

Track Surveillance Drone - System Requirements Document

Project: Track Surveillance Drone

Version: 1.0

Date: [Current Date]

Classification: [As Required]

1. Introduction

1.1 Purpose

This System Requirements Document (SRD) defines the technical system requirements for the Track Surveillance Drone system, derived from the User Requirements Document (URD) version 1.0. The document establishes the technical foundation for design, development, and verification activities in accordance with CENELEC standards for railway applications.

1.2 Scope

This document covers all technical aspects of the Track Surveillance Drone system including hardware, software, interfaces, safety, security, and operational requirements. The system shall operate as an autonomous inspection platform for railway track infrastructure.

1.3 Definitions and Abbreviations

- **CENELEC:** European Committee for Electrotechnical Standardization
- **RaSTA:** Safe Transport Protocol for Railway Applications
- **GNSS:** Global Navigation Satellite System
- **EMC:** Electromagnetic Compatibility
- **MTBF:** Mean Time Between Failures
- **MTTR:** Mean Time To Repair
- **SIL:** Safety Integrity Level
- **THR:** Tolerable Hazard Rate
- **LRU:** Line Replaceable Unit
- **BIT:** Built-In Test
- **FMECA:** Failure Mode, Effects and Criticality Analysis

1.4 Requirements Obligation Levels

- **Requirement:** Mandatory requirements (equivalent to Shall/Must)
- **Recommendation:** Recommended requirements (equivalent to Should)
- **Info:** Informative statements (equivalent to Will)

2. Referenced Documents

2.1 CENELEC Standards

- EN 50126: Railway Applications - Specification and Demonstration of Reliability, Availability, Maintainability and Safety (RAMS)
- EN 50128: Railway Applications - Communication, Signalling and Processing Systems - Software for Railway Control and Protection Systems
- EN 50129: Railway Applications - Communication, Signalling and Processing Systems - Safety Related Electronic Systems for Signalling
- EN 50155: Railway Applications - Electronic Equipment used on Rolling Stock
- EN 50701: Railway Applications - Cybersecurity

2.2 Other Standards

- IEC 61508: Functional Safety of Electrical/Electronic/Programmable Electronic Safety-related Systems
- DO-178C: Software Considerations in Airborne Systems and Equipment Certification
- ISO 26262: Road Vehicles - Functional Safety
- 2011/65/EU: RoHS Directive

2.3 Project Documents

- Track Surveillance Drone URD v1.0
- System Architecture Document
- Safety Plan
- Verification and Validation Plan

3. General Description

3.1 System Overview

The Track Surveillance Drone system consists of an autonomous unmanned aerial vehicle equipped with inspection sensors, communication systems, and ground-based control infrastructure for railway track monitoring and defect detection.

3.2 System Architecture

The system architecture comprises:

- Aerial Platform (drone hardware)
- Sensor Suite (cameras, LIDAR, thermal imaging)
- Flight Control System
- Communication Subsystem

- Ground Control Station
- Data Processing and Analysis System
- Asset Management Interface

4. System Requirements

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|------------|
| SR-001 | Technical Requirements | Info | - |
| SR-002 | The following section defines the technical requirements for the drone system including environmental, hardware, functional, and navigation specifications. | Info | - |
| SR-003 | The drone flight control system shall operate continuously at altitudes from 0 to 2000 meters above sea level without performance degradation. | Requirement | UR-01 |
| SR-004 | The barometric pressure compensation system shall maintain flight stability across the operational altitude range with accuracy of ± 1 meter. | Requirement | UR-01 |
| SR-005 | All electronic components shall maintain specified performance parameters within the temperature range of -20°C to $+50^{\circ}\text{C}$ during operation. | Requirement | UR-02 |
| SR-006 | The thermal management system shall maintain critical component temperatures within 5°C of optimal operating range during maximum performance operations. | Requirement | UR-02 |
| SR-007 | Battery performance shall maintain minimum 80% capacity throughout the operational temperature range. | Requirement | UR-02 |
| SR-008 | The drone shall maintain stable flight operations in precipitation up to 2.5 mm/hour and wind speeds up to 40 km/h (11 m/s). | Requirement | UR-03 |
| SR-009 | Water ingress protection shall prevent operational degradation during 30 minutes of continuous light rain exposure. | Requirement | UR-03 |
| SR-010 | The flight control system shall automatically compensate for wind gusts up to 60 km/h (17 m/s) peak velocity. | Requirement | UR-03 |
| SR-011 | All system components shall survive storage temperatures from -40°C to $+60^{\circ}\text{C}$ without permanent damage or performance degradation. | Requirement | UR-04 |
| SR-012 | Storage humidity tolerance shall extend from 5% to 95% relative humidity without condensation effects. | Requirement | UR-04 |
| SR-013 | The weather monitoring subsystem shall measure and report wind speed (± 1 m/s accuracy), temperature ($\pm 2^{\circ}\text{C}$ accuracy), humidity ($\pm 5\%$ accuracy), and precipitation intensity every 10 seconds. | Requirement | UR-05 |
| SR-014 | Weather data shall be transmitted to ground control station with maximum 5-second latency. | Requirement | UR-05 |
| SR-015 | The flight control system shall automatically abort mission and execute safe landing procedure when weather conditions exceed operational limits. | Requirement | UR-05 |
| SR-016 | Weather monitoring sensors shall self-calibrate every 24 hours of operation. | Requirement | UR-05 |

