.
6 %
        08-May-2022 16:02:29
8 L_2 = params.L2;
9 L_3 = params.L3;
10 theta_2 = q(3);
11 theta_3 = q(6);
12 theta_dot_2 = q_dot(3);
13 theta_dot_3 = q_dot(6);
14 x_2 = q(1);
x_{dot_2} = q_{dot_1}
y_2 = q(2);
17
   y_dot_2 = q_dot(2);
19 t2 = cos(theta_2);
20 	 t3 = cos(theta_3);
t4 = \sin(theta_2);
12 	ext{t5} = \sin(\text{theta}_3);
```



```
23 t6 = L_2.^2;

24 t7 = theta_dot_2.^2;

25 t8 = theta_dot_3.^2;

26 t9 = (L_3.*t3.*t8)./2.0;

27 t10 = (L_3.*t5.*t8)./2.0;

28 gamma = ...

[-theta_dot_2.*(theta_dot_2.*((t2.^2.*t6)./2.0+(t4.^2.*t6)./2.0+L_2.*t2.*(x_2-(L_2.*t2)./2.0)+L_2.*t4.*]
```

6.7 Φ

```
1 function phi = Phi(G, L_1, L_2, L_3, theta_2, theta_3, x_2, x_3, y_2, y_3)
2 %PHI
3 %
        PHI = PHI(G, L_1, L_2, L_3, THETA_2, THETA_3, X_2, X_3, Y_2, Y_3)
5 %
        This function was generated by the Symbolic Math Toolbox version 8.7.
6
   응
        08-May-2022 16:02:28
s t2 = cos(theta_2);
   t3 = cos(theta_3);
10 	 t4 = sin(theta_2);
11 t5 = sin(theta_3);
12 	 t6 = (L_2.*t2)./2.0;
13 	 t7 = (L_3.*t3)./2.0;
14 t8 = (L_2.*t4)./2.0;
15 	 t9 = (L_3.*t5)./2.0;
16 phi = [(t6-x_2).^2+(t8-y_2).^2-L_1.^2;t6+t7+x_2-x_3;t8+t9+y_2-y_3;-G+t7+x_3;t9+y_3];
```

6.8 Newton Raphton

```
1 function w = NewtonRaphston(t,q)
_{2} es = 0.00000000001;
                           % Tolerance
3 \text{ maxit} = 100000;
                          % Maximum iteration
4 iter = 0; % Defining iteration
5 	ext{ driving = zeros}(1,6);
6 driving(3) = 1;
7 while(1)
       Phi1 = [Phi(params,t,q);q(3)*1*t];
       Jac = [Phi_q(q,params); driving];
9
       dq = Jac\Phi1; % Defining the change
10
       q = q-dq; % Removing change from previous guess
11
       iter = iter+1; % Counting Iteration
12
       ea = 100*max(abs(dq./q)); % Looking at error-value
13
       if(iter≥maxit || ea≤es) % See if conditions are satisfied
14
           w = q; % Saving data if satisfied
15
16
           break
       end
17
18 end
```

