```
"Turbo"
"HW3 P4"
"Zhaoyi Jiang(.1364)"
"Given"
p01=6.5[bar]
t01=800[k]
p03=1.3[bar]
psi=1.1
phi=0.58
alpha_1=-5[degree]
alpha_3=alpha_1
psi=2*(1-R-phi*tan(alpha 3))
"Anales"
psi=phi*(tan(alpha_2)-tan(alpha_3))
beta_1=beta_3
beta_2=arctan(-(R-psi/2)/phi)
beta_3=arctan(-(R+psi/2)/phi)
"Work in normail stage"
eta tt=0.94
h01=enthalpy(Air, T=t01)
s01=entropy(Air,P=p01,T=t01)
h03ss=enthalpy(Air, P=p03, s=s01)
eta_tt=(h01-h03)/(h01-h03ss)
W=h01-h03
psi=W/u^2*convert(km,m)
phi=cz/u
"Velocities"
c1=c3
w1=w3
c2=cz/cos(alpha 2)
w2=((c2*sin(alpha_2)-u)^2+cz^2)^.5
c3=cz/cos(alpha 3)
w3=((c3*sin(alpha 3)-u)^2+cz^2)^.5
```

SOLUTION

Unit Settings: SI K bar kJ mass deg

```
\alpha^1 = -5 [Degree]

\beta_1 = -61.1 [degree]

c_1 = 295 [m/s]

c_2 = 293.9 [m/s]

c_3 = 539.9 [kj/kg]

c_3 = 1.3 [bar]

c_3 = 0.5007

c_3 = 506.7 [m/s]

c_4 = 294.9 [m/s]
```

```
\alpha^2 = 61.07 \text{ [degree]}
\beta^2 = 4.854 \text{ [degree]}
c^2 = 607.5 \text{ [m/s]}
\eta t = 0.94
t^2 = 0.58
t^2 = 0.58
t^2 = 0.184 \text{ [kj/kg-k]}
t^2 = 0.184 \text{ [kj/kg]}
```

```
\alpha^3 = -5 [degree]

\beta^3 = -61.1 [degree]

\alpha^3 = 295 [m/s]

\alpha^3 = 295 [m/s]

\alpha^3 = 295 [m/s]

\alpha^3 = -61.1 [bar]

\alpha^3 = -5 [degree]

\alpha^3 = -5 [degree]

\alpha^3 = -5 [degree]

\alpha^3 = -5 [degree]

\alpha^3 = -61.1 [degree]

\alpha^3 = -61.1
```

No unit problems were detected.