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|------------|
| SR-017 | All electronic components and materials shall comply with EU RoHS Directive 2011/65/EU with maximum concentrations: Lead (0.1%), Mercury (0.1%), Cadmium (0.01%), Hexavalent chromium (0.1%), PBB (0.1%), PBDE (0.1%). | Requirement | UR-06 |
| SR-018 | Material certificates shall be maintained for all components with hazardous substance declarations. | Requirement | UR-06 |
| SR-019 | The drone shall support hot-swappable battery packs with standardized electrical and mechanical interfaces. | Requirement | UR-07 |
| SR-020 | Battery pack electrical interface shall provide reverse polarity protection and over-current protection. | Requirement | UR-07 |
| SR-021 | Battery packs shall provide minimum 45 minutes flight time at maximum payload configuration. | Requirement | UR-07 |
| SR-022 | Battery management system shall monitor individual cell voltages, temperatures, and charge/discharge currents. | Requirement | UR-07 |
| SR-023 | Battery pack mechanical interface shall withstand 1000 insertion/removal cycles without degradation. | Requirement | UR-07 |
| SR-024 | The primary housing shall achieve minimum IP54 rating with ingress protection against dust and water splashing from any direction. | Requirement | UR-08 |
| SR-025 | Critical electronic components shall be housed in IP67 rated enclosures. | Requirement | UR-08 |
| SR-026 | Housing materials shall demonstrate UV resistance for minimum 5 years outdoor exposure. | Requirement | UR-08 |
| SR-027 | Gasket and seal materials shall maintain integrity through 500 thermal cycles from -20°C to +50°C. | Requirement | UR-08 |
| SR-028 | The drone shall display high-visibility markings compliant with aviation regulations including registration number, weight class, and emergency contact information. | Requirement | UR-09 |
| SR-029 | Marking materials shall maintain visibility and adhesion for minimum 2 years outdoor exposure. | Requirement | UR-09 |
| SR-030 | Emergency contact information shall be displayed in characters minimum 10mm height. | Requirement | UR-09 |
| SR-031 | The emergency shut-off switch shall be accessible within 5 seconds and shall immediately terminate all motor functions and enable emergency landing mode. | Requirement | UR-10 |
| SR-032 | Emergency shut-off shall be operable with protective gloves and in low-light conditions. | Requirement | UR-10 |

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|------------|
| SR-033 | Emergency shut-off activation shall be logged with timestamp and transmitted to ground control. | Requirement | UR-10 |
| SR-034 | The defect detection system shall identify track irregularities including: rail breaks, bolt failures, ballast displacement, vegetation encroachment, and surface defects with 95% detection accuracy. | Requirement | UR-11 |
| SR-035 | Track defect classification shall distinguish between severity levels: Critical (immediate attention), Major (within 24 hours), Minor (scheduled maintenance). | Requirement | UR-11 |
| SR-036 | The defect detection algorithm shall process imagery at minimum 30 frames per second with real-time analysis. | Requirement | UR-11 |
| SR-037 | False positive rate for defect detection shall not exceed 5% under normal operating conditions. | Requirement | UR-11 |
| SR-038 | The system shall detect rail surface defects as small as 5mm in any dimension. | Requirement | UR-11 |
| SR-039 | Defect location accuracy shall be within ± 0.5 meters along track centerline. | Requirement | UR-11 |
| SR-040 | The autonomous navigation system shall follow pre-programmed flight paths with ± 2 meter lateral accuracy using GNSS and visual odometry. | Requirement | UR-12 |
| SR-041 | The flight path planning system shall automatically generate optimized routes considering track geometry, obstacles, and restricted airspace. | Requirement | UR-12 |
| SR-042 | Navigation system shall maintain position accuracy during GNSS signal loss for minimum 2 minutes using inertial navigation. | Requirement | UR-12 |
| SR-043 | Flight path deviation alerts shall be generated when lateral position exceeds ± 5 meters from planned route. | Requirement | UR-12 |
| SR-044 | The system shall automatically adjust flight altitude to maintain constant clearance above track infrastructure. | Requirement | UR-12 |
| SR-045 | Navigation waypoints shall be updatable during flight without interrupting autonomous operation. | Requirement | UR-12 |
| SR-046 | Real-time data transmission shall deliver video streams at minimum 1080p resolution with maximum 500ms latency to the control center. | Requirement | UR-13 |
| SR-047 | Sensor data transmission shall include GPS coordinates, timestamps, and sensor readings with 100ms maximum latency. | Requirement | UR-13 |
| SR-048 | Data transmission shall maintain 99.9% packet delivery rate under normal operating conditions. | Requirement | UR-13 |
| SR-049 | The system shall automatically adjust data transmission quality based on available bandwidth. | Requirement | UR-13 |

| ID | Requirement | Obligation | URD Source |
|--------|--|----------------|------------|
| SR-050 | Transmitted data shall include integrity checksums and error correction codes. | Requirement | UR-13 |
| SR-051 | The manual control interface shall provide full flight control authority with maximum 200ms response time from operator input to drone response. | Requirement | UR-14 |
| SR-052 | Manual override shall be available within 2 seconds from any autonomous flight mode. | Requirement | UR-14 |
| SR-053 | Manual control inputs shall be validated for safety limits before execution. | Requirement | UR-14 |
| SR-054 | The system should provide haptic feedback for manual control operations. | Recommendation | UR-14 |
| SR-055 | Manual control authority shall be transferred seamlessly without flight disruption. | Requirement | UR-14 |
| SR-056 | All inspection data shall be geo-referenced with accuracy better than 1 meter using GNSS and inertial navigation systems. | Requirement | UR-15 |
| SR-057 | Time synchronization accuracy shall be within ± 10 milliseconds of UTC. | Requirement | UR-15 |
| SR-058 | Geo-referencing data shall include altitude, heading, and roll/pitch attitude information. | Requirement | UR-15 |
| SR-059 | The return-to-home function shall activate automatically when battery level reaches 25% remaining capacity or communication loss exceeds 30 seconds. | Requirement | UR-16 |
| SR-060 | Return-to-home shall execute a pre-programmed safe landing sequence at the designated home position with ± 5 meter accuracy. | Requirement | UR-16 |
| SR-061 | Multiple return-to-home locations shall be programmable with automatic selection based on current position. | Requirement | UR-16 |
| SR-062 | Return-to-home flight path shall avoid known obstacles and restricted airspace. | Requirement | UR-16 |
| SR-063 | The system shall support redundant controller levels for availability with automatic failover capability. | Requirement | UR-17 |
| SR-064 | Controller redundancy switching shall occur within 50ms without loss of flight control. | Requirement | UR-17 |
| SR-065 | Redundant controllers shall continuously synchronize state information. | Requirement | UR-17 |
| SR-066 | Health monitoring shall detect controller failures within 100ms. | Requirement | UR-17 |

| ID | Requirement | Obligation | URD Source |
|--------|---|----------------|------------|
| SR-067 | Interface Requirements | Info | - |
| SR-068 | The following section defines the interface requirements for communication protocols, data exchange, and user interfaces. | Info | - |
| SR-069 | The transport protocol for communication with the interlocking shall be RaSTA compliant with EN 50159. | Requirement | UR-18 |
| SR-070 | RaSTA communication shall implement safety code SC-1 with 32-bit CRC and sequence numbering for data integrity. | Requirement | UR-18 |
| SR-071 | RaSTA protocol stack shall support message authentication and replay attack prevention. | Requirement | UR-18 |
| SR-072 | The system shall support remote diagnosis for any component fault with detailed fault isolation. | Requirement | UR-19 |
| SR-073 | Remote diagnostic access shall be secure with multi-factor authentication. | Requirement | UR-19 |
| SR-074 | Diagnostic data shall be transmitted in real-time during flight operations. | Requirement | UR-19 |
| SR-075 | The system shall include redundant communication links to prevent data loss using different frequency bands. | Requirement | UR-20 |
| SR-076 | Communication range shall extend to minimum 5 kilometers line-of-sight with automatic handover between base stations. | Requirement | UR-20 |
| SR-077 | Redundant communication systems shall automatically switch within 1 second of primary link failure. | Requirement | UR-20 |
| SR-078 | The system shall support secure over-the-air updates using encrypted channels with digital signature verification. | Requirement | UR-21 |
| SR-079 | Update packages shall be cryptographically signed and verified before installation with rollback capability. | Requirement | UR-21 |
| SR-080 | Software updates shall not interrupt ongoing flight operations. | Requirement | UR-21 |
| SR-081 | Update verification shall include integrity checks and compatibility validation. | Requirement | UR-21 |
| SR-082 | The operator shall be provided with a user-friendly interface for mission planning and monitoring with graphical map display. | Requirement | UR-32 |
| SR-083 | Mission planning interface shall support waypoint editing, flight path visualization, and restricted area management. | Requirement | UR-32 |
| SR-084 | The interface should provide flight simulation capability for mission validation before execution. | Recommendation | UR-32 |

| ID | Requirement | Obligation | URD Source |
|--------|---|-------------|------------|
| SR-085 | The interface shall show live status and alerts including battery level, GPS position, altitude, speed, weather conditions, and system health indicators. | Requirement | UR-33 |
| SR-086 | Alert notifications shall be presented with visual and audible indicators for critical conditions. | Requirement | UR-33 |
| SR-087 | Status information shall be updated with maximum 1-second refresh rate. | Requirement | UR-33 |
| SR-088 | The system shall log operator actions for traceability including timestamps, user identification, and command details. | Requirement | UR-34 |
| SR-089 | Operator action logs shall be tamper-evident and maintain complete audit trail. | Requirement | UR-34 |
| SR-090 | The interface shall support multi-user access with role-based permissions including Administrator, Operator, and Viewer roles. | Requirement | UR-35 |
| SR-091 | User authentication shall require minimum 8-character passwords with complexity requirements and session timeout after 30 minutes of inactivity. | Requirement | UR-35 |
| SR-092 | RAMS Requirements | Info | - |
| SR-093 | The following section defines the Reliability, Availability, Maintainability, and Safety requirements for the drone system. | Info | - |
| SR-094 | The overall system shall achieve minimum Mean Time Between Failures (MTBF) of 2000 flight hours for all mission-critical components. | Requirement | UR-23 |
| SR-095 | The flight control system shall implement dual-redundant processors with hot-standby switching capability within 50ms. | Requirement | UR-17 |
| SR-096 | Power systems shall provide triple redundancy with automatic load balancing and fault isolation. | Requirement | UR-17 |
| SR-097 | The system shall maintain 99.5% operational availability during scheduled inspection periods. | Requirement | UR-17 |
| SR-098 | Planned maintenance activities shall not exceed 2% of total operational time. | Requirement | UR-23 |
| SR-099 | The system shall support condition-based maintenance with predictive algorithms for component life estimation. | Requirement | UR-23 |
| SR-100 | Component replacement procedures shall be completed within maximum 15 minutes for field-replaceable units. | Requirement | UR-07 |
| SR-101 | Built-in test equipment shall provide fault isolation to Line Replaceable Unit (LRU) level with 95% accuracy. | Requirement | UR-26 |
| SR-102 | The drone system shall comply with Safety Integrity Level (SIL) 2 requirements per EN 50129 for safety-critical functions. | Requirement | UR-31 |

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|------------|
| SR-103 | Emergency landing procedures shall be executable within 60 seconds from fault detection to safe touchdown. | Requirement | UR-29 |
| SR-104 | The system shall implement fail-safe behavior for all safety-critical functions with automatic transition to safe state. | Requirement | UR-29 |
| SR-105 | Propulsion system failures shall not result in uncontrolled flight or ground impact with kinetic energy exceeding 79 Joules. | Requirement | UR-29 |
| SR-106 | The drone shall include obstacle detection and collision avoidance using LIDAR and stereo cameras. | Requirement | UR-28 |
| SR-107 | Obstacle detection shall identify objects larger than 10cm at distances up to 50 meters. | Requirement | UR-28 |
| SR-108 | Collision avoidance shall execute automatic evasive maneuvers maintaining minimum 5-meter clearance from detected obstacles. | Requirement | UR-28 |
| SR-109 | The system shall detect and avoid power lines, communication cables, and other overhead hazards. | Requirement | UR-28 |
| SR-110 | Emergency landing shall be controllable to avoid populated areas and critical infrastructure. | Requirement | UR-29 |
| SR-111 | The drone shall have geofencing to prevent entry into restricted airspace with 10-meter buffer zones. | Requirement | UR-30 |
| SR-112 | Geofencing boundaries shall be updateable remotely and during flight operations. | Requirement | UR-30 |
| SR-113 | The system shall automatically execute return-to-home when approaching geofence boundaries. | Requirement | UR-30 |
| SR-114 | The system shall comply with aviation safety regulations for unmanned flights including DO-178C Level C for flight-critical software. | Requirement | UR-31 |
| SR-115 | Security Requirements | Info | - |
| SR-116 | The following section defines the cybersecurity requirements for the drone system in accordance with railway security standards. | Info | - |
| SR-117 | The system shall comply with railway cybersecurity requirements EN 50701 including network segmentation, access controls, and intrusion detection. | Requirement | UR-22 |
| SR-118 | All communication channels shall use AES-256 encryption with perfect forward secrecy and mutual authentication. | Requirement | UR-22 |
| SR-119 | The system shall maintain security event logs with tamper-evident storage and automatic alerting for suspicious activities. | Requirement | UR-22 |
| SR-120 | Security patches shall be deployable within 24 hours of availability through the secure update mechanism. | Requirement | UR-22 |

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|---------------------------|
| SR-121 | Physical security measures shall prevent unauthorized access to flight control systems and data storage devices. | Requirement | UR-22 |
| SR-122 | Network intrusion detection shall monitor all communication interfaces continuously. | Requirement | UR-22 |
| SR-123 | Security key management shall implement automatic key rotation every 30 days. | Requirement | UR-22 |
| SR-124 | The system shall implement certificate-based authentication for all secure communications. | Requirement | UR-22 |
| SR-125 | Security audit logs shall be forwarded to a centralized security monitoring system in real-time. | Requirement | UR-22 |
| SR-126 | The system shall implement secure boot procedures with verified signatures for all firmware components. | Requirement | UR-22 |
| SR-127 | Manufacturing Requirements | Info | - |
| SR-128 | The following section defines the manufacturing requirements for quality control, testing, and production processes. | Info | - |
| SR-129 | Manufacturing processes shall comply with ISO 9001 quality management standards with full traceability of components and assemblies. | Requirement | UR-06 |
| SR-130 | All safety-critical components shall undergo 100% functional testing and burn-in procedures before integration. | Requirement | UR-36 |
| SR-131 | Environmental stress screening shall be performed on all electronic assemblies including thermal cycling, vibration, and humidity testing. | Requirement | UR-02, UR-03, UR-04 |
| SR-132 | Component marking and labeling shall include part numbers, serial numbers, manufacturing dates, and compliance certifications. | Requirement | UR-09 |
| SR-133 | Manufacturing documentation shall maintain complete configuration control with version management and change tracking. | Requirement | UR-36 |
| SR-134 | Production testing shall verify all functional requirements before final assembly. | Requirement | UR-36 |
| SR-135 | Quality control inspection points shall be established at each critical manufacturing stage. | Requirement | UR-36 |
| SR-136 | Manufacturing equipment shall be calibrated and maintained according to documented procedures. | Requirement | UR-36 |
| SR-137 | Non-conforming products shall be identified, segregated, and processed according to quality control procedures. | Requirement | UR-36 |

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|------------|
| SR-138 | Manufacturing records shall be maintained for minimum 10 years for traceability purposes. | Requirement | UR-36 |
| SR-139 | Diagnosis and Maintenance Requirements | Info | - |
| SR-140 | The following section defines the requirements for system diagnosis, maintenance planning, and asset management. | Info | - |
| SR-141 | No periodical or preventive maintenance shall be necessary under normal conditions for minimum 1000 flight hours. | Requirement | UR-23 |
| SR-142 | Maintenance intervals shall be automatically calculated based on actual usage patterns and environmental conditions. | Requirement | UR-23 |
| SR-143 | All relevant information related to asset management shall be accessible remotely by a centralized asset management system. | Requirement | UR-24 |
| SR-144 | Asset management interface shall provide real-time status updates and historical trend analysis. | Requirement | UR-24 |
| SR-145 | Configuration management data shall be synchronized with the asset management system automatically. | Requirement | UR-24 |
| SR-146 | The system shall log all flight and inspection data for at least 12 months with automatic archival. | Requirement | UR-25 |
| SR-147 | Data storage shall provide redundancy and error correction to prevent data loss. | Requirement | UR-25 |
| SR-148 | Historical data shall be searchable and retrievable within 30 seconds for any specific flight or time period. | Requirement | UR-25 |
| SR-149 | The system shall generate automatic diagnostic reports after each flight mission. | Requirement | UR-26 |
| SR-150 | Diagnostic reports shall include component health status, performance metrics, and anomaly detection. | Requirement | UR-26 |
| SR-151 | The diagnostic system shall monitor all safety-critical parameters continuously with real-time health assessment and trend analysis. | Requirement | UR-19 |
| SR-152 | Remote diagnostic capabilities shall provide access to all system parameters, fault codes, and performance metrics without physical access to the drone. | Requirement | UR-19 |
| SR-153 | Automated diagnostic reports shall be generated after each flight mission and transmitted to the asset management system within 5 minutes of landing. | Requirement | UR-26 |
| SR-154 | The diagnostic system shall predict component failures with minimum 100-flight-hour advance warning for scheduled replacement. | Requirement | UR-27 |

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|-----------------|
| SR-155 | The system shall provide alerts for upcoming component replacements with minimum 7-day advance notice. | Requirement | UR-27 |
| SR-156 | Component life prediction shall be based on usage patterns, environmental exposure, and manufacturer specifications. | Requirement | UR-27 |
| SR-157 | Maintenance scheduling shall optimize component replacement timing to minimize operational disruption. | Requirement | UR-27 |
| SR-158 | The maintenance planning system shall automatically schedule preventive maintenance based on flight hours, calendar time, and condition monitoring data. | Requirement | UR-23, UR-27 |
| SR-159 | Maintenance alerts shall be issued with minimum 7-day advance notice for scheduled maintenance and immediate notification for unscheduled repairs. | Requirement | UR-27 |
| SR-160 | The system shall maintain spare parts inventory recommendations based on failure rates and lead times. | Requirement | UR-27 |
| SR-161 | Diagnostic data correlation shall identify patterns and trends across multiple system components. | Requirement | UR-19 |
| SR-162 | The system shall support remote firmware updates for diagnostic and maintenance systems. | Requirement | UR-21 |
| SR-163 | Maintenance procedures shall be documented and accessible through the asset management interface. | Requirement | UR-24 |
| SR-164 | The system shall track component serial numbers and revision levels throughout the operational lifecycle. | Requirement | UR-24 |
| SR-165 | Verification Requirements | Info | - |
| SR-166 | The following section defines the verification requirements for system validation and compliance testing. | Info | - |
| SR-167 | All requirements shall be verifiable by test, analysis, or inspection in compliance with the V-Model (EN 50126). | Requirement | UR-36 |
| SR-168 | The supplier shall deliver a Verification and Validation plan with complete traceability matrix. | Requirement | UR-37 |
| SR-169 | Safety-critical functions shall undergo independent verification and validation by qualified third-party assessors. | Requirement | UR-36 |
| SR-170 | Performance testing shall demonstrate compliance with all operational requirements under specified environmental conditions. | Requirement | UR-36 |
| SR-171 | Integration testing shall verify all interface requirements and system-level functionality. | Requirement | UR-36 |

| ID | Requirement | Obligation | URD Source |
|--------|--|-------------|---------------------------|
| SR-172 | Environmental testing shall validate operation across full temperature, humidity, and altitude ranges. | Requirement | UR-02, UR-03, UR-04 |
| SR-173 | EMC testing shall demonstrate compliance with railway electromagnetic compatibility requirements. | Requirement | UR-06 |
| SR-174 | Software verification shall comply with DO-178C Level C requirements for flight-critical functions. | Requirement | UR-31 |
| SR-175 | Hardware verification shall include worst-case analysis and derating verification. | Requirement | UR-36 |

